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TURBINE ENGINE CONTROL SYNTHESIS. VOLUME II.
SIMULATION AND CONTROLLER SOFTWARE

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Honeywell, Incorporated

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20. Abstract (Continued)

A command controller is synthesized and wind tunnel tested. This controller is a good approximation to time optimal with surge-stall, TT4, and flameout constraints. Small-amplitude control responses are precise. There is strong stability. Volume II contains three Appendices. Appendix A contains the details of engine math models. The software for the wind tunnel controller is presented in Appendix B. Appendix C contains a derivation of rate model following. Volume III presents results of frequency response tests of a J85-13 engine operating in the APL wind tunnel. The data are reduced and models identified.

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APPENDIX A COMPONENT MODEL SOFTWARE

Software models of the J85 engine are presented in this appendix. Two computer programs are discussed:

- Linearization
- Nonlinear engine simulation

Fortran listings of the two programs are presented in Tables A-1* and A-2. Both programs were written for the SDS-9300 computer.

The linearization program, discussed in the first part of this appendix, was used to generate linear engine models for the synthesis of linear optimal controllers reported in Section IV of Volume I.

The nonlinear engine simulation package is discussed in the second half of this appendix. This computer program is basically a Fortran version of the J85 NASA component model of Reference A-1.

LINEARIZATION PROGRAM

The function of this program is to generate linear models of the J85 engine. A Fortran listing of the program is presented in Table A-2 and discussed in the following paragraphs.

* For the convenience of the reader, all figures and tables are provided at the end of each appendix.

The discussion is divided into four sections which correspond with the main parts of the program:

- Trim point calculation
- Engine dynamics
- Linearization
- Input data

Trim point calculations are discussed in the first subsection, labeled Trim Routine, where steady-state set points for the engine are computed. This section of the program calculates the fuel flow required to maintain the nominal operating condition specified by the input parameters. Trim values of engine responses are also calculated in this subsection.

Engine dynamics are discussed in the next subsection. A nonlinear dynamic model of the engine is contained in a subroutine called DYNAMIC. The model is a reduced order-version of the NASA component model of Reference A-1. All gas dynamics have been removed from the model so that it contains only two states, spool speed and engine case temperature.

The linearization procedure is presented in the third subsection of this appendix, under Linearization Routine. Engine dynamics are linearized about a steady-state trim point.

Input data are discussed in the last subsection. Two sets of data are required to run the program. One set defines the nominal operating conditions, i. e., steady-state spool speed, geometry control positions, compressor inlet pressure, and rotor torque load. The other set contains steady-state engine component data, i. e., compressor stage data and turbine map data.

Computations in the linearization program proceed in the following order. First, engine component data are read in. Then input parameters defining the nominal operating condition are read. Next a steady-state trim point corresponding to the input parameters is computed. Finally, a linear model of the engine is obtained by linearizing the nonlinear engine dynamics about the trim point.

Trim Routine

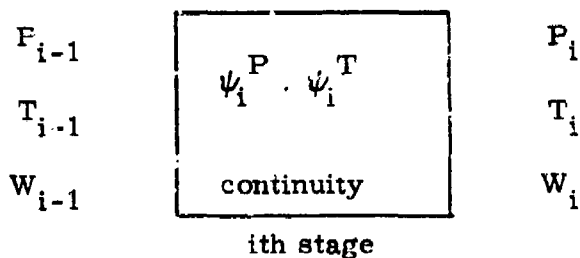
This section of the program generates steady-state trim solutions of the engine dynamic equations. Inputs to the routine include steady-state spool speed (N), geometry control position (IGV, BLD, A_g), compressor inlet pressure and temperature (P_o , T_o), exhaust nozzle discharge pressure (P_g), and external rotor torque load (SPLC). Given this set of inputs, the program iteratively computes a steady-state operating point. The computational procedure is summarized in the next paragraph.

First, steady-state values of the eight input parameters, N , A_g , IGV, BLD, P_o , T_o , P_g , and SPLC, are read in. Then initial guesses for fuel flow (WF), burner temperature (TB), turbine discharge pressure (PT), and inlet airflow (W_o) are made. Steady-state values of these four parameters are iteratively computed in four nested iteration loops. Steady-state values of all of the other engine variables are computed closed form.

Computations in the routine proceed in a manner analogous to the path followed by a particle of air entering the inlet; i. e., the compressor section is trimmed first, followed by the burner, the turbine, and finally the nozzle section.

The compressor is modeled by the stage stacking technique. Each stage is individually represented by a pair of experimental functions (ψ^P , ψ^T) which

are used to compute the pressure rise and temperature rise across the stage. Airflow through the stage is computed from the steady-state continuity relation.



$$T_i = T_{i-1} + f_1 [\psi_i^T, N]$$

$$P_i = P_{i-1} \cdot f_2 [\psi_i^P, N, T_{i-1}] \quad (A-1)$$

$$W_i = W_{i-1}$$

The stages are interconnected, or stacked, to form the compressor model where the discharge conditions of one stage are the inlet conditions of the following stage. Compressor bleed (BLD) and inlet guide vane (IGV) effects are included in the appropriate stages.

Thus, steady-state values of all the compressor variables can be computed closed form from a knowledge of the input parameters η , N , IGV, BLD, P_o , T_o , and W_o . All of these inputs are specified, except for inlet airflow (W_o). This variable is computed iteratively in the outer iteration loop.

Burner performance is represented by three experimental relations, pressure drop across the burner (ΔP_B), burner enthalpy (HB), and burner efficiency (η_B), together with the steady-state continuity relation. The pressure drop across the combustor is a function of compressor discharge pressure (PCD), burner inlet airflow (WB), compressor discharge temperature (TCD) and burner temperature (TB).

$$\Delta PB = f_1 [PCD, WB, TCD, TB] \quad (A-2)$$

Burner enthalpy is computed from a real gas experimental relationship which is a function of burner temperature (TB), burner airflow (WB) and fuel flow (WF).

$$HB = f_2 [TB, WB, WF] \quad (A-3)$$

Combustor efficiency is defined as the portion of the heat of combustion that is available for a gas temperature rise. It is computed from an experimental correlation of the form

$$\eta_B = f_3 [PB \cdot \Delta TB] \quad (A-4)$$

where

$$PB = PCD + \Delta PB$$

$$\Delta TB = TB - TCD$$

These three functions, f_1 , f_2 , and f_3 , are functions of the three burner variables, WB, TB, and WF. One of these parameters, WB, is computed closed form (from the continuity relation), as

$$WB = WCD - WTC \quad (A-5)$$

where WTC is airflow which is bled from the compressor to cool the turbine.

The other two parameters, TB and WF, are computed iteratively in the inner two iteration loops.

The turbine is modeled by two performance maps, together with the steady-state continuity relation. Turbine enthalpy drop (ΔHT) and turbine airflow (WT) are represented as functions of burner temperature (TB), burner pressure (PB), spool speed (N), and turbine pressure ratio (PR_T).

$$HT = f_1 [N, TB, PR_T] \quad (A-6)$$

$$WT = f_2 [N, TB, PB, PR_T]$$

These functions cannot be evaluated until the variable PR_T is computed. Although WT is established by the continuity relation

$$WT = WB + WF \quad (A-7)$$

PR_T cannot be obtained closed form because the second function, f_2 , cannot be inverted to solve for PR_T . Thus, PR_T is calculated in the third iteration loop.

The exhaust nozzle is represented as a variable area flow passage capable of choking. The mathematical relation is

$$\frac{WN \sqrt{TT}}{PT} = KNA_8 \cdot A_8 \cdot f \left[\frac{P_8}{PT} \right] \quad (A-8)$$

where

$$f \left[\frac{P_8}{PT} \right] = \left(\frac{P_8}{PT} \right)^{\frac{1}{\gamma}} \sqrt{1 - \frac{P_8}{PT}}^{\frac{\gamma-1}{\gamma}}$$

This expression is used to compute the nozzle airflow (WN).

Compressor inlet airflow (W_o) is systematically changed in the outer iteration loop until the nozzle airflow computed from the above relation agrees with nozzle airflow computed from the steady-state continuity relation

$$WN = WT + WTC \quad (A-9)$$

Details of the trim routine are presented in the flowchart of Figure A-1.

First, the input variables N , A_g , IGV , BLD , P_o , T_o , P_g , and $SPLC$ are read in. The last variable, $SPLC$, is a fictitious external torque load applied to the rotor shaft. If this variable is set to zero, the routine will identify a steady-state operating point on the engine equilibrium line. Non-zero values of $SPLC$ cause the routine to identify quasi-steady-state operating points off of the equilibrium line. Quasi-steady state means that both $N = \text{constant}$ and $\dot{N} = \text{constant}$; the nonzero \dot{N} is balanced by the external torque lead $SPLC$.

Then, initial guesses of fuel-to-air-ratio in the burner (FAB), burner temperature (TB), and inlet air flow (W_o) are made. The parameter FAB is defined as

$$FAB \triangleq WF/WB \quad (A-10)$$

Thus, guessing a value of FAB is equivalent to guessing fuel flow.

Next, the integer variables which count the number of iterations are initialized to zero. The variables are defined as:

ITER1 -- number of iterations of the W_o loop

ITER2 -- number of iterations of the PT loop

ITER3 -- number of iterations of the TB loop

A counter is not assigned to the inner loop, the WF iteration. The variable III is a switch which is maintained as zero during the iteration process and set equal to one when all loops have converged.

In the next section of the program steady-state compressor variables are computed. Individual calculations are made for each compressor stage; the outlet conditions of one stage are inlet conditions of the next stage.

First, the pressure at the outlet of the inlet guide vanes is computed from the equation

$$P_{IGV} = P_o \cdot PR_{IGV} - 0.005 P_o \quad (A-11)$$

where the IGV pressure ratio (PR_{IGV}) is calculated as a function of spool speed.

$$PR_{IGV} = PR_{IGV} [N]$$

Temperature and airflow, which are constant across the inlet guide vanes, are computed as

$$\begin{aligned} T_{IGV} &= T_o \\ W_{IGV} &= W_o \end{aligned} \quad (A-12)$$

The outlet conditions of the inlet guide vanes are the inlet conditions of the first compressor stage. Airflow in the first compressor stage is computed from the continuity equation

$$WC_1 = W_{IGV} \quad (A-13)$$

This airflow, together with T_{IGV} and P_{IGV} , are used to compute the axial component of velocity in the stage:

$$v_{z_1} = v [WC_1, T_{IGV}, P_{IGV}] \quad (A-14)$$

which in turn is used to compute the flow coefficient

$$\phi_1 = K_{\phi 1} \cdot v_{z_1} / N \quad (A-15)$$

The constant K_{ϕ_1} in this expression is a function of the geometry of the stage. Next, pressure rise and temperature rise coefficients are determined from ϕ_1 .

$$\begin{aligned}\psi_1^P &= \psi_1^P [\phi_1, \text{IGV}] \\ \psi_1^T &= \psi_1^T [\phi_1, \text{IGV}]\end{aligned}\tag{A-16}$$

Note the effect of inlet guide vane position is included in the first-stage coefficients. Finally, the pressure and temperature at the outlet of the stage are computed.

$$\begin{aligned}BC_1 &= P_{\text{IGV}} \cdot \left(1 + \psi_1^P \cdot K_{\psi_1} \cdot N^2 / T_{\text{IGV}} \right)^{\frac{\gamma}{\gamma-1}} \\ TC_1 &= T_{\text{IGV}} + K_{\psi_1} \cdot N^2 \cdot \psi_1^T\end{aligned}\tag{A-17}$$

The constant K_{ψ_1} in these expressions is also a function of the geometry of the first stage.

Pressure, temperature, and airflow in the other compressor stages are computed in the same manner as the first-stage data. Calculations for the second and third stages are shown explicitly in the Figure A-1 flowchart.

Compressor bleed effects are included in the third, fourth, and fifth compressor stages. Bleed airflow in the third stage (WBL_3) is computed from the relation

$$WBL_3 = KBLD_3 \cdot BLD \cdot PC_3 / TC_3\tag{A-18}$$

where

$KBLD_3$ is the bleed flow coefficient

BLD is the bleed area

PC_3 is the third-stage discharge pressure

TC_3 is the third-stage discharge temperature

The third-stage bleed airflow is subtracted from the third-stage inlet airflow (WC_3) to determine the inlet airflow to the fourth stage (WC_4)

$$WC_4 = WC_3 - WBL_3 \quad (A-19)$$

Bleed effects in the fourth and fifth stages are evaluated in the same way.

Compressor discharge relations are evaluated at the end of the compressor simulation section. The pressure, temperature, and airflow at the compressor discharge are the values at the outlet guide vanes.

$$\begin{aligned} PCD &= P_{OGV} \\ TCD &= T_{OGV} \\ WCD &= W_{OGV} \end{aligned} \quad (A-20)$$

Compressor discharge enthalpy (HCD) is evaluated as a function of TCD from a real gas model,

$$HCD = HCD [TCD] \quad (A-21)$$

Finally, the net change in airflow times enthalpy across the compressor is evaluated.

$$\begin{aligned} \Delta(WH)_{CD} &= WCD \cdot HCD + c_p (WBL_3 \cdot TC_3 + WBL_4 \cdot TC_4 + WBL_5 \cdot TC_5) \\ &\quad - c_p \cdot W_o \cdot T_o + SPLC \end{aligned} \quad (A-22)$$

This change is proportional to the compressor torque load on the rotor shaft.

Next, steady-state airflow into the burner (WB) is evaluated as

$$WB = WCD - WTC \quad (A-23)$$

where WTC is airflow which is extracted from the compressor discharge to cool the turbine.

Then a test is made to determine if ITER1 equals one. If ITER1 = 1, indicating that this is the first pass through the W_o iteration loop, an initial value is assigned to turbine discharge pressure (PT).

$$PT = 0.35 PCD \quad (A-24)$$

If ITER1 > 1, the routine goes directly to step 2 since a value for PT has already been calculated in this case.

Turbine inlet airflow is then computed from the steady-state continuity relation

$$WT_{OLD} = WB (1 + FAB) \quad (A-25)$$

This is the sum of burner inlet airflow and fuel flow.

Next, the three experimental relations which model burner performance are evaluated. First, the pressure drop across the burner is computed

$$\Delta PB = \frac{KB \cdot WB^2}{PCD} (0.771 TCD - 0.35 TB) \quad (A-26)$$

where KB is a constant. Pressure losses due to both fluid friction and momentum changes from the addition of heat are included in this expression.

The pressure at the burner discharge (PB) is

$$PB = PCD - \Delta PB \quad (A-27)$$

Next, burner efficiency η_B is evaluated as a function of the parameter PBDTB where

$$PBDTB = \frac{\Delta}{PB} (TB - TCD) \quad (A-28)$$

Burner efficiency is defined as the portion of the heat of combustion that is available for a gas temperature rise. Finally, burner enthalpy (HB) is determined from the real-gas functional relationship

$$HB = HB [FAB, TB] \quad (A-29)$$

The Figure A-1 flow chart shows that turbine enthalpy drop (ΔHT) is actually calculated between the computation of burner efficiency and burner enthalpy. Turbine enthalpy drop is determined from experimental turbine data, i. e., from $\frac{\Delta HT}{N \sqrt{TB}}$, $\frac{PT}{PB}$, and $\frac{N}{\sqrt{TB}}$. Thus the enthalpy drop is

$$\Delta HT = \frac{\Delta HT}{N \sqrt{TB}} \cdot N \cdot \sqrt{TB} \quad (A-30)$$

At this point in the routine, sufficient data are available to recalculate turbine inlet airflow from the heat equation as applied to the burner. The heat equation specifies that under steady flow conditions

$$\Sigma \dot{Q}_{BURNER} = \Sigma (WH)_{BURNER} = 0$$

The amount of heat which enters the burner must equal the heat which exits from the burner. In terms of the parameters previously identified

$$(WH)_{in} = (WH)_{out}$$

or (A-31)

$$WB \cdot HCD + W_f \cdot h_{FUEL} \cdot \eta_B = (WB + W_f) \cdot HB$$

This equation is solved for the term $(WB + W_f)$ which is the burner discharge flow. The result is

$$WT_1 = \frac{WB (h_{FUEL} \cdot \eta_B - HCD)}{(h_{FUEL} \cdot \eta_B - HB)} \quad (A-32)$$

where $WT_1 = (WB + W_f)$ is the burner discharge or turbine inlet airflow. Note that fuel flow does not appear explicitly in this equation, but rather as the difference $WT_1 - WB$.

Next, the difference between turbine inlet airflow as determined from the heat equation (WT_1) and as determined from the continuity relation (WT_{OLD}) is computed. The result is termed turbine airflow error, WT_{ERROR} .

$$WT_{ERROR} = |WT_1 - WT_{OLD}| \quad (A-33)$$

The magnitude of this error is the convergence criterion for the fuel flow iteration loop. If $|WT_{ERROR}| \leq 0.0005$, the iteration loop is converged. If $|WT_{ERROR}| > 0.0005$, fuel flow is updated according to the following scheme:

$$\begin{aligned} WT_{OLD} &= 1/2 (WT_{OLD} + WT_1) \\ WF &= WT_{OLD} - WB \\ FAB &= WT/WB \end{aligned} \quad (A-34)$$

and the routine is returned to step 4. The fuel flow iteration continues until the criterion $|WT_{ERROR}| \leq 0.0005$ is satisfied.

After the fuel flow iteration converges, the airflow out of the turbine is computed. This airflow is called the nozzle airflow, WN .

$$WN = WT_1 + WTC \quad (A-35)$$

It is assumed that the cooling airflow, WTC , is added back into the flow at the turbine discharge.

Next, turbine enthalpy is computed from the equation

$$HT = \frac{WT_1(HB - \Delta HT) + WTC \cdot HCD}{WN} \quad (A-36)$$

Then the steady-state rotor torque relation,

$$\dot{N} = \Sigma_{TORQUE} = 0$$

is used to recalculate burner enthalpy. The airflow-enthalpy change across the compressor is subtracted from the airflow-enthalpy change across the turbine to determine the net rotor torque.

$$\Sigma_{TORQUE} = \Delta(WH)_{TURBINE} - \Delta(SH)_{COMPRESSOR} = 0$$

This equation is solved for a new estimate of burner enthalpy, called HB_R .

$$HB_R = \frac{\Delta(WH)_{CD} + WN \cdot HT - WTC \cdot HCD}{WT_1} \quad (A-37)$$

The difference between burner enthalpy as calculated from the above equation (HB_R) and burner enthalpy as previously determined from the real gas model (HB) is termed burner temperature error.

$$TB_{ERROR} \triangleq HB_R - HB \quad (A-38)$$

A non-zero value of TB_{ERROR} indicates that the burner temperature estimate, TB , is inaccurate. The magnitude of this error is the convergence criterion for the TB iteration loop. If $|TB_{ERROR}| \leq 0.0005$ the iteration is converged; if $|TB_{ERROR}| > 0.0005$ the estimate of burner temperature, TB , is updated and the routine returns to step 3.

The change in TB depends on the algebraic sign of TB_{ERROR} . If HB_R is greater than HB , TB is increased. If HB_R is less than HB , TB is decreased. The magnitude of the change in TB , called ΔTB , is regulated in the routine

such that if the algebraic sign of TB_{ERROR} changes in successive iterations, the step size is halved. This procedure guarantees convergence.

Flow conditions in the exhaust nozzle are computed after the TB iteration is converged. First, the nozzle pressure ratio (PR_N) is evaluated

$$PR_N = \frac{P_8}{P_T} \quad (A-39)$$

where P_8 is discharge pressure at the nozzle exit.

