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Evaluation of 120 glued laminated (glulam) beams provided criteria for improved utilization of lumber in such beams. Objectives were: (1) to determine if lumber grade can be somewhat reduced on the compression side of beams without significantly changing design strength; (2) to establish analytic procedures for incorporating lumber having had its modulus of elasticity (E) determined (E-rated lumber) into glulam beams; and (3) to determine the effect on beam properties of using lumber with limited wane.

Test results indicated that wane of up to 1/6 of the lumber width along either or both edges did not result in large shear weaknesses; thus design levels in shear equal

#### Abstract

to 2/3 that of wane-free lumber appear justified. In addition, a design procedure was developed for reducing grade of lumber in the compression side of glulam beams. Such use of lower grade material was not found to change the breaking level of near minimum strength beams. Testing was also done of Erated lumber in glulam combinations. Procedures for incorporating E-rated lumber into beams are presented so that results will be in line with those for beams made entirely of visually graded lumber.

> The procedures developed will provide those preparing specifications a wider raw material base for glulam timber, resulting in more efficient use of our timber resource.

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# IMPROVED UTILIZATION OF LUMBER IN GLUED LAMINATED BEAMS<sup>1/</sup>

By

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#### INTRODUCTION

Efficient utilization of America's forest resources must be a primary concern of all forestry-oriented research if the Nation's needs for wood are to be met in the future. Revised design of glued iaminated (glulam) beams may permit lower grade materials to be used without seriously affecting the design strength, thus extending national supplies of high-grade lumber materials.

A design concept to estimate bending properties for glulam beams from mixed species and different lumber grades, developed at the U.S. Forest Products Laboratory (FPL), was published in 1974 (<u>15</u>).<sup>3/</sup> That approach used transformed section analysis, the  $I_{\mathbf{K}}/I_{\mathbf{G}}$  concept, and tension lamination requirements to predict near minimum bending strength and average stiffness.

Three approaches to conserving lumber in beams, not investigated in the 1974 study (15), are explored here: (a) reducing grade of lumber on the compression side of beams; (b) using lumber with wane for inner lamination; and (c) extending criteria developed for visually graded lumber to lumber which has had its modulus of elasticity (E) determined (E-rated lumber). The present study has extended criteria developed in (15) to unsymmetric glulam beams using visually graded or a combination of E-rated and visually graded lumber. Criteria were evaluated by tests of 12or 18-in,-deep beams designed to fail first in compression or to have near equal likelihood of failing in either tension or compression. In addition, the effect of introducing limited wane in the interior laminations was evaluated by subjecting beams containing such material to high shear stresses during tests.

#### Compression Side Overdesign

Beams designed by both the original IK/IG concept and the concept as modified with transformed section analysis will generally fail in tension, not compression. Glulam beams are usually overdesigned in compression; this was noted by Moe (14) and Madsen (13), and has been verified since by test results at various laboratories (8,12,15,19). The current standard AITC 117-74 (1) recognizes compression overdesign in that lamination grades are lower for the compression than fc the tension side. The basis for AITC 117-74 is an averaging approach with the IK/IG concept. However, derivation of these combinations does not consider a shift in the neutral axis due to the unbalanced design.

To compensate for compression overdesign, higher clear wood stress values may be assumed for the compression side than for the tension side. Bohannan's study of prestressing ( $\underline{7}$ )indicates how much higher the compressive stress could be. He found that 1,300 lb/in.<sup>2</sup> prestress on the compression side induced a significant number of compression failures in tests of L3 grade Douglas-fir lumber. If this, 1,300 lb/in.<sup>2</sup> value is added to the 6,350 lb/in.<sup>2</sup> clear wood design stress value for Douglas-fir (appendix I), an increase of approximately 20 pct in clear wood stress for the compression side is indicated. This

<sup>1/</sup> Research conducted in cooperation with the American Institute of Timber Construction.

<sup>2/</sup> Maintained at Madison, Wis., in cooperation with the University of Wisconsin.

<sup>3/</sup> Underlined numbers in parentheses refer to literature cited at end of this report.

procedure interprets the 1,300 lb/in.2 value as an increased strength capability for the compression side. This increase will probably not be valid for clear lumber because small clear beams will usually fail in compression, but it should be valid for structural beams.

Analysis of data from three studies—FPL 236 (15) and two test series conducted by Johnson (8,9)—indicates that if this 20 pct value is applied on the compression side in all cases, no beam strengths will fall below the calculated strength levels. One of the goals of the present study has been to determine the amount that the clear wood stress for the compression side of glulam timbers might be increased for various structural grades of lumber.

#### Using E-Rated Lumber

Since the development of mechanical stress rating systems, one promising area of application frequently discussed has been glulam timber ( $\underline{9}$ ). Such a system might be used to select high-grade lumber for outer laminations and lower grade for the inner, less stressed lamination. Koch et al, (<u>11</u>), working with veneer and thin lumber, demonstrated its potential.

Aplin (6) considered utilizing E-rating for manufacturing gluiam beams using structural spruce lumber from Eastern Canada. He concluded from tests of 32 beams that strength and stiffness levels higher than with visually graded material were possible. However, he noted that stiffness grading alone was not sufficient and recommended additional visual assessment of each board. Subsequent work on five southern pine beams confirms the potential but also suggests the need for visual assessment, at least for the highly stressed tension zone material (16).

Based on tests conducted by Johnson (8,9), an AITC Standard was developed on the use of E-rated lumber for glulam beams— AITC 120-71 (2). Unfortunately, the combinations are limited to only those evaluated, and any variation from the standard must be supported by further testing. The promise of E-rating combined with proof loading is further confirmed by Strickler and Pellerin (for one example see (21)). Also, Littleford (12) has noted that E-rating offers potential for developing design stresses 30 pct higher than with visual grading alone for spruce and pine in Western Canada.

We believe that the modified  $I_K/I_G$  concept developed in (15) can be applied to combinations in AITC 120-71 as a theoretical mechanism for studying variations. This standard assumes balanced stiffness combinations, and test results verify that the lower strength beams will fail in tension, not compression. A variation on this standard introducing unbalanced E grades may increase the probability of compression side failures and approach a "balanced strength" design. We believe that failures of this type will result in less variable strength data and more efficient use of material when strength is the principal design limitation.

#### Lumber with Wane

Wane is not permitted in current lumber grades for laminating, which eliminates much otherwise acceptable material. This restriction assures a full-width gluing surface to develop shear strength in the beams. However, by permitting limited wane and reducing the design level in shear, satisfactory beams for dry uses may be obtained.

Effect of wane on the shear strength of glulam beams will be assumed to depend on the amount of wane present. For members subjected to continuous dry use, wane of up to one-sixth of the lumber width at each edge will be assumed to reduce shear strength by up to one-third (i.e., reduced proportional to the maximum shear area lost due to wane). Note: permitting uncontrolled wane along one edge in amounts larger than this is not directly comparable.

Wane may introduce problems due to stress concentrations at the gluelines in beams subjected to repeated wetting and drying. Such exposure, until thoroughly researched, is not recommended; beams made with wany lumber are proposed for dry use only.

#### DEVELOPMENT OF DESIGN CRITERIA

Design criteria in  $(\underline{15})$  are limited in several respects. They neither account for compression overdesign nor provide for use of E-rated lumber. In the present study, the criteria presented in  $(\underline{15})$  were modified to analyze unsymmetric beams by considering the neutral axis shift.

Many changes are necessitated by the neutral axis shift and, initially, its position must be determined from the transformed section (figs. 1a and 1b).

$$z = \frac{\sum_{n=1}^{N} A_n y_n}{\sum_{n=1}^{N} A_n}$$
(1)

where <u>A</u> is the cross-sectional area of the lamination and where <u>y</u> is the distance from the base to the centroid of the different areas. Once the neutral axis is positioned, the beam stiffness can be determined as

$$EI = \sum_{n=1}^{N} E_n I_n$$
(2)

Compression and tension zones of the beam are thus defined and stresses can be determined by independent analysis of each zone as half of a symmetrical beam (figs. 1d and 1e). Stress capabilities at each change in stiffness of the cross section,  $f_n$ , can then be determined following procedures given in appendix 1 of (15).

Changes in notation from (15) are necessary because of the change in method of designating zone depths. For the tension zone,  $n_1$  becomes 2z,  $n_2$  becomes  $2(z - d_1)$ , and  $n_3$ becomes  $2(z - d_1 - d_2)$ . For the compression zone,  $n_1$  becomes 2(d - z) while  $n_2$  becomes  $2(d - z - d_4)$ . Also, for the compression zone the subscript notation must be modified in terms of whether zones 3 or 4 are being considered.

The procedure for checking inner lamination overstress also changes to the following:



Figure 1.—Transformed section showing stress distribution within a beam having four stiffness zones. (M 145 167)

$$F_2 > \frac{(z - d_1)E_2}{z - E_1} f_1$$
 (3)

f

$$f_3 > \frac{(z - d_1 - d_2)E_3}{z - E_1}f_1$$
 (4)

$$f_4 > \frac{(d-z-d_4)E_4}{z}E_1 = f_1$$
 (5)

$$f_5 > \frac{(d-z)E_5}{z}E_1$$
  $f_1$  (6)

If these inequalities are not satisfied,  $f_1$  is limited to a value that will satisfy an equality.

For design use, the bending moment capacity should be determined using easily measured quantities; probably the most feasible are the physical dimensions. The conversion to readily usable stress values is best made by utilizing a transformed section factor  $\underline{I}$ , defined as the ratio of the transformed moment of inertia of the cross section(fig. 1 b) to the actual value (fig. 1 a). For unbalanced combinations,

$$T = \frac{I_1}{I}$$
(7)

Where I<sub>1</sub> is calculated from figure 1b,

 $\underline{T}$  is transformed section factor, and

<u>I</u> is moment of inertia based on actual dimension (fig. la). The stress value,  $\underline{f_1}$  in figure 1c, can be determined as:

$$f_1 = \frac{Mz}{I_1}$$
(8)

where  $\underline{M}$  is a given applied moment. But for design purposes it is useful to define a value of  $\underline{f}$  such that

$$T = \frac{M\frac{d}{2}}{I} = \frac{M}{S}$$
(9)

where <u>d</u> is the beam depth and <u>S</u> is the section modulus. By solving both equations (8) and (9) for <u>M</u> and equating, it can be shown that

$$\mathbf{f} = \frac{\mathbf{f}_1 \mathbf{I}_1 \mathbf{d}}{2\mathbf{z}\mathbf{I}} = \mathbf{f}_1 \left(\frac{\mathbf{T}\mathbf{d}}{2\mathbf{z}}\right) \tag{10}$$

Also, because E1I1 - EI,

$$\mathbf{E} = \mathbf{E}_{1}\mathbf{T} \tag{11}$$

#### E-Rated Lumber

To extend the developed  $I_K/I_G$  concept to E-rated lumber, it was necessary to develop clear wood stress (CWS) values and knot properties for the respective grades. If these values are known, the different E grades can be treated much the same way as species might be for visually graded lumber. Using the beam strength data presented by Johnson (8, 9) and, in addition, supplemental unpublished knot information, the following CWS values were estimated for various grades:

Nominal	Estimated near-minimum
E-grade	ultimate CWS
	Lb/in.2
1.4	4,200
1.6	5,250
1.8	6,300
2.0	7,350
2.2	8.400

These values, if correct and used with appropriate knot data, can be used to predict near-minimum modulus of rupture (MOR) values for the 5-min laboratory test of 12-in.-deep uniformly loaded beams having a 21:1 span-to-depth ratio.

#### **Beam Design and Manufacture**

Eight different beam combinations were designed—four having all visually graded lumber and four having the outer few laminations both E-rated and visually graded while the inner laminations were visually graded only. Six of the eight combinations were 12 in. deep. The other two had lumber with wane in the inner laminations; they were made 18 in. deep to impose high shear stresses on these inner laminations during test. Four combinations utilized E-rated lumber but the compression side laminations all had large knots characteristic of the inner laminations.

The eight combinations were all of nominal 2- by 4-in. lumber, and all were of unbalanced design, having lower grade material on the compression side. Fifteen replicates were evaluated for each beam group. Species and grades of lumber, along with the arrangement of lamina, are illustrated in figure 2 for all groups. Estimated strength values for these beams, calculated using the developed design concept, are presented in table 1 and figure 3. In addition, assumed lumber properties are given in appendix I and detailed information on source and properties of materials in appendix II.

#### Beams of Visually Graded Lumber

Group A beams incorporated four Douglas-fir outer laminations and five Engelmann spruce inner laminations; Engelmann spruce is a near-minimum strength and stiffness species included in the western or white wood group (23,24). These inner laminations were lower grade than commonly used for laminating with western species. Instead of an L3 structural lamination, it was No. 3 structural light framing as deter-



Figure 2.—Diagram showing composition of beam combinations evaluated. (M 145 171)





mined by a grader certified by the Western Wood Products Association and a representative of the American Institute of Timber Construction. The main difference between the L3 and No. 3 was a larger allowable centerline knot and steeper slope-of-grain.

Group B was similar to group A in lumber species. However, the beams were designed as 13 laminations deep instead of 9 so that higher shear stresses could be imposed on the No. 3 inner laminations. These inner laminations were specially selected such that lumber having wane up to one-sixth of the wide face on one or both edges was concentrated in one-half of the length of the inner three or four laminations (fig. 4). The higher shear stress, coupled with the wany lumber, allowed probable design levels in shear and bending to be approached at about the same rate during tests. Any shear weakness in the wany lumber beyond assumed values was expected to become apparent during tests.

<u>Group C</u> also explored the effect of wane on shear strength, but in this case the wany inner laminations were of No. 3 Douglas-fir. This material is stiffer and stronger than the Engelmann spruce of group B, and thus was used more extensively in group C beams to yield predicted ultimate strength nearly equal to beams of group B.

<u>Group D</u> utilized visually graded hem-fir lumber. Johnson (10) demonstrated the potential for developing a high-strength combination using this type of lumber. (Data developed in the present study could increase confidence in the strength of beams built from visually graded hem-fir lumber.) A nine-lamination beam was designed using somewhat conservative assumptions for CWS values in bending. The L1 tension lamination was graded to have a specific gravity of at least 0.39—somewhat above average.

	Design $E^{1/2}$		Million 1b/in. <sup>2</sup>	1.67	1.65	1.69	1.43	1.82	1.67	1.68	1.42	
	Compression	bonus required <sup>2/</sup>		1.10	1.18	1.16	1.23	1.25	1.23	1.34	1.45	
w į	Strength	ratio <sup>2/</sup>		0.685	.722	.720	.647	.717	.720	.812	.624	
ropertie	4/z/d			0.49	87.	.49	67.	67.	.49	.49	67.	
umber p	AC			1.17	1.17	1.16	1.12	1.11	1.11	1.12	1.18	
on as	re 3 <sup>27</sup>	f 7	Lb/in. <sup>2</sup>	1	4,900	1	1	1	1	1	1	
gn based o	n in figur	f 6	Lb/in. <sup>2</sup>	1	3,060	4,740	1	I	I	ł	1	
Deam G SI	ions shown	f 5	Lb/in. <sup>2</sup>	4,580	1,510	2,760	2,960	5,790	5,010	5,510	3,430	
mmary of	at locat	f4	Lb/in. <sup>2</sup>	1,440	920	2,010	1,470	3,630	3,270	2,230	1,300	
le 1Su	get MOR <sup>1/</sup>	f <sub>3</sub>	Lb/in. <sup>2</sup>	1,310	2,770	2,780	1,390	2,380	2,160	2,130	1,190	
Tab	es at tar	f <sub>2</sub>	Lb/in. <sup>2</sup>	3,540	4,060	4,060	2,210	4,220	3,670	4,160	2,530*	
	Stress	fl	Lb/in. <sup>2</sup>	5,070*	5, 340*	5,320*	3,340*	6,020*	5,290*	\$,960*	3,930	
	Target near	minimum MOR <sup>1/</sup>	Lb/in. <sup>2</sup>	4,350	4,580	4,600	2,980	5,410	4,760	5, 340	3, 340	
	roup			A	22	C	D	Е	EL.	9	ж	-

 $\frac{1}{2}$  MOR = modulus of rupture; target moment resistance divided by gross section modulus.

