

11-77

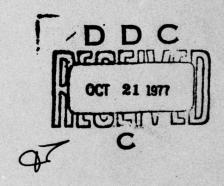
à (



RADC-TR-77-270, Volume III (of three) Final Technical Report August 1977

UNATTENDED/MINIMALLY ATTENDED RADAR STUDY Radar Costs

ITT Gilfillan



Approved for public release; distribution unlimited.

DOC FILE COPY

ROME AIR DEVELOPMENT CENTER
Air Force Systems Command
Griffiss Air Force Base, New York 13441

Some of the figures in this report are not of the highest printing quality but because of economical consideration, it was determined in the best interest of the government that they be used in this publication.

This report has been reviewed by the RADC Information Office (OI) and is releasable to the National Technical Information Service (NTIS). At NTIS it will be releasable to the general public, including foreign nations.

This report has been reviewed and is approved for publication.

APPROVED: adrian S. Briggs

Project Engineer

APPROVED:

JOSEPH L. RYERSON Technical Director Surveillance Division

ment & Rewon

FOR THE COMMANDER:

CARLO P. CROCETTI Chief, Plans Office

Your address has changed or if you wish to be removed from the RADC mailing list, or if the addressee is no longer employed by your organization, please notify RADC (DAP) Griffiss AFB NY 13441. This will assist us in maintaining a current mailing list.

Do not return this copy. Retain or destroy.

UNCLASSIFIED
SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

REPORT DOCUMENTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
REPORT NUMBER 2. GOVT ACCESS	ION NO. 3. RECIPIENT'S CATALOG NUMBER
RADC-TR-77-270; Vol (of three)	
TITLE (and Subtitle)	5. TYPE OF REPORT & PERIOD COVERS
UNATTENDED/MINIMALLY ATTENDED RADAR STUDY,	9 Final Technical Repeat.
2 5	6. PERFORMING ORG. REPORT NUMBER
(14) 17	TG-TP-0002/L-7018-Vol-3 /
. AUTHOR(s)	S. SONTRACT OR GRANT NUMBER(s)
Land Based Air Defense Systems Group	F30602-76-C-0383
PERFORMING ORGANIZATION NAME AND ADDRESS	10. PROGRAM ELEMENT, PROJECT, TAS
ITT Gilfillan	10. PROGRAM ELEMENT, PROJECT, TAS AREA & WORK UNIT NUMBERS
7821 Orion Avenue	63101F
	E2330102
Van Nuys CA 91409	12. REPORT DATE
1. CONTROLLING OFFICE NAME AND ADDRESS	
Rome Air Development Center (OCDE)	13. NUMBER OF PAGES
Griffiss AFB NY 13441	51 V2 650
4. MONITORING AGENCY NAME & ADDRESS(if different from Controlling	
Same (16) E233)	UNCLASSIFIED
	15a DECLASSIFICATION/DOWNGRADING
(17) 01	N/A SCHEDULE
Approved for public release; distribution unli	ímít
Approved for public release; distribution unli	
Approved for public release; distribution unli	
Approved for public release; distribution unli	
Approved for public release; distribution unli	
Approved for public release; distribution unling. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 diff. Same.	
Approved for public release; distribution unling. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer:	
Approved for public release; distribution unling. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE)	ferent from Report)
Approved for public release; distribution unling. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the	ferent from Report)
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 ditt.) Same 18. Supplementary notes RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731.	Electronic Systems Division/XR,
Approved for public release; distribution unling. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. Supplementary notes RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block)	Electronic Systems Division/XR,
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. Supplementary notes RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR - Unattended Radar Surveillance	Electronic Systems Division/XR,
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability	Electronic Systems Division/XR,
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability DEWLine	Electronic Systems Division/XR,
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability	Electronic Systems Division/XR,
Approved for public release; distribution units 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability DEWLine Life Cycle Cost	Electronic Systems Division/XR,
Approved for public release; distribution unline 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 19. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block Dewline Cost Continue on reverse side if necessary and identify by block Dewline Cycle Cost	Electronic Systems Division/XR, k number)
Approved for public release; distribution unline 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 19. KEY WORDS (Continue on reverse side if necessary and identify by block UAR — Unattended Radar Surveillance MAR — Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block The volume presents the Life Cycle Cost analy.	Electronic Systems Division/XR, k number) e
Approved for public release; distribution unline 17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 19. KEY WORDS (Continue on reverse side if necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal	Electronic Systems Division/XR, k number) e number) ses of alternative radar design ly satisfy requirements for un-
Approved for public release; distribution unline 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 19. KEY WORDS (Continue on reverse side if necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal attended and minimally attended radar operations.	Electronic Systems Division/XR, number) e number) ses of alternative radar design ly satisfy requirements for un- ons. The requirements, detailed
Approved for public release; distribution unline 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 19. KEY WORDS (Continue on reverse side if necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal attended and minimally attended radar operation the study statement of work, consist of two	Electronic Systems Division/XR, number) e number) ses of alternative radar design ly satisfy requirements for un- ons. The requirements, detailed o sets of nominal radar performa
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side 11 necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side 11 necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal attended and minimally attended radar operation the study statement of work, consist of two parameter goals; one set for the unattended reserved.	Electronic Systems Division/XR, number) e number) ses of alternative radar design ly satisfy requirements for un- ons. The requirements, detailed o sets of nominal radar performa adar (UAR) and a second for the
Approved for public release; distribution unline 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 ditts Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side if necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side if necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal attended and minimally attended radar operation the study statement of work, consist of two parameter goals; one set for the unattended reminimally attended radar (MAR). Based on these	Electronic Systems Division/XR, number) e number) ses of alternative radar design ly satisfy requirements for un- ons. The requirements, detailed o sets of nominal radar performa adar (UAR) and a second for the se requirements, alternative rada
Approved for public release; distribution unline. 7. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, 11 diff. Same 18. SUPPLEMENTARY NOTES RADC Project Engineer: Adrian S. Briggs (OCDE) The effort reported here was sponsored by the Hanscom Air Force Base MA 01731. 9. KEY WORDS (Continue on reverse side 11 necessary and identify by block UAR - Unattended Radar Surveillance MAR - Minimally Attended Radar Reliability DEWLine Life Cycle Cost 10. ABSTRACT (Continue on reverse side 11 necessary and identify by block The volume presents the Life Cycle Cost analy approaches that could be developed to optimal attended and minimally attended radar operation the study statement of work, consist of two parameter goals; one set for the unattended reserved.	Electronic Systems Division/XR, number) ses of alternative radar design ly satisfy requirements for un- ons. The requirements, detailed o sets of nominal radar performa adar (UAR) and a second for the se requirements, alternative rada AR and MAR, and evaluated for

SECURITY CLASSIFICATION OF THIS PAGE(When Date Entered)

The radar designs recommended for UAR nad MAR displayed the greatest potential for optimally satisfying all stated requirements. These designs are detailed in Volume II of the report. The UAR, a 60 nmi 2D radar, automatically outputs target track data that can be remoted to manned logistics nodes and/or ROCCs via narrowband communications links. It provides all altitude surveillance coverage of aircraft targets between 100 and 100,000 feet, and is configurable to be sufficiently reliable to guarantee, to a >90 percent confidence level, failure-free system operation for periods of time from three months to one year. The MAR, a 200 nmi 3D radar, also automatically outputs target track data to logistics nodes and/or ROCCs. It provides all altitude 3D coverage of aircraft targets between 100 and 100,000 feet, and is configurable to be sufficiently reliable to guarantee, to a > 90 percent confidence level, failure-free system operation for periods of time from five days to 0.5 month.

Volume I provides an Executive Summary of the Unattended/Minimally Attended Radar Study. Volume II presents alternative approaches investigated and details the radar designs recommended. Volume III presents data estimating radar acquisition and Life Cycle costs.

TABLE OF CONTENTS

SECTION		PAGE
1.0	Introduction	1
2.0	Life Cycle Cost (LCC)	1
2.1	UAR Life Cycle Cost (LCC) Analysis	i
2.1.1	Cost Elements Considered	2
2.1.2	Configuration Alternatives Considered	2
2.2	Program Variables	3
2.2.1	Customer Factors	4
2.2.2	Contractor Factors	4
2.3	Maintenance and Support Philosophy	4
2.3.1	On-Site Maintenance	8
2.3.2	Node Maintenance	9
2.3.3	Depot Maintenance	9
2.3.4	Maintenance and Support Manning	9
2.4	Equipment Installation	9
2.4.1	Installation Planning	10
2.4.2	UAR Installation Sequence and Schedule	10
2.4.3	Installation Support	12
2.5	LRU Stockage Model	13
2.5.1	Support Variations	13
2.5.2	Initial Spares	13
2.6	Life Cycle Cost Computation	13
2.6.1	Life Cycle Cost Summary	14
2.6.2	Acquisition Costs	14
2.6.3	Initial Logistic Cost	17
2.6.4	Recurring Logistic Cost	19
3.0	MAR Life Cycle (LCC) Costs	21
3.1	MAR Maintenance Philosophy	21
3.1.1	Maintenance Structure	21
3.1.2	Organizational Maintenance	21
3.1.3	Depot Maintenance	22
3.1.4	Maintenance Manning	23
3.2	Built-In Test Equipment (BITE) Concept	23
3.2.1	BITE Objectives	23
3.2.2	System Status Panels	24
3.2.3	Subsystem Status Panel	25
3.2.4	Display Console	25

TABLE OF CONTENTS (Cont'd)

SECTION		PAGE
3.3	Supportability	25
3.3.1	Modularity	27
3.3.2	Commonality	27
3.3.3	Repairability	27
3.3.4	Maintainability	28
3.3.5	Reliability	28
3.3.6	Test Equipment	28

Vancound ten? St. 23 (25)

St. 25 (25)

St.

EVALUATION

The effort reported is one of three parallel study contracts performed under Project E233 by direction of ESD/XR. These reports identify alternative concepts and activity necessary to support the development of a short-range, unattended radar and a long-range minimally attended radar. The short-range radar is being viewed for application in DEW Line to replace the AN/FPS-19 and the long-range radar is being viewed for application by the Alaskan Air Command to replace the AN/FPS-93. These studies provide the assurance that current technology can support the development of unattended/minimally attended radars that offer improved performance and can significantly reduce operating and maintenance costs.

These efforts were performed in accordance with 1978-1982 TPO III,

Thrust C Advanced Sensor Technology. The results will be used by ESD to

develop system acquisition strategy for SEEK FROST (Project 2448), PE

12412F. It also provides supplemental data supporting SEEK IGL00 (Project 968H), PE 12325F.

adrian S. Briggs ADRIAN S. BRIGGS

Project Engineer

1.0 INTRODUCTION

This volume discusses life cycle cost (LCC) analyses and data pertaining to a short range, 2-dimension (2D) unattended radar (UAR) and a long-range, 3-dimension (3D) minimally attended radar (MAR).

LCC emphasis has been placed on the UAR since it represents an extreme departure from current radar support philosophies and involves a unique design approach. During evolution of the UAR design, LCC analyses were performed on design alternatives utilizing two computer math models: The RCA PRICE model and the ITTG OPSTOCK/CONSUME model. The PRICE model was used to assess the radar's acquisition costs and the ITTG model was used to assess logistics effects of the designs. Results of these analyses are combined in this LCC study.

The UAR LCC data furnished is limited to that indicated by the study statement-of-work (SOW). The SOW indicated that personnel logistics, power requirements, transportation (surface and air), and disposal should not be included in the analysis. Reference has been made to some of these items, but they are not included as cost elements.

The LCC data was developed, in part, by cost estimating relationships and, in part, by direct pricing analysis of the specific program elements. The derivation of system reliability is documented in Volume II. Unit level reliability data are based on unit planning documents using MIL-STD approaches. Maintenance and personnel requirements were obtained from the math models and maintenance engineering analysis of the equipment in its intended operational environment.

The LCC analysis data on the MAR is based on concepts and techniques employed on a current ITTG product line radar which is optimized for low life cycle costs.

2.0 LIFE CYCLE COST (LCC)

2.1 UAR Life Cycle Cost (LCC) Analysis

A modified version of the standard LCC algorithm is used to describe the life cycle cost aspects of the UAR program.

$$LCC = A_T + I_L + \sum_{i=1}^{N} RL$$

where:

AT = Acquisition Cost Total

I_I = Initial Logistics Cost

R_L = Recurring Logistic Cost (Annual)

N = Number of Years in the Life Cycle.

2.1.1 Cost Elements Considered

The cost elements considered in the LCC analysis consist of the following:

- Acquisition Cost Total
 - Research and Development
 - Prototype Development
 - Performance, Environmental, Reliability Testing
 - Production
- Initial Logistics Cost
 - Spares
 - Inventory Entry
 - Personnel
 - Training
 - Tools and Test Equipment
 - Technical Documentation
 - Installation and Checkout
- Recurring Logistic Cost (Annual)
 - Spares Replenishment
 - Inventory Management
 - Personnel
 - Training
 - Depot Maintenance Manhours
 - Maintenance Administration
 - Tools and Test Equipment Maintenance
 - Preventive Maintenance

2.1.2 Configuration Alternatives Considered

The UAR study considered three reliability configurations which are defined by reliability data furnished in Volume II, and the equipment breakdown and LCC data listing furnished in Section 2.2.2 of this volume. The three configurations are:

- Three-month system where the probability of zero failures in a three-month interval is 0.931.
- Six-month system where the probability of zero failures in a six-month interval is 0.908.
- And, a twelve-month system where the probability of zero failures in a twelvemonth interval is 0.909.

The LCC calculations are given in Paragraph 2.6. The three-month system has merit not only on the basis of lower life cycle costs but also because it more effectively utilizes the support organization. As indicated by Figure 2.1-1, higher reliability configurations make fewer demands on the support structure, particularly in relationship to on-site maintenance requirements. If there is a problem associated with high reliability, it is the resulting inactivity of maintenance personnel which causes boredom and gradual degradation of skills. The three-month system is not a completely effective solution to this problem, but does improve personnel utilization.