The flow condition in the nozzle is determined by the magnitude of PR_N . If $PR_N > 1$, ambient pressure is greater than nozzle pressure and thus zero flow is assumed. If $PR_N < 0.528$, the nozzle is choked and if $0.528 < PR_N < 1$, the nozzle is unchoked. A nozzle coefficient is assigned depending on the flow condition.

$$\begin{aligned} K_{NOZ} &= 0 \quad \text{if } PR_N > 1, \quad \text{zero flow} \\ K_{NOZ} &= 0.2588 \quad \text{if } PR_N > 0.528, \quad \text{choked flow} \\ K_{NOZ} &= \left(\frac{P_8}{P_T}\right)^{\frac{1}{\gamma}} \sqrt{1 - \left(\frac{P_8}{P_T}\right)^{\frac{\gamma-1}{\gamma}}} \quad \text{if } 0.528 < PR_N < 1, \quad \text{unchoked flow} \end{aligned} \quad (A-40)$$

Next, turbine airflow is recalculated from experimental turbine data which is a correlation of the three parameters $\frac{WT \cdot TB}{N \cdot PB}$, $\frac{PT}{PB}$, $\frac{N}{TB}$. Turbine airflow computed from this data is

$$WT_2 = \left(\frac{WT \cdot TB}{N \cdot PB}\right) \cdot \frac{N \cdot PB}{TB} \quad (A-41)$$

The symbol WT_2 is used to differentiate this airflow from the two expressions for turbine airflow previously obtained, WT_{OLD} and WT_1 .

The difference between WT_2 and WT_1 is then computed.

$$PT_{ERROR} \triangleq WT_2 - WT_1 \quad (A-42)$$

This error is called turbine pressure error because a mismatch between WT_2 and WT_1 indicates that turbine pressure is not correct. The magnitude of this error determines if the iteration on PT is converged. If $|PT_{ERROR}| \leq 0.0005$, the iteration is converged. If $|PT_{ERROR}| > 0.0005$ the estimate of PT is updated and the routine returned to step 2 for another iteration.

The algebraic sign of PT_{ERROR} determines how the value of PT is adjusted. If PT_{ERROR} is positive, the value of PT is increased; and if PT_{ERROR} is negative, PT is decreased. Mechanization of the PT iteration is identical to the TB iteration (refer to Figure A-1 flow chart).

Following the convergence of the PT iteration, the turbine temperature (TT) is evaluated as a function of turbine enthalpy (HT) and fuel-to-air ratio in the turbine (FAT). Turbine temperature is then used to recompute nozzle air-flow from the isentropic relation

$$WN_X = \frac{KNA8 \cdot K_{NOZ} \cdot PT \cdot A_8}{\sqrt{TT}} \quad (A-43)$$

The constant KNA8 is a contraction coefficient which is a function of spool speed. The subscript X on WN is used in this expression to differentiate between the nozzle airflow computed here and the nozzle airflow previously computed from the continuity relation, WN.

The difference between WN_X and WN is then computed

$$W_{ERROR} \triangleq WN_X - WN \quad (A-44)$$

This error is a measure of the accuracy achieved by the outer loop iteration for inlet airflow, W_o . If $|W_{\text{ERROR}}| \leq 0.0005$, the iteration is sufficiently converged. If $|W_{\text{ERROR}}| > 0.0005$, the value of W_o is updated and the routine returns to step 1. Inlet airflow, W_o , is increased if W_{ERROR} is positive and decreased if W_{ERROR} is negative. The logic associated with this iteration loop is identical to the logic used in the TB and PT iterations.

Logic for the trim completion switch III is also found in this section of the routine. The switch controls printout of results obtained from intermediate steps in the program. Until all four iteration loops have converged to within the specified tolerances, the value of III is zero. Once the loops have all converged, III is set equal to one and the routine is sent back to the beginning, station 1. Values of the parameters of interest are then printed out during this final pass through the iteration loops.

Dynamic Subroutine

This section of the program computes derivatives with respect to time of spool speed (N) and case temperature (TM) given the following set of initial conditions: compressor inlet pressure and temperature (P_o, T_o), nozzle discharge pressure (P_g), current spool speed (N), current case temperature (TM), fuel flow (W_f), and geometry control positions ($A_g, \text{IGV}, \text{BLD}$).

The structure of this routine closely parallels that of the TRIM routine. Computations begin at the engine inlet and proceed through the engine to the exhaust nozzle. Parameters associated with the compressor section are evaluated first, followed in order by burner, turbine and finally exhaust nozzle parameters.

Initial conditions are specified by the nine input parameters, $P_o, T_o, P_g, N, \text{TM}, W_f, A_g, \text{IGV}$, and BLD . Initial estimates of turbine pressure (PT), inlet

airflow (W_o), and burner efficiency (η_B) are also required. Actual values of these three parameters, PT , W_o and η_B , are computed iteratively in the subroutines.

The compressor is modeled by the same set of mathematical relations which are included in the TRIM routine. Inputs to the compressor section include spool speed, inlet parameters W_o , P_o , and T_o , and compressor geometry control positions IGV and BLD. Steady-state pressure and temperature rise maps are used to compute individual stage parameters. The stages are stacked to form the compressor model; i. e., the discharge conditions of one stage are the inlet conditions of the next stage. Flow conditions at the compressor discharge are defined in terms of pressure (PCD), temperature (TCD), airflow (WCD), and enthalpy (HCD).

Steady-state burner performance is modeled by the same three experimental relations which are included in the TRIM routine. Two of these relations are used to compute burner temperature (TEB) and burner pressure (PB). The third relation is used in an iteration loop to determine burner efficiency (η_B).

Thermal capacitance effects are included at the end of the burner section. The rate of change of temperature of the engine case metal is calculated from the equation

$$\dot{T}_M = K_{TM} (TEB - TM) \quad (A-45)$$

where TM is the average temperature of the metal and K_{TM} is a constant of proportionality (a function of thermal conductivity and geometric measurements). The temperature of the gas discharged from the burner (TB) is computed from

$$TB = TEB - K_{TB} \cdot \dot{T}_M \quad (A-46)$$

where K_{TB} is a constant similar to K_{TM} . Note that in thermal equilibrium these equations reduce to

$$TEB = TM = TB$$

Turbine and exhaust nozzle performance are also modeled by the steady-state relations which are included in the TRIM routine. These functions relate nozzle airflow (WN), turbine temperature (TT), turbine enthalpy drop (ΔHT), and turbine airflow (WT) with spool speed, burner discharge pressure and temperature (TB and PB), nozzle area (A_g), and ambient nozzle pressure (P_g).

Rotor dynamics are considered at the end of the nozzle section. Angular acceleration of the rotor shaft is computed as a function of enthalpy change,

$$\dot{N} = \frac{K_N \cdot [\Delta(WH)_T - \Delta(WH)_{CD}]}{N} \quad (A-47)$$

where $\Delta(WH)_T$ is the airflow · enthalpy change across the turbine, and $\Delta(WH)_{CD}$ is the airflow · enthalpy change across the compressor. K_N is a constant relating rotor speed in radians per second to rotor speed in revolutions per minute.

Inlet airflow (W_o) and turbine discharge pressure (PT) are computed iteratively in the last section of the program. A gradient search procedure, Newton's method, is used to find W_o and PT since they cannot be obtained directly from the model equations.

A flow chart of the DYNAMIC subroutine is presented in Figure A-2. Details of the procedure are discussed in the following paragraphs.

First, the initial conditions P_o , T_o , P_g , N , TM , W_f , A_g , IGV and BLD and initial estimates W_o , PT , and η_E are read in. These variables are obtained either directly from the TRIM routine or from a previous call to this subroutine.

Then the iteration loop counters are initialized. Both ITER1 and ITER2 are set to zero. ITER2 counts the number of outer loop iterations and ITER1 is a loop counter within the gradient search procedure.

Compressor variables are computed in the next section of the program. The compressor model included in this subroutine is identical to the model included in the TRIM routine.

Inputs to the compressor section include inlet conditions W_o , P_o , and T_o and compressor geometry control positions IGV and BLD. Discharge airflow, WCD, pressure, PCD, temperature, TCD, and enthalpy, HCD, are evaluated in the model. Details of the compressor simulation are shown in the Figure A-2 flow chart and discussed in the TRIM routine documentation.

Next burner inlet airflow is computed from the continuity relation

$$WB = WCD - WTC \quad (A-48)$$

where WTC is the airflow which is extracted from the compressor discharge airflow to cool the turbine vanes. The fuel-to-air ratio in the burner is also evaluated,

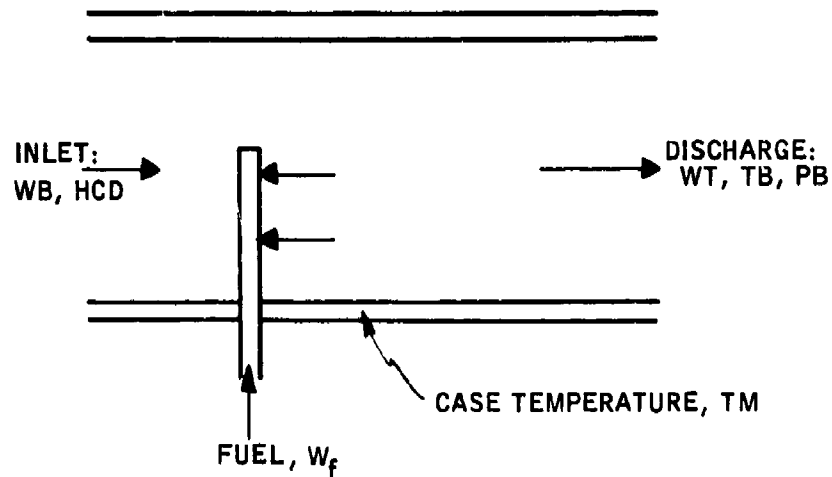
$$FAB = W_f / WB \quad (A-49)$$

Burner enthalpy is calculated from the heat equation.

$$HB = HCD + h_{FUEL} \cdot \eta_{B_0} \cdot FAB \quad (A-50)$$

The term $h_{FUEL} \cdot \eta_{B_0} \cdot FAB$ is the enthalpy increase due to burning of the fuel. The current value of burner efficiency, η_B , is also stored as the variable η_{B_0} in this step.

The time derivative of burner case temperature, \dot{T}_M , is determined in the next step from the thermal capacitance model:



First, the combustion temperature of the gas is computed as a function of FAB and HB,

$$T_{EB} = T_{EB} [FAB, HB] \quad (A-51)$$

Then the rate of change of case temperature, \dot{T}_M , is computed from the heat transfer relation, Equation (A-45):

$$\dot{T}_M = K_{TM} \cdot (T_{EB} - T_M)$$

The constant K_{TM} is a function of the thermal properties of the case material and the term $(T_{EB} - T_M)$ is the temperature gradient at the gas-metal interface. Finally, the temperature of the gas discharged from the burner is computed from Equation (A-46):

$$T_B = T_{EB} - K_{TB} \cdot \dot{T}_M$$

K_{TB} is a constant in this equation. Note that if the burner is not in thermal equilibrium, i. e., $T_M \neq T_{EB}$, the temperature of the gas discharged from the burner, T_B , will not equal the combustion temperature, T_{EB} .

Burner pressure is calculated in the next step.

$$P_B = P_{CD} - \frac{K_B \cdot W_B^2}{P_{CD}} (0.771 T_{CD} - 0.085 T_B) \quad (A-52)$$

This relation was also used to calculate burner pressure in the TEM routine.

Then the value of burner efficiency is recalculated from the experimental data relating efficiency to the variables P_B , T_B and T_{CD} .

$$\eta_B = \eta_B [P_B (T_B - T_{CD})] \quad (A-53)$$

The updated value η_B is compared with the previous value η_{B_0} to determine if the burner simulation is converged. If the error $|\eta_B - \eta_{B_0}|$ is less than $E-10$, the routine proceeds to the turbine simulation. If $|\eta_B - \eta_{B_0}|$ is greater than $E-10$, η_{B_0} is replaced by η_B and the routine returns to step 2.

The first parameter calculated in the turbine section is turbine inlet airflow, W_T . It is computed from the continuity relation

$$W_T = W_B + W_F \quad (A-54)$$

Fuel-to-air ratio in the turbine is also computed at this time.

$$FAT = W_F / (W_T + W_{TC}) \quad (A-55)$$

Note that the turbine cooling airflow, W_{TC} , has been added to turbine inlet airflow in this equation.

Next, turbine inlet airflow is recalculated from the experimental data relating airflow with turbine pressure ratio, burner temperature, and spool speed.

$$WT_{CAL} = \frac{N \cdot PB}{TB} \cdot \left(\frac{WT \cdot TB}{N \cdot PB} \left[\frac{PT}{PB}, \frac{N}{\sqrt{TB}} \right] \right) \quad (A-56)$$

The subscript CAL is attached to this airflow to differentiate between it and turbine airflow computed from the continuity relation.

The difference between WT_{CAL} and WT is then taken.

$$PT_{ERROR} = WT_{CAL} - WT \quad (A-57)$$

The variable name PT_{ERROR} is assigned to this difference because it represents an error in the estimation of turbine discharge pressure, PT . This error is used in the gradient search portion of the program to obtain a better estimate of PT .

Nozzle airflow, WN , is computed from the continuity relation in the next step,

$$WN = WT + WTC \quad (A-58)$$

This parameter is used to evaluate turbine enthalpy, HT , from the heat equation,

$$HT = \frac{WT(HB - \Delta HT) + WTC \cdot HCD}{WN} \quad (A-59)$$

where turbine enthalpy drop, ΔHT , is obtained from the experimental relation

$$\Delta HT = N \cdot \sqrt{TB} \cdot \frac{\Delta HT}{N \sqrt{TB}} \frac{PT}{PB} \cdot \frac{N}{\sqrt{TB}} \quad (A-60)$$

Turbine temperature is then computed from the real gas relation

$$TT = TT [FAT, HT] \quad (A-61)$$

Airflow in the exhaust nozzle is computed in the next subsection. First, the pressure ratio across the nozzle opening is computed,

$$PR_N = \frac{P_8}{P_T} \quad (A-62)$$

The value of this coefficient determines if the nozzle is choked, unchoked or operating under conditions of reversed flow. This information is conveyed to the nozzle airflow equation through the coefficient K_{NOZ} .

$$K_{NOZ} = 0 \quad \text{if } PR_N > 1, \quad \text{reversed flow}$$

$$K_{NOZ} = 0.2588 \quad \text{if } PR_N < 0.528, \quad \text{choked flow} \quad (A-63)$$

$$K_{NOZ} = \left(\frac{P_8}{P_T} \right)^{\frac{1}{\gamma}} \sqrt{1 - \left(\frac{P_8}{P_T} \right)^{\frac{\gamma-1}{\gamma}}} \quad \text{if } 0.528 < PR_N < 1, \quad \text{normal flow}$$

Reversed flow is not allowed in the simulation. If $PR_N > 1$, nozzle airflow is set to zero by assigning $K_{NOZ} = 0$.

After the nozzle coefficient is computed, nozzle airflow is recalculated from the isentropic relation

$$WN_{CAL} = \frac{K_{NOZ} \cdot K_{NA8} \cdot P_T \cdot A_8}{TT} \quad (A-64)$$

This expression is also used in the TRIM routine. The subscript CAL is used to differentiate between nozzle airflow computed from the continuity relation, WN , and airflow computed from this expression, WN_{CAL} .

Next, the difference between WN_{CAL} and WN is calculated

$$W_{ERROR} = WN_{CAL} - WN \quad (A-65)$$

The name W_{ERROR} is assigned to this difference since it represents the error in the estimation of inlet airflow, W_o . This error, together with PT_{ERROR} , is used in the gradient search procedure to obtain better estimates of the parameters W_o and PT .

Rotor acceleration, \dot{N} , is computed next from the conservation of angular momentum.

$$\dot{N} = \frac{K_N \cdot [\Delta(WH)_T - \Delta(WH)_{CD}]}{N} \quad (A-66)$$

The symbols $\Delta(WH)_T$ and $\Delta(WH)_{CD}$ represent the airflow · enthalpy changes across the turbine and the compressor respectively. They are defined as

$$\Delta(WH)_T = WT \cdot \Delta HT$$

$$\begin{aligned} \Delta(WH)_{CD} = H_{CD} \cdot W_{CD} - c_p \cdot T_o \cdot W_o \\ + c_p (WBL_3 \cdot TC_3 + WBL_4 \cdot TC_4 + WBL_5 \cdot TC_5) \end{aligned} \quad (A-67)$$

Finally, the errors PT_{ERROR} and W_{ERROR} are interrogated to determine if the outer iteration loop on the parameters PT and W_o is converged. If the magnitudes of both errors are less than the maximum allowable error, e , the iteration is converged and the subroutine returns to the main program. If the test is not passed, new estimates of the parameters PT and W_o are computed by Newton's method and the subroutine starts over at step 1.

In Newton's method the k+1 gradient step is

$$\underline{Z}^{k+1} = \underline{Z}^k - (\nabla h_{[\underline{Z}^k]})^{-1} \cdot \underline{h}_{[\underline{Z}^k]} \quad (\text{A-68})$$

where \underline{Z} is the vector of unknowns and \underline{h} is the vector of errors. Thus the k+1 estimate of \underline{Z} is computed from the kth estimate of \underline{Z} , the value of the error function \underline{h} evaluated at \underline{Z}^k , and the gradient of the error function ∇h evaluated at \underline{Z}^k . In terms of the parameters PT , W_o , PT_{ERROR} and W_{ERROR} the vectors \underline{Z} , \underline{h} and ∇h are

$$\begin{aligned} \underline{Z}^T &\triangleq \{PT, W_o\} \\ \underline{h}^T &\triangleq \{PT_{\text{ERROR}}, W_{\text{ERROR}}\} \\ \nabla h &\triangleq \begin{bmatrix} \frac{\partial PT_{\text{ERROR}}}{\partial PT} & \frac{\partial PT_{\text{ERROR}}}{\partial W_o} \\ \frac{\partial W_{\text{ERROR}}}{\partial PT} & \frac{\partial W_{\text{ERROR}}}{\partial W_o} \end{bmatrix} \end{aligned} \quad (\text{A-69})$$

Since the partial derivatives in ∇h cannot be computed analytically, they are approximated by finite difference equations in the computer program. For example,

$$\frac{\partial PT_{\text{ERROR}}}{\partial PT} = \frac{PT_{\text{ERROR}}[PT + \Delta PT, W_o] - PT_{\text{ERROR}}[PT - \Delta PT, W_o]}{2 \Delta PT} \quad (\text{A-70})$$

Thus, both positive and negative perturbations in the unknown variable PT are considered. Similar expressions could be written for the other partial derivatives.

The gradient calculation consists of five intermediate steps. In the first step, the errors in the h vector are evaluated, PT_{ERROR} and W_{ERROR} . The partial derivatives with respect to PT , $\partial PT_{\text{ERROR}}/\partial PT$ and $\partial W_{\text{ERROR}}/\partial W_o$, are computed in the second and third steps. These calculations require two steps because both positive and negative perturbations in PT are considered. The other two partial derivatives, $\partial PT_{\text{ERROR}}/\partial W_o$ and $\partial W_{\text{ERROR}}/\partial PT$, are evaluated in the final two steps. New estimates of PT and W_o are also obtained in the last step from the equation

$$\begin{bmatrix} PT_S \\ W_{oS} \end{bmatrix} = \begin{bmatrix} PT \\ W_o \end{bmatrix} + \begin{bmatrix} \frac{\partial PT_{\text{ERROR}}}{\partial PT} & \frac{\partial PT_{\text{ERROR}}}{\partial W_o} \\ \frac{\partial W_{\text{ERROR}}}{\partial PT} & \frac{\partial W_{\text{ERROR}}}{\partial W_o} \end{bmatrix}^{-1} \begin{bmatrix} PT_{\text{ERROR}} \\ W_{\text{ERROR}} \end{bmatrix} \quad (\text{A-71})$$

where the subscript S is used to denote the updated values.

