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LOCALIZED HYDROGEN ANALYSIS IN SIMULATED GRAIN BOUNDARIES OF ALUMINUM ALLOYS EXPOSED TO STRESS CORROSION

RE-551

Final Report (28 April 1975 to 28 September 1976)

February 1978

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Philip N. Adler

Prepared Under Contract N00019-75-C-0394

for

Department of the Navy Naval Air Systems Command Washington, D.C. 20361

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Research Department Grumman Aerospace Corporation Bethpage, New York 11714

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removal. A substantial hydrogen concentration increase was observed in the interface region of MgZn₂ film/7075 substrate samples exposed to stress corrosion; smaller increases were introduced in the interface region of 7475 film/7075 substrate samples; no increase occurred in aluminum film/7075 substrate samples. The increase in hydrogen concentration was found to be stress dependent. Upon removal of stress, hydrogen was reduced to its initial concentration in the 7475 film/7075 substrate interface but remained trapped in the MgZn₂ film/7075 substrate interface. The relatively strong affinity for hydrogen in the interface region of MgZn₂ film/7075 substrate samples during stress corrosion exposure coupled with the actual stress corrosion failure of two of these samples suggest that stress corrosion attack in 7000 series aluminum alloys is hydrogen related.



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A major impetus for this work was supplied by Mr. Sam Goldberg of the Naval Air Systems Command (AIR 52031), who over the past several years has actively supported the development and application of the Lithium Nuclear Microprobe for Hydrogen (LNMH) technique. The helpful suggestions and review of this work by Mr. E. Balmuth of the Naval Air System Command (AIR 52031G) are gratefully acknowledged.

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ABSTRACT

The role of hydrogen in the stress corrosion attack of 7000 series aluminum alloys was investigated using sputtered film/ substrate samples to simulate the grain boundary constituents in these alloys. Localized hydrogen measurements were made in the film, interface region, and substrate using the Lithium Nuclear Microprobe for Hydrogen (LNMH) technique. Hydrogen concentrations were measured in samples of pure aluminum film/7075 substrate, 7475 film/7075 substrate, and MgZn, film/ 7075 substrate for as-sputtered, corrosion exposed, and stress corrosion exposed conditions as well as after stress removal. A substantial hydrogen concentration increase was observed in the interface region of MgZn, film/7075 substrate samples exposed to stress corrosion; smaller increases were introduced in the interface region of 7475 film/7075 substrate samples; no increase occurred in aluminum film/7075 substrate samples. The increase in hydrogen concentration was found to be stress dependent. Upon removal of stress, hydrogen was reduced to its initial concentration in the 7475 film/7075 substrate interface but remained trapped in the MgZn₂ film/7075 substrate interface. The relatively strong affinity for hydrogen in the interface region of MgZn₂ film/7075 substrate samples during stress corrosion exposure coupled with the actual stress corrosion failure of two of these samples suggest that stress corrosion attack in 7000 series aluminum alloys is hydrogen related.

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1. INTRODUCTION

Stress corrosion is one of the most serious problems encountered with high-strength aluminum alloys of the Al-Zn-Mg type (Refs. 1 and 2). To avoid this problem, these alloys are used in a lower strength, overaged temper (Refs. 3 and 4). If the full strength capabilities of Al-Zn-Mg type alloys are to be realized, greater understanding of the mechanism of stress corrosion attack must be attained.

Recent evidence suggests that stress corrosion in these alloys is a form of hydrogen embrittlement. Speidel has shown that stress corrosion crack growth of 7079-T651 aluminum alloy in an aqueous salt environment is temperature dependent, with an activation energy for subcritical crack growth comparable to that observed in the embrittlement of nickel and iron-base alloys in distilled water (Ref. 5). Speidel suggests that similar rate controlling steps for hydrogen entry are involved in all these material environment combinations. Green, Hayden, and Montague have found that stress corrosion cracking susceptibility of 7075-T6 aluminum alloy is far greater under tensile (mode I) loading than torsional (mode III) loading (Ref. 6). They conclude that under torsion loading the component of hydrostatic stress is insufficient to cause a concentration of hydrogen ahead of the crack tip whereas hydrogen damage is a dominant mechanism leading to stress corrosion failure under tensile loading.

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Another indication that hydrogen is an important factor in stress corrosion attack concerns the effect of pre-exposure. Gruhl and Brungs have found that uncracked specimens of Al-Zn-Mg alloy exhibited a reduction from 20 to 13 percent elongation

after stress corrosion exposure in an aqueous chloride environment (Ref. 7). These specimens showed evidence of intergranular attack after testing. However, if exposed samples were reheat-treated to their original temper, no ductility loss was observed. The loss in ductility was interpreted as grain boundary embrittlement by hydrogen that was eliminated in subsequent heat treatment.

In direct observations of stress corrosion attack using transmission electron microscopy, Montgrain and Swann exposed an A1-7% Zn-3%Mg alloy to a saturated water vapor environment. They found that intergranular crack propagation takes place at the interface between incoherent MgZn₂ grain boundary precipitates and one of the grains (Ref. 8). No evidence of grain boundary precipitate dissolution was observed during crack propagation but evolution of molecular hydrogen was measured during deformation and fracture.

One of the major arguments against a hydrogen-related mechanism of stress corrosion attack in high strength aluminum alloys has been that cathodic polarization acts to reduce stress corrosion attack. Since copious amounts of hydrogen exist at the cathode, one would expect enhanced attack if hydrogen was significant to the attack mode. However, recent work of Gest and Troiano (Ref. 9) show that hydrogen permeability in 7075-T651 in a 3 percent NaCl solution goes through a minimum over the -800 to -1200 mV applied potential range (relative to Standard Calomel Electrode). Since the open circuit potential is approximately -800 mV, hydrogen would be relatively immobile until cathodic potentials in excess of -1200 mV were applied. Indeed, Gest and Troiano found a significant increase in stress corrosion crack growth velocity at an applied potential of -1500 mV.