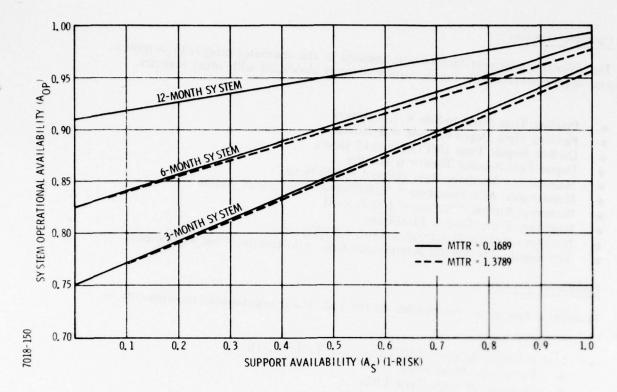


Figure 2.1-1. Support Availability (As) (1-Risk)

Figure 2.1-1 graphically depicts how operational availability (Aop) is achieved through a combination of reliability and support availability. The higher reliability equipments require less external support to achieve availability objectives for short term. However, the use of redundancy to achieve high reliability increases the amount of hardware, which translates into higher acquisition costs and more hardware to support.

On the graph in Figure 2.1-1, the Aop vs A_s curves for the three configurations are shown with solid lines representing the MTTR exclusive of travel time to the site, while dashed lines include the 1.21-hour mean transportation time. Assuming an Aop objective of 0.9 and the inclusive travel time MTTR, A_s is 0.75 for the 3-month system and 0.5 for the 6-month system. When this is translated into risk for shortage of 0.25 and 0.5 respectively, it can be used to estimate the cost of initial spares required. (Reference Figure 2.1-5.)

2.2 Program Variables

The LCC computation is based on the following program variables:

- Number of Years in Life Cycle = 20 Years
- Number of Operating Systems = 80 Systems
- Operating Time per System = Full Time = 168 Hours/Week
- Operational Availability (Aop) Objective ≥ 90 Percent

2.2.1 Customer Factors

The following customer factors were inserted in the computer models to determine support spare requirements and in the derivation of costs associated with other logistics elements:

- Pipeline Time Node-to-Site = 1.21 Hours
- Pipeline Time Depot-to-Node = 1 Month
- On-Site Repair Time (MTTR) = 0.17 Hours
- Depot Turn-Around Time = 6 Months
- Maintenance Manhour Rate (Depot) = \$20/Hour
- Maintenance Administration = \$250/Depot Maintenance Action
- Personnel Annual Attrition = 100 Percent
- Inventory Entry Cost = \$100/Item
- Inventory Annual Maintenance Cost = \$100/Item
- Test Equipment Annual Maintenance Cost = Acquisition Cost x 5 Percent

2.2.2 Contractor Factors

Contractor factors are correlatable to the LRU's and maintenance requirements as follows:

- Line Replaceable Unit (LRU) Data (see Table 2.2-1).
 - Equipment Breakdown
 - Quantity of Each Type LRU
 - LRU Failure Rate
 - Node NRTS (Not-Repairable-This-Station) Rate
 - Depot NRTS Rate
 - On-Line MTTR
 - Off-Line MTTR
 - Unit Price
 - Cost-to-Repair
- Personnel and Skill Requirements
- Preventive Maintenance Requirements
- Tools and Test Equipment Requirements
- Codification Requirements

2.3 Maintenance and Support Philosophy

To support 80 unattended operating locations, a network of intermediate repair locations (nodes) and a single depot is recommended. Six nodes will be located at existing military sites which can accommodate airlift, as well as helicopter operations. A single new node site is required.

TABLE 2.2-1. EQUIPMENT ASSEMBLY BREAKDOWN AND LIFE CYCLE COST DATA LISTING SYSTEM NAME: UAR — UNATTENDED SHORT RANGE RADAR, SURVFILLANCE, 2D

					Quantity	ıtity		Failure	NRTS	NRTS Rate	A	MTTR	Price Data)ata
Item	Part	Ident		3 Mo	0	6 Mo.	12 Mo	Rate					Unit	RPR
No.	Number	2	Description	Active	Stdby	Stdby	Stdby	106	Node	Depot	On Line	Off Line	Price	Cost
10	00009	-	Radar Set, UAR	1										
05	61000	2	Antenna Group	-	0	0	0	0.842	ı	1	1	1	136,625.00	1
03	61100	es	Antenna Assy	3	0	0	0	0.001	1	1	1	1	42,275.00	1
8	61110	4	Antenna Array	12	0	0	0	0.001	1	1	1	1	2,000.00	1
90	61120	4	Antenna Reflector, Dual	ဗ	0	0	0	0.001	i	1	1	1	20,775.00	1
90	61150	4	Diplexer Radar/IFF	12	0	0	0	0.001	1	1	1	1	750.00	1
07	61200	9	Antenna Array, IFF	12	0	0	0	0.001	1	1	i	1	750.00	1
88	61300	3	Antenna Assy, ONMI	-	0	0	0	0.001	1	1	1	ı	800.00	ı
8	62000	2	Microwave Group	-	0	0	0	0.800	1	1	Ī	1	40,813.00	1
10	62100	8	Waveguide	-	0	0	0	0.001	1	1	1	1	31,725.00	1
=	62150	4	· Drier Assy	13	0	0	0	0.001	1.00	1.00	0.1	0.0	30.00	30.00
12	62200	8	Switch Driver Assy	-	0	0	0	0.619	0.10	0.15	0.1	6.0	7,338.00	75.00
13	62300	8	*Switch Module R.F.	13	0	0	0	0.157	0.10	0.01	0.2	1.1	100.00	00.09
14	62400	8	Diplexer IFF/ECM OMNI	-	0	0	0	0.00	1.00	1.00	0.5	0.0	450.00	450.00
15	63000	2	Equipment Mounting GP	-	0	0	0	0.425	1	1	1	1	16,963.00	1
16	63100	0	Mounting Assy	-	0	0	0	0.010	1	ī	1	ı	2 500.00	1
17	63200	8	Cabling Installation	-	0	0	0	0.317	1	1	1	ı	9,463.00	1
18	64000	2	Radar Transmitter	,	0	0	0	2.152	0.00	0.00	0.3	0.0	54,562.00	1,050.00
19	64100	9	Cabinet Transmitter	-	0	0	0	0.131	1	1	1	1	812.00	1
20	64200	8	*Amplifier Medule RF	22	2	2	2	1,915	0.20	0.05	0.1	2.0	750.00	140.00
21	64300	8	Pwr Divider Preamp, 3 Way	-	0	0	0	0.001	1.00	0.10	0.4	2.0	1,500.00	75.00
22	64400	8	Pwr Combiner, Preamp, 3 Way	-	0	0	0	900'0	1.00	0.10	0.4	2.0	1,500.00	75.00
23	64410	3	Monitor Assy	-	0	0	0	0,259	0.10	0.01	0.1	1,0	13,500.00	20.00
24	64500	8	Power Divider, 24 Way	-	0	0	0	0,001	1.00	0.10	0.4	2.0	8,500.00	200.00
25	64600	8	Power Combiner, 24 Way	-	0	0	0	900'0	1.00	0.10	0.4	2.0	8,500.00	200.00
56	65000	7	· Radar Receiver, R.F.	2	0	0	-	1,505	0.01	0.0	9.0	1.1	4,994.00	1,150.00
27	65050	8	Cabinet Radar Rec, R.F.	7	0	0	-	0.001	1	1	ı	1	400.00	ı
28	65100	9	Limiter STC, Substrate	7	0	0	-	0.280	0.5	0:0	0.0	2.0	1,094.00	62.00
53	65200	8	RF Preamp, Substrate	2	0	0	-	0.205	0.20	0.10	0.0	2.0	1,500.00	84.00
30	65300	8	Mixer Assy, RF	7	0	0	-	0.200	0.20	1.00	1.0	1.0	1,250.00	20.00
31	65400	8	Lin-Log Amplifier	2	0	0	-	0.715	06.0	0.10	0.0	2,5	750.00	62.00
32	65500	2	*Radar Receiver, IF	-	0	0	-	1,234	0.01	0.0	0.5	1.0	5,232.00	1,500.00
33	65550	8	Cabinet Radar Rec, 1F	-	0	0	-	0.001	1	1	1	1	400.00	1
¥	65600	6	IF Amplifier Substrate	-	0	0	-	0.378	0.95	0.10	0.0	2.0	1,500.00	00.49
35	65700	8	Chirp Line Assy	-	0	0	-	0.524	06'0	0.15	0.0	1.2	1,750.00	67.00
36	65800	8	Quad Detector Assy	-	0	0	-	0.241	06'0	0.05	0.0	2.5	1,582.00	25.00
37	00099	2	Freq Synthesizer	-	0	-	-	4.280	00'0	0.00	0.3	0.0	10,371.00	3,200.00
38	66100	3	*Course Loop Assy	-	0	-	-	2.400	0.10	0.20	9.0	1.1	3,600.00	1,040.00
39	66110	4	Enclosure, Course Loop	-	0	-	-	0.001	1	1	ı	1	400.00	1
														-

TABLE 2.2-1. EQUIPMENT ASSEMBLY BREAKDOWN AND LIFE CYCLE COST DATA LISTING
SYSTEM NAME: UAR UNATTENDED SHORT RANGE RADAR, SURVEILLANCE, 2D (Conf'd)