 $\underline{2}^{\prime}$  Critical stress and location denoted by \*.

 $\underline{3}^{\prime}$  Ratio of maximum outer fiber stress to target MOR.

 $\frac{4}{2}$  fraction of the depth, measured from the tension side, to the neutral axis.

 $\frac{5}{2}$  Ratio of beam strength to that consisting of clear material of the type in the outer tension lamination.

 $\frac{6}{10}$  Tension stress assumed to control, but compression bonus shown needed to satisfy this condition.

 $\underline{1}/$  Design modulus of elasticity (E) is 95 pct of the  $\Sigma$  EI  $\div$  I.

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#### Beams of E-Rated Lumber

Four combinations were designed to use E-rated material of four different species groups for the outer laminations. Tension laminations of three different nominal Egrades were used. One requirement of all Erated lumber was that it meet the visual requirements of the L3 laminating grade (or No. 2 for southern pine).

<u>Group E</u> used all Douglas-fir material with six of the nine laminations visually graded L3. The 2.2E outer tension laminations met the visual requirements for 2,700f-2.2E machine stress-rated (MSR) lumber (<u>24</u>) and was referred to as E2.2T. The second tension lamination met the 2,400f-2.0E MSR visual grade requirement (E2.0T). The outer compression lamination—a 2.0E grade—met visual requirements, not of the 2,400f MSR grade but of the L3 grade; it was referred to as E2.0C.

<u>Group F</u> used all southern pine material with six of the nine laminations of No. 2MG (<u>20</u>). The 2.0E outer tension lamination met visual requirements for the 2,400f-2.0E MSR grade (<u>20</u>) and was referred to as E2.0T. The second tension lamination met the 2,100f-1.8E visual grade requirements (E1.8T). The outer compression laminate was a I.8E grade but had visual characteristics permitted in the No. 2 lumber (EI.8C).

<u>Group G</u> closely resembled group F with hem-fir lumber except that two 1.8E compression laminations were used. Inner laminations were L3 hem-fir as used in group D.

Group H was to explore a potential species-independent system for manufacturing glulam beams. The one requirement was that all material should come from a grade and species having an average nominal E of at least 1.0 x 10<sup>6</sup> lb/in.<sup>2</sup>. The outer tension lamination was about 1.8E, had the edge knot restriction of 2;100f-1.8 MSR grade (24), and was referred to as E1.8T. The second tension lamination was about 1.5E and had the edge knot requirements of the 1,650f-1.5E grade (E1.5T). The outer two compression laminations were also 1.5E but had visual characteristics of the L3 grade (E1.5C). Inner laminations were L3. Lodgepole pine lumber comprised the outer two top and bottom laminations for group H. This material has been evaluated (8) and used to a limited extent (2) for laminating, is readily available in the E ranges selected, and is known to have knots representing near maximum size permitted in the grade. Engelmann spruce was used for the inner lamination because an L3 grade has an estimated average E of about the minimum required (1 million lb/in.2).



Figure 4.—A group C Douglas-fir beam 3-1/8 in. by 18 in. by 20 ft, showing full-span yoke deflectometer and lateral supports, also used for 12-in.-deep beams. Beam span was 19 ft with 4 ft between the loading heads. For these 18-in.-deep beams, additional lateral support bracing was added near the top of the shown supports to prevent lateral buckling. Note the wane apparent in the inner No. 3 Douglas-fir laminations.
(M 143 179)

#### **Tension Lamination Criteria**

Previous work has repeatedly indicated a need to assure visual quality for the outer tension lamination. Although visual criteria are not currently required for beams less than 16 in. deep, specific guidelines were used to select a visual tension lamination grade for the various combinations. The indicator of required visual quality was the strength ratio of the tension side of the beam as determined by the  $I_{\rm K}/I_{\rm G}$  concept.

The grades developed closely approximated AITC grades (1) except that the different effect of edge and centerline knots was recognized for all grades. Edge knots were defined as those with associated grain deviations closer than 1/2 in. from the edge. For grading the tension laminations, a 1-ft length was assumed to constitute a cross section. In addition, limitation on low density pieces was applied to the visually graded tension laminations. For the E-rated tension lamination, no large change in properties from one end to the other was allowed.

#### Material preparation

The L3 and No. 3 Douglas-fir and No. 2MG southern pine were finger jointed following grading and the lumber evaluated as 20.3ft-long pieces. All other lumber was processed prior to finger jointing as follows:

1. Each piece of lumber was identified and approximate dimensions and moisture content measured. Weight and E were then determined using an E-computer.

2. The lumber was finger jointed into 20.3-ft-long laminations. For some of the beams, end joints occurred near midlength of the outer tension laminations.

3. Laminations were assigned to beams according to their grade and classification; PS 56-73 was followed for end-joint spacing.

Lumber used as midlength tension laminations was especially selected to have a strength-reducing characteristic near the maximum described in table 2. The requirements for pith-associated wood currently used (1), which limit the amount to one-eighth of the cross section, were applied to the ends of all of the tension laminations. For the visually graded tension laminations of groups A-D, an attempt was made to obtain material of near average density. Douglas-fir pieces (groups A, B, and C) with a low specific gravity (below 0.45) or a high specific gravity (above 0.53) were excluded as not typical of near average material. A similar range for hem-fir (group D) was 0.39 to 0.42.

All group A tension laminations were from L1 material. For group B, four tension laminations (for B-01, -02, -03, and -05) were

Group	Nominal grade		Grade desci	ription	Approximate equivalent AITC visual grade (1)
	8	Limita on l grain	ation based knots and n deviation	Maximum slope-of-grain	
		Edge	Centerline		
		Pct	Pct		
A,D,H	65	43	50	1:12	301-20
B,C,E	70	36	42	1:16	301-22
F	75	30	35	1:16	301-24
G	80	24	28	1:16	301-26

Table 2.--Grading requirements used for tension laminations of the eight combinations

from L2 dense with the remainder from L1. For group C, one tension lamination was L1 (C-04) and all others were L2D. All group D tension laminations were from L1 (dense) visually graded hem-fir.

Tension laminations for groups E and G were to be selected from material that was Erated by the CLT-1 machine. However, shortages developed in both groups so that to obtain desired visual characteristics it was necessary to select material from visually graded lumber using E-computer stiffness values. Thus, two beams in group E (E-24 and E-27) and four beams in group G (G-01, -02, -03, and -05) had tension laminations not processed through the CLT-1 machine but with stiffness values determined by the E-computer.

The two group E tension laminations not CLT-1 graded were from L1 material. Based on E-computer data, these two had stiffness values near to the remaining 13. The four group G tension laminations were from visually graded L1 (dense) hem-fir. These four were intentionally selected with E values lower than the target average E of 2.0 million lb/in.<sup>2</sup> so as to balance the distribution. The four with lower values brought the overall distribution and average close to the minimum target specification.

For groups F and H, tension laminations were selected by E-computer results. Frequency distributions similar to those obtained for lumber used in groups E and G were simulated with the E-computer.

Group F tension laminations were selected from No. 1 and No. 2 material that had been E-rated with an E-computer: Seven beams had No. 1D, another seven had No. 2D, and one had a No. 1MG visual grade. All group H tension laminations were first selected based on static E determined over full span with a weight and dial gage at the lumbermill. Final selection was by E-computer. The resulting E distribution approximated the desired one.

A special attempt was made to obtain at least one E-rated tension lamination each in groups E, F, G, and H which had an E slightly below 200,000 lb/in.<sup>2</sup> less than the nominal E. Conversely, not more than one exceeding the nominal E by this amount was permitted. No tension laminations with E values by Ecomputer that differed from the average by more than 13 pct were used. All lumber for the tension side met or exceeded the same minimum criteria as the outer midlength tension laminations but did not necessarily have near-maximum strengthreducing characteristics. All tension laminations were manufactured such that near-maximum allowable strength-reducing characteristics were placed within 2 ft of midlength. Also, 30 to 40 pct of the beams intentionally had finger joints in the highly stressed midlength region.

#### Beam Manufacture

Except for the differing lumber grades, manufacture followed PS 56-73 (22). Group F beams were manufactured by a commercial laminator who commonly processes southern pine material. All other beams were manufactured by a laminator who commonly processes Douglas-fir.

For group F material, a phenol-resorcinol adhesive was used with a finger joint whose profile was visible on the wide face. For all other material, a melamine-urea adhesive was used with a finger joint whose profile was visible on the narrow face. A common finger joint profile was used: 1.1-in. length, 1/4-in. pitch, 0.030-in. fingertip width.

A major problem developed when processing the No. 3 Engelmann spruce material. Considerable twist and bow in much of this lumber made it incompatible with the automatic finger-jointing equipment. Thus many of the finger joints were not properly alined during mating. Also, end pressure had to be nearly eliminated during curing to prevent buckling of the laminations. As a result, quality of many of the resulting finger joints in the No. 3 Engelmann spruce was questionable upon visual inspection. However, considering the axial and bending stresses to which finger joints would be subjected in the inner laminations of group A and B beams, the relative quality of the material, and the time and machine modifications necessary to correct the problem, it was decided to use the material as initially manufactured. Except for adjustments of end pressure for the other species of wood, the remaining material was processed with few problems.

Following finger jointing, all beams were assembled from the various grades, and the location of each piece of lumber was noted. At this time, the No. 3 and L3 Douglas-fir and the No. 2MG southern pine laminations were Erated, weighed, and their moisture content determined. Knots were also measured in the midlength 10 ft of all laminations.

Before gluing, all laminations were surfaced to a thickness of about 1-3/8 in. The beams were then immediately glued using a phenol-resorcinol adhesive. After removal from clamps, they were surfaced to a 3-1/8-in. width and trimmed to a 20-ft length. They were then shipped to Madison, Wis., and stored indoors for 1 to 2 mo prior to testing.

#### **RESEARCH METHODS**

#### Equipment

Test equipment and procedures conformed to ASTM D 198 (5). All beams were loaded on a 19-ft span with 4 ft between loading heads (fig. 4). Lateral supports at about 4 ft from each end had roller contacts to minimize frictional force. Supports shown in figure 4 were used for the 12-in.-deep beams while additional bracing was added near the top of the 18-in.-deep beams.

To obtain a complete record of the full beam deflection, a yoke deflectometer was developed. Using an electrical transducer, deflection measurements over the 19-ft full span were continuously recorded to failure with no threat of damage to the equipment. Deflection measurements were also made over the 4-ft short span between load points using a different yoke arrangement (fig. 4). An electrical transducer was also used to measure this short span deflection.

A two-channel scanning X-Y recorder was used to record test machine load along with the two deflection measurements.

#### Procedure

After each beam was properly alined in the test equipment, a preload of about 200lb/in? maximum stress was applied to assure proper contact. The X-Y plotting equipment was properly adjusted and loading was continuous to failure. Machine head speeds of about 0.8 in./min were used for the 12-in.deep beams and 0.5 in./min for the 18-in.deep beams. At about 40 to 50 pct of anticipated minimum strength, the equipment used to record deflection on the 4-ft span was removed to prevent possible damage at failure.

Many failures were expected to be sudden and complete. However, because some compression-type failures were also expected, test machine head movement continued until the machine load dropped below about 50 pct maximum.

#### Data Obtained

Dimensions of each beam were determined by measuring the cross-section at each load point and the total length. Each beam was also weighed before test and a photograph taken of the center 6-8 ft of the beam bottom. A continuous record of machine load versus both full-span and short-span deflections was obtained.

During test, audible cracking and visible splitting were recorded. Details of failure were noted and probable initiation points estimated. Moisture content was measured for each lamination near failure with a resistance type meter having 1-1/2-in.-long probes. Each beam was then photographed; if failure appeared to begin in a specific region, this was cut from the beam for further inspection.

#### PRESENTATION OF RESULTS

#### Lumber Properties

Lumber properties are found in appendix Il along with detailed information on individual tension laminations for each group. Average properties of tension laminations are summarized in table 3.

#### Beams

Average properties of each beam group as determined by test are in table 4. Distribution of MOR and E data are shown in figures 5 and 6. Individual test results are in appendix III.

Group	Nominal grade	Specific gravity <sup>2/</sup>	Moisture content 3/	E <sup>4/</sup>	Average reducin	size of strength g characteristics
					Knots	Knots and grain deviation
				Million		
			Pct	$1b/in.^2$	Pct	Pct
A	65	0.50	12	2.21	23	40
В	70	.52	12	2.41	17	35
С	70	.50	11	1.94	24	37
D	65	.39	11	1.63	26	41
E	70	.49	11	2.17	23	40
F	75	.52	15	2.00	13	31
G	80	.42	10	1.99	17	28
Н	65	.44	10	1.78	24	38

## Table 3.--Summary of average properties of midlength tension laminations<sup>1/</sup>

 $\frac{1}{}$  Averages are for the 15 pieces of lumber used.

 $\frac{2}{2}$  Based on weight adjusted to ovendry and volume at time of test.

 $\frac{3}{2}$  Determined with power-loss type moisture meter.

 $\frac{4}{-}$  Modulus of elasticity (E) determined with E-computer.

#### **Test Beam Failures**

Failures were expected in several different categories: Shear; compression; and tension involving finger joints, knots, or slopeof-grain. Shear failures were expected in several group B and C beams; however, only one (B-10) failed in that manner. One beam in group A also failed in shear (A-05).

Compression failures were expected in each group but occurred to a significant extent only in groups D, G, and H. Three beams each in groups G and H and four beams in group D developed compression wrinkles located away from the loading heads which were severe enough to significantly modify the stress distributions. Subsequent loading resulted in a tension mode of failure in all beams but one. For this beam, G-13, the load had decreased to less than 75 pct ultimate when the limit of machine head travel was reached. Although no tension failure occurred, compression wrinkles were apparent at several locations throughout the upper half of the beam (fig. 7). Most beam failures appeared to begin in Table 4.--Summary of average beam properties<sup>1/</sup>

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Group	Dimens	itons <sup>2/</sup>	Moisture	Specific	Modulus	s of rupture	Modulus of	elasticity	Shear	Strength
	Width	Depth	content	gravity <sup>1</sup>	Average	Coefficient <sub>5</sub> / of variation	Full span	Short span	stress at failure	factor 6
	In.	In.	Pct		Lb/in. <sup>2</sup>	Pct	$\frac{M11110n}{1b/1n.^2}$	<u>M11110n</u> 1b/1n. <sup>2</sup>	<u>Lb/1n.<sup>2</sup></u>	
A	3.08	12.39	6	0.46	5,040	24	1.80	1.94	173	1.09
89	3.08	17.89	10	.47	5,560	14	1.78	1.87	276	1.18
U	3.08	17.89	6	67.	5,880	17	1.81	1.82	292	1.20
D	3.07	12.39	6	.37	5,110	12	1.31	1.37	176	1.38
ы	3.08	12.39	11	.52	6,170	16	2.05	2.19	212	1.19
E.	3.14	12.35	11	.50	6,590	17	1.69	1.92	226	1.32
U	3.08	12.38	6	.40	6,210	15	1.70	1.84	214	1.55
н	3.08	12.37	10	.42	5,220	19	1.42	1.56	179	1.24
			11 12							

<sup>1</sup> Averages are for 15 tests.

 $\frac{2}{2}$  Average of measurements made at load points.

 $\frac{3}{2}$  betermined following test using resistance-type meter with 1-1/2-in. needles. Recommended species corrections were applied.

 $\frac{4}{2}$  Based on weight and volume at time of test (weight adjusted to ovendry).

5/ Standard devlation ÷ average.

. /9

Average modulus of rupture ÷ (specific gravity x 10,000).

-13-



Figure 5.—Distribution of beam modulus-of-rupture values, also showing average values (MOR). (M 145 169)





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Figure 7.—Compression failures occured throughout the top one-half of beam G-13 at two locations within the highly stressed region. Large knots in the top and third-from-top laminations were involved near both the 9- and 11-ft locations. Load was applied at the 8- and 12-ft locations. A total of 10 beams developed significant compression wrinkles; the other 9 were generally limited to the top 2 or 3 laminations.

