

AD-A019 538

**A COMPUTER CODE FOR FULLY-COUPLED ROCKET NOZZLE FLOWS
(FULLNOZ)**

H. S. Pergament, et al

AeroChem Research Laboratories, Incorporated

Prepared for:

Air Force Office of Scientific Research

April 1975

DISTRIBUTED BY:

NTIS

**National Technical Information Service
U. S. DEPARTMENT OF COMMERCE**

AFOSR - TR - 75 - 1568

AFOSR Scientific Report

AFOSR-TR-

AeroChem TP-322

ADA019538

023123

**A COMPUTER CODE FOR FULLY-COUPLED
ROCKET NOZZLE FLOWS (FULLNOZ)**

**H. S. Pergament
R. D. Thorpe**

**AeroChem Research Laboratories, Inc.
Princeton, New Jersey 08540**

April 1975

**AIR FORCE OFFICE OF SCIENTIFIC RESEARCH (AFOSR)
OFFICE OF OPERATIONAL AND TO DO
This technical report has been reviewed and is
approved for public release under FAR 190-12 (7b).
Distribution is unlimited.
A. B. BROWN
Technical Information Officer**

**Approved for Public Release;
Distribution Unlimited**

**DDC
RECEIVED
JAN 22 1976
REGISTRY
D**

Prepared for

AIR FORCE OFFICE OF SCIENTIFIC RESEARCH

Arlington, Virginia 22209

**Reproduced by
NATIONAL TECHNICAL
INFORMATION SERVICE
U S Department of Commerce
Springfield VA 22151**

**Copy available to DDC does not
permit fully legible reproduction**

20

**Best
Available
Copy**

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER AFOSR - TR - 75 - 1563	GOVT ACCESSION NO.	2. RECIPIENT'S CATALOG NUMBER
3. TITLE (and Subtitle) A COMPUTER CODE FOR FULLY-COUPLED ROCKET NOZZLE FLOWS (FULLNOZ)		4. TYPE OF REPORT & PERIOD COVERED INTERIM 1 Sept 73 - 30 Apr 75
7. AUTHOR(s) H.S. PERGAMENT R.D. THORPE		6. PERFORMING ORG. REPORT NUMBER TP-322
9. PERFORMING ORGANIZATION NAME AND ADDRESS AERO-CHEM RESEARCH LABORATORIES, INC. PO BOX 12 PRINCETON, NEW JERSEY 08540		8. CONTRACT OR GRANT NUMBER(s) F44620-74-C-0006
11. CONTROLLING OFFICE NAME AND ADDRESS AIR FORCE OFFICE OF SCIENTIFIC RESEARCH/NA 1400 WILSON BOULEVARD ARLINGTON, VIRGINIA 22209		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS 681308 9711-01 611025
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		12. REPORT DATE April 1975
		13. NUMBER OF PAGES 120
		15. SECURITY CLASS (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report) DDC DECLASSIFIED JAN 22 1978 REGISTRY D		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) ROCKET NOZZLE FLOWS COMPUTER CODES CHEMICALLY REACTING FLOWS GAS/PARTICLE NONEQUILIBRIUM FLOWS		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) A comprehensive computer code (FULLNOZ) has been developed to perform detailed calculations of rocket nozzle flows downstream of the sonic line. The code uses the streamtube method to integrate the hyperbolic governing equations of steady supersonic flow. The program represents a significant advance in nozzle flow predictions through the coupling of gas/particle nonequilibrium effects, non-equilibrium chemistry, turbulent boundary layers (to determine the effects of wall heat transfer and shear stress) and turbulent mixing across streamtubes.		

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

This report describes the analytical and numerical techniques employed by the code, presents results of a sample calculation for the Minuteman Stage 2 nozzle and gives complete instructions on the preparation of input data and a full FORTRAN listing.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

Conditions of Reproduction

Reproduction, translation, publication, use and disposal in whole or in part by or for the United States Government is permitted.

Qualified requestors may obtain additional copies from the Defense Documentation Center, all others should apply to the National Technical Information Service.

SUMMARY

This report describes the analytical and numerical techniques utilized in the development of a fully-coupled rocket nozzle flow computer code (FULLNOZ). The code uses the streamtube method to integrate the governing equations of steady supersonic flow. The elliptic Navier-Stokes equations are reduced to hyperbolic form by assuming diffusional effects along streamlines are small compared to those across streamlines. Finite difference techniques are then used to solve the hyperbolic equations along and perpendicular to streamlines.

FULLNOZ represents a significant advance in nozzle flow calculations by coupling the effects of nonequilibrium chemistry, gas/particle thermal and dynamic nonequilibrium, turbulent mixing across streamtubes and turbulent boundary layers. Turbulent mixing is treated via a phenomenological eddy viscosity model, while the turbulent boundary layer analysis utilizes the experimental data of Keener and Hopkins (which relates the compressible skin friction coefficient to measured velocity/temperature profiles in flows with favorable pressure gradients), the Van Driest transformations and the momentum integral equation. The operation of FULLNOZ requires the specification of initial gas and particle properties just downstream of the sonic line, the nozzle wall contour and temperature, and a chemical reaction mechanism and rate coefficients. The marching scheme proceeds downstream computing flow properties and composition along surfaces orthogonal to a specified number of streamtubes. A mixed explicit/implicit differencing scheme is used to obtain the most favorable integration step size.

Sample calculations are presented for the Minuteman Stage 2 nozzle. In addition, this report describes the preparation of input data and gives a full FORTRAN listing of the program. Also included is an analysis of heterogeneous electron/ion recombination on particles, although it has not as yet been incorporated into the code.

ACCESSION for	
NTIS	White Section <input checked="" type="checkbox"/>
DDC	Buff Section <input type="checkbox"/>
UNANNOUNCED	<input type="checkbox"/>
JUSTIFICATION	
.....	
BY	
DISTRIBUTION AVAILABILITY CODES	
DATE OF SPECIAL	
A 23	

D D C
RECEIVED
JAN 22 1974
RECEIVED
D

Copy available to DDC does not permit fully legible reproduction

PREFACE

This is a report on the work performed on Contract F44620-74-C-0006 covering the period 1 September 1973 to 30 April 1975. It is basically a program user's manual for FULLNOZ, although some additional results obtained during this period, e. g. a preliminary study of how to incorporate heterogeneous electron-ion recombination into the code, have also been included.

The authors would like to thank J. T. Kelly for his initial work in the development of FULLNOZ and Capt. L. R. Lawrence, AFOSR Program Manager, for his interest, encouragement and support during the development of FULLNOZ.

This code is available for public use and may be obtained by forwarding a request and a tape to Lt. Robert Sperlein, DYSP, at the Air Force Rocket Propulsion Laboratory, Edwards Air Force Base, CA 93523.

This scientific report has been reviewed and is approved.

LIST OF SYMBOLS

A	streamtube area; also pre-exponential term in rate coefficient equations
$a_{i, jj}$	enthalpy-temperature polynomial coefficients, see Eq. (20)
B	activation energy
C_D	particle drag coefficient
\bar{C}_D	normalized drag coefficient, see Eq. (34)
C_{DFM}	free molecular drag coefficient
C_{DI}	high Reynolds number drag coefficient, see Eq. (35)
C_i	mass fraction of i th species
C_F	skin friction coefficient, see Eqs. (54) and (59)
C_p	gas specific heat
C_s	particle specific heat
D	diffusion flux, defined by Eq. (23)
\bar{D}	diffusion coefficient
F_i	defined as C_i/M_i
f_p	ratio of actual particle drag coefficient to the drag coefficient for Stokes flow
g_p	ratio of actual particle heat transfer coefficient to heat transfer coefficient for Stokes flow
H	gas stagnation enthalpy
H_{12}	shape factor, defined by Eq. (69)
H_f°	heat of formation at 298°K
h	gas static enthalpy
h'	particle/gas heat transfer coefficient
I	total number of gas species
J	total number of particle groups
K_p	equilibrium constant
k	thermal (eddy) conductivity
k_f	forward reaction rate coefficient
k_g	molecular thermal conductivity in particle/gas interaction terms

L	total number of species on left or right side of reaction
L₁-L₃	specific heat polynomial coefficients
L₆	enthalpy constant of integration, see Eq. (31)
L₇	entropy constant of integration, see Eq. (32)
Le	turbulent Lewis number, $Le = \frac{C \bar{D}\rho}{k}$
M	Mach number
M_i	molecular weight of ith species
m	streamtube mass flow, defined by Eq. (13)
N	number of reactions
Nu	Nusselt number, $Nu = \frac{h' r}{k_g}$
n	distance normal to streamline, also exponent in Eqs. (57) and (58)
Pr	turbulent Prandtl number, $Pr = \frac{\mu C}{k}$
Pr_g	laminar Prandtl number
p	static pressure
q	heat flux, defined by Eq. (22)
q_w	wall heat transfer, defined by Eq. (62)
R	universal gas constant, also value of nozzle radius at each x
Re	Reynolds number based on streamtube width
Re_p	particle Reynolds number
Re_x	Reynolds number based on distance along nozzle wall
Re_θ	Reynolds number based on boundary layer momentum thickness
r	radial distance from axis; also recovery factor, Eq. (49)
r_p	particle radius
s	distance along streamline
S	entropy
S_t	Stanton number, defined by Eq. (62)
T	static temperature
T_r	recovery temperature

u	velocity along streamline
U	velocity in boundary layer
U_T	friction velocity, defined by Eq. (63)
V_p	particle velocity normal to gas streamline
\dot{W}_i	production rate of i th species
x	axial distance from nozzle starting line
y	distance from nozzle wall
Y	molar concentration

Greek Letters

α	numerical stability coefficient for marching scheme, see Eq. (29)
γ	ratio of specific heats
ΔG	Gibbs free energy
$\Delta_k(\Phi)$	change in Φ across streamtube
δ	boundary layer thickness
δ^*	displacement thickness
δn	finite difference mesh spacing in n direction
δs	finite difference mesh spacing in s direction
Θ	angle between streamline and plume axis; also boundary layer momentum thickness
$\Theta_s, \Theta_s', \Theta_s''$	respectively; final, initial and intermediate streamline radii of curvature
Φ	$T/1000$, used in Eq. (30)
ν	kinematic viscosity
ν'	stoichiometric coefficient on left hand side of reaction
ν''	stoichiometric coefficient on right hand side of reaction
μ	eddy viscosity
μ_g	molecular viscosity in particle/gas interaction terms
ρ	gas density
ρ_p	particle cloud density
ρ_s	density of liquid or solid particle
τ	shear stress, defined by Eq. (21) and Eq. (60)
\cdot	multiplication

Σ summation

Subscripts

aw adiabatic wall
c compressible
e wall streamtube
FM free molecular
g gas
i ith species; also, incompressible
j particle group identification index
k streamtube index
l orthogonal surface index; also refers to species on left or right hand side of reaction in Eq. (6)
m mth reaction
n differentiation in direction normal to streamlines
o stagnation value
p particle
s differentiation in streamline direction
t total
w wall

TABLE OF CONTENTS

	<u>Page</u>
LIST OF SYMBOLS	3
I. INTRODUCTION	9
II. GOVERNING EQUATIONS	13
A. Differential Equations	13
1. Gas Phase	13
2. Particles	15
B. Finite-Difference Equations	17
1. Gas Phase	17
2. Particles	20
3. Integration Step Size	22
4. Solution Procedure	22
C. Auxiliary Calculations	23
1. Thermodynamic Input Data	23
2. Chemical Kinetic Input Data	24
3. Particle/Gas Drag and Heat Transfer Coefficients	26
D. Turbulent Boundary Layer Equations	28
III. PREPARATION OF INPUT DATA	32
IV. PRELIMINARY RESULTS	47
A. Sample Calculations	48
B. Heterogeneous Electron Ion Recombination	54
V. REFERENCES	56

LIST OF ILLUSTRATIONS

<u>Figure</u>		
1	FULLNOZ SCHEMATIC	11
2	INPUT DATA DEFINITIONS	33

LIST OF ILLUSTRATIONS (continued)

<u>Figure</u>		<u>Page</u>
3	COMPARISON BETWEEN METHOD OF CHARACTERISTICS (MOC) AND FULLNOZ CALCULATIONS OF MM-STAGE 2 NOZZLE PRESSURE DISTRIBUTIONS	47
4	COMPARISON BETWEEN METHOD OF CHARACTERISTICS (MOC) AND FULLNOZ CALCULATIONS OF MM-STAGE 2 NOZZLE GAS TEMPERATURE DISTRIBUTIONS	48
5	INFLUENCE OF NOZZLE BOUNDARY LAYER ON WALL STREAMTUBE TEMPERATURE	49
6	INFLUENCE OF NOZZLE BOUNDARY LAYER ON WALL STREAMTUBE VELOCITY	50
7	BOUNDARY LAYER PROFILES AT MM-STAGE 2 NOZZLE EXIT PLANE	51
8	INFLUENCE OF PARTICLE DRAG AND HEAT TRANSFER COEFFICIENTS ON EXIT PLANE GAS PROPERTIES	52
9	INFLUENCE OF PARTICLE DRAG AND HEAT TRANSFER COEFFICIENTS ON EXIT PLANE PARTICLE TEMPERATURES	53

LIST OF TABLES

<u>Table</u>		
1	NOZZLE EXIT PLANE BOUNDARY LAYER PARAMETERS	51
	APPENDIX A: SAMPLE INPUT DATA	
	APPENDIX B: SAMPLE OUTPUT	
	APPENDIX C: SAMPLE OUTPUT - EXPLANATIONS	
	APPENDIX D: FORTRAN LISTING	

I. INTRODUCTION

This report describes a new fully-coupled rocket nozzle code (FULLNOZ) which treats, simultaneously, gas/particle nonequilibrium, nonequilibrium chemistry, diffusion across streamlines, and turbulent boundary layers within axisymmetric and two-dimensional nozzles. Because of its fully-coupled capability FULLNOZ can be applied to a wider range of problems than such existing rocket nozzle codes,[†] as TDK¹ (which does not account for gas/particle nonequilibrium effects), the nozzle portion of the CONTAM² code, the original constant γ gas/particle nonequilibrium code of Nickerson and Kliegel³ and the recently-developed gas/particle and chemical nonequilibrium code developed at Lockheed/Huntsville.⁴ The emphasis in the present code has been on an accurate calculation of nozzle exit plane gas and particle properties (particularly major and minor neutral and charged species concentrations), rather than on the determination of specific impulse, although FULLNOZ is well equipped to calculate I_{sp} .

[†] e.g. none of these codes calculate turbulent boundary layers.

1. "ICRPG Two-Dimensional Kinetic (TDK) Nozzle Analysis Computer Program," Dynamic Science Corp., December 1973 (revised version).
2. Hoffman, R.J., English, W.D., Oeding, R.G., and Webber, W.T., "Plume Contamination Effects Prediction: The CONTAM Computer Program," Final Report, Air Force Rocket Propulsion Laboratory, AFRPL-TR-71-109, December 1971.
3. Nickerson, G.R. and Kliegel, J.R., "Axisymmetric Two-Phase Perfect Gas Performance Program," TRW Systems Report No. 02874-6006-R000, Vols. I and II, April 1967.
4. Penny, M.M. and Smith, S.D., "Supersonic Gas-Particle Flows, Including Reacting Chemistry," JANNAF 8th Plume Technology Meeting, Colorado Springs, CO, July 1974.

FULLNOZ is based on the MULTITUBE code developed by Boynton,⁵ which incorporates the streamtube method† (described by Boynton and Thomson⁶) to integrate the hyperbolic governing equations of steady supersonic flow. The major routines incorporated into FULLNOZ which are not in MULTITUBE include (1) particle/gas nonequilibrium, (2) nonequilibrium chemistry and (3) turbulent wall boundary layers. Briefly, in the streamtube method the elliptic Navier-Stokes equations‡ are reduced to hyperbolic form by assuming that diffusional effects along streamlines are small compared to diffusion across streamlines. This assumption is very good for rocket nozzle (and plume) flows and enables one to solve an initial value problem (where a marching procedure can be used) rather than the more difficult boundary value problem. The gas flow equations, in finite-difference form, are solved along and perpendicular to streamlines while a full continuum particle cloud system of equations is incorporated for the condensed phase.

The advantages of using the streamtube method over the method of characteristics in calculating rocket nozzle (and plume) flows are:

- Species diffusion, shear, and heat transfer normal to streamlines are easily included.
- Chemical reactions or internal relaxations are easily incorporated since the calculation follows streamlines.
- Bounding surfaces and gradients normal to streamlines are treated without difficulty; and
- A wide variety of boundary conditions including mass transfer, shear and heat transfer can readily be incorporated.

† In contrast to the abovementioned codes,¹⁻⁴ which all use the method of characteristics.

‡ In using the Navier-Stokes equations as a base the technique can readily be extended, with the proper boundary conditions, to low density nozzle flows.

5. Boynton, F.P., "The MULTITUBE Supersonic Flow Computer Code," General Dynamics/Convair GDC-DBB 67-003, February 1967.
6. Boynton, F.P. and Thomson, A., "Numerical Computation of Steady, Supersonic, Two-Dimensional Gas Flow in Natural Coordinates," J. Computational Phys. 3, 379-398 (1969).

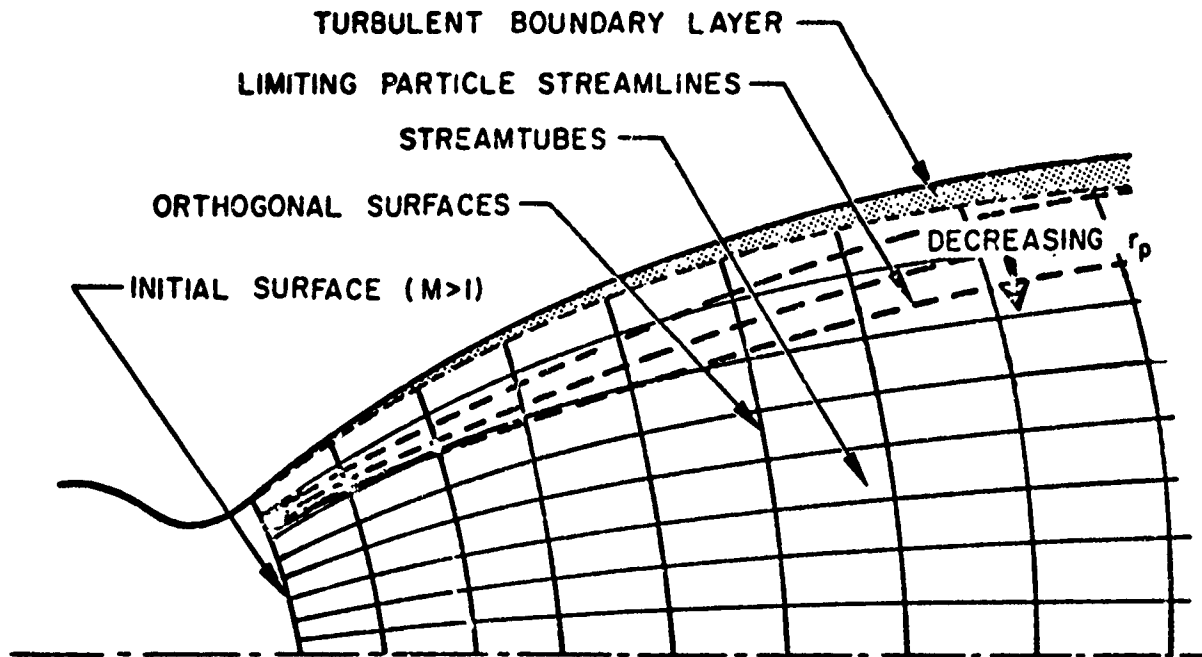


Figure 1. FULLNOZ schematic.

Operation of the code is achieved by specifying (1) initial gas and particle properties in the supersonic region just downstream of the nozzle throat, † (2) the nozzle wall contour, (3) a chemical reaction mechanism and rate coefficients, (4) physical properties of the particles, and (5) the nozzle wall temperature (for the boundary layer calculations). With the above input data the marching scheme steps from one orthogonal surface to the next (Fig. 1) computing gas and particle properties within a specified number of streamtubes. The (essentially) continuous particle size distribution is represented by up to a maximum of 8 discrete particle sizes. All particles cross the gas streamlines, but the lighter particles follow the gas streamlines more closely than the heavy particles. As depicted in Fig. 1, limiting particle streamlines are computed for each particle size. At each orthogonal surface (i.e. each integration

† It would be useful to incorporate an "initializing" scheme which utilizes combustion chamber properties as initial conditions, computes flow properties through the transonic region and establishes an initial supersonic data line. Such a scheme, developed by Nickerson and Kliegel,³ is contained in the codes of Refs. 1-3, and with additional programming could be inserted into FULLNOZ.

step) the code calculates wall shear stress and heat transfer, boundary layer displacement thickness and velocity and temperature profiles. The shear stress and heat transfer are coupled to the main nozzle flow via their effect on the wall streamtube properties.[†] The turbulent boundary layer analysis utilizes the Van Driest⁷ transformations and the experimental data of Keener and Hopkins,⁸ which relates the compressible skin friction coefficient to measured velocity/temperature profiles in flows with favorable pressure gradients (see Section II.C.4). The boundary layer momentum thickness is computed via the momentum integral equation, using the wall streamtube pressure, velocity, etc., as the boundary layer 'edge' conditions. The displacement thickness is then determined from the velocity profile and momentum thickness.

The code will not handle shocks that might originate from the nozzle wall in turning the flow. The flow in the region of the shock will be treated as a strong compression wave.[‡]

[†] This implicitly assumes that the boundary layer momentum and energy thicknesses are smaller than the wall streamtube thickness. This will generally be true for high Reynolds number nozzle flows. Provision is in the program, however, to transfer momentum, heat and mass across streamtubes, so that if an appropriate transfer coefficient can be defined the boundary layer effects can be "felt" throughout the flow.

[‡] Nozzle shocks could be incorporated into FULLNOZ in a manner similar to that employed in the AIPP code^{9,10} to detect internal plume shocks, but this would require additional programming.

7. Van Driest, E.R., "Turbulent Boundary Layer in Compressible Fluids," *J. Aeron. Sci.* 18, 145-160 (1951).
8. Keener, E.R. and Hopkins, E.J., "Van Driest Generalization Applied to Turbulent Skin Friction and Velocity Profiles Measured on the Wall of a Mach 7.4 Wind Tunnel," *ALAA J.* 11, 1784-1785 (1973).
9. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program. (The AIPP Code). Part I. Analytical and Numerical Techniques," *AeroChem TP-302a, AFPRL-TR-74-59*, November 1974.
10. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program (The AIPP Code). Part II. Program User's Manual," *AeroChem TP-* (in preparation).

The code was written for operation on a CDC 6400 computer and requires approximately 122 K (octal) of core storage. For operation on machines with larger core storage the user may want to increase the dimension of some parameters, such as total number of species (25 maximum) and total number of reactions (40 maximum).

This report serves primarily as a program user's manual,[†] but also contains all the governing equations (Section II) and the results of some preliminary parametric calculations for the Minuteman, second stage nozzle (Section IV.A). The user should be cautioned that, although the code has been formally debugged for several test cases, extensive calculations have not (as of April 1975) been made; therefore some operational problems may be experienced. If these do occur, please contact the authors.

II. GOVERNING EQUATIONS

This section gives the governing differential and finite-difference equations used in the code. Also included are the auxiliary equations for calculating thermodynamic and chemical kinetic properties of the system, gas/particle drag and heat transfer coefficients and turbulent boundary layer properties.

A. Differential Equations

1. Gas Phase

For most high Reynolds number nozzle flows of interest the turbulent transport of mass, momentum and energy throughout the main nozzle flow will be negligible. Thus in the equations that follow all transport terms will be identically zero. In practice this is achieved by setting ITURB (Card 4, Cols. 56-60) equal to zero. If turbulent diffusion across streamtubes is to be included due either to initial non-uniformities or to the propagation of boundary layer effects across the flow, appropriate (constant) values of the eddy transport terms, μ , Pr and Le will have to be input on Card 10. If a suitable expression for eddy viscosity in terms of local properties can be developed for transport across streamtubes within nozzles, this expression can easily be added to the program in subroutine TRANSP.

[†] The same numerical integration techniques and similar program logic is incorporated into the AIFP code,^{9,10} to which the reader is referred for additional information on program subroutines.

Global Continuity

$$(\rho u)_s + \rho u \frac{\sin \theta}{r} + \rho u \dot{\theta}_n = 0 \quad (1)$$

s-Momentum

$$\rho u u_s + p_s = \frac{1}{r} \left[r \mu u_n \right]_n - \sum_{j=1}^J \frac{g}{2} \frac{\rho_{p_j}^g \rho_{p_j}^f \mu_g}{\rho_s r p_j^2} (u - u_{p_j}) \quad (2)$$

n-Momentum

$$\rho u^2 \theta_s + p_n = \sum_{j=1}^J \frac{g}{2} \frac{\mu_g \rho_{p_j}^g \rho_{p_j}^f}{\rho_s r p_j^2} v_{p_j} \quad (3)$$

Species Continuity[†]

$$\rho n(C_i)_s = \rho \dot{W}_i + \frac{1}{r} \left[r \mu \frac{Le}{Pr} (C_i)_n \right]_n \quad (4)$$

Energy

$$\begin{aligned} \rho u H_s = & \frac{1}{r} \left[r \frac{\mu}{Pr} H_n \right]_n + \frac{1}{r} \left[\left(1 - \frac{1}{Pr}\right) \mu \left(\frac{u^2}{2}\right)_n \cdot r \right]_n \\ & + \frac{1}{r} \left[\sum_{i=1}^I \left\{ \frac{\mu}{Pr} (Le - 1) r h_i (C_i)_n \right\} \right]_n \\ & + \sum_{j=1}^J \frac{g}{2} \frac{\mu_g \rho_{p_j}^g \rho_{p_j}^f}{\rho_s r p_j^2} \left\{ v_{p_j}^2 + (u - u_{p_j})^2 \right. \\ & \left. + \frac{2}{3} \frac{g_{p_j} C_p}{f_{p_j} Pr_g} (\tau_{p_j} - \tau) \right\} \quad (5) \end{aligned}$$

[†]In this formulation, \dot{W}_i is expressed in units of sec^{-1} .

The drag (f_{pj}) and heat transfer (g_{pj}) factors are defined as, $f_{pj} = C_D Re_p / 24$ and $g_{pj} = (Nu/2) (T_p - T_r) / (T_p - T)$, where T_r is the recovery temperature based on the relative velocity between the gas and particle.

The following auxiliary expressions are required:

Species Production

$$\dot{W}_i = \frac{M_i}{\rho} \sum_{m=1}^N (v_{i,m}'' - v_{i,m}') \left[k_{fm} \prod_{l=1}^L Y_l^{v_{l,m}'} - \frac{k_{fm}}{K_p} \prod_{l=1}^L Y_l^{v_{l,m}''} \right] \quad (6)$$

Equation of State

$$\rho = \frac{P}{RT} \sum_i \frac{C_i}{M_i} \quad (7)$$

Stagnation Enthalpy

$$H = \frac{u^2}{2} + h(T) \quad (8)$$

2. Particles

For the condensed phases present within the flow a continuum particle cloud assumption is made and therefore field conservation equations for continuity, momentum and energy can be written for the particles. The (essentially) continuous distribution of particle sizes at each point in the flow is modeled by several groups of constant size particles representative of the distribution. For a given group, j , the conservation equations, written in a streamline oriented coordinate system are:

Continuity

$$(r \rho_{p_j} u_{p_j})_s + (r \rho_{p_j} v_{p_j})_n + r \rho_{p_j} u_{p_j} \Theta_n - r \rho_{p_j} v_{p_j} \Theta_s = 0 \quad (9)$$

s-Momentum

$$u_{p_j} (u_{p_j})_s + v_{p_j} (u_{p_j})_n - v_{p_j} (v_{p_j} \Theta_n - u_{p_j} \Theta_s) =$$
$$- \frac{9}{2} \frac{\mu_g^f p_j}{\rho_s r_{p_j}^2} (u_{p_j} - u) \quad (10)$$

n-Momentum

$$u_{p_j} (v_{p_j})_s + v_{p_j} (v_{p_j})_n + u_{p_j} (v_{p_j} \Theta_n + u_{p_j} \Theta_s) =$$
$$- \frac{9}{2} \frac{\mu_g^f p_j}{\rho_s r_{p_j}^2} v_{p_j} \quad (11)$$

Energy[†]

$$u_{p_j} (C_s T_{p_j})_s + v_{p_j} (C_s T_{p_j})_n = - \frac{3 k_g g_{p_j}}{r_{p_j}^2 \rho_s} (T_{p_j} - T) \quad (12)$$

[†]The effects of chemical reactions on the surface of particles are not included in this analysis.

When the particle undergoes a phase change (liquid to solid) it is kept at the solidification temperature until the total heat of solidification is released (via radiative and convective heat transfer) to the gas.

B. Finite-Difference Equations

A finite difference formulation of the gas phase and particle cloud governing equations is utilized on a grid which lies along and perpendicular to the streamlines.

1. Gas-Phase

The momentum and energy equations (Eqs. (2), (3) and (5)) are solved via an explicit finite-difference marching technique, whereas the species continuity equation (Eq. (4)), utilizes an implicit finite difference formulation developed in an earlier study at AeroChem.¹¹ This mixed form of the difference equations is necessary for the economical operation of the present code since, for near-equilibrium chemistry, the explicit finite-difference form of Eq. (4) leads to stability-limited (impractically small) integration step sizes. Equations (2), (3) and (5) are left in explicit form since the required integration step sizes for stability are reasonable; an implicit formulation of these equations would unnecessarily complicate the calculations.

To minimize the effects of large tube-to-tube property variations on the calculation of streamline curvature, a VonMises-type transformation is employed, i. e.

$$\dot{m} = 2\pi \int \rho u r \, dn \quad (13)$$

Using Eq. (13) the finite difference forms of the differential equations become:

Global Continuity

$$\dot{m}_k = \rho_{k,l+1} u_{k,l+1} A_{k,l+1} \quad (14)$$

11. Mikatarian, R.R., Kau, C.J., and Pergament, H.S., "A Fast Computer Program for Nonequilibrium Rocket Plume Predictions," Final Report, AeroChem TP-282, AFRPL-TR-72-94, August 1972.

s-Momentum

$$\begin{aligned}
 \dot{m}_k (u_{k,l+1} - u_{k,l}) + \frac{1}{2} (A_{k,l+1} + A_{k,l}) (p_{k,l+1} - p_{k,l}) = \\
 \Delta_k (r_k \tau_k \delta s_k) 2\pi - \frac{1}{2} (A_{k,l+1} + A_{k,l}) \\
 \times \sum_{j=1}^J \frac{g}{2} \left(\frac{\rho_{p_j} f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} (u_{k,l} - u_{p_j,k,l})
 \end{aligned} \tag{15}$$

n-Momentum

$$\begin{aligned}
 \left(\frac{\partial \theta}{\partial s} \right)_{k,l} = - \frac{8\pi r_{k,l}}{u_{k,l} + u_{k+1,l}} \left(\frac{p_{k+1,l} - p_{k,l}}{\dot{m}_k + \dot{m}_{k+1}} \right) \\
 + \frac{2}{((\rho u^2)_{k+1,l} + (\rho u^2)_{k,l})} \\
 \times \sum_{j=1}^J \frac{g}{2} \left(\frac{\rho_{p_j} f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} v_{p_j,k,l}
 \end{aligned} \tag{16}$$

Species Continuity[†]

$$\dot{m}_k (C_{i,k,l+1} - C_{i,k,l}) = \Delta_k (r_k D_{i,k} \delta s_k) 2\pi + \frac{2 \dot{m}_k \delta s_k}{(u_{k,l+1} + u_{k,l})} \dot{W}_{i,k} \tag{17}$$

[†] The species mass fractions at station $k, l + 1$ ($C_{i, k, l+1}$) are determined by linearizing the chemistry terms ($(\dot{W}_i)_k$) and inverting the resulting matrix (see Ref. 11).

Energy

$$\begin{aligned}
 \dot{m}_k \left(h_{k,l+1} - h_{k,l} + \frac{1}{2} u_{k,l+1}^2 - \frac{1}{2} u_{k,l}^2 \right) = \\
 \Delta_k \left(r_k \left[q_k + \left(\frac{1}{2} (u_k^2 + u_{k+1}^2) \right)^{1/2} \bar{r}_k + \sum_{i=1}^I h_{i,k} D_{i,k} \right] \delta s_k \right) 2\pi \\
 + \frac{1}{2} (A_{k,l+1} + A_{k,l}) \sum_{j=1}^J \left(\frac{q}{2} \left(\frac{\rho_{p_j} f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} \left[v_{p_j k,l}^2 \right. \right. \\
 \left. \left. + (u_{k,l}^2 - u_{p_j k,l}^2) + \frac{2}{3} \left(\frac{g_{p_j} C_{p_g}}{f_{p_j} Pr} \right)_{k,l} (T_{p_j k,l} - T_{k,l}) \right] \right) \quad (18)
 \end{aligned}$$

State

$$\rho_{k,l} = \frac{P_{k,l}}{R T_{k,l} \sum_i \left(\frac{C_i}{M_i} \right)_{k,l}} \quad (19)$$

Enthalpy

$$H = \frac{u_{k,l}^2}{2} + \sum_i \sum_{jj} a_{i,jj} T_{k,l}^{jj-1} C_{i,k,l} \quad (20)$$

where in Eqs. (15), (17) and (18),

$$\tau_k = -2\pi r_k^2 (u_{k+1,l}^2 - u_{k,l}^2) \frac{(\rho_{k,l} \mu_{k,l} + \rho_{k+1,l} \mu_{k+1,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (21)$$

and

$$q_k = -2\pi r_k^2 (u_{k+1,l} + u_{k,l}) \times \frac{(\rho_{k,l} k_{k,l} + \rho_{k+1,l} k_{k+1,l})(T_{k+1,l} - T_{k,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (22)$$

and

$$D_{i,k} = -\frac{2\pi r_k^2 (u_{k+1,l} + u_k) (\rho_{k+1,l}^2 \bar{D}_{k+1,l} + \rho_{k,l}^2 \bar{D}_{k,l}) (C_{i,k+1,l} - C_{i,k,l})}{(\dot{m}_{k+1} + \dot{m}_k)} \quad (23)$$

2. Particles

The finite difference forms of Eqs. (9-12) are:

Continuity

$$\begin{aligned} & (r\rho_{p_j} u_{p_j})_{k,l+1} - (r\rho_{p_j} u_{p_j})_{k,l} + ((r\rho_{p_j} V_{p_j})_{k+1,l} - (r\rho_{p_j} V_{p_j})_{k,l}) \frac{\delta s_k}{\delta n_k} \\ & + \delta s_k (r\rho_{p_j} u_{p_j})_{k,l} \Theta_{n_{k,l}} - \delta s_k (r\rho_{p_j} V_{p_j})_{k,l} \Theta_{s_{k,l}} = 0 \end{aligned} \quad (24)$$

s-Momentum

$$\begin{aligned}
 & u_{p_{jk,l}} (u_{p_{jk,l+1}} - u_{p_{jk,l}}) + v_{p_{jk,l}} (u_{p_{jk+1,l}} - u_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} \\
 & - \delta s_k v_{p_{jk,l}} (v_{p_j \Theta_n} - u_{p_j \Theta_s})_{k,l} = \\
 & - \frac{9}{2} \left(\frac{f_{p_j} \mu_g}{\rho_s r_{p_j}^2} \right)_{k,l} (u_{p_{jk,l}} - u_{k,l}) \delta s_k
 \end{aligned} \tag{25}$$

n-Momentum

$$\begin{aligned}
 & u_{p_{jk,l}} (v_{p_{jk,l+1}} - v_{p_{jk,l}}) + v_{p_{jk,l}} (v_{p_{jk+1,l}} - v_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} \\
 & + u_{f_{jk,l}} (v_{p_j \Theta_n} + u_{p_j \Theta_s})_{k,l} \delta s_k = \\
 & - \frac{9}{2} \left(\frac{\mu_g f_{p_j}}{\rho_s r_{p_j}^2} \right)_{k,l} v_{p_{jk,l}} \delta s_k
 \end{aligned} \tag{26}$$

Energy

$$\begin{aligned}
 & u_{p_{jk,l}} C_s (T_{p_{jk,l+1}} - T_{p_{jk,l}}) + v_{p_{jk,l}} C_s (T_{p_{jk+1,l}} - T_{p_{jk,l}}) \frac{\delta s_k}{\delta n_k} = \\
 & - \frac{3\epsilon\sigma}{r_{p_{jk,l}} \rho_s} T_{p_{jk,l}}^4 \delta s_k - 3 \left(\frac{k_g g_{p_j}}{r_{p_j}^2 \rho_s} \right)_{k,l} \\
 & \quad \times (T_{p_{jk,l}} - T_{k,l}) \delta s_k
 \end{aligned} \tag{27}$$

3. Integration Step Size

The integration step size must be limited in order to perform a stable calculation. The stability of the explicit finite difference scheme for solving the gas dynamic equations is discussed by Boynton and Thomson,⁶ who show that the stable step size for laminar flow is determined from,

$$\delta s \leq \frac{\delta n}{2} \left[\frac{1}{\text{Re}} + \frac{1}{(M^2 - 1)^{1/2}} \right]^{-1} \quad (28)$$

For turbulent flow the eddy viscosity replaces the laminar viscosity in the expression for Re.[†] The step size determined from Eq. (28) may not be sufficiently small if the chemistry is very "fast". Thus, an additional "chemistry" control has been incorporated into the code: at each integration step (orthogonal surface) the species mass fractions are tested for sign. If any C_i goes negative the integration step size is halved and the calculations are repeated until the C_i in question becomes positive or the step size becomes less than the minimum allowable step size. In the latter case the program terminates.

The particle cloud system of equations (Eqs. (9-12)) contains derivatives related only to the convection of mass, momentum and energy. In the momentum and energy equations, the convection terms are equal to the particle/gas interaction terms. Unlike the gas flow equations the particle equations have no wave or diffusive nature. Consequently when these equations are solved explicitly there is no stability limitation on the integration step size and the step size can be determined from Eq. (28).

4. Solution Procedure

The gas phase and particle cloud finite difference equations are solved on a grid consisting of the streamtubes and the surfaces orthogonal to them, as illustrated in Fig. 1. All gas and particle flow properties, streamline positions and angles must be known along the initial orthogonal surface downstream of the throat ($M > 1$). Starting with the first streamtube (adjacent to the axis) the streamtubes are extended a distance δs to a downstream surface, utilizing the radius of curvature, $(\Theta_S)'$ obtained from the normal momentum equation (Eq. (16)) and the known initial pressure distribution along the

[†] Eq. (28) is not, of course, valid in the limit, $\text{Re} \rightarrow 0$. In that case (i. e. the situation for most practical nozzle flows), $\delta s \leq \frac{\delta n}{2} [M^2 - 1]^{1/2}$.

surface. The resulting streamtube areas are then used to determine all the necessary gas properties at the downstream surface from Eqs. (14), (15), and (17)-(23). The transfer of mass, momentum and energy into or out of each streamtube is arranged such that what is lost from a given streamtube is gained by the adjacent tube. Thereby mass, momentum and energy are automatically conserved.

In order to render this marching scheme conditionally stable a single iteration on the radius of curvature calculation is required,⁶ with the new value, $(\Theta_g)''$, determined using the downstream surface properties. $(\Theta_s)''$ is then combined with $(\Theta_s)'_k$ in the following expression,⁶

$$(\Theta_g) = (1 - \alpha)(\Theta_s)' + \alpha(\Theta_s)'' \quad (29)$$

This new value for the radius of curvature is then used in the calculation. A value for α of 0.55 is used in the present code. (The scheme is conditionally stable for $\alpha \geq 1/2$).

After the gas phase equations have been solved for the first streamtube the particle properties are determined by sequentially applying the particle momentum, continuity and energy equations. The calculation of gas and particle properties then moves outward to the next streamtube and the procedure is repeated up to the last streamtube[†] where boundary conditions must be applied.

C. Auxiliary Calculations

1. Thermodynamic Input Data

The thermodynamic data are input via curve fits of specific heats¹² of individual species in the JANNAF tables.¹³ These curve fits have the form,

$$C_{p_i} = L_{1_i} + L_{2_i} \Phi + L_{3_i} \Phi^2 + L_{4_i} \Phi^3 + L_{5_i} \Phi^{-2} \quad \text{cal/mole-}^\circ\text{K} \quad (30)$$

[†] In general the limiting particle streamlines for each particle size will not extend to the last streamtube.

12. Cruise, D.R., "Information Manual for the Theoretical Propellant Evaluation Program," Naval Weapons Center PEP NOTE TN-U-1 (plus additions), December 1964.
13. JANNAF Thermochemical Tables (Dow Chemical Company, Midland, Mich.), continuously updated.

where $\Phi = T(^{\circ}\text{K})/1000$. The enthalpy is then expressed as,

$$h_i = \int_0^T C_{P_i} dT + L_{6_i} \quad \text{kcal/mole} \quad (31)$$

where $L_{6_i} = H_f^0 - \int_0^{298} C_{P_i} dT$; H_f^0 being the heat of formation at 298°K .

The entropy is expressed as,

$$S_i = \int_0^T C_{P_i} \frac{dT}{T} + L_{7_i} \quad \text{cal/mole-}^{\circ}\text{K} \quad (32)$$

where L_{7_i} is the entropy integration constant. The coefficients $L_1 - L_7$ are input on Card group 8.

2. Chemical Kinetic Input Data

Ten possible reaction types are included in the program:

Reaction Type

(1)	A + B	\rightleftharpoons	C + D
(2)	A + B + M	\rightleftharpoons	C + M
(3)	A + B	\rightleftharpoons	C + D + E
(4)	A + B	\rightleftharpoons	C
(5)	A + M	\rightleftharpoons	C + D + M
(6)	A + B	\rightarrow	C + D
(7)	A + B + M	\rightarrow	C + M
(8)	A + B	\rightarrow	C + D + E
(9)	A + B	\rightarrow	C
(10)	A + M	\rightarrow	C + D + M

Reaction types (6)-(10) correspond to reaction types (1)-(5), but proceed in the forward direction only. In Reactions (2), (5), (7) and (10), M is an arbitrary third body. In this program, all species are assumed to have equal third body efficiencies.

The forward rate coefficient, k_f , is input to the code as one of the following 8 types

Rate Coefficient Type†

- | | |
|-----|----------------------------|
| (1) | $k_f = A$ |
| (2) | $k_f = AT^{-1}$ |
| (3) | $k_f = AT^{-2}$ |
| (4) | $k_f = AT^{-\frac{1}{2}}$ |
| (5) | $k_f = A \exp(B/RT)$ |
| (6) | $k_f = AT^{-1} \exp(B/RT)$ |
| (7) | $k_f = AT^{-\frac{3}{2}}$ |
| (8) | $k_f = AT^N \exp(B/RT)$ |

The equilibrium constant, K_p , is determined from

$$\ln K_p = - \Delta G/RT \quad (33)$$

where the Gibbs free energy, ΔG , for individual reactions is computed from the input thermodynamic data.

† Rate coefficient data for typical rocket nozzle and plume reactions may be found, e.g., in Ref. 14.

14. Jensen, D.E. and Jones, G.A., "Gas-Phase Reaction Rate Coefficients for Rocketry Applications," Rocket Propulsion Establishment Technical Report No. 71/9, October 1971.

3. Particle/Gas Drag and Heat Transfer Coefficients

The momentum and energy exchange (via convective heat transfer) between the small diameter particles and the combustion products in the nozzle cannot adequately be described by simple theoretical expressions (e.g. Stokes law).¹⁵ Empirical correlations of drag and heat transfer coefficients developed by Crowe¹⁶ have therefore been incorporated in the particle/gas interaction terms (see, e.g. Eqs. (2), (3), and (5)).

a. Drag Coefficient - The drag coefficient has been correlated by Crowe¹⁶ in terms of a normalized value,

$$\bar{C}_D = (\dot{C}_D - C_{D_I}) / (C_{D_{FM}} - C_{D_I}) \quad (34)$$

where C_{D_I} is the drag coefficient at very large Reynolds number and $C_{D_{FM}}$ is the free molecular drag coefficient. Note that for $Re_p \ll 1$, $\bar{C}_D \rightarrow 1$, while for $Re_p \gg 1$, $\bar{C}_D \rightarrow 0$.

The expressions needed to evaluate C_D from Eq. (32) are:

$$C_{D_I} = 0.66 + 0.26 [\exp(4 \ln M_p) - 1] + 0.17 \exp[-2.5(\ln M_p / 1.4)^2] \quad (35)$$

$$C_{D_{FM}} = \frac{\exp(-S_1^2/2)}{\sqrt{\pi} S_1^3} (1 + 2S_1^2) + \frac{4(S_1^4 + S_1^2) - 1}{2S_1^4} \operatorname{erf}(S_1) + \frac{2}{3} \frac{\sqrt{\pi}}{\sqrt{T/T_p}} \quad (36)$$

$$\text{where, } S_1 = \sqrt{\gamma/2} M_p \quad (37)$$

$$\bar{C}_D = G(\text{Kn}) D(\text{Kn}, Re_p) \quad (38)$$

$$G(\text{Kn}) = \frac{\text{Kn}^{0.4} \exp(1.2 \text{Kn}^{0.5})}{1 + \text{Kn}^{0.4} \exp(1.2 \text{Kn}^{0.5})} \quad (39)$$

15. Soo, S. L., Fluid Dynamics of Multiphase Systems (Blaisdell Publ. Co, Waltham, Mass., 1967).

16. Crowe, C. T., "On the Momentum and Heat Transfer Equations for Two-Phase Plumes," Washington State Univ., March 1971.

and

$$D(\text{Kn}, \text{Re}_p) = 1 - \exp\left[-\frac{\text{Re}_p}{8} \text{Kn}^{0.6} \exp(\text{Kn}) (C_{D_0} - 0.4)\right] \quad (40)$$

$$\text{Kn} = 1.26 \sqrt{\gamma} \frac{M_p}{\text{Re}_p} \quad (41)$$

and, $C_{D_0} = 24/\text{Re}_p$ (42)

The expression used in the particle/gas interaction terms is then,

$$f_p = \text{FFF} \frac{C_D}{C_{D_0}} \quad (43)$$

where FFF is a factor (input on Card 15, Cols. 1-10) used to arbitrarily vary C_D to account for uncertainties in the above analysis.

b. Heat Transfer Coefficient - Heat transfer from the particle to the gas is expressed in terms of a Nusselt number as,

$$q = 2\pi r_p \text{Nu} k(T_p - T_r) \quad (44)$$

where T_r is the recovery temperature. From Crowe¹⁶ we get the following expression,

$$\text{Nu} = \text{Nu}_{\text{KD}} + \frac{\gamma + 1}{\gamma} \text{Re}_p \text{Pr}_g \exp(-\text{Re}_p/2M_p) \quad (45)$$

$$\text{Nu}_{\text{KD}} = \text{Nu}_0 / \left(1 + \frac{5\gamma^{1.5}}{\gamma + 1} (M_p/\text{Re}_p \text{Pr}_g) \text{Nu}_0\right) \quad (46)$$

Nu_0 is the Nusselt number in incompressible flow, expressed as

$$\text{Nu}_0 = 2.0 + 0.459 \text{Re}_p^{0.55} \text{Pr}_g^{0.33} \quad (47)$$

The recovery temperature is defined as,

$$T_r = T + r \left((u - u_p)^2 + v_p^2 \right) / 2C_p \quad (48)$$

where the recovery factor r is¹⁶;

$$r = 0.9 + (r_{FM} - 0.9) \exp(-Re_p / 2M_p) \quad (49)$$

and

$$r_{FM} = \frac{\gamma}{\gamma + 1} \left(2 + 0.67 \exp(-M^2/\beta) \right) \quad (50)$$

The expression used in the particle/gas interaction term is,

$$g_p = FFG \frac{Nu}{2} \frac{T_p - T_r}{T_p - T} \quad (51)$$

where FFG is a factor (input on Card 15, Cols. 11-20) used to arbitrarily vary g_p to account for uncertainties in the above analysis.

D. Turbulent Boundary Layer Equations[†]

The turbulent compressible boundary layer analysis is initiated by calculating the corresponding adiabatic flat plate, zero pressure gradient, incompressible boundary layer properties, based on a 1/7 power law¹⁷

[†] The "free stream" properties (subscript "e") for the boundary layer analysis are taken to be the properties in the last (wall) streamtube. It is implicitly assumed that the boundary layer displacement thickness is smaller than the width of the wall streamtube.

17. Schlichting, H., Boundary Layer Theory, 6th Ed. (McGraw-Hill, New York, 1968), p. 599.

velocity profile.[†] Expressions for the incompressible momentum and boundary layer thickness are,

$$\theta_i(x) = 0.036/Re_x^{0.2} \quad (52)$$

$$\delta_i(x) = 10.286 \theta_i(x) \quad (53)$$

The incompressible skin friction coefficient is evaluated from the Karman-Schoenherr relation.²⁰

$$C_{F_i} = [17.08 (\log_{10} Re_{\theta_i})^2 + 25.11 \log_{10} Re_{\theta_i} + 6.012]^{-1} \quad (54)$$

Transformation from the compressible to the incompressible boundary layer is accomplished via the Van Driest equations,⁷ which relate the velocity and temperature profiles as follows,

$$\bar{T}_t = \frac{T_e - T_w}{T_{t_e} - T_w} = f\left(\frac{U}{U_e}\right) \quad (55)$$

[†] It was originally anticipated that the complete boundary layer equations including nonequilibrium chemistry would be solved by finite differences and coupled to the nozzle flow solution. A number of boundary layer codes are available, including those of Herring and Mellor¹⁸ and the Aerotherm BLIMP code,¹⁹ which would be very useful for this purpose. However, initial attempts to incorporate the BLIMP code into FULLNOZ showed that it would take more effort than was warranted at this time. Consequently, a more simplified analysis was incorporated into the present code.

18. Herring, H. J. and Mellor, G. L., "A Method of Calculating Compressible Turbulent Boundary Layers," NASA CR-1144, September 1968.
19. Tong, H., Buckingham, A. C., and Morse, H. L., "Nonequilibrium Chemistry Boundary Layer Integral Matrix Procedure," Aerotherm Final Report No. 73-67, July 1973.
20. Hopkins, E. J., Keener, E. R., and Louie, P. T., "Direct Measurements of Turbulent Skin Friction on a Nonadiabatic Flat Plate at Mach Number 6.5 and Comparisons with Eight Theories," NASA TN D-5675, February 1970.

The general functional relation used in the analysis is

$$\bar{T}_t = \left(\frac{U}{U_e} \right)^n \quad (56)$$

Equation (56) becomes the Crocco relation for $n = 1$ and the quadratic for $n = 2$.

A general expression between T/T_e and U/U_e can then be written

$$\frac{T}{T_e} = A \left(\frac{U}{U_e} \right)^2 + B \left(\frac{U}{U_e} \right)^n + C \quad (57)$$

where

$$A = \left(\frac{T_{te}}{T_e} - 1 \right); B = \left(\frac{T_{te}}{T_e} - \frac{T_w}{T_e} \right); C = \frac{T_w}{T_e}$$

From Back and Cuffel,²¹ wall cooling ($T_w < T_{aw}$) tends to make the exponent n closer to 1.0, while wall heating causes n to approach 2 or more. The value of n selected here was 1.2. It was determined by fitting recent data of Keener and Hopkins⁸ for which, $T_w/T_{aw} = 0.32$. (This corresponds to a wall temperature of 1000-1300 K.) Equation (57) was used to generate a table of T/T_e vs. U/U_e , which is then used to determine the compressible skin friction coefficient via the Van Driest transformation

$$\frac{C_{f_c}}{C_{f_i}} = \left\{ \int_0^1 \left(\frac{T_e}{T} \right)^{1/2} d \left(\frac{U}{U_e} \right) \right\}^2 \quad (58)$$

The Stanton number was determined from the relation

$$S_t = 0.35 C_{f_c} \quad (59)$$

which is supported by the experimental data of Back and Cuffel²¹ for accelerating flows.

