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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK. VOLUME 40. OV-10A IN--ETC(U)  
NOV 77 J F ROSE, N A FARINACCI, R W GORMAN

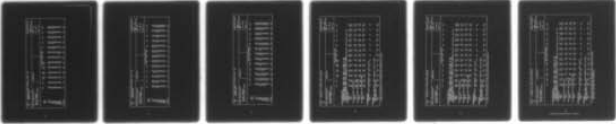
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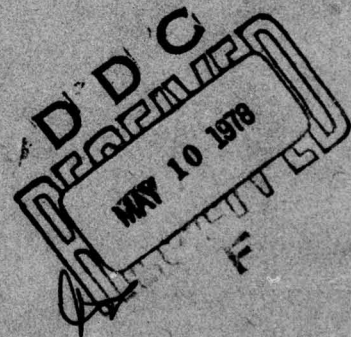
**USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK**

V.41-AD-A048 831

**Volume 40**

**OV-10A In-Flight Crew Noise**

NOVEMBER 1977



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AEROSPACE MEDICAL RESEARCH LABORATORY  
AEROSPACE MEDICAL DIVISION  
AIR FORCE SYSTEMS COMMAND  
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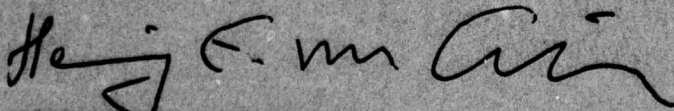
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FOR THE COMMANDER



HENNING E. VON GIERKE  
Director  
Biodynamics and Bioengineering Division  
Aerospace Medical Research Laboratory

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) The OV-10A is a USAF multi-purpose counter-insurgency aircraft whose uses include forward air control, strike reconnaissance, and light logistics transport. This report provides measured data defining the bioacoustic environments at flight crew locations inside this aircraft during normal flight operations. Data are reported for 1 location in a wide variety of physical and psychoacoustic measures: overall and band sound pressure levels, C-weighted and A-weighted sound levels, preferred speech interference level, perceived noise level, and limiting times for total daily exposure of personnel with and with-		

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out standard Air Force ear protectors. Refer to Volume 1 of this handbook, "USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application," AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

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### PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 72310418, Measurement of Noise and Vibration Environments of Air Force Operations. Col Justus F. Rose, Jr. conducted the field measurements and performed the data analysis; Capt Nick Farinacci prepared this report.

The authors acknowledge the efforts of Mr. John N. Cole who established the data analysis requirements and assisted in the preparation of this report, and Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton who assisted in the mechanics of data processing.

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## INTRODUCTION

The OV-10A is a USAF multi-purpose counter-insurgency aircraft whose uses include forward air control, strike reconnaissance, and light logistics transport. This aircraft, which is manufactured by North American Aviation, Incorporated, Columbus Division, is powered by two T76-G-10/12 turboprop engines rated at 715 shp at 41,730 rpm maximum take-off power. Each engine drives a Hamilton Standard three-blade constant-speed, 2.6 m diameter propeller through a 0.048 gear reduction. The engines are manufactured by The Garrett Corporation, AiResearch Manufacturing Company.

This volume provides measured data defining the bioacoustic environments produced inside the aircraft. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with operations of the OV-10A aircraft. Additional data on the OV-10A have been published (reference 1). Noise measurements are described for internal and near-field during engine starting and pre-takeoff phases of the OV-10A operations and for internal during airborne operations.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and aerospace ground equipment. The far-field, community-type, noise data in the handbook describe the noise produced during *ground operations* of aircraft, aerospace ground equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. *Refer to Volume 1* (reference 2) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., in-flight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published, and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of the updated index as it is generated.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; Autovon 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

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1. Gasaway, Donald, *Noise Associated with Operation of Air Force OV-10A*, SAM-TR-70-51 (AD 713882), USAF School of Aerospace Medicine, Brooks Air Force Base, Texas, 1970.
  2. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 1: Organization, Content and Application*, AMRL-TR-75-50 (1) (AD A-031865), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975



## IN-FLIGHT NOISE

### MEASUREMENTS

All noise measurements were made on-board two standard-configured OV-10A aircraft during typical speed, altitude, and flight maneuver conditions. These levels describe the standard OV-10A environments, but may not be representative of those levels encountered if the aircraft has been configured differently (e.g., major equipment or structural changes).

