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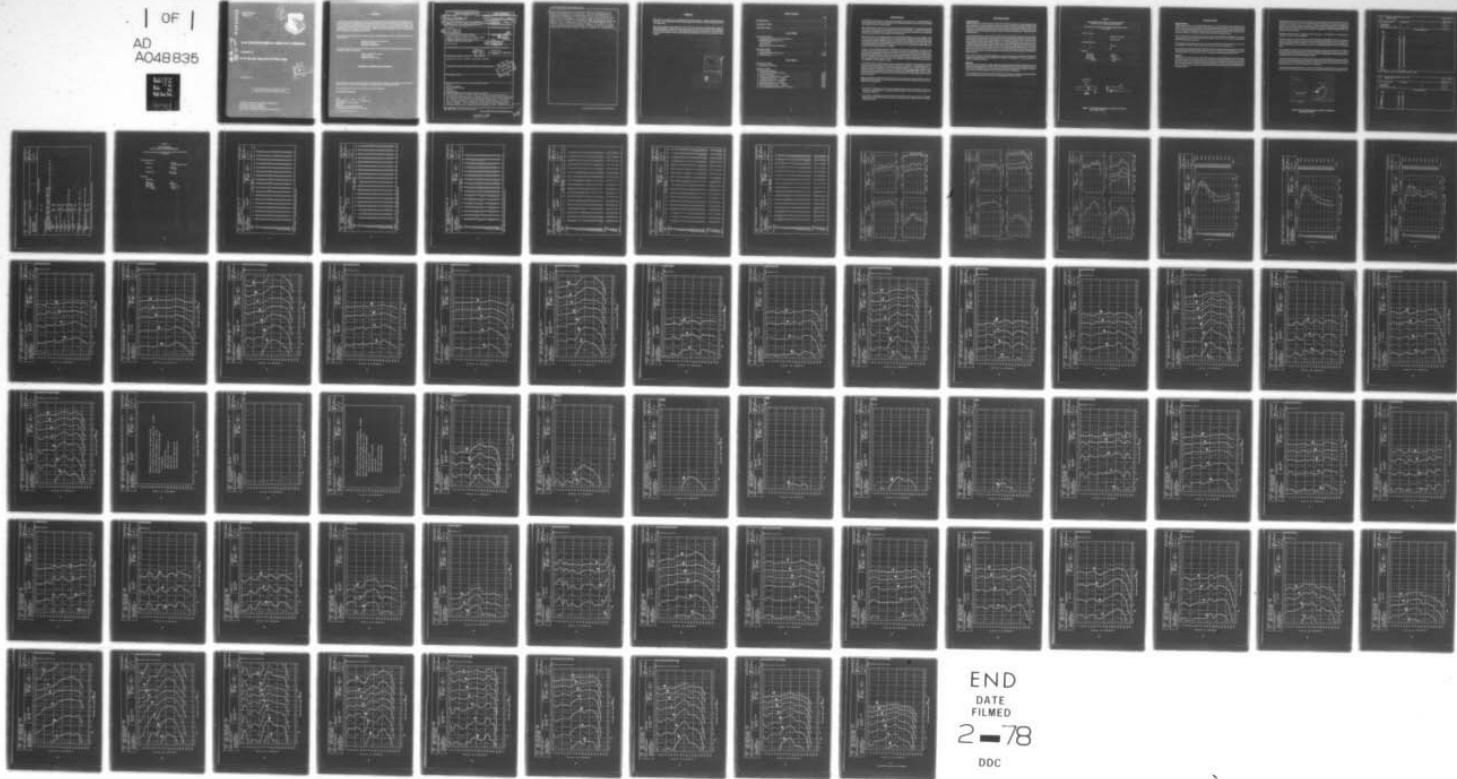
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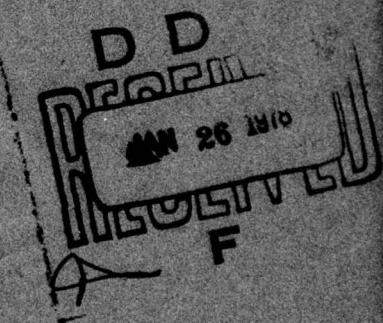
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USAF BIOENVIRONMENTAL NOISE DATA HANDBOOK

Volume 77

A-1E Aircraft, Near and Far-Field Noise

FEBRUARY 1977



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AEROSPACE MEDICAL RESEARCH LABORATORY
AEROSPACE MEDICAL DIVISION
AIR FORCE SYSTEMS COMMAND
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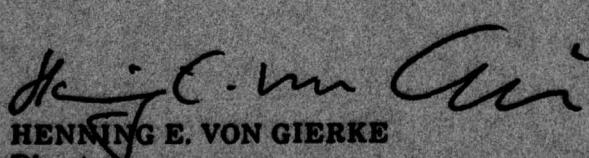
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FOR THE COMMANDER


HENNING E. VON GIERKE
Director
Biodynamics and Bionics Division
Aerospace Medical Research Laboratory

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level, and limiting times for total daily exposure of personnel with and without standard Air Force ear protectors. Far-field data measured at 19 locations are normalized to standard meteorological conditions and extrapolated from 50-8000 meters to derive sets of equal-value contours for these same seven acoustic measures as functions of angle and distances from the source. Refer to Volume 1 of this handbook, 'USAF Bioenvironmental Noise Data Handbook, Vol 1: Organization, Content and Application', AMRL-TR-75-50(1) 1975, for discussion of the objective and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc.

PREFACE

This report was prepared by the Biodynamic Environment Branch, Aerospace Medical Research Laboratory, under Project/Task 723104, Measurement and Prediction of Noise Environments of Air Force Operations.

The author gratefully acknowledges Mr. John Cole for his assistance in preparing this report, Mr. Robert England for his assistance in acquiring the raw data, Mr. Keith Kettler, Mr. Henry Mohlman and Mr. David Eilerman of the University of Dayton for assistance in the mechanics of data processing, and Mrs. Norma Peachey and Mr. Mike Patterson for assistance in typing and preparation of the graphics.

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INTRODUCTION

The USAF A-1E Skyraider is a tactical close-air-support aircraft powered by an R3350-26WD reciprocating engine. The aircraft was manufactured by McDonnell Douglas and the engine by the Wright Aeronautical Division of Curtiss Wright.

This volume provides measured and extrapolated data defining bioacoustic environments produced by this aircraft during ground runup operations. Such data are essential to evaluate ear protection requirements, limiting personnel exposure times, voice communication capabilities, and annoyance problems associated with ground runups of the A-1E aircraft.

This volume is one of a series published by the Aerospace Medical Research Laboratory (AMRL) under the same report number (AMRL-TR-75-50) as a multi-volume handbook that quantifies the noise environments produced at flight/ground crew locations and in surrounding communities by operations of Air Force aircraft and ground support equipment. The far-field, community-type, noise data in the handbook describe the noise produced during *ground operations* of aircraft, ground support equipment, and other ground-based equipment or facilities.

Volume 1 of this handbook discusses the objectives and design of the handbook, the types of data presented, measurement procedures, instrumentation, data processing, definitions of quantities, symbols, equations, applications, limitations, etc. Volume 2 provides a method and data for adjusting the handbook's far-field noise data, which are for standard meteorological conditions (15°C temperature, 70% rel humidity, 0.760 meters Hg barometric pressure), to derive comparable data for other meteorological conditions. Refer to Volumes 1 and 2 (references 2 and 3) for such information because it is not repeated in other handbook volumes.

A cumulative index lists those aerospace systems contained in the handbook, and identifies the specific volumes containing each type of environmental noise data available (i.e., inflight/flight crew and passenger noise, near-field/ground crew noise, far-field/community noise). Volume numbers are assigned sequentially as individual volumes are published. This index is periodically updated as individual volumes are published and is available upon request from AMRL/BBE, Wright-Patterson AFB, OH 45433. Organizations on the distribution list for the handbook will automatically receive a copy of each updated index.

Direct any questions concerning the technical data in this report and other handbook volumes to: AMRL/BBE, Wright-Patterson AFB, OH 45433; AUTOVON 78-53675 or 78-53664; Commercial (513) 255-3675 or (513) 255-3664.

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1. Cole, John N., *USAF Bioenvironmental Noise Data Handbook Volume 1: Organization, Content and Application*, AMRL-TR-75-50 (1), Aerospace Medical Research Laboratory, Wright-Patterson Air Force Base, Ohio, 1975.
 2. Cole, John N., *USAF Bioenvironmental Noise Data Handbook, Volume 2: Procedure to Evaluate Effects of Non-standard Meteorological Conditions on Far-Field Noise*, AMRL-TR-75-50 (2), AMRL, WPAFB, OH, 1975.

NEAR-FIELD NOISE

MEASUREMENTS

AMRL acquired near-field noise data on the A-1E aircraft during ground runup operations of its reciprocating engine. For these tests the aircraft was located on a concrete runup pad at Hurlburt Field, Eglin AFB, with no significant reflecting surfaces in the vicinity except the ground plane. Table 1 gives the surface meteorological conditions and the engine condition. The ground-crew chief selected power conditions and near-field locations generally used during routine maintenance or engine runup for preflight checks.

At each near-field location a test engineer randomly moved a hand-held microphone in and around each location, probing all areas where a crew member's head would normally be located. He recorded all the noise samples on magnetic tape. During analysis of each sample, he determined the one-third octave band root-mean-square sound pressure using a 4- or 8-second integration time to derive a power-averaged level for each location. Figure 1 shows the two near-field locations where ground crews are usually located for maintenance and/or preflight checkout operations. Estimates of noise levels at other locations are difficult in the near-field since the noise source is spatially distributed, i.e., not a point source. The noise levels at near-field locations can vary widely depending upon relative distances from each noise source (intake noise, exhaust noise, panel resonances, internal engine noise through the engine wall, etc.).

Table 1 lists the numeric/alphabetic designators used on the data pages in this report to identify the measurement locations and test conditions. For example, the designator 1/A means ground crew location 1 and test condition A.

RESULTS

The measured data presented in Table 2 define the sound pressure levels (SPL) produced by the A-1E aircraft at the two ground crew locations. This table includes the overall, 1/3 octave band, and octave band levels. From these data one can calculate the variety of measures given in Table 3, which are widely used to assess the effects of noise on personnel and their performance.

All near-field data are for the meteorological conditions at the time of test but are valid for all typical airbase meteorology because of the short sound propagation distances involved.

TABLE 1
MEASUREMENT LOCATIONS AND TEST CONDITIONS
FOR NEAR-FIELD NOISE MEASUREMENTS

A-1E Aircraft, Ground Runup, Hurlburt Field, Eglin AFB, 6 Aug 1971
Tail # 52436

Ground Crew Location

1	Engine Start, Fire Guard
2	Wheel Chock Pull

Aircraft Engine Operation

A	Taxi Power
B	Idle

Meteorology

Temperature	28.9 C
Bar Pressure	0.763 M Hg
Rel Humidity	73 %
Wind — Speed	1.5 M/Sec (3 kt)
— Direction	55 Deg

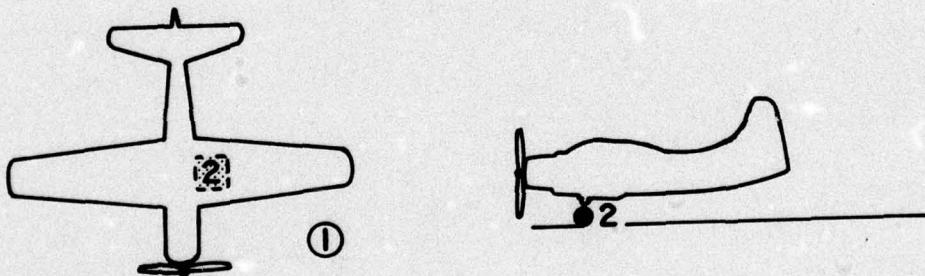


Figure 1. Near-Field Measurement Locations at Hurlburt Field, Eglin AFB FL

FAR-FIELD NOISE

MEASUREMENTS

AMRL acquired both near- and far-field data during a 1- 2-hour test period, thus keeping similar meteorological conditions. Figure 2 shows the ground runup pads, ground cover, aircraft orientation and the 19 microphone measurement sites on the semicircle. The center of the 30 meter radius semicircle used in surveying the R3350-26WD engine was on the ground directly below the intersection of the aircraft's centerline and the plane passing through the engine's propeller plane.

Table 4 provides cockpit readouts of some engines characteristics (RPM and manifold pressure) for each power setting used in the far-field tests. Also listed in this table are the surface meteorological conditions during data acquisition.

All microphone measurement sites are in the acoustic far-field of the source where the sound wave-fronts spherically diverge and the noise source may be regarded as a point source.

Test personnel acquired far-field noise data at Eglin AFB by using a hand-held microphone(1.7 meters/ 5½ feet above the ground plane and pointed at the source, 0° incidence) and sequentially recording 5 to 10 seconds of data at each far-field location on a portable microphone/tape recorder system.

RESULTS

Table 5 lists the overall and 1/3 octave band SPL measured at the far-field locations under meteorological conditions at the time of the test. Data in all other figures and tables are based on these levels. These data were normalized to 100 meters distance and standard meteorological conditions (15°C temperature, 70% relative humidity, 0.760 meter Hg barometric pressure) and used to derive the graphic data in Figure 3, which provides a compact summary of the far-field noise characteristics of the A-1E aircraft in a standard format.

Figure 4 and Table 6 present two acoustic measures, the acoustic power levels and the directivity index, respectively. The acoustic power level describes the power radiated by the source as a function of frequency. The directivity index is a standard acoustical engineering measure that describes the geometric way in which the source radiates this power as a function of both frequency and angle from source. These basic source measures are primarily of interest for acoustical engineers and noise generation/control specialists.

Estimates of noise levels for intermediate power settings (e.g., 1800 RPM) can be determined as explained in Volume 1 of this handbook.

Figures 5 through 11 are sets of equal noise contours describing seven different measures of noise as a function of angle and distance from the source for standard day meteorology. They are, respectively, overall sound pressure level, C-weighted sound level, A-weighted sound level, perceived noise level, speech interference level, permissible exposure times for personnel and octave band sound pressure levels.

Data excessively influenced by spurious background/electronic noise were eliminated from all figures and tables. No data are presented at the 170 and/or 180 degree locations for the highest power settings because of turbulent air flow behind the aircraft. Typically, the A-weighted levels for these angles are 10 to 20 dBA below the level at the 160 degree location.

Test personnel performed noise surveys during quiet periods when the background noise was minimal, e.g., early in the morning when no other aircraft or engine test stands were operating. Data eliminated because they were near the background/electronic noise were generally not significant because the levels were so low (e.g., Table 5 and Figure 11 at 1200 RPM).

Volume 2 of the handbook describes the influence of meteorology on far-field noise environments, and provides, if required, the factors necessary to adjust the handbook's standard meteorological day data.

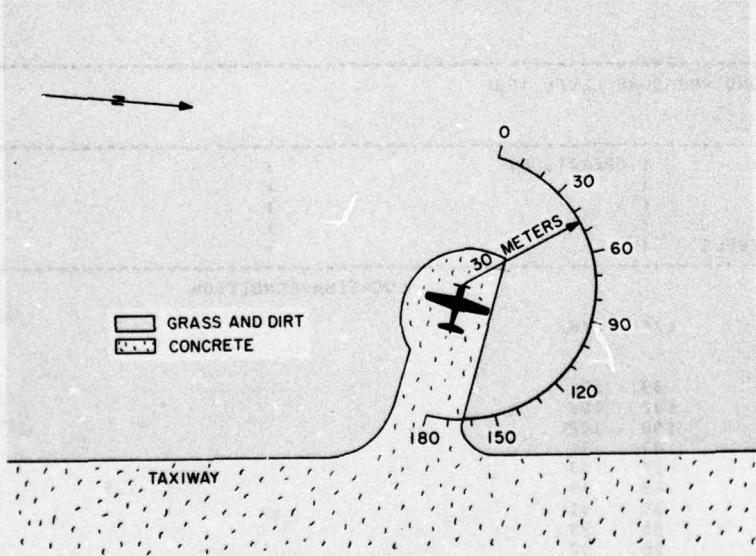


Figure 2. Far-Field Measurement Locations at Hurlburt Field, Eglin AFB FL

TABLE 2 MEASURED SOUND PRESSURE LEVEL (DB) 1/3 OCTAVE BAND			IDENTIFICATION
NOISE SOURCE/SUBJECT: (OPERATION:)) OMEGA 3.2) TEST 71-019-001) RUN 01) 04 DEC 74) PAGE F1
A-1E AIRCRAFT ())
GROUND CREW ())
NEAR FIELD NOISE LEVELS ())
LOCATION/CONDITION			
FREQ (HZ)	1/A	2/B	
25	72	95	
31.5	83	91	
40	88	100	
50	91	99	
63	99	106	
80	98	101	
100	97	100	
125	95	97	
160	90	90	
200	85	86	
250	83	88	
315	83	91	
400	81	90	
500	82	87	
630	83	86	
800	82	84	
1000	80	84	
1250	80	85	
1600	81	86	
2000	80	87	
2500	81	87	
3150	79	85	
4000	83	85	
5000	81	85	
6300	81	88	
8000	82	89	
10000	80	87	
OVERALL	104	110	

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE 2 MEASURED SOUND PRESSURE LEVEL (DB) OCTAVE BAND			IDENTIFICATION
NOISE SOURCE/SUBJECT: (OPERATION:)) OMEGA 3.2) TEST 71-019-001) RUN 01) 04 DEC 74) PAGE J1
A-1E AIRCRAFT ())
GROUND CREW ())
NEAR FIELD NOISE LEVELS ())
LOCATION/CONDITION			
FREQ (HZ)	1/A	2/B	
31.5	89	102	
63	102	108	
125	100	102	
250	88	94	
500	87	93	
1000	85	89	
2000	85	91	
4000	86	90	
8000	86	93	
OVERALL	104	110	

TABLE I MEASURES OF HUMAN NOISE EXPOSURE

3

		IDENTIFICATION	
		OMEGA	3.2
		TEST 71-019-001	
		RUN 01	
		04 DEC 74	
		PAGE H4	
NOISE SOURCE/SUBJECT:		OPERATION:	
A-1E AIRCRAFT GROUND CREW			
NEAR FIELD NOISE LEVELS			
		LOCATION/CONDITION	
1/A	2/B		
HAZARD/PROTECTION			
C-WEIGHTED OVERALL SOUND LEVEL (OASLC IN DBC) AT EAR			
A-WEIGHTED OVERALL SOUND LEVEL (OASLA IN DBA) AT EAR			
MAXIMUM PERMISSIBLE TIME (T IN MINUTES) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)			
NO PROTECTION			
OASLC	104	109	
OASLA	93	98	
T	101	42	
MINIMUM QPL EAR MUFFS			
OASLA*	81	85	
T	807	404	
AMERICAN OPTICAL 1700 EAR MUFFS			
OASLA*	77	82	
T	960	679	
V-51R EAR PLUGS			
OASLA*	69	74	
T	960	960	
AMERICAN OPTICAL 1700 EAR MUFFS PLUS V-51R EAR PLUGS			
OASLA*	59	64	
T	960	960	
H-133 GROUND COMMUNICATION UNIT			
OASLA*	71	76	
T	960	960	
COMMUNICATION PREFERRED SPEECH INTERFERENCE LEVEL (PSIL IN DB)			
PSIL	86	91	
ANNOYANCE			
PERCEIVED NOISE LEVEL, TONE CORRECTED (PNLT IN PN08)			
TONE CORRECTION (C IN DB)			
PNLT	110	114	
C	6	0	

* BASED ON CALCULATED SPL SPECTRUM UNDER PROTECTIVE DEVICE.

TABLE 4

TEST CONDITIONS
FOR FAR-FIELD NOISE MEASUREMENTS

A-1E Aircraft, Ground Runup, Hurlburt Field, Eglin AFB, 6 Aug 1971
Tail # 52436

Aircraft Engine Operation

Idle Power	650 RPM 22 Inches Manifold Pressure
Taxi Power	1200 RPM 20" MAP
Military Power	2800 RPM 52.5" MAP

Meteorology

Meteorology	
Temperature	28.9 C
Bar Pressure	0.763 M Hg
Rel Humidity	73 %
Wind — Speed	1.5 M/Sec (3 kt)
— Direction	55 Deg

TABLE 5 MEASURED SOUND PRESSURE LEVEL (DB) AT 1/3 OCTAVE BAND DISTANCE = 30 METERS

TABLE 5 MEASURED SOUND PRESSURE
1/3 OCTAVE BAND
DISTANCE = 30 METERS

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)											
5 1/3 OCTAVE BAND DISTANCE = 30 METERS											
NOISE SOURCE/SUBJECT: A-1E AIRCRAFT R-3350-26MD ENGINE FAR FIELD NOISE											
OPERATION: IDLE POWER 650 RPM				METEOROLOGY: TEMP = 29 C BAR PRESS = .763 Hg REL HUMID = 73 %							
FREQ (HZ)				ANGLE (DEGREES)							
25	69<	72<	71<	74<	72<	68<	70<	69<	73<	71<	73<
31.5	75<	74<	73<	77<	75<	74<	77	77	76<	76<	76<
40	80	83	81	82	81	82	82	81	83	82	83
50	81	83	82	82	83	82	84	81	83	83	83
63	80	81	82	84	85	84	86	85	85	84	83
80	80	82	81	82	83	82	81	81	82	82	82
100	82	83	82	81	82	83	84	83	84	85	84
125	81	81	80	80	80	80	79	78	79	80	81
160	79	79	78	78	77	77	77	78	78	79	80
200	74	72<	75	75	74	71<	73	72<	73	75	75
250	69<	67<	69<	68<	68<	67<	69<	66<	67<	68<	69<
315	66	65	66	65	66	64<	67	67	67	68	69
400	66	66	68	66	68	67	72	71	70	71	72
500	67	66	66	67	69	67	71	68	69	70	71
630	64	62	62	64	62	64	61	60	63	64	66
600	61	62	62	65	63	60	63	64	65	63	68
1000	62	61	61	66	62	63	64	62	63	65	65
1250	62	60	60	65	59	60	64	63	65	65	63
1600	61	60	60	63	62	61	66	63	66	65	63
2000	61	60	62	63	61	62	65	63	66	67	62
2500	60	60	61	64	61	62	66	67	66	67	64
3150	57	59	58	63	61	64	67	67	67	68	64
4000	56	57	57	62	59	59	64	66	68	69	61
5000	55	56	56	59	60	61	66	63	67	62	60
6300	56	57	57	58	59	61	64	67	68	69	56
6000	55	57	58	58	60	61	65	70	71	64	60
10000	53	55	54	55	58	59	63	68	69	62	63
OVERALL	89	90	90	90	91	91	92	90	91	92	91

◀ LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)
5 1/3 OCTAVE BAND
DISTANCE = 30 METERS

NOISE SOURCE/SUBJECT:		(OPERATION:		METEOROLOGY:																
		1200 RPM				TEMP = 29 C		BAR PRESS = .763 Hg		REL HUMID = 73%		05 MAY 75								
														PAGE 2						
FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180	
25	67<	67<	67<	67<	67<	66<	66<	67<	67<	66<	66<	66<	66<	66<	66<	66<	66<	66<	66<	
31.5	76<	79	63	75<	80	74<	76	73<	70<	74<	76<	73<	76<	73<	74<	74<	77	79	78	63
40	62	61	62	60	62	79	60	79	80	80	81	82	82	81	79	77	77	79	81	
50	74	76	79	81	82	83	63	64	65	66	67	67	66	66	63	79	77	77	80	
63	87	90	89	91	91	91	91	93	93	93	94	95	96	93	91	86	86	84	84	
80	69	90	90	90	90	90	90	91	91	90	89	90	90	91	92	91	89	86	86	
100	92	93	92	91	92	91	92	91	91	92	93	93	95	96	92	93	91	92	85	
125	89	90	91	90	91	90	89	89	89	88	87	88	88	90	90	89	86	83	88	
160	90	90	90	89	88	88	87	90	91	90	89	89	90	90	90	89	87	83	79	
200	80	83	63	82	81	60	60	62	61	62	61	62	63	64	64	65	82	86	73	
250	76	80	80	79	79	79	79	79	79	79	78	78	80	81	79	81	79	74	69<	
315	79	77	79	78	78	77	76	77	76	76	76	76	76	76	76	79	82	79	76	
400	80	78	61	80	80	82	81	80	81	82	81	80	76	80	81	82	81	77	73	
500	79	80	79	79	77	77	78	77	77	79	75	78	78	80	79	80	87	83	79	
630	73	76	73	72	73	72	71	70	71	72	74	74	77	77	78	76	75	73	66	
800	74	75	74	73	74	74	74	74	71	72	72	74	76	76	78	78	75	71	67	
1000	72	72	72	71	72	73	73	71	70	73	70	73	76	77	76	75	74	68	63	
1250	72	73	71	70	71	73	72	72	72	73	74	74	76	76	76	72	66	62	62	
1600	71	73	70	69	71	71	71	70	72	73	76	72	73	75	74	73	69	64	59	
2000	71	71	71	70	71	72	71	72	71	71	73	75	73	74	74	75	71	66	60	
2500	69	71	70	69	69	71	71	72	72	72	73	71	73	73	72	71	69	63	59	
3150	67	66	66	68	66	68	66	68	69	70	73	74	72	73	66	71	69	66	57	
4000	67	68	66	66	66	66	66	66	70	71	74	74	73	69	69	69	65	61	56	
5000	65	66	66	66	67	66	66	67	66	69	70	71	72	67	66	67	65	62	54	
6300	65	66	66	66	67	67	67	67	69	70	72	73	72	67	66	67	65	61	55	
8000	65	66	66	66	67	69	69	71	73	73	73	73	68	69	67	65	61	59	56	
10000	63	64	64	65	65	67	69	71	72	71	70	66	66	67	65	63	59	57	53	
OVERALL	97	98	98	98	98	98	98	98	99	99	99	99	99	99	100	101	99	99	92	

< LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE: MEASURED SOUND PRESSURE LEVEL (DB)
5 1/3 OCTAVE BAND
DISTANCE = 30 METERS

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26ND ENGINE
FAR FIELD NOISE

OPERATION:
MILITARY POWER
2800 RPM

METEOROLOGY:
TEMP = 29 C
BAR PRESS = .763 HG
REL HUMID = 73 %

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 03
05 MAY 75
PAGE 2

FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180
25	82	80	84	87	88	88	90	91	93	94	95	95	96	96	97	97	96	96	97
31.5	81	81	81	82	81	82	82	82	85	84	85	85	85	84	86	85	84	84	89
40	87	89	90	92	92	93	93	94	95	96	97	97	97	98	98	99	99	99	91
50	88	90	95	99	100	99	101	103	103	104	105	106	106	106	106	106	106	107	68
63	89	92	95	99	100	102	101	101	101	102	102	104	104	107	107	105	106	101	94
80	96	96	102	103	109	109	107	112	116	116	119	116	117	117	111	106	102	97	97
100	96	96	96	98	98	99	98	99	99	102	104	105	105	105	105	105	102	103	97
125	99	95	94	94	97	100	101	100	96	98	99	102	104	104	103	104	102	99	99
160	105	111	106	100	104	102	98	107	115	116	117	115	114	114	114	114	114	110	91
200	96	100	100	96	98	94	97	97	100	103	104	103	104	104	105	102	102	98	98
250	96	101	100	102	101	97	95	105	106	103	105	111	111	112	107	105	105	98	98
315	97	97	97	98	100	100	99	99	106	107	103	107	109	108	107	106	107	104	91
400	102	96	104	102	101	101	101	102	106	106	105	106	106	106	106	105	104	104	88
500	104	99	100	102	101	101	101	104	104	105	105	105	103	106	106	106	106	103	90
630	97	99	99	97	98	100	99	101	101	101	101	101	103	103	103	102	102	100	98
800	99	95	99	99	99	97	97	101	102	103	103	102	104	105	105	104	105	105	95
1000	97	96	96	97	97	97	99	100	101	103	101	103	103	103	103	101	101	97	97
1250	98	96	97	97	98	98	101	103	104	106	104	105	105	105	105	103	102	98	98
1600	99	97	98	99	100	99	101	103	107	109	107	109	107	107	107	103	101	96	96
2000	102	100	100	101	100	101	100	103	106	108	110	109	105	105	105	103	102	97	97
2500	100	98	101	101	100	102	105	106	109	107	105	105	105	103	102	96	97	97	97
3150	100	97	99	100	100	99	102	105	106	108	107	103	103	103	101	101	95	93	93
4000	100	98	100	101	101	103	106	107	109	107	109	107	103	103	103	102	102	98	98
5000	98	95	96	99	100	98	101	104	104	106	104	106	104	104	104	101	100	98	98
6300	97	94	97	98	98	99	101	102	104	105	103	100	100	100	100	99	96	99	99
8000	96	93	96	97	98	96	96	100	102	103	105	102	102	102	102	99	97	98	98
10000	93	90	93	95	96	96	98	100	100	103	100	100	100	100	100	95	95	92	95
OVERALL	113	114	113	113	115	114	115	115	118	121	123	123	122	121	120	118	115	106	106

LEVEL CORRECTED TO REMOVE BACKGROUND/ELECTRONIC NOISE.