21. Back, L.H. and Cuffel, R.F., "Relationship Between Temperature and Velocity Profiles in a Turbulent Boundary Layer along a Supersonic Nozzle with Heat Transfer," AIAA J. 8, 2066-2069 (1970).

Wall shear stress and heat transfer are then calculated from

$$\tau_w = \frac{1}{2} \rho_e U_e^2 C_{f_c} \quad (60)$$

$$\dot{q}_w = -S_t \rho_e U_e C_p (T_{t_e} - T_w) \quad (61)$$

In order to obtain the velocity profiles the friction velocity profile, $(U/U_\tau)_i = f(U/U_e)$ is first determined from another Van Driest transformation,⁷

$$\left(\frac{U}{U_\tau}\right)_i = \left(\frac{2}{C_{f_c}}\right)^{1/2} \int_0^{U/U_e} \left(\frac{T_e}{T}\right)^{1/2} d\left(\frac{U}{U_e}\right) \quad (62)$$

where the friction velocity is defined as,

$$U_{\tau_i} = \left(\rho_e U_e^2 C_{f_i} / 2\rho_w\right)^{1/2} \quad (63)$$

$(y/\delta)_i$ is then obtained from the standard incompressible boundary layer profiles²² †

$$\frac{yU_\tau}{\nu} = \frac{U}{U_\tau}; \quad \frac{U}{U_\tau} < 5 \quad (64)$$

$$\frac{yU_\tau}{\nu} = \exp\left\{\frac{U/U_\tau + 3.05}{2.5}\right\}; \quad 5 \leq \frac{U}{U_\tau} \leq 13.96 \quad (65)$$

$$\frac{yU_\tau}{\nu} = \exp\left\{\frac{U/U_\tau - 5.05}{2.5}\right\}; \quad \frac{U}{U_\tau} > 13.96 \quad (66)$$

† The values of yU_τ/ν are normalized by $\exp\{[U/U_\tau \text{ max} - 5.05]/2.5\}$ to avoid the calculation of the ν profile in the boundary layer.

22. Kays, W.M., Convective Heat and Mass Transfer (McGraw-Hill, New York, 1966).

Another Van Driest transformation yields $(y/\delta)_c = (y/\delta)_i$. Once the compressible profiles are known, Θ/δ , δ^*/δ , and the shape factor are determined from

$$\Theta/\delta = \int_0^1 \left(\frac{U}{U_e} \right) \left(1 - \frac{U}{U_e} \right) d\left(\frac{y}{\delta} \right) \quad (67)$$

$$\delta^*/\delta = \int_0^1 \left(1 - \frac{U}{U_e} \right) d\left(\frac{y}{\delta} \right) \quad (68)$$

$$H_{12} = \frac{\delta^*/\delta}{\Theta/\delta} \quad (69)$$

The variation of momentum thickness along the nozzle is determined via integration of the momentum integral equation,

$$\frac{C_f}{2} = \frac{d\Theta}{dx} + [H_{12} + 2]\Theta \frac{1}{U_e} \frac{dU_e}{dx} + \frac{\Theta}{\rho_e} \frac{d\rho_e}{dx} + \frac{\Theta}{R} \frac{dR}{dx} \quad (70)$$

by the use of backward differences.[†] δ and δ^* are then evaluated from Eqs. (67) and (68).

III. PREPARATION OF INPUT DATA

All necessary information for preparing input data is given below; Fig. 2 defines some of the input for a sample case. Many of the input parameters that are left blank are used in the companion rocket plume code¹⁰ (AIPP), but not in FULLNOZ.

[†] This requires that a throat value of Θ be assumed.

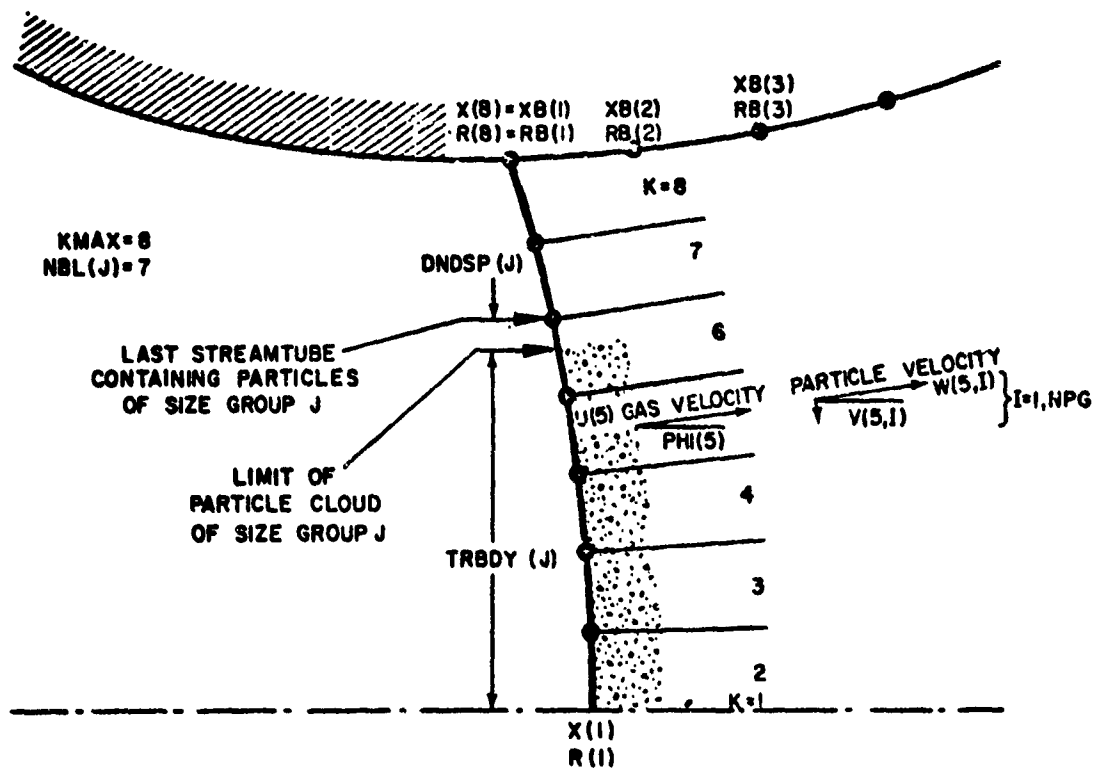


Figure 2. Input data definitions.

Card No.	Cols.	Fortran Name	Description	Format
1	1-10	NDATA	Number of data sets; in present version of code data sets cannot be stacked; therefore NDATA = 1	I.10
2	1-80	ID	Run identification	20A4
3	1-10	NSEC	Maximum run time (sec); when run time reaches NSEC nozzle properties at last orthogonal surface are punched and can be used to continue the calculation. (Not operational in present version of code)	I.10
	11-22	DXLSS	Approximate distance (cm) along axis between orthogonal surface print stations. For control of print increment by number of integration steps (KP, Card 5, Cols. 11-15) instead of axial distance, make DXLSS larger than XLMAX (below)	E12.5

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
3	23-34	XLMAX	Maximum distance (cm) along axis for which calculations will be made. XLMAX should be set somewhat larger than axial distance to nozzle exit plane in order to complete calculation at last wall point	E12.5
4	1- 5	ITYPE	Inner boundary condition (nozzle axis) Set ITYPE = 1	15
	6-10	IKIND	Outer boundary condition; set IKIND = 1 for wall boundary condition	
	11-15		Not used; leave blank	
	16-20		Not used; leave blank	
	21-25		Not used; leave blank	
	26-30	IBUGSH	Debug printout index; 0 - No debug printout 1 - Extensive printout for debugging	15
	31-35		Not used; leave blank	
	36-40		Not used; leave blank	
	41-45		Not used; leave blank	
	46-50	IPART	Particle indicator; 0 - No particles 1 - Particles in flow	15
	51-55		Not used; leave blank	
	56-60	ITURB	Turbulent flow indicator; 0 - Inviscid 3 - Constant values of turbulent mixing parameters, μ , Pr and Le are input on Card 10	15
5	1- 5	KMAX	Initial number of streamtubes plus one (axis is counted as K = 1; maximum is 40)	15
	6-10	NN	Number of terms for C_p polynomial curve fit plus one; for curve fits supplied with program, ¹² NN = 6. If fewer coefficients are used for a given species NN is left at 6 and zeroes are input for the missing coefficients (see Card 8)	

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
5	11-15	KP	Output control; number of orthogonal surfaces between print stations. To control print via axial distance (DXLSS, Card 3, Cols. 11-22) instead of orthogonal surfaces set KP larger than LPLANE (Card 5, Cols. 36-40)	
	16-20	MMAX	Number of points needed to describe shape of inner boundary; always set equal to 2 for axis boundary	I5
	21-25	NMAX	Number of points required to define nozzle wall contour (50 maximum)	I5
	26-30	NDS	Total number of gas species in flow (25 maximum)	I5
	31-35	NITER	Maximum number of iterations allowed for iterative solutions (e.g. calculation of streamtube properties for variable Y) Recommended value: NITER = 50	I5
	36-40	LPLANE	Maximum number of integration steps for entire calculation	I5
	41-45	IKINE	Number of chemical reactions (40 maximum)	I5

NOTE: IF NO BOUNDARY LAYER CALCULATIONS ARE TO BE MADE CARD 6 MAY BE LEFT BLANK, BUT MUST BE INCLUDED

6	1- 5	TWALL	Nozzle wall temperature ($^{\circ}\text{K}$) (assumed constant)	F5.0
	8	IBLFLG	Boundary layer property printout indicator; 0 - Print Re , δ_1 , θ_1 , C_{f_1} , U_T , δ_c , δ_c^* , θ_c , C_{f_c}/C_{f_1} , H_{12} , q_w , T_w 1 - Print above plus velocity and temperature profiles	I1

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
6	11	IBL	Boundary layer calculation indicator; 0 - Do not calculate boundary layer 1 - Calculate boundary layer	I1
7	1- 9	ALPHAH	Factor that multiplies maximum stable step size (see Eq. (28)); Recommended value: ALPHAH = 0.8	F9.4
	10-18	EPSLON	Amount by which streamtube Mach numbers must exceed one in order for the calculation to continue. Recommended value: EPSLON = 0.01	F9.4
	19-27	TOL	Convergence tolerance for iterative solutions; Recommended value: TOL = 1×10^{-4}	F9.4
	28-36	DELTA	Metric exponent [†] ; 0 - Two-dimensional flow 1 - Axially symmetric flow	F9.4
	37-45	ATOL	Maximum allowable fractional change in streamtube area per step. Recommended value: ATOL = 0.1	F9.4

NOTE: CARD GROUP 8 DEFINES THE THERMODYNAMIC DATA FOR EACH SPECIES. THERE ARE 2 CARDS PER SPECIES AND NDS SPECIES

8.1.1	1-13	A(1)	} L ₁ L ₂ L ₃ L ₄	Specific heat polynomial constants for first species, see Section II.C.1 (cal/mole °K)	E13.5
	14-26	A(2)			
	27-39	A(3)			
	40-52	A(4)			

[†] The particle conservation equations are written only for axially symmetric flow. Thus two-dimensional solutions can only be obtained for flows without particles.

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
8.1.2	1-13	A(5)	L ₅ , Specific heat polynomial constant for first species	E13.5
	14-26	A(6)	L ₆ , Enthalpy constant of integration for first species $\left(\Delta H_f^\circ - \int_0^{298} C_p dT \right) \text{ (kcal/mole)}$	E13.5
	27-39	CS(1)	L ₇ , entropy constant of integration for first species (cal/mole °K)	
	⋮			
8.NDS.1 8.NDS.2			Repeat thermodynamic data for NDS species	

NOTE: CARD GROUP 9 IDENTIFIES EACH SPECIES AND SPECIFIES VARIOUS TRANSPORT PROPERTIES. MUST BE IN SAME ORDER AS CARD GROUP 8

9.1	1- 4	IDENT(1)	Species name; first species (the remaining data on Card 9.1 also apply to the first species)	A4
	13-24	MUO(1)	Viscosity at reference temperature † (g/cm-sec); only used when flow contains particles. FOR IPART = 0, MUO MUST = 0 FOR ALL SPECIES	E12.4
	25-36	TO(1)	Reference temperature for viscosity (°K)	E12.4
	37-48	OMEGA(1)	Exponent describing viscosity/temperature relation; $\mu \propto T^\omega$	E12.4
	49-60	PR(1)	Reciprocal of species Prandtl number	E12.4
	61-72	SC(1)	Reciprocal of species Schmidt number	E12.4
	73-80	MW(1)	Species molecular weight	E8.2
	⋮			

† Values for common gases can be found in most physics and chemistry handbooks.

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
9.NDS			Repeat for each species	
			NOTE: CARD 10 IS NOT NEEDED IF ITURB = 0 (CARD 4, COLS. 56-60)	
10	1-10	TLE	Turbulent Lewis number (constant)	
	11-20	TPR	Turbulent Prandtl number (constant)	
	21-30	EDDYK	Eddy viscosity, g/cm-sec (constant)	
			NOTE: CARD GROUPS 11 AND 12 DEFINE THE LOCATION OF THE INITIAL ORTHOGONAL SURFACE, AND THE FLOW ANGLE, PRESSURE, TEMPERATURE, VELOCITY AND SPECIES MASS FRACTIONS WITHIN EACH STREAMTUBE	
11	1-12	X(1)	Initial axial position on inner boundary (cm)	E12.4
	13-24	R(1)	Initial radial position on inner boundary (cm); usually equal to 0 for axis boundary	E12.4
	25-36	PHI(1)	Initial flow angle [†] on inner boundary (radians); usually equal to 0 for axis boundary	E12.4
12.1.1	1-12	X(2)	Axial position at outer boundary of first streamtube (cm)	E12.4
	13-24	R(2)	Radial position at outer boundary of first streamtube (cm)	E12.4
	25-36	PHI(2)	Flow angle at outer boundary of first streamtube (radians)	E12.4
	37-48	P(2)	Average pressure in first streamtube (atm)	E12.4
	49-60	T(2)	Average temperature in first streamtube (°K)	E12.4
	61-72	U(2)	Average velocity in first streamtube (cm/sec)	E12.4
12.1.2	1-10	C(1, 2)	Average mass fraction of first species in first streamtube	E10.3

[†] Flow angle is defined as the angle between the flow velocity vector and the axis.

Card No.	Cols.	Fortran Name	Description	Format
12.1.2	11-20	C(2,2)	Average mass fraction of second species in first streamtube	E10.3
	⋮			⋮
	71-80	C(8,2)	Average mass fraction of eighth species in first streamtube	E10.3

⋮
12.1.I($\frac{NDS}{8}$)

NOTE: I($\frac{NDS}{8}$) DENOTES THE NEXT INTEGER LARGER

THAN ($\frac{NDS}{8}$). REPEAT CARD 12.1.2 TO INCLUDE NDS SPECIES, 8 SPECIES/CARD

12.2.1	1-12	X(3)	Axial position at outer boundary of second streamtube (cm)	E12.4
	⋮			⋮
	61-72	U(3)	Average velocity in second streamtube (cm/sec)	E12.4

12.2.2	1-10	C(1,3)		E10.3
	⋮			⋮
	71-80	C(8,3)		⋮

12.2.I($\frac{NDS}{8}$)		C(NDS,3)	Average mass fraction of last species in second streamtube	E10.3
---------------------------	--	----------	--	-------

12.(KMAX-1).1	1-12	X(KMAX)	Axial position at outer boundary of last (KMAX-1) streamtube (cm)	E12.4
---------------	------	---------	---	-------

12.(KMAX-1).I($\frac{NDS}{8}$)		C(NDS, KMAX)	Average mass fraction of last species in last streamtube	E10.3
----------------------------------	--	--------------	--	-------

NOTE: CARD 13 DEFINES THE POSITION OF THE INNER BOUNDARY

13.1	1-12	XW(1)	Initial axial position (cm); repeat of X(1) on Card 10	E12.4
	13-24	RW(1)	Initial radial position (cm); repeat of R(1) on Card 10	E12.4

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
13.1	25-36	PH1W(1)	} Leave rest of card blank; program will determine these inner boundary properties	E12.4
	37-48	PW(1)		E12.4
	49-60	SW(1)		E12.4
13.2	1-12	XW(2)	Axial distance greater than XLMAX (Card 3, Cols. 23-34)(cm). Since the inner boundary is an axis the program will interpolate linearly between XW(1) and XW(2) to get axial locations of orthogonal surfaces. Therefore, set $XW(2) = 1 \times 10^{10}$	E12.4
	13-24	RW(2)	Final radial position of inner boundary (cm); since inner boundary is an axis set RW(2) equal to R(1) on Card 11	E12.4
	25-36	PHIW(2)	} Leave rest of card blank; program will determine these inner boundary properties	E12.4
	37-48	PW(2)		E12.4
	49-60	SW(2)		E12.4
NOTE: CARD GROUP 14 DEFINES THE POSITION OF THE OUTER BOUNDARY (NOZZLE WALL CONTOUR)				
14.1	1-12	XB(1)	Initial axial position of nozzle wall (cm); equal to X(KMAX) on Card 12. (KMAX-1).1	E12.4
	13-24	RB(1)	Initial radial position of nozzle wall (cm); equal to R, KMAX) on Card 12. (KMAX-1).1	E12.4
14.2	1-12	XB(2)	Axial position (cm) of second point along nozzle wall	E12.4
	13-24	RB(2)	Radial position (cm) of second point along nozzle wall	E12.4
14.NMAX	1-12	XB(NMAX)	Final axial position (cm) of nozzle wall	E12.4
	13-24	RB(NMAX)	Final radial position (cm) of nozzle wall	E12.4
NOTE: CARD GROUPS 15-19 AND 20-24 (OR 25-32) ARE INCLUDED ONLY IF PARTICLES ARE IN FLOW, i.e. IPART = 1 (CARD 4, COLS. 46-50)				
15	1-10	FFF	Factor which multiplies particle/gas drag coefficient (see Section II.C.3)	E10.3

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
15	11-20	FFG	Factor which multiplies particle/gas heat transfer coefficient (see Section II. C. 3)	E10.3
	21-30	CL	Liquid particle specific heat (cal/g-°K)	E10.3
	31-40	CS	Solid particle specific heat (cal/g-°K)	E10.3
	41-50	HTRAN	Particle heat of solidification (cal/g)	E10.3
	51-60	WT	Particle molecular weight	E10.3
16	1-10	RHSS	Particle density (g/cm ³)	E10.3
	11-20		Not used; leave blank	
	21-30		Not used; leave blank	
	31-40	TPS	Particle solidification temperature (°K)	E10.3
17	1- 5	NPG	Number of particle groups (8 maximum)	I5
	6-10	NC	Index noting whether particle properties are constant or variable along initial orthogonal surface 0 - constant 1 - variable	I5
<p>NOTE: NBL() INDICATES RADIAL EXTENT OF EACH PARTICLE GROUP ALONG INITIAL ORTHOGONAL SURFACE (SEE FIG. 2); DEFINED AS LAST STREAMTUBE NUMBER CONTAINING PARTICLES PLUS 2. FOR EXAMPLE, IF THE FIRST PARTICLE GROUP EXTENDS TO STREAMTUBE NO. 12, NBL(1) = 14</p>				
18	1- 5	NBL(1)	Last streamtube containing first particle group along initial orthogonal surface (streamtube number + 2)	I5
	:			
	:			
		NBL(NPG)	Last streamtube containing NPG particle group (streamtube number + 2)	I5
19	1-10	RP(1)	Radius of first particle group (cm)	8E10.3
	:	:		
	:	:		
		RP(NPG)	Radius of NPG particle group (cm)	

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
-----------------	--------------	---------------------	--------------------	---------------

NOTE: IF PARTICLE PROPERTIES ALONG INITIAL ORTHOGONAL SURFACE ARE CONSTANT (NC = 0) CARDS 20-24 ARE INPUT; IF NC = 1 THE NEXT CARD GROUP IS NO. 25

20	1-10	WI(1)	Velocity of first particle group in stream-line direction (cm/sec)	E10.3
	11-20	WI(2)	Velocity of second particle group in streamline direction (cm/sec)	E10.3

		WI(NPG)	Velocity of NPG particle group in stream-line direction (cm/sec)	E10.3
21	1-10	VI(1)	Velocity of first particle group normal to streamline (cm/sec)	E10.3
	11-20	VI(2)	Velocity of second particle group normal to streamline (cm/sec)	E10.3

		VI(NPG)	Velocity of NPG particle group normal to streamline (cm/sec)	E10.3
22	1-10	TPI(1)	Temperature of first particle group ($^{\circ}$ K)	E10.3
	11-20	TPI(2)	Temperature of second particle group ($^{\circ}$ K)	E10.3

		TPI(NPG)	Temperature of NPG particle group ($^{\circ}$ K)	E10.3
23	1-10	RHP(1)	Particle cloud density of second particle group (g/cm^3)	E10.3
	11-20	RHPI(2)	Particle cloud density of second particle group (g/cm^3)	E10.3

		RHPI(NPG)	Particle cloud density of NPG particle group (g/cm^3)	E10.3

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
-----------------	--------------	---------------------	--------------------	---------------

NOTE: IF A PARTICLE GROUP IS AT THE SOLIDIFICATION TEMPERATURE THE AMOUNT OF HEAT WHICH HAS BEEN TRANSFERRED FROM THE LIQUID PARTICLES (i. e. SOME FRACTION OF THE TOTAL HEAT OF SOLIDIFICATION) IS INPUT ON CARD 24. IF THE PARTICLE TEMPERATURE IS ABOVE OR BELOW THE SOLIDIFICATION TEMPERATURE, DENG(J) = 0 (J = 1, NPG)

24	1-10	DENGI(1)	Heat transferred from first particle group at solidification temperature (cal)	E10.3
	11-20	DENGI(2)	Heat transferred from second particle group at solidification temperature (cal)	E10.3

		DENGI(NPG)	Heat transferred from NPG particle group at solidification temperature (cal)	E10.3

NOTE: IF PARTICLE PROPERTIES ALONG INITIAL ORTHOGONAL SURFACE ARE VARIABLE (NC = 1) CARDS 25-32 ARE INPUT (SEE FIG. 2)

25.1	1-10	W(1,1)	Streamwise velocity of first particle group at r = 0 (cm/sec)	E10.3
	11-20	W(1,2)	Streamwise velocity of second particle group at r = 0 (cm/sec)	E10.3

		W(1,NPG)	Streamwise velocity of NPG particle group at r = 0 (cm/sec)	E10.3
26.1	1-10	V(1,1)	Normal velocity of first particle group at r = 0 (cm/sec); generally = 0	E10.3
	11-20	V(1,2)	Normal velocity of second particle group at r = 0 (cm/sec); generally = 0	E10.3

		V(1,NPG)	Normal velocity of NPG particle group at r = 0 (cm/sec); generally = 0	E10.3

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
27.1	1-10	TP(1,1)	Temperature of first particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
	11-20	TP(1,2)	Temperature of second particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
	.	.		
	.	.		
		TP(1,NPG)	Temperature of NPG particle group at $r = 0$ ($^{\circ}\text{K}$)	E10.3
28.1	1-10	RHP(1,1)	Particle cloud density of first particle group at $r = 0$ (g/cm^3)	E10.3
	11-20	RHP(1,2)	Particle cloud density of second particle group at $r = 0$ (g/cm^3)	E10.3
	.	.		.
	.	.		.
		RHP(1,NPG)	Particle cloud density of NPG particle group at $r = 0$ (g/cm^3)	E10.3
29.1	1 -5	ICOND(1,1)	Index which indicates whether first particle group is at solidification temperature at $r = 0$ ICOND = 0; No = 1; Yes	I5
	6 -10	ICOND(1,2)	Same as above for second particle group	I5

		ICOND(1,NPG)	Same as above for NPG particle group	I5
30.1	1-10	DENG(1,1)	Heat transferred from first particle group at $r = 0$ at solidification temperature (cal) (See NOTE on page)	E10.3
	11-20	DENG(1,2)	Same as above for second particle group	E10.3

		DENG(1,NPG)	Same as above for NPG particle group	E10.3
25.2				
26.2				
27.2				
28.2				
29.2				
30.2				
			Same as 25.1, 26.1, 27.1, 28.1, 29.1, and 30.1, except for first streamtube	
			.	
			.	

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>	
25.(NBL-1)	}		.		
26.(NBL-1)			.		
27.(NBL-1)			Same as above except for last stream-tube containing particles		
28.(NBL-1)					
29.(NBL-1)					
30.(NBL-1)					

NOTE: CARDS 31 AND 32 LOCATE THE INITIAL BOUNDARY OF EACH PARTICLE GROUP

31	1-10	TRBDY(1)	Distance from axis, along initial orthogonal surface, to boundary of first particle group (cm) (see Fig. 2)	E10.3
	11-20	TRBDY(2)	Same as above for second particle group	E10.3
	.	.		.
	.	.		.
		TRBDY(NPG)	Same as above for NPG particle group	E10.3
32	1-10	DNDSP(1)	Distance from outer boundary of last streamtube containing particle to boundary of first particle group at initial orthogonal surface (cm) (see Fig. 2)	E10.3
	11-20	DNDSP(2)	Same as above for boundary of second particle group	E10.3
	.	.		.
	.	DNDSP(NPG)	Same as above for boundary of NPG particle group	.

NOTE: THE FOLLOWING CARDS CONTAIN THE REACTION MECHANISM AND RATE COEFFICIENTS. USE ONLY IF IKINE (CARD 5, COLS. 41-45) IS GREATER THAN 0. (SEE SECTION II.C.2 FOR REACTION AND RATE COEFFICIENT TYPES)

33.1	1- 4	IZD(1)	Species A	A4
	7		+ sign	
	8-11	IZD(2)	Species B	A4
	14		+ sign	
	15-20		Blank or M	A4

<u>Card No.</u>	<u>Cols.</u>	<u>Fortran Name</u>	<u>Description</u>	<u>Format</u>
33.1	21		= sign	
	22-25	IZD(3)	Species C	A4
	28		+ sign (if needed)	
	29-32	IZD(4)	Species D	A4
	35		+ sign (if needed)	
	36-39	IZD(5)	Species E	A4
	49-50	IRR	Reaction type (1 to 10)	I2
	51	IRT	Rate coefficient type (1 to 8)	I1
	52-59	RC(1)	Pre-exponential factor, A (cm-molecule-sec units)	E8.2
	60-63	RC(2)	Temperature exponent, N	F4.1
	64-72	RC(3)	Activation energy, B (cal/mole)	F9.1
.				
.				
.				

33. IKINE

Same as above for IKINE reaction

IV. PRELIMINARY RESULTS

This section gives the results of several sample calculations made with FULLNOZ and an analysis performed to determine heterogeneous electron recombination rates in solid propellant nozzle flows.

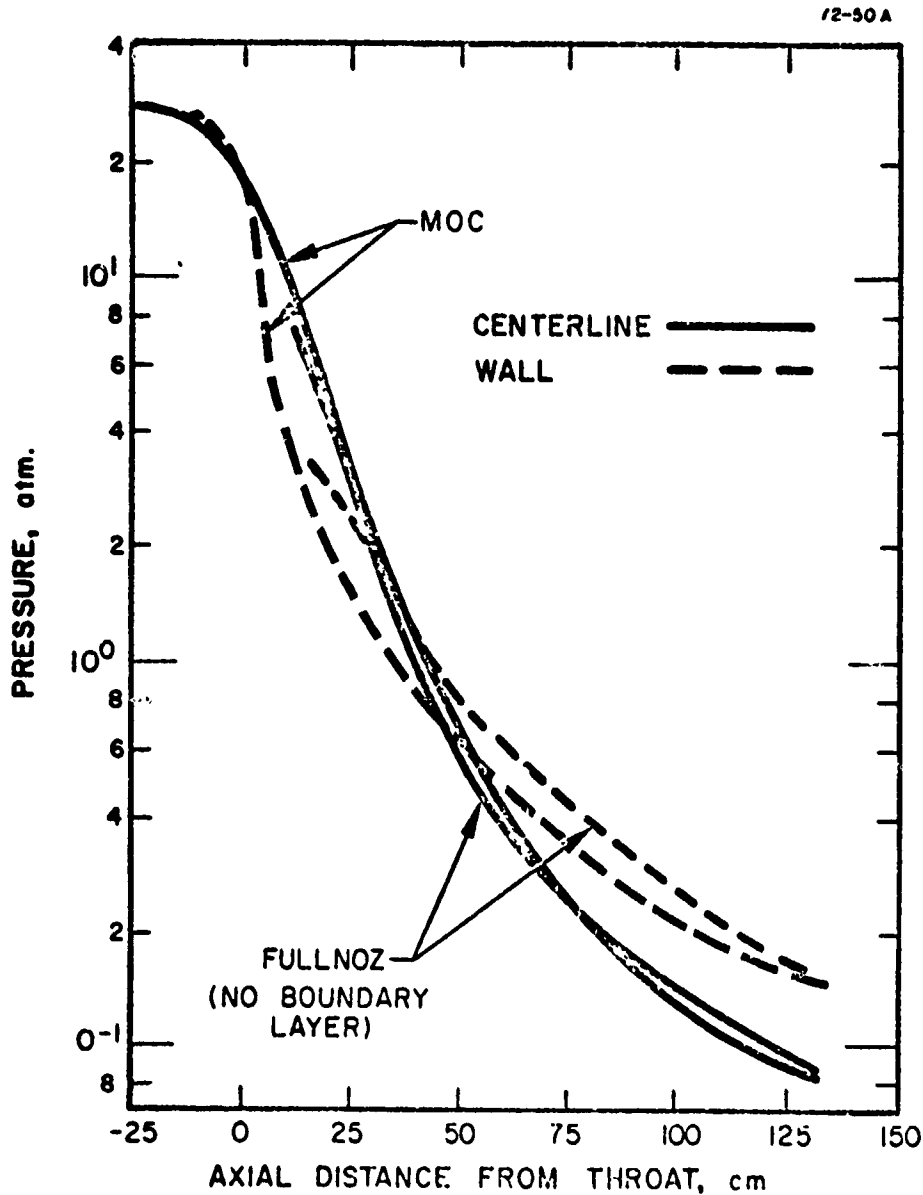


Figure 3. Comparison between Method of Characteristics (MOC) and FULLNOZ calculations of MM-Stage 2 nozzle pressure distributions.

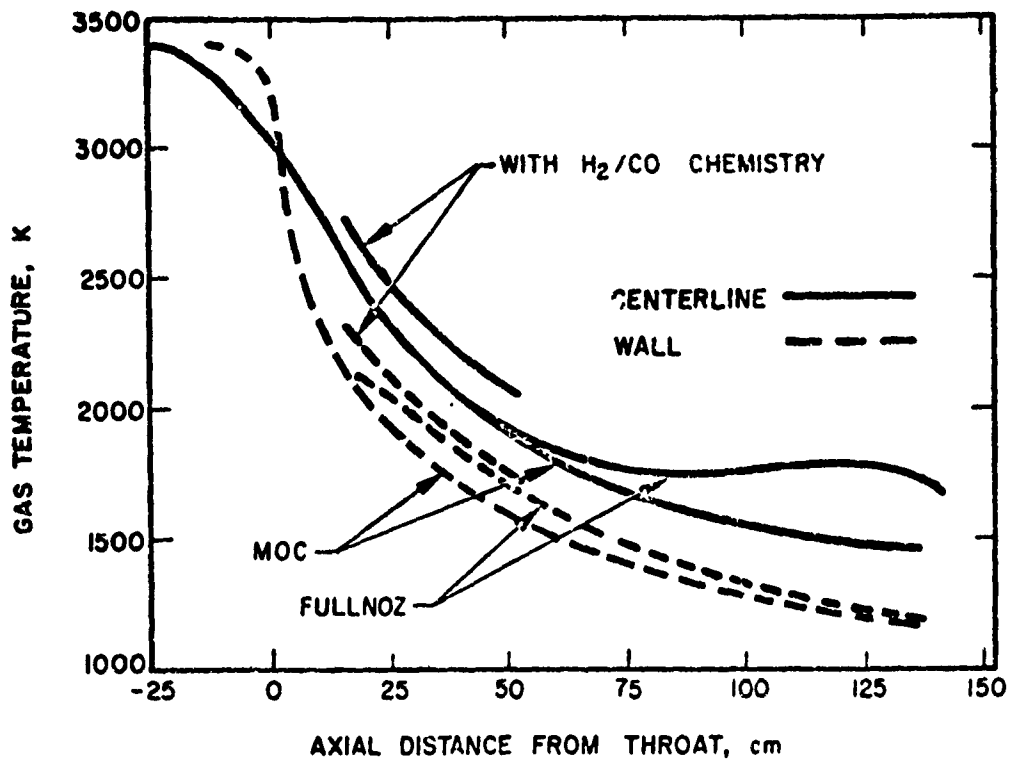


Figure 4. Comparison between Method of Characteristics (MOC) and FULLNOZ calculations of MM-Stage 2 nozzle gas temperature distributions.

A. Sample Calculations

Initial calculations with FULLNOZ were compared with calculations for the Minuteman, Stage 2 nozzle using a two-phase constant γ method of characteristics (MOC) code.³ The purpose of this comparison was to check the numerical accuracy of the code. Input data for the calculations are given in Ref. 23.

Figures 3 and 4 show the pressure and temperature distributions along the centerline and wall (with no boundary layer effects). The pressure distributions compare very well, but FULLNOZ temperatures are slightly higher than those calculated via the MOC code. The centerline gas temperature calculated with FULLNOZ shows a more pronounced effect of gas/parti-

23. Pergament, H.S. and Mikatariyan, R.R., "Predictions of Minuteman Exhaust Plume Electrical Properties," AeroChem TP-281, July 1972.

cle interactions than was demonstrated by the MOC code. The results of a short run with FULLNOZ, including a set of 10 reactions involving H_2/CO chemistry, (see Ref. 23 for the reaction mechanism and rate coefficients) are also shown on Fig. 4.

Figures 5 and 6 show the influence of boundary layer heat transfer and shear stress on the temperature and velocity within the wall streamtube. The largest effects are observed for the cold ($500^\circ K$) wall; when the wall is near the adiabatic wall temperature ($\approx 3000^\circ K$) the results are similar to those for no boundary layer.

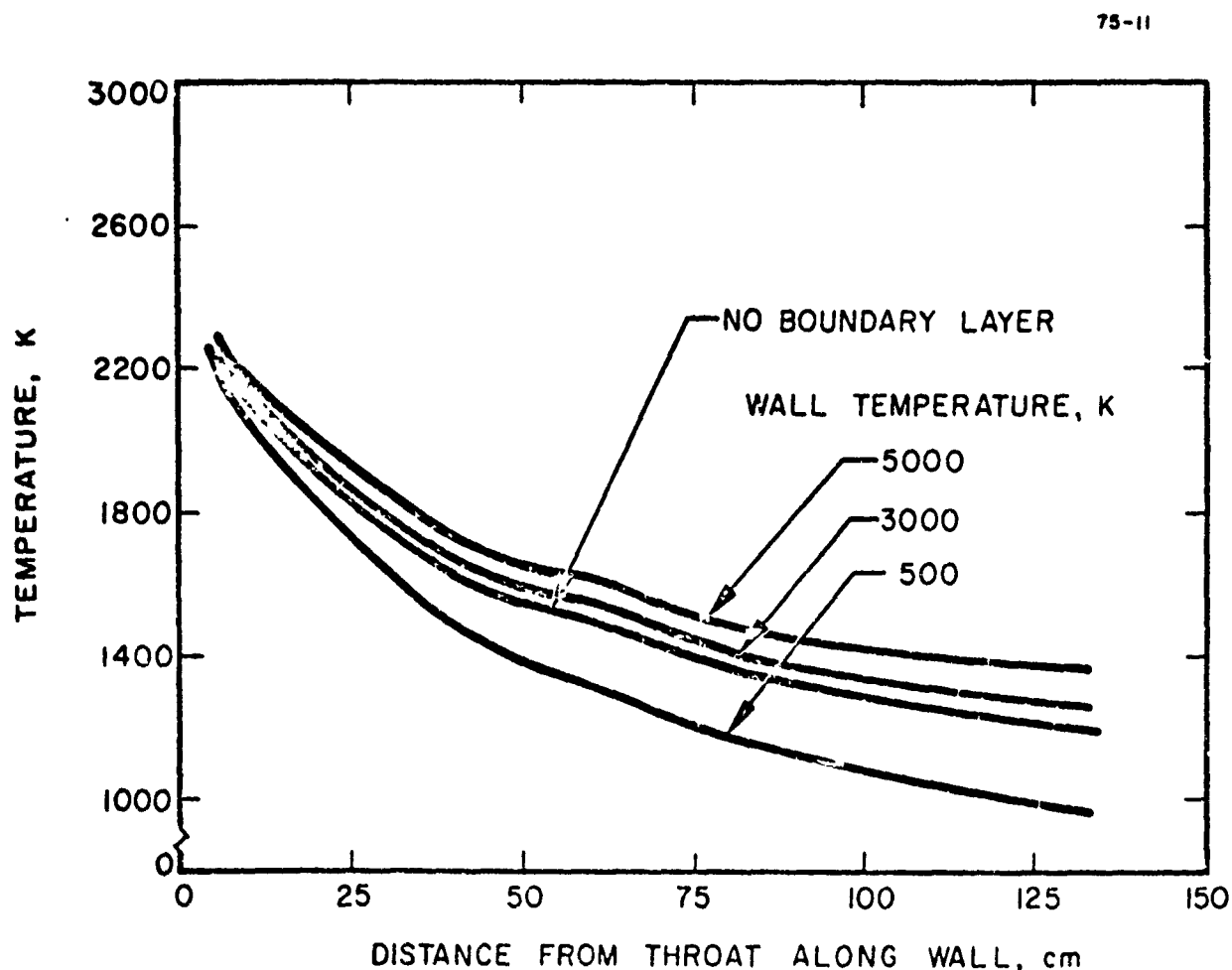


Figure 5. Influence of nozzle boundary layer on wall streamtube temperature.

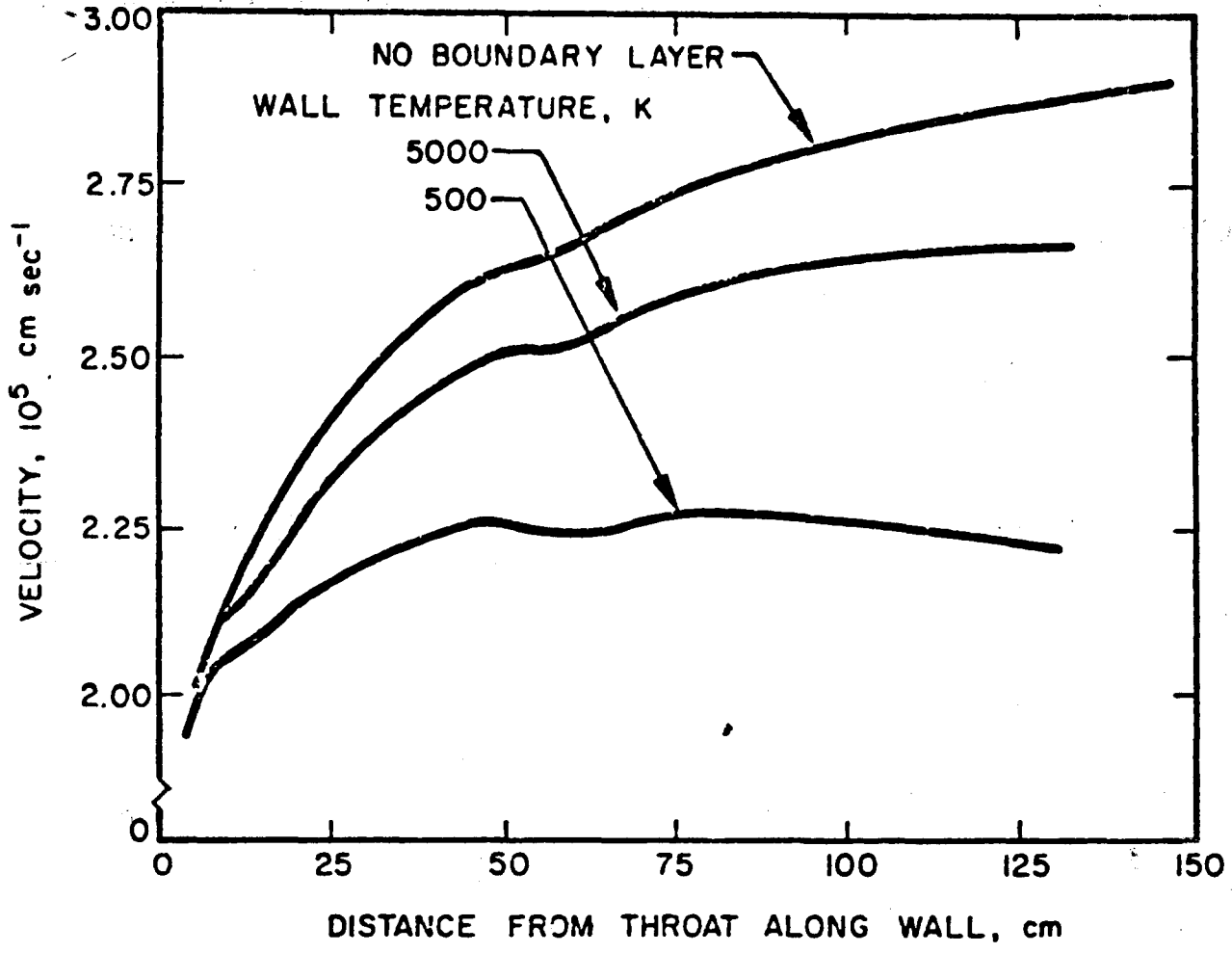


Figure 6. Influence of nozzle boundary layer on wall streamtube velocity.

Figure 7 shows the boundary layer velocity and temperature profile at the nozzle exit plane; Table 1 demonstrates that the boundary layer displacement and momentum thicknesses are much less than the wall streamtube thickness, i. e. all boundary layer effects are confined to the wall streamtube.

Figures 8 and 9 show the results of a parametric series of calculations in which the gas/particle drag and heat transfer coefficients were arbitrarily varied (via the factors FFF and FFG on Card 15) over their approximate ranges of uncertainty to test the effect on exit plane and particle properties. Figure 8 shows that varying FFF and FFG can have significant effects on gas temperatures and velocities. Figure 9 demonstrates that the 4μ diameter particles can be either at the solidification temperature or completely solidified, depending on the value chosen for FFG.

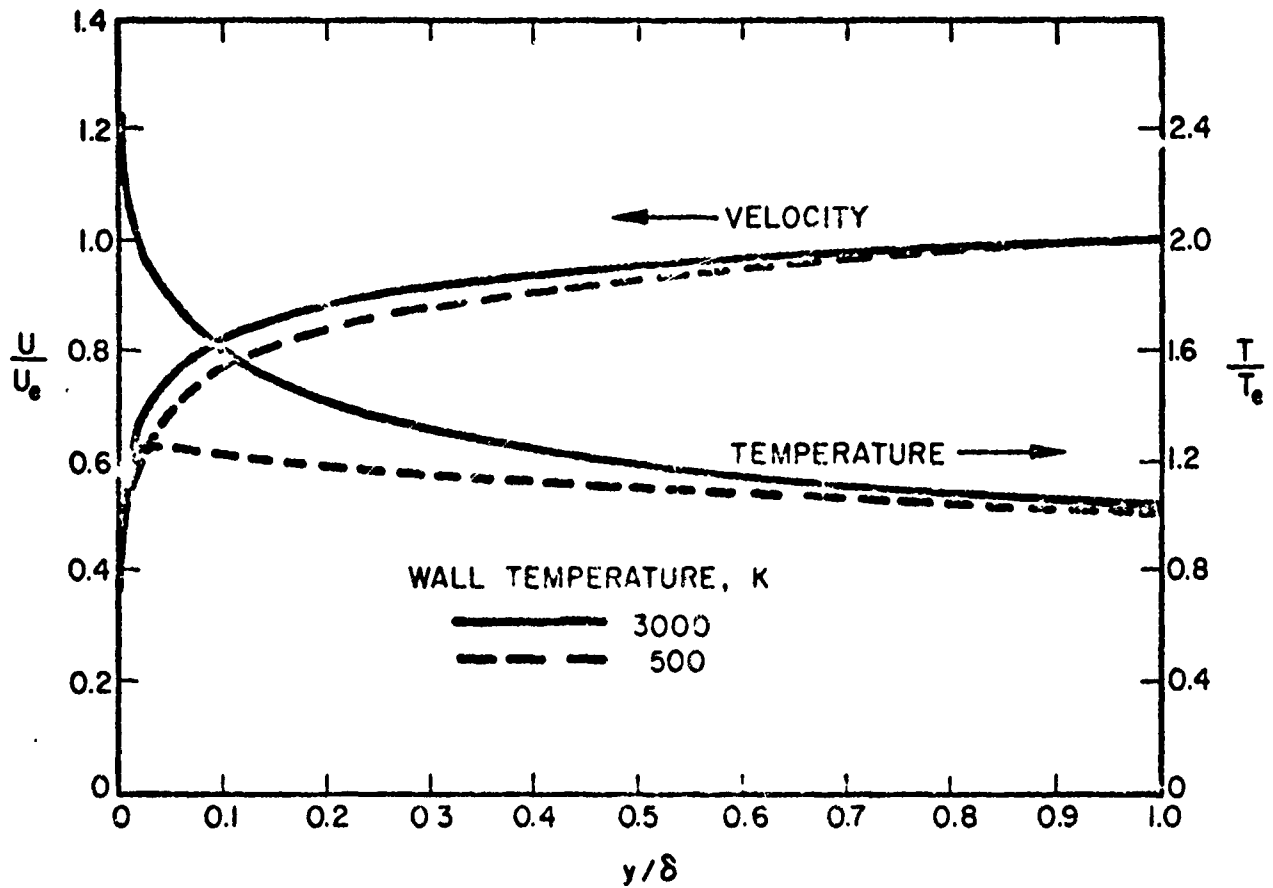


Figure 7. Boundary layer profiles at MM-Stage 2 nozzle exit plane.

TABLE 1

NOZZLE EXIT PLANE BOUNDARY LAYER PARAMETERS

MM-Stage 2 (Ref. 23)

Nozzle Exit Radius = 60.9 cm

Axial Distance From Throat = 134 cm

	Wall Temperature, °K	
	500	3000
Wall Streamtube Width, cm	1.24	1.38
Boundary Layer Thickness, cm	4.45	2.50
Displacement Thickness, cm	0.451	0.196
Momentum Thickness, cm	0.355	0.161

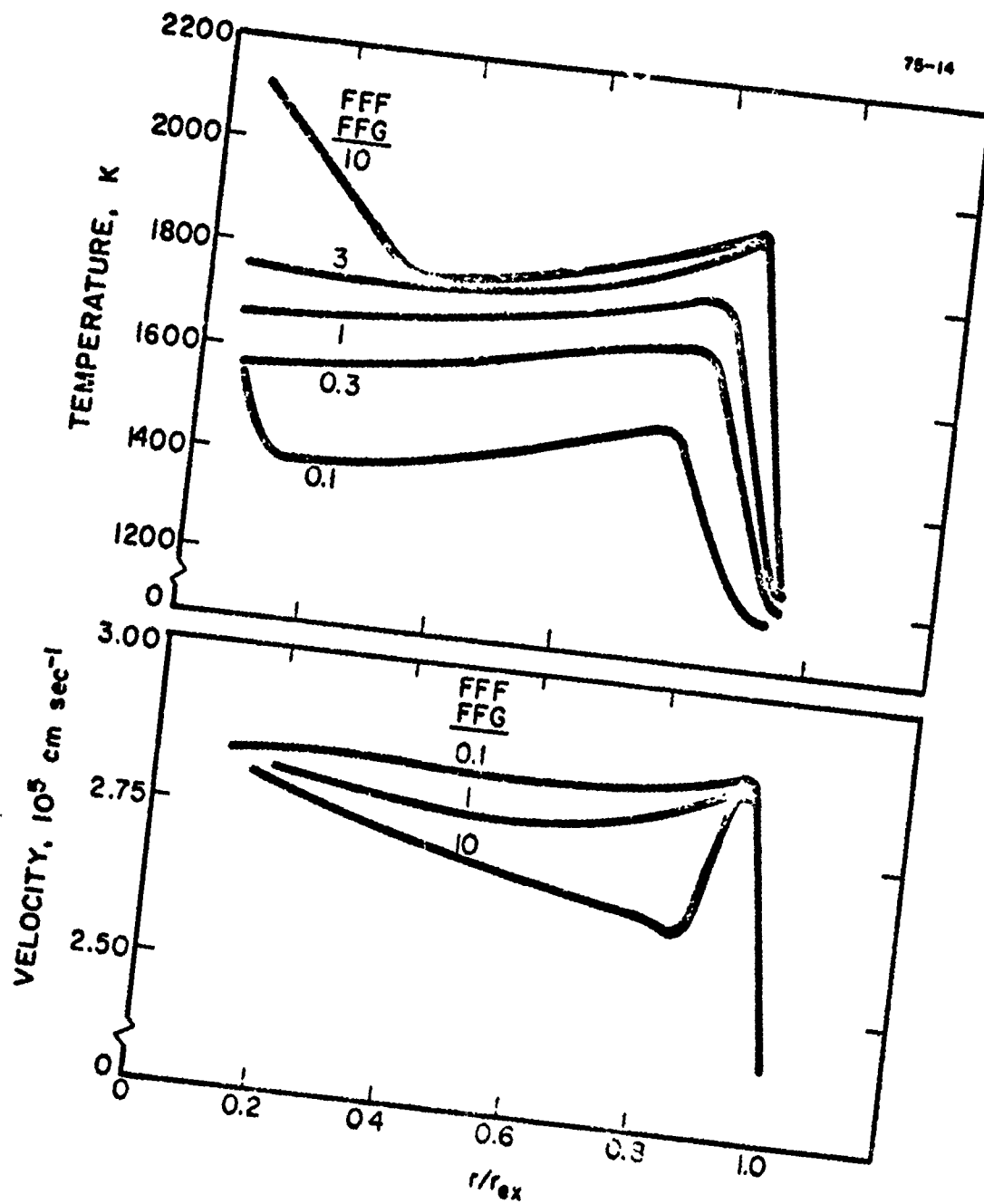


Figure 8. Influence of particle drag and heat transfer coefficients on exit plane gas properties.

