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A STUDY OF PRESSURE-VOLUME RATES AND PLENUM
MEMBRANE ADDITIONS TO THE CAPTURED AIR
BUBBLE SURFACE EFFECT SHIP XR-3 DIGITAL
COMPUTER LOADS AND MOTION PROGRAM

John Martin Boggio

NAVAL POSTGRADUATE SCHOOL

Monterey, California



THESIS

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AIR BUBBLE SURFACE EFFECT SHIP XR-3
DIGITAL COMPUTER LOADS AND MOTION PROGRAM

by

John Martin Boggio

June 1976

Thesis Advisor:

A. Gerba, Jr.

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A Study of Pressure-Volume Rates and Plenum
Membrane Additions to the Captured Air
Bubble Surface Effect Ship XR-3 Digital
Computer Loads and Motion Program

by

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Submitted in partial fulfillment of the
requirements for the degree of

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June 1976

ABSTRACT

A study was conducted of modeling changes to the XR-3 Captured Air Bubble Surface Effect Ship digital computer Loads and Motion program. These changes included the addition of pressure rate and volume rate to the existing 6 degrees of freedom equations. Additional equations were developed to simulate the application of a nonpermeable membrane to the plenum of the XR-3 test craft. The objectives of this study were to determine whether the addition of pressure rate and volume rate equations would improve computer execution time and to test some simplified models of the plenum membrane. Computer timing improvements were demonstrated and membrane modeling results are presented.

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I. INTRODUCTION

A. BACKGROUND

The Captured Air Bubble Surface Effect Ship (CAB SES) offers a dynamic new approach to improve surface ship performance. As with any new approach, extensive testing at the design stage of development is required. The U.S. Navy has been conducting sea trials on two 100 ton Captured Air Bubble (CAB) test craft, the 100-A and the 100-B [Ref. 1]. In addition to scale model testing, another approach to design evaluation involves the use of high speed digital computers to simulate the dynamic performance of the craft.

A digital computer Loads and Motion (L&M) program for the CAB SES was developed under government contract by the Oceanics Corp. [Ref. 2]. The L&M program installed at the Naval Postgraduate School, W. R. Church, computer facility on an IBM 360/67 computer was for a 100 ton displacement test craft, the 100-B. Leo and Boncal [Ref. 3] modified this computer program to represent a smaller 3 ton craft, the XR-3. The XR-3 is an operational test craft currently maintained and operated by the Naval Postgraduate School [Ref. 4].

Various modifications and changes to Leo and Boncal's basic work have been implemented. Finley, Forbes, and Menzel, in Refs. 5-7, provides program changes to obtain better representation of the Fan maps, bow and stern seal dynamics and pitch and roll damping.

One adverse characteristic of both the 100-B computer program and the XR-3 computer program was the extensive use of the digital computer time when operating in simulated sea states. Since the 100-B and the XR-3 programs are similiar, the computer time analysis provided by Mitchell [Ref. 8] for the 100-B also applies to the XR-3 program.

Another aid to design is the testing done in towing tanks in which scale models are used. Towing tests were conducted at NSRDC using a rubber membrane installed on the forward portion of a CAB SES model in order to better scale the pressure-volume relationship [Ref. 9]. The addition of this membrane improved the pitch and heave characteristics of the scale model in sea state operation.

B. OBJECTIVES

It is the objective of this work to examine the L&M digital computer program and make any changes that would improve the program with particular attention to changes that would reduce the amount of time required to execute the program.

A second objective is to model the plenum membrane, add the required program changes and study the effects of the membrane on simulated craft performance.

II. PROGRAMMING CHANGES

A. RATE EQUATIONS FOR PRESSURE AND VOLUME

1. Introduction

An examination of the dynamics of the XR-3 L&M program indicate that the key parameters of this type of ship design are those which describe the dynamic behavior of the bubble of the air in the plenum. This pressure of air provides approximately 75% of the lift force when the craft is "on the bubble." Small changes in bubble volume and pressure therefore cause significant changes in total ship response. These variables were examined in an effort to decrease the digital computer simulation execution time.

2. SUBROUTINES RHS and INTGRL

The force and moment equations for the craft are the six degrees of freedom equations plus four auxillary equations contained in SUBROUTINE RHS [Ref. 2].

The integration of these equations is performed in SUBROUTINE INTGRL. INTGRL uses the Runga-Knutta-Merson numerical integration technique with an automatic variable step size. If the error tolerances specified are exceeded the step size is reduced and the step repeated until all error tolerances are met or the step size is reduced to 1×10^{-6} sec., at which point the program stops. Thus the integration values are forced to converge to their correct values. All ten equations are integrated serially at the

same time step. When all integration error tolerances are met INTGRL returns the integrated values and increases the simulation time one time-step increment. Since pressure and volume are key variables, integrating their rates should cause the system of equations to converge more rapidly to their true values and avoid any problems associated with the feedback loops in the pressure-volume calculation [Ref. 10].

3. Formulation of Volume Rate and Pressure Rate

The bubble pressure equation is for an adiabatic process [Ref. 2].

$$P_b = P_a \left(\frac{\rho_b}{\rho_a} \right)^\gamma \quad (1)$$

where the subscripts a and b represent atmosphere and plenum values respectively

P = pressure absolute

$$\rho = \text{density } \frac{M}{V} \quad (2)$$

γ = adiabatic exponent for air

Substituting (2) into (1)

$$P_b = P_a \left(\frac{\frac{M}{V} \rho_a}{\rho_a} \right)^\gamma \quad (3)$$

Differentiating (3) with respect to time where P_a and ρ_a are assumed to be constant yields

$$\dot{P}_b = \frac{P_a}{\rho_a} \gamma \left(\frac{\frac{M}{V} \rho_a}{\rho_a} \right)^{\gamma-1} \left(\frac{\dot{V}M - M\dot{V}}{V^2} \right) \quad (4)$$

Factoring $\frac{M}{V}$ from the last term of (4)

$$\dot{P}_b = \gamma P_a \left(\frac{M}{\rho_a V} \right)^{\gamma-1} \left(\frac{M}{\rho_a V} \right) \left(\frac{\dot{M}}{M} - \frac{\dot{V}}{V} \right) \quad (5)$$

Noting $\left(\frac{M}{\rho_a V} \right)^{\gamma-1} \left(\frac{M}{\rho_a V} \right) = \left(\frac{M}{\rho_a V} \right)^\gamma$

and

$$P_a \left(\frac{M}{\rho_a V} \right)^\gamma = P_b$$

$$\dot{P}_b = \gamma P_b \left(\frac{\dot{M}}{M} - \frac{\dot{V}}{V} \right) \quad (6)$$

The volume equation for the bubble is computed in RHS in four parts and summed together

$$V = V_{nom} - D - WA + WT \quad (7)$$

The four terms that determine volume are

- (1) V_{nom} = the empty plenum volume
- (2) D = draft term-change in volume due to draft
- (3) WA = Wave term-change in volume due to the presence of waves
- (4) WT = WATSLP term-- correction term for the added volume wedge which is a function of speed
[Ref. 6].

Differentiating equation (4)

$$\dot{V} = -\dot{D} - \dot{WA} \quad (8)$$

where

\dot{W}_A = rate of change of volume due to waves
 \dot{D} = rate of change of volume due to displacement
 \dot{W}_T = 0.0
 \dot{V}_{nom} = 0.0

\dot{W}_A is computed in SUBROUTINE WAVES by taking the difference in change of volume due to waves at successive time steps and dividing by the time step. Since WATSLP is a function of speed, the differentiated value of WATSLP, \dot{W}_T , is a function of acceleration. Forward accelerations are normally small; therefore, the differentiated value of WATSLP was approximated to be zero. Equations (4) and (5) were programmed as follows:

```
VALUE (11) = -(((XL*WIDTH)-ABW)*.5*W)-DVWDOT
VALUE (12) = GAM*VAL(13)*(VALUE(10)/VAL(11)-VALUE(11)/
                  VAL(12))
```

where

DVWDOT = rate of change of volume due to waves
VALUE(10) = Bubble mass flow rate
VALUE(11) = Volume rate
VALUE(12) = Pressure rate
VAL(11) = Bubble Mass
VAL(12) = Bubble Volume
VAL(13) = Bubble Pressure
GAM = Adiabatic Exponent for air

B. ELIMINATION OF SUBROUTINE DMINV

In an effort to improve program timing, the suggestions given in Ref. 7 were undertaken. One recommendation suggested the removal of SUBROUTINE DMINV in the 100B L&M program by making changes to SUBROUTINES INCON and RHS. An examination of the L&M program indicated that these changes also could be applied to the XR-3 program. Consequently the following was deleted:

```
212    DO 211 I = 1,6
          DO 211 N = 1,6
211    A(I,N)    = 0.0
          DO 213 N = 1.3
213    A(N,N)    = AM
          A(4,4)    = AIXX
          A(5,5)    = AIYY
          A(6,6)    = AIZZ
          A(4,6)    = -AIXZ
          A(6,4)    = -AIXZ
          AIMAX     = AMAXI(AM, AIXX,AIYY,AIZZ,ABS(AIXZ))
          DO 214 I = 1,6
          DO 214 J = 1,6
214    A(I,J)    = A(I,J)/AIMAX
          CALL DMINV(A,G,D, )
          DO 215I = 1,6
215    A(I,J)    = A(I,J)/AIMAX
          IF (D.NE.0.0.) GO TO 10
          WRITE (6,216)
          STOP
```

In place of the above the following was added:

```
215  AMASSI = 1.0/AM  
      D      = 1.0/(AIXX*AIZZ-AIXZ*AIXZ)  
      DIXX   = AIXX*D  
      DIXZ   = AIZZ*D  
      DIZZ   = AIZZ*D  
      AIYYI  = 1.0/AIYY  
      GO TO 10
```

Linkage between INCON and RHS was provided by the following

```
COMMON/ATRIX/AMASSI,AIYYI,DIXX,DIXZ,DIZZ
```

In SUBROUTINE RHS the six element matrix GF(J) was deleted and the following identifiers were substituted for the summation of forces: SUMX, SUMY, SUMZ, SUMK, SUMM, SUMN. In addition, the following deletion was made.

```
DO 1 I = 1,6  
    VALUE(I) = 0.0  
DO 1 J = 1,6  
    VALUE (I) = VALUE (I) + A(I,J)*GF(J)  
1 CONTINUE
```

Substituted for the above DO LOOPS was the following

```
VALUE (1) = SUMX*AMASSI  
VALUE (2) = SUMY*AMASSI -R*U  
VALUE (3) = SUMZ*AMASSI  
VALUE (4) = SUMK*DIZZ + SUMN*DIXZ
```

VALUE (5) = SUMM*AIYYI

VALUE (6) = SUMN*DIXX + SUMK*DIXZ

These changes eliminated SUBROUTINE DMINV and several DO LOOPS.

C. EXPANSION OF WAVE COMPONENTS

Since the program was being tested with sea state, the method used for the wave generation was studied. SUBROUTINE INCON provides a means of introducing sea-state by individual wave components and amplitudes. In addition, the subroutine will accept an average height and lowest and highest wave frequency and/or wave period and generate up to 10 wave components.

These wave components represent a sampling of the spectral energy density of the given sea state condition. Reference 11 suggests for irregular sea, a minimum of 15 - 20 components are required to simulate irregular sea conditions. The program dimension statements were accordingly increased from 10-20 to determine if any significant changes could be observed by increasing the number of wave components.

D. MISCELLANEOUS

Each subroutine was examined with regard to efficiency of coding and changes were implemented such as multiplication in lieu of the use of the exponential. For example, IBM 360/67 fixed point exponentiation requires that the natural logarithm be generated and then a logarithm be computed, as compared

to straight forward multiplication. When such exponentiation is nested in DO LOOPS, a significant time savings can be realized by the use of multiplication.

III. TEST PRODECURES

A. COMPUTER SYSTEM

The digital computer used throughout this study is an IBM 360/67 model, VERSION I, located at the NPS W. R. Church Computer Facility. The system's hardware configuration provides 2 M bytes of core. This main core is composed of IBM core storage devices (IBM 2365 Mod. 12) and compatable but slightly different Lockheed core storage devices (MM 365). The Lockheed core devices are about 18% slower in execution time than IBM core devices. Since the system is time-shared and the core is contiguous, there is no practical way to determine in which portion of the main core the program is residing. Therefore, exact timing of a program is not easily achieved. In this report all timing values must be interpreted with this anomaly in mind. Among the options available under this operating system are the FORTRAM G and FORTRAN H compilers [Ref. 12]. Mitchell reported substantial time savings using the H compiler [Ref. 8]. Since one of the main objectives was to reduce execution time, computer model testing was done using both compiler options. Runs made under different compilers will be noted.

A word of caution on the use of the H compiler. The H compiler produces its timing improvements and core reduction by coding optimization. This means the compiler re-arranges

the Fortran code. This re-arrangement may move computations outside of DO LOOPS for example. Applicable portions of Ref. 13 should be read and understood before using this feature.

B. TEST CONDITIONS

The L&M program for the XR-3 used in this report is the one given in Ref. 7. This program was then modified to incorporate the changes previously described. The basic program Ref. 7 computes pressure using eq. (1) and volume using eq. (7) and is referred to as PROGRAM ONE. PROGRAM ONE was used as reference for comparative purposes. The addition of the pressure rate and volume rate equations (6) and (8) and the elimination of SUBROUTINE DMINV constituted major modifications to the basic program. This version of the Loads and Motion program was called PROGRAM TWO. PROGRAM THREE was PROGRAM TWO compiled and executed with the FORTRAN H compiler.

Five test conditions for a given speed were established to provide a full range of disturbances for testing the L&M programs. These test conditions are tabulated in Table I.

In Condition One small disturbances were introduced by changing the initial conditions of draft (DS) and pitch (θ). These changes consisted of decreasing the draft and pitch from previously computed steady state values by the magnitude shown in Table I. In addition, the rudder was held at zero and the sea was calm. In Condition Two, the perturbations were the same as Condition One except that now the rudder

TABLE I

TEST PERTURBATIONS

| | ΔDS | $\Delta \theta$ | RUDDER | SEA STATE |
|-----------------|-------------|-----------------|--------------|---------------|
| Condition One | 0.1 inch | .03 | 0.0 | Calm water |
| Condition Two | 0.1 | .03 | 20° | Calm water |
| Condition Three | 0.0 | 0.0 | 0.0 | Regular One |
| Condition Four | 0.0 | 0.0 | 0.0 | Regular Two |
| Condition Five | 0.0 | 0.0 | 0.0 | Irregular One |

was displaced to the right 20 degrees. The rudder was offset at the start of the simulation and held through the run as this value. In Condition Three, Four and Five the only disturbances used were the sea state conditions.

Sea state is generated in SUBROUTINE WAVES. SUBROUTINE INCON generates wave components and amplitudes based on several different type input parameters. *[See users manual Ref. 7]* Regular sea one and two are simple sinusoidal waves. Irregular sea state conditions are simulated by the addition of several regular sinusoidal wave components. Sea state used throughout this study is true sea state and is not scaled in any manner.

The L&M program provides two options for propulsion, either constant speed or constant thrust. The constant

thrust option was selected because this is the normal operating mode for the XR-3 test craft.

Integrator tolerance levels were maintained at the values given in Ref. 7. However two new error tolerances had to be determined for the integration of the pressure and volume rates. These new levels were determined by first selecting a very tight value .000001 which was increased in increments until a change in output could be noticed. The value was then decreased by a factor of 10. Pressure and volume tolerance levels were then set at this value which was found to be .0001.

All other variables such as draft and pressure were initialized from the steady state undisturbed conditions for that specific starting speed. The steady state conditions were evaluated by first using the constant speed propulsion option and the initial values in Ref. 7. The program was then run until the key input variables - pitch, draft, thrust and pressure - reached steady state. These values were then read in and propulsion was switched to constant thrust. The program was run again to insure that steady state conditions existed. These steady state values are shown in Table II.

TABLE II
STEADY STATE CONDITIONS

| Speed(Knots) | 10 | 20 | 30 |
|----------------------|--------|--------|--------|
| Pressure | 23.93 | 24.84 | 24.84 |
| Draft | 8.17 | 6.12 | 5.34 |
| Thrust (each engine) | 200.31 | 218.17 | 287.22 |
| Pitch | 1.62 | .48 | .26 |

Draft in inches Pitch in degrees

IV. MEMBRANE ADDITION

A. INTRODUCTION

During towing tank tests conducted at NSRDC, a rubber membrane was installed on the forward bow of the XR-1 scale model [Ref. 9]. The installation of this membrane improved the pitch and heave characteristics of the scale model during simulated sea conditions. Reference 9 provided the following empirical expression which relates the Pressure and Volume of the membrane.

$$V = KP_b \quad (9)$$

K was experimentally determined and has the value .317 when scaled to the XR-3 test craft dimensions. (See Appendix A).

B. MATHEMATICAL MODELING

The approach used to include the effects of this membrane was to develop mathematical expressions which could be added to the already existing pressure, volume and air mass equations in the L&M program. Five mathematical expressions were developed, tested and referred to as Models One, Two, Three, Four and Five. They are explained below and the results discussed in Section V.

1. Model One

Model One is based on the premise that by correcting the plenum volume with the membrane volume, the equations

in SUBROUTINE RHS would show the effect of the membrane.

Model One added the following equation to PROGRAM ONE;

$$V_m = K P_m \quad (10)$$

Then using (10) to correct the volume of the plenum and

assuming $P_m = P_b$.

$$V = V_b + V_m \quad (11)$$

where V_b is given by eq. (7).

2. Model Two

Model Two is an extension of Model One. Based on the premise that by correcting the rate of change of plenum volume with the rate of change of membrane volume, the rate equations given in SUBROUTINE RHS and subsequent integration of these terms would show the effect of the membrane. Model Two added the following rate equation to PROGRAM TWO.

$$\dot{V}_m = K \dot{P}_m \quad (12)$$

Then using (12) to correct the rate of change of plenum volume and assuming

$$\dot{P}_m = \dot{P}_b$$

$$\dot{V} = \dot{V}_b + \dot{V}_m$$

where \dot{V}_b is given by eq. (8).

3. Model Three

Model Three was based on the premise that the volume rate eq. (8) and the pressure rate eq. (6) as computed without the membrane could be corrected for the membrane by adjusting these terms by appropriate factors within PROGRAM TWO.

Correction terms were sought that would be a function of the membrane. A logical candidate for this correction factor for the pressure was the ratio of rates of change of the membrane volume to plenum volume multiplied by plenum pressure rate.

$$PCT = \frac{\dot{V}_m}{\dot{V}_b} \cdot \dot{P}_b \quad (13)$$

where PCT is the correction factor to be added to the plenum pressure rate term \dot{P}_b before integration. It was assumed that the pressure and pressure rate of the plenum and membrane were equal.

$$P_m = P_b \quad (14)$$

$$\dot{P}_m = \dot{P}_b \quad (15)$$

It was also assumed that the membrane volume rate and pressure rate were related by

$$\dot{V}_m = K \dot{P}_m \quad (16)$$

Substituting (16) into (13) and using (15)

$$PCT = K \frac{\dot{P}_b^2}{\dot{V}_b} \quad (17)$$

The volume rate correction term was assumed to be the ratio of the rate of change of membrane pressure to the rate of change of plenum pressure times the rate of change of membrane volume.

$$VCT = \frac{\dot{P}_m}{\dot{P}_b} \quad \dot{V}_m \quad (18)$$

where VCT is the correction factor to be added to the plenum volume rate term \dot{V}_b before integration

$$\text{Since } \dot{P}_m = \dot{P}_b \text{ and } \dot{V}_m = K \dot{P}_b$$

Then from (18)

$$VCT = K \dot{P}_b \quad (19)$$

Then using equations (17) and (19) the rate equations are

$$\dot{V} = \dot{V}_b + VCT \quad (20)$$

$$\dot{P} = \dot{P}_b + PCT \quad (21)$$

where \dot{V}_b is given by eq. (8) and \dot{P}_b by eq. (6).

Consider the following. Since K is a function of the elasticity of the membrane, if $K = 0.0$. Equations (20) and (21) reduce to the no membrane case.

If the membrane expands at the same rate that the volume of the plenum contracts the system volume rates is zero and the system pressure rate is also zero.

4. Model Four

Again using PROGRAM TWO, Model Four was based on the premise that the mass of the membrane air and mass flow rate of membrane air would be important factors in any correction term to be applied to plenum volume rate and pressure rate.

Again it was assumed that the membrane effect could be described as an adiabatic reversible process. Then using Equation (6) applied to the membrane,

$$\dot{P}_m = P_m \gamma \left(\frac{\dot{M}_m}{M_m} - \frac{\dot{V}_m}{V_m} \right) \quad (22)$$

where m denotes membrane

\dot{M}_m = Mass flow rate of the membrane air

M_m = Mass of the membrane air

Again it was assumed that the pressure and pressure rate of the membrane and plenum were equal.

$$P_m = P_b \quad (23)$$

$$\dot{P}_m = \dot{P}_b \quad (24)$$

It was also assumed that the membrane volume rate and pressure rate were related by

$$\dot{V}_m = K \dot{P}_m \quad (25)$$

Equation (25) was obtained by differentiating

$$V_m = K P_m \quad (26)$$

Substituting (25) and (26) into (22) yields

$$\frac{\dot{M}_m}{M_m} = \left(\frac{1+\gamma}{\gamma} \right) \frac{\dot{P}_m}{P_m} \quad (27)$$

One term is not available, M_m . Two possibilities were considered to determine this value. The first possibility would be to integrate \dot{M}_m and use the result of the integration from the preceding time step. The other possibility would be to hold temperatures constant in the membrane, and use the ideal gas law to determine M_m for varying pressure and volume. This last condition would be in error; however, as a first approximation, it would produce acceptable results.

Because of the past history of difficulty with integrator 10, plenum air mass rate, the second possibility was chosen. Evaluation of the terms in the ideal gas law for a temperature of 68° degrees excluding mass, pressure and volume produced the constant.

$$4.9 \times 10^{-7} \text{ slugs/psf - ft}^3$$

therefore;

$$\dot{M}_m = (4.9 \times 10^{-7}) (P_m) (V_m) \quad (28)$$

Equations (22), (25) and (27) describe the membrane. These equations are used to correct the mass rate and volume rate of the plenum without the membrane.

$$\dot{M}_s = \dot{M} + \dot{M}_m \quad (29)$$

$$\dot{V}_s = \dot{V} + \dot{V}_m \quad (30)$$

where

s = denotes the system of plenum plus membrane

\dot{M} = mass flow rate of plenum

\dot{V} = volume rate of change of plenum

Using (29) and (30) in equation (6) the corrected plenum pressure rate is

$$\dot{P}_s = P_b \gamma \left(\frac{\dot{M}_s}{M} - \frac{\dot{V}_s}{V} \right)$$

where P_b , M and V are values obtained from the previous time-step calculation.

5. Model Five

Model Five is the application of equation (10) to PROGRAM TWO. It is based on the premise that the volume of the membrane could be introduced into the plenum pressure rate equation, and represent the membrane effect.

$$V_m = K P_m \quad (31)$$

$$V = V_b + V_m \quad (32)$$

Equation (31) and (32) are computed as in Model One. Then equation (32) is used in the pressure rate equation.

$$\dot{P} = P_b \gamma \left(\frac{\dot{M}}{M} - \frac{\dot{V}}{V} \right) \quad (33)$$

V. DISCUSSION OF RESULTS

A. PRESSURE RATE AND VOLUME RATE CHANGE

The addition of the pressure rate and volume rate equations was implemented without any major difficulty. In order to provide a basis of comparison for computer time analysis a quality factor "Q" was established. "Q" is defined to be the ratio of CPU execution time to problem time. For example, if the problem specified 1 minute of simulation and the CPU execution time (GO STEP in IBM COMPUTER SYSTEMS) was 5 minutes $Q = 5.0$. PROGRAMS ONE, TWO and THREE were tested with identical input parameters. The results are summarized in Table III and Table IV.