Name									all die		AL CITAL			LICE Data	Cara
Mumber Log Description Active Sidby Sidby 106 Note Depot 1 On Line Off Line	tem	Part	Ident		3 N	10	6 Mo	12 Mo	Rate					Unit	RPR
66120 4 Power Splitter Driver 1 0 1 1 0.200 0.00 0.00 2.6 6.6	No.	Number	7	Description	Active	Stdby	Stdby	Stdby	106	Node	Depot	On Line	Off Line	Price	Cost
66130 4 Loop Filter Assy 1 0.800 100 0.80 0.0 2.6 66200 3 Fine Loop Assy 1 0 1 1 0.800 10 0.8 0.5 1.1 3 66200 4 Programmable Divider 1 0 1 1 0.000 1.0 0.8 0.0 2.6 66200 4 Mixer Asy, Loop 1 0 1 1 0.00 0.0 0.5 1.1 3 66230 4 Programmable Divider 1 0 1 1 0.00 0.0 2.6 1.1 0 0.0 2.6 0.0 2.6 1.5 0 0.0 2.6 0.0 2.6 0.0	04	66120	4		-	0	1	1	0.200	0.70	0.01	0.0	2.5	1,050.00	75.00
66210 4 Programmable Divider 1 1 1 1 1 1 1 1 1	41	66130	4	Loop Filter Assy	-	0	-	-	0.800	1.00	0.80	0.0	2.6	1,030.00	80.00
66200 3 Fine Loop Assy 1 1 2,000 0.10 0.2 0.5 1.1 3 6,620 4 Fine Loop Assy 1 1 2,000 0.10 0.2 0.5 1.1 3 66220 4 Loop Fine Assy 1 0.000 0.2 1.0 0.0 2.5 1.1 3 66230 3 4 Loop Fine Assy 1 0.000 1.00 0.00 0.0 2.5 1.1 3 66230 3 Frequency Multiplier 1 1 0.000 0.0 0.0 0.0 2.5 1.5 3 5 3 5 3 5 3 4 F. Multiplier 1 1 0.000 0.0 0.0 2.5 0.0<	42	66140	4	Programmable Divider	-	0	-	-	0.800	1.00	0.80	0.0	2.5	1,120.00	90.00
68220 4 Enclosure Fine Loop 1 0.001 —<	43	66200	3	* Fine Loop Assy	-	0	-	-	2.000	0.10	0.20	0.5	1.1	3,471.00	1,000.00
68220 4 Mixer Asy, Loop 1 1 0.300 0.25 1 68220 4 Programmable Divider 1 1 0.300 0.20 0.26 68230 4 Programmable Divider 1 1 0.800 1.00 0.00 2.5 1 68310 4 Programmable Divider 1 0 1 1 0.800 1.00 0.05 0.5 0	4	66210	4	Enclosure, Fine Loop	-	0	-	-	0.001	1	1	1	1	400.00	1
66230 4 Loop Filter Assy 1 0.800 1.00 2.6 66230 4 Programmable Divider 1 0 1 1 0.800 1.00 2.6 66300 3 Frequency Mutiplier 0 1 1 0.800 0.0 0.0 2.5 3.5 66310 4 Reference Gen 1 0	45	66220	4	Mixer Assy, Loop	-	0	-	-	0.300	0.20	1.00	0.0	2.5	1,000.00	64.00
68240 4 Programmable Divicier 1 1 0.800 100 0.89 0.00 2.5 1 66310 4 Prequency Multiplier 1 0.001 0.001 0.00 0.05 0.0	9	66230	4	Loop Filter Assy	-	0	-	-	0.800	1.00	0.80	0.0	5.6	995.00	80.00
65300 3 Frequency Multiplier 1 2,060 0,40 0,05 0,5 3,5 3,6 65300 4 Frequency Multiplier 1 0,01 1 1,00 0,00 0,0 0,5 0,0 3,2 4 65300 4 Fleterence Gen 1 0,00 0,0 0,0 0,0 0,0 3,2 4 65300 2 A COHO Amplifier 1 0,00 0,0	47	66240	4	Programmable Divider	-	0	-	-	0.800	1.00	0.80	0.0	2.5	1,076.00	90.00
68310 4 Enclosure, Multiplier 1 0.0001 — <th< td=""><td>8</td><td>96300</td><td>3</td><td>* Frequency Multiplier</td><td>-</td><td>0</td><td>1</td><td>-</td><td>2.060</td><td>0.40</td><td>0.05</td><td>0.5</td><td>3.5</td><td>3,300.00</td><td>1,075.00</td></th<>	8	96300	3	* Frequency Multiplier	-	0	1	-	2.060	0.40	0.05	0.5	3.5	3,300.00	1,075.00
68320 4 Reference Gen 1 0 1 1 0860 0.20 0.05 <th< td=""><td>49</td><td>66310</td><td>4</td><td>Enclosure, Multiplier</td><td>-</td><td>0</td><td>-</td><td>-</td><td>0.001</td><td>1</td><td>i</td><td>1</td><td>1</td><td>400.00</td><td>1</td></th<>	49	66310	4	Enclosure, Multiplier	-	0	-	-	0.001	1	i	1	1	400.00	1
66330 4 IF Multiplier 1 0 600 0.40 0.05 0.03 3.2 66340 2 COHO Amplifier 1 0 1 1 6500 0.50 0.05 0.05 0.03 3.4 66410 3 Enclosure Radar Exciter 1 0 0 1 6500 0.50 0.00 0.01 0	00	66320	4	Reference Gen	-	0	-	-	0.860	0.20	0.05	0.0	2.5	1,375.00	92.00
66240 4 COHO Amplifier 1 1 0500 0.55 0.05 <	15	66330	4	IF Multiplier	-	0	-	-	0.600	0.40	0.05	0.0	3.2	970.00	64.00
68400 2 Radar Exciter 1 1,682 0.10 0.01 0.3 1,2 6 68410 3 Pulse Modulator 1 0 0 1 0,001 - - - - - - 6 -	25	66340	4	COHO Amplifier	-	0	-	-	0.500	0.50	0.05	0.0	3.4	555.00	53.00
66410 3 Enclosure Radar Exciter 1 0.001 - - - 66420 3 Pulse Modulator 1 0.500 0.560 0.56 0.05	53	66400	2	* Radar Exciter	-	0	0	-	1.682	0.10	0.01	0.3	1.2	5,407.00	1,710.00
66420 3 Pulse Modulator 1 0.500 0.50 0.05	Z	66410	3	Enclosure Radar Exciter	-	0	0	-	0.001	1	1	1	1	400.00	1
66430 3 Chirpline Assy 1 0 0 1 0.500 0.90 0.25 0.0 2.5 66440 3 Miscer- Driver 1 0 0 0.500 0.90 0.25 0.0 0.0 67200 3 Forecer Drivder IFF Trans 1 0 0 0 0.01 0.0 0	35	66420	က	Pulse Modulator	-	0	0	-	0.500	0.50	0.05	0.0	2.5	1,600.00	45.00
66440 3 Mixer - Driver 1 0 0 0.562 0.15 0.01 0.0 2.0 67200 2 • IF Transmitter 1 0 0 0 0.01 0.01 0.0 0 67210 3 Amplifier Asy IFF 1 0 0 0 0.01 0.0 0<	99	66430	က	Chirpline Assy	-	0	0	-	0.500	0.90	0.25	0.0	2.5	1,600.00	45.00
67200 2 IFF Transmitter 1 0 0 2.351 0.01 0.01 0.05 1.0 10 67210 3 Enclosure IFF Trans 1 0 0 0.076 — — — — 67220 3 Power Divider IFF 1 0 0 0.076 —	22	66440	က	Mixer - Driver	-	0	0	0	0.582	0.15	0.01	0.0	2.0	1,807.00	120.00
67210 3 Enclosure IFF Trans 1 0 0 0 0.076 -<	8	67200	2	* IFF Transmitter	1	0	0	0	2.351	0.01	0.01	0.5	1.0	10,888.00	1,130.00
67220 3 Amplifier Assy IFF 3 0 0 0.729 0.10 0.01 0.0 0.0 0.729 0.10 0.01 0.0 0	66	67210	က	Enclosure IFF Trans	-	0	0	0	0.076	1	1	1	1	00.009	1
67230 3 Power Divider IFF 1 0 0 0.001 1.00 0.10	8	67220	က	Amplifier Assy IFF	က	0	0	0	0.729	0.10	0.01	0.0	2.0	820.00	100.00
67240 3 Power Combiner IFF 1 0 0 0.006 1.00 0.10 0.10 0.10 1.0 1 67250 3 Monitor Assy IFF 1 0 0 0 0.086 0.25 0.05 0.0 1.0 4 67300 2 *IFF Receiver 1 0 0 1 0.086 0.25 0.05 0.0 1.0 4 67320 3 Limitan First L.O. 1 0 0 1 0.001 0.0 0.0 0.0 0.0 0.0 0.0 0 </td <td>15</td> <td>67230</td> <td>က</td> <td>Power Divider IFF</td> <td>-</td> <td>0</td> <td>0</td> <td>0</td> <td>0.001</td> <td>1.00</td> <td>0.10</td> <td>0.0</td> <td>1.0</td> <td>1,750.00</td> <td>70.00</td>	15	67230	က	Power Divider IFF	-	0	0	0	0.001	1.00	0.10	0.0	1.0	1,750.00	70.00
67250 3 Monitor Asy IFF 1 0 0 0.086 0.25 0.05 0.0 1.0 4 67300 2 • IFF Receiver 1 0 0 1 2.248 0.01 0.00 0.5 0.10 0 0 67310 3 Limiter - Filter 1 0 0 1 0.001 - <t< td=""><td>32</td><td>67240</td><td>က</td><td>Power Combiner IFF</td><td>-</td><td>0</td><td>0</td><td>0</td><td>900'0</td><td>1.00</td><td>0.10</td><td>0.0</td><td>1.0</td><td>1,750.00</td><td>70.00</td></t<>	32	67240	က	Power Combiner IFF	-	0	0	0	900'0	1.00	0.10	0.0	1.0	1,750.00	70.00
67300 2 • IFF Receiver 1 0 0 1 2.248 0.01 0.00 0.5 1.0 5 67310 3 Enclosure IFF Receiver 1 0 0 1 0.001 —	n	67250	က	_	-	0	0	0	0.086	0.25	0.05	0.0	1.0	4,238.00	83.00
67310 3 Enclosure IFF Receiver 1 0 0 1 0.001 — <th< td=""><td>X</td><td>67300</td><td>7</td><td>* IFF Receiver</td><td>1</td><td>0</td><td>0</td><td>-</td><td>2.248</td><td>0.01</td><td>0.00</td><td>0.5</td><td>0.1</td><td>5,157.00</td><td>1,400.00</td></th<>	X	67300	7	* IFF Receiver	1	0	0	-	2.248	0.01	0.00	0.5	0.1	5,157.00	1,400.00
67320 3 Limiter - Filter 1 0 0 1 0.828 1.00 1.00 0.0 1 67330 3 FF Amplifier - L.O. 1 0 0 1 0.800 0.05 0.05 0.0 3.0 1 6740 3 AGC - IF Amp 1 0 0 1 0.800 0.05 0.05 0.0 2.5 1 6740 2 IF Exciter 1 0 0 1 0.836 0.10 0.01 0.25 0.12 0.0 2.5 1 6740 3 Pulse Generator 1 0 0 1 0.250 0.05 0.01 0.0 2.5 1 6740 3 Pulse Generator 1 0 0 1 0.250 0.05 0.0 0.0 2.5 1 6740 3 Driver 1 0 0 1 0.250 0.00 0.0 0	35	67310	က		1	0	0	-	0.001	1	1	1	1	400.00	1
67330 3 RF Amplifier - L.O. 1 0 0 1 0.800 0.30 0.05 0.05 3.0 1 67340 3 AGC - IF Amp 1 0 0 1 0.700 0.05 0.05 0.05 0.0 2.5 1 67400 2 IF Exciter 1 0 0 1 0.03 0.01 0.0 2.5 1 67410 3 Fulse Generator 1 0 0 1 0.05 0.01 0.0 2.5 1 67420 3 Pulse Mcdulator 1 0 0 1 0.250 0.01 0.0 2.5 1 67440 3 Drivise Mcdulator 1 0 0 1 0.250 0.05 0.05 0.0 2.5 1 68100 2 1 0 0 1 0.23 0.0 0.0 0 0 0 0 0 0 <td>98</td> <td>67320</td> <td>က</td> <td></td> <td>-</td> <td>0</td> <td>0</td> <td>-</td> <td>0.828</td> <td>1.00</td> <td>1.00</td> <td>0.0</td> <td>0.0</td> <td>1,700.00</td> <td>1,700.00</td>	98	67320	က		-	0	0	-	0.828	1.00	1.00	0.0	0.0	1,700.00	1,700.00
67340 3 AGC - IF Amp 1 0 0 1 0.700 0.20 0.15 0.0 2.5 1 67400 2 • IF Exciter 1 0 0 1 0.836 0.10 0.01 0.3 1.2 4 67410 3 Fulse Generator 1 0 0 1 0.250 0.05 0.01 0.0 2.5 1 67430 3 Pulse Modulator 1 0 0 1 0.250 0.05 0.05 0.01 0.0 2.5 1 67440 3 Driver 1 0 0 1 0.256 0.05 0.01 0.0 2.5 1 1 0.0 0	22	67330	က		-	0	0	-	0.800	0.30	0.05	0.0	3.0	1,640.00	28.00
67400 2 • IFF Exciter 1 0 0 1 0.836 0.10 0.01 0.3 1.2 4 67410 3 Enclosure, IFF Exciter 1 0 0 1 0.001 —	92	67340	က	AGC - IF Amp	-	0	0	-	0.700	0.20	0.15	0.0	2.5	1,417.00	20.00
67410 3 Enclosure, IFF Exciter 1 0 0 1 0.001 — <th< td=""><td>8</td><td>67400</td><td>7</td><td>* IFF Exciter</td><td>-</td><td>0</td><td>0</td><td>-</td><td>0.836</td><td>0.10</td><td>0.01</td><td>0.3</td><td>1.2</td><td>4,407.00</td><td>1,310.00</td></th<>	8	67400	7	* IFF Exciter	-	0	0	-	0.836	0.10	0.01	0.3	1.2	4,407.00	1,310.00
67420 3 Pulse Generator 1 0 0 1 0.250 0.05 0.01 0.0 2.5 67430 3 Pulse Mcdulator 1 0 0 1 0.250 0.30 0.05 0.0 2.5 67440 3 Driver 1 0 0 1 0.256 0.30 0.05 0.0 2.5 68100 2 Ander Signal Processor 1 0 0 91.756 0.00 0.0 0.0 2.0 68110 3 Enclosure, Wired 1 0 0 0 91.756 0.00 0.0 0.0 6.0 68120 3 A/D Converter - Encoder 1 1 1 3.976 0.30 0.01 0.2 2.5 68130 3 Central Timing Unit 1 1 1 1 5.476 0.20 0.05 0.2 2.5	0	67410	က	Enclosure, IFF Exciter	-	0	0	-	0.001	ı	1	1	1	400.00	1
67430 3 Pulse Modulator 1 0 0 1 0.250 0.30 0.05 0.0 2.5 67440 3 Driver 1 0 0 1 0.256 0.20 0.01 0.0 2.0 68100 2 Radar Signal Processor 1 0 0 0 91.750 0.00 0.0 0 0 68110 3 Enclosure, Wired 1 0 0 0 5.787 - - - - 68120 3 A/D Converter - Encoder 1 1 1 3,438 0,40 0.25 0.3 3.0 68130 3 * Central Timing Unit 1 1 1 1 3,478 0,20 0.05 0.2 2.5 68140 3 * Control/Mon - Proc Interface 1 1 1 5,476 0.20 0.05 0.2 2.5	-	67420	က	Pulse Generator	-	0	0	-	0.250	0.05	0.01	0.0	2.5	1,260.00	49.00
67440 3 Driver 1 0 0 1 0.236 0.20 0.01 0.0 2.0 68100 2 0 0 0 91.750 0.00 0.3 0.0 6.0 6.0 0 <td>12</td> <td>67430</td> <td>က</td> <td>Pulse Modulator</td> <td>-</td> <td>0</td> <td>0</td> <td>-</td> <td>0.250</td> <td>0.30</td> <td>0.05</td> <td>0.0</td> <td>2.5</td> <td>1,330.00</td> <td>23.00</td>	12	67430	က	Pulse Modulator	-	0	0	-	0.250	0.30	0.05	0.0	2.5	1,330.00	23.00
68100 2 Radar Signal Processor 1 0 0 0 1750 0.00 0.03 0.0 68110 3 Enclosure, Wired 1 0 0 0 5,787 - - - - 68120 3 *A/D Converter - Encoder 1 1 1 3,438 0,40 0,25 0,3 3,0 68130 3 *Central Timing Unit 1 1 1 1 5,476 0,20 0,05 0,2 2.0 68140 3 *Control/Mon - Proc Interface 1 1 1 5,476 0,20 0,05 0,2 2.5	3	67440	8	Driver	-	0	0	-	0.236	0.20	0.01	0.0	2.0	1,417.00	54.00
68110 3 Enclosure, Wired 1 0 0 0 5.787 – – – – – 68120 3 * A/D Converter - Encoder 1 0 1 1 3,438 0,40 0,25 0,3 3.0 4, 68130 3 * Central Timing Unit 1 1 1 1 3,976 0,30 0,01 0,2 2.0 1, 68140 3 * Control/Mon - Proc Interface 1 1 1 1 5,476 0,20 0,05 0,2 2.5 3,	4	68100	7	Radar Signal Processor	-	0	0	0	91,750	0.00	0.00	0.3	0.0	64,587.00	3,260.00
68120 3 * A/D Converter - Encoder 1 0 1 1 3,438 0.40 0.25 0.3 3.0 68130 3 * Central Timing Unit 1 1 1 1 3,976 0.30 0.01 0.2 2.0 68140 3 * Control/Mon - Proc Interface 1 1 1 5.476 0.20 0.05 0.2 2.5	2	68110	က	Enclosure, Wired	-	0	0	0	5.787	1	1	1	1	900.00	1
68130 3 *Central Timing Unit 1 1 1 3.976 0.30 0.01 0.2 2.0 68140 3 *Control/Mon-Proc Interface 1 1 1 5.476 0.20 0.05 0.2 2.5	9	68120	8		-	0	-	-	3,438	0.40	0.25	0.3	3.0	4,820.00	253.00
68140 3 *Control/Mon-Proc Interface 1 1 1 1 5.476 0.20 0.05 0.2 2.5	11	68130	8	* Central Timing Unit	-	-	-	-	3.976	0.30	0.01	0.2	2.0	1,930.00	20.00
	8	68140	3	* Control/Mon - Proc Interface	-	-	-	-	5.476	0.20	0.05	0.2	2.5	3,040.00	32.00

