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DAVID W. TAYLOR NAVAL SHIP RESEARCH AND DEVELOPMENT CENTER

Bethesda, Md. 20084

CAVITATION PERFORMANCE CHARACTERISTICS OF

SUPERCAVITATING PROPELLERS 4698 AND 4699

by

James G. Peck

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SHIP PERFORMANCE DEPARTMENT DEPARTMENTAL REPORT



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December 1977

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NOTATION

Dimensions

Blade-section length at 0.7 R	ft, m
Propeller diameter	ft, m
Advance coefficient $J = V_a/nD$	
Torque coefficient $K_Q = Q/\rho n^2 D^5$	
Thrust coefficient $K_T = T/\rho n^2 D^4$	
Loading coefficient	
Revolutions per unit time	rps
Ambient static pressure	$1b/ft^2$, N/m^2
Ambient vapor pressure	1b/ft ² , N/m ²
Torque	ft-1b, N·m
Thrust	1b, N
Velocity of boat	ft/sec, m/sec
Speed of advance $V(1 - w)$	ft/sec, m/sec
Thrust deduction fraction	
Wave fraction	
Propeller efficiency $n = K_T \cdot J/K_Q \cdot 2\pi$	
Mass density of water	lb-sec ² /ft ⁴ , Kg/m ³
Cavitation number $\sigma = (P - P_v)/(\frac{1}{20}v^2)$	
	Blade-section length at 0.7 R Propeller diameter Advance coefficient $J = V_a/nD$ Torque coefficient $K_Q = Q/\rho n^2 D^5$ Thrust coefficient $K_T = T/\rho n^2 D^4$ Loading coefficient Revolutions per unit time Ambient static pressure Ambient vapor pressure Torque Thrust Velocity of boat Speed of advance V(1 - w) Thrust deduction fraction Wave fraction Propeller efficiency $n = K_T \cdot J/K_Q \cdot 2\pi$ Mass density of water Cavitation number $\sigma = (P - P_V)/(\frac{1}{20}V^2)$



*

ABSTRACT

Two model supercavitating propellers were characterized over a range of cavitation numbers and advance coefficients. The results of these characterizations are presented in curves and tabular form. The propeller performance as predicted from these experiments is compared to design prediction. It was found that one of the propellers would reach 95 percent of design speed at design rpm whereas the other propeller would reach 109 percent of its design speed at design rpm.

ADMINISTRATIVE INFORMATION

This work was performed under Naval Material Command funding. Program identification at the David W. Taylor Naval Ship R&D Center was Work Unit 1-1500-200, Task Area ZF 43421001, and Project Element 62543N.

INTRODUCTION

David Taylor Naval Ship Research and Development Center (DTNSRDC) conducted an experimental program to evaluate two present day methods of supercavitating propeller design.

Two supercavitating propellers were designed, independently, for a typical 200 ton (181 metric tons) hydrofoil craft. The propeller design conditions are given in Table 1. The design methods are published in References 1 and 2.* The propeller design characteristics are given in Table 2.

Two model propellers were manufactured at DTNSRDC from these designs. The principal dimensions of these propellers are given in Table 3, and drawings of the propellers are shown in Figures 1 and 2. After manufacture the propellers were measured to verify the manufacturing process. Computer plots of the measured values of blade section offsets

* References are listed on page 4

compared to design values are given for the 0.7 R section of one blade of each propeller in Figures 3 and 4. The sections are typical of all the sections of all the blades.

EXPERIMENTAL PROCEDURE

Cavitation performance characteristics of the propellers were obtained in the 36" Variable Pressure Water Tunnel. Tunnel water velocities were measured on the tunnel venturi system. The scope of the experiments is listed in Table 4.

Tunnel pressure and water velocity were set to establish each cavitation number and then propeller revolutions were varied to cover a range of speed coefficients. Propeller thrust and torque were measured at each condition, and sketches were made of the propeller cavitation present. The Reynolds numbers during the experiments ranged from 7.5 x 10^5 to 5.6 x 10^6 .

PRESENTATION OF DATA AND DISCUSSION

The thrust and torque data were reduced to non-dimensional coefficients K_T and K_Q . Propeller efficiencies were calculated from faired values of K_T and K_Q . The cavitation performance characteristics of the two propellers are presented in Tables 5 and 6.

Curves representing the faired data from Tables 5 and 6 at three cavitation numbers for each propeller are shown in Figures 5 and 6. These

curves show the performance of Propellers 4698 and 4699 to be typical of supercavitating propellers. Photographs of the propellers, at their design-operating condition, are shown in Figure 7 and Figure 8.