Runs were compared using PROGRAM ONE as a reference. No differences in computed output were noted for Condition One thru Four and only small changes in computed values noted with Condition Five. The integrator error tolerance levels were maintained constant for all runs.

As can be seen from the tabulated values, the addition of pressure rate and volume rate equations reduces the CPU execution time considerably for small disturbances. However, as the sea state is increased the CPU time increases. Apparently other factors in the program begin to dominate the total execution time.

TABLE III

TIMING QUALITY FACTORS20 Knots

| | Program One | Program Two | Program Three |
|-----------------|-------------|-------------|---------------|
| Condition One | 11.25 | 4.8 | 4.50 |
| Condition Two | 13.85 | 6.0 | 5.50 |
| Condition Three | 23.0 | 20.5 | 19.2 |
| Condition Four | 42.8 | 28.01 | 22.0 |
| Condition Five | 71.0 | 58.1 | 54.0 |

TABLE IV

TIMING QUALITY FACTORS30 Knots

| | Program One | Program Two | Program Three |
|-----------------|-------------|-------------|---------------|
| Condition One | 10.8 | 7.5 | 4.5 |
| Condition Two | 18.0 | 14.25 | 8.7 |
| Condition Three | 24.0 | 22.0 | 21.0 |
| Condition Four | 45.0 | 30.0 | 28.0 |
| Condition Five | 75.0 | 59.2 | 53.0 |

This supports the results of Reference (7) that for sea conditions the bulk of computation time is apparently in the SUBROUTINE WAVES.

It was observed that at irregular sea state two and regular sea state three water came in contact with the top of the plenum. The L&M program does not include the effect of this phenomenon in the program simulation; therefore, testing was not conducted at higher sea state conditions. The fact that water hits the top of the plenum is a reasonable condition under certain circumstances. For example, if the pitch angle is 3 degrees up the stern sinks over 6 inches. If the draft of the craft at that time is 11 inches and if the preceding wave passing the stern station is 5 inches, water should hit the top of the plenum at the stern station. The effect of this condition on total craft behavior in terms of forces and moments is not known. The effect of water contact would depend upon such factors as the length of time the wave is in contact with the top of the plenum and the area of the plenum in contact with the wave.

B. INCREASING WAVE COMPONENTS

Tests were made for irregular seas one and two using 8, 10, 15, and 20 wave components. No apparent change in output could be determined. The time at which changes in output occurred changed, but no significant reduction in output values were noted; that is the time when wave components were in-phase or out-of-phase was different, but the craft response

was the same. It was felt that if more components were used, the energy (in this case, the magnitude and frequency of the disturbing force) would be distributed over many components and that the irregular combination of these components would spread the energy of the disturbance over a larger time base. This spreading of the energy would then reduce the magnitude of the disturbance and that in turn would provide less rapid rates of change. With less rapid changes in the disturbances, the integration would take less time. The over-all time increase due to more wave components would be offset by the reduction in integration time. This is not the case; using more wave components increased CPU time. Whether 10 or less wave components are sufficient to accurately represent irregular seas was not tested.

C. MEMBRANE MODELS

A large amount of time was expended in attempting to provide adequate modeling of the plenum membrane. Model One using $V_m = K P_b$ in PROGRAM ONE did not work at all. The program would begin to execute and then stop when the integration step size was less than 1×10^{-6} .

As an aid to find the reason for this failure a computational flow diagram was drawn (See Figure 1). The addition of the correction term V_m creates an algebraic inner loop as shown by the dashed line in Fig. 1. Using a small perturbation of D in calm water and using the print switches in RHS as well as using the DEBUG option available with the G compiler, it

was observed that with Model One the values of pressure were divergent. It is not known why pressure values diverge; however, it was noted that at certain times the values of pressure and volume were increasing or decreasing together, that is when pressure was increasing while volume was increasing.

The computational flow diagram for Model Two (Figure 2) shows the inner connections of the pressure, volume and mass variables. Model Two exhibited the same symptoms and the same results as Model One, except in some cases pressure rate and volume rate rose or fell together at the same time step.

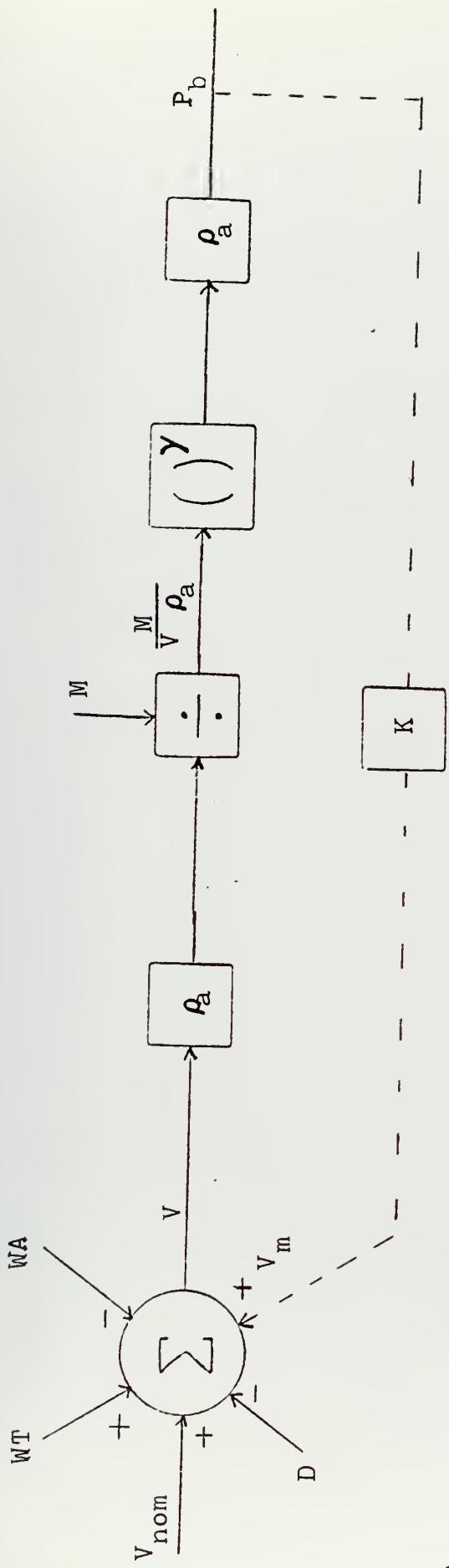
The failures of Model One and Two led to Model Three. It was felt that by correcting both pressure rate and volume rate the divergence exhibited in Models One and Two could be corrected. Model Three also failed; however, it provided the motivation to consider mass rate into the membrane. The computations performed in Model Three is shown in Figure 3.

Model Four was marginally successful. The program executes and slightly reduces the center of gravity acceleration. the reduction of the center of gravity acceleration was chosen as the criterion of whether the model was representing the membrane. This model was tested in sea state one and sea state two and provides slight reduction in the maximum values in the center of gravity acceleration. It did not reduce all of the center of gravity values. The computational flow diagram

of Model Four is given in Figure 4 and shows no algebraic loops in the RHS computations of pressure and volume rates.

Model Five was the most successful model. The computational flow is shown in Figure 5. Model Five reduced the center of gravity acceleration and for small disturbances, the pitch angle and pitch rate. For sea state one a significant reduction was observed. (See Figures 6 and 7). However, for sea state two the center of gravity accelerations were greater for the membrane when compared to the no membrane case. (See Figures 8 and 9). No reason for this could be found. The only differences in these two runs was the sea state.

One possible reason for the sea state two results, though not yet proven, could be a resonance effect at the sea state encounter frequency. Concurrent work in this area [Ref. 14] investigates the conditions for the existence of any resonance phenomena. Preliminary investigations indicate that at a wave encounter frequency of approximately 4 rad / second resonance in pitch angle occurs. The encounter frequency for regular sea state two is 5 rad / second. Whether resonance phenomena could cause this effect, however, was not determined.



(All values shown for the Kth iteration in RHS)

Figure 1. Computational Flow Diagram Model I

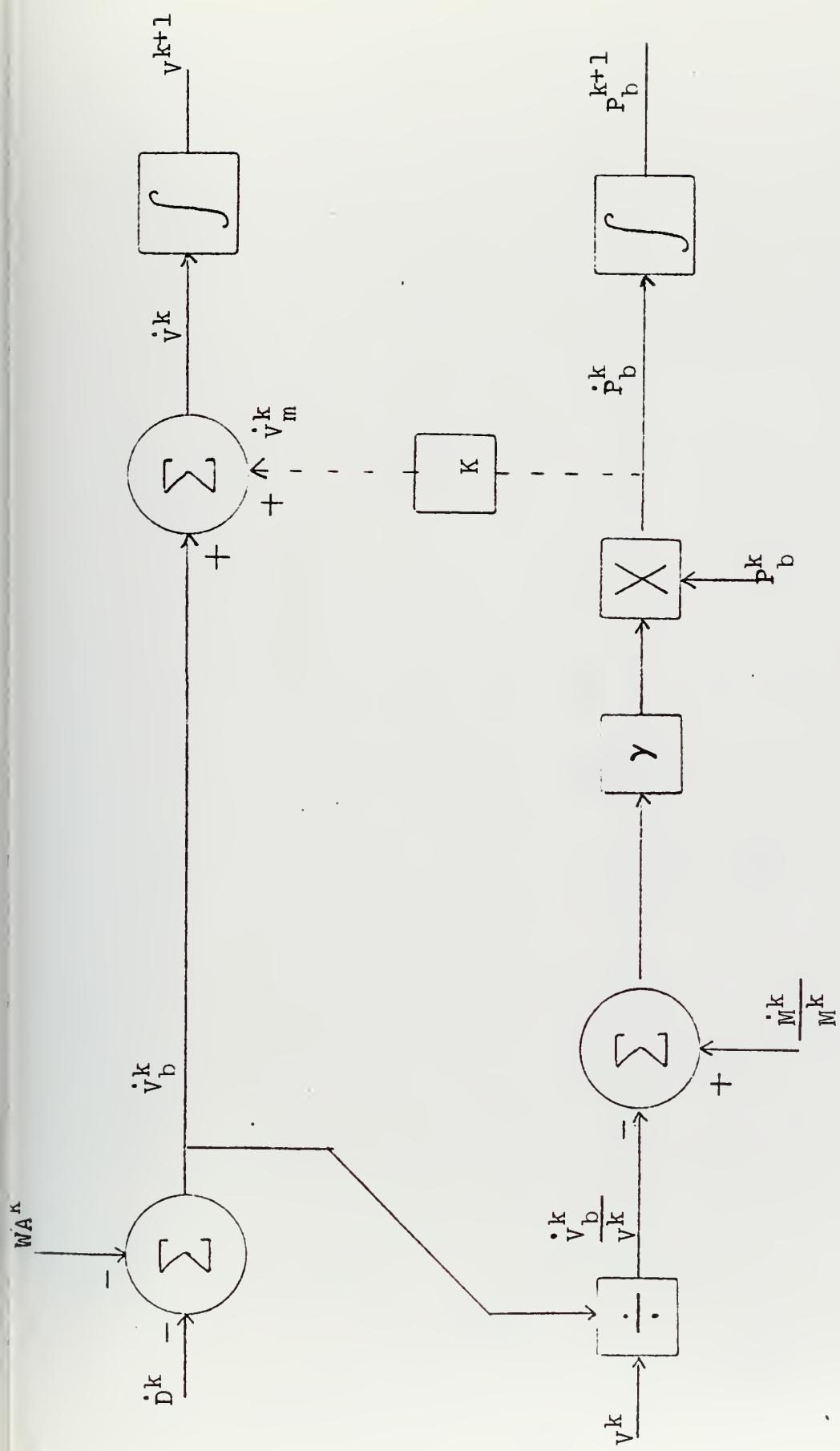


Figure 2. Computational Flow Diagram Model II

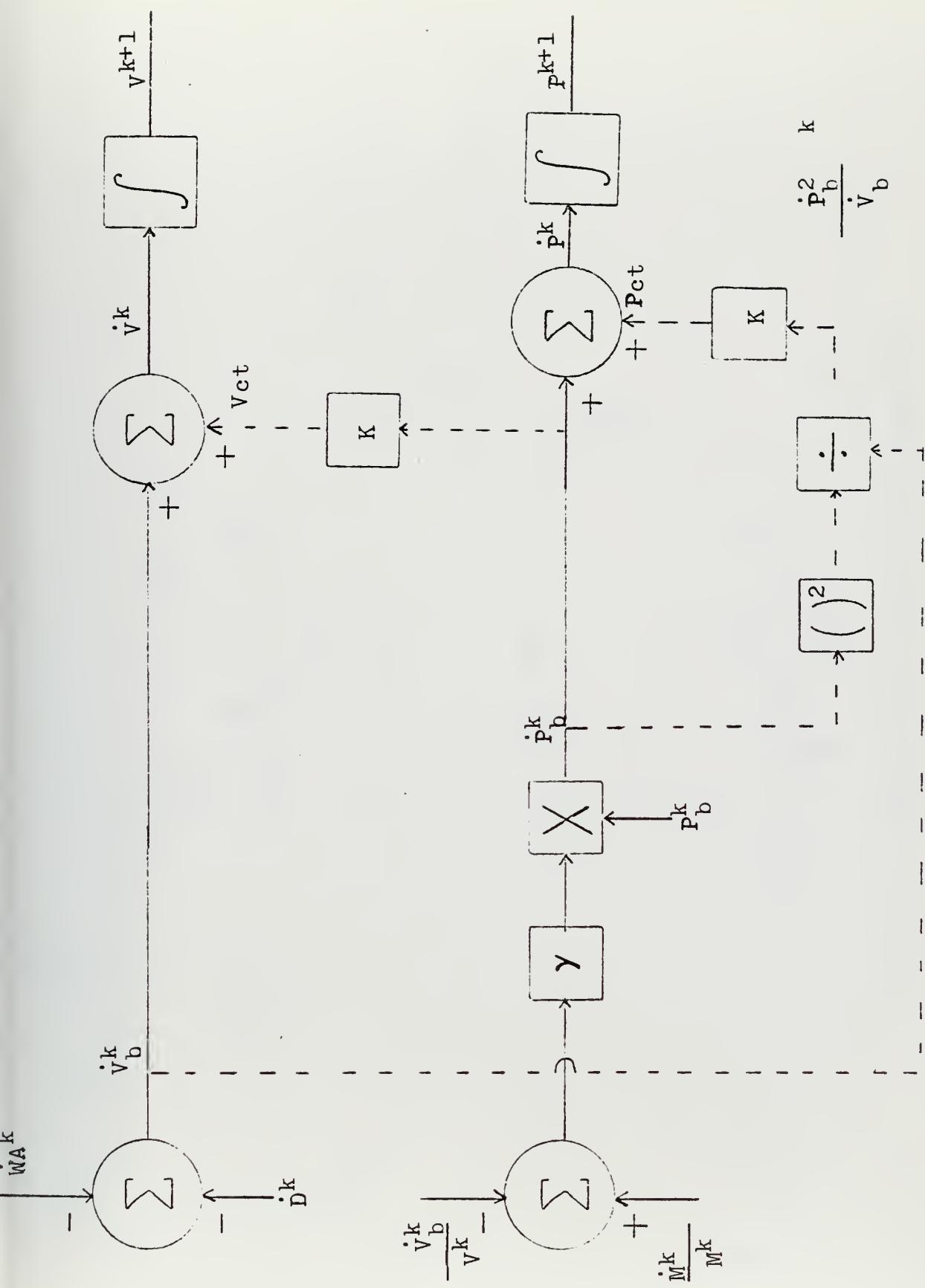


Figure 3. Computational Flow Diagram Model III

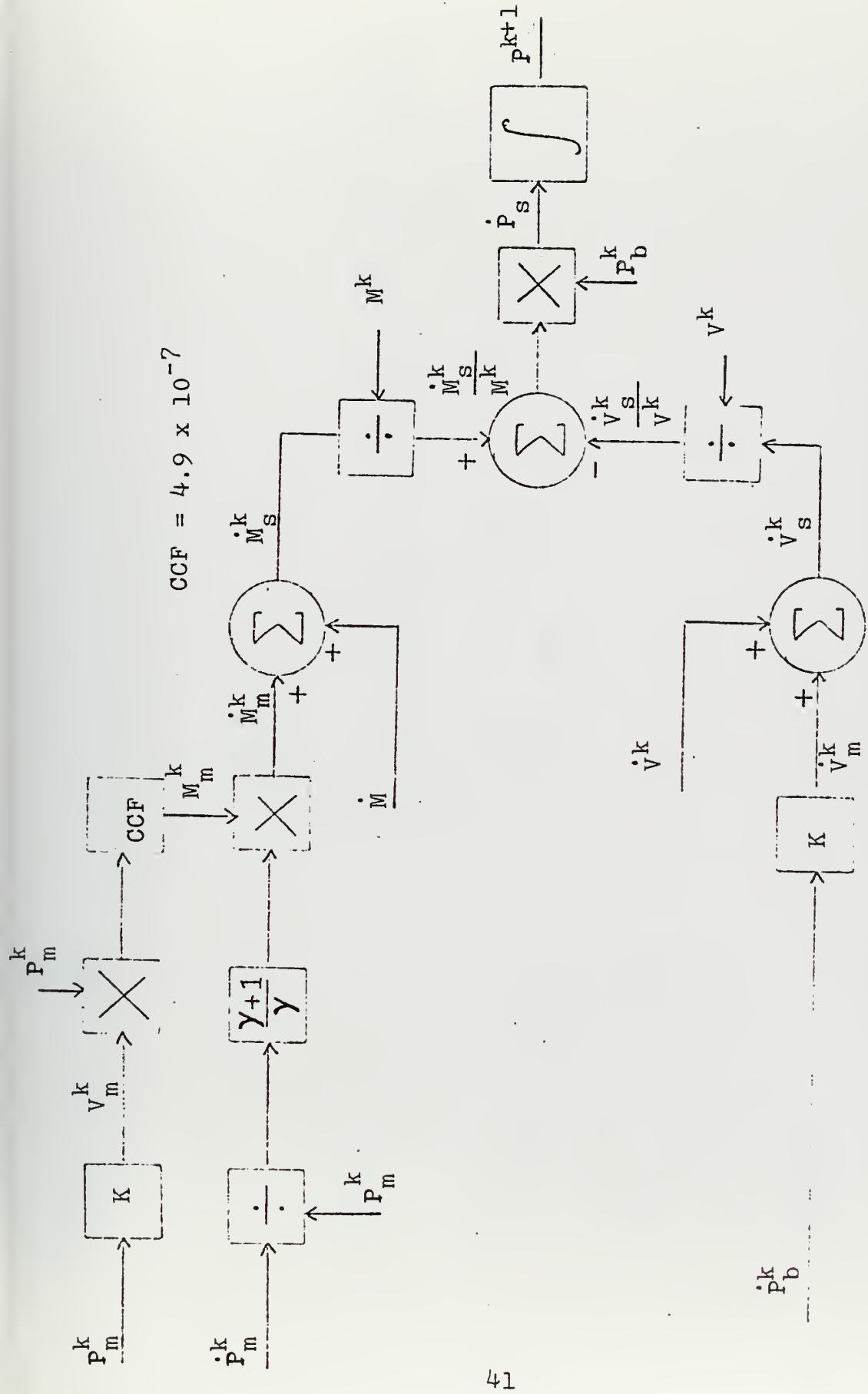


Figure 4. Computational Flow Diagram Model IV

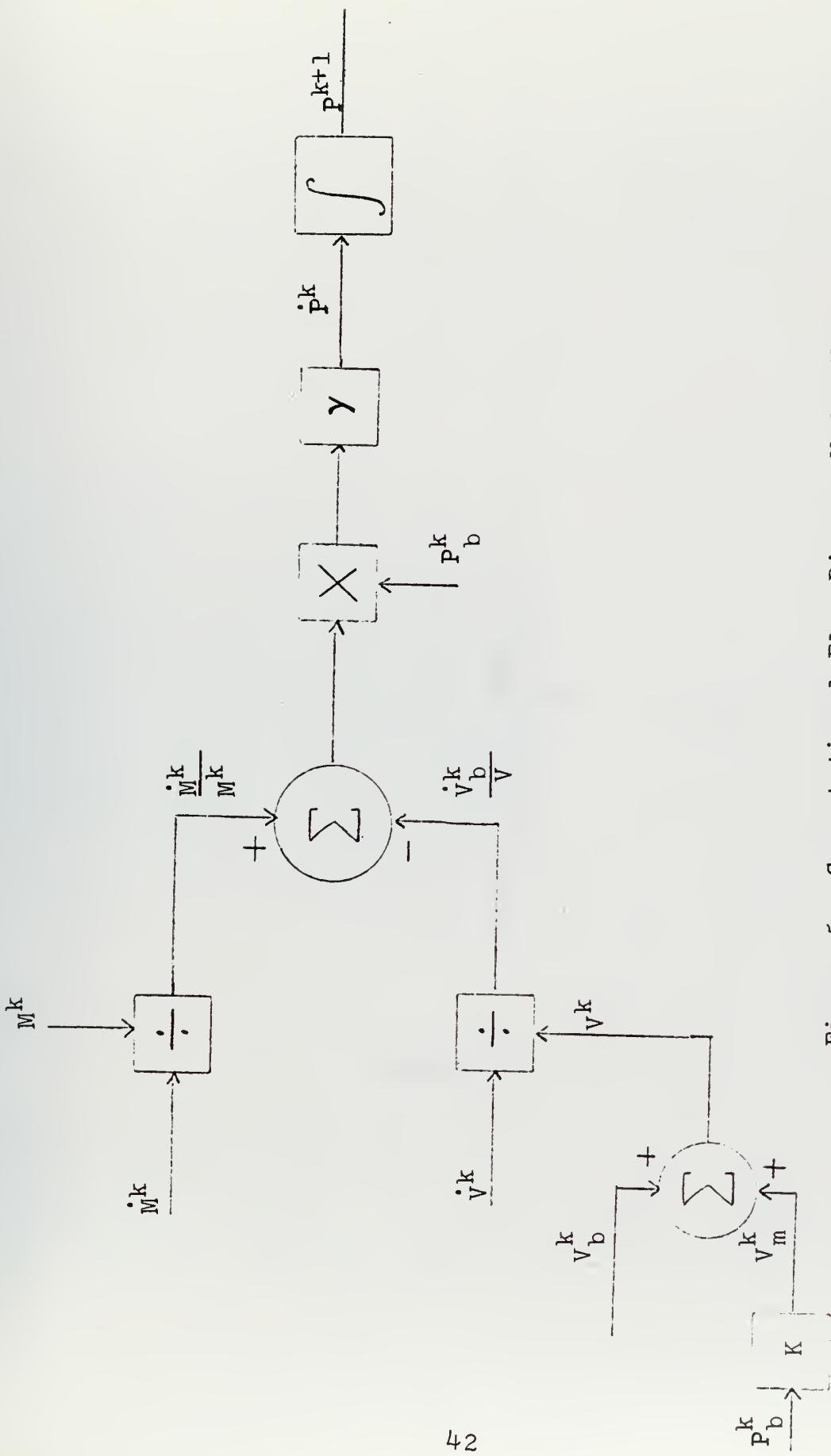


Figure 5. Computational Flow Diagram Model V

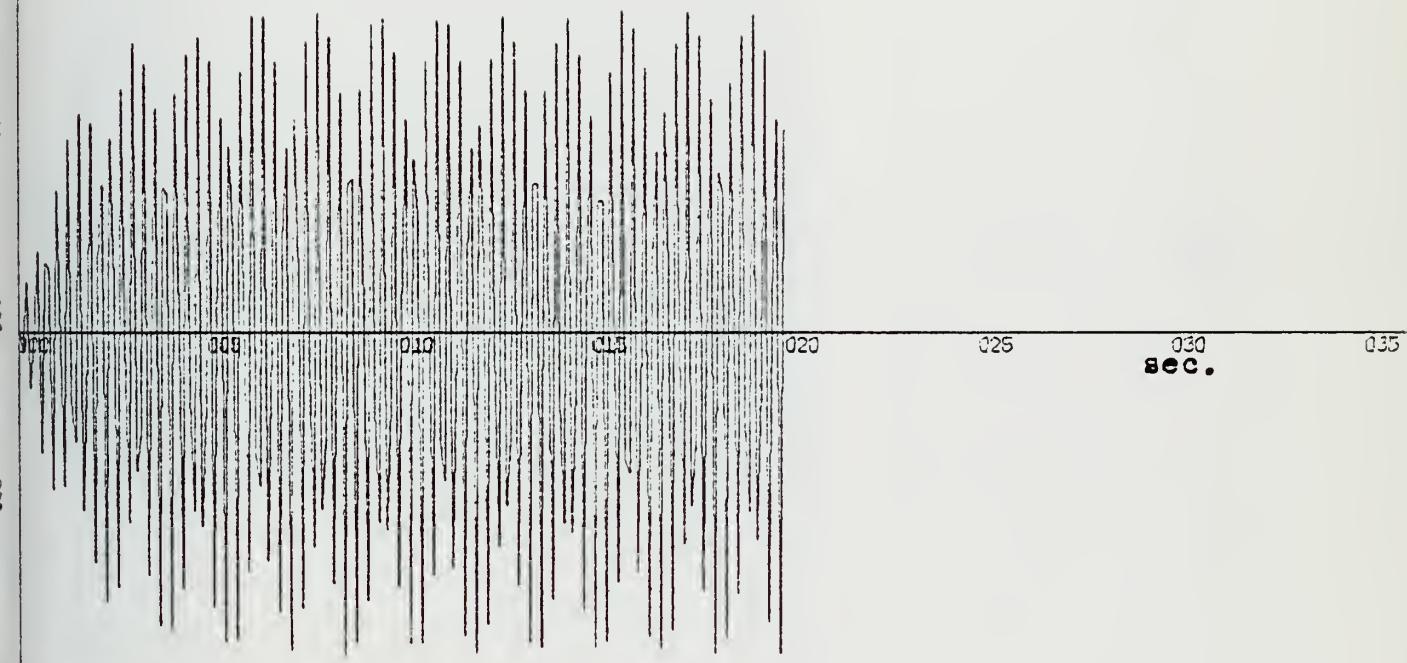


FIGURE 6

X-SCALE=5.00E+00 UNITS INCH.