Acoustic measurements were made at one flight crew location. Table 1 lists the measurement location and test conditions as numeric/alphabetic designators which are used on the data pages. The designator 1/A means measurement location 1 and test condition A.

The microphone was randomly moved external to the headgear in a region 0.2-0.3 meter from the head and the resultant samples analyzed using a 4- or 8-second integration time to obtain a power-averaged level that effectively smooths out short-duration fluctuations and best describes the exposure.

### RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced inside the OV-10A aircraft at the specified location. This table includes the overall, 1/3 octave band, and octave band levels. From these data, C-weighted and A-weighted sound levels, maximum permissible time for one exposure per day (AFR 161-35) with and without standard Air Force ear protectors, preferred speech interference level, and perceived noise level are calculated and presented in Table 3. These measures are widely used to assess the effects of noise on personnel and their performance.

TABLE 1  
MEASUREMENT LOCATION AND TEST CONDITIONS

OV-10A, Eglin AFB, 28 Jul 1971, Serial # 66-13553;  
Hurlburt Fld., 5 Aug 1971, Serial # 67-14605

LOCATION	POSITION	HEIGHT ABOVE DECK
1	Aft Cockpit	Seated Head Level
CONDITION	DESCRIPTION	
A	Ground power unit operating, right rear canopy closed, front right canopy open.	
B	Ground power unit operating, both right canopies open.	
C	Left engine start, ground power unit operating, front right canopy open.	
D	Right engine start, left engine idle, ground power unit operating, front right canopy open.	
E	Idle power (both engines), front right canopy open.	
F	Taxiing, Torque below 600#-ft, 70% RPM, front right canopy open.	
G	Takeoff, 1390-1440#-ft. torque, 101% RPM.	
H	Initial acceleration, gear and flaps up, condition level — T/O and Land.	
I	Climb, 1500#-ft. torque, 120 KIAS, condition lever — T/O and Land.	
J	Climb, 2.0M PA ↗, 1150#-ft, torque, 120 KIAS, condition lever — T/O and Land.	

TABLE 1 (Continued)  
MEASUREMENT LOCATION AND TEST CONDITIONS

OV-10A, Eglin AFB, 28 Jul 1971, Serial # 66-13553;  
Hurlburt Fld., 5 Aug 1971, Serial # 67-14605