TABLE 2.2-1. EQUIPMENT ASSEMBLY BREAKDOWN AND LIFE CYCLE COST DATA LISTING SYSTEM NAME: UAR UNATTENDED SHORT RANGE RADAR, SURVEILLANCE, 2D (Cont'd)

					Qua	Quantity		Failure	NRTS	NRTS Rate	LM	MTTR	Price Data	Data
Item	Part	Ident		3 Mo	No	6 Mo	12 Mo	Rate					Unit	RPR
No.	Number	2	Description	Active	Stdby	Stdby	Stdby	106	Node	Depot	Depot On Line	Off Line	Price	Cost
-	68150	3	Comm Interface Unit	-	-	-	-	4.419	0.05	0.01	0.2	2.6	3,225.00	34.00
_	68160	8		80	-	2	8	6.344	0.10	0.01	0.2	2.1	4,630.00	49.00
	68170	(7)		2	2	2	2	3.773	0.10	0.01	0.2	2.0	2,760.00	29.00
~	68180	0	* Correlation Detector	-	-	-	-	2.910	0.10	0.01	0.2	2.0	2,120.00	22.00
83	68190	3	• IFF Target Det Unit	-	0	0	-	3.342	0.15	0.01	0.2	2.0	2,430.00	26.00
	68191	6	* IFF Target Buffer	-	0	0	-	3.442	0.15	0.01	0.2	2.0	2,510.00	26.00
	68192	8	* Jamming Detector	1	0	0	-	0.662	0.10	0,01	0.2	1.5	1,052.00	11.00
"	68200	7	Data Processor	1	0	0	0	111.208	0.00	0.00	0.3	0.0	17,075.00	1,100.00
-	68210	8	Enclosure Wired	-	0	0	0	3.680	1	1	1	1	800.00	1
~	68220	8	* CPU Module No. 1	-	-	2	2	16.000	0.25	0.10	0.5	4.2	1,025.00	25.00
_	68230	8	• CPU Module No. 2	1	-	2	2	16.000	0.25	0.10	0.5	4.2	1,200.00	25.00
_	68240	8		-	-	2	2	15,000	0.25	0.10	0.5	4.2	850.00	25.00
	68250	3	* Programmed Memory Assy	4	4	4	4	6.464	0.10	0.01	0.5	3.0	1,500.00	15.00
~	68260	3		4	2	2	4	8.839	0.05	0.01	0.5	3.0	1,800.00	18.00
_	00069	7	Power Supply GP	1	-		-	0.532	0.00	0.00	9.0	0.0	16,513.00	7,200.00
_	69100	0	Inverter - Regulator	-	-	-	-	0.325	0.20	0.02	0.5	1.5	9,013.00	90.00
	69110	4	Transformer	4	4	4	4	0.007	1.00	1.00	0.0	0.5	1,750.00	1,750.00
	69120	4	* Rectifier Assy	-	-	-	-	0,160	0.05	0.01	0.0	8.0	2,013.00	45.00
	00000	~	* Requisitor Assy V	15	7	1,	15	0 101	0 15	0.05	20	11	500 00	00 6

·LRU

Each node will be a self-contained facility capable of supporting a permanently manned maintenance and supply environment. The following node assignments have been made:

Node Designator	Node Location	Number of Sites Supported		Mean Travel Time To Site
N1	Point Barrow	14		1.27
N2	Tuktoyaktuk	14		1.4
N3	Cambridge Bay	13		1.3
N4	Hall Beach	11		1.15
N5	Point Dyer	11		1.2
N6	Saglek (New Site)	8		0.96
<u>N7</u>	Goose Bay	9		1.26
7		80	System Mean	1.21

Generally, nodes are centrally located in the cluster of operational sites they serve. The average distance between stations is less than 50 nmi and the maximum distance from node to an operating site is 350 nmi. Helicopters, the prime means of node-to-site transportation, are assumed to have an operational range of approximately 500 nmi. UAR depot repair actions will be performed at existing facilities at McClellan Air For Base.

2.3.1 On-Site Maintenance

Demands for maintenance action will be initiated by on-line, built-in-test-equipment (BITE) and reported through the communications link to the node. At this time a maintenance team of two intermediate level electronic technicians and a power equipment technicians will travel by helicopter to the site. A complete complement of line replaceable unit (LRU) spares and assorted discrete parts will be stowed on the helicopter for use on site as required. The helicopter will also contain a complete set of common test equipment and tools necessary to perform organizational maintenance. No tools, test equipment or spares will be stored on-site.

The fault isolation process involves use of the on-line BITE system. The BITE design objectives are compatible with LRU packaging which is to isolate 95 percent or more of all system failures to an LRU module(s) concept. These objectives are:

- System level BITE to report system performance status
- Subsystem level BITE to detect a fault in a primary equipment element and to activate switching to backup equipments.
- BITE penetration to a single LRU for 95 percent of all system failures
- Maximum BITE penetration within each LRU to the shop replaceable unit (SRU) level to assist off-line repair activity.

At arrival on-site, the electronic technician(s) will check the subsystem status indicators and replace defective LRU's. All individual subsystem tests will be activated to determine the need to replace other LRU's which failed since the last visit causing a "failed item" report but which, because of redundancies, did not result in a "failed system" maintenance demand.

Before departing the site, technician(s) will perform all preventive maintenance (PM) normally scheduled for the near term to eliminate the need for the next scheduled visit if possible. The waveguide drier (desiccant) elements will be replaced at six-month intervals or during unscheduled visits.

2.3.2 Node Maintenance

All defective LRU's for a given complex of sites will be evacuated to the node for repair, disposal, or further evacuation to the depot. Each node serves as a repair and supply support center and will be fully equipped to perform those functions. A two-year supply of LRU's/SRU's and repair parts will be stocked at each node to support its complex of sites. The repair facility consists of a bench environment and will be equipped with tools, test equipment and test fixtures necessary to test and repair the LRU's.

2.3.3 Depot Maintenance

LRU's/SRU's not repairable at the node will be returned to the depot for repair or disposal. The recommended UAR places no special demands on the depot. Special tools and test equipment will be CFE furnished.

2.3.4 Maintenance and Support Manning

Nodes N1 and N2 which have the largest number of satellites, will each support 14 operating sites. Based on system failure rates and standard Poisson distribution (see Figure 2.3-1), the number of yearly demands from 14 operating sites will be less than one-per-month for the three-month reliability configurations. This low demand rate and a PM schedule having a minimum three-month interval indicates that a single maintenance team can easily perform all on-site and node maintenance tasks. The extremely low statistical probability of simultaneous failure of two UARs further supports the single team concept.

A three-man node electronics team is recommended. Two men would travel to the sites when needed to perform maintenance. The third man would remain at the node to monitor the status of the remaining operating systems. A three-man team provides its own back-up in event of illness.

Electronic technicians will have LRU repair skills involving the use of bench environments and hand tools as well as the system skills necessary to accomplish on-site maintenance for all electronic equipments.

Power equipment will be maintained by a power equipment specialist assigned to the node maintenance team.

2.4 Equipment Installation

Installation of UAR equipment is complicated by the number of sites involved and the hostile environmental conditions that prevail in the Arctic. With a time window of only

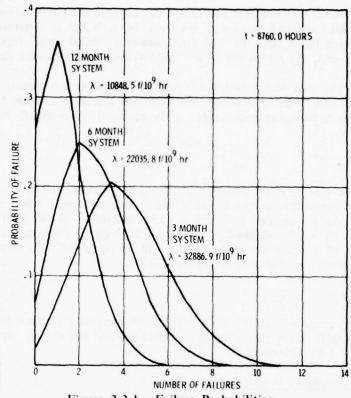


Figure 2.3-1. Failure Probabilities

24 months in which to accomplish this task, a great deal of emphasis will be placed on developing an installation plan compatible with program requirements and schedules. A preliminary schedule is described in Figure 2.4-1.

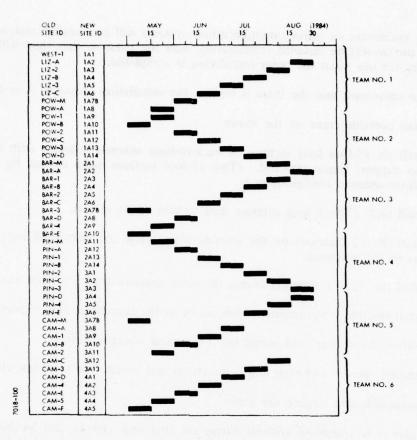
2.4.1 Installation Planning

The installation plan has been segregated into smaller planning elements to provide greater cost and schedule visibility. Costs associated with installation elements are contained in Paragraph 2.6. A final plan will be developed early in the RDT&E program that includes customer recommendations and guidance.

2.4.2 UAR Installation Sequence and Schedule

The UAR site installation sequence is based upon the following assumed conditions:

- a. The towers and other structures to support the antennas and other radar equipment are completed and in place.
- b. Each tower will be able to support and provide mounting interfaces for a lightweight modularized crane.
- c. Prime power sources and distribution systems will be available to support installation activities.



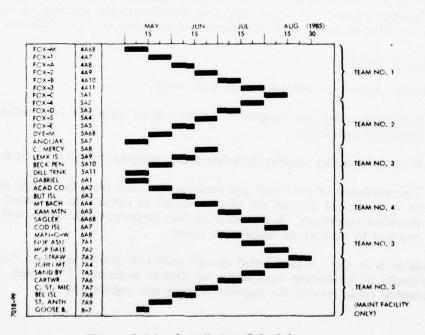


Figure 2.4-1. Installation Schedule

d. The microwave or other communications systems will be installed and ready for use or will be installed concurrently with the radar so that they will be ready for use when the radar installation is completed.

When the equipment and the team is on-site, the installation sequence is as follows:

- a. In portable crane on the tower.
- b. Install six 15-foot long sections of back-to-back antenna reflector with feed array support beams attached. (Two 15-foot sections make up one leg of the tri-form antenna configuration.)
- c. Install each 27-foot long antenna feed section on its mount.
- Install the 12 diplexers on the antenna tower deck and associated waveguide runs to the antennas.
- e. Install the 12 IFF antenna horns, the omni antenna and associated waveguide.
- f. Install electronic equipment shelter on or in the support structure provided.
- g. Position the shelter, and install the 13 vertical waveguide runs.
- h. Complete power and remoting connections and initiate the electrical checkout.
- i. Disassemble and remove the crane.

The preliminary plan is to install 46 systems during the first year (1984), and 34 during the second year (1985), as shown in Figure 2.4-1. These schedules are ambitious, but feasible with sufficient logistics support. It is projected that a five-man crew working 12 hours a day can complete an installation in 15 days. Checkout will take two men an additional five days.

2.4.3 Installation Support

Installation support encompasses three basic areas:

- a. Preliminary tools and equipment choices will be made and later refined during prototype installation.
- b. Personnel logistics support including safety equipments will be provided.
- c. Transportation of personnel and equipment to and from the Arctic areas will be accomplished through use of commercial air carriers on scheduled or chartered operations. A minimum of four cargo-type helicopters will be required to support six installation teams.

An option available is to use a demothballed aircraft carrier or similar ship as a floating support base for the majority of installation work along the Eastern seaboard. This vessel, equipped with two helicopters, could provide the required housing and logistics support.

2.5 LRU Stockage Model

The initial and recurring investment in LRU's is a logistics LCC element. The selection and allocation of main spare units was accomplished using math models to optimize the dollars invested to achieve an operational availability $(A_{op}) \ge 90$ percent. Three models were used in the optimization process.

- OPSTOCK Optimum Stockage for multilevel maintenance environment.
 Optimization based on risk for spares shortage vs operational availability
- CONSUME Consumable item requirement based on stockage objectives and support pipelines
- Q-FACTOR A subroutine based on maintenance and logistic time distributions which establishes a risk for spares factor.

2.5.1 Support Variations

The math models accommodate variations in the number of sites supported by each node and provide separate printed stockage lists for each node. The models also consider depot turn-around time and the need to fill the pipelines. Identical kits of on-board spares for the node helicopters has been assumed.

The models determine stockage requirements on basis of the detailed LRU data described in Paragraph 2.2.2. Not-repairable-this-station (NRTS) rates for the node and depot have also been assigned which influence stockage quantities and account for attrition of the most vulnerable items.

2.5.2 Initial Spares

The cost of initial spares, repairables and consumables, is based on the Operational Availability (A_{op}) objective. Assuming a risk for shortage of 0.25 (see Paragraph 2.1.2) for a 3-month configuration, the cost of initial spares can be determined using the graph in Figure 2.5-1. A 0.25 risk translates to an LRU investment of approximately 470,000.00 dollars. A 6-month configuration with the same A_{op} objective would require a 220,000.00 dollar investment.

Figure 2.5-1 indicates the dollar amount and the computer program prints a corresponding list of LRUs. For a two-year stockage objective this initial spares dollar amount is added to the cost of consumables expended in two years as determined by the CONSUME model.

2.6 Life Cycle Cost Computation

All LCC dollar amounts shown are in constant January 1977 dollars without discounting. Prices used in these computations are based on an 80 system build, with a concurrent build of all spares for the life cycle. Common test equipment cost computations were based on current catalog prices.

The system contains a single life limited item, the waveguide desiccant elements, and the demands for this item were processed at the replacement rate. All other items were processed at the failure rate. Redundant systems were assumed to be in "cold" standby.