Although the studies described above strongly suggest that stress corrosion attack in Al-Zn-Mg type alloys is related to hydrogen embrittlement, other mechanisms have been more generally accepted, and the role of hydrogen is considered quite controversial (Ref. 10). The purpose of the present work is to clarify the role of hydrogen in the intergranular attack mode characteristic of stress corrosion in these alloys by direct measure of the hydrogen concentration in simulated grain boundary regions exposed to stress corrosion. Previous work has shown the importance of the grain boundary precipitate in stress corrosion attack (Refs. 8, 11-15). In the present work, localized hydrogen concentration was measured using the recently developed Lithium Nuclear Microprobe for Hydrogen (LNMH) technique that has a depth resolution of approximately ±0.10 µm in the materials studied (Refs. 16-18). To capitalize on this depth resolution capability. a film/substrate sample configuration was utilized for the grain boundary simulation. In this configuration, the constituents at a grain boundary are represented by the film and substrate. with the interface that separates the film and substrate representing the grain boundary. Hydrogen concentrations were measured in the film, interface, and substrate regions of samples exposed to corrosion and stress corrosion environments as well as in control samples.

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2. EXPERIMENTAL

LOCALIZED HYDROGEN ANALYSIS

For this study, the LNMH technique was utilized to determine localized hydrogen concentrations. This technique is based on measuring the characteristic 14.7- and 17.6-MeV gen a rays emitted during the resonant ${}^{1}H({}^{7}Li,\gamma)^{8}$ Be nuclear reaction, as illustrated in Fig. 1. The gamma-ray yield is proportional to hydrogen concentration. In practice, ⁷Li ions are accelerated in a Van de Graaff accelerator and allowed to strike a target sample. For accelerated ⁷Li ions at the resonance energy for this reaction, 3.07 MeV, only the hydrogen atoms at the very outer surface of the sample will be measured. As the energy of the accelerated 'Li ions is increased, the resonant reaction occurs at deeper lying regions below the surface via slowing down of the incident ⁷Li ions to the resonance energy. Thus, hydrogen concentration depth profiles can be measured by varying the accelerator voltage. Since the resonance width is very narrow (0.08 MeV), the depth resolution of the LNMH for metals is typically ± 0.05 to ± 0.15 um, depending on sample composition. For the aluminum and MgZn, samples herein studied, the depth resolution is approximately ±0.10 um.





Localized hydrogen analysis with the LNMH can be done effectively using the depth resolution capability of this approach. For study of hydrogen concentration at grain boundary regions associated with stress corrosion exposure, the film/substrate sample configuration illustrated in Fig. 2 was utilized. This sample configuration enabled simulation of the chemical constituents typical of the grain boundary in Al-Zn-Mg type alloys and permitted depth profiling in the film, interface, and substrate regions. By in situ measurements in these regions, hydrogen concentrations associated with different conditions of environment exposure were determined.





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SAMPLE FREPARATION

Samples were prepared by sputtering specific films onto 7075-T651 aluminum alloy substrates. Films of pure aluminum. 7475 aluminum alloy, and MgZn2 were sputtered from appropriate The pure aluminum target was zone-refined material targets. having a nominal purity of 99.999+%; the 7475 aluminum alloy target was prepared from 3/16-inch thick plate that contained 5.60 wt% Zn, 2.44% Mg, 1.43% Cu, and less than 0.01% of Cr, Fe, Si, Mn or Ti; the MgZn, target was prepared from vacuum-hot-pressed powder that contained no binders. These targets were assembled by the Materials Research Corporation (MRC). The 7475 aluminum alloy is a high purity form of 7075 that was specially supplied by the Metallurgical Research Division of the Reynolds Metals Company. Substrates, of 0.040-inch thickness x 3/8-inch width x 2-inch length, were sliced from 3-inch thick 7075-T651 bar stock; the 2-inch substrate length was in the short transverse direction of the bar. These substrates were polished to a 0.065 \pm 0.015 μ m rms finish (as measured with a profilometer) using standard metallographic techniques and a 0.05 μ m final grit size. Substrate polishing was utilized to avoid the surface roughness that was previously found to limit depth resolution capability in the interface region (Ref. 19). Prior to film deposition, substrates were sputter-etched to remove surface contaminants and promote adherence with the sputtered film.

Both sputter etching and film sputtering were done in a MRC Model SEM-8620 RF Sputter-Etch Unit. During preliminary experiments, relatively high hydrogen concentrations, of the order of 100 wppm, were measured by LNMH in pure aluminum sputtered films. Contamination introduced by the argon gas used in sputtering was suspected. Therefore, Research Grade Argon of 99.9995% purity, the

highest purity argon commercially available, was substituted for normally used High Purity Argon. In addition, the argon gas was purified by a titanium-getter heater prior to entry into the sputtering unit. These changes reduced the hydrogen concentration in the pure aluminum sputtered films is less than 40 wppm.

Another problem encountered during preliminary sputtering experiments concerned substrate softening caused by heating. A reduction in the yield strength of 7075-T651 substrates from 67 to 16 ksi was encountered. Such softening was undesirable since only minimal elastic loadings could be applied during stress corrosion exposure. Therefore different methods to avoid substrate heating during sputtering were evaluated. It was found that use of a high vacuum silicone grease on the back surface of the substrate served to enable sufficient heat transfer during sputtering to avoid softening. In fact, no change in substrate hardness was experienced when using the silicone grease. Therefore this procedure was used uniformly in preparing film/substrate samples.

FILM CHARACTERIZATION

Both gravimetric and nuclear resonance techniques were utilized in measuring the thickness and thickness uniformity of sputtered films. Scanning Electron Microscopy (SEM) and X-ray diffraction were also used for film characterization. Film surface topography was examined using an Advanced Metals Research Corporation (AMR) Model 1000 Scanning Electron Microscope (SEM). Films were examined after sputtering and each condition of environmental exposure. In conjunction with SEM examination, Energy Dispersive Analysis of X-Rays (EDAX) was used in identifying the elemental constituency in localized film regions. The crystallographic character of the sputtered films was measured using a Picker X-ray Diffractometer with a copper target.

Accurate measurement of film thickness was important in locating the interface region of each sample. Film thickness was determined by weight difference before and after sputtering. No significant substrate weight loss was measured for the sputter etching treatment utilized. Careful attention was paid to remove all the silicone high vacuum grease that was used to transfer heat from the substrate during sputtering. Before final weighing, each sample was cleaned in an ultrasonic bath of trichloroethylene for 1 min followed by successive 1 min rinses in separate ultrasonic baths of Freon. This procedure was effective since no further weight loss was observed by additional cleaning.