Y-SCALE=5.00E-02 UNITS INCH.

PROGRAM 2 MODEL 5 20KTS REG. SEA 1

PLOT IS C.G. ACCELERATION VERSUS TIME

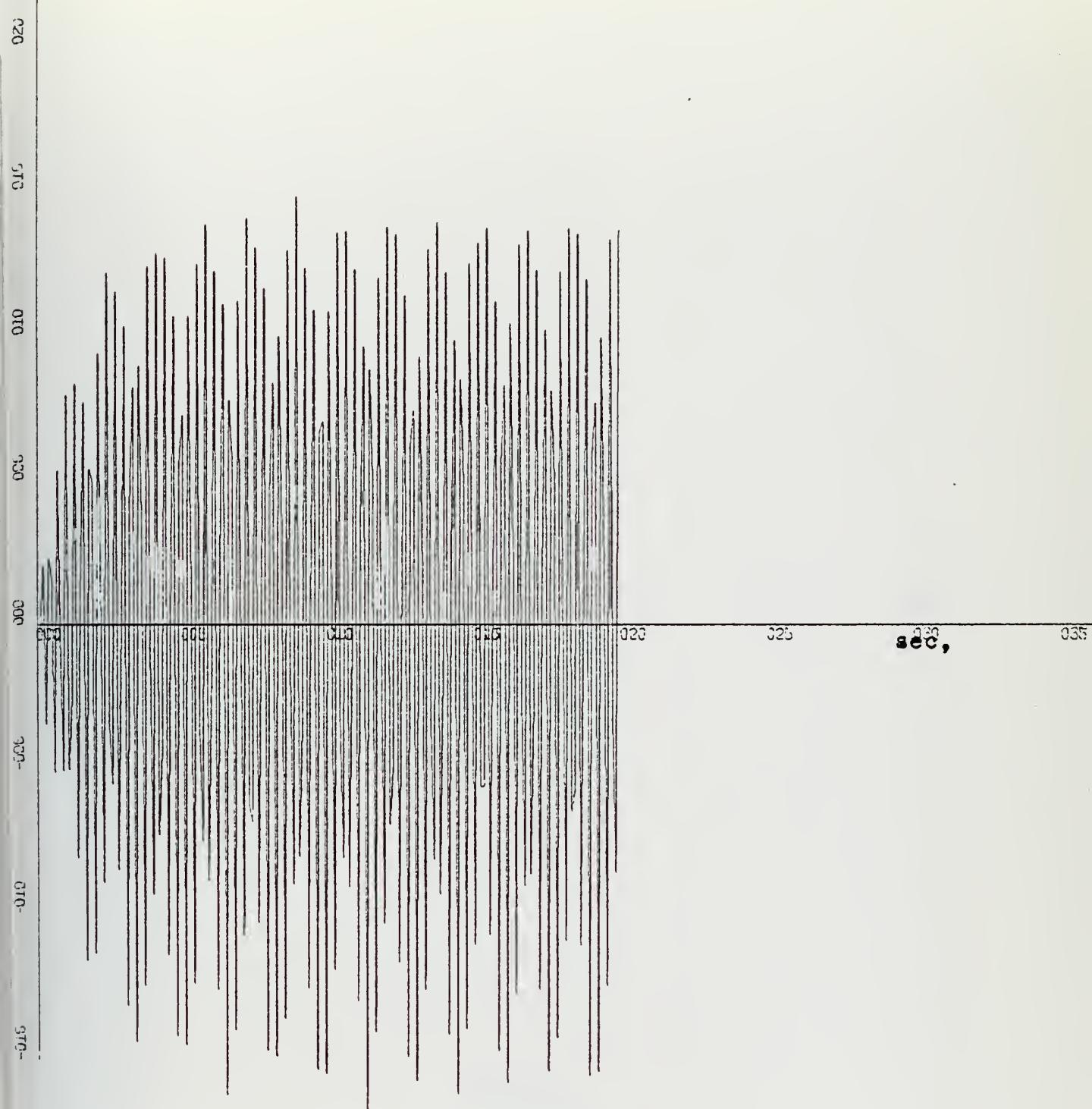


FIGURE 7

X-SCALE=5.00E+00 UNITS INCH,

Y-SCALE=5.00E-02 UNITS INCH,

PROGRAM 2 WITHOUT MEMBRANE REG. SEA 1
PLOT IS C.G. ACCELERATION VERSUS TIME

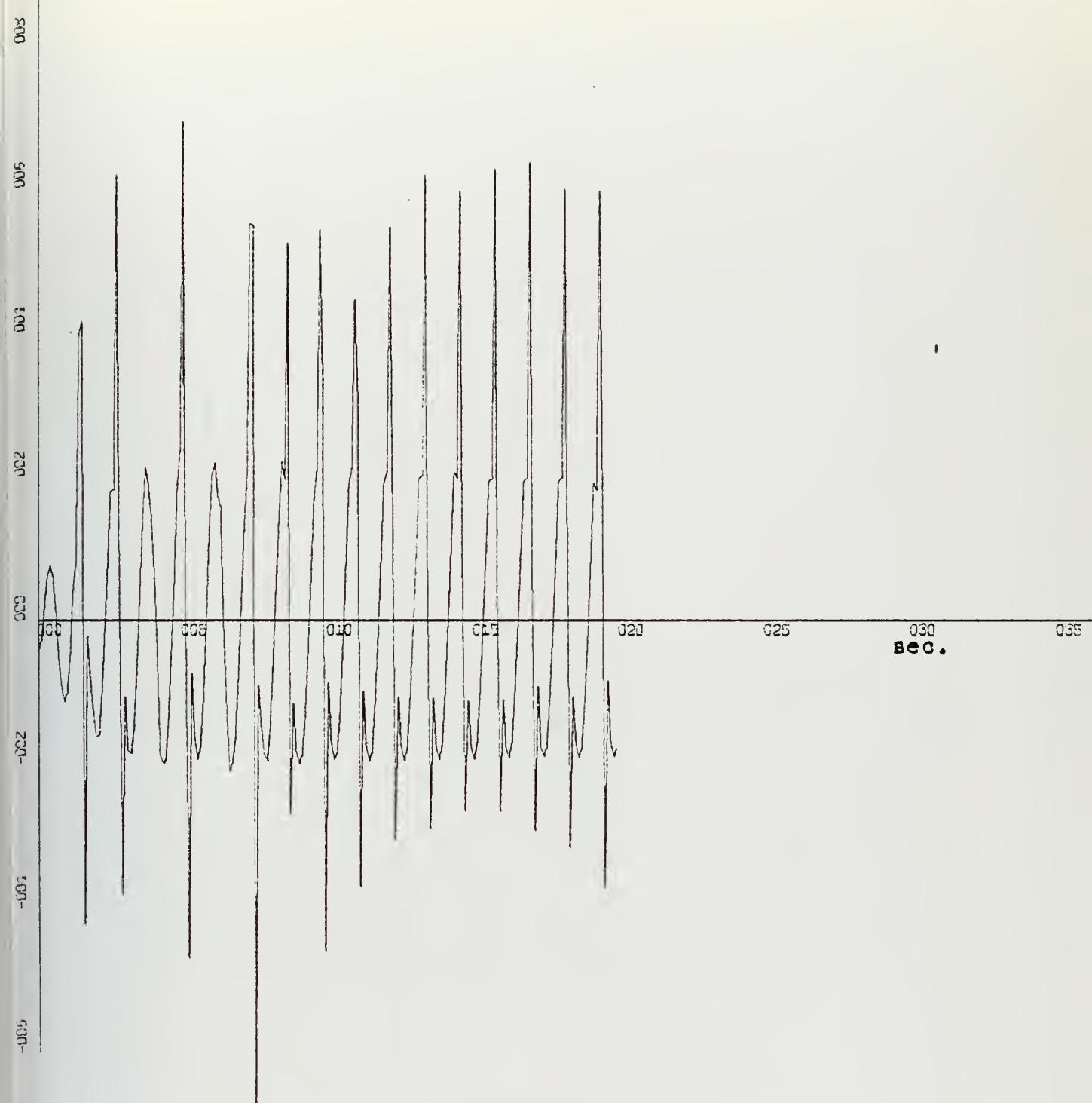


FIGURE 8

X-SCALE=5.00E+00 UNITS INCH,
Y-SCALE=2.00E-01 UNITS INCH.
PROGRAM 2 MODEL 5 20KTS REG. SEA 2
PLOT IS C.G. ACCELERATION VERSUS TIME

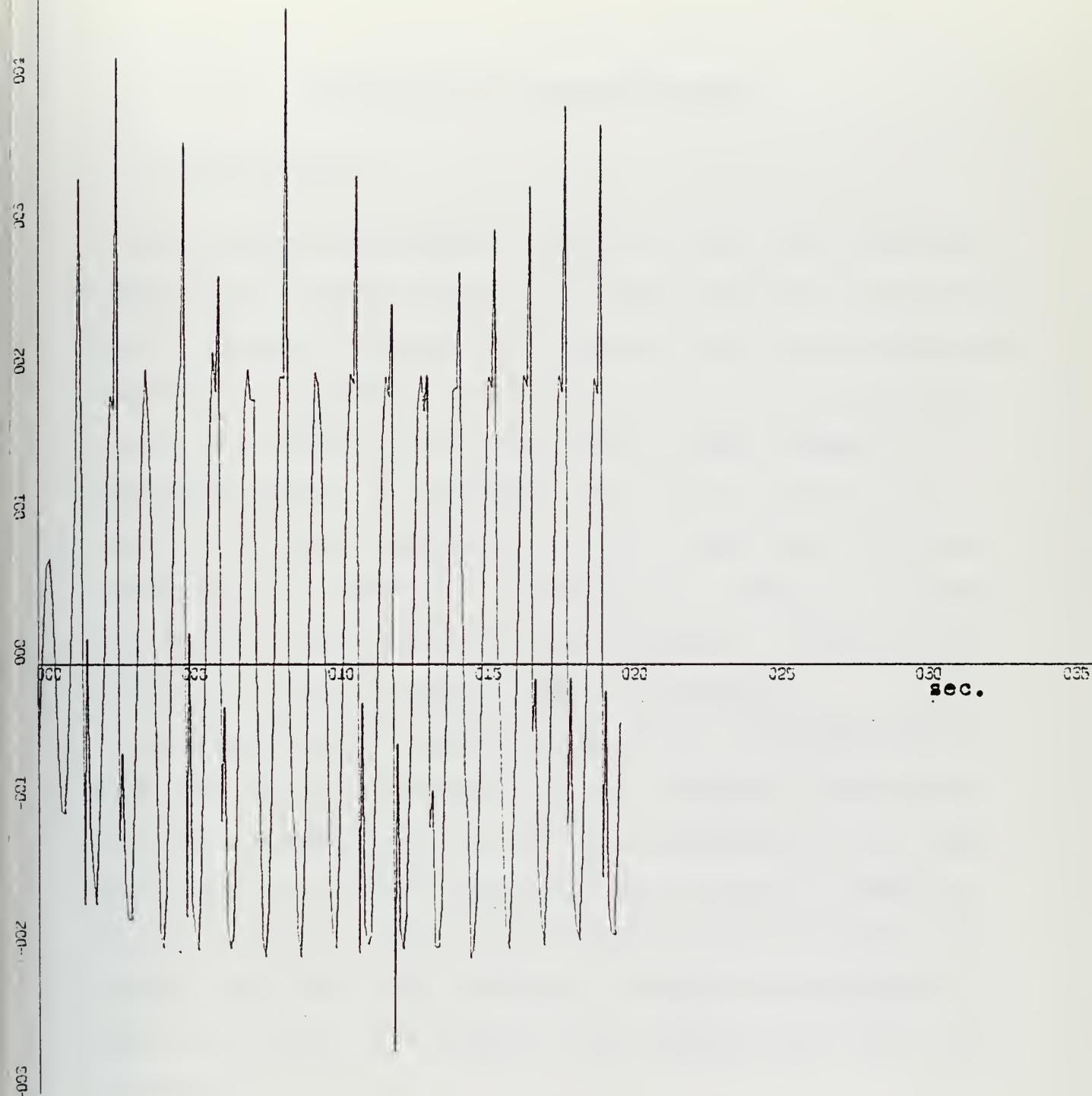


FIGURE 9

X-SCALE=5.00E+00 UNITS INCH.

Y-SCALE=1.00E-01 UNITS INCH.

PROGRAM 2 WITHOUT MEMBRANE REG. SEA 2
PLOT IS C.G. ACCELERATION VERSUS TIME

VI. CONCLUSIONS AND RECOMMENDATIONS

A. PROGRAM TIMING

The inclusion of pressure rate and volume rate equations improved the execution time of the L&M program without decreasing the program accuracy. The changes to the program were not difficult to implement. Further effort to improve execution time of the program should be directed toward changes to SUBROUTINE WAVES. In connection with this, a study of the number of required sidewall and bow and stern seal stations should also be conducted. By reducing the number of sidewall stations the execution time should decrease. A study of the losses in accuracy, however, would be necessary.

Another facet for investigation within SUBROUTINE WAVES would be a study to determine if the trigonometric functions could be represented by the in-line programming of the expansion of sine and cosine functions, vice the use of computer library functions. Since some sine and cosine functions are nested in DO LOOPS small savings in time would accumulate. Again, the losses in accuracy to this change would have to be determined.

The elimination of SUBROUTINE DMINV did not produce any noticeable time savings but did save approximately 2 K bytes of core.

The expansion of the number of wave components did not demonstrate any time savings; however the optimum number of wave components required to simulate irregular sea conditions should be determined. A possible course of action would be to generate wave components, calculate the magnitude and frequency and compute the energy spectral density. This energy density spectrum could then be compared to the Pierson and Neumann or Moskowitz model [Ref. 11]. By determining the optimum number of wave components, a savings in CPU execution time could result.

B. ADDITIONAL MODELING

The fact that waves can hit the top of the plenum was discussed in Section V. Whether this effect produces significant forces and moments is unknown. Since the time that the wave is in contact with the plenum and the location (station numbers) is known and since the area of the plenum in contact with the waves could be approximated, (no greater than the difference between wetted stations) additional correction terms could be developed to determine whether significant forces and moments are generated by wave contact. The drag of the wetted plenum top could also be considered.

C. MEMBRANE MODELING

Of the five membrane models developed in this study, Model Five is the most usable. The accuracy of this model must be determined and the model must be validated. Whether

mass and mass flow rate terms could improve this model also needs to be determined. In this regard a detailed study of adiabatic-reversible unsteady flow conditions would be useful. This would mean the establishment of a suitable control volume and the solution of energy-work equations.

The membrane modeling in this study was for a continuous non-permeable membrane. Additional modeling for the case of a hole cut into the membrane should be attempted. The equation for a sharp-edge orifice could be used to determine air loss. The area (size) of the hole can be determined by considering as a first approximation the ratio of surface area of a hemisphere to volume of the hemisphere. As an alternative, the experimental results of the towing tank tests conducted at NSRDC, on the XR-1, using suitable scale factors, could be used to determine empirical correction factors to be applied to the Loads and Motion program.

The occurrences of algebraic loops and summation of feed forward terms as shown in the computational flow diagrams and the failure of the models in which they occur indicate that further modeling of the membrane is required. To prevent model failures, elimination of the effects of the computational loops appears to be necessary.

APPENDIX A

MEMBRANE SCALE FACTOR

The empirical relationship between pressure and volume for the membrane given in Ref. 9 is for the XR-1 test craft.

$$V_m = K P_m \quad (A1)$$

where

$$K = \frac{40 \text{ in}^3}{\text{Psf}}$$

It was therefore necessary to scale the constant K for the XR-3 test craft.

In the following development the subscript 1 refers to the XR-1 and subscript 3 refers to XR-3.

Since XR-1 and XR-3 are similar it was assumed that all dimensions scale by the ratio of plenum lengths

$$\frac{L_3}{L_1} = \lambda \quad (A2)$$

where

$$L_3 = 20 \text{ ft.}$$

$$L_1 = 5.4 \text{ ft.}$$

It was assumed that the ratio of mass to volume was constant

$$\frac{M_1}{V_1} = \frac{M_3}{V_3} = \rho \quad (A3)$$

The ratio of craft volume is

$$\frac{V_3}{V_1} = \lambda^3 \quad (A4)$$

The ratio of craft pressure is

$$\frac{P_3}{P_1} = \frac{\frac{M_3 g / A_3}{M_1 g / A_1}}{} \quad (A5)$$

where g is the gravitational constant.

Substituting (A2) and (A3) into (A5)

$$\frac{P_3}{P_1} = \frac{\frac{V_3 A_1}{V_1 A_3}}{} = \frac{\lambda^3}{\lambda^2} = \lambda \quad (A6)$$

Using (A4) and (A6) in (A1)

$$K_1 = \frac{V_1}{P_1} \frac{V_3 / \lambda^3}{P_3 / \lambda} = \frac{K_3}{\lambda^2} \quad (A7)$$

$$K_3 = \lambda^2 K_1 \quad (A8)$$

Converting in. ³ to ft. ³ and using (A2)

$$K_3 = \left(\frac{20}{5.4} \right)^2 \frac{40}{1728} = .3175 \text{ ft.}^3/\text{psf} \quad (A9)$$

APPENDIX B

LISTING OF FORTRAN XR-3

L&M COMPUTER PROGRAM

PROGRAM TWO MODEL FIVE

```

INTEGER ON
COMMON /AIR/ PINF, RHOINF, GAM
COMMON /BMC0/ IMM, IMNX, IIMNY, IRMFIL, BTIME, INT, XMI(10), YMI(7), IX, IY
COMMON /CONST/ PI, RAD, UO
COMMON /ENGINE/ NPS, NPP, THSTS(25), THSTP(25), XP, YP, ZP, STHS, STHP, TIPMAIN
1(25) TIS(25)
1 COMMON /EQNCO/ NEQS, TOL(20), JQQ
1 COMMON /FPROP/ FXP, FYP, FZP, FKP, FNP
1 COMMON /FRDDE/ FNCRIT
1 COMMON /PRIME/ STIME, DELT, DELPNT, TPRINT
1 COMMON /PROMOD/ PROMO1, PROMO2, PROMO3, PROMO4, PROMOS, PROMO6, PROM07
1 COMMON /PRTINT/ ON, IACCEL, IVEL, ITraj, ISIDWL, IBQSL, ISTNSL, IWAVES, IMAINT
1 RUD, IPROP, IAERO, IRHS
1 COMMON /RDLR/ PHIMAX, TROLL
1 COMMON /RUDDR/ NPRO, DELPUD(25), XR, YR, ZR, IRDS, TL, RSPAN, RAREA, RASPR, RMAINT
1 CLB, RTC, RUDANG, TIR{25}
1 COMMON /VALOLD/ YOLD(20)
1 COMMON /WAVE/ ET A(4,11), AW(20), OMEGA(20), DVOLW, NWAVE, BETA, FXWAV, FYMAINT
1 COMMON /FWAV, FKWAV, FMWAV, FNWAV, ZBAR, PHIBAR, THERBAR, TC, COSBBET, SINBET, PBMAIN
1 2 BAR
1 EQUIVALENCE (VAL(2), VAL(1)), (VAL(3), VAL(8)), (VAL(4), VAL(9)), (VAL(5), VAL(10)), (VAL(6),
1 1), (VAL(7), VAL(12)), (VAL(13), VAL(21)), (VAL(14), VAL(22)), (VAL(15), VAL(23)), PSI), (VAL(24), PE), MAIN
1 3 (VAL(12), VOL)
3 DIMENSION DUMMY(20)
TC = 1.0
ON = 1
PI = 4.*ATAN(1.)
RAD = 180./PI
WRITE(6,12)
1 READ(44,13,END=2) DUMMY
1 WRITE(6,14) DUMMY
GO TO 1
2 REWIND 5
3 CALL INCON(TIME)
IF(INM.EQ.3) GO TO 11
VAL(13) = VAL(24)
C
4 DO 4 J=1,20
      YOLD(J) = VAL(J+1)
C
5 GO TO 8
      CONTINUE
      TOLD = TIME
      PBBAR = PBBAR*(1.-DELT/TC)+DELT*(PB-PINF)/TC