TABLE I DIRECTIVITY INDEX (DB)

6

NOISE SOURCE/SUBJECT:		OPERATION:		METEOROLOGY:		ANGLE (DEGREES)		OCTAVE		IDENTIFICATION:												
A-1E AIRCRAFT R-3350-26HD ENGINE FAR FIELD NOISE		IDLE POWER 650 RPM		TEMP = 29 C BAR PRESS = .763 H HG REL HUMID = 73 %		0 10 20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180		31.5 63 125 250 500 630 800 1000 1250 1600 2000 2500 3150 4000 5000 6300 8000 10000		OMEGA 1.4 TEST 75-002-001 RUN 01 05 MAY 75 PAGE 4												
FREQ (HZ)	1/3 OCTAVE	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180	OCTAVE	OVERALL
25	31.5	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	31.5	-2
40	50	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	50	-2
63	63	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	63	-2
80	125	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	125	-2
100	200	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	200	-2
125	250	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	250	-2
160	315	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	315	-2
200	400	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	400	-2
250	500	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	500	-2
300	630	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	630	-2
400	800	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	800	-2
500	1000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	1000	-2
630	1250	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	1250	-2
800	1600	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	1600	-2
1000	2000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	2000	-2
1250	2500	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	2500	-2
1600	3150	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	3150	-2
2000	4000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	4000	-2
2500	5000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	5000	-2
3000	6300	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	6300	-2
4000	8000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	8000	-2
5000	10000	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	10000	-2

TABLE: DIRECTIVITY INDEX (DB)
6

NOISE SOURCE/SUBJECT:		OPERATION:		ANGLE (DEGREES)												IDENTIFICATION:			
A-1E AIRCRAFT R-3350-26WD ENGINE FAR FIELD NOISE		1200 RPM		METEOROLOGY:			TEST 75-002-001			TEST 75-002-001			TEST 75-002-001			OMEGA 1.4			
FREQ (HZ)	0	10	20	30	40	50	60	70	80	90	100	110	120	130	140	150	160	170	180
1/3 OCTAVE																			
25	-0	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
31.5	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19
40	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20
50	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21
63	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22
80	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23
100	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24
125	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25
160	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26
200	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27
250	-10	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28
315	-11	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29
400	-12	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30
500	-13	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31
630	-14	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32
800	-15	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33
1000	-16	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34
1250	-17	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35
1600	-18	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36
2000	-19	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37
2500	-20	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38
3150	-21	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39
4000	-22	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39	-40
5000	-23	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39	-40	-41
6300	-24	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39	-40	-41	-42
8000	-25	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39	-40	-41	-42	-43
10000	-26	-27	-28	-29	-30	-31	-32	-33	-34	-35	-36	-37	-38	-39	-40	-41	-42	-43	-44
OCTAVE	31.5	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	63	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	125	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	250	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	500	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	1000	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	2000	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	4000	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
	8000	-1	-2	-3	-4	-5	-6	-7	-8	-9	-10	-11	-12	-13	-14	-15	-16	-17	-18
OVERALL	-2	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1	-1

TABLE I: DIRECTIVITY INDEX (DB)

FIGURE: NORMALIZED FARFIELD NOISE LEVELS

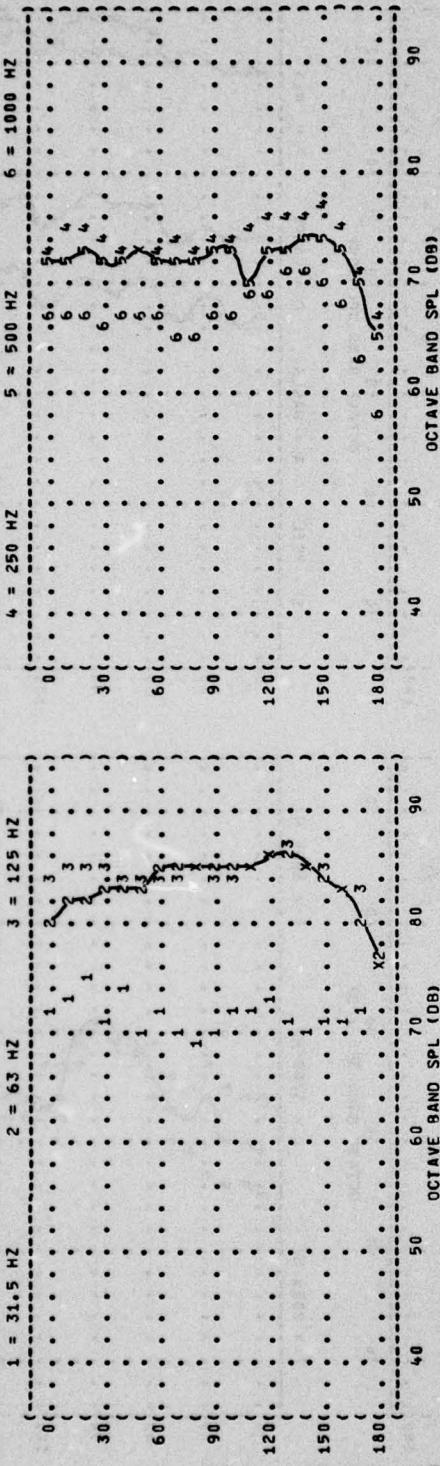
3 DISTANCE = 100 METERS

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT R-3350-26ND ENGINE FAR FIELD NOISE

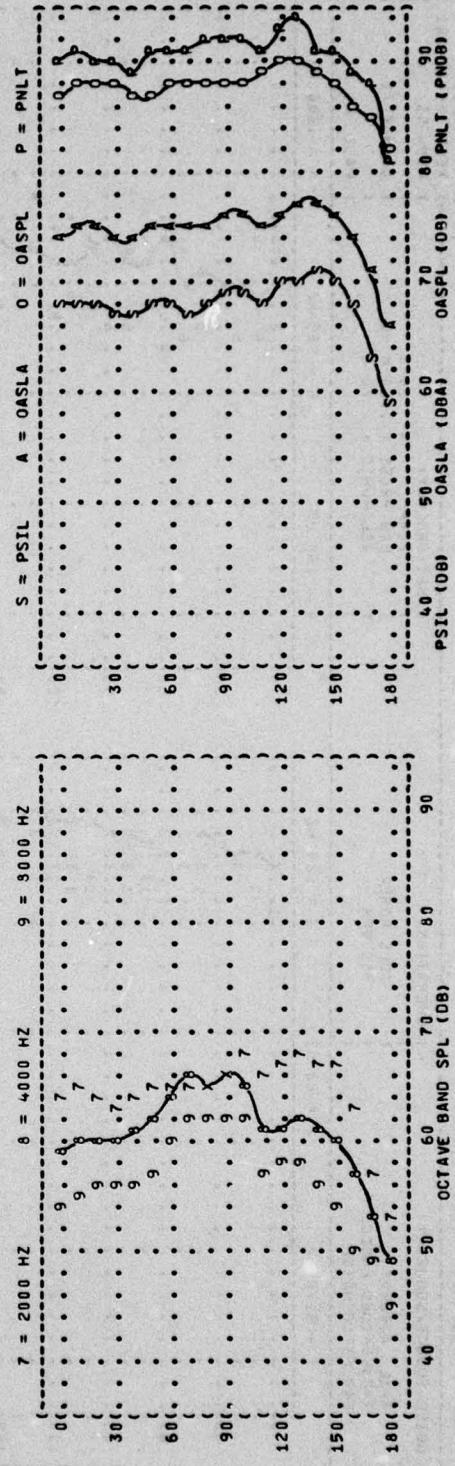
OPERATION: 1200 RPM

METEOROLOGY: TEMP = 15 C BAR PRESS = .760 MM HG REL HUMID = 70 %

TEST 75-002-001 RUN 02 05 MAY 75 PAGE 6



—



S = PSIL A = OASLA O = OASPL P = PNLT

FIGURE: NORMALIZED FARFIELD NOISE LEVELS

3 DISTANCE = 100 METERS

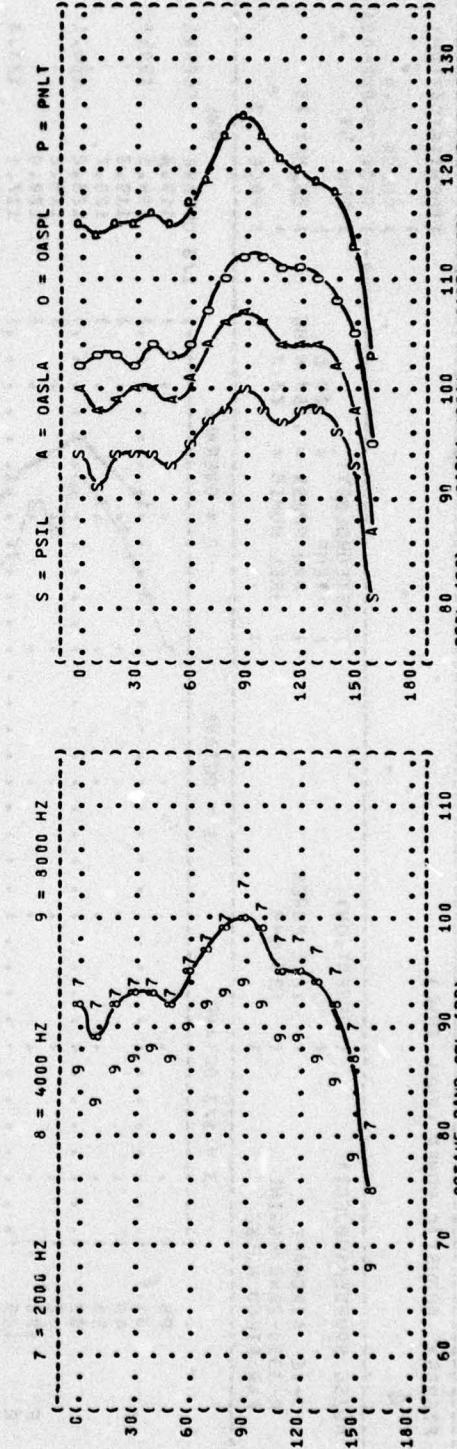
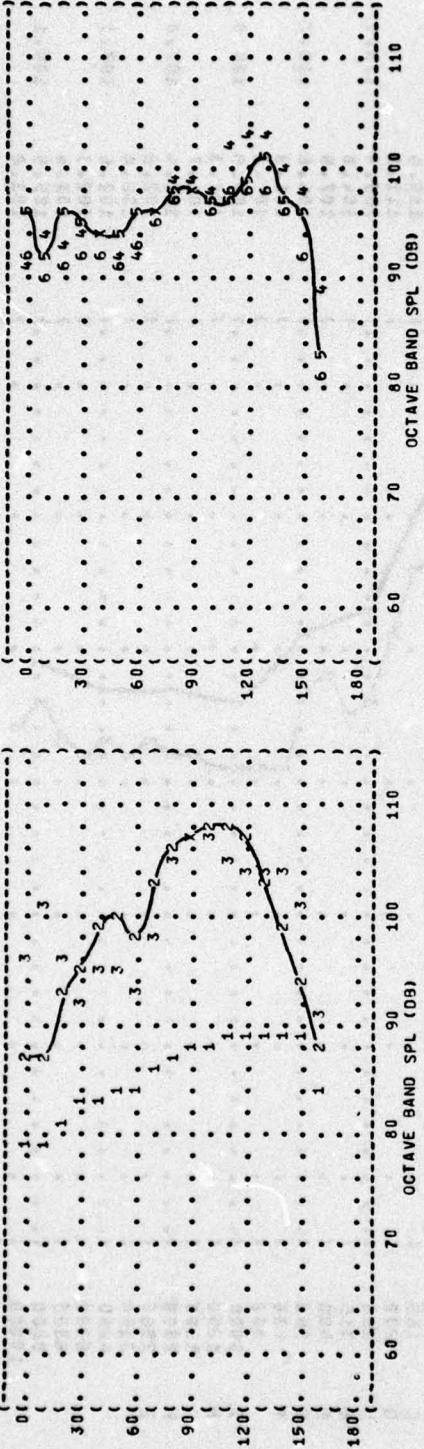
NOISE SOURCE/SUBJECT:

A-1E AIRCRAFT
R-3350-26HD ENGINE
FAR FIELD NOISE

OPERATION:

MILITARY POWER

2800 RPM



IDENTIFICATION: OMEGA 1-4 TEST 75-002-001 RUN 03

METEOROLOGY: TEMP = 15 C BAR PRESS = .760 MM HG REL HUMID = 70%

PAGE 6

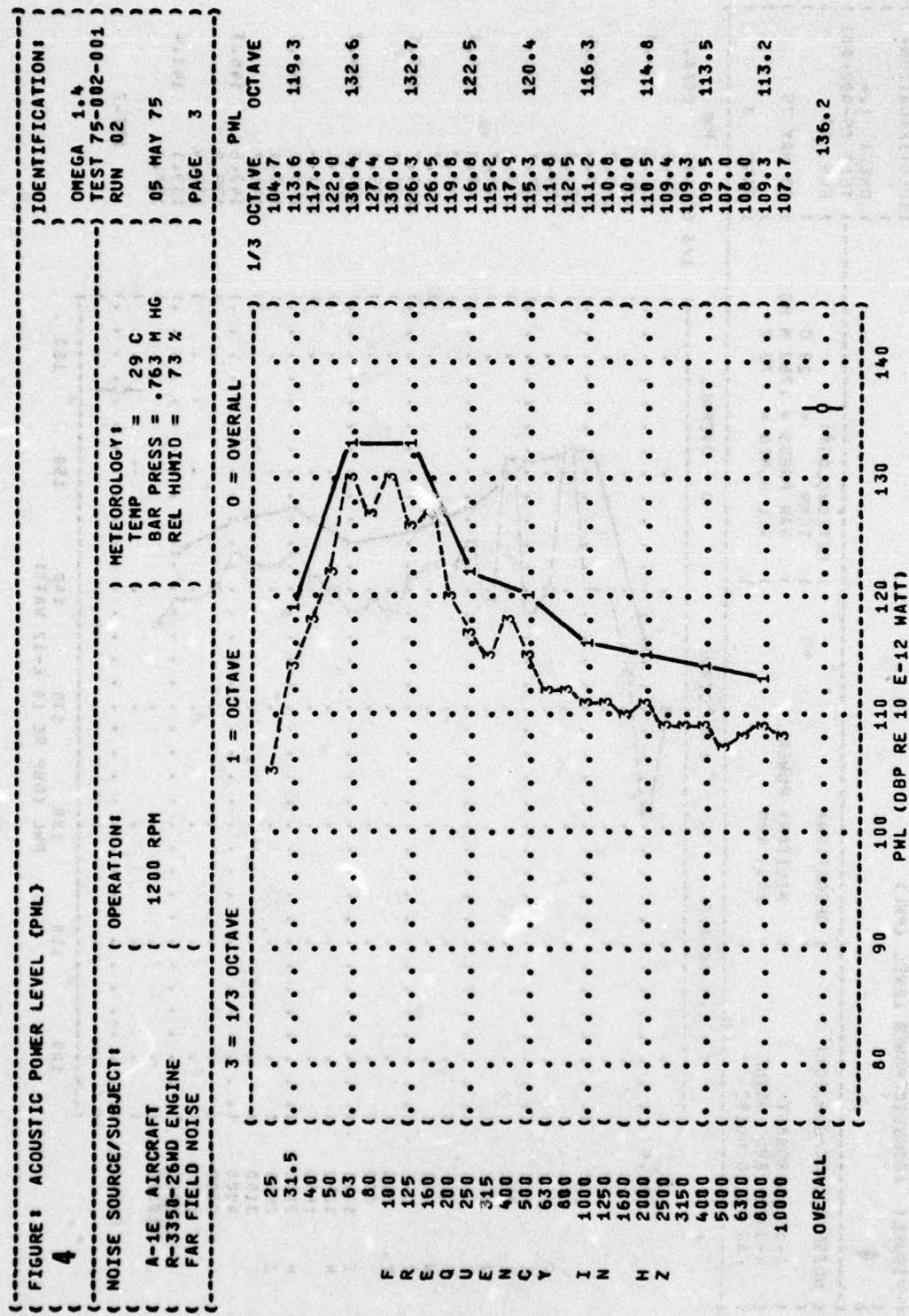
S = PSIL O = OASLA P = PNLT

D = 2000 Hz 8 = 4000 Hz 9 = 8000 Hz

S = PSIL A = OASLA O = OASPL P = PNLT

PSIL (dB) 90 100 110 120 130

OASLA (dB) OASPL (dB) PNLT (PNLdB)



{ FIGURE 4 ACOUSTIC POWER LEVEL (PWL)

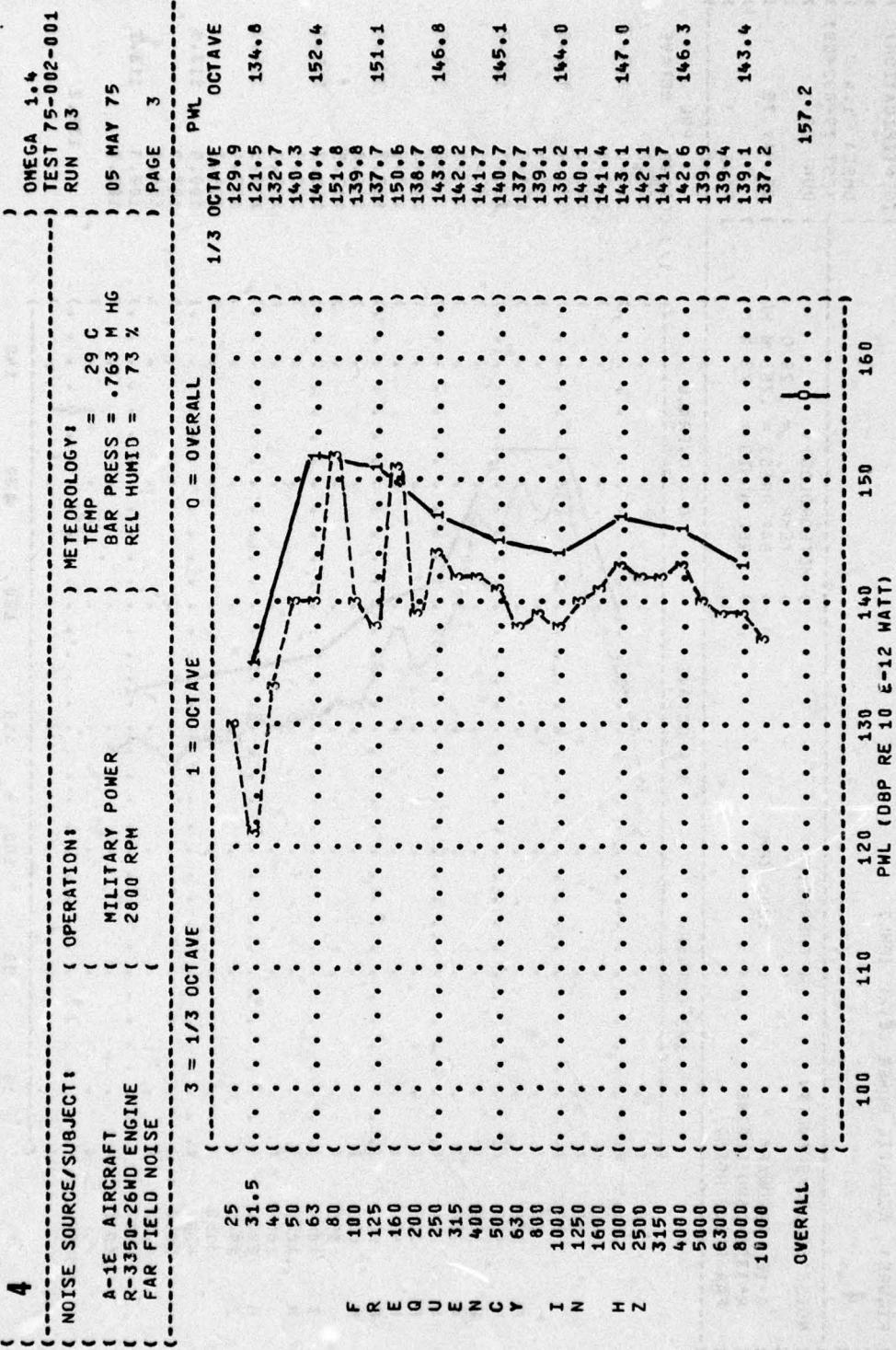


FIGURE: OVERALL SOUND PRESSURE LEVEL (OASPL)
5 EQUAL LEVEL CONTOURS (DB)

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

OPERATION: IDLE POWER
 650 RPM

METEOROLOGY: TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %

TEST 75-002-001
 RUN 01
 PAGE 13

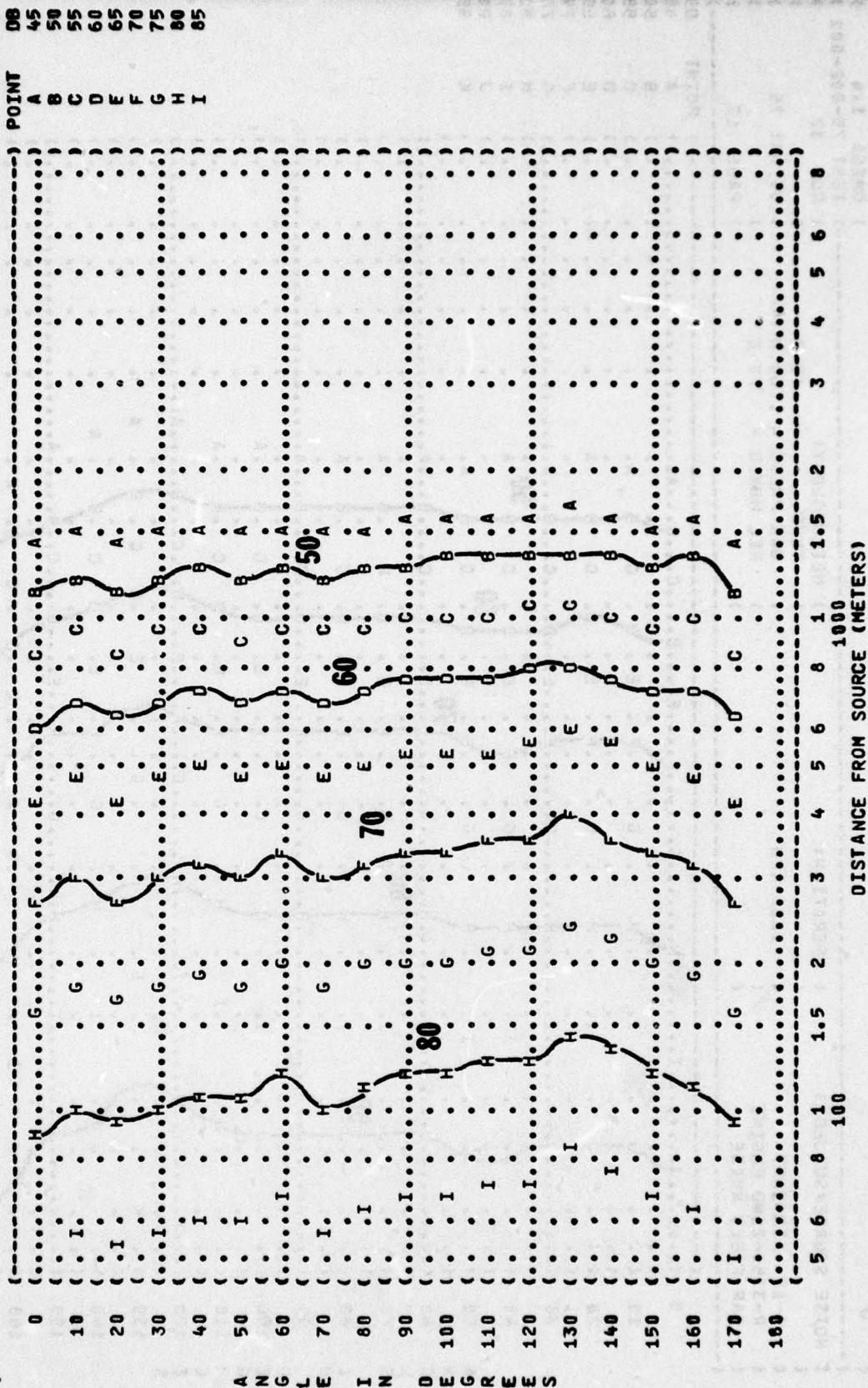


FIGURE 1 OVERALL SOUND PRESSURE LEVEL (OASPL)
EQUAL LEVEL CONTOURS (dB)

IDENTIFICATION!