Film thickness will be expressed in units of mg/cm^2 throughout. When thickness is indicated in microns, a density value of 2.71 g/cm³ has been used for the pure aluminum and 7475 films, and 5.20 g/cm³, based on the density of crystalline material (Ref. 20), has been used for the MgZn₂ films. Since the samples contained a hole and were not of uniform width, an accurate measure of surface area was obtained for each substrate by dividing the substrate weight by the product of the thickness and density. Substrates were 0.038 \pm 0.002-inch thick and were measured to the nearest 0.001 inch. Weighings were made to the nearest 0.01 mg. For the films prepared, weight increases were of the order of 4-6 mg. The overall error in the determination of film thickness is estimated as 3 percent and is primarily related to the precision in substrate thickness measurement.

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In addition to the gravimetric approach, the thickness of selected sputtered films was measured using a nuclear technique based on stimulation of proton-induced resonances. This approach was used to corroborate gravimetric measurements and provide a means of evaluating the uniformity of film thickness over an



Fig. 3 Sample Surface Masking (a) As-Sputtured Sample (b) Masked Sample Before Environmental Exposure (c) Sample After Environmental Exposure and LNMH Measurement

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area equivalent to that used with the LNMH. Two variations of the technique were necessary. For MgZn, films, the resonance reaction between fast protons and aluminum nuclei in the substrate, ${}^{27}A1(p,\gamma){}^{28}Si$ at $E_{res} = 0.992$ MeV, was utilized. For pure aluminum and 7475 aluminum alloy films, the small aluminum composition difference between film and substrate made use of this reaction impracticable. Therefore, a surface "marker" consisting of a very thin layer of LiF was evaporated onto six selected polished 7075 substrates prior to sputtering, and the resonance reaction between fast protons and fluorine nuclei, ${}^{19}F(p,\alpha\gamma){}^{16}Q$ at $E_{res} = 0.874$ MeV, was utilized. Film thickness was calculated from the measured shift of resonance energy. The width of the resonance energy profile, corrected for natural line width, proton-beam energy dispersion, and straggling, provided a measure of the variation in film thickness within the 0.10×0.10 -inch area probed.

ENVIRONMENTAL EXPOSURE

Samples of each film/substrate combination were exposed to a corrosion or stress corrosion environment at room temperature prior to the LNMH measurement. An aqueous chloride solution containing 3 percent NaCl and 0.5 percent $AlCl_3.6H_20$ and having a 3.5 pH was used as the corrosion medium. The solution was constantly stirred during the 500 min exposure time. Earlier, it had been found that a corrosion product at least 3 µm thick formed during similar environmental exposure (Refs. 16, 18). Formation of such a corrosion product would obscure desired LNMH measurements. Therefore, masking was used to prevent contact between the solution and film surface region to be probed. Masking consisted of a combination of an inner 0.200-inch x 1/2inch Teflon tape, ar outer 0.300-inch x 3/4-inch electrical insulating tape seal, and a perimeter seal of Collodion. The

masks were placed over the center portion of the sample film surface as shown in Fig. 3, and were removed before LNMH measurement.

A test device was designed and built for stress corrosion exposure. The device, shown in Fig. 4, was capable of accommodating eight samples (five are shown in Fig. 4). Stress was applied using four-point bending and was maintained during exposure and subsequent LNMH measurement. The inner-two loading points were specific to each sample and are shown at the three unoccupied sample positions in Fig. 4; the outer-two loading points were continuous near the top and bottom of the device. By displacement of the inner contact points, a constant elastic stress was produced on the sample in the 1/2-inch length between inner contacts. Two devices were fabricated and used for both stress corrosion and corrosion exposures as well as for holding samples during the LNMH measurements. The devices were machined from 321 stainless steel plate and were coated with an epoxy fuel tank paint to avoid galvanic interaction with the samples. The strain-displacement characteristics of each sample position was calibrated using an SR-4 strain gage on the outer sample surface. Each position was found to be independent of the loading on other samples. Based on repeated strain measurements and a 10.4 x 10⁶ psi elastic modulus, known displacements resulted in an average deviation of \pm 4.5% to 60 ksi.

Samples were stress corrosion exposed under either an elastic or plastic strain in the substrate surface. For elastic loading, a stress of 29 ksi was applied; for plastic loading, the stress was 64 ksi. The latter value is based on an applied strain of 0.675 percent and the stress strain behavior of the 7075 aluminum alloy substrate. For MgZn₂ films at this



Fig. 4 Test Device for Loading and Environmental Exposure. Five Sample Positions are Occupied:

- 1 Sputter-Etched Substrate
- 2 As Sputtered Al Film/Substrate Sample 3 Corrosion Exposed Al Film/Substrate Sample
- 4 Stress Corrosion Exposed (Plastic-Loading)
- 7475 Film/Substrate Sample 5 Stress Corrosion Exposed (Plastic-Loading)
- MgZn2 Film/Substrate Sample with Partial Al Overlay Mask



strain, the stress was approximately 74 ksi since both aluminum and $MgZn_2$ have almost the same modulus (Ref. 21) and $MgZn_2$ would not be expected to undergo plastic deformation.

LNMH MEASUREMENTS

Hydrogen concentration was measured in the film, interface, and substrate regions of unexposed and exposed film/substrate samples as well as in the surface of polished and sputter etched substrates. LNMH measurements were made at the Nuclear Structure Laboratory, SUNY, Stony Brook, New York using the doubly ionized ⁷Li beam of a High Voltage Engineering Corporation (HVEC) Model FN Tandem Van de Graaff Accelerator. To facilitate measurements in the numerous samples examined, ⁷Li beam energies of 4.25, 4.75, and 5.25 MeV were utilized to examine the film, interface and substrate regions, respectively, and samples of specific film thickness were prepared so that the LNMH resonance reaction occurred in the interface region at a 4.75 MeV beam energy. The actual depths of the LNMN measurement are given in Table 1 and are based on accurately known stopping powers for the elemental constituency of the sample (Ref. 22).