CONDITION	DESCRIPTION
K	Level flight, 3.5M PA, 1100#-ft. torque, 160 KIAS, 90-95% RPM, condition lever — Normal Flight.
L	Cruise, 3.8M PA, 1100#-ft. torque, 175 KIAS, 90-95% RPM, condition lever — Normal Flight.
M	Cruise, 3.0M PA, 1200#-ft. torque, 185 KIAS, 90-95% RPM, condition lever — Normal Flight.
N	Climb, 3.0M PA ↗, 1050#-ft. torque, 135 KIAS, condition lever, Normal Flight.
P	Climb, 6.0M PA ↗, 1000#-ft. torque, 120 KIAS, 92% RPM, condition lever — Normal Flight.
Q	Cruise, 7.0M PA, 1000#-ft. torque, 150 KIAS, 93% RPM, condition lever — Normal Flight.
R	Cruise, 7.0M PA, 1000#-ft. torque, 165 KIAS, 95% RPM, condition lever — Normal Flight.
S	Descent, 7.0M PA ↘, 1100#-ft. torque, 200 KIAS.
T	Descent, 2.5M PA ↘, 1300#-ft. torque, 210 KIAS.
U	Descent, 2.0M PA ↘, 600#-ft. torque, 150 KIAS, condition lever — T/O and Land.
V	Formation join-up, 1100#-ft. torque, 130 KIAS, 98% RPM, condition lever — T/O and Land.
W	Climb, 4.0M PA ↗, 1250#-ft. torque, 130 KIAS, 98% RPM, condition lever — T/O and Land.
X	Cruise, 5.0M PA, 900#-ft. torque, 160 KIAS, 98% RPM, condition lever — T/O and Land.
Y	BDU High Angle dive bomb pattern, 3500' entry, 2300' release, release airspeed — 255 KIAS, 45° dive angle, recovery 1300' minimum, 600-800#-ft. torque, 98% RPM, condition lever — T/O and Land.
Z	Rocket pass, 3500' entry, 1800' release, recovery 1300' minimum, release airspeed 255 KIAS, 30° dive angle, condition lever — T/O and Land.
AA	Same as K — recovery (gear warning horn actuated), condition lever — T/O and Land.
BB	Strafing run, 3000' entry, 2000/1500' fire, recovery 1000' minimum, 220 KIAS, 20° dive angle, condition lever — T/O and Land.
CC	BDU Low Angle dive bomb pattern, 2500' entry, 700' release, 300' minimum recovery, 220 KIAS, 10° dive angle, condition lever — T/O and Land.
DD	Cruise, 3.5M PA, 1000#-ft. torque, 160 KIAS, 98% RPM, condition lever — T/O and Land.
EE	Descent (Clean), 3.5M PA ↘, 650#-ft. torque, 170 KIAS, 97% RPM, condition lever — T/O and Land.
FF	VFR overhead pattern, initial 1500', 1100#-ft. torque, 160 KIAS, condition lever — T/O and Land.
GG	GCA pattern, gear and flaps down, 700#-ft. torque, condition lever — T/O and Land, final approach.
HH	Pitchout, gear warning horn actuated, condition lever — T/O and Land.
II	Final approach (including turn to final) gear and flaps down, condition lever — T/O and Land.
JJ	Touchdown and landing roll.
KK	Prop. reverse during landing roll.
LL	Taxiing, both right canopies open.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB) 1/3 OCTAVE BAND													IDENTIFICATION:									
FREQ (HZ)	1/A		1/B		1/C		1/D		1/E		1/F		1/G		1/H		1/I		1/J		1/K	
	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX	MIN	MAX
25	59	63	80	91	85	101	88	82	100	105	98	96	96	100	105	98	96	96	96	96	100	100
31.5	72	71	80	79	90	87	88	82	103	94	87	82	94	95	94	89	94	95	94	96	96	96
40	73	81	92	81	99	87	89	85	93	100	89	85	89	93	100	89	87	89	87	91	91	91
50	77	74	86	77	90	82	81	78	86	96	84	82	84	86	96	84	82	84	82	86	86	86
63	76	75	86	88	93	97	96	97	97	99	88	87	87	86	90	87	87	87	87	90	90	90
80	82	78	84	104	87	106	85	86	96	99	106	85	86	96	99	96	100	100	100	102	102	102
100	90	74	83	85	85	86	81	79	114	114	86	81	79	114	114	113	114	113	114	115	115	115
125	74	74	83	83	88	93	93	93	90	97	93	93	93	97	97	95	93	95	93	96	96	96
160	70	86	89	86	89	96	89	85	95	95	96	89	85	95	95	93	97	97	97	97	97	97
200	76	79	82	86	84	91	86	85	115	113	91	86	85	115	113	110	111	110	111	110	110	110
250	73	80	83	90	84	89	84	80	98	98	89	84	80	98	98	99	100	99	100	101	101	101
315	67	72	77	86	84	91	86	88	104	108	91	86	88	104	108	109	105	109	105	103	103	103
400	63	66	77	86	84	91	89	87	98	103	91	89	87	98	103	103	101	103	101	102	102	102
500	62	66	76	85	83	93	93	92	98	99	93	93	92	98	99	98	101	98	101	102	102	102
630	64	72	78	87	84	94	91	88	97	102	94	91	88	97	102	100	99	100	99	97	97	97
800	64	68	74	83	82	93	88	88	91	92	93	88	88	91	92	93	88	91	88	91	91	91
1000	62	69	73	81	83	94	92	89	90	93	94	92	89	90	93	91	89	91	89	89	89	89
1250	61	66	72	80	85	92	90	88	88	88	90	90	88	88	90	90	89	91	88	89	91	91
1600	61	66	71	79	83	93	90	89	87	89	93	90	89	87	89	89	88	89	88	92	92	92
2000	62	67	72	79	86	94	94	94	93	88	94	94	94	93	88	89	88	88	88	90	90	90
2500	60	63	70	77	86	93	94	94	86	87	93	94	94	86	87	86	85	86	85	89	89	89
3150	59	63	71	77	86	93	94	94	94	94	93	94	94	85	87	87	86	87	86	91	91	91
4000	59	62	69	78	87	100	96	97	97	86	100	96	97	86	86	86	84	86	84	87	87	87
5000	55	60	67	74	88	98	94	96	82	83	98	94	96	82	83	83	81	81	81	85	85	85
5300	54	58	67	73	82	92	91	93	63	63	93	91	93	63	63	62	60	62	60	63	63	63
8000	53	57	66	73	82	96	96	101	83	81	96	96	101	83	81	81	79	81	79	83	83	83
10000	50	54	62	73	85	95	92	94	81	81	94	92	94	81	81	80	80	80	80	82	82	82
12500	49	54	62	69	81	92	93	93	82	84	92	93	93	82	84	82	81	81	81	81	81	81
16000	50	55	59	68	78	91	90	91	81	80	91	90	91	81	80	79	78	79	78	79	79	79
OVERALL	91	90	96	105	103	111	106	107	118	118	116	116	117	118	118	116	117	117	117	117	117	117