The actual calculations performed in the subroutine are presented in the Figure A-2 flow chart beginning with the computation

$$\text{ITER1} = \text{ITER1} + 1$$

Logic which differentiates between the five steps of the gradient procedure is provided through this variable.

In the first step ($\text{ITER1}=1$), nominal values of the errors PT_{ERROR} and W_{ERROR} are stored under the names F and G . Then the nominal value of PT is increased by the amount ΔPT and the routine is sent back to location number 1.

In the second step (ITER1=2), new values of the errors PT_{ERROR} and W_{ERROR} evaluated for a positive perturbation in PT are stored as FX_+ and GX_+ . Then the current value of PT is decreased by the amount $2\Delta PT$ and the routine is sent back to location 1. This is equivalent to decreasing the nominal value of PT by ΔPT .

New values of the errors PT_{ERROR} and W_{ERROR} evaluated for a negative perturbation in PT are stored as FX_- and GX_- in the third step (ITER1=3). The partial derivatives with respect to PT are evaluated from the finite difference approximations,

$$\frac{\partial PT_{ERROR}}{\partial PT} \triangleq FX = \frac{(FX_+ - FX_-)}{2\Delta PT} \quad (A-72)$$

$$\frac{W_{ERROR}}{PT} \triangleq GX = \frac{(GX_+ - GX_-)}{2\Delta PT}$$

Then PT is returned to its nominal value by adding ΔPT to the current value, and the nominal value of W_0 is increased by ΔW_0 . Finally, the routine is sent to location 1.

Partial derivatives with respect to W_0 are evaluated in steps four and five in the same manner as derivatives with respect to PT were obtained in steps two and three. The resulting finite difference approximations are

$$\frac{\partial PT_{ERROR}}{\partial W_0} \triangleq FY = \frac{(FY_+ - FY_-)}{2\Delta W_0} \quad (A-73)$$

$$\frac{\partial W_{ERROR}}{\partial W_0} \triangleq GY = \frac{(GY_+ - GY_-)}{2\Delta W_0}$$

These partial derivatives, together with the nominal errors F and G, are then used to compute the incremental gradient step defined by

$$\Delta PT_S = \frac{(-F \cdot GY + G \cdot FY)}{D} \tag{A-74}$$

$$\Delta W_{oS} = \frac{(-G \cdot FX + F \cdot GX)}{D}$$

where ΔPT_S is the incremental change in PT and ΔW_{oS} is the incremental change in W_o . The symbol D represents the determinant of the partial derivative matrix

$$D = FX \cdot GY - GX \cdot FY \tag{A-75}$$

Before the gradient step defined by the increments ΔPT_S and ΔW_{oS} is taken, the magnitude of the increments is tested and reduced, if necessary. First the magnitude of ΔPT_S is tested.

$$|\Delta PT_S| < 2\Delta PT$$

If this test is failed, the magnitudes of both ΔPT_S and ΔW_{oS} are reduced by the ratio, $2\Delta PT/|\Delta PT_S|$.

This adjustment reduces only the magnitude of the gradient step; the gradient direction is preserved. If $|\Delta PT_S|$ is smaller than $2\Delta PT$, this adjustment is bypassed.

The magnitude of ΔW_{oS} is also tested in a similar manner. If $|\Delta W_{oS}|$ is greater than $2\Delta W_o$, the gradient step is further reduced by the ratio, $2\Delta W_o/|\Delta W_{oS}|$. If $|\Delta W_{oS}|$ is smaller than $2\Delta W_o$, this magnitude adjustment is bypassed.

Finally, the current values of PT and W_o are updated,

$$\begin{aligned}PT &= PT + \Delta PT_S \\ W_o &= W_o + \Delta W_{oS}\end{aligned}\tag{A-76}$$

the counter ITER1 is reinitialized, and the routine is started anew from location number 1.

Linearizer

This section of the program extracts linear models from the nonlinear engine model. Inputs to the program include steady-state spool speed (N), steady-state engine case temperature (TM), fuel flow (W_f), geometry control positions (A_g , IGV, BLD), inlet pressure and temperature (P_o , T_o), exhaust nozzle discharge pressure (P_g), and perturbation step size (DPERT). The nonlinear engine model is linearized about the equilibrium operating point defined by the first nine input parameters. The tenth input parameter (DPERT) determines the magnitude of the perturbations considered in constructing the linear model.

The linear models obtained are of the form

$$\begin{aligned}\Delta \dot{x} &= F\Delta x + G1\Delta u + G2\Delta \eta \\ \Delta r &= H\Delta x + D1 \Delta u + D2 \Delta \eta\end{aligned}\tag{A-77}$$

where x is the state vector, u is the control vector, η is the disturbance vector, r is the response vector and F , $G1$, $G2$, H , $D1$, $D2$ are coefficient matrices. The Δ symbol is used in these equations to emphasize the fact that the linear models represent perturbations from equilibrium operating conditions.

Engine variables included in the x , u , η , or r vectors are:

- x = N (spool speed)
TM (engine case temperature)
- u = WF (fuel flow)
IGV (inlet guide vane angle)
 A_8 (exhaust area)
BLD (compressor bleed position)
- η = P_c (inlet pressure) (A-78)
 T_o (inlet temperature)
 P_8 (exhaust nozzle discharge pressure)
- r = PCD (compressor discharge pressure)
PT (turbine discharge pressure)
TB (burner temperature)
TT (turbine discharge temperature)

It should be noted that additional variables can be added to the response vector by the user, if desired. The x , u , and η vectors cannot be enlarged as they already contain all the states, controls, and disturbances which are included in the nonlinear engine model.

Coefficients in the matrices F , G_1 , G_2 , H , D_1 and D_2 are computed in the program by a procedure based on the linearization method described in Reference A-2. Briefly, the procedure consists of expanding the nonlinear engine model represented by the nonlinear matrix functions f and h ,

$$\begin{aligned}\dot{x} &= f(x, u, \eta) \\ r &= h(x, u, \eta)\end{aligned}\tag{A-79}$$

in a Taylor series about the equilibrium operating point defined by the steady-state input parameters N , TM , WF , A_g , IGV , BLD , P_o , T_o and P_g . This equilibrium point is denoted as (x_o, u_o, η_o) . Note that substitution of these variables, x_o , u_o , and η_o , into the nonlinear system equations gives

$$f(x_o, u_o, \eta_o) = \dot{x}_o = 0 \quad (A-80)$$

$$h(x_o, u_o, \eta_o) = r_o$$

The result of the Taylor series expansion is

$$\begin{aligned} f(x, u, \eta) - f(x_o, u_o, \eta_o) &= \frac{\partial f}{\partial x}(x_o, u_o, \eta_o)\Delta x + \frac{\partial f}{\partial u}(x_o, u_o, \eta_o)\Delta u \\ &\quad + \frac{\partial f}{\partial \eta}(x_o, u_o, \eta_o)\Delta \eta \end{aligned} \quad (A-81)$$

$$\begin{aligned} h(x, u, \eta) - h(x_o, u_o, \eta_o) &= \frac{\partial h}{\partial x}(x_o, u_o, \eta_o)\Delta x + \frac{\partial h}{\partial u}(x_o, u_o, \eta_o)\Delta u \\ &\quad + \frac{\partial h}{\partial \eta}(x_o, u_o, \eta_o)\Delta \eta \end{aligned}$$

which is equivalent to the linear representation of Equations (A-77) if the following definitions are made.

$$\begin{aligned} \Delta \dot{x} &\triangleq \dot{x} - \dot{x}_o = f(x, u, \eta) - f(x_o, u_o, \eta_o) \\ \Delta r &\triangleq r - r_o = h(x, u, \eta) - h(x_o, u_o, \eta_o) \\ F &\triangleq \frac{\partial f}{\partial x}(x_o, u_o, \eta_o) \\ G1 &\triangleq \frac{\partial f}{\partial u}(x_o, u_o, \eta_o) \\ G2 &\triangleq \frac{\partial f}{\partial \eta}(x_o, u_o, \eta_o) \end{aligned} \quad (A-82)$$

$$H \triangleq \frac{\partial h}{\partial x} (x_0, u_0, \eta_0)$$

$$D1 \triangleq \frac{\partial h}{\partial u} (x_0, u_0, \eta_0)$$

(A-82)

$$D2 \triangleq \frac{\partial h}{\partial \eta} (x_0, u_0, \eta_0)$$

Thus the coefficient matrices are actually matrices of partial derivatives evaluated at the equilibrium point. For example, the F matrix is

$$F = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} (x_0, u_0, \eta_0) & \frac{\partial f_1}{\partial x_2} (x_0, u_0, \eta_0) \dots & \frac{\partial f_1}{\partial x_\eta} (x_0, u_0, \eta_0) \\ \frac{\partial f_2}{\partial x_1} (x_0, u_0, \eta_0) & \frac{\partial f_2}{\partial x_2} (x_0, u_0, \eta_0) \dots & \frac{\partial f_2}{\partial x_\eta} (x_0, u_0, \eta_0) \\ \vdots & \vdots & \vdots \\ \frac{\partial f_n}{\partial x_1} (x_0, u_0, \eta_0) & \frac{\partial f_n}{\partial x_2} (x_0, u_0, \eta_0) \dots & \frac{\partial f_n}{\partial x_\eta} (x_0, u_0, \eta_0) \end{bmatrix}$$

(A-83)

where n is the dimension of the state vector, x.

Written in terms of engine variables, this matrix is

$$F = \begin{bmatrix} \frac{\partial \dot{N}}{\partial N} (x_0, u_0, \eta_0) & \frac{\partial \dot{N}}{\partial TM} (x_0, u_0, \eta_0) \\ \frac{\partial \dot{TM}}{\partial N} (x_0, u_0, \eta_0) & \frac{\partial \dot{TM}}{\partial TM} (x_0, u_0, \eta_0) \end{bmatrix}$$

(A-84)

Similar expressions could be written for the other coefficient matrices.

Since the partial derivatives in these matrices cannot be evaluated analytically, they are computed from finite difference approximations in the computer program. The method is illustrated below for the (1, 1) element in the F matrix.

$$\frac{\partial f_1}{\partial x_1}(x_0, u_0, \eta_0) = \frac{f_1(x_{1c} + \Delta x_1, x_{20}, \dots, x_{\eta_0}, u_0, \eta_0) - f_1(x_{1c} - \Delta x_1, x_{20}, \dots, x_{\eta_0}, u_0, \eta_0)}{2\Delta x_1} \quad (\text{A-85})$$

Thus the procedure involves evaluating the nonlinear-dependent function [$f_1(x, u, \eta)$ in the example] for small perturbations in the independent variable (Δx_1) about the equilibrium point (x_0, u_0, η_0). Both positive and negative perturbations in the independent variable are considered. The results are averaged to compute the final answer.

In the notation used in the computer program, the partial derivatives associated with the coefficient matrices are denoted as

$$\frac{\partial DX_i}{\partial X_j} = \frac{DX2_i - DX1_i}{2\Delta X_j} \quad (\text{A-86})$$

where

$$DX^T = (\dot{N}, \dot{TM}, PCD, PT, TB, TT)$$

$$X^T = (N, TM, WF, IGV, A_8, BLD, P_0, T_0, P_8)$$

Thus the engine variables associated with the nonlinear functions f and h (i. e., time derivatives of the states and responses) are lumped together in the DX vector. The independent variables (i. e., states, controls, and disturbances) are lumped together in the X vector. The symbol DX2 is used in these equations to denote the DX vector evaluated for a positive perturbation in X_j . Similarly, DX1 denotes the DX vector evaluated for a negative perturbation in X_j .

computations in the program proceed in the following order. First, all the derivatives with respect to X_1 are computed,

$$\frac{\partial DX_i}{\partial X_1} \quad i = 1, 2, \dots, NXR$$

where NXR is the dimension of the DX vector. Then all the derivatives with respect to X_2 are computed,

$$\frac{\partial DX_i}{\partial X_2} \quad i = 1, 2, \dots, NXR$$

This procedure continues until all the derivatives have been computed. The last set evaluated is

$$\frac{\partial DX_i}{\partial X_{NXUE}} \quad i = 1, 2, \dots, NXR$$

where $NXUE$ is the dimension of the X vector.

A flowchart of the linearization program is presented in Figure A-3. This flowchart corresponds to the portion of the fortran listing beginning at statement number 511 in the main program (see listing in Table A-2).

First the parameters N , TM , W_f , IGV , A_g , BLD , P_o , T_c and P_g specifying the operating point are input. These variables are obtained from the TRIM section of the main program.

Then the perturbation step size $DPERT$ is read in. The units on $DPERT$ are percent.

Next the integer variable J which corresponds to the subscript j in Equation (A-86) is initialized. It is set to zero.

Then nominal values of the variables in the DX vector are computed in subroutine DYNAMIC. The nominal values obtained are stored in the vector DXN.

In the next step the value of J is increased by one. This means that the partial derivatives with respect to X_1 are to be computed first.

Values of the variables in the DX vector are recalculated for a negative perturbation in X_j , in the following steps. However, before the actual calculations are made, the variable X_j is tested to determine if it is zero. A zero value of X_j implies that a negative perturbation step in X_j cannot be taken, since all of the variables in the X vector must always be positive. Thus if $X_j = 0$, the calculations for a negative perturbation in X_j are bypassed. This condition will be discussed later.

If X_j is nonzero, a negative perturbation in X_j is computed from the relation,

$$\begin{aligned} \text{PERT} &= X_j \cdot \text{DPERT} \\ X_j &= X_j - \text{PERT} \end{aligned} \tag{A-87}$$

Then new values of the variables in the DX vector are calculated in subroutine DYNAMIC. The new values are stored in the vector DX1 and the vector DX is reloaded with the nominal values stored in DXN. Finally, the independent variable X_j is restored to its nominal value by adding PERT back on X_j .

$$X_j = X_j + \text{PERT}$$

At this point the values of variables in the DX vector have been computed for a negative perturbation in X_j . In the next steps the variables in the DX vector are recomputed for a positive perturbation in X_j . However, before these calculations can be made, the value of X_j is again tested. This time X_j is tested to determine if its value is close to one, i. e., if $|X_j - 1|$ is less than PERT.

The condition $X_j = 1$ is important because two of the independent variables, IGV and BLD, are scaled to be in the range 0 - 1.0. Thus if X_j corresponds to one of these variables ($J = 4$ or 6) and X_j is one, then a positive perturbation in X_j cannot be computed since it would give $X_j > 1$. In this case the calculations for a positive perturbation in X_j are bypassed. It should be noted that this test does not affect the other independent variables since they are always much greater than one. The calculations performed if $|X_j - 1|$ is less than PERT will be discussed later.

If $|X_j - 1|$ is greater than PERT, then a positive perturbation in X_j is calculated,

$$X_j = X_j + \text{PERT}$$

Values of the variables in the DX vector are recomputed in subroutine DYNAMIC. The results are stored in DX2 and the vector DX is reloaded with nominal values stored in DXN. Finally, X_j is restored to its nominal value by subtracting PERT from X_j .

$$X_j = X_j - \text{PERT}$$

At this point if both the tests on X_j ,

$$X_j = 0 \quad \text{and} \quad |X_j - 1| < \text{PERT}$$

were failed, the values of the variables in the DX vector for a negative perturbation in X_j are stored in DX1 and the values of the variables for a positive perturbation in X_j are stored in DX2. In this case the values of the partial derivatives with respect to X_j are computed from the finite difference equation,

$$\frac{\partial DX_i}{\partial X_j} = \frac{DX2_i - DX1_i}{2 \text{ PERT}} \quad i = 1, 2, \dots, \text{NXR} \quad (\text{A-88})$$

However, if either of the tests on X_j were passed, then the partial derivatives must be calculated from a different equation because only one of the vectors DX1 or DX2 can be computed.

First consider the case $X_j=0$. In this case only positive perturbations in X_j can be computed. Thus in the calculations beginning at station 3, first a positive perturbation in X_j is computed from

$$\text{PERT} = \text{DPERT}$$

$$X_j = X_j + \text{PERT}$$

(Note that a perturbation in X_j cannot be computed from $\text{PERT} = X_j \cdot \text{DPERT}$ because $X_j=0$.) Then the values of the variables in the DX vector are computed and stored in DX2. Next, X_j is restored to its nominal value

$$X_j = X_j - \text{PERT}$$

and finally the partial derivatives with respect to X_j are computed from the one-sided finite difference equation

$$\frac{\partial DX_i}{\partial X_j} = \frac{DX2_i - DXN_i}{\text{PERT}} \quad i=1, 2, \dots, \text{NXR} \quad (\text{A-89})$$

Similarly, in the case $|X_j - 1| < \text{PERT}$ (corresponding to station 4) the partial derivatives are calculated from the one-sided finite difference equation

$$\frac{\partial DX_i}{\partial X_j} = \frac{DXN_i - DX1_i}{\text{PERT}} \quad i=1, 2, \dots, \text{NXR} \quad (\text{A-90})$$

since values of DX2 cannot be obtained.

After the partial derivatives with respect to X_j have been calculated, control of the routine is transferred to station 2. The variable J is tested to determine if all the partial derivatives have been computed ($J = \text{NXUE}$). If J is less than NXUE the routine returns to station 1 to compute the partial derivatives with respect to X_{j+1} . If $J = \text{NXUE}$, the linearization procedure is finished.

Input Data

The input data required to run the linearization program are described in this subsection. Two groups of data are necessary, the program control group and the component description group.

The program control group includes parameters which define the nominal operating condition for the engine and parameters which control the linearization procedure. This information is input on the four data cards identified below, cards A-E.

Card A

- (1) ERROR This parameter determines the accuracy of the iterations in subroutine DYNAMIC

Card B

- (1) NX Dimension of the state vector
- (2) NU Dimension of the control vector
- (3) NE Dimension of the disturbance vector
- (4) NR Dimension of the response vector
- (5) DPERT Perturbation step size used in the LINEARIZATION routine

Card C

- (1) N Nominal value of spool speed
- (2) WINGS Initial guess for inlet airflow in the TRIM routine
- (3) SPLC Rotor torque load
- (4) IGv Inlet guide vane position
- (5) BLD Compressor bleed position

Card D

- (1) P_o Compressor inlet pressure
- (2) T_o Compressor inlet temperature

The component description group consists of tabulated experimental data which models the steady-state operating characteristics of the engine components. This data is stored on magnetic tape and read into dummy arrays at the beginning of the program. Two function subroutines, FUN1 and FUN2, are used in the program to interpolate between the data points.

The experimental functions contained in this data group are presented in Tables A-3a through A-3x and identified below.

Table Number	Function ID	Experimental Function
A-3a	F11	ABLB = f(BVOB)
A-3b	F12	IGVPR = f(N/N _{max})
A-3c	F13	OGVPR = f(N/N _{max})
A-3d	F15	$\psi_2^P = f(\phi_2)$
A-3e	F16	$\psi_2^T = f(\phi_2)$
A-3f	F17	$\psi_3^P = f(\phi_3)$
A-3g	F18	$\psi_3^T = f(\phi_3)$
A-3h	F19	$\psi_4^P = f(\phi_4)$
A-3i	F110	$\psi_4^T = f(\phi_4)$
A-3j	F111	$\psi_5^P = f(\phi_5)$
A-3k	F112	$\psi_3^T = f(\phi_5)$
A-3l	F113	$\psi_6^P = f(\phi_6)$
A-3m	F114	$\psi_6^T = f(\phi_6)$
A-3n	F115	$\psi_7^P = f(\phi_7)$
A-3o	F116	$\psi_7^T = f(\phi_7)$
A-3p	F117	$\psi_8^P = f(\phi_8)$
A-3q	F118	$\psi_8^T = f(\phi_8)$
A-3r	F119	$\eta_B = f[PB \cdot (TB-TCD)]$
A-3s	F120	KWB = f(N/N _{max})
A-3t	F1	BVOB = f(N/N _{max} , T _o)

Table Number	Function ID	Experimental Function
A-3u	F2	$\psi_2^P = f(\phi_2, \text{IGV})$
A-3v	F3	$\frac{WT \cdot TB}{N \cdot PB} = f\left(\frac{PT}{PB}, \frac{N}{\sqrt{TB}}\right)$
A-3w	F4	$\frac{\Delta HT}{N \sqrt{TB}} = f\left(\frac{PT}{PB}, \frac{N}{\sqrt{TB}}\right)$
A-3x	F5	$\psi_2^T = f(\phi_2, \text{IGV})$

Nominal schedules for the two compressor geometry controls are contained in functions F1 and F11. F1 gives the nominal setting for the IGV (BVOB) as a function of spool speed and compressor inlet temperature. F11 gives the nominal setting for the BLD (ABLB) as a function of BVOB. These actuator schedules were obtained from the NASA component model (Reference A-1). They were not used in the linearization program. Nominal settings for the IGV and BLD are read in on card C of the program control group.