```

```

IF (NWAVE.LE.0) GO TO 6
ZBAR = (1.-DELT/TC)*ZBAR+DELT*Z/TC
PHIBAR = (1.-DELT/TC)*PHIBAR+DELT*PHI/TC
THEFAR = (1.-DELT/TC)*THEBAR+DELT*THETA/TC
CALL WAVES (TIME)
CALL SIDEWL
6 CALL PROP
CALL RUDDER
CALL AEROD
CALL INTGRL (TIME)
IF (TIME.GT.FNCRIT) GO TO 7
PRINT 17
GO TO 10
7 DELOLD = TIME-TOLD
PSI = PSI+DELOLD*R
X = X+DELOLD*(U*COS(PSI)-V*SIN(PSI))
Y = Y+DELOLD*(U*SIN(PSI)+V*COS(PSI))
IF (ABS(TIME-TPRINT).LT.1.E-6) GO TO 8
GO TO 5
8 CONTINUE
IF (ITRAJ.EQ.0) GO TO 9
DPHI = PHI*RAD
DPSI = PSI*RAD
DTHETA = THETA*RAD
DP = P*RAD
DQ = Q*RAD
DR = R*RAD
VEL = 0.5925*U
WRITE (6,15) TIME, VEL, V, W, DP, DQ, DR, Z, DPHI, DTHTA, X, Y, DPSI
BETS = (-V/U)*RAD
DELRS = RUDANG*RAD
WRITE (6,16) BETS, DELRS, FXP
9 CONTINUE
IMMTAG = (IMM+1)/2
IF (IMMTAG.FQ.1.*AND.*TIME.GE.*BTIME-1.E-8) IMT=1
TPRINT = TPRINT+DELNT
CN = 1
GO TO 5
10 CALL COLFIL
IF (IMM.LT.1) GO TO 3
IF (IMM.NE.1) GO TO 11
END FILE IBMFIL
GO TO 3
11 CALL SAM
GO TO 3
12 FORMAT (1H1//35X,22H LISTING OF INPUT DECK//)

```

```

13 FORMAT (20A4)
14 FORMAT (5X,20A4)
15 FORMAT (//10X,13H TIME (SEC) = F6.2 // 10X,33H TRANSLATIONAL VELS (KTS)
16 1)(FT/SEC)/10X,2H U=F6.2,5X,2HV=F6.2,5X,2HW=F6.3//10X,31H ROTATIONAL MAIN
17 2VELOCITIES (DEG/SEC)/10X,2H U=F6.2,5X,2H O=F6.2,5X,2HR=F6.2//10X,30MAIN
3H DISPLACEMENTS (FT AND DEGREES)/10X,2H Z=F7.3,5X,4H PHI=F6.2,3X,6H THM
4ETA=F6.2//10X,27H TRAJECTORY (FT AND DEGREES)/10X,2H X=F8.2,4X,2HY=F7.3,5X,4H
58.2,4X,4H PSI=F8.2
16 23HSIDESLIP ANGLE (DEG) = F8.2,10X,21H RUDDER ANGL E.
17 (DEG) = F8.2,10X,15H THRJST (LBS) = F12.1
17 FORMAT (//25X,28HCRAFT SPEED BELOW HUMP SPEED)
END

```

SUBROUTINE AERO

```
1 FORMAT (/10X,23HAERO,FX,FY,FZ,FK,FM,FN/6E15.4)
2 END
```

| | |
|--------------|--------|
| BLOCK DATA | |
| COMMON /AIR/ | Z1(3) |
| COMMON /BMC/ | Z2(25) |
| BL DA | 10 |
| BL DA | 20 |
| BL DA | 30 |

| | |
|-------------------|----------|
| COMMON //COLUMN/ | Z3(2) |
| COMMON //CONST/ | Z4(3) |
| COMMON //CNTRL/ | Z5(10) |
| COMMON //ENGINE/ | Z6(107) |
| COMMON //EQNCO/ | Z7(22) |
| COMMON //FAERO/ | Z8(6) |
| COMMON //FAIR/ | Z9(2) |
| COMMON //FANMAP/ | Z10(262) |
| COMMON //FORBS/ | Z11(7) |
| COMMON //FORSS/ | Z12(8) |
| COMMON //FROP/ | Z13(6) |
| COMMON //FRDDE/ | Z14(2) |
| COMMON //FRUD/ | Z15(6) |
| COMMON //GBOW/ | Z16(1) |
| COMMON //GEOMBS/ | Z17(138) |
| COMMON //GEOMSSW/ | Z18(62) |
| COMMON //GEOMSW/ | Z19(62) |
| COMMON //GEOMSWH/ | Z20(11) |
| COMMON //GEOMSWI/ | Z21(1) |
| COMMON //GEOMSWR/ | Z22(4) |
| COMMON //KSWTCH/ | Z23(817) |
| COMMON //LEAKER/ | Z24(36) |
| COMMON //MASSSES/ | Z25(55) |
| COMMON //MATRIX/ | Z26(14) |
| COMMON //MSIDW/ | Z27(12) |
| COMMON //MWAVEE/ | Z28(7) |
| COMMON //OPTION/ | Z29(5) |
| COMMON //PRIME/ | Z30(12) |
| COMMON //PRINT/ | Z31(2) |
| COMMON //PWAVE/ | Z32(1) |
| COMMON //RISER/ | Z33(2) |
| COMMON //ROLL/ | Z34(62) |
| COMMON //RUDDP/ | Z35(22) |
| COMMON //SICE/ | Z36(20) |
| COMMON //SOFTBS/ | Z37(19) |
| COMMON //SOFTSS/ | Z38(5) |
| COMMON //STABLE/ | Z39(2) |
| COMMON //STSRL/ | Z40(20) |
| COMMON //VALLD/ | Z41(40) |
| COMMON //VARBLE/ | Z42(100) |
| COMMON //WAVEF/ | Z43(80) |
| COMMON //SLOPE/ | Z44(5) |
| COMMON //PROBD/ | Z45(7) |
| DATA Z1/3*0.0/ | |
| DATA Z2/25*0.0/ | |
| DATA Z3/2*0.0/ | |
| DATA Z4/3*0.0/ | |
| DATA Z5/10*0.0/ | |

```

DATA Z6/107*0.0/
DATA Z7/21*0.0/
DATA Z8/6*0.0/
DATA Z9/2*0.0/
DATA Z10/2.62*0.0/
DATA Z11/7*0.0/
DATA Z12/8*0.0/
DATA Z13/6*0.0/
DATA Z14/2*0.0/
DATA Z15/6*0.0/
DATA Z16/0.0/
DATA Z17/1.38*0.0/
DATA Z18/62*0.0/
DATA Z19/62*0.0/
DATA Z20/1*0.0/
DATA Z21/0.0/
DATA Z22/4*0.0/
DATA Z23/817*0.0/
DATA Z24/36*0.0/
DATA Z25/55*0.0/
DATA Z26/12*0.0/
DATA Z27/4*0.0/
DATA Z28/7*0.0/
DATA Z29/5*0.0/
DATA Z30/12*0.0/
DATA Z31/2*0.0/
DATA Z32/0*0.0/
DATA Z33/2*0.0/
DATA Z34/62*0.0/
DATA Z35/22*0.0/
DATA Z36/20*0.0/
DATA Z37/19*0.0/
DATA Z38/5*0.0/
DATA Z39/2*0.0/
DATA Z40/20*0.0/
DATA Z41/40*0.0/
DATA Z42/100*0.0/
DATA Z43/80*0.0/
DATA Z44/5*0.0/
DATA Z45/7*0.0/
END

SUBROUTINE BWSL
  INTEGER CN
  COMMON /AIR/ PINF, RHINF, GAM
  COMMON /CONST/ PI, RAD, U0
  COMMON /FORBS/ FX, FY, FZ, FK, FM, FN, QL
  BWSL 10
  BWSL 20
  BWSL 30
  BWSL 40
  BWSL 50
  BWSL 60

```

```

COMMON /GEOM/ WIDTH,XL,XX(4,11),YY(4,11),NSTA(4),AB,VOLNOM,DELS(4,BWSL
110),XCP(2CP,BWSL),GEOMBS/ DETABX(11),DETABT(11),ARM1B(10),ARM2B(10),BWSL
110,TSKIB(10),BWSL
1,COMMON /LEAKER/ ALEAK,BLEAK,CFSS,CFBS,BWSL
1,COMMON /MASSES/ AM,AIXX,AIYY,AIZZ,AIXZ,AIMAX,G,WEIGHT,RHO,NMASS,AMBSL
11(201),XI(201),Y(201),Z(201),XS,ZS,HRH0,BWSL
1,COMMON /PRTINT/ ONIAACEL,IVEL,IISIDLW,IBOWSL,ISTNSL,IWAVES,I BWSL
1,IRUD,PROP1AEROD,TIRHS,BWSL
1,COMMON /PROMOD/ PROM01,PROM02,PROM03,PROM04,PROM05,PROM06,PROM07,BWSL
COMMON /SLOPE/ WATSLP,XPWV,XLX,PWV,PWHT,XPWV,XS,BWSL
COMMON /SOFTBS/ XBF,PRS,SINBS,COSBS,XBS,ZBS,DELYBS,DPBS,ELMAXB,YAVBWSL
1GB(10),CENCAAB,BWSL
1,COMMON /VARABLE/ VAL(40)
COMMON /WAVE/ ETA(4,11),AW(20),OMEGA(20),DVOLW,NWAVE,BETA,FXWAY,FYBWSL
1IWAV,FZIWAV,FMWAV,FNWAV,ZBAR,PHIBAR,THEBAR,TC,COSBRET,SINBET,PBRBWSL
2BAR,DIMENSION FLSKID(11),WETLEN(11),BWSL(6,24),GAP(11),ELSKI(11),
1DPFT(11),BWSL
1,DATA ENU,HINGHT/1.28E-5,1.875/
1,DATA BWSL,HINGHT/0.0,3.8E-6,9,9,12/3,15.0,0,4,1.7*3,10.5,13.2,15.7,0
1,4,3,7,7,11.0,13.7,16.7,0,0,4,7,8,2,11.5,14.5,17.4,0,0,5,2,1.8,9,12
1,24,15,4,18,3,0,0,5,5,9,6,13,3,16,1,19,6,0,0,5,8,9,13,8,17,1.2,0
30,0,6,1,10,6,14,7,18,4,2,1,9,0,3,0,1,6,4,1,1,0,1,5,3,19,3,0,0,1,6,6,1,1
4,7,16,4,20,5,24,5,0,3,0,2,1,7,6,1,2,1,2,5,7,0,7,2,1,3,0,5,1,9,6,2,4
5,0,1,28,9,3,0,0,8,2,7,3,2,1,7,2,6,3,3,1,0,3,0,0,9,6,1,7,5,2,4,1,2,9,7,34,6,0
6,0,1,11,9,2,0,9,7,2,6,0,3,0,14,0,3,0,0,7,4,2,0,2,2,5,3,0,0,7,35,0,8,4,1,0,0
6,7,0,5,5,3,2,5,3,9,7,4,4,5,0,0,3,0,0,7,4,8,7,0,0,3,0,0,4,7,0,0,4,6,1,7,5,2,3,0
8,8,4,5,0,4,8,7,0,0,2,9,4,3,8,5,4,5,0,0,4,8,7,0,0,3,0,0,4,5,0,3,0,0,4,8,7,4,8,7,35,0
9,7,0,0,4,0,0,4,8,7,4,8,7,4,8,7,4,8,7,0,0,4,8,7,4,8,7,4,8,7,0,0,4,8,7,4,8,7,36,0
$8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,4,8,7,37,0
DATA CORLEN/3.75/
1,EQUIVALENCE (VAL(1),TIME),(VAL(2),U),(VAL(3),V),(VAL(4),W),
1L(5),P,(VAL(6),Q),(VAL(7),R),(VAL(8),PHI),(VAL(9),THETA),
2L(10),Z,(VAL(11),BMASS),(VAL(21),X),(VAL(22),Y),(VAL(23),PSI),
2L,(VAL(24),PBI)
CONTINUUF
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FK = 0.0
FN = 0.0
DELPBG = PBS-PB
IF (DELPBG.LT.0.0) DELPBG=0.0
PBAR = PB-PINF
DELP = PBAR
IF (DELP.LT.0.0) DELP=0.0
ARGO = ARSIN(ARGO)
ANGO = XBS*THETA-CORLEN*COS(ANGO)
X1 = XBS+ZBS*THETA-ELMAXB*COS(THFTA)
Z1 = -2-ZBS+XBS*THETA-ELMAXB*COS(THFTA)
DPTHFT = (5.5/(1+(U/25.))**2)*0.0833
IF (CENCA.BGT.1.1875) CENCA=1.1875
N = NSTA(3)

DO 3 K=1,N
  DPFT(K) = DPTHFT
  ELSKI(K) = (ETA(3,K)-DETABX(K)*(XX(3,K)-X1)-Z1)+YY(3,K)*PHI+XLXPWV
  1*WATSLP
  IF ((HINGHT-ELSKI(K).GT.HINGHT) ELSKI(K)=HINGHT
  1 IF ((HINGHT-ELSKI(K)+DPFT(K)).GE.ELMAXB) DPFT(K)=ELMAXB-HINGHT+ELS
  1 KI(K)
  IF (DPFT(K).LT.0.0) DPFT(K)=0.0
  ELSKID(K)=(ELSKI(K)-DPFT(K))*12.0
  IF ((HINGHT-ELSKID(K)/12.0).GE.ELMAXB) ELSKID(K)=(HINGHT-ELMAXB)*1
  12.0
  DPIN = DPFT(K)*12.0
  MM = DPIN
  MM1 = MM+1
  MM2 = MM1+1
  DINC = DPIN-MM
  GAP(K) = -ELSKI(K)+(HINGHT-ELMAXB)
  IF (GAP(K).LT.0.0) GAP(K)=0.0
  IF ((ELSKID(K).GE.0.0) GO TO 2
  1 WETLEN(K) = ELSKI(K)
  GO TO 3
  MN3 = ELSKID(K)
  MM4 = MM3+1
  MM5 = MM4+1
  DLINC = ELSKID(K)-MM3
  BWSL1 = BWSL(MM1,MM4)
  BWSL2 = BWSL(MM1,MM5)
  BWSL3 = BWSL(MM2,MM4)
  BWSL4 = BWSL(MM2,MM5)
  BWSLA1 = (BWSL2-BWSL1)*DLINC+BWSL1
  BWSLA2 = (BWSL4-BWSL3)*DLINC+BWSL3
  WETLEN(K) = ((BWSLA2-BWSLA1)*DINC+BWSLA1)*0.08333
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3 CONTINUE

C C

N = NSTA(3)-1

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DO 10 J=1 N
WETLAV = WETLEN(J+1)+WETLEN(J)*0.5
IF(WETLAV .LE. 0.01) GO TO 8
DPFTAV = (DPFT(J+1)+DPFT(J))*0.5
ELSKIA = (ELSKID(J+1)+ELSKID(J))*0.5
SEALHT = HINGHT-ELSKDA
DIFF = 2.0*CENTAB-(SEALHT+0.5)
IF(DIFF .GT. 0.5) DIFF=0.5
ARM1B(J) = X1+WETLAV*0.5
ARM2B(J) = ZS-ELSKIA+DPFTAV*0.5
IF(DIFF .LT. 0.5) GO TO 4
DFBS(J) = -DELPH*WETLAV
GO TO 7
```

4 FORLEN = XBF-WETLAV
IF(FORLEN .EQ. 0.0) GO TO 5
ARGW = (HINGHT-ELSKIA)/FORLEN
IF(ARGW .GT. 1.0) ARGW=1.0
ANGW = ARSIN(ARGW)
FURCOS = COS(ANGW)
GU TO 6

5 FORCOS = 0.0
6 DFB(S(J)) = -DELPH*DELYBS*WETLAV-DELPH*FORLEN*DELYBS*(FORLEN*FORCOS*(DIFF-0.25)*4.0)

7 RESKI = U*WETLAV/ENU
CDTSKIB(J) = 0.427*(ALOG10(RESKI)-0.407)**2.64
CDTSKIB(J) = -ARG*CDTSKIB(J)

8 DFBS(J) = 0.0
9 CONTINUE

FX = FX+TSKIB(J)
FZ = FZ+DFBS(J)
FK = FK+DFBS(J)*YAVGB(J)
FM = FM-DFBS(J)*ARMB(J)+TSKIB(J)*ARM2B(J)
FN = FN-TSKIB(J)*YAVGB(J)
ALBS = ALBS+(GAP(J)+GAP(J+1))*DELYBS*0.5

10 CONTINUE

C

ALBS = ALBS+BLEAK
SQFAC = SQRT(2*ABS(PBAR)/RHOINF)
QL = CFB*ALBS*SQFAC*SIGN(1.,PBAR)
IF(I3CWSL.NE.ON) RETURN

WRITE (6,11) GAP,WETLEN,FX,FY,FZ,FK,FM,FN

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IF ( JQQ .NE. 2 ) GO TO 1
1 WRITE(6,23) STEP2
ENDFILE 1
REWIND 1
TITLE(7) = LINE2(1)
TITLE(10) = LINE2(2)
IF (NGRAF.EQ.0) GO TO 10
J = 1
NGF = NGRAF
INDEX = NGRAF*2
C
DO 6 I=1,INDEX,2
  INDX = NXY(I)
  INDY = NXY(I+1)
  IGE = 0
  READ (1,END=8) TIME,ETA,Z,THETA,PB,BOWACC,ACC,FANPWR,PHI,BETAS,
  1ICCLAT,U,TRADUS,VOLP,X,Y,QIN,QOUT,GXXX,XPWAV,THSTS(1),THSTP(1),
  2FG,PDEG,TRDEG,DELRS
  IF (IQ.GE.900) GO TO 3
  IQ = IQ+1
  XOUT(IQ) = PVQQ(INDX)
  YOUT(IQ) = PVQQ(INDY)
  GO TO 2
3 REWIND 1
  INX = INDEX*2
 INY = INDY*2
  NAMEX(1) = NAMES(INX-1)
  NAMEX(2) = NAMES(INX)
  NAMEY(1) = NAMES(INY-1)
  NAMEY(2) = NAMES(INY)
  IF (NDRW.EQ.1) GO TO 4
  CALL PLOTP(XOUT,YOUT,-IQ,0)
  C
  WRITE(6,24) NAMEX,NAMEY
  GO TO 6
4 TITLE(8) = NAMEY(1)
  TITLE(9) = NAMEY(2)
  TITLE(11) = NAMEX(1)
  TITLE(12) = NAMEX(2)
  NCUR = NCURV(J)
  IF (NCUR.NE.0) GO TO 5
  LABEL = LAB(1)
  C
  CALL DRAW(IQ,XCUT,YCUT,NCUR,0,LABEL,TITLE,0,0,0,0,0,8,8,0,LAST),
  C
  J = J+1
  GO TO 1
C
C

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5 WRITE (2) IQ,NCUR,NGF,LABEL,(TITLE(K),K=1,12),(XOUT(L),L=1CFL
1 I(G)
1 NGF = NGF-1
6 CONTINUE
C   IF ((NGF.EQ.0).AND.(INCUR.EQ.3)) GO TO 16
7 IF (IVERT.NE.1) GO TO 16
    WRITE (6,25)
    K = 0
C   DO 8 I=1,8
    IF (ISUM1(I).NE.0) K=K+1
8 CONTINUE
C   NUM1 = K
    IF (K.EQ.0) GO TO 16
    N = K*2
    J = 1
C   DO 9 I=1,NUM1
    INDEX = I*SUM1(I)*2
    INAME(J) = NAMES(INDEX-1)
    INAME(J+1) = NAMES(INDEX)
    J = J+2
9 CONTINUE
C   WRITE (6,26) (INAME(I),I=1,N)
    GO TO 14
10 END FILE 2
    REWIND 2
    NGF = NGRAF
    J = 1
11 READ (2,END=38) IQ,NCUR,NGRAF,LABEL,(TITLE(K),K=1,12),(XOUT(L),Y
    ICUT(L)=1,IQ)
    IF ((NGF.EQ.NGRAF).AND.(INCUR.EQ.J)) GO TO 12
    GO TO 11
12 LABEL = LAB(J+1)
C   CALL DRAW (IQ,XOUT,YOUT,NCUR,TITLE,0,0,0,0,0,0,8,8,0, LAST)
C   REWIND 2
    J = J+1
    IF (J.EQ.4) GO TO 13
    GO TO 11
13 NGF = NGF-1
    IF (NGF.EQ.0) GO TO 7
    J = 1
    GO TO 11

```

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14 READ (1,END = 13) (PVQQ(I),I=1,26)
C
C DO 15 I=1,NUM1
C J = ISUM1(I)
15 AFIL(E(I)) = PVQQ(J)
C
C WRITE (6,27) (AFILE(I),I=1,NUM1)
C GO TO 14
C REWIND 1
C IF (ILATRL*NE.1) GO TO 22
C WRITE (6,28)
C K = 0
C
C DO 17 I=1,8
C IF (ISUM2(I)*NE.0) K=K+1
17 CONTINUE
C
C NUM2 = K
C IF (K*EE.0) GO TO 21
C N = K*2
C J = 1
C
C DO 18 I=1,NUM2
C INDEX = ISUM2(I)*2
C INAME(J) = NAMES(INDEX-1)
C INAME(J+1) = NAMES(INDEX)
C J = J+2
18 CONTINUE
C
C WRITE (6,26) (INAME(I),I=1,N)
C 19 PREAD (1,END = 16) (PVQQ(I),I=1,26)
C
C DO 20 I=1,NUM2
C J = ISUM2(I)
20 AFIL(E(I)) = PVQQ(J)
C
C WRITE (6,27) (AFILE(I),I=1,NUM2)
C GO TO 19
C REWIND 1
C RETURN
C
C 23 FORMAT ('0!4X,' THIS RUN USED VARIABLE STEP SIZE ',0',4X,' THE M
C 1NIMUM STEPSIZE RECORDED DURING THE RUN WAS ',2XE30.5)
C 24 1FORMAT ('0!20X,2A8,0 IS THE INDEPENDENT VARIABLE')
C 1DEPENDENT VARIABLE')
C 25 1FORMAT ('0!50X,0**SUMMARY ONE***,/,00')
C 26 1FORMAT ('0!16A8')
C 27 1FORMAT ('0!,8(3X,F10.2,3X)')