Curves of maximum speed thrust loading (K_T/J^2) have been added to the performance curves (Figure 5 and 6). The intersection of the K_T/J^2 curve and the K_T curve at the design sigma determines the predicted operational point for each propeller. A comparison between the design operational points and the points determined by the experimental data are listed in Table 7.

Sketches of back cavitation present on the propellers at two cavitation numbers are given in Figures 9 and 10. These sketches cover a range of advance coefficient from partially cavitating to fully cavitating conditions. The propellers at any advance coefficient lower than those shown, at the same cavitation number, would also be fully cavitating. There was no face cavitation present on either propeller over the range of cavitation numbers and advance coefficients covered in these experiments.

CONCLUSIONS

Although both propellers had no face cavitation and had essentially full back cavitation at the design operational point neither propeller performed as predicted. The method used in designing Propeller 4698 over predicted the available thrust. Propeller 4698 would require 5 percent more rpm and 19 percent more power than predicted to reach design speed. The method used in designing Propeller 4699 under predicted the available

thrust. Propeller 4699 would reach design speed with 8 percent less rpm and 17 percent less power than predicted. If the operating point is defined as the speed and rpm where the propellers absorb the available maximum power, Propeller 4698 would operate at V = 55.6 knots and rpm = 1000; while the operating point for Propeller 4699 would be V = 61.0 and rpm = 970. The propeller efficiencies at these conditions are 52% and 73%, respectively.

REFERENCES

1. Bohn, J. and Altmann, R., "Two Supercavitating Propeller Designs for Hydrofoil Ships," Hydronautics, Inc. Technical Report 7607.01 (May 1976).

2. Baker, Elwyn S., "Engineering Design of Two Supercavitating Propellers Using Modified Lifting Surface Programs," DTNSRDC Report SPD-680-01 (Aug 1977).









1.3 5 2-* 1.2 : 1.0 ò = 0.088 6.0 0.7 0.8 ADVANCE COEFFICIENT J TK_1/J2 12.0 0.34 0.34 0.34 9.0 12.0 0.5 12.0 ..4 3.0 3.01 3.0 0.3 0.7 9.0 0.2 0.5 .. 0.3 1.0 0 " Pue 0, 01 . 1

Figure 5 - Cavitation Performance of Propeller 4698 at Three Cavitation Numbers

1.3 1.2 5 2 = 0.087 Figure 6 - Cavitation Performance of Propeller 4699 at Three Cavitation Numbers 2 K7/3 1.0 1 6.0 ADVANCE COEFFICIENT J 0.8 26.2 0.7 0.34 1.0 NE.0 9.0 0.5 29.5 26.2 •.0 3.0 3.0 3.0 0.3 0.2 0.6 0.5 .. 0.3 1.0 0.7 0

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 $\sigma = 3.0$



 $\sigma = 0.34$

Figure 9 - Sketches of Cavitation Present on the Back of Propeller 4698 at Two Cavitation Numbers





 $\sigma = 0.34$

Figure 10 - Sketches of Cavitation Present on the Back of Propeller 4699 at Two Cavitation Numbers

TABLE 1 PROPELLER DESIGN CONDITIONS

Power, max. cont.	16000 HP (16213.3 Metric HP) for two propellers
1 - t	0.925
1 - w _t	0.870
Shaft Centerline at Design Speed	6.82 ft (2.079 m)
Maximum D	5 ft (1.524 m)
No. of Blades	4
Design Objective	maximum speed
Minimum Rotative Speed	750 rpm
Hump Thrust Margin	20%

TABLE 2

PROPELLER DESIGN CHARACTERISTICS

Propeller Number Designed by V _S	<u>4698</u> Hydronautics, Inc. 58.5 kts	<u>4699</u> DTNSRDC 59.0 kts
N _S	1020	1000
К _Т	0.090	0.0943
J	1.01	1.04
٥	0.34	0.34
n (147) (17)	0.62	0.64

TABLE 3

MODEL PROPELLER GEOMETRY

Propeller Number	4698	4699
Number of Blades	4	4
Diameter, Inches (m)	16.000 (.4064)	16.000 (.4064)
Pitch at 0.7R, Inches (m)	22.389 (.5687)	22.683 (.5761)
P/D at 0.7R	1.399	1.418
Expanded Area Ratio	0.350	0.492
Chord Length at 0.7R, Inches (m)	3.421 (.0869)	4.547 (.1155)

TABLE 4

SCOPE OF EXPERIMENTS

	Propel1	er 4698		Propeller 4699	
σ	V fps	J	σ	V fps	J
12.0	15.0	0.5-1.3	26.2	15.0	0.45-1.5
6.0	15.0	0.4-1.5	3.0	15.0	0.3-1.2
3.0	30.0	0.337-1.2	1.5	30.0	0.5-1.2
0.4	35.0	0.7-1.2	0.75	30.0	0.5-1.2
0.34	35.0	0.7-1.2	0.4	35.0	0.7-1.2
0.3	35.0	0.7-1.2	0.34	35.0	0.7-1,2
			0.3	35.0	0.7-1.2