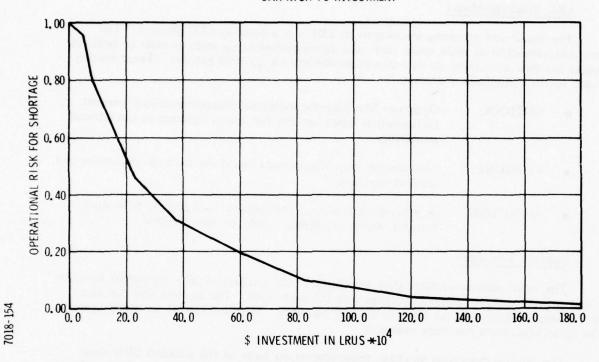


Figure 2.5-1. UAR Risk versus Investment

All support calculations were based on all systems operating in a full performance mode, full time for the life cycle.

2.6.1 Life Cycle Cost Summary

This summary includes three equipment options and two node manning options. The data is presented in matrix format in Table 2.6-1.

2.6.2 Acquisition Costs

The RCA PRICE model was used to obtain acquisition cost data for the UAR. This is a parametric cost modeling technique that provides reliable estimates of system acquisition costs especially during the conceptural phase of a program. PRICE inputs are primarily physical characteristics such as size, weight, type of componetry, power dissipation, and construction type. Prototype and production quantities are also required inputs. PRICE model outputs include recurring and nonrecurring costs for development and production phases. It also provides an engineering schedule, and it has numerous internal checks to verify reasonableness of the various results.

Printouts from the UAR PRICE Model are given in Appendix A. Greatest accuracy is obtained from the model when a system is broken down into small constituent parts. In addition, purchased items are handled most effectively by being considered separately. This was done in the UAR modeling with the receiver, exciter, and synthesizer purchase parts lumped into a single purchased item. As a consequence, it is difficult to determine the total

TABLE 2.6-1. LIFE CYCLE COST SUMMARY MATRIX

	3-Month Co	onfiguration	6-Month C	onfiguration	12-Month	Configuration
LCC Cost Category	USAF Technicians	Contractor Technicians	USAF Technicians	Contractor Technicians	USAF Technicians	Contractor Technicians
Acquisition Cost	51,124,952.00	51,124,952.00	54,058,768.00	54,058,768.00	59,283,659.00	59,283,659.00
Initial Logistics	17,306,806.00	17,924,862.00	16,956,806.00	17,674,862.00	16,736,806.00	17,454,862.00
Recurring Logistics x 18 Yrs	12,627,144.00	23,595,912.00	12,627,144.00	23,595,912.00	12,627,144.00	23,595,912.00
Totals	\$80,958,902.00	92,645,726.00	\$83,642,718.00	\$95,329,542.00	\$88,647,609.00	\$100,334,433.00

costs of these individual items from the printouts. Table 2.6-2 gives the cost breakdown of the units with the appropriate purchase item cost added to the PRICE estimate of the unit.

The baseline system is the minimum system capable of performing required radar operations. It contains no redundancy other than that inherent in designs of the individual modules. The platform factor (PLATFM) of 1.95 used for the PRICE runs implies a very high reliability design, using components that roughly satisfy the reliability requirements discussed in Volume II. In order to meet the 3-, 6-, and 12-month reliability goals, module redundancy is also needed. The required redundancy for these three cases and their cost impacts are given in Tables 2.6-3 through 2.6-5. In each case, the needed redundancy is specified relative to the baseline system. Also, an additional 15 percent of the cost of redundant items is added to the total to account for increased switching and other miscellaneous equipment that may be required.

TABLE 2.6-2. BASELINE UAR PRICE BREAKDOWN (\$K)

	Development	Production	Total Cost	Avg Prod Cost/Unit
RF Receiver (2 Req'd)	632	800	1432	4.994
IF Receiver	338	419	757	5.232
Radar Exciter	178	432	610	5.407
IFF Receiver	254	413	667	5.157
IFF Exciter	178	353	531	4.407
Synthesizer	440	830	1270	10.371
Signal Processor	4171	5167	9337	64.588
Power Supplies	1020	1177	2197	14.713
Radar Transmitter	1821	4365	6186	54.563
IFF Transmitter	660	871	1531	10.888
Waveuigde Switch	667	727	1394	9.088
Waveguide	63	2538	2601	31.725
Antenna	2053	10930	12984	136.625
Data Processor	34	1366	1401	17.075
Integration and Test	348	1357	1705	16.963
Total Baseline	12859	31744	44603	396.800

There are several points which should be mentioned in regards to these data. The cost figures given are factory cost and are not the actual sell price of the equipment. Also, these are radar costs only, and they do not include costs of communication equipment, shelter, prime power, environmental control equipment, or purchase and installation of the antenna tower. In addition, the costs related to a Reliability Verification and Modification Program during and after system deployment are not included. It is impossible at this time

TABLE 2.6-3. UAR THREE-MONTH CONFIGURATION

Redundant Items

1 Central Timing Unit	\$ 1,930
1 Control Monitor/Proc. Interface	3,040
1 Communication Interface	3,225
1 Doppler CFAR Proc.	4,630
2 Clutter Maps	5,520
1 Correlation Detector	2,120
1 CPU Assembly	3,875
4 PROM Boards	6,000
2 RAM Boards	3,600
1 Power Supply	16,513
Cost of Redundant Items	50,453
15% of Cost for Switching and Misc.	7,568
Total for Redundancy	58,021
Baseline	396,800
Total for 3 Mo. Configuration	\$454,821
System Acquisition Costs (80 Units)	
Development	\$14,739,272
Production	36,385,680
Total	\$51,124,952

TABLE 2.6-4. UAR SIX-MONTH CONFIGURATION

Redundant Items

1 Central Timing Unit	\$ 1,930
1 Control Monitor/Proc. Interface	3,040
1 Communication Interface	3,225
2 Doppler CFAR Proc.	8,260
2 Clutter Maps	5,520
1 Correlation Detector	2,120
2 CPU Assemblies	7,750
4 PROM Boards	6,000
2 RAM B ards	3,600
1 Power Supply	16,513
1 Frequency Synthesizer	10,371
1 A/D Converter	4,820
Cost of Redundant Items	73,149
15% of Cost for Switching and Misc.	10,972
Total for Redundancy	84,121
Baseline	396,800
Total for 6 Mo. Configuration	480,921
System Acquisition Costs (80 Units)	
Development	\$15,585,088
Production	38,473,680
Total	\$54,058,768

ral	Timing Unit		

Redundant Items

1 Central Timing Unit	\$ 1,930
1 Control Monitor/Proc. Interface	3,040
1 Communication Interface	3,225
3 Doppler CFAR Proc.	13,890
2 Clutter Maps	5,520
1 Correlation Detector	2,120
2 CPU Assemblies	7,750
4 PROM Boards	6,000
4 RAM Boards	7,200
1 Power Supply	16,513
1 Frequency Synthesizer	10,371
1 A/D Converter	4,820
1 Radar RCVR, RF	4,994
1 Radar RCVR, IF	5,232
1 Radar Exciter	5,407
1 IFF Receiver	5,157
1 IFF Exciter	4,407
1 IFF Target Detection Unit	2,430
1 IFF Buffer	2,510
1 Jamming Detector	1,052
Cost of Redundant Items	113,568
15% of Cost for Switching and Misc.	17,035
Total for Redundancy	130,603
Baseline .*	396,800
Total for 12 Mo. Configuration	\$527,403
System Acquisition Costs (80 Units)	
Development	\$17,091,419
Production	42,192,240
Total	\$59,283,659

to evaluate the extent of such a program; however, it is almost inevitable that design changes will be required in order to meet the reliability requirements. Depending on the deployment schedule, the reliability confidence level desired, and amount of redesign required, a possible increase in the system acquisition cost by 25 percent seems reasonable.

2.6.3 Initial Logistic Cost

The cost elements directly related to the initial deployment of 80 UAR systems are:

2.6.3.1	Spares (24-Month)	3-Month	6-Month	12-Month
	LRU/SRU (Node & Depot)	\$482,870.00	\$232,870.00	\$12,870.00
	Repair Parts	36,496.00	36,496.00	36,496.00
	Total	\$519,366.00	\$269,366,00	\$49.366.00

Initial quantities of LRU/SRU spares was determined by the OPSTOCK computer model. LRU/SRU repair parts were determined by a maintenance demand calculation and summed with a percent allocation for other piece-parts.

2.6.3.2 Inventory Entry (Codification)

LRU/SRUs (66 Line Items) \$ 6,600.00 Nonstandard Parts (750 Estimated) 75,000.00

Total \$81,600.00

Inventory entry was calculated at \$100.00/item.

2.6.3.3 Personnel (Node)

USAF Electronic Technicians (24 Men) \$414,000.00

Contractor Electronic Technicians (24 Men) \$1,154,400.00

USAF Technicians are in the E4E6 pay and allowance categories (\$525.00/mo. + \$30.00/day)

Contractor Technicians calculations includes base pay, allowances and contractor support & transportation.

A 56-hour work week is assumed.

2.6.3.4 Training (2 each, 14 week Courses-Contractor Training)

Training Data, Instruction, Facilities \$256,000.00 USAF Technician Pay & Allowances (24 Men) 135,240.00 Total \$391,240.00

Training Data, Instruction, Facilities \$256,000.00 Contractor Technicians (24 Men) 112,896.00 Total \$368,896.00

Contractor technicians are not paid allowances during training. Attrition of 3-men prior to or during assignment is assumed.

Tools and Test Equipment 2.6.3.5

Common Tools at Nodes (7 nodes) \$479,000.00 Test Equipment (7 nodes) 630,000.00 Common Tools at Depot 7,300.00 Test Equipment & Fixtures, Depot 85,700.00 Total

\$1,202,000.00

Node quantities include helicopter kits.

Technical Documentation (TOs)

Preparation and 100 sets \$600,000.00

Consists of Operations, Service, Operation, IPB To USAF Standards

2.6.3.7 Installation and Checkout (80 Systems)

Labor	\$2,114,400.00
Special Tools & Installation Equipt.	1,280,000.00
Personnel Logistics	610,000.00
Transportation (Installation & Personnel)	
Transportation of Radar Sets	7,358,000.00
Total	,,

\$13,998,600.00

Eighty (80) sets, each having 10,875 cubic feet & weighs 16,200 lbs (does not include spares or test equipment). Six Contractor teams will perform the installation.

2.6.3.8 Total of All Initial Logistics Costs

	3-Month	6-Month	12-Month
All Items & USAF			
Tech. Option All Items & CTS Option	17,206,806.00 17,924,862.00	16,956,806.00 17,674,862.00	16,736,806.00
The recine of CIB Option	17,724,002.00	17,074,002.00	17,454,862.00

2.6.4 Recurring Logistic Cost

The cost elements directly related to sustaining operational performance of UAR systems on an annual basis are:

2.6.4.1 Spares Replenishment

LRU/SRUs (Node & Depot)	\$ 6,435.00	
Repair Parts (Node & Depot)	18,248.00	
Total		\$24,863.00

LRU/SRU quantities were determined by the CONSUME computer model. LRU/SRU repair parts were determined by a maintenance demand calculation and summed with a percent allocation for other piece-parts.

2.6.4.2 Inventory Management

LRU/SRU (66-Line Items)	\$ 6,600.00	
Nonstandard Parts (750 Estimated)	75,000.00	
Total		\$81,600.00

Inventory Management calculated at \$100.00/Item/Year

2.6.4.3 Personnel (Node)

USAF Electronic Technicians (24 Men)	\$414,000.00
or	
Contractor Electronic Technician (20 Men)	\$962,000,00

Refer to Item 1.6.3.3 for rates Contractor technician attrition is \approx 83 percent.

2.6.4.4 Training (2 each, 14-week Courses – USAF Training)

Training Equipment & Support	\$23,553.00	
Training Instructors USAF (4 Men*)	7,360.00	
USAF Technician Pay & Advances (24 Men) Total	44,160.00	

\$75,073.00

*E4-E6 Rate

or

Training Equipment & Support	23,553.00	
Contractor Training (20 Men)	112,896.00	
Total		\$136,449.00

A training system should be procured: Training equipment calculated at 1/20 of acquisition cost. Support cost based on 16 hours per day equipment utilization. Training set, of contractors site, is "bailed" equipment. Contractor training of 20 men includes four instructors.

2.6.4.5 Depot Maintenance

Manhours	\$ 1,148.00
	• -,

Maintenance calculated at \$20/manhour

2.6.4.6 Maintenance Administration

Depot	\$ 7,174.00

Maintenance Administration calculated at \$250.00/repair action at depot.

2.6.4.7 Tools and Test Equipment Maintenance

Nodes and Depot	\$35,250,00

Maintenance cost calculated at 5 percent of acquisition cost/year.

2.6.4.8 Preventive Maintenance (Materials)

On Site (80 sites)	\$62,400.00

2.6.4.9 Total of All Recurring Logistics Costs (Annual)

All Items and U.S.A.F. Technician ar	id Training	\$701,508.00
or All Items and Contractor Technicians and	and Training	\$1,310,884.00

3.0 MAR LIFE CYCLE (LCC) COSTS

Two of the three factors in the LCC algorithm are logistics dependent, e.g.,

$$LCC = A_T + I_L + \sum_{i=1}^{N} R_L$$

where

A_T = acquisition cost total

I_I = initial logistics cost

R_I = recurring logistics cost (annual)

N = number of years in the life cycle.

In order to quantify these logistics factors, a specific systems deployment and maintenance philosophy must be defined. Such a plan has not been developed for MAR, consequently its LCC cannot presently be determined. An estimate of MAR's acquisition cost, however, has been made based on a production run of 20 systems. Furthermore, certain aspects of a suggested maintenance philosophy, based on achieving a low LCC, have been addressed.

The acquisition cost of MAR is estimated as follows:

Average unit cost based on a quantity of 20 units = \$2,110K (for the 5-day reliability configuration)

The acquisition cost total = $20 \times \$2,110K = \$42.2M$

The maintenance philosophy suggested is presented in the following paragraphs.