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Hydrogen concentrations were measured in two samples of each film/substrate type for each of the following conditions:

As-Sputtered (S) Corrosion Exposed (C) Stress Corrosion Under Elastic Stress (SCE) Above With Stress Removed (SCE-U) Stress Corrosion Under Plastic Stress (SCP) Above With Stress Removed (SCP-U)

One as-polished and two sputter etched substrates were also examined at the 4.25, 4.75 and 5.25 MeV energies. LNMH measurements

TABLE 1 DEPTH OF LNMH MFASUREMENT

Sample	Depth of Measurement [†]					
(Film/Substrate)	4.25 MeV (Film)		4.75 MeV (Interface)		5.25 MeV (Substrate)	
	mg/cm^2	<u>um</u> #	mg/cm ²	μ m #	mg/cm^2	um#
Pure A1/7075	0.61	2.25		3.3610.11	1.23	4.54
7475/7075	0.61	2.25	0.91±0.03	3.36±0.11	1.23	4.54
MgZn ₂ /7075	0.95	1.63	1.25±0.05	2.40±0.09	1.58	3.58
 ⁺ Deviations are based on LNMH reaction resonance width and straggling and are approximately the same at the three energies investigated. [*] Based on density of 2.71 g/cm³ for A1, 7075 and 7475 and 5.20 g/cm³ for MgZn₂. 						

were made 19 \pm 1 days after environmental exposure. One of the MgZn₂ film/substrate samples exposed to SCP failed while under load less than one day after being removed from the solution environment. This sample was examined in the unloaded condition (SCP-U), and a third MgZn₂ film/substrate SCE sample was substituted. (During preliminary exposure evaluation, a similar failure occurred in another MgZn₂ film/substrate sample exposed under SCP shortly after removal from the solution environment.) These were the only two sample failures that were encountered during the course of this investigation.

A 0.10 x 0.10-inch beam collimator was used throughout. The area examined was always within the inner-two contact points of the test device. Alignment for the different sample positions of

the test device relative to the ⁷Li beam was established prior to LNMH measurement. A cathetometer was used to align the beam collimator with a reference cross-hair on the LNMH target chamber. The test device was mounted on a X - Y positioning stage within the target chamber and specific positions were indexed at the areas to be measured. Insertion of the X-Y positioning stage into the LNMH target chamber is shown in Fig. 5. Although surface masking during solution exposure was generally effective in preventing corrosion product formation on the film area to be examined, a limited number of samples did undergo attack near their edges. For these cases, a partial overlay mask of pure aluminum sheet was used to eliminate any possibility of interaction between the corrosion product and the 'Li beam. (A sample with an overlay mask as well as four unmasked samples, are shown after LNMH measurement in Fig. 4. The area probed with the beam is evidenced by the darkened "beam spot".)

A total of 40 samples were examined in five separate target chamber loadings (eight samples per loading). For each sample, measurement was made at three depths, so that in total, 120 LNMH measurements were made. A specially prepared NBS 354 H-in-Ti Standard, 215 \pm 6 wppm, (Ref. 16) was used for calibrating each loading at each of the three energies examined. A vacuum annealed pure iron standard was also examined periodically as a measure of beam-dependent-background. Beam currents from 70-190 na were used and were generally in the 130-150 na range. Measurement times were of the order of 400-800 sec depending on the hydrogen concentration being measured and the desired count statistics. Measurement times were subdivided into equivalent charge collection periods to monitor possible time dependent concentration change during the measurement. In no case was

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any time dependent concentration change observed for the range of beam currents utilized. Each sample loading required approximately 9 hours of measurement time.

In addition to LNMH measurements at 4.25, 4.75 and 5.25 MeV, hydrogen concentration-depth profiling was done in the surface of polished and sputter-etched substrates over the 3.08 to 3.85 MeV energy range. Surface profiling was done at Grumman's 4 MV Van de Graaff Accelerator facility using singly ionized ⁷Li.



Fig. 5 LNMH Target Chamber Assembly. Test Device is Mounted on X-Y Positioning Stage

3. RESULTS AND DISCUSSION

FILM DESCRIPTION

Accurate measurement of film thickness was important to establish the depth at which the film/substrate interface was located in each sample. Film thickness was determined from the weight difference before and after sputtering and the substrate surface area. The accuracy of this gravimetric method was evaluated using a nuclear technique of film thickness measurement on selected samples. Comparison of film thickness results for specific film/substrate samples is made in Table 2. As indicated, the agreement between these two techniques is quite good.

Film	Thickness (mg/cm ²)			
		Nuclear Method		
	Gravimetric Method	Average and Maximum Deviation [†]	Maximum Local Variation#	
Pure Aluminum	1.36 ± 0.02 1.25 ± 0.02 1.23 ± 0.02	1.22 🗄 0.09	± 0.06 ± 0.06 ± 0.06	
7475 Aluminum Alloy	1.48 ± 0.02 1.45 ± 0.02 1.50 ± 0.02	1.43 ± 0.00	± 0.05 ± 0.06 ± 0.06	
MgZn2	1.13 ± 0.02	1.10 ± 0.00	± 0.06	

TABLE 2 FILM THICKNESS MEASUREMENT COMPARISON

^T Based on three to five separate areas in same sample

Within 0.10 x 0.10 inch area — calculated from broadening of resonance width

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Variation in film thickness within the 0.10 x 0.10-inch area probed was also determined from the nuclear measurements. Based on the extent of broadening in the resonance width of the ${}^{27}\text{Al}(p,\gamma){}^{28}\text{Si}$ reaction for the Mg2n₂/7075 sample and of the ${}^{19}\text{F}(p,\alpha\gamma){}^{16}$ O reaction in the pure aluminum/7075 and 7475/7075 samples, the variation was \pm 0.06 mg/cm², as indicated in Table 2. Since this variation is approximately three times larger than the precision of the average film thickness measured using the gravimetric method, an uncertainty of \pm 0.06 mg/cm² will be used in describing sample film thickness.

Both X-ray diffraction and SEM were used in characterizing sputtered films. Initially, films were sputtered onto glass substrates to determine their crystalline nature as a function of film thickness. As indicated in Table 3, the films were crystalline with the exception of one $MgZn_2$ film of 0.17 mg/cm² thickness. Diffraction patterns could be indexed with expected lattice parameters and crystal structures (Ref. 20). These films generally had a preferred orientation that changed with film thickness.

Subsequently, films were sputtered onto 7075-T6 substrates. The topographies of these films are illustrated in the SEM micrographs of Fig. 6. Films of pure aluminum and 7475 aluminum alloy appear similar and have a cubic crystallite morphology with an approximate 3-6 μ m size; crystallites of the MgZn₂ film have a spherical morphology and an approximate 0.5-2 μ m size. Since the thickness of these films is of the order of 2-4 μ m, it is likely that the films are a polycrystalline array only one or two crystals thick.