(M 143 194)

the outer tension lamination. Many beams failed through knots and associated grain deviation in the midlength of the outer tension lamination. Forty-seven of the beam failures were attributed directly to strength-reducing characteristics identified prior to test (fig. 8).

Thirty-four beams had finger joints within the most highly stressed (midlength) region of the tension lamination while another 48 had joints where the stress was at least 75 pct maximum. An additional 26 had joints where the stress level was between about 50 and 75 pct maximum. For 26 beams, failure was attributed solely to the finger joint in the tension lamination. Most were in the most highly stressed 4 ft (the midlength); all but two of the others were in the zone with at least 75 pct maximum stress. Two beams—A-09 and B-14—failed through finger joints in the tension lamination where the stress was only about 50 pct maximum. An additional 17 beams failed such that finger joints in combination with other characteristics were involved (fig. 9).

The remaining 18 beams failed through knots or grain deviation in the tension lamination which appeared less severe than the selected characteristics (fig. 10). Beam failure types are summarized as follows:

Туре	Number of beams	Pct
Shear	2	2
Compression	10	8
Tension		
Selected tension lamination	n	
characteristics	47	39
Finger joint alone	26	22
Finger joint and other defe	cts 17	14
Other tension lamination		
characteristics	18	15



Figure 8.—Some of the midlength tension laminations from beams which failed through the selected visual characteristics; 47 of the 120 beams failed in this manner. (M 143 729, M 143 732)





Figure 9.—Some tension laminations from beams which failed through portions of a finger joint and also through either slope of grain or grain deviations nearby; 17 beams failed in this manner. characteristics; 18 beams failed in this manner. (M 143 731)

Figure 10.—Some midlength tension laminations from beams which failed through slope of grain or grain deviations and not at the selected (M 143 730)

#### ANALYSIS OF RESULTS

#### Modulus of Elasticity (E)

Table 5 compares average full span E of each beam group with the design E. Design E was assumed to be 95 pct of predicted E based on a transformed section analysis. For groups A, B, and C—those with visually graded Douglasfir outer laminations—test values exceeded the nominal design goals by 7 or 8 pct. This reflects the 10-15 pct above-average stiffness of the L1 and L2 visually graded Douglas-fir available for the study. For the visually graded hem-fir beams (group D), the average test E was 8 pct below the design goal. This difference can be explained by the lower than anticipated stiffness of the L1 and L2 hem-fir lumber. Three E-rated combinations (groups F, G, and H) had average test E essentially equal to the design goal. This was expected because the outer laminations which contribute most to beam stiffness were selected on the basis of their modulus of elasticity. For group E, the average test value exceeded the design goal by 13 pct. This difference was larger than expected and no explanation could be found for it.

Actual E values for all beams are compared in figure 11 to values predicted using known properties. Predicted E for the 120 beams ranged between 1.25 and 2.31 million lb/in.<sup>2</sup> while test values ranged from 1.20 to 2.26 million lb/in.<sup>2</sup>. A regression analysis suggested a line of best fit as

Group	Design E	Acti	ual test $E^{1/2}$	Actual E ÷ design E
		Average	Coefficient of variation	
	Million	Million		
	$\frac{1b/in.^2}{}$	$1b/in.^2$	Pct	
A	1.67	1.80	6.5	1.08
В	1.65	1.78	9.1	1.08
с	1.69	1.81	5.2	1.07
D	1.43	1.31	4.8	.92
Е	1.82	2.05	6.0	1.13
F	1.67	1.69	4.0	1.01
G	1.68	1.70	2.4	1.01
н	1.42	1.42	4.2	1.00

Table 5.--Comparison of design and actual modulus of elasticity (E) values

 $\frac{1}{}$  Based on 15 replicates in each group.

-18-

$$Y = 0.80X + 0.24 \tag{12}$$

where  $\underline{Y}$  is actual full span E from test (million  $\mathsf{lb}/\mathsf{in}.^2)$  and

 $\underline{X}$  is predicted E (million lb/in.<sup>2</sup>) with a coefficient of determination (R<sup>2</sup>) of 0.96.

The resulting intercept limits the usefulness of the equation. Overall, actual fullspan E averaged 93.3 pct of the predicted values, suggesting an equation of the form

Y = 0.933X (13)

where factors are as described for (12). This compares favorably with previous results (15) and confirms the use of the 0.95 factor proposed.

As expected, short-span E values were generally somewhat larger than full-span values—averaging 7 pct higher. This difference is consistent with differences previously found (<u>15</u>). The short-span E value also had larger variability; the coefficient of variation averaged about twice that of the fullspan E.

#### Modulus of Rupture

Predictability of the MOR can be measured by comparing the target strength with actual ultimate strengths (table 6). At least one beam in five of the eight groups was below the near-minimum predicted (target) strength. In groups A and E, 4 of the 15 beams were below the target values.



Figure 11.—Comparison of actual and predicted full-span modulus of elasticity. Letters denote beam group for each specimen.

(M 145 166)

#### Group A

The low strength beam (A-12) attained only 73 pct of the near-minimum predicted value. The failure of this beam, along with another (A-14) which attained 88 pct of the desired value, seems to have been caused by a finger joint in the tension lamination. Examination of these two finger joints indicated low percentage of wood failure. Improved bonding, especially in the latewood, would undoubtedly have improved the strength of these two beams along with beams A-07, A-09, and A-10.

One beam failed in shear at 79 pct of the design goal due to shake in an inner Engelmann spruce 2 by 4 not detected during grading. Before the beam test, the shake was apparent over the end 8-9 ft. Laminating grades of lumber do not permit this type of defect and the No. 3 grade for use in glulam should prohibit it also.

The fourth beam below desired values appeared to fail due to combined effects of slope-of-grain and a finger joint at 98 pct of the desired value.

Reanalysis of the combination using actual knot data for the lumber (appendix II) revealed that the predicted near-minimum MOR would be 2 pct lower at 4,270 lb/in.<sup>2</sup> (table 7), a value affected by the nominal "65" grade tension lamination which was assumed to limit the outer fiber strength ratio to 0.674. The 4,270 lb/in.<sup>2</sup> value is essentially equal to the fourth lowest strength, leaving three beams significantly below it.

These three beams failed such that, had the strength reducing characteristic in the tension lamination been smaller, there is no reason to believe they would have been stronger. Three other beams failed between the 4,270 lb/in.<sup>2</sup> value as limited by the tension lamination guality and the 4,660 lb/in.<sup>2</sup> value

Group <sup>1/</sup>	Target near	S	trength da	Number of	Average ÷	
		Moisture content	Average MOR	Low strength beam	beams below target MOR	target
	Lb/in. <sup>2</sup>	Pct	Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>		
A	4,350	9	5,040	3,190	4	1.16
В	4,580	10	5,560	4,030	2	1.21
С	4,600	9	5,880	3,500	1	1.28
D	2,980	9	5,110	4,230	0	1.71
Е	5,420	11	6,170	5,120	4	1.14
F	4,760	11	6,590	4,780	0	1.38
G	5,340	9	6,210	3,810	1	1.16
Н	3,340	10	5,220	4,240	0	1.56

Table 6.--Comparison of target and actual beam strengths

 $\frac{1}{}$  Fifteen beams in each group.

as limited by the  $I_{K}/I_{Q}$  analysis; tension lamination defects appeared to be the primary cause. These observations strongly indicate that the strength as limited by the tension lamination quality is close to that assumed.

#### Group B

Two beams failed below the desired strength, one (B-12) at 88 pct and another (B-04) at 97 pct of the desired value. Beam B-12 failed near a finger joint but only about 5 pct of the actual joint was involved. The remainder of the cross-section failed in an unusual manner (fig. 9). Although low wood failure was apparent over the small portion of the finger joint, the primary weakness seemed to be in wood quality near the joint. Beam B-04 failed through a maximum sized tension lamination characteristic.

The reanalysis (table 7) shows that a design MOR about 5 pct higher may have been

attained with a better quality tension lamination. However, the nominal, "70" grade tension lamination limits the overall strength ratio to 0.724, essentially the same ratio as derived from the preliminary design. Thus, no change was found upon reanalysis. Note that the one beam failing at 97 pct of the desired value and another just 2 pct over the desired value both failed at near-maximum sized tension lamination defects. This also indicates that the beam strength as controlled by the tension lamination is as assumed.

#### Group C

One beam, C-14, failed at 76 pct of the desired MOR of 4,600 lb/in.<sup>2</sup> apparently due to a maximum defect in the tension lamination. Reevaluation of this lamination revealed slope-of-grain within the failure region steeper than estimated (fig. 8). Another beam failed at exactly the design value due to a maximum ten-

Group <sup>1/</sup>	Strength ratio by I <sub>K</sub> /I <sub>G</sub>	Compression bonus required	Strength ratio limited by tension lamination quality	$\frac{2}{\frac{Td}{2Z}}$	Target near minimum MOR by reanalysis	Comparison with target MOR from table l
					Lb/in. <sup>2</sup>	Pct
A	0.734	1.41	0.674	0.858	4,270	98
В	.761	1.43	.724	.857	4,590	100
с	.756	1.30	.724	.865	4,630	101
D	. 589	1.20	. 674	.894	2,710	91
E	.686	1.48	.724	.898	5,190	95
F	.752	1.44	.774	.900	4,980	104
G	.771	1.64	.824	.895	5,080	95
н	.640	1.49	.674	. 848	3,420	102

Table 7.--Results of reanalysis using actual knot data

1/ Fifteen beams in each group.

2/ Factor for use in equation (10).

sion lamination defect. The strength of all other beams greatly exceeded the desired value.

A reanalysis with actual data would result in an increased predicted MOR. However, the nominal "70" grade tension lamination limited the design MOR to within 1 pct of the preliminary assumption. Thus, the reanalysis results in no change.

#### Overview of Groups A, B, and C

Overall, these three groups had 7 of 45 beams which failed to meet desired levels. Two of these 7 beams had strengths within 2 or 3 pct of the desired value but the other 5 were more than 10 pct below the desired value.

Two of these five lower strength beams failed through finger joints in the outer tension lamination; improved finger joint quality would undoubtedly have increased their strength. A third beam failed due to shake in an inner lamination. The other two beams failed due to characteristics in the tension laminations. With one, the extent of grain deviation associated with a knot was considerably greater than had been assumed during selection. For the other, wood quality near the finger joint appeared to limit strength but with no visual indication of the reason. These last two beams suggest limitations to visual grading in assessing lumber strength.

Average MOR values for groups A, B, and exceeded the near-minimum predicted С MOR by between 16 and 28 pct (table 6). For more favorable results, this should have been at least 30 and more likely 40 pct. Test results indicate that (a) several of the minimum values are lower than desired, and (b) overall average strength values are low. In the analysis it was assumed that the outer tension laminations were L1 material. Actually, a significant number of the near-minimum quality tension laminates were from material graded L2D by the plant (2 for A, 4 for B, and 14 for C). This was obviously some of the better L2D material; knot analysis indicated it to be typical of the L1, and not the L2.

#### Group D

All beams greatly exceeded the nearminimum design MOR of 2,980 lb/in.<sup>2</sup>. In fact, the lowest strength beam exceeded this value by 42 pct. Reanalysis resulted in yet a lower strength with a predicted minimum MOR of only 91 pct of the 2,980 lb/in.<sup>2</sup> value. The uniformly high strength values suggest that the CWS value used in preliminary design was too conservative. These results are consistent with Johnson's in that CWS values considerably higher than obtained from ASTM procedures appear applicable.

#### Group E

Beams E-04, E-24, and E-27 all failed below the desired level (fig. 8) through maximum defects in the tension lamination. One other (E-05) failed through a finger joint with a high percentage of wood failure. These four all ranged from 94 to 98 pct of the 5,420 lb/in.<sup>2</sup> target value.

Reanalysis resulted in a 4 pct lowering of the predicted near-minimum MOR. Two beams remained below this value. Both failed at tension lamination maximum defects but were within about 1 pct of the adjusted target value (5,190 lb/in.<sup>2</sup>).

#### Group F

No beams of this group were below the desired MOR but two had strength only about 1 pct above this 4,760-lb/in.<sup>2</sup> target value. One of these failed through a finger joint and one through a tension lamination defect.

By reanalysis, predicted minimum strength increased about 5 pct to 4,980 lb/in.<sup>2</sup>. The two beams previously just over the desired value then fell to 96 pct of that value.

#### Group G.

The one beam below the desired value was significantly so at 71 pct of the 5,300*lb/in.*<sup>2</sup> target value. This beam, G-01, failed through a maximum tension lamination defect and along slope-of-grain (fig. 8). It is uncertain why this beam had low strength. However, the midlength tension lamination was one of only four pieces not processed through the CLT-1 machine. The source of these four pieces was also different in that they were grand fir material from Idaho.

Reanalysis resulted in a 5 pct decrease in design MOR for group G but essentially did not change the analysis. The low beam averaged 75 pct of the 5,080-lb/in.<sup>2</sup> desired value.

#### Group H

All beams exceeded the near-minimum desired value of 3,340 lb/in.<sup>2</sup>. The low strength beam greatly exceeded this (by 27 pct), indicating that conservative assumptions were

used in design. Reanalysis increased the nearminimum MOR by only 2 pct.

#### **Overall Comments**

Except for group D, reanalysis using actual knot data had little effect (5 pct or less) on the predicted near-minimum MOR. For group D, the reanalysis decreased the predicted value by 9 pct. Thus, in general, knot data used in the preliminary design were fairly representative of the actual lumber used.

In only two groups, D and H, did the average MOR exceed the near minimum predicted by over 40 pct (table 6). For these two groups, high MOR values suggest that the CWS values used in the analysis are too conservative. In group F, for which the average MOR exceeded the estimated near minimum by 38 pct, the assumed CWS value may be near the desired value. For the other groups, which had Douglas-fir and E-rated hem-fir outer lamination, the CWS value should be reexamined.

#### Effect of Procedure for Selecting Tension Laminations

The original visual lumber grade of the tension lamination had no obvious effect on the strength of beams in groups B, C, and F, where visual grading was used exclusively. For groups E and G, those beams where non-CLT tension laminations were used because of material shortage were among the lower strength ones. Four minimum strength beams in group E had MOR values between 5,100 and 5,300 lb/in.<sup>2</sup>, and the only two beams with non-CLT graded tension laminations were within this range.

The three lowest strength beams in group G were represented by non-CLT graded tension laminations; the other beam with non-CLT material was above average. Similar comparisons within groups F and H are not possible because no tension laminations were CLT graded—all were selected either by Ecomputer (group F) or by dual criteria of static load and E-computer (group H).

This apparent difference in lumber quality with method of E-rating warrants closer examination. It is partially explained for group G because the E values (vibration) of the four non-CLT graded tension laminations were considerably lower than the others—intentionally so to simulate a 2.0E grade.

However, the same explanation cannot be

applied to group E because tension laminations selected by E-computer had near average modulus of elasticity values. Also, in groups F and H the modulus of elasticity of the tension lamination did not seem related to beam strength.

Differences between the E-computer and the CLT-1 machine might be examined to explain low strength beams in groups E and G. However, all group F and H tension laminations were selected by the E-computer and all met or exceeded the desired strength values.

#### Finger Joint Quality

Several beams failed through finger joints in the tension lamination with a low percentage of wood failure. Some finger joints, however, exhibited high percentages of wood failure and probably developed the full potential strength of the finger joint design. Many of the finger joint failures showed excellent bonding in the earlywood with poor bonding in the latewood, a condition not restricted to any species or growth rate. The joint strength, and consequently the beam strength, appeared to be limited by the amount and strength of earlywood present.

If higher strength joints in tension laminations are desired, improved joint design, better adhesive systems, and improved quality control techniques all appear to offer potential.

#### Comparison of Strength-Weight Factors

The several different species and two grading methods used provided an opportunity to determine the relative strength-weight efficiencies of the different beam groups. This factor has little to do with design and probably relates most closely to ease of handling and shipping.

The ratio of the average MOR to specific gravity was divided by 10<sup>5</sup> to yield factors near 1 for the eight groups (last column, table 4). Higher factors denote "more strength per lb of material."