TABLE 5

CAVITATION PERFORMANCE OF PROPELLER 4698

	STOMA	= 12,000			SIGMA	- 6,000		
J	KTOUT	1 OKnOUT	EFFIC	J	KTOUT	IOKONUT	EPPIC	
.5000	. 3240	. 7790	.3310	.4000	.1916	.4952	.2463	
.5500	.3479	. 8364	.358A	.4500	.2195	.5456	.2001	
.6500	. 3410	.8125	. 3874	.5000	.2430	.5953	. JZ4A	
.7000	. 1050	.7680	.4424	.6000	.2734	.6726	.3887	
.7500	,2807	.7177	.4669	.6500	.2797	.6940	.4169	
	.2559	.6679		.7000	.2799	.7022	.4441	
.9000	.7090	.5800	.5162	.8000	2634	6795	.4936	
.9500	.1867	.5411	.5214	.8500	.2477	.6510	.514A	
1.0000	. 1642	.5028	.5197	.9000	.2279	.6136	.5320	
1.1000	.1160	.4190	.4847	1.0000	1790	.5214	.8465	
1.1500	.0898	.3711	.4429	1.0500	.1516	.4714	.5373	
1.2600	.0631	. 3207	.3759	1.1000	.1231	.4216	.5110	
1.3000	.0140	.2350	.1585	1.2000	.0656	. 1286	.3812	
				1.2500	.0376	.2865	.2612	
				1.3000		.2464	.0481	
				1.4000	0412	.1627	564"	
	STGMA	- 3,000			STGMA	400		
J	KTOUT	INKOOUT	EFFIC	J.	KTOUT	1050001	EFPIC	
. 3000	.0779	.2220	.1675	.7000	.0970	.2620	.4124	
.3500	.1007	.2806	.1991	.7500	.0987	.2645	4454	
.4500	.1703	.3302	.2520		OAAA	.2580	.4637	
.5000	.1572	.4147	.301A	.9000	.0848	.2513	.4631	
.5500	.1747	.4524	.33An	.9500	.0867	.2417	.5679	
.6501	2055	5183	4102	1.0500	.0896	.2416	.6194	
.7000	.2172	.5442	.4448	1.1000	.0972	.2441	.6613	
.7500	.2252	.5633	.4773	1.2000	.0977	.2405	.6987	
.8500	.2763	.5747	.5327					
.9000	.2102	.5647	.5934					
.9500	.1847	-5434	.5076					
1.0500	.1601	.4714	.5677					
1.1000	.1734	.4255	.5486					
1.2000	.0441	.3389	.473A					
	SIGMA	- ,340			21644	• .300		
•	KTMIT	IRKONUT	EFFIC	J	KTOUT	IOKONUT	EFFIC	
.7000	.0919	.2600	. 3934	.7600	.0940	.2603	4190	
.7500	.0957	.2610	4650	.8000	.0939	.2607	.4586	
	.0917	.2578	.4815	.8500	.0441	. 2574	.4948	
.9000	.0869	.2508	.4963			.2406	.5303	
1.0000	.0795	,2330	5435	1.0000	.0791	.2304	.5461	
1.0500	.0791	.7274	.5017	1,0500	.0773	.2217	.5028	
1.1000	.OAOA	. 2260	.6263	1.1500		.2143	.7110	
1.2000	.0975	.2292	.6960	1.2000	.0740	.2151	.7017	
1.2500	.0770	.2211	.6923					
1.3000		.1404	.0193					
				DECT			-	
				RF7	AVA	ILAB		IPY
			1	7				