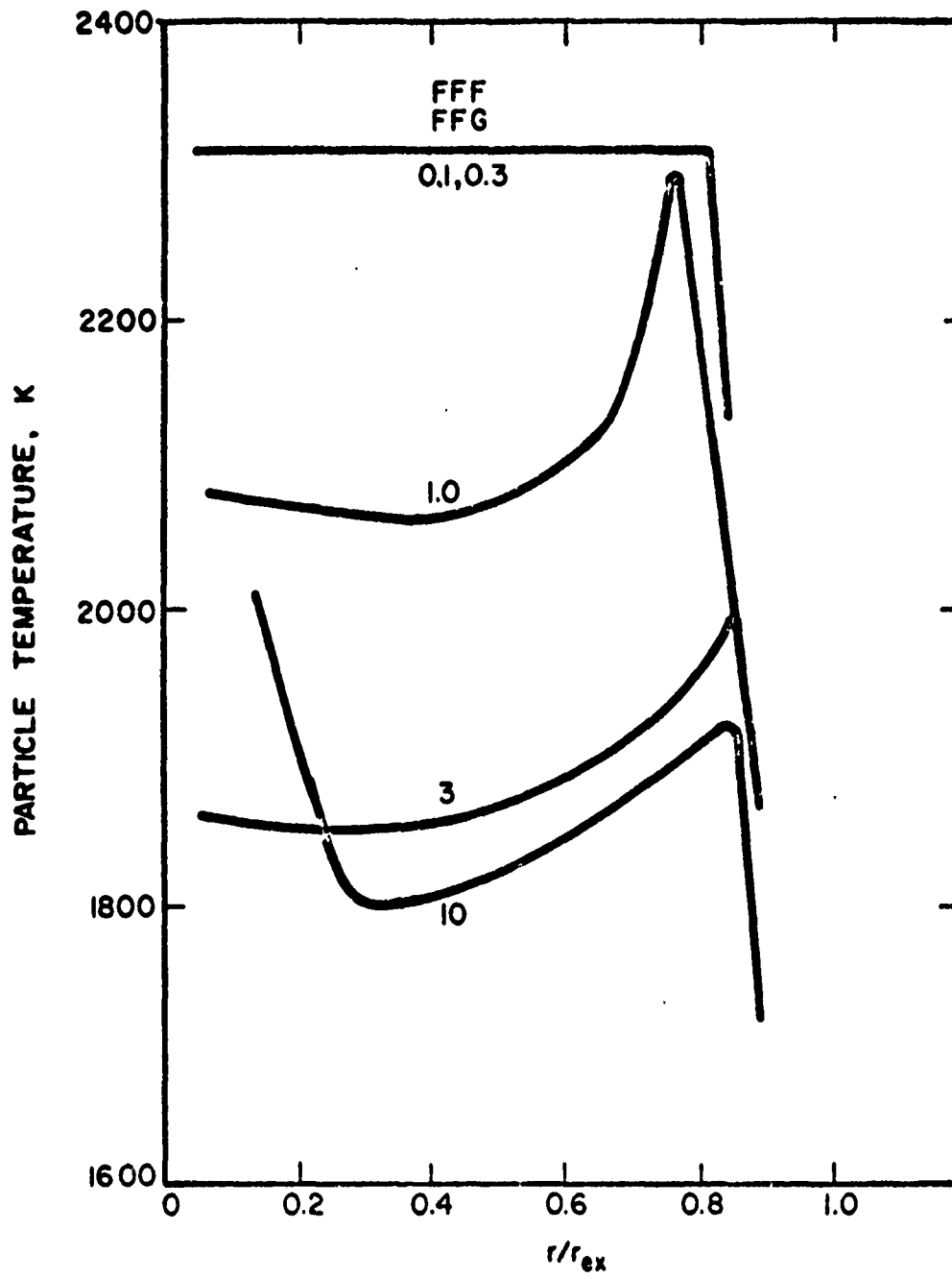


Figure 9. Influence of particle drag and heat transfer coefficients on exit plane particle temperatures.

B. Heterogeneous Electron Ion Recombination

One of the unanswered questions in determining nozzle exit plane electron mole fractions is the extent to which heterogeneous electron/ion recombination on the surface of solid particles can enhance homogeneous electron/ion recombination rates. Towards this end we have adopted a model developed at AeroChem by Calcote, Kurzius and Silla²⁴ in which a negatively charged particle (particles are negatively charged because the electrons will reach the surface more rapidly than positive ions) is neutralized by positive ions striking the surface. Thus, instead of requiring a three-body collision for recombination, only a two-body collision between the solid particle and positive ion need occur. To a first approximation the particle electron recombination rate can be equated to the rate at which positive ions strike the solid particles. However, if negative ions are present (for typical solid propellants mole fractions of Cl^- are from 2 to 3 orders of magnitude greater than electron mole fractions) the electron recombination rate will decrease, since some of the positive ions will react with negative ions rather than electrons.

The solid particle electron recombination coefficient, α_{pe} , is defined from

$$\left(\frac{dn_e}{dt}\right)_p = -\alpha_{pe} n_+ n_p \quad (71)$$

where n_e is electron density, t is time, n_+ is positive ion density and n_p is the particle number density. If particle diameters are small compared to the gas mean free path[†] (free molecular flow) and the electron densities are sufficiently high that the particles remain negatively charged, α_{pe} is essentially the random ion flux to the particle (with a correction factor for negative ions),

$$\alpha_{pe} = \pi r^2 \left(\frac{8kT}{\pi m_+}\right)^{1/2} \left[1 + \left(\frac{n_-}{n_e}\right) \left(\frac{m_e}{m_-}\right)^{1/2}\right]^{-1} \quad (72)$$

[†] If particle diameters are not small compared to the mean free path corrections will have to be made to the expression for α_{pe} .

24. Calcote, H. F., Kurzius, S. C., and Silla, H., "Solid Propellant Flame Ionization and the Effect of Chemical Additives," Third Radar Attenuation Symposium, CPIA Publ. No. 46 (Applied Physics Lab., Johns Hopkins Univ., Silver Spring, 1964), pp. 17-40.

where r is the particle radius, k is the Boltzmann constant, T is the gas temperature, m_+ , m_e , and m_- are the masses of positive ions, electrons and negative ions, respectively, and n_- is the negative ion density.

Equations (71) and (72) represent the formal method for incorporating heterogeneous electron/ion recombination into FULLNOZ. However, we must also account for the possibility that, at the high temperatures near the nozzle throat, the particle is emitting electrons via thermionic emission. Under these conditions a steady state is achieved by balancing the random current density to the particle (from the plasma) by thermionic emission. Equating these electron currents results in a "critical" electron density for which the net current flow to the particle is zero and an initially neutral particle will remain neutral. This critical electron density is defined by,

$$(n_e)_{cr} = B \left(\frac{T_p}{T_g} \right) T_g^{3/2} \exp \left[-11,605 E_w / T_p \right] \quad (73)$$

where B is the thermionic emission constant, T_p is the particle temperature, T_g is the gas temperature and E_w is the effective work function of the particle in volts. Thus, heterogeneous electron/ion recombination is only of potential importance for $n_e > (n_e)_{cr}$.

The technique adopted to incorporate the above equations into FULLNOZ is:

1. At each integration step determine whether the local electron density is greater than the critical electron density. (During the initial stages of the expansion process, where particle temperatures are very high it is likely that $n_e < (n_e)_{cr}$.)
2. If $n_e < (n_e)_{cr}$ then electron/ion recombination will not be significant and Eqs. (71) and (72) will not be employed.
3. If $n_e > (n_e)_{cr}$ Eqs. (71) and (72) will be incorporated directly into the general kinetic scheme (with possible corrections to Eq. (72) due to non-free molecular flow effects).

The above procedure has not as yet been incorporated into the code.

V. REFERENCES

1. "ICRPG Two-Dimensional Kinetic (TDK) Nozzle Analysis Computer Program," Dynamic Science Corp., December 1973 (revised version).
2. Hoffman, R.J., English, W.D., Oeding, R.G., and Webber, W.T., "Plume Contamination Effects Prediction: The CONTAM Computer Program," Final Report, Air Force Rocket Propulsion Laboratory, AFRPL-TR-71-109, December 1971.
3. Nickerson, G.R. and Kliegel, J.R., "Axisymmetric Two-Phase Perfect Gas Performance Program," TRW Systems Report No. 02874-6006-R000, Vols. I and II, April 1967.
4. Penn, M.M. and Smith, S.D., "Supersonic Gas-Particle Flows, Including Reacting Chemistry," JANNAF 8th Plume Technology Meeting, Colorado Springs, CO, July 1974.
5. Boynton, F.P., "The MULTITUBE Supersonic Flow Computer Code," General Dynamics/Convair GDC-DBB 67-003, February 1967.
6. Boynton, F.P. and Thomson, A., "Numerical Computation of Steady, Supersonic, Two-Dimensional Gas Flow in Natural Coordinates," J. Computational Phys. 3, 379-398 (1969).
7. Van Driest, E.R., "Turbulent Boundary Layer in Compressible Fluids," J. Aeron. Sci. 18, 145-160 (1951).
8. Keener, E.R. and Hopkins, E.J., "Van Driest Generalization Applied to Turbulent Skin Friction and Velocity Profiles Measured on the Wall of a Mach 7.4 Wind Tunnel," AIAA J. 11, 1784-1785 (1973).
9. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program. (The AIPP Code). Part I. Analytical and Numerical Techniques," AeroChem TP-302a, AFPRL-TR-74-59, November 1974.
10. Pergament, H.S. and Kelly, J.T., "A Fully-Coupled Underexpanded Rocket Plume Program (The AIPP Code). Part II. Program User's Manual," AeroChem TP- (in preparation).
11. Mikatarian, R.R., Kau, C.J., and Pergament, H.S., "A Fast Computer Program for Nonequilibrium Rocket Plume Predictions," Final Report, AeroChem TP-282, AFRPL-TR-72-94, August 1972.

12. Cruise, D.R., "Information Manual for the Theoretical Propellant Evaluation Program," Naval Weapons Center PEP NOTE TN-U-1 (plus additions), December 1964.
13. JANNAF Thermochemical Tables (Dow Chemical Company, Midland, Mich.), continuously updated.
14. Jensen, D.E. and Jones, G.A., "Gas-Phase Reaction Rate Coefficients for Rocketry Applications," Rocket Propulsion Establishment Technical Report No. 71/9, October 1971.
15. Soo, S.L., Fluid Dynamics of Multiphase Systems (Blaisdell Publ. Co, Waltham, Mass., 1967).
16. Crowe, C.T., "On the Momentum and Heat Transfer Equations for Two-Phase Plumes," Washington State Univ., March 1971.
17. Schlichting, H., Boundary Layer Theory, 6th Ed. (McGraw-Hill, New York, 1968), p. 599.
18. Herring, H.J. and Mellor, G.L., "A Method of Calculating Compressible Turbulent Boundary Layers," NASA CR-1144, September 1968.
19. Tong, H., Buckingham, A.C., and Morse, H.L., "Nonequilibrium Chemistry Boundary Layer Integral Matrix Procedure," Aerotherm Final Report No. 73-67, July 1973.
20. Hopkins, E.J., Keener, E.R., and Louie, P.T., "Direct Measurements of Turbulent Skin Friction on a Nonadiabatic Flat Plate at Mach Number 6.5 and Comparisons with Eight Theories," NASA TN D-5675, February 1970.
21. Back, L.H. and Cuffel, R.F., "Relationship Between Temperature and Velocity Profiles in a Turbulent Boundary Layer along a Supersonic Nozzle with Heat Transfer," AIAA J. 8, 2066-2069 (1970).
22. Kays, W.M., Convective Heat and Mass Transfer (McGraw-Hill, New York, 1966).
23. Pergament, H.S. and Mikatarian, R.R., "Predictions of Minuteman Exhaust Plume Electrical Properties," AeroChem TP-281, July 1972.
24. Calcote, H.F., Kurzius, S.C., and Silla, H., "Solid Propellant Flame Ionization and the Effect of Chemical Additives," Third Radar Attenuation Symposium, CPIA Publ. No. 46 (Applied Physics Lab., Johns Hopkins Univ., Silver Spring, 1964), pp. 17-40.

APPENDIX A

SAMPLE INPUT DATA

CARD NO

1
2 *** FULLY COUPLED NOZZLE FLOW PROGRAM - FULLNOZ - SAMPLE TEST CASE 9/75 ***
3 7200 2.000E+2 1.524E+2
4 1 1 0 1 0
5 25 6 10 2 22 10 50 900 10
6 1000 1 1
7 0.8 0.01 1.0E-4 1.0 0.1
8.11 64108562 E+1 18575287 E+1 44865856E+0 36280951 E-1 CO 1
2 -5981864 E-2 28443584E+2 54410327 E+2 CO 2
8.21 38741331E+1 1.13592404E+02 -86056609E1 21500497E+1 CO2 1
2 -30567063E+1 -96439872E+2 54321459E+2 CO2 2
8.31 49683629 E+1 -23301268E-3 59667528 E-4 -51872545E-5 H 1
2 -18467323E-4 50618655 E+2 33404654 E+2 H 2
8.41 60596451 E+1 13412152 E+1 16508752E+0 88760457 E-2 H2 1
2 46616242 E-1 -17005029E+1 38443271 E+2 H2 2
8.51 58818265 E+1 17395375 E+1 32536218E+0 22313547 E-1 OH 1
2 75177541 E-1 79212009 E+1 50938996 E+2 OH 2
8.61 60311014 E+1 47049758 E+1 96291245E+0 69087996 E-1 H2O 1
2 60421152 E-1 59593534E+2 51387265 E+2 H2O 2

CARD NO

8.71 62643469 E+1 19328092 E+1 46158274E+0 37083135 E-1 N2 1
2 91852768 E-2 19242408E+1 52824774 E+2 N2 2
8.81 50617760 E+1 11736656E+0 39405358 E-1 -20531577E-2 O 1
2 18127099 E-1 58115304 E+2 44728172 E+2 O 2
8.91 72417589 E+1 12398122 E+1 18448469E+0 10507580 E-1 O2 1
2 59604067E-1 24219968E+1 57031740 E+2 O2 2
8.101 48485464 E+1 32006164 E+0 28545471E+0 83218696 E-1 K 1
2 43725018 E-2 19867370 E+2 44106314 E+2 K 2

91 CO	139. E-6	300.	0.75	1.4	1.4	28.
2 CO2	139. E-6	300.	0.75	1.4	1.4	44.
3 H	83.5 E-6	300.	0.75	1.4	1.4	1.
4 H2	83.5 E-6	300.	0.75	1.4	1.4	2.
5 OH	125.5 E-6	300.	0.75	1.4	1.4	17.
6 H2O	125.5 E-6	300.	0.75	1.4	1.4	18.
7 N2	160. E-6	300.	0.75	1.4	1.4	28.
8 O	166. E-6	300.	0.75	1.4	1.4	16.
9 O2	189. E-6	300.	0.75	1.4	1.4	32.
10 K	129.7 E-6	300.	0.75	1.4	1.4	39.1

~NOTE~ CARD 10 OMITTED FOR ITURB=0(CARD 4, COL 56-60)



FORTRAN Coding Form

FULLNOZ									3	14
CARD NO	FORTRAN STATEMENT									
11	12.668	0.0	0.0							
12.11	12.663	0.6053	0.01804	8.500	2761.	1.534E+5				
.2	3.0E-1	6.0E-2	6.3E-4	2.4E-2	2.6E-3	1.9E-1	4.2E-1	7.8E-5		
3	5.6E-5	1.1E-6								
12.21	12.646	1.2110	0.03607	8.473	2759.	1.534E+5				
2	NOTE	CARDS	12.22	TO 12.24.2	ARE IDENTICAL TO CARD	12.1.2				
3		CARDS	12.23	TO 12.24.3	ARE IDENTICAL TO CARD	12.1.3				
12.31	12.619	1.8158	0.05411	8.443	2758.	1.534E+5				
.2	SEE NOTE									
3										
12.41	12.581	2.4199	0.07214	8.400	2757.	1.536E+5				
2	SEE NOTE									
3										
12.51	12.532	3.0238	0.09018	8.343	2755.	1.538E+5				
2	SEE NOTE									
3										
12.61	12.472	3.6262	0.10821	8.273	2751.	1.541E+5				
.2	SEE NOTE									
3										



FORTRAN Coding Form

FULLNOZ									4	14
CARD NO	FORTRAN STATEMENT									
12.71	12.401	4.2274	0.12625	8.193	2745.	1.555E+5				
3	SEE NOTE									
12.81	12.319	4.8278	0.14428	8.103	2739.	1.550E+5				
2	SEE NOTE									
3										
12.91	12.227	5.4260	0.16232	8.003	2732.	1.556E+5				
.2	SEE NOTE									
3										
12.01	12.123	6.0225	0.18035	7.895	2726.	1.562E+5				
2	SEE NOTE									
3										
12.11	12.009	6.6176	0.19839	7.775	2719.	1.570E+5				
2	SEE NOTE									
3										
12.12	11.885	7.2100	0.21642	7.643	2712.	1.578E+5				
2	SEE NOTE									
3										



FORTRAN Coding Form

FULL NOZ

5 14

CARD No

FORTRAN STATEMENT

12131	11.749	7.8000	0.23446	7.498	2708.	1.588E+5
.2	SEE NOTE					
.3						
12141	11.603	8.3880	0.25249	7.328	2700.	1.600E+5
.2	SEE NOTE					
.3						
12151	11.447	8.9728	0.27053	7.138	2688.	1.613E+5
.2	SEE NOTE					
.3						
12161	11.280	9.5546	0.28856	6.935	2671.	1.629E+5
.2	SEE NOTE					
.3						
12171	11.102	10.1340	0.30660	6.715	2648.	1.647E+5
.2	SEE NOTE					
.3						
12181	10.914	10.7090	0.32463	6.480	2624.	1.670E+5
.2	SEE NOTE					
.3						



FORTRAN Coding Form

FULL NOZ

6 14

CARD No

FORTRAN STATEMENT

12191	10.716	11.2810	0.34267	6.225	2594.	1.676E+5
.2	SEE NOTE					
.3						
12201	10.507	11.8500	0.36070	5.950	2563.	1.725E+5
.2	SEE NOTE					
.3						
12211	10.289	12.4150	0.37874	5.680	2550.	1.761E+5
.2	SEE NOTE					
.3						
12221	10.060	12.9750	0.39677	5.390	2533.	1.798E+5
.2	SEE NOTE					
.3						
12231	9.8200	13.5300	0.41481	5.065	2502.	1.847E+5
.2	SEE NOTE					
.3						
12241	9.5715	14.0840	0.43284	4.602	2401.	1.904E+5
.2	SEE NOTE					
.3						



FULLNOZ

7 14

CARD NO

FORTRAN STATEMENT

13.1	12.668	0.0							
14.1	9.5715	14.084							
15.1	9.957	14.275							
16.1	10.312	14.503							
17.1	11.582	15.215							
18.1	12.852	15.850							
19.1	14.122	16.434							
20.1	16.662	17.729							
21.1	19.202	18.999							
22.1	21.742	20.218							
23.1	26.822	22.606							
24.1	31.902	24.994							
25.1	36.982	27.305							
26.1	44.602	30.683							
27.1	52.222	33.985							
28.1	59.842	37.084							
29.1	73.914	42.062							
30.1	87.173	46.838							
31.1	99.095	50.749							
32.1	111.150	54.508							
33.1	123.490	58.039							
34.1	134.420	60.909							
35.1	152.400	65.634							



FULLNOZ

8 14

CARD NO

FORTRAN STATEMENT

15.1	1.0	1.0	0.339	0.323	255.	102.			
16.1	4.0			2320.					
17.1	4.	1.							
18.1	19.	19.	19.	19.					
19.1	1.0E-4	2.0E-4	3.0E-4	4.0E-4					

NC(CARD 17, COL. 10) = 1, THEREFORE CARD NO 25 IS NEXT

25.1	0.147 E+6	0.139 E+6	0.131 E+6	0.124 E+6					
26.1	-0.276 E+3	-0.559 E+3	-0.768 E+3	-0.101 E+4					
27.1	2813.	2893.	2913.	2973.					
28.1	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4					
29.1	0.	0.	0.	0.					
30.1	0.0	0.0	0.0	0.0					

25.2	0.147 E+6	0.139 E+6	0.132 E+6	0.125 E+6					
26.2	-0.292 E+3	-0.592 E+3	-0.814 E+3	-0.107 E+4					
27.2	2813.	2893.	2913.	2973.					
28.2	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4					
29.2	0.	0.	0.	0.					
30.2	0.0	0.0	0.0	0.0					

NOTE: CARDS 25.2 TO 29.18 ARE IDENTICAL TO 29.1
CARDS 30.2 TO 30.18 ARE IDENTICAL TO 30.1

25.3	0.148 E+6	0.139 E+6	0.132 E+6	0.125 E+6					
26.3	0.310 E+3	-0.625 E+3	-0.864 E+3	-0.114 E+4					



FORTRAN Coding Form

NAME	FULLINOZ	DATE		PROGRAM		CLASS		NO.	914
------	----------	------	--	---------	--	-------	--	-----	-----

CARD NO	STATEMENT
273	2813.
28.3	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.3	
30.3	
25.4	0.148 E+6 0.139 E+6 0.132 E+6 0.125 E+6
26.4	0.831 E+3 0.671 E+3 0.922 E+3 0.121 E+4
27.4	2813.
28.4	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.4	
30.4	
25.5	0.148 E+6 0.140 E+6 0.132 E+6 0.125 E+6
26.5	0.356 E+3 0.719 E+3 0.989 E+3 0.130 E+4
27.5	2813.
28.5	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.5	
30.5	
25.6	0.148 E+6 0.140 E+6 0.132 E+6 0.125 E+6
26.6	0.382 E+3 0.775 E+3 0.106 E+4 0.140 E+4
27.6	2813.
28.6	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.6	



FORTRAN Coding Form

NAME	FULLINOZ	DATE		PROGRAM		CLASS		NO.	1014
------	----------	------	--	---------	--	-------	--	-----	------

CARD NO	STATEMENT
30.6	
25.7	0.149 E+6 0.140 E+6 0.132 E+6 0.125 E+6
26.7	0.414 E+3 0.839 E+3 0.115 E+4 0.151 E+4
27.7	2803.
28.7	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.7	
30.7	
25.8	0.149 E+6 0.141 E+6 0.133 E+6 0.126 E+6
26.8	0.451 E+3 0.915 E+3 0.126 E+4 0.165 E+4
27.8	2803.
28.8	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.8	
30.8	
25.9	0.150 E+6 0.141 E+6 0.133 E+6 0.126 E+6
26.9	0.497 E+3 0.107 E+4 0.138 E+4 0.182 E+4
27.9	2803.
28.9	0.279 E-4 0.143 E-3 0.182 E-3 0.554 E-4
29.9	
30.9	



FORTRAN Coding Form

FULLNOZ 11 14

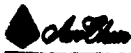
CARD NO	FORTRAN STATEMENT			
25.10	0.150 E+6	0.142 E+6	0.134 E+6	0.126 E+6
26.10	0.552 E+3	0.112 E+4	0.154 E+4	0.202 E+4
27.10	2793.	2873.	2903.	2933.
28.10	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.10				
30.10				
25.11	0.151 E+6	0.143 E+6	0.134 E+6	0.127 E+6
26.11	0.621 E+3	0.126 E+4	0.173 E+4	0.27 E+4
27.11	2793.	2873.	2903.	
28.11	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.11				
30.11				
25.12	0.152 E+6	0.143 E+6	0.135 E+6	0.127 E+6
26.12	0.709 E+3	0.144 E+4	0.198 E+4	0.260 E+4
27.12	2783.	2873.	2893.	2943.
28.12	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.12				
30.12				
25.13	0.153 E+6	0.144 E+6	0.136 E+6	0.128 E+6
26.13	0.828 E+3	0.168 E+4	0.231 E+4	0.303 E+4
27.13	2783.	2873.	2893.	2943.



FORTRAN Coding Form

FULLNOZ 12 14

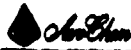
CARD NO	FORTRAN STATEMENT			
28.13	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.13				
30.13				
25.14	0.154 E+6	0.145 E+6	0.136 E+6	0.128 E+6
26.14	0.993 E+3	0.201 E+4	0.277 E+4	0.363 E+4
27.14	2763.	2853.	2883.	2933.
28.14	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.14				
30.14				
25.15	0.155 E+6	0.146 E+6	0.137 E+6	0.129 E+6
26.15	0.124 E+4	0.252 E+4	0.346 E+4	0.454 E+4
27.15	2763.	2853.	2883.	2933.
28.15	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.15				
30.15				
25.16	0.156 E+6	0.147 E+6	0.138 E+6	0.130 E+6
26.16	0.166 E+4	0.336 E+4	0.461 E+4	0.606 E+4
27.16	2743.	2823.	2873.	2923.
28.16	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4
29.16				
30.16				



FORTRAN Coding Form

FULLNOZ						13 14	
---------	--	--	--	--	--	-------	--

CARD NO	FORTRAN STATEMENT							
25.17	0.159 E+6	0.149 E+6	0.139 E+6	0.131 E+6				
26.17	-0.260 E+4	-0.504 E+4	-0.692 E+4	-0.909 E+4				
27.17	2749.	2823.	2873.	2923.				
28.17	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
29.17								
30.17								
25.18	0.159 E+6	0.140 E+6	0.140 E+6	0.132 E+6				
26.18	-0.497 E+4	-0.101 E+5	-0.138 E+5	-0.182 E+5				
27.18	2713.	2793.	2853.	2903.				
28.18	0.279 E-4	0.143 E-3	0.182 E-3	0.554 E-4				
29.18								
30.18								
31	10.7	10.7	10.7	10.7				
32	0.0	0.0	0.0	0.0				



FORTRAN Coding Form

FULLNOZ						14 14	
---------	--	--	--	--	--	-------	--

CARD NO	FORTRAN STATEMENT							
33.10	+0	+M	=D2	+M	21 1 0E-29	0		
20	+H	+M	=OH	+M	22 1 0E-29	1.0		
3H	+H	+M	=H2	+M	22 5 0E-29	1.0		
4H	+OH	+M	=H20	+M	22 2 0E-28	1.0		
5C0	+0	+M	=CO2	+M	24 1 0E-29	1.0	-2500	
6DH	+OH		=OH2	+0	15 1 0E-11		-780	
7DH	+H2		=H20	+H	15 3 6E-11		-5200	
80	+H2		=OH	+H	15 2 9E-11		-9460	
9H	+02		=OH	+0	15 3 7E-10		-16800	
10C0	+OH		=CO2	+H	15 9 0E-13		-1080	

APPENDIX B

SAMPLE OUTPUT

B-i

FULLY COUPLED NOZZLE FLOW PROGRAM

AERUCHEM RESEARCH LABORATORIES, INC.
PRINCETON, NEW JERSEY 08540

*** FULLY COUPLED NOZZLE FLOW PROGRAM - FULLNOZ - SAMPLE TEST CASE 9/75 ***

11. IN DATA SETS 1
MSET= 7200 DALSSE 7.000E+02 DLR= 1.520E+07 DLR= 0.000E+00 IPUS = 0 NUIS = 0

NMAX 25 N 0 NP 10 IP 0 ITPF 1 IRIU 1 NMAX 22 NUS 10 IVO -0 NLEN 50 IUFF 0 LPLAN 900
IPIL 0 IPCM 0 IUMK 0 IUGSM 0 IIO -0 ICSM 0 ITAPD 0 IAME 10 IPANT 1 IKI -0 ITUMD 0 IPTU -0

NETAP 1.0130E+00 HJ 0.1020E+01 H(LIN) 1.0000E+00

ALPHAN 0.000E+00 LPSLON 0.010E+03 IOL 0.000E+03 DELTA 1.0 ATOL 0.100 FSTEP -0.000 GRAD -0.000 FRAC -0.000 FRACTN -0.000

ITER-I	U	V	W	UMEA	PM	SC	CA
CU	1.500E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	2.000E+01
CU2	1.500E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	4.000E+01
N	1.500E+05	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	1.000E+00
M2	1.500E+05	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	2.000E+00
UM	1.500E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	1.700E+01
M2U	1.500E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	1.400E+01
M2	1.000E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	2.000E+01
U	1.000E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	1.000E+01
U2	1.000E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	3.200E+01
K	1.207E+00	3.000E+02	7.500E-01	1.400E+00	1.400E+00	1.400E+00	4.000E+01

K	X	U	V	W	P	T	H
1	1.207E+01	0.	0.	0.	0.	0.	0.
2	1.200E+01	0.053E+01	1.000E+02	0.500E+00	2.701E+03	1.530E+05	1.530E+05
3	1.205E+01	1.211E+00	3.007E+02	0.473E+00	2.759E+03	1.534E+05	1.534E+05
4	1.202E+01	1.010E+00	5.011E+02	0.400E+00	2.750E+03	1.535E+05	1.535E+05
5	1.250E+01	0.400E+00	7.010E+02	0.400E+00	2.757E+03	1.537E+05	1.537E+05
6	1.255E+01	3.000E+00	9.010E+02	0.300E+00	2.750E+03	1.530E+05	1.530E+05
7	1.247E+01	3.000E+00	1.002E+01	0.273E+00	2.751E+03	1.541E+05	1.541E+05
8	1.240E+01	0.227E+00	1.003E+01	0.193E+00	2.705E+03	1.540E+05	1.540E+05
9	1.232E+01	0.020E+00	1.003E+01	0.103E+00	2.730E+03	1.551E+05	1.551E+05
10	1.225E+01	0.000E+00	1.003E+01	0.000E+00	2.732E+03	1.550E+05	1.550E+05
11	1.212E+01	0.000E+00	1.000E+01	7.000E+00	2.720E+03	1.560E+05	1.560E+05
12	1.201E+01	0.010E+00	1.000E+01	7.775E+00	2.710E+03	1.571E+05	1.571E+05
13	1.180E+01	7.010E+00	2.100E+01	7.000E+00	2.712E+03	1.570E+05	1.570E+05
14	1.175E+01	7.000E+00	2.305E+01	7.000E+00	2.700E+03	1.500E+05	1.500E+05
15	1.100E+01	0.300E+00	2.525E+01	7.300E+00	2.700E+03	1.000E+05	1.000E+05
16	1.105E+01	0.973E+00	2.705E+01	7.130E+00	2.600E+03	1.010E+05	1.010E+05
17	1.120E+01	0.505E+00	2.000E+01	0.935E+00	2.071E+03	1.030E+05	1.030E+05
18	1.110E+01	1.013E+01	3.000E+01	0.715E+00	2.000E+03	1.040E+05	1.040E+05
19	1.091E+01	1.071E+01	3.000E+01	0.000E+00	2.020E+03	1.071E+05	1.071E+05
20	1.072E+01	1.120E+01	3.027E+01	0.205E+00	2.500E+03	1.070E+05	1.070E+05
21	1.051E+01	1.105E+01	3.007E+01	5.000E+00	2.500E+03	1.720E+05	1.720E+05
22	1.029E+01	1.042E+01	3.075E+01	5.000E+00	2.550E+03	1.702E+05	1.702E+05
23	1.000E+01	1.000E+01	3.000E+01	5.300E+00	2.533E+03	1.700E+05	1.700E+05
24	0.982E+01	1.053E+01	4.100E+01	5.000E+00	2.502E+03	1.000E+05	1.000E+05
25	0.957E+01	1.000E+01	0.500E+01	4.002E+00	2.001E+03	1.000E+05	1.000E+05

I	X	U	V	W	P	S
1	1.207E+01	0.	0.	0.	0.	0.
2	1.000E+01	0.	0.	0.	0.	0.

A	NO	U	V	W	P	S
1	0.572E+00	1.400E+01	0.	0.	0.	0.
2	0.457E+00	1.020E+01	0.	0.	0.	0.
3	1.031E+01	1.050E+01	0.	0.	0.	0.
4	1.150E+01	1.522E+01	0.	0.	0.	0.
5	1.025E+01	1.505E+01	0.	0.	0.	0.
6	1.012E+01	1.000E+01	0.	0.	0.	0.
7	1.000E+01	1.773E+01	0.	0.	0.	0.
8	1.020E+01	1.000E+01	0.	0.	0.	0.
9	2.174E+01	2.020E+01	0.	0.	0.	0.
10	2.000E+01	2.000E+01	0.	0.	0.	0.
11	3.100E+01	2.000E+01	0.	0.	0.	0.
12	3.000E+01	2.731E+01	0.	0.	0.	0.
13	4.000E+01	3.000E+01	0.	0.	0.	0.
14	5.000E+01	3.300E+01	0.	0.	0.	0.
15	5.000E+01	3.700E+01	0.	0.	0.	0.
16	7.000E+01	4.000E+01	0.	0.	0.	0.
17	0.717E+01	4.000E+01	0.	0.	0.	0.
18	0.000E+01	0.000E+01	0.	0.	0.	0.
19	1.112E+02	0.000E+01	0.	0.	0.	0.
20	1.255E+02	0.000E+01	0.	0.	0.	0.
21	1.300E+02	0.000E+01	0.	0.	0.	0.
22	1.500E+02	0.000E+01	0.	0.	0.	0.

	1st 0.	2nd 0.	3rd 0.	4th 0.	5th 0.	6th 0.	7th 0.
2	As 1.2000E+01 -As .1270E+01 Pis .1710E+02	Ns 0.0530E+01 UCLV .0053E+00 MNS .7000E+03	PHis 1.0000E+02 Ns .1007E+03 Sum 0.	Ts 2.7010E+03 Mts .1000E+03 SUMMITS .1310E+03	Pt 0.5000E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			
3	As 1.2000E+01 -As .1270E+01 Pis .1710E+02	Ns 1.2110E+00 UCLV .0050E+00 MNS .7020E+03	PHis 1.0070E+02 Ns .1057E+03 Sum 0.	Ts 2.7500E+03 Mts .1071E+03 SUMMITS .5250E+03	Pt 0.0730E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			
4	As 1.2010E+01 -As .1270E+01 Pis .1700E+02	Ns 1.0150E+00 UCLV .0050E+00 MNS .7000E+03	PHis 5.0110E+02 Ns .1057E+03 Sum 0.	Ts 2.7500E+03 Mts .1000E+03 SUMMITS .1170E+00	Pt 0.0050E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			
5	As 1.2010E+01 -As .1200E+01 Pis .1700E+02	Ns 2.0100E+00 UCLV .0053E+00 MNS .7300E+03	PHis 7.2100E+02 Ns .1047E+03 Sum 0.	Ts 2.7500E+03 Mts .1000E+03 SUMMITS .2000E+00	Pt 0.0000E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			
6	As 1.2030E+01 -As .1200E+01 Pis .1600E+02	Ns 1.0250E+00 UCLV .0050E+00 MNS .7320E+03	PHis 9.0100E+02 Ns .1057E+03 Sum 0.	Ts 2.7500E+03 Mts .1000E+03 SUMMITS .3050E+00	Pt 0.0000E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			
7	As 1.2040E+01 -As .1200E+01 Pis .1600E+02	Ns 1.0020E+00 UCLV .0050E+00 MNS .7270E+03	PHis 1.0020E+01 Ns .1010E+03 Sum 0.	Ts 2.7510E+03 Mts .1010E+03 SUMMITS .0070E+00	Pt 0.0000E+00 Tans 0.	Us 1.5300E+05 Us 0.	
	SPECIE CU	Cs .3000E+00	Xs .2125E+00	ADITS 0.			
	SPECIE CUD	Cs .0000E+01	Xs .2700E+01	ADITS 0.			
	SPECIE M	Cs .0300E+03	Xs .1249E+01	ADITS 0.			
	SPECIE M2	Cs .2000E+01	Xs .2300E+00	ADITS 0.			
	SPECIE UM	Cs .2000E+02	Xs .3033E+02	ADITS 0.			
	SPECIE M2U	Cs .1900E+00	Xs .2093E+00	ADITS 0.			
	SPECIE M2	Cs .0200E+00	Xs .2975E+00	ADITS 0.			
	SPECIE U	Cs .7000E+00	Xs .9000E+00	ADITS 0.			
	SPECIE U2	Cs .5000E+00	Xs .3071E+00	ADITS 0.			
	SPECIE A	Cs .1100E+05	Xs .5050E+00	ADITS 0.			

8 AX 1.2401E+01 MS 2.2274E+00 PHIS 1.2425E-01 IS 2.7490E+03 PS 8.1930E+00 US 1.5494E+05
 HX .1291E+01 DELTA .6054E+00 MS .9040E+02 HX .3042E+03 TANN 0. US 0.
 PIX .1670E+02 MMUS .7214E+03 SUMMU 0.6341E+04

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

9 AX 1.2519E+01 MS 2.2274E+00 PHIS 1.4424E-01 IS 2.7190E+03 PS 8.1930E+00 US 1.5505E+05
 HX .1291E+01 DELTA .6040E+00 MS .9550E+02 HX .3015E+03 TANN 0. US 0.
 PIX .1659E+02 MMUS .7152E+03 SUMMU 0.6253E+04

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

10 AX 1.2727E+01 MS 2.2274E+00 PHIS 1.4424E-01 IS 2.7320E+03 PS 8.0030E+00 US 1.5461E+05
 HX .1291E+01 DELTA .6052E+00 MS .9420E+02 HX .3015E+03 TANN 0. US 0.
 PIX .1647E+02 MMUS .7062E+03 SUMMU 0.1000E+05

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

11 AX 1.2123E+01 MS 2.0225E+00 PHIS 1.4035E-01 IS 2.7260E+03 PS 7.8950E+00 US 1.5620E+05
 HX .1359E+01 DELTA .6054E+00 MS .8899E+02 HX .3010E+03 TANN 0. US 0.
 PIX .1634E+02 MMUS .7002E+03 SUMMU 0.1270E+05

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

12 AX 1.2409E+01 MS 2.0170E+00 PHIS 1.4034E-01 IS 2.7190E+03 PS 7.7750E+00 US 1.5707E+05
 HX .1310E+01 DELTA .6050E+00 MS .8545E+02 HX .3004E+03 TANN 0. US 0.
 PIX .1619E+02 MMUS .6913E+03 SUMMU 0.1500E+05

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

13 AX 1.1605E+01 MS 2.2100E+00 PHIS 2.1042E-01 IS 2.7120E+03 PS 7.6630E+00 US 1.5740E+05
 HX .1320E+01 DELTA .6042E+00 MS .8192E+02 HX .3000E+03 TANN 0. US 0.
 PIX .1604E+02 MMUS .6813E+03 SUMMU 0.1423E+05

SPECL1 CU Cx .3000E+00 Xx .2125E+00 ADUITS 0.
 SPECL1 C02 Cx .6000E+01 Xx .2704E+01 ADUITS 0.
 SPECL1 H Cx .6300E+03 Xx .1249E+01 ADUITS 0.
 SPECL1 M2 Cx .2400E+01 Xx .2300E+00 ADUITS 0.
 SPECL1 UM Cx .2400E+02 Xx .3033E+02 ADUITS 0.
 SPECL1 M2U Cx .1900E+00 Xx .2093E+00 ADUITS 0.
 SPECL1 M2 Cx .4200E+00 Xx .2975E+00 ADUITS 0.
 SPECL1 U Cx .7800E+04 Xx .4660E+04 ADUITS 0.
 SPECL1 U2 Cx .5000E+04 Xx .3471E+04 ADUITS 0.
 SPECL1 A Cx .1100E+05 Xx .5454E+06 ADUITS 0.

Reproduced from best available copy.

14 NO 1.1744E+01 NO 7.0000E+00 PHIS 2.5000E+01 TS 2.7000E+03 PS 7.0700E+00 UN 1.5000E+05
MAA .1335E+01 DELTA .0050E+00 MA .7000E+02 MTA .3010E+03 TAMA 0.
PIA .1500E+02 MMA .0000E+03 SMA 0. SUMMATA .2100E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

15 NO 1.1000E+01 NO 0.3000E+00 PHIS 2.5000E+01 TS 2.7000E+03 PS 7.1200E+00 UN 1.0000E+05
MAA .1307E+01 DELTA .0050E+00 MA .7500E+02 MTA .3030E+03 TAMA 0.
PIA .1500E+02 MMA .0001E+03 SMA 0. SUMMATA .2000E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

16 NO 1.1000E+01 NO 0.9720E+00 PHIS 2.7000E+01 TS 2.0000E+03 PS 7.1500E+00 UN 1.0150E+05
MAA .1300E+01 DELTA .0050E+00 MA .0000E+02 MTA .3010E+03 TAMA 0.
PIA .1500E+02 MMA .0000E+03 SMA 0. SUMMATA .2700E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

17 NO 1.1000E+01 NO 0.9500E+00 PHIS 2.0000E+01 TS 2.0710E+03 PS 0.9500E+00 UN 1.0000E+05
MAA .1370E+01 DELTA .0050E+00 MA .0120E+02 MTA .1700E+03 TAMA 0.
PIA .1510E+02 MMA .0277E+03 SMA 0. SUMMATA .3100E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

18 NO 1.1100E+01 NO 1.0130E+01 PHIS 3.0000E+01 TS 2.0000E+03 PS 0.7150E+00 UN 1.0070E+05
MAA .1000E+01 DELTA .0001E+00 MA .0071E+02 MTA .3700E+03 TAMA 0.
PIA .1000E+02 MMA .0131E+03 SMA 0. SUMMATA .3531E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

19 NO 1.0000E+01 NO 1.0700E+01 PHIS 3.2000E+01 TS 2.0700E+03 PS 0.0000E+00 UN 1.0700E+05
MAA .1000E+01 DELTA .0050E+00 MA .3707E+02 MTA .5710E+03 TAMA 0.
PIA .1000E+02 MMA .0070E+03 SMA 0. SUMMATA .3000E+05

SPECIF CU CA .3000E+00 ZA .2125E+00 NUITA 0.
SPECIF CU2 CA .0000E+01 ZA .2700E+01 NUITA 0.
SPECIF M CA .0300E+03 ZA .1200E+01 NUITA 0.
SPECIF M2 CA .2000E+01 ZA .2300E+00 NUITA 0.
SPECIF UM CA .2000E+02 ZA .3033E+02 NUITA 0.
SPECIF M2U CA .1900E+00 ZA .2093E+00 NUITA 0.
SPECIF M2 CA .0200E+00 ZA .2075E+00 NUITA 0.
SPECIF U CA .7000E+00 ZA .9000E+00 NUITA 0.
SPECIF U2 CA .5000E+00 ZA .3071E+00 NUITA 0.
SPECIF A CA .1100E+05 ZA .5050E+00 NUITA 0.

20 AN 1.4716E+01 NA 1.1201E+01 PHIS 3.4207E-01 IS 2.5900E+03 PA 6.2050E+00 UN 1.0702E+05
 VAS .1430E+01 DELTA .0037E+00 NA .7207E+02 MTS .9500E+03 TANS 0.0
 PIS .1027E+02 NMOB .4002E+03 SUM 0.0 SUMMITS .4300E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

21 AN 1.0507E+01 NA 1.1050E+01 PHIS 3.0070E-01 IS 2.5050E+03 PA 5.9000E+00 UN 1.7700E+05
 VAS .1000E+01 DELTA .0002E+00 NA .7201E+01 MTS .3030E+03 TANS 0.0
 PIS .1020E+02 NMOB .5012E-03 SUM 0.0 SUMMITS .4700E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

22 AN 1.0200E+01 NA 1.0215E+01 PHIS 3.704E-01 IS 2.5500E+03 PA 5.0000E+00 UN 1.7010E+05
 VAS .1574E+01 DELTA .0040E+00 NA .7203E+00 MTS .3710E+03 TANS 0.0
 PIS .1500E+02 NMOB .5305E-03 SUM 0.0 SUMMITS .5107E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

23 AN 1.0000E+01 NA 1.0275E+01 PHIS 3.9677E-01 IS 2.5330E+03 PA 5.3000E+00 UN 1.7001E+05
 VAS .1550E+01 DELTA .0040E+00 NA .7225E+01 MTS .3700E+03 TANS 0.0
 PIS .1500E+02 NMOB .5100E-03 SUM 0.0 SUMMITS .5000E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

24 AN 4.0200E+00 NA 1.3530E+01 PHIS 4.1001E-01 IS 2.5000E+03 PA 5.0000E+00 UN 1.0070E+05
 VAS .1011E+01 DELTA .0030E+00 NA .7231E+02 MTS .3000E+03 TANS 0.0
 PIS .1331E+02 NMOB .0000E-03 SUM 0.0 SUMMITS .0100E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

25 AN 4.5715E+00 NA 1.0100E+01 PHIS 4.5200E-01 IS 2.4010E+03 PA 4.0000E+00 UN 1.0000E+05
 VAS .1000E+01 DELTA .0050E+00 NA .7203E+02 MTS .3000E+03 TANS 0.0
 PIS .1000E+02 NMOB .0000E-03 SUM 0.0 SUMMITS .0000E+05

SPECIE LU CA .3000E+00 XX .2125E+00 ADITS 0.
 SPECIE LU2 CA .6000E+01 XX .2700E-01 ADITS 0.
 SPECIE M CA .6300E+03 XX .1200E-01 ADITS 0.
 SPECIE M2 CA .2000E+01 XX .2300E+00 ADITS 0.
 SPECIE UM CA .2000E+02 XX .3033E+02 ADITS 0.
 SPECIE M2U CA .1900E+00 XX .2093E+00 ADITS 0.
 SPECIE M2 CA .4200E+00 XX .2975E+00 ADITS 0.
 SPECIE U CA .7000E+04 XX .9000E-04 ADITS 0.
 SPECIE U2 CA .5000E+00 XX .3471E-04 ADITS 0.
 SPECIE A CA .1100E-05 XX .5450E-06 ADITS 0.

```

GRAM1
OFF = .3E+01.
PFG = .3E+01.
UMI = -0.0.
TL = .330E+00.
CS = .273E+00.
TPO = .230E+00.
MMS = .4E+01.
OT = .10E+03.
MTRM = .250E+03.
SIG = -0.0.
EP = -0.0.
MPS = 0.
MC = 1.
MMI = 10.
MP = .1E-03, .2E-03, .3E-03, .4E-03, 0.0, 0.0, 0.0, 0.0.
MI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0.
VI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0.
TPI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0.
MFI = 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0, 0.0.

```

```

SEND
STREAMLINE NO. 1
DOWNSTREAM VELOCITY 1.470E+05 1.390E+05 1.310E+05 1.200E+05
CROSS-STREAM VELOCITY -2.700E+02 -5.500E+02 -7.000E+02 -1.010E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 2
DOWNSTREAM VELOCITY 1.470E+05 1.390E+05 1.320E+05 1.200E+05
CROSS-STREAM VELOCITY -2.920E+02 -5.920E+02 -6.140E+02 -1.070E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 3
DOWNSTREAM VELOCITY 1.400E+05 1.390E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.100E+02 -6.270E+02 -6.040E+02 -1.100E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 4
DOWNSTREAM VELOCITY 1.400E+05 1.390E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.310E+02 -6.710E+02 -6.220E+02 -1.210E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 5
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.550E+02 -7.190E+02 -6.040E+02 -1.300E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 6
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -3.820E+02 -7.750E+02 -6.040E+03 -1.400E+03
PARTICLE TEMPERATURE 2.013E+03 2.003E+03 2.013E+03 2.073E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 7
DOWNSTREAM VELOCITY 1.400E+05 1.400E+05 1.320E+05 1.250E+05
CROSS-STREAM VELOCITY -4.140E+02 -8.390E+02 -6.150E+03 -1.510E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 8
DOWNSTREAM VELOCITY 1.400E+05 1.410E+05 1.330E+05 1.200E+05
CROSS-STREAM VELOCITY -4.510E+02 -9.150E+02 -6.260E+03 -1.650E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 9
DOWNSTREAM VELOCITY 1.500E+05 1.410E+05 1.330E+05 1.200E+05
CROSS-STREAM VELOCITY -4.970E+02 -1.010E+03 -6.300E+03 -1.820E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.013E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05
STREAMLINE NO. 10
DOWNSTREAM VELOCITY 1.500E+05 1.420E+05 1.340E+05 1.200E+05
CROSS-STREAM VELOCITY -5.520E+02 -1.120E+03 -6.540E+03 -2.070E+03
PARTICLE TEMPERATURE 2.003E+03 2.003E+03 2.003E+03 2.063E+03
PARTICLE DENSITY 2.700E+05 1.430E+04 1.020E+04 5.500E+05

```

STREAMLINE NO.	11				
DOWNSTREAM VELOCITY	1.510E+05	1.430E+05	1.340E+05	1.270E+05	
CROSS-STREAM VELOCITY	-0.210E+02	-1.200E+03	-1.730E+03	-2.270E+03	
PARTICLE TEMPERATURE	2.703E+03	2.673E+03	2.643E+03	2.613E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	12				
DOWNSTREAM VELOCITY	1.520E+05	1.430E+05	1.350E+05	1.270E+05	
CROSS-STREAM VELOCITY	-7.090E+02	-1.040E+03	-1.960E+03	-2.600E+03	
PARTICLE TEMPERATURE	2.703E+03	2.673E+03	2.643E+03	2.613E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	13				
DOWNSTREAM VELOCITY	1.530E+05	1.400E+05	1.300E+05	1.200E+05	
CROSS-STREAM VELOCITY	-0.200E+02	-1.000E+03	-2.310E+03	-3.030E+03	
PARTICLE TEMPERATURE	2.703E+03	2.673E+03	2.643E+03	2.613E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	14				
DOWNSTREAM VELOCITY	1.540E+05	1.450E+05	1.360E+05	1.280E+05	
CROSS-STREAM VELOCITY	-0.930E+02	-2.010E+03	-2.770E+03	-3.630E+03	
PARTICLE TEMPERATURE	2.703E+03	2.653E+03	2.623E+03	2.593E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	15				
DOWNSTREAM VELOCITY	1.550E+05	1.400E+05	1.370E+05	1.290E+05	
CROSS-STREAM VELOCITY	-1.700E+03	-2.570E+03	-3.400E+03	-4.500E+03	
PARTICLE TEMPERATURE	2.703E+03	2.653E+03	2.623E+03	2.593E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	16				
DOWNSTREAM VELOCITY	1.560E+05	1.470E+05	1.380E+05	1.300E+05	
CROSS-STREAM VELOCITY	-1.000E+03	-3.300E+03	-4.010E+03	-4.800E+03	
PARTICLE TEMPERATURE	2.703E+03	2.623E+03	2.673E+03	2.623E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	17				
DOWNSTREAM VELOCITY	1.580E+05	1.400E+05	1.390E+05	1.310E+05	
CROSS-STREAM VELOCITY	-4.000E+03	-5.040E+03	-6.020E+03	-6.990E+03	
PARTICLE TEMPERATURE	2.703E+03	2.623E+03	2.673E+03	2.623E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	18				
DOWNSTREAM VELOCITY	1.590E+05	1.500E+05	1.400E+05	1.320E+05	
CROSS-STREAM VELOCITY	-0.470E+03	-1.010E+04	-1.300E+04	-1.620E+04	
PARTICLE TEMPERATURE	2.713E+03	2.703E+03	2.653E+03	2.603E+03	
PARTICLE DENSITY	2.700E+05	1.430E+04	1.020E+04	5.540E+03	
STREAMLINE NO.	19				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	20				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	21				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	22				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	23				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	24				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
STREAMLINE NO.	25				
DOWNSTREAM VELOCITY	0.	0.	0.	0.	
CROSS-STREAM VELOCITY	0.	0.	0.	0.	
PARTICLE TEMPERATURE	0.	0.	0.	0.	
PARTICLE DENSITY	0.	0.	0.	0.	
1	U	H	H2	H	22 10E+2A 1.0 =0.0
2	U	H	H2	H	22 10E+20 1.0 =0.0
3	H	H	H2	H	22 10E+20 1.0 =0.0
4	H	UH	H2U	H	22 10E+27 1.0 =0.0
5	LU	L	LU2	L	26 10E+2A 1.0 =2500.0
6	UH	UH	H2U	H	15 10E+10=0.0 =1000.0
7	UH	H2	H2U	H	15 50E+10=0.0 =5700.0
8	U	H2	UH	H	15 50E+10=0.0 =8400.0
9	H	U2	UH	U	15 30E+09=0.0 =10500.0
10	LU	UH	LU2	H	15 50E+12=0.0 =600.0

1 20 100 .2200E+02 1000 0. PH10 0.
 2 10 2.2000E+01 10 2.2000E+01 PH10 2.2000E+02 10 2.4500E+03 10 3.1135E+00 10 2.0000E+05
 100 .1777E+01 100 .0225E+00 100 .7110E+07 100 .0117E+03 1000 0. 1000 0.
 100 .9207E+01 100 .3077E+03 100 .1021E+02 100 .1310E+03

SPECIE CU Co .2601E+00 10 .2100E+00 1000 0.
 SPECIE CU2 Co .0017E+01 10 .2095E+01 1000 0.
 SPECIE H Co .3123E-03 10 .0221E+02 1000 0.
 SPECIE H2 Co .2000E-01 10 .2000E+00 1000 0.
 SPECIE UM Co .0500E+03 10 .9900E-03 1000 0.
 SPECIE MDU Co .1095E+00 10 .2090E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1277E+04 10 .1590E+04 1000 0.
 SPECIE L2 Co .1100E+04 10 .7123E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5477E+06 1000 0.