Use of high vacuum silicone grease on the back surface of each substrate during sputtering resulted in cooler substrates



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TABLE 3 SPUTTERED FILM CRYSTALLOGRAPHIC CHARACTERISTICS

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Film	Thickness		Crystallographic	
	(mg/cm ²)	(µ.m) #	Nature‡	
Pure Aluminum	1.17	4.32	FCC [*]	
7475 Aluminum Alloy	0.10 0.17 0.42 0.99	0.63	FCC, Fine Grained FCC FCC, [200] FCC, [200]	
MgZn ₂	0.17 0.65 0.92 1.83 14.4	$\begin{array}{c} 1.3 \\ 1.8 \end{array}$	Amorphous HCP, *(10.0) HCP, [10.0] HCP, [00.2] HCP, [00.2]	
+ Sputtered onto glas	s substrat	e	*********	
* Based on a density films, and 5.20 g/c			the aluminum and 7475	
* Planes of preferred orientation indicated where observed				
* FCC - Face Centered Cubic HCP - Hexagonal Close Packed				

that significantly affected the crystallographic nature of the sputtered films. For pure aluminum, the crystallite morphology remained comparable to that on hotter substrates, but the size of the cubic crystallites was reduced from 4~6 µm to approximately 0.8 µm. The crystallite morphology of the 7475 aluminum alloy film was difficult to discern but appeared less cubic than on hotter substrates, and crystallite size was reduced from 3-6 µm

to approximately 0.5 µm. No discrete crystalline morphology was apparent in the topography of sputtered MgZn, films. X-ray diffraction results substantiated the crystalline nature of the pure aluminum and 7475 aluminum alloy films but indicated that the $MgZn_2$ films were amorphous. Diffraction patterns from the pure aluminum and 7475 film/substrate samples differed from those of the 7075-T6 substrate in the relative intensity of [311] planes. To corroborate the crystalline nature of the pure aluminum and 7475 aluminum alloy films, these films were sputtered onto cooled copper substrates. In both cases, diffraction results showed the films to be crystalline. Diffraction results from MgZn, film/substrate samples, however, exhibited none of the peaks observed from films sputtered onto hotter substrates. Extensive broadening, extending over 9-10°, was apparent over ranges of 2θ where three or more diffraction peaks had been observed in crystalline films. Therefore, it can be concluded that only short range ordering existed in the MgZn₂ films that were sputtered onto the cooler substrates.

LOCALIZED HYDROGEN MEASUREMENTS

Results of LNMH measurements in as-polished and sputter etched substrates at depths of 0.61, 0.91 and 1.23 mg/cm² (4.25, 4.75, and 5.25 MeV energies, respectively) were below the limit of detectability, 19 wppm, used for these measurements. Longer measurement times or increased beam currents would have been required to reduce the detectability limit further, and this was not considered necessary for the scope of the present work. These samples were examined in different target chamber loadings, and the low hydrogen concentrations, consistent with earlier work (Ref. 16), attest to the absence of background problems throughout these measurements.

The hydrogen concentration results for each of the film/ substrate combinations are summarized in Tables 4, 5, and 6. Hydrogen concentrations in individual samples are indicated for the film, interface region, and substrate for the different conditions of exposure. For pure aluminum film/substrate samples. see Table 4, the hydrogen concentrations in the film and substrate are generally below the limit of detectability used for these samples. The low concentrations measured in the film are indicative of an absence of surface contamination and of relatively little hydrogen trapping during sputtering; the low concentrations in the substrate are as expected for bulk 7075 aluminum alloy and confirm that no contamination problem existed in any of these samples. High hydrogen concentrations exist in the interface region for all conditions examined, but no significant increases appear to be associated with stress corrosion exposure. In fact, the disparity in concentration for a specific condition is comparable to the total range of concentrations observed, i.e., as-sputtered samples have interface concentrations of 121 \pm 20 and 230 \pm 32 wppm, and the total range of hydrogen concentrations measured for this group of samples is 85 to 230 wppm.

For 7475 film/substrate samples, see Table 5, the hydrogen concentration in the substrates is below the limit of detectability; the alloy films have measurable concentrations that are higher than those observed in the pure aluminum films. These higher concentrations are likely a result of hydrogen trapping

	Hydrogen Concentration (wppm)		
Condition	Aluminum	Interface	7075
	Film* ¹	Region* ²	Substrate* ³
As-Sputtered (S)	<39	121 ± 20	<37
	<39	230 ± 32	<24
Corrosion Exposure	<39	116 ± 20	<31
(C)	<39	85 ± 18	<24
Stress Corrosion	50 ± 16	189 ± 25	<30
Elastic* ⁴ (SCE)	<40	169 ± 24	<19
Unloaded SCE (SCE-U)	<39	155 ± 23	67 ± 17
	<36	137 ± 22	<37
Stress Corrosion	<38	229 ± 28	<26
Plastic* ⁵ (SCP)		123 ± 21	<40
Unloaded SCP (SCP-U)	<32	108 ± 17	<28
	60 ± 15	102 ± 19	<19

TABLE 4 HYDROGEN CONCENTRATION IN ALUMINUM FILM/SUBSTRATE SAMPLES

*¹ Depth of 0.61 mg/cm² (2.25μm) below surface *² Depth of 0.91 mg/cm² (3.36μm) below surface *³ Depth of 1.22 mg/cm² (4.50μm) below surface *⁴ Elastic stress of 29 ksi *⁵ Plastic stress of 64 ksi

	Hydrogen Concentration (wppm)			
Condition	7475	Interface	7075	
	Film* ¹	Region* ²	Substrate* ³	
As-Sputtered (S)	54 ± 15	167 ± 23	<19	
	90 ± 16	68 ± 15	<25	
Corrosion	44 ± 12	199 ± 29	<38	
Exposure (C)	84 ± 16	45 ± 13	<38	
Stress Corrosion	52 ± 14	192 ± 25	<19	
Elastic* ⁴ (SCE)	66 ± 17	105 ± 19	<26	
Unloaded SCE	<44	145 ± 22	<19	
(SCE-U)	75 ± 16	104 ± 19	<35	
Stress Corrosion	67 ± 17	248 ± 28	<40	
Plastic * ⁵ (SCP)	71 ± 15	82 ± 17	<19	
Unloaded SCP	72 ± 15	162 ± 23	<25	
(SCP-U)	64 ± 15	58 ± 16	<23	