TABLE 6

CAVITATION PERFORMANCE OF PROPELLER 4699

	-	26,200			576ma	. 3,000		
L	KTOUT	INKONUT	EFFIC	J	KTOUT	1040001	EFFIC	
		1.0128	. 1194	. 3000	.1001	.2448	.1955	
.5000	.4271	.9514	. 3977	. 3500	.1229	.2470	.2304	
	.4035	.4963	. 3940	.4000	.1495	. 3576	.2661	
	. 3401	, 8457	.0294	5000	2048		3377	
.7880	.1119	.7546	.4929	.5500	,2794	.5385	.3732	
.7500	.3101	.7124	.5196	.6000	.2504	.5859	.4081	
	.2857	.6714	.5410		.2000			
. 9000	2351	5906	.5766	.7500	.2790	.6568	.5070	
.9=00	.2087	.5501	.\$737	.8000	.2755	.4537	.5367	
1.0000	-1417	.5087			.2497	60%	.505A	
1.1000	.1259	.4223	.5221		,227A	.5719	.6027	
1.1500	.097)	.3768	.4718	1.0000	,2019	.5771	.60%6	
1.2000	.0677	. 3294	. 3423	1.1000	.14 10	.4 105	.5014	
1. 3000	.0064	. 2288	.0579	1,1500	,1137	. 3877	. 5366	
1.3500	075A	.1752	3166	1.7000	.0A75	. 3556	.400-	
1.4500	0950	.0610						
1.5000	1331	.0002						
	\$16ma				-	. 750		
	310-1	• 1.500			310			
,	KTOUT	1040001	FFFIC	J	KTMIT	IOKOOUT	EFFIC	
.5500	.1350	1368	.2510	5500	.1141	.2835	.3521	
.6001	.1510	. 3713	. 3907	.6000	,1160	.2910	. 3467	
. 7000	-1698			.7000	-1150	.2963	.4100	
.7500	.2027	.4773	.5070	.7500	.1240	. 3066	.445A	
	.2144	.5014	.5444		.1357	. 3733	.5326	
.9000	.2224	5198	.6128	.9000	.1592	. 3635	.6273	
	.2144	.5119	.6393	.9500	.1669	. 1780	.6475	
1.0000	.2034	.6929		1,0000	1612	. 3828		
1.1000	.1431	.4241	.6453	1.1000	.1464	.355A	.7203	
1.1500	.1441	. 3909	.6550	1.1500	.1262	. 3277	.7047	
					Sel les			
	5164A				STGMA	. :340		
1-2-5-5				1000		147-017		
,	RTOUT	1040001	EPPIC	.7.440	.1120	. 2820		
.7400	.1196	.78.39	.502A	.7500	.1131	.2418	.4792	
	-1149	.2010	.5205	.8000	,1172	.2611	.5082	
	.1040	2660	.5630	.9000	.1033	.2644	. 5594	
	.1064	.2621	.6142	.9580	.099A	.7545	.5929	
1.0000	-1126	. 2742		1.0500	1054	.2548		
1.1000	.1734	.7849	. 1594	1.1000	.1124	.7645	. 1439	
1.1500	.1186	.2853	.1610	1,1500	,113A	. 2680	1975	
1.5000	.1000	.2530		1.200		North Mark		
	STOMA							
.7000	RTOUT	1040007	tric					
.7400	.1087	.2735	.4745		mar x			
	.1170	.2784	.5207	Selet.	10 24-			
	.1097	.2664	SARA			That Profile		
	.1415	.7510	.6117					
1.0400	.0975	.2353	.6927					
1.1000	.1054	.7429	.7595		. Hereit			
1,2000		1505	.7730	DICT	11/	A	11	
				Dr/	AV		11	FIDV
						NLAI	JLE	LUPT

Propeller Number	4	4698	4699	
	Design	Model Data	Design	Model Data
V kts	58.5	58.5	59.0	59.0
J	1.01	0.96	1.04	1.13
N	1020	1073	1000	920
η	0.62	0.52	0.64	0.77

TABLE 7 PROPELLER OPERATIONAL POINTS

DTNSRDC ISSUES THREE TYPES OF REPORTS

1. DTNSRDC REPORTS, A FORMAL SERIES, CONTAIN INFORMATION OF PERMANENT TECH-NICAL VALUE. THEY CARRY A CONSECUTIVE NUMERICAL IDENTIFICATION REGARDLESS OF THEIR CLASSIFICATION OR THE ORIGINATING DEPARTMENT.

2. DEPARTMENTAL REPORTS, A SEMIFORMAL SERIES, CONTAIN INFORMATION OF A PRELIM-INARY, TEMPORARY, OR PROPRIETARY NATURE OR OF LIMITED INTEREST OR SIGNIFICANCE. THEY CARRY A DEPARTMENTAL ALPHANUMERICAL IDENTIFICATION.

3. TECHNICAL MEMORANDA, AN INFORMAL SERIES, CONTAIN TECHNICAL DOCUMENTATION OF LIMITED USE AND INTEREST. THEY ARE PRIMARILY WORKING PAPERS INTENDED FOR IN-TERNAL USE. THEY CARRY AN IDENTIFYING NUMBER WHICH INDICATES THEIR TYPE AND THE NUMERICAL CODE OF THE ORIGINATING DEPARTMENT. ANY DISTRIBUTION OUTSIDE DTNSRDC MUST BE APPROVED BY THE HEAD OF THE ORIGINATING DEPARTMENT ON A CASE-BY-CASE BASIS.