3 10 2.2000E+01 10 1.0019E+00 PH10 0.4010E+02 10 2.0530E+03 10 3.1100E+00 10 2.0121E+05
 100 .1777E+01 100 .0220E+00 100 .7332E+02 100 .4107E+03 1000 0. 1000 0.
 100 .9207E+01 100 .3077E+03 100 .1020E+02 100 .5253E+03

SPECIE CU Co .2000E+00 10 .2100E+00 1000 0.
 SPECIE CU2 Co .0570E+01 10 .2070E+01 1000 0.
 SPECIE H Co .3095E-03 10 .0005E+02 1000 0.
 SPECIE H2 Co .2007E-01 10 .2037E+00 1000 0.
 SPECIE UM Co .0000E+03 10 .9053E-03 1000 0.
 SPECIE MDU Co .1090E+00 10 .2090E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1250E+04 10 .1507E+04 1000 0.
 SPECIE L2 Co .1090E+04 10 .0019E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5470E+06 1000 0.

4 10 2.2705E+01 10 2.0595E+00 PH10 0.7190E+02 10 2.4530E+03 10 3.1100E+00 10 2.0100E+05
 100 .1777E+01 100 .0100E+00 100 .7272E+02 100 .4107E+03 1000 0. 1000 0.
 100 .9207E+01 100 .3002E+03 100 .1020E+02 100 .1170E+04

SPECIE CU Co .2007E+00 10 .2110E+00 1000 0.
 SPECIE CU2 Co .0520E+01 10 .2050E+01 1000 0.
 SPECIE H Co .3055E-03 10 .0000E+02 1000 0.
 SPECIE H2 Co .2000E-01 10 .2035E+00 1000 0.
 SPECIE UM Co .0502E+03 10 .1003E-02 1000 0.
 SPECIE MDU Co .1090E+00 10 .2100E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1300E+04 10 .1020E+04 1000 0.
 SPECIE L2 Co .1090E+04 10 .0790E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5470E+06 1000 0.

5 10 2.2705E+01 10 3.2740E+00 PH10 0.0310E+02 10 2.4502E+03 10 3.1000E+00 10 2.0100E+05
 100 .1777E+01 100 .0100E+00 100 .7242E+02 100 .4100E+03 1000 0. 1000 0.
 100 .9207E+01 100 .3077E+03 100 .1010E+02 100 .2091E+04

SPECIE CU Co .2007E+00 10 .2110E+00 1000 0.
 SPECIE CU2 Co .0517E+01 10 .2050E+01 1000 0.
 SPECIE H Co .3000E-03 10 .0100E+02 1000 0.
 SPECIE H2 Co .2000E-01 10 .2030E+00 1000 0.
 SPECIE UM Co .0000E+03 10 .1010E+02 1000 0.
 SPECIE MDU Co .1090E+00 10 .2100E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1320E+04 10 .1050E+04 1000 0.
 SPECIE L2 Co .1100E+04 10 .0050E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5470E+06 1000 0.

6 10 2.2000E+01 10 4.0097E+00 PH10 1.1270E+01 10 2.4530E+03 10 3.0975E+00 10 2.0100E+05
 100 .1777E+01 100 .0100E+00 100 .7240E+02 100 .4105E+03 1000 0. 1000 0.
 100 .9207E+01 100 .3005E+03 100 .1017E+02 100 .3250E+04

SPECIE CU Co .2072E+00 10 .2110E+00 1000 0.
 SPECIE CU2 Co .0440E+01 10 .2010E+01 1000 0.
 SPECIE H Co .3037E-03 10 .0050E+02 1000 0.
 SPECIE H2 Co .2000E-01 10 .2030E+00 1000 0.
 SPECIE UM Co .0733E-03 10 .1023E+02 1000 0.
 SPECIE MDU Co .1090E+00 10 .2100E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1307E+04 10 .1102E+04 1000 0.
 SPECIE L2 Co .1070E+04 10 .0000E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5470E+06 1000 0.

7 10 2.2500E+01 10 4.0000E+00 PH10 1.3520E+01 10 2.4527E+03 10 3.0000E+00 10 2.0000E+05
 100 .1777E+01 100 .0100E+00 100 .7332E+02 100 .4093E+03 1000 0. 1000 0.
 100 .9170E+01 100 .3057E+03 100 .1015E+02 100 .0070E+04

SPECIE CU Co .2070E+00 10 .2110E+00 1000 0.
 SPECIE CU2 Co .0400E+01 10 .2010E+01 1000 0.
 SPECIE H Co .3010E-03 10 .0010E+02 1000 0.
 SPECIE H2 Co .2000E-01 10 .2030E+00 1000 0.
 SPECIE UM Co .0750E-03 10 .1020E+02 1000 0.
 SPECIE MDU Co .1090E+00 10 .2100E+00 1000 0.
 SPECIE H2 Co .0200E+00 10 .2000E+00 1000 0.
 SPECIE U Co .1370E+04 10 .1100E+04 1000 0.
 SPECIE L2 Co .1050E+04 10 .0050E+05 1000 0.
 SPECIE A Co .1100E+05 10 .5470E+06 1000 0.

0 XN 2.4027E+01 Nn 5.7091E+00 PMin 1.5023E-01 Tn 2.4517E+03 Pn 3.0722E+00 Un 2.0092E+05
 Max .1779E+01 DELTA .0175E+00 Mx -.7302E+02 Ntn .4000E+03 TAnn 0. Gm 0.
 Pta .9132E+01 MMOn .3043E+03 SIn .1013E+02 SUMDUtn .6341E+04

SPECIE CU Cn .2976E+00 Xn .2117E+00 NDUtn -.1299E-05
 SPECIE CO2 Cn .0373E-01 Xn .2409E-01 NDUtn .1299E-05
 SPECIE H Cn .3005E-03 Xn .5905E-02 NDUtn .2159E-03
 SPECIE H2 Cn .2030E-01 Xn .2429E+00 NDUtn -.2102E-03
 SPECIE OH Cn .0000E-03 Xn .1032E-02 NDUtn -.2177E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2171E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1307E-04 Xn .1702E-04 NDUtn -.0430E-06
 SPECIE U2 Cn .1030E-04 Xn .0405E-05 NDUtn -.2197E-07
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

0 XN 2.2200E+01 Nn 6.5153E+00 PMin 1.4107E-01 Tn 2.4400E+03 Pn 3.0500E+00 Un 2.0111E+05
 Max .1777E+01 DELTA .0181E+00 Mx -.7557E+02 Ntn .4000E+03 TAnn 0. Gm 0.
 Pta .9091E+01 MMOn .3075E+03 SIn .1011E+02 SUMDUtn .6253E+04

SPECIE CU Cn .2976E+00 Xn .2117E+00 NDUtn -.1510E-05
 SPECIE CO2 Cn .0370E-01 Xn .2007E-01 NDUtn .1510E-05
 SPECIE H Cn .2976E-03 Xn .5920E-02 NDUtn .2109E-03
 SPECIE H2 Cn .2039E-01 Xn .2429E+00 NDUtn -.2109E-03
 SPECIE OH Cn .0000E-03 Xn .1010E-02 NDUtn -.2205E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2197E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1307E-04 Xn .1702E-04 NDUtn -.0977E-06
 SPECIE U2 Cn .1010E-04 Xn .0323E-05 NDUtn -.1917E-07
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

10 XN 2.2130E+01 Nn 7.3192E+00 PMin 2.0620E-01 Tn 2.4400E+03 Pn 3.0177E+00 Un 2.0127E+05
 Max .1780E+01 DELTA .0193E+00 Mx -.7700E+02 Ntn .4000E+03 TAnn 0. Gm 0.
 Pta .9012E+01 MMOn .2990E+03 SIn .1000E+02 SUMDUtn .1000E+05

SPECIE CU Cn .2975E+00 Xn .2117E+00 NDUtn -.2200E-05
 SPECIE CO2 Cn .0300E-01 Xn .2093E-01 NDUtn .2200E-05
 SPECIE H Cn .2903E-03 Xn .5002E-02 NDUtn .2030E-03
 SPECIE H2 Cn .2000E-01 Xn .2430E+00 NDUtn -.2027E-03
 SPECIE OH Cn .0500E-03 Xn .1002E-02 NDUtn -.2000E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2030E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1330E-04 Xn .1657E-04 NDUtn -.9030E-06
 SPECIE U2 Cn .9937E-05 Xn .0103E-05 NDUtn -.1390E-07
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

11 XN 2.1952E+01 Nn 6.1207E+00 PMin 2.5153E-01 Tn 2.4400E+03 Pn 2.9001E+00 Un 2.0107E+05
 Max .1700E+01 DELTA .0211E+00 Mx -.6071E+02 Ntn .4050E+03 TAnn 0. Gm 0.
 Pta .0030E+01 MMOn .2967E+03 SIn .1005E+02 SUMDUtn .1270E+05

SPECIE CU Cn .2975E+00 Xn .2117E+00 NDUtn -.2430E-05
 SPECIE CO2 Cn .0390E-01 Xn .2090E-01 NDUtn .2430E-05
 SPECIE H Cn .2900E-03 Xn .5700E-02 NDUtn .2030E-03
 SPECIE H2 Cn .2000E-01 Xn .2431E+00 NDUtn -.2030E-03
 SPECIE OH Cn .0370E-03 Xn .0909E-03 NDUtn -.2077E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2000E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1290E-04 Xn .1590E-04 NDUtn -.9000E-06
 SPECIE U2 Cn .9000E-05 Xn .5400E-05 NDUtn -.1120E-07
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

12 XN 2.1753E+01 Nn 6.9213E+00 PMin 2.5095E-01 Tn 2.4400E+03 Pn 2.9300E+00 Un 2.0207E+05
 Max .1700E+01 DELTA .0249E+00 Mx -.6237E+02 Ntn .4050E+03 TAnn 0. Gm 0.
 Pta .0030E+01 MMOn .2927E+03 SIn .1001E+02 SUMDUtn .1900E+05

SPECIE CU Cn .2975E+00 Xn .2117E+00 NDUtn -.2001E-05
 SPECIE CO2 Cn .0390E-01 Xn .2097E-01 NDUtn .2001E-05
 SPECIE H Cn .2900E-03 Xn .5730E-02 NDUtn .2030E-03
 SPECIE H2 Cn .2000E-01 Xn .2431E+00 NDUtn -.2030E-03
 SPECIE OH Cn .0270E-03 Xn .0907E-03 NDUtn -.2030E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2030E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1201E-04 Xn .1570E-04 NDUtn -.9171E-06
 SPECIE U2 Cn .9000E-05 Xn .5470E-05 NDUtn -.9000E-06
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

13 XN 2.1550E+01 Nn 4.7100E+00 PMin 2.6297E-01 Tn 2.4400E+03 Pn 2.8800E+00 Un 2.0290E+05
 Max .1700E+01 DELTA .0277E+00 Mx -.6000E+02 Ntn .4050E+03 TAnn 0. Gm 0.
 Pta .0010E+01 MMOn .2910E+03 SIn .9900E+01 SUMDUtn .1000E+05

SPECIE CU Cn .2970E+00 Xn .2110E+00 NDUtn -.2200E-05
 SPECIE CO2 Cn .0400E-01 Xn .2090E-01 NDUtn .2200E-05
 SPECIE H Cn .2800E-03 Xn .5670E-02 NDUtn .2030E-03
 SPECIE H2 Cn .2000E-01 Xn .2430E+00 NDUtn -.2030E-03
 SPECIE OH Cn .0125E-03 Xn .0900E-03 NDUtn -.2030E-03
 SPECIE H2O Cn .1900E+00 Xn .2107E+00 NDUtn .2030E-03
 SPECIE H2 Cn .4200E+00 Xn .2900E+00 NDUtn 0.
 SPECIE U Cn .1200E-04 Xn .1520E-04 NDUtn -.9500E-06
 SPECIE U2 Cn .9150E-05 Xn .5090E-05 NDUtn -.9000E-06
 SPECIE A Cn .1100E-05 Xn .5470E-06 NDUtn 0.

20	AX 1.9200E+01 MAX .2030E+01 MIN .0511E+01	MA 1.5217E+01 DELTA .7621E+00 MINOR .2610E+03	PHIS 0.2513E-01 MS -.2100E+03 SRS .9420E+01	TS 2.1901E+03 MTS .3500E+03 SUMDTS .0332E+05	PS 2.3500E+00 TANS 0.	US 2.1005E+05 US 0.
	SPECIE C1 SPECIE C2 SPECIE M SPECIE M2 SPECIE UM SPECIE M2U SPECIE U SPECIE U2 SPECIE X	C1 .2950E+00 C2 .0700E-01 M .1125E-03 M2 .2472E-01 UM .2260E+03 M2U .1040E+00 U .4200E+00 U2 .1414E-05 X .1000E-05 X .1100E-05	X1 .2103E+00 X2 .3000E-01 X3 .2240E-02 X4 .2400E+00 X5 .2600E-03 X6 .2101E+00 X7 .2995E+00 X8 .1700E-05 X9 .0700E-06 X0 .5400E-06	ADU1A -.2103E-06 ADU1B .2103E-06 ADU1C .5400E-04 ADU1D -.5400E-04 ADU1E .5400E-04 ADU1F 0. ADU1G -.1000E-06 ADU1H -.0000E-09 ADU1I 0.		
21	AX 1.9000E+01 MAX .1910E+01 MIN .9747E+01	MA 1.5020E+01 DELTA .8663E+00 MINOR .3125E+03	PHIS 3.4350E-01 MS -.1653E+03 SRS .9391E+01	TS 2.2906E+03 MTS .3625E+03 SUMDTS .4759E+05	PS 2.9415E+00 TANS 0.	US 2.1000E+05 US 0.
	SPECIE C1 SPECIE C2 SPECIE M SPECIE M2 SPECIE UM SPECIE M2U SPECIE U SPECIE U2 SPECIE X	C1 .2960E+00 C2 .6507E-01 M .1207E-05 M2 .2450E-01 UM .2650E-03 M2U .1905E+00 U .4200E+00 U2 .1655E-05 X .1153E-05 X .1100E-05	X1 .2110E+00 X2 .2452E-01 X3 .2410E-02 X4 .2450E+00 X5 .3111E-03 X6 .2113E+00 X7 .2440E+00 X8 .2000E-05 X9 .7195E-06 X0 .5400E-06	ADU1A .2753E-05 ADU1B .2753E-05 ADU1C .0012E-04 ADU1D .0500E-04 ADU1E .0000E-04 ADU1F .0599E-04 ADU1G 0. ADU1H .1799E-07 ADU1I -.7000E-09 ADU1J 0.		
22	AX 1.8710E+01 MAX .1974E+01 MIN .9520E+01	MA 1.6050E+01 DELTA .6900E+00 MINOR .2913E+03	PHIS 4.5705E-01 MS -.1000E+03 SRS .9356E+01	TS 2.2551E+03 MTS .3700E+03 SUMDTS .5147E+05	PS 2.7000E+00 TANS 0.	US 2.1450E+05 US 0.
	SPECIE C1 SPECIE C2 SPECIE M SPECIE M2 SPECIE UM SPECIE M2U SPECIE U SPECIE U2 SPECIE X	C1 .2965E+00 C2 .6570E-01 M .1503E-03 M2 .2450E-01 UM .3550E-03 M2U .1402E+00 U .4200E+00 U2 .2950E-05 X .2202E-05 X .1100E-05	X1 .2112E+00 X2 .2403E-01 X3 .2990E-02 X4 .2450E+00 X5 .4171E-03 X6 .2100E+00 X7 .2443E+00 X8 .3000E-05 X9 .1423E-05 X0 .5400E-06	ADU1A .0400E-05 ADU1B .0400E-05 ADU1C .2003E-03 ADU1D .1940E-03 ADU1E .2020E-03 ADU1F .1450E-03 ADU1G 0. ADU1H .5532E-07 ADU1I .5499E-08 ADU1J 0.		
23	AX 1.8000E+01 MAX .2010E+01 MIN .8940E+01	MA 1.7040E+01 DELTA .7043E+00 MINOR .2750E+03	PHIS 4.5620E-01 MS -.1920E+03 SRS .9512E+01	TS 2.2512E+03 MTS .3700E+03 SUMDTS .5043E+05	PS 2.5000E+00 TANS 0.	US 2.1020E+05 US 0.
	SPECIE C1 SPECIE C2 SPECIE M SPECIE M2 SPECIE UM SPECIE M2U SPECIE U SPECIE U2 SPECIE X	C1 .2950E+00 C2 .6742E-01 M .1422E-03 M2 .2470E-01 UM .3103E-03 M2U .1893E+00 U .4200E+00 U2 .2490E-05 X .2010E-05 X .1100E-05	X1 .2102E+00 X2 .3001E-01 X3 .2437E-02 X4 .2405E+00 X5 .3090E-03 X6 .2090E+00 X7 .2990E+00 X8 .3117E-05 X9 .1250E-05 X0 .5400E-06	ADU1A -.2301E-05 ADU1B .2301E-05 ADU1C .2400E-03 ADU1D .2053E-03 ADU1E .2070E-03 ADU1F .2050E-03 ADU1G 0. ADU1H .4011E-06 ADU1I .3100E-08 ADU1J 0.		
24	AX 1.8000E+01 MAX .2010E+01 MIN .9147E+01	MA 1.7707E+01 DELTA .6947E+00 MINOR .2740E+03	PHIS 4.6525E-01 MS -.1849E+03 SRS .9277E+01	TS 2.2401E+03 MTS .3430E+03 SUMDTS .6100E+05	PS 2.5033E+00 TANS 0.	US 2.1010E+05 US 0.
	SPECIE C1 SPECIE C2 SPECIE M SPECIE M2 SPECIE UM SPECIE M2U SPECIE U SPECIE U2 SPECIE X	C1 .2950E+00 C2 .6600E-01 M .1374E-03 M2 .2465E-01 UM .3070E-03 M2U .1849E+00 U .4200E+00 U2 .2371E-05 X .1740E-05 X .1100E-05	X1 .2107E+00 X2 .3033E-01 X3 .2700E-02 X4 .2400E+00 X5 .3010E-03 X6 .2100E+00 X7 .2990E+00 X8 .2490E-05 X9 .1114E-05 X0 .5400E-06	ADU1A .1199E-05 ADU1B .1199E-05 ADU1C .1205E-03 ADU1D .1149E-03 ADU1E .1212E-03 ADU1F .1203E-03 ADU1G 0. ADU1H .2717E-06 ADU1I .1007E-08 ADU1J 0.		
25	AX 1.7000E+01 MAX .1940E+01 MIN .8901E+01	MA 1.8303E+01 DELTA .6885E+00 MINOR .2934E+03	PHIS 4.6420E-01 MS -.2313E+03 SRS .9252E+01	TS 2.1905E+03 MTS .2910E+03 SUMDTS .6563E+05	PS 2.6014E+00 TANS .2977E+05	US 2.0957E+05 US -.1010E+03

WALL TEMPERATURE 1000. KELVIN

WALL SURFACE DATA PARAMETERS

WALL	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00
TEMP	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00
AREA	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00	1.0000E+00

BOUNDARY LAYER PROFILES

U/1E	V/1E	U/UF 1nC	V/DL
.1000E+01	.4694E+00	.1045E+00	.2100E+03
.2000E+01	.4771E+00	.1067E+00	.4197E+03
.3000E+01	.4855E+00	.1066E+00	.6270E+03
.4000E+01	.4943E+00	.1243E+01	.8320E+03
.5000E+01	.5031E+00	.1490E+01	.1030E+04
.6000E+01	.5127E+00	.1799E+01	.1230E+04
.7000E+01	.5221E+00	.2070E+01	.1430E+04
.8000E+01	.5317E+00	.2305E+01	.1630E+04
.9000E+01	.5413E+00	.2609E+01	.1830E+04
.1000E+00	.5510E+00	.2930E+01	.2027E+04
.1900E+00	.5996E+00	.4296E+01	.2972E+04
.2900E+00	.6474E+00	.5679E+01	.3908E+04
.3900E+00	.6936E+00	.6875E+01	.4935E+04
.4900E+00	.7375E+00	.8101E+01	.6010E+04
.5900E+00	.7780E+00	.9292E+01	.7165E+04
.6900E+00	.8172E+00	.1045E+02	.8380E+04
.7900E+00	.8575E+00	.1159E+02	.9650E+04
.8900E+00	.8945E+00	.1270E+02	.1095E+05
.9900E+00	.9130E+00	.1380E+02	.1231E+05
.1000E+00	.9300E+00	.1480E+02	.1372E+05
.1500E+00	.9594E+00	.1595E+02	.1520E+05
.2000E+00	.9769E+00	.1700E+02	.1674E+05
.2500E+00	.9907E+00	.1805E+02	.1830E+05
.3000E+00	.1001E+01	.1909E+02	.1990E+05
.3500E+00	.1006E+01	.2013E+02	.2150E+05
.4000E+00	.1008E+01	.2117E+02	.2310E+05
.4500E+00	.1006E+01	.2220E+02	.2470E+05
.5000E+01	.1000E+01	.2324E+02	.2630E+05

SPECIE	CU	Ca	XB	ADU1E	ADU1E
SPECIE CU	Ca	.2401E+00	XB	.2101E+00	ADU1E -.1007E+05
SPECIE CU2	Ca	.6940E+01	XB	.4106E+01	ADU1E .1007E+05
SPECIE H	Ca	.4236E+04	XB	.1044E+02	ADU1E .7007E+04
SPECIE H2	Ca	.2476E+01	XB	.2072E+00	ADU1E -.7007E+04
SPECIE H2O	Ca	.1791E+03	XB	.2100E+03	ADU1E -.7007E+04
SPECIE H2O	Ca	.1893E+00	XB	.2100E+00	ADU1E .7007E+04
SPECIE H2	Ca	.4200E+00	XB	.2495E+00	ADU1E U.
SPECIE U	Ca	.4244E+00	XB	.1150E+04	ADU1E -.1173E+06
SPECIE H2	Ca	.1730E+00	XB	.4512E+00	ADU1E -.4391E+09
SPECIE H	Ca	.1110E+05	XB	.4492E+00	ADU1E U.

```

=====STREAMLINE 1 SE 10.20905
-- LOCAL GAS PROPERTIES
UR 2.010E+05 TR 2.450E+03 DENSITY 3.077E+04 REYNOLDS NO. 6.911E+21 Y20 6.225E+01 H20 6.220E+01
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.890E+05 1.801E+05 1.530E+05 1.421E+05
CROSS-STREAM VELOCITY -3.750E+01 -2.959E+02 -5.505E+02 -6.415E+02
PARTICLE REYNOLDS NO. 1.222E+00 6.675E+00 1.453E+01 2.373E+01
PARTICLE TEMPERATURE 2.522E+03 2.630E+03 2.719E+03 2.792E+03
PARTICLE DENSITY 1.201E+05 7.015E+05 9.411E+05 3.200E+05
PARTICLE MASS FLOW 2.650E+00
=====STREAMLINE 2 SE 10.20905
-- LOCAL GAS PROPERTIES
UR 2.010E+05 TR 2.450E+03 DENSITY 3.043E+04 REYNOLDS NO. 6.902E+21 Y20 1.642E+00 H20 1.642E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.892E+05 1.801E+05 1.535E+05 1.421E+05
CROSS-STREAM VELOCITY -1.103E+02 -5.506E+02 -4.011E+02 -1.096E+03
PARTICLE REYNOLDS NO. 1.244E+00 6.700E+00 1.449E+01 2.366E+01
PARTICLE TEMPERATURE 2.523E+03 2.627E+03 2.710E+03 2.786E+03
PARTICLE DENSITY 1.207E+05 6.740E+05 9.200E+05 2.942E+05
PARTICLE MASS FLOW 2.522E+00
=====STREAMLINE 3 SE 10.20905
-- LOCAL GAS PROPERTIES
UR 2.010E+05 TR 2.450E+03 DENSITY 3.042E+04 REYNOLDS NO. 6.877E+21 Y20 2.461E+00 H20 2.450E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.891E+05 1.801E+05 1.530E+05 1.420E+05
CROSS-STREAM VELOCITY -2.396E+02 -5.070E+02 -1.250E+03 -1.525E+03
PARTICLE REYNOLDS NO. 1.233E+00 6.718E+00 1.447E+01 2.366E+01
PARTICLE TEMPERATURE 2.523E+03 2.620E+03 2.710E+03 2.788E+03
PARTICLE DENSITY 1.222E+05 6.615E+05 9.195E+05 2.902E+05
PARTICLE MASS FLOW 2.522E+00
=====STREAMLINE 4 SE 10.20905
-- LOCAL GAS PROPERTIES
UR 2.010E+05 TR 2.450E+03 DENSITY 3.075E+04 REYNOLDS NO. 6.851E+21 Y20 3.280E+00 H20 3.279E+00
----- PARTICLE PHASE PROPERTIES -----
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.891E+05 1.801E+05 1.530E+05 1.420E+05
CROSS-STREAM VELOCITY -2.602E+02 -5.207E+02 -1.270E+03 -1.501E+03
PARTICLE REYNOLDS NO. 1.230E+00 6.690E+00 1.443E+01 2.355E+01
PARTICLE TEMPERATURE 2.523E+03 2.620E+03 2.711E+03 2.788E+03
PARTICLE DENSITY 1.224E+05 6.620E+05 9.180E+05 2.888E+05
PARTICLE MASS FLOW 2.522E+00

```



```

--STREAMLINE 13 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.032E+05 IS 2.424E+03 DENSITY 2.423E-04 REYNOLDS NO.= 0.430E+21 V2= 1.000E+01 M2= 1.041E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.900E+05 1.703E+05 1.550E+05 1.440E+05
CROSS-STREAM VELOCITY -3.580E+03 -0.750E+03 -0.142E+03 -0.099E+03
PARTICLE REYNOLDS NO. 1.237E+00 0.293E+00 1.347E+01 2.701E+01
PARTICLE TEMPERATURE 2.502E+05 2.012E+05 2.090E+05 2.709E+05
PARTICLE DENSITY 1.204E-05 7.143E-05 9.024E-05 5.001E-05
PARTICLE MOMENTUM FLUX 2.710E+00
--STREAMLINE 14 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.040E+05 IS 2.443E+03 DENSITY 2.751E-04 REYNOLDS NO.= 0.501E+21 V2= 1.152E+01 M2= 1.131E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.911E+05 1.700E+05 1.555E+05 1.441E+05
CROSS-STREAM VELOCITY -4.370E+03 -0.050E+03 -0.503E+03 -1.040E+04
PARTICLE REYNOLDS NO. 1.251E+00 0.243E+00 1.391E+01 2.210E+01
PARTICLE TEMPERATURE 2.494E+05 2.000E+05 2.077E+05 2.750E+05
PARTICLE DENSITY 1.274E-05 7.210E-05 9.722E-05 5.101E-05
PARTICLE MOMENTUM FLUX 7.747E+00
--STREAMLINE 15 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.052E+05 IS 2.411E+03 DENSITY 2.600E-04 REYNOLDS NO.= 0.270E+21 V2= 1.237E+01 M2= 1.211E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.917E+05 1.710E+05 1.567E+05 1.459E+05
CROSS-STREAM VELOCITY -5.500E+03 -0.400E+03 -1.105E+04 -1.307E+04
PARTICLE REYNOLDS NO. 1.245E+00 0.242E+00 1.324E+01 2.151E+01
PARTICLE TEMPERATURE 2.492E+05 2.003E+05 2.090E+05 2.760E+05
PARTICLE DENSITY 1.277E-05 7.330E-05 1.003E-04 5.245E-05
PARTICLE MOMENTUM FLUX 2.453E+00
--STREAMLINE 16 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.067E+05 IS 2.390E+03 DENSITY 2.500E-04 REYNOLDS NO.= 0.192E+21 V2= 1.320E+01 M2= 1.201E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.920E+05 1.721E+05 1.572E+05 1.461E+05
CROSS-STREAM VELOCITY -7.040E+03 -1.201E+04 -1.894E+04 -1.002E+04
PARTICLE REYNOLDS NO. 1.353E+00 0.242E+00 1.323E+01 2.145E+01
PARTICLE TEMPERATURE 2.470E+05 2.502E+05 2.072E+05 2.740E+05
PARTICLE DENSITY 1.270E-05 7.535E-05 1.059E-04 5.563E-05
PARTICLE MOMENTUM FLUX 5.002E+00
--STREAMLINE 17 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.070E+05 IS 2.302E+03 DENSITY 2.414E-04 REYNOLDS NO.= 0.123E+21 V2= 1.414E+01 M2= 1.173E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.930E+05 1.730E+05 1.580E+05 1.473E+05
CROSS-STREAM VELOCITY -8.450E+03 -1.496E+04 -2.007E+04 -2.402E+04
PARTICLE REYNOLDS NO. 1.575E+00 0.171E+00 1.313E+01 2.124E+01
PARTICLE TEMPERATURE 2.407E+05 2.500E+05 2.070E+05 2.703E+05
PARTICLE DENSITY 1.290E-05 8.100E-05 1.245E-04 4.700E-05
PARTICLE MOMENTUM FLUX 5.503E+00
--STREAMLINE 18 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.133E+05 IS 2.511E+03 DENSITY 2.521E-04 REYNOLDS NO.= 0.035E+21 V2= 1.500E+01 M2= 1.053E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 1.950E+05 0. 0. 0.
CROSS-STREAM VELOCITY -8.430E+03 0. 0. 0.
PARTICLE REYNOLDS NO. 1.573E+00 0. 0. 0.
PARTICLE TEMPERATURE 2.435E+05 0. 0. 0.
PARTICLE DENSITY 1.300E-05 0. 0. 0.
PARTICLE MOMENTUM FLUX 2.500E+01
--STREAMLINE 19 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.140E+05 IS 2.190E+03 DENSITY 2.410E-04 REYNOLDS NO.= 5.007E+21 V2= 1.500E+01 M2= 1.522E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 0. 0. 0. 0.
CROSS-STREAM VELOCITY 0. 0. 0. 0.
PARTICLE REYNOLDS NO. 0. 0. 0. 0.
PARTICLE TEMPERATURE 0. 0. 0. 0.
PARTICLE DENSITY 0. 0. 0. 0.
PARTICLE MOMENTUM FLUX 0.
--STREAMLINE 20 SR 10.20405
-- LOCAL GAS PROPERTIES
US 2.100E+05 IS 2.270E+03 DENSITY 3.125E-04 REYNOLDS NO.= 5.071E+21 V2= 1.007E+01 M2= 1.503E+01
==== PARTICLE PHASE PROPERTIES ====
PARTICLE GROUP 1 2 3 4
DOWNSTREAM VELOCITY 0. 0. 0. 0.
CROSS-STREAM VELOCITY 0. 0. 0. 0.
PARTICLE REYNOLDS NO. 0. 0. 0. 0.
PARTICLE TEMPERATURE 0. 0. 0. 0.
PARTICLE DENSITY 0. 0. 0. 0.
PARTICLE MOMENTUM FLUX 0.
BOUNDARY OF PARTICLE PHASE AT SR 10.01204 1.470E+01 1.417E+01 1.370E+01 1.302E+01
BOUNDARY OF PARTICLE PHASE AT SR 11.01090 1.490E+01 1.431E+01 1.389E+01 1.353E+01
BOUNDARY OF PARTICLE PHASE AT SR 11.01900 1.513E+01 1.447E+01 1.403E+01 1.360E+01
BOUNDARY OF PARTICLE PHASE AT SR 11.02986 1.530E+01 1.462E+01 1.417E+01 1.370E+01
BOUNDARY OF PARTICLE PHASE AT SR 12.05150 1.547E+01 1.470E+01 1.431E+01 1.391E+01
BOUNDARY OF PARTICLE PHASE AT SR 12.07000 1.560E+01 1.483E+01 1.445E+01 1.405E+01

```

APPENDIX C

SAMPLE OUTPUT - EXPLANATIONS

C-i

APPENDIX C

SAMPLE OUTPUT - EXPLANATIONS

The first page of output contains all the input data on Cards 1-5 and 7, followed by the species identification data (Card 9) and the distributions of properties along the initial orthogonal surface. The units of each variable are identical with those used for the input data. These are followed by the axis location and the nozzle wall contour.

The standard printed output at each value of KP (Card 5, Cols. 11-15) or DXLSS (Card 3, Cols. 11-22) gives properties for each streamtube, from K (streamtube index) = 2 to K = KMAX. In addition to the usual output* i. e. X, R, PHI, T, P, U, the following properties are printed:

MA	Mach number				
DELY	streamtube width (cm)				
H	enthalpy (cal/g)				
HT	stagnation enthalpy (cal/g)				
TAW	shear stress at tube interface (dynes/cm ²)				
Q	heat flux at tube interface (cal/cm ² -sec)				
PT	dynamic pressure (atm)				
RHO	density (g/cm ³)				
SX	distance along streamtube (cm)				
SUMDOT	total mass flow bounded by streamtube (g/sec)				
C	species mass fraction				
X	species mole fraction				
WDOT	species production rate (g/cm ³ sec)				
only for viscous flows	<table border="0" style="margin-left: 20px;"> <tr> <td style="padding-right: 10px;"> F1</td> <td>total mass flow of species up to present streamtube (g/sec)</td> </tr> <tr> <td style="padding-right: 10px;"> ZJ</td> <td>species flux at tube interface</td> </tr> </table>	F1	total mass flow of species up to present streamtube (g/sec)	ZJ	species flux at tube interface
F1	total mass flow of species up to present streamtube (g/sec)				
ZJ	species flux at tube interface				

* X and R refer to the coordinates of the outer boundary of streamtube while the flow properties are average values across the streamtube.

Boundary layer parameters, if computed, are printed between the tube properties and the composition data in the last streamtube (at downstream print stations). The following parameters are printed:

RES	Reynolds number based on distance along wall
DELTAI	boundary layer thickness (incompressible)
THETA I	momentum thickness (incompressible)
CFI	skin friction coefficient (incompressible)
UFI	friction velocity (incompressible)
DISP	displacement thickness
DELTAC	boundary layer thickness
THETA C	momentum thickness
CFC	skin friction coefficient
H12	shape factor (= DISP/THETA C)
QWALL	heat flux at wall (cal/cm ² -sec)
TAUWALL	shear stress at wall (dynes/cm ²)
U/UE	B. L. velocity profile (UE = velocity in last tube)
T/TE	B. L. temperature profile (TE = temperature in last tube)
Y/DEL	location in B. L. (DEL ≡ DELTAC)

If IPART = 1, particle properties are printed; first, a NAMELIST of the input data on Cards 15 through 19 (and 20 through 23 for NC = 0), followed by the initial particle properties in each streamtube. The standard particle print gives the downstream and cross-stream velocity, temperature, Reynolds number and particle cloud density for each particle group and total particle momentum flux, for each streamtube. Since, typically, the limiting particle streamlines are within streamtube KMAX the outer tubes will contain no particles. Limiting particle streamlines are noted as, 'BOUNDARY OF PARTICLE PHASE AT S = X. XXXXX', followed by the radial position of the limiting particle streamline for each particle group. If the particles are initially in the liquid phase the program will print where they start to solidify.

APPENDIX D

FORTRAN LISTING

51	FUNPAT (1M, 10, 17MAD, OF DATA SETS ,15)	MAIN	56
	LU 5 (1, 1, 1, 1)	MAIN	57
105	CALL PUTIN	MAIN	58
	CALL PUTIN	MAIN	59
	CALL PUTIN	MAIN	60
	CALL PUTIN	MAIN	61
	CALL PUTIN	MAIN	62
	CALL PUTIN	MAIN	63
110	FUNPAT (110, 3, 12, 5, 2, 12)	MAIN	64
50	FUNPAT (110, 3, 12, 5, 2, 12)	MAIN	65
52	FUNPAT (110, 3, 12, 5, 2, 12)	MAIN	66
	ALPHADALSS	MAIN	67
	ALPHADALSS	MAIN	68
115	FORLNDSE	MAIN	69
	TULDR0	MAIN	70
	AMG81	MAIN	71
	CALL PUTIN	MAIN	72
120	CALL PUTIN	MAIN	73
	CALL PUTIN	MAIN	74
125	CALL PUTIN	MAIN	75
	CALL PUTIN	MAIN	76
	CALL PUTIN	MAIN	77
	CALL PUTIN	MAIN	78
	CALL PUTIN	MAIN	79
	CALL PUTIN	MAIN	80
	CALL PUTIN	MAIN	81
	CALL PUTIN	MAIN	82
130	CALL PUTIN	MAIN	83
	CALL PUTIN	MAIN	84
	CALL PUTIN	MAIN	85
	CALL PUTIN	MAIN	86
	CALL PUTIN	MAIN	87
	CALL PUTIN	MAIN	88
135	CALL PUTIN	MAIN	89
	CALL PUTIN	MAIN	90
	CALL PUTIN	MAIN	91
	CALL PUTIN	MAIN	92
	CALL PUTIN	MAIN	93
140	CALL PUTIN	MAIN	94
	CALL PUTIN	MAIN	95
	CALL PUTIN	MAIN	96
	CALL PUTIN	MAIN	97
	CALL PUTIN	MAIN	98
145	CALL PUTIN	MAIN	99
	CALL PUTIN	MAIN	100
	CALL PUTIN	MAIN	101
	CALL PUTIN	MAIN	102
	CALL PUTIN	MAIN	103
150	CALL PUTIN	MAIN	104
	CALL PUTIN	MAIN	105
	CALL PUTIN	MAIN	106
	CALL PUTIN	MAIN	107
	CALL PUTIN	MAIN	108
155	CALL PUTIN	MAIN	109
	CALL PUTIN	MAIN	110
	CALL PUTIN	MAIN	111
	CALL PUTIN	MAIN	112
	CALL PUTIN	MAIN	113
160	CALL PUTIN	MAIN	114
	CALL PUTIN	MAIN	115
	CALL PUTIN	MAIN	116
	CALL PUTIN	MAIN	117
165	CALL PUTIN	MAIN	118
	CALL PUTIN	MAIN	119
	CALL PUTIN	MAIN	120
	CALL PUTIN	MAIN	121
	CALL PUTIN	MAIN	122
	CALL PUTIN	MAIN	123
170	CALL PUTIN	MAIN	124
	CALL PUTIN	MAIN	125
	CALL PUTIN	MAIN	126
	CALL PUTIN	MAIN	127
	CALL PUTIN	MAIN	128
175	CALL PUTIN	MAIN	129
	CALL PUTIN	MAIN	130
	CALL PUTIN	MAIN	131
	CALL PUTIN	MAIN	132
	CALL PUTIN	MAIN	133
180	CALL PUTIN	MAIN	134
	CALL PUTIN	MAIN	135
	CALL PUTIN	MAIN	136
	CALL PUTIN	MAIN	137
	CALL PUTIN	MAIN	138
185	CALL PUTIN	MAIN	139
	CALL PUTIN	MAIN	140
	CALL PUTIN	MAIN	141
	CALL PUTIN	MAIN	142
	CALL PUTIN	MAIN	143
190	CALL PUTIN	MAIN	144
	CALL PUTIN	MAIN	145
	CALL PUTIN	MAIN	146
	CALL PUTIN	MAIN	147
	CALL PUTIN	MAIN	148
195	CALL PUTIN	MAIN	149
	CALL PUTIN	MAIN	150

	PSIBMTA	MAJN	151
	PSIBMTA	MAJN	152
	PSIBMTA(1,4,4)	MAJN	153
200	PSIBMTA(1,4,4)	MAJN	154
	PSIBMTA(1,4,4)	MAJN	155
	PSIBMTA(1,4,4)	MAJN	156
	PSIBMTA(1,4,4)	MAJN	157
	PSIBMTA(1,4,4)	MAJN	158
205	PSIBMTA(1,4,4)	MAJN	159
	PSIBMTA(1,4,4)	MAJN	160
	PSIBMTA(1,4,4)	MAJN	161
	PSIBMTA(1,4,4)	MAJN	162
	PSIBMTA(1,4,4)	MAJN	163
210	PSIBMTA(1,4,4)	MAJN	164
	PSIBMTA(1,4,4)	MAJN	165
	PSIBMTA(1,4,4)	MAJN	166
	PSIBMTA(1,4,4)	MAJN	167
	PSIBMTA(1,4,4)	MAJN	168
215	PSIBMTA(1,4,4)	MAJN	169
	PSIBMTA(1,4,4)	MAJN	170
	PSIBMTA(1,4,4)	MAJN	171
	PSIBMTA(1,4,4)	MAJN	172
	PSIBMTA(1,4,4)	MAJN	173
220	PSIBMTA(1,4,4)	MAJN	174
	PSIBMTA(1,4,4)	MAJN	175
	PSIBMTA(1,4,4)	MAJN	176
	PSIBMTA(1,4,4)	MAJN	177
	PSIBMTA(1,4,4)	MAJN	178
225	PSIBMTA(1,4,4)	MAJN	179
	PSIBMTA(1,4,4)	MAJN	180
	PSIBMTA(1,4,4)	MAJN	181
	PSIBMTA(1,4,4)	MAJN	182
	PSIBMTA(1,4,4)	MAJN	183
230	PSIBMTA(1,4,4)	MAJN	184
	PSIBMTA(1,4,4)	MAJN	185
	PSIBMTA(1,4,4)	MAJN	186
	PSIBMTA(1,4,4)	MAJN	187
235	PSIBMTA(1,4,4)	MAJN	188
	PSIBMTA(1,4,4)	MAJN	189
	PSIBMTA(1,4,4)	MAJN	190
	PSIBMTA(1,4,4)	MAJN	191
	PSIBMTA(1,4,4)	MAJN	192
240	PSIBMTA(1,4,4)	MAJN	193
	PSIBMTA(1,4,4)	MAJN	194
	PSIBMTA(1,4,4)	MAJN	195
	PSIBMTA(1,4,4)	MAJN	196
	PSIBMTA(1,4,4)	MAJN	197
	PSIBMTA(1,4,4)	MAJN	198
245	PSIBMTA(1,4,4)	MAJN	199
	PSIBMTA(1,4,4)	MAJN	200
	PSIBMTA(1,4,4)	MAJN	201
	PSIBMTA(1,4,4)	MAJN	202
	PSIBMTA(1,4,4)	MAJN	203
250	PSIBMTA(1,4,4)	MAJN	204
	PSIBMTA(1,4,4)	MAJN	205
	PSIBMTA(1,4,4)	MAJN	206
	PSIBMTA(1,4,4)	MAJN	207
	PSIBMTA(1,4,4)	MAJN	208
255	PSIBMTA(1,4,4)	MAJN	209
	PSIBMTA(1,4,4)	MAJN	210
	PSIBMTA(1,4,4)	MAJN	211
	PSIBMTA(1,4,4)	MAJN	212
	PSIBMTA(1,4,4)	MAJN	213
260	PSIBMTA(1,4,4)	MAJN	214
	PSIBMTA(1,4,4)	MAJN	215
	PSIBMTA(1,4,4)	MAJN	216
	PSIBMTA(1,4,4)	MAJN	217
265	PSIBMTA(1,4,4)	MAJN	218
	PSIBMTA(1,4,4)	MAJN	219
	PSIBMTA(1,4,4)	MAJN	220
	PSIBMTA(1,4,4)	MAJN	221
	PSIBMTA(1,4,4)	MAJN	222
	PSIBMTA(1,4,4)	MAJN	223
270	PSIBMTA(1,4,4)	MAJN	224
	PSIBMTA(1,4,4)	MAJN	225
	PSIBMTA(1,4,4)	MAJN	226
	PSIBMTA(1,4,4)	MAJN	227
	PSIBMTA(1,4,4)	MAJN	228
275	PSIBMTA(1,4,4)	MAJN	229
	PSIBMTA(1,4,4)	MAJN	230
	PSIBMTA(1,4,4)	MAJN	231
	PSIBMTA(1,4,4)	MAJN	232
	PSIBMTA(1,4,4)	MAJN	233
280	PSIBMTA(1,4,4)	MAJN	234
	PSIBMTA(1,4,4)	MAJN	235
	PSIBMTA(1,4,4)	MAJN	236
	PSIBMTA(1,4,4)	MAJN	237
	PSIBMTA(1,4,4)	MAJN	238
285	PSIBMTA(1,4,4)	MAJN	239
	PSIBMTA(1,4,4)	MAJN	240
	PSIBMTA(1,4,4)	MAJN	241
	PSIBMTA(1,4,4)	MAJN	242
290	PSIBMTA(1,4,4)	MAJN	243
	PSIBMTA(1,4,4)	MAJN	244
	PSIBMTA(1,4,4)	MAJN	245
	PSIBMTA(1,4,4)	MAJN	246

	SIGNS(100)	0410	207
	SIGNS(100)	0410	208
205	(SIN(100))	0410	209
	(SIN(100))	0410	210
	(SIN(100))	0410	211
	(SIN(100))	0410	212
	(SIN(100))	0410	213
300	(SIN(100))	0410	214
	(SIN(100))	0410	215
	(SIN(100))	0410	216
	(SIN(100))	0410	217
	(SIN(100))	0410	218
	(SIN(100))	0410	219
	(SIN(100))	0410	220
305	(SIN(100))	0410	221
	(SIN(100))	0410	222
	(SIN(100))	0410	223
	(SIN(100))	0410	224
	(SIN(100))	0410	225
	(SIN(100))	0410	226
	(SIN(100))	0410	227
	(SIN(100))	0410	228
	(SIN(100))	0410	229
	(SIN(100))	0410	230
310	(SIN(100))	0410	231
	(SIN(100))	0410	232
	(SIN(100))	0410	233
	(SIN(100))	0410	234
	(SIN(100))	0410	235
	(SIN(100))	0410	236
	(SIN(100))	0410	237
	(SIN(100))	0410	238
	(SIN(100))	0410	239
	(SIN(100))	0410	240
315	(SIN(100))	0410	241
	(SIN(100))	0410	242
	(SIN(100))	0410	243
	(SIN(100))	0410	244
	(SIN(100))	0410	245
	(SIN(100))	0410	246
	(SIN(100))	0410	247
	(SIN(100))	0410	248
	(SIN(100))	0410	249
	(SIN(100))	0410	250
320	(SIN(100))	0410	251
	(SIN(100))	0410	252
	(SIN(100))	0410	253
	(SIN(100))	0410	254
	(SIN(100))	0410	255
	(SIN(100))	0410	256
	(SIN(100))	0410	257
	(SIN(100))	0410	258
	(SIN(100))	0410	259
	(SIN(100))	0410	260
325	(SIN(100))	0410	261
	(SIN(100))	0410	262
	(SIN(100))	0410	263
	(SIN(100))	0410	264
	(SIN(100))	0410	265
	(SIN(100))	0410	266
	(SIN(100))	0410	267
	(SIN(100))	0410	268
	(SIN(100))	0410	269
	(SIN(100))	0410	270
330	(SIN(100))	0410	271
	(SIN(100))	0410	272
	(SIN(100))	0410	273
	(SIN(100))	0410	274
	(SIN(100))	0410	275
	(SIN(100))	0410	276
	(SIN(100))	0410	277
	(SIN(100))	0410	278
	(SIN(100))	0410	279
	(SIN(100))	0410	280
335	(SIN(100))	0410	281
	(SIN(100))	0410	282
	(SIN(100))	0410	283
	(SIN(100))	0410	284
	(SIN(100))	0410	285
	(SIN(100))	0410	286
	(SIN(100))	0410	287
	(SIN(100))	0410	288
	(SIN(100))	0410	289
	(SIN(100))	0410	290
340	(SIN(100))	0410	291
	(SIN(100))	0410	292
	(SIN(100))	0410	293
	(SIN(100))	0410	294
	(SIN(100))	0410	295
	(SIN(100))	0410	296
	(SIN(100))	0410	297
	(SIN(100))	0410	298
	(SIN(100))	0410	299
	(SIN(100))	0410	300
345	(SIN(100))	0410	301
	(SIN(100))	0410	302
	(SIN(100))	0410	303
	(SIN(100))	0410	304
	(SIN(100))	0410	305
	(SIN(100))	0410	306
	(SIN(100))	0410	307
	(SIN(100))	0410	308
	(SIN(100))	0410	309
	(SIN(100))	0410	310
350	(SIN(100))	0410	311
	(SIN(100))	0410	312
	(SIN(100))	0410	313
	(SIN(100))	0410	314
	(SIN(100))	0410	315
	(SIN(100))	0410	316
	(SIN(100))	0410	317
	(SIN(100))	0410	318
	(SIN(100))	0410	319
	(SIN(100))	0410	320
355	(SIN(100))	0410	321
	(SIN(100))	0410	322
	(SIN(100))	0410	323
	(SIN(100))	0410	324
	(SIN(100))	0410	325
	(SIN(100))	0410	326
	(SIN(100))	0410	327
	(SIN(100))	0410	328
	(SIN(100))	0410	329
	(SIN(100))	0410	330
360	(SIN(100))	0410	331
	(SIN(100))	0410	332
	(SIN(100))	0410	333
	(SIN(100))	0410	334
	(SIN(100))	0410	335
	(SIN(100))	0410	336
	(SIN(100))	0410	337
	(SIN(100))	0410	338
	(SIN(100))	0410	339
	(SIN(100))	0410	340
365	(SIN(100))	0410	341
	(SIN(100))	0410	342
	(SIN(100))	0410	343
	(SIN(100))	0410	344
	(SIN(100))	0410	345
	(SIN(100))	0410	346
	(SIN(100))	0410	347
	(SIN(100))	0410	348
	(SIN(100))	0410	349
	(SIN(100))	0410	350
370	(SIN(100))	0410	351
	(SIN(100))	0410	352
	(SIN(100))	0410	353
	(SIN(100))	0410	354
	(SIN(100))	0410	355
	(SIN(100))	0410	356
	(SIN(100))	0410	357
	(SIN(100))	0410	358
	(SIN(100))	0410	359
	(SIN(100))	0410	360