All factors for the Douglas-fir groups (A, B, C, and E) were between 1.09 and 1.20, a range lower than for the other four groups. Group G, made using E-rated hem-fir, had the highest factor: 1.55. Group D, which also contained hem-fir, but all visually graded, had a somewhat lower factor of 1.38. The southern pine (group F) and white wood (group H) fell between the hem-fir and Douglas-fir; values were 1.32 for group F and 1.24 for group H.

Although many other factors influence material selection, these data indicate that hem-fir lumber provides the greatest return in average bending strength per lb of material used in manufacture.

#### Influence of Wane

Groups B and C were designed to have a higher probability of horizontal shear failure than the other groups. However, only one of these 30 beams was believed to have failed in shear, i.e, beam B-10 at a calculated shear stress of 280 lb/in.<sup>2</sup>. This was near the average calculated shear stress for the other 29 beams which failed in bending.

Design goals (about 1/2 target nearminimum value) for group B were 90 and for group C 110 lb/in.<sup>2</sup> in horizontal shear. These represent two-thirds of the nominal design values for lumber without wane or splits. All 30 beams can be analyzed by dividing the calculated horizontal shear stress at failure by the design goals and examining the ratios. Individual ratios varied from 1.58 to 3.70 with an average of 2.86. An estimated fifth percentile would be about 2.1.

It is extremely difficult to arrive at any conclusions on shear strength of the beams with wany lumber in the interior laminations because 29 of the 30 failed such that this lumber did not appear to influence failure. What is significant is that no large shear weaknesses were apparent due to the wany lumber. Thus, the one-third reduction in design strength due to wane amounting to one-sixth of the width on either or both edges appears to be adequate.

#### Compression Bonus

One purpose of this study was to determine whether the grade of lumber on the compression side of glulam beams could be significantly reduced without developing compression failure. The amount by which nominal design stress on the tension side exceeded that on the compression side was denoted by the ratio of these two stress values and called a "compression bonus."

No compression failures were apparent in any of the Douglas-fir or southern pine groups. Discounting slight wrinkles in the top lamination at the loading head, only three groups had beams which developed obvious compression failures: Three beams each in groups G and H and four beams in group D developed compression wrinkles to the extent as to likely change the stress distribution in the cross section.

The required compression bonus in the different groups to maintain a tension mode of failure is shown in table 8. Based on assumed properties, a compression bonus of at least 1.2 (group D) was required to induce a small incidence; a factor of 1.5 (group H) did not create any predominance of compression failure. Using actual knot data for the beam analysis, compression bonus factors of 1.3-1.48 did not induce any failure in the Douglas-fir or southern pine. The test definitely indicates that a compression bonus of some value above unity is justified.

Although the results do not suggest any specific value of compression bonus for design of glulam beams, a factor of at least 1.3 appears to be justified. If 1.3 were used with the design concept proposed, low incidence of compression failure would probably result in any beams tested. A 1.4 or 1.5 factor may introduce more compression failures and lower average strength but yet is not likely to result in any beam strengths low enough to affect design levels.

#### Analysis of CWS Values

Many CWS values in bending for the different grades and species were assumed in order to design the various groups. The CWS values used were estimated near-minimum ultimate bending stresses for 12-in.-deep clear wood glulam beams consisting entirely of the described material (appendix I).

Results of this study along with results of many other recent large glulam beam tests provided an opportunity to analyze these estimated values. Details of the analysis are given in appendix IV and only the general trends summarized in table 9 will be discussed here.

For dense, visually graded Douglas-fir, the data indicate that when all grading and manufacturing variables in the research beams are considered, the CWS value at the 5th percentile should be less than the 7,390lb/in.<sup>2</sup> value assumed. At 75 and 90 pct levels of confidence, values ars 6,460 and 6,310

Group <sup>1</sup> /	Compression bonus	factor required	Compression	n-side failures
	Assumed knot data	Actual knot data	Number	Frequency
				Pct
Α	1.10	1.41		0
В	1.18	1.43		0
С	1.16	1.30		0
D	1.23	1.20	4	27
Е	1.25	1.48		0
F	1.23	1.44		0
G	1.34	1.64	3	20
Н	1.53	1.49	3	20
TOTAL			10	8

#### Table 8.--Compression bonus factor

1/ Fifteen beams in each group.

Ib/in.<sup>2</sup>. Dense hem-fir resulted in quite similar values. This type of analysis does not account for the selection of near-minimum quality tension lamination in the test samples. Thus, the true value of the lower percentile for a representative population is probably less than the 5th; further analysis is necessary to determine its absolute value.

For E-rated material, the data do not show justification for any difference in CWS values for the 2.0E through the 2.6E grade. CWS values of 7,000 and 6,800 lb/in.<sup>2</sup> for 75 and 90 pct levels of confidence could probably be justified for this range of E-grade. These are somewhat higher than for dense visually graded Douglas-fir.

Both the 1.6E and 1.8E grades have somewhat lower CWS values. Values of 5,600 and 5,200 lb/in<sup>2</sup> are suggested for the 1.8E grade for the two levels of confidence but data are insufficient to develop a recommendation for the 1.6E grade. Some of the beams with Erated laminations also included near-minimum quality tension laminations. As with the visually graded material, the exact effect of this technique on the statistical analysis is yet to be determined.

Description of	Number of	Clear wood bending stress							
material <sup>17</sup>	beams	Mean	Standard deviation	Estimated fifth percentile (tolerance limit)					
				75% confidence	90% confidence				
		Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>				
		VISUALLY G	RADED MATERIA	AL.					
Dense Douglas-fir	88	8,760	1,300	6,460	6,310				
Dense southern pine	28	10,790	1,860	7,280	6,870				
Dense hem-fir	30	8,980	1,290	6,570	6,300				
		E-RATE	MATERIAL						
2.6 + 2.4	21	10,520	1,720	7,220	6,760				
2.2	31	10,030	1,760	6,760	6,400				
2.0	95	9,880	1,630	7,010	6,840				
1.8	35	9,540	2,120	5,620	5,220				
1.6	6	7,850	890	5,780	5,110				

#### Table 9.--Results of clear wood bending stress analysis

 $\underline{1}/$  Many of the beams were intended to be near minimum quality in that near minimum quality critical tension laminations were especially selected.

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and the second se

#### CONCLUSIONS

In considering the absolute values of beam strengths from this study, bear in mind that samples were selected to represent the near minimum quality for each grade. This was done by selecting near minimum quality outer tension laminations for the most highly stressed region.

Specific conclusions are:

(a) Design of unbalanced glulam beams using a "compression bonus" of 1.3 appears justified. The resulting reduced grade on the compression side will probably result in only a slight increase in the number of test beam failures in compression. A higher factor of 1.4 or 1.5 will introduce more compression failure in test beams; although a higher factor could lower the average bending strength, it would probably not affect the near-minimum strength.

(b) The procedure used to design visually graded glulam beams can be extended to beams with E-rated outer faminations. Clear wood stress values associated with 2.0E material are somewhat higher than for dense visually graded Douglas-fir. Data presented provide information on which to base the design of beams using E-rated outer laminations.

(c) Shear weaknesses larger than assumed were not apparent in beams made with lumber having wane occupying up to onesixth of the width at either edge. Under dry conditions, design of such beams to a level of shear stress equal to two-thirds of that with wane-free lumber would appear justified. However, such lumber must be graded following the rules regarding splits and shakes.

Generally, results confirm previous findings in that the performance of some lower strength beams appeared to be limited by finger joint quality. Other beams below the target strengths indicated the importance of carefully grading tension laminations for the amount of grain deviation and the lower grade inner laminations for shake, which is not now permitted in laminating grades. Examination of tension laminations following failure suggested that, when questions existed regarding the amount of grain deviation, it was usually greater than assumed.

#### **APPENDIX I**

Appendix	I Assumed	lumber	properties	for	designing
	glula	n combin	nations		

Species group	Grade	$E^{\frac{1}{2}}$	Knot	$data^{2/}$	Clear wood
and grading method			x	h	bending stres
		Million			
		1b/in. <sup>2</sup>	Pct	Pct	Lb/in. <sup>2</sup>
Douglas-fir 3/					
Visually graded	Ll	2.1	6.9	32.4	7,390
	L2D	1.9	10.3	38.1	7,390
	L2	1.8	10.3	38.1	6,350
	L3	1.6	11.6	46.4	6,350
4.1	No. 3	1.5	15	50	5,040
E-rated-	E2.2T	2.2	5.2	32.6	8,400
	E2.OT	2.0	8.1	37.4	7,350
	E2.0C	2.0	10	36.9	7,350
Southern pine 5/					
Visually graded	No. 2MG	1.5	7.6	43.3	6,350
E-rated4/	E2.0T	2.0	3.6	32.9	7,350
	E1.8T	1.8	7.6	43.3	7,350
	E1.8C	1.8	7.6	43.3	6,300
Hem-fir 6/					
Visually graded	LLD	1.71	6.6	38.2	5,150
	L2	1.47	10.2	46.9	4,480
41	L3	1.30	12.2	53.5	4,480
E-rated-	E2.0T	2.0	3.2	26.2	7,350
	E1.8T	1.8	3.7	27.4	6,300
	E1.8C	1.8	7.2	39.2	6,300
Engelmann spruce7/					
Visually graded	L3	1.0	20	55	3,290
	No. 3	1.0	23	55.8	3,290
Lodgepole pine4/					
E-rated	E1.8T	1.8	6	34	6,300
	E1.5T	1.5	10	40	4,720
	E1.5C	1.5	15	45	4.720

1/ E = modulus of elasticity.

 $2/\overline{X}$  is average sum of knot sizes and <u>h</u> is the difference between the estimated near maximum and average sum of knot sizes.

3/ For L1, L2D, L2, L3 grades, E data are from AITC 117-74 (1), knot data from an industry-wide survey, and clear wood bending stress data obtained by multiplying values in USDA Bull. No. 1069 by 2.1 following adjustment to 10-year loading. For No. 3 grade, E from National Design Specification for lumber at 15 pct maximum moisture content, knot data assumed to be slightly larger than for L3, and clear wood bending stress assumed to be 80 pct of medium grain to account for possible occurrence of occasional wide-ringed material.

 $\frac{4}{8}$  Knot and clear wood bending stress data based on analysis of unpublished data supplied by Johnson for beams reported in  $(\underline{8},\underline{9})$ .

5/ E is based on information collected in several previous studies on No. 2 medium grain lumber, knot data from partial results of an industry-wide survey, and clear wood bending stress obtained from USDA Bull. No. 1069 as described in footnote 3.

6/ E data from ASTM D 245 procedure and assuming a 5 pct increase if specific gravity greater than 0.39, knot data supplied by AITC, and clear wood bending stress data based on a 5 pct lower exclusion limit from ASTM D 2535 data adjusted to 12-inch deep, uniformly loaded condition. A 17 pct increase in bending stress was then assumed to apply to "dense" material having a specific gravity greater than 0.39.

7/ Same as footnote 6 except that knot data estimated.

#### APPENDIX II

#### Source and Properties of Lumber

#### Material Source

#### Douglas-fir

All of the visually graded Douglas-fir material was from stock available at a glulam manufacturing plant in the northern California area. It was graded following kiln drying at the plant using normal plant procedures according to the laminating grades (23,24). The exception was the No. 3 material used for the inner laminations of group C. This material was graded as L3 except the following characteristics were permitted:

(a) Met No. 3 requirement for knots and slope-of-grain for Structural Light Framing except that knot holes were the same size as knots. All material meeting No. 2 Structural Light Framing was removed to assure the material would be representative of No. 3.

(b) Wane was permitted along either edge to a maximum of one-sixth of the wide face. This resulted in a central portion of the lumber equal to two-thirds the total width, which provided a continuous glue bond.

(c) White speck or a combination of white speck and a knot did not occupy more than one-half of the cross section.

Hereafter, this shall be referred to as No. 3, but bear in mind that it was subject to the above grading criteria. For all grades and species, no material meeting a higher grade was permitted in that grade for the test beams.

The E-rated Douglas-fir came from the Willamette Valley region of Oregon and was graded by a CLT-1 machine. The 2.2E Douglas-fir was machine graded with the machine set to select all material meeting or exceeding 2.2E. This lumber was sorted a second time with the machine set to select 2.4E material, which was then removed and not used for this study. For the material to be used on the tension side of beams, the edge knot requirement (one-fourth for 1650f-1.5E and onesixth for higher grades) was imposed. Material not meeting the edge knot requirement but meeting L3 grade was used on the compression side.

The goal for describing the E-grades used in the test beams was as follows:

1. Average E of the grade as determined by the E-computer was to be at least 95 pct of the nominal value, i.e., the 2.0E material averaged at least 1.9 million lb/in.<sup>2</sup>.

2. The E of at least 90 pct of the material exceeded the nominal grade less 200,000 lb/in.<sup>2</sup>, i.e., 1.8 million lb/in.<sup>2</sup> for a 2.0E material.

#### Hem-fir

The visually graded hem-fir material was obtained in the vicinity of Boise, Idaho, and was graded by a representative of the American Institute of Timber Construction to meet the laminating grade requirements (23,24). It was selected at a lumber mill and its maximum moisture content checked during grading. Many pieces were found to exceed the maximum 16 pct moisture content desired and were therefore discarded. This sorting operation may have eliminated some of the heavier material which dried more slowly, thus biasing the sample toward the lighter and perhaps less stiff material. The degree, if any, to which this selection procedure affected the properties of the sample was impossible to determine. E-rated hem-fir was also obtained in Oregon using the same equipment and general principles used to obtain the Douglasfir E-rated lumber.

Material used in beams D-03 and D-04 was anatomically identified as belonging to the white fir (Abies) group and was probably either grand fir or California red fir. Knowing it was purchased in the Boise, Idaho area suggested that it was grand fir (Abies grandis).

Midlength tension laminations for beams G-01, -02, -03, -05, and -10 were identified as belonging to the fir group. All of these five except G-10 were probably grand fir because they were purchased in Idaho; G-10, along with the second lamination of G-02, may have been white fir (Abies Concolor) from Oregon. Material used for the outer two tension laminations in beam G-11 was identified as belonging to the hemlock (Tsuga) group and, as it was purchased in Oregon, it was probably western hemlock (<u>Tsuga heterophylla</u>). The other tension laminations were also identified as hemlock (<u>Tsuga</u>) and were likely western hemlock.

#### Southern pine

All of the southern pine began as a mixture of No. 1, No. 1D, No. 2D, and No. 2MG material at a glulam plant as graded according to southern pine rules (19). Its origin was unknown. From one group, enough No. 2MG was sorted for the inner lamination of group F beams. E-rated material was obtained from a different group by use of E-computer. As southern pine was not readily available in Egrades or no commercial equipment could be found in a location convenient to the southern pine lumber industry, the E-grades for the outer lamination were selected using the Ecomputer. The MSR Douglas-fir and hem-fir lumber grades had been previously evaluated by the E-computer and their distribution form was used as a guideline.

#### Western woods

Visually graded Engelmann spruce was obtained from Colorado for the inner lamination of groups A, B, and H beams. L3 material for group H was graded according to the rules for laminating lumber (22,23). No. 3 material was graded as previously described for Douglas-fir. Both L3 and No. 3 were obtained from Standard and Utility grade light framing material.

The tension laminations of beams H-02 and H-11 were anatomically identified as belonging to the pine (Pinus) group and were probably lodgepole (Pinus contorta). The wood's known source supports this. The inner laminations were identified as belonging to the spruce (Picea) group and it was purchased in Colorado as Engelmann spruce (Picea engelmannii).

E-rated material was obtained from lodgepole pine in central Oregon. As with the southern pine, it was necessary to simulate the E-grades. Rather than an E-computer, midspan deflections under a known weight were used as the criteria for selection. Edge knot criteria were followed, i.e., material for the tension side of beams had a maximum of onefourth edge knot, while compression side material was permitted up to 50 pct knot as in L3. The lodgepole pine was selected from construction and standard light framing grade lumber.

#### LUMBER PROPERTIES

Properties of the various grades of lumber are summarized in table II-1, and midlength tension lamination data are given in table II-2.