OMEGA 14
TEST 75-002-001
RUN 02

A-1E AIRCRAFT R-3350-26WD ENGINE
EAR FTEARD MOTSE

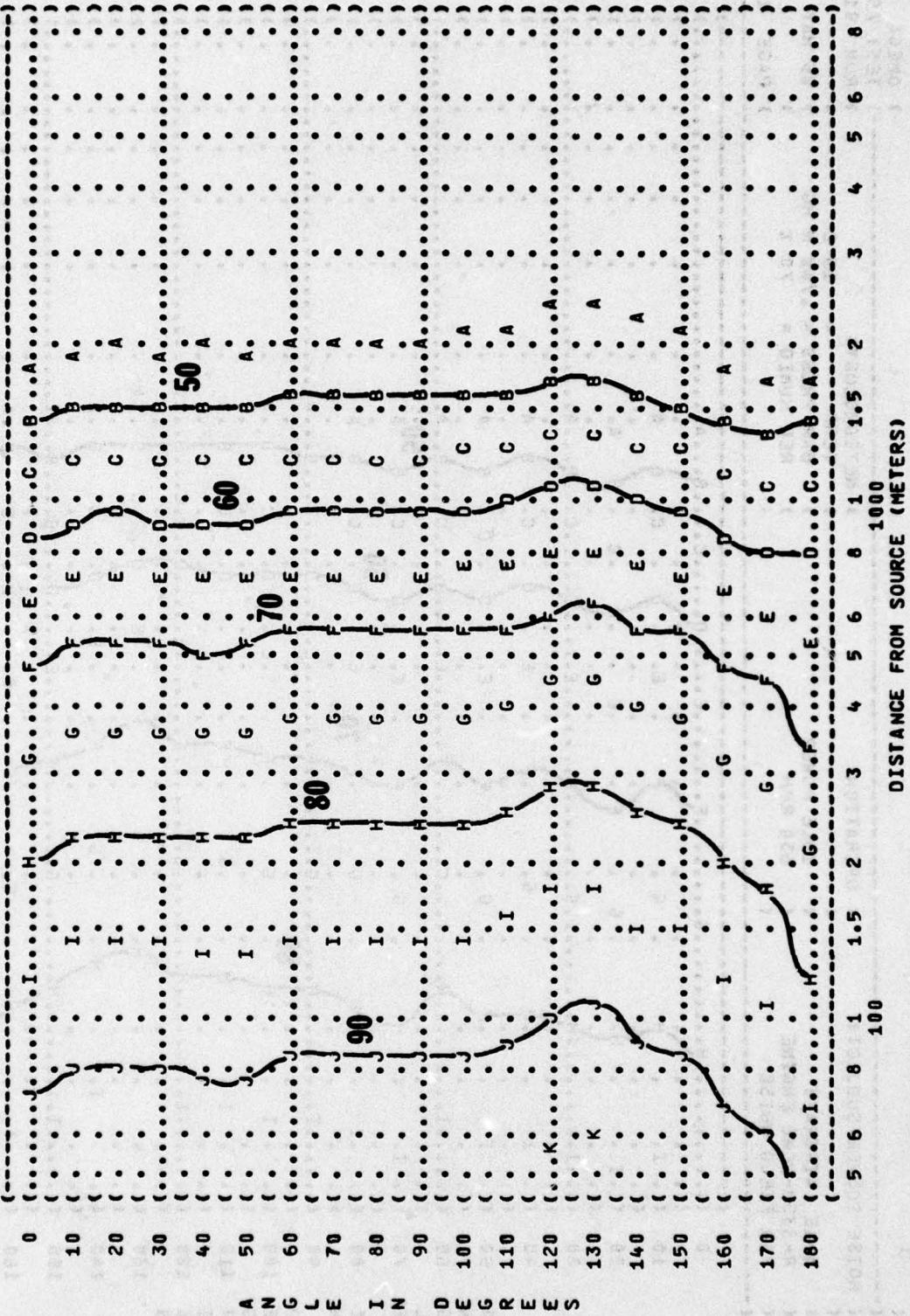


FIGURE 5
OVERALL SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)

NOISE SOURCE/SUBJECT:	OPERATION:	
	MILITARY POWER	2800 RPM
A-1E AIRCRAFT R-3350-26WD ENGINE FAR FIELD NOISE		

TEST 75-002-001
RUN 03
05 MAY 75
PAGE 13

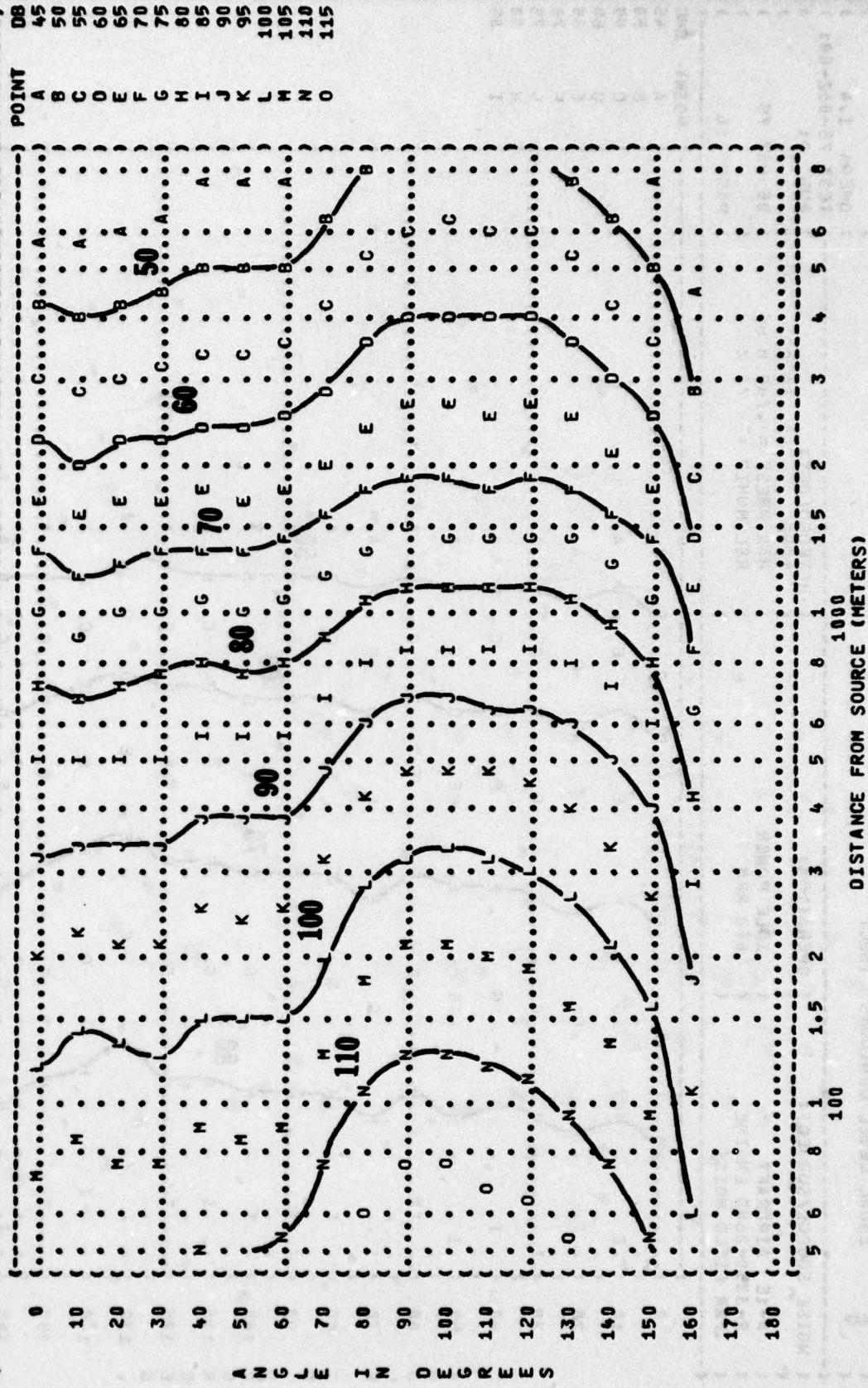


FIGURE: C-WEIGHTED OVERALL SOUND LEVEL (DB(C))
6 EQUAL LEVEL CONTOURS (DB(C))

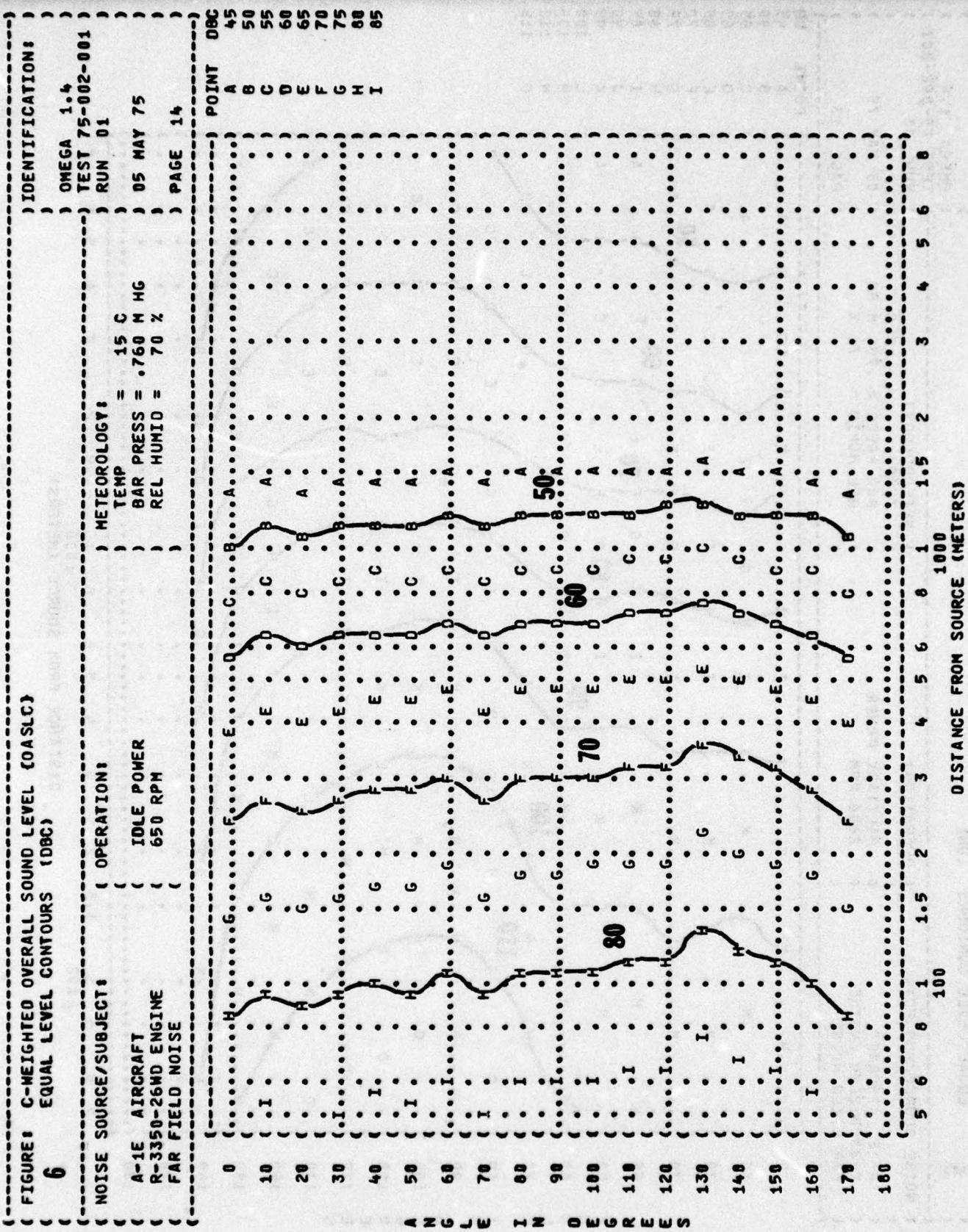


FIGURE 1 C-WEIGHTED OVERALL SOUND LEVEL (OASLC)
6 EQUAL LEVEL CONTOURS (DBC)

NOISE SOURCE/SUBJECT: (OPERATION:)
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

1200 RPM

TEST 75-002-001
RUN 02

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

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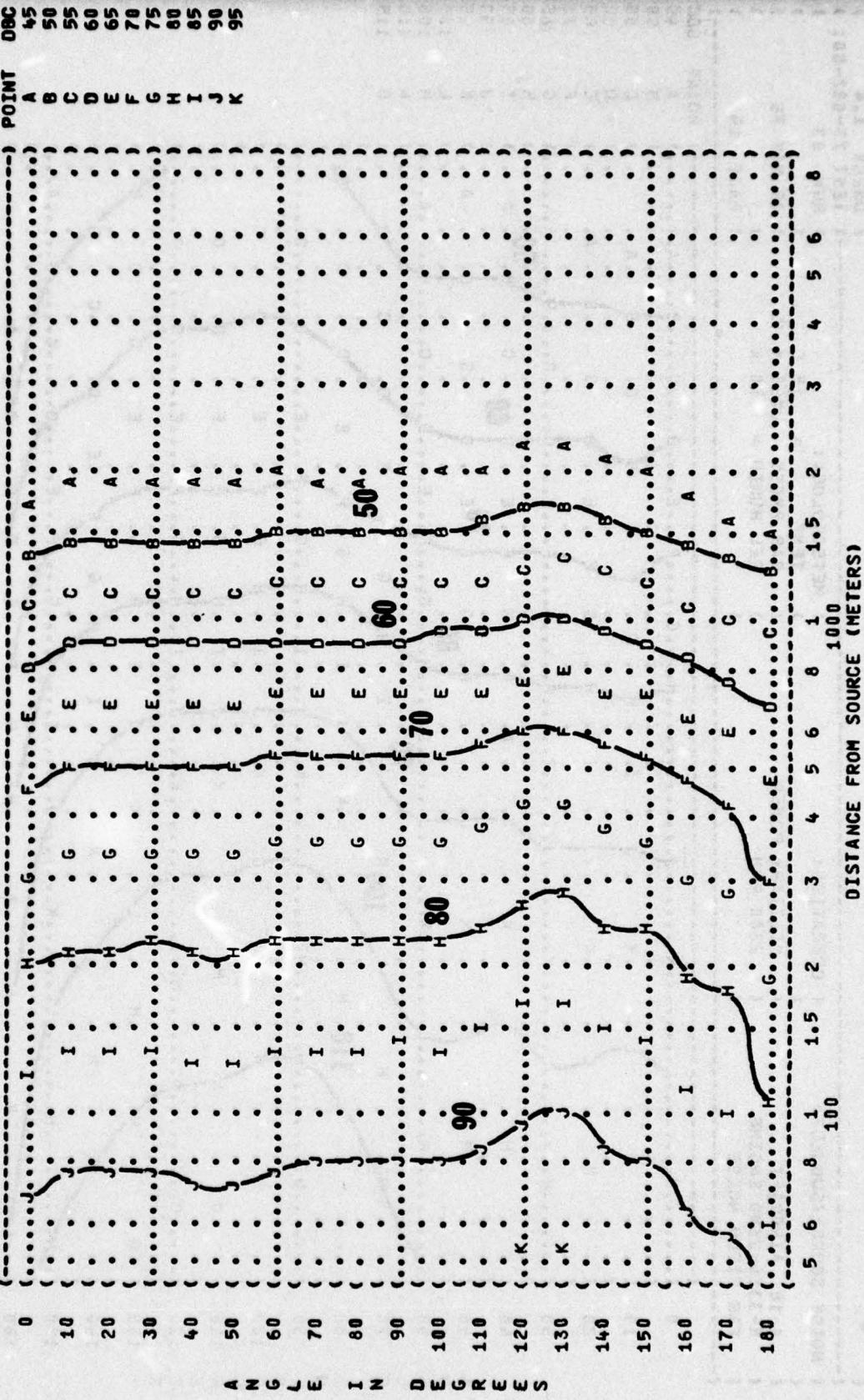


FIGURE 6 C-WEIGHTED OVERALL SOUND LEVEL (OASLC)
EQUAL LEVEL CONTOURS (DBC)

IDENTIFICATIONS

OMEGA 1.4
TEST 75-002-001
RUN 03

05 MAY 75

DISSE SOURCE/SUBJECT	OPERATION!	MILITARY POWER
A-1E AIRCRAFT	R-3350-26MD ENGINE	2800 RPM

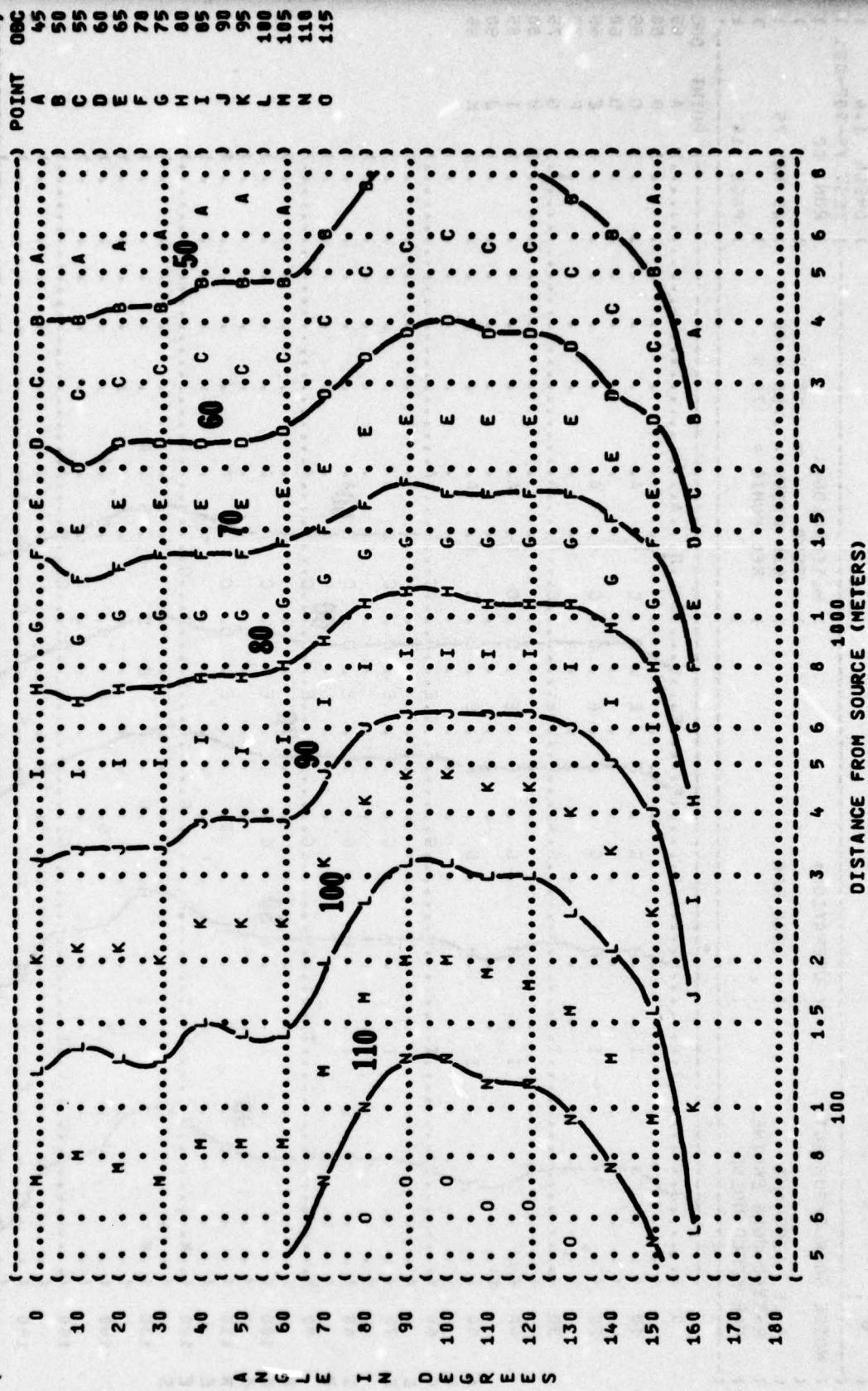


FIGURE 7 A-WEIGHTED OVERALL SOUND LEVEL (OASLA) EQUAL LEVEL CONTOURS (DBA)

7

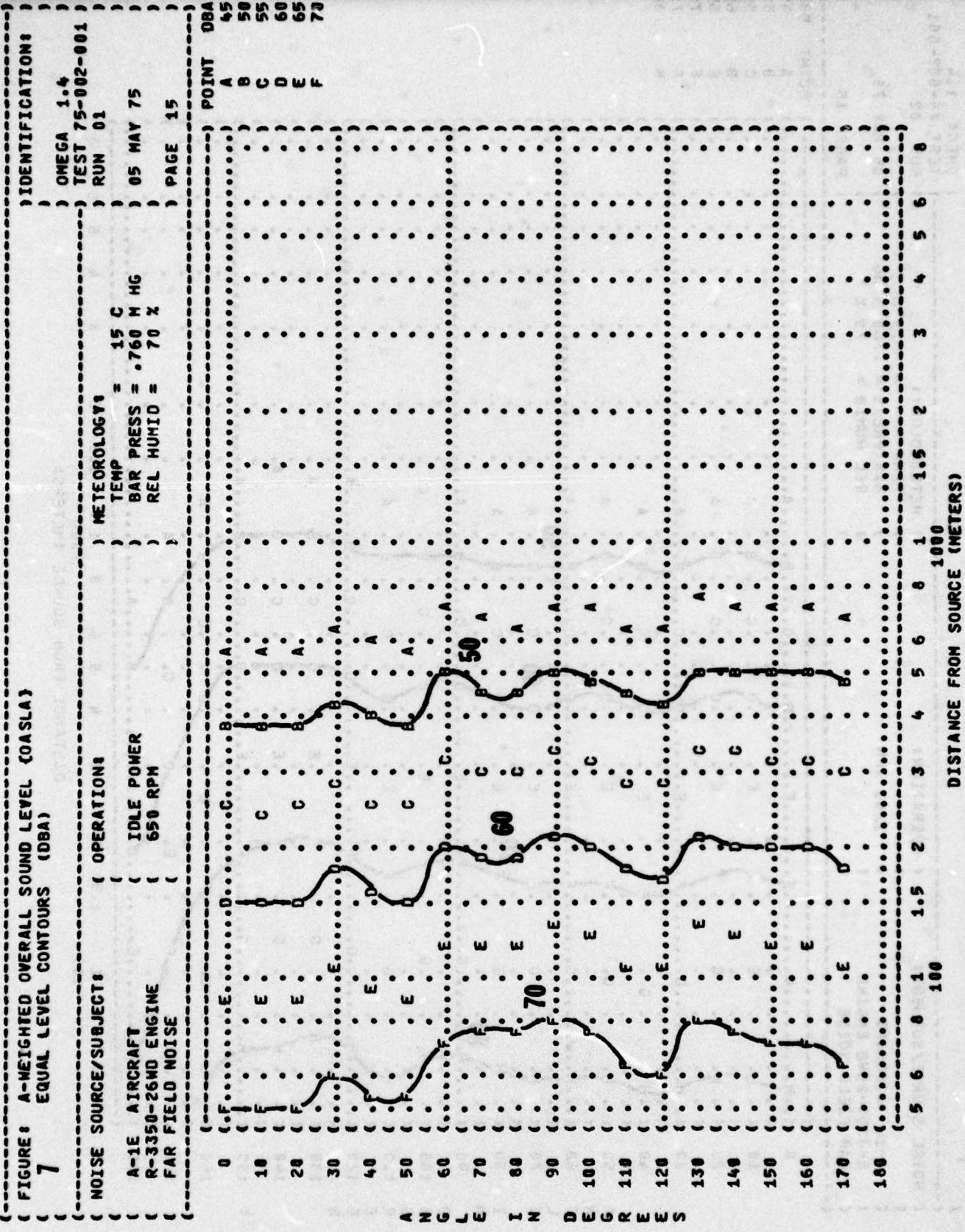


FIGURE 1 A-WEIGHTED OVERALL SOUND LEVEL (DBA)
7 EQUAL LEVEL CONTOURS (DBA)

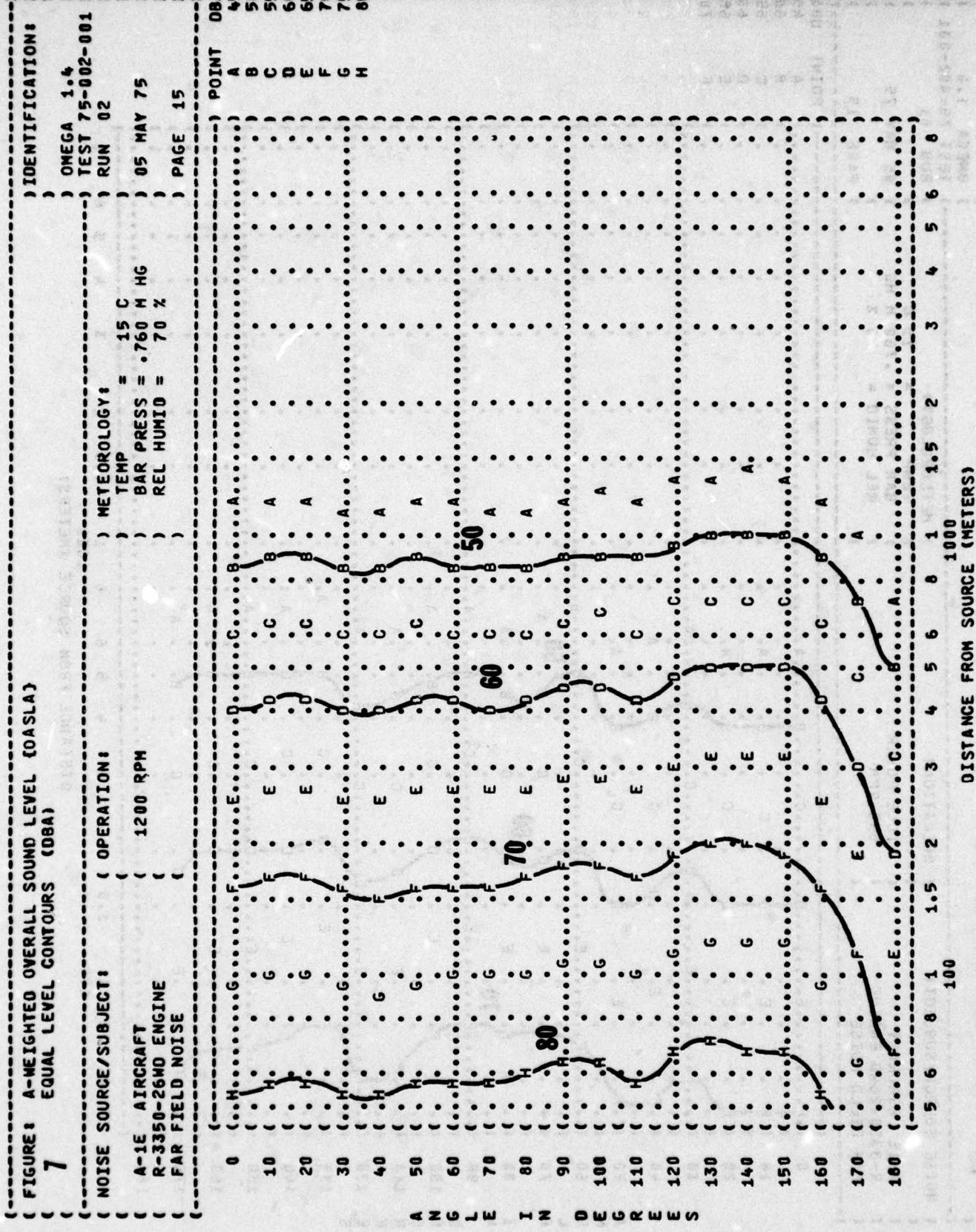


FIGURE 1 A-WEIGHTED OVERALL SOUND LEVEL (DBA)

7

NOISE SOURCE/SUBJECT:

A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:

MILITARY POWER
2800 RPM

IDENTIFICATION:

OMEGA 1.4
TEST 75-002-001
RUN 03

05 MAY 75

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METEOROLOGY:
TEMP = 15 C
BAR PRESS = 760 MM HG
REL HUMID = 70 %

POINT DBA

B
C
D
E
F
G
H
I
J
K
L
M
N

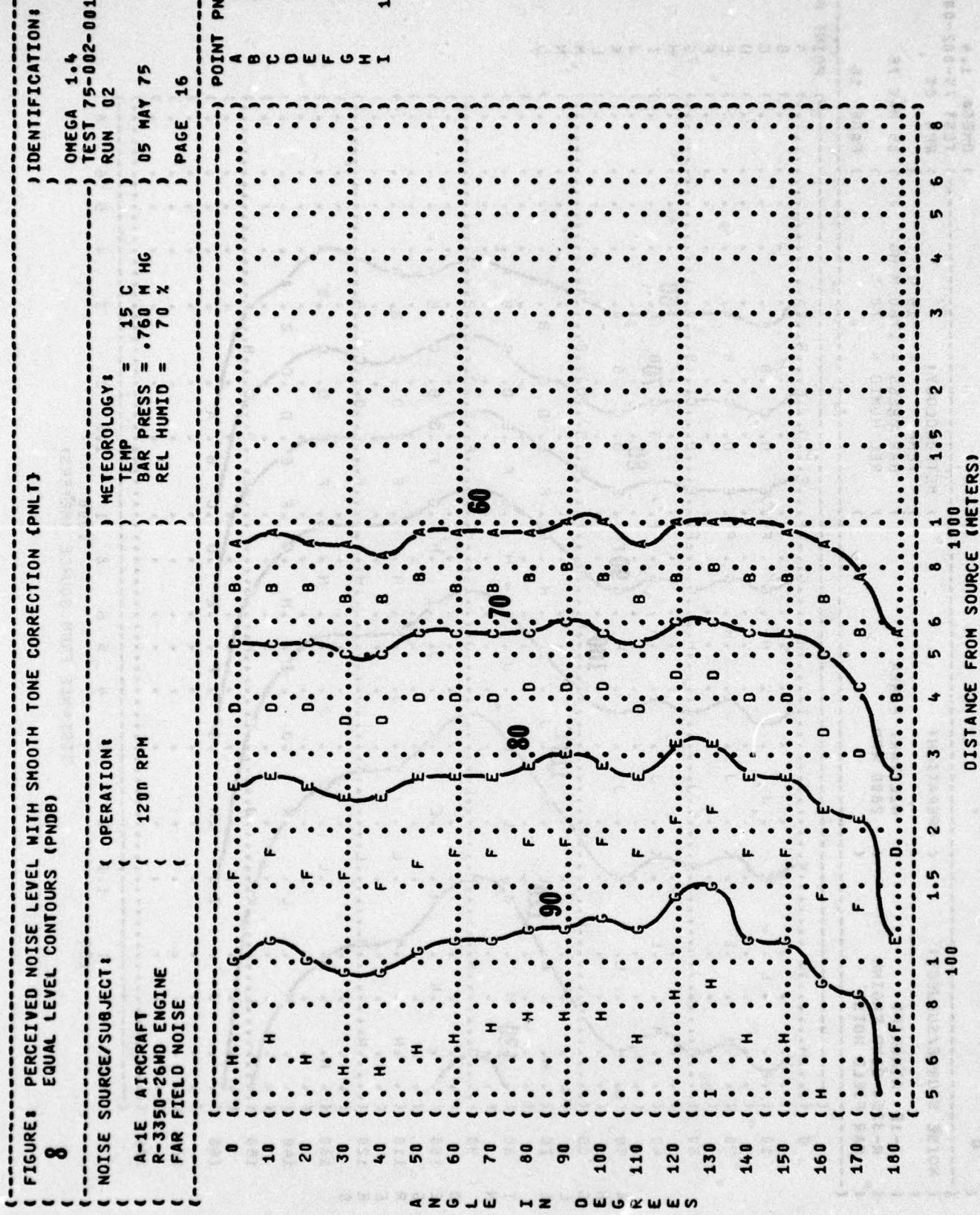
50
60
70
80
90
100
110

5 6 8 1 1.5 2 3 4 5 6 8 1 1.5 2 3 4 5 6 8
100 110 120 130 140 150 160 170 180

DISTANCE FROM SOURCE (METERS)