		AMIT(7,22)A(1),M(1),R(1),M(1),H(MAX),H(MAX),R22,H(MAX)	MAIN	347
305	221	FUMPA(2,12,5/2,5/2,12,5/2,12,5)	MAIN	348
		IF(I=1,0,1,0) GO TO 247	MAIN	349
		AMIT(7,22)H(MAX),H(MAX),PBI,PDI,TBI,UBI	MAIN	350
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	351
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	352
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	353
		IF(I=1,0,1) GO TO 247	MAIN	354
400		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	355
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	356
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	357
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	358
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	359
405	200	IF(I=1,0,1) GO TO 211	MAIN	360
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	361
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	362
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	363
410		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	364
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	365
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	366
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	367
415	C	AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	368
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	369
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	370
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	371
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	372
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	373
420		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	374
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	375
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	376
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	377
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	378
425		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	379
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	380
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	381
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	382
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	383
430	42	AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	384
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	385
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	386
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	387
435		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	388
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	389
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	390
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	391
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	392
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	393
440		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	394
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	395
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	396
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	397
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	398
445		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	399
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	400
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	401
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	402
450	247	AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	403
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	404
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	405
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	406
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	407
455		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	408
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	409
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	410
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	411
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	412
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	413
460		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	414
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	415
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	416
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	417
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	418
465		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	419
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	420
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	421
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	422
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	423
470		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	424
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	425
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	426
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	427
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	428
475		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	429
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	430
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	431
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	432
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	433
480		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	434
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	435
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	436
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	437
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	438
485		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	439
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	440
		AMIT(7,22)R(STRM(1),I(1),M(1))	MAIN	441
		AMIT(7,22)C(STRM(1),I(1),M(1))	MAIN	442

```

505 070 AA(N)M(I)M2(N)M2(I=1)MDELTOOGLY(N)      MAIN      530
      V2(N)M2(I=1)MDELTY(N)                   MAIN      530
      CALL STP                                  MAIN      540
      IF ((IFLAG,NE,0).AND.(N=1),LE,MM2),AND.(IPART,NE,0)) CALL PUNT
      IF (M2ALANM) 005,005,077                MAIN      542
500 077 M2ALANM                                MAIN      543
      IF (ICOUNT=25)250,0010,0010            MAIN      546
      005 IF (MUCASE) 0060,000,0060           MAIN      545
      000 IF (MUNE) 1000,000,1000             MAIN      546
      000 IF (MUGSM) 000,020,000             MAIN      547
      000 CALL MPTOUT(3)                       MAIN      548
505 JNYZ010                                    MAIN      549
      IF (IPART,NE,0) CALL PPTOUT(3)          MAIN      550
070 NAME1                                      MAIN      551
      IF (M2ALANM) 500,1000,1000             MAIN      552
      M2ALANM COMPANY (M2ALANM)              MAIN      553
000 1000 M2ALANM                              MAIN      554
      M2ALANM                                MAIN      555
      M2ALANM,0                               MAIN      556
      CALL M2ALANM                             MAIN      557
      IF (M2ALANM) 001,1020,001              MAIN      558
005 1020 M2ALANM                              MAIN      559
      IF (M2ALANM) 1030,1030,077             MAIN      560
      1030 IF (MUCASE) 0060,1050,0060        MAIN      561
      1050 V2(N)M2(I=1)MDELTY(N)             MAIN      562
      IF (MUGSM) 1000,1070,1000             MAIN      563
010 1000 CALL MPTOUT(3)                       MAIN      564
      JNYZ011                                  MAIN      565
      IF (IPART,NE,0) CALL PPTOUT(3)          MAIN      566
      1070 IF (IFLAG) 1000,1150,1000         MAIN      567
      1500 M2ALANM,001                         MAIN      568
015 001 1100 L=2,M=2                          MAIN      569
      J=1                                       MAIN      570
      M(J)M2(I)                                MAIN      571
      A(J)M2(I)                                MAIN      572
      M(I)M2(J)                                MAIN      573
020 IF (J=1) 1100,1100,1062                  MAIN      574
      M(I)M2(I)                                MAIN      575
      M(J)M2(I)                                MAIN      576
      M(I)M2(I)                                MAIN      577
      M(I)M2(I)                                MAIN      578
025 M2ALANM,0                               MAIN      579
      001 1000 L=1,M=1                          MAIN      580
      1000 M2ALANM(2,1,0,0)M2ALANM(1)       MAIN      581

      M2ALANM,0/200                             MAIN      582
      M2ALANM,0/200                             MAIN      583
030 001 1000 L=1,M=1                          MAIN      584
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      585
      1000 M2ALANM(1,1,0,0)M2ALANM(1,1)       MAIN      586
      M2ALANM,0                               MAIN      587
      001 1000 L=1,M=1                          MAIN      588
      001 1000 L=1,M=1                          MAIN      589
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      590
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      591
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      592
040 1100 L=1,M=1                          MAIN      593
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      594
      1000 M2ALANM(1,1,0,0)M2ALANM(1,1)       MAIN      595
      IF ((IFLAG,NE,0).AND.(IPART,NE,0))CALL PUNT
      M2ALANM,0                               MAIN      596
      M2ALANM,0                               MAIN      597
      M2ALANM,0                               MAIN      598
045 1000 M2ALANM                              MAIN      599
      IF (M2ALANM) 201,1130,201              MAIN      600
      201 IF (M2ALANM) 60 TO 200              MAIN      601
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      602
050 111. M2ALANM                              MAIN      603
      CALL MPTOUT(3)                           MAIN      604
      IF ((IFLAG,NE,0).AND.(IPART,NE,0))CALL PPTOUT(3)
      001 1000 L=1,M=1                          MAIN      605
      1150 IF (IPART,NE,0) CALL M2ALANM
      M2ALANM,0                               MAIN      606
055 001 1000 L=1,M=1                          MAIN      607
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      608
      001 1000 L=1,M=1                          MAIN      609
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      610
      001 1000 L=1,M=1                          MAIN      611
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      612
      001 1000 L=1,M=1                          MAIN      613
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      614
      001 1000 L=1,M=1                          MAIN      615
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      616
      001 1000 L=1,M=1                          MAIN      617
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      618
065 001 1000 L=1,M=1                          MAIN      619
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      620
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      621
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      622
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      623
070 001 1000 L=1,M=1                          MAIN      624
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      625
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      626
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      627
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      628
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      629
075 001 1000 L=1,M=1                          MAIN      630
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      631
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      632
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      633
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      634
080 001 1000 L=1,M=1                          MAIN      635
      M2ALANM(1,1,0,0)M2ALANM(1,1)           MAIN      636

```


100	C	FJAEU WALL	BNOMY	52
		300 IF (S-SM(N+1)) 500,500,400	BNOMY	53
		400 NBM+1	BNOMY	54
		IF (N-MMAX) 300,300,4030	BNOMY	55
		500 N=1	BNOMY	56
		Y(K)ZU,0	BNOMY	57
105		Y2(K)ZU,0	BNOMY	58
		PH12(K)XPH1(K)+(PH1(N+1)-PH1(N))*(S-SM(N))/(SM(N+1)-SM(N))	BNOMY	59
		IF (ABS(PH1(N+1)-PH1(N))-10.00(-4)) 525,525,550	BNOMY	60
		525 X2(N)XN(N)+(XN(N+1)-XN(N))*(S-SM(N))/(SM(N+1)-SM(N))	BNOMY	61
		X2(N)XN(N)+(XN(N+1)-XN(N))*(S-SM(N))/(SM(N+1)-SM(N))	BNOMY	62
110		GU TO 575	BNOMY	63
		550 NADU=U2LSU/(PH1(N)-PH12(N))	BNOMY	64
		XUB(N)=NADU*SIN(PH1(N))	BNOMY	65
		XUB(N)=NADU*COS(PH1(N))	BNOMY	66
		X2(N)XU=NADU*SIN(PH12(N))	BNOMY	67
115		X2(N)XADU=(U*(PH12(N)))/NO	BNOMY	68
		575 IF (IPLAG) 000,400,000	BNOMY	69
		000 NNAK(N)=S*(N(K)+N2(K))	BNOMY	70
		N2(K)=S*(N(K+1)+N2(K+1))	BNOMY	71
		YANAK=5*(Y(N+1)+Y2(N+1))	BNOMY	72
120		YBAN=0	BNOMY	73
		UBAK(N+1)=SGWT(.5*(U(N+1)+U2(N+1)+2))	BNOMY	74
		RMBAN=5*(NMU(N+1)+NMU2(K+1))	BNOMY	75
		TABAN=5*(T(N+1)+T2(N+1))	BNOMY	76
		PABAN=5*(P(N+1)+P2(N+1))	BNOMY	77
125		NA2=0,0	BNOMY	78
		OU 700 I=1,NUS	BNOMY	79
		CABAN(I)=S*(C(I,N+1)+C2(I,N+1))	BNOMY	80
		700 NA2=PM2+CABAN(I)/NA(I)	BNOMY	81
		NA2=1,0/N=2	BNOMY	82
130		OU 800 I=1,NUS	BNOMY	83
		XABAN(I)=S*(X(I,N+1)+X2(N+1))	BNOMY	84
		M2P(I)=0,0	BNOMY	85
		M2P(I)=0,0	BNOMY	86
		OU 800 J=1,NH	BNOMY	87
135		M2P(I)=S*(M2P(I)+A(I,J)+TABAN(N+J=2))	BNOMY	88
		GU TO 1050	BNOMY	89
		900 N2AN(N)=S*(N)	BNOMY	90
		N2AN(N+1)=S*(N+1)	BNOMY	91
		Y2AN(N+1)=S*(Y(N+1)+Y2(N+1))	BNOMY	92
140		YBAN=0	BNOMY	93
		UBAN(N+1)=S*(U(N+1)+U2(N+1))	BNOMY	94
		N2AN(N+1)=S*(N(N+1)+N2(N+1))	BNOMY	95
		TABAN(N+1)=S*(T(N+1)+T2(N+1))	BNOMY	96
		PABAN(N+1)=S*(P(N+1)+P2(N+1))	BNOMY	97
145		OU 1000 I=1,NUS	BNOMY	98
		CABAN(I)=S*(C(I,N+1)+C2(I,N+1))	BNOMY	99
		XABAN(I)=S*(X(I,N+1)+X2(N+1))	BNOMY	100
		M2P(I)=0,0	BNOMY	101
		M2P(I)=0,0	BNOMY	102
150		OU 1000 J=1,NH	BNOMY	103
		M2P(I)=S*(M2P(I)+A(I,J)+TABAN(N+J=2))	BNOMY	104
		1050 IF (IEXIA(I)) 1100,1200,1100	BNOMY	105
		1100 MUL=0	BNOMY	106
		1060 I=1,NUS	BNOMY	107
155		CALL TRANSF (TAN,PHAN,AK,MUL,I,IG)	BNOMY	108
		1200 TAN(K)=0,0	BNOMY	109
		G(K)=0,0	BNOMY	110
		OU 1300 I=1,NUS	BNOMY	111
160		J(I,A)=0,0	BNOMY	112
		1500 IF (NMAA+AA) 1510,1510,1520	BNOMY	113
		1510 AA=2	BNOMY	114
		GU TO 10000	BNOMY	115
		AA=3	BNOMY	116
165		OU TL 10000	BNOMY	117
		C	BNOMY	118
		OU TL 10000	BNOMY	119
		5100 N=1	BNOMY	120
		IF (LL) 5110,5110,5190	BNOMY	121
		5110 USINH((X(N)-X(N+1))*2*(N(N)-N(N+1))+2)	BNOMY	122
170		IF (ABS(PH1(N)-PH1(N+1))-1.0(-04)) 5120,5120,5130	BNOMY	123
		5120 S=(P)ZU	BNOMY	124
		GU TO 5190	BNOMY	125
		5130 S=(P)ZU*(PH1(N)-PH1(N+1))/(2.0*SIN(0.5*(PH1(N)-PH1(N+1))))	BNOMY	126
		5140 GU TO (3/00,5000,5100,5700), 1K1AD	BNOMY	127
175		C	BNOMY	128
		FJAEU WALL	BNOMY	129
		5200 N=1	BNOMY	130
		SSX=X(N)+0.5*(N-1)	BNOMY	131
		5300 IF (SSX-SM(N+1)) 5500,5500,5600	BNOMY	132
		5400 N=1	BNOMY	133
180		IF (L=NA) 5500,9050,9050	BNOMY	134
		5500 IF (ABS(PH1(N+1)-PH1(N))-1.0(-04)) 5600,5600,5700	BNOMY	135
		5600 PH1(N)=S*(PH12(N+1)+PH1(N))	BNOMY	136
		PH1(N)=S*(PH1(N)+PH12(N))	BNOMY	137
		IF (ABS(PH1(N)-PH1(N+1))-1.0(-04)) 5620,5620,5640	BNOMY	138
185		5620 X2(N)X2(N+1)	BNOMY	139
		GU TO 5600	BNOMY	140
		5640 X2(N)X2(N+1)=X(N)+I*(PH1(N)-PH1(N+1))*X2(N+1)/I*(PH1(N)+PH1(N+1))	BNOMY	141
		X2(N)X2(N+1)=I*(PH1(N)+PH1(N+1))*X2(N+1)/I*(PH1(N)-PH1(N+1))	BNOMY	142
190		5660 X2(N)X2(N+1)=I*(PH1(N)+PH1(N+1))*X2(N+1)/I*(PH1(N)-PH1(N+1))	BNOMY	143
		PH12(N)=S*(PH1(N)+PH12(N))	BNOMY	144
		GU TO 4200	BNOMY	145
		5700 NADU=(S*(N+1)-S(N))/(PH1(N)-PH1(N+1))	BNOMY	146
		ULLRSU=I*(X(N+1)-X(N))*2*(N(N+1)-N(N))	BNOMY	147
		ULLRSU=I*(NADU+2*ULLRSU/2,0)	BNOMY	148

195	P IUTANATAN((NB(N+1)-NB(N))/(NB(N+1)+NB(N)))	BNDRY	148
	IF (NADD) 3740,3740,3720	BNDRY	149
3720	XNDB(N)+.5*(NB(N+1)-NB(N))*ELL2*SBIN(F IUTA)	BNDRY	150
	YNDB(N)+.5*(NB(N+1)-NB(N))*ELL2*CBIN(F IUTA)	BNDRY	151
	GU TL 3800	BNDRY	152
200	3740 XNDB(N)+.5*(NB(N+1)-NB(N))*ELL2*SBIN(F IUTA)	BNDRY	153
	YNDB(N)+.5*(NB(N+1)-NB(N))*ELL2*CBIN(F IUTA)	BNDRY	154
	5800 L0	BNDRY	155
	ENDB(N)+ELL2*(N+1)*COS(PH1(N))	BNDRY	156
	IF (NADD) 3820,3840,3840	BNDRY	157
205	3820 AMMHN=SUMI(NADD*2*(ENB=XX)**2)	BNDRY	158
	GU TL 3900	BNDRY	159
3840	AMMHN=SUMI(NADD*2*(ENB=XX)**2)	BNDRY	160
3900	PH1BAN=.5*(PH12(N-1)+ATAN((ENB=XX)/(NB-AMR)))	BNDRY	161
	TPMHN=TAN(PH1BAN)	BNDRY	162
210	GE((N2(N-1)-NN-(ENB=2(N-1)))/TPMHN)**2-NADD*2*(ENB=XX)**2	BNDRY	163
	DNDB(ENB=XX)/SIN(NADD*2*(ENB=XX)**2)	BNDRY	164
	IF (NADD) 3910,3910,3905	BNDRY	165
3905	DNDB=DNDB	BNDRY	166
3910	DPND=.5*((AMH-NN)*(ENB=XX)*DNDB)/NADD*2	BNDRY	167
215	DPND*PH1/(CUS(PH1BAN))**2	BNDRY	168
	GE((ENB=2*(ENB=XX)-(N2(N-1)-NN-(ENB=2(N-1)))/TPMHN)**2)/	BNDRY	169
	TPMHN -(ENB=2*(ENB=XX)-N2(N-1)-NN-(ENB=2(N-1)))/TPMHN	BNDRY	170
	DNDB=GE((ENB=2*(ENB=XX)-N2(N-1)-NN-(ENB=2(N-1)))/TPMHN	BNDRY	171
	ELL2)	BNDRY	172
220	ENDB(N)+ELL2	BNDRY	173
	IF (NADD) 3920,3940,3940	BNDRY	174
3920	AMMHN=SUMI(NADD*2*(ENB=XX)**2)	BNDRY	175
	GU TL 4000	BNDRY	176
3940	AMMHN=SUMI(NADD*2*(ENB=XX)**2)	BNDRY	177
225	4000 IF (L=4) 3900,4100 3100	BNDRY	178
4100	N2(N)=AMH	BNDRY	179
	PH12(N)=ATAN((ENB=XX)/(AMH-NN))	BNDRY	180
		BNDRY	181
		BNDRY	182
230	4200 U=SGN((N2(N)-N(N))**2+(N2(N)-N(N))**2)	BNDRY	183
	IF (ABS(PH12(N)-PH1(N))-1.E-04) 4300,4300,4400	BNDRY	184
4300	DEL(N)=0	BNDRY	185
	GU TL 4500	BNDRY	186
4400	DEL(N)=((PH12(N)-PH1(N))/(2.*SIN(0.5*(PH12(N)-PH1(N))))	BNDRY	187
4500	PH12(N)-N2(N-1))**2+(N2(N)-N2(N-1))**2)	BNDRY	188
235	IF (ABS(PH12(N)-PH12(N-1))-1.E-04) 4600,4600,4700	BNDRY	189
4600	DEL(N)=0	BNDRY	190
	GU TL 4800	BNDRY	191
4700	DEL(N)=((PH12(N)-PH12(N-1))/(2.*SIN(0.5*(PH12(N)-PH12(N-1))))	BNDRY	192
4800	ABS(PH12(N)-PH12(N-1))**2+(PH12(N)-PH12(N-1))**2)	BNDRY	193
240	DEL(N)=DEL(N)+DEL(N)	BNDRY	194
	GU TL 5000	BNDRY	195
4900	DEL(N)=0	BNDRY	196
	GU TL 10000	BNDRY	197
245	PH1 BAN=ATAN(ENB=XX)	BNDRY	198
5000	PH1 BAN=ATAN(ENB=XX)	BNDRY	199
	PH1BAN(N)+PH1BAN(N+1)-PH1BAN(N)+S2X*SB(N)/(SB(N+1)+SB(N))	BNDRY	200
	GU TL 5700	BNDRY	201
250	PH1BAN=ATAN(ENB=XX)	BNDRY	202
	PH1BAN=0	BNDRY	203
	PH1BAN=0	BNDRY	204
	PH1BAN=0	BNDRY	205
	PH1BAN=0	BNDRY	206
	PH1BAN=0	BNDRY	207
255	5400 IF (L=4) 5520,5500,5525	BNDRY	208
5500	PH1BAN=0	BNDRY	209
	GU TL 5550	BNDRY	210
5525	PH1BAN=0	BNDRY	211
5550	IF (SIGMA) 5575,5575,5600	BNDRY	212
5575	PH1BAN=0	BNDRY	213
260	GU TL 5700	BNDRY	214
5600	IF (SIGMA=0.5*PI) 5650,5650,5675	BNDRY	215
5675	PH1BAN=0	BNDRY	216
	GU TL 5700	BNDRY	217
265	PH1BAN=0	BNDRY	218
	IF (PH1BAN=0) ANGLE(0.990) K, FLAG, Y(N), Y(N+1), MM(N), U(N),	BNDRY	219
	Y2(N), Y2(N-1), MM(N), U2(N), SIGMA, PH1	BNDRY	220
	PH1BAN=0	BNDRY	221
	PH1BAN=0	BNDRY	222
270	5800 LPH1(N)=LPH1(N)+LPH1(N)/(Y(N)-Y(N-1))	BNDRY	223
	LPH1(N)=LPH1(N)+LPH1(N)/(MM(N)+U(N)**2)	BNDRY	224
	GU TL 6000	BNDRY	225
5900	LPH1(N)=LPH1(N)+LPH1(N)/(Y2(N)-Y2(N-1))	BNDRY	226
	LPH1(N)=LPH1(N)+LPH1(N)/(MM(N)+U2(N)**2)	BNDRY	227
	LPH1(N)=LPH1(N)+LPH1(N)/(MM(N)+U(N)**2)	BNDRY	228
275	6000 MU=1	BNDRY	229
	RETCN	BNDRY	230
	PH1BAN=0	BNDRY	231
280	PH1BAN=0	BNDRY	232
	GU TL 4000	BNDRY	233
4000	PH1BAN=0	BNDRY	234
	GU TL 4000	BNDRY	235
4150	PH1BAN=0	BNDRY	236
4300	RETCN	BNDRY	237
285	10000 RETCN	BNDRY	238

END

BNDRY 239

```

SUBROUTINE CHEM(IRINE)
COMMON A(25,7) ,AA(40) ,ALFA(25,25),ALPHAM ,ALPHAP ,CHEM
1 ,ATOL ,BETAP ,BMR ,C(1,36) ,C(25,40)
2 ,C12(25) ,C2(25,40) ,CABAR(25) ,COP=9(25) ,CP
3 ,CP8(25) ,CP3M ,LBM(25) ,C2=1(25) ,CSTEM(25)
4 ,MASTMA ,OZIN(25,25) ,OZFF(25) ,OZTFF(25) ,DELTA
5 ,D11 ,DELBB(40) ,DELB ,DELBO ,DLB(40)
6 ,DIM(25,25) ,D12 ,DEL(40) ,D13 ,DPBY(40)
7 ,D14 ,DPHOB(40) ,EPCOM
10 ,EPBLUN ,EATNA(50) ,FSTEP ,FNAB ,GAD
4 ,H11 ,H(40) ,HM ,HJ ,HMH(25)
,HPM(25) ,HPM(25) ,ILUNB ,ILUUN ,IUEIN(25)
1 ,IENNOB ,IEITNA(50) ,IFLAG ,IRISOB
2 ,IPYUC ,ISOCR ,ITYPE ,IPO
15 ,IOIFF ,R ,RAY ,RAYB ,RAYZ
4 ,ALO ,ANA)
5 ,AUP ,AR ,L ,LPLANE ,MA
6 ,MASH ,MLUT ,MAX ,MUO ,MU
7 ,MU2 ,MUB ,PM ,RMZ ,MMOM
20 ,MBOUND ,MDS ,NITEM ,NMAX ,NM
1 ,COMMON NUCABE ,OMEGA(25) ,P11 ,P140 ,P12
2 ,P2(40) ,PB(50) ,PABAR ,PGBAR
3 ,PMS ,P15 ,PHI(40) ,PHI(50)
4 ,PHI51 ,PHSTIM ,PH ,PH(40,50) ,P18
25 ,PHI2(40) ,PHIB ,PI ,PH(25) ,P20
5 ,PBM ,P31 ,PBTREM ,P40 ,PH(50)
6 ,P11 ,PATR1 ,QTRZ
7 ,QA(50) ,P11 ,R(40) ,RB(50)
30 ,MHS ,P13 ,RBAH(40) ,M14 ,R2(40)
4 ,MCON ,P15 ,RE(40) ,REB ,R16
,MMO(40) ,R17 ,RMQZ(40) ,RMB(25) ,RMBH
1 ,MMS(M) ,MABAB ,RMBAB ,RU ,RN
2 ,RSTREM ,RY ,R19 ,RN(100)
3 ,SB(50) ,SC(25) ,SH(50)
35 ,S13 ,SX(40) ,T11 ,T(40)
4 ,T2(40) ,TABAN ,TBAN ,T(25) ,T14
6 ,TAN(40) ,TINI ,TXTRZ ,TUL ,TBM
7 ,TBM ,TSTREM ,Tn(50) ,TNO
8 ,TS ,US1 ,U(40) ,U12 ,U2(40)
40 ,COMMON UTR1 ,UTR2 ,U14 ,UBAR(40) ,USH
1 ,USH1 ,USTREM ,UN(50) ,UNS
2 ,X11 ,X(40) ,X12 ,X2(40)
3 ,XB(50) ,XBAN(25) ,XBAN(25) ,X15
45 ,XB(25,40) ,XBH ,XSTREM ,XB(50)
5 ,XMB(40) ,X11 ,T(40) ,Y12
6 ,Y2(40) ,YABAR ,YBAN ,Y2(25) ,Y11(25)
7 ,YJ(25,40) ,ZMA ,ZAX ,ZB ,ZBH
9 ,Y(2,40) ,Y(2,40) ,YMH(2,40) ,Y(2,40) ,Y1(2,40)
50 ,YU(2,40) ,YNDL(2,40) ,YNDR(2,40) ,YFC(2,40,25),CAMD1
,
NEAL ,RAY ,RAYB(25) ,RAYZ ,RN ,MDO(40)
,MA(40) ,MUB(25) ,MUS(25) ,MUO(25) ,MU2
,MA(25) ,MAY ,MASH ,MASH1 ,MASH
MAP
55 DIMENSION L(25),MP(25),HM(25)
COMMON/CHEM1/ ZIG(5), INR(40),INI(40),MC(40,3),IARR(40,5),
1 AV,CM(20,20),F1(20),NP(20),
CHEM
2
2 AV(20),F0U(20,40),C1(20),WA(20)
COMMON/INUGS1/ IPUGM
CHEM
3 COMMON/XYZ/XYZ,Z,JYZ
CHEM
4 GAS CONSTANT IN 1.907 CAL/G-MOLE=
XYZB3
CHEM
5 JYZB0
CHEM
6 HM(1,90)=1(R)
CHEM
7 GAS CONSTANT IN 02.00 CM3-ATM/G-MOLE=R
HM(1,90)=1(R)
CHEM
8 ALGTRALOG(1(R))
CHEM
9 ALGTRALOG(1(R)/1000.0)
CHEM
10 UU 3 INR1,ADS
CHEM
11 F1(1,1)UC(1,1,1)/HM(1)
CHEM
12 G1(1)RU,0
CHEM
13 P1(1)RU,0
CHEM
14 HM(1)RU,0
CHEM
15 ADU(1,1)RU,0
CHEM
16 CU 4 JHE1,ADS
CHEM
17 CM(1,1,1)RU,0
CHEM
18 3 CONTINUE
CHEM
19 CALCULATE GIBBS ENERGY
CHEM
20 DU 6 INR1,ADS
CHEM
21 G(1)=AA(1,1)+.5/T(R)+(A(1,3)+(.1-ALLU))=LS1(1)+T(R)+A(1,2)
CHEM
22 IF (AA,1,4) GO TO 7
CHEM
23 G(1)=G(1)+A(1,4)+T(R)+.2+.5*A(1,5)+T(R)+.3=A(1,6)/S,OT(R)+.4
CHEM
24 7 G(1)= G(1)+MM(1)
CHEM
25 C CALCULATE GIBBS ENERGY IN CAL/MOLE
CHEM
26 D CONTINUE
CHEM
27 DU 1 INR1,IPINE
CHEM
28 N1IR(1)(1)
CHEM
29 GU TO (0),042,043,044,045,046,047,048),RINI
CHEM
30 HRFAC(1,1)=T(R)+NC(1R,2)*EXP(NC(1R,3)/RRT)
CHEM
31 GU TO 049
CHEM
32 HRFAC(1,1)
CHEM
33 DU TO 049
CHEM
34 HRFAC(1,1)/T(R)
CHEM
35 GU TO 049
CHEM
36 HRFAC(1,1)/T(R)/T(R)
CHEM
37 GU TO 049
CHEM
38 HRFAC(1,1)/50-T(1(R))
CHEM
39 (1)TU 049
CHEM
40

```

100	005 NAFARC(IN,1)*EXP(ARC(IN,3)/NRT)	CHEM	52
	GO TO 009	CHEM	53
	006 NAFARC(IN,1)*EXP(ARC(IN,3)/NRT)/T(N)	CHEM	54
	GO TO 009	CHEM	55
	007 NAFARC(IN,1)/T(N)/SQRT(T(N))	CHEM	56
	009 CONTINUE	CHEM	57
105	A1MM=MM(IN)	CHEM	58
	GO TO (051,052,053,054,055,056,057,058,059,000),A1MM	CHEM	59
	051 J1=IMM(IN,1)	CHEM	60
	J2=IMM(IN,2)	CHEM	61
	J3=IMM(IN,3)	CHEM	62
	J4=IMM(IN,4)	CHEM	63
110	E=(G(J1)+G(J2)-G(J3)-G(J4))/NRT	CHEM	64
	IF (E.LT.-40.0) E=-40.0	CHEM	65
	IF (E.GT.+40.0) E=40.0	CHEM	66
	WRITE(E)	CHEM	67
115	CHNHRFAMU(K)	CHEM	68
	MP(IN)=CHNRF(J1)+F(J2)	CHEM	69
	NM(IN)=CHNRF(J3)+F(J4)/E	CHEM	70
	DO 773 J=1,4	CHEM	71
	SIGN=1.0	CHEM	72
120	IF (J.GT. 2) SIGN=-1.0	CHEM	73
	IMU=IMM(IN,J)	CHEM	74
	CM(INU,J1)=CM(INU,J1)+SIGN*CHNRF(J2)	CHEM	75
	CM(INU,J2)=CM(INU,J2)+SIGN*CHNRF(J1)	CHEM	76
	CM(INU,J3)=CM(INU,J3)-SIGN*CHNRF(J4)/E	CHEM	77
125	CM(INU,J4)=CM(INU,J4)-SIGN*CHNRF(J3)/E	CHEM	78
	UX(INU)=UX(INU)+SIGN*(MP(IN)-NM(IN))	CHEM	79
	773 CONTINUE	CHEM	80
	GO TO 007	CHEM	81
	052 J1=IMM(IN,1)	CHEM	82
130	J2=IMM(IN,2)	CHEM	83
	J3=IMM(IN,3)	CHEM	84
	E=(G(J1)+G(J2)-G(J3))/NRT	CHEM	85
	IF (E.LT.-40.0) E=-40.0	CHEM	86
	IF (E.GT.+40.0) E=40.0	CHEM	87
135	WRITE(E)	CHEM	88
	CHNHRFAMU(K)/Z=0.0	CHEM	89
	MP(IN)=CHNRFAMU(K)+F(J1)+F(J2)	CHEM	90
	NM(IN)=CHNRF(J3)/E	CHEM	91
140	DO 774 J=1,3	CHEM	92
	SIGN=1.0	CHEM	93
	IF (J.GT. 2) SIGN=-1.0	CHEM	94
	IMU=IMM(IN,J)	CHEM	95
	CM(INU,J1)=CM(INU,J1)+SIGN*CHNRFAMU(K)+F(J2)	CHEM	96
	CM(INU,J2)=CM(INU,J2)+SIGN*CHNRFAMU(K)+F(J1)	CHEM	97
145	CM(INU,J3)=CM(INU,J3)-SIGN*CHNRFAMU(K)/E	CHEM	98
	UX(INU)=UX(INU)+SIGN*MP(IN)	CHEM	99
	774 CONTINUE	CHEM	100
	GO TO 006	CHEM	101
	053 J1=IMM(IN,1)	CHEM	102
150	J2=IMM(IN,2)	CHEM	103
	J3=IMM(IN,3)	CHEM	104
	J4=IMM(IN,4)	CHEM	105
	J5=IMM(IN,5)	CHEM	106
	E=(G(J1)+G(J2)-G(J3)-G(J4)-G(J5))/NRT	CHEM	107
155	IF (E.LT.-40.0) E=-40.0	CHEM	108
	IF (E.GT.+40.0) E=40.0	CHEM	109
	WRITE(E)	CHEM	110
	CHNHRFAMU(K)	CHEM	111
	MP(IN)=CHNRF(J1)+F(J2)	CHEM	112
160	NP(IN)=CHNRF(J3)+F(J4)+F(J5)+AMU(K)/NRT/E	CHEM	113
	DO 775 J=1,5	CHEM	114
	SIGN = 1.0	CHEM	115
	IF (J.GT. 2) SIGN =-1.0	CHEM	116
	IMU=IMM(IN,J)	CHEM	117
165	CM(INU,J1)=CM(INU,J1)+SIGN*CHNRF(J2)	CHEM	118
	CM(INU,J2)=CM(INU,J2)+SIGN*CHNRF(J1)	CHEM	119
	CM(INU,J3)=CM(INU,J3)-SIGN*CHNRF(J4)+F(J5)+AMU(K)/NRT/E	CHEM	120
	CM(INU,J4)=CM(INU,J4)-SIGN*CHNRF(J5)+F(J3)+AMU(K)/NRT/E	CHEM	121
	CM(INU,J5)=CM(INU,J5)-SIGN*CHNRF(J3)+F(J4)+AMU(K)/NRT/E	CHEM	122
170	UX(INU)=UX(INU)+SIGN*(MP(IN)+2.0*NM(IN))	CHEM	123
	775 CONTINUE	CHEM	124
	GO TO 001	CHEM	125
	054 J1=IMM(IN,1)	CHEM	126
175	J2=IMM(IN,2)	CHEM	127
	J3=IMM(IN,3)	CHEM	128
	E=(G(J1)+G(J2)-G(J3))/NRT	CHEM	129
	IF (E.LT.-40.0) E=-40.0	CHEM	130
	IF (E.GT.+40.0) E=40.0	CHEM	131
	WRITE(E)	CHEM	132
180	CHNHRFAMU(K)	CHEM	133
	MP(IN)=CHNRF(J1)+F(J2)	CHEM	134
	NM(IN)=CHNRF(J3)/E	CHEM	135
	DO 776 J=1,3	CHEM	136
	SIGN=1.0	CHEM	137
185	IF (J.GT. 2) SIGN=-1.0	CHEM	138
	IMU=IMM(IN,J)	CHEM	139
	CM(INU,J1)=CM(INU,J1)+SIGN*CHNRFAMU(K)+F(J2)	CHEM	140
	CM(INU,J2)=CM(INU,J2)+SIGN*CHNRFAMU(K)+F(J1)	CHEM	141
	CM(INU,J3)=CM(INU,J3)-SIGN*CHNRFAMU(K)/E	CHEM	142
190	UX(INU)=UX(INU)+SIGN*MP(IN)	CHEM	143
	776 CONTINUE	CHEM	144
	GO TO 000	CHEM	145
	055 J1=IMM(IN,1)	CHEM	146

195	J2=ALD+1	CHEM	147
	J3=IHHH(IH,3)	CHEM	148
	J4=IHHH(IH,4)	CHEM	149
	EM(E(J1)-G(J3)-G(J4))/MMT	CHEM	150
	IF (E .LT. +80.0) EM =80.0	CHEM	151
	IF (E .GT. +80.0) EM=0.0	CHEM	152
200	ENDP (E)	CHEM	153
	CH=ANF*OHMU(R)/ZM	CHEM	154
	MP(IH)CCH=OF(I(J1))	CHEM	155
	MM(IH)CCH=OHRTI=OHMU(R)+F1(J3)+F1(J4)/E	CHEM	156
	DU 772 J=1.4	CHEM	157
205	IF (J .EQ. 2) GO TO 772	CHEM	158
	SIGN=1.0	CHEM	159
	IF (J .GT. 2) SIGN=-1.0	CHEM	160
	IMUP=IHHH(IH,J)	CHEM	161
210	CM(IHUN,J)ICCP(IHUN,J1)+SIGN*CMH	CHEM	162
	CM(IHUN,J2)ICCP(IHUN,J3)+SIGN*CHRORTI=OHMU(R)+F1(J4)/E	CHEM	163
	CM(IHUN,J4)ICCP(IHUN,J4)+SIGN*CHRORTI=OHMU(R)+F1(J3)/E	CHEM	164
	GR(IHUN)GR(IHUN)+SIGN*MP(IH)	CHEM	165
	772 CONTINUE	CHEM	166
	GO TO 807	CHEM	167
215	850 J1=IHHH(IH,1)	CHEM	168
	J2=IHHH(IH,2)	CHEM	169
	J3=IHHH(IH,3)	CHEM	170
	J4=IHHH(IH,4)	CHEM	171
	CH=ANF*OHMU(R)	CHEM	172
220	MP(IH)CCH=OF(I(J1))+F1(J2)	CHEM	173
	MM(IH)CCH=0	CHEM	174
	DU 770 J=1.4	CHEM	175
	SIGN=1.0	CHEM	176
	IF (J .GT. 2) SIGN=-1.0	CHEM	177
225	IMUP=IHHH(IH,J)	CHEM	178
	CM(IHUN,J1)ICM(IHUN,J1)+SIGN*CH=OF(I(J2))	CHEM	179
	CM(IHUN,J2)ICM(IHUN,J2)+SIGN*CH=OF(I(J1))	CHEM	180
	GR(IHUN)GR(IHUN)+SIGN*MP(IH)	CHEM	181
	770 CONTINUE	CHEM	182
	GO TO 807	CHEM	183
230	857 J1=IHHH(IH,1)	CHEM	184
	J2=IHHH(IH,2)	CHEM	185
	J3=IHHH(IH,3)	CHEM	186
	CH=ANF*OHMU(R)+AV/ZM	CHEM	187
235	MP(IH)CCH=OHMU(R)+F1(J1)+F1(J2)	CHEM	188
	MM(IH)CCH=0	CHEM	189
	DU 779 J=1.3	CHEM	190
	SIGN=1.0	CHEM	191
	IF (J .GT. 2) SIGN=-1.0	CHEM	192
240	IMUP=IHHH(IH,J)	CHEM	193
	CM(IHUN,J1)ICM(IHUN,J1)+SIGN*CH=OHMU(R)+F1(J2)	CHEM	194
	CM(IHUN,J2)ICM(IHUN,J2)+SIGN*CH=OHMU(R)+F1(J1)	CHEM	195
	GR(IHUN)GR(IHUN)+SIGN*MP(IH)	CHEM	196
	779 CONTINUE	CHEM	197
	GO TO 808	CHEM	198
245	850 J1=IHHH(IH,1)	CHEM	199
	J2=IHHH(IH,2)	CHEM	200
	J3=IHHH(IH,3)	CHEM	201
	J4=IHHH(IH,4)	CHEM	202
250	J5=IHHH(IH,5)	CHEM	203
	CH=ANF*OHMU(R)	CHEM	204
	MP(IH)CCH=OF(I(J1))+F1(J2)	CHEM	205
	MM(IH)CCH=0	CHEM	206
	DU 780 J=1.5	CHEM	207
255	SIGN=1.0	CHEM	208
	IF (J .GT. 2) SIGN=-1.0	CHEM	209
	IMUP=IHHH(IH,J)	CHEM	210
	CM(IHUN,J1)ICCP(IHUN,J1)+SIGN*CH=OF(I(J2))	CHEM	211
	CM(IHUN,J2)ICCP(IHUN,J2)+SIGN*CH=OF(I(J1))	CHEM	212
	GR(IHUN)GR(IHUN)+SIGN*MP(IH)	CHEM	213
	780 CONTINUE	CHEM	214
	GO TO 808	CHEM	215
260	850 J1=IHHH(IH,1)	CHEM	216
	J2=IHHH(IH,2)	CHEM	217
	J3=IHHH(IH,3)	CHEM	218
	CH=ANF*OHMU(R)	CHEM	219
	MP(IH)CCH=OHMU(R)+F1(J1)+F1(J2)	CHEM	220
	MM(IH)CCH=0	CHEM	221
	DU 776 J=1.3	CHEM	222
270	SIGN=1.0	CHEM	223
	IF (J .GT. 2) SIGN=-1.0	CHEM	224
	IMUP=IHHH(IH,J)	CHEM	225
	CM(IHUN,J1)ICM(IHUN,J1)+SIGN*CH=OHMU(R)+F1(J2)	CHEM	226
	CM(IHUN,J2)ICM(IHUN,J2)+SIGN*CH=OHMU(R)+F1(J1)	CHEM	227
	GR(IHUN)GR(IHUN)+SIGN*MP(IH)	CHEM	228
	776 CONTINUE	CHEM	229
	GO TO 808	CHEM	230
275	800 J1=IHHH(IH,1)	CHEM	231
	J2=IHHH(IH,2)	CHEM	232
	J3=IHHH(IH,3)	CHEM	233
	J4=IHHH(IH,4)	CHEM	234
	CH=ANF*OHMU(R)/ZM	CHEM	235
	MP(IH)CCH=OF(I(J1))	CHEM	236
	MM(IH)CCH=0	CHEM	237
280	DU 777 J=1.4	CHEM	238
	SIGN=1.0	CHEM	239
	IF (J .GT. 2) SIGN=-1.0	CHEM	240
	IMUP=IHHH(IH,J)	CHEM	241

```

      INUM = IARR(I,R,J)
      LC(I,INUM,J) = CM(I,INUM,J) * SIGN = CMN
299 777 CONTINUE
      GO TO 867
      861 NP(J5) = NP(J5) + NP(I)
      NM(J5) = NM(J5) + NM(I)
295 867 NP(J6) = NP(J6) + NP(I)
      NM(J6) = NM(J6) + NM(I)
      868 NP(J3) = NP(J3) + NP(I)
      NM(J3) = NM(J3) + NM(I)
      NP(J2) = NP(J2) + NP(I)
      NM(J2) = NM(J2) + NM(I)
300 NP(J1) = NP(J1) + NP(I)
      NM(J1) = NM(J1) + NM(I)
      1 CONTINUE
      DO 897 J=1,NUS
305 C PRODUCTION RATE IN MOLE/GM-CM
      ADDT(J,R) = (NP(J) - NM(J)) / U(R)
      897 CONTINUE
      USUM = DLB(R) / U(R)
      DO 10 I=1,NUS
310 USUM = USUM + PA(I)
      QX(I,R) = ((I,R) * HNS(I) / MDWT(R) + USUM * PA(I))
      L(I) = JN(I) * NUS
      CM(I,R,J) = CM(I,R,J) * USUM
      P(I,EU,J) = CM(I,R,J) * 1.0 * CM(I,R,J)
315 11 CONTINUE
      IF (IBUGM .NE. 0)
      WRITE(6,100) R,IN,(CM(I,R,J),JN(I),NUS),QX(I)
100 FORMAT(I2,2I5,1P8E12,3)
      10 CONTINUE
320 CALL SLOP(OX,LP,NDS)
      DO 12 I=1,NUS
      IF (IBUGM .NE. 0)
      WRITE(6,100) R,IN,(CM(I,R,J),JN(I),NUS),QX(I)
325 C CALCULATE NEW MASS FRACTION OF SPECIES
      12 CONTINUE
      RETURN
      END

```

```

CHEM 247
CHEM 248
CHEM 249
CHEM 250
CHEM 251
CHEM 252
CHEM 253
CHEM 254
CHEM 255
CHEM 256
CHEM 257
CHEM 258
CHEM 259
CHEM 260
CHEM 261
CHEM 262
CHEM 263
CHEM 264
CHEM 265
CHEM 266
CHEM 267
CHEM 268
CHEM 269
CHEM 270
CHEM 271
CHEM 272
CHEM 273
CHEM 274
CHEM 275
CHEM 276
CHEM 277
CHEM 278
CHEM 279
CHEM 280
CHEM 281

```

```

1 SUBROUTINE CPUTIN(I,IDENT,IKINE,NUS)
C -----
2 DIMENSION IDENT(25),IZU(5)
3 COMMON/CHEM1/ ZI(5), IRR(40), IRT(40), HCF(40,3), IHHH(40,5),
5 1 AV,CM(26,26),F1(26),NP(26),
6 2 AN(26),ADOT(26,40),LSI(26),OX(26)
7 COMMON/XYZ/XYZ1,XYZ2
8 XYZ=0
9 XYZ=0
10 AV=0,USE22
11 DO 2 I=1,IKINE
12 HEAD(5,100) = (IZU(J),J=1,5),IHHH(1),IRT(1),(NC(1,AN),K=1,3)
13 K=1(6,101) 1, (IZU(J),J=1,5),IHHH(1),IRT(1),(NC(1,AN),K=1,3)
14 DO 5 J=1,5
15 IHHH(1,J) = 0
16 DO 3 L=1,NDS
17 * (IZU(J) = IDENT(L)) 3,4,3
18 4 IHHH(1,J) = 0
19 3 CONTINUE
20 5 CONTINUE
21 HCL(1) = HCL(1) + AV
22 2 CONTINUE
23 100 FORMAT (A4,3I,4A,10X,4A,3X,4A,4 12,1I,4A,2F,4,1P,9,1)
24 101 FORMAT (2X,12,2I,4A,3X,4A,10X,4A,3I,4, 18,4A,9I,12,1I,4E,2,
25 1P,4,1P,4,1)
26 RETURN
27 END
28 CPUTIN 2
CPUTIN 3
CPUTIN 4
CPUTIN 5
CPUTIN 6
CPUTIN 7
CPUTIN 8
CPUTIN 9
CPUTIN 10
CPUTIN 11
CPUTIN 12
CPUTIN 13
CPUTIN 14
CPUTIN 15
CPUTIN 16
CPUTIN 17
CPUTIN 18
CPUTIN 19
CPUTIN 20
CPUTIN 21
CPUTIN 22
CPUTIN 23
CPUTIN 24
CPUTIN 25
CPUTIN 26
CPUTIN 27
CPUTIN 28

```

```

1 SUBROUTINE UNAG
2 COMMON A(25,7)
3 1 A10L ABA(40) ALFA(25,2),ALPHAH ALPHAY A
4 2 A12(25) AC(25,40) A
5 3 A18(25) ACBAH(25) CBAR(25) CP A
6 4 A19(25) ACPM ACSI(25) CSTREM(25) A
7 5 A21 ADELSS(40) DELS DELSU DLS(40) A
8 6 A22(25,25) ADEL ADEL(40) A
9 7 A24 APMI(50) EPCUN A
10 8 A25 AEXTNA(50) FSTEP FMAX GNAO A
11 9 A26 APM APM APM A
12 10 A27 A2PM(25) A2PM(25) ACONST ICOUNT IOENT(25) A
13 11 A28 AEXTNA(50) IFLAG IIND A
14 12 A29 AISMUCK IITPI IPU A
15 13 A30 A A AAYS KATZ A
16 14 A31 A A A A A
17 15 A32 A A A A A
18 16 A33 A A A A A
19 20 A34 A A A A A
21 1 A35 A A A A A
22 2 A36 A A A A A
23 3 A37 A A A A A
24 4 A38 A A A A A
25 5 A39 A A A A A
26 6 A40 A A A A A
27 7 A41 A A A A A
28 UNAG 2
UNAG 3
UNAG 4
UNAG 5
UNAG 6
UNAG 7
UNAG 8
UNAG 9
UNAG 10
UNAG 11
UNAG 12
UNAG 13
UNAG 14
UNAG 15
UNAG 16
UNAG 17
UNAG 18
UNAG 19
UNAG 20
UNAG 21
UNAG 22
UNAG 23
UNAG 24
UNAG 25
UNAG 26
UNAG 27
UNAG 28

```

```

      INUM = INRR(IR,J)
      CP(IRON,J) = CP(INUM,J) * SIGN = CMN
290 777 CONTINUE
      GO TO 807
      801 NP(J) = NP(J) * NP(IR)
      NM(J) = NM(J) * NM(IR)
295 807 NP(J) = NP(J) * NP(IR)
      NM(J) = NM(J) * NM(IR)
      800 NP(J) = NP(J) * NP(IR)
      NM(J) = NM(J) * NM(IR)
300 NP(J) = NP(J) * NP(IR)
      NM(J) = NM(J) * NM(IR)
      PP(J) = PP(J) * PP(IR)
      AP(J) = AP(J) * AP(IR)
      1 CONTINUE
      DU 897 J01, NUS
305 C PRODUCTION RATE IN MOLE/GM-CM
      PDOT(J,R) = P(J) * NM(J) / U(R)
      897 CONTINUE
      USUR = DLS(K) / L(K)
310 DU 10 INR1, NUS
      USUR = USUR * A(K)
      QX(IN) = ((IN, R) * NUS(IN) / MUOT(K) * USUR * A(K) * IN)
      L1 11 JNR1, NUS
      CM(IN, JR) = CM(IN, JN) * USUR
      P(IN, R, JN) = P(IN, JN) * 1.0 * CM(IN, JN)
315 11 CONTINUE
      IF (IBUGSH .NE. 0)
      *WRITE(6,100) K, IN, (CM(IN, JN), JN=1, NDB), QX(IN)
100 FORMAT(1X, 2I5, 1PBE12.3)
      10 CONTINUE
      CALL SLOW(QX, LP, NDS)
320 DU 12 INR1, NUS
      IF (IBUGSH .NE. 0)
      *WRITE(6,100) K, IN, (CM(IN, JN), JN=1, NDB), QX(IN)
      C2(IN, R) = QX(IN)
325 C CALCULATE NEW MASS FRACTION OF SPECIES
      12 CONTINUE
      RETURN
      END

```

```

SUBROUTINE CPUTIN(JUENT, JLINE, NUS)
.....
DIMENSION JUENT(25), IZU(5)
COMMON/CHEN1/ ZI(5), INR(40), INT(40), MC(40,3), INRR(40,3),
5 1 AV, CM(20,20), F1(20), NP(20),
  2 AN(20), ADOT(20,40), CS(20), QX(20)
  COMMON/XYZ/ IXYZ, JXYZ
  IXYZ=0
  JXYZ=0
10 AV=0.3E25
  DU 2 INR1, INR4
  HEAD(5,100) = (IYZ(J), J=1,5), INR(1), INT(1), MC(1,1), KRB(1,3)
  KRT(6,101) = (IZU(J), J=1,5), INR(1), INT(1), MC(1,1), AN(1,3)
  DU 5 J=1,5
15 INR(1, J) = 0
  DU 3 L=1, NDB
  * IZU(J) = IDENT(L) * 3.4, 3
  4 INR(1, J) = 0
  3 CONTINUE
20 5 CONTINUE
  MC(1,1) = MC(1,1) * AV
  2 CONTINUE
100 FORMAT(1X, 3I5, A0, 10X, A0, 3X, A0, 3X, A0, 9 12, 11, 10, 2, F4, 1, F, 0, 1)
25 101 FORMAT(2X, 12, 2V, A0, 3X, A0, 10X, A0, 3X, 4, 3, A0, 9X, 12, 11, 10, 2,
  1 F4, 1, F, 0, 1)
  RETURN
  END

```

```

SUBROUTINE UNAG
COMMON A(25,7)
1 1 A1L1L A1L1P A1L1A A1L1C(25,40) A1L1P
5 2 C12(25) C2(25,40) CABAN(25) CUBAR(25) CP
  3 CPS(25) CPM CSM(25) CBSH(25) CSTREH(25) A
  4 MSTRM PPIH(25,25), DIFP(25) D2EFF(25) OFLIA A
  5 D11 D1L5(40) D1L5 D1L5D D1L5(40) A
  6 D11(25,25) D12 D1L5(40) D15 D1UDY(40) A
  7 D1W DPH(10) DPGU A
10 8 DFLON DETHA(50) DSTLP DPAK DSHAD A
  9 DMI D(40) DM DMJ DMRH(25) A
  4 DMPH(25) MSPH(25) ILONSI ICOUNT IOENT(25) A
  1 DETHA(50) DFLAG DIND A
  2 D1L1L D1SMCK D1YPI D1U A
15 3 D1YF D1Y D1YAYS D1YAYS D1Y2 A
  4 D1U D1PAX A
  5 D1U D1L D1PLANE D1MA A
  6 D1ASH D1U1 D1PAX D1U D1MASH A
  7 D12 D1US D1M D12 D1MASH A
20 8 D1UVAL D1NS D1N1EN D1MAX D1N A
  1 DUMUN NUCASE D1MEGA(25) D1I D1(40) D12 A
  2 D1S D1P15 D1P1(40) D1P1(40) D1P1(50) A
  3 D1M1S1 D1P1S1 D1P1N(50) D1P1R A
  4 D1M1C(40) D1M1S D1P1 D1P1(25) D1PSH A
  5 D1S1 D1S1 D1S1KEM D1P1(50) A
  6 D1I D1(40) D1XTM1 D1XTM2 A
  7 D1(50) D1I D1N(40) D1R(50) A