```

28 FORMAT ('0',50X,'***SUMMARY TWO***,,/,00')

CFL 1830
CFL 1840

SUBROUTINE FAN

```

C
      INTEGER ON
      COMMON /AIR/ PINF,RHOINF,GAM
      COMMON /FANMAP/ QIN,QBFFAN(25),QMFAN(25),ENBFAN,ENFAN
      COMMON /BRPM/ SRPM,NPTSB,NPTSM,NP1SM,NPTSB,NPTSM
      1 SFAN,BRPM,EMRPM,DELB(25),NBTMES(25),DELS(25),NS
      25),TMEB(25),DELB(25),NBTMES(25),DELS(25),NS
      COMMON /PROMOD/ PROMO1,PROMO2,PRDM03,PRDM04,PROMO5,PROMO6,PROMO7
      COMMON /PRTINT/ ON,IACCEL,IVEL,IWAVES,IFAN
      IRUD,IPROP,IAEROD,TIRHS
      COMMON /SOFTBS/ XBF,PBS,SINRS,COSBS,XBS,ZBS,DELYSS,DPBS,ELMAXB,YAVFAN
      16B(10)
      COMMON /SOFTSS/ XLF,PSS,SINTH,COSTH,XSS,ZSS,DELYSS,DPSS,ELMAXS,YAVFAN
      16S(10)
      COMMON /VARIABLE/ VAL(40)
      DIMENSION QB(1),QM(1),QS(1),PBOW(1),PM(1),PS(1),HP(8)
      EQUIVALENCE (VAL(1),T1ME),(VAL(2),Y),(VAL(3),V),(VAL(4),W),
      1 L(5),P),(VAL(6),Q),(VAL(7),R),(VAL(8),PHI),(VAL(9),THETA),
      2 L(10),Z),(VAL(11),BMASS),(VAL(21),X),(VAL(22),Y),(VAL(23),PSI)
      3 ,(VAL(24),PB)
      EQUIVALENCE (VAL(18),FANPWR)
      EQUIVALENCE (QBFFAN(1),QBF(1)),(QMFAN(1),QMF(1)),
      1 (PBFAN(1),PBOW(1)),(PMFAN(1),FM(1)),(QSFAN(1),QS(1)),
      DATA HP/2.9,2.3,1.95,1.77,1.85,2.05,1.93,1.62/
      C
      BRA = 8000/BRPM
      EMRAT = 8000/EMRPM
      TL = VAL(1)
      IF (NB.EQ.0.0) GO TO 1
      DPBS = FGI(TL,NB,TMEB,ILB)
      PBS = PB+DPBS
      1 IF (NS.EQ.0.0) GO TO 2
      DPSS = FGI(TL,NS,TMES,DELS,ILS)
      PSS = PB+DPSS
      2 CONTINUE
      PB1 = PB-S-PINF
      PB2 = PB-P-INF
      PB3 = PSS-P-INF
      PBAR = PB1*BRA*T*EMRAT
      PBARM = PB2*EMRAT*SRAT
      PBARS = PB3*SRAT*SRAT
      QBOW = ENBFAN*FG1(PBARB,NPTSB,PBOW,QB,IB)/BRAT
      QMAIN = ENMFAN*FG1(PBARM,NPTSM,PM,QM,IS)/EMRAT
      QSTN = ENSFAN*FG1(PBARS,NPTSS,PS,QS,IS)/SRAT
      C

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SUBROUTINE INCON (TIME)

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REAL * 8 TICRD
INTEGER ON
COMMON /AIR/ PINF, RHOINF, GAM
COMMON /AXIS/ NXY(26)
COMMON /ATRIX/ AMASSI, AIXYI, DIXX, DIXZ, DIZ, IMT, XMI(10), YMI(7), IX, IY
COMMON //BMCO/ IMM, IMN, IMNY, IBMFIL, BTIME, IMT, XMI(10), YMI(7), IX, IY
COMMON //COLUMN/ IVERT, ILATRL
COMMON //CONST/ PIRADUO
COMMON //CNTRL/ CONTW, CONTQ, CONTH, QMULT, LOUVER, ACONTZ, ACONTW, ZEQUI
COMMON //THEQL/ ACBASE
COMMON //ENGINE/ NPS, NPP, THSTS(25), THSTP(25), XP, YP, ZP, STHS, STHP, TIP
COMMON //EQNCO/ NEQS, TOL(20), JQQ
COMMON //FAIR/ RHOA, XLAERO
COMMON //FANMAP/ QIN, QBFAN(25), QMFAN(25), ENBfan, ENMfan, EN
COMMON //BRPM/ SRPM, NPTSBNP, NPTSM, NPTSS, PBfan(25), PSfan(25)
COMMON //BPM/ SRPM, NPTSM, NPTSS, PBfan(25), PSfan(25)
COMMON //TMEB(25), DELB(25), NB, TMES(25), DETS(25), NS
COMMON //FROUDE/ FN, FNCRIT
COMMON //GBOW/ XBOW
COMMON //GEOM/ WIDTH, XL, XX(4,11), YY(4,11), NSTA(4), AB, VOLNOM, DELS(4,
10), XC, ZCP, GEOMSW, XAVG(10), DS
COMMON //GRAF/ NGRAF, ND RW
COMMON //FEADG/ TICRD(6)
COMMON //PWAVE/ FNCON, PWVCON
COMMON //LEAKER/ ALFAK, BLEAK, CFSS, CFBS
COMMON //MASSFS/ AM, AIXX, AIYY, AIZZ, AIIMAX, G, WEIGHT, RHO, NMAS S, AM
11(201), X(201), Y(201), Z(201), XS, ZS, HRHO
COMMON //MATRIX/ A(6,6)
COMMON //OPTION/ I3DOF, ISRG, ITRIM, IDIA
COMMON //PLENUM/ XBBW, ABW, BUBHGT, DVWDOT, VMDOT, CMM
COMMON //PLVCQQ/ NYD, NLD, NLD
COMMON //PRIME/ STIME, FTIME, DELT, DELPNT, PRINT
COMMON //PRTINT/ ON, IACCEL, IVEL, ITraj, ISIDLW, IBOWSL, IWAVES, I
1 RUD, IPOP, IAERO, IRHS
COMMON //PROMOD/ PROMO1, PROMO2, PROMO3, PROMO4, PROMO5, PROMO6, PROM07
COMMON //ROLL/ PHIMAX, TROLL
COMMON //RUDDR/ NPK, DELRUD(25), XR, YR, ZR, IRDS, TL, RSPAN, RAREA, RASPR, R
1 CLB, RT, C, RUDDANG, TIR(25)
COMMON //RISER/ AMPTC
COMMON //SOFTBS/ XBF, PBS, SINBS, COSBS, XBS, DELYBS, DPBS, ELMAXB, YAV
1 GB(10), CENCA B
COMMON //SOFTSS/ XLF, PSS, SINTH, COSTH, XSS, ZSS, DELYSS, DPSS, ELMAXS, YAV
1 GS(10)

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COMMON /SIDE/ FYSW,FZSW,FMSW,FNSW,ALSW,YSW,XLSW,CFSW,CDSINCN
1W,VAREA,VCHORD,VSPAN,VANGLE,VGOS,VX,VY,VZ,AVBMSW,DELX,VTC
COMMON /SLOPE/ WATSLP,XPPWV,XLXPWV,PWVHT,XPWVXS
COMMON /STABLE/ S(4)ISTAB
COMMON /STSRL/ CPHI,CPHID
COMMON /SUM/ ISUM1(8)ISUM2(8)
COMMON /VALOLD/ YOLD(20)
COMMON /VARBLE/ VAL(40)
COMMON /WAVE/ ET A(4,11)AW(20),OMEGA(20),DVOLW,NWAVE,BETA,FXWAY,FY
COMMON /WAVEN/ WAVELEN(WAVELN(20),OMEGA(20),WAVSLP(20),ENC PFR(20),
COMMON /WAVTAB/ NAL,SDS,NDS,SDS,NTH,DTH,NBB,DBB,SB,B,AC
1 1(20,5,7),AC2(20,5,7),AC3(20,5,7),AC4(20,5,7),AC5(20,5,7),AC6(20,5,7),AC
2 2(20,5,7),AC7(20,5,7),AC0(20,5,7),AC8(20,5,7),AS1(20,5,7),AS2(20,5,7),AS
3 S2(20,5,7),AS3(20,5,7),AS4(20,5,7),AS5(20,5,7),AS6(20,5,7),AS7(20,5,7),AS
4 5(20,5,7),AS0(20,5,7),AS8(20,5,7),BB(36),XREF,RX
DIMENSION ZZ(14,50)
EQUIVALENCE (ZZ,NAL)
EQUIVALENCE (VAL(21),U),(VAL(3),V),(VAL(4),W),(VAL(5),P),
1 1,Q),(VAL(7),R),(VAL(8),P),(VAL(9),THETA),(VAL(10),PSI),
2 2,1),BMASS),(VAL(21),X),(VAL(22),Y),(VAL(23),PSI),(VAL(24),PR),
3 (VAL(12),VCL)
DIMENSION TEMP(7),XMO(10)
DIMENSION TITLC(20)
DATA BEAM,BETAD,DELO,DELPPI,DLRDO,DSO,ISYS,RMAX0,RONO,RRATO,RREVO,T
1 HETO,THSSI,TPRINO,U,VZO,XBSI,XCP0,XLTOT,XPO,XSSI,YPO,ZPO,
2 ZR0,ZSSI/6*0.0,0,2*0.0/
INITIAL CONDITIONS WITH WATSLP

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DO 1 I=1,8
1 ISUM1(I)=0
1 ISUM2(I)=0
C PINF = 2116.
C RHOINF = .002378
C GAM = 1.4
C GO TO 3
2 READ (5,98) NGRAF,NDRW
2 READ (5,99) NXYS
2 READ (5,100) TICRD
3 READ (5,107) TISYSLOPT(TEMP(I))=107
3 IF (1SYSLEQ.1SYS.AND.1SYS.EQ.1SYS) 97
1 SYS = 1SYS
1 IF ((1SYSLE 0) OR ((ISYS.GT.22)) GO TO 97
GO TO (4,14,23,24,27,28,31,34,37,41,42,52,53,74,75,82,89,92,93,6,7

```

1,2), I SYS

C PROGRAM CONTROL PARAMETERS

4 CONTINUE
5 GO TO (5,8,9,10,11,12), IOPT

5 STIME = TEMP(1)
STIME = TEMP(2)

DELO = TEMP(3)
DELPNT = TEMP(4)

TPRIN = TEMP(5)
IF (TPRINO .GT. STIME+DELPNT) TPRINO=STIME+DELPNT

IF (DELO.GT.DELPNT) DELO=DEL PNT

IF (DELO.EQ.0.0) GO TO 13

GO TO 3 NCURV

6 READ (5,101) NCURV
GO TO 3

7 READ (5,102) ISUM1
READ (5,102) ISUM2

8 READ (5,108) IACCEL, IVEL, ITRAJ, ISIDYL, IBOWSL, ISTNSL, IWAVES, IRUD, IP

1 IROP, IAEROD, IRHS
GO TO 3

9 READ (5,111) NEQS, JQQ, (TOL(J), J=1, NFQS)

10 READ (5,108) IVERT, ILATRL, NVD, NVI, NLD, NL

11 GO TO 3

11 CONTINUE

12 I3DOF = TEMP(1)
ISRGE = TEMP(2)

ITRIM = TEMP(3)
IDIA = TEMP(4)

GO TO 3

12 CONTINUE

PRCM01 = TEMP(1)
PRCM02 = TEMP(2)

PRCM03 = TEMP(3)
PRCM04 = TEMP(4)

PRCM05 = TEMP(5)
PRCM06 = TEMP(6)

PRCM07 = TEMP(7)

GO TO 3

13 WRITE (6,110)

STOP

C MASS DISTRIBUTION

CC

INCN 960
INCN 980
INCN 990
INCN 1000
INCN 1010
INCN 1020
INCN 1030
INCN 1040
INCN 1050
INCN 1060
INCN 1070
INCN 1080
INCN 1090
INCN 1100
INCN 1110
INCN 1120
INCN 1130
INCN 1140
INCN 1150
INCN 1160
INCN 1170
INCN 1180
INCN 1190
INCN 1200
INCN 1210
INCN 1220
INCN 1230
INCN 1240
INCN 1250
INCN 1260
INCN 1270
INCN 1280
INCN 1290
INCN 1300
INCN 1310
INCN 1320
INCN 1330
INCN 1340
INCN 1350
INCN 1360
INCN 1370
INCN 1380
INCN 1390
INCN 1400
INCN 1410
INCN 1420
INCN 1430

```

14   6 = 32.17
    RHO = 1.99
    RRHO = RHO*.5
    GO TO (15,17,22), INPT
15   IMM = 0
    WEIGHT = TEMP(1)
    AX = WEIGHT/G
    XS = TEMP(2)
    ZS = TEMP(3)
    AIXX = TEMP(4)
    AIYY = TEMP(5)
    AIZZ = TEMP(6)
    AIXZ = TEMP(7)

C     C     INERTIA MATRIX OPERATIONS
16   AMASSI = 1.0/AM
    D = 1.0/(AIXX*AIZZ-AIXZ*AIXX)
    DIXX = AIXX*D
    DIXZ = AIXZ*D
    DIZZ = AIZZ*D
    AIYYI = 1.0/AIYY
    GO TO 3

C     C     READ WEIGHT DISTRIBUTION - ASSUME TRANSVERSE (PORT/STBD) SYMMETRY
17   I = 1
    READ (5,109) AMI(I),XI(I),YI(I),ZI(I)
    IF (AMI(I).LT.0.0) GO TO 19
    IF (I.GT.201) GO TO 97
    GO TO 18
18   NMASS = I-1
    SUM = 0.0
    SUX = 0.0
    SUZ = 0.0
    DO 20 I=1,NMASS
        AMI(I) = AMI(I)/G
        SUM = SUM+AMI(I)
        SUX = SUX+AMI(I)*XI(I)
        SUZ = SUZ+AMI(I)*ZI(I)
20   AM = SUM*2.0
    WEIGHT = AM*G

```

```

ZS = SUZ/SUM
SUM = 0.0
SUX = 0.0
SUY = 0.0
SUZ = 0.0
C DO 21 I=1,NMASS
      XI(I) = X(I)-XS
      ZI(I) = -Z(I)+ZS
      AMK = AM(I)*2.0
      SUX = SUX+AMK*X(I)*XI(I)
      SUY = SUY+AMK*Y(I)*YI(I)
      SUZ = SUZ+AMK*Z(I)*ZI(I)
      SUM = SUM+AMK*X(I)*ZI(I)
21 C AIXX = SUY+SUZ
      AIYY = SUX+SUZ
      AIZZ = SUX+SUY
      AIYZ = SUM
      GO TO 16
      GO TO 3
22 C XX AND YY TABLES
C 23 CONTINUE
      NSTA(1) = TEMP(1)
      NSTA(2) = TEMP(2)
      NSTA(3) = TEMP(3)
      NSTA(4) = TEMP(4)
      XLTOT = TEMP(5)
      GO TO 3
C C SIDEWALL ( INCLUDING APPENDAGES )
C 24 CONTINUE
      GO TO (25,26), IOPT
25 YSW = TEMP(1)
      XLSW = TEMP(2)
      CFSW = TEMP(3)
      CDSW = TEMP(4)
      AVBMSW = TEMP(5)
      READ (10) ZZ
      REWIND 10
      GO TO 3
C C BLOCK 4 OPTION 2 REMOVED.
C C NO APPENDAGES
C C INCN1920
C C INCN1930
C C INCN1940
C C INCN1950
C C INCN1960
C C INCN1970
C C INCN1980
C C INCN1990
C C INCN2000
C C INCN2010
C C INCN2020
C C INCN2030
C C INCN2040
C C INCN2050
C C INCN2060
C C INCN2070
C C INCN2080
C C INCN2090
C C INCN2100
C C INCN2110
C C INCN2120
C C INCN2130
C C INCN2140
C C INCN2150
C C INCN2160
C C INCN2170
C C INCN2180
C C INCN2190
C C INCN2200
C C INCN2210
C C INCN2220
C C INCN2230
C C INCN2240
C C INCN2250
C C INCN2260
C C INCN2270
C C INCN2280
C C INCN2290
C C INCN2300
C C INCN2310
C C INCN2320
C C INCN2330
C C INCN2340
C C INCN2350
C C INCN2360
C C INCN2370
C C INCN2380
C C INCN2390

```

26 CONTINUE
GO TO 3

C C STERNSEAL

27 CONTINUE
XSSI = TEMP(1)
ZSSEAK = TEMP(2)
CFSS = TEMP(3)
ELMAXS = TEMP(4)
DPSS = TEMP(5)
XLFS = TEMP(6)
ARCA = ELMAXS/XLF
COSTH = ARCA*SARGA
THSSI = SIN(THSSI)
GO TO 3

C C BOWSEAL

28 GO TO (29,30), IOPT
XBSI = TEMP(1)
CFBS = TEMP(2)
DPBS = TEMP(3)
ZBSI = TEMP(4)
ELMAXB = TEMP(5)
XBFB = TEMP(6)
BLFEAK = TEMP(7)
GO TO 3
30 CENCAB = TEMP(1)
GO TO 3

C C PLENUM

31 CONTINUE
GO TO (32,33), IOPT
32 XLBW = TEMP(1)
XBBW = TEMP(2)
XPWV = TEMP(3)
WIDTH = TEMP(4)
XL = TEMP(5)
XCP0 = TEMP(6)
BUBHGT = TEMP(7)
XLXPWV = XLBW-XPWV
PWVHT = (XPWV*XPWV-XLXPWV)*XLXPWV

PWVHT = (XPWV*XPWV-XLXPWV)*XLXPWV

```

XPPWVXS = XPPWV-XS
ABW = XB BW*XLBW
AB = WIDTH*X
VCLNOM = (ABW+AB)*BUBHGT*.5
GO TO 3
CONTINUE
FNCRIT = TEMP(1)
GO TO 3

C     PROPELLSION
34    CONTINUE
      GO TO (35,36), IOPT
35    CONTINUE
      XPO = TEMP(1)
      YPO = TEMP(2)
      ZPO = TEMP(3)
      GO TO 3

C     BLOCK 8 OPTION 2 REMOVED. ENGINE OUT INPUT IN BLOCK 16
36    CONTINUE
      GO TO 3

C     RUDDER
37    CCNTINUE
      GO TO (38,39;40), IOPT
38    XRO = TEMP(1)
      YR = TEMP(2)
      ZRO = TEMP(3)
      RSPAN = TEMP(4)
      RASPR = TEMP(5)
      RAAREA = TEMP(6)
      RCLB = 2.*PI*RASPR/(RASPR+3.)
      RTC = TEMP(7)
      GO TO 3

C     910 NOT USED
39    CONTINUE
      GO TO 3
40    CONTINUE
      GO TO 3

C     AERODYNAMICS
41    CONTINUE

```

```

XL AERO = TEMP(1)
BEAM = TEMP(2)
RHOA = .5*RHOINF*XLAERO*BEAM
GO TO 3

C C C WAVES
C C C

42 CONTINUE
IWAWSW = ICPT
IF(IWAWSW.GT.4) GO TO 97
NWAVE = TEMP(1)
IF(NWAVE.EQ.0.) GO TO 3
IF(NWAVE.GT.20) GO TO 97
BETAD = TEMP(2)
BETA = BETAD/RAD
COSBET = COS(BETA)
SINBET = SIN(BETA)
TC = 1.0
GO TO (43,45,47,47), IWAWSW

C 43 DO 44 I=1,NWAVE
44 PREAD (5,116) OMFGA(I),AW(I)
C GO TO 3

C 45 DC 46 I=1,NWAVE
46 READ (5,116) WAVLEN(I),AW(I)
C GO TO 3

C 47 SHIWV = TEMP(3)
G1G = 32.17
G2 = G1G*G1G
G4 = G2*G2
GO TO (3,3,48,49), IWAWSW

C 48 CONTINUE
PERL = TEMP(4)
PERH = TEMP(5)
WWN = (2.0*3.141592)*(1.0/PERH)
WWX = (2.0*3.141592)*(1.0/PERL)
GO TO 50
CONTINUE
WWN = TEMP(4)
WWX = TEMP(5)

C 49 CONTINUE
WWN = TEMP(4)
WWX = TEMP(5)

50 CONTINUE
UUU=SQRT(SHT WV*54.0)*1.6878
UU4 = UUU*UUU*UUU*UUU
CCC = (WWX/WWN)**(1./NWAVE)
WWPO = WWN

```

```

C DO 51 I=1,NWAVE
      WWPN = WWPNC $\ddot{C}$ CCC
      WW = (WWPN+WWPO)*.5
      DDW = WWPN-WPO
      WWPO = WWPN
      WW4 = WW*WW*WW*WW
      WW5 = WW*WW4
      SS = 0.0081*G2/(EXP(0.74*G4/(WW4*U14))*WW5)
      GMEGA(I) = WW
      AAW(I) = SQRT(2.*SS*DDW)
      CONTINUE
51    GO TO 3
C INITIAL CONDITIONS
52    CONTINUE
      UC = TEMP(1)
      THETO = TEMP(2)
      DSO = TEMP(3)
      DELPI = TEMP(4)
      DPHI = TEMP(5)
      GO TO 3
53    CONTINUE
C INPUT COMPLETED. 1) PRINT ALL INPUT
      WRITE(6,123) TITLE
      WRITE(6,124) STIME,FTIME,DELO,TPRINO,DELPT
      WRITE(6,125) IACCEL,IVEL,ITRAJ,ISIDL,ISTNSL,IWAVES,IRUD,I
      1 FPROP,IAROD,Irhs
      WRITE(6,135) IDOF,ISRGF,ITRIM,IDA
      WRITE(6,142) PROMO1,PROMO2,PROMO3,PROMO4,PROMO5,PROMO6,PROMO7
      WRITE(6,126) NEQS('TOL(J),J=1,NEQS')
      WRITE(6,113) WEIGHT,XS,ZS,AIXX,AIYY,AIZZ,AIXZ
      WRITE(6,112) AIMAX
      WRITE(6,133) NSTA
      WRITE(6,114) YSW,XLSW,CFSW,CDSW,VCHORD,VXO,VY,VZO,AV
      1 BMSN,VT
      WRITE(6,115) NALDAL,SAL,NDSD,SDS,NTH,DTH,STH,NBB,DBB,SBB
      IF (IM*6T*0) WRITE(6,121) (XMO(J),J=1,IMNX)
      WRITE(6,119) IMM,IMNX,IMNY,IBMFIL,BTIME,IMT
      IF (IMM*GT*0) WRITE(6,122) (YMI(J),J=1,IMNY)
      WRITE(6,128) XLBW,XBBW
      WRITE(6,129) XL,WIDHT,XCPO,VOLNOM,BUBHG T
      WRITE(6,134) DELPI
      WRITE(6,127) FNCRIT,XLTOT

```

```

      WRITE(6,141) ENBFAN, BRPM, ENMFAN, EMPM, ENSFAN, SRPM,
      WRITE(6,131) XRO, YR, ZRO, RONO, RMAX0, RRATO, RREVO, DLRDO, RSPAN, RASPR, INCN4320
      1 RAREA, RCLBRTIC
      WRITE(6,130) XPO, YPO, ZPO
      WRITE(6,140) XLAERO, BEM
      WRITE(6,139) XBSI, CFBSS, DPBS, ELMAXB
      WRITE(6,138) XSSI, ZSSI, ALLEAK, CFSS, ELMAXS, DPS, XLF
      WRITE(6,132) UN, THETO, DSO
      INCN4330
      INCN4350
      INCN4360
      INCN4370
      INCN4380
      INCN4390
      INCN4400
      INCN4410
      INCN4420
      INCN4430
      INCN4440
      INCN4450
      INCN4460
      INCN4470
      INCN4480
      INCN4490
      INCN4500
      INCN4510
      INCN4520
      INCN4530
      INCN4540
      INCN4550
      INCN4560
      INCN4570
      INCN4580
      INCN4590
      INCN4600
      INCN4610
      INCN4620
      INCN4630
      INCN4640
      INCN4650
      INCN4660
      INCN4670
      INCN4680
      INCN4690
      INCN4700
      INCN4710
      INCN4720
      INCN4730
      INCN4740
      INCN4750
      INCN4760
      INCN4770
      INCN4780
      INCN4790

      AND 2) INITIALIZE VARIABLES FOR CALCS.

C
C      DO 54 I=1,40
C      VAL(I) = 0.0
C
C      U = U0*1.6889
C      XSS = -(XS-XSSI)
C      ZSS = ZS-ZSSI
C      THETA = THETO/RAD
C      PHI = DPHI/RAD
C      THEQL = THETA
C      DS = DSO*.0833
C      Z = -ZS+DS
C      ZEQUIL = Z
C      PHIMAX = 0.
C      TROLL = 0.
C      IRDS = 0
C      TL = 0.0

      WAVE PARAMETERS TABLE
      IF (NWAVE.EQ.0) GO TO 61
      AMPTC = 1.30287
      GO TO (55,57), IWAVSW

C      DO 56 I=1,NWAVE
C      WAVLEN(I) = 2.*PI*G/(OMEGA(I)*WAVLEN(I))
C      GO TO 59

C      DO 58 I=1,NWAVE
C      CMEGA(I) = SQRT(2.*PI*G/WAVLEN(I))
C      58 CONTINUE
C      CALCULATE INITIAL FREQUENCIES OF ENCOUNTER
C