TABLE 5 HYDROGEN CONCENTRATION IN 7475 FILM/ SUBSTRATE SAMPLES

 $*^1$ Depth of 0.61 mg/cm² (2.25µm) below surface

 $*^2$ Depth of 0.91 mg/cm² (3.36 μ m) below surface

 $*^3$ Depth of 1.22 mg/cm² (4.50µm) below surface

*⁴ Elastic stress of 29 ksi

*⁵ Plastic stress of 64 ksi

Condition	MgZn ₂	Interface	7075
	Film*1	Region* ²	Substrate* ³
As-Sputtered (S)	26 ± 3	214 ± 24	<26
	36 ± 5	203 ± 24	<38
Corrosion	50 ± 6	140 ± 20	<38
Exposure (C)	39 ± 6	101 ± 17	<38
Stress Corrosion Elastic* ⁴ (SCE)	31 ± 4 47 ± 6 53 ± 6	274 ± 29 400 ± 41 170 ± 22	<40 54 ± 14 78 ± 17
Unloaded SCE (SCE-U)	30 ± 4 53 ± 6 58 ± 7	183 ± 23 457 ± 49 168 ± 22	<39 86 ± 18 64 ± 17
Stress Corrosion Plastic* ⁵ (SCP)	52 ± 6	426 ± 47 	70 ± 17
Unloaded SCP	60 ± 7	355 ± 38	41 ± 15
(SCP-U)	48 ± 6	538 ± 54	71 ± 17

TABLE 6 HYDROGEN CONCENTRATION IN MgZn₂ FILM/ SUBSTRATE SAMPLES

*¹ Depth of 0.85 mg/cm² (1.63 μ m) below surface *² Depth of 1.25 mg/cm² (2.40 μ m) below surface *³ Depth of 1.56 mg/cm² (3.54 μ m) below surface *⁴ Elastic stress of 29 ksi *⁵ Plastic stress of 64 ksi

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during sputtering and indicate that the alloy has a greater affinity for hydrogen than pure aluminum. High hydrogen concentrations exist in the interface region, and there appears to be some increase for specific exposure conditions.

For the MgZn₂ film/substrate samples, see Table 6, both film and substrates have comparatively low hydrogen concentrations; slightly higher concentrations exist in most of the substrates exposed to stress corrosion. In the interface region, there is clear indication that stress corrosion exposure, especially under plastic loading, leads to hydrogen enrichment. Only one of three samples has a significantly higher hydrogen concentration after exposure to stress corrosion under elastic loading, SCE, whereas both samples exposed under plastic loading, SCP, have exceptionally high hydrogen concentrations. For one of the SCP samples, failure occurred in a typical stress corrosion manner while stress was maintained before LNMH measurement. This is illustrated by the intergranular cracking and failure mode shown in Fig. 7.

For each of the previously described film/substrate samples, the hydrogen concentration in the interface region was much higher than in the film or substrate. Since the high concentrations exist in as-sputtered samples, they appear inherent to the interface region. To determine whether these high hydrogen concentrations could be attributed to a substrate surface condition, hydrogen concentration depth profiling was done in the surface of as-polished and sputter-etched substrates. These profiles, which are shown in Fig. 8, clearly indicate that the hydrogen concentration within the first 0.20 mm of the substrate is well in excess of 100 wppm. Since the resonance width for the LNMH measurements at the interface

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Fig. 7 Stress Corrosion Failure in MgZn₂ Film/Substrate Sample (a) Intergranular Cracking Through 0.040-inch Thickness (b) Etched Section Adjacent to Fracture Surface The second second


region is approximately 0.20 µm, a significant portion of the substrate surface volume is sampled. Therefore, the high hydrogen concentrations measured in the interface region appear to be directly related with the hydrogen concentration in the surface of the substrate.

Some explanation should also be given for the large differences in hydrogen concentration in the interface region between samples of the same condition. For example, concentrations of 121 ± 20 and 230 ± 32 wppm were measured in the interface region of two as-sputtered pure aluminum film/substrate samples. Throughout these measurements, a constant 'Li beam energy of 4.75 MeV was utilized for determining the hydrogen concentration in the interface region. However, the film thickness of each sample differed so that the LNMH resonance reaction did not occur at the same position relative to the interface in all cases. As a result, the interface region sampled was not always comparable. Consideration of the effect of the relative position of the resonance energy on the hydrogen concentration is given in Table 7 for pure aluminum film/substrate samples. There appears to be no correlation between the relative position of the resonance energy and the measured hydrogen concentration. For the assputtered condition, a lower concentration was indicated when the resonance occurred in the substrate, whereas for stress corrosion under plastic loading, SCP, the reverse was true. In general, the variation in film thickness within the area probed, ± 0.06 mg/cm^2 , coupled with the resonance width, averaged as \pm 0.04 mg/cm², assured that the film, the singularity at the interface, as well as the substrate were sampled. However, because of this uncertainty it does not appear that the contribution of the separate sample volumes can be sufficiently resolved to explain the measured differences.

Condition	Film Thickness (mg/cm ²)	∆* ² (mg/cm ²)	Position of Resonance Energy	Measured Hydrogen Concentration (wppm)
S	0.79 ± 0.06	-0.12	substrate	121 ± 20
S	0.91 ± 0.06	0.00	interface	230 ± 32
C	0.95 ± 0.06	+0.04	interface	116 ± 20
C	0.92 ± 0.06	-0.09	substrate	85 ± 18
SCE SCE	0.92 ± 0.06 0.95 ± 0.06	+0.01 -0.06	interface interface/ substrate	189 ± 25 169 ± 24
SCP	0.79 ± 0.06	-0.12	substrate	229 ± 28
SCP	0.92 ± 0.06	+0.05	interface	123 ± 21

TABLE 7 POSITION OF RESONANCE ENERGY RELATIVE TO INTERFACE IN ALUMINUM FILM/SUBSTRATE SAMPLES*1

 $*^1$ 0.91 ± 0.03 mg/cm² depth of resonance at 4.75 MeV

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 $*^2 \triangle$ = average film thickness - resonance energy depth