#### Visual Grades

Visually graded Douglas-fir lumber was 10-25 pct stiffer than assumed while the visually graded hem-fir was 5-15 pct lower in stiffness than assumed. L2 and L3 grade Douglasfir had smaller average knot size but larger near maximum knot sizes than assumed—the net effect probably being to cancel one another in their effect upon beam design. L1 hem-fir had a larger knot size than assumed which can be attributed to the specific selection of relatively low quality L1 pieces for tension lamination. L2 and L3 hem-fir had knot properties close to those assumed.

No. 2MG southern pine lumber had properties close to those assumed, while Engelmann spruce was slightly stiffer and had slightly smaller knot size.

#### E-Rated Grades

As expected, the E of all the E-rated grades was close to that assumed. Knot properties of both Douglas-fir and hem-fir were quite near those assumed for the material used on the tension side but the compression side material had a considerably higher near maximum knot size. It would appear that this material should be assumed to have knot properties similar to L3 material of the same species group.

The outer tension lamination southern pine material had a smaller near maximum sized knot than assumed but the other mate ial was quite close. As with the Douglas-fir and hem-fir, the E-rated compression side material should probably be considered to have knot properties similar to the inner laminations (No.

#### 2MG).

The outer tension lamination grade of the lodgepole pine had knot properties larger than assumed, probably due to the specific selection of low quality pieces. The other grade used as a second lamination was close to that assumed while the outer compression lamination had larger knots. As with the other species groups, the lodgepole pine pieces probably should be assumed to be similar to an L3 grade of the same species (15).

#### **Tension Laminations**

All tension laminations had significant

strength reducing characteristics within the constant moment section (table II-2). Up to 0.3 feet may have been sawn from the ends of the beams during manufacture so that the location of characteristics might not correspond exactly with failure descriptions given in appendix III.

All measurements of knot size and amount of grain deviation were conducted prior to assembly of beams. It was obvious after the tests that the amount of grain deviation and slope-of-grain had been underestimated in several instances.

Species group	Crade	Number of	Moisture	Specific	Modulus	of elasticity	Knot da	nta <sup>1/</sup>	
and grading method		pieces	content	gravity	Mean	Coefficient of variation	Lineal feet of 2 x 4 lumber measured	x	h
					Million				
			Pct		1b/in. <sup>2</sup>	Pct	Ft	Pct	Pct
Douglas-fir									
Visually graded	L1	134	12	0.53	2.33	20.1	450	4.3	30.7
, ,	L2D	138	12	.51	2.17	21.4	450	6.0	42.4
	L2	90	10	. 49	2.02	18.6	900	5.4	55.3
	L3	237	12	. 51	2.02	18.6	2,400	8.2	58.9
	No. 3	120	10	. 47	1.84	19.7			
E-rated	E2.2T	54	11	. 48	2.24	4.0	300	6.3	36.5
	F2.0T	52	11	.47	2.08	3.9	300	8.4	36.6
	E2.0C	63	11	. 51	2.08	5.9	300	9.7	58.8
Southern pine									
Visually graded	No. 2MG	180			1.53	17.8	900	8.6	49.1
E-rated	E2.OT	57			2.01	5.7	300	2.4	19.4
	E1.8T	56			1.79	5.7	300	5.6	41.5
	E1.8C	52			1.79	5.8	300	9.4	49.1
Hem-fir									
Visually graded	L1D	77	10	. 39	1.62	12.5	150	11.2	38.1
	L2	92	9	. 36	1.27	14.6	450	11.1	50.2
	1.3	273	9	. 36	1.20	14.8	1,500	13.0	54.1
E-rated	E2.0T	48	10	.45	2.22	7.8	150	5.4	23.9
	E1.8T	35	9	. 41	1.88	3.6	150	5.6	35.4
	E1.8C	59	11	.43	1,88	3.8	300	9.9	52.7
Engelmann spruce									
Visually graded	L3	163	13	. 40	1.20	14.9	750	16.0	45.4
	No. 3	342	11	. 39	1.22	16.8			
Lodgepole pine									
E-rated	E1.8T	45	10	. 46	1.80	6.6	150	11.4	34.
	E1.5T	36	10	. 43	1.49	7.2	150	11.8	35.
	E1.5C	58	10	. 45	1,49	8.2	300	17.3	54.1

Table II-1.--Properties of lumber grades used in beam manufacture

 $1/\overline{X}$  is average sum of knot sizes and <u>h</u> is the difference between the estimated near maximum and average sum of knot sizes.

Room		Lumbe	er data		(	Critical	knot	
No.	Length	Specific gravity <sup>1/</sup>	Moisture content <sup>2/</sup>	E <sup>3/</sup>	Location <sup>4/</sup>	Knot type <sup>5/</sup>	Knot size	Grain devia- tion
				Million				
	Ft		Pct	1b/in.2	<u>Ft</u>		Pct	Pct
			GROUP A	-DOUGLAS-F	IR			
A01	12	0.47	11	2.25	10.1-10.3	Ed	23	45
A02	12	.52	14	2.37	9.6	Ed	23	41
A03	10	.52	13	2.18	10.5	С	27	52
A04	14	.54	12	2.49	10.3	Ed	23	43
A05	14	. 44	10	1.88	11.1	Ed	20	36
A06	8	.48	9	1.82	10.2	С	21	41
A07	16	.49	12	1.84	10.3	Ed	21	43
A08	16	.53	12	1.97	10.3	Ed	23	39
A09	14	.53	13	2.49	10.1	Ed	25	41
A10	14	.51	13	2.49	10.2	Ed	20	39
A11	16	. 51	12	2.27	9.7	Ed	20	34
A12	10	. 58	13	2.76	10.6	Ed	29	41
A13	10	. 52	12	2.46	11.6	Ed	21	38
A14	14	.46	10	2.12	11.3	Fd	27	38
A15	10	.48	11	1.80	9.3	Ed	23	32
Avera	ige	.51	12	2.21		C-2	24	46
						Ed-13	23	39
			GROUP B	-DOUGLAS-F	FIR			
B01	8	0.54	11	2.74	10.0	Ed	21	32
B02	14	.50	12	2.23	11.0	Ed	14	32
B03	14	.48	10	2.10	8.6	С	25	46
B04	14	.52	13	1.81	10.1	Ed	11	36
B05	16	.48	11	1.86	11.1	С	25	46
B06	16	.51	11	2.49	9.8	Ed	18	34
B07	14	. 54	13	2.60	10.2	Ed	20	34
B08	14	. 55	15	2.68	10.2	Ed	16	36
B09	16	.54	12	2.15	10.3	Ed	7	34
B10	14	.47	10	2.04	10.1	Ed	18	30
B11	14	.53	12	2.87	9.7	Ed	16	30
B12	14	. 54	13	1.52	11.2	Ed	18	29
B13	14	.48	11	1.62	10.3	Ed	0	34
B14	16	. 56	13	2.93	9.8	Ed	25	36
B15	10	.56	15	2.23	10.2	Ed	18	32
Avera	ige	.52	12	2.26		C-2	25	46
						Ed-13	16	33

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D		Lumbe	er data		(	Critical	knot	
Beam No.	Length	Specific gravity <sup>1/</sup>	Moisture content <sup>2/</sup>	E <u>3/</u>	Location <sup>4/</sup>	Knot type <sup>5/</sup>	Knot size	Grain devia- tion
				Million				
	Ft		Pct	$\frac{1b}{in.^2}$	Ft		Pct	Pct
			GROUP C	DOUGLAS-F	IR			
C01	14	0.52	10	2.02	9.8	Ed	23	30
C02	16	.50	11	1.92	9.0-9.2	Ed	20	43
C03	14	.51	13	2.10	9.1	Ed	25	36
C04	14	.49	11	1.94	10.0	Ed	29	39
C05	16	. 52	13	1.72	10.4	Ed	29	39
C06	8	.46	9	1.87	9.6	C	29	43
C07	16	.53	11	2.43	9.2	Ed	20	34
C08	14	.50	11	1.88	10.5	Ed	20	41
C09	8	.48	10	1.71	10.6	С	25	43
C10	16	.52	11	2.08	10.8	Ed	20	30
C11	16	.49	10	2.30	9.0	Ed	16	27
C12	8	.49	10	1.87	9.4	Ed	20	34
C13	16	.50	13	2.10	9.2	Ed	21	34
C14	16	.47	12	1.74	10.2	Ed	29	41
C15	14	.50	10	1.49	9.9-10.2	Ed	21	38
Avera	ige	.50	11	1.94		C-2	27	43
						Ed-13	23	36
			GROUP I	DHEM-FIF	2			
D01	16	0.40	18	1.65	10.1	С	25	38
D02	16	. 39	13	1.83	9.5	С	21	36
D03	16	. 39	12	1.79	10.2	Ed	27	41
D04	14	.37	14	1.43	10.0	С	27	41
D05	16	.40	13	1.93	10.0	Ed	29	43
D06	16	. 38	13	1.66	9.9	С	34	50
D07	16	. 39	13	1.41	10.2	Ed	27	41
D08	16	. 39	12	1.78	10.6	С	34	48
D09	16	. 39	15	1.60	9.6-10.0	Ed	21	38
D10	16	. 39	13	1.39	11.0	С	21	32
D11	16	. 39	14	1.49	10.5	Ed	27	39
D12	16	. 37	14	1.59	9.8	С	27	36
D13	16	.41	15	1.78	10.1	С	29	45
D14	16	.40	12	1.46	10.0	Ed	16	38
D15	16	.40	14	1.66	10.2	С	27	45
Avera	ige	. 39	14	1.63		C-9	27	41
						Ed-6	24	40

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Design		Lumbe	er data		C	Critical	knot	
Beam No.	Length	Specific gravity $\frac{1}{}$	Moisture content <sup>2/</sup>	E <u>3/</u>	Location4/	Knot type <sup>5/</sup>	Knot size	Grain devia- tion
				Million				
	Ft		Pct	$1b/in.^2$	<u>Ft</u>		Pct	Pct
			GROUP E	-DOUGLAS-F	IR			
E04	12	0.49	11	2.13	11.6	С	27	45
E05	12	. 51	11	2.32	10.6-10.9	C	23	43
E08	12	.49	11	2.32	10.3	C	30	43
E09	12	.45	10	2.12	9,1-9,3	c	36	43
E10	12	.48	12	2.25	11.1	C	27	46
E12	12	. 51	10	2.25	10.3	Ed	27	41
E15	12	. 50	13	2.37	9.6	Ed	11	29
E16	12	.45	10	2.23	9.9	C	16	39
E19	12	.46	10	2.34	11.4	Ed	21	30
E21	12	.44	9	2.16	9.2	Ed	25	46
E24	14	.51	13	2.18	11.4	C	32	46
E26	12	.44	9	1.95	10.3	Ed	18	27
E27	16	. 52	11	2.16	9.2	Ed	25	38
E28	12	.51	11	2.32	9.9	Ed	18	39
E29	12	.47	11	2.25	10.8	Ed	14	39
Avera	ge	.48	11	2.22		C-7	27	44
						Ed-8	20	36
			GROUP FS	SOUTHERN P	INE			
F02	14			1.77	9.8	С	11	32
F03	14			2.02	10.2	Ed	14	27
F04	14			1.81	9.3	С	12	36
F07	14			2.01	11.8	С	9	27
F08	14			2.16	9.8	Ed	9	27
F11	14			2.04	8.7	Ed	21	29
F12	14			2.20	9.5	Ed	14	29
F13	14			1.81	9.8	С	12	29
F17	14			2.12	10.1	С	12	34
F20	16			1.82	10.0	С	14	39
F21	14			1.97	10.2	Ed	18	32
F22	16			2.14	8.2	С	12	29
F23	14			2.03	8.9	Ed	11	34
F27	16			2.01	10.0	Ed	11	27
F28	14			2.18	10.0	С	21	38
Avera	ge			2.01		C-8 Ed-7	13 14	33 29

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Poar		Lumbe	er data		(	Critical	knot	
No.	Length	Specific gravity <sup>1/</sup>	Moisture content <sup>2/</sup>	E <sup>3/</sup>	Location4/	Knot type <sup>5/</sup>	Knot size	Grain devia- tion
				Million				
	Ft		Pct	$1b/in.^2$	Ft		Pct	Pct
			GROUP (	GHEM-FIR				
G01	16	0.40	10	1.74	9.8	Ed	18	25
G02	16	. 38	12	1.86	10.0	С	12	22
G03	16	.37	10	1.76	10.0	С	27	34
G04	14	.41	12	2.10	10.0	С	16	27
G05	16	. 38	13	1.80	10.0	Ed	18	27
G06	14	.39	12	2.16	10.1	Ed	18	25
G07	12	.42	13	2.03	10.0	С	18	30
G08	12	.42	10	2.04	10.0	Ed	11	27
G09	12	.47	14	2.24	8.1	Ed	18	32
G10	12	.41	12	2.11	10.0	C	20	32
G11	14	.41	9	1.97	10.0	Ed	16	29
G12	14	.45	12	2.19	10.1	С	18	30
G13	12	.45	14	1.91	10.1	Ed	16	27
G14	14	.45	12	2.09	10.2	Ed	18	32
G15	14	.40	13	1.99	10.3	Ed	16	29
Avera	age	.41	12	2.00		C-6 Ed-9	18 17	29 28
			GROUP H	LODGEPOLE	PINE	Du y		20
H01	12	0.47	12	1.71	9.8-10.2	C	25	38
H02	12	.42	12	1.87	10.1-10.3	Ed	30	46
HO3	14	. 44	11	1.83	10.1	Ed	25	36
H04	16	•41	11	1.88	10.1	C	23	38
H05	14	.49	11	1.92	10.1	C	14	38
107	16	.43	11	1.72	10.0	C	21	32
H07	14	.42	11	1.80	0.0	Ed	23	39
100	14	.44	11	1.39	9 9-10 0	Ed	23	32
110	14	.45	12	1.90	10.0	Ed	25	40
u11	16	.44	11	1.76	11. 2-11 7	C	21	30
u12	16	.43	11	1.78	11.0	C	21	41
H12	10	.43	11	1.70	11.0	U	21	41

		Lumbe	r data		C	Critical	knot		
Beam No.	Length	Specific gravity $\frac{1}{}$	Moisture content <sup>2/</sup>	E <sup>3/</sup>	Location 4/	Knot type <sup>5/</sup>	Knot size	Grain devia- tion	
				Million					
	Ft		Pct	$1b/in.^2$	<u>Ft</u>		Pct	Pct	
		GRC	UP HLODGE	POLE PINE	cont.				
H13	14	0.44	11	1.65	10.0	С	29	43	
H14	14	.41	10	1.62	10.7-11.0	Ed	30	45	
H15	14	.49	11	2.03	11.0	Ed	20	34	
Avera	ge	.44	11	1.78		C-8 Ed-7	22 27	37 40	

1/ Based on ovendry weight and volume at time of test.

2/ Average of three values taken with a surface-type meter.

3/ Modulus of elasticity determined with an E-computer.

4/ Location of defect in beam measured from reference end of beam.

5/ Edge (Ed) or centerline (C).