Functions F12 and F13 are correlations of inlet guide vane pressure ratio and outlet guide vane pressure ratio with spool speed.

Pressure and temperature rise coefficients for compressor stages 2 through 8 are contained in functions F15 - F118. These coefficients are functions of a single variable, the flow coefficient ϕ_1 .

Pressure and temperature rise coefficients for the first compressor stage are given by functions F2 and F5.

The coefficients for this stage are functions of both flow coefficient ϕ_1 and inlet guide vane position.

Burner efficiency is presented as a function of the parameter $PB \cdot (TB - TCD)$ in F119 where PB is burner pressure, TB is burner temperature, and TCD is compressor discharge temperature.

The constant KWB is determined as a function of spool speed in F120. This constant is used to determine the pressure loss in the burner.

The function F3 and F4 contain steady-state turbine performance data. The parameter $WT \cdot TB/N \cdot PB$ where WT is turbine airflow is given as a function of turbine pressure ratio and the parameter N/\sqrt{TB} in F3. Turbine enthalpy drop ΔHT divided by $N \cdot \sqrt{TB}$ is given as a function of the same two parameters, PT/PB and N/\sqrt{TB} , in F4.

NONLINEAR ENGINE SIMULATION

The nonlinear engine simulation program is discussed in this subsection. This program is a fortran version of the NASA component model of Reference A-1. A Fortran listing of the program is presented in Table A-1. A listing of the reduced-order component model is presented in Table A-2.

The function of this program is to simulate the transient response of the engine to changes in full flow, exhaust area, inlet guide vane position and compressor bleed position.

A flowchart of the program is presented in Figure A-4. Computations performed in the program are summarized in the following paragraphs. A detailed description of the software is contained in Reference A-1 and in Section II, Volume I of this report.

First, nominal values of spool speed (N), geometry control positions (A_g , IG, BLD), compressor inlet pressure and temperature (P_o , T_o), nozzle

discharge pressure (P_g) and rotor torque load (SPLC) are read in. These parameters define the nominal operating condition for the engine.

A steady-state trim point corresponding to these nominal input parameters is computed next. The fuel flow required to maintain nominal spool speed is calculated in addition to initial values of all the engine states $X(0)$ and responses $r(0)$. It should be noted that the section of the program which performs these calculations is identical to the TRIM routine included in the linearization program. A detailed discussion of the TRIM routine is included in the documentation of the linearization program.

Next the control positions $u(T)$ defining the transient to be simulated are read in. The u vector includes fuel flow, exhaust area, inlet guide vane position and compressor bleed position.

The time increment ΔT and simulation stop time FJNTIME are defined in the following step. Then time is initialized and the time corresponding to the first integration step is computed.

$$T = T + \Delta T$$

Engine dynamics are computed in the next two steps from the nonlinear engine model contained in subroutine DYNAMIC. This nonlinear model is described in detail in Section II, Volume I of this report. Time derivatives of the engine states are computed from the nonlinear function f ,

$$\dot{x}(T) = f[x(T), r(T), u(T)]$$

and updated values of the responses are computed from the nonlinear function h

$$r(T + \Delta T) = h[x(T), r(T), u(T)]$$

The derivatives are then integrated with a four point Runge Kutta integration routine to determine the value of the states at time $T + \Delta T$.

$$x(T + \Delta T) = x(T) + \int_T^{T+\Delta T} \dot{x}(T)dT$$

In the final step in the program, the current value of time is compared with the stop time. If $T \geq \text{FINTIME}$, the program exits from the integration loop. If $T < \text{FINTIME}$, the routine returns to station 1 for an additional integration step.

Table A-1. Nonlinear Engine Simulation Program

```

AF3RTRAN LS,CO
11: DIMENSION IGV(42),A(20),PV(20),TV(20),YY1(14),XX1(17),ZZ1(196)
21: DIMENSION L(20),V(20),KGAL(20),KVBL(20),KNR(8),RAD(8),KRAD(8)
31: DIMENSION KA(30)
41: DIMENSION THV(20),WD(20),WV(20),KBLD(8),DPRB(14)
51: COMMON/DTATA/TIME,DT,ISTEP,NICST
61: COMMON/DATA/X(39),I(3),ETA(3),DXN(40),DX(40),DX7(40),CLM(40),
71: KVBL,KGAL,KVBLG,KVBLGD,KGALGD,RTHO,ABL,K9GV,WTC,KVBLCD,KGALCD,TR,THD
81: K3ALB,KVBLB,HT,TT,PT,K4,K2,WT,MCD,PU,WN,KNAB,KSPED,KFIGV,K3,IGV
91: 3R,TVO
101: REAL ICTWVO,ICWVO
111: REAL K5,K8,NC1,ICN,NC1N,IGVPR,IGV,K1,K3,K4,K6,K7,L,KGAL,KVBL
121: REAL KVBLGD,KGALGD,KVBLCD,KGALCD,KGALB,KWB,KVBLB,KVBLT,KSPED
131: REAL KFIGV,NCX,KNR,KRAD,KA,K9GV,KNAB,NRTTB,ICPYO,ICWD0,ICWD1,ICWV1
141: REAL ICTWV1,ICWD2,ICWV2,ICTWV2,ICWD3,ICWV3,ICTWV3,ICWD4,ICWV4
151: REAL ICTWV4,ICWD5,ICWV5,ICTWV5,ICWD6,ICWV6,ICTWV6,ICWD7,ICWV7
161: REAL ICTWV7,ICWD8,ICWV8,ICTWV8,ICWD9,ICWV9,ICWD0G,ICW0G,ICWCD
171: REAL ICWCD0,ICWB,ICPB,ICHB,ICRY,ICRHT,ICW5DP,NC2,NC3,NC4,NC5,NC6
181: REAL NC7,NC8,NDEMD,ITER,IMPL,INTGR,KIC
191: REAL KA8,ICWD0G,N,NDT
201: REAL KBLD,K2,NRAT,KG/LIG,KVBLIG
211: EQUIVALENCE (ICWD0,W0D0,X(1)),(ICWV0,WV0,X(2)),(ICTWV0,TW0,X(3))
221: 1),(ICWD1,W1D1,X(4)),(ICWV1,WV1,X(5)),(ICTWV1,TW1,X(6))
231: 2),(ICWD2,W2D2,X(7)),(ICWV2,WV2,X(8)),(ICTWV2,TW2,X(9))
241: 3),(ICWD3,W3D3,X(10)),(ICWV3,WV3,X(11)),(ICTWV3,TW3,X(12))
251: 4),(ICWD4,W4D4,X(13)),(ICWV4,WV4,X(14)),(ICTWV4,TW4,X(15))
261: 5),(ICWD5,W5D5,X(16)),(ICWV5,WV5,X(17)),(ICTWV5,TW5,X(18))
271: 6),(ICWD6,W6D6,X(19)),(ICWV6,WV6,X(20)),(ICTWV6,TW6,X(21))
281: 7),(ICWD7,W7D7,X(22)),(ICWV7,WV7,X(23)),(ICTWV7,TW7,X(24))
291: 8),(ICWD8,W8D8,X(25)),(ICWV8,WV8,X(26)),(ICTWV8,TW8,X(27))
301: 9),(ICWD9,W9D9,X(28)),(ICWV9,WV9,X(29)),(ICTWV9,TW9,X(30))
311: A),(ICWCD0,WCD0,X(31)),(ICWCD,WCD,X(32)),(ICTWCD,TWCD,X(33))
321: B),(ICWB,WB,X(34)),(ICPB,PB,X(35))
331: C),(ICHB,HB,X(36)),(ICRHT,RHT,X(37)),(ICRY,RT,X(38))
341: D),(ICN,N,X(39)),(WFO,U(1)),(BVB,U(2)),(AB,U(3)),(P2,ETA(1))
351: E),(T2,ETA(2)),(PB,ETA(3))
361: EQUIVALENCE (WD0DT ,DX(1)),(WV0DT ,DX(2)),(TW0DT ,DX(3)),
371: 1,(WD1DT ,DX(4)),(WV1DT ,DX(5)),(TW1DT ,DX(6)),
381: 2,(WD2DT ,DX(7)),(WV2DT ,DX(8)),(TW2DT ,DX(9)),
391: 3,(WD3DT ,DX(10)),(WV3DT ,DX(11)),(TW3DT ,DX(12)),
401: 4,(WD4DT ,DX(13)),(WV4DT ,DX(14)),(TW4DT ,DX(15)),
411: 5,(WD5DT ,DX(16)),(WV5DT ,DX(17)),(TW5DT ,DX(18)),
421: 6,(WD6DT ,DX(19)),(WV6DT ,DX(20)),(TW6DT ,DX(21)),
431: 7,(WD7DT ,DX(22)),(WV7DT ,DX(23)),(TW7DT ,DX(24)),
441: 8,(WD8DT ,DX(25)),(WV8DT ,DX(26)),(TW8DT ,DX(27)),
451: 9,(WD9GDT ,DX(28)),(WV9GDT ,DX(29)),(TW9GDT ,DX(30)),
461: A,(WCD0DT ,DX(31)),(WCDDT ,DX(32)),(TWCD0DT ,DX(33)),
471: B,(WB0DT ,DX(34)),(PBDT ,DX(35)),
481: C,(HBDT ,DX(37)),(RHTDT ,DX(38)),(RTDT ,DX(39)),
491: D,(NDT ,DX(39))
501: REWIND 3
511: 8099 CONTINUE
521: RND1=0
531: RND2=0
541: READ(5,8030)NX,NU,NE,DPERT
551: 8030 FORMAT(3I2,G12.5)
561: REWIND 7
571: DATA (DPRB(1),1=1,14)/.60,7.26E-4,.70,7.07E-4,.80,6.98E-5,.85,6.9E-5
581: 1,4,.50,6.96E-4,.97,6.56E-4,1.0,7.38E-4/
591: DATA (KBLD(1),1=1,8)/2.0,1.1025,1.0572,1.0411,3.0,7

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

60: DATA (KVAL(1),I=1,8)/25542.,27942.,27247.,26407.,240R.,21877.,221
61: 155.,22439./
62: DATA (KVAL(1),I=1,8)/1.9107,3.3711,4.9797,7.0830,9.3017,11.2953,13
63: 1.,727.15,1219/
64: DATA (TV(1),I=1,20)/5.1600.,15.,518.7/
65: DATA (TV(1),I=1,20)/20.,10./
66: DATA (WC(1),I=1,20)/20.,30./
67: DATA (WC(1),I=1,20)/20.,01/
68: READ(7)(IGV(I),I=1,18)
69: F11=FN1SET(1,IGV,19,1,1)
70: READ(7)(IGV(I),I=1,38)
71: F14=FN1SET(4,IGV,19,4,5)
72: READ(7)(IGV(I),I=1,20)
73: F12=FN1SET(2,IGV,10,2,2)
74: READ(7)(IGV(I),I=1,18)
75: F13=FN1SET(3,IGV,9,3,3)
76: READ(7)(IGV(I),I=1,40)
77: F15=FN1SET(5,IGV,20,6,7)
78: READ(7)(IGV(I),I=1,42)
79: READ(5,677)(IGV(I),I=1,42)
80: R77 FOR:AT(10F3.4)
81: F16=FN1SET(6,IGV,21,8,9)
82: READ(7)(IGV(I),I=1,34)
83: F17=FN1SET(7,IGV,17,10,11)
84: READ(7)(IGV(I),I=1,38)
85: READ(5,677)(IGV(I),I=1,38)
86: F18=FN1SET(8,IGV,19,12,13)
87: READ(7)(IGV(I),I=1,36)
88: F19=FN1SET(9,IGV,18,14,15)
89: READ(7)(IGV(I),I=1,36)
90: IGV(3)=53
91: READ(5,677)(IGV(I),I=1,40)
92: F110=FN1SET(10,IGV,20,16,17)
93: READ(7)(IGV(I),I=1,32)
94: F111=FN1SET(11,IGV,16,18,19)
95: READ(7)(IGV(I),I=1,32)
96: READ(5,677)(IGV(I),I=1,36)
97: F112=FN1SET(12,IGV,18,20,21)
98: READ(7)(IGV(I),I=1,26)
99: F113=FN1SET(13,IGV,13,22,23)
100: READ(7)(IGV(I),I=1,26)
101: READ(5,677)(IGV(I),I=1,26)
102: F114=FN1SET(14,IGV,13,24,25)
103: READ(7)(IGV(I),I=1,30)
104: F115=FN1SET(15,IGV,15,26,27)
105: READ(7)(IGV(I),I=1,24)
106: READ(5,677)(IGV(I),I=1,26)
107: F116=FN1SET(16,IGV,13,28,29)
108: READ(7)(IGV(I),I=1,30)
109: F117=FN1SET(17,IGV,15,30,31)
110: READ(7)(IGV(I),I=1,32)
111: READ(5,677)(IGV(I),I=1,32)
112: F118=FN1SET(18,IGV,16,32,33)
113: READ(7)(IGV(I),I=1,28)
114: F119=FN1SET(19,IGV,14,34,35)
115: READ(7)(IGV(I),I=1,26)
116: F120=FN1SET(20,IGV,7,36,37)
117: READ(7)(IGV(I),I=1,40)
118: F121=FN1SET(21,IGV,20,38,39)
119: READ(7)(A(I),I=1,20)
120: READ(7)(L(I),I=1,20)

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

121:      HEAD(7)(YV(I),I=1,20)
122:      HEAD(7)(IGV(I),I=1,20)
123:      HEAD(7)(IGV(I),I=1,20)
124:      HEAD(7)(KAD(I),I=1,8)
125:      HEAD(7)(KAC(I),I=1,8)
126:      HEAD(7)(FV(I),I=1,20)
127:      HEAD(7)(IGV(I),I=1,20)
128:      HEAD(7)(YY1(I),I=1,5)
129:      HEAD(7)(AX1(I),I=1,13)
130:      HEAD(7)(ZZ1(I),I=1,65)
131:      F1 = FNPSET(1,XX1,YY1,ZZ1,17,5,1,2)
132:      HEAD(7)(KA(I),I=1,30)
133:      HEAD(7)(YY1(I),I=1,4)
134:      HEAD(7)(AX1(I),I=1,17)
135:      HEAD(7)(ZZ1(I),I=1,64)
136:      F2 = FNPSET(2,XX1,YY1,ZZ1,17,4,3,4)
137:      HEAD(7)(YY1(I),I=1,14)
138:      HEAD(7)(AX1(I),I=1,14)
139:      HEAD(7)(ZZ1(I),I=1,196)
140:      DJ = 9773 I=1,43
141:      IJ = 196 * I
142:      JJ = IJ + 1
143:      9773 ZZ1(JJ) = ZZ1(I)
144:      ZZ1(148) = .0544
145:      WRITE(9,1602)
146:      WRITE(9,1601)(YY1(I),I=1,14)
147:      WRITE(9,1603)
148:      WRITE(9,1601)(XX1(I),I=1,14)
149:      WRITE(9,1603)
150:      WRITE(9,1601)(ZZ1(I),I=1,196)
151:      WRITE(9,1602)
152:      F3 = FNPSET(3,XX1,YY1,ZZ1,14,14,5,6)
153:      HEAD(7)(YY1(I),I=1,14)
154:      HEAD(7)(AX1(I),I=1,14)
155:      HEAD(7)(ZZ1(I),I=1,196)
156:      1601 FORMAT(13F9.4)
157:      1602 FORMAT(1H)
158:      1603 FORMAT(//)
159:      ZZ1(94) = .09 * I
160:      F4 = FNPSET(4,XX1,YY1,ZZ1,14,14,7,8)
161:      READ(5,8066)(YY1(I),I=1,4)
162:      READ(5,8067)(XX1(I),I=1,17)
163:      READ(5,8067)(ZZ1(I),I=1,65)
164:      F5 = FNPSET(5,XX1,YY1,ZZ1,17,4,9,10)
165:      C
166:      C SET PARAMETERS
167:      C
168:      KAK=0
169:      N5SP=1
170:      TIGS=1700.
171:      READ(5,8066)IRAT,AB,TIGS,WINGS,FABIS,SPLC
172:      8066 FORMAT(6G12.5)
173:      WRITE(9,8066)IRAT,AB,TIGS,WINGS,FABIS,SPLC
174:      F2=14.7
175:      T2=512.7
176:      WDEL=.1
177:      <1 = 3.14159/160.
178:      <2 = 7.032 * 17.5 * 35 / (K1 * K1)
179:      <3 = SQRT(5) * 4.7
180:      <4 = (53 * 35 * 12) / 17400.
181:      <5 = 17.5 * 37.