```

```

DO 60 I=1,NWAVE
WAVSLP(I) = 360.0*NWAVE(I)/WAVLEN(I)
CMEGAE(I) = 2.*PI*(SQR(G*WAVLEN(I)/(2.*PI))-U*COSBET)/WAVLEN(I)
ENCPER(I) = 2.0*PI/OMEGA(I)
CONTINUE
C
      WRITE(6,117) NWAVE,BETAD,OMEGA(I),OMEGA(I),WAVLEN(I),AW(I),WAUS(I)
1 LP(I),ENCPER(I),I=1,NWAVE}
      GO TO 62
      WRITE(6,118)
61 CONTINUE
C
      DO 63 I=1,4
C
      DO 63 N=1,1
63 ETA(I,N)=0.0
C
      DVOLW = 0.0
      FXWAV = 0.0
      FYWAV = 0.0
      FZWAV = 0.0
      FKWAV = 0.0
      FMWAV = 0.0
      FNWAV = 0.0
      ZBAR = Z
      PHIBAR = PHI
      THEBAR = THE TA
      TIME = STIME
      DELT = DELC
      TPRINT = TPRINT-DELPNT
      PWYCON = 4.*WEIGHT/(RHO*G*XLBW)
      FNCON = SQR(XLBW*G)
      VY = VY0-XS
      VZ = VS-VZO
      XP = XPO-XS
      XR = XRO-XS
      YP = YPO
      ZP = ZS-ZPO
      ZR = ZS-ZRO
      IF (IRM.EQ.0) GO TO 65
C
      DO 64 J=1,IMNX
64 XM1(J) = XM0(J)-XS
C
      65 CONTINUE
      XCP = XCPO-XS
      ZCP = ZS-BUBHGT
      XBS = XBSI-XS

```

```

N = NSTA(3)
ZBS = ZS-ZBSI
C
DO 66 J=1,N
DELYBS = XBBW/(N-1)
XX(3,J) = XBS-XSS
YY(3,J) = -0.5*XBBW*(J-1)*DELYBS
66 CONTINUE
C
N = N-1
C
DO 67 J=1,N
YAVGB(J) = (YY(3,J+1)+YY(3,J))*0.5
67 CONTINUE
C
N = NSTA(4)
DELYSS = XBBW/(N-1)
C
DO 68 J=1,N
XX(4,J) = -XSS
YY(4,J) = -0.5*XBBW*(J-1)*DELYSS
68 CONTINUE
C
N = N-1
C
DO 69 J=1,N
YAVGS(J) = (YY(4,J+1)+YY(4,J))*0.5
69 CONTINUE
C
XBOW = XLTOT-XS
N = NSTA(1)
DELX = XBSI/(N-1)
C
DO 70 J=1,2
C
DO 70 I=1,N
XX(J,I) = (I-1)*DELX-XS
70 YY(J,I) = YSW*(2*I-3)
C
WRITE(6,103) ((XX(J,N),N=1,11),(YY(J,N),N=1,11),J=1,4)
C
DO 71 I=1,N
XAVG(I) = DELX*(2*I-1)*0.5-XS
71
C
CALL WAVES (TIME)

```

INITIALIZE BUBBLE PRESSURE, ABSOLUTE (PSF)

```

C PB = PINF+DELP1
C PBBAR = DELPI
C PBAR = DELPI
C PSS = PB+DPSS
C PBS = PB+DPBS
C AB = ABW-(ABW-AB)**(ZS+Z/BUBHGT)
C CF = .37/((U/FNCN)**1.5655981)
C WATSLP = PBBAR*CF*PWVCN/WFLIGHT
C IF (IDIA.EQ.1) GO TO 72
C VOL = VOLNOM-.5*(AB+ABW)*(Z+ZS)-DVOLW+.5*WATSLP*XL*AB
C GO TO 73
C VOL = VOLNOM-.5*(AB+ABW)*(Z+ZS)-DVOLW+PBAR*.3175+.5*WATSLP*XL*AB
C CONTINUE
C BMASS = (PB/PINF)**(1./GAM)*VOL*RHOINF
C WRITE (6,137)
C DVWDOT = 0.0
C RETURN

C RUN TERMINATOR
C 74 WRITE (6,106)
C STOP

C BENDING MOMENT
C 75 GO TO (76,77,79,81), IOPT
C 76 IMM = TEMP(1)
C IF (IMM.GT.3) GO TO 97
C IMNX = TEMP(2)
C IF (IMNX.GT.10) GO TO 97
C IMNY = TEMP(3)
C IF (IMNY.GT.7) GO TO 97
C IBMFIL = TEMP(4)
C BTIME = TEMP(5)
C IF (IMM.EQ.3) IMT = TEMP(6)
C GO TO 3

C 77 DO 78 J=1,7
C 78 XM0(J) = #EMP(J)

C IF (IMNX.LE.7) GO TO 3
C READ 120, (XM0(J), J=8, IMNX)
C GO TO 3

C 79 DO 80 J=1,IMNY
C 80 YM1(J) = #EMP(J)

```

```

C     GO TO 3
81    CONTINUE
C     GO TO 3
82    CONTINUE
C     GO TO (83,85,87), IOPT
83    CONTINUE
C     VALUES INPUT FOR STD SCREW
C
C     THST1 = TEMP(1)
NPS = TEMP(2)
C     STHS = TEMP(3)
IF (NPS.EQ.0.0) GO TO 84
READ (5,104) (TIS(j),j=1,NPS)
GO TO 3
C     THSTS(1) = THST1
84    CONTINUE
C     VALUES INPUT FOR PORT SCREW
C
C     THST2 = TEMP(1)
NPP = TEMP(2)
C     STHP = TEMP(3)
IF (NPP.EQ.0.0) GO TO 86
READ (5,104) (TIP(j),j=1,NPP)
READ (5,104) (THSTP(j),j=1,NPP)
GO TO 3
C     THSTP(1) = THST2
86    CONTINUE
C     VALUES INPUT FOR RUDDER
C
C     DELR = TEMP(1)
NPR = TEMP(2)
C     IF (NPR.EQ.0.0) GO TO 88
READ (5,104) (TR(j),j=1,NPR)
READ (5,104) (DELRUD(j),j=1,NPR)
GO TO 3
C     DELRUD(1) = DELR
88    CONTINUE
C     GO TO 3
89    GO TO (90,91), IOPT
C     NB = TEMP(1)
90    NB = TEMP(1)
READ (5,104) (TMEB(I),I=1,NB)
READ (5,104) (DFLB(I),I=1,NB)
GO TO 3
C     NS = TEMP(1)
91    NS = TEMP(1)

```

```

READ (5,104) (TME$({I}), I=1,NS)
READ (5,104) (DETS({I}), I=1,NS)
C      TITLE CARD (ALL 80 COLUMNS )
C      READ (5,136) TITLC
GO TO 3
C      FAN MAPS
C      93 CONTINUE
      GO TO (94,95,96), IOPT
C      94 CONTINUE
      ENBFAN = TEMP(1)
      BRPM = TEMP(2)
      NPTSB = TEMP(3)
      READIN = TEMP(4)
      IF (READIN.EQ.0.0) GO TO 3
      READ (5,104) (PBFAN({J}), J=1,NPTSB)
      READ (5,104) (QBFAN({J}), J=1,NPTSB)
      GO TO 3
C      95 CONTINUE
      ENMFAN = TEMP(1)
      EMRPM = TEMP(2)
      NPTSM = TEMP(3)
      READIN = TEMP(4)
      IF (READIN.EQ.0.0) GO TO 3
      READ (5,104) (PMFAN({J}), J=1,NPTSM)
      READ (5,104) (QMFMAN({J}), J=1,NPTSM)
      GO TO 3
C      96 CONTINUE
      ENSFAN = TEMP(1)
      SRP1 = TEMP(2)
      NPPTS = TEMP(3)
      READIN = TEMP(4)
      IF (READIN.EQ.0.0) GO TO 3
      READ (5,104) (PSFAN({J}), J=1,NPTSS)
      READ (5,104) (QSFFAN({J}), J=1,NPTSS)
      GO TO 3
C      97 CONTINUE
      WRITE (6,105) ISYS
      STOP
C      ERROR IN INPUT
C      98 CONTINUE
      INCN6720
      INCN6730
      INCN6740
      INCN6750
      INCN6760
      INCN6770
      INCN6780
      INCN6790
      INCN6800
      INCN6810
      INCN6820
      INCN6830
      INCN6840
      INCN6850
      INCN6860
      INCN6870
      INCN6880
      INCN6890
      INCN6900
      INCN6910
      INCN6920
      INCN6930
      INCN6940
      INCN6950
      INCN6960
      INCN6970
      INCN6980
      INCN6990
      INCN7000
      INCN7010
      INCN7020
      INCN7030
      INCN7040
      INCN7050
      INCN7060
      INCN7070
      INCN7080
      INCN7090
      INCN7100
      INCN7110
      INCN7120
      INCN7130
      INCN7140
      INCN7150
      INCN7160
      INCN7170
      INCN7180
      INCN7190

```

```

98 FORMAT (2I2)
99 FORMAT (26I2)
100 FORMAT (6A8)
101 FORMAT (10I1)
102 FORMAT (8I2)
103 FORMAT (//17H XX AND YY ARRAYS /14H PORT SIDEWALL /2(11F10.2/),1H STERN /2(11F10.2/),1H S/15H SEAL /2 INCN 7250
104 FORMAT (11F10.2/)
105 FORMAT (8F10.0)
106 FORMAT (1H120(/)50X,19H COMPLETE ALL RUNS)
107 FORMAT (13I2,7F10.0)
108 FORMAT (16I5)
109 FORMAT (5F10.0)
110 FORMAT (//10X65HEX IN INPUT --- DELT AND/OR DELPNT EQUALS ZER INCN 7340
110 --- JOB ADAPTED)
111 FORMAT (2I2/(8F10.0))
112 FORMAT (22H INERTIA MATRIX, A1MAX6E15.4/(22X,6E15.4))
113 FORMAT (30H WEIGHT, C6.0! INERTIA MOMENT S7F12.3)
114 FORMAT (15H SIDEWALL INPUT 12(F8.31X))
115 FORMAT (26H SIDEWALL TABLE PARAMETERS(14,F7.3,F7.3))
116 FORMAT (2F10.0)
117 FORMAT (12HONO OF WAVES12,10H BETA(DEG) F5.0/15H OMEGA(RAD/SEC) 5X
1,16H OMEGA(RAD/SEC) 5X16H WAVE LENGTH(FT) 5X14H AMPLITUDE(FT) 5X,1
2,6HMAX SLOPE(DEG) 5X13HPERIOD,E(SEC)/(F8.4,12X,F8.4,4F20.3))
118 FORMAT (/11HOCALM WATER)
119 FORMAT (32HOMOMENT CALC. CONTROL PARAMETERS(15,F8.3,15)
120 FORMAT (5X,7F10.0)
121 FORMAT (22H MOMENT CALCS. AT X OF 11F10.3)
122 FORMAT (22H MOMENT CALCS. AT Y OF 11F10.3)
123 FORMAT (33H1SES MOTIONS AND LOADS PROGRAM - 20A4*/)
124 FORMAT (23H START AND FINISH TIMES 2F10.2/22H INITIAL TIME INTERVAL
1F12.4/18H START PRINTING AT F8.2,17H IN INCREMENTS OFF8.2)
125 FORMAT (24H INTERMEDIATE PRINT TAGS! 615)
126 FORMAT (39H NO. OF STATE EQUATIONS, AND TOLERANCES 15/(10X, 10E12.2))
127 FORMAT (23HOCRITICAL FROUDE NUMBER F15.4, 5X, 19H TOTAL CRAFT LENGTH INCN 7560
115.4)
128 FORMAT (34HOPLENUM, LENGTH AND WIDTH AT WATER2F12.4)
129 FORMAT (34HOPLENUM, LENGTH AND WIDTH AT HULL2F12.4/35H PLENUM; CF INCN 7570
1INTER OF PRESSURE AT HULLF12.4/23H PLNUM, NOMINAL VOLUME F12.1,10X, INCN 7610
26HHEIGHT F12.4)
130 FORMAT (33H PROPELLSION, X, Y, Z COORDINATES 3F12.4*)
131 FORMAT (/28HORRUDER X, Z COORDINATES 3F12.4/41H RUDDER ON, MAX INCN 7630
1, RATE, REVERSE, INITIAL 5F12.4/33H Rudder, SPAN, ASPECT, AREA, CLB, T/ INCN 7640
2, C5F12.4)
132 FORMAT (/39H0 INITIAL CONDITIONS, VELOCITY (KNOTS) = F7.2, 5X, 13HPITC INCN 7650
11H (DEG) = F8.3, 5X, 12HDRAFT (IN) = F8.2) INCN 7660

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133 FORMAT (49H NUMBER OF STATIONS, SIDEWALLS (P+S) ,4 15 ) INCN7680
134 FORMAT (38F PLENUM, INITIAL PRESSURE GAGE (PSF) F8.2) INCN7690
135 FORMAT (79H PROGRAM OPTION SWI TCH SETTINGS (LATERAL PLANE, CONSTANT INCN7700
136 SPEED, TRIN, MEMBRANE) 715) INCN7710
137 FORMAT ((20A4)) INCN7720
138 FORMAT (1H1) INCN7730
139 FORMAT (16H STERNSEAL INPUT 7F12.4) INCN7740
140 FORMAT (16H BOWSEAL INPUT 7F12.4) INCN7750
141 FORMAT (19H AERODYNAMICS INPUT 7F12.4) INCN7760
142 FORMAT (33H OFANS, NO + RPM, BOW, MAIN, STERN 3(F10.0, F10.1)) INCN7770
143 FORMAT (32H PROGRAM MODIFICATION SETTINGS 7(F12.4,1X)) INCN7780
144 FORMAT ((1H1)) INCN7790

133 FORMAT (49H NUMBER OF STATIONS, SIDEWALLS (P+S) ,4 15 ) INCN7680
134 FORMAT (38F PLENUM, INITIAL PRESSURE GAGE (PSF) F8.2) INCN7690
135 FORMAT (79H PROGRAM OPTION SWI TCH SETTINGS (LATERAL PLANE, CONSTANT INCN7700
136 SPEED, TRIN, MEMBRANE) 715) INCN7710
137 FORMAT ((20A4)) INCN7720
138 FORMAT (1H1) INCN7730
139 FORMAT (16H STERNSEAL INPUT 7F12.4) INCN7740
140 FORMAT (16H BOWSEAL INPUT 7F12.4) INCN7750
141 FORMAT (19H AERODYNAMICS INPUT 7F12.4) INCN7760
142 FORMAT (33H OFANS, NO + RPM, BOW, MAIN, STERN 3(F10.0, F10.1)) INCN7770
143 FORMAT (32H PROGRAM MODIFICATION SETTINGS 7(F12.4,1X)) INCN7780
144 FORMAT ((1H1)) INCN7790

SUBROUTINE INTGRL (TIME)
  INTEGER CN
  COMMON /BMC0/ PINF, RHOINF, GAM
  COMMON /EQNC0/ IMN, IMN, IBMFIL, RTIME, IMT, XM1(10), YM1(7), IX, IY
  COMMON /KSWTCH/ ITHRST, NFOQS, TOL(20), JQQ
  COMMON /MASSES/ AMAXXX, AIXZ, AIZZ, AIMAX, G, WFIGHT, RHO, NMASS, AM
  11(2C1), XOPTION, Y(201), ZXS, ZSHRH0
  COMMON /OPTION/ I3DOF, ISRGE, ITIM, IDIA
  COMMON /PLENUM/ XBBW, XBRW, ABW, BUBHG, DVWDOT, VMDOT, PMDOT
  COMMON /PRIME/ STIME, FTIME, DELT, DELTPNT, TPRINT
  COMMON /PROMOD/ PROMOL, PROMO2, PROM03, PROM04, PROM05, PROM06, PROM07, I
  COMMON /PRTINT/ ON, IACCEL, IVFL, ISTNSL, IWAVES, I
  1RUD, IPROPAEROD, IRHS
  COMMON /STAB/ S(4), ISTAB
  COMMON /STEP/ STEP2
  COMMON /VALOLD/ YOLD(20)
  COMMON /VARIABLE/ VAL(40)
  EQUIVALENCE (VAL(1), X)(1)(VAL(2), Y(1))
  DIMENSION Y(20), FRROR(20)
  REAL K1(20), K2(20), K3(20), K5(20)
  DATA IPASS/0/
  STEP2 = 1.0
  PB = VAL(24)
  BMASS = Y(10)
  IF ((TIME+DELT).LE.TPRINT) GO TO 1
  DELT = DELT
  DELT = TPRINT-TIME
  IPASS = 1
  1 X = TIME
  DO 2 J=1,NEQS
    Y(J) = YOLD(J)
  2 CONTINUE

```



```

DC 11 J=1,NEQS
11 CONTINUE
C
12      DELT = 2.*DELT
13      IF (DELT.GT.DELPNT) DELT=DEL PNT
14      STEP2 = DELT
15      GO TO 12
16      IPASS = 0
17      GO TO 4
18      WRITE(6,21) TIME,DELT,(K1(J),J=1,NEQS),VAL
      STOP
C
19      FORMAT (/10X,23HINTGRL TIME,DELT,K1,VAL/2E15.4/),5(8E15.4
1/ )
20      FORMAT (1H0,9X,33HTOTAL LATERAL ACCELERATION (6) = F12.4,12X,5HD T
1=E15.4)
21      FORMAT (1H1'10X,44HDELT TIME LESS THAN 1.OE-6 - - JOB STOPS)
22      FORMAT (/10X,5HINT-J2E30.5,I5,2E20.5)
      END
C
      SUBROUTINE PROP
C
      INTEGER ON
      COMMON /CCNST/ PI,RAD,UO
      COMMON /FPROP/ FX,FY,FZ,FK,FM FN
      COMMON /ENGINE/ NPS,NPP,THSTS(25),THSTP(25),XP,YP,ZP,STHS,STHP,TIPRC
      1(25)T IS(25)
      COMMON /PRTNT/ ON,IACCEL,IVEL,ITRAJ,ISIDWL,IBOWSL,ISTNSL,IWAVES,IPROP
      1RUD,IPROP,IAEROD,IRHS
      COMMON /PROMOD/ PROMOD1,PROMOD2,PROMOD3,PROMOD4,PROMOS5,PROMOD6,PROMOD7
      1CLB,RTC,RUDANG,TIR(25)
      COMMON /VARIABLE/ VAL(40)
      EQUIVALENCE (VAL(1),T'ME),(VAL(2),U),(VAL(3),V),(VAL(4),W),
      1L(5),P),(VAL(6),Q),(VAL(7),R),(VAL(8),PHI),(VAL(9),THETA),(VAL(10),
      150
      PROP 10
      PPROP 20
      PPROP 30
      PPROP 40
      PPROP 50
      PPROP 60
      PPROP 70
      PPROP 80
      PPROP 90
      PPROP 100
      PPROP 110
      PPROP 120
      PPROP 130
      PPROP 140
      VAPROP 150
      VAPROP 150

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```

2L(10),Z),(VAL(11),RMASS), (VAL(21),X), (VAL(22),Y), (VAL(23),PSI) PR UP
3,(VAL(24),PB) PR UP 160
DIMENSION THS(1) THP(1) TS(1) { THSTP(1) RUD(1) } TR(1)
EQUIVALENCE (THS(1),THS(1)) { THSTP(1) THP(1) } { TIS(1) }
1(TIP(1),TP(1)), (TR(1),TR(1)), (DELRUD(1),RUD(1)), PR UP 170
PR UP 180
PR UP 190
PR UP 200
PR UP 210
PR UP 220
PR UP 230
PR UP 240
PR UP 250
PR UP 260
PR UP 270
PR UP 280
PR UP 290
PR UP 300
PR UP 310
PR UP 320
PR UP 330
PR UP 340
PR UP 350
PR UP 360
PR UP 370
PR UP 380
PR UP 390
PR UP 400
PR UP 410
PR UP 420
PR UP 430
PR UP 440
PR UP 450
PR UP 460
PR UP 470
PR UP 480
PR UP 490
PR CP 500
PR CP 510
PR CP 520
PR CP 530
PR CP 540
PR CP 550
PR CP 560
PR CP 570
PR CP 580
PR CP 590
PR CP 600
PR CP 610
PR CP 620
PR CP 630
C
FX = 0.0
FY = 0.0
FZ = 0.0
FK = 0.0
FM = 0.0
FN = 0.0
TL = TIME
IF (NPR.EQ.0) GO TO 1
RUDANG = FG1(TL,NP,TR,RUD,IR)
RUEANG = RUDANG/RAD
C
C CALCULATE THRUSTS AND MOMENTS INDIVIDUALLY
C
GO TO 2
1 RUDANG = DELRUD(1)
2 CD = CCS(RUDANG)
SD = SIN(RUDANG)
IF (NPS.EQ.0) GO TO 3
THSS = FG1(TL,NPS,TS,THS,IS)
GC TO 4
3 THSS = THSTS(1)
IF (NPP.EQ.0) GO TO 5
THSP = FG1(TL,NPP,TP,THP,IP)
GC TO 6
5 THSP = THSTP(1)
STHSTS = STHS*THSS
STHSTP = STHP*THSP
FXS = THSS*CD - STHSTS*SD
FXP = THSP*CD + STHSTP*SD
FYS = -STHSTS*CD - THSS*SD
FYP = STHSTP*CD + SD*THSP
FZS = -THSS*THETA*CD + STHSTS*SD*PHI
FZP = -THSP*THETA*CD - STHSTP*SD*PHI
FX = FXP+FXS
FY = FYP+FYS
FZ = FZP+FZS
FKP = -FZP*YP-FYP*ZP
FKS = FZS*YP-FYS*ZP
FK = FK+FKP
FMS = FZS*(-XP)+FYS*ZP
FMP = FZP*(-XP)+FYP*ZP

```

C

C

```

FM = FMS+FMP
FNS = -FXS*YP-FYS*(-XP)
FNP = FXP*YP-FYP*(-XP)
FN = FNS+FNP
IF (I PROP*NE*ON) RETURN
WRITE (6,7) FX,FY,FZ,FK,FM,FN
RETURN
C    7 FORMAT (/10X,22HPP OP FX,FY,FZ,FK,FM,FN/6F15.4)
END

```

SUBROUTINE RUDDER

```

C
      INTEGER CN
      COMMON /CONST/ PI,RAD,U0
      COMMON /FRUD/ FX,FY,FZ,FK,FM,FN
      COMMON /MASSES/ AM,AIXX,AIZZ,AIXZ,AIMAX,G,WEIGHT,RHO,NMASS,AMRUD
      1 I(201),X,I(201),Y,I(201),XS,ZS,HRHO
      COMMON /PRCMOD/ PROMOD1,PROMOD2,PROMOD3,PROMOD4,PROMOD5,PROMOD6,PROMOD7
      COMMON /PRTINT/ ON,IACCFL,IVEL,ITRAJ,ISIDLW,IBOWSL,ISTNSL,IWAVES,IRUD
      IRUD,IPROP,IAEROD,IRHS
      COMMON /RUDDR/ NPR,DELRUD(25),XR,YR,ZR,IRDS,TL,RSPAN,RAREA,RASPR,RUD
      1CLB,RTC,RUDANG,TIR(25)
      COMMON /VARBLE/ VAL(40)
      EQUIVALENCE (VAL(1),TIME),(VAL(2),U),(VAL(3),V),(VAL(4),W),
      1L(5),P),(VAL(6),G),(VAL(7),R),(VAL(8),E),(VAL(9),PHI),
      2L(10),Z),(VAL(11),BMASS),(VAL(21),X),(VAL(22),Y),(VAL(23),PSI)
      3 VAL(24),PB
      EQUIVALENCE (DELPUD(1),RUD(1)),(TR(1),TR(1))
      EQUIVALENCE (RUD(1),TR(1))
      EQUIVALENCE (VAL(18),FANPWR)
      DATA ENU/1.28E-5/
C
      C      CALCULATE PROGRAMMED RUDDER DEFLECTION
      TL = TIME
      IF (NPR.EQ.0.0) GO TO 1
      GO TO 2
      1 RUDANG = DELRUD(1)
      GO TO 3
      2 RUDANG = FG1(TL,NPR,TR,RUD,IR)
      C      SIDE FORCE ON RUDDER
      C
      3 DSR = Z+ZS-XR*THETA
      ENDFAC = (1.+DSR/(DSR+RSPAN))
```

```

VH = V+XR*R-ZR*P
CQ = HRHO*U*U*RAREA
EFFANG = RUDANG-ENDFAC*VH/U
FY = 2.*QQ*ENDFAC*RCLB*EFFANG

DRA G FORCE ON RUDDER