In connection with the lack of certainty in the position of the resonance energy, the method of determining the hydrogen concentration in the $MgZn_2$ film/substrate interface region will be described. To calculate the hydrogen concentration in wppm from the measured γ -ray count rate, the elemental constituency of the host material must be known, as indicated by the following (Ref. 16):

$$w_{H} = \begin{bmatrix} \frac{\kappa \left(\frac{N}{O}\right)^{-1} - \varepsilon_{H}}{\sum \frac{M_{H}}{M_{j}} w_{j}^{\circ} \varepsilon_{j}} \end{bmatrix}$$

where w_H = hydrogen concentration

0

K - calibration constant

 $\mathbf{N} = \gamma$ -ray count/unit charge

 $\xi_{\rm H}$ = atomic stopping cross-section of hydrogen at resonance energy

M₁ = atomic weight of species j

j = weight fraction of species j (without inclusion of hydrogen)

stomic stopping cross-section of species j at resonance energy

Because of the difficulty in resolving the position of the resonance energy in relation to the interface, the weight fractions of specific elements in the above equation could not be quantitatively identified. This was not a problem for the pure aluminum film/substrate and 7475 film/substrate samples because the difference introduced by using pure aluminum or the 7475 and 7075 alloys was relatively insignificant. For MgZn₂ film/substrate samples, a comparison is given in Table 8 between considering 100% MgZn₂ and 100% 7075 aluminum alloy for the same measured count rate, i.e., N/Q. As indicated, the concentrations for the 7075 aluminum alloy would be approximately 36 percent higher than for MgZn₂. Rather than assume that the reliability in the film thickness measurement was sufficient in each case to be used in calculating the concentration, a constant average concentration was utilized. In this way, the hydrogen concentration is a direct reflection of the measured count rate, N/Q, instead of a floating value that is dependent on the uncertainty in film thickness. The average values indicated in Table 8 are those used in Table 6.

Although the hydrogen concentration results in the interface region are limited by the number of samples examined and the extent of variation that exists, specific trends become apparent. These trends can be identified by considering the average hydrogen concentration in the interface, normalized to the as sputtered condition, as shown in Table 9. No significant hydrogen enrichment is evident for any of the pure aluminum film/ substrate exposures. In fact, there is a reduction in hydrogen concentration following corrosion exposure as well as accompanying removal of applied stress after stress corrosion exposure. No change occurs in the 7475 film/substrate interface following corrosion exposure, but increases of 26 and 40 percent are indicated for stress corrosion exposure under elastic and plastic loading, respectively. Upon removal of stress, the hydrogen concentration is reduced to the original level. For the MgZn₂ film/substrate interface, a decrease in hydrogen concentration, equivalent to that observed in the aluminum film/substrate interface, accompanies corrosion exposure. However, there is an increase of 34 percent following stress corrosion exposure under elastic loading, SCE, and a very large increase, greater than

	Hydrogen Concentration (wppm)			
Condition	Based on 100% MgZn ₂	Based on 100% 7075	Average	
S	181 ± 20	247 ± 28	$214 \pm 24 \\ 203 \pm 24$	
S	171 ± 20	234 ± 27		
C	118 ± 17	161 ± 23	140 ± 20	
C	85 ± 14	116 ± 20	101 ± 17	
SCE	231 ± 24	316 ± 33	274 ± 29	
SCE	338 ± 34	462 ± 47	400 ± 41	
SCE	143 ± 18	196 ± 25	170 ± 22	
SCE-U	154 ± 19	211 ± 26	183 ± 23	
SCE-U	386 ± 41	527 ± 56	457 ± 49	
SCE-U	142 ± 18	193 ± 25	168 ± 22	
SCP	360 ± 39	492 ± 54	426 ± 47	
SCP-U	300 ± 32	410 ± 43	355 ± 38	
SCP-U	455 ± 45	621 ± 62	538 ± 54	

TABLE 8HYDROGEN CONCENTRATION EVALUATION
FOR MgZn2FILM/SUBSTRATE INTERFACE
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TABLE 9COMPARISON OF AVERAGE HYDROGEN
CONCENTRATION IN INTERFACE REGION

	Normalized Hydrogen Concentration*1		
Condition	Al Film/ 7075 Substrate	7475 Film/ 7075 Substrate	MgZn ₂ Film/ 7075 ² Substrate
As-Sputtered (S)	1.00 (176 wppm)	1.00 (118 wppm)	1.00 (209 wppm)
Corrosion Exposure (C)	0.57	1.03	0.58
Stress Corrosion Elastic (SCE)	1.02	1.26	1.34
Unloaded SCE(SCE-U)	0.83	1.06	1.29
Stress Corrosion Plastic (SCP)	1.00	1.40	2.04* ²
Unloaded SCP(SCP-U)	0.60	0.93	2.14 1.70* ²

*¹ Normalized to indicated as-sputtered average concentration

*² Based on single sample

100%, associated with the stress corrosion exposure under plastic loading, SCP. Upon unloading, a reduction in hydrogen concentration occurred in one of the three SCE samples (see Table 6) and in the SCP sample that did not fail (indicated by asterisk in Table 9). Based on these observations, it appears that three levels of hydrogen enrichment accompany stress corrosion exposure:

1. No change - aluminum film/substrate (all exposures)

2. An approximate 30 - 40 percent increase -7475 film/substrate (SCE & SCP) MgZn₂ film/substrate (SCE)

3. An approximate 100 percent increase -

MgZn, film/substrate (SCP)

In addition, unloading following stress corrosion exposure significantly reduces the existing hydrogen concentration in both aluminum film/substrate and 7475 film/substrate interfaces. Comparable reductions occur in the only half of the MgZn₂ film/ substrate interfaces examined.

STRESS CORROSION MECHANISM

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The intent of the present investigation was to determine whether the constituents at the grain boundary of high strength aluminum alloys had a preferential affinity for hydrogen when exposed to an aqueous chloride stress corrosion environment. If such an affinity exists, a hydrogen related mechanism that leads to the intergranular cracking characteristic of stress corrosion attack in these alloys can be inferred. Since direct measurement of the hydrogen concentration in the grain boundary region is not possible, a simulation was made to enable use of the LNMH technique in measuring the extent of the hydrogen

enrichment associated with stress corrosion exposure in grain boundary constituents and in the interface region between these constituents. Although the properties of the grain boundary state were not achieved by the sputtered film/substrate samples used in this simulation and the sputtered MgZn₂ films were not crystalline, control samples were utilized throughout to enable comparison and discern enrichment. Pure aluminum film/substrate samples provided comparison with a sample configuration that was not susceptible to stress corrosion attack, and the as-sputtered and corrosion exposure conditions served as a normalization for the stress corrosion exposure.

The results obtained in this investigation clearly show that hydrogen enrichment accompanies stress corrosion exposure in both 7475 film/substrate and MgZn, film/substrate interface regions. In addition, the extent of enrichment appears related to the magnitude of the applied stress for the MgZn, film/substrate interface. There was no indication of comparable increases in hydrogen concentration in either the film or substrate of any of the samples examined or in the interface region of the aluminum film/substrate samples. Moreover, the extent of enrichment was particularly high in MgZn, film/substrate interfaces and remained relatively high even when stress was removed. It appears that hydrogen diffused from the aqueous chloride solution into the interface region under the influence of stress and remained trapped when the solution environment was removed; for some of the MgZn, film/substrate samples, hydrogen remained trapped even after stress was removed.