```


Table A-1. Nonlinear Engine Simulation Program (Continued)

```

182:      <VALIG=3.31
183:      <SALRG = 2243.
184:      <VALRG = 15.11
185:      <SALCO = 6730.
186:      <VALCO = 1.981
187:      <SALR = 5470.
188:      <VALR = 2.649
189:      <NB = .0004445
190:      <VALT = 14.15
191:      <SPEED = 138400.
192:      ICN,NRAT,16500.
193:      P2,P2
194:      P8,P2
195:      TVO = T2
196:      RTHO = SQRT(TVO/518.7)
197:      NC1 = ICN/RTHO
198:      NC1N = NC1/16500.
199:      BV0 = FUN2(1,NC1N,TVO,1)
200:      ABL,FUN1(1,BV0,1)
201:      IGVPR,FUN1(2,NC1N,2)
202:      BSVPR,FUN1(3,NC1N,3)
203:      TV(10) = T2
204:      III=0
205:      ITER1=0
206:      ITER2=0
207:      ITER3=0
208:      DO 40 K=1,NSSP
209:      FAB,FABLS
210:      T3,TRGS
211:      WIN,WINGS
212:      DAINX=.1
213:      PV(10) = P2+IGVPR = .005*P2
214:      99 CONTINUE
215:      WD(10)=WIN-FLB*AT(K-1)*WDEL
216:      ITER1=ITER1+1
217:      <F1GV,KCALIG=(P2-PV(10))/(WD(10)*WD(10))
218:      WBL=0.
219:      WBLTBL=0.
220:      IF(GENSE SWITCH 3)8073,8074
221:      8074 CONTINUE
222:      IF(III.NE.1) GOTO 5901
223:      8073 CONTINUE
224:      J = 0
225:      WRITE(6,50) ICN,ABL,BV0,IGVPR,BSVPR
226:      WRITE(6,51)
227:      WRITE(6,52) J,PV(10),TV(10),WD(10)
228:      5901 CONTINUE
229:      DO 20 I=1,18
230:      J = I-10
231:      DELX = FV(I-1)/14.7
232:      RTHX = SQRT(TV(I-1)/518.7)
233:      NCX = ICN/RTHX
234:      WD(I)=WD(I-1)*WBL
235:      FPX=WD(I)*RTHX/(DELX*A(I))
236:      VZTX=KA(J)+FPX*(KA(J+10)+FPX*KA(J+20))
237:      <KAD(J) = K1*RAD(J)
238:      PHIX = VZTX/(K*RAD(J)*NCX)
239:      GB TA (1,2,3,4,5,6,7,R),J
240:      1 CONTINUE
241:      PSIPX = FUN2(2,PHIX,BV0,3)
242:      PSITX=FUN2(5,PHIX,BV0,9)

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

243:      GOTO 10
244:      2 CONTINUE
245:      PSIPX=FUN1(5,PHIX,8)
246:      PSITX=FUN1(6,PHIX,8)
247:      GOTO 10
248:      3 CONTINUE
249:      PSIPX=FUN1(7,PHIX,10)
250:      PSITX=FUN1(8,PHIX,12)
251:      GOTO 10
252:      4 CONTINUE
253:      PSIPX=FUN1(9,PHIX,14)
254:      PSITX=FUN1(10,PHIX,16)
255:      GOTO 10
256:      5 CONTINUE
257:      PSIPX=FUN1(11,PHIX,18)
258:      PSITX=FUN1(12,PHIX,20)
259:      GOTO 10
260:      6 CONTINUE
261:      PSIPX=FUN1(13,PHIX,22)
262:      PSITX=FUN1(14,PHIX,24)
263:      GOTO 10
264:      7 CONTINUE
265:      PSIPX=FUN1(15,PHIX,26)
266:      PSITX=FUN1(16,PHIX,28)
267:      GOTO 10
268:      8 CONTINUE
269:      PSIPX=FUN1(17,PHIX,30)
270:      PSITX=FUN1(18,PHIX,32)
271:      10 CONTINUE
272:      KNR(J)=(ICNRAD(J))*2/KP
273:      PV(I) = PV(I-1)*(1+PSIPX*KNR(J)/TV(I-1))*3.5
274:      TV(I) = TV(I-1)+KNR(J)*PSITX
275:      TWV(I)=FV(I)/KVL(J)
276:      WV(I)=TWV(I)/TV(I)
277:      PH = PV(I)/PV(I-1)
278:      WBL=KBLD(J)*ABL*PV(I)/SQRT(TV(I))
279:      IF(J.EQ.3) ABL3=WBL
280:      IF(J.EQ.4) ABL4=WBL
281:      IF(J.EQ.5) ABL5=WBL
282:      WBLTBL=WBLTFL+WBL*TV(I)
283:      IF(SENSE SWITCH 3)8075,8076
284:      8076 CONTINUE
285:      IF(III.NE.1) GOTO 5902
286:      8075 CONTINUE
287:      WRITE(6,52)J,PV(I),TV(I),WD(I),WBL,PSITX,VZTX,PHIX,PSIPX,PR
288:      5902 CONTINUE
289:      20 CONTINUE
290:      J = 10
291:      P0GV = PV(1R)*0.0VPR
292:      T0GV = TV(1R)
293:      W0GV=WD(10)
294:      K0GV = *GALRG*(FV(1R)-P0GV)/W0GV**2
295:      T*0G = P0GV/KVBLRG
296:      W0GG = T*0G/T0GV
297:      PR23=P0GV/P2
298:      TR23=T0GV/TP
299:      EFF23=(PR23**285-1.)/(TR23-1.)
300:      IF(SENSE SWITCH 3)8077,8078
301:      8078 CONTINUE
302:      IF(III.NE.1) GOTO 5903
303:      8077 CONTINUE

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

304:      WRITE(6,53)
305:      WRITE(6,52)J,PBGV,TBGV,KBGV,WBLTBL,EFF23,PR23,TR23
306: 5903 CONTINUE
307:      J = 11
308:      PCD = PBGV
309:      TCD = TBGV
310:      CALL PRBCBM(0,TCD,CPCD,GMCD,GMCDX,HCD,IFA)
311:      WCD = KBGV
312:      TWCD = FCD/KVRLCD
313:      WUCD = TWCD/TCD
314:      WTC = .033*WD(10)
315:      DLWHC = HCD*WCD*.24*(WBLTBL-WD(10)*TV(10))+SPLC
316:      KNAR = FUN1(21,ICN,38)*.975
317:      IF (SENSE SWITCH 3) 8170,8171
318: 8171 CONTINUE
319:      IF (III*NE.1) GOTO 5904
320: 8170 CONTINUE
321:      WRITE(6,54)
322:      WRITE(6,52)J,PCD,TCD,WCD,WTC,DLWHC,KNAB
323: 5904 CONTINUE
324:      J = 12
325:      KWB = FUN1(20,NRAT,36)
326:      WB = WCD*WTC
327:      IF (ITER1.EQ.1) PT = .35*PCD
328:      DPTX = 1
329: 220 CONTINUE
330:      ITER2 = ITER2 + 1
331:      DTBX = .25
332:      WTOLD = (1.+FAB)*WB
333: 221 NRAT = ICN/SQRT(TB)
334:      ITER3 = ITER3 + 1
335:      DELPB = KWB*WB**2/PCD*(.771*TCD+.085*TB)
336:      PB = PCD-DELPB
337:      PSDLTR = PB*(TB-TCD)
338:      ETAB = FUN1(19,PSDLTR,36)
339:      PTPB = PT/PB
340:      DPTNTB = FUN2(4,PTPB,NRAT,7)
341:      DPT = DPTNTB*(ICN/1000)*SQRT(TB)
342: 222 CALL PRBCBM(FAB,TB,CPS,GMB,SMBX,H4,IFA)
343:      IF (SENSE SWITCH 4) 8098,8091
344: 8091 CONTINUE
345:      WT1 = WB*(1.650+.05*ETAB-HCD)/(1.650+.05*ETAB-H4)
346:      IF (IFA.GT.0) GOTO 8071
347:      WTERR = ABS(WT1-WTOLD)
348:      IF (WTERR.GT..0005) GOTO 223
349: 8072 CONTINUE
350:      WF = WT1*WB
351:      FAB = WF/WB
352:      KN = WT1*WTC
353:      GOTO 224
354: 8071 IF (WT1.LT.WB) WT1 = WB
355:      IF (WT1.GT.(WB*1.067623)) WT1 = .067623*WB
356:      WTOLD = WT1
357:      GOTO 8072
358: 223 WTOLD = (WT1+.TOLD)*.5
359:      WF = WTOLD*WB
360:      FAB = WF/WB
361:      GOTO 222
362: 224 CONTINUE
363:      WTC = (WB-DPT)/WCD + WTC/WB*HCD
364:      WBR = (DLWHC+KN*HT-HCD*WTC)/WT1

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

365:      TBERR=HBR-WH
366:      IF(TBERR.GT..0005) GOTO 225
367:      IF(TBERR.LT..0005) GOTO 228
368:      GOTO 229
369: 225 IF(DTBX)226,226,227
370: 226 DTBX=DTBX*.5
371: 227 TB=TB+DTBX
372:      GOTO 221
373: 228 IF(DTBX)227,227,226
374: 229 CONTINUE
375:      POPT=PO/PT
376:      IF(POPT=.528)233,233,230
377: 230 IF(POPT=1.)232,231,231
378: 231 WNTKNP=0
379:      GOTO 234
380: 232 WNTKNP=POPT*(1./1.4)*SQRT(1.-POPT*(.4/1.4))
381:      GOTO 234
382: 233 WNTKNP=.2588
383: 234 CONTINUE
384:      WITNPB=FUN2(3,PTPB,NRTTB,5)
385:      WITNPB=PB/TB*ICN
386:      PTERR=WT2-WT1
387: 240 IF(PTERR.GT..0005) GOTO 241
388: 241 IF(PTERR.LT..0005) GOTO 245
389:      GOTO 250
390: 241 IF(DPTX)242,242,243
391: 242 DPTX=DPTX*.5
392: 243 PT=PT+DPTX
393:      GOTO 220
394: 245 IF(DPTX)243,243,242
395: 250 CONTINUE
396:      FAT=WF/WN
397:      TT=TFNH(1,FAT,HT,TV)
398:      IF(SENSE SWITCH 4)8092,8092
399: 8092 CONTINUE
400:      WNX=(KNAB*PT+WNTKNP*AB)/SQRT(TT)
401:      WNERR=WNX-WN
402:      IF(III.EQ.1) GOTO 60
403:      IF(WNERR.GT..005) GOTO 5951
404:      IF(WNERR.LT..005) GOTO 5955
405:      III=1
406:      GOTO 99
407: 5951 IF(DWINX)5952,5952,5953
408: 5952 DWINX=DWINX*.5
409: 5953 WIN=WIN+DWINX
410:      GOTO 99
411: 5955 IF(DWINX)5953,5953,5952
412: 60 CONTINUE
413:      DLWMT=HE*WT1+HCD*WT2+HT*WN
414:      WFM = 3600*WF
415:      WRITE(6,56)
416:      WRITE(6,52) J,FB,TB,WB,ETAB,WB,PTPB,NRTTB,DHTNTB,WITNPB
417:      J = 13
418:      WRITE(6,57)
419:      WRITE(6,52)J,PT,TT,WT1,WF,HT,DLWMT,WN,WNTKNP,AB
420:      WRITE(6,2108)ITER1,ITER2,ITER3
421: 2108 FORMAT(1H0,5X,8HITER1 = 15,5X,8HITER2 = 15,5X,8HITER3 = 15)
422: 50 FORMAT(1H1,4X,4MN = ,F8.2,4X,6HABL = ,F8.4,4X,6HGV0 = ,F8.4,4X,
423: 8HIGVPR = ,F8.4,4X,8HOGVPR = ,F8.4)
424: 51 FORMAT(1H0,3X,1HJ,5X,5HPV(I),9X,5HTV(I),9X,5HWD(J),8X,6HMBL(J),8X
425: 6HPSIT J,8X,6HVZT(J),8X,6HPI(J),7X,7HPSIP(J),9X,5HPR(J))

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

424: 52 FORMAT(1H0,14,9G14.5)
427: 53 FORMAT(1H0,3X,1FJ,6X,4HPRGV,10X,4HT0GV,10X,4HW0GV,8X,6HBLTBL,9X,5
428: 1HEFF23,10X,4HPR23,10X,4HTR23)
429: 54 FORMAT(1H0,3X,1FJ,7X,3HPCD,11X,3HTCD,11X,3HWCD,11X,3HATC,9X,5HDLWH
430: 1C,10X,4HKNAK)
431: 56 FORMAT(1H0,3X,1FJ,8X,2HPB,12X,2HTB,12X,2HAB,10X,4HETAB,12X,2HAB,
432: 110X,4HPTP,9X,5HRTTB,8X,6HDHTNB,8X,6HHTNPB)
433: 57 FORMAT(1H0,3X,1FJ,8X,2HPT,12X,2HTT,12X,2HNT,12X,2HNT,9X,5
434: 1HDLWHT,12X,2HNT,8X,6HNTKNP,12X,2HAB)
435: ICPVO = PV(10)
436: ICWD0=WD(10)
437: ICWD1=WD(11)
438: ICWV1=VW(11)
439: ICTWV1=TWV(11)
440: ICWD2=WD(12)
441: ICWV2=VW(12)
442: ICTWV2=TWV(12)
443: ICWD3=WD(13)
444: ICWV3=VW(13)
445: ICTWV3=TWV(13)
446: ICWD4=WD(14)
447: ICWV4=VW(14)
448: ICTWV4=TWV(14)
449: ICWD5=WD(15)
450: ICWV5=VW(15)
451: ICTWV5=TWV(15)
452: ICWD6=WD(16)
453: ICWV6=VW(16)
454: ICTWV6=TWV(16)
455: ICWD7=WD(17)
456: ICWV7=VW(17)
457: ICTWV7=TWV(17)
458: ICWD8=WD(18)
459: ICWV8=VW(18)
460: ICTWV8=TWV(18)
461: ICTW0G = TW0G
462: ICW0GV = W0GV
463: ICW00G = W00G
464: ICTWCD = TWCD
465: ICWCD = WCD
466: ICWDCD = WDCD
467: ICWB = WB
468: ICPB = PB
469: ICMB = MB
470: ICRT = PT/K4/TT
471: ICRT = HT/ICRT
472: ICTWVO = ICPVO/KVBLIG
473: ICWVO = ICTWVO/TVO
474: DO 298 I=1,N
475: II=I+10
476: KGAL(II)=KGAL(I)
477: 298 KVBL(II)=KVBL(I)
478: READ(5,9011)(DX(I),I=1,NX)
479: 9011 FORMAT(8E10.4)
480: IF(DX(1).NE.0.) GOTO 9701
481: CCC=0.
482: GOTO 9702
483: 9701 CONTINUE
484: DO 9012 I=1,NX
485: 9012 DX(I)=2.*DX(I)
486: CCC=.01*X(1)/DX(1)

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

487: 9702 CONTINUE
488:   DO 9013 I=1,NX
489:     DX(I)=DX(I)+CCC
490:   9013 X(I)=X(I)+DX(I)
491:     WRITE(3)(X(I),I=1,NX)
492:   READ(5,9014)DELT,FINTIME,PRDEL,BUTDEL
493:   9014 FORMAT(4G12.5)
494:     DT=DELT
495:     TIME=0.
496:     DTM=DELT*.5
497:     NXUE=NX,NU=NE
498:     SIGN=1.
499:     WRITE(9,8061)
500:     WRITE(9,8060)(X(I),I=1,NX)
501:   8060 FORMAT(E20.8)
502:   8061 FORMAT(1H1)
503:   3333 CONTINUE
504:     ISTEP=0
505:     NICOT=0
506:     CALL DYNAM(A,PV,TV,KGAL,KV9L,KNR,RAD,KRAD,KA,TAV,WD,HV,KBLD,OPRR)
507:     TIME=TIME+DELT
508:     IF(ABS(TIME-PRDEL).GT..000001) GOTO 3333
509:     PRDEL=PRDEL+BUTDEL
510:     WRITE(3)(X(I),I=1,NX)
511:     IF(ABS(TIME-FINTIME).GT..000001) GOTO 3333
512:   8098 CONTINUE
513:     PAUSE
514:     GOTO 8099
515:     END

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1:      FUNCTION INTGRL(IC,DXDT)
2:      COMMON/TDATA/TIME,DT,ISTEP,NICOT
3:      DIMENSION G(40,4),XK(40)
4:      REAL IC,INTGRL
5:      IF(ISTEP.NE.0) GOTO 2
6:      IF(NICOT.NE.0) GOTO 1
7:      NI=39
8:      1 NICOT=NICOT+1
9:      G(NICOT,1)=DT*DXDT
10:     XK(NICOT)=IC
11:     INTGRL=XK(NICOT)+.5*G(NICOT,1)
12:     IF(NICOT.EQ.NI) ISTEP=1
13:     RETURN
14:     2 NICOT=NICOT+1
15:     GOTO(3,4,5),ISTEP
16:     3 G(NICOT,2)=DT*DXDT
17:     INTGRL=XK(NICOT)+.5*G(NICOT,2)
18:     IF(NICOT.EQ.NI) ISTEP=2
19:     RETURN
20:     4 G(NICOT,3)=DT*DXDT
21:     INTGRL=XK(NICOT)+G(NICOT,3)
22:     IF(NICOT.EQ.NI) ISTEP=3
23:     RETURN
24:     5 G(NICOT,4)=DT*DXDT
25:     INTGRL=XK(NICOT)+(G(NICOT,1)+2.*G(NICOT,2)+2.*G(NICOT,3)+G(NICOT,4
26:     1))/6.
27:     IF(NICOT.EQ.NI) ISTEP=4
28:     RETURN
29:     END

```

```

1:      FUNCTION TFMH(NX,FAH,HX,TV)
2:      DIMENSION TV(20)
3:      DTX=50.
4:      TX=TV(NX)
5:      51 CALL PRBCGM(FAH,FX,CPX,GMX,GMXX,HX1,IFA)
6:      IF(IFA.GT.1) GOTO 70
7:      TXERR=HX-HX1
8:      IF(TXERR.GT..001) GOTO 52
9:      IF(TXERR.LT..001) GOTO 55
10:     GOTO 60
11:     52 IF(DTX)53,53,54
12:     53 DTX=-DTX*.5
13:     54 TX=TX+DTX
14:     GOTO 51
15:     55 IF(DTX)54,54,53
16:     60 CONTINUE
17:     70 CONTINUE
18:     TFMH=TX
19:     TV(NX)=TX
20:     RETURN
21:     END

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1:      SUBROUTINE PRACGM(FARX,TEX,CP,GM,GMX,H,IFA)
2:      IFA=0
3:      IF(FARX.GT.0.) GOTO 2
4:      FARX=0.
5:      IFA=1
6:      GOTO 3
7:      2 IF(FARX.LT..067623) GOTO 3
8:      FARX=.067623
9:      IFA=1
10:     3 IF(TEX*1500.) 20,10,5
11:     5 IF(TEX.LT.4000.) GOTO 7
12:     TEX=4000.
13:     IF(IFA.EQ.1) GOTO 50
14:     IFA=2
15:     GOTO 16
16:     50 IFA=3
17:     GOTO 16
18:     7 IF(TEX*2300.) 9,14,8
19:     8 IF(TEX*2500.) 14,16,16
20:     9 IF(TEX*2000.) 10,12,12
21:     10 CPA = .26*4*2.6E-5*(TEX-1500.)
22:     WA = (.22519*1.292E-5*TEX)*TEX+2.3733
23:     GM TH 40
24:     12 CPA = .2773R+1.82E-5*(TEX-2000.)
25:     WA = (.22519*1.292E-5*TEX)*TEX+2.3733
26:     GM TH 40
27:     14 CPA = .2773R+1.82E-5*(TEX-2000.)
28:     WA = (.25987*5.36E-6*TEX)*TEX-37.404
29:     GM TH 40
30:     16 CPA = .2865*1.17E-5*(TEX-2500.)
31:     WA = (.25987*5.36E-6*TEX)*TEX-37.404
32:     GM TH 40
33:     20 IF(TEX.GT.300.) GOTO 21
34:     TEX=300.
35:     IF(IFA.EQ.1) GOTO 51
36:     IFA=2
37:     GOTO 24
38:     51 IFA=3
39:     GOTO 24
40:     21 IF(TEX*900.) 23,28,22
41:     22 IF(TEX*1200.) 26,30,30
42:     23 IF(TEX*700.) 24,26,26
43:     24 CPA = .2392*1.1E-5*(TEX-500.)
44:     WA = (.22623*1.126E-5*TEX)*TEX+3.5214
45:     GM TH 40
46:     26 CPA = .241*2.4E-5*(TEX-700.)
47:     WA = (.22623*1.126E-5*TEX)*TEX+3.5214
48:     GM TH 40
49:     28 CPA = .2458*3.1E-5*(TEX-900.)
50:     WA = (.22623*1.126E-5*TEX)*TEX+3.5214
51:     GM TH 40
52:     30 CPA = .2458*3.1E-5*(TEX-900.)
53:     WA = (.22519*1.292E-5*TEX)*TEX+2.3733
54:     40 CPF = .9333*(5.87E-5+3.27E-8*(3500.-TEX))*(3500.-TEX)
55:     HF = (.50699*6.180E-5*TEX)*TEX-132.20
56:     CP = (CPA+FARX*CPF)/(1.+FARX)
57:     H = (WA+FARX*HF)/(1.+FARX)
58:     AM = 28.97-.9461R6*FARX
59:     REX = 1.98637/AM
60:     GM = CP/(CP*REX)
61:     GMX = (GM-1.)/GM
62:     RETURN
63:     END