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE 1 MEASURED SOUND PRESSURE LEVEL (DB) 1/3 OCTAVE BAND													IDENTIFICATION:			
2													OMEGA 3-2			
NOISE SOURCE/SUBJECT:													TEST 74-093-001			
( OPERATION:													RUN 02			
( OV-10A AIRCRAFT													17 JAN 75			
( INFLIGHT NOISE LEVELS													PAGE F2			
LOCATION/CONDITION																
FREQ (HZ)	1/L	1/M	1/N	1/P	1/Q	1/R	1/S	1/T	1/U	1/V	1/W	1/X	1/Y			
25	99	100	95	96	99	98	100	101	108	91	90	94	96			
31.5	95	94	93	93	93	94	95	95	91	84	85	87	89			
40	91	89	88	88	90	89	92	93	89	77	78	80	83			
50	87	88	85	84	85	85	88	88	85	76	76	79	83			
63	91	92	87	85	89	90	93	94	100	85	84	88	92			
80	98	104	98	100	97	97	102	104	113	97	93	94	99			
100	110	117	111	113	110	108	115	116	94	115	111	111	115			
125	96	98	93	93	95	95	98	97	97	93	92	93	96			
160	94	98	92	98	94	96	97	97	104	91	94	91	100			
200	105	110	105	112	106	109	107	106	96	109	112	109	118			
250	104	102	100	102	102	100	101	105	98	96	96	95	107			
315	106	104	102	105	104	103	103	107	92	105	104	103	116			
400	101	103	99	102	101	102	104	103	92	99	100	98	111			
500	99	99	102	100	100	99	101	101	90	99	100	96	106			
630	99	97	98	102	95	98	99	99	90	100	101	99	108			
800	91	93	89	90	90	92	95	95	88	91	92	92	101			
1000	90	92	88	89	89	91	94	96	88	88	89	90	101			
1250	90	92	89	89	90	91	93	94	90	88	89	91	98			
1600	90	93	88	88	90	91	94	95	89	87	88	91	98			
2000	91	92	88	88	89	91	94	94	89	87	88	91	98			
2500	90	91	86	86	89	90	94	94	87	85	86	90	96			
3150	93	94	87	86	90	92	96	97	89	85	86	90	96			
4000	88	89	84	84	87	88	93	93	86	84	85	88	95			
5000	85	87	81	81	83	86	90	91	84	81	82	85	93			
6300	84	85	81	80	82	84	88	89	83	80	81	84	92			
8000	83	85	79	79	82	84	88	89	82	79	80	82	92			
10000	82	84	79	79	80	82	85	86	81	79	79	80	89			
12500	81	82	79	79	79	81	84	85	78	82	81	81	88			
16000	80	81	77	77	78	79	84	84	78	78	78	79	87			
OVERALL	114	119	114	117	114	114	117	118	115	116	115	114	122			

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.