```



```

30 8 .RHS .R13 .RMAN(40) .R10 .R2(40) A 29
9 .RCU4 .R15 .RE(40) .RESM .R16 A 30
0 .RMU(40) .R17 .RMU2(40) .RMS(25) .RMS4 A 31
1 .RMSUM .RMANAR .RMSMAN .RU .RSM A 32
2 .RSTPEM .RY .R19 .RM(100) .S A 33
3 .S .S(50) .SC(25) .S(50) A 34
35 4 .S15 .S(40) .T11 .T(40) .T12 A 35
5 .T2(40) .TARAN .TSMAN .T0(25) .T14 A 36
6 .TAP(40) .TATH1 .TATH2 .TUL .TSM A 37
7 .TSM1 .TSTHEM .T(50) .TAS A 38
8 .T3 .T11 .T(40) .T12 .T2(40) A 39
40 COMMON UTHM1 .UTHM2 .U14 .UMAN(40) .UM8 A 40
1 .USH1 .USTHEM .U12 .UM(50) .UM8 A 41
2 .U11 .U(40) .U12 .U2(40) A 42
3 .UM(50) .UM5 .RMAN(25) .RMAN(25) .U15 A 43
4 .US(25,40) .USH .USTHEM A 44
45 5 .ZMM(40) .T11 .T(40) .T12 A 45
6 .Z2(40) .YAMAN .YMAN .ZA(25) .Z11(25) A 46
7 .ZJ(25,40) .ZMM .ZXX .ZSM A 47
8 .ZP(2,40) .ZP(2,40) .ZPM1(2,40) .ZP(2,40) .ZT(2,40) A 48
9 .ZU(2,40) .ZADL(2,40) .ZADR(2,40) .ZL(2,40,25) .ZARD1 A 49
50 0 .Z A 50
REAL RAY .RAY3(25) .RAY2 .RN .RNU1(40) DRAG 4
1 .RA(40) .RU .RUS(25) .RUS(25) .RU2 DRAG 5
2 .RA(25) .RHZ .RASH .RASH1 .RASH DRAG 6
3 .RAB DRAG 7
55 COMMON P11,ZAMM(40),ZMM(40),ZEMPI(40),ZEP(40,40),V(40,40),n(40,40),
:V2(40,40),A2(40,40),IMRDY(40),TP(40,40),TP2(40,40),MP(40),
ZMP(40,40),NMP2(40,40),ZEMP2(40),NOL(40),UM(40) UNAG 8
UNAG 9
UNAG 10

3,VULP(40),AIP(40),AC(40),UENG(40,40),ICUM(40,40) UNAG 11
COMMON P12,ZF1,ZF2,ZAMA,UM1,PU,NM,MM,DM,DM(40) UNAG 12
COMMON P13,LL,CS,TPS,NMS,OT,MINAN,BIG,EP,IPANT,IRINE,TH(40) UNAG 13
1,SUMP,SUMPV,SUMPE UNAG 14
COMMON/EP/PIU UNAG 15
COMMON/RYZ/IAVZ,JAVZ UNAG 16
IAVZ5 UNAG 17
JAVZ6 UNAG 18
AZZ4H UNAG 19
NMR=1 UNAG 20
GAMP1,GAMA+1, UNAG 21
SOGAPSS1,*(GAMA) UNAG 22
SUTP1=1.774 UNAG 23
DU 1 J=1, N4 UNAG 24
IF (N4.GT. NOL(J)) GO TO 6 UNAG 25
2 CONTINUE UNAG 26
ARX18 ((U(N)-*(N,J))**2 +V(N,J)**2) UNAG 27
ARX1 SUM(ARX1) UNAG 28
UM(N)=0.0 UNAG 29
U(1)=1.0,US UNAG 30
7 UM(N)=MIN(1.0,5*(1.0+UMU(1)+T(N)/10(1))**UMTGA(1)) UNAG 31
C (CALCULATE DRAIN COEFFICIENT) UNAG 32
HEP(P,J)=MIN(1.0,0.5*ANP(J)*2.0/UM(N)) UNAG 33
NAMP NA(N)=0.5*H(N) UNAG 34
NMPAP/HEP(N,J) UNAG 35
LUM124,HEP(N,J) UNAG 36
IF (HEP(N,J).GT.0.0) CUURU,40 UNAG 37
TANG62,ALUG(NAP) UNAG 38
TANT6(EXP(2.*TANG)-1.)/(EXP(2.*TANG)+1.) UNAG 39
LUM1,60,20*TANT6,17*EXP(-2.5*(ALUG(NAP)/1.4)**2) UNAG 40
IF (NAP.LT.1.3) (I=,40 UNAG 41
LUM1,0 UNAG 42
CUR(1) UNAG 43
ARX1,26 0 SOGAPSS1 UNAG 44
IF (ARX1.GT.4.) GO TO 4 UNAG 45
IF (ARX1.LT. 1.4) GO TO 5 UNAG 46
ARX1,ARX1,4 UNAG 47
ARX1,ARX1,4 UNAG 48
GAMP1,ARX1/(1.0+ARX1**2) UNAG 49
ARX1,ARX1,4 UNAG 50
ARX1,ARX1,4 UNAG 51
UM(1)=0.5*EXP(-7.5*ARX1*(CUURU,40)+HEP(N,J)/4.) UNAG 52
C (DRAINAGE) UNAG 53
4 CONTINUE UNAG 54
S18SUNT(,SOGAPSS1)+MAP UNAG 55
S28S1S1 UNAG 56
S38S2S1 UNAG 57
S48S3S1 UNAG 58
S58S4S1 UNAG 59
S68S5SUM1(1) (N) /IP(N,J) UNAG 60
LUMPE EXP(-532)*((1.0+2.0*S2)/S1)P1/53 UNAG 61
1 *(4.0*(S4+S2)-1.0)/2.0/SRATRF(S1) UNAG 62
2 *(0.67*SUIP1/50 UNAG 63
C (DRAINAGE) UNAG 64
5 P18(1/24,40) (N,J)P18 UNAG 65
C (CALCULATE RUSSELL NUMBER) UNAG 66
ARX1 2.0*0.459*(HEP(N,J) **0.55)*(PU**0.33) UNAG 67

115 P18 5.0*(GAPAAA1.5)+MMH/GAMP1/PU*RU UNAG 68
RUM(1)=RU/(1.0+P18) UNAG 69
RUPR RUM*(GAMP1/GAPAAKEH(N,J)+PUMER*(0.5/MMH)) UNAG 70
LUM 0.5*RU*(P18) UNAG 71
RZ 0.45 0 UM(N) 0 RU / (NMS+MP(J)+NMP(J)) UNAG 72
LUM1(J) 0 RZ*(N-1,J) UNAG 73
RUM1(J) 0 RZ*(N-1,J) UNAG 74
HEP1(J) 0 RZ*(N-1,J) UNAG 75
HEP2(J) 0 RZ*(N-1,J) UNAG 76
HEP3(J) 0 RZ*(N-1,J) UNAG 77
HEP4(J) 0 RZ*(N-1,J) UNAG 77

```

```

85      DU 20 [1],NBS          FLUX 24
      20 ZJ(1,1)NS,0          FLUX 25
      CO 10 10000            FLUX 26
      C ----- SETUP OF AVERAGE FLUX QUANTITIES NEEDED ----- FLUX 27
      30 T[MAKSTAB]          FLUX 28
      T[MAKSTAB]          FLUX 29
      T[MAKSTAB]          FLUX 30
      T[MAKSTAB]          FLUX 31
      DU 50 [1],NBS          FLUX 32
      L[AN]([1],C[AN]([1])   FLUX 33
      50 AN[1]([1],C[AN]([1]) FLUX 34
      T[MAKSTAB]          FLUX 35
      AN[1]([1],C[AN]([1]) FLUX 36
      IF ([1],C[AN]([1]) * 100, AN[1]([1],C[AN]([1]) FLUX 37
      100 T[MAKSTAB,50]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 38
      T[MAKSTAB,50]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 39
      C ----- FLUX QUANTITIES ----- FLUX 40
      U[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 41
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 42
      110 AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 43
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 44
      130 T[MAKSTAB]          FLUX 45
      DU 200 [1],NBS         FLUX 46
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 47
      200 T[MAKSTAB]          FLUX 48
      T[MAKSTAB]          FLUX 49
      DU 300 [1],NBS         FLUX 50
      300 L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 51
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 52
      CO 10 10000            FLUX 53
      400 T[MAKSTAB]          FLUX 54
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 55
      U[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 56
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 57
      120 AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 58
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 59
      DU 500 [1],NBS         FLUX 60
      500 L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 61
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 62
      T[MAKSTAB]          FLUX 63
      600 CALL TRANS[AN]([1],AN[AN]([1],C[AN]([1]),P[UL],1016) FLUX 64
      C ----- CALCULATION OF MEAT FLUX AND SHEAR ----- FLUX 65
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 66
      T[MAKSTAB]          FLUX 67
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 68
      T[MAKSTAB]          FLUX 69
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 70
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 71
      C ----- CALCULATION OF MASS FLUX ----- FLUX 72
      DU 1000 [1],NBS        FLUX 73
      L[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 74
      125 AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 75
      1000 ZJ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 76
      IF ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 77
      IF ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 78
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 79
      130 L[AN]          FLUX 80
      DISTANCE[AN]          FLUX 81
      IF ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 82
      140 L[AN]          FLUX 83
      AN[AN]([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 84
      L[AN]          FLUX 85
      IF ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 86
      1000 ZJ([1],C[AN]([1]) * 5, AN[1]([1],C[AN]([1]) * 2, AN[1]([1],C[AN]([1]) FLUX 87
      1000 N[AN]          FLUX 88
      END          FLUX 89
      I      SUBROUTINE LOCATE (AM, TO, AT, T1, X, Y, Z, NS, IS, IF, G)          LOCATE 2
      RETURN          LOCATE 3
      C          LOCATE          LOCATE 4
      I      SUBROUTINE MATZ (M, N, A, B, C, D, E, F, G, H, I, J, K, L, M, N, O, P, Q, R, S, T, U, V, W, X, Y, Z, AA, AB, AC, AD, AE, AF, AG, AH, AI, AJ, AK, AL, AM, AN, AO, AP, AQ, AR, AS, AT, AU, AV, AW, AX, AY, AZ, BA, BB, BC, BD, BE, BF, BG, BH, BI, BJ, BK, BL, BM, BN, BO, BP, BQ, BR, BS, BT, BU, BV, BW, BX, BY, BZ, CA, CB, CC, CD, CE, CF, CG, CH, CI, CJ, CK, CL, CM, CN, CO, CP, CQ, CR, CS, CT, CU, CV, CW, CX, CY, CZ, DA, DB, DC, DD, DE, DF, DG, DH, DI, DJ, DK, DL, DM, DN, DO, DP, DQ, DR, DS, DT, DU, DV, DW, DX, DY, DZ, EA, EB, EC, ED, EE, EF, EG, EH, EI, EJ, EK, EL, EM, EN, EO, EP, EQ, ER, ES, ET, EU, EV, EW, EX, EY, EZ, FA, FB, FC, FD, FE, FF, FG, FH, FI, FJ, FK, FL, FM, FN, FO, FP, FQ, FR, FS, FT, FU, FV, FW, FX, FY, FZ, GA, GB, GC, GD, GE, GF, GG, GH, GI, GJ, GK, GL, GM, GN, GO, GP, GQ, GR, GS, GT, GU, GV, GW, GX, GY, GZ, HA, HB, HC, HD, HE, HF, HG, HH, HI, HJ, HK, HL, HM, HN, HO, HP, HQ, HR, HS, HT, HU, HV, HW, HX, HY, HZ, IA, IB, IC, ID, IE, IF, IG, IH, II, IJ, IK, IL, IM, IN, IO, IP, IQ, IR, IS, IT, IU, IV, IW, IX, IY, IZ, JA, JB, JC, JD, JE, JF, JG, JH, JI, JJ, JK, JL, JM, JN, JO, JP, JQ, JR, JS, JT, JU, JV, JW, JX, JY, JZ, KA, KB, KC, KD, KE, KF, KG, KH, KI, KJ, KK, KL, KM, KN, KO, KP, KQ, KR, KS, KT, KU, KV, KW, KX, KY, KZ, LA, LB, LC, LD, LE, LF, LG, LH, LI, LJ, LK, LL, LM, LN, LO, LP, LQ, LR, LS, LT, LU, LV, LW, LX, LY, LZ, MA, MB, MC, MD, ME, MF, MG, MH, MI, MJ, MK, ML, MM, MN, MO, MP, MQ, MR, MS, MT, MU, MV, MW, MX, MY, MZ, NA, NB, NC, ND, NE, NF, NG, NH, NI, NJ, NK, NL, NM, NN, NO, NP, NQ, NR, NS, NT, NU, NV, NW, NX, NY, NZ, OA, OB, OC, OD, OE, OF, OG, OH, OI, OJ, OK, OL, OM, ON, OO, OP, OQ, OR, OS, OT, OU, OV, OW, OX, OY, OZ, PA, PB, PC, PD, PE, PF, PG, PH, PI, PJ, PK, PL, PM, PN, PO, PP, PQ, PR, PS, PT, PU, PV, PW, PX, PY, PZ, QA, QB, QC, QD, QE, QF, QG, QH, QI, QJ, QK, QL, QM, QN, QO, QP, QQ, QR, QS, QT, QU, QV, QW, QX, QY, QZ, RA, RB, RC, RD, RE, RF, RG, RH, RI, RJ, RK, RL, RM, RN, RO, RP, RQ, RR, RS, RT, RU, RV, RW, RX, RY, RZ, SA, SB, SC, SD, SE, SF, SG, SH, SI, SJ, SK, SL, SM, SN, SO, SP, SQ, SR, SS, ST, SU, SV, SW, SX, SY, SZ, TA, TB, TC, TD, TE, TF, TG, TH, TI, TJ, TK, TL, TM, TN, TO, TP, TQ, TR, TS, TT, TU, TV, TW, TX, TY, TZ, UA, UB, UC, UD, UE, UF, UG, UH, UI, UJ, UK, UL, UM, UN, UO, UP, UQ, UR, US, UT, UY, UV, UW, UX, UY, UZ, VA, VB, VC, VD, VE, VF, VG, VH, VI, VJ, VK, VL, VM, VN, VO, VP, VQ, VR, VS, VT, VY, VZ, WA, WB, WC, WD, WE, WF, WG, WH, WI, WJ, WK, WL, WM, WN, WO, WP, WQ, WR, WS, WT, WY, WZ, XA, XB, XC, XD, XE, XF, XG, XH, XI, XJ, XK, XL, XM, XN, XO, XP, XQ, XR, XS, XT, XU, XV, XW, XX, XY, XZ, YA, YB, YC, YD, YE, YF, YG, YH, YI, YJ, YK, YL, YM, YN, YO, YP, YQ, YR, YS, YT, YZ, ZA, ZB, ZC, ZD, ZE, ZF, ZG, ZH, ZI, ZJ, ZK, ZL, ZM, ZN, ZO, ZP, ZQ, ZR, ZS, ZT, ZY, ZV, ZW, ZX, ZY, ZZ, AA, AB, AC, AD, AE, AF, AG, AH, AI, AJ, AK, AL, AM, AN, AO, AP, AQ, AR, AS, AT, AU, AV, AW, AX, AY, AZ, BA, BB, BC, BD, BE, BF, BG, BH, BI, BJ, BK, BL, BM, BN, BO, BP, BQ, BR, BS, BT, BU, BV, BW, BX, BY, BZ, CA, CB, CC, CD, CE, CF, CG, CH, CI, CJ, CK, CL, CM, CN, CO, CP, CQ, CR, CS, CT, CU, CV, CW, CX, CY, CZ, DA, DB, DC, DD, DE, DF, DG, DH, DI, DJ, DK, DL, DM, DN, DO, DP, DQ, DR, DS, DT, DU, DV, DW, DX, DY, DZ, EA, EB, EC, ED, EE, EF, EG, EH, EI, EJ, EK, EL, EM, EN, EO, EP, EQ, ER, ES, ET, EU, EV, EW, EX, EY, EZ, FA, FB, FC, FD, FE, FF, FG, FH, FI, FJ, FK, FL, FM, FN, FO, FP, FQ, FR, FS, FT, FU, FV, FW, FX, FY, FZ, GA, GB, GC, GD, GE, GF, GG, GH, GI, GJ, GK, GL, GM, GN, GO, GP, GQ, GR, GS, GT, GU, GV, GW, GX, GY, GZ, HA, HB, HC, HD, HE, HF, HG, HH, HI, HJ, HK, HL, HM, HN, HO, HP, HQ, HR, HS, HT, HU, HV, HW, HX, HY, HZ, IA, IB, IC, ID, IE, IF, IG, IH, II, IJ, IK, IL, IM, IN, IO, IP, IQ, IR, IS, IT, IU, IV, IW, IX, IY, IZ, JA, JB, JC, JD, JE, JF, JG, JH, JI, JJ, JK, JL, JM, JN, JO, JP, JQ, JR, JS, JT, JY, JZ, KA, KB, KC, KD, KE, KF, KG, KH, KI, KJ, KK, KL, KM, KN, KO, KP, KQ, KR, KS, KT, KU, KV, KW, KX, KY, KZ, LA, LB, LC, LD, LE, LF, LG, LH, LI, LJ, LK, LL, LM, LN, LO, LP, LQ, LR, LS, LT, LU, LV, LW, LX, LY, LZ, MA, MB, MC, MD, ME, MF, MG, MH, MI, MJ, MK, ML, MM, MN, MO, MP, MQ, MR, MS, MT, MU, MV, MW, MX, MY, MZ, NA, NB, NC, ND, NE, NF, NG, NH, NI, NJ, NK, NL, NM, NN, NO, NP, NQ, NR, NS, NT, NU, NV, NW, NX, NY, NZ, OA, OB, OC, OD, OE, OF, OG, OH, OI, OJ, OK, OL, OM, ON, OO, OP, OQ, OR, OS, OT, OU, OV, OW, OX, OY, OZ, PA, PB, PC, PD, PE, PF, PG, PH, PI, PJ, PK, PL, PM, PN, PO, PP, PQ, PR, PS, PT, PU, PV, PW, PX, PY, PZ, QA, QB, QC, QD, QE, QF, QG, QH, QI, QJ, QK, QL, QM, QN, QO, QP, QQ, QR, QS, QT, QU, QV, QW, QX, QY, QZ, RA, RB, RC, RD, RE, RF, RG, RH, RI, RJ, RK, RL, RM, RN, RO, RP, RQ, RR, RS, RT, RU, RV, RW, RX, RY, RZ, SA, SB, SC, SD, SE, SF, SG, SH, SI, SJ, SK, SL, SM, SN, SO, SP, SQ, SR, SS, ST, SU, SV, SW, SX, SY, SZ, TA, TB, TC, TD, TE, TF, TG, TH, TI, TJ, TK, TL, TM, TN, TO, TP, TQ, TR, TS, TT, TU, TV, TW, TX, TY, TZ, UA, UB, UC, UD, UE, UF, UG, UH, UI, UJ, UK, UL, UM, UN, UO, UP, UQ, UR, US, UT, UV, UW, UX, UY, UZ, VA, VB, VC, VD, VE, VF, VG, VH, VI, VJ, VK, VL, VM, VN, VO, VP, VQ, VR, VS, VT, VY, VZ, WA, WB, WC, WD, WE, WF, WG, WH, WI, WJ, WK, WL, WM, WN, WO, WP, WQ, WR, WS, WT, WY, WZ, XA, XB, XC, XD, XE, XF, XG, XH, XI, XJ, XK, XL, XM, XN, XO, XP, XQ, XR, XS, XT, XU, XV, XW, XX, XY, XZ, YA, YB, YC, YD, YE, YF, YG, YH, YI, YJ, YK, YL, YM, YN, YO, YP, YQ, YR, YS, YT, YZ, ZA, ZB, ZC, ZD, ZE, ZF, ZG, ZH, ZI, ZJ, ZK, ZL, ZM, ZN, ZO, ZP, ZQ, ZR, ZS, ZT, ZY, ZV, ZW, ZX, ZY, ZZ

```

```

25 C      IND      MUST BE A FIXED OR FLOATING POINT VARIABLE DESIGNATING
C      A ONE-DIMENSIONAL ERASABLE ARRAY OF LENGTH AT LEAST
C      EQUAL TO THE NUMBER OF ROWS IN MATRIX A. IT IS IN THIS
C      AREA SIMG KEEPS A RECORD OF THE COLUMN PERMUTATIONS.
C      NEG1      IS A FIXED POINT VARIABLE WHICH WILL BE ASSIGNED THE
30 C      PIPED POINT CONSTANTS.
C      1 IF THE SOLUTION WAS SUCCESSFUL.
C      3 IF THE MATRIX A WAS SINGULAR.
C      DIMENSIONS
C      A      IJIM = IJIM IF MM IS LESS THAN OR EQUAL TO IJIM
C      IJIP = MM IF MM IS GREATER THAN IJIM
35 C      P      IJIP = MM
C      IND      IJIM
C      NOTE
C      FOR MATRICES THE MM DIMENSION MUST BE THE SAME AS
C      THAT GIVEN ABOVE. THE COLUMN DIMENSION MUST AT LEAST
40 C      BE THAT GIVEN ABOVE.
C      NROW
C      FOR THE VECTOR THE DIMENSION MUST AT LEAST BE THAT
C      GIVEN ABOVE
C      EXECUTION OF THIS ROUTINE DESTROYS THE ORIGINAL A AND
45 C      N MATRICES
C      AFTER A SUCCESSFUL EXIT FROM THIS SUBROUTINE, THE
C      ANSWERS ON THE A MATRIX REPLACE THE A MATRIX.
C      THIS REPLACEMENT IS DONE ACCORDING TO THE SCHEME.
C      A(I,J) IS REPLACED BY R(I,J) .
50 C      ROWS
C      SET INITIAL CONSTANTS
C      NUG(1)
C      SPECIAL CONSIDERATION WHEN THE ORDER OF MATRIX A IS 1.
55 C      IF (N,NE,1) GO TO 3
C      IF (A(1,1),EQ,0.) GO TO 110
C      MUL(1,1)
C      DU 1 I=1,N
C      A(I,1)R(I,1)/MUL
C      DETERMINUL
60 C      N(I,1)
C      S A(I,1)=1
C      A(I,1)=1
C      INITIALIZE DETERMINANT
C      DETERMUL
65 C      INITIALIZE COLUMN INDICATORS
C      DU 5 I=1,N
C      > IND(I,1)
C      P(I,1) IN TRIANGULARIZATION TO GET UPPER TRIANGLE
C      DU 70 I=1,N-1
C      R(I,1)=1
C      A(I,1)
C      P(I,1)
70 C      SEARCH FOR PIVOTAL ELEMENT
C      NIGARANS(I,P)
C      DU 10 I=1,N
C      DU 10 J=1,N
C      IF (NIGARANS(I,P),ABS(A(I,J))) GO TO 110
C      NIGARANS(I,J)
80 C      R(I,J)
C      R(I,J)
85 C      10 CONTINUE
C      TEST FOR SINGULAR MATRIX
C      IF (NIGARANS(I,P),EQ,0.) GO TO 110
C      UPDATE DETERMINANT
85 C      UPDATE INVERSE
C      INTERCHANGE ROWS
C      IF (NIGARANS(I,P),GT,0) GO TO 30
C      DU 20 I=1,N
C      MUL(1,1)
90 C      A(I,1)R(I,1)
C      20 A(I,1)R(I,1)
C      INTERCHANGE ELEMENTS OF RIGHT HAND SIDES
C      DU 25 I=1,N
C      MUL(1,1)
95 C      D(I,1)R(I,1)
C      25 A(I,1)R(I,1)
C      CHANGE SIGN OF DETERMINANT DUE TO ROW INTERCHANGE
C      DETERMUL
C      INTERCHANGE COLUMNS
100 C      30 INTERCHANGE COLUMNS
C      DU 40 I=1,N
C      MUL(1,1)
C      D(I,1)R(I,1)
105 C      40 A(I,1)R(I,1)
C      INTERCHANGE COLUMN INDICATORS
C      I(I,1)
C      IND(I,1)
C      IND(I,1)
110 C      CHANGE SIGN OF DETERMINANT DUE TO COLUMN INTERCHANGE
C      DETERMUL
C      DIVIDE REDUCED EQUATION-ROW BY LEADING ELEMENT
115 C      55 DU 10 I=1,N
C      60 A(I,1)R(I,1)/A(I,1)
C      115 A(I,1)
120 C      65 A(I,1)R(I,1)/A(I,1)
C      120 A(I,1)R(I,1)
C      125 A(I,1)
C      130 A(I,1)

```

```

25 C     IND     MUST BE A FIXED OR FLOATING POINT VARIABLE DESIGNATING
C     A ONE-DIMENSIONAL ERASABLE ARRAY OF LENGTH AT LEAST
C     EQUAL TO THE NUMBER OF ROWS IN MATRIX A. IT IS IN THIS
C     AREA SIBS NEEDS A RECORD OF THE COLUMN PERMUTATIONS.
C     ADOG     IS A FIXED POINT VARIABLE WHICH WILL BE ASSIGNED THE
C     FIXED POINT CONSTANTS.
30 C     1- IF THE SOLUTION WAS SUCCESSFUL.
C     2- IF THE MATRIX A WAS SINGULAR.
C     DIMENSIONS
C     A     IDIM * IDIM IF NM IS LESS THAN OR EQUAL TO IDIM
C     IDIM * NM IF NM IS GREATER THAN IDIM
35 C     B     IDIM * NM
C     IND     IDIM
C     NOTE
C     FOR MATRICES THE ROW DIMENSION MUST BE THE SAME AS
C     THAT GIVEN ABOVE. THE COLUMN DIMENSION MUST AT LEAST
40 C     BE THAT GIVEN ABOVE.
C     NROW     FOR THE VECTOR THE DIMENSION MUST AT LEAST BE THAT
C     GIVEN ABOVE.
C     EXECUTION OF THIS ROUTINE DESTROYS THE ORIGINAL A AND
45 C     B MATRICES.
C     AFTER A SUCCESSFUL EXIT FROM THIS SUBROUTINE, THE
C     ANSWERS ON THE Y MATRIX REPLACE THE A MATRIX.
C     THIS REPLACEMENT IS DONE ACCORDING TO THE SCHEME:
C     A(I,J) IS REPLACED BY X(I,J).
50 C     NMM     SET INITIAL CONSTANTS
C     NODUM     SPECIAL CONSIDERATION WHEN THE ORDER OF MATRIX A IS 1.
C     IF (N.EQ.1) GO TO 3
55 C     IF (A(1,1).EQ.0.) GO TO 110
C     MOLDW(1,1)
C     DO 1 I=1,N
1 A(I,1)=A(1,1)/MOLD
C     DET=DETMOLD
C     NETUM=N
60 C     5 NMIN=1
C     NP(1)=1
C     INITIALIZE DETERMINANT
C     DET=DETM
65 C     INITIALIZE COLUMN INDICATORS
C     DO 5 I=1,N
C     5 IND(I)=1
C     BEGIN TRIANGULARIZATION TO GET UPPER TRIANGLE
70 C     DO 70 I=1,N-1
C     NP(I)=I
C     NMIN=N
C     PC=N
C     SEARCH FOR PIVOTAL ELEMENT
75 C     NIGABABS(A(I,N))
C     DO 10 I=N,N
C     DO 10 J=N,N
C     IF (NIGA.GE.ABS(A(I,J))) GO TO 10
C     NIGA=ABS(A(I,J))
C     NMI=I
C     NCI=J
80 C     10 CONTINUE
C     TEST FOR SINGULAR MATRIX
C     IF (NIGA.EQ.0.) GO TO 110
C     UPDATE DETERMINANT
85 C     DET=DETM*(A(N,N)-N)
C     INTERCHANGE NIMS
C     IF (N.NE.N) GO TO 30
C     DO 20 I=N,N
C     MOLDW(A(I,1))
90 C     A(I,1)=A(N,1)
C     A(N,1)=MOLDW
C     20 INTERCHANGE ELEMENTS OF RIGHT HAND SIDES
C     DO 25 I=N,N
C     MOLDW(A(I,1))
95 C     A(I,1)=A(N,1)
C     A(N,1)=MOLDW
C     CHANGE SIGN OF DETERMINANT DUE TO ROW INTERCHANGE
C     DET=-DET
C     INTERCHANGE COLUMNS
100 C     30 IF (N.EQ.1) GO TO 35
C     DO 40 J=1,N
C     MOLDW(A(I,J))
C     A(I,J)=A(N,J)
C     A(N,J)=MOLDW
105 C     40 INTERCHANGE COLUMN INDICATORS
C     I=IND(I)
C     J=IND(N)
C     IND(I)=J
C     IND(N)=I
C     CHANGE SIGN OF DETERMINANT DUE TO COLUMN INTERCHANGE
110 C     DET=-DET
C     DIVIDE REDUCED EQUATION-N, BY LEADING ELEMENT
C     DO 55 I=1,N-1
C     DO 55 K=1,N-1
C     A(I,K)=A(I,K)/A(N,N)
115 C     55 A(I,K)=A(I,K)/A(N,N)
C     55 A(I,K)=A(I,K)/A(N,N)
120 C     65 A(I,K)=A(I,K)/A(N,N)
C     REDUCE MATRIX AND RIGHT HAND SIDES
C     DO 70 I=N,N
C     65 A(I,K)=A(I,K)/A(N,N)

```

```

120      65-ALL(JJRA(L),J)A(1,K)*KIN(J)
      DO 70 L=1,M
125      70-ALL(CIDR(L),L)*A(1,K)*KIN(L)
      C      FINAL TEST FOR SINGULAR MATRIX
      IF (A(1,M)) GO TO 110
      COMPUTE SINGULAR DETERMINANT
130      DELTA=ABS(D)
      C      CHECK SUBSTITUE TO OBTAIN SOLUTION VECTORS
      DO 80 L=1,M
      W(L)=ABS(D)/A(L,M)
      DO 90 J=1,N
135      80-ALL(W(J),J)*A(L,K)
      90-ALL(W(J),J)*A(L,K)
      MULDEL
      J=J+1
      GO TO 80
      75-ALL(W(L),L)*A(J,K)
      85-ALL(W(L),L)*A(J,K)
      C      REARRANGE SOLUTION VECTORS TO ORIGINAL ORDER
      DO 90 J=1,N
      J=J+1
140      90-ALL(W(L),L)*A(J,K)
      RETURN
      C      SINGULAR MATRIX = REDUNDANT SET OF EQUATIONS
110 KUGURS
      DELTA=0
145      RETURN
      END

```

```

MATE7 120
MATE7 121
MATE7 122
MATE7 123
MATE7 124
MATE7 125
MATE7 126
MATE7 127
MATE7 128
MATE7 129
MATE7 130
MATE7 131
MATE7 132
MATE7 133
MATE7 134
MATE7 135
MATE7 136
MATE7 137
MATE7 138
MATE7 139
MATE7 140
MATE7 141
MATE7 142
MATE7 143
MATE7 144
MATE7 145
MATE7 146
MATE7 147

```

```

1      SUBROUTINE GINTHG(SIDE,LIM1,LIM2)
      RETURN
      END
      UNTHOG 2
      UNTHOG 3
      UNTHOG 4

```

```

1      SUBROUTINE PBDY
      COMMON A(25,7)
      1      A(1)UL
      2      A(2)C(25)
      3      A(3)CPS(25)
      4      A(4)KXSTK
      5      A(5)D(1)
      6      A(6)UIM(25,25)
      7      A(7)UIN
      8      A(8)EPSLUB
      9      A(9)M(1)
      10     A(10)M2PM(25)
      11     A(11)IKRUK
      12     A(12)IFUL
      13     A(13)IDIFF
      14     A(14)LU
      15     A(15)MUP
      16     A(16)MASH
      17     A(17)MUC
      18     A(18)MBOUND
      19     A(19)MUMIN
      20     A(20)MUCAS
      21     A(21)P(40)
      22     A(22)P(40)
      23     A(23)P(40)
      24     A(24)P(40)
      25     A(25)P(40)
      26     A(26)P(40)
      27     A(27)P(40)
      28     A(28)P(40)
      29     A(29)P(40)
      30     A(30)P(40)
      31     A(31)P(40)
      32     A(32)P(40)
      33     A(33)P(40)
      34     A(34)P(40)
      35     A(35)P(40)
      36     A(36)P(40)
      37     A(37)P(40)
      38     A(38)P(40)
      39     A(39)P(40)
      40     A(40)P(40)
      41     A(41)P(40)
      42     A(42)P(40)
      43     A(43)P(40)
      44     A(44)P(40)
      45     A(45)P(40)
      46     A(46)P(40)
      47     A(47)P(40)
      48     A(48)P(40)
      49     A(49)P(40)
      50     A(50)P(40)
      51     A(51)P(40)
      52     A(52)P(40)
      53     A(53)P(40)
      54     A(54)P(40)
      55     A(55)P(40)
      56     A(56)P(40)
      57     A(57)P(40)
      58     A(58)P(40)
      59     A(59)P(40)
      60     A(60)P(40)
      61     A(61)P(40)
      62     A(62)P(40)
      63     A(63)P(40)
      64     A(64)P(40)
      65     A(65)P(40)
      66     A(66)P(40)
      67     A(67)P(40)
      68     A(68)P(40)
      69     A(69)P(40)
      70     A(70)P(40)
      71     A(71)P(40)
      72     A(72)P(40)
      73     A(73)P(40)
      74     A(74)P(40)
      75     A(75)P(40)
      76     A(76)P(40)
      77     A(77)P(40)
      78     A(78)P(40)
      79     A(79)P(40)
      80     A(80)P(40)
      81     A(81)P(40)
      82     A(82)P(40)
      83     A(83)P(40)
      84     A(84)P(40)
      85     A(85)P(40)
      86     A(86)P(40)
      87     A(87)P(40)
      88     A(88)P(40)
      89     A(89)P(40)
      90     A(90)P(40)
      91     A(91)P(40)
      92     A(92)P(40)
      93     A(93)P(40)
      94     A(94)P(40)
      95     A(95)P(40)
      96     A(96)P(40)
      97     A(97)P(40)
      98     A(98)P(40)
      99     A(99)P(40)
      100    A(100)P(40)

```

```

1XZ410
JAY(40)
KXKX
DD=5, JBI, NPG
NBLJ=NBL(J)
NBLJ=NBL(J)
DNDSP(J)=DNDSP(J)+DLS(NBL1)+V2(NBLJ,J)/V2(NBLJ,J)
HNDY(J)=Y2(NBL1)+HNDSP(J)
DD=12, LK=1, NMAX
IF (IHPOT(J) .LT. Y2(IEM+1)) GO TO 15
12. CONTINUE
  IEM(IEM+1)
15. CONTINUE
  IF (IEM.NE.NBL(J)) DNDSP(J)=HNDY(J)+Y2(IEM+1)
  IF (IEM.GT.NMAX) NMAX=IEM
  DD=14, LK=1, IEM
  IF (LX2.GT.NBL(J)) GO TO 17
  GO TO 16
17. V2(LA,J)=V2(LA+1,J)
  V2(LA,J)=V2(LA+1,J)
  IP(LA,J)=IP(LA+1,J)
  NMP2(LA,J)=NMP2(LA+1,J)
18. CONTINUE
  V(LA,J)=V(LA+1,J)
  W(LA,J)=W(LA+1,J)
  IP(LA,J)=IP(LA+1,J)
11. NMP(LA,J)=NMP2(LA,J)
14. CONTINUE
  NBL(J)=IEM
5. CONTINUE
  AM11(6,102)=5, (IHPOT(J),JBI,NPG)
103. FORMAT (I, '33M BOUNDARY OF PARTICLE PHASE AT SE, F1125, IPB11, J)
103. FORMAT (I, '-----NO PARTICLES IN THIS PLG. FIELD //')
RETURN
END

```

1	SUBROUTINE=PPUTIN	PPUTIN	2
	COMMON A(25,7)	ALFA(25,25), ALPHAM	ALPHAP(25,40)
2	ATBL	BETAP	C11(25)
3	C12(25)	CABAR(25)	CHUAR(25)
4	CPS(25)	CPSH(25)	CSINH(25)
5	MAXIM	DEFF(25)	DEFF(25)
6	D11	DELSS(40)	DELSS(40)
7	D1M(25,25)	D12	DELTA(40)
8	D1N	DPMDS(40)	DELTA(40)
9	FPSLN	FEXTRA(50)	FSTEP
10	F11	F12(40)	F13
11	F1M(25)	F1N(25)	F1O(25)
12	F1L(40)	F1XTRA(50)	F1Y(40)
13	F1P(40)	F1SHOCK	F1TYPE
14	F1DIFF	F1N	F1AYS
15	F1L(40)	F1MAX	F1AYS
16	F1OP	F1NA	F1LPLANE
17	F1MASH	F1MDO1	F1MU
18	F1M2	F1MUS	F1M2
19	F1MOUND	F1MUS	F1M2
20	F1MOUND	F1MUS	F1M2
21	F1MOUND	F1MUS	F1M2
22	F1MOUND	F1MUS	F1M2
23	F1MOUND	F1MUS	F1M2
24	F1MOUND	F1MUS	F1M2
25	F1MOUND	F1MUS	F1M2
26	F1MOUND	F1MUS	F1M2
27	F1MOUND	F1MUS	F1M2
28	F1MOUND	F1MUS	F1M2
29	F1MOUND	F1MUS	F1M2
30	F1MOUND	F1MUS	F1M2
31	F1MOUND	F1MUS	F1M2
32	F1MOUND	F1MUS	F1M2
33	F1MOUND	F1MUS	F1M2
34	F1MOUND	F1MUS	F1M2
35	F1MOUND	F1MUS	F1M2
36	F1MOUND	F1MUS	F1M2
37	F1MOUND	F1MUS	F1M2
38	F1MOUND	F1MUS	F1M2
39	F1MOUND	F1MUS	F1M2
40	F1MOUND	F1MUS	F1M2
41	F1MOUND	F1MUS	F1M2
42	F1MOUND	F1MUS	F1M2
43	F1MOUND	F1MUS	F1M2
44	F1MOUND	F1MUS	F1M2
45	F1MOUND	F1MUS	F1M2
46	F1MOUND	F1MUS	F1M2
47	F1MOUND	F1MUS	F1M2
48	F1MOUND	F1MUS	F1M2
49	F1MOUND	F1MUS	F1M2
50	F1MOUND	F1MUS	F1M2
51	F1MOUND	F1MUS	F1M2
52	F1MOUND	F1MUS	F1M2
53	F1MOUND	F1MUS	F1M2
54	F1MOUND	F1MUS	F1M2
55	F1MOUND	F1MUS	F1M2
56	F1MOUND	F1MUS	F1M2
57	F1MOUND	F1MUS	F1M2
58	F1MOUND	F1MUS	F1M2
59	F1MOUND	F1MUS	F1M2
60	F1MOUND	F1MUS	F1M2

	1. SUPP, SUPPV, SUPPE	PPUTIN	14
	LD=DN/XYZ/XYZ, JYZ	PPUTIN	15
	GIMENSION: VI(B); AI(B); TPI(B); NMP1(B) , DENG1(B)	PPUTIN	16
	NAMELIST/NAME1/ FFF, FFG, UMI ; CL, CS, IPS, NMB, AT, HTHAN, SIG, EP, NPG,	PPUTIN	17
65	1. NC, KPA, RP, SI, VI, TPI, NMP1	PPUTIN	18
	XYZ=1	PPUTIN	19
	JYZ=0	PPUTIN	20
	HEAD (5,100) FFF, FFG, CL, CS, HTHAN, AT	PPUTIN	21
	HEAD (5,100) NMB, SIG, EP , IPS , UMI	PPUTIN	22
70	HEAD (5,107) NPG, AC	PPUTIN	23
	HEAD (5,107) (NCL(J), J1, NPG)	PPUTIN	24
	DO 5 J1, NPG	PPUTIN	25
75	25 NCL(J)=NCL(J)+1	PPUTIN	26
	KMABNBL(1)	PPUTIN	27
	IF (NPG, LE, 1) GO TO 245	PPUTIN	28
	DO 241 J2, NPG	PPUTIN	29
	IF (NMA, LT, NBL(J)) KMABNBL(J)	PPUTIN	30
	241 CONTINUE	PPUTIN	31
80	245 CONTINUE	PPUTIN	32
	HEAD (5,100) (NPI(J), J1, NPG)	PPUTIN	33
	IF (NCL, EQ, 1) GO TO 2	PPUTIN	34
	HEAD (5,100) (N1(J), J1, NPG)	PPUTIN	35
	HEAD (5,100) (V1(J), J1, NPG)	PPUTIN	36
	HEAD (5,100) (TPI(J), J1, NPG)	PPUTIN	37
85	HEAD (5,100) (NMP1(J), J1, NPG)	PPUTIN	38
	HEAD (5,100) (DEFG1(J), J1, NPG)	PPUTIN	39
	DO 5 J1, NPG	PPUTIN	40
	KMABNBL(J)	PPUTIN	41
90	DO 4 K1, KPA	PPUTIN	42
	K(R, J)=K1(J)	PPUTIN	43
	V(K, J)=K1(J)*V1(J)/(K+1)	PPUTIN	44
	(R(K, J)=R1(J)	PPUTIN	45
	NMP(K, J)=NMP1(J)	PPUTIN	46
	DEFG(K, J)=DEFG1(J)	PPUTIN	47
95	V2(K, J)=V2(K, J)	PPUTIN	48
	V2(K, J)=V2(K, J)	PPUTIN	49
	IP2(K, J)=IP2(K, J)	PPUTIN	50
	KMP2(K, J)=KMP(K, J)	PPUTIN	51
100	4. CONTINUE	PPUTIN	52
	HEAD (5,100) 0	PPUTIN	53
	HEAD (5,100) (NMA, J1)	PPUTIN	54
	5. CONTINUE	PPUTIN	55
	DO 21 J1, NPG	PPUTIN	56
105	KMABNBL(J)	PPUTIN	57
	DO 21 K1, KPA	PPUTIN	58
	TCM(K, J)=0	PPUTIN	59
	21 CONTINUE	PPUTIN	60
	GO TO 3	PPUTIN	61
110	2. CONTINUE	PPUTIN	62
	DO 1 K1, KPA	PPUTIN	63
	HEAD (5,100) (K(K, J), J1, NPG)	PPUTIN	64
	HEAD (5,100) (V(K, J), J1, NPG)	PPUTIN	65
	HEAD (5,100) (IP(K, J), J1, NPG)	PPUTIN	66
	HEAD (5,100) (NMP(K, J), J1, NPG)	PPUTIN	67
115	HEAD (5,107) (IC(K, J), J1, NPG)	PPUTIN	68
	HEAD (5,100) (DEFG(K, J), J1, NPG)	PPUTIN	69
	DO 6 J1, NPG	PPUTIN	70
	K2(K, J)=K(K, J)	PPUTIN	71
	V2(K, J)=V(K, J)	PPUTIN	72
120	IP2(K, J)=IP(K, J)	PPUTIN	73
	KMP2(K, J)=KMP(K, J)	PPUTIN	74
	6. CONTINUE	PPUTIN	75
	1. CONTINUE	PPUTIN	76
125	HEAD (5,100) (FFDY(J), J1, NPG)	PPUTIN	77
	HEAD (5,100) (IP(LP(J), J1, NPG)	PPUTIN	78
	3. CONTINUE	PPUTIN	79
	K1TE(6, NMA)	PPUTIN	80
	DO 7 J1, NPG	PPUTIN	81
130	QCL(J)=1.333333*PI*(K(J)+3)	PPUTIN	82
	IP(J)=QCL(J)*KMP(J)	PPUTIN	83
	7. CONTINUE	PPUTIN	84
	DO 12 J1, NPG	PPUTIN	85
	IF (KMAX=NBL(J)) 12, 12, 11	PPUTIN	86
135	11. KMA1=NBL(J)+1	PPUTIN	87
	DO 10 K1=NMA1, NMA2	PPUTIN	88
	V(K, J)=0	PPUTIN	89
	IP(K, J)=0	PPUTIN	90
	KMP(K, J)=0	PPUTIN	91
140	K2(K, J)=0	PPUTIN	92
	V2(K, J)=0	PPUTIN	93
	IP2(K, J)=0	PPUTIN	94
	KMP2(K, J)=0	PPUTIN	95
	DEFG(K, J)=0	PPUTIN	96
145	IC(K, J)=0	PPUTIN	97
	IC(K, J)=0	PPUTIN	98
	10. CONTINUE	PPUTIN	99
	12. CONTINUE	PPUTIN	100
	DO 22 K1, KPA	PPUTIN	101
	K1TE(6, 101) K	PPUTIN	102
150	K1TE(6, 102) (K(K, J), J1, NPG)	PPUTIN	103
	K1TE(6, 103) (V(K, J), J1, NPG)	PPUTIN	104
	K1TE(6, 105) (IP(K, J), J1, NPG)	PPUTIN	105
	K1TE(6, 106) (KMP(K, J), J1, NPG)	PPUTIN	106
	22. CONTINUE	PPUTIN	107
155	100. FORMATT(16, 3)	PPUTIN	108
	101. FORMATT(16, 15) STREAMLINE NO. , 15)	PPUTIN	109
	102. FORMATT(16, 14) STREAMLINE VELOCITY , 24, IP(12, 3)	PPUTIN	110
	103. FORMATT(16, 13) STREAMLINE VELOCITY , IP(12, 3)	PPUTIN	111
	104. FORMATT(16, 12) COMPRESSIBILITY , 12, IP(12, 3)	PPUTIN	112
160	105. FORMATT(16, 11) COMPRESSIBILITY , 54, IP(12, 3)	PPUTIN	113
	106. FORMATT(16, 10) COMPRESSIBILITY , 54, IP(12, 3)	PPUTIN	114
	107. FORMATT(16, 9) COMPRESSIBILITY , 54, IP(12, 3)	PPUTIN	115
	108. FORMATT(16, 8) COMPRESSIBILITY , 54, IP(12, 3)	PPUTIN	116

```

SUBROUTINE PART
COMMON A(25,7)
1 1 .SATOL .SATA(40) .ALFA(25,25) .ALPHAM .ALPHAP PART 2
2 .C12(25) .C2(25,40) .CANAM(25) .C11(25) .C(25,40) PART 3
3 .EPS(25) .CPM .CANAM(25) .CBAM(25) .CP PART 4
4 .SRSTM .C2IM(25,25) .DEFF(25) .C3M(25) .C3THEM(25) PART 5
5 .D11 .DELBO(40) .DELB .DELEFF(25) .DELTA PART 6
6 .DIM(25,25) .D12 .DELY(40) .D13 .DOPY(40) PART 7
7 .D14 .DPHDS(40) .EPCIM PART 8
8 .EPBLUM .EXTNA(50) .E3IEP .FMAX .GNAD PART 9
9 .H11 .H(40) .HM .HU .MNM(25) PART 10
10 .H2PM(25) .H3PM(25) .ICNBT .ICOUNT .IOCNT(25) PART 11
11 .JEMOM .JEXTRA(50) .JFLAG .JINNO PART 12
12 .JPTUC .JSHOCK .JTYPE .JPD PART 13
13 .JRIFF .K .KAY .KAYS .KAY2 PART 14
14 .KLO .KPAK .K PART 15
15 .KUP .K PART 16
16 .KABM .KDO1 .KLL .LPLANE .MA PART 17
17 .KUZ .KMS .KMU .MUB PART 18
18 .KABUND .KMS .KMU .MUB PART 19
19 .KABUND .KMS .KMU .MUB PART 20
20 .KABUND .KMS .KMU .MUB PART 21
21 .KABUND .KMS .KMU .MUB PART 22
22 .KABUND .KMS .KMU .MUB PART 23
23 .KABUND .KMS .KMU .MUB PART 24
24 .KABUND .KMS .KMU .MUB PART 25
25 .KABUND .KMS .KMU .MUB PART 26
26 .KABUND .KMS .KMU .MUB PART 27
27 .KABUND .KMS .KMU .MUB PART 28
28 .KABUND .KMS .KMU .MUB PART 29
29 .KABUND .KMS .KMU .MUB PART 30
30 .KABUND .KMS .KMU .MUB PART 31
31 .KABUND .KMS .KMU .MUB PART 32
32 .KABUND .KMS .KMU .MUB PART 33
33 .KABUND .KMS .KMU .MUB PART 34
34 .KABUND .KMS .KMU .MUB PART 35
35 .KABUND .KMS .KMU .MUB PART 36
36 .KABUND .KMS .KMU .MUB PART 37
37 .KABUND .KMS .KMU .MUB PART 38
38 .KABUND .KMS .KMU .MUB PART 39
39 .KABUND .KMS .KMU .MUB PART 40
40 .KABUND .KMS .KMU .MUB PART 41
41 .KABUND .KMS .KMU .MUB PART 42
42 .KABUND .KMS .KMU .MUB PART 43
43 .KABUND .KMS .KMU .MUB PART 44
44 .KABUND .KMS .KMU .MUB PART 45
45 .KABUND .KMS .KMU .MUB PART 46
46 .KABUND .KMS .KMU .MUB PART 47
47 .KABUND .KMS .KMU .MUB PART 48
48 .KABUND .KMS .KMU .MUB PART 49
49 .KABUND .KMS .KMU .MUB PART 50
50 .KABUND .KMS .KMU .MUB PART 51
51 .KABUND .KMS .KMU .MUB PART 52
52 .KABUND .KMS .KMU .MUB PART 53
53 .KABUND .KMS .KMU .MUB PART 54
54 .KABUND .KMS .KMU .MUB PART 55
55 .KABUND .KMS .KMU .MUB PART 56
56 .KABUND .KMS .KMU .MUB PART 57
57 .KABUND .KMS .KMU .MUB PART 58
58 .KABUND .KMS .KMU .MUB PART 59
59 .KABUND .KMS .KMU .MUB PART 60
60 .KABUND .KMS .KMU .MUB PART 61
61 .KABUND .KMS .KMU .MUB PART 62
62 .KABUND .KMS .KMU .MUB PART 63
63 .KABUND .KMS .KMU .MUB PART 64
64 .KABUND .KMS .KMU .MUB PART 65
65 .KABUND .KMS .KMU .MUB PART 66
66 .KABUND .KMS .KMU .MUB PART 67
67 .KABUND .KMS .KMU .MUB PART 68
68 .KABUND .KMS .KMU .MUB PART 69
69 .KABUND .KMS .KMU .MUB PART 70
70 .KABUND .KMS .KMU .MUB PART 71
71 .KABUND .KMS .KMU .MUB PART 72
72 .KABUND .KMS .KMU .MUB PART 73
73 .KABUND .KMS .KMU .MUB PART 74
74 .KABUND .KMS .KMU .MUB PART 75
75 .KABUND .KMS .KMU .MUB PART 76
76 .KABUND .KMS .KMU .MUB PART 77
77 .KABUND .KMS .KMU .MUB PART 78
78 .KABUND .KMS .KMU .MUB PART 79
79 .KABUND .KMS .KMU .MUB PART 80
80 .KABUND .KMS .KMU .MUB PART 81
81 .KABUND .KMS .KMU .MUB PART 82
82 .KABUND .KMS .KMU .MUB PART 83
83 .KABUND .KMS .KMU .MUB PART 84
84 .KABUND .KMS .KMU .MUB PART 85
85 .KABUND .KMS .KMU .MUB PART 86
86 .KABUND .KMS .KMU .MUB PART 87
87 .KABUND .KMS .KMU .MUB PART 88
88 .KABUND .KMS .KMU .MUB PART 89
89 .KABUND .KMS .KMU .MUB PART 90
90 .KABUND .KMS .KMU .MUB PART 91
91 .KABUND .KMS .KMU .MUB PART 92
92 .KABUND .KMS .KMU .MUB PART 93
93 .KABUND .KMS .KMU .MUB PART 94
94 .KABUND .KMS .KMU .MUB PART 95
95 .KABUND .KMS .KMU .MUB PART 96
96 .KABUND .KMS .KMU .MUB PART 97
97 .KABUND .KMS .KMU .MUB PART 98
98 .KABUND .KMS .KMU .MUB PART 99
99 .KABUND .KMS .KMU .MUB PART 100