3.1 MAR Maintenance Philosophy

3.1.1 Maintenance Structure

Based on MAR architecture, BITE penetration, (number of systems involved), and assuming full-time operation (168 hours per week) a two-tier maintenance structure is recommended: these two levels are the organizational level (on-site) and the depot level.

Investigations, involving life-cycle cost (LCC) analyses, indicate that consolidation of the first three echelons of repair at the organizational level is a viable cost-effective option. On-site maintenance of certain LRU repairables that involve removal and replacement of low-cost consumable materials is recommended, instead of a flow of these repairables through a repair loop. All LRU's will be individually considered to determine how they best "fit" into stockage and repair aspects of the maintenance system.

3.1.2 Organizational Maintenance

All organizational maintenance can be performed by a single trained technician. Certain maintenance activities may require more than one man because of safety or physical size/weight of an item involved in the maintenance action. However, this assistance to the technician does not require any system skills. The on-site maintenance will be an intermediate level technician (USAF Level 5) with certain bench skills involving use of soldering tools and standard test equipment. The on-site technician will be responsible for all preventive maintenance, corrective maintenance, certain LRU repairs, and performance testing of the system subsequent to any corrective maintenance actions or modification of equipment.

- a. Preventive maintenance (PM) The off-line radar preventive maintenance intervals are greater than 168 hours. This preventive maintenance (PM) features is in keeping with an operational availability objective exceeding 90 percent for one week of continuous operation. A small amount of daily on-line PM consisting primarily of monitoring BITE indicators or metering devices is recommended.
- b. <u>Corrective Maintenance (CM)</u> The BITE minimizes skill requirements, need for special purpose test equipment on-site, and reduces the requirement for standard test equipments.

All faulty digital modules and digital portions of the hybrid modules used throughout the MAR are repaired on-site to the maximum extent feasible. Each site is furnished a digital test set to facilitate rapid identification of failed integrated circuit devices. The on-site technicians remove and replace defective items using repair materials from on-site stock. Certain repairable analog-type LRU's used in the transmitter are repaired on-site when possible.

LRU's in antenna or transmitter high-voltage areas which cannot be safely repaired and tested on-site are evacuated to the depot facility. Analog modules of the receiver are repaired at the depot where the required analog test station and test fixtures are available. All electro-mechanical LRU's removed from major subsystems are returned to depot facility for repair and subsequent testing.

3.1.3 Depot Maintenance

The prime responsibility of the depot is to provide operating units with a constant source of equipment spares. The depot meets this responsibility by maintaining stockage quantities which meet all demands and fill all pipelines to the operating satellites. To perform this function in a cost-effective manner, the depot is fully equipped and manned to effect repair of items returned from the field. Pipeline time between the depot and operating sites is a factor in determining spares requirements.

The depot is equipped with a digital dual disc base console to facilitate repair of all digital modules used in the radar. This test console provides the depot facility with the backup capability of fault isolating and repairing all digital modules. The digital test console at the depot also serves to support the portable digital tester used at the sites.

The analog test station includes a complement of interfacing test fixtures necessary to facilitate fault isolating, alignment, and testing of all analog modules. Those test fixtures required to fault isolate, align, and test modules in the transmitter high voltage area, which cannot be tested safely on-site, are available at the depot.

3.1.4 Maintenance Manning

On-site maintenance, including repair of certain modules, can be accomplished by one man per maintenance shift. Assuming full-time operations of 168 hours per week, the work week could be divided into three, 56-hour shifts which would indicate the need for three technicians.

However, high system reliability results in few maintenance demands for both on-line and off-line actions, and the total "downtime" due to preventive maintenance is less than 25 hours per year.

Based on those system maintenance requirements and the remote status/alarm system (reference BITE, paragraph 2-2) the equipment does not require a man in constant attendance. A technician must be on-call however, and the distance between the radar set and standby area should be compatible with the 0.5 hour MTTR objective. If this condition is satisfied, a single, standard, 40-hour maintenance shift is recommended during which scheduled daily PM and other nonscheduled PM is performed and all bench repair activities accomplished. The work schedule could consist of one standard 40-hour maintenance shift and a "standby" maintenance shift for the remaining time.

Recommended MAR Site manning is: 2 men, AFSC 5. With modernization of the Dew Line installations, the USAF has an option open to them that could reduce their electronic maintenance manpower costs by as much as 25 percent. The maintainability and packaging design of all new equipment is rapidly reducing the technicians' requirement for higher levels of system knowledge. Under the LRU remove and replacement concept, a technician need only be capable of interpreting the system BITE indications. Therefore, by crosstraining electronic technicians, further manpower reductions can be realized.

3.2 BUILT-IN TEST EQUIPMENT (BITE) CONCEPT (Figure 3.2-1.)

The MAR BITE design is basically a two-tier system that is automatic and on-line. The first tier collects status information from the subsystems, relays it to a remote center and displays it on the local maintenance console control and status panel.

The second-tier monitors all critical subsystem parameters, reports to the first tier and displays the information on local equipment group panels.

3.2.1 BITE Objectives

The BITE penetration design objectives are to isolate faults to:

- A single LRU, on-line, for 75 percent of system failures
- Three LRU's, on-line, for 90 percent of system failures
- A single LRU, off-line, for 95 percent of system failures.

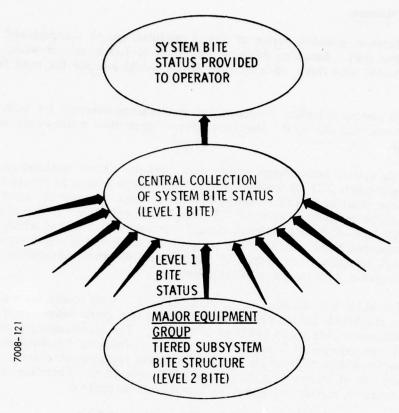


Figure 3.2-1. General BITE Concept

3.2.2 System Status Panels

The system status panels will use green, amber, and red indicators for the following conditions:

- Green = Status Satisfactory
- Amber = Subsystem in Maintenance Mode

or

Subsystem Requires Maintenance

• Red = Status Unsatisfactory.

If the system includes backup equipment, each channel will have a separate set of indicators. Should a fault occur, the BITE system will initiate an automatic switch-over sequence to the standby channel. The indicators will show which channel is on-line.

Status of the following equipment groups will be shown on the system status panel: Antenna, Receiver/Synthesizer/Processor, Transmitter, and IFF.

3.2.3 Subsystem Status Panel

Each subsystem will have a local second-tier BITE status panel. The antenna, because it is physically separated from the remaining equipment will have its status remoted to the receiver/processor panel.

The Receiver/Synthesizer/Processor panel (Figure 3.2-2) is a sample of an equipment group panel. As an example, consider the following:

- 1. A fault in the RCVR/SYNTH/PROC group is reported on the system status panel.
- The technician checks the local panel and observes that FAULT 2 SYN A is illuminated.
- 3. Using the thumbwheel selector, he sets the fault code to 02.
- 4. The ENTER/SEQUENCE switch is activated and the LRU designator appears.
- 5. If the sequence involved two or more LRU's, the multiple fault indicator would be illuminated. The LRU with the highest probability of failure would appear in the FAULT REF DESIG window.
- 6. Depressing the sequence switch would enter the next LRU designator.

This sequence is summarized in Figure 3.2-3.

3.2.4 Display Console

The Maintenance Display Console is provided for the purpose of monitoring MAR's output target message data quality. The display is not considered mission essential hardware. In the event of a remoting system malfunction, the display will assist in the fault isolation process.

3.3 Supportability

Support cost could be a significant element in the life cycle cost for the MAR. These costs, which are a consequence of design, can be determined for a given equipment assessed in a given operational environment and when taken cumulatively, define the supportability characteristic of the design. To quantify this characteristic, however, requires considerable customer and contractor data not presently available in the form necessary to complete this study.

Supportability can also be measured in less specific terms by evaluating various aspects of design and their logistic effect. In following paragraphs, the consequence of our selected MAR design approach have been summarized in supportability terms.

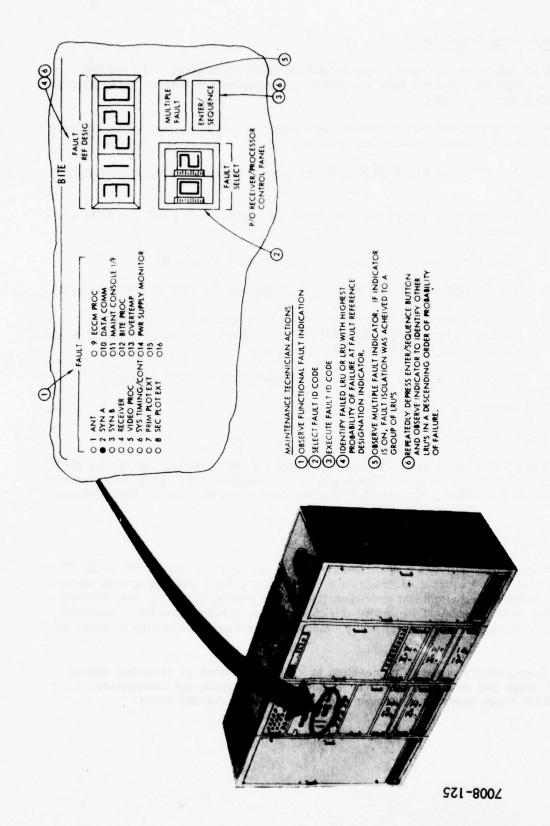


Figure 3.2-2. Fault Isolation Example

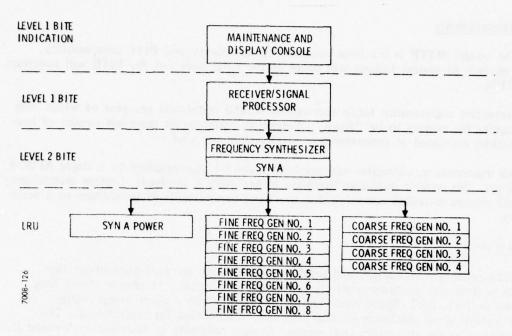


Figure 3.2-3. Fault Isolation Example Summary

3.3.1 Modularity

The MAR baseline maintenance design involves modular packaging of the equipment to facilitate removal and replacement maintenance on-site. The radar set will consist of approximately 200 LRU's. These 200 LRU's will represent 95 percent or more of the systems failure rate.

3.3.2 Commonality

All like items will be directly interchangeable. Where possible, common solutions will be used to reduce the number of different items. The processor for example, will contain approximately 60 boards of five different types. Power supply commonality has been achieved by using the best tolerance, regulation, or capacity where common voltages are required. Blower motor commonality is achieved in the same manner. There are approximately 160 different LRU types in the radar set.

3.3.3 Repairability

Approximately 75 to 78 percent of all failures on site will be corrected without involving the depot. This is achieved through a combination of on-site repair of digital LRU's and discard of consumables on failure. Twenty-two to twenty-five percent of system failures involve electronic LRU, and three percent of less are mechanical item failures. These LRU's will be repaired at depot.

3.3.4 Maintainability

The on-site MTTR is 0.5 hour achieved by modularity and BITE compatibility. Ninety-five percent of system failures will occur at the LRU level and the BITE will penetrate to these LRU's.

Preventive maintenance totals approximately 450 man-hours per year of which only 25 hours require the system to be "down". Preventive maintenance materials consist of low-cost consumables estimated at approximately \$100/system per year.

All maintenance, corrective and preventive, can be accomplished by a single AFSC-5 level technician. No outside assistance involving higher skills is involved. Certain maintenance actions could require unskilled assistance due to safety requirements or weight/size of a maintenance item.

3.3.5 Reliability

High reliability techniques and solid-state technology are used throughout the design. The system also contains relatively few life-limited items. At present, these have been defined as TWT, CRT, Spark gaps, TR limiter, blower and coolant pump motors. All these items exhibit some condition of wear and can be scheduled for replacement. The TWT is repairable as are all motors and pumps. System reliability is described in Volume II, the technical description.

3.3.6 Test Equipment

All common test equipment required exist in the USAF inventory. No special tools and test equipment are required on-site. A commercially available digital tester will be furnished as CFE for each site. Depot special hi-cost test equipment, consisting of six items will be CFE. No special tools are required at depot.

APPENDIX A

PRICE MODEL PRINTOUTS FOR THE BASELINE UNATTENDED RADAR

INPUT DATA			
QTY 160. PROTOS QTYSYS 2. INTEGE	4.0 UT 0.500 IHTEGS	1.500 VOL 0.037 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 0.580 MCPLXS	6.002 PRODS	4.473 HENST 1.000	DESRPS 0.000
ELECTRONICS USEVOL 0.697 MCPLXE PUR 2.500 CMPNTS	8.451 PRODE	4.722 MENEL 1.000	DESRPE 0.000
ENGINEERING ENMTHS 1.0 ENMTHP			
PRODUCTION PRMTHS 34.0 PRMTHF			
GLOBAL YEAR 1978, ESC PLATFM 1.950 SYSTEM	9.00% PROJET	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000
PROGRAM COST ENGINEERING DRAFTING DESIGN SYSTEMS PROJ MGMT DATA SUBTOTAL (ENG)	85. 312. 96. 62. 34. 589.	6. 21. 6. 23. 2. 52.	91. 333. 96. 85. 35. 640.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EQ SUBTOTAL (MFG)	0. 24. 11. 35.	409. 0. 12. 421.	489. 24. 24. 456.
	524.		
UOL 0.037 AUCOST NT 1.500 ECHE	2.55 TOTAL 0.092 ECNS	AV PROD COST 2.95 0.026 DESRPE 0.000	LCURVE 0.950 DESRPS 0.000
MECH/STRUCT NS 0.520 WSCF ELECTRONICS	14.054 MECID	0.000 PRODS 4.473	MCPLXS 6.002
WE 0.980 WECF PWR 2.500 CMPNTS	38.000 CMPID 219.		MCPLXE 8.451 CMPEFF -6.909
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	23.225 ENMTH1 66.000 AVER.	「 23.225 ECMPLX 1.752 PROD RATE PER MONTH	PRNF 0.000 5.000
COST RANGES FROM CENTER TO	DEVELOPMENT 563. 624. 706.	PRODUCTION 407. 472. 556.	TOTAL COST 970. 1096. 1262.