```


Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1: SUBROUTINE DYNAM(A,PV,TV,KGAL,KVBL,KNR,RAD,KRAD,KA,TWV,WD,WV,KBLD,
2: IDPRB)
3: DIMENSION A(20),PV(20),V(20),KGAL(20),KVBL(20),KNR(8),RAD(8)
4: DIMENSION KRAD(8),KA(30),TWV(20),WD(20),WV(20),KBLD(8),IDPRB(10)
5: COMMON/TDATA/TIME,DT,ISTEP,NICST
6: COMMON/DATA/X(39),U(3),ETA(3),DXN(40),DX(40),DX1(40),CLM(40)
7: 1KVBLIG,KGALIG,KVBLGG,KGALGG,RTHO,ABL,KSGV,HTC,KVBLCD,KGALCD,TB,KNB
8: 2,KGALB,KVBLB,HT,TT,PT,K6,K2,HT,MCD,PO,WN,KNAB,KSPED,KFIGV,K3,ISVP
9: 3R,TVO
10: REAL ICTHVO,ICWV
11: REAL K5,K6,NC1,ICN,NC1,IGVPR,IGV,K1,K3,K6,K7,L,KGAL,KVBL
12: REAL KVBLGG,KGALGG,KVBLCD,KGALCD,KGALB,KWB,KVBLB,KVBLT,KSPED
13: REAL KFIGV,NCX,KNR,KRAD,KA,KSGV,KNAB,NRYTB,ICPV,ICWD,ICW,ICWV
14: REAL ICTW1,ICWD2,ICW2,ICTW2,ICWD3,ICW3,ICTW3,ICW4,ICWV
15: REAL ICTW4,ICWD5,ICW5,ICTW5,ICWD6,ICW6,ICTW6,ICW7,ICWV
16: REAL ICTW7,ICWD8,ICW8,ICTW8,ICW9,ICWD9,ICWCD,ICWCD
17: REAL ICWCD,ICWB,ICPB,ICMB,ICRT,ICRMT,ICWOP,NCB,NCB,NCB,NCB,NCB
18: REAL NC7,NC8,NOEMD,ITER,IMPL,INTGR,KIC
19: REAL KAS,ICWGG,INDT
20: REAL KBLD,K2,NRAT,KGALIG,KVBLIG
21: EQUIVALENCE (ICWD0,WD0,X(1)),(ICW0,WV0,X(2)),(ICTW0,TW0,X(3))
22: 1),(ICWD1,WD1,X(4)),(ICW1,WV1,X(5)),(ICTW1,TW1,X(6))
23: 2),(ICWD2,WD2,X(7)),(ICW2,WV2,X(8)),(ICTW2,TW2,X(9))
24: 3),(ICWD3,WD3,X(10)),(ICW3,WV3,X(11)),(ICTW3,TW3,X(12))
25: 4),(ICWD4,WD4,X(13)),(ICW4,WV4,X(14)),(ICTW4,TW4,X(15))
26: 5),(ICWD5,WD5,X(16)),(ICW5,WV5,X(17)),(ICTW5,TW5,X(18))
27: 6),(ICWD6,WD6,X(19)),(ICW6,WV6,X(20)),(ICTW6,TW6,X(21))
28: 7),(ICWD7,WD7,X(22)),(ICW7,WV7,X(23)),(ICTW7,TW7,X(24))
29: 8),(ICWD8,WD8,X(25)),(ICW8,WV8,X(26)),(ICTW8,TW8,X(27))
30: 9),(ICWD9,WD9,X(28)),(ICW9,WV9,X(29)),(ICTW9,TW9,X(30))
31: A),(ICWCD,WCDC,X(31)),(ICWCD,WCD,X(32)),(ICWCD,TWCD,X(33))
32: B),(ICWB,WB,X(34)),(ICPB,PB,X(35))
33: C),(ICMB,MB,X(36)),(ICRMT,RMT,X(37)),(ICRT,RT,X(38))
34: D),(ICN,N,X(39)),(HF,U(1)),(BVO,U(2)),(AS,U(3)),(PB,ETA(1))
35: E),(T2,ETA(2)),(PB,ETA(3))
36: EQUIVALENCE (WD0DT ,DX(1)),(WV0DT ,DX(2)),(TW0DT ,DX(3))
37: 1,(WD1DT ,DX(4)),(WV1DT ,DX(5)),(TW1DT ,DX(6))
38: 2,(WD2DT ,DX(7)),(WV2DT ,DX(8)),(TW2DT ,DX(9))
39: 3,(WD3DT ,DX(10)),(WV3DT ,DX(11)),(TW3DT ,DX(12))
40: 4,(WD4DT ,DX(13)),(WV4DT ,DX(14)),(TW4DT ,DX(15))
41: 5,(WD5DT ,DX(16)),(WV5DT ,DX(17)),(TW5DT ,DX(18))
42: 6,(WD6DT ,DX(19)),(WV6DT ,DX(20)),(TW6DT ,DX(21))
43: 7,(WD7DT ,DX(22)),(WV7DT ,DX(23)),(TW7DT ,DX(24))
44: 8,(WD8DT ,DX(25)),(WV8DT ,DX(26)),(TW8DT ,DX(27))
45: 9,(WD9DT ,DX(28)),(WV9DT ,DX(29)),(TW9DT ,DX(30))
46: A,(WCDDT ,DX(31)),(WCDT ,DX(32)),(TWCDT ,DX(33))
47: E,(WBDT ,DX(34)),(RBDT ,DX(35))
48: C,(HBDT ,DX(36)),(RHTDT ,DX(37)),(RTDT ,DX(38))
49: D,(NDT ,DX(39))
50: 999 CONTINUE
51: TVO = TWV/WVO
52: RTHO = SORT(TVO/S18.7)
53: ABL = FUN1(1,BVO,1)
54: DO 299 I=1,8
55: 299 KNR(I) = (NRAD(I))**2/K2
56: C
57: C DYNAMICS
58: C
59: C IN[ ] AND STAGE ONE
60: C

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

61: AC1 = N/RTH1
62: AC1V = AC1/16500
63: IGVP1 = FUN1(1,AC1,2)
64: PD1 = P2 - IGVP1 - 0.05 * PP
65: TJO = T2
66: PV1 = KVBL1G * TV1
67: WDOT = KUAL1 * (PD1 - PV1)
68: WDOT = WDOT * W1
69: TAVDOT = 1.0 * (TJO - WDOT - TV1 * W1)
70: PV1 = KVBL(11) * TV1
71: DEL1 = PV1/14.7
72: FP1 = WDOT * RTH1 / (DEL1 * A(11))
73: VZT1 = KA(1) + KA(11) * FP1 + KA(21) * FP1 * FP1
74: PHI1 = VZT1 / (KRAD(1) * AC1)
75: PSIP1 = FUN2(2,PHI1,BV0,4)
76: PSIT1 = FUN2(5,PHI1,BV0,10)
77: PD1 = PV1 * (1.0 + PSIP1 * KNR(1) / TV1) ** 3.5
78: WDOT = KUAL(11) * (PD1 - PV1)
79: WDOT = WDOT * W2
80: TV1 = TAV1 / W1
81: TJO = TV0 * KNR(1) * PSIT1
82: TAVDOT = 1.0 * (TJO - WDOT - TV1 * W2)
83: C
84: C STAGE TWO
85: C
86: PV2 = KVBL(12) * TV2
87: FTH1 = SQRT(TV1/518.7)
88: AC2 = N/RTH1
89: DEL1 = PV1/14.7
90: FP2 = WDOT * RTH1 / (DEL1 * A(12))
91: VZT2 = KA(2) + KA(12) * FP2 + KA(22) * FP2 * FP2
92: PHI2 = VZT2 / (KRAD(2) * AC2)
93: PSIP2 = FUN1(5,PHI2,7)
94: PD2 = PV1 * (1.0 + PSIP2 * KNR(2) / TV1) ** 3.5
95: WDOT = KUAL(12) * (PD2 - PV2)
96: WDOT = WDOT * W3
97: TV2 = TAV2 / W2
98: PSIT2 = FUN1(1,PHI2,9)
99: TJO = TV1 * KNR(2) * PSIT2
100: TAVDOT = 1.0 * (TJO - WDOT - TV2 * W3)
101: C
102: C STAGE THREE
103: C
104: PV3 = KVBL(13) * TV3
105: FTH2 = SQRT(TV2/518.7)
106: AC3 = N/RTH2
107: DEL2 = PV2/14.7
108: FP3 = WDOT * RTH2 / (DEL2 * A(13))
109: VZT3 = KA(3) + KA(13) * FP3 + KA(23) * FP3 * FP3
110: PHI3 = VZT3 / (KRAD(3) * AC3)
111: PSIP3 = FUN1(7,PHI3,11)
112: PD3 = PV2 * (1.0 + PSIP3 * KNR(3) / TV2) ** 3.5
113: WDOT = KUAL(13) * (PD3 - PV3)
114: WDOT = WDOT * W4
115: TV3 = TAV3 / W3
116: FTH3 = SQRT(TV3/518.7)
117: WBL3 = KBLU(3) * APL * PV3 / (K3 * RTH3)
118: WDOT = WDOT * W5
119: PSIT3 = FUN1(1,PHI3,13)
120: TJO = TV2 * KNR(3) * PSIT3
121: TAVDOT = 1.0 * (TJO - WDOT - TV3 * (W4 + WBL3))

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

122: C STAGE FOUR
123: C
124: PV4 = KVBL(14)*TV4
125: NC4 = N/RTM3
126: DEL3 = FV3/14.7
127: FP4 = WD4*RTM3/(DEL3*A(14))
128: VZT4 = KA(4) + KA(14)*FP4 + KA(24)*FP4*FP4
129: PH14 = VZT4/(K*RAD(4)*NC4)
130: PSIP4 = FUN1(9,PH14,15)
131: PD4 = PV3*(1+PSIP4*KNR(4)/TV3)**3.5
132: TV4DT = KGAL(14)*(PD4-PV4)
133: TV4 = TV4/AV4
134: RTM4 = SQRT(TV4/518.7)
135: KBL4 = KBL(4)*ABL*PV4/(K3*RTM4)
136: WV4DT = WD4*WD5*WBL4
137: PSIT4 = FUN1(10,PH14,17)
138: TD4 = TV3*KNR(4)*PSIT4
139: TV4VDT = 1+*(TD4*WD4-TV4*(WD5+WBL4))
140: C
141: C STAGE FIVE
142: C
143: PV5 = KVBL(15)*TV5
144: NC5 = N/RTM4
145: DEL4 = FV4/14.7
146: FP5 = WD5*RTM4/(DEL4*A(15))
147: VZT5 = KA(5) + KA(15)*FP5 + KA(25)*FP5*FP5
148: PH15 = VZT5/(K*RAD(5)*NC5)
149: PSIP5 = FUN1(11,PH15,19)
150: PD5 = PV4*(1+PSIP5*KNR(5)/TV4)**3.5
151: WD5DT = KGAL(15)*(PD5-PV5)
152: TV5 = TV5/AV5
153: RTM5 = SQRT(TV5/518.7)
154: KBL5 = KBL(5)*ABL*PV5/(K3*RTM5)
155: WV5DT = WD5*WD6*WBL5
156: PSIT5 = FUN1(12,PH15,21)
157: TD5 = TV4*KNR(5)*PSIT5
158: TV5VDT = 1+*(TD5*WD5-TV5*(WD6+WBL5))
159: C
160: C STAGE SIX
161: C
162: PV6 = KVBL(16)*TV6
163: NC6 = N/RTM5
164: DEL5 = FV5/14.7
165: FP6 = WD6*RTM5/(DEL5*A(16))
166: VZT6 = KA(6) + KA(16)*FP6 + KA(26)*FP6*FP6
167: PH16 = VZT6/(K*RAD(6)*NC6)
168: PSIP6 = FUN1(13,PH16,23)
169: PD6 = PV5*(1+PSIP6*KNR(6)/TV5)**3.5
170: WD6DT = KGAL(16)*(PD6-PV6)
171: WV6DT = WD6*WD7
172: TV6 = TV6/AV6
173: PSIT6 = FUN1(14,PH16,25)
174: TD6 = TV5*KNR(6)*PSIT6
175: TV6VDT = 1+*(TD6*WD6-TV6*WD7)
176: C
177: C STAGE SEVEN
178: C
179: PV7 = KVBL(17)*TV7
180: RTM6 = SQRT(TV6/518.7)
181: NC7 = N/RTM6
182: DEL6 = FV6/14.7

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

183:     FH7 = WD7*RTW6/(DEL6*A(17))
184:     VZT7 = KA(7) + KA(17)*FP7 + KA(27)*FP7*FP7
185:     PHI7 = VZT7/(KRAD(7)*NCR7)
186:     PSIP7 = FUN1(15,PHI7,27)
187:     PD7 = PV6*(1+PSIP7*KNR(7)/TV6)**3.5
188:     WDRDT = KGAL(17)*(PD7-PV7)
189:     WVD7 = WD7*WDR
190:     TV7 = TW7/WV7
191:     PSIT7 = FUN1(16,PHI7,29)
192:     TD7 = TV6*KNR(7)*PSIT7
193:     TWV7 = T1+*(TD7*WD7-TV7*WDR)
194: C
195: C STAGE EIGHT
196: C
197:     PVR = KVBL(18)*TWV8
198:     RTH7 = SQRT(TV7/518.7)
199:     NCR8 = N/RTH7
200:     DEL7 = PV7/14.7
201:     FPR = WDR*RTW7/(DEL7*A(18))
202:     VZT8 = KA(8) + KA(18)*FPR + KA(28)*FPR*FPR
203:     PHI8 = VZT8/(KRAD(8)*NCR8)
204:     PSIP8 = FUN1(17,PHI8,31)
205:     PD8 = PV7*(1+PSIP8*KNR(8)/TV7)**3.5
206:     WDRDT = KGAL(18)*(PD8-PV8)
207:     WDR = WDR + WDRGV
208:     TV8 = TWV8/WV8
209:     PSIT8 = FUN1(18,PHI8,33)
210:     TD8 = TV7*KNR(8)*PSIT8
211:     TWV8 = T1+*(TD8*WDR-TV8*WDRGV)
212: C
213: C OUTLET GUIDE VANES
214: C
215:     TGGV = TWGG/WGG
216:     TWGGDT = T1+*(TV8*WGGV-TGGV*WCD)
217:     PGGV = KVBLPG*TWGG
218:     WGGVDT = KGALPG*(PVG-PGGV)-KGGV*WGGV*WGGV
219:     WGGDT = WGGV*WCD
220: C
221: C COMPRESSOR DISCHARGE
222: C
223:     WTC = 1033*...
224:     TCD = TWCD/WCD
225:     TWCDT = (TGGV*WCD-TCD*(WB+WTC))*1.4
226:     PCD = KVBLCD*TWCD
227:     WCDDT = KGALCD*(PVG-PCD)
228:     WCDT = WCD*WB+WTC
229: C
230: C BURNER
231: C
232:     FAB = WF/WB
233:     TB = T2NH(2,FA,MB,TV)
234:     DELPB = KWB*W1+2/PCD*(.771*TCD+.085*TB)
235:     WBDT = KGALB*(PCD-PB-DELPB)
236:     NRTTB = N/SQRT(TB)
237:     HT = RHT/HT
238:     FAT = WF/(WB+WF+WTC)
239:     TT = T2NH(3,FAT,HT,TV)
240:     PT = K4*RT*TT
241:     PTPB = PT/PB
242:     WTTNPH = FUN2(3,PTPB,NRTTB,4)
243:     WT = WTTNPH*W1*PB/TB

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

244: PBDLTB*PB*(1B-TCD)
245: ETAB=FUN1(19,PBDLTB,35)
246: CALL PROC0M(0,TCD,CPCD,GMCD,GMCDX,HCD,IFA)
247: PBDT=KVBLD*(HCD*WB+18650*ETAB*WF*HB)
248: PBDT = HB/PB*(PBDT=KVBLB/1.4*HB*(WB*WF*WT))
249: POPT=PB/PT
250: WNTKNP=HB*KEY(POPT)
251: WN = WNTKNP*AG*KNA2*PT/SQRT(TT)
252: DHTNTB = FUN2(4,PTPB,NRTTB,8)
253: DHTN/100**DHTNTB*SQRT(1B)
254: RHTDT=(HB*WT*HCD*WTC*DHT*WT*T*WN)/1.53
255: RTDT = (WT*WTC*WN)/1.53
256: WLTLB = *BL3*TV3*WBL+TV4*WBL5*TV5
257: DLWHC=HCD*WCD+.2*(WBL3L*TV0*WDO)
258: DLWHT=HB*WT*HCD*WTC*HT*WN
259: KDT=KSPEED/NA*(DLWHT=DLWHC)
260: WD0 = INTGRL(ICWD0,WDDT)
261: WV0 = INTGRL(ICV0,WVODT)
262: TWV0=INTGRL(ICTV0,TWVODT)
263: WD1=INTGRL(ICWD1,WD1DT)
264: WV1=INTGRL(ICV1,WV1DT)
265: TWV1=INTGRL(ICTV1,TWV1DT)
266: WD2=INTGRL(ICWD2,WD2DT)
267: WV2=INTGRL(ICV2,WV2DT)
268: TWV2=INTGRL(ICTV2,TWV2DT)
269: WD3=INTGRL(ICWD3,WD3DT)
270: WV3=INTGRL(ICV3,WV3DT)
271: TWV3=INTGRL(ICTV3,TWV3DT)
272: WD4=INTGRL(ICWD4,WD4DT)
273: WV4=INTGRL(ICV4,WV4DT)
274: TWV4=INTGRL(ICTV4,TWV4DT)
275: WD5=INTGRL(ICWD5,WD5DT)
276: WV5=INTGRL(ICV5,WV5DT)
277: TWV5=INTGRL(ICTV5,TWV5DT)
278: WD6=INTGRL(ICWD6,WD6DT)
279: WV6=INTGRL(ICV6,WV6DT)
280: TWV6=INTGRL(ICTV6,TWV6DT)
281: WD7=INTGRL(ICWD7,WD7DT)
282: WV7=INTGRL(ICV7,WV7DT)
283: TWV7=INTGRL(ICTV7,TWV7DT)
284: WDB=INTGRL(ICDB,WDBDT)
285: WVB=INTGRL(ICVB,WVBDT)
286: TWVB=INTGRL(ICTVB,TWVBDT)
287: WAG=INTGRL(ICAG,WAGDT)
288: WGV=INTGRL(ICVG,WGVDT)
289: WDG=INTGRL(ICWG,WDDGT)
290: WCD=INTGRL(ICCD,WCCDT)
291: WCD=INTGRL(ICCD,WCCDT)
292: WCD=INTGRL(ICCD,WCCDT)
293: WB = INTGRL(ICWB,WBDT)
294: PB = INTGRL(ICPB,PBDT)
295: HB = INTGRL(ICHE,HBDT)
296: RHT = INTGRL(ICRHT,RHTDT)
297: RT = INTGRL(ICRT,RTDT)
298: NT = INTGRL(ICNT,NTDT)
299: NICHT=0
300: SHTA(999,999,999,999),IST,P
301: 998 CONTINUE
302: RETURN
303: END

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1:      FUNCTION H0KEY(PBPT)
2:      IF(PBPT.GE.1.) GOTO 1
3:      IF(PBPT.GE..53) H0KEY=PBPT**(1./1.4)*SQRT(1.-PBPT**(1.4/1.4))
4:      IF(PBPT.GE.0..AND.PBPT.LE..53) H0KEY=.2588
5:      RETURN
6:      1 H0KEY=0.
7:      RETURN
8:      END

```

```

1:      FUNCTION FN1SET(N,ZX,NP,N1,N2)
2:      COMMON XX(17,5),YY(17,5),NX(5),NY(5),X(17,17,5),YDEL(40),
3:      1I(40),JU(40),SLP1(40),SLP2(40),ZPT1(40),ZPT2(40)
4:      COMMON X1(21,21),Z1(21,21),KK(50),MX(23),XDIF(50),SLP(50),
5:      1ZPT(50)
6:      DIMENSION ZX(1)
7:      NX(N) = NP
8:      DO 10 J=1,NP
9:      K=2*J
10:     X1(J,K) = ZX(K-1)
11:     Z1(J,N) = ZX(K)
12:     FN1SET = 1.
13:     DO 30 NR=N1,N2
14:     KK(NR) = 2
15:     XDIF(NR) = X1(2,N)-X1(1,N)
16:     ZPT(NR) = Z1(1,N)
17:     SLP(NR) = (Z1(2,N)-Z1(1,N))/XDIF(NR)
18:     30 CONTINUE
19:     RETURN
20:     END