## APPENDIX III

### Beam Test Results

Table III-1. -- Results of bending tests

Beam	Dimens	ion91/	Moisture	Specific	Modulus	Modul	us of	Shear		Failure comment	<mark>s<sup>4/</sup></mark>
NO.	Width	Depth	content <sup>2/</sup>	gravity3/	rupture	Full	Con-	at failure	Selected	Tension	Other
						span	stant moment section		lamination knot	finger joint	
						Million	Million				
	In.	In.	Pct		Lb/in. <sup>2</sup>	1b/in. <sup>2</sup>	1b/in.2	Lb/in. <sup>2</sup>			
				GROUP A:	OUTER LA INNER LA	MINATIONS	5VISUALL 5NO. 3 E	Y GRADED	DOUGLAS-FIR, SPRUCE		
A01	3.08	12.38	9	0.45	6,550	1.64	1.80	225	MAJ. at 10.1		
A02	3.07	12.36	9	.47	7,720	1.72	1.85	265	MAJ. at 9.8		
A03	3.07	12.39	10	. 50	6,000	1.83	1.87	207		10 pct at 6.0	S.O.G. 4 to 6
A04	3.07	12.39	11	.45	4,260	1.96	2.07	147		40 pct at 8.6	S.O.G.
A05	3.08	12.40	11	.43	3,450	1.73	1.84	119			Shear
A06	3.08	12.37	8	.47	4,560	1.85	1.78	157	INV.		S.O.G. 8 to 11
A07	3.09	12.37	10	.44	5,590	1.76	1.85	192		MAJ. at 7.5	
A08	3.08	12.38	10	.46	4,830	1.94	2.04	165	MAJ. at 10.0		
A09	3.08	12.38	9	. 47	5,780	1.97	1.99	199		MAJ. at 4.0	
A10	3.09	12.40	11	.46	4,920	1.93	2.39	169		MAJ. at 8.0	
A11	3.08	12.39	10	.44	4,940	1.58	1.77	170	MAJ. at 9.3		
A12	3.09	12.39	10	.44	3,190	1.75	2.13	110		MAJ. at 9.2	
A13	3.08	12.43	9	.46	5.350	1.73	1.88	185			G.D. at 9.5
A14	3.08	12.39	10	.44	3,860	1.72	1.79	133		MAJ. at 9.2	
A15	3.08	12.38	9	.47	4,520	1.82	1.98	155			G.D. at 11.4
Av.	3.08	12.39	10	.46	5,040	1.80	1.94	173			
c.o.1	1.2/				23.7	6.6	8.8				
				GROUP B:	OUTER LA	MINATION	SVISUALI	Y GRADED	DOUGLAS-FIR.		
					INNER LA	MINATION	5NO. 3 E	NGELMANN	SPRUCE		
B01	3.07	17.86	9	0.47	6.280	2.04	2.20	312	MAJ. at 10.0		
B02	3.08	17.91	10	.46	6.460	1.83	1.95	322		MAJ. at 8.5	
B03	3.08	17.90	10	.45	. 4.670	1.71	1.68	232	MAJ. at 8.8		
B04	3.09	17.92	10	.46	4,440	1.63	1.63	221	MAJ. at 10.1		
B05	3.09	17.90	10	.46	5,690	1.60	1.86	283	MAJ. at 9.8		G.D. at 9.0
806	1.08	17 89	10	46	6 080	1.97	1.86	302	MAL at 9.8		
B07	3.08	17.86	8	.46	6,710	1.79	2.16	333	MAI. at 10.2		
808	3.08	17.89	10	.47	4,720	1.72	1.79	235	and at it.	MAI. at 9.1	
B09	3.08	17.91	10	46	5.620	1.67	1.68	279	MAJ. at 10.0		
B10	3.07	17.86	10	.47	5,660	1.78	1.71	281			Shear
	2.00		10	10	6 210	1 00	2.10				
811	3.08	17.91	10	.48	6,310	1.89	2.12	314		MAJ. at 10.5	C D at 12 5
812	3.07	17.90	10	.40	4,030	1.58	1.55	201	MAT 10.0	5 pct at 13.1	G.D. at 12.5
813	3.07	17.00	10	.47	5,170	2.09	2.15	200	A.J. at 10.0	MAT at 4.4	
815	3.00	17.00	10	. 50	5 780	1 79	1.07	289		1 pot at 10 3	S.O.C. 10 to 12
515	3.07	17.91	10	.4/	3,780	1.79	1.9/	200		r per at 10.3	5.0.0. 10 10 12
Av.	3.08	17.89	10	.47	5,560	1.78	1.87	276			
C.O.1	1.2/				14.3	9.1	11.6				

Line constants       are there is a constant of the constant	Beam	Dimens	ions <sup>1/</sup>	Moisture	Specific	Modulus	Modul	us of	Shear		Failure comment	s <mark></mark>
$ \begin{array}{c c c c c c c c c c c c c c c c c c c $		Width	Depth	content <sup>27</sup>	gravity <sup>27</sup>	rupture	Full span	Con- stant moment section	at failure	Selected tension lamination knot	Tension lamination finger joint	Other
GROUP C: OUTER LAMINATIONSVISUALLY GRADED DOUCLAS-FIR.         CONTRET CAMINATIONSND. 3 DOUCLAS-FIR.         CONTRET		In.	In.	Pct		Lb/in, <sup>2</sup>	$\frac{\text{Million}}{1\text{b/in.}^2}$	$\frac{\text{Million}}{1\text{b/in.}^2}$	Lb/in. <sup>2</sup>			
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$					GROUP C:	OUTER LA INNER LA	MINATIONS MINATIONS	VISUAL1	Y GRADED OUGLAS-FI	DOUGLAS-FIR, R		
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	C01	3.07	17.91	9	0.48	6,270	1.83	1.90	312	MAJ. at 9.6		
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	C02	3.08	17.88	9	.49	5,890	1.88	2.00	292	MAJ. at 9.1		
$\begin{array}{cccccc} & 3.08 & 17.88 & 8 & .49 & 5,990 & 1.79 & 1.70 & 288 & MAJ, at 10.1 \\ \hline 005 & 3.08 & 17.89 & 9 & .50 & 5,900 & 1.78 & 1.61 & 293 & MAJ, at 10.0 \\ \hline 006 & 3.06 & 17.86 & 9 & .50 & 5,230 & 1.83 & 2,10 & 260 & & & & & & & & & & & & & & & & & & &$	003	3.08	17.91	9	. 50	6,770	1.79	1.86	337	MAJ. at 8.9		
C05 3.08 17.89 9 .50 5,200 1.78 1.61 293 MAJ. at 10.0 C06 3.06 17.86 9 .50 5,230 1.83 2.10 260 S.0.6.7 to 9 C07 3.08 17.89 9 .50 7,770 1.92 1.87 386 INV. at 10.4 10 pct at 9.5 S.0.6. 7 to 9 C08 3.08 17.88 10 .47 5,570 1.67 1.54 282 C13 3.07 17.89 9 .48 6,860 1.76 1.54 283 C13 3.07 17.87 9 .49 5,320 1.93 2.06 264 MAJ. at 9.0 C11 3.07 17.87 9 .49 5,320 1.93 2.06 264 MAJ. at 9.0 C12 3.08 17.86 10 .49 6,180 1.89 1.91 306 S.0.6. 7 to 9 C13 3.07 17.91 9 .49 5,320 1.83 1.57 174 MAJ. at 10.0 C14 3.08 17.91 9 .49 5,880 1.81 1.82 292 C.0.V. $\frac{5}{2}$ I.7.9 I.6.9 5.2 11.2 CROUP D: OUTER LAMINATIONSVISUALLY GRADED HEM-FIR, INNER LAMINATIONSVISUALLY GRADED HEM-FIR, INNER LAMINATIONSVISUALLY GRADED HEM-FIR, INNER LAMINATIONSL3 HEM-FIR D01 3.06 12.41 10 0.37 5,250 1.37 1.38 181 MAJ. at 9.9 G.D. at 13.0 G.D. At 8.5 G.D. at 13.0 G.D. At 8.5 G.D	C04	3.08	17.88	8	.49	5,990	1.79	1.70	298	MAJ. at 10.1		
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	C05	3.08	17.89	9	. 50	5,900	1.78	1.61	293	MAJ. at 10.0		
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	C06	3.06	17.86	9	. 50	5.230	1.83	2.10	260			S.O.G. 7 to 9
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	C07	3.08	17.90	9	. 50	5,960	1.89	2.15	296	MAI at 9.2		0.0.0. / 00 /
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	C08	3.08	17.89	9	50	7 770	1 92	1 87	386	INV at 10 /	10 not at 0 5	S 0 C
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	000	3.08	17 88	10	47	5 670	1 67	1.54	282	1117. at 10.4	to per ar 3.5	S.O.C. 7 to 9
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	C10	3.07	17.89	9	.48	6,860	1.76	1.68	341	MAJ. at 10.7		3.0.0. / 10 9
Cil 3.06 17.87 9		2 07	17 97	0	10	5 220	1 02	2.06	244			
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	C11	3.07	17.0/	10	.49	5,320	1.93	2.00	204	MAJ. at 9.0		C O C 0 - 11
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	012	3.00	17.00	10	.49	6,180	1.09	1.91	300			S.U.G. 8 to 11
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	C13	3.07	17.91	9	. 50	0,240	1.77	1.77	311	MAX 10.0		G.D. at 7.3
Av. 3.08 17.89 9 .49 5,880 1.81 1.82 292 c.o.v. $\frac{5^{f}}{2}$ .60,v. $\frac{5^{f}}{2}$ .70,v. $\frac{5^{f}}{2}$ .70	C14	3.08	17.91	10	. 49	4,600	1.58	1.57	229	MAJ. at 10.0		G.D. at 7.5
Av. 3.08       17.89       9       .49       5,880       1.81       1.82       292         C.O.V. $\frac{5}{2}$ 16.9       5.2       11.2          GROUP D:       OUTER LAMINATIONSVISUALLY GRADED HEM-FIR, INNER LAMINATIONSL3 HEM-FIR       MAJ. at 9.5         D01       3.06       12.41       10       0.37       5,250       1.37       1.38       181       MAJ. at 9.5         D02       3.06       12.37       9       .36       4,720       1.30       1.35       162       G.D. at 13.0         D03       3.06       12.37       9       .38       5,750       1.38       1.41       170       30 pct at 13.7       S.0.6.10 to 14         D05       3.06       12.37       10       .38       4,550       1.30       1.46       157       MAJ. at 9.9       G.D. at 10.6         D06       3.08       12.40       10       .38       4,550       1.30       1.46       157       MAJ. at 9.8       G.D. at 11.9         D07       3.07       12.39       9       .36       4,720       1.37       1.53       146       MAJ. at 10.5       Compression         D08       3.08       12.39       9       .37       <												
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Av.	3.08	17.89	9	.49	5,880	1.81	1.82	292			
$ \begin{array}{c c c c c c c c c c c c c c c c c c c $	C.O.V	.=				16.9	5.2	11.2				
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$					GROUP D	OUTER	LAMINATIC LAMINATIC	NSVISUA	ALLY GRADE	D HEM-FIR,		
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	D01	3.06	12.41	10	0.37	5,250	1.37	1.38	181		MAJ. at 9.5	
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	D02	3.06	12.37	9	. 36	4,720	1.30	1.35	162			G.D. at 13.0
$\begin{array}{cccccccccccccccccccccccccccccccccccc$	D03	3.06	12.39	9	. 38	5,750	1.38	1.43	198	MAJ. at 9.9		G.D. at 8.5
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	D04	3.07	12.39	9	. 38	4,920	1.35	1.41	170		30 pct at 13.7	S.O.G. 10 to 14
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	D05	3.06	12.37	10	. 38	5,720	1.28	1.29	197	MAJ. at 9.9		G.D. at 10.6
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	D06	3.08	12.40	10	. 38	4,550	1.30	1.46	157	MAJ. at 9.8		
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	D07	3.07	12.39	9	. 39	4.720	1.24	1.39	163			G.D. at 11.9
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	D08	3.08	12.39	9	. 36	4.230	1.37	1.53	146	MAJ. at 10.5		
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	209	3.07	12.35	9	. 37	5.360	1.20	1.32	184			Compression
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	D10	3.08	12.40	10	. 37	4,510	1.29	1.34	155		MAJ. at 8.8	
$\begin{array}{cccccccccccccccccccccccccccccccccccc$		3.07	12 18	9	18	4 820	1 22	1 40	166	MAT at 10 3		
D13 $3.07$ $12.36$ $9$ $.38$ $6.310$ $1.31$ $124$ $124$ $1000$ $100$ $100$	D12	3.08	12.30	10	37	6,210	1 39	1.41	214	. at 10.3		Compression
D14       3.08       12.39       10       .37       4.790       1.24       1.33       165       Compression         D15       3.09       12.39       9       .39       4.710       1.32       1.20       162       MAJ. at 9.9         Av.       3.07       12.39       9       .37       5.110       1.31       1.37       176 $C.o.v.^{5/}$ 12.4       4.8       5.6	013	3.07	12.36	9	38	6 310	1 38	1 31	216			Compressi
Av.       3.07       12.39       9       .39       4,710       1.32       1.20       162       MAJ. at 9.9         Av.       3.07       12.39       9       .37       5,110       1.31       1.37       176         C.o.v. <sup>5/</sup> 12.4       4.8       5.6	014	3.08	12 30	10	37	4 790	1 24	1 33	165			Compression
Av. $3.07$ 12.39       9       .37       5,110       1.31       1.37       176         c.o.v. <sup>5/</sup> 12.4       4.8       5.6	D15	3.09	12.39	9	39	4 710	1 32	1.33	162	MAT at 9 9		compression
Av. $3.07$ $12.39$ $9$ $.37$ $5,110$ $1.31$ $1.37$ $176$ c.o.v. <sup>5/</sup> 12.4 $4.8$ $5.6$ $$	015	1.09	12.39	,	. 37	4,710	1.52	1.20	102	. a. a. 3.9		
c.o.v. <sup>2</sup> / 12.4 4.8 5.6	Av.	3.07	12.39	9	. 37	5,110	1.31	1.37	176			
	C.O.V	1.2/				12.4	4.8	5.6				

#### Table III-1.--Results of bending tests--continued

Table	III-1, Results	of	bending	testscontinu	ied
				and the second of the second of the second sec	

Seim	Dimens	tons <sup>1/</sup>	Moisture	Specific 3/	Modulus	Modul	us of	Shear		Failure comments	<u>4</u> /
	Width	Depth	content <sup>27</sup>	gravity <sup>27</sup>	rupture	Full span	Con- stant moment section	at failure	Selected tension lamination knot	Tension lamination finger joint	Other
						Million	Million		*************		
	In.	In.	Pet		Lb/in. <sup>2</sup>	1b/in, <sup>2</sup>	1b/tn. <sup>2</sup>	Lb/in. <sup>2</sup>			
				GROUP	E: OUTE INNE	R LAMINAT R LAMINAT	IONSE-F IONSL3	ATED DOUG DOUGLAS-F	las-fir, IR		
E04	3.07	12.38	11	0.49	5,120	1.91	1.90	176	MAJ. at 11.7		
:05	3.09	12.40	12	.51	5,300	2.02	2.32	183		MAJ. at 8.7	
08	3.08	12.40	8	. 52	6,760	1.98	2.14	233		40 pct at 9.1	S.O.G. 8 to 9
9	3.07	12.40	10	. 52	6,110	2.02	2.22	210		90 pct at 12.5	S.O.G.
10	3.07	12.40	10	. 51	7,250	2.04	2.14	250	MAJ. at 11.0		G.D. 9 to 10
e1.2	3.09	12.40	11	. 50	6.420	1.97	1.81	221	MAJ at 10.1		G.D. at 8.4
15	3.08	12.39	10	. 53	5,500	2.26	2.37	189		50 pct at 10.2	G.D. at 9.6
16	3.06	12.40	10	. 51	6,990	2.22	2.40	241	TNV at 9.8	5 net at 7 4	0. D. at 7.0
19	3.07	12.41	11	54	5 820	2 21	2 48	201	MAI at 11 5	J per ar 7.4	SOC 8 to 11
21	3.08	12.42	10	. 52	5,680	1.98	2.20	196	MAJ. at 9.1		5.0.0. 0 10 11
	2 00	12.20	10	- 1	5 000	1.00	2.12	100			
24	3.00	12.39	10	. 54	5,220	1.90	2.12	180			G.D. 9 to 10
20	3.08	12.39	10	. 35	5,690	2.07	2.48	196		MAJ. at 6.1	
27	3.08	12.40	12	. 56	5,140	1.97	2.02	1//	MAJ. at 9.1		
29	3.09	12.40	10	.53	8,740	2.21	2.32	299	INV. at 10.8		S.O.G. 6 to 11
v.	3.08	12.39	10	. 52	6.170	2.05	2.19	212			
	5/				16.3	6.0	9.4				
				GROUP	F: OUTER INNER	LAMINATI	ONSE-RA	2MG SOUTH	ERN PINE, ERN PINE		
702	3.11	12.33	11	0.53	8,380	1.73	1.96	287	MAJ. at 9.6		
03	3.13	12.34	10	.51	7,060	1.72	2.07	242	MAJ. at 9.9		
04	3.15	12.37	10	. 50	7,280	1.70	1.99	250		MAJ. at 8.1	
07	3.11	12.4C	10	.49	6,530	1.65	1.78	225		MAJ, at 13.2	
08	3.11	12.38	12	. 51	7,040	1.80	1.89	242		MAJ. at 12.8	
11	3.11	12.32	11	. 48	5.620	1.60	1.73	192	INV. at 8.5	30 pct at 7.5	
12	5.14	12.32	10	.49	6.500	1.69	1.98	223		MAJ, at 11.6	
13	3.14	12.31	10	. 51	7.310	1.67	1.81	250		10 nct at 7.0	G. D. at 8.0
17	3.15	12.35	12	. 50	5,900	1.70	1.94	202		5 pct at 13.5	S.O.G. 12 to 1
20	3.14	12.36	11	. 49	5,420	1.62	1.67	186		MAJ. at 11.8	
21	3.17	12.37	12	.51	4.800	1.75	1.95	165	MAL at 10.0		
:22	3.17	12.37	13	48	4,780	1.67	2 17	164	. at 10.0	MAT at 9 3	
222	3 18	12.37	11	40	6 890	1.55	2.04	236		20 not at 12 2	C D 12 to 13
227	2.16	12.33	11	50	8 710	1.33	2.04	230	MA1	20 pet at 12.2	G.D. 12 to 13
278	3 13	12.30	11	.50	6,660	1 76	1.83	299	MAL at 9.8		a. D. ac 9.0
20	5.15	12.33	11	. 50	0,000	1.74	1.03	228	na. at 9.8		
۱۷.	3.14	12.35	11	.50	6,590	1.69	1.92	226			
C 0 1	1 21				17.4	4.0	7.2				