Since stress corrosion cracking in high strength aluminum alloys has been shown to occur at the interface between a MgZn₂ grain boundary precipitate and the adjacent grain (Ref. 8) and

since the distribution of MgZn₂ grain boundary precipitates has been related with stress corrosion susceptibility (Refs. 11-15), the extent of hydrogen enrichment and trapping observed in the simulated MgZn₂/matrix interface of this investigation strongly suggests that hydrogen is of significance in the grain boundary embrittlement associated with stress corrosion.

Finally, it should be pointed out that two actual stress corrosion failures were encountered during this investigation. In both cases, failure occurred while stress alone was being maintained on MgZn₂ film/substrate samples following stress corrosion exposure. All substrates were susceptible to attack since they were prepared from 7075-T6 bar material and were stressed in their short transverse direction. However, only the two samples indicated underwent failure. It appears that the extend of hydrogen enrichment in the MgZn₂ film/substrate interface was of sufficient magnitude to cause attack in the substrate. These failures lend supporting evidence to the hypothesis that the extensive hydrogen enrichment at MgZn₂/ matrix interfaces which accompanies stress corrosion in an aqueous chloride environment promotes stress corrosion attack.

4. CONCLUSIONS

- Use of a sputtered film/substrate sample configuration provides a suitable means for simulating the grain boundary constituency of 7000 series aluminum alloys.
- 2. There is significant hydrogen enrichment in the interface region of MgZn₂ film/substrate samples exposed to stress corrosion in an aqueous chloride solution; a lesser degree of enrichment occurs in the interface of 7475 film/7075 substrate samples; and no enrichment occurs in pure aluminum film/7075 substrate samples.
- 3. The extent of hydrogen concentration increase during stress corrosion exposure appears related to the magnitude of the applied stress.
- 4. The affinity for hydrogen in the simulated grain boundary interface between MgZn₂ and the matrix appears significant to the mechanism of stress corrosion attack in 7000 series aluminum alloys.

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5. REFERENCES

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1.	Mondolfo, L.R., <u>Metals Mater</u> ., Vol. 5, p. 95, 1971.
2.	Ostermann, F. G. and Reimann, W. H., <u>ASTM STP 467.</u> p. 169, 1970.
3.	Sprowls, D. O. and Brown, R. H., <u>Metals Progr</u> ., Vol. 81, No. 4, p. 79, 1962.
4.	Sprowls, D. O. and Brown, R. H., <u>Metals Progr</u> ., Vol. 81, No. 5, p. 77, 1962.
5.	Speidel, M. A., in <u>Hydrogen in Metals</u> , ed. by I. M. Bernstein and A. W. Thompson, ASM, p. 249, Metals Park, Ohio, 1974.
б.	Green, J.A.S., Hayden, H. W., and Montague, W.G., in <u>Effect of Hydrogen on Behavior of Materials</u> , ed. by A. W. Thompson and I. M. Bernstein, TMS-AIME, p. 200, New York, 1976.
7.	Gruhl, W. and Brungs, D., <u>Metall</u> ., Vol. 23, p. 1020, 1969.
8.	Montgrain, L. and Swann, P.R., in <u>Hydrogen in Metals</u> , ed. by I. M. Bernstein and A. W. Thompson, ASM, p. 575, Metals Park, Ohio, 1974.
9.	Gest, R. J. and Troiano, A.R., <u>Corrosion</u> , Vol. 30, p. 274, 1974.
10.	Sprowls, D.O. and Brown, R.H., <u>Proc. of Conference on</u> <u>Fundamental Aspects of Stress Corrosion Cracking</u> , ed. by R.W. Staehle, A. J. Forty, and D. van Rooyen, NACE, p. 498, Houston, Texas, 1969.
11.	Adler, P. N., DeIasi, R., and Geschwind, G., <u>Met. Trans.</u> , Vol. 3, p. 3191, 1972. 43 <i>Preceding Page Blank</i>

- Geschwind, G. and Adler, P.N., International Corrosion Conference - 73, Paper No. 97, NACE, Houston, Texas, 1973.
- Unwin, P.N.T. and Nicholson, R.B., <u>Acta Met.</u>, Vol. 17, p. 1379, 1969.
- 14. Kent, K.G., J. Aust. Inst. Metals, Vol. 15, p. 171, 1970.
- Poulose, P.K., Morral, J.E., and McEvily, A.J., <u>Met. Trans.</u>, Vol. 5, p. 1393, 1974.
- 16. Padawer, G.M., Adler, P.N., Kamykowski, E.A., Schneid, E.J., Kuehne, F.J. Jr., Stauber, M.C., and D'Agostino, M.D., "Development of a Nuclear Microprobe Technique for Hydrogen Analysis in Selected Materials," Final Report on NASC Contract N00019-72-C-0404, Grumman Research Department Report RE-464, Oct. 1973.
- 17. Padawer, G.M., Adler, P.N. and Stauber, M.C., "Further Development and Application of the Lithium Nuclear Microprobe for Hydrogen," Final Report on NASC Contract N00019-74-C-0176, Grumman Research Department Report RE-522, August 1976.
- 18. Adler, P.N., Padawer, G.M., and Kamykowski, E.A., Proc. of Tri-Service Corrosion of Military Equipment Conference. Vol. II, p. 25. AFML-TR-75-42, WPAFB, Dayton, Ohio, October 1974.
- 19. Adler, P.N., and Padawer, G.M., in <u>Effect of Hydrogen in</u> <u>Behavior of Materials</u>, ed. by A.W. Thompson and I. M. Bernstein, TMS-AIME, p. 698, New York, 1976.
- 20. Pearson, W.B., <u>Lattice Spacings and Structures of Metals</u> and <u>Alloys</u>, Peragemon Press, New York, 1958.

21. Guillet, L and LeRoux, R., in <u>Intermetallic Compounds</u>, ed. by J. H. Westbrook, P. 456, John Wiley and Sons, New York, 1967.

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22. Whaling, W., The Energy Loss of Charged Particles in Matter, Handbuch der Physik, Springer-Verlag, Vol. 34, p. 193, 1958.

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