8 FIGURE 8

PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PNLT)
EQUAL LEVEL CONTOURS (PNDB)



8

FIGURE 1 PERCEIVED NOISE LEVEL WITH SMOOTH TONE CORRECTION (PNLT)
EQUAL LEVEL CONTOURS (PNDB)

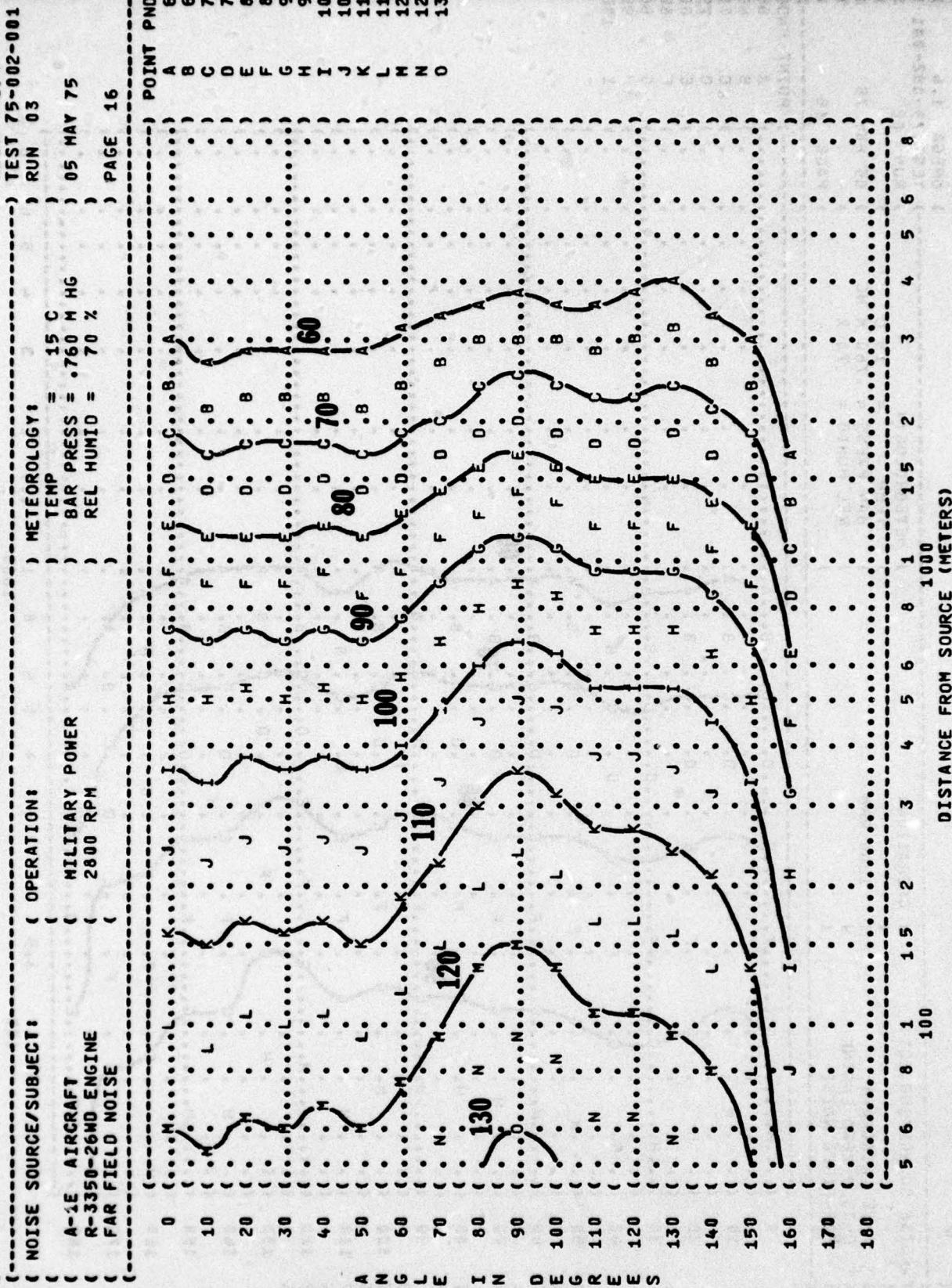


FIGURE 9 PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)
EQUAL LEVEL CONTOURS (DB)

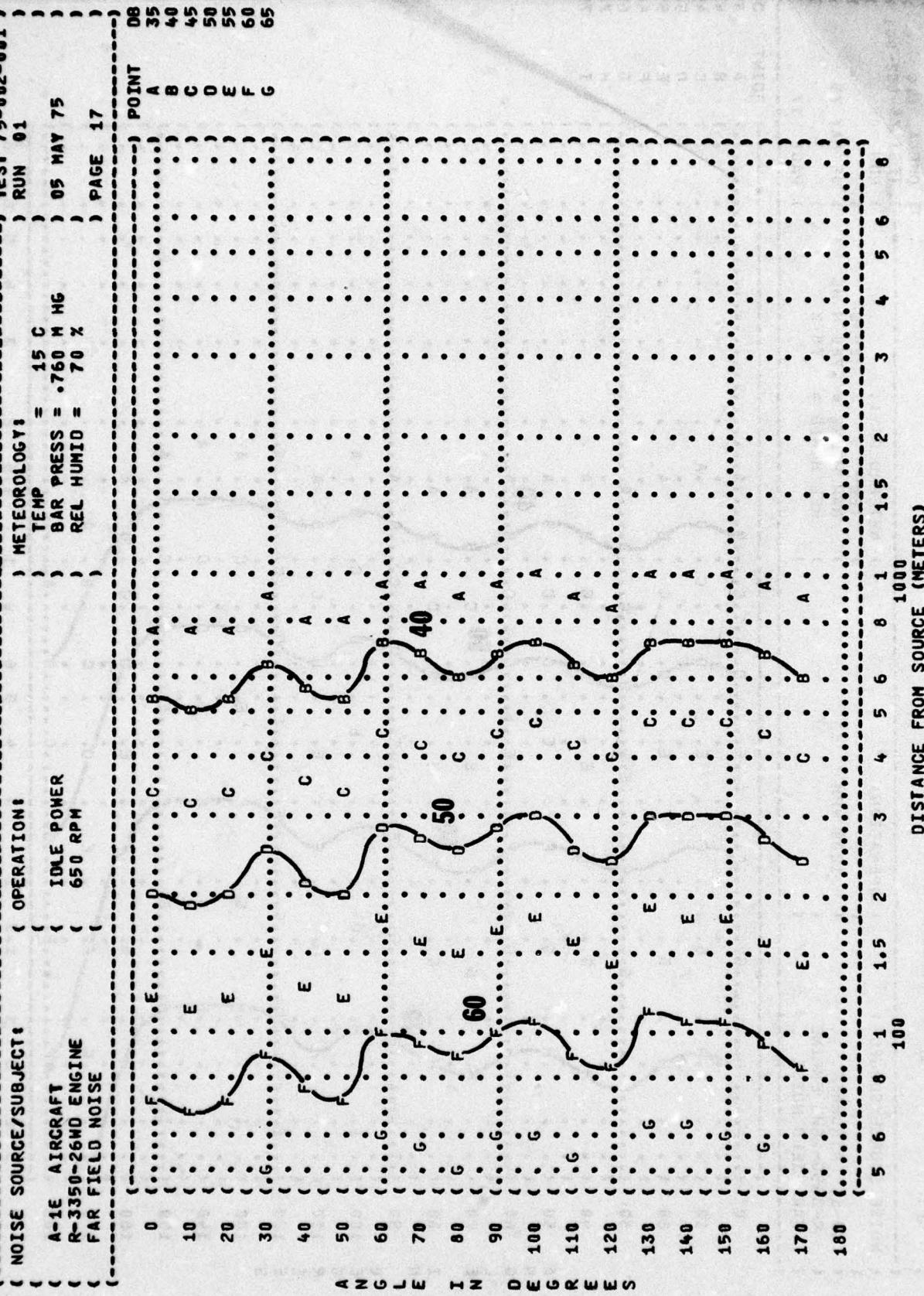


FIGURE 9 PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)

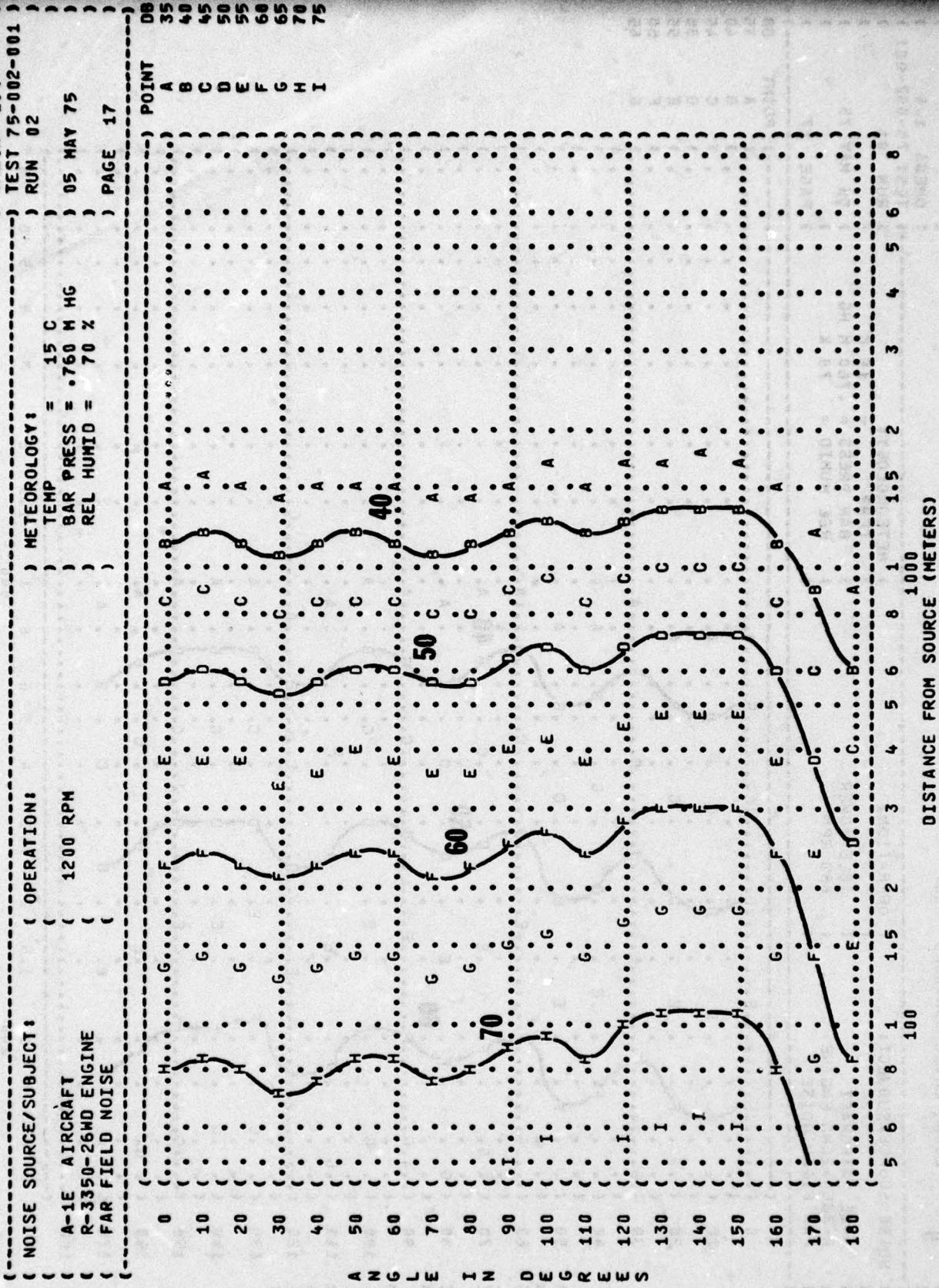


FIGURE: PREFERRED SPEECH INTERFERENCE LEVEL (PSIL)
9 EQUAL LEVEL CONTOURS (DB)

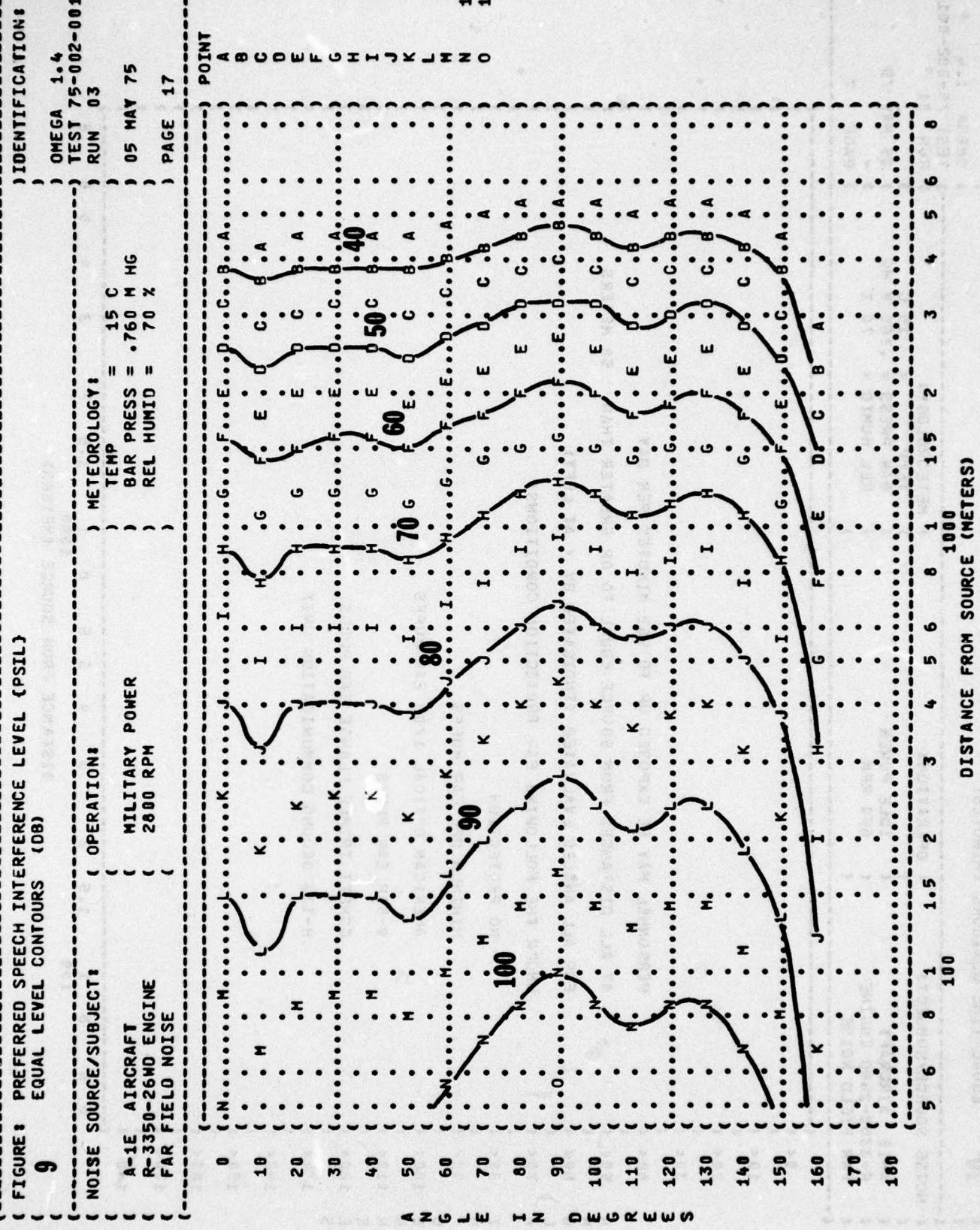
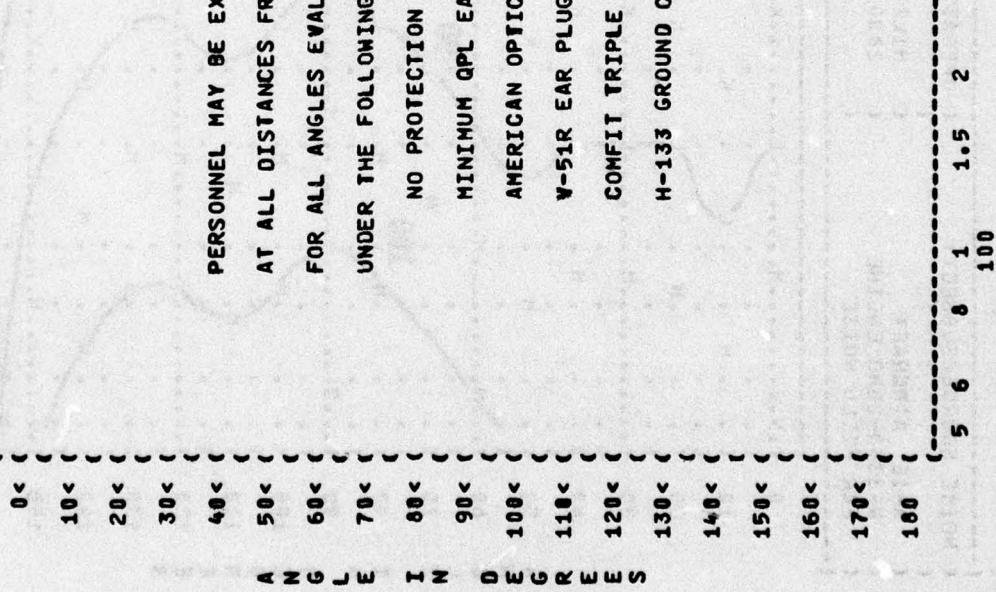


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
10 EQUAL TIME CONTOURS (MINUTES)

NOISE SOURCE/SUBJECT:	OPERATION:	METEOROLOGY:	IDENTIFICATION:
A-1E AIRCRAFT R-3350-26WD ENGINE FAR FIELD NOISE	IDLE POWER 650 RPM	TEMP = 15 C BAR PRESS = .760 Hg REL HUMID = 70 %	OMEGA 1.4 TEST 75-002-001 RUN 01
0<			
10<			



PERSONNEL MAY BE EXPOSED UP TO 960 MINUTES PER DAY
AT ALL DISTANCES FROM SOURCE EQUAL TO OR GREATER THAN 50 METERS
FOR ALL ANGLES EVALUATED (INDICATED BY < AT LEFT)
UNDER THE FOLLOWING EAR PROTECTION CONDITIONS:

NO PROTECTION

MINIMUM QPL EAR MUFFS

AMERICAN OPTICAL 1700 EAR MUFFS

V-51R EAR PLUGS

COMFIT TRIPLE FLANGE EAR PLUGS

H-133 GROUND COMMUNICATION UNIT

FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
10
 EQUAL TIME CONTOURS (MINUTES)
 NO PROTECTION

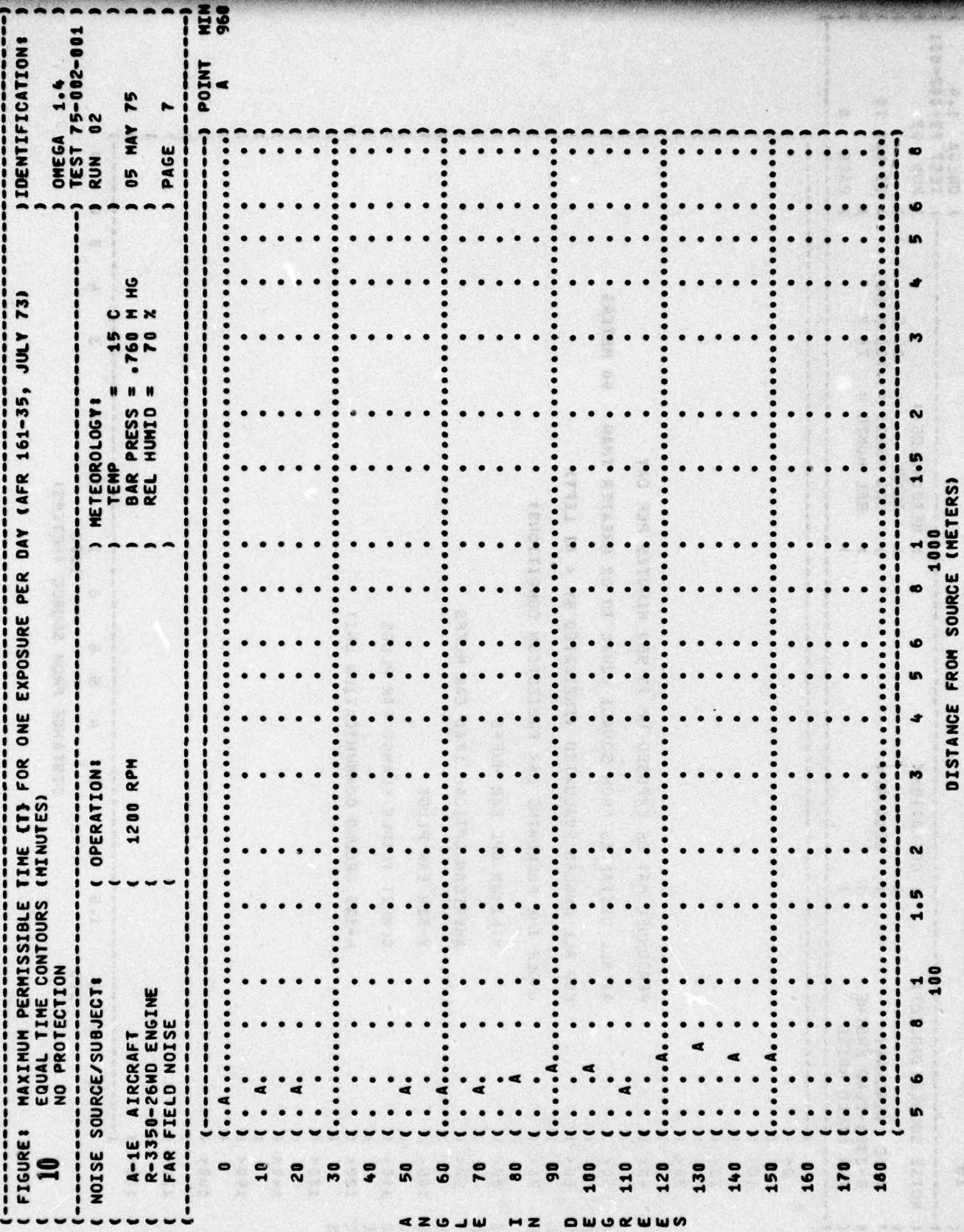


FIGURE 8 MAXIMUM PERMISSIBLE TIME (t_1) FOR ONE EXPOSURE
10 EQUAL TIME CONTOURS (MINUTES)

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

0< 10< 20<

PERSONNEL M AT ALL DIST FOR ALL ANG

MINIMUM
AMERICAN
N-512

COMFIT
H-133

REVERSE PER DAY (AFR 161-35, JULY 73) IDENTIFICATION:

TEST 72-002-001
RUN 02
15 C M HG
05 MAY 75
PAGE 6

60 MINUTES PER DAY
AL TO OR GREATER THAN 50 METERS
TESTED BY < AT LEFT)

THE JOURNAL OF CLIMATE

— — — — —
UGS UNIT

IDENTIFICATION

TEST 73-002-001
RUN 02
05 MAY 75
PAGE 6

—
—
—
—
—
—

10

104

10

161-35, JULY 73) IDENTIFICATION

TEST 72-002-001
RUN 02
05 MAY 75
PAGE 6

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DAY THAN 50 METERS

THE JOURNAL OF CLIMATE

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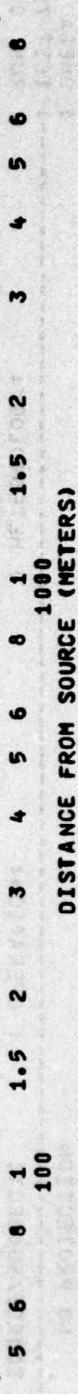


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
10 EQUAL TIME CONTOURS (MINUTES)
NO PROTECTION

NOISE SOURCE/SUBJECT:
**A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE**