Beam	Dimens	ions <sup>1/</sup>	Moisture	Specific 2/	Modulus	Modul	us of	Shear		Failure comment	s <sup>4</sup> /
	Width	Depth	content <sup>2</sup>	gravity <sup>_1</sup>	or rupture	Full span	Con- stant moment section	at at failure	Selected tension lamination knot	Tension lamination finger joint	Other
						Million	Million				
	In.	In.	Pct		Lb/in. <sup>2</sup>	1b/in. <sup>2</sup>	<u>1b/in.<sup>2</sup></u>	Lb/in. <sup>2</sup>			
				GR	OUP G: OU IN	TER LAMIN	ATIONSE	-RATED HE	M-FIR,		
G01	3.09	12.40	9	0.40	3,810	1.66	1.79	131	MAJ. at 9.6		G.D. at 12.5
G02	3.09	12.39	8	.40	5,450	1.69	1.86	188	MAJ. at 9.7		
G03	3.09	12.41	9	. 39	6,480	1.71	2.04	224		MAJ. at 8.3	
G04	3.08	12.36	9	.40	6,340	1.72	2.09	218			S.O.G. 9 to 13
G05	3.07	12.39	9	.40	5,440	1.75	1.73	187	MAJ. at 10.1		
				10			2.01	227		00	
.00	3.08	12.38	9	.40	6,610	1.73	2.04	227		80 pct at 6.4	5.0.6.
607	3.08	12.37	9	.41	7,850	1.69	1.69	270			Compression
308	3.07	12.36	9	.40	6,890	1.71	1.84	237			Compression
309	3.08	12.36	8	. 42	7,290	1.73	1.81	250			S.O.G. 6 to 8
310	3.07	12.37	8	.41	6,140	1.72	1.83	211		MAJ. at 8.3	
311	3.07	12.37	9	.40	5,600	1.70	1.78	192		MAJ. at 8.4	
312	3.08	12.42	10	. 39	6.170	1.65	1.71	213		MAJ. at 9.4	
313	3.08	12.40	10	.40	6.730	1.61	1.81	232			Compression
314	3.08	12,40	9	.40	6,640	1.71	1.84	229	MAJ. at 10.0		
515	3.08	12.29	9	.40	5,690	1.78	1.80	194	INV. at 10.1	60 pet at 9.0	
Av.	3.08	12.38	9	.40	6,210	1.70	1.84	214			
c.o.v	.5/				15.3	2.4	6.6				
			G	ROUP H: OU	TER LAMINA	TIONSE-	RATED WHI	TE WOOD	(LODGEPOLE PINE	.),	
				IN	NER LANINA	11104313	, whill we	OD (ENGII	In STRUCE)		
H01	3.06	12.33	10	0.43	7,580	1.45	1.52	260			Compression
H02	3.06	12.36	10	. 42	6,430	1.44	1.66	221	INV. at 10.0		G.D. at 11.5
H03	3.08	12.38	11	. 41	4,540	1.48	1.54	156	INV. at 9.9	40 pct at 9.0	
H04	3.07	12.35	10	.43	5,490	1.38	1.54	188		MAJ. at 8.2	
405	3.07	12.35	10	. 42	5,480	1.45	1.65	188			Compression
106	3.08	12.36	11	.42	4.500	1.37	1.44	154		MAJ. at 8.8	
H07	3.07	12.34	9	. 43	5.440	1.47	1.58	186			G.D. at 10.6
HOS	3.08	12.34	9	. 43	4.420	1.36	1.50	152	MAJ. at 9.2		
	3.09	12.38	9	. 43	4.510	1.38	1.57	155	MAJ. at 10.0		
H09	3 09	12.38	11	. 42	4,400	1.45	1.58	152	MAJ. at 10.9		G.D. at 11.5
H09 H10	3.08					1 11	1 10	100	MAT		
H09 H10	3.08	12 35	11	4.1	5 700		1. 37	177	inter de lise		
H09 H10 H11	3.08	12.35	11	.41	5,790	1.34	1 74	218			Compression
H09 H10 H11 H12	3.08	12.35	11 10	.41 .43	5,790 6,360 4,300	1.51	1.74	218	MAL at 9.9		Compression
H09 H10 H11 H12 H13	3.08 3.09 3.09	12.35 12.37 12.38	11 10 9	.41 .43 .43	5,790 6,360 4,300	1.34 1.51 1.40	1.74	218 148	MAJ. at 9.9	MAI at 2.0	Compression
H09 H10 H11 H12 H13 H14	3.08 3.08 3.09 3.09 3.08	12.35 12.37 12.38 12.39	11 10 9 10	.41 .43 .43 .42	5,790 6,360 4,300 4,240 4,240	1.34 1.51 1.40 1.34	1.74 1.51 1.44	218 148 146	MAJ. at 9.9	, MAJ. at 7.0	Compression
H09 H10 H11 H12 H13 H14 H15	3.08 3.09 3.09 3.09 3.08 3.08	12.35 12.37 12.38 12.39 12.39	11 10 9 10 10	.41 .43 .43 .42 .42	5,790 6,360 4,300 4,240 4,780	1.34 1.51 1.40 1.34 1.51	1.74 1.51 1.44 1.71	218 148 146 164	MAJ. at 9.9 MAJ. at 10.8	, MAJ. at 7.0	Compression G.D. at 9.4
H09 H10 H11 H12 H13 H14 H15 Av.	3.08 3.09 3.09 3.09 3.08 3.08 3.08	12.35 12.37 12.38 12.39 12.39 12.39	11 10 9 10 10	.41 .43 .43 .42 .42 .42	5,790 6,360 4,300 4,240 4,780 5,220	1.34 1.51 1.40 1.34 1.51	1.74 1.51 1.44 1.71 1.56	218 148 146 164 179	MAJ. at 9.9 MAJ. at 10.8	, MAJ. at 7.0	Compression G.D. at 9.4

#### Table III-1. -- Results of bending tests -- continued

1/ Dimensions are averages of measurements made at--and 4 ft both sides of--midlength.

2/ Determined immediately following test using a resistance-type meter with 1-1/2-in.-long needles. Data given are averages of readings taken for each lamination at point of failure. Readings were corrected using factors published by the manufacturer.

3/ Based on weight and volume of complete beam at time of test. Weight was adjusted to ovendry.

4/ Locations are given in feet with reference to one end of the beam. Midlength was at 10.0 and constant moment section was from 8.0 to 12.0. MAJ. - major cause; INV. - involved in failure; S.O.G. - slope of grain; and G.D. - grain deviation.

5/ Coefficient of variation - (standard deviation + average) x 100.

#### APPENDIX IV

#### Determination of Clear Wood Bending Stress (CWS) Value from Beam Test Data

To attain CWS values for different species and grades of lumber, data from past experiments conducted at FPL, Oregon State University, and both Canadian Laboratories were analyzed. Strength ratios were estimated by the  $I_K/I_G$  concept using knot data obtained either from analysis of the lumber used or from similarly graded lumber. Strength ratios were expressed as the ratio of the anticipated near minimum strength to that of a clear beam consisting entirely of the density and/or stiffness of material in the tension lamination. Based on concepts discussed in this report:

$$MOR = (CWS)(SR) \left(\frac{Td}{2z}\right)$$

MOR - expected near minimum beam strength CWS - expected near minimum clear wood stress for tension lamination quality material

SR - strength ratio

- t transformed section factor
- d/2z ratio of half depth to neutral axis positions.

By redefining SR as

$$(SR)_{\mathbf{E}ff} = (SR)\left(\frac{Th}{2z}\right)$$

clear wood design stress can be expressed as

$$CWS = \frac{MOR}{(SR)} Eff$$

Values of CWS were calculated for all beam tests for which strength ratios could be estimated.

MOR values were adjusted to a 12-in. common depth, a 21:1 span-to-depth ratio and uniform loading, and a 12 pct moisture content (ASTM D 2915). Also, the dead load stresses of the beams were added if they were not considered in the initial analysis.

Strength ratios calculated using the unbalanced  $I_K/I_G$  concept are included in table IV-1. Also, the strength ratio as believed to be limited by the tension lamination grade is included. The lower of these two strength ratios was multiplied by  $\frac{Td}{2z}$  to determine the effective strength ratio. CWS values were thus calculated for each beam and are given in table IV-2. Each group was then statistically analyzed and results presented in the form of averages and standard deviations (table IV-3).

Source	Beam identification	Number of	Tensic strengt	on side ch ratio	Td/2z <sup>3/</sup>	Effective strength
		Deding	Knots <sup>1/</sup>	Tension		ratio-
				lamina-		
				tion <sup>2/</sup>		
	VISUA	LLY GRAI	DED DOUGLA	AS-FIR		
Present study	A01-A15	15	0.734	0.674	0.858	0.578
	B01-B15	15	.761	.724	.857	.620
	C01-C15	15	.756	.724	.865	.626
FPL 113	1,3,6-10	7	.735	21.774	.934	.686
	2,4,5	3	.735	674	.934	.630
	21-23	3	.770	.824	.934	.719
FPL 146	41-45	5	.735	.774	.934	.686
	46-50	5	.739	.774	.935	.691
FPL 236 ( <u>15</u> )	86-90	5	. 633	.674	.843	.534
	91-95	5	.637	.674	.928	. 591
	96-105	10	. /3/	. 174	.922	.680
Total		88				
	VISUAI	LY GRADI	ED SOUTHER	RN PINE		
FPL 113	11-20	10	.791	.674	.865	.583
FPL 113	24-26	3	.819	.774	.838	.649
FPL 146	36-40	5	.822	.774	.877	.679
FPL 151	51-60	10	.822	.774	.877	.679
Total		28				
	۲۷	SUALLY (	GRADED HEN	1-FIR		
Present study	D01-D15	15	. 589	.674	.894	.527
RP 18 (10)	Comb. 1	5	.732	.814	.930	.681
	Comb. 2	5	.718	.724	.901	.647
	Comb. 3	5	.718	.674	.897	.604
Total		30				
	E-RATED DOUGLAS	S-FIR (2	.4-2.6E TH	ENSION LAN	MINATION)	
$T_{-26}(9)$	D01-D05	5	. 766	. 724	.887	.642
1 20 ())	D06-D11	6	. 844	.724	.918	.665
T-27 (8)	D01-D06	6	. 829	.774	.898	.695
Total		17				

Table IV-1.--Beam groups used in clear wood stress analysis

Source	Beam identification	Number of	Tensic	on side th ratio	$Td/2z^{3/}$	Effective
		beams	Knots <sup>1/</sup>	Tension lamina- tion <sup>2/</sup>		ratio4/
	E-RATED DOUGI	AS-FIR (	(2.2E TEN	SION LAMIN	NATION)	
Present study T-27 ( <u>8</u> ) Total	E04-E29 D07-D12  E-RATED SOUTH	15 6 21 ERN PINE	.686 .696 	.724 .724 	.898 .933 	.616 .649 
T-27 ( <u>8</u> )	SP07-SP12	6	. 766	.774	.901	.690
	E-RATED SOUTH	ERN PINE	(2.0E TE	NSION LAMI	INATION)	
Present study T-27 ( <u>8</u> ) Total	F02-F28 SP01-SP06	15 6 21	. 752 . 770	.774 .724	.900 .933 	.677 .675
	E-RATED HEN	M-FIR (2.	OE TENSI	ON LAMINAT	TION)	
Present study T-26 ( <u>9</u> ) T-27 ( <u>8</u> )	G01-G15 H01-H06 W01-W06 HDF13-HDF18 H19-H24	15 6 6 6	.771 .870 .871 .860 .855	.824 .824 .824 .774 .774	.895 .965 .896 .933 .927	.690 .795 .738 .722 .717
Total		39				
	E-RATED HE	M-FIR (1.	.8E TENSI	ON LAMINA	FION)	
T-27 ( <u>8</u> )	H25-H30	6	.848	.724	.916	.663
	E-RATED "WHIT	E WOODS"	(2.4E TE	NSION LAM	INATION)	
VP-X-132 ( <u>12</u> )	9,11	4	.825	.824	.862	.710
	E-RATED "WHIT	e woods"	(2.2E TE	NSION LAM	INATION)	
VP-X-132 (12)	7,12	4	.747	.774	.861	.643

### Table IV-1.--Beam groups used in clear wood stress analysis

Source	Beam identification	Number of	Tensio strengt	n side h ratio	Td/2z <sup>3/</sup>	Effective strength
		beams	Knots <sup>1/</sup>	Tension lamina- tion <sup>2/</sup>		ratio <sup>47</sup>
	E-RATED "WHITE	WOODS"	(2.0E TEN	SION LAMI	NATION)	
VP-X-132 (12)	3.4	4	.729	.774	.920	.670
·	8.10	4	. 768	.774	.905	.695
Aplin (6)	A1-A8, B1-B8	14	.715	.824	.952	.681
	D1-D8	7	.802	.824	.955	.766
	Е2-Е8	6	.801	.824	.937	.751
Total		35				
	E-RATED "WHITE	woods"	(1.8E TEN	SION LAMI	NATION)	
Present study	H01-H15	15	.640	.674	.848	. 543
T-27 (8)	LP31-LP36	6	.659	.724	.911	.600
VP-X-132 (12)	1,2,5,6	8	. 649	.774	.909	.590
Total		29				
	E-RATED "WHITE	WOODS"	(1.6E TEN	SION LAMI	NATION)	
T-27 ( <u>8</u> )	LP37-LP42	6	.633	.674	.897	.549

Table IV-1.--Beam groups used in clear wood stress analysis

1/ Based on unbalanced  $I_K/I_G$  analysis using knot data either from lumber used in manufacture or similarly graded lumber. Tension side assumed to control.

2/ Estimated limiting strength ratio for outer tension lamination.

3/ Transformed section factor--see "Development of Design Criteria" in main text of this report.

 $\frac{4}{1}$  Lowest tension-side strength ratio multiplied by transformed section factor.