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1:      FUNCTION FUN1(N,XIN,NR)
2:      COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZZ(17,17,5),YDEL(40),
3:      1II(40),JJ(40),SLP1(40),SLP2(40),ZPT1(40),ZPT2(40)
4:      COMMON      X1(21,21),Z1(21,21),KK(50),MX(23),XDIF(50),SLP(50),
5:      1ZPT(50)
6:      C
7:      10LD = KK(NR)
8:      NXP = MX(N)
9:      IF(XIN=X1(10LD,N)) 105,105,120
10:     105 IF(XIN=X1(10LD-1,N)) 140,140,110
11:     110 I = 10LD
12:     GO TO 250
13:     C
14:     120 IF(XIN=X1(NXP,N)) 125,180,300
15:     125 NF = 10LD + 1
16:     DO 130 I = NF,NXP
17:     IF(XIN=X1(I,N)) 200,200,130
18:     130 CONTINUE
19:     GO TO 200
20:     C
21:     140 IF(XIN=X1(1,N)) 300, 190,145
22:     145 NL = 10LD - 2
23:     DO 150 K = 1,NL
24:     I = 10LD - K
25:     IF(XIN=X1(I-1,N)) 150,150, 200
26:     150 CONTINUE
27:     GO TO 200
28:     180 I = NXP
29:     GO TO 200
30:     190 I = 2
31:     200 XDIF(NR) = X1(I,N)-X1(I-1,N)
32:     ZPT(NR) = Z1(I-1,N)
33:     SLP(NR) = (Z1(I,N)-ZPT(NR))/XDIF(NR)
34:     250 XINC = XIN-X1(I-1,N)
35:     FUN1 = ZPT(NR)+XINC*SLP(NR)
36:     KK(NR) = I
37:     RETURN
38:     300 CONTINUE
39:     IF(XIN*LT*X1(1,N))FUN1=Z1(1,N)
40:     IF(XIN*GT*X1(NXP,N))FUN1=Z1(NXP,N)
41:     RETURN
42:     END

```

Table A-1. Nonlinear Engine Simulation Program (Continued)

```

1:  FUNCTION FN2SET(N,X,Y,Z,NXP,NYP,N1,N2)
2:  COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZZ(17,17,5),YDEL(40),
3:  1I(40),JJ(40),SLP1(40),SLP2(40),ZPT1(40),ZPT2(40)
4:  COMMON      X1(21,21),Z1(21,21),KK(30),MX(23),XDIF(50),SLP(50),
5:  1ZPT(50)
6:  DIMENSION X( 1),Y(1),Z(1)
7:  10 NX(N) = NXP
8:  NY(N) = NYP
9:  DO 15 J=1,NYP
10:  YY(J,N) = Y(J)
11:  DO 15 I=1,NXP
12:  K = I+(J-1)*NXP
13:  15 ZZZ(I,J,N) = Z(K)
14:  DO 20 I=1,NXP
15:  20 XX(I,N) = X(I)
16:  FN2SET = 1.0
17:  DO 30 NR=N1,N2
18:  1I(NR) = 2
19:  JJ(NR) = 2
20:  XDEL = XX(2,N)-XX(1,N)
21:  YDEL(NR) = YY(2,N)-YY(1,N)
22:  ZPT1(NR) = ZZZ(1,1,N)
23:  ZPT2(NR) = ZZZ(1,2,N)
24:  SLP1(NR) = (ZZZ(2,1,N)-ZPT1(NR))/XDEL
25:  SLP2(NR) = (ZZZ(2,2,N)-ZPT2(NR))/XDEL
26:  30 CONTINUE
27:  RETURN
28:  END

```


Table A-1. Nonlinear Engine Simulation Program (Concluded)

```

1:      FUNCTION FLN2(N,XIN,YIN,NR)
2:      COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZZ(17,17,5),YDEL(40)
3:      II(40),JJ(40),SLP1(40),SLP2(40),ZPT1(40),ZPT2(40)
4:      COMMON      X1(21,21),Z1(21,21),KK(50),MX(23),XDIF(50),SLP(50)
5:      IZPT(50)
6:      C      TEST FOR X IN PREVIOUS INTERVAL
7:      NXP = NX(N)
8:      IOLD = II(NR)
9:      IF(XIN = X(IOLD,N)) 105,105,120
10:     105 IF(XIN = X(IOLD-1,N)) 140,140,110
11:     110 I = IOLD
12:     GO TO 200
13:     C      COUNT UP
14:     120 IF(XIN = XX(NXP,N)) 125,180,180
15:     125 NF = IOLD + 1
16:     DO 130 I = NF,NXP
17:     IF(XIN = XX(I,N))200,200,130
18:     130 CONTINUE
19:     GO TO 200
20:     C      COUNT DOWN
21:     140 IF(XIN=XX(1,N)) 190,190,145
22:     145 NL = IOLD - 2
23:     DO 150 K = 1,NL
24:     I = IOLD - K
25:     IF(XIN=XX(I,N))150,150,200
26:     150 CONTINUE
27:     GO TO 200
28:     180 I=NXP
29:     XIN=XX(NXP,N)
30:     GO TO 200
31:     190 I = 2
32:     XIN=X(I,N)
33:     C      TEST FOR Y IN PREVIOUS INTERVAL
34:     200 NYP = NY(N)
35:     JOLD = JJ(NR)
36:     IF(YIN = YY(JOLD,N)) 205, 205, 220
37:     205 IF(YIN = YY(JOLD-1,N)) 240,240,210
38:     210 J = JOLD
39:     IF(I=IOLD) 300,400,300
40:     C      COUNT UP
41:     220 IF(YIN = YY(NYP,N)) 225, 280,280
42:     225 NF = JOLD + 1
43:     DO 230 J = NF,NYP
44:     IF(YIN = YY(J,N)) 300,300,230
45:     230 CONTINUE
46:     GO TO 300
47:     C      COUNT DOWN
48:     240 IF(YIN = YY(1,N))290,290,245
49:     245 NL = JOLD - 2
50:     DO 250 K = 1,NL
51:     J = JOLD - K
52:     IF(YIN =YY(J,N)) 250,250,300
53:     250 CONTINUE
54:     GO TO 300
55:     280 J = NYP
56:     YIN=YY(NYP,N)
57:     GO TO 300
58:     290 J = 2
59:     YIN=YY(1,N)
60:     C      COMPUTE Z(Y) INTERCEPTS AND SLOPES
61:     300 XDEL = XX(1,N)*XX(I-1,N)
62:     YDEL(NR) = YY(J,N)-YY(J-1,N)
63:     ZPT1(NR) = ZZZ(I-1,J-1,N)
64:     ZPT2(NR) = ZZZ(I-1,J,N)
65:     SLP1(NR) = (ZZZ(I,J-1,N)-ZPT1(NR))/XDEL
66:     SLP2(NR) = (ZZZ(I,J,N)-ZPT2(NR))/XDEL
67:     C      INTERPOLATE FOR ANSWER
68:     400 II(NR) = I
69:     JJ(NR) = J
70:     XINC = XIN-XX(I-1,N)
71:     P1ZZ = ZPT1(NR)+XINC*SLP1(NR)
72:     P2ZZ = ZPT2(NR)+XINC*SLP2(NR)
73:     YFRAC = (YIN-YY(J-1,N))/YDEL(NR)
74:     FUN2 = P1ZZ + YFRAC*(P2ZZ-P1ZZ)
75:     RETURN
76:     END

```

Table A-2. Reduced-Order Component Model

```

1 00000000 DIMENSION IGV(42),A(20),PV(20),TV(20),YY1(14),XX1(17),ZZ1(196)
2 00000001 DIMENSION L(20),V(20),KGAL(20),KVBL(20),KNR(8),RAD(8),KRAD(8)
3 00000002 DIMENSION KA(30)
4 00000003 DIMENSION TWV(20),WD(20),WV(20),KBLD(8),DPRB(14)
5 00000004 COMMON/TA/TA/TIME,DTH
6 00000005 COMMON/DATA/X( 2),U(4),ETA(3),DX( 6),DX( 6),DX1( 6),CLM( 6),KVBLI
7 00000006 IG, RTHO, KVBL0G,KGAL0G,K0GV,WTC,KVBLCD,KGALCD, KWB,KGALB,
8 00000007 ZHT, K4,K2,WY,HCD,KVBLB,PO,WN,KNAB,KSPPEED,KGALIG,KFIGV,K3,
9 00000010 3WBL3,WBL4,WBL5,WCD,WCCD,TWCD,ETAB,DHT
10 00000011 4,TCD, WB,MB,PB,ERROR
11 00000012 REAL K5,K8,NC1,ICN,NC1N,IGVPR,IGV,K1,K3,K4,K6,K7,L,KGAL,KVBL
12 00000013 REAL KVBL0G,KGAL0G,KVBLCD,KGALCD,KGALB,KWB,KVBLB,KVBLT,KSPPEED
13 00000014 REAL KFIGV,NCX,KNR,KRAD,KA,K0GV,KNAB,NRTTB,ICPVO,ICWDO,ICWD1,ICWV1
14 00000015 REAL ICTWV1,ICWD2,ICWV2,ICTWV2,ICWD3,ICWV3,ICTWV3,ICWD4,ICWV4
15 00000016 REAL ICTWV4,ICWD5,ICWV5,ICTWV5,ICWD6,ICWV6,ICTWV6,ICWD7,ICWV7
16 00000017 REAL ICTWV7,ICWD8,ICWV8,ICTWV8,ICTW0G,ICW0GV,ICW0G,ICWCD,ICWCD
17 00000020 REAL ICWDCD,ICW5,ICPB,ICMB,ICRT,ICRMT,ICWFOF,NC2,NC3,NC4,NC5,NC6
18 00000021 REAL NC7,NC8,NOEND,ITER,IMPL,INTGRL,KIC
19 00000022 REAL KAS,ICW0G,N,NDT
20 00000023 REAL KBLD,K2,NRAT,KGALIG,KVBLIG
21 00000024 EQUIVALENCE (ICN,N,X(1))
22 00000025 EQUIVALENCE
23 00000026 2 (NDT,DX( 1)),(WF,U(1)),(BVB,U(2)),(A6,U(3)),(P2,ETA(1
24 00000027 3)),(TV0,T2,ETA(2)),(P8,ETA(3)),(U(4),ABL)
25 00000030 EQUIVALENCE (TMIC,TH,X(2)),(TMDT,DX(2))
26 00000031 EQUIVALENCE (PCD,DX(3)),(PT,DX(4)),(TB,DX(5)),(TT,DX(6))
27 00000032 REWIND 3
28 00000033 8099 CONTINUE
29 00000034 READ(5,6306) ERROR
30 00000035 IF(ERROR.EQ.0.0) GO TO 8098
31 00000036 6306 FORMAT(G12.5)
32 00000037 READ(5,8040) NX,NU,NE,NR,DPERT
33 00000040 8030 FORMAT(4I2,G12.5)
34 00000041 REWIND 7
35 00000042 DATA (DPRB(I),I=1,14)/.60,7.26E-4,.70,7.07E-4,.80,6.98E-4,.85,6.9E
36 00000043 1,.90,6.96E-4,.97,6.96E-4,1.0,7.38E-4/
37 00000044 DATA (KBLD(I),I=1,8)/20.0,1.1025,1.0572,1.0411,3.0,7
38 00000045 DATA (KGAL(I),I=1,8)/25542.27942,27247.26407,24084.21872,221
39 00000046 156.22439.7/
40 00000047 DATA (KVBL(I),I=1,8)/1.9107,3.3711,4.9797,7.0839,9.3087,11.2953,13
41 00000050 1.7727,15.1219/
42 00000051 DATA (TV(I),I=1,20)/5.1600,15.518.7/
43 00000052 DATA (TWV(I),I=1,20)/20.10.7/
44 00000053 DATA (WD(I),I=1,20)/20.30.7/
45 00000054 DATA (WV(I),I=1,20)/20.01.7/
46 00000055 READ(7)(IGV(I),I=1,18)
47 00000056 F11 =FN1SET(1,IGV ,9,1,1)
48 00000057 READ(7)(IGV(I),I=1,38)
49 00000060 F14 =FN1SET(4,IGV ,19,4,5)
50 00000061 READ(7)(IGV(I),I=1,20)
51 00000062 F12 =FN1SET(2,IGV,10,2,2)
52 00000063 READ(7)(IGV(I),I=1,18)
53 00000064 F13 =FN1SET(3,IGV,9, 3,3)
54 00000065 READ(7)(IGV(I),I=1,40)
55 00000066 F15 =FN1SET(5,IGV ,20,6,7)
56 00000067 READ(7)(IGV(I),I=1,42)
57 00000070 F16 =FN1SET(6,IGV ,21,8,9)
58 00000071 READ(7)(IGV(I),I=1,34)
59 00000072 F17 =FN1SET(7,IGV ,17,10,11)
60 00000073 READ(7)(IGV(I),I=1,38)

```

Table A-2. Reduced-Order Component Model (Continued)

61	00000074	F18 = FN1SET(8, IGV, 19, 12, 13)
62	00000075	READ(7)(IGV(I), I=1, 36)
63	00000076	F19 = FN1SET(9, IGV, 18, 14, 15)
64	00000077	READ(7)(IGV(I), I=1, 40)
65	00000100	F110 = FN1SET(10, IGV, 20, 16, 17)
66	00000101	READ(7)(IGV(I), I=1, 32)
67	00000102	F111 = FN1SET(11, IGV, 16, 18, 19)
68	00000103	READ(7)(IGV(I), I=1, 36)
69	00000104	F112 = FN1SET(12, IGV, 15, 20, 21)
70	00000105	READ(7)(IGV(I), I=1, 26)
71	00000106	F113 = FN1SET(13, IGV, 13, 22, 23)
72	00000107	READ(7)(IGV(I), I=1, 26)
73	00000110	F114 = FN1SET(14, IGV, 13, 24, 25)
74	00000111	READ(7)(IGV(I), I=1, 30)
75	00000112	F115 = FN1SET(15, IGV, 15, 26, 27)
76	00000113	READ(7)(IGV(I), I=1, 26)
77	00000114	F116 = FN1SET(16, IGV, 13, 28, 29)
78	00000115	READ(7)(IGV(I), I=1, 30)
79	00000116	F117 = FN1SET(17, IGV, 15, 30, 31)
80	00000117	READ(7)(IGV(I), I=1, 32)
81	00000120	F118 = FN1SET(18, IGV, 16, 32, 33)
82	00000121	READ(7)(IGV(I), I=1, 28)
83	00000122	F119 = FN1SET(19, IGV, 14, 34, 35)
84	00000123	F120 = FN1SET(20, DPRB, 7, 36, 37)
85	00000124	READ(7)(IGV(I), I=1, 40)
86	00000125	IGV(6) = 1.39
87	00000126	IGV(8) = 1.395
88	00000127	DB 6301 I=9, 39, 2
89	00000130	I1 = I + 1
90	00000131	XSQ = IGV(I) * IGV(I)
91	00000132	F8FX = XSQ * 1.031964679E-8 = IGV(I) * 1.735930756E-4 + 2.129761925
92	00000133	IGV(I1) = F8FX
93	00000134	F121 = FN1SET(21, IGV, 20, 38, 39)
94	00000135	READ(7)(A(I), I=1, 20)
95	00000136	READ(7)(L(I), I=1, 20)
96	00000137	READ(7)(V(I), I=1, 20)
97	00000140	READ(7)(IGV(I), I=1, 27)
98	00000141	READ(7)(IGV(I), I=1, 23)
99	00000142	READ(7)(RAD(I), I=1, 8)
100	00000143	READ(7)(KRAD(I), I=1, 8)
101	00000144	READ(7)(PV(I), I=1, 20)
102	00000145	READ(7)(IGV(I), I=1, 20)
103	00000146	READ(7)(YY1(I), I=1, 5)
104	00000147	READ(7)(XX1(I), I=1, 13)
105	00000150	READ(7)(ZZ1(I), I=1, 65)
106	00000151	F1 = FN2SET(1, XX1, YY1, ZZ1, 13, 5, 1, 2)
107	00000152	READ(7)(KA(I), I=1, 30)
108	00000153	READ(7)(YY1(I), I=1, 4)
109	00000154	READ(7)(XX1(I), I=1, 17)
110	00000155	READ(7)(ZZ1(I), I=1, 68)
111	00000156	F2 = FN2SET(2, XX1, YY1, ZZ1, 17, 4, 3, 4)
112	00000157	READ(7)(YY1(I), I=1, 14)
113	00000160	READ(7)(XX1(I), I=1, 14)
114	00000161	READ(7)(ZZ1(I), I=1, 196)
115	00000162	F3 = FN2SET(3, XX1, YY1, ZZ1, 14, 14, 5, 6)
116	00000163	READ(7)(YY1(I), I=1, 14)
117	00000164	READ(7)(XX1(I), I=1, 14)
118	00000165	READ(7)(ZZ1(I), I=1, 196)
119	00000166	F4 = FN2SET(4, XX1, YY1, ZZ1, 14, 14, 7, 8)
120	00000167	READ(7)(YY1(I), I=1, 4)
121	00000170	READ(7)(XX1(I), I=1, 17)

Table A-2. Reduced-Order Component Model (Continued)

122	00000171	READ(7) (ZZ1(1),I=1,68)
123	00000172	F5=FN2SET(5,XX1,YY1,ZZ1,17,4,9,10)
124	00000173	C
125	00000174	C SET PARAMETERS
126	00000175	C
127	00000176	KAB=0
128	00000177	NSSP=1
129	00000200	TTGS=1700.
130	00000201	READ(5,8066) NRAT,WINGS,SPLC,BVB,ADL
131	00000202	WRITE(9,8066) ERROR
132	00000203	WRITE(9,8070) NX,NU,NE,NR,DPERT
133	00000204	WRITE(9,8066) NRAT,WINGS,SPLC,BVB,ABL
134	00000205	READ(5,9903) P2,T2
135	00000206	9903 FORMAT(2G12,5)
136	00000207	WRITE(9,9903) P2,T2
137	00000210	8066 FORMAT(6G12,5)
138	00000211	AB=AXFN(NRAT)
139	00000212	AJ=AJ
140	00000213	TBGS=2100.
141	00000214	FABGS=.02
142	00000215	T2=518.7
143	00000216	WDEL=.1
144	00000217	DTM=DEL*.5
145	00000220	TIME=0.
146	00000221	K1 = 3.14159/360.
147	00000222	K2=7.*32*17.23.35/(K1*K1)
148	00000223	K3=SQRT(518.7)
149	00000224	K4=(53.35*12.)/17600.
150	00000225	KGALIG=51837.
151	00000226	K/PLIG=3*31
152	00000227	KGALRG = 22430.
153	00000230	KVFLRG = 1E-10
154	00000231	KGALCD = 8730.
155	00000232	KVFLCD = 1.981
156	00000233	KGALB=54*7
157	00000234	KVFLB = 2.659
158	00000235	KWB = .0054445
159	00000236	KVFLT = 14.15
160	00000237	KSPEED = 178400.
161	00000240	IC=NRAT*16500.
162	00000241	P2=P2
163	00000242	P8=P2
164	00000243	TV = T2
165	00000244	RTHO = SQRT(TV./518.7)
166	00000245	NC1 = IC/RTHO
167	00000246	NC1N = NC1/16500.
168	00000247	BVVB=FUN2(1,NC1N,TV0,1)
169	00000250	ABLB=FUN1(1,BVVB,1)
170	00000251	IGVPR=FUN1(2,NC1N,2)
171	00000252	RGVPR=FUN1(3,NC1N,3)
172	00000253	TV(1) = T2
173	00000254	III=0
174	00000255	ITER1=0
175	00000256	ITER2=0
176	00000257	ITER3=0
177	00000260	DB 6=K=1,NSSP
178	00000261	FAB=FABGS
179	00000262	TB=TBGS
180	00000263	WIN=WINGS
181	00000264	DELNX=.01
182	00000265	PV(10) = P2*IGVPR = .005*P2

Table A-2. Reduced-Order Component Model (Continued)