OPERATIONS:
**MILITARY POWER
 2800 RPM**

METEOROLOGY:
**TEMP = 15 C
 BAR PRESS = 760 MM HG
 REL HUMID = 70 %**

TEST 75-002-001
 RUN 03
 PAGE 7

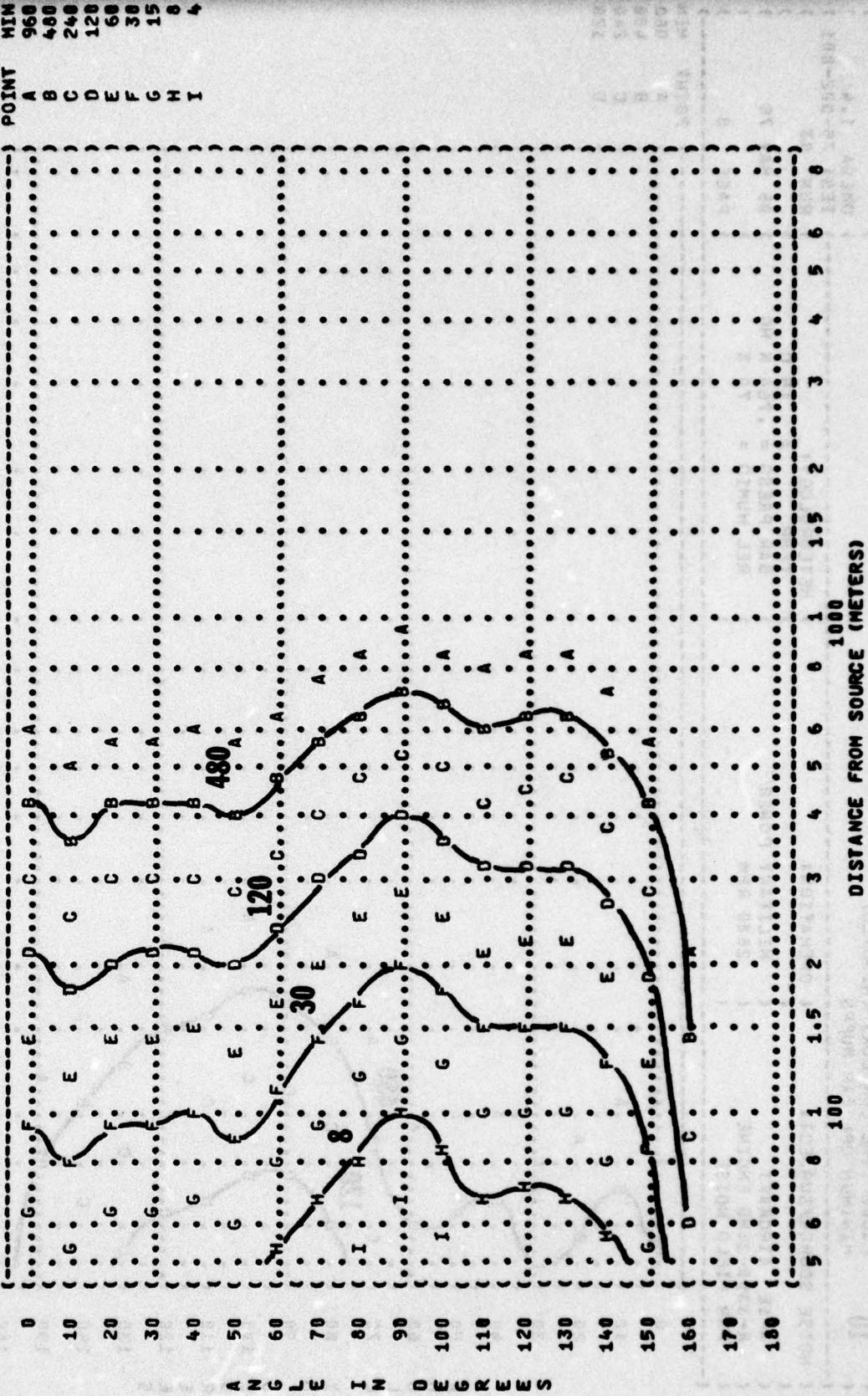


FIGURE: MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
 EQUAL TIME CONTOURS (MINUTES)
10 MINIMUM QPL EAR MUFFS

NOISE SOURCE/SUBJECT: **A-1E AIRCRAFT**
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
MILITARY POWER
2800 RPM

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 Hg
REL HUMID = 70 %

TEST 75-002-001
 RUN 03
 PAGE 8

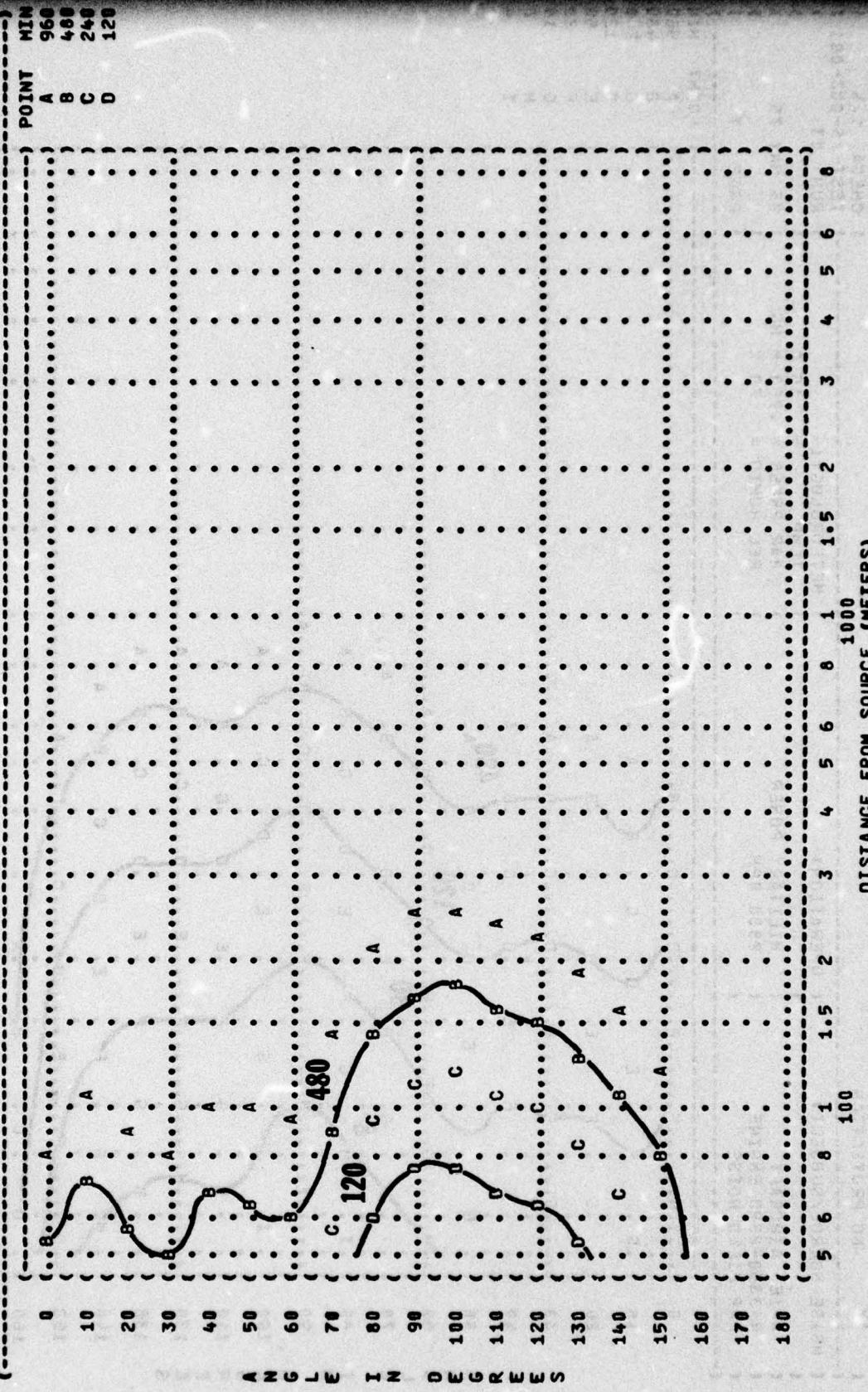


FIGURE 1 MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
 EQUAL TIME CONTOURS (MINUTES)
10
 V-51R EAR PLUGS

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

OPERATION: MILITARY POWER
 2800 RPM

METEOROLOGY: TEMP = 15 C
 BAR PRESS = .760 Hg
 REL HUMID = .70 %

TEST 75-002-001
 RUN 03
 PAGE 10

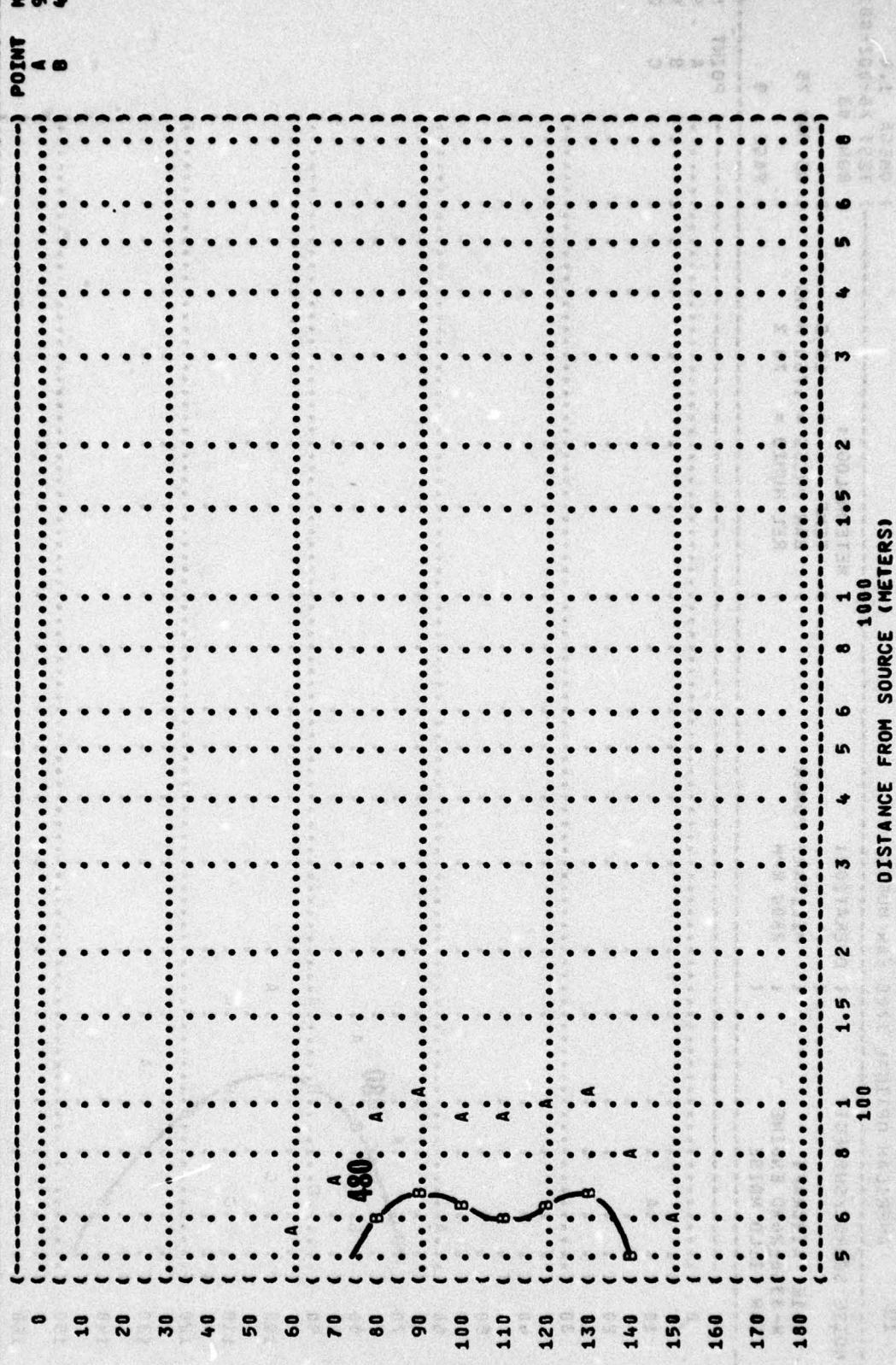


FIGURE 8 MAXIMUM PERMISSIBLE TIME (T) FOR ONE EXPOSURE PER DAY (AFR 161-35, JULY 73)
 10 EQUAL TIME CONTOURS (MINUTES)
 H-133 GROUND COMMUNICATION UNIT

NOISE SOURCE/SUBJECT	OPERATION	MILITARY POWER 2800 RPM	METEOROLOGY: TEMP = 15 C BAR PRESS = .760 MM HG REL HUMID = 70 %	TEST 75-002-001 RUN 03 PAGE 12
A-1E AIRCRAFT R-3350-26WD ENGINE FAR FIELD NOISE				

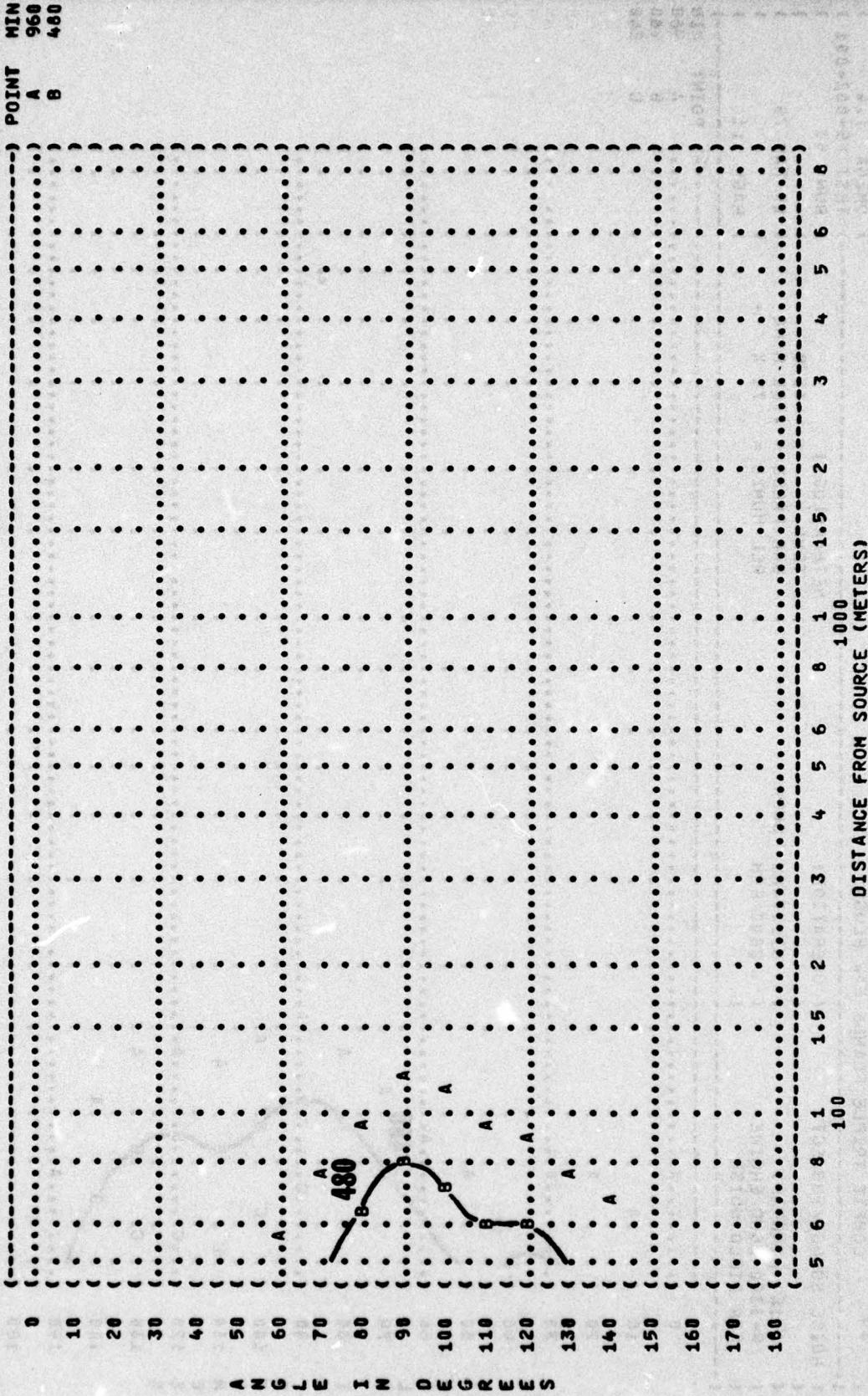


FIGURE 8 SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
31.5 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:

IDLE POWER
650 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 01

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
PAGE 18

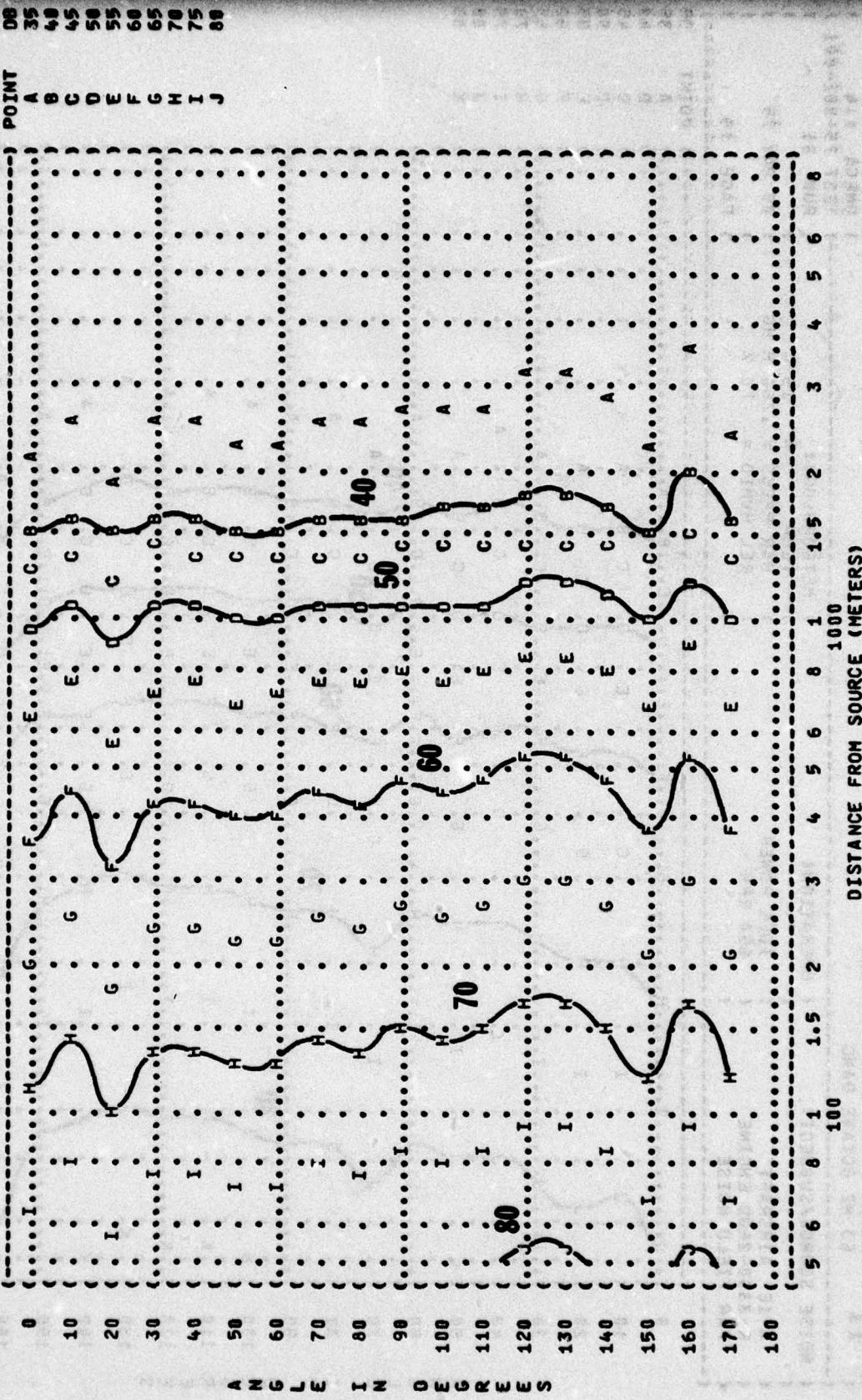


FIGURE 11
 SOUND PRESSURE LEVEL (SPL)
 EQUAL LEVEL CONTOURS (DB)
 125 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT

R-3350-26WD ENGINE
FAR FIELD NOISE

IDLE POWER
650 RPM

IDENTIFICATION:

OMEGA 1.4
TEST 75-002-001
RUN 01

PAGE 20

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
05 MAY 75

POINT 08
A 35
B 40
C 45
D 50
E 55
F 60
G 65
H 70
I 75
J

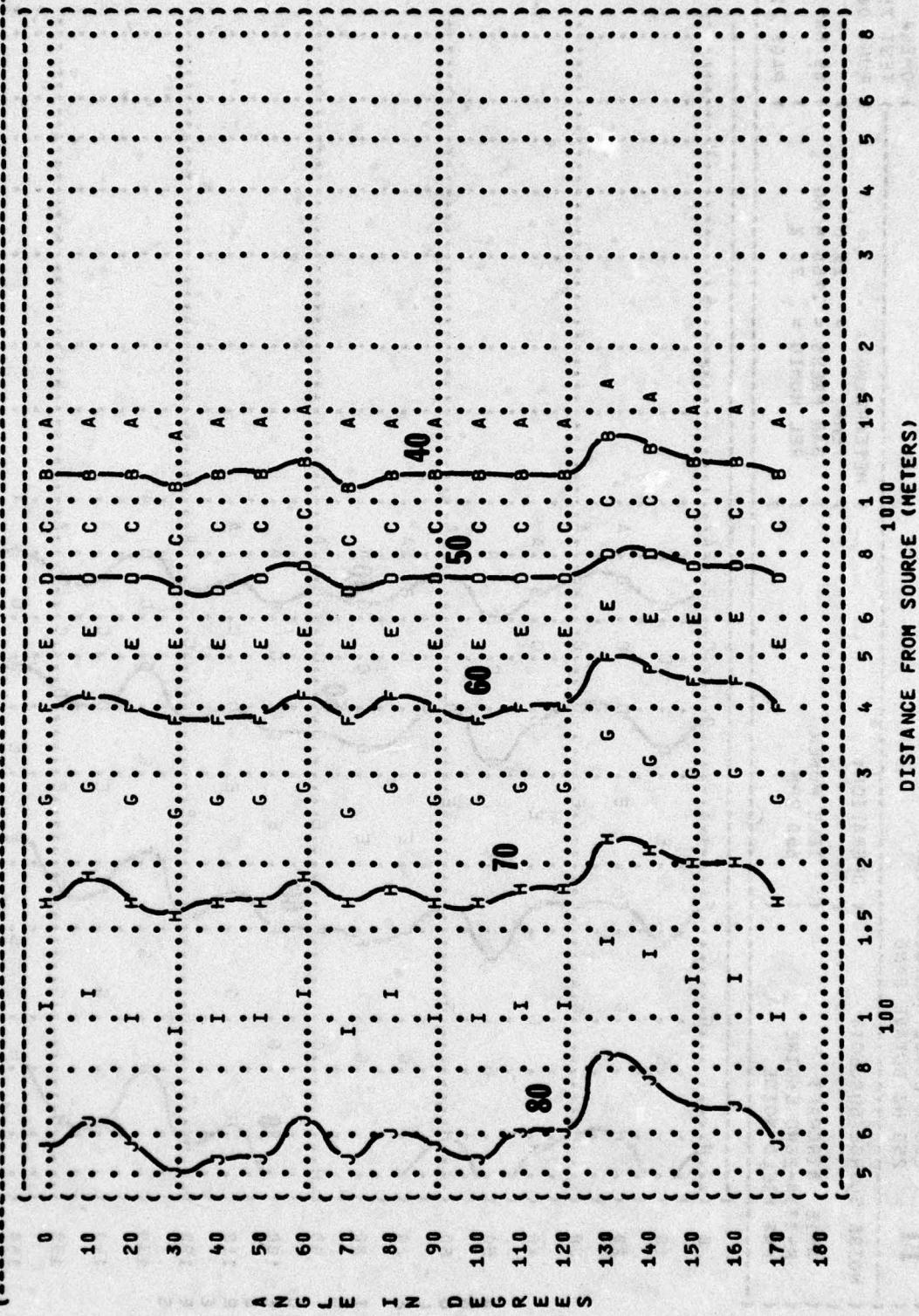


FIGURE: SOUND PRESSURE LEVEL (SPL)
11
 EQUAL LEVEL CONTOURS (0dB)
 250 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT
 R-3350-26MD ENGINE
 FAR FIELD NOISE

OPERATIONS: IDLE POWER
 650 RPM

IDENTIFICATION: OMEGA 1-4
 TEST 75-002-001
 RUN 01

METEOROLOGY: TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 PAGE 21

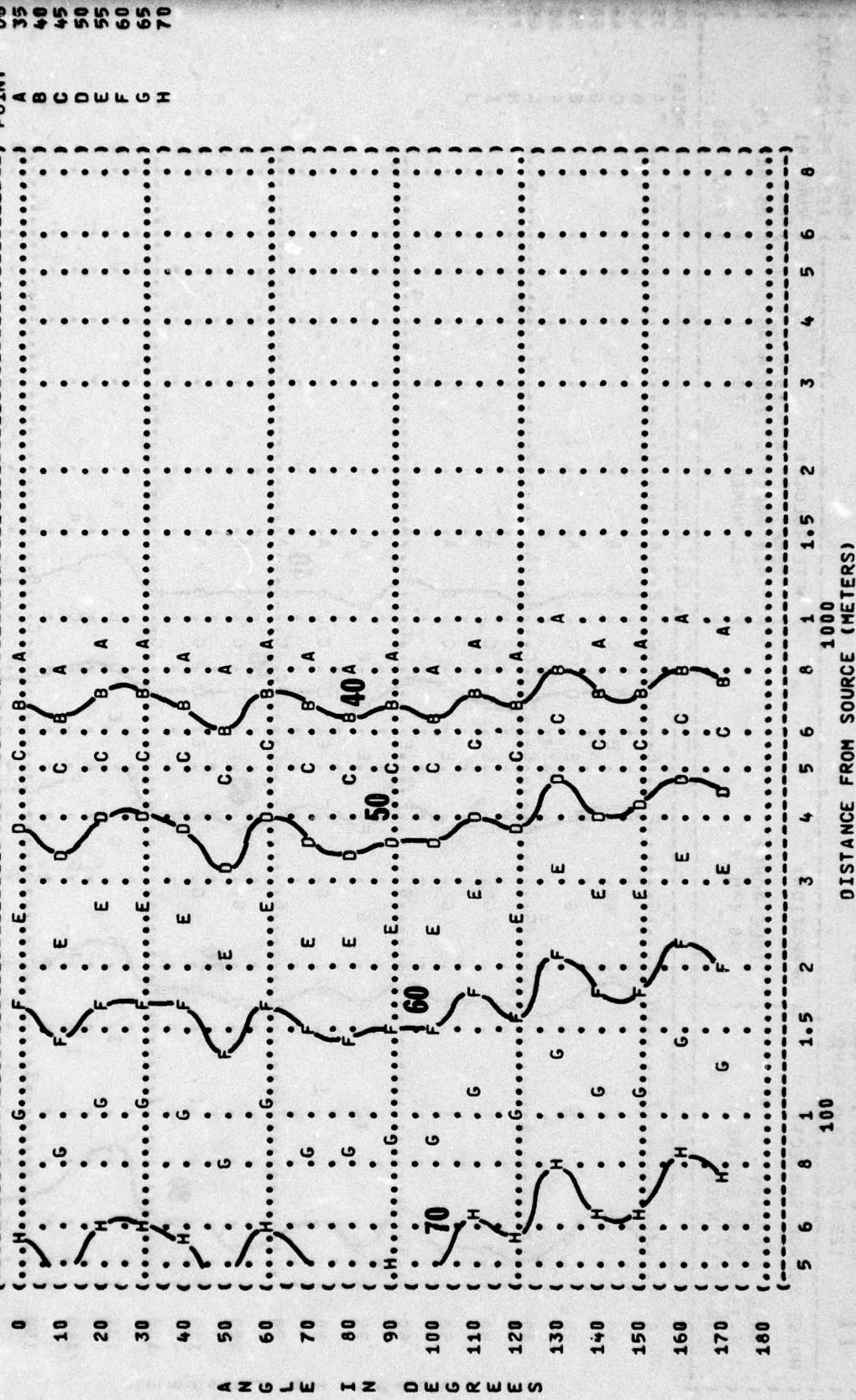


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
500 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
IDLE POWER
650 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 01