Beam identifi-	Adjusted	Calculated CWS	Beam identifi-	Adjusted	Calculated CWS	Beam identifi-	Adjusted	Calculated CWS
cation			cation			cation		
	Lb/in.	Lb/in. <sup>2</sup>		Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>		Lb/in. <sup>2</sup>	Lb/in.2
VISUALLY	GRADED DO	UGLAS-FIR	VISUALLY	GRADED DOU continued	GLAS-FIR	VISUALLY	GRADED DOU continued	GLAS-FIR
A01	6,200	10,730	C01	6,050	9,660	41	7,080	10,320
A02	7,300	12,640	C02	5,680	9,070	42	5,050	7,360
A03	5,820	10,070	C03	6,520	10,410	43	6,010	8,760
A04	4,250	7,360	C04	5,650	9,020	44	5,940	8,650
A05	3,450	5,980	C05	5,690	9,090	45	6,460	9,420
A06	4,240	7,340	C06	5,050	8,070	46	7,170	10,370
A07	5,430	9,390	C07	5,740	9,170	47	3,920	5,670
A08	4,690	8,120	C08	7,480	11,950	48	5,900	8,550
A09	5,480	9,480	C09	5,600	8,950	59	5,360	7,760
A10	4,890	8,470	C10	6,610	10,560	50	6,180	8,950
A11	4,810	9,320	C11	5,130	8,200			
A12	3,120	5,400	C12	6,090	9,730	86	4,570	8,550
A13	5,080	8,780	C13	6,020	9,610	87	4,850	9,080
A14	3,770	6,510	C14	3,390	5,420	88	4,710	8,810
A15	4,290	7,430	C15	4,550	7,270	89	5,230	9,800
						90	5,640	10,560
B01	6,050	9,760	1	6,200	9.050	91	4,610	7,800
B02	6,380	10,280	3	5,840	8,520	92	6,270	10,610
BO3	4.610	7.440	6	6.250	9,100	93	5,930	10.030
B04	4.390	7.090	7	5.840	8.510	94	4.670	17,910
B05	5.620	9,060	8	6,110	8,900	95	4.680	7,910
B06	6,000	9 670	9	6, 380	9,300		.,	.,
807	6 320	10,190	10	5,310	7 730	96	6.170	9.080
808	4 670	7 530	2	5 520	8 760	97	6 900	10,150
BOG	5,550	8,940	7	4,600	7 300	98	5 880	8 640
B10	5,590	9 020	5	4,000	7,790	99	5 140	7 550
B11	6,230	10,040	1	4, 510	1,190	100	5 830	8 570
B12	3,000	6,640	21	5 960	8 200	100	5.940	8,370
p11	5,100	9,220	21	5,760	8,290	102	6,100	8 970
BIJ	5,100	0,250	22	5,700	0,000	102	5,100	0,370
D14	5,740	9,250	23	6,470	9,000	103	5,930	9,750
815	5,710	9,200				104	7,000	10,290
						105	7,000	10,290
VISUALLY	GRADED SOU	THERN PINE	VISUALLY (	GRADED SOUT	HERN PINE	VISUALLY (	RADED SOUT	HERN PINE
11	5,370	9,220	24	5,790	8,920	51	8,040	11,850
12	6,800	11,660	25	5,190	8,000	52	7,270	10,710
13	5,410	9,270	26	5,420	8,350	53	7,410	10,910
14	4,390	7,530				54	8,430	12,420
15	5,380	9,230	36	7,400	10,900	55	8,230	12,120
16	4.540	7,790	37	8,220	12,110	56	7,610	11,210
17	4,920	8,440	38	9,530	14,030	57	7,670	11,290
18	7.940	13,620	39	8,690	12,800	58	7,720	11,370
19	5,330	9.140	40	8.930	13,160	59	8,330	12,270
20	7,050	12,100				60	7,870	11,590
								mu hrn
VISUAL	LY GRADED	HEM-FIR	VISUAL	continued	EM-FIR	VISUALI	continued	EM-FIR
D01	5 090	9 660	D11	4.570	8.670	2-1	4.800	7.420
002	4 480	8,500	D12	6 010	11,410	2-2	5.140	7.940
002	5 440	10, 330	D12	5 960	11, 320	2-3	4.620	7,130
003	5,440	9 960	D13	4 650	8 820	2-4	4 460	6,890
004	4,670	0,800	014	4,050	8 490	2-5	5 370	8 300
005	5,540	10,520	015	4,470	0,400		3,570	0,000
006	4,420	8,390		6 700	0 970	7-1	5 770	0 550
007	4,480	8,500	1-1	7, 200	10 570	3-2	5,220	8 660
008	4,020	7,620	1-2	6,200	10,010	3-2	6,190	10,250
009	5,080	9,640	1-3	6,820	7,650	3-3	5 770	9 550
010	4,380	8,310	1-4	3,210	10, 370	3-4	3,770	6.640
			1-5	7,000	10,270	3-3	3,890	0,440

## Table IV-2.--Adjusted modulus of rupture (MOR) and clear wood stress (CWS) values for glued, laminated beams

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Beam	Adjusted	Calculated	Beam	Adjusted	Calculated	Beam	Adjusted	Calculated
identit	$1 - MOR^{1/2}$	CWS	identifi	- MOR_1/	CWS	identifi-	MOR1/	CWS
cation			cation			cation	non	
		2			2			
	Lb/1n.	Lb/in.		Lb/in.	Lb/in.		Lb/in.	Lb/in.
E-RATED	DOUGLAS-FIR	(2.4-2.6E	E-RATED	DOUGLAS-FIR	(2 4-2 6F	F-RATED D	DUCLAS-FIR	(2 4-2 6F
TENSION	LAMINATION)	(	TENSION	LAMINATION)-	-continued	TENSION L	AMINATION)-	(2.4-2.0E
			renoren	autinition)	-concinaed	ILHOION L	ATTRATION)-	-concinded
D01	5,990	9.330	D06	6.810	10.250	D01	8.080	11 620
D02	7.440	11,580	D07	6.210	9.340	D02	8,360	12,020
D03	4,990	7.780	D08	5.940	8,930	003	8 150	11 730
D04	8,080	12,580	D09	6,000	9,030	D04	8 550	12,300
D05	7.370	11,480	D10	8,360	12,570	D05	5 270	7 580
			D11	8,180	12,300	D06	6 670	9 590
				-,	,	000	0,0/0	,,,,,,
E-RATED	DOUGLAS-FIR	(2.2E	E-RATED	DOUGLAS-FIR	(2.2E	E-RATED DO	DUGLAS-FIR	(2.2E
TENSION	LAMINATION)		TENSION	LAMINATION) -	-continued	TENSION L	AMINATION) -	-continued
E04	5,090	8,270	E19	5,780	9.380	D07	5.140	7.920
E05	5,400	8,770	E21	5,510	8,940	D08	7.440	11.470
E08	6,260	10,160	E24	5.070	8.240	D09	5.300	8,170
E09	5,930	9.620	E26	5.520	8,960	D10	5.860	9.020
E10	7.020	11.390	E27	5.240	8.510	D11	6.510	10,040
E12	6.370	10.340	E28	5.240	8.510	D12	5.780	8,900
E15	5.340	8,670	E29	8,870	14,400			.,
E16	6,770	11,000		0,070	11,100			
E-RATED	SOUTHERN PIN	NE (2.2E	E-RATED	SOUTHERN PIN	E (2.2E	E-RATED SO	DUTHERN PIN	E (2.2E
TENSION	LAMINATION)		TENSION	LAMINATION) -	-continued	TENSION LA	MINATION) -	-continued
SP07	6,430	9,320	SP09	9,040	13,100	SP11	6,490	9,410
SP08	6,900	10,000	SP10	7,110	10,300	SP12	7,150	10,360
E-RATED	SOUTHERN PIN	NE (2.0E	E-RATED	SOUTHERN PIN	NE (2.0E	E-RATED SC	DUTHERN PIN	IE (2.0E
TENSION	LAMINATION)		TENSION	LAMINATION)-	-continued	TENSION LA	AMINATION) -	-continued
F02	8,310	12,270	F17	6,000	8,870	SPOI	5,570	8,250
F03	6,830	10,090	F20	5,390	7,960	SP02	6,000	8,880
F04	7,050	10,410	F21	4,900	7,240	SP03	7,340	10,870
F07	6,330	9,350	F22	5,000	7,390	SP04	7,320	10,850
F08	7,160	10,570	F23	6,830	10,100	SP05	8,300	12,300
F11	5,580	8,250	F27	8,630	12,740	SP06	7,410	10,970
F12	6,300	9,310	F28	6,610	9,770			
F13	7,080	10,450						
						-		P. TENCION
E-RATED	HEM-FIR (2.0	JE TENSION	E-RATED	HEM-FIR (2.0	DE TENSION	E-RATED HI	M-FIR(2.0)	E TENSION
LAMINAT	ION)		LAMINATI	ON)continu	led	LAMINATIO	()continu	ied
COL	2 620	5 250	101	7 2/0	0 110	UDE1 2	7 800	10 930
COI	5,020	3,230	101	9 160	9,110	HDF1/	7,050	0,990
602	5,050	7,310	102	0,160	10,270	HDF14	7,130	10,690
GOS	6,130	0,090	H03	0,000	10, 270	HDF15	6 750	0,000
604	6,000	8,690	HOA	8,240	10,370	HDF10	6,730	9,330
G05	5,150	7,470	HUS	8,670	10,900	HDF17	6,810	9,430
606	6,250	9,060	H06	8,240	10,380	HDF 18	8,020	11,100
007	7,420	10,750	1101	7	0.010	1110	5 000	9 220
608	6,510	9,440	WOI	7,340	9,940	119	6,900	0,230
G09	6,730	9,760	W02	6,740	9,130	H20	6,920	9,650
G10	5,680	8,240	W03	7,340	9,940	H21	6,680	9,310
G11	5,300	7,680	W04	6,840	9,270	H22	7,090	9,890
G12	5,970	8,650	W05	7,110	9,630	H23	7,760	10,820
G13	6,510	9,430	W06	6,630	8,980	H24	7,440	10,380
G14	6,280	9,100						
G15	5,390	7,810						

#### Table IV-2.--Adjusted modulus of rupture (MOR) and clear wood stress (CWS) values for glued, laminated beams--continued

Beam identification	Adjusted MOR <sup>1/</sup>	Calculated CWS	Beam identifi cation	- $\frac{\text{Adjusted}}{\text{MOR}^{1/2}}$	Calculated CWS	Beam identifi- cation	Adjusted MOR <sup>1/</sup>	Calculated CWS
	Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>		Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>		Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>
E-RATED LAMINATI	HEM-FIR (1.8E	TENSION	E-RATED LAMINATI	HEM-FIR (1.8) ON)continue	E TENSION	E-RATED H	HEM-FIR (1.8 DN)continu	E TENSION ed
H25	5,680	8,560	H27	4,420	6,670	H29	4,770	7,190
H26	5,680	8,570	H28	6,740	10,160	H30	6,580	9,920
E-RATED TENSION	"WHITE WOODS" LAMINATION)	(2.4E	E-RATED TENSION	"WHITE WOODS' LAMINATION)	' (2.2E			
9A	7,210	10,150	7A	7,180	11,170			
9B	6,360	8,950	7B	9,090	14,140			
11A	6,180	8,710	12A	8,250	12,830			
11B	9,270	13,060	128	4,850	7,550			
E-RATED TENSION	"WHITE WOODS" LAMINATION)	(2.0E	E~RATED TENSION	"WHITE WOODS'	' (2.0E	E-RATED '	WHITE WOODS	" (2.0E
								concentred
3A	9,270	13,830	A5	7,320	10,750	D1	7.970	10.410
3B	6,970	10,400	A7	10,240	15,030	D3	8.330	10.870
4A	7,460	11,140	A8	6,220	9,130	D4	7.820	10,210
4B	8,700	12,980				D5	7.140	9.330
			B1	6.820	10.010	D6	6.780	8,850
8A	6,610	9,510	B2	4.500	6 600	D7	6,280	8 200
88	5,150	7.420	83	6.860	10,070	D8	7 380	9 630
10A	6.640	9.550	R4	6 230	9 140	100	1,500	5,050
108	8 660	12 460	85	6,760	9 930	F.2	6 680	8 900
100	0,000	12,400	87	5 620	8,260	EZ EA	7,200	0,300
41	9 260	13 500	19.8	7 750	11 280	E4 ES	9 520	11, 260
12	7,640	11, 210	bo	1,150	11,300	ED	8,530	11,360
A2	7,640	11,210				ED	7,520	10,010
AS	9,350	13,720				E/	6,280	8,360
A4	8,690	12,760				E8	8,390	11,170
E-RATED	"WHITE WOODS"	(1.8E	E-RATED	WHITE WOODS'	(1.8E	E-RATED '	WHITE WOODS	" (1.8E
TENSION	LAMINATION)		TENSION	LAMINATION)	-continued	TENSION I	LAMINATION) -	-continued
H01	7 330	13 500	u11	5 750	10 580	1.5	8 360	7 580
402	6 220	11,460	u12	6,150	11 220	10	7,760	11 180
102	6,220	8 320	u13	4 080	7 520	24	5 270	15 220
H0/	5,320	9, 790	H13	4,080	7,520	28	6 540	11,510
405	5,310	9,790	u15	4,120	8 540	20	4,470	14, 170
105	6,680	9,770	hij	4,040	0, 340	5R	6,470	13,160
100	5 150	0,240	1 0 2 1	5 910	0 600	56	0,000	9 030
409	4 200	7 730	1 0 2 2	5 220	9,090	68	6,980	11 080
100	4,200	7,750	1 0 2 2	4,080	8,710	0.0	0,790	11,080
109	4,280	7,870	1024	4,980	8,300			
H10	4,380	0,070	1025	4,470	7,450			
			LP35 LP36	5,500	9,160			
E-RATED	"WHITE WOODS"	(1.6E	E-RATED	"WHITE WOODS'	' (1.6E	E-RATED '	WHITE WOODS	" (1.6E
TENSION	LAMINATION)		TENSION	LAMINATION)	-continued	TENSION I	AMINATION) -	-continued
1	4.370	7.960	1.239	4.940	9.000	LP41	4.580	8,350
1 238	3 920	7 150	LP40	4 470	8 140	1 P42	3 580	6 520
LP 38	3,920	7,150	LP40	4,470	8,140	LP42	3,580	6,5

#### Table IV-2.--Adjusted modulus of rupture (MOR) and clear wood stress (CWS) values for glued, laminated beams--continued

1/ Adjusted to a 12-in.-deep, uniformly loaded beam with a 21:1 span-to-depth ratio and to 12 pct moisture content. Dead-load stresses of beam during test also added.

Group description	Number of beams	Average CWS	Standard deviation	Coeffi- cient of variation
		Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>	Lb/in. <sup>2</sup>
Dense visual grade				
Douglas-fir	88	8,760	1,300	14.8
Southern pine	28	10,790	1,860	17.2
Hem-fir	30	8,980	1,290	14.4
E-rated grade				
2.4E + 2.6E Douglas-fir	17	10,590	1,700	16.1
2.2E Douglas-fir	21	9,660	1,530	15.8
2.2E Southern pine	6	10,410	1,390	13.4
2.0E Southern pine	21	9,850	1,560	15.8
2.0E Hem-fir	39	9,390	1,220	13.0
1.8E Hem-fir	6	8,510	1,400	16.5
2.4E White wood	4	10,220	2,000	19.6
2.2E White wood	4	11,420	2,850	25.0
2.0E White wood	35	10,450	1,910	18.3
1.8E White wood	29	9,760	2,200	22.5
1.6E White wood	6	7,850	890	11.3

Table IV-3.--Summary of clear wood stress (CWS) analysis

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<ul> <li>L 2-2</li> <li>U.S. Forest Products Laboratory.</li> <li>Improved utilization of lumber in glued laminated beams, by Russell C. Moody, Madison, Wis., FPL, 1977. 49 p. (USDA For. Serv. Res. Pap. FPL 292)</li> <li>To provide criteria for improving lumber util-ization in glued laminated beams, 120 specimen</li> </ul>	beams were evaluated. KEYWORDS: E-rated lumber, laminated beams, lumber grade, unsymmetric beam design, wane.	L 2-2 U.S. Forest Products Laboratory.	Improved utilization of lumber in glued laminated beams, by Russell C. Moody, Madison, Wis., FPL, 1977. 49 p. (USDA For. Serv. Res. Pap. FPL 292)	To provide criteria for improving lumber util- ization in glued laminated beams, 120 specimen beams were evaluated.	KEYWORDS: E-rated lumber, laminated beams, lumber grade, unsymmetric beam design, wane.
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