```

```

REY = U*(RAREA/RSPAN)/ENU
CFR = .427/(ALOG10(REY)-.407)*2.64
PI8 = PI/8*CFR+PI8*RTC*RTCA*(1.+G*RSAN/(U*U))+RCLB*EFFANG
CD = 2.*CD*RAREA*HRHO*U*U
FX = -2.*CD*EFFANG
FZ = 0.
FK = -ZR*FY
FM = FX*ZR
FN = XR*FY
IF (IRUD.NE.0) RETURN
WRITE (6,4) FX,FY,FZ,FK,FM,FN

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RETURN

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4 FORMAT (/10X,24H RUDDER FX,FY,FZ,FK,FM,FN/6E15.4)

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SUBROUTINE RHS (VALUF)

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INTEGER CN
COMMON /AIR/ PINF,RHOINF,GAM
COMMON /ATPIX/ AMASSI,AIYYI,DIXX,DIZZ
COMMON /BMCOLMN/ IMM,IMMX,IMNY,IBMFIL,BTIME,IMT,XMI(10),YMI(7),IX,IY
COMMON /COLUMN/ IVERT,ILATRL
COMMON /CONST/ PIRAD,UO
COMMON /CNTRL/ CONTW,CONTQ,CONTTH,QMULT,LOUVER,ACONTZ,ACONTW,ZQUIRH
1L,THEQL,ACBASE/NPS,NPP,THSTS(25),THSTP(25),XP,YP,ZP,STHS,STHP,TIPRHS
1(25),TIS(25)
COMMON /FANMAP/ QIN,QBFAN(25),QSFAN(25),ENMFAN,ENRFAN,ENRHS
1SFAN,BRPM,SQRM,NPTSM,NPTS,PBFAN(25),PMFAN(25),PSFAN(25RHS
2),TMEB(25),DELR(25),NB,TESS(25),DETS(25),NS
COMMON /FAERO/ FXAED,FYAED,FZAED,FKAED,FNAED
COMMON /FORBS/ FXBS,FZBS,FKBS,FMBS,QLBS
COMMON /FORSS/ FXSS,FYSS,FZSS,FKSS,FMSS,QLSS,FMS
COMMON /FPROP/ FXP,FYP,FZP,FKP,FNP
COMMON /FRDUD/ FN,FNCRIT
COMMON /FFUD/ FXFUD,FYRUD,FZKUD,FKPUD,FMRUD,FNRUD
COMMON /GBDM/ XBDW
COMMON /GEOM/ WIDTH,XL,XX(4,11),YY(4,11),NSTA(4),AB,VOLNOM,DELS(4,RHS
110),XCP,ZCP

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COMMON /GEOMBS/ DE TABX(11), DETABT(11), ARM1B(10), ARM2B(10), DFBS(10) RHS
1 TSKIE(10) RHS
1 COMMON /GEOMSS/ DETADX(11), DETADT(11), ARMS(10), DFSS(10), TSKIS(10) RHS
1 ARM2S(10) RHS
1 COMMON /KSWATCH/ ITHR ST RHS
1 COMMON /MASSES/ AM,AIXXX,AIYY,AIZZ,AIXZ,AIMAX,G,WEIGHT,RHO,NMASS,AMRHS
1 (201),XI(201),YI(201),XSS,ZS,HRHO RHS
1 COMMON /MSIDW/ DF(2,10),DSWA(2,10),FXH(2),FYH(2),FZH(2),FMH(2),FNRHS
1 H(2)VY(2)VFZ(2)FXV RHS
1 COMMON /MWAVE/ FXW(2),FYW(2),FZW(2),FKW(2),FMW(2),FNW(2) RHS
1 COMMON /OPTION/ 1300F,ISRGE,ITRIM,INDIA RHS
1 COMMON /PLENUM/ XLBW,XBBW,ABW,BUBHG,DVWDOT,VMDOT,CMM RHS
1 COMMON /PRIME/ STIME,FTIME,DEL,TDEL,TDELPNT,PRINT RHS
1 COMMON /PRTINT/ PRINT,IVEL,ITRAJ,ISIDL,IOWSL,ISTNSL,IWAVES,I RHS
1 RUD,IPROP,IAEROD,IRHS RHS
1 COMMON /PROMOD/ PROMO1,PROMO2,PROMO3,PROMO4,PROMO5,PROMO6,PROMO7 RHS
1 COMMON /Pwave/ FNCN,PWCON RHS
1 COMMON /RUDDR/ NPR,DELFRD(25),XR,YR,ZR,IRDS,TL,RSpan,RAREA,RASPR,RKHS RHS
1 CLBL,RT,RC,RUDANG,TIF(25) RHS
1 COMMON /SIDE/ FXSW,FZSW,FKSW,FMSW,FNSW,ALSW,YSW,XLSW,CFSW,CDSRHS
1 W,VARERA,VCHORD,VSPAN,VANGLE,VCONS,VX,VZ,AVBMSW,DELX,VTC RHS
1 COMMON /SLOPE/ WATSL,PXPWV,XL,PXPWV,XL,PXPWV,XL,PXPWV,XS RHS
1 COMMON /SOFTBS/ XBF,PBS,SINBS,COSBS,XBS,DELYBS,PBSS,ELMAXB,YAVRHS RHS
1 GB(10),CENCAB RHS
1 COMMON /SOFTSS/ XLF,PSS,SINTH,COSTH,XSS,ZSS,DELYSS,DPSS,ELMAXS,YAVRHS RHS
1 GS(10) RHS
1 COMMON /VALOLD/ YOLD(20) RHS
1 COMMON /VARIABLE/ VAL(40) RHS
1 COMMON /WAVE/ ETA(4,11),AW(20),OMEGA(20),DVOLW,NWAVE,BETA,FXWAY,FYRHS
1 WAV,F2WAV,FMWAV,FNWAV,ZBAR,PHIBAR,THEBAR,TC,COSBET,SINBET RHS
1 2 BAR RHS
1 EQUIVALENCE (VAL(1),TIME),(VAL(2),U),(VAL(3),V),(VAL(4),W),(VAL(5),P) RHS
1 L(10),Z),(VAL(6),Q),(VAL(7),R),(VAL(8),PHI),(VAL(9),THETA),(VAL(10),PSI) RHS
1 3,(VAL(24),PB) RHS
1 EQUIVALENCE (VAL(18),FANPWR) RHS
1 DIMENSION ACC(3),ANGACL(3) RHS
1 DATA CCF/4.9E-07/ RHS
1 DO 1 J=1,20 RHS
1 VALUE(J)=0.0 RHS
1 CALCULATION OF BUBBLE WAVE MAKING DRAG RHS
AB = ABW-(ABW-(XL*WIDTH))*(ZS+Z)/BUBHGT RHS

```

```
1 DO VALUE(J=1),20
```

CALCULATION OF BUBBLE WAVE MAKING DRAG
 $AB = ABW - (ABW - (XL*WIDTH)) * (ZS+Z) / BUBHGT$

```

FN = U/FNCON
CF = 37/(FN**1.5655981)
FXPWAV = -PWXCON*PBBAR*CF
WATSLP = -FXPWAV/WEIGHT
VOL = VAL(12)
PB = VAL(13)
PBAR = VAL(13)-P INF
PBS = FE+DPE S
PSS = PB+DPSS
ABPB = PBAR*AB
C      CALCULATION OF BUBBLE WAVE MAKING DRAG
FLOW = SQRT(2.*ABS(PBAR)/RHOINF)*SIGN(1.,PBAR)
GLSW = CFSW*ALSW*FLOW
C      CALL BOWSL
CALL STNSL
C      SUMX = FXBS+FXSS+FXSW+FXRUD+FX P+FXWAV+FX AED+FXPWAV
IF (IT_HFST.NE.ITRM) GO TO 2
THSTS(1) = THSTP(1)-SUMX**5
THSTP(1) = THSTS(1)-SUMX**5
THST = THSTS(1)+THSTP(1)
SUMX = 0.0
2     SUMY = FYBS+FYS +FYRUD+FYP+FYWAV+FY AED
SUMZ = WEIGHT-ABPB+FZBS+FZSS+FZRJD+FZP+FZWAV+FZ AED
SUMK = FKBS+FKSS+FKSW+FKRUD+FKP+FKWAV+FK AED+ABPB*PHI*(-Z)
C      CALCULATION OF EFFECTIVE CENTER OF PRESSURE
UU = U*0.5921-30.0
XCPU = XCP+0*001975*UU*UU-0.974
XCP = SHXYA(XCPU,ZCP,THETA,PI)
FMBUB = ABPB*XCP
SUMM = FMBS+FMSS+FMSW+FMRUD+FM P+FMAED+FBUB+FX PWAV*ZS
FWAVZ = FXPWAV*ZS
SUMN = FNBS+FNSS+FNSW+FN RUD+FNP+FNWAV+FNAED
IF (13DOF.NE.1) GO TO 3
SUMZ = 0.0
SUMM = 0.0
CONTINUE
3     VALUE(1) = SUMX*AMASSI
      VALUE(2) = SUMY*AMASSI-R*U
      VALUE(3) = SUMZ*AMASSI
      VALUE(4) = SUMK*DIZZ+SUMN*DIXZ
      VALUE(5) = SUMM*AIYY
      VALUE(6) = SUMN*DIXX+SUMK*DIZZ

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```

C      VALUE(7) = P
C      VALUE(8) = Q
C      VALUE(9) = W
C      IF (I3DOF.EQ.1) GO TO 6
C
C      BUBBLE PRESSURE EQUATION
C
C      QOUT = QLBS+QLSS+QLSW
C
C      CALL FAN
C      VALUE(10) = RHOINF*(QIN-QOUT)
C      VALUE(11) = -(XL*WIDTH)*.5)*W-DVWDT
C      VALUE(12) = VAL(13)*GAM*((VALUE(10)/VAL(11))-(VALUE(11)/VAL(12)))
C      IF (I1D.EQ.1) GO TO 4
C      GO TO 5
C      4 CONTINUE
C
C      MEMBRANE STUDY
C
C      VAL(12)=VAL(12)+0.3175*VAL(13)
C      VALUE(12)=VAL(13)*GAM*((VALUE(10)/VAL(11))-(VALUE(11)/VAL(12)))
C      5 CONTINUE
C      GO TO 7
C      6 CONTINUE
C      VALUE(10) = 0.0
C      7 CONTINUE
C
C      WRITE DATA FILE FOR MOMENT AND SHEAR CALC'S., IF REQUIRED
C
C      IF (IMT.NE.1) GO TO 8
C      NBS = NSTA(3)-1
C      NSS = NSTA(4)-1
C      NSSL = NSS/2+1
C      WRITE (IBMFIL)(VAL(I),I=1,24),ZBAR,PHIBAR,THEBAR,FYW,FZW,FKW,
C      1,FMW,FMW2,(VALUE(I),I=1,10),DF,DSWA,V,FXH,FZH,FMH,VFY,VVFZ,
C      2,DXRUD,FYRUD,FXP,FZP,FZSS,FXBSS,FXBSS,FXBSS,FXBSS,FXBSS,
C      3,DFMAED,FNAED,FXPWAV,FXSS,FKRS,FKRS,FKRS,FKRS,FKRS,FKRS,
C      4,NSSL,NSS),(TSKIB(I),DFBS(I),ARMIB(I),ARMIB(I)),I=1,NBS)
C      8 CONTINUE
C
C      CONSTANT LONGITUDINAL VFLOCITY ( U )
C
C      IF (ISRG.EQ.1) VALUE(1)=0.0
C      IF (ON.NE.1) RETURN
C
C      DO 9 I=1,3
C      ACCEL(I) = VALUE(I)/G
C      ANGACL(I) = VALUE(I+3)*RAD
C
C      9 CONTINUE

```

9 CONTINUE

C BOWACC = ACCEL(3)-XROW*VALUE(5)/6
 STNACC = ACCEL(3)+XS*VALUE(5)/6
 IF (INVERT.NE.0.N) GO TO 10
 ZD = Z+ZS
 VOLP = VCL
 THEtar = THEta*RAD
 QDEG = Q*RAD
10 IF (ILATRL.NE.ON) GO TO 13
 DEFSI = PSI*RAD
 PDEG = P*RAD
 RDEG = R*RAD
 BETAS = -V*U*RAD
 ACCLAT = (VALUE(2)+U*R)/G
 DPHI = PHI*RAD
 DRFT = 1.2*0*ZD
 VEL = 0.5925*U
 DELRS = RUDANG*RAD
 DEF(R*EQ.0.0) GO TO 11
 TRADIUS = U/R
GO TO 12

11 TRADIUS = 1.E 8
12 WRITEPS (1) TIME, VAL(16), DRFT, THETAR, PBAR, BOWACC, ACCEL(3), FANPWR, THSTS(11), FXPWAV, TPHRHS
13 THSTP(1), QDEG, PDEG, RDEG, DELRS
14 IF (IRHS.NE.ON) RETURN
15 WRITE(6,14) FMBS, FMSS, FMRUD, FMP, FMWAV, FMAED, FMRBUB, FWAVZ
16 WRITE(6,15) FXPWAV
17 WRITE(6,16) PBAR, FANPWR, QIN, QLBS, QLSW, QLSS
18 WRITE(6,17) AE, VOL
19 WRITE(6,18) VALUE, VAL
20 WRITE(6,19) GF, ACCFL, ANGACL
21 WRITE(6,20) BOWACC, STNACC

C RETURN

C 14 FORMAT ('0', 6X, 5HFMBSS=, E16.6, 2X, 5HFMSSE=, F16.6, 2X, 5HFMSW=,
1 E16.6, /, 0, 6X, 6X, 6HFMRUD=, E16.6, 2X, 4HFMP=, E16.6, 2X, 6HFMMWA^V=, E16.6,
2 /, 0, 6X, 6HFMaed=, E16.6, 2X, 6HFBUBB=, E16.6, 2X, 6HFWAVZ=, E16.6),
15 FORMAT ('0', 6X, 7HFXPWA^V=, E16.6)
16 FORMAT ('/10X, 3HHS20HGA^GEPRFSS• (PSF)=F7 • 2.5X, 21HFAN POWER REQD (HRH
1 P)=F8 • 2.5X, 27HFAN FLOW RATE (CU FT/SEC)=F9 • 2 / 31H LEAKAGE FLOW RH
2 RATES ((CU FT/SEC)/11H BOW SEAL =F9.2, 11H SIDEWALL =F9.2, 13H STFRNRHS
3 SEAL =F9.2)
17 FORMAT ('/13H PLENUM AREA=F9.2, 10X, 14HPLNUM VOLUME=F10.2)
18 FORMAT ('/12H VALUE ARRAY2(/10E13.4)/10H VAL ARRAY4(/10E13.4)')

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19 FORMAT ('/10X',24HTOTAL FORCES AND MOMENTS 6E12.4/10X,24HACCELERATIONRHS 2170
20 1$G,DEG/SEC2.6E12.4) 20$H80 ACCEL. (G) =F12.4,21H STERN ACCEL. (G) = E12.RHS 2180
21 FORMAT ('/10X',16HBOW END 2190
22 14) 2200
23 RHS 2210

SUBROUTINE SAM
C      WRITE(6,1)
C      RETURN
C
1 1 FORMAT ('1H1','YOU HAVE CALLED A DUMMY SAM SUBROUTINE. /'
          '110X,CHANGE TO USE THE SAM SUBROUTINE.')
END

SUBROUTINE SIDEWL
C      INTEGER ON
COMMON /AIR/ PINF, RHOINF, GAM
COMMON /BMC0/ IMM, IMNX, IMNY, IBMFIL, BTIME, IMT, XMII(10), YMI(7), IX, IY
COMMON /CONST/ PI, RAD, UO
COMMON /GEOM/ WIDTH, XL, XX(4,11), YY(4,11), NSTA(4), AB, VOLNOM, DELS(4),
110, XCP, ZCP, GEOMSW/ XAVG(10), DS
COMMON /KSWATCH/ ITHRST
COMMON /MASSES/ AM, AIXX, AIYY, AIZZ, AIXZ, AIMAX, G, WEIGHT, RHO, NMASS, AM
1I(201), XI(201), YI(201), ZI(201), XS, ZS, HRHO
COMMON /MSIDW/ DF(2,10), DSWAV(2,10), FZH(2), FYH(2), FZM(2), FNM(2)
1H(2), VFY(2), VFZ(2), FXV
COMMON /PLENUM/ XLBW, XBBW, ABW, RUBHGT, DVW, DOT
COMMON /PRIME/ STIME, FTIME, DELT, PRTNT, PRINT
COMMON /PRTINT/ ON, TACCEL, IVEL, ITRAJ, ISIDLW, IBOWSL, ISTNSL, IWAVES, ISDNL
1 RUD, IPDF, IAEROD, TRHS
COMMON /PROMOD/ PROMO1, PROMO2, PROMO3, PROMO4, PROMO5, PROMO6, PROMO7
COMMON /SIDE/ FX, FY, FZ, FK, FM, FN, ALSW, YSW, XLSW, CFSW, VAREA, VCHO, SDWL
1 RD, VS, PAN, VANGLE, VCOS, VV, VZ, AVBMSW, DFL, X, VTC
COMMON /SLOPE/ WATSLP, XPWV, XLXPWV, PWVHT, XPMVXS
COMMON /VARBLE/ VAL(40)
COMMON /WAVE/ ET(41), AW(20), OMEGA(20), DVOLW, NWAVE, BETA, FXWAV, FYSDWL
1 WAV, FZWAV, FKWAV, FMWAV, ZBAR, PHIBAR, THEBAR, TC, COSBET, SINBET
2BAR
COMMON /WAVTAB/ NAL, DAL, SAL, NDS, DDS, SDS, NTH, DTH, STH, NBB, DBB, SB, ACSDWL
11(20,5,7), AC2(20,5,7), AC3(20,5,7), AC4(20,5,7), AC5(20,5,7), AC6(20,5,7)
12(20,5,7), AC7(20,5,7), AC0(20,5,7), AC00(20,5,7), AC8(20,5,7), AS1(20,5,7)
3S2(20,5,7), AS3(20,5,7), AS4(20,5,7), AS5(20,5,7), AS6(20,5,7), AS7(20,5,7)
45(7), ASO(20,5,7), ASOO(20,5,7), AS8(20,5,7), BB(36), XREF, RX
1 EQUIVALENCE (VAL(12), VAL(3)), V, VAL(4), VAL(5), VAL(6)
1, Q), (VAL(7), R), (VAL(8), PHI), (VAL(9), THETA), (VAL(10), Z), (VAL(11), P)

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211) BMASS : (VAL(21),X), (VAL(22),Y), (VAL(23),PSI), (VAL(24),PB)
DIMENSION GAP(2,1), DSW(2,1)
DIMENSION FZHOLD(2), FZHDRP(2)
DATA ENU/1.28E-5/
DATA PBAR = PB-PINF
DATA PBHEAD = PBAR/(RHO*G)

CROSS-FLOW DRAG ON SIDEWALLS
CONTINUE
6   CO 6   I=1,N
      ALSW = ALSW + (GAP(J,I)+GAP(J,I+1))*DELX/2.
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

LEAKAGE AREA
ALSH = 0.0
CO 6   J=1,N
      N = NSTA(J)-1
      SDWL 640
      SDWL 650
      SDWL 660
      SDWL 670
      SDWL 680
      SDWL 690
      SDWL 700
      SDWL 710
      SDWL 720
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
1   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
      SDWL 490
      SDWL 500
      SDWL 510
      SDWL 520
      SDWL 530
      SDWL 540
      SDWL 550
      SDWL 560
      SDWL 570
      SDWL 580
      SDWL 590
      SDWL 600
      SDWL 610
      SDWL 620
      SDWL 630
      SDWL 640
      SDWL 650
      SDWL 660
      SDWL 670
      SDWL 680
      SDWL 690
      SDWL 700
      SDWL 710
      SDWL 720
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
2   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
      SDWL 490
      SDWL 500
      SDWL 510
      SDWL 520
      SDWL 530
      SDWL 540
      SDWL 550
      SDWL 560
      SDWL 570
      SDWL 580
      SDWL 590
      SDWL 600
      SDWL 610
      SDWL 620
      SDWL 630
      SDWL 640
      SDWL 650
      SDWL 660
      SDWL 670
      SDWL 680
      SDWL 690
      SDWL 700
      SDWL 710
      SDWL 720
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
3   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
      SDWL 490
      SDWL 500
      SDWL 510
      SDWL 520
      SDWL 530
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      SDWL 570
      SDWL 580
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      SDWL 610
      SDWL 620
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      SDWL 640
      SDWL 650
      SDWL 660
      SDWL 670
      SDWL 680
      SDWL 690
      SDWL 700
      SDWL 710
      SDWL 720
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
4   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
      SDWL 490
      SDWL 500
      SDWL 510
      SDWL 520
      SDWL 530
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      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
5   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
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      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

CONTINUE
6   DO 5   J=1,N
      K = NSTA(J)
      SDWL 420
      SDWL 430
      SDWL 440
      SDWL 450
      SDWL 460
      SDWL 470
      SDWL 480
      SDWL 490
      SDWL 500
      SDWL 510
      SDWL 520
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      SDWL 670
      SDWL 680
      SDWL 690
      SDWL 700
      SDWL 710
      SDWL 720
      SDWL 730
      SDWL 740
      SDWL 750
      SDWL 760
      SDWL 770
      SDWL 780
      SDWL 790
      SDWL 800
      SDWL 810

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C DO 7 I=1,N
N = NSTA(I)-1
C
DO 7 J=1,N
DSWAV(I,J) = (DSW(I,J)+DSW(I,J+1))/2.
VREL = V+XAVG(J)*R-(ZS-DSWAV(I,J))/2.*P
DF(I,J) = -HRHO*CDSW*VREL*ABSS(VREL)*DELX*DSWAV(I,J)
FYD = FYD+DF(I,J)
FND = FND+DF(I,J)*XAVG(J)
FKD = FKD-(ZS-DSWAV(I,J))/2.*DF(I,J)
7
C SET UP STERN LIMIT OF FORCE DETERMINATION
XSS = -XS
GO TO 8
ENTRY SIDWLM
XSS = XM1(TX)
8 IP = 1.+((THETA*PAD-STH)/DTH
IP = MAX(MINO(IP,NTH),1)
IP1 = MINO(IP+1,NTH)
DTHETA = ((IP-1)*DTH+STH)
CIP = (THETA*RAD-DTHETA)/DTH
C CALC REYNOLDS NO. AND DRAG COEFF.
PEY = U*XL SW/ENU
CDT = .427/(ALOG10(RFY)-.407)*2.64
C SIDEWALL FORCES, P/S
DC 11 J=1,2
WARFA = 0.0
N = NSTA(J)-1
NI = (XSS+XS)*N/XLSW+1.5
C
DC 9 I=N1,N
ZOR1 = 1.
IF (DSWAV(J,I).EQ.0.0) ZOR1=0.
WARFA = WAREA+DELX*(2.*DSWAV(J,I))+ZOR1*AVBMSW
9
C FXH(J) = -HRHO*CDT*WARFA*U*U
FM1 = 2.*J-3
YLSW = PM1*YSW
DS = Z+ZS+YLSW*PHI
DSS = DS-XSS*THETA