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
PAGE 22

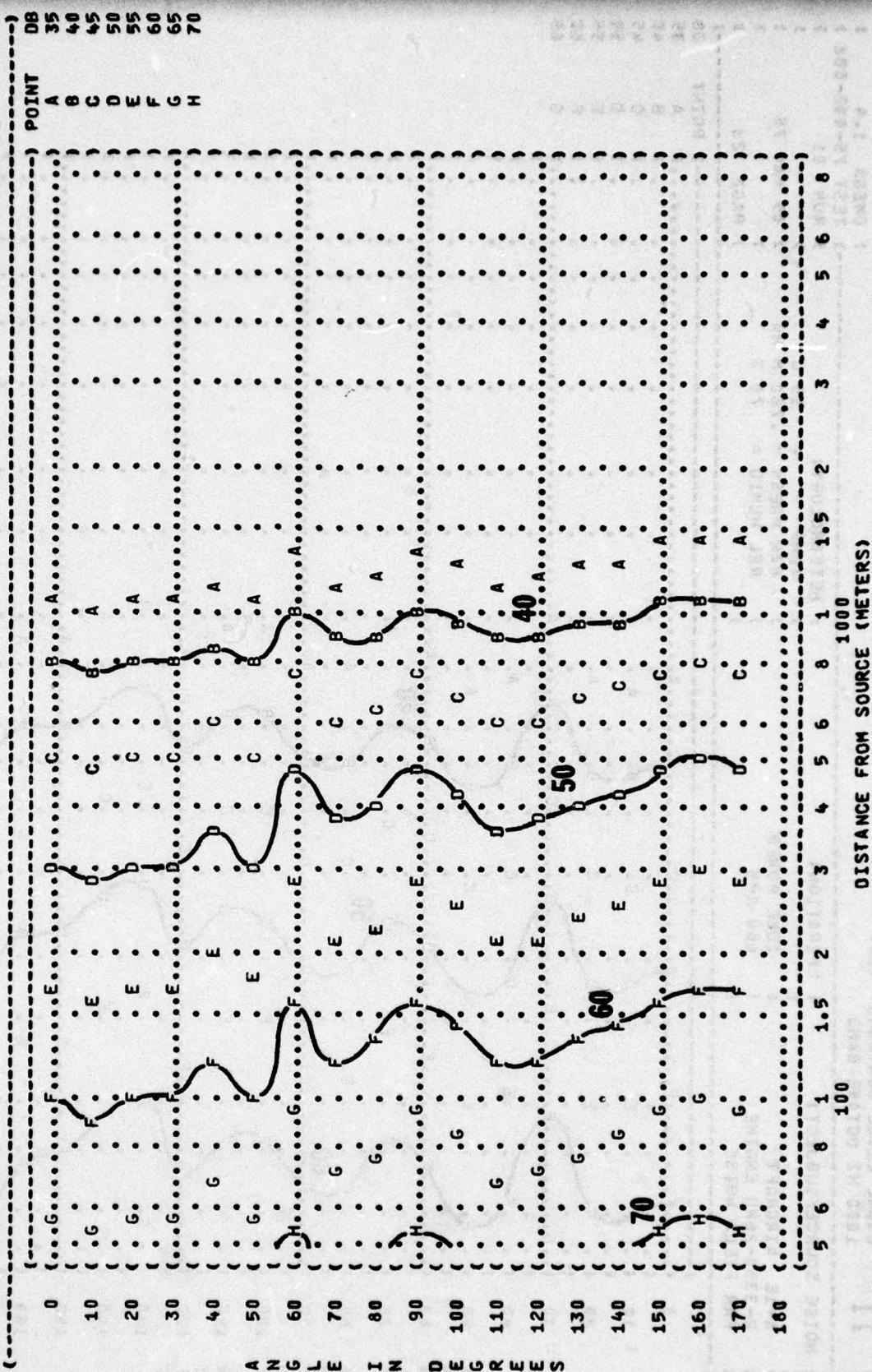


FIGURE 11
SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DB)
1000 Hz OCTAVE BAND

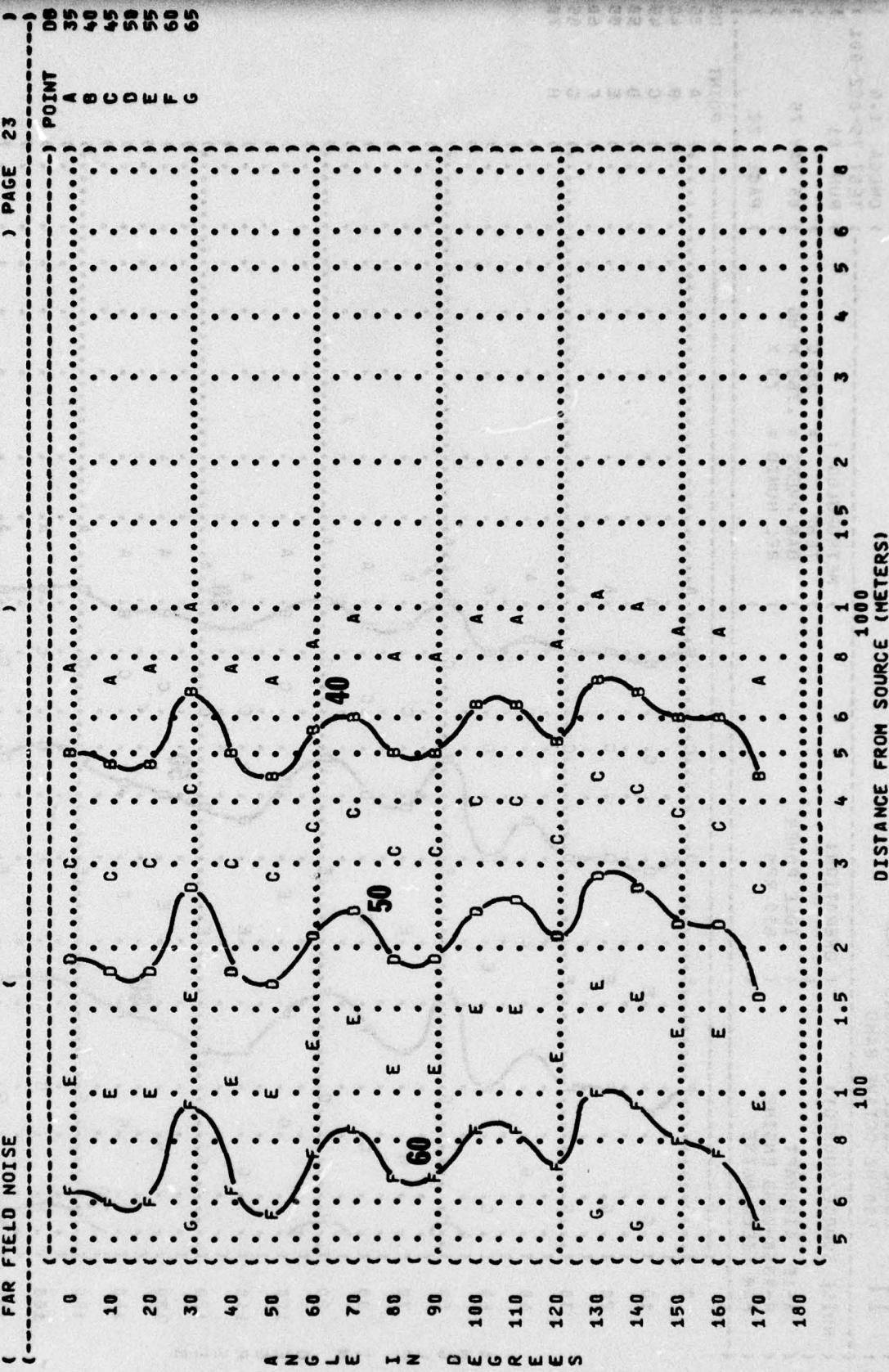
A-1E AIRCRAFT
R-3350-26MD ENGINE
FAR FIELD NOISE

NOISE SOURCE/SUBJECT:

OPERATION:
IDLE POWER
650 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 01

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
PAGE 23



{ FIGURE 1 SOUND PRESSURE LEVEL (SPL)
 11 EQUAL LEVEL CONTOURS (DB)
 2000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

A-1E AIRCRAFT
 R-3350-26ND ENGINE
 FAR FIELD NOISE

OPERATION:

IDLE POWER
 650 RPM

) IDENTIFICATION:
) OMEGA 1-4
) TEST 75-002-001
) RUN 01
) PAGE 24

) METEOROLOGY:
) TEMP = 15 C
) BAR PRESS = .760 M HG
) REL HUMID = 70 %

) POINT DB
 A 35
 B 40
 C 45
 D 51
 E 55
 F 60
 G 65

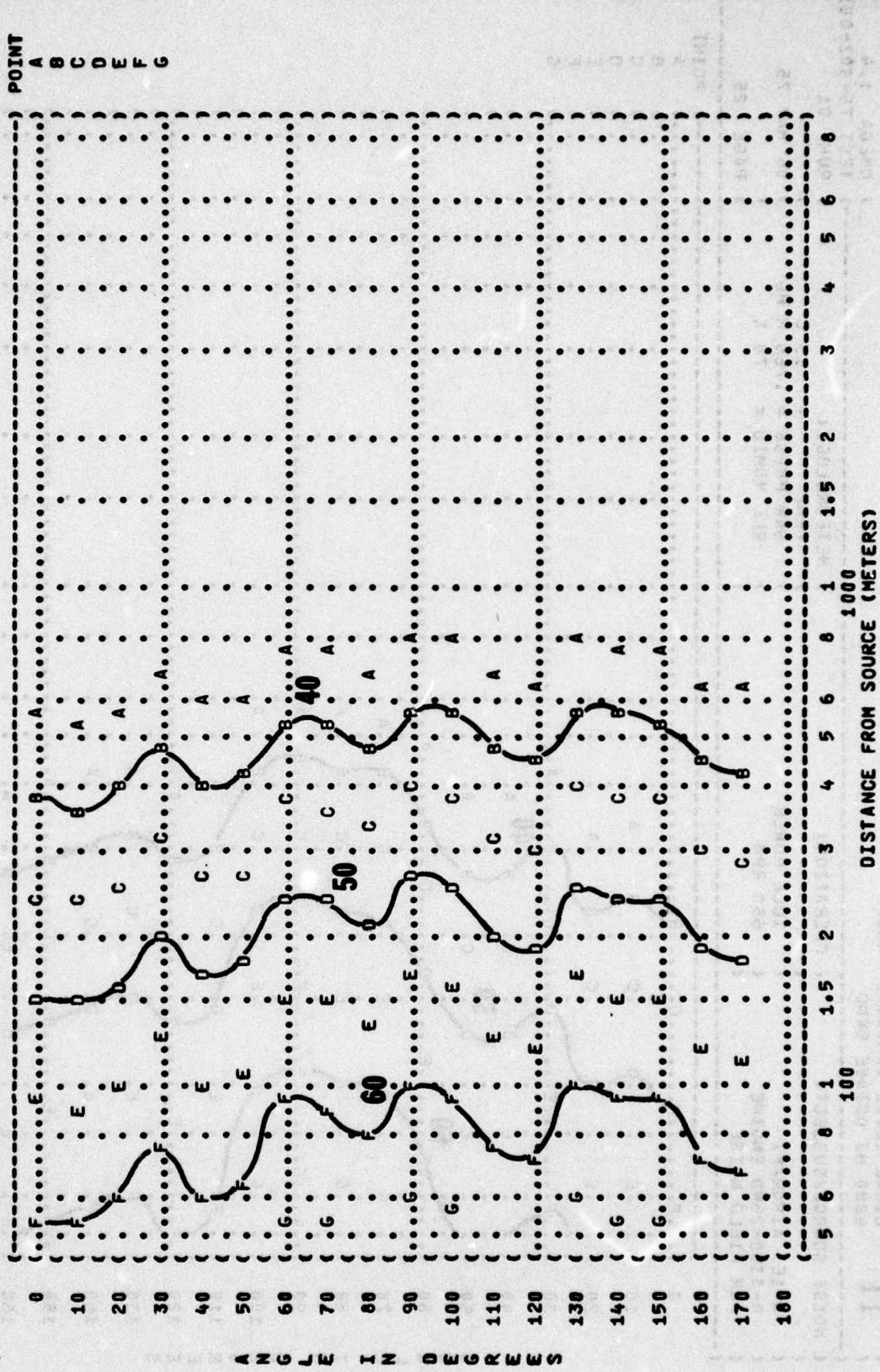


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
4000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
IDLE POWER
650 RPM

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

TEST 75-002-001
RUN 01

PAGE 25

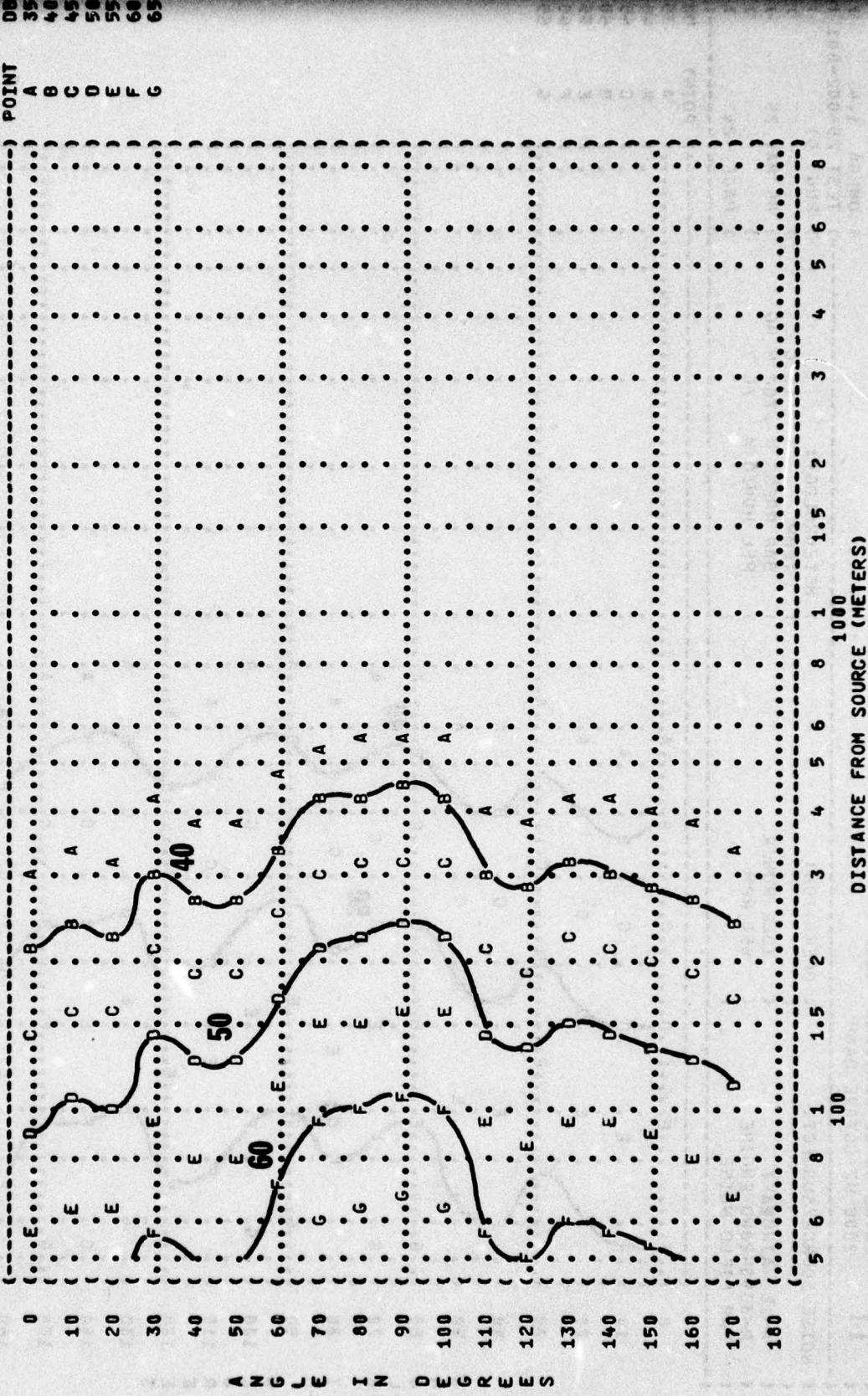


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
 8000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
 A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

IDENTIFICATION:
 OMEGA 1-4
 TEST 75-002-001
 RUN 01

OPERATION:
 IDLE POWER
 650 RPM

METEOROLOGY:
 TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 PAGE 26

POINT DB
 A 35
 B 40
 C 45
 D 50
 E 55
 F 60
 G 65

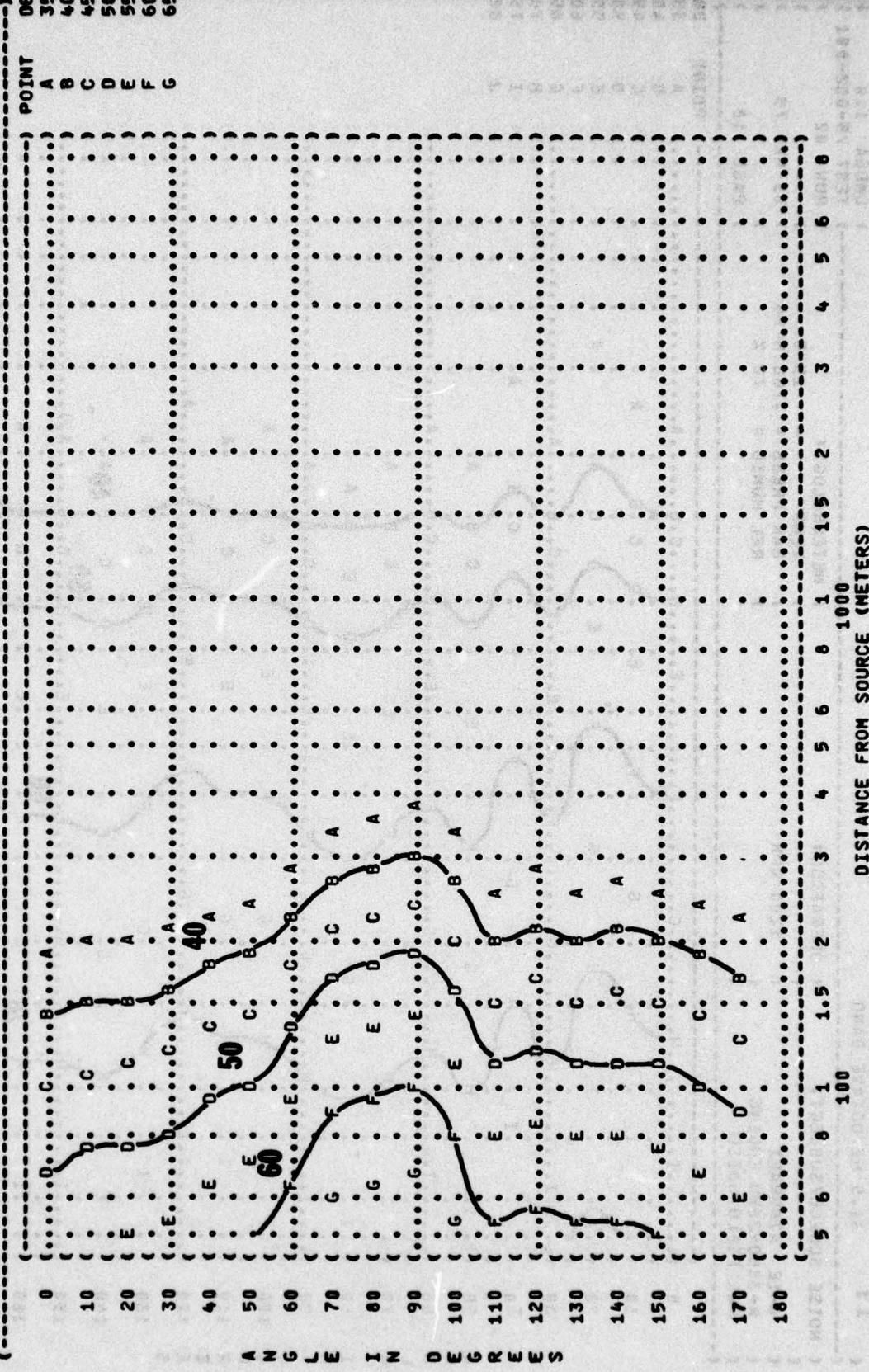


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (dB)
31.5 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:

A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:

1200 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 02
PAGE 18

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 N HG
REL HUMID = 70 %

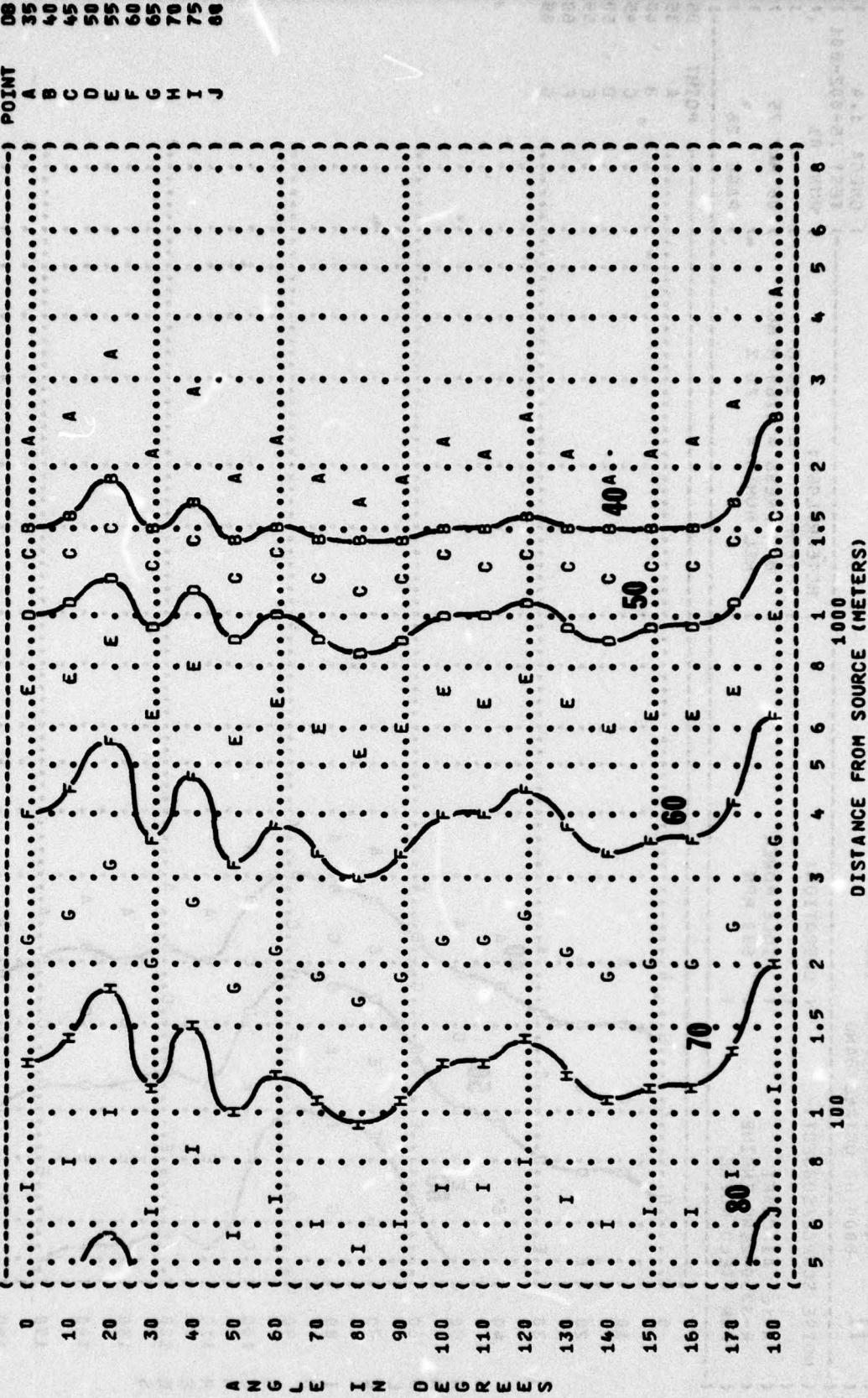


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (dB)
 63 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
**A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE**

OPERATION:
 1200 RPM

IDENTIFICATION:
 OMEGA 1.4
 TEST 75-002-001
 RUN 02
 METEOROLOGY:
 TEMP = 15 C
 BAR PRESS = .760 MM HG
 REL HUMID = 70 %
 PAGE 19

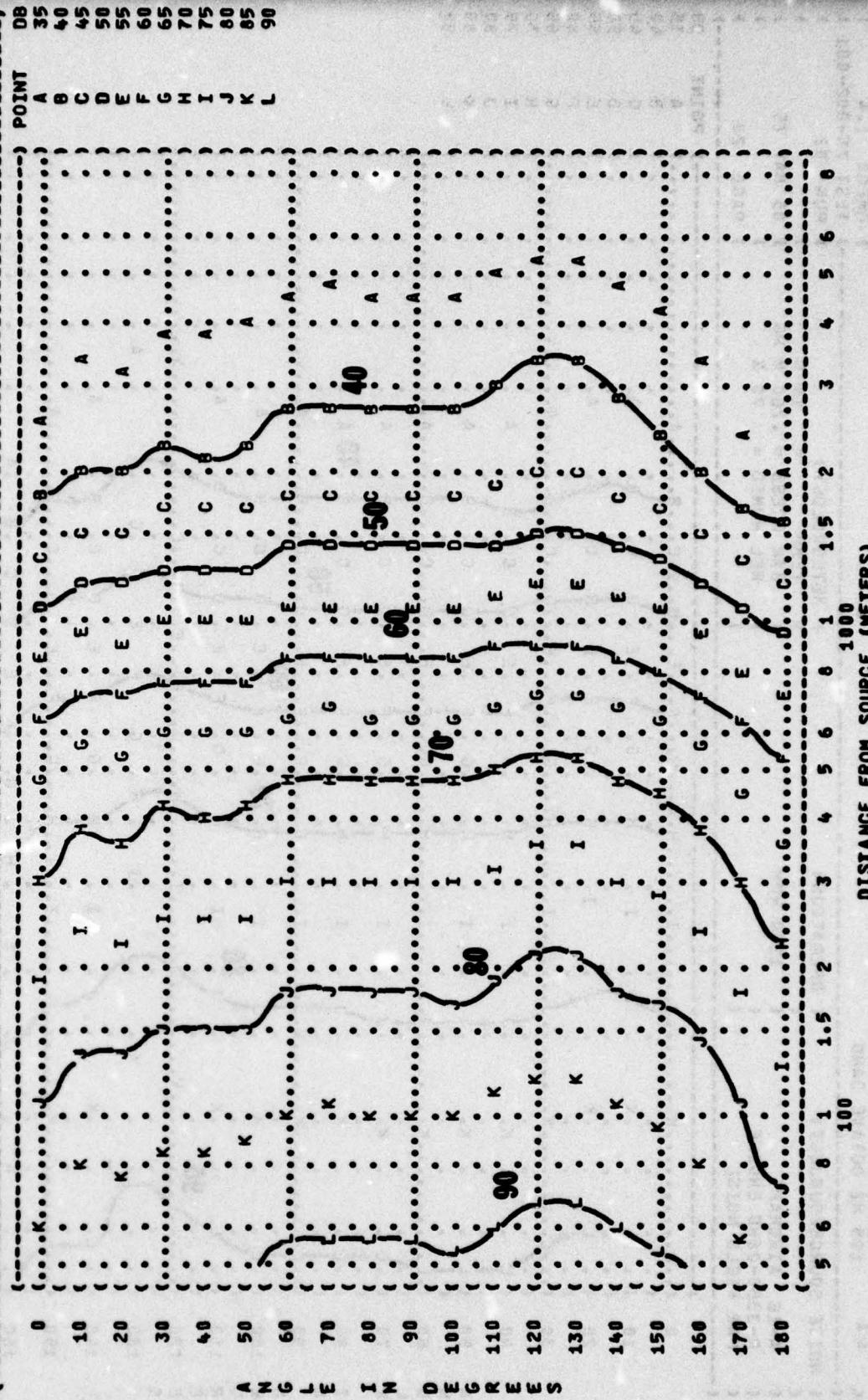


FIGURE 1 SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS
125 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26ND ENGINE
FAR FIELD NOISE