Figure A-1. RF Receiver

INPUT DATA ATY 80. PROTOS ATYSYS 1. INTEGE	2.0 MT 0.700 INTEGS	1.500 UOL 0.037 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 0.520 MCPLXS	6.002 PRODS	4.473 MEMST 1.000	DESRPS 0.000
ELECTRONICS USEVOL 0.697 MCPLXE PWR 2.500 CMPNTS	8.451 PRODE 0. CMPID	4.722 MENEL 0.800 0.000 PURFAC 2.500	DESRPE 6.200 CMPEFF 6.000
ENGINEERING ENMTHS 1.0 ENMTHR	0.0 ENMTHT	0.0 ECMPLX 1.546	PRMF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECME 0.000	ECNS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
DRAFTING DESIGN SYSTEMS PROJ MGNT DATA SUBTOTAL (EMG)	49. 175. 46. 30. 15. 315.	5. 17. 9. 13. 1. 36.	55. 192. 46. 42. 16. 551.
MANUFACTURING PRODUCTION PROTOTYPE TOOL—TEST EO SUBTOTAL (MFG)	0. 12. 7. 19.	212. 0. 7. 219.	212. 12. 14. 238.
TOTAL COST	334.	255.	589.
UOL 6.837 AUCOST NT 1.500 ECNE	2.65 TOTAL 0.092 ECNS	AU PROD COST 3.19 0.026 DESRPE 0.200	LCURUE 0.950 DESRPS 0.000
MECH/STRUCT MS 0.520 MSCF	14.054 MECID	0.000 PRODS 4.473	MCPLXS 6.002
ELECTRONICS WE 0.980 WECF PWR 2.500 CMPMTS		0.000 PRODE 4.722 PWRFAC 2.500	MCPLXE 8.451 CMPEFF -6.909
SCHEDULES ENMTHS 1.000 ENMTHP PPMTHS 34.000 PRMTHF			PRNF 0.000 2.500
COST PANGES FROM CENTER TO	DEVELOPMENT 301. 334. 379.	PRODUCTION 220. 255. 299.	TOTAL COST 521. 589. 677.

Figure A-2. IF Receiver

IMPUT DATA OTY 80. PROTOS OTYSYS 1. INTEGE	2.0 NT 0.500 INTEGS	1.300 VOL 0.037 0.300 AMULTE 106.50%	MODE AMULTH	1. 196.50%
MECH/STRUCT NS 0.520 MCPLXS	6.002 PRODS	4.473 HENST 1.000	DESRPS	0.000
ELECTRONICS USEVOL 0.555 MCPLXE PWR 3.000 CMPNTS	8.451 PRODE 8. CMPID	4.722 NEWEL 1.000 0.000 PWRFAC 2.500	DESRPE CMPEFF	0.000 0.000
ENGINEERING ENMTHS 1.0 EHMTHP	0.0 EMMTHT	0.0 ECMPLX 0.825	PRHE	0.000
PRODUCTION PRMTHS 34.0 PRMTHE	é6.0 LCUPUE	0.950 ECHE 0.800	ECHS	0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST PTLGTS	2.000 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL	COST
ENGINEERING DRAFTING DESIGN SYSTEMS PROJ NGNT DATA SUBTOTAL (ENG)	34. 101. 12. 11. 5. 164.	6. 21. 0. 11. 1. 40.		41. 122. 12. 23. 6. 203.
		183. 0. 6. 189.		193. 8. 9. 200.
TOTAL COST	174.	229.		403.
UOL 0.037 AUCOST NT 1.300 ECME	2.23 TOTAL 0.092 ECNS	AV PROD COST 2.86 0.026 DESRPE 0.000	LOURUE DESRPS	0.950 0.000
MECH/STRUCT US 0.520 WSCF	14.054 MECID	0.000 PRODS 4.473	MOPLES	6.002
ELECTRONICS WE 0.780 WECF PWR 3.000 CMPNTS		0.000 PRODE 4.722 PWRFAC 2.500	MCPLXE CMPEFF	8.451 1.194
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	10.928 ENMTH 66.000 AVER.	T 10.928 ECMPLX 0.825 PROD RATE PER MONTH	PRNF	0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 157. 174. 197.	PRODUCTION 198. 229. 267.	TOTAL	COST 356. 403. 464.

Figure A-3. Radar Exciter

INPUT DATA OTY 80. PROTOS OTYSYS 1. INTEGE	2.0 WT 0.500 INTEGS	1.400 UOL 0.037 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 0.520 MCPLXS	6.002 PRODS	4.473 NEWST 1.000	DESRPS 0.000
ELECTRONICS USEVOL 0.626 MCPLXE PUR 1.000 CMPNTS			
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 ENMTHT	0.0 ECMPLX 1.031	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECNE 0.000	ECMS 0.000
GLOBAL YEAR 1978, ESC PLATEM 1.950 SYSTEM			
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
EMGINEERING DRAFTING DESIGN SYSTEMS PROJ MGNT DATA DUSTOTAL (ENG)	47. 146. 23. 15. 7. 237.	7. 22. 0. 12. 1. 42.	53. 162. 23. 27. 8. 279.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EO SUBTOTAL (MFG)	0. 9. 4. 13.	2 00. 8. 7. 207.	200. 9. 11. 320.
TOTAL COST	250.	249.	499.
NOL 0.037 AUCOST NT 1.400 ECNE	2.50 TOTAL 0.092 ECHS	AV PROD COST 3.11 0.826 DESRPE 0.000	LCURVE 0.950 DESRPS 0.000
MECH STRUCT US 0.520 USCF ELECTPONICS	14.054 MECID	0.000 PPODS 4.473	MCPLXS 6.002
WE 0.380 WECF PWP 1.900 CMPNTS			MCPLXE 8.451 CMPEFF-22.860
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF		T 13.661 ECMPLX 1.031 PROD RATE PER MONTH	PRMF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 225. 250. 284.	PRODUCTION 216. 249. 291.	TOTAL COST 441. 499. 575.

Figure A-4. IFF Receiver

IMPUT DATA 0TY SO. PROTOS 0TYSYS 1. INTEGE	2.0 WT 0.500 INTEGS	1.300 MOL 0.037 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT NS 3.520 MCPLXS	6.002 PRODS	4.473 HEWST 1.000	DESRPS 0.000
ELECTRONICS USEVOL 6.555 NOPLXE PWR 2.600 CMPNTS	8.451 PRODE 8. CMPID	4.722 NEWEL 1.000 0.000 PWRFAC 2.500	DESRPE 0.008 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHR	0.0 ENMTHT	0.0 ECMPLX 0.825	PRMF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECNE 0.000	ECNS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
ENGINEERING DRAFTING DESIGN SYSTEMS PROJ NGMT DATA SUBTOTAL(ENG)	34. 101. 12. 11. 5. 164.	6. 21. 6. 11. 1. 40.	41. 122. 12. 23. 6. 283.
MANUFACTURING . PRODUCTION PROTOTYPE TOOL—TEST EG SUBTOTAL (MFG)	0. 0. 3. 11.	183. 0. 6. 189.	183. G. 9. 200.
TOTAL COST	174.	229.	403.
UOL 0.937 AUCOST NT 1.300 ECHE	2.28 TOTAL 0.092 ECMS	AV PROD COST 2.86 0.826 DESRPE 0.000	LCURVE 0.950 DESRPS 0.000
MECH/STRUCT WS 0.520 WSCF	14.054 MECID	0.000 PRODS 4.473	MCPLXS 6.002
ELECTRONICS WE 0.780 WECF PUR 3.000 CMPNTS		0.000 PRODE 4.722 PWRFAC 2.500	MCPLXE 8.451 CMPEFF -6.805
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	10.928 ENMTH 66.000 AVER.	T 10.928 ECMPLX 0.825 PROD RATE PER MONTH	PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 157. 174. 197.	PRODUCTION 198. 229. 267.	TOTAL COST 356. 403. 464.

Figure A-5. IFF Exciter

IMPUT DATA OTY 80. PROTOS OTYSYS 1. INTEGE	2.0 MT 0.500 INTEGS	2.510 VOL 0.100 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 1.230 MCPLXS	5.997 PRODS	4.508 HENST 1.000	DESRPS 0.500
ELECTRONICS USEVOL 0.337 MCPLME PWR 10.000 CMPNTS	8.451 PRODE 0. CMPID	4.722 MENEL 0.700 0.000 PNRFAC 2.500	DESRPE 0.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 EMMTHT	0.0 ECMPLX 1.546	PRMF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LOURUE	0.950 ECME 0.900	ECHS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPPOJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
DRAFTING DESIGN SYSTEMS PROJ MGMT DATA SUBTOTAL (ENG)	64. 225. 59. 38. 20. 405.	PRODUCTION 3. 25. 0. 17. 1. 51.	71. 250. 59. 55. 21. 437.
MANUFACTURING PRODUCTION PROTOTYPE TOOL—TEST EG SUBTOTAL (MFG)		279. 0. 9.	
TOTAL COST	430.	339.	769.
UOL 8.100 AUCOST NT 8.510 ECNE	3.49 TOTAL 0.092 ECHS	AU PROD COST 4.24 0.026 DESRPE 0.000	LCURUE 0.950 DESRPS 0.500
MECH/STRUCT NS 1.230 NSCF	12.300 MECID	0.000 PRODS 4.508	MCPLXS 5.997
ELECTRONICS WE 1.280 WECF PWP 10.000 CMPNTS	38.000 CMPID 875.		MCPLXE 8.451 CMPEFF 15.166
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF			PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 388. 430. 487.	PRODUCTION 293. 339. 397.	TOTAL COST 682. 769. 884.

Figure A-6. Synthesizer

INPUT DATA

OTY ,LCURUE	00. 0.850	PROTOS	2.00	BÚCOST	16.000	AMULTM 115.00%
ИТ	50.000	QTYSYS	1.			
MECH/STRUCT						
NS ELECTRONICS	30.000	INTEGS	0.300	HCPLXS	6.055	
NE NE	20.000	INTEGE	0.300	HOPLNE	7.900	
PROGRAM COS PUPCH ITEM	Т	DEVEL	LOPMENT		ODUCTION	TOTAL COST 1509.

Figure A-7. Receiver/Exciter/Synthesizer Purchase Parts

INPUT DATA OTY 80. PROTOS OTYSYS 1. INTEGE	2.0 NT 0.400 INTEGS	94.000 UOL 5.670 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 55.000 MCPLXS	5.725 PRODS	4.371 NEWST 0.500	DESRPS 0.500
ELECTRONICS USEVOL 0.181 MCPLXE PNR 90.000 CMPNTS			DESRPE 0.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 ENMTHT	0.0 ECMPLX 1.240	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECHE 0.000	ECNS 0.000
GLOSAL YEAR 1978. ESC PLATFN 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
DRAFTING DESIGN SYSTEMS PROJ MGMT DATA SUBTOTAL (EMG)	684. 2238. 458. 300. 145. 3825.	76. 240. 0. 246. 17. 579.	760. 2478. 458. 546. 162. 4404.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EO SUBTOTAL (MFG)	0. 248. 98. 345.	4509. 0. 79. 4588.	4509. 248. 176. 4933.
TOTAL COST	4171.		9337.
UOL 5.670 AUCOST NT 94.000 ECME	56.37 TOTAL 0.085 ECNS	AU PROD COST 64.58 0.024 DESRPE 0.000	LCURVE 0.950 DESRPS 0.500
MECH/STRUCT US 55.000 HSCF	9.700 MECID	0.000 PPODS 4.371	MCPLXS 5.725
ELECTRONICS WE 39.000 WECF PWR 90.000 CMPNTS			MCPLXE 8.173 CMPEFF -7.418
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	15.860 ENMTH 66.000 AVER.	T 15.860 ECMPLX 1.240 PROD RATE PER MONTH	PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEUELOPMENT 3747. 4171. 4758.	PRODUCTION 4408. 5167. 6171.	TOTAL COST 8155. 9337. 10929.

Figure A-8. Signal Processor

INPUT DATA OTY SO. PROTOS OTYSYS 1. IHTEGE	2.0 MT 20.00 0.400 INTEGS 0.30	0 VOL 0.700 0 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT NS 9.400 MCPLMS	5.730 PRODS 4.88	3 NEWST 1.000	DESRPS 2.000
ELECTRONICS USEVOL 0.303 MCPLXE PWR 100.000 CMPNTS	7.810 PRODE 4.17 0. CMPID 0.00	6 NEWEL 1.000 0 PNRFAC 0.700	DESRPE 2.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 EMMTHT 0.	0 ECMPLX 1.250	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE 0.95	10 ECME 0.000	ECNS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.00 1.000 PPROJ 1.00	00 DATA 1.300 00 PBATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST	DEVELOPMENT	PRODUCTION	TOTAL COST
ENGINEERING DRAFTING DRAFTING DESIGN SYSTEMS PROJ MGMT DATA SUBTOTAL(ENG)	167. 541. 116. 76. 37. 938.	16. 45. 0. 56. 4. 121.	183. 586. 116. 132. 41. 1959.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EQ SUBTOTAL(MFG)	0. 58. 25. 82.	1030. 0. 27. 1056.	1939. 58. 51. 1139.
TOTAL COST	1020.	1177.	
UOL 0.700 AUCOST NT 20.000 ECHE	12.87 TOTAL AU A	POD COST 14.72 024 DESRPE 0.333	LCURVE 0.950 DESRPS 0.233
MECH/STRUCT US 9.400 WSCF	13.429 MECID 0.	.000 PRODS 4.283	MCPLXS 5.730
ELECTRONICS WE 10.600 WECF PWR 100.000 CMPNTS	50.000 CMPID 0.	.000 PRODE (4.176 PNRFAC 0.700	MCPLXE 7.810 CMPEFF 10.085
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	15.193 ENMTHT 15 66.000 AVER. PRO	,193 ECMPLX 1.250) RATE PER MONTH	PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 908. 1020. 1185.	PRODUCTION 998. 1177. 1440.	TOTAL COST 1905. 2197. 2625.