```



```

100 C ENERGY EQUATION PART 52
CDEL PART 53
IF (TP(N,J).LE.TPS) CDEB PART 54
IF (ICOM(N,J).EQ.0) GO TO 14 PART 55
XNUM=TP(J)+EMEP2(J) DEL /M(N,J)/M PART 56
XNUM=TP(J)+EMEP2(J) DEL /M(N,J)/M PART 57
DEB(N,J)=DEB(N,J)+XNUM PART 58
IF (ABS(DEB(N,J)).GE.C(J)) GO TO 15 PART 59
GO TO 12 PART 60
15 CONTINUE PART 61
ICOM(N,J)= PART 62
WRITE (6,102) M(J),S,K,DEB(N,J) PART 63
110 14 CONTINUE PART 64
XNUM=EMEP2(J)/M(J)/CC PART 65
XNUM=TP(J)+EMEP2(J) DEL /M(N,J)/M PART 66
TP2(N,J)=TP(N,J)+XNUM*OTON=(XNUM+XNUM)*DEL PART 67

115 IF ((TP2(N,J)-TP(N,J))>(TP(N,J)-TP(N,J)).GE.0.0) GO TO 12 PART 68
ICOM(N,J)= PART 69
TP2(N,J)=TP(N,J) PART 70
WRITE (6,103) M(J),S,K PART 71
103 FORMAT (1M,1PE10.3,3M,CM,PARTICLE GROUP START SOLIDIFYING AT SR PART 72
1,OPP10.0,1M,CM,STREAMLINE NO.,15/) PART 73
120 12 CONTINUE PART 74
C CONTINUITY EQUATION PART 75
XN1=(S*(R(N+1)+R(N)))>>DELTA PART 76
XN2=(S*(M2(N+1)+M2(N)))>>DELTA PART 77
XN10=XN1-EMEP(N,J)*R(N,J) PART 78
XN10=XN1-EMEP(N,J)*V(N,J) PART 79
XN2(N,J)=1.0/(M2(N,J)+XN10(1.0-DEL.ATHEM)) PART 80
1 XN10=XN1-DEL PART 81
2 XN10=XN1-DEL PART 82
130 GO TO 11 PART 83
3 V2(N,J)=0 PART 84
M2(N,J)=0 PART 85
TP2(N,J)=0 PART 86
XN2(N,J)=0.0 PART 87
M2(N,J)=0.0 PART 88
135 11 CONTINUE PART 89
102 FORMAT (1M,1PE10.3,4M,CM,SIZE PARTICLE GROUP SOLIDIFIED AT SR, PART 90
1,OPP10.0,1M,CM,STREAMLINE NO.,15/,3M,1PE10.3,3M,CM,/) PART 91
M(KK) PART 92
140 M(KK) PART 93
RETURN PART 94
END PART 95

```

```

SUBROUTINE PUTOUT(10UT) PUTOUT 2
COMMON A(25,7) AAL(40) ALFA(25,25) ALPHAM ALPHAM PUTOUT 2
1 ATUL METAP MP1X C11(25) C(25,40) PUTOUT 3
2 C12(25) J2(25,40) LABAH(25) CHAH(25) CP PUTOUT 4
3 CP3(25) CP3M(25) CCM(25) CCM1(25) CSTEM(25) PUTOUT 5
4 INSTAP J2TH(25,25) DEFF(25) DEFF(25) DELTA PUTOUT 6
5 U11 DELS(40) UELB DELBU DLS(40) PUTOUT 7
6 J1M(25,25) D12 DELY(40) D13 DPOV(40) PUTOUT 8
7 U14 GPH1D(40) PCUN PUTOUT 9
8 EPBLUM EXTHA(50) PSTEP PMAI PHAN PUTOUT 10
9 PH1 H(40) PHM PHJ PHM(25) PUTOUT 11
0 M2PM(25) MSPM(25) MCONB ICOUNT ISENT(25) PUTOUT 12
1 ERMUM EXTHA(50) IFLAG IIND PUTOUT 13
2 ITUC ISMOCK ITYPE IMPD PUTOUT 14
3 IUFF R RAY RAYS RAY2 PUTOUT 15
4 SALD RMAX PUTOUT 16
5 RUP RKA ALL PLANE MA PUTOUT 17
6 MASH MDUT MMAX MUO MU PUTOUT 18
7 MUB MUS MO M2 MMBH PUTOUT 19
8 MUMUND MDS MITER MMAX MN PUTOUT 20
COMMON KUCASE (MEGAT25) P11 H(40) P12 P12 PUTOUT 21
1 P2(40) PH(50) PABAH PHAN PUTOUT 22
2 PHS P15 PH1(40) PH1(50) PH1(50) PUTOUT 23
3 PH1M1 PMSTM PH1(50) PH1(50) PH1(50) PUTOUT 24
4 PH12(40) PH1B P1 PH1(25) PH1(50) PUTOUT 25
5 PH1 P11 PSTEM PH(50) PUTOUT 26
6 U11 U(40) UXTM1 UXTM2 PH(50) PUTOUT 27
7 PH(50) PH1 H(40) H(40) H(50) PUTOUT 28
8 RMS PH1 MHAH(40) H14 H2(40) PUTOUT 29
9 MUM H15 HE(40) HESH H14 PUTOUT 30
0 MUM(40) M17 M12(40) MMS(25) MMSA PUTOUT 31
1 MMSUM MMBAH MMBAH MU BSM PUTOUT 32
2 MSTEM MV M19 M(100) S PUTOUT 33
3 SH(50) SC(25) SM(50) PUTOUT 34
4 SX(40) T11 T(40) T12 PUTOUT 35
5 T2(40) TABAH TABAH T9(25) T14 PUTOUT 36
6 TAR(40) TATH TATH TUL TSM PUTOUT 37
7 TSM TSTEM T(50) TMS PUTOUT 38
8 TS U11 U(40) U12 U2(40) PUTOUT 39
9 C(MPH) UXTM1 UXTM2 U14 UMAM(40) USM PUTOUT 40
1 UBM1 UXTM UN(50) UMB PUTOUT 41
2 X11 X(40) X12 X2(40) PUTOUT 42
3 X10(50) XMS XAHAR(25) XMAH(25) X15 PUTOUT 43
4 X3(25,40) XSM XSTEM XN(50) PUTOUT 44
5 XMAH(40) X11 Y(40) Y12 PUTOUT 45
6 Y2(40) YABAH YBAH ZA(25) Z11(25) PUTOUT 46
7 ZJ(25,40) ZM ZX Z2M PUTOUT 47
8 FX(2,40) FK(2,40) FPH1(2,40) FP(2,40) FT(2,40) PUTOUT 48
9 FU(2,40) FNDL(2,40) FNDM(2,40) FC(2,40,25) CARD PUTOUT 49
0 M M PUTOUT 50
REAL RAY RAYS(25) RAY2 RA MOUT(40) PUTOUT 4

```

```

55  COMMON/PT1/ANOMY(8), ANOMP(8), ENEP1(8), REP(40,8), Y(40,8), W(40,8),
    1V2(40,8), K2(40,8), TRDY(8), TP(40,8), TP2(40,8), HY(8),
    2RNP(40,8), RNP2(40,8), ENEP2(8), HML(8), UM(40)
    PUTOUT 5
    PUTOUT 6
    PUTOUT 7
    PUTOUT 8
    PUTOUT 9
    PUTOUT 10

60  3. VOLP(8), NTF(8), MC(8), DENG(40,8), ICOMO(40,8)
    PUTOUT 11
    COMMON/PT2/FFF,FFG,GAHA,UMI,PO,MPG,KMX,DNDSP(8)
    PUTOUT 12
    COMMON/PT3/CL,CS,TPS,RMSB,RT,HTAN,STG,EP,IPANT,IAIAE,TR(40)
    PUTOUT 13
    1. SUMP, SUMPY, SUMPY
    PUTOUT 14
    COMMON/CHM1/ ZID(5), INH(40), INT(40), MC(40,3), INNR(40,5)
    PUTOUT 15
    1. AV, CM(26,26), FT(26), AP(26),
    PUTOUT 16
    2. RM(26), HDOT(26,40), CSI(26), OK(26)
    PUTOUT 17
65  COMMON/SHOUT/ XSMN, RSMN, PBR, PRMN, ISMN, USMN, CSN(25)
    PUTOUT 18
    COMMON/CF/ IPTO
    PUTOUT 19
    COMMON/XYZ/ XYZ, JYZ
    PUTOUT 20
    COMMON/BLF/ INFLG, IOL
    PUTOUT 21
    COMMON/MALL/ TRALL
    PUTOUT 22
    COMMON/BLOUT/ MEXI, THETA1, CFI, UFI, CPCCF1, M12, DISP, THETA2(2), DELTI,
    DELTAC, CFC, OFALL, TAUALL, FTA(140), TETC(140), UHUF(140), YDELI(140)
    PUTOUT 23
    DIMENSION: SUM1(25)
    PUTOUT 24
    DATA I01/AMC1 /
    PUTOUT 25
    DATA I02/AMC2 /
    PUTOUT 26
75  DATA I03/AMC3 /
    PUTOUT 27
    DATA I04/AMC4 /
    PUTOUT 28
    DATA I05/AMC5 /
    PUTOUT 29
    DATA I06/AMC6 /
    PUTOUT 30
    DATA I07/AMC7 /
    PUTOUT 31
80  IYZ=13
    PUTOUT 32
    JYZ=0
    PUTOUT 33
    GO TO (50,1300,10), IOUT
    PUTOUT 34
    10 L&K
    PUTOUT 35
    GO TO 250
    PUTOUT 36
85  50 L&I
    PUTOUT 37
    SUMG1=0
    PUTOUT 38
    DO 75 I=1,NUS
    PUTOUT 39
    75 SUM1(I)=0
    PUTOUT 40
90  WRITE (6,100) LL,X2(L),R2(L),PHI2(L)
    PUTOUT 41
    100 FORMAT ('100:IS,AR,3MNR, E11.4,3X,3MNR, E11.4,3X,5MNHAB,
    AT11.4)
    PUTOUT 42
    101
    PUTOUT 43
    250 IF (LL) 225,260,260
    PUTOUT 44
    225 WRITE (6,300) U(X(L),A(L),PHI(L),T(L),M(L),U(L)
    PUTOUT 45
    GO TO 325
    PUTOUT 46
95  260 WRITE (6,300) L,X2(L),R2(L),PHI2(L),T2(L),P2(L),U2(L)
    PUTOUT 47
    300 FORMAI (100,5X,15,3X,2MNR,1P1E11.4,4X,2MNR,1P1E11.4,4X,4MNHAB,1P1E11.4,4X,4MNHAB,1P1E11.4)
    PUTOUT 48
    325 HTRM(L)=U(L)**2/(2.0*H)
    PUTOUT 49
    PTRM(L)=RML(L)*U(L)**2/(2.0*HETAP)
    PUTOUT 50
    100 SUMG1=SUMG1+MDOT(L)
    PUTOUT 51
    375 IF (LL) 350,360,360
    PUTOUT 52
    350 WRITE (6,400) A(L),DFLY(L),M(L),HTR, TAN(L),U(L),PIA,MMO(L),SR(L)
    PUTOUT 53
    400
    PUTOUT 54
105  400 FORMAI (10X,3MNR, E11.4,3X,5MDELTA, E11.4,3X,2MNR, E11.4,4X,
    3MNHAB, E11.4,3X,4MNHAB, E11.4,2X,2MNR, E11.4,4,14X,3MPTM,
    E11.4,3X,4MNHAB, E11.4,2X,3MNR, E11.4,3X,7MSUMDOT, E11.4
    )
    PUTOUT 55
    GO TO 410
    PUTOUT 56
110  360 WRITE (6,400) A(L),L2LY(L),M(L),HTR, TAN(L),U(L),PTR,MMO2(L),SR(L)
    PUTOUT 57
    400
    PUTOUT 58
    IF (L2LY,AMAX) GO TO 410
    PUTOUT 59
    WRITE (6,1001)
    PUTOUT 60
    1001 FORMAI (//12X(//))
    PUTOUT 61

115  IF (IHL,EO=0) GO TO 408
    255EPT 11
    WRITE (6,1007) TRALL
    255EPT 12
    1007 FORMAI (50X,17HALL TEMPERATURE,1X,FS,0,1X,6MFLV1)
    255EPT 13
    WRITE (6,1002)
    PUTOUT 14
120  1002 FORMAI (50X,25HSECONDARY LAYER PARAMETERS)
    255EPT 15
    WRITE (6,1003) RETI, DELTI, THETA1, CFI, UFI, DISP, DELTAC, THETA2(2),
    CFC, CPCCF1, M12, GALL, TAUALL
    PUTOUT 16
    1003 FORMAI (15,4MRES, E11.4,124, *DELTA1, E11.4,146, *THETA1, E11.4,
    1E11.4,106, *CFI, E11.4,117, *UFI, E11.4,
    215, *DISP, E11.4,124, *DELTAC, E11.4,116, *THETA2, E11.4,
    316, *CFC, E11.4,117, *CFC/CFI, E11.4,
    415, *M12, E11.4,124, *GALL, E11.4,146, *TAUALL, E11.4)
    PUTOUT 17
    IF (IHL,FLG) 1004,410,1004
    PUTOUT 18
    1004 WRITE (6,1005)
    PUTOUT 19
130  1005 FORMAI (//15, *SECONDARY LAYER PHIFILES, //126, 0U/0L, 107,
    10, 17E, 107, 0U/0F, JACA, JAH, 0Y/0EL)
    PUTOUT 20
    WRITE (6,1006) (FA(J), TETC(J), UHUF(J), YDELI(J), J=5,100,5)
    PUTOUT 21
    1006 FORMAI (125, E11.4, 145, E11.4, 145, E11.4, 145, E11.4, 145, E11.4)
    PUTOUT 22
    408 WRITE (6,409)
    255EPT 15
    409 FORMAI (45X, 27HSECONDARY LAYER NOT COMPUTED)
    255EPT 16
135  410 WRITE (6,1000)
    PUTOUT 17
    1000 FORMAI (100)
    PUTOUT 18
    IF (MUS=1) 450,450,1050
    PUTOUT 19
    1050 DO 1299 I=1,NUS
    PUTOUT 20
    IF (EXTRA(I)) 1120,1110,1120
    PUTOUT 21
140  1110 IF (L) 1105,1115,1115
    PUTOUT 22
    1105 WRITE (6,1200) IREX(I),C(I),L),AS(I,L),ADOT(I,L)
    PUTOUT 23
    GO TO 1299
    PUTOUT 24
    1115 WRITE (6,1200) IREX(I),C2(I,L),AS(I,L),ADOT(I,L)
    PUTOUT 25
145  1200 FORMAI (10X,6HSPTCIE,2X, 4X,2X,2MNR, E11.4,4X,2MNR, E11.4,4X,
    5MNDOT, E11.4)
    PUTOUT 26
    GO TO 1299
    PUTOUT 27

```

```

1120 SUM1(I)=SUM1(I)+C01(L)*C(I,L)          PUTOUT 94
1150 IF (LL) 1125,1160,1160                PUTOUT 95
1125 WRITE (6,1225) IDENT(I),C(I,L),X(I,L),MMU(I,L),SUM1(I),Z(I,L) PUTOUT 96
1225 FORMAT (1X,6H0PCIE,2X, A4,7X,2MC0,1P1E11,4X,2MX,1P1E11,4,4X, PUTOUT 97
      * 5M0DT0,E11,4,3X,3MF10,
      * 1P1E11,4,3X,3M230,1P1E11,4)        PUTOUT 98
      GO TO 1299                             PUTOUT 100
1160 WRITE (6,1225) IDENT(I),C2(I,U),XS(I,L),MMU(I,L),SUM1(I),Z(I,L) PUTOUT 101
1299 CONTINUE                               PUTOUT 102
450 IF (IP10 .EQ. 0) GO TO 457             PUTOUT 103
      XL=Z(L)/M2                             PUTOUT 104
      AL=Z(L)/M2                             PUTOUT 105
1220Z5                                       PUTOUT 106
1230Z5                                       PUTOUT 107
1240Z5                                       PUTOUT 108
1250Z5                                       PUTOUT 109
1260Z5                                       PUTOUT 110
1270Z5                                       PUTOUT 111
105 DO 0U J=1,MS                             PUTOUT 112
      IF (IDENT(J) .EQ. 101) 1710J         PUTOUT 113
      IF (IDENT(J) .EQ. 102) 1720J         PUTOUT 114
      IF (IDENT(J) .EQ. 103) 1730J         PUTOUT 115
      IF (IDENT(J) .EQ. 104) 1740J         PUTOUT 116
170 IF (IDENT(J) .EQ. 105) 1750J         PUTOUT 117
      IF (IDENT(J) .EQ. 106) 1760J         PUTOUT 118

      IF (IDENT(J) .EQ. 107) 1770J         PUTOUT 119
      CONTINUE                               PUTOUT 120
175 ENDO,733E22*P2(L)/I2(L)+C2(I27,L)     PUTOUT 121
      VE=0,215*SUM1(I(L))                  PUTOUT 122
      SPC=C2(I27,L)+I2,00E-23+VEE+2,00E-16)+C2(I27,L)+0,7E-0/VEE+C2(I23, PUTOUT 123
      AL)+0,9/(VEE+2)+C2(I28,L)+0,05/(VEE+VEE)+C2(I29,L)+0,29E-23+VEE+ PUTOUT 124
      C2(I28,L)+0,1,05E-23+VEE+0,9E-16)    PUTOUT 125
      ENDO,57E27*P2(L)/SQRT(I2(L))+300     PUTOUT 126
180 SGM=7,10549E-04E1E/ENUE              PUTOUT 127
      IF (IP10,0,4) GO TO 01                PUTOUT 128
      WRITE (6,1400) XL,AL,ENE,ENUE,SGM    PUTOUT 129
1400 FORMAT (2X,6H1/HEX,1P1E11,4,5X,6H1/HEX,E11,4,5X,3M0E,0,11,4,5X, PUTOUT 130
      1 4H0F0,E11,4,5X,6H1G0AN,E11,4)     PUTOUT 131
185 IF (IP10 .EQ. 1 OR IP10 .EQ. 3)       PUTOUT 132
      WRITE (7,1600) XL,AL,ENE,ENUE,SGM    PUTOUT 133
1600 FORMAT (4E14,17/3E19,12)            PUTOUT 134
      IF (IP10 .EQ. 2 OR IP10 .EQ. 4)     PUTOUT 135
      WRITE (8,1601) XL,AL,ENE,ENUE,SGM    PUTOUT 136
190 1601 FORMAT (7E19)                    PUTOUT 137
      GO WRITE (6,1400) XL,AL,ENE,ENUE,SGM PUTOUT 138
1500 FORMAT (5X,6H1/HEX,1P1E11,4,5X,6H1/HEX,E11,4,4X,2M0E,0,11,4,9E, PUTOUT 139
      1 2H0F0,E11,4,10X,6H1C02)=E11,4,4X,6H1M2U)=E11,4,7X,5M1C0)=E11,4) PUTOUT 140
      IF (IP10 .EQ. 5 OR IP10 .EQ. 3)     PUTOUT 141
195 WRITE (7,1600) XL,AL,ENUE,ENUE,SGM    PUTOUT 142
      IF (IP10 .EQ. 6 OR IP10 .EQ. 3)     PUTOUT 143
      WRITE (8,1601) XL,AL,ENUE,ENUE,SGM    PUTOUT 144
457 IF (IP10 .EQ. 0) GO TO 600,600        PUTOUT 145
500 LE=1                                    PUTOUT 146
200 GO TO 250                               PUTOUT 147
1300 IF (NBOUND) 1320,1310,1320          PUTOUT 148
1310 K=1                                    PUTOUT 149
      DELP=IPMI(K)-PMISM                    PUTOUT 150
      WRITE (6,1401) LL,K,X2(K),M2(K),PMI2(A),PMI,DELPMI PUTOUT 151
      K=2                                    PUTOUT 152
205 GO TO 600                               PUTOUT 153
1520 DELP=IPMI(KMAX)-PMISM                PUTOUT 154
      WRITE (6,1401) LL,KMAX,X(KMAX),M(KMAX),PMI(KMAX),PMI,DELPMI PUTOUT 155
1401 FORMAT (1H0,5H0PCIE,2X,3ML0,14,2X,2M0E,12,2X,2M0E,1P1E11,4,2X,2M0E PUTOUT 156
      2X,1P1E11,4,2X,6H1M2U),1P1E11,4,2X,4H0P50,1P1E11,4,2X,7H0LPH0,1P1E PUTOUT 157
      11,4)                                  PUTOUT 158
      GO TO 1400                             PUTOUT 159
      END                                     PUTOUT 160

```

	SUBROUTINE	MMOUT (NAME)	PPRINT	
1	COMMON A(25,7)	AA(40)	ALPHA(25,25),ALPHAM	ALPHAM
	1	ATOL	BUMY	C11(25)
	2	LI2(25)	C2(25,40)	CCAR(25)
5	3	CS(25)	CSM(25)	CSTHEM(25)
	4	CS14	DEFF(25)	DELTA
	5	DELSS(40)	DELS	DELS(40)
	6	DELS(25,25)	DELY(40)	DDPY(40)
	7	DI4	DPHDS(40)	EPCUN
10	8	FPS(10)	FTRK(40)	FSTEP
	9	MI1	MM	MJ
	10	M2PH(25)	M3PH(25)	ICDAST
	11	IKH04	LEATHA(50)	IFLAG
	12	IP10C	ISM0CA	ITYPE
15	13	IP10F	K	KAYS
	14	KAX	KAX	KAY2
	15	KLL	KAX	KAY2
	16	KAP	K	LL
	17	KASH	M001	MMA2
	18	K02	MUS	MMA2
	19	K02	MUS	MMA2
20	20	NBOUND	ADS	NIIEH
	21	CPH04	NOCASE	CFGA(25)
	22	P2(40)	PH(50)	PBAR

1	ALPHA(25,7)	ALPHA(25,7)	ALPHA(25,25)	ALPHAM	ALPHAM	A	7
2	ATOL	ATOL	ATOL	ATOL	ATOL	A	8
3	C12(25)	C12(25,40)	CONAN(25)	C11(25)	C1(25,40)	A	9
4	CPS(25)	CPS	CON(25)	CON(25)	CUM(25)	A	10
5	DELTA	DELTA(25,25)	DEFF(25)	DEFF(25)	DELTA(25)	A	11
6	DI(25,25)	DELTA(40)	DELTA	DELTA	DELTA(40)	A	12
7	DI	DI(25,25)	DI(40)	DI(40)	DI(40)	A	13
8	FPSLCA	FPSLCA	FSTEP	FSTEP	FSTEP	A	14
9	MI	MI	MI	MI	MI	A	15
10	MP(25)	MP(25)	MP(25)	MP(25)	MP(25)	A	16
11	ILMCH	ILMCH	IFLAG	IFLAG	IFLAG	A	17
12	ITUC	ITUC	ITYPE	ITYPE	ITYPE	A	18
13	DIFF	DIFF	DIFF	DIFF	DIFF	A	19
14	RLU	RLU	RLU	RLU	RLU	A	20
15	KUP	KUP	KUP	KUP	KUP	A	21
16	ASH	ASH	ASH	ASH	ASH	A	22
17	U2	U2	U2	U2	U2	A	23
18	NOUGH	NOUGH	NOUGH	NOUGH	NOUGH	A	24
19	COMPH	COMPH	COMPH	COMPH	COMPH	A	25
20	NUCLASE	NUCLASE	NUCLASE	NUCLASE	NUCLASE	A	26
21	P2(40)	P2(40)	P2(40)	P2(40)	P2(40)	A	27
22	PS	PS	PS	PS	PS	A	28
23	PHI(50)	PHI(50)	PHI(50)	PHI(50)	PHI(50)	A	29
24	PHI2(40)	PHI2(40)	PHI2(40)	PHI2(40)	PHI2(40)	A	30
25	PSH	PSH	PSH	PSH	PSH	A	31
26	U1	U1	U1	U1	U1	A	32
27	U(50)	U(50)	U(50)	U(50)	U(50)	A	33
28	MS	MS	MS	MS	MS	A	34
29	CON	CON	CON	CON	CON	A	35
30	MO(40)	MO(40)	MO(40)	MO(40)	MO(40)	A	36
31	RSHCH	RSHCH	RSHCH	RSHCH	RSHCH	A	37
32	RSTRE	RSTRE	RSTRE	RSTRE	RSTRE	A	38
33	SP(50)	SP(50)	SP(50)	SP(50)	SP(50)	A	39
34	BR(40)	BR(40)	BR(40)	BR(40)	BR(40)	A	40
35	T2(40)	T2(40)	T2(40)	T2(40)	T2(40)	A	41
36	TAK(40)	TAK(40)	TAK(40)	TAK(40)	TAK(40)	A	42
37	TSH	TSH	TSH	TSH	TSH	A	43
38	TSHM	TSHM	TSHM	TSHM	TSHM	A	44
39	U1	U1	U1	U1	U1	A	45
40	U(40)	U(40)	U(40)	U(40)	U(40)	A	46
41	U1	U1	U1	U1	U1	A	47
42	U2(40)	U2(40)	U2(40)	U2(40)	U2(40)	A	48
43	USM	USM	USM	USM	USM	A	49
44	USM	USM	USM	USM	USM	A	50
45	X1	X1	X1	X1	X1	A	51
46	X(50)	X(50)	X(50)	X(50)	X(50)	A	52
47	X(25,40)	X(25,40)	X(25,40)	X(25,40)	X(25,40)	A	53
48	Z(40)	Z(40)	Z(40)	Z(40)	Z(40)	A	54
49	Y(40)	Y(40)	Y(40)	Y(40)	Y(40)	A	55
50	J(25,40)	J(25,40)	J(25,40)	J(25,40)	J(25,40)	A	56
51	F(2,40)	F(2,40)	F(2,40)	F(2,40)	F(2,40)	A	57
52	H(2,40)	H(2,40)	H(2,40)	H(2,40)	H(2,40)	A	58
53	CA	CA	CA	CA	CA	A	59
54	CA(40)	CA(40)	CA(40)	CA(40)	CA(40)	A	60
55	CA(25)	CA(25)	CA(25)	CA(25)	CA(25)	A	61
62	MAP	MAP	MAP	MAP	MAP	A	62
63	ALPHA(25,7)	ALPHA(25,7)	ALPHA(25,25)	ALPHAM	ALPHAM	A	63
64	ATOL	ATOL	ATOL	ATOL	ATOL	A	64
65	C12(25)	C12(25,40)	CONAN(25)	C11(25)	C1(25,40)	A	65
66	CPS(25)	CPS	CON(25)	CON(25)	CUM(25)	A	66
67	DELTA	DELTA(25,25)	DEFF(25)	DEFF(25)	DELTA(25)	A	67
68	DI(25,25)	DELTA(40)	DELTA	DELTA	DELTA(40)	A	68
69	DI	DI(25,25)	DI(40)	DI(40)	DI(40)	A	69
70	FPSLCA	FPSLCA	FSTEP	FSTEP	FSTEP	A	70
71	MI	MI	MI	MI	MI	A	71
72	MP(25)	MP(25)	MP(25)	MP(25)	MP(25)	A	72
73	ILMCH	ILMCH	IFLAG	IFLAG	IFLAG	A	73
74	ITUC	ITUC	ITYPE	ITYPE	ITYPE	A	74
75	DIFF	DIFF	DIFF	DIFF	DIFF	A	75
76	RLU	RLU	RLU	RLU	RLU	A	76
77	KUP	KUP	KUP	KUP	KUP	A	77
78	ASH	ASH	ASH	ASH	ASH	A	78
79	U2	U2	U2	U2	U2	A	79
80	NOUGH	NOUGH	NOUGH	NOUGH	NOUGH	A	80
81	COMPH	COMPH	COMPH	COMPH	COMPH	A	81
82	NUCLASE	NUCLASE	NUCLASE	NUCLASE	NUCLASE	A	82
83	P2(40)	P2(40)	P2(40)	P2(40)	P2(40)	A	83
84	PS	PS	PS	PS	PS	A	84
85	PHI(50)	PHI(50)	PHI(50)	PHI(50)	PHI(50)	A	85
86	PHI2(40)	PHI2(40)	PHI2(40)	PHI2(40)	PHI2(40)	A	86
87	PSH	PSH	PSH	PSH	PSH	A	87
88	U1	U1	U1	U1	U1	A	88
89	U(50)	U(50)	U(50)	U(50)	U(50)	A	89
90	MS	MS	MS	MS	MS	A	90
91	CON	CON	CON	CON	CON	A	91
92	MO(40)	MO(40)	MO(40)	MO(40)	MO(40)	A	92
93	RSHCH	RSHCH	RSHCH	RSHCH	RSHCH	A	93
94	RSTRE	RSTRE	RSTRE	RSTRE	RSTRE	A	94
95	SP(50)	SP(50)	SP(50)	SP(50)	SP(50)	A	95
96	BR(40)	BR(40)	BR(40)	BR(40)	BR(40)	A	96
97	T2(40)	T2(40)	T2(40)	T2(40)	T2(40)	A	97
98	TAK(40)	TAK(40)	TAK(40)	TAK(40)	TAK(40)	A	98
99	TSH	TSH	TSH	TSH	TSH	A	99
100	TSHM	TSHM	TSHM	TSHM	TSHM	A	100


```

200      DZLFF(I)=DELFF(I)                STEP 153
      DO 130M J=1,NDZ
      DZ1M(I,J)=DZ1M(I,J)                STEP 154
130M  CONTINUE                             STEP 155
130M  CP=0.                                STEP 156
205      X=1310 I=1,MAX                    STEP 157
1310  CP=CP+L0*(1-P)+Z*(1)+P*(K)+*(1-3) STEP 158
      IF (1VISC) 1315,1360,1315        STEP 160
1315  IF (RAYZ=MU2*LP) 1350,1350,1320  STEP 161
1320  FMAX=AYZ/(LP*MU2)                 STEP 162
210      1330 IF (MULT.MI.O.AN1.AS(1810,K+1).GT.O.75) GO TO 1352 STEP 163
      JMAX=MS-1                          STEP 164
      G=1350 I=1,1+MAX                    STEP 165
      J=1                                  STEP 166
1355  J=J+1                               STEP 167
215      IF (DZ1M(I,J)+MMOZ(K)/MU2-FMAX) 1345,1345,1340 STEP 168
1340  IF (MULT.MI.O.AN1.AS(1810,K+1).GT.O.75) GO TO 1342 STEP 169
      FMAX=DZ1M(I,J)+MMOZ(K)/MU2        STEP 170
      GO TO 1345                          STEP 171
1342  FMAX=DZLFF(J)+MMOZ(K)/MU2        STEP 172
220      1345 IF (J.LI.NDS) GO TO 1335   STEP 173
1350  CONTINUE                             STEP 174
      GO TO 1355                          STEP 175
1352  GO 1354 I=1,NDZ                    STEP 176
225      IF (DZLFF(I)+MMOZ(K)/MU2.GT.FMAX) FMAX=DZLFF(I)+MMOZ(K)/MU2 STEP 177
1355  CONTINUE                             STEP 178
1355  IF (1HUGSH.MI.O) =M1E (6,1375) MMOZ(K),UZ(K),DELY(K),MU2,KATZ,FMA STEP 179
      13,LP                                STEP 180
1375  FMAX=1 (6+MMOZ2,F11.4,7H H2,F11.4,7H DELY2,F11.4,75H MU2,F11.4 STEP 181

230      1,7H MAY2,F11.4,7H FMAX,F11.4,5H CP=F11.4) STEP 182
      M(K)=MMOZ(K)+UZ(K)+DELY(K)/(MU2+FMAX) STEP 183
      IF (1HUGSH.MI.O) =M1E (6,135H) M(K) STEP 184
1350  FMAX=F1 (6+M(F11.4))              STEP 185
1360  M(K)=MUZ(K)+SUN1((CP+ZMA=H(0))/(CP+Z(K)+MU)) STEP 186
      GAMACP/(LP=H(0)/ZMA)              STEP 187
235      MUZ=4.0*ZMA*ZMA/9.0*GAMA=5.0) STEP 188
      GO TO 10000                          STEP 189
1400  CONTINUE                             STEP 190
      ENRPF MESSAGES                       STEP 191
240      4500 MUCASE=2250                  STEP 192
      GO TO 9900                          STEP 193
      MUCASE=750                          STEP 194
      WRITE (67888M) MUCASE                STEP 195
      GO TO 9900                          STEP 196
      MUCASE=1500                          STEP 197
245      8880 FURFAL (1M1,157.0) ***** MUCASE=,157//) STEP 198
      9900 JENDP=3                          STEP 199
10000  M11=MA                             STEP 201
      GOTO 101                             STEP 202

1      FUNCTION ENF(I)                      ENF 2
1      ***** THIS CALCULATES THE ERROR FUNCTION AND ITS COMPLEMENT ***** ENF 3
      DATA EPSIL/1.0E-10,Z/100,CONST/1.1283791674550/ ENF 4
      M11=Z*Z ENF 5
      GO TO 5 ENF 6
      U=13Y ENF 6
      M11=U-1 ENF 7
      5 IF (U) 1,55,15 ENF 8
      10 Z=Z ENF 10
      15 Z=Z ENF 11
      20 IF (Z=5.) 25,25,45 ENF 12
      25 /M11 ENF 13
      15 ENF 14
      ENF 15
      10 ENF 16
      ENF 17
      ENF 18
      ENF 19
      ENF 20
      ENF 21
      ENF 22
      ENF 23
      35 IF (100=EPSIL/M11) 40,30,30 ENF 24
      40 IF (0=EPSIL,0) 50,50,45 ENF 25
      25 ENF 26
      30 IF (0.1) 35,55,65 ENF 27
      55 ENF 28
      GO TO 65 ENF 29
      60 ENF 30
      65 IF (0.05) 70,90,90 ENF 31
      ***** COMPLETE CALCULATION OF ERROR FUNCTION END ***** ENF 32
      ENF 33
      ENF 34
      ENF 35

1      SUBROUTINE TRANS (I1,PP,ARR,MULT,IND)  TRANS 2
1      ** THIS SUBROUTINE CALCULATES THE THERMODYNAMIC TRANSPORT PROPERTIES - TRANS 3
      CONST=402.77) AA(100) ALFA(25,25),ALPHAM ALPHAV 4
      I ALPH 1BETAP AMIX AC11(25) AC(25,100) A 5
      2 ALP(25) Z(25,100) GAMAR(25) GAMAR(25) ZP A 6
      5 ALP(25) EPSU ALM(25) ALM(25) ALM(25) ALM(25) A 7
      6 ALP(25) ALP(25,25),ALFF(25) DELTA DELTA A 8
      7 ALP ALFSS(100) DELS DELS A 9
      8 ALP(25,25) ALP(25,25) DELY(100) DELS DELY(100) A 10
      9 ALP ALM(100) EPCEP A 11
      10 ALP ALM(100) EPCEP A 12
      11 ALP ALM(100) EPCEP A 13
      12 ALP(100) EPCEP A 14

```



```

500 DU 500 J=1,NN
500 CPS(I)=CPS(I)+FLOAT(J-2)*A(I,J)+TI*(J-3)

115 IF (LL.LI. 0) GO TO 550
    CP=CP+CPS(I)+CABAR(I)
    GO TO 600
550 CP=CP+CPS(I)+C(I,ANK)
600 CONTINUE
120 KAY=CM*U/TPK
    IF (IHUGSM.NE.0)
125 IAKITE(6,702) CP, KAY, KHABAR
    IF (LL.LI. 0) GO TO 650
    DIM(I,2)=KAY+ILE/(RHABAR*CP)
125 GO TO 10000
650 DIM(I,2)=KAY+ILE/(HML(ANK)*CP)
    I1 FIKHAT(1M ,CHPUPAAA,1PE9,2,2X,OMPUMIN,19,2,2X,7MODELIX,EG,2,2X
    1 ,3MUP,EG,2,2X,4HMOE,EG,2,2X,2MU,EG,2,2X,7HMHABAR,EG,2)
    1 700 FORMAT(1M ,4HKKK=13,2X,3MLL=13,2X,5HDDIT,1PE9,2,2X,2MU,EG,2,
    1 2X,7HMO(H)=,EG,2,2X,4HMHU(H)=,EG,2,2X,6HIFLAG=,12)
    1 702 FIKHAT(1M ,3HCP=,1PE9,2,2X,4HKAY=,EG,2,2X,7HMHABAR=,EG,2)
10000 KETUPA
    END

SUBROUTINE VISCO
THIS SUBROUTINE CALCULATES THE FLUX CONTRIBUTIONS TO THE
CONSERVATION EQUATIONS
COMMON A(25,7)
1 A(40) ,ALFA(25,25),ALPHAM ,ALPHAP
2 A(40) ,ATOL ,REIAP ,MMIX ,C11(25) ,C(25,40)
3 A(40) ,C12(25) ,C2(25,20) ,CABAR(25) ,CBBAR(25) ,CP
4 A(40) ,CPS(25) ,CPSH ,CSH(25) ,CSH1(25) ,CSTHE(25)
5 A(40) ,XSTH ,D2TH(25,25),DEFF(25) ,DEFF1(25) ,DELTA
6 A(40) ,D11 ,DELSS(40) ,DELS ,DELS1 ,DLS(40)
7 A(40) ,D1H(25,25) ,D12 ,DELT(40) ,D15 ,DUPDT(40)
8 A(40) ,D14 ,LWINDS(40) ,EPCUM
9 A(40) ,EPLSLA ,FXIA(50) ,FSTEP ,FMAX ,GNAD
10 A(40) ,F11 ,H(40) ,HM ,HMJ ,HMH(25)
11 A(40) ,F2PM(25) ,H3PH(25) ,ICONS1 ,ICUUM1 ,IDENT(25)
12 A(40) ,IEMHIV ,IFXIA(50) ,IFLAG ,IAD
13 A(40) ,IPLOC ,ISHOCK ,I1TYPE ,IPD
14 A(40) ,IOIFF ,K ,KAY ,KAYS ,KAY2
15 A(40) ,KLU ,KMAX
16 A(40) ,KUP ,KA ,LL ,LMLATL ,MA
17 A(40) ,MASH ,MUI ,MMAX ,M10 ,MU
18 A(40) ,M2 ,MLS ,M ,M2 ,MASH
19 A(40) ,MHUUM ,MDS ,METER ,MMAX ,MH
20 A(40) ,COMPON VUCASE ,UMEGA(25) ,P11 ,P(40) ,P12
21 A(40) ,P2(40) ,P(50) ,PABAR ,PBHAR
22 A(40) ,PMS ,P15 ,PHI(40) ,PHI(50) ,PHI(50)
23 A(40) ,PMSH ,PMSH
24 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
25 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
26 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
27 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
28 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
29 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
30 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
31 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
32 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
33 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
34 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
35 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
36 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
37 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
38 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
39 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
40 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
41 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
42 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
43 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
44 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
45 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
46 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
47 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
48 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
49 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
50 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
51 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
52 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
53 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
54 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
55 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
56 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
57 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
58 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
59 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
60 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
61 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
62 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
63 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
64 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
65 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
66 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
67 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
68 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
69 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
70 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
71 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
72 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
73 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
74 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
75 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
76 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
77 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
78 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
79 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
80 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
81 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
82 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
83 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
84 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
85 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
86 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
87 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
88 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
89 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
90 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
91 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
92 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
93 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
94 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
95 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
96 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
97 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
98 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
99 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH
100 A(40) ,PMS1 ,PMS1 ,PMSH ,PMSH

```

	900	DUMMHBAK(K-1)*DELTA	VISCO	25
	910	GO TO 1200	VISCO	26
75	1000	IF (LELTA) 900,1100,900	VISCO	27
	1100	CONTINUE	VISCO	28
	1200	MHSPOM=(HBAK(K)*DELTA+TAN(K)*DLS(K)-DUMMHBAK(K-1)*DLS(K-1))	VISCO	29
		*PI*2,0*DELTA	VISCO	31
80	C	KINETIC ENERGY TRANSFER	VISCO	32
		IF (K-2) 1300,1300,1400	VISCO	33
	1300	DSSPIN=HBAK(K)*DELTA+TAN(K)*(UBAK(K)+UBAK(K+1))/2.0*DLS(K)-DUMMHBAK(K-1)*DLS(K-1)	VISCO	34
		*PI*2,0*DELTA	VISCO	35
	1400	*SSPIN=HBAK(K)*DELTA+TAN(K)*(UBAK(K)+UBAK(K+1))/2.0*DLS(K)-DUMMHBAK(K-1)*DLS(K-1)	VISCO	36
85		*AA(K-1)*(UBAK(K)+UBAK(K+1))/2.0*DLS(K-1)	VISCO	38
	C	ENERGY TRANSFER BY DIFFUSION	VISCO	39
	1500	DIFFLX=0	VISCO	40
		DO 1700 I=1,NHS	VISCO	41
		HSP(I)=HSP(I)	VISCO	42
90		H2P(I)=H2P(I)	VISCO	43
		H3P(I)=H3P(I)	VISCO	44
		IF (K+1-NHSA) 1550,1650,1650	VISCO	45
	1550	IF (IFLAG) 1570,1560,1570	VISCO	46
	1560	HFM(I)=CPS(I)+I*(K-1)	VISCO	47
95		GO TO 1650	VISCO	48
	1570	HFM(I)=LPS(I)*.5*(I*(K-1)+I*(K+1))	VISCO	49
	1650	DIFFLX=DIFFLX+.5*HBAK(K)*DELTA+ZJ(I,K)*DLS(K)*(HFM(I)+H2PM(I))	VISCO	50
		*.5*DUMMHBAK(K-1)*DLS(K-1)*(H2PM(I)+H3PM(I))	VISCO	51
100	C	SPECIES TRANSFER	VISCO	52
	1700	HMS(I)=HBAK(K)*DELTA+ZJ(I,K)*DLS(K)-DUMMHBAK(K-1)*DLS(K-1)*PI	VISCO	53
		*2.0*DELTA	VISCO	54
	C	INITIAL ENERGY TRANSFER	VISCO	55
		HMSEN=(DIFFLX+DSSPIN+H2HBAK(K)*DELTA+DLS(K)-DUMMHBAK(K-1)*DLS(K-1))	VISCO	56
105		*K-1)*PI*2.0*DELTA	VISCO	57
		HHEE(K)=ABS(HMSPO)+HHEE(K)	VISCO	58
		IF (IHUSHM(10,0)) GO TO 10000	VISCO	59
		HRITE (6,2001)	VISCO	60
	2001	FORMAT (100,7HNSUM)	VISCO	61
110		DO 2100 I=1,NHS	VISCO	62
		HRITE (6,2002) HMS(I)	VISCO	63
	2100	CONTINUE	VISCO	64
	2002	FORMAT (3F,1PIE11.4)	VISCO	65
		HRITE (6,2003) HMS(0),DIFFLX,DSSPIN,HMSEN	VISCO	66
	2003	FORMAT (100,7HNSUM,1PIE11.4,9H DIFFLX,1PIE11.4,9H DSSPIN,1PI	VISCO	67
115		*E11.4,9H HMSEN,1PIE11.4)	VISCO	68
	10000	RETURN	VISCO	69
		END	VISCO	70
1		SUBROUTINE SLDP(A,N)	SLDP	2
		THIS PROGRAM FINDS THE SOLUTIONS TO A SET OF N SIMULTANEOUS LINEAR	SLDP	3
		EQUATIONS BY USING THE GAUSS-ORDAN REDUCTION ALGORITHM WITH THE	SLDP	4
		DIAGONAL PIVOT STRATEGY	SLDP	5
5		DIMENSION A(26,26),X(26)	SLDP	6
		DO 9 J=1,N	SLDP	7
		IF (ABS(A(J,J)) .GT. 1.E-10) GO TO 5	SLDP	8
		HRITE (6,101) A(J,J)	SLDP	9
		GO TO 101	SLDP	10
10		*F11(N,100)(A(I,J),J=1,N),X(I)	SLDP	11
		DO 100 I=1,N	SLDP	12
	100	FORMAT(1X, 10F12.5)	SLDP	13
	101	FORMAT(22H ***** SMALL PIVOT ,15, E12.5)	SLDP	14
		STOP	SLDP	15
15		*PI*2	SLDP	16
		DO 8 J=1, N	SLDP	17
		A(I,J)=A(I,J)/A(I,I)	SLDP	18
		X(I)=X(I)/A(I,I)	SLDP	19
20		A(I,I)=1.0	SLDP	20
		DO 9 I=1,N	SLDP	21
		IF (1.E-10 .LE. ABS(A(I,I))) GO TO 4	SLDP	22
		DO 8 J=I+1,N	SLDP	23
		A(I,J)=A(I,J)-A(I,I)*A(J,I)	SLDP	24
25		X(I)=X(I)-A(I,I)*X(J)	SLDP	25
		A(I,I)=0.	SLDP	26
		CONTINUE	SLDP	27
	99	CONTINUE	SLDP	28
		RETURN	SLDP	29
		END	SLDP	30
1		SUBROUTINE MURMEL	MURMEL	2
		COMMON A(25,7)	MURMEL	3
	1	*ALFA(25,25),ALPHAM	MURMEL	4
	2	*ALFA(25,25),ALPHAM	MURMEL	5
	3	*ALFA(25,25),ALPHAM	MURMEL	6
5		*ALFA(25,25),ALPHAM	MURMEL	7
	4	*ALFA(25,25),ALPHAM	MURMEL	8
	5	*ALFA(25,25),ALPHAM	MURMEL	9
	6	*ALFA(25,25),ALPHAM	MURMEL	10
	7	*ALFA(25,25),ALPHAM	MURMEL	11
10		*ALFA(25,25),ALPHAM	MURMEL	12
	8	*ALFA(25,25),ALPHAM	MURMEL	13
	9	*ALFA(25,25),ALPHAM	MURMEL	14
	0	*ALFA(25,25),ALPHAM	MURMEL	15
	1	*ALFA(25,25),ALPHAM	MURMEL	16
	2	*ALFA(25,25),ALPHAM	MURMEL	17
15		*ALFA(25,25),ALPHAM	MURMEL	18
	3	*ALFA(25,25),ALPHAM	MURMEL	19
	4	*ALFA(25,25),ALPHAM	MURMEL	20
	5	*ALFA(25,25),ALPHAM	MURMEL	21
	6	*ALFA(25,25),ALPHAM	MURMEL	22
	7	*ALFA(25,25),ALPHAM	MURMEL	23

```

20      8      ,NBOUND      ,NDS      ,NITEM      ,NMAX      ,NN      A      20
COMMON NUCCASE      ,OMEGA(25) ,P11      ,P(40)      ,P12      A      21
1      ,P2(40)      ,PB(50)      ,PABAR      ,PBBAR      A      22
2      ,PBS      ,P15      ,P1(40)      ,PHI(50)      ,PHIB(50)      A      23
3      ,PHISH1      ,PHSTRM      ,PHI(50)      ,PHI      A      24
25      4      ,PH12(40)      ,PH13      ,P1      ,PH(25)      ,PSH      A      25
5      ,PSH1      ,PS1      ,PSTREM      ,PH(50)      A      26
6      ,U11      ,Q(40)      ,QXTR1      ,QXTR2      A      27
7      ,Gn(50)      ,R11      ,R(40)      ,RB(50)      A      28
8      ,RBS      ,R13      ,RBAR(40)      ,R14      ,R2(40)      A      29
30      9      ,RCUN      ,R15      ,RE(40)      ,RESM      ,R16      A      30
      ,RHU(40)      ,R17      ,RHU2(40)      ,RHS(25)      ,RMSEN      A      31
1      ,RHSUM      ,RHABAR      ,RHBBAR      ,RHU      ,RSM      A      32
2      ,RSTHEM      ,RV      ,R19      ,RM(100)      ,S      A      33
3      ,SB(50)      ,SC(25)      ,S      ,SM(50)      A      34
35      4      ,S13      ,SX(40)      ,T11      ,T(40)      ,T12      A      35
5      ,T2(40)      ,TABAR      ,TUBAR      ,T(25)      ,T14      A      36
6      ,TAA(40)      ,TXTR1      ,TXTR2      ,TUL      ,TSM      A      37
7      ,TSH1      ,TSTEM      ,Tm(50)      ,Tm3      A      38
8      ,TS      ,U11      ,U(40)      ,U12      ,U2(40)      A      39
40      COMMON UXTR1      ,UXTR2      ,U14      ,UBAR(40)      ,USH      A      40
1      ,USH1      ,USTREM      ,U14      ,U1(50)      ,USH      A      41
2      ,X11      ,X(40)      ,X12      ,X2(40)      ,UNS      A      42
3      ,XB(50)      ,XBS      ,XBAR(25)      ,XBHAR(25)      ,X15      A      43
4      ,XS(25,40)      ,XSM      ,XSTEM      ,XM(50)      A      44
45      5      ,ZMAX(40)      ,Y11      ,Y(40)      ,Y12      A      45
6      ,Y2(40)      ,YABAR      ,YBBAR      ,ZA(25)      ,Z11(25)      A      46
7      ,ZJ(25,40)      ,ZMH      ,SZX      ,ZSM      A      47
8      ,FX(2,40)      ,FX(2,40)      ,FPHI(2,40)      ,FP(2,40)      ,FT(2,40)      A      48
9      ,FU(2,40)      ,INDL(2,40)      ,INDH(2,40)      ,FC(2,40,25),CANOL      A      49
50      ,H      ,H      A      50
COMMON/ALL/ TALL
LONLW/BLQU/ REJ1,THEI1,CF1,UF1,CFCCF1,M12,UISM,THEIAC(2),DELT1,
DELTA2,CF2,WAALL,TAUALL,ETA(140),TETC(140),UBUFI(140),YOFI(140),
DJMPSJOM (1)INT(140),H2INT(140)
55      REAL      KAT      ,KAYS(25)      ,KAY2      ,KN      ,MUOT(40),
      ,MA(40)      ,MU      ,MUS(25)      ,MUO(25)      ,MU2      BOUNDL      8
      ,MA(25)      ,MH2      ,MASH      ,MASH1      ,MNSM      BOUNDL      9
      ,MASH1      ,MNSM      BOUNDL      10
      ,MAP      BOUNDL      11
60      REAL      MUEHEF      BOUNDL      12
      IF (LL) 90,90,91      BOUNDL      13
90      UNALL=0.0      BOUNDL      14
      TAU=ALX=0      BOUNDL      15
      LU TU 500      BOUNDL      16
65      91      CGATTIUD      BOUNDL      17
      MUEHEF=91.E=06      BOUNDL      18
      TE=T2(KMAX)      BOUNDL      19
      UE=U2(KMAX)      BOUNDL      20
      MHUF=MHU2(KMAX)      BOUNDL      21
      VE=V2(KMAX)      BOUNDL      22
70      XE=X2(KMAX)      BOUNDL      23
      PAU=H2(KMAX)      BOUNDL      24
      SE=SX(KMAX)      BOUNDL      25
      ANE=1.2      BOUNDL      26
75      AMU=AMUFREF*SGHT(TE/1034.)      BOUNDL      27
      AMU=AMUFREF*SGHT(TAALL/1034.)      BOUNDL      28
      HEXI=UE*SE*AMU/AMUF      BOUNDL      29
      THEI1=.036*SE/(REX1**2)      BOUNDL      30
      DELT1=10.286*THEI1      BOUNDL      31
      C....CALC SKIN FRICTION X=S EU., FLAT PLATE, ZERU PRES GRAD      BOUNDL      32
      REI1=UE*THEI1*AMU/AMUE      BOUNDL      33
      CFI=1./(.17,UN*(ALU610*(REI1)**2+25.11*ALU610*(REI1)**0.012)      BOUNDL      34
80      C....CALC FRICTION VELOCITY      BOUNDL      35
      RC=RE2.1      BOUNDL      36
      C.....REPLACE ZHAR(1) WITH CONST MOLECULAR WGT FUN NOM      BOUNDL      37
      NHU=TE*AMHOL/TALL      BOUNDL      38
      UFI=(.5*AMHOL*UE*HECF1/NHUA)**.5      BOUNDL      39
90      C....GENERATE TABLE OF TET VS U/UF      BOUNDL      40
      CCONSTROU UE=.4973RE=UK      BOUNDL      41
      TET=RE*U/US1      BOUNDL      42
      ACR=(TET/TE-1.)      BOUNDL      43
      CCR=TALL/TE      BOUNDL      44
      H=TIU-TE/TE-CC      BOUNDL      45
95      C....FINEN RESULC ILLU FUN VETIA=1      BOUNDL      46
      I1=50      BOUNDL      47
      I2=100      BOUNDL      48
      I3=1101      BOUNDL      49
      DO 100 J=1,11      BOUNDL      50
100      FIA(J)=J/(11*10.)      BOUNDL      51
      DO 101 J=13,12      BOUNDL      52
101      FIA(J)=FIA(11)*(J-11)/100.      BOUNDL      53
      UETA1=FIA(1)      BOUNDL      54
      UETA2=FIA(13)-FIA(11)      BOUNDL      55
      DO 102 J=1,12      BOUNDL      56
102      TETC(J)=1./((AL*(FIA(J)+FIA(J))+UETA(J)**ANE)+CL)**.5      BOUNDL      57
105      C....CALC (MUS CF/IN(CAP CF      BOUNDL      58
      SUM1=0      BOUNDL      59
      DO 110 J=1,11      BOUNDL      60
110      SUM1=SUM1+TETC(J)*DETA1      BOUNDL      61
      DO 111 J=13,12      BOUNDL      62
111      SUM1=SUM1+TETC(J)*DETA2      BOUNDL      63
      CFCF=SUM1*SUM1      BOUNDL      64
      CFC=FCFCF1*CF1      BOUNDL      65
      C....CALL REAL TRANSFER AND SHEAR STRESS      BOUNDL      66
      STAPX,35=CF      BOUNDL      67