OPERATION:
1200 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 02

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %
PAGE 20

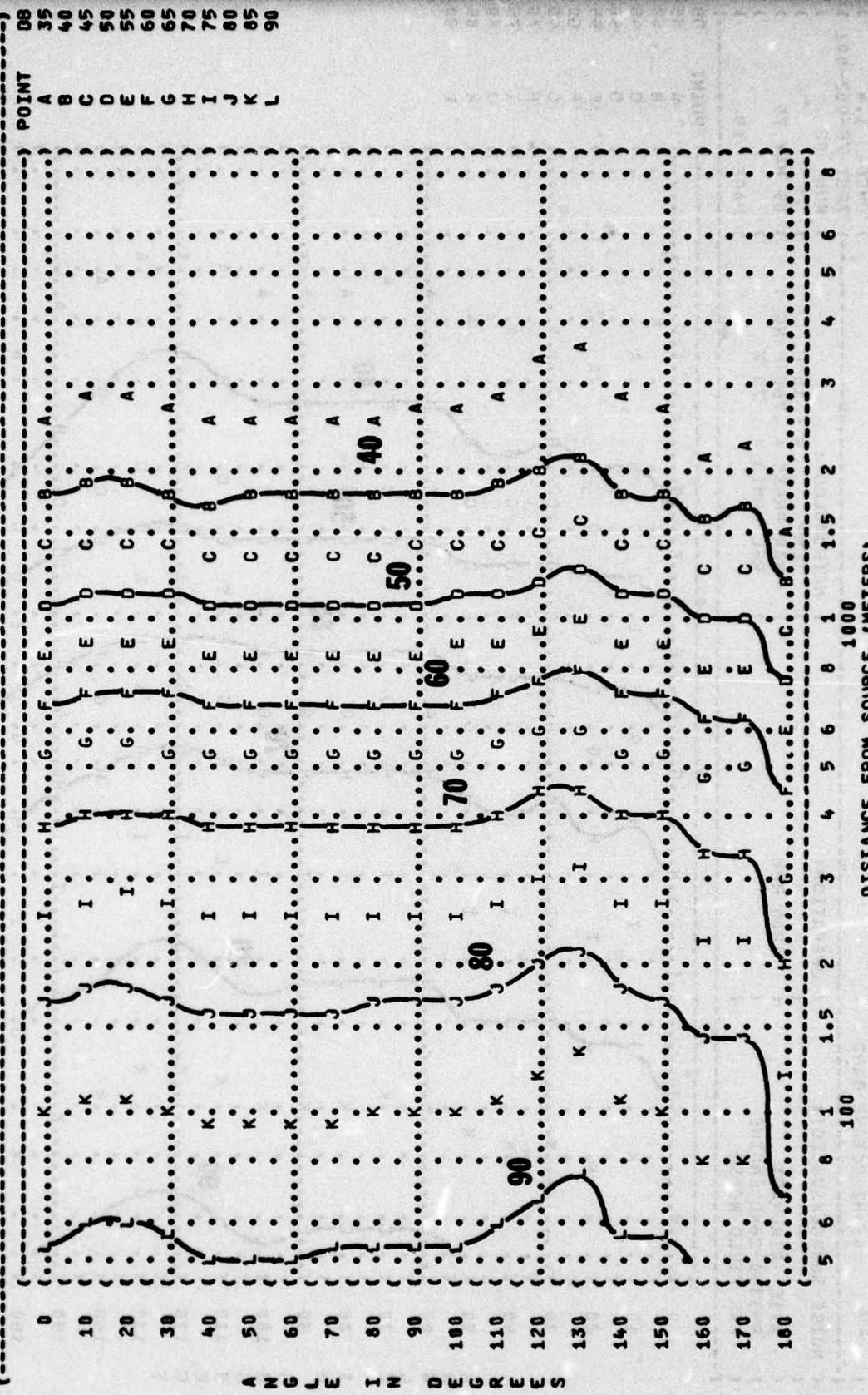


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
250 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT

R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
1200 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 02
PAGE 21

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 HG
REL HUMID = 70 %

POINT DB
A 35
B 40
C 45
D 50
E 55
F 60
G 65
H 70
I 75
J 80

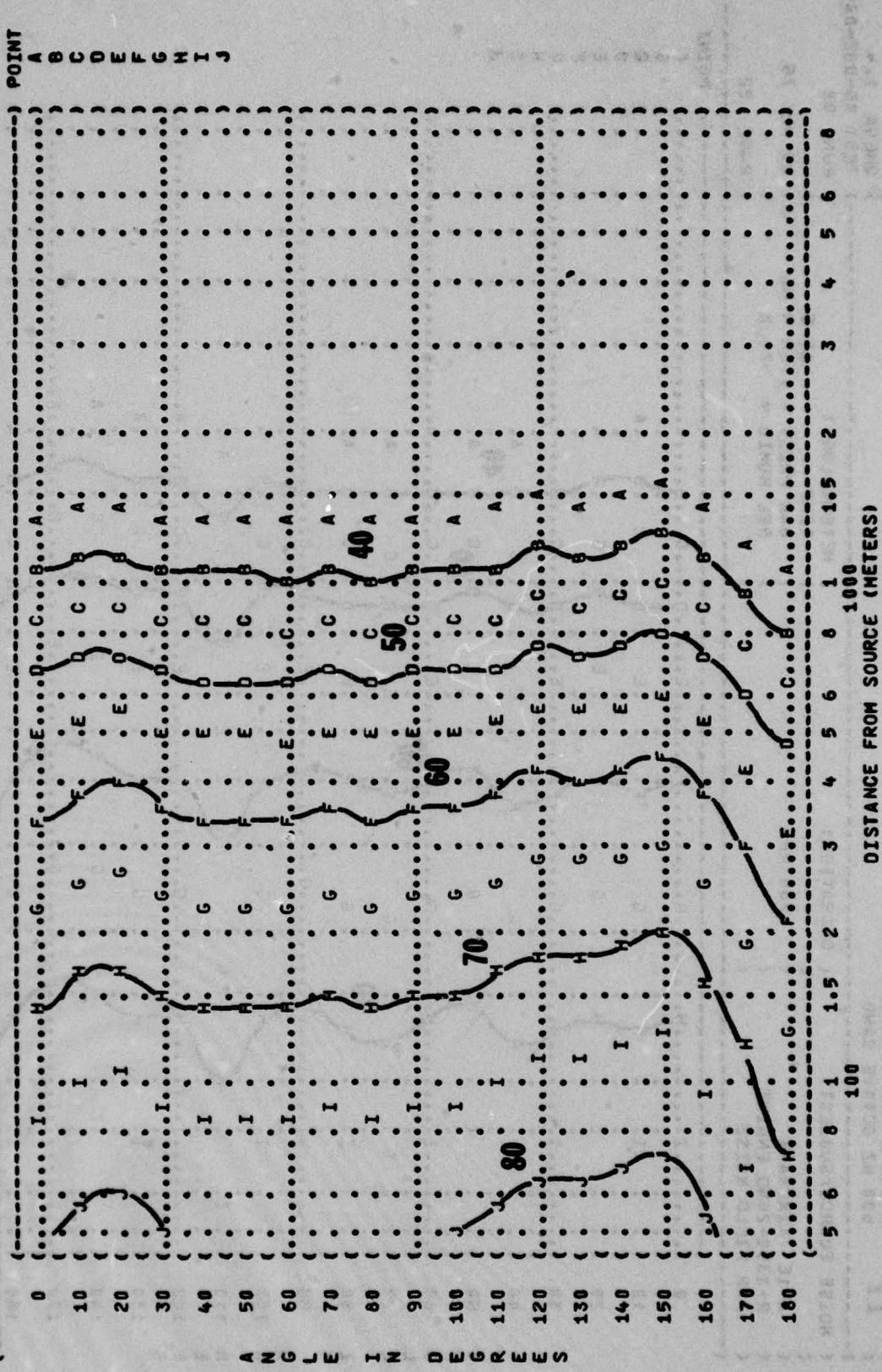


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS
500 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26ND ENGINE
FAR FIELD NOISE

OPERATION:
1200 RPM

IDENTIFICATIONS:
OMEGA 1.4
TEST 75-002-001
RUN 02
MAY 75
PAGE 22

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

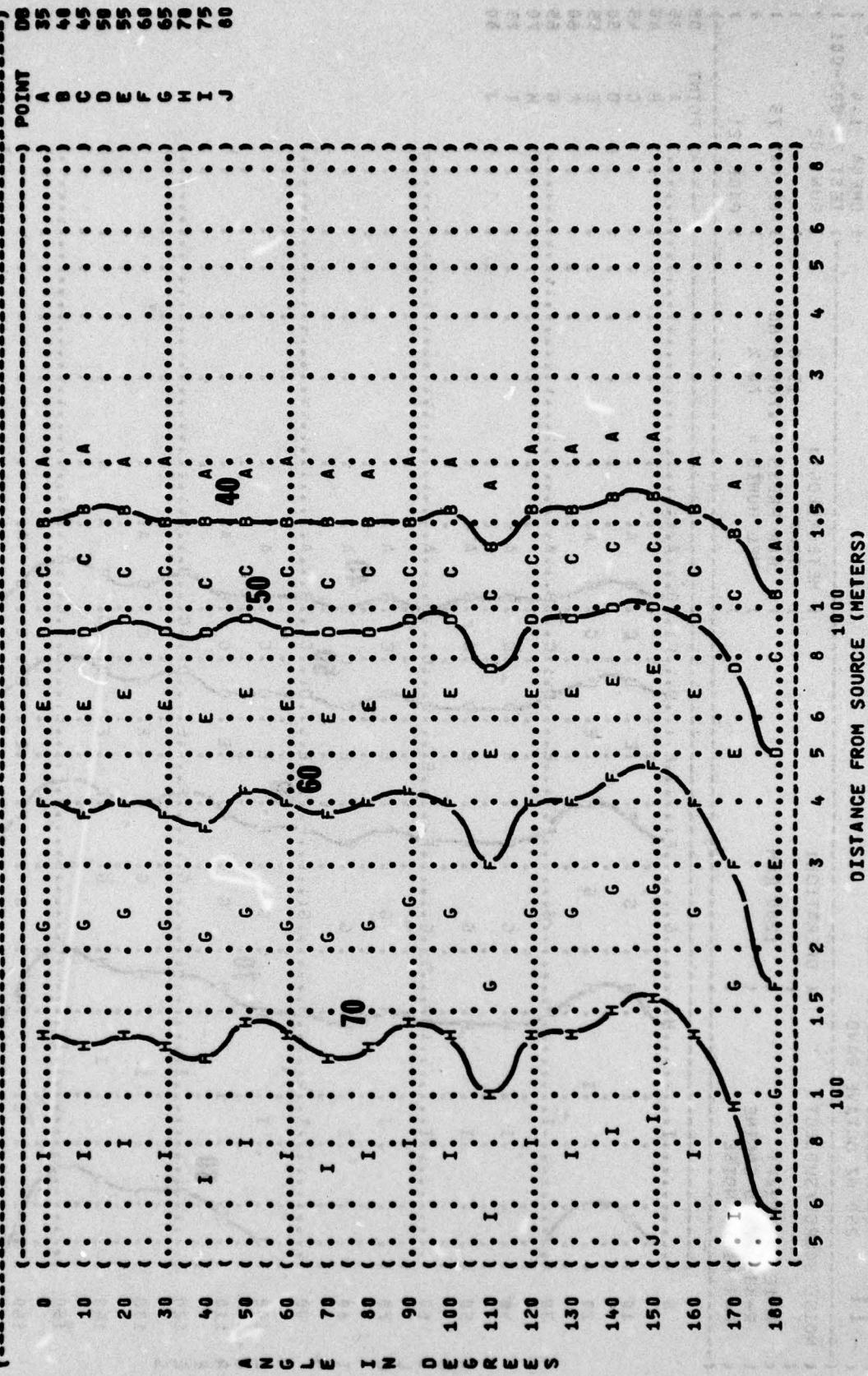


FIGURE:	SOUND PRESSURE LEVEL (SPL) EQUAL LEVEL CONTOURS 11 1000 Hz OCTAVE BAND	NOISE SOURCE/SUBJECT	OPERATOR
A-1E AIRCRAFT			
R-3350-26WD ENGINE			
FAR FIELD NOISE			

IDENTIFICATION:
 OMEGA 1.4
 TEST 75-002-001
 RUN 02
 PAGE 23
 METEOROLOGY:
 TEMP = 15 C
 BAR PRESS = .760 M HG
 REL HUMID = 70 %
 OPERATIONS:
 1200 RPM

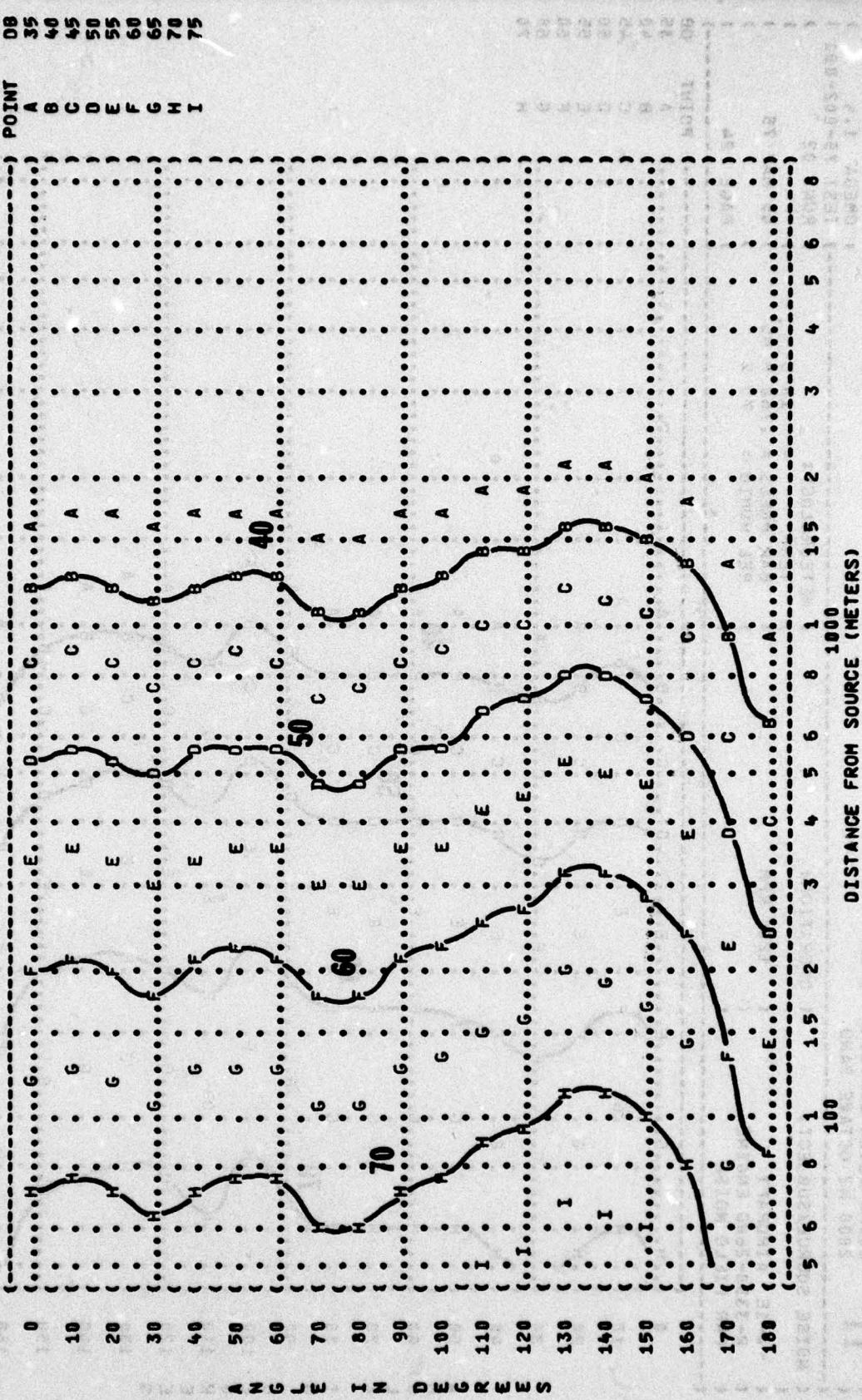


FIGURE 1 SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
 2000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
 A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

OPERATION:
 1200 RPM

IDENTIFICATION:

OMEGA 1.4
 TEST 75-002-001
 RUN 02

METEOROLOGY:
 TEMP = 15 C
 BAR PRESS = 760 M HG
 REL HUMID = 70 %
 PAGE 24

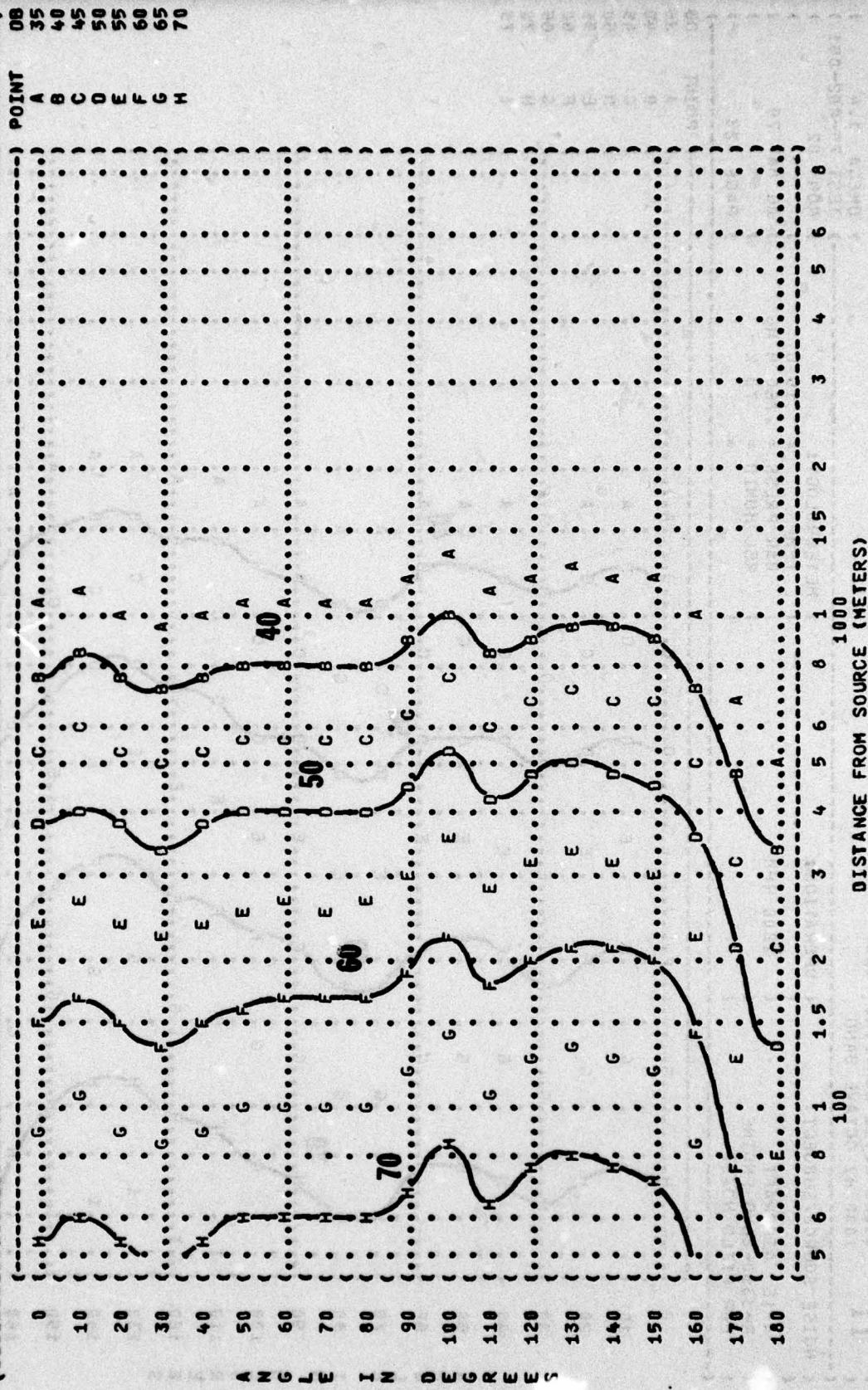


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
4000 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26ND ENGINE
FAR FIELD NOISE

OPERATION:
1200 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 02

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 MM HG
REL HUMID = 70 %
PAGE 25

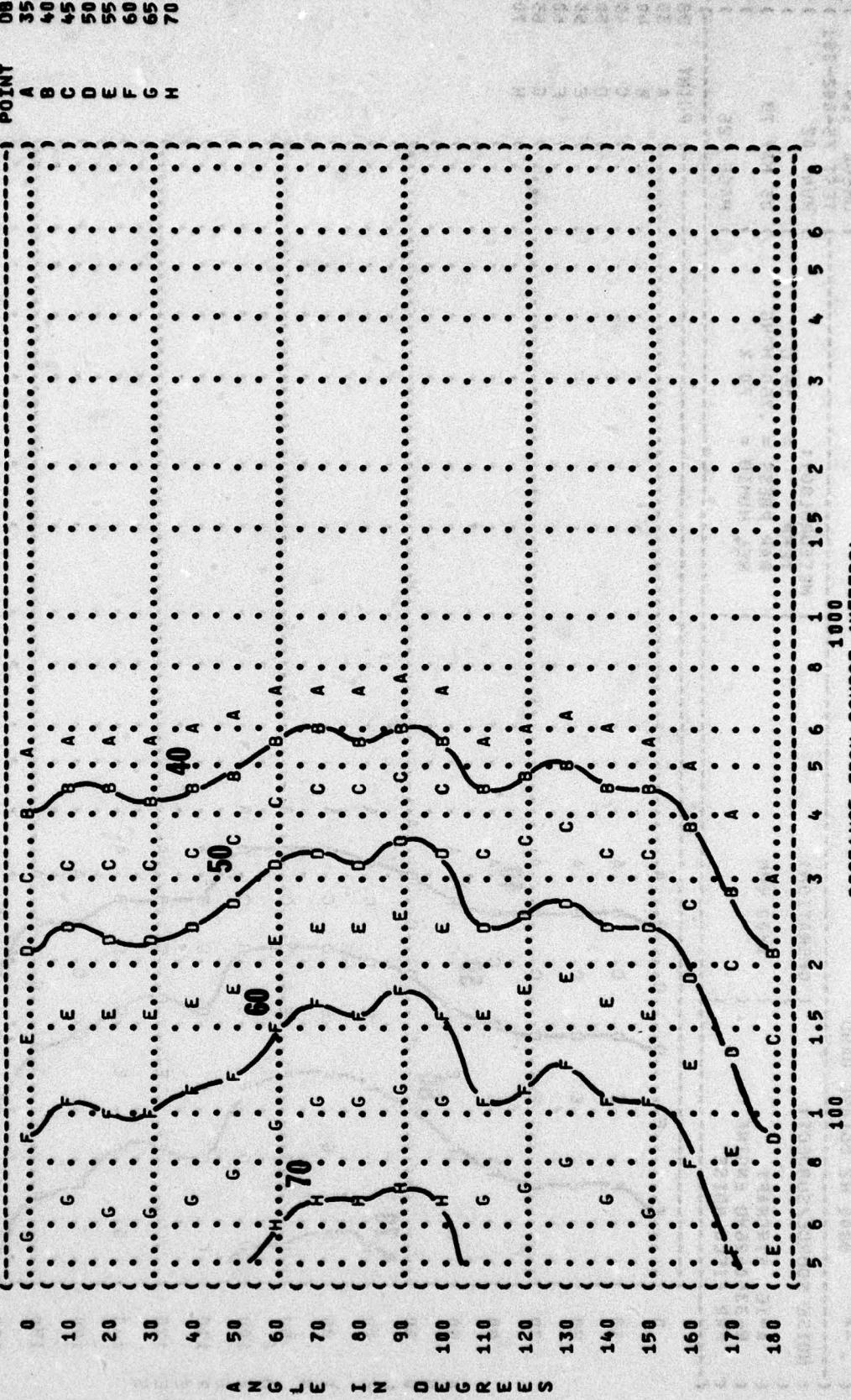


FIGURE 1 SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
8000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
1200 RPM

IDENTIFICATION:
OMEGA 1,4
TEST 75-002-01
RUN 02
METEOROLOGY:
TEMP = 15 C
BAR PRESS = 760 MM HG
REL HUMID = 70 %
PAGE 26

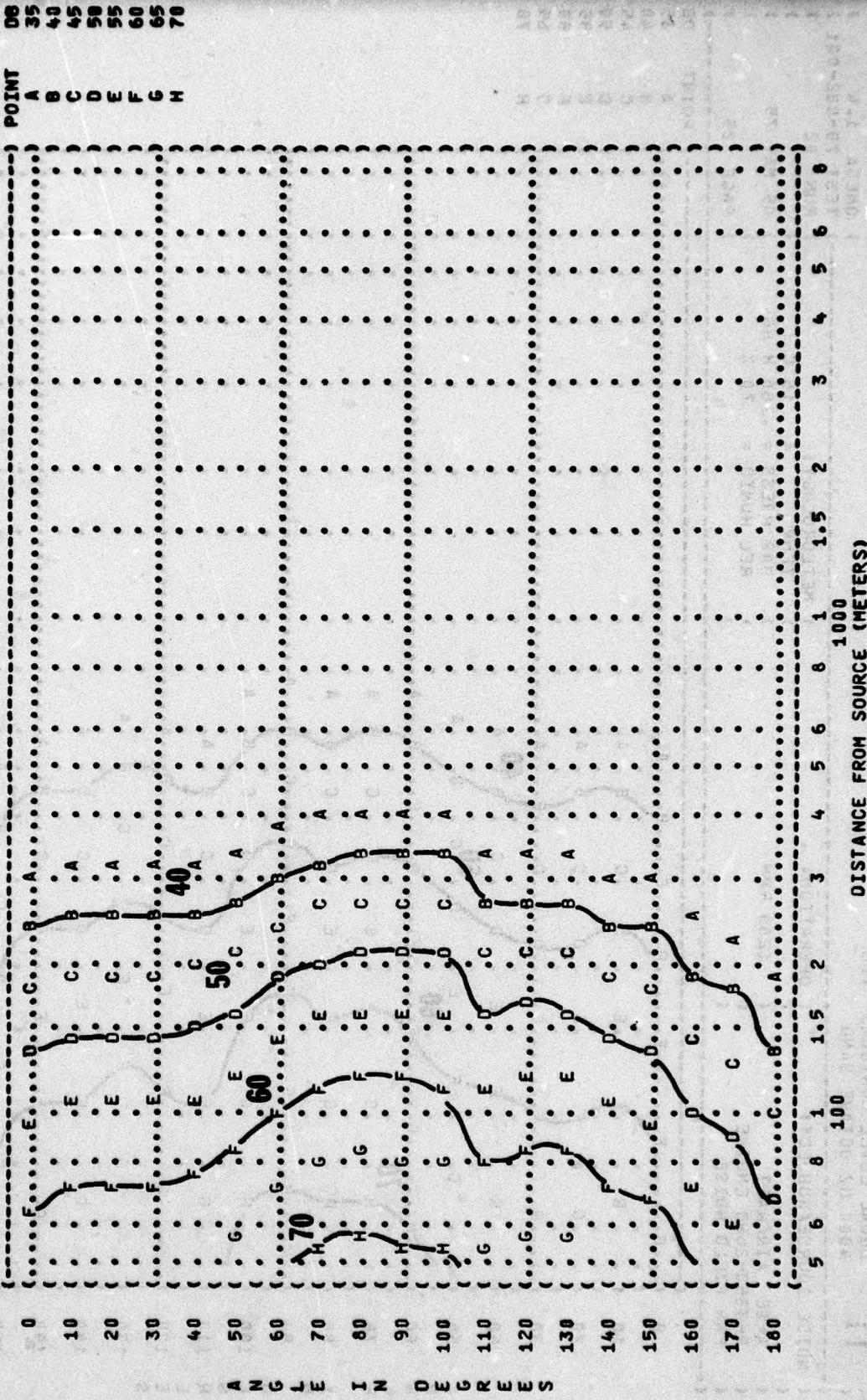


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (0B)
31.5 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION:
MILITARY POWER
2800 RPM

IDENTIFICATION:

OMEGA 1.4
TEST 75-002-001
RUN 03
PAGE 18

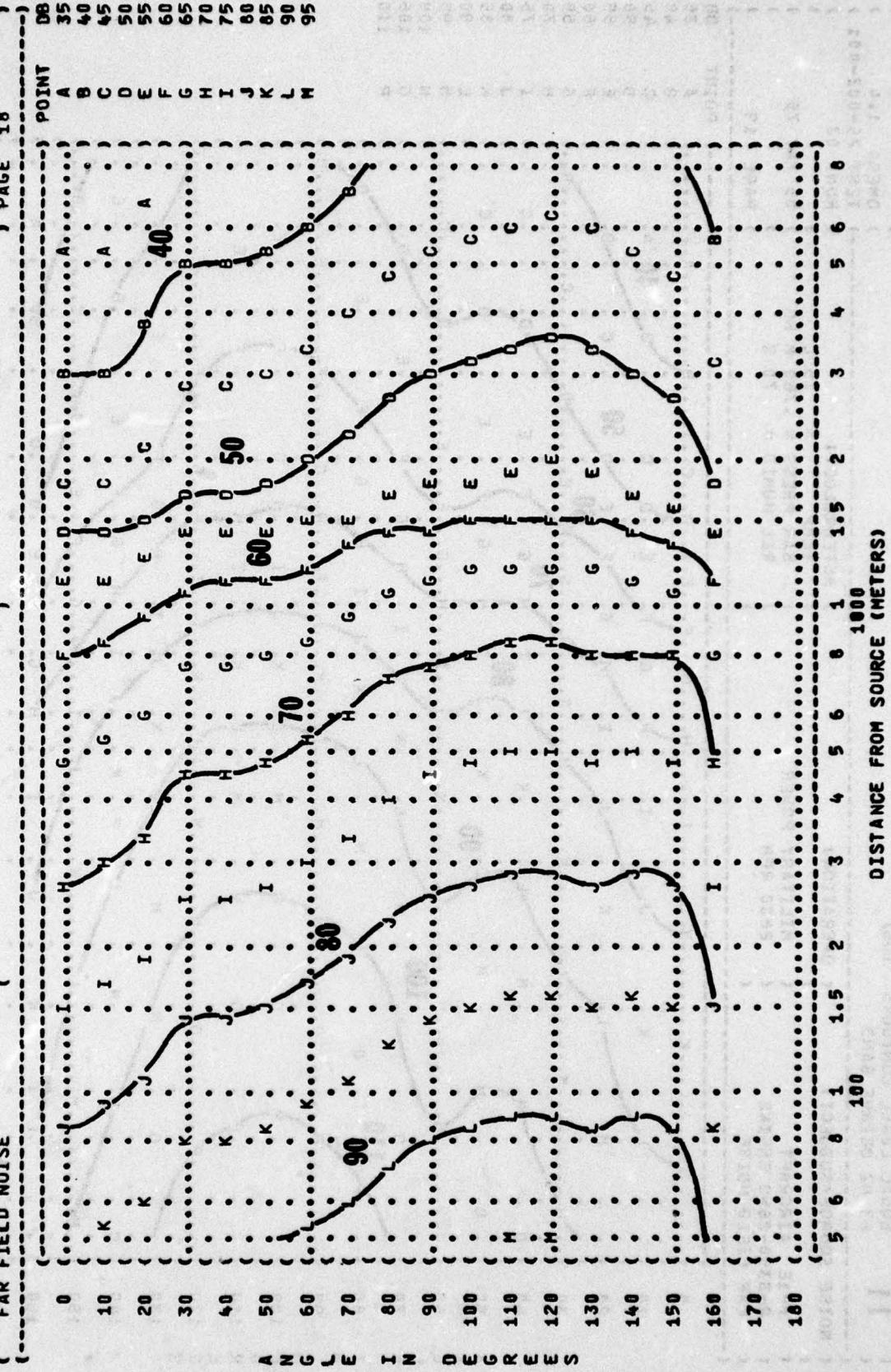


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (dB)
63 Hz OCTAVE BAND

NOISE SOURCE/SUBJECT:
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

MILITARY POWER
2800 RPM

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 03

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 Hg
REL HUMID = 70 %

05 MAY 75

PAGE 1.9

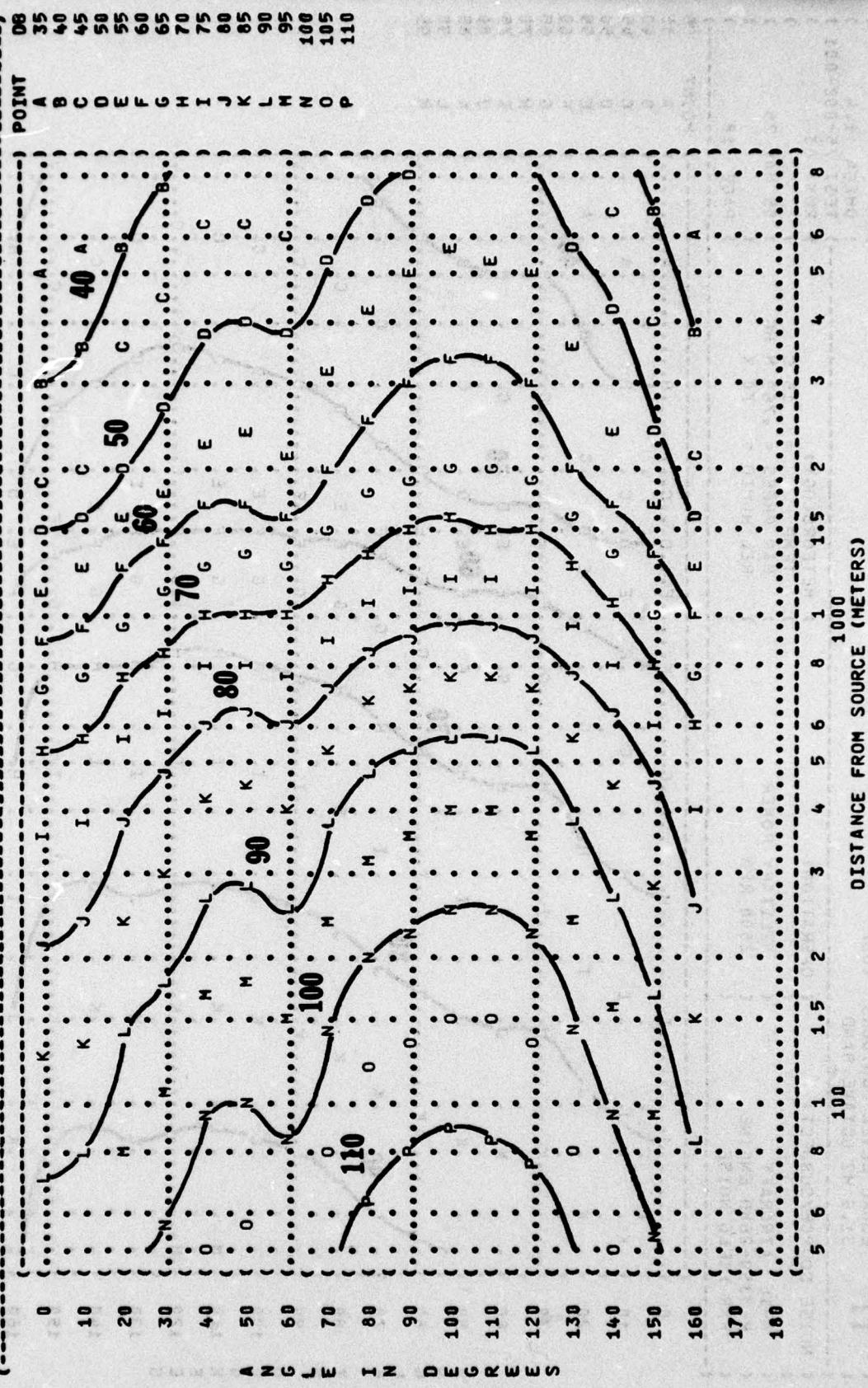


FIGURE 11 EQUAL LEVEL CONTOURS (DB)
125 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT: A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION: MILITARY POWER
2800 RPM

METEOROLOGY: TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

TEST 75-002-001
RUN 03

PAGE 20

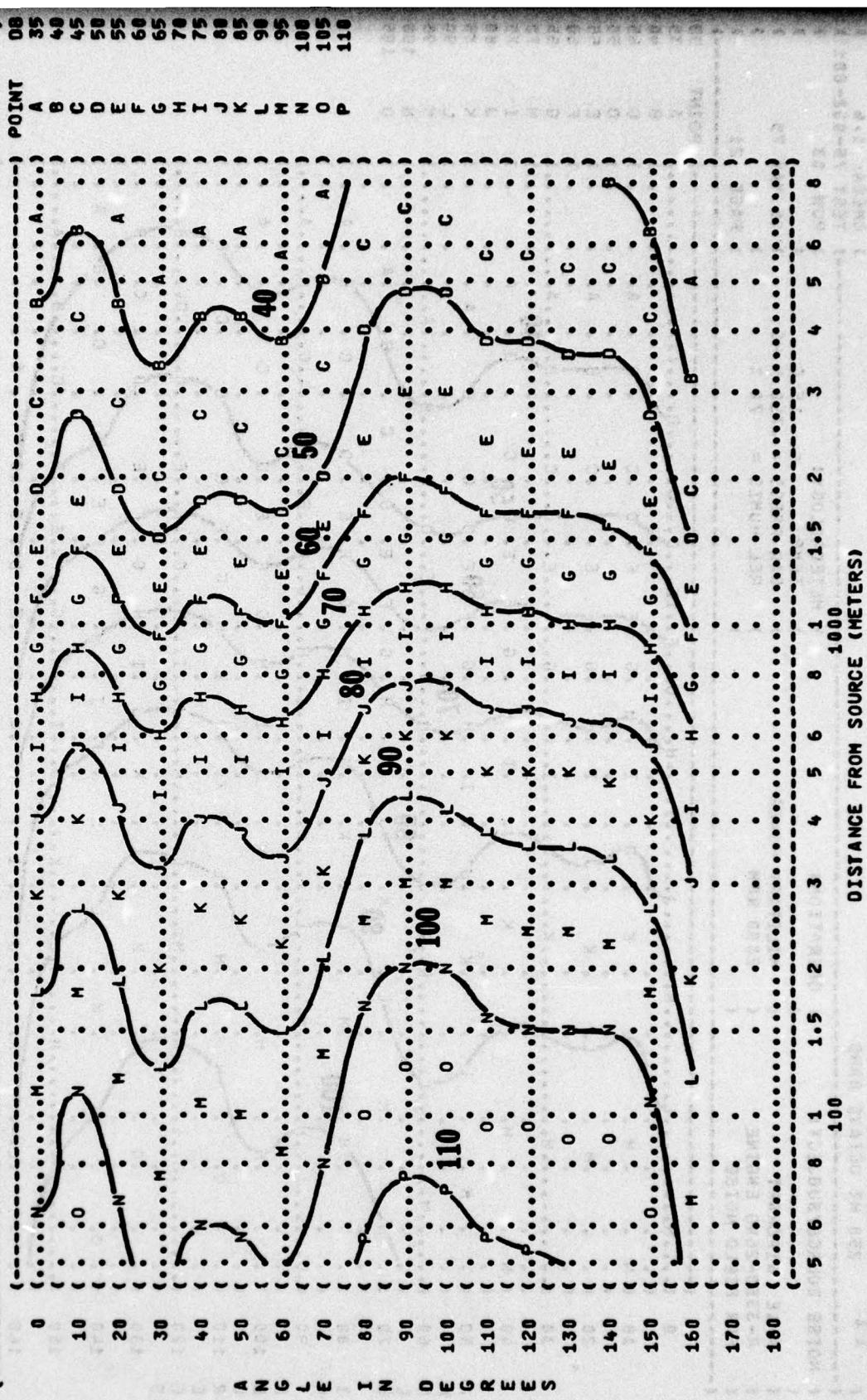


FIGURE 11
SOUND PRESSURE LEVEL (SPL)
EQUAL LEVEL CONTOURS (DBA)

11
250 Hz OCTAVE BAND
NOISE SOURCE/SUBJECT :
A-1E AIRCRAFT
R-3350-26WD ENGINE
FAR FIELD NOISE

OPERATION :
MILITARY POWER
2800 RPM

TEST 75-082-081
RUN 03
OMEGA 1.4
05 MAY 75
BAR PRESS = .760 HG
REL HUMID = 70 %
PAGE 21

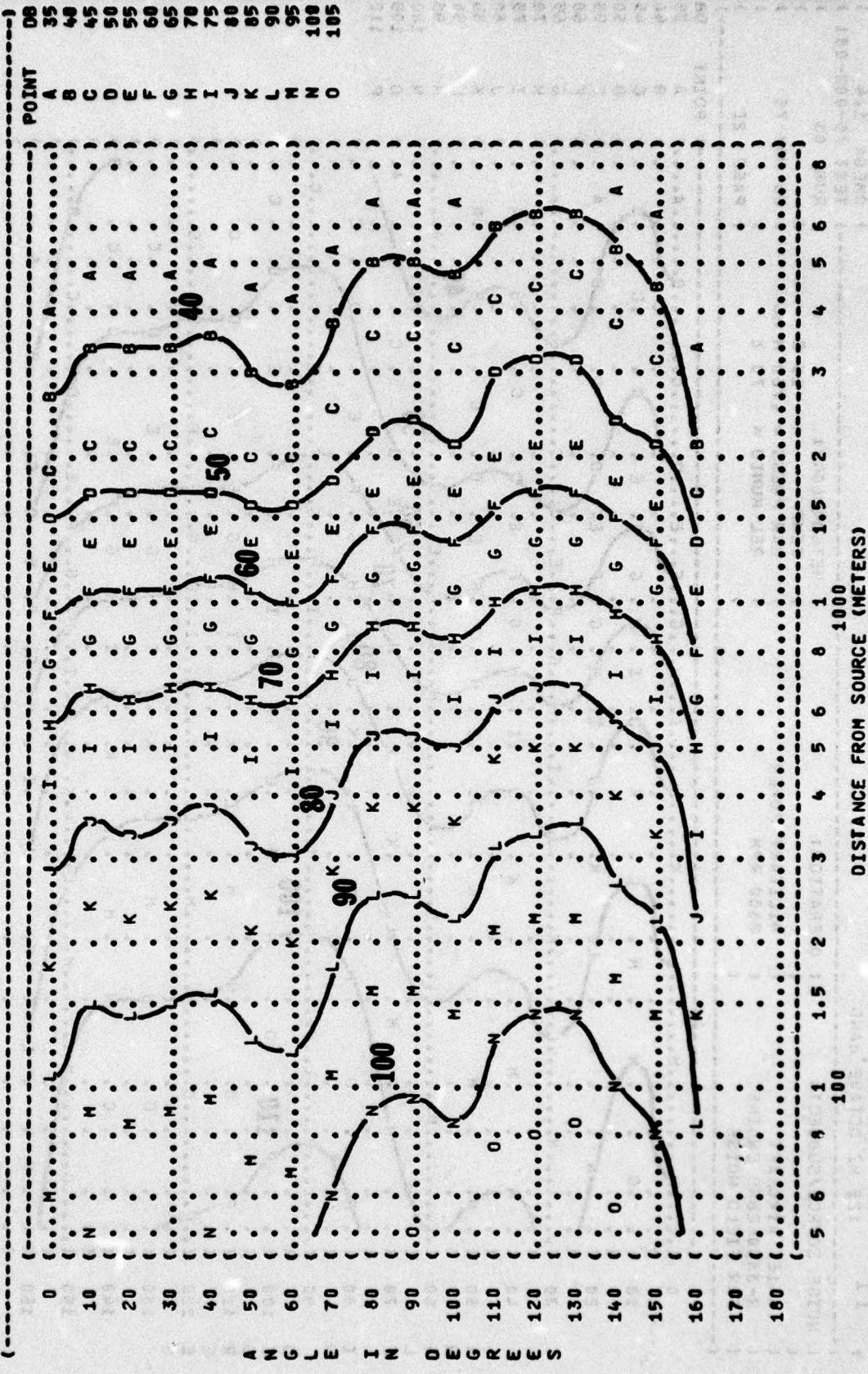


FIGURE 1 SOUND PRESSURE LEVEL (SPL)

11

EQUAL LEVEL CONTOURS
500 Hz OCTAVE BAND

A-1E AIRCRAFT
R-3350-26ND ENGINE
FAR FIELD NOISE

MILITARY POWER
2800 RPM

OPERATION:

METEOROLOGY:
TEMP = 15 C
BAR PRESS = .760 M HG
REL HUMID = 70 %

IDENTIFICATION:
OMEGA 1.4
TEST 75-002-001
RUN 03
PAGE 22

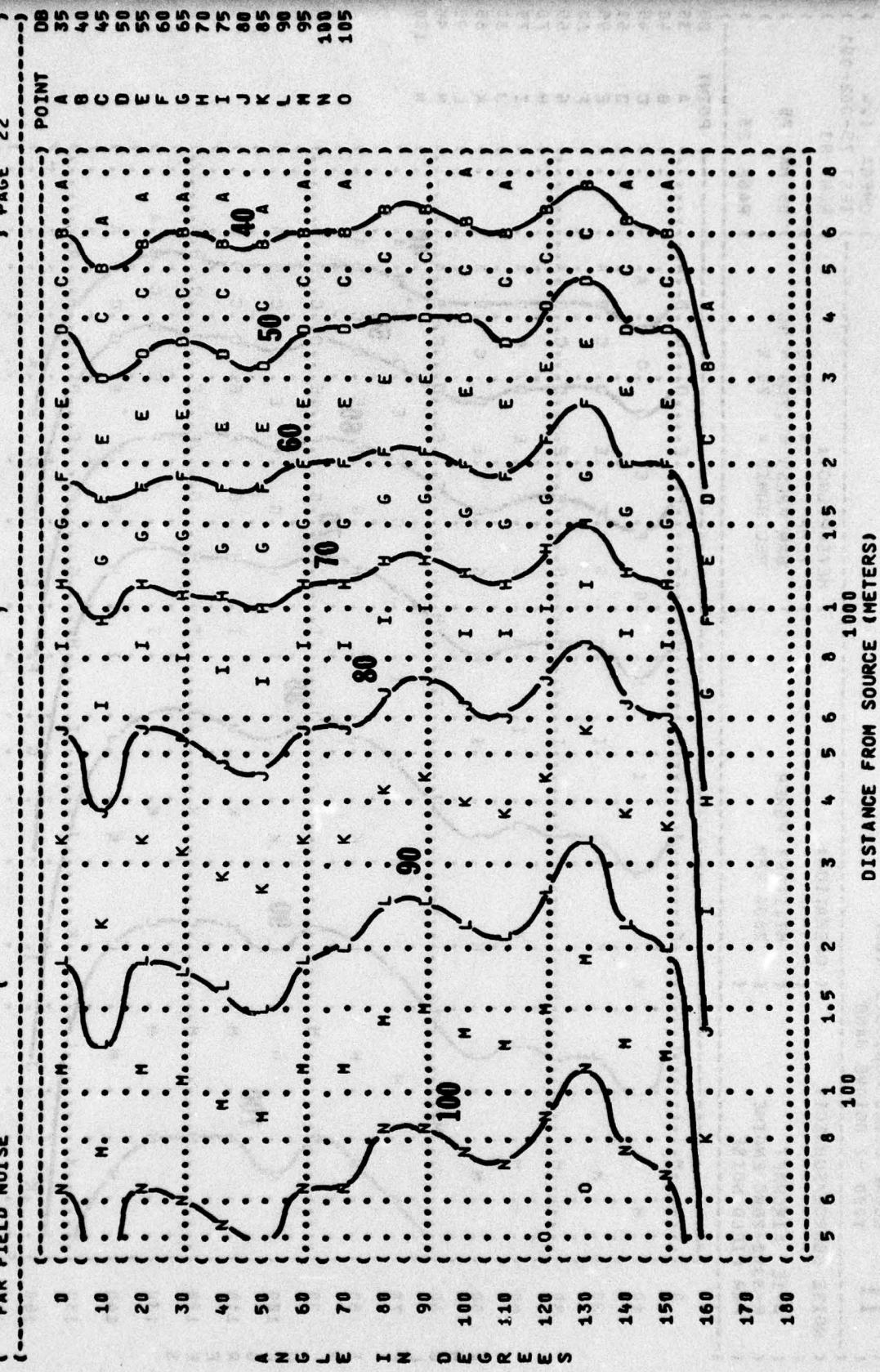


FIGURE:	SOUND PRESSURE LEVEL (SPL)	(SPL)	OPERATOR
11	EQUAL LEVEL CONTOURS	(0B)	MILITARY
	1000 HZ OCTAVE BAND		2000
NOISE SOURCE/SUBJECT:			
A-1E AIRCRAFT			
R-3360-26MD ENGINE			
FAR FIELD NOISE			

IDENTIFICATION:
OMEGA 14
TEST 75-002-001
RUN 03
05 MAY 75
PAGE 23

Detailed description: This is a contour map showing sound pressure levels (dB) as a function of distance from a source. The vertical axis represents the point where measurements were taken, ranging from 0 to 100. The horizontal axis represents the distance from the source in meters, ranging from 0 to 180. Contour lines are labeled with values 35, 40, 45, 50, 55, 60, 65, 70, 75, 80, 85, 90, 95, and 100 dB. Points A through N are marked along the contours. The map shows that sound pressure levels generally increase with distance from the source, with higher levels occurring at greater distances.

FIGURE 11
 SOUND PRESSURE LEVEL (SPL)
 EQUAL LEVEL CONTOURS (DB)
 2000 Hz OCTAVE BAND
 NOISE SOURCE/SUBJECT:
 A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

OPERATIONS
 MILITARY POWER
 2800 RPM

IDENTIFICATIONS
 OMEGA 1.4
 TEST 75-002-001
 RUN 03
 05 MAY 75
 BAR PRESS = .760 Hg
 REL HUMID = 70 %
 PAGE 24

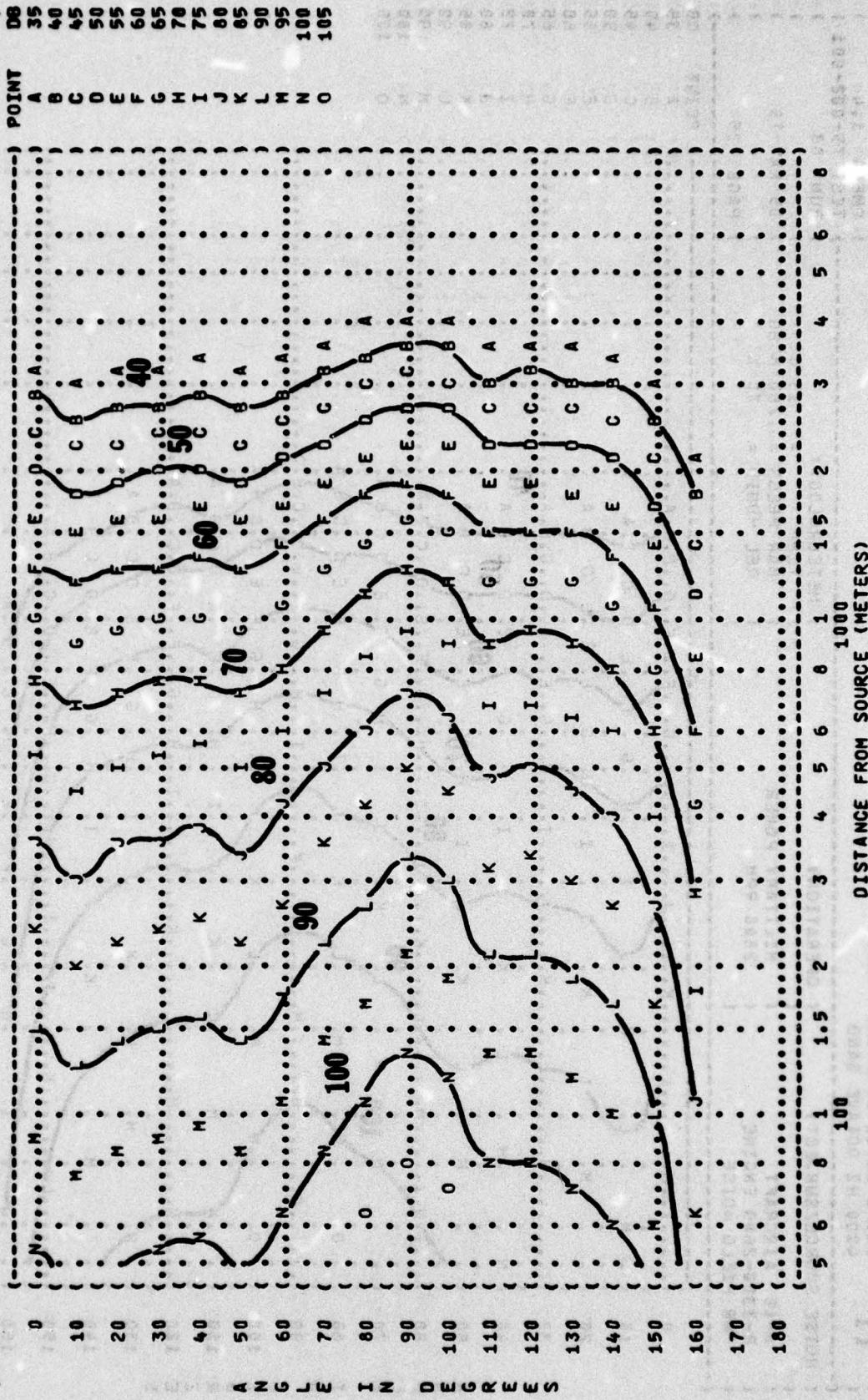
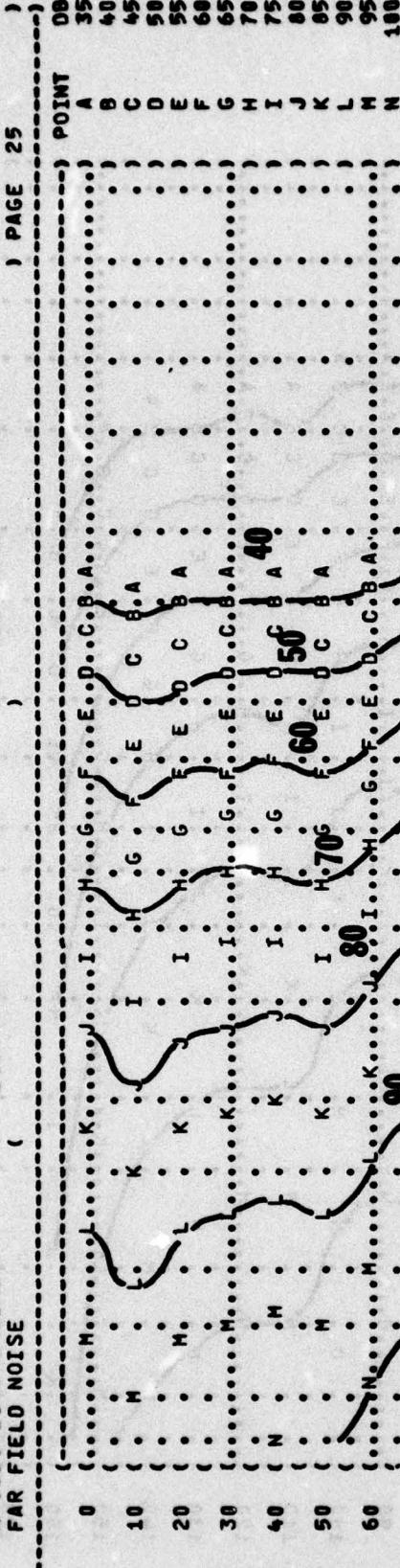


FIGURE: SOUND PRESSURE LEVEL (SPL)
11 EQUAL LEVEL CONTOURS (DB)
 4000 HZ OCTAVE BAND

NOISE SOURCE/SUBJECT:
 A-1E AIRCRAFT
 R-3350-26WD ENGINE
 FAR FIELD NOISE

OPERATION:
 MILITARY POWER
 2800 RPM

IDENTIFICATION:
 OMEGA 1.4
 TEST 75-002-001
 RUN 03



105

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