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```

ZOR1 = SIGN(1.,DSS)*ZOR1
DSS = 1.5*(DSS-SBB)/DBB
IDSS = MINO(NBB, IDSS)
BS = BB(IDSS)
ZOR1 = SIGN(1., DSS)+1.)*0.5
DSS = DSS*ZOR1
DRBOW = DSS-(XX(J,N+1)-XSS)*THETA
1F(DRBOW*LT*0.0, DRBOW*0.0
A33S = (RHO*PI*BSS**2)*0.125
A22S = (RHO*4*PI*DSS**2)*0.5
IF(THETA*LT*0.0) A22S=4*RHO*PI*DRBOW*DRBOW*0.5
DSR = DS-(XREF-XS)*THETA
DSR = 1.+(DSR*-SDS)/SDS
ID = MAX(MINO(ID, NDS), 1)
DDSR = (ID-1)*DD+SDS
ID1 = MINO(ID+1, NDS)
DID = (DSR*12.-SDS)/NDS
BC0 = AC0(1, ID, IP)
BC00 = AC0(1, ID, IP)
BC2 = AC2(1, ID, IP)
BC5 = AC5(1, ID, IP)
BC6 = AC6(1, ID, IP)
BC0 = BC0+DID*(AC0(1, ID, IP)-AC0(1, ID, IP)+BC0)-BC0+DID*(AC0(1, ID, IP)-BC0)+DID*(AC0(1, ID, IP)+BC0)
1*ID1, BC00 = BC00+DID*(AC00(1, ID1, IP)-BC00)+DID*(AC00(1, ID1, IP)+BC00)
1*(AC00(1, ID1, IP)-AC00(1, ID1, IP)-AC00(1, ID1, IP)+BC0)
BC2 = BC2+DID*(AC2(1, ID1, IP)-BC2)+DID*(AC2(1, ID1, IP)+BC2)
1*ID1, IP1)-AC2(1, ID1, IP)-AC2(1, ID1, IP)+BC2)
1*BC5 = BC5+DID*(AC5(1, ID1, IP)-BC5)+DID*(AC5(1, ID1, IP)+RC5)
1*ID1, IP1)-AC5(1, ID1, IP)-AC5(1, ID1, IP)+RC5)
BC6 = BC6+DID*(AC6(1, ID1, IP)-BC6)+DID*(AC6(1, ID1, IP)+BC6)
1, ID1, IP1)-AC6(1, ID1, IP)-AC6(1, ID1, IP)+BC6)

SHIFT MOMENT CENTER FRCM XREF TO C.G.
BC00 = BC00-(XS-XREF)*BC0
BC6 = BC6-(XS-XREF)*BC5

HYDROSTATIC AND HYDRODYNAMIC FORCES
1*P*YLSW = -G*BC0-U*A33S*THETA-U*A33S*Q+G*BC0+U*(-BC2+A33S*XSS)-U*A33S*XSS
1*P) FMH(J) = -U*XSS*XSS*A33S*Q+G*BC0+U*(A33S*XSS+BC2)*(W+U*THETA+YL SW)
1*P) FYH(J) = -A22S*U*(V+XSS*R-ZS*S)
FNH(J) = FYH(J)*XSS-U*((V-ZS*S)*BC5+R*BC6)

```

C ADD VERTICAL FORCE DUE TO DEADRISE PROJECTION OF LATERAL FORCE
 C
 CTNDR = 0.0
 IF (DSS .LE. 0.0) GO TO 10
 CTNDR = (BS-BB(1))/DSS
 IF (THETA.LT.0.0) CTNDR=.39391
 10 CONTINUE
 FZHOLD(J) = FZH(J)
 FZHDRP(J) = PM1*FYH(J)*CTNDR*PROM01
 FZH(J) = FZH(J)+FZHDPP(J)
 IF (IMT.EQ.2) GO TO 11
 C CALC OF FORCE ON VENTRAL FINS REMOVED
 C
 11 CONTINUE
 C IF (IMT.EQ.2) GO TO 12
 C TOTAL SIDEWALL FORCES AND MOMENTS
 FX = FXH(1)+FXH(2)
 FY = FYH(1)+FYH(2)
 FZ = FZH(1)+FZH(2)
 FK = (FZH(2)-FZH(1))*YSW+FKD-FY*ZS
 FY = FY+FYD
 FM = FMH(1)+FMH(2)+Z*FX
 FN = FND+FNH(1)+FNH(2)+(FXH(1)-FXH(2))*YSW
 C DRAG FORCE ON FINS REMOVED
 C
 12 CONTINUE
 C ADD ROLL DAMPING DUE TO VERTICAL WAVE GENERATION
 C
 DSS = Z+ZS-XSS*THETA
 ZOR1 = (SIGN(1.,DSS)+1.)/2.
 DSS = DSS*ZOR1
 DS = Z+ZS
 DSR = DS-(XREF-XS)*THEFTA
 ID = 1+(DSR*12.-SDS)/DDS
 ID = MAX0(MIN0(ID,ND\$),1)
 DDSR = (ID-1)*DD\$+SD\$
 ID1 = MIN0(ID+1,ND\$)
 DID = (DSR*(ID-12.-DDSR))/DD\$
 BC2 = AC2(1,IP)
 BC2 = BC2+DI**AC2(1,IDL,IP)-BC2+DI**AC2(1,IDL,IP)+BC2
 1 ID1, IP1)-AC2(1, ID, IP1)-AC2(1, ID1, IP)+BC2
 FKCLD = FK

```

FK = FK - PROM02*YSW*BC2*P/P
FZH(1) = FZH(1) + PROM02/2.*YSW*BC2*P/P
FZH(2) = FZH(2) - PROM02/2.*YSW*BC2*P/P
IF (PROM03.EQ.1.0) WRITE (6,15) VAL(1),FZHOLD(1),FZHOLD(2),FZHOLD(1)
IF (FZHCRP(2).EQ.1.0) WRITE (6,15) VAL(1),FZH(1),FKOLD,FK
IF (ISIDWL.NE.ON) RETURN

DO 13 I=1,2
C      DO 13 J=1,11
C      GAP(I,J) = 12.0*GAP(I,J)
13 DSW(I,J) = 12.0*DSW(I,J)

C      WRITE (6,16) ((GAP(I,J), J=1,11), I=1,2), ((DSW(I,J), J=1,11), I=1,2), F
C      1X,FY,FZ,FK,FM,FN

C      RETURN

C      14 FCORMAT (/10X,43HWATER CONTACT WITH TOP OF BUBBLE CHAMBER AT F7•2, 14
C      1H FT TIME=F7•2,19H SEC•T TIME=F7•2,4H FT•)
C      15 FCORMAT (/2X,TIME•,1X,E15•4•2X,•OLD VERTICAL FORCES•,2(5X,E15•4•)/
C      1 25X,E15•4•2(5X,E15•4•)/25X,•NEW VERTICAL FORCES•,2(5X,E15•4•)/
C      2 ESS•,2(5X,E15•4•)/25X,•OLD AND NEW ROLL MOMENTS•,2(5X,E15•4•)/
C      16 FORMAT (/10X,8HSIDEWALL/25H GAP (FT.) (STERN TO BOW)/14H PORT SIDEWALL/11F10.2
C      1 WALL/11F10.5/14H STBD SIDEWALL/11F10.5/37H IMMERSION DEPTH (FT.) (SDWL 2510
C      2 STERN TO BOW)/14H PORT SIDEWALL/11F10.5/14H STBD SIDEWALL/11F10.5/
C      3 10X,26HSIDEWALL FX,FY,FZ,FK,FM,FN/6E15.4)
C      END

FUNCTION SHXYAX (X,Z,ANGYAX,PI)
H = SQRT (X*X+Z*Z )
IF (X.EQ.0.0) GO TO 1
ARG = Z/X
ANGOLD = ATAN(ARG)
IF (ANGOLD.GE.0.0) GO TO 2
ANGNEW = ANGOLD+PI-ANGYAX
GO TO 3
1 ANGNEW = PI*.50-ANGYAX
2 ANGNEW = ANGOLD-ANGYAX
3 SHXYAX = H*COS(ANGNEW)
RETURN
END

SUBROUTINE STNSL

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ELSKIL(J) = 0.0
AIRTLEN(J) = 0.0
CONTINUE

*EFDEP IS THE EFFECTIVE LIFTING DEPTH OF THE STERN SEAL

```
EFDEP = 6.0
MM = EFDEP
ALSS = 0.0
FX = 0.0
FZ = 0.0
FK = 0.0
FM = 0.0
FN = 0.0
AGAP1 = 0.0
AGAP2 = 0.0
AGAPA1 = 0.0
DEL_P = PSS-PB
TF(DEL_P*LT*0.0) DEL_P=0.0
PBAR = PB-PINF

CALCULATE ELSKI HERE.

SINDIF = SINTH-COSTH*THETA
COSDIF = COSTH+SINTH*THETA
X1 = XSS+ZSS*THETA-XLF*SINDIF
Z1 = (-Z-ZSS+XSS*THETA-ELMAXS*COS(THETA))

CALCULATE GAP HERE.

N = NSTA(4)

FC2,K=1,N = (ETA(4,K)-DETADX(K)*(XX(4,K)-X1)-Z1)+YY(4,K)*PHI-XPW*W
1ATSLP
IF(ELSKIL(K)*GT*HIGHT) ELSKIL(K)=HIGHT
IF(ELSKIL(K)=ELSKIL(K)+GPS
IF(ELSKIL(K)*GT*HIGHT) ELSKIL(K)=HIGHT
IF(ELSKIL(K)*LT*(HIGHT-ELMAXS)) ELSKIL(K)=HIGHT-ELMAXS
GAP(K)=-ELSKIL(K)+(HIGHT-ELMAXS)
IF(GAP(K)*LT*0.0) GAP(K)=0.0
MM1 = ELSKIL(K)*12.0
MM2 = MM1+1
MM3 = MM2+1
DLINC = ELSKIL(K)*12.0-MM1
STNSL1 = CTNSL(MM,MM2)
STNSL2 = CTNSL(MM,MM3)
```

```

AIRLEN(K) = ((STNSL2-STNSL1)*DLINC+STNSL1)/12.0
2 CCONTINUE
C N = NSTA(4)-1
C DO 5 J=1,N
ELS1 = (ELSKI(J+1)+ELSKI(J))*0.5
ELS2 = (ELSKI(J+1)+ELSKI(J))*0.5
AIRLEN = (AIRLEN(J+1)+AIRLEN(J))*0.5
AGAP = ELSKIA-ELSKIA
AGAP1 = AGAP
IF ((AGAP.LT.GPS) AGAP=GPS
IF ((AGAP1.GT.GPS) AGAP1=GPS
ARM1S(J) = X*(4-J)+ELSKIA*0.5
ARM2S(J) = Z-ELSKIA
DFSS(J) = -DELP*DELYSS*AIRLAV/(GPS/AGAP) **2
IF ((AIRLAV.LE.0.0) GO TO 3
ARG = 5*RHO*U*AIRLAV**DELYSS
RESKI = U*AIRLAY/ENJ
CDTSK1 = 427/(LOG10(RESKI)-.407)**2.64
CDTSK1(J) = -ARG*CDTSK1
C THE FOLLOWING CARD REMOVES WATER DRAG EFFECTS OF STERN SEAL
C TSKIS(J) = 0.0
GO TO 4
3 TSKIS(J) = 0.0
4 CCONTINUE
FX = FX+TSKIS(J)
FZ = FZ+DFSS(J)
FK = FK+DFSS(J)*YAVGS(J)
FM = FM-DFSS(J)*ARM1S(J)+TSKIS(J)*ARM2S(J)
FN = FN-TSKIS(J)*YAVGS(J)
ALSS = ALSS+(GAP(J)+GAP(J+1))*DELYSS*0.5
AGAP2 = AGAP2+AGAP1
AGAP1 = AGAP2/J
5 CCONTINUE
C ALSS = ALSS+ALEAK*(AGAP1/GPS)
SQFAC = SQRT(2.*ABS(PBAR)/RHOINF)
QL = CFSS*ALSS*SQFAC*SIGN(1.,PBAR)
IF (ISTNSL.NE.ON) RETURN
WRITE (6,6) GAP,AIRLEN,FX,FY,FZ,FK,FM,FN
C RETURN
C 6 FORMAT //12H STERN SEAL/26H GAP (FT.) PORT TO STBD. /11E11.3/28SS
C

```



```

3 S2(20,5,7) AS3(20,5,7) AS4(20,5,7) AS5(20,5,7) AS6(20,5,7) AS7(20,5,7)
4 S3(20,5,7) AS0(20,5,7) AS0(20,5,7) AS8(20,5,7) BB(36) XREF RX
5 DIMENSION WC0(2), WC0(2), WC1(2), WC2(2), WC3(2), WC4(2), WC5(2),
1 WC6(2), WC7(2), WC8(2)
1 DIMENSION WS0(2), WS0(2), WS1(2), WS2(2), WS3(2), WS4(2), WS5(2),
1 WS6(2), WS7(2), WS8(2)
1 EQUIVALENCE (VAL(2),U), (VAL(3),V), (VAL(4),W), (VAL(5),P)
1 ) Q) (VAL(7),R), (VAL(8),PHI), (VAL(9),THETA), (VAL(10),Z),
2 ) B(MASS), (VAL(21),X), (VAL(22),Y), (VAL(23),PSI), (VAL(24),PB)
2 EQUIVALENCE (VAL(16),ETACG)

C IF (NWAVE.EQ.0) RETURN

C CALCULATION OF SHIFT OF XC PO

XCPU = XCPO+0.001975*(U*0.5921-30.0)**2-0.974
XCP = SHXYAX(XCPU,ZCP,TETA,PI)
GAMMA = BETA-PSI
SIGAM = SIN(GAMMA)
COGAM = COS(GAMMA)
FO = -X*COSRET-Y*SINRET
DVOLW = 0.0
FTACG = 0.0
N = NSTA(3)

DO 1 J=1,N
1 DETABX(J) = 0.0
CONTINUE
N = NSTA(4)
DO 2 J=1,N
2 DETADX(J) = 0.0
CONTINUE
DO 3 K=1,N
3 DETA(J,K) = 0.0
CONTINUE
DO 4 J=1,2
4 FXW(J) = 0.0
FYW(J) = 0.0
FZW(J) = 0.0
FKW(J) = 0.0

```

$F_{MW}(J) = 0.0$
 $F_{NW}(J) = 0.0$

4

CONTINUE

C XSS = -XS
IF (IMT*EQ*2) XSS = XM1(IX)
IP = 1+(THEBAR*RAD-STH)/DTH
IP = MAXO(MINO(IP,NTH),1)
IP1 = MINO(IP+1,NTH)
DTHETA = (IP-1)*DTH+STH
DIP = (THETA*RAD-DTHETA)/DTH
TIME_RISE_FACTOR FOR WAVE AMPLITUDE
AMPFAC = 1.-EXP(-TIME/AMPTC)

DO 11 I=1,NWAVE
OM1 = CMFGAT(I)
OM2 = CM1*CM1
XWK = CM2/G
AA = AW(1)*AMPFAC
FT = CM1*TIME*XWK*FO
AL = XWK*COGAM
IAA = 1+(ABS(AL)-SAL)/DAL
IAA = MAXO(MINO(IAA,NAL),1)
IAA1 = MINO(IAA+1,NAL)
DAA = (IAA-1)*DAL+SAL
DIA = (ABS(AL)-DAA)/DAL
SALP = SIGN(1.,AL)

WAVE FORCES AND MOMENTS ON THE SIDEWALLS

DO 7 J=1,2
YLSW = (2*J-3)*YSW
WE = FT+XWK*SIGAM*YLSW
ST = SIN(WE)
CT = COS(WE)
DSR = ZBAR+ZS+YLSW*PHIBAR
ID = DS-(XREF-XS)*THEBAR
ID = 1.+(DSF*1.2-SDS)/DDS
ID = MAXO(MINO(ID,NDI),1)
DDSR = (ID-1)*DDS+SDS
CID = (DSR*1.2-DDSR)/DDS
ID1 = MINO(ID+1,NDI)
DSS = DS-XSS*THEBAR
ZORI = (SIGN(1.,DSS)+1.)*0.5
DSS = DSS*ZORI
IDSS = 1.5+(DSS-SBBI)/DBB
IDSS = MINO(NBB,NDI)

CCC

$BS = BB(1DSS)$
 $CK = CCS(XWK*COGAM**SS)$
 $A33S = (RHO*PI*BSS)*0.125$
 $SK = SIN(XWK*COGAM**SS)$
 $A22S = (RHO*.4*PI*DSS)*0.5$
 $A42S = 0.0$

INTERPOLATION OF WAVE TABLES

CCC

$K = \frac{1}{16}AA$
 $L = \frac{1}{16}A$
 $COUNTINUF(L, ID, IP)$
 $BC00 = AC00(L, ID, IP)$
 $BC1 = AC1(L, ID, IP)$
 $BC2 = AC2(L, ID, IP)$
 $BC3 = AC3(L, ID, IP)$
 $BC4 = AC4(L, ID, IP)$
 $BC5 = AC5(L, ID, IP)$
 $BC6 = AC6(L, ID, IP)$
 $BC7 = AC7(L, ID, IP)$
 $BC8 = AC8(L, ID, IP)$
 $BS00 = AS00(L, ID, IP)$
 $BS01 = AS1(L, ID, IP)$
 $BS2 = AS2(L, ID, IP)$
 $BS3 = AS3(L, ID, IP)$
 $BS4 = AS4(L, ID, IP)$
 $BS5 = AS5(L, ID, IP)$
 $BS6 = AS6(L, ID, IP)$
 $BS7 = AS7(L, ID, IP)$
 $BS8 = AS8(L, ID, IP)$
 $WC00(K) = BC00 + DID*(AC00(L, ID, IP) - AC0(L, ID, IP)) + DIP*(AC0(L, ID, IP) - BC0(L, ID, IP)) + BC0(L, ID, IP)$
 $1 DID*(AC00(L, ID, IP) - AC0(L, ID, IP)) + DIP*(AC00(L, ID, IP) - BC00(L, ID, IP)) + BC00(L, ID, IP)$
 $WC1(K) = BC1 + DID*(AC1(L, ID, IP) - BC1(L, ID, IP)) + DIP*(AC1(L, ID, IP) - BC1(L, ID, IP)) + BC1(L, ID, IP)$
 $1C1(L, ID, IP) - AC1(L, ID, IP) + BC1(L, ID, IP)$
 $WC2(K) = BC2 + DID*(AC2(L, ID, IP) - BC2(L, ID, IP)) + DIP*(AC2(L, ID, IP) - BC2(L, ID, IP)) + BC2(L, ID, IP)$
 $1C2(L, ID, IP) - AC2(L, ID, IP) + BC2(L, ID, IP)$
 $WC3(K) = BC3 + DID*(AC3(L, ID, IP) - BC3(L, ID, IP)) + DIP*(AC3(L, ID, IP) - BC3(L, ID, IP)) + BC3(L, ID, IP)$
 $1C3(L, ID, IP) - AC3(L, ID, IP) + BC3(L, ID, IP)$
 $WC4(K) = BC4 + DID*(AC4(L, ID, IP) - BC4(L, ID, IP)) + DIP*(AC4(L, ID, IP) - BC4(L, ID, IP)) + BC4(L, ID, IP)$
 $1C4(L, ID, IP) - AC4(L, ID, IP) + BC4(L, ID, IP)$
 $WC5(K) = BC5 + DID*(AC5(L, ID, IP) - BC5(L, ID, IP)) + DIP*(AC5(L, ID, IP) - BC5(L, ID, IP)) + BC5(L, ID, IP)$
 $1C5(L, ID, IP) - AC5(L, ID, IP) + BC5(L, ID, IP)$
 $WC6(K) = BC6 + DID*(AC6(L, ID, IP) - BC6(L, ID, IP)) + DIP*(AC6(L, ID, IP) + BC6(L, ID, IP)) + BC6(L, ID, IP)$
 $1C6(L, ID, IP) - AC6(L, ID, IP) + BC6(L, ID, IP)$

9

C SHIFT MOMENT CENTER FROM XREF TO C.G.

```

BC00 = BC00-(XS-XREF)*BC0
BC3 = BC3-(XS-XREF)*BC1
BC4 = BC4-(XS-XREF)*BC2
BC6 = BC6-(XS-XREF)*BC5
BS00 = BS00-(XS-XREF)*BS0
BS3 = BS3-(XS-XREF)*BS1
BS4 = BS4-(XS-XREF)*BS2
BS6 = BS6-(XS-XREF)*BS5

```

C CALCULATE WAVE FORCES AND MOMENTS

```

FZC = BS1-XWK*G*(BS2+BS0)-U*OM1*(-A33S*CK-AL*BS2)
FZS = BC2+BC0)+U*OM1*(-A33S*SK+AL*RC2)
FMC = BS3-XWK*G*(BS4+BS0)-U*OM1*(-A33S*XS*CK-BC2-AL*BS4)
FMS = BC3-XWK*G*(BC4+PC00)+U*OM1*(-A33S*XS*SK-BS2+AL*BC4)
FYC = XWK*G*(BC5+BC0)-U*OM1*(-A22S*SK+AL*BC5)
FYS = -XWK*G*(BS5+BS0)-U*OM1*(-A22S*CK-AL*BS5)
FNC = XWK*G*(BC6+BC0)-U*OM1*(-A22S*XS*SK-BS5+AL*BC6)
FNS = -XWK*G*(BS6+BS0)-U*OM1*(-A22S*XS*CK-BC5-AL*BS6)
FKC = XWK*G*(BC7-B(C8)+U*OM1*(-A2S*SK+AL*BC8)
FKS = -XWK*G*(BS7-BS8)+U*OM1*(-A42S*CK-AL*BS8)
FZW(J) = FZU(J)-AA*(FZC*CT+FZS*ST)
FNW(J) = FNU(J)+AA*(FMC*CT+FM*S*ST)**SIGAM
FYW(J) = FYU(J)-AA*(FYC*CT+FYS*ST)**SIGAM
FNW(J) = FNU(J)-AA*(FNC*CT+FN*S*ST)**SIGAM
FKW(J) = FKU(J)-AA*(FKC*CT+FKS*ST)**SIGAM
FXW(J) = FXU(J)-2.*AA*RHO*G*BS*DSS*SK*CT

```

7 CONTINUE

IF (IMT.EQ.2) GO TO 11

C WAVE ELEVATION AROUND THE SIDEWALLS AND SEALS

```

DO 8 J=1,4
N = NSTA(J)

```

```

DO 8 K=1,N
ETA(J,K) = ETA(J,K)+SIN(XWK*(-XX(J,K)*CDGAM-YY(J,K)*SIGAM)+FT)*AA
8 CONTINUE
ETACG = ETACG+AA*SIN(FT)
N = NSTA(3)

```

```

DO 9 J=1,N

```

| | |
|------------|--|
| WA VS 2190 | |
| WA VS 2200 | |
| WA VS 2210 | |
| WA VS 2220 | |
| WA VS 2230 | |
| WA VS 2240 | |
| WA VS 2250 | |
| WA VS 2260 | |
| WA VS 2270 | |
| WA VS 2280 | |
| WA VS 2290 | |
| WA VS 2300 | |
| WA VS 2310 | |
| WA VS 2320 | |
| WA VS 2330 | |
| WA VS 2340 | |
| WA VS 2350 | |
| WA VS 2360 | |
| WA VS 2370 | |
| WA VS 2380 | |
| WA VS 2390 | |
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| WA VS 2490 | |
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| WA VS 2610 | |
| WA VS 2620 | |
| WA VS 2630 | |
| WA VS 2640 | |
| WA VS 2650 | |
| WA VS 2660 | |

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