```

115	UNALL=STANARMOE+UE+CP*110TE=TWALL	BOUNDL	68
	TAUNAL=,SARMOE+UE+CP	BOUNDL	69
	C.....RETURN POINT FROM FIRST PASS	BOUNDL	70
	IF (IFLAG) 112,500,112	BOUNDL	71
	C.....CALC U INCOMP/U FRICT INCOMP	BOUNDL	72
120	112 UBUF(1)=SGN(2./CFC)*TEFC(1)*DELTA1	BOUNDL	73
	DO 120 J=2,11	BOUNDL	74
	120 UBUF(J)=UBUF(J-1)+SQRT(2./CFC)*TEFC(J)*DELTA1	BOUNDL	75
	DO 121 J=13,12	BOUNDL	76
	121 UBUF(J)=UBUF(J-1)+SQRT(2./CFC)*TEFC(J)*DELTA2	BOUNDL	77
125	C.....CALL COMRESPONDING Y/DELTA, INCOMP	BOUNDL	78
	FACT=1./EXP((UBUF(12)-5.05)/2.5)	BOUNDL	79
	DO 132 J=1,12	BOUNDL	80
	IF (UBUF(J).LT.5.0) GO TO 130	BOUNDL	81
	IF (UBUF(J).LT.13.96) GO TO 131	BOUNDL	82
130	YDEL(J)=EXP((UBUF(J)-5.05)/2.5)*FACT	BOUNDL	83
	GO TO 132	BOUNDL	84
	130 YDEL(J)=UBUF(J)*FACT	BOUNDL	85
	GO TO 132	BOUNDL	86
	131 YDEL(J)=EXP((UBUF(J)+5.05)/5.0)*FACT	BOUNDL	87
135	132 CONTINUE	BOUNDL	88
	C.....NOW HAVE Y/DELTA INCOMP = Y/DELTA COMP VS U/UE & TE/T:COMP	BOUNDL	89
	C.....CALL THEIA/DELTA & DISP/DELTA BY TRAPEZOIDAL INTEGRATION	BOUNDL	90
	C.....IF U/UE(1)=C/UE; (1-V/UE) & Y/DELTA	BOUNDL	91
140	SUMD2=0.0	BOUNDL	92
	SUMD1=0.0	BOUNDL	93
	IY=12-1	BOUNDL	94
	DO 140 J=1,12	BOUNDL	95
	O1INT(J)=1.-TEA(J)	BOUNDL	96
	O2INT(J)=O1INT(J)*TEA(J)	BOUNDL	97
145	DO 141 J=1,12	BOUNDL	98
	SUMD1=SUMD1+(YDEL(J+1)-YDEL(J))*S*(O1INT(J)+O1INT(J+1))	BOUNDL	99
	SUMD2=SUMD2+(YDEL(J+1)-YDEL(J))*S*(O2INT(J)+O2INT(J+1))	BOUNDL	100
	M12=SUMD1/SUMD2	BOUNDL	101
150	C.....CALL METAC FROM MOMENTUM INTEGRAL	BOUNDL	102
	IF (LLEQ.0) GO TO 142	BOUNDL	103
	METAC(1)=METAC(2)	BOUNDL	104
	GO TO 143	BOUNDL	105
	142 METAC(1)=.1	BOUNDL	106
155	143 DELX=SQRT((X2(KMAX)-X(KMAX))**2+(R2(KMAX)-R(KMAX))**2)	BOUNDL	107
	DUXX=(UE-U(KMAX))/DELX	BOUNDL	108
	DDXX=(HMO-H(KMAX))/DELX	BOUNDL	109
	DDXX=(HMO-H(KMAX))/DELX	BOUNDL	110
	AD=(M12+.2)/UE+DDXX+DRJDX/RMOE+DRDX/RAD	BOUNDL	111
160	METAC(2)=(LFC/2.+DELX*METAC(1))/(1.+AD*DELX)	BOUNDL	112
	DELTA2=METAC(2)/SUMD2	BOUNDL	113
	DISP=SUMD1+DELTA2	BOUNDL	114
	DO 150 J=1,12	BOUNDL	115
	TEFC(J)=1./TEFC(J)*TEFC(J)	BOUNDL	116
165	500 RETURN	BOUNDL	117
	END	BOUNDL	118
1	SUBROUTINE STABLE	STABLE	2
	C THIS SUBROUTINE DETERMINES STABLE STEPPING DISTANCE AND PUNCHES	STABLE	3
	C OUTPUT DATA WHEN CALLED FOR	STABLE	4
	C	STABLE	5
5	COMMON A(25,7), AA(40), ALFA(25,25), ALPHAN, ALPHAP	A	2
	1 ATUL, ATIAM, AMIX, C11(25), C(25,40)	A	3
	2 C12(25), C2(25,40), CAHAK(25), CUHAM(25), CP	A	4
	3 CFS(25), CFSH, CSH(25), CSN1(25), CSTREH(25)	A	5
	4 CSTRM, C2IM(25,25), CFFF(25), CDELTA	A	6
10	5 O11, OELSS(40), OELLS, OELLSU, OLS(40)	A	7
	6 O1M(25,25), O12, OELLY(40), O13, OPMY(40)	A	8
	7 O14, OPMIDS(40), OEPCH, OMAX, OMO	A	9
	8 OPLCN, OEXTRA(50), OSTEP, OMAX, OMO	A	10
	9 O11, OMO, OMH, OMJ, OPM(25)	A	11
15	4 O2PM(25), O3PM(25), ICUNST, ICOUNT, IOEM(25)	A	12
	1 IERADN, IEXTRA(50), IFLAG, IKIND	A	13
	2 IFRIC, ISMCK, ITYPE, IPI	A	14
	3 IOIFF, K, KAY, KAYS, KAY2	A	15
	4 KLU, KMAX, LL, LPLANE, MA	A	16
20	5 KUP, KX, LLM, MUO, MU	A	17
	6 MASH, MDUI, MMAX, MUO, MSH	A	18
	7 M02, MUS, M, M02, MSH	A	19
	8 M03(40), MLS, MIFR, MMAX, MN	A	20
25	COMMON YULCAST, YLEGA(25), P11, P(40), P12	A	21
	1 P2(40), P3(50), PABAN, PBBAN	A	22
	2 PHS, P15, P1(40), P1M(50), P1R(50)	A	23
	3 P1SH1, P1STRM, P1, P1M(50), P1R	A	24
	4 P12(40), P1HS, P1, P1(25), P1S	A	25
	5 P1M1, P15, P1STRM, P1M(50), P1R(50)	A	26
30	6 O11, O(40), OXTR1, OXTR2	A	27
	7 OR(50), R11, R(40), RB(50)	A	28
	8 RBS, R13, RBAR(40), R14, R2(40)	A	29
	9 RCUN, R15, RE(40), RESH, R16	A	30
	4 RMO(40), R17, RMO2(40), RNS(25), RNSFN	A	31
35	1 RNSHM, RMAHAK, RMBHAK, RU, RSM	A	32
	2 RSTRM, RY, R19, RNS(10), RS	A	33
	3 S, SH(50), SC(25), SM(50)	A	34
	4 S13, SX(40), S11, S(40)	A	35
	5 TP(40), TABAR, TBBAR, T(25), T14	A	36
40	6 TA(40), TXTR1, TXTR2, TUL, TSM	A	37
	7 TSH1, TSTRM, TR(50), TAS	A	38
	8 T13, T11, T(40), T2(40)	A	39
	COMMON UXTM1, UXTM2, U14, UHAM(40), USH	A	40
	1 USM1, USTRM, U14, UAM(50), UMS	A	41
45	2 X11, X(40), X12, X2(40)	A	42
	3 XAM(50), XHS, XAHAK(25), XHAM(25), X15	A	43
	4 X2(25,40), XSM, XSTRM, XAM(50)	A	44
	5 X2AM(40), Y11, Y(40), Y12	A	45
	6 Y(40), YBBAN, ZA(25), Z11(25)	A	46
50	7 ZJ(25,40), ZNA, SZX, RZSH	A	47
	8 FX(2,40), FPHI(2,40), FFP(2,40), FFI(2,40)	A	48

	9	,FU(2,40)	,INDL(2,40):	,INCR(2,40)	,FC(2,40,25),CARD1	A	49
	9	,M	,N			A	50
55	9	REAL KAY	,KAYS(25)	,KAYZ	,KN	,MDDT(40)	STABLE 7
	9	,MA(40)	,MU	,MUB(25)	,MUD(25)	,MUZ	STABLE 8
	9	,Mn(25)	,Mn2	,MASH	,MASH1	,MASH	STABLE 9
	9	,MAP					STABLE 10
		COMMON/PI1/AMOMV(8), AMOMV(8), ENPE1(8), HEP(40,8), V(40,8), n(40,8),					STABLE 11
		1V2(40,8), N2(40,8), TRODY(8), TP(40,8), TP2(40,8), KP(8),					STABLE 12
60		2KMP(40,8), RMP2(40,8), ENP2(8), NBL(8), UM(40)					STABLE 13
		3,VULP(8), WTP(8), HC(8), DENG(40,8), ICOND(40,8)					STABLE 14
		LC(MN/PI2/FFF, FFL, GAVA, UMI, PU, NPG, KMX, DNGSP(8)					STABLE 15
		CURVON/PI3/LL, CS, IFS, RMSB, NT, MIBAN, SIG, EP, IPART, IAIN, TN(40)					STABLE 16
		1, SUPP, SUPPV, SUPPE					STABLE 17
65		COMMON/IBUGS1/ IBUGSM					STABLE 18
		DIMENSION INDEX2(120)					STABLE 19
		COMMON/INIL/XLX, JKI, HEX					STABLE 20
		COMMON/XYZ/XYZ, JXYZ					STABLE 21
		DIMENSION CA(30)					STABLE 22
70		XYZ#20					STABLE 23
		JXYZ#0					STABLE 24
		IF (LL) 10,10,40					STABLE 25
		INDEXING					STABLE 26
75		10 IF X1MA(6)#0					STABLE 27
		IN#0					STABLE 28
		DO 20 I=1, NMAX					STABLE 29
		20 INDEX2(I)=1					STABLE 30
		NMAX=NMAX+1					STABLE 31
		GO 30 I=NMAX, 120					STABLE 32
80		50 INDEX2(I)=0					STABLE 33
		40 K#2					STABLE 34
		I#2					STABLE 35
		DELS=1.0E10					STABLE 36
85		45 IF (IEXTRA(5)=LPLANE) 50,50,90					STABLE 37
		50 IF (INDEX2(I)) 75,90,90					STABLE 38
		75 I=I+1					STABLE 39
		GO TO 50					STABLE 40
		VISCOUS STABILITY CRITERION					STABLE 41
90		90 DEL1=DELY(K)*ANE(K)/2.0					STABLE 42
		IF (IBUGSM, NE, 0) WRITE (6,95) DEL1					STABLE 43
		95 FORMAT (6H DEL1=,E11.4)					STABLE 44
		INITIAL STABILITY CRITERION					STABLE 45
		IF (MA(K)-1.0-EPSLN) 100,100,600					STABLE 46
95		100 WRITE (6,200) K					STABLE 47
		200 FORMAT (25HIFLOW IS SUBSONIC IN TUBE, I5)					STABLE 48
		CALL PUMP(A(1,1), N, I)					STABLE 49
		CALL EXIT					STABLE 50
		600 DEL2=.5*DELY(K)*(MA(K)**2-1.0)**(.5)					STABLE 51
		500 FORMAT (6H DEL2=,E11.4)					STABLE 52
100		IF (IBUGSM, NE, 0) WRITE (6,500) DEL2					STABLE 53
		IF (DEL1.EQ.0.0) GO TO 590					STABLE 54
		DELSS(K)=ALPHA/(1.0/DEL1+1.0/DEL2)					STABLE 55
		GO TO 595					STABLE 56
		590 DELSS(K)=ALPHA*(DEL2					STABLE 57
105		545 CONTINUE					STABLE 58
		COMBINING SMALL TUBES					STABLE 59
		IF (LL) 608,602,602					STABLE 60
		602 FLELL					STABLE 61
		FPLANE=LPLANE					STABLE 62
110		FMULT=FL/FPLANE					STABLE 63
		GSTEP=FSSTEP					STABLE 64
		IF (EXTRA(4), LT, 1.E-5) GO TO 604					STABLE 65
		FSSTEP=GSTEP*(GSTEP-EXTRA(3))*(1.-FMULT)/EXTRA(4)					STABLE 66
		GO TO 606					STABLE 67
115		604 FSSTEP=GSTEP					STABLE 68
		608 L#K					STABLE 69
		CALL COMB(L)					STABLE 70
		FSSTEP=GSTEP					STABLE 71
		IF (L#K) 610,604,602					STABLE 72
120		610 K=L					STABLE 73
		IF (IEXTRA(5)=LPLANE) 630,630,90					STABLE 74
		620 K=L-1					STABLE 75
		I=I+1					STABLE 76
		IF (IEXTRA(5)=LPLANE) 630,630,90					STABLE 77
125		630 J=1					STABLE 78
		640 J=J+1					STABLE 79
		IF (INDEX2(IJ)) 640,650,650					STABLE 80
		650 INDEX2(IJ)=1					STABLE 81
		IEXTRA(K)=IEXTRA(6)+1					STABLE 82
130		GO TO 50					STABLE 83
		ANGLE CHANGE LIMITATION					STABLE 84
135		640 ULNA=(SIN(.5*(PI*(K)+PI*(K-1)))/(1.5*(K(K)+K(K-1)))*DELTA+(PI(K)-				STABLE 85	
		PI(K-1))/E(LY(K))*DELSS(K)					STABLE 86
		IF (IBUGSM, NE, 0) WRITE (6,695) ULNA, ATUL					STABLE 87
		695 FORMAT (6H ULNA=,E11.4, 7H ATUL=,E11.4)					STABLE 88
		IF (ABS(ULNA)-ATUL) MUO, 900, 700					STABLE 89
		700 DELSS(K)=DELSS(K)+ATUL/ABS(ULNA)					STABLE 90
		640 IF (DELSS(K)=DELS) 900,1000,1000					STABLE 91
		900 DELSS(DELSS(K))					STABLE 92
140		EXTRA(7)=K					STABLE 93
		1000 IF (IEXTRA(5)=LPLANE) 1025,1025,1050					STABLE 94
		1025 I=I+2(I)=K					STABLE 95
		IEXTRA(K)=1					STABLE 96
		I=I+1					STABLE 97
145		1050 K=K+1					STABLE 98
		IF (K=NMAX) 65,65,1055					STABLE 99
		1055 IF (KMAX=32) 1057,1057,1060					STABLE 100

	C	(COMBINING SMALLEST TUBE AREA NUMBR OF TUBES GETS TOO LARGE	STABLE	101
150	1057	TUBE=FSIIP	STABLE	102
		IF (KMAX=ITUM) 1400,1400,1060	STABLE	103
	1060	L=EXTNA(7)	STABLE	104
		NXL	STABLE	105
		STONE=GRAD	STABLE	106
		STONE=ZFSTF	STABLE	107
155		STONE=ZY(NMAX)	STABLE	108
		GNAL=10,000	STABLE	109
		FSTF=10,000	STABLE	110
		Y(NMAX)=10,000	STABLE	111
		CALL COMHU(L)	STABLE	112
160		GNAL=STONE1	STABLE	113
		FSTF=STONE2	STABLE	114
		Y(NMAX)=STONE3	STABLE	115
		N=NMAX+1	STABLE	116
	1063	IF (L=EXTNA(5)-LPLANE) 1065,1065,2000	STABLE	117
165	1065	IF (L=EXTNA(7)) 1065,1060,1060	STABLE	118
	1060	L=EXTNA(N)	STABLE	119
		GO TO 1090	STABLE	120
	1085	L=EXTNA(6)-1	STABLE	121
170	1090	L=EXTNA(6)+EXTNA(6)+1	STABLE	122
		JMAX=NMAX+EXTNA(6)	STABLE	123
		J=J+1	STABLE	124
		J=JMAX	STABLE	125
175	1095	INDEX2(J)=INDEX2(J-1)	STABLE	126
		J=J-1	STABLE	127
		IF (J=JMAX) 1097,1095,1095	STABLE	128
	1097	INDEX2(J)=1	STABLE	129
	1400	IF (L=EXTNA(5),G,I,PLANE) GO TO 2000	STABLE	130
		IF (X2(2),L1, XL) GO TO 2000	STABLE	131
		IF (X2(2),L6, XL) GO TO 1500	STABLE	132
180		IF (INDEX2(L),EXTNA(5)) 2000,1500,2000	STABLE	133
		FUNCTIONG OUTPUT (ARGUS)	STABLE	134
	1500	I=1	STABLE	135
		IF (I=1,0,0)	STABLE	136
185		NR=1/(7,1000) X(I),H(I),PH(I),P(I),T(I),U(I),I,I	STABLE	137
		IF (I=1,0,0,0,0,0,0,0)	STABLE	138
		NR=1/(10,1000) X(I),R(I),PH(I),P(I),T(I),U(I),I,I	STABLE	139
	1600	FORMAT(6E12,5,2I6)	STABLE	140
		IF (I=1,0,0)	STABLE	141
		NR=1/(7,1700) ((J,I),J=1,NDS)	STABLE	142
190		IF (I=1,0,0,0,0,0,0,0)	STABLE	143
		NR=1/(10,1700) ((J,I),J=1,NDS)	STABLE	144
	1700	FORMAT(1E10,3)	STABLE	145
		NR=NRMAX+EXTNA(6)	STABLE	146
		IF (I=1,0,0) GO TO 1802	STABLE	147
195		DO 1701 I=2,NR	STABLE	148
		Y=0,0	STABLE	149
		X=0,0	STABLE	150
		NR=0,0	STABLE	151
200		NR=0,0	STABLE	152
		NR=0,0	STABLE	153
		NR=0,0	STABLE	154
		I=I+1	STABLE	155
		GO 51 J=1,NR	STABLE	156
		XMP=XP+RMP2(I,J)+R2(I,J)+R2	STABLE	157
205		XMP=XMP+RMP2(I,J)+R2(I,J)+R2(I,J)	STABLE	158
		XMP=XMP+RMP2(I,J)+R2(I,J)+R2	STABLE	159
		U=U+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	160
		NR=NR+RMP2(I,J)	STABLE	161
210		IF (XMP+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	162
		UP1=(XMP+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	163
		VP1=(U+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	164
		VP=VP1+(VP1+VP1+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	165
		UP=UP1+(UP1+VP1+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	166
215		RMP2=(XMP+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	167
		RMP2=(RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J))	STABLE	168
		U=U+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	169
		VP=VP1+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	170
		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	171
		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	172
220		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	173
		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	174
	202	FORMAT(1E12,5)	STABLE	175
		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	176
		NR=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	177
225		GO 112 J=NR+RMP2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)+R2(I,J)	STABLE	178
		J=J+NR	STABLE	179
	102	(A(J)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ))	STABLE	180
		GO 105 J=1,NDS	STABLE	181
230		IF (A(J)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ))	STABLE	182
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	183
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	184
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	185
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	186
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	187
235		GO 105 J=1,NDS	STABLE	188
		GO 105 J=1,NDS	STABLE	189
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	190
	105	NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	191
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	192
240		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	193
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	194
		NR=NR+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)+RMP2(I-1,JJ)	STABLE	195

	106	UNPROMPTED CS=CA(J)	STABLE	196	
		CONTINUE	STABLE	197	
245		IF (MS(G), I1, I2) GO TO 107	STABLE	198	
		CONTINUE	STABLE	199	
		TA=IA-U/DG	STABLE	200	
	104	CONTINUE	STABLE	201	
		WRITE (N, 106) I1, IA, MPP/DGM, UMP	STABLE	202	
250		104 FURNISH(SOME) INFORMATION DOES NOT CONVERGE TO CORRECT TEMPERATURE	STABLE	203	
		1 10, 2, 3, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52, 53, 54, 55, 56, 57, 58, 59, 60, 61, 62, 63, 64, 65, 66, 67, 68, 69, 70, 71, 72, 73, 74, 75, 76, 77, 78, 79, 80, 81, 82, 83, 84, 85, 86, 87, 88, 89, 90, 91, 92, 93, 94, 95, 96, 97, 98, 99, 100	STABLE	204	
		2 10, 5	STABLE	205	
	107	CONTINUE	STABLE	206	
		Z=Z+Q	STABLE	207	
255		GO TO 9, J=1, I=5	STABLE	208	
	109	Z=Z+Q*(A(J)/Y(J))	STABLE	209	
		CONTINUE	STABLE	210	
		WRITE (I1, 100) X(I), Y(I), P(I), U(I), A(I), U(I), I, I	STABLE	211	
		WRITE (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	212	
260		101 (CONTINUE)	STABLE	213	
		GO TO 100	STABLE	214	
	1002	(I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	215	
		IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	216	
	1000	J=J+1	STABLE	217	
265		1025 J=J+1	STABLE	218	
		1025 J=J+1	STABLE	219	
		IF (I1) 1025, 1025, 1000	STABLE	220	
	1050	1025 J=J+1	STABLE	221	
	1900	IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	222	
270		WRITE (I1, 100) (X(I), Y(I), U(I), P(I), I(1), I(2), I(3), I(4), I(5), I(6), I(7), I(8), I(9), I(10), I(11), I(12), I(13), I(14), I(15), I(16), I(17), I(18), I(19), I(20), I(21), I(22), I(23), I(24), I(25), I(26), I(27), I(28), I(29), I(30), I(31), I(32), I(33), I(34), I(35), I(36), I(37), I(38), I(39), I(40), I(41), I(42), I(43), I(44), I(45), I(46), I(47), I(48), I(49), I(50), I(51), I(52), I(53), I(54), I(55), I(56), I(57), I(58), I(59), I(60), I(61), I(62), I(63), I(64), I(65), I(66), I(67), I(68), I(69), I(70), I(71), I(72), I(73), I(74), I(75), I(76), I(77), I(78), I(79), I(80), I(81), I(82), I(83), I(84), I(85), I(86), I(87), I(88), I(89), I(90), I(91), I(92), I(93), I(94), I(95), I(96), I(97), I(98), I(99), I(100))	STABLE	223	
		IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	224	
		WRITE (I1, 100) (X(I), Y(I), U(I), P(I), I(1), I(2), I(3), I(4), I(5), I(6), I(7), I(8), I(9), I(10), I(11), I(12), I(13), I(14), I(15), I(16), I(17), I(18), I(19), I(20), I(21), I(22), I(23), I(24), I(25), I(26), I(27), I(28), I(29), I(30), I(31), I(32), I(33), I(34), I(35), I(36), I(37), I(38), I(39), I(40), I(41), I(42), I(43), I(44), I(45), I(46), I(47), I(48), I(49), I(50), I(51), I(52), I(53), I(54), I(55), I(56), I(57), I(58), I(59), I(60), I(61), I(62), I(63), I(64), I(65), I(66), I(67), I(68), I(69), I(70), I(71), I(72), I(73), I(74), I(75), I(76), I(77), I(78), I(79), I(80), I(81), I(82), I(83), I(84), I(85), I(86), I(87), I(88), I(89), I(90), I(91), I(92), I(93), I(94), I(95), I(96), I(97), I(98), I(99), I(100))	STABLE	225	
		IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	226	
275		WRITE (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	227	
		IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	228	
		WRITE (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	229	
	1950	CONTINUE	STABLE	230	
	1960	IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	231	
		GO TO 1970 (X(I), Y(I), U(I), P(I))	STABLE	232	
280		1970 IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	233	
		CONTINUE	STABLE	234	
		GO TO 1980 (X(I), Y(I), U(I), P(I))	STABLE	235	
	1980	IF (I1, 100) (X(I), Y(I), U(I), P(I))	STABLE	236	
285		2000 IF (I1, 100) (X(I), Y(I), U(I), P(I)) RETURN	STABLE	237	
		GO TO 2200 (X(I), Y(I), U(I), P(I))	STABLE	238	
		WRITE (N, 2100) U(LSS(I))	STABLE	239	
290		2100 FURNISH (I1) U(LSS(I), I(1), I(2))	STABLE	240	
		2200 CONTINUE	STABLE	241	
		GO TO 2	STABLE	242	
290		END	STABLE	243	
1		SOME OF THE FILES	PUITN	2	
		FILES SOME OF THE FILES IN THE INITIALIZES ALL DATA LAUNCH THAT	PUITN	3	
		MAINTAIN THE FILES AT THE INITIALIZES ALL DATA LAUNCH THAT	PUITN	4	
		CONVERSION A(25,7)	ALPHA(25,25)	ALPHAB	5
5		1 A(10)	BE(10)	CM(25,40)	6
		2 C(2,25)	C(25,40)	CAHAR(25)	7
		3 CP(2,5)	CP(25)	CSH(25)	8
		4 S(10)	S(25,25)	UFF(25)	9
		5 U(1)	U(1,1)	U(1,1)	10
10		6 U(1,25,25)	U(1,2)	U(1,2)	11
		7 U(1)	U(1,1)	U(1,1)	12
		8 U(1)	U(1,1)	U(1,1)	13
		9 U(1)	U(1,1)	U(1,1)	14
15		10 U(1)	U(1,1)	U(1,1)	15
		11 U(1)	U(1,1)	U(1,1)	16
		12 U(1)	U(1,1)	U(1,1)	17
		13 U(1)	U(1,1)	U(1,1)	18
20		14 U(1)	U(1,1)	U(1,1)	19
		15 U(1)	U(1,1)	U(1,1)	20
		16 U(1)	U(1,1)	U(1,1)	21
		17 U(1)	U(1,1)	U(1,1)	22
		18 U(1)	U(1,1)	U(1,1)	23
		19 U(1)	U(1,1)	U(1,1)	24
		20 U(1)	U(1,1)	U(1,1)	25
25		21 U(1)	U(1,1)	U(1,1)	26
		22 U(1)	U(1,1)	U(1,1)	27
		23 U(1)	U(1,1)	U(1,1)	28
		24 U(1)	U(1,1)	U(1,1)	29
30		25 U(1)	U(1,1)	U(1,1)	30
		26 U(1)	U(1,1)	U(1,1)	31
		27 U(1)	U(1,1)	U(1,1)	32
		28 U(1)	U(1,1)	U(1,1)	33
		29 U(1)	U(1,1)	U(1,1)	34
		30 U(1)	U(1,1)	U(1,1)	35
35		31 U(1)	U(1,1)	U(1,1)	36
		32 U(1)	U(1,1)	U(1,1)	37
		33 U(1)	U(1,1)	U(1,1)	38
		34 U(1)	U(1,1)	U(1,1)	39
		35 U(1)	U(1,1)	U(1,1)	40
		36 U(1)	U(1,1)	U(1,1)	41
		37 U(1)	U(1,1)	U(1,1)	42
		38 U(1)	U(1,1)	U(1,1)	43
		39 U(1)	U(1,1)	U(1,1)	44
		40 U(1)	U(1,1)	U(1,1)	45
45		41 U(1)	U(1,1)	U(1,1)	46
		42 U(1)	U(1,1)	U(1,1)	47
		43 U(1)	U(1,1)	U(1,1)	48
		44 U(1)	U(1,1)	U(1,1)	49
		45 U(1)	U(1,1)	U(1,1)	50
50		46 U(1)	U(1,1)	U(1,1)	51
		47 U(1)	U(1,1)	U(1,1)	52
		48 U(1)	U(1,1)	U(1,1)	53
		49 U(1)	U(1,1)	U(1,1)	54
		50 U(1)	U(1,1)	U(1,1)	55

	0	,FU(2,40)	,INCL(2,40)	,INDM(2,40)	,FC(2,40,25),CARD1	A	49
	5	,M	,M			A	50
	10	,REAL	,RAY1	,RAY2	,RAY	,MUI(40)	POTIN
55	15	,MA(40)	,MU	,MUS(25)	,MUO(25)	,MUZ	POTIN
	20	,MA(25)	,M2	,MASH	,MASH1	,M2SH	POTIN
	25	,M2					POTIN
	30	CUMPL,P11/P2(40),	X2UM(40),	ENP1(40),	NEP(40,5),	V(40,40),	POTIN
	35						POTIN
	40	IV2(40,5),	A2(40,5),	IRNDY(8),	IP(40,4),	IP2(40,4),	POTIN
	45	IRNDY(4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	50	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	55	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	60	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	65	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	70	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	75	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	80	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	85	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	90	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	95	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	100	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	105	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	110	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	115	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	120	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	125	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	130	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	135	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	140	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN
	145	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	IRNDY(40,4),	POTIN

	REAL(5,000) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	POTIN	100
	PHI(1) (4,000) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	POTIN	101
150	5502 PHIE(n,n1) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	POTIN	102
	LU 5501 J=1,0	POTIN	103
	AM(J)EA(1,JA)	POTIN	104
	5501 CONTINUE	POTIN	105
	A(1,1)=AM(5)*1.0E+06/MA(1)	POTIN	106
	A(1,2)=AR(6)/MA(1)+1000.0	POTIN	107
155	A(1,3)=AR(11)/MA(1)	POTIN	108
	A(1,4)=AB(2)/12.0E+06/MA(1)	POTIN	109
	A(1,5)=AN(3)/(5.0E+06/MA(1))	POTIN	110
	A(1,6)=AN(4)/10.0E+06/MA(1)	POTIN	111
	LSI(1)=CSI(1)/A(1,1)	POTIN	112
160	550 CONTINUE	POTIN	113
	LU 540 J=0,NA	POTIN	114
	LU 560 I=1,ABS	POTIN	115
	IF(A(1,1)) 330,351,330	POTIN	116
165	351 IF(A(1,3)) 330,340,330	POTIN	117
	330 CONTINUE	POTIN	118
	340 CONTINUE	POTIN	119
	400 FUPMAT(A0,R),SE1E,0,1E0,2)	POTIN	120
	DELTA=0	POTIN	121
170	IF(11000,LU,0) GO TO 450	POTIN	122
	REAL(5,000) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	POTIN	123
	PHI(1) (4,000) IDEFAT(1),MUO(1),FO(1),OMEGA(1),PHI(1),SL(1),MA(1)	POTIN	124
	425 FUPMAT(0F 12,5)	POTIN	125
	IVISC=1	POTIN	126
175	LU 11 050	POTIN	127
	450 LU 000 I=1,ABS	POTIN	128
	IF(MUO(1)) 500,600,500	POTIN	129
	500 IVISC=1	POTIN	130
	600 CONTINUE	POTIN	131
180	IF(11000,LU,0,AR,0,IPANT,HE,0) IVISC=0	POTIN	132
	PHIE(n,n1)=20	POTIN	133
	MUO=10.000(-20)	POTIN	134
	C J=NEW STREAMLINE POSITION	POTIN	135
	1100 K=1	POTIN	136
185	IF(111,FO,C) REAL(5,200) X(K),H(K),PHI(K)	POTIN	137
	PHIE(n,n1)	POTIN	138
	IF(111,AL,U) REAL(9,200) X(K),H(K),PHI(K)	POTIN	139
	PHIE(n,n20) K,X(K),H(K),PHI(K)	POTIN	140
	FORX(K)	POTIN	141
	Y(K)=0.0	POTIN	142
190	C J=NEW STREAMLINE POSITION AND STREAMTUBE PROPER	POTIN	143
	K=2	POTIN	144
	1200 CONTINUE	POTIN	145
	IF(111,LU,0) REAL(5,200) X(K),H(K),PHI(K),P(K),I(K),U(K)	POTIN	146
	IF(111,HE,C) REAL(9,200) X(K),H(K),PHI(K),P(K),I(K),U(K)	POTIN	147
195	PHIE(0,020) K,Y(K),H(K),PHI(K),P(K),I(K),U(K)	POTIN	148
	H=SLRT((X(K)-X(K-1))**2+(H(K)-H(K-1))**2)	POTIN	149
	IF(ABS(PHI(K)-PHI(K-1))=1.0E-06) 1300,1300,1400	POTIN	150
	1300 DELTA=H	POTIN	151
	GO TO 1500	POTIN	152
200	1400 DELTA=H*(PHI(K)-PHI(K-1))/H*(2.0+SIN(.5*(PHI(K)-PHI(K-1))))	POTIN	153
	1500 AA(K)=P(K)*H(K)+PHI(K-1)*DELTA*DELTA	POTIN	154
	Y(K)=Y(K-1)+PHI(K)*DELTA	POTIN	155
	C SIN(AMOUNT) (SPLICES MASS FRACTIONS)	POTIN	156
205	IF(111,LU,0) REAL(5,1000) C(1,NA),I=1,ABS	POTIN	157
	IF(111,HE,C) REAL(9,1000) C(1,NA),I=1,ABS	POTIN	158
	1000 FUPMAT(0F 10,3)	POTIN	159
	C SETUP CALCULATIONS	POTIN	160
	1900 Z=0.0	POTIN	161
	LU 2000 I=1,ABS	POTIN	162
210	2000 Z=Z+PHI(I)*Z(I)	POTIN	163
	Z=Z+U(I)/Z	POTIN	164
	Z=Z+K(I)/Z	POTIN	165
	LU 2100 I=1,ABS	POTIN	166
215	2100 ZS(1,NA)=C(1,NA)*Z/MA(1)	POTIN	167
	2000 H=Z+PHI(I)*Z(I)/H+V(I)*Z	POTIN	168
	H=Z+PHI(I)*Z(I)/H+V(I)*Z	POTIN	169
	PHI(I)=H*PHI(I)/H	POTIN	170
	P=Z+U(I)	POTIN	171
	P=Z+U(I)	POTIN	172
220	LU 2005 I=1,ABS	POTIN	173
	LU 2005 J=1,NA	POTIN	174
	I=I+1)Z=0	POTIN	175
	2005 CONTINUE	POTIN	176
225	2405 CONTINUE	POTIN	177
	IF(IVISC) 2406,2402,2406	POTIN	178
	2402 LU 2404 I=1,ABS	POTIN	179
	UPS(1)=0	POTIN	180
	LU 2401 J=1,NA	POTIN	181
	2404 UPS(1)=UPS(1)+C(1,NA)*Z(I)*Z(I)+Z(I)-Z(I)	POTIN	182
230	GO TO 2408	POTIN	183
	2408 I=I+1	POTIN	184
	ADL=0	POTIN	185
	(ALL I=NA,SP(I,NA),P(K),K,MULT,1810)	POTIN	186
	2408 P=0	POTIN	187
235	LU 2410 I=1,ABS	POTIN	188
	2410 I=Z+PHI(I)*Z(I)/H	POTIN	189
	IF(NA=1) 2430,2430,2420	POTIN	190
	2420 I=Z+PHI(I)*Z(I)/H	POTIN	191
	2430 LU 2450 I=1,ABS	POTIN	192
240	2450 I=Z+PHI(I)*Z(I)/H	POTIN	193
	LU 2450 J=1,NA	POTIN	194

```

IF (OIM(I,J)AMH(F)/MU=FMAX) 2450,2450,2440
2440 FMAX=OIM(I,J)AMH(F)/MU
2450 CUMINUL
HE(N)RH(N)RU(N)=DELY/(MU*FMAX)
KATN)MUA)MSH((CP+ZMN=NCUN)/(CP+T(K)AKU))
GAMERCP/(CP+MUA/ZKA)
MUE 4.0*SDMA/(Q.064A=5.0)
M(N)=0
250 GO 2500 I=1,MUS
GO 2500 J=1,NH
2500 M(N)M(A)+A(1,J)+C(1,K)+I(A)+*(J=2)
IF (A=AMAX) 2600,2700,2700
2600 M(N)+1
GO TO 1200
2700 TORY(KMAX)
MEXN(KMAX)
C
CALL CONDITIONS
IF (I=1,NE,0) GO TO 2801
M=11(0,621)
GO 2800 I=1,MAX
READ (5,200) XN(1),RN(1),PHN(1),PN(1),SB(1)
2800 WRITE (6,620) I,XN(1),RN(1),PHN(1),PN(1),SB(1)
GO TO 3700
285 2801 READ (4,200) XN(1),RN(1),XN(2),RN(2)
3700 S=ASU
M=5*ASU
M=1
P=SMRPA(M)
PHN=PHN(M)
IF (PHN ) 3900,3900,3900
3900 PHN(M)=PHN(M)
3900 M=+1
PHN=PHN(M)
IF (PHN ) 4300,4000,4300
4000 IF (A=AMAX) 4100,4200,4200
4100 PHN(M)=XN(1)A((XN(M+1)-XN(M-1)))/(XN(M+1)+XN(M-1)))
GO TO 4400
4200 PHN(M)=XN(2)A((RN(M)-RN(M-1)))/(RN(M)+RN(M-1)))
PHN(M)=PHN(M)
280 4300 IS=ASU(M)
IF (IS ) 4800,4200,4800
4400 FMAX(SI(PH(N))=SI(PH(N-1)))/(XN(M)+XN(M-1))
IF (ABS(FMAX/(XN(M)+XN(M-1)))-1,E=06) 4600,4600,4500
285 4500 ILSFA=(PH(N)-PH(N-1))/FMAX
GO TO 4700
4600 ULSFA=SI(XN(N)-XN(N-1))+2*(RN(N)-RN(N-1))+2)
4700 S=ASU+ULSFA
S(M)=SFA
290 4800 IF (A=AMAX) 4900,4900,4900
4900 IF (I=1,NE,0) GO TO 4901
WRITE (6,623)
GO 5000 I=1,MAX
READ (5,200) XN(N),RN(N),PHN(N),PN(N),SB(N)
295 5000 WRITE (6,620) N,XN(N),RN(N),PHN(N),PN(N),SB(N)
GO TO 5700
4901 READ (4,200) XN(1),RN(1),XN(2),RN(2)
5700 S=ASU
C
CALL CONDITIONS
M=1
PHN=PHN(M)
IF (PHN ) 5900,5900,5900
5900 PHN(M)=PHN(M)
5900 M=+1
PHN=PHN(M)
IF (PHN ) 6300,6000,6300
6000 IF (A=AMAX) 6100,6200,6200
6100 PHN(M)=XN(1)A((XN(M+1)-XN(M-1)))/(XN(M+1)+XN(M-1)))
GO TO 6500
6200 PHN(M)=XN(2)A((RN(M)-RN(M-1)))/(RN(M)+RN(M-1)))
PHN(M)=PHN(M)
310 6300 IS=ASU(M)
IF (IS ) 6750,6400,6750
6400 FMAX(SI(PH(N))=SI(PH(N-1)))/(XN(N)+XN(N-1))
IF (ABS(FMAX/(XN(N)+XN(N-1)))-1,E=06) 6600,6600,6500
315 6500 ULSFA=(PH(N)-PH(N-1))/FMAX
GO TO 6700
6600 ULSFA=SI(XN(N)-XN(N-1))+2*(RN(N)-RN(N-1))+2)
6700 S=ASU+ULSFA
S(M)=SFA
320 6750 IF (A=AMAX) 6900,6800,6800
6800 IF (I=1,NE,3) 6905,6850,6850
C
CALL CONDITIONS
M=1
PHN=PHN(M)
IF (PHN ) 7100,7000,7100
7000 IF (I=1,NE,0) READ(5,200) XSM,NSM,PSI,PSM,ISM,USM
IF (I=1,NE,0) READ(5,1000) (CSM(I),I=1,MUS)
IF (I=1,NE,0) READ(4,200) XSM,NSM,PSI,PSM,ISM,USM
WRITE (6,624) XSM,NSM,PSI,PSM,ISM,USM
IF (I=1,NE,0) READ(4,1000) (LSM(I),I=1,MUS)
330 6900 FMAX(I=1,3) 7000,6910,6910
PHN=PHN(M)
C
CALL CONDITIONS
M=1
PHN=PHN(M)
IF (PHN ) 7200,7100,7200
7100 FMAX(SI(PH(N))=SI(PH(N-1)))/(XN(N)+XN(N-1))
IF (ABS(FMAX/(XN(N)+XN(N-1)))-1,E=06) 7300,7300,7200
7200 ULSFA=(PH(N)-PH(N-1))/FMAX
GO TO 7400
7300 ULSFA=SI(XN(N)-XN(N-1))+2*(RN(N)-RN(N-1))+2)
7400 S=ASU+ULSFA
S(M)=SFA
335 7450 IF (I=1,NE,3) 7500,6910,6910
7500 FMAX(I=1,3) 7600,6910,6910
C
CALL CONDITIONS
M=1
PHN=PHN(M)
IF (PHN ) 7700,7600,7700
7600 FMAX(SI(PH(N))=SI(PH(N-1)))/(XN(N)+XN(N-1))
IF (ABS(FMAX/(XN(N)+XN(N-1)))-1,E=06) 7800,7800,7700
7700 ULSFA=(PH(N)-PH(N-1))/FMAX
GO TO 7900
7800 ULSFA=SI(XN(N)-XN(N-1))+2*(RN(N)-RN(N-1))+2)
7900 S=ASU+ULSFA
S(M)=SFA

```

```

PUTIN 195
PUTIN 196
PUTIN 197
PUTIN 198
PUTIN 199
PUTIN 200
PUTIN 201
PUTIN 202
PUTIN 203
PUTIN 204
PUTIN 205
PUTIN 206
PUTIN 207
PUTIN 208
PUTIN 209
PUTIN 210
PUTIN 211
PUTIN 212
PUTIN 213
PUTIN 214
PUTIN 215
PUTIN 216
PUTIN 217
PUTIN 218
PUTIN 219
PUTIN 220
PUTIN 221
PUTIN 222
PUTIN 223
PUTIN 224
PUTIN 225
PUTIN 226
PUTIN 227
PUTIN 228
PUTIN 229
PUTIN 230
PUTIN 231
PUTIN 232
PUTIN 233
PUTIN 234
PUTIN 235
PUTIN 236
PUTIN 237
PUTIN 238
PUTIN 239
PUTIN 240
PUTIN 241
PUTIN 242
PUTIN 243
PUTIN 244
PUTIN 245
PUTIN 246
PUTIN 247
PUTIN 248
PUTIN 249
PUTIN 250
PUTIN 251
PUTIN 252
PUTIN 253
PUTIN 254
PUTIN 255
PUTIN 256
PUTIN 257
PUTIN 258
PUTIN 259
PUTIN 260
PUTIN 261
PUTIN 262
PUTIN 263
PUTIN 264
PUTIN 265
PUTIN 266
PUTIN 267
PUTIN 268
PUTIN 269
PUTIN 270
PUTIN 271
PUTIN 272
PUTIN 273
PUTIN 274
PUTIN 275
PUTIN 276
PUTIN 277
PUTIN 278
PUTIN 279
PUTIN 280
PUTIN 281
PUTIN 282
PUTIN 283
PUTIN 284
PUTIN 285
PUTIN 286
PUTIN 287
PUTIN 288
PUTIN 289
PUTIN 290
PUTIN 291
PUTIN 292

```

```

IF (DIM(I,J) .AND. (F)/MU = FMAX) 2450,2450,2440
2440 FMAX = DIM(I,J) * RHO(R)/MU
2450 CONTINUE
NE(N) = RHO(N) * ODEL(Y(N)/(MU * FMAX)
NA(N) = MU(N) * ASQRT((CP * ZMA - HCON) / (CP * T(K) * NU))
GAMA = CP / (CP * CUM(ZMA) / ZMA)
RUE = 4.0 * SAMA / (C. * GA * A = 5.0)
M(N) = 0.0
250 DO I = 1, AUS
DO J = 1, JMAX
2500 M(K) = M(K) + A(I, J) * C(I, K) * T(K) * (J = 2)
IF (M = MMAX) 2600, 2700, 2700
2600 K = K + 1
GO TO 1200
2700 YOUT(KMAX)
WRITE(KMAX)
C
CALL CONDITIONS
IF (IK1, NE, 0) GO TO 2801
WRITE(16, A21)
DO I = 1, IMA
READ (5, 290) X(I), H(I), PH(I), P(I), S(I)
2800 WRITE (6, 020) I, X(I), H(I), PH(I), P(I), S(I)
GO TO 3700
285 READ (9, 200) X(1), H(1), X(2), H(2)
3700 S = 0.0
WRITE(16, A20)
I = 1
S = S * M(N)
PH(I) = PH(I) * M(N)
IF (PH(I) ) 3900, 3800, 3900
3800 PH(I) = PH(I) * M(N)
3900 K = K + 1
PH(I) = PH(I) * M(N)
IF (PH(I) ) 4300, 4000, 4300
4000 IF (M = MMAX) 4100, 4200, 4200
4100 PH(I) = (PH(I) * A(I, H(I) * M(N) - 1)) / (X(I) * M(N) - 1)
GO TO 4300
4200 PH(I) = (PH(I) * A(I, H(I) * M(N) - 1)) / (X(I) * M(N) - 1)
PH(I) = PH(I) * M(N)
4300 IS = S * M(N)
IF (IS ) 4600, 4400, 4600
4400 FAX = (SIN(PH(I) * M(N)) - SIN(PH(I) * M(N) - 1)) / (X(I) * M(N) - X(I) * M(N) - 1)
IF (ABS(FAX) / (X(I) * M(N) - 1)) = 1.0, I = 06) 4600, 4600, 4500
285 I = I * A(I, H(I) * M(N) - 1) / FAX
GO TO 4700
4600 U = (S * A(I, H(I) * M(N) - 1)) * 2 + (M(N) * H(I) * M(N) - 1) * 2)
4700 S = S * A(I, H(I) * M(N) - 1)
S(I) = S * A
290 IF (M = MMAX) 3900, 4900, 4900
4900 IF (IK1, NE, 0) GO TO 4901
WRITE (6, 023)
GO 5000 I, IMA
READ (5, 290) X(I), H(I), PH(I), P(I), S(I)
295 WRITE (6, 024) X, H(I), H(I), PH(I), P(I), S(I)
GO TO 5700
4901 READ (9, 200) X(1), H(1), X(2), H(2)
5700 S = 0.0
C
WRITE(16, A20)
I = 1
PH(I) = PH(I) * M(N)
IF (PH(I) ) 5900, 5800, 5900
5800 PH(I) = PH(I) * M(N)
5900 K = K + 1
PH(I) = PH(I) * M(N)
IF (PH(I) ) 6300, 6000, 6300
6000 IF (M = MMAX) 6100, 6200, 6200
6100 PH(I) = (PH(I) * A(I, H(I) * M(N) - 1)) / (X(I) * M(N) - 1)
GO TO 6300
6200 PH(I) = (PH(I) * A(I, H(I) * M(N) - 1)) / (X(I) * M(N) - 1)
PH(I) = PH(I) * M(N)
6300 IS = S * M(N)
IF (IS ) 6750, 6400, 6750
6400 FAX = (SIN(PH(I) * M(N)) - SIN(PH(I) * M(N) - 1)) / (X(I) * M(N) - X(I) * M(N) - 1)
IF (ABS(FAX) / (X(I) * M(N) - 1)) = 1.0, I = 06) 6600, 6600, 6500
6500 U = (S * A(I, H(I) * M(N) - 1)) * 2 + (M(N) * H(I) * M(N) - 1) * 2)
GO TO 6700
6600 U = (S * A(I, H(I) * M(N) - 1)) * 2 + (M(N) * H(I) * M(N) - 1) * 2)
6700 S = S * A(I, H(I) * M(N) - 1)
S(I) = S * A
320 IF (M = MMAX) 5900, 6800, 6800
6800 IF (IK1, NE, 0) 6900, 6850, 6850
C
WRITE(16, A20)
325 IF (IK1, NE, 0) READ(5, 200) XSM, NSM, PSI, PSM, ISM, USM
IF (IK1, NE, 0) READ(5, 1600) (CSM(I), I = 1, AUS)
IF (IK1, NE, 0) READ(9, 200) XSM, NSM, PSI, PSM, ISM, USM
WRITE (6, 024) XSM, NSM, PSI, PSM, ISM, USM
IF (I = 1, I = AUS) READ (9, 1600) (LSM(I), I = 1, AUS)
330 WRITE(16, A21)
PHSM = PSM
C
WRITE(16, A20)
335 READ (5, 200) XSTREM, NSTREM, PSTREM, PSSTREM, ISSTREM, USTREM
WRITE (6, 025) XSTREM, NSTREM, PSTREM, PSSTREM, ISSTREM, USTREM
READ (5, 6900) (LSSTREM(I), I = 1, AUS)
WRITE (6, 026)
C
WRITE(16, A20)
340 WRITE(16, A20)
IF (IK1, NE, 0) READ (5, 200) XSM, NSM, PSI, PSM, ISM, USM

```

```

PUTIN 195
PUTIN 196
PUTIN 197
PUTIN 198
PUTIN 199
PUTIN 200
PUTIN 201
PUTIN 202
PUTIN 203
PUTIN 204
PUTIN 205
PUTIN 206
PUTIN 207
PUTIN 208
PUTIN 209
PUTIN 210
PUTIN 211
PUTIN 212
PUTIN 213
PUTIN 214
PUTIN 215
PUTIN 216
PUTIN 217
PUTIN 218
PUTIN 219
PUTIN 220
PUTIN 221
PUTIN 222
PUTIN 223
PUTIN 224
PUTIN 225
PUTIN 226
PUTIN 227
PUTIN 228
PUTIN 229
PUTIN 230
PUTIN 231
PUTIN 232
PUTIN 233
PUTIN 234
PUTIN 235
PUTIN 236
PUTIN 237
PUTIN 238
PUTIN 239
PUTIN 240
PUTIN 241
PUTIN 242
PUTIN 243
PUTIN 244
PUTIN 245
PUTIN 246
PUTIN 247
PUTIN 248
PUTIN 249
PUTIN 250
PUTIN 251
PUTIN 252
PUTIN 253
PUTIN 254
PUTIN 255
PUTIN 256
PUTIN 257
PUTIN 258
PUTIN 259
PUTIN 260
PUTIN 261
PUTIN 262
PUTIN 263
PUTIN 264
PUTIN 265
PUTIN 266
PUTIN 267
PUTIN 268
PUTIN 269
PUTIN 270
PUTIN 271
PUTIN 272
PUTIN 273
PUTIN 274
PUTIN 275
PUTIN 276
PUTIN 277
PUTIN 278
PUTIN 279
PUTIN 280
PUTIN 281
PUTIN 282
PUTIN 283
PUTIN 284
PUTIN 285
PUTIN 286
PUTIN 287
PUTIN 288
PUTIN 289
PUTIN 290
PUTIN 291
PUTIN 292

```