TABLE: MEASURED SOUND PRESSURE LEVEL (DB)		IDENTIFICATION:											
OCTAVE BAND		OMEGA 3.2											
NOISE SOURCE/SUBJECT:		TEST 74-093-001											
OPERATION:		RUN 02											
OV-10A AIRCRAFT		17 JAN 75											
INFLIGHT NOISE LEVELS		PAGE J2											
		LOCATION/CONDITION											
FREQ (HZ)	1/L	1/M	1/N	1/P	1/Q	1/R	1/S	1/T	1/U	1/V	1/W	1/X	1/Y
31.5	100	102	98	98	100	100	101	102	108	92	91	95	97
63	99	105	98	100	98	98	103	104	113	97	94	95	99
125	110	117	111	113	110	108	115	116	105	115	111	111	115
250	110	111	107	113	109	110	109	111	100	110	112	110	120
500	104	105	105	106	104	104	106	106	95	104	105	103	113
1000	95	97	93	94	94	96	99	100	93	94	95	96	105
2000	95	97	92	92	94	95	99	99	93	91	92	95	102
4000	94	96	89	89	92	94	98	99	91	89	89	93	100
8000	88	89	84	84	86	86	92	93	86	84	84	87	96
16000	83	84	81	81	82	83	87	87	81	83	83	83	90
OVERALL	114	119	114	117	114	114	117	118	115	116	115	114	122









MEASURES OF HUMAN NOISE EXPOSURE												
IDENTIFICATION:												
NOISE SOURCE/SUBJECT: ( OPERATION: )												
OV-10A AIRCRAFT ( )												
INFLIGHT NOISE LEVELS ( )												
LOCATION/CONDITION												
1/Z	1/AA	1/BB	1/CC	1/DD	1/EE	1/FF	1/GG	1/HH	1/II	1/JJ	1/KK	1/LL
HAZARD/PROTECTION												
C-WEIGHTED OVERALL SOUND LEVEL (OASLC IN DBC) AT EAR												
A-WEIGHTED OVERALL SOUND LEVEL (OASLA IN DBA) AT EAR												
MAXIMUM PERMISSIBLE TIME (T IN MINUTES) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)												
NO PROTECTION												
OASLC	119	119	120	115	112	112	113	115	116	117	108	120
OASLA	113	116	114	109	105	106	107	105	117	111	100	109
T	3.2	P	2.7	6	13	11	9	13	P	4.5	30	6
HGU-2A/P HELMET WITH H-154												
OASLA*	107	107	109	103	99	99	100	102	105	106	95	104
T	9	9	6	18	36	36	30	21	13	11	71	15
HGU-2A/P HELMET WITH H-154(A)												
OASLA*	104	103	105	99	95	94	95	98	99	102	91	101
T	15	16	13	36	71	85	71	42	36	21	143	25
HGU-2A/P HELMET WITH CUSTOM LINER												
OASLA*	110	110	112	106	102	102	103	104	108	109	97	107
T	5	5	3.8	11	21	21	16	15	8	6	50	9
COMMUNICATION												
PREFERRED SPEECH INTERFERENCE LEVEL (PSIL IN DB)												
PSIL	106	107	108	103	99	101	102	96	108	102	93	102
ANNOYANCE												
PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PNDB)												
TONE CORRECTION (C IN DB)												
PNLT	129	134	130	125	122	122	122	122	134	126	116	127
C	3	3	2	3	3	3	3	3	3	3	2	3

\* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE.  
P ADDITIONAL EAR PROTECTION REQUIRED.