Figure A-9. Power Supplies

IMPUT DATA OTY 80. PROTOS OTYSYS 1. IMTEGE	2.0 NT 0.600 INTEGS	40.000 VOL 1.800 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT WS 19.000 MCPLMS	5.984 PRODS	4.543 MEWST 1.000	DESRPS 2.000
ELECTRONICS USEVOL 0.259 MCPLXE PNR 150.000 CMPMTS	8.661 PRODE 8. CMPID	4.710 NEWEL 1.000 0.000 PWRFAC 0.500	DESRPE 2.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 ENMTHT	0.0 ECMPLX 1.237	PRMF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECNE 0.000	ECNS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING DRAFTING DESIGN SYSTEMS PROJ MGNT DATA SUBTOTAL (ENG)	DEVELOPMENT 272. 906. 176. 122. 57. 1533.	PRODUCTION 33. 107. 0. 200. 13. 352.	
MANUFACTURING PRODUCTION PROTOTYPE TOOL—TEST EO SUBTOTAL (MFG)	0. 202. 35. 287.	3938. 0. 75. 4013.	
TOTAL COST	1821.	4365.	6186.
UOL 1.800 AUCOST NT 40.000 ECME	49.22 TOTAL 0.098 ECNS	AV PROD COST 54.56 0.027 DESRPE 0.536	LCURVE 0.950 DESRPS 0.395
MECH/STRUCT WS 19.000 NGCF ELECTRONICS	10.556 MECID	0.000 PRODS 4.543	MCPLXS 5.984
NE 21.000 NECF PNR 150.000 CMPNTS	45.000 CMPID 1188.	0.000 PRODE 4.710 PURFAC 0.500	MCPLNE 8.661 CMPEFF -3.400
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	16.871 ENMTH 66.000 AVER.	T 16.871 ECMPLX 1.237 PROD RATE PER MONTH	PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 1631. 1821. 2030.	PRODUCTION 3705. 4365. 5214.	TOTAL COST 5336. 6186. 7293.

Figure A-10. Radar Transmitter

INPUT DATA OTY 80. PROTOS OTYSYS 1. INTEGE	2.0 WT 0.500 INTEGS	5.000 VOL 0.050 0.300 AMULTE 106.50%	MODE 1. AMULTM 106.50%
MECH/STRUCT NS 0.800 MCPLXS	5.737 PRODS	4.240 MENST 1.000	DESRPS 2.000
ELECTRONICS USEVOL 1.867 MCPLXE PWR 2.000 CMPNTS	8.462 PRODE 25. CMPID	4.602 MENEL 1.000 0.000 PNRFAC 0.000	DESRPE 2.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 ENMTHT	0.0 ECMPLX 1.238	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURVE	0.950 ECNE 0.000	ECNS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST			
		13. 41. 0. 41. 3. 98.	
MANUFACTURING PRODUCTION PPOTOTYPE TOOL—TEST EQ SUBTOTAL(NFG)	0. 38. 17. 55.	754. 9. 20. 774.	754. 38. 37. 829.
TOTAL COST	660.	871.	
UOL 0.950 AUCOST WT 5.000 ECHE	9.48 TOTAL 0.098 ECNS	AU PROD COST 10.89 0.024 DESRPE 0.347	LCURVE 0.950 DESRPS -0.190
MECH/STRUCT WS 0.800 WSCF	16.000 MECID	0 000 PPOTS 4 240	MCPLYS 5 737
ELECTRONICS WE 4.200 WECF			
PWR 2.000 CMPNTS			CMPEFF-52.769
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	16.448 ENMTH 66.000 AVER.	T 16.448 ECMPLX 1.238 PROD RATE PER MONTH	PRNF 0.000 2.500
COST RANGES FROM CENTER TO	BEVELOPMENT 591. 660. 755.	PRODUCTION 743. 871. 1640.	TOTAL COST 1334. 1531. 1794.

Figure A-11. IFF Transmitter

INPUT DATA ATY SO. PROTOS ATYSYS 1. INTEGE	ટ.૭ ЫТ છ.500 INTEGS	15.000 VOL 2.000 0.300 ANULTE 106.50%	MODE 1. ANULTM 106.50%
MECH/STRUCT US 12.100 MCPLKS	5.399 PRODS	4.250 NEWST 1.000	DESRPS 2.000
ELECTRONICS USEVOL 0.036 NCFLME PNR 5.300 CMPNTS	8.455 PRODE 0. CMPID	4.686 NEWEL 1.000 0.000 PWRFAC 0.500	DESRPE 2,000 CNPEFF 0,000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 EMMINT	0.0 ECMPLX 1.340	PRMF 0.000
PRODUCTION FEMTHS 34.6 PRMTHE	66.0 LCUPUE	0.950 ECHE 0.900	ECHS 0.008
GLOBAL YEAR 1978. ESC PLATEM 1.950 SYSTEM	U.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST	DEVELOPMENT	PRODUCTION	TOTAL COST
EHGINEERING DRAFTING DESIGN SYSTEMS PROJ MGMT DATA SUBTOTAL(EMG)	105. 355. 78. 51. 25. 614.	f. •	38 3. 78. 36.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EO SUBTOTAL (MFG)	0. 36. 17. 53.	630. 0. 17. 647.	630. 36. 34. 701.
TOTAL COST	667.	727.	1394.
NOL 2.000 AUCOST NT 15.000 CCHE	7.88 TOTA 0.092 ECNS	L AV PROD COST 9.09 0.020 DESRPE 0.326	LOURUE 0.950 DESRPS 0.254
MECH/STRUCT NS 12.100 NSCF	6,050 HECI	0 0.000 PRODS 4.250	MCPLXS 5.399
ELECTRONICS NE 2.900 NECF PNR 5.300 CMPHTS	48.800 CMPI 42.	D 0.000 PRODE 4.686 PWRFAC 0.500	MCPLME 8.455 CMPEFF-29.165
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF	17.786 ENMT 66.000 AVER	HT 17.786 ECMPLX 1.340 . PROD RATE PER MONTH) PRNF 0.000 2.500
COST RANGES FROM CENTER TO	DEVELOPMENT 599. 667. 762.	625. 727.	TOTAL COST 1224. 1394. 1620.

Figure A-12. Waveguide Switch

IMPUT DATA

QTY 80. PROTOS 2.00 BUCOST 26.000 AMULTM 122.00%.

NT 2570.000 0TYSYS 1.

MECH/STRUCT US 2570.000 INTEGS 0.300 MCPLMS 4.731

PROGRAM COST DEVELOPMENT PRODUCTION TOTAL COST PURCH ITEM 63. 2538. 2601.

Figure A-13. Waveguide

INPUT DATA OTY SO. PPOTOS OTYSYS 1. INTEGE	2.0 HT 50: 0.000 INTEGS	37.000 VOL 65000.000 0.700 AMULTE 106.50%	MODE 2. AMULTM 106.50%
MECH/STRUCT US 5087.000 NCPLXS	5.114 PRODS	5.341 MEWST 0.300	DESRPS 0.600
ENGINEERING ENMTHS 1.0 ENMTHR	0.0 ENMTHT	0.0 ECMPLX 1.227	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.0 LCURUE	0.950 ECME 0.000	ECHS 0.000
GLOBAL YEAR 1978, ESC PLATFM 1.950 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
DRAFTING	132.	12.	144.
DESIGN SYST EMS	369.	43.	412.
Print R. J. J. Land 1988	109.	0. 467.	109. 617.
PROJ NGMI DATA SUBTOTAL(EMG)	57.	29.	86.
SUBTOTAL (EHG)	819.	550.	1369.
MANUFACTURING PRODUCTION PROTOTYPE TOOL-TEST EQ SUDTOTAL (MFG)	0. 1039. 195. 1234.	10321. 0. 60. 10391.	10321. 1039. 255. 11615.
TOTAL COST	2053.		12984.
UOL 65000.000 AUCOST WT 5007.000 ECHE	129.01 TOTAL 0.001 ECMS	AU PROD COST 136.63 0.018 DESRPE 0.000	LCURVE 0.950 DESRPS 0.600
MECH/STRUCT WS 5087.000 WSCF	0.078 MECID	0.000 PRODS 5.341	MCPLXS 5.114
SCHEDULES ENMTHS 1.000 ENMTHP PRMTHS 34.000 PRMTHF			
COST RANGES FROM CENTER TO	DEVELOPMENT 1876. 2053. 2274.	9748. 10930.	TOTAL COST 11624. 12984. 14449.

Figure A-14. Antenna

INPUT DATA

OTY •LCURUE UT	80. 0.850 25.000	PROTOS OTYSYS	2.00 1.	BUCOST	14.000	AMULTM 122.00%
MECH/STRUCT NS ELECTRONICS NE	3.600	INTEGS	0.300	HCPLXS	5.693	
PROGRAM COST	21.400 T	INTEGE DEVEL	0.500 OPMEHT		7.850 ODUCTION	TOTAL COST
PURCH ITEM		34.		1366.		1401.

Figure A-15. Data Processor

IMPUT DATE OTY SO. PEOTOS OTYSYS 1. IMTEGE	2.0 IWT 22 0.000 INTEGS	31.270 IVOL 14.150 8.800 AMULTE 106.58%	MODE 5. AMULTM 106.50%
MECH/STRUCT IWS 818.847 MCPLXS	4.658 PRODS	0.000 NEWST 0.300	DESRPS 0.000
ELECTROHICS I-UUOL 0.018 MCPLXE PWR 0.000 CMPHTS	7.688 PRODE 0.000 CMPID	0.000 NEWEL 0.300 0.000 PURFAC 0.000	DESRPE 0.000 CMPEFF 0.000
ENGINEERING ENMTHS 1.0 ENMTHP	0.0 EMMTHT	0.0 ECMPL% 0.900	PRNF 0.000
PRODUCTION PRMTHS 34.0 PRMTHF	66.8 LOURUE	0.000 ECME 0.000	ECMS 0.000
GLOBAL YEAR 1978. ESC PLATFM 1.800 SYSTEM	0.00% PROJET 1.000 PPROJ	1.000 DATA 1.000 1.000 PDATA 1.000	TLGTST 2.000 PTLGTS 1.00
PROGRAM COST	DEVELOPMENT	PRODUCTION	
ENGINEERING DRAFTING DESIGN SYSTEMS PROJ HGHT DATA SUBTOTAL(ENG)	50. 145. 21. 20. 9. 244.	19. 57. 6. 65. 3. 144.	69. 201. 21. 35. 12. 388.
MANUFACTURING PRODUCTION PPOTOTYPE TOOL-TEST EO SUBTOTAL (MFG)	Ñ.	1184.	1184.
PPOTOTYPE	74.	0.	74.
SUBTOTAL (MFG)	30. 194.	0. 29. 1813.	59 . 1317 .
TOTAL COST	348.		
COST RANGES FROM CENTER TO		PRODUCTION	TOTAL COST 1440.
TO CEMILER	348. 411.	1357. 1722.	17 05. 2133.

Figure A-16. Integration and Test

TOTAL COST, LESS I	NTECRATION COST		
FRUGERAN CUST	DEVELOPMENT	Discount to the second	
ENGINEERING	are the factor of the fit	PRODUCTION	TOTAL COST
DRAFTING DESIGN	1780.	199.	4.00
SYSTEMS	5824.	636 .	1979. 6460.
PPOJ INGHT	1250.	0.	125s.
DATA	914.	1132.	2046.
SUBTOTAL (ENG)	431. 10207.	79.	504.
	* Salas al I ·	2040.	12246.
MANUFACTURING PRODUCTION			
PROTOTYPE	0.	22646.	22646.
TOOL-TEST EQ	1697.	EI.	1697.
PURCH ITEMS	473.	325.	797 .
SUBTOTAL (MFG)	134.	5376.	5510.
	2305.	28347.	30651.
TOTAL COST	12511.	30386.	
			42398.
COST RANGES	DEVELOPMENT		
FROM	11279.	PRODUCTION	TOTAL COST
CENTER	12511.	27136.	38415.
TO	14198.	30386. 34349.	42898.
		VT242.	48547.
TOTAL COST, WITH IN	TEODOTTAL		
PROGRAM COST	DEUKHITUH COST		
ENGINEERING	DEVELOPMENT	PRODUCTION	TOTAL COST
DRAFTING	. 1830 .		
DESIGN	5968.	218.	2048.
SYSTEMS	1278.	693 .	6661.
PROJ MGMT	935.	0. 1196.	1278.
DATA	439.	76.	2131.
SUBTOTAL (ENG)	19450.	2183.	516.
MANUFACTURING			12634.
PRODUCTION			
PROTOTYPE	0.	23831.	23831.
TOOL-TEST EQ	1771. 503.	ð.	1771.
PURCH ITEMS	134.	353.	856.
SUBTOTAL (NFG)	2409.	5376.	5510.
70701		29560.	31969.
TOTAL COST	12959.	31744.	incon
			44603.
COST PANGES	DEVELOPMENT	Different to The Table	
FROM	11586.	PRODUCTION	TOTAL COST
CENTER	12859.	282 68. 31 744.	39354.
TO	14608.	347 44. 36072.	44603.
			50680.

Figure A-17. Total Cost

MISSION

Of

Rome Air Development Center

RADC plans and conducts research, exploratory and advanced development programs in command, control, and communications (c³) activities, and in the C³ areas of information sciences and intelligence. The principal technical mission areas are communications, electromagnetic guidance and control, surveillance of ground and aerospace objects, intelligence data collection and handling, information system technology, ionospheric propagation, solid state sciences, microwave physics and electronic reliability, maintainability and compatibility.

action content on the content of the