```

183 0000266          99 CONTINUE
184 0000267          WD(I) = WIN - FLOAT(K-1) * DEL
185 0000270          ITER1 = ITER1 + 1
186 0000271          KFIGV = KGALIG * (P2 - PV(10)) / (AD(10) * WD(10))
187 0000272          KBL = C
188 0000273          KBLTHL = C
189 0000274          IF (SENSE $: ITC = 3) 8073, 8074
190 0000275      8074 CONTINUE
191 0000276          IF (I11 = E.1) GOTO 5901
192 0000277      8073 CONTINUE
193 0000300          J = 0
194 0000301          WRITE(6,50) ICN, ABL, BVB, IGVPR, CGVPR
195 0000302          WRITE(6,51)
196 0000303          WRITE(6,52) J, PV(10), TV(10), WD(10)
197 0000304      5901 CONTINUE
198 0000305          DO 20 I = 1, 10
199 0000306          J = I - 10
200 0000307          DELX = PV(I-1) / 14.7
201 0000310          RTX = SQRT(TV(I-1) / 518.71)
202 0000311          NCX = ICN / RTX
203 0000312          WD(I) = D(I-1) * BBL
204 0000313          FPX = AD(I) * RTX / (DELX * A(I))
205 0000314          VZTX = KA(J) * FPX * (KA(J+10) * FPX * KA(J+20))
206 0000315          KRAD(J) = 41. * RAD(J)
207 0000316          PHIX = VZTX / (KRAD(J) * NCX)
208 0000317          GO TO (1, 2, 3, 4, 5, 6, 7, 8) J
209 0000320          1 CONTINUE
210 0000321          PSIPX = FUN2(2, PHIX, BVB, 3)
211 0000322          PSITX = FUN2(5, PHIX, BVB, 9)
212 0000323          GOTO 1
213 0000324          2 CONTINUE
214 0000325          PSIPX = FUN1(5, PHIX, 6)
215 0000326          PSITX = FUN1(4, PHIX, 8)
216 0000327          GOTO 10
217 0000330          3 CONTINUE
218 0000331          PSIPX = FUN1(7, PHIX, 10)
219 0000332          PSITX = FUN1(8, PHIX, 12)
220 0000333          GOTO 10
221 0000334          4 CONTINUE
222 0000335          PSIPX = FUN1(9, PHIX, 14)
223 0000336          PSITX = FUN1(10, PHIX, 16)
224 0000337          GOTO 10
225 0000340          5 CONTINUE
226 0000341          PSIPX = FUN1(11, PHIX, 18)
227 0000342          PSITX = FUN1(12, PHIX, 20)
228 0000343          GOTO 10
229 0000344          6 CONTINUE
230 0000345          PSIPX = FUN1(13, PHIX, 22)
231 0000346          PSITX = FUN1(14, PHIX, 24)
232 0000347          GOTO 10
233 0000350          7 CONTINUE
234 0000351          PSIPX = FUN1(15, PHIX, 26)
235 0000352          PSITX = FUN1(16, PHIX, 28)
236 0000353          GOTO 10
237 0000354          8 CONTINUE
238 0000355          PSIPX = FUN1(17, PHIX, 30)
239 0000356          PSITX = FUN1(18, PHIX, 32)
240 0000357          10 CONTINUE
241 0000360          KNR(J) = (ICN * RAD(J)) * 2 / K2
242 0000361          PV(I) = PV(I-1) * (1 + PSIPX * KNR(J) / TV(I-1)) * 3.5
243 0000362          TV(I) = TV(I-1) * KNR(J) * PSITX

```

Table A-2. Reduced-Order Component Model (Continued)

```

244 00000363      TWV(I)=PV(I)/KVBL(J)
245 00000364      WV(I)=TWV(I)/TV(I)
246 00000365      PR = PV(I)/PV(I-1)
247 00000366      WBL=KBLD(J)*ABL*PV(I)/SQRT(TV(I))
248 00000367      IF(J.EQ.3) WBL3=WBL
249 00000370      IF(J.EQ.4) WBL4=WBL
250 00000371      IF(J.EQ.5) WBL5=WBL
251 00000372      WBLTHL=WBLTBL+WBL*TV(I)
252 00000373      IF(SENSE SWITCH 3)8075,8076
253 00000374      8076 CONTINUE
254 00000375      IF(III.NE.1) GOTO 5902
255 00000376      8075 CONTINUE
256 00000377      WRITE(9,52)J,PV(I),TV(I),WD(I),WBL,PSITX,V7,TX,PHIX,PSIPX,PR
257 00000400      5902 CONTINUE
258 00000401      20 CONTINUE
259 00000402      J = 10
260 00000403      P0GV = PV(18)*0GVPR
261 00000404      T0GV = TV(18)
262 00000405      W0GV=WD(18)
263 00000406      K0GV = K*GALOG*(PV(18)-P0GV)/W0GV**2
264 00000407      TW0G = P0GV/KVBL0G
265 00000410      WD0G = TW0G/T0GV
266 00000411      PR23=P0GV/P2
267 00000412      TR23=T0GV/T2
268 00000413      EFF23=(PR23**285-1.)/(TR23-1.)
269 00000414      IF(SENSE SWITCH 3)8077,8078
270 00000415      8078 CONTINUE
271 00000416      IF(III.NE.1) GOTO 5903
272 00000417      8077 CONTINUE
273 00000420      WRITE(6,53)
274 00000421      WRITE(6,52)T,P0GV,T0GV,W0GV,WBLTBL,EFF23,PR23,TR23
275 00000422      5903 CONTINUE
276 00000423      J = 11
277 00000424      PCD = P0GV
278 00000425      TCD=T0GV
279 00000426      CALL PR0COM(0.,PCD,CPCD,GMCD,GMCDX,HCD,IFA)
280 00000427      WCD = W0GV
281 00000430      TWCD = PCD/KVBLCD
282 00000431      WDCD = TWCD/TCD
283 00000432      WTC=.033*WD(10)
284 00000433      DLWMC=HCD,WCD**2*(WBLTBL=WD(10),TV(10))+SPLC
285 00000434      KNAB=5.0*05=.0429772,A8=.000126664,A8**2
286 00000435      IF(SENSE SWITCH 3)8170,8171
287 00000436      8171 CONTINUE
288 00000437      IF(III.NE.1) GOTO 5904
289 00000440      8170 CONTINUE
290 00000441      WRITE(6,54)
291 00000442      WRITE(6,52)J,PCD,TCD,WCD,WTC,DLWMC,KNAB
292 00000443      5904 CONTINUE
293 00000444      J=12
294 00000445      KWB=FUN1(20,NRAT,36)
295 00000446      WB=WCD-WTC
296 00000447      IF(ITER1.EQ.1) PT=.35*PCD
297 00000450      DPTX=1.
298 00000451      220 CONTINUE
299 00000452      ITER2=ITER2+1
300 00000453      DTBX=25.
301 00000454      WTBLD=(1.+FAB)*WB
302 00000455      221 NRTTB=|CN/SQRT(TB)
303 00000456      ITER3=ITER3+1
304 00000457      DELPB = KWB*WB**2/PCD*(.771*TCD+.085*TB)

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Table A-2. Reduced-Order Component Model (Continued)

```

305 00000460          PB = PCD*DELPH
306 00000461          PBCLTB = PB*(TR=TCO)
307 00000462          ETAB =FUN1(19,PBDLTB,34)
308 00000463          PTPB = PT/PB
309 00000464          DHTNTB = FUN2(4,PTPB,NRTTB,7)
310 00000465          DHT=DHTNTB*ICN/1000.*SQRT(TB)
311 00000466          227 CALL PROCAM(FAR,TB,CPB,GMB,GMBX,HB,IFA)
312 00000467          IF(SENSE SWITCH 4)8098,8091
313 00000470          8091 CONTINUE
314 00000471          WT1=WB*(18650.*ETAB=HCD)/(18650.*ETAB+WB)
315 00000472          IF(IFA.GT.0) GOTO 8071
316 00000473          WTERR=ABS(WT1-WTOLD)
317 00000474          IF(WTERR.GT..0005) GOTO 223
318 00000475          8072 CONTINUE
319 00000476          WF=WT1-WB
320 00000477          FAB=WF/WB
321 00000500          KN=WT1+WTC
322 00000501          GOTO 224
323 00000502          8071 IF(WT1.LT.WB) WT1=WB
324 00000503          IF(WT1.GT.(WB*1.067623)) WT1=1.067623*WB
325 00000504          WTOLD=WT1
326 00000505          GOTO 8072
327 00000506          223 WTOLD=(WT1+WTOLD)*.5
328 00000507          WF=WTOLD-WB
329 00000510          FAR=WF/WB
330 00000511          GOTO 222
331 00000512          224 CONTINUE
332 00000513          HT=WT1/KN*(HB-DHT)+WTC/KN*HCD
333 00000514          HBR=(DL*HC+KN*HT-HCD=NTC)/WT1
334 00000515          TBERR=HBR-HB
335 00000516          IF(TBERR.GT..0005) GOTO 225
336 00000517          IF(TBERR.LT..-0005) GOTO 228
337 00000520          GOTO 229
338 00000521          225 IF(DTBX)226,226,227
339 00000522          226 DTBX=-DTBX*.5
340 00000523          227 TB=TB+DTBX
341 00000524          GOTO 221
342 00000525          228 IF(DTBX)227,227,226
343 00000526          229 CONTINUE
344 00000527          POPT=PO/PT
345 00000530          IF(POPT=.528)233,233,230
346 00000531          23 IF(POPT=1.1232,231,231)
347 00000532          231 WTKNP=0
348 00000533          GOTO 234
349 00000534          232 WTKNP= POPT*(1./1.4)*SQRT(1.-POPT*(.4/1.4))
350 00000535          GOTO 234
351 00000536          233 WTKNP=.25RR
352 00000537          234 CONTINUE
353 00000540          WTTNPB=FUN2(3,PTPB,NRTTB,5)
354 00000541          WTP=TTNPB*PB/TB*ICN
355 00000542          PTERR=WT2-WT1
356 00000543          24 IF(PTERR.GT..0005) GOTO 241
357 00000544          IF(PTERR.LT..-0005) GOTO 245
358 00000545          GOTO 250
359 00000546          241 IF(DPTX)242,242,243
360 00000547          242 DPTX=-DPTX*.5
361 00000550          243 PT=PT+DPTX
362 00000551          GOTO 220
363 00000552          245 IF(DPTX)243,243,242
364 00000553          25 CONTINUE
365 00000554          FAT=WF/KN

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Table A-2. Reduced-Order Component Model (Continued)

```

366 00000555      TT=TEMP(1,FAT,WT,Tv)
367 00000556      IF(SENSE S-ITCM #)R098,8092
368 00000557      8092 CONTINUE
369 00000560      KX=(KXAB,PT,WTKNP,AB)/SQRT(TT)
370 00000561      WERR=KX-1
371 00000562      IF(III.EQ.1) GOTO 60
372 00000563      IF(WERR.GT..005) GOTO 5951
373 00000564      IF(WERR.LT.-.005) GOTO 5955
374 00000565      III=1
375 00000566      GOTO 99
376 00000567      5951 IF(DW1,X)5952,5952,5953
377 00000570      5952 DW1X=DW1X+.5
378 00000571      5953 W1=X+DW1X
379 00000572      GOTO 99
380 00000573      5955 IF(DW1,X)5953,5953,5952
381 00000574      61 CONTINUE
382 00000575      DLWHT=HB*WT1+MCD*WTC-WT*W
383 00000576      WFM = 3600*AA*F
384 00000577      WRITE(6,56)
385 00000600      WRITE(6,52) J,PB,TB,WB,ETAB,HB,PTP6,WRTTB,DHTNTB,WTTNPB
386 00000601      J = 13
387 00000602      WRITE(6,57)
388 00000603      WRITE(6,52) J,PT,TT,W1,WFM,HT,DLWHT,WK,WTKNP,AB
389 00000604      ACD=50.7334498
390 00000605      CFMCD=ACD*SQRT(53*3*TC/(1+.4*32*2))/(PCD*ACD)
391 00000606      FMCD=RNFM(CFMCD)
392 00000607      CFMN=W1*SQRT(53*3*TT/(1+.4*32*2))/(PT*AJ)
393 00000608      FMN=RNFM(CFMN)
394 00000611      FMNS=FMN*FMN
395 00000612      AAA=1+.2*FMNS
396 00000613      AAA=AAA+.3*5
397 00000614      AAA=1./AAA
398 00000615      AAA=(1+.4*FMNS+1.)*AAA
399 00000616      THRUST=(PT*AAA-P2)*AJ
400 00000617      SPFC=WFM/THRUST
401 00000620      WRITE(9,6791)FMCD,FMN,WFM,THRUST,SPFC
402 00000621      6791 FORMAT(1H,2X,5H MCD=G13.5,4H MN=G13.5,5H WFM=G13.5,8H THRUST=G13.
403 00000622      15,6H SPFC=G13.5)
404 00000623      PCDP2=PCD/P2
405 00000624      WRITE(9,6792)PCDP2
406 00000625      6792 FORMAT(1H,7H P3/P2=G13.5)
407 00000626      WRITE(6,2108)ITER1,ITER2,ITER3
408 00000627      2108 FORMAT(1H,5X,8HITER1 = 15,5X,8HITER2 = 15,5X,8HITER3 = 15)
409 00000630      50 FORMAT(1H,4X,4HN = ,F8.2,4X,6HABL = ,F8.4,4X,6HVB0 = ,F8.4,4X,
410 00000631      8HIGVPR = ,F8.4,4X,8HOGVPR = ,F8.4)
411 00000632      52 FORMAT(1H,13,9G13.5)
412 00000633      51 FORMAT(1H,2X,1HJ,4X,5HPV(J),8X,5HTV(J),8X,5HWD(J),7X,6HMBL(J),7X,
413 00000634      18HPSIX(J),6X,7HV7TX(J),6X,7HPHIX(J),6X,8HPSIPX(J),5X,5HPR(J))
414 00000635      53 FORMAT(1H,3X,1HJ,6X,4HPGV,10X,4HTGV,10X,4HMBL,9X,5
415 00000636      1HEFF23,10X,4HPR23,10X,4HTR23)
416 00000637      54 FORMAT(1H,3X,1HJ,7X,3HPCD,11X,3HTCD,11X,3HMCD,11X,3HWTC,9X,5HDLWH
417 00000640      1C,10X,4HKHAB)
418 00000641      56 FORMAT(1H,3X,1HJ,8X,2HPB,12X,2HTB,12X,2HWB,10X,4HETAB,12X,2HMB,
419 00000642      110X,4HPTPB,9X,5HRTTB,8X,6HDHTNB,8X,6HWTNPB)
420 00000643      57 FORMAT(1H,3X,1HJ,8X,2HPT,12X,2HTT,12X,2HWT,12X,2HWF,12X,2HHT,9X,5
421 00000644      1HDLWHT,12X,2HMW,8X,6HWTKNP,12X,2HAB)
422 00000645      ICPV = PV(10)
423 00000646      ICWD = WD(10)
424 00000647      ICWD1 = WD(11)
425 00000650      ICW1 = WV(11)
426 00000651      ICTW1 = TWV(11)

```


Table A-2. Reduced-Order Component Model (Continued)

```

427 00000652      ICWD2=WD(12)
428 00000653      ICWV2=V(12)
429 00000654      ICTWV2=TWV(12)
430 00000655      ICWD3=WD(13)
431 00000656      ICWV3=V(13)
432 00000657      ICTWV3=TWV(13)
433 00000660      ICWD4=WD(14)
434 00000661      ICWV4=V(14)
435 00000662      ICTWV4=TWV(14)
436 00000663      ICWD5=WD(15)
437 00000664      ICWV5=V(15)
438 00000665      ICTWV5=TWV(15)
439 00000666      ICWD6=WD(16)
440 00000667      ICWV6=V(16)
441 00000670      ICTWV6=TWV(16)
442 00000671      ICWD7=WD(17)
443 00000672      ICWV7=V(17)
444 00000673      ICTWV7=TWV(17)
445 00000674      ICWD8=WD(18)
446 00000675      ICWV8=V(18)
447 00000676      ICTWV8=TWV(18)
448 00000677      ICTW8G = TW8G
449 00000700      ICW8GV = W8GV
450 00000701      ICWD8G = W8DG
451 00000702      ICTWCD = TWCD
452 00000703      ICWCD = WCD
453 00000704      ICWDCD = WDCD
454 00000705      ICWB = WB
455 00000706      ICPB = PB
456 00000707      ICMB = MB
457 00000710      ICRT = PT/K4/TT
458 00000711      ICRHT = HT*ICRT
459 00000712      DB 298 I=1,B
460 00000713      II=I+IC
461 00000714      KGAL(II)=KGAL(I)
462 00000715      298 KVBL(II)=KVBL(I)
463 00000716      TMIC=TB
464 00000717      TBSS=TB
465 00000720      TM=TB
466 00000721      3333 CONTINUE
467 00000722      NXUE=NX+NU+NE
468 00000723      NX1=NX+1
469 00000724      NXR=NX+NR
470 00000725      SIGN=1.
471 00000726      WRITE(9,8061)
472 00000727      WRITE(9,8060)(X(I),I=1,NX)
473 00000730      8060 FORMAT(E20.8)
474 00000731      8061 FORMAT(1H1)
475 00000732      WT=WT1
476 00000733      CALL DYNAM(A,PV,TV,KGAL,KVBL,KNR,RAD,KRAD,K,A,TWV,WD,WV,KBLD,DPB,1
477 00000734      1)
478 00000735      WB= .4BL3
479 00000736      WBL= .4BL4
480 00000737      WBL5S=.4BL5
481 00000740      WTS=WT
482 00000741      WTS=.4TC
483 00000742      WRITE(9,8061)
484 00000743      WRITE(9,9000)
485 00000744      9000 FORMAT(////,5X,'+1HSTEADY STATE DATA FROM SUBROUTINE DYNAMIC,////)
486 00000745      WRITE(9,9001)
487 00000746      9001 FORMAT(8X,'4X,14X,24WF,13X,24AB,13X,34BV8,12X,34ABL)

```

Table A-2. Reduced-Order Component Model (Continued)

```

488 00000747 WRITE(9,9010) N,NF,AB,BV3,ABL
489 00000750 WRITE(9,9002)
490 00000751 9002 FORMAT(//,4X,4HWBL3,11X,4HWBL4,11X,4HWBL5)
491 00000752 WRITE(9,9010) WBL3,WBL4,WBL5
492 00000753 WRITE(9,9003)
493 00000754 9003 FORMAT(//,7X,3HWCD,12X,3HTCD,12X,3HPCD,12X,3HMCD)
494 00000755 WRITE(9,9010) WCD,TCD,PCD,HCD
495 00000756 WRITE(9,9004)
496 00000757 9004 FORMAT(//,7X,2HWB,13X,2HTB,13X,2HPB,13X,2HMB,13X,2MKWB,11X,4HETAB)
497 00000760 WRITE(9,9010) AB,TB,PB,HB,KWB,ETAB
498 00000761 WRITE(9,9005)
499 00000762 9005 FORMAT(//,7X,2HWT,13X,2HTT,13X,2HPT,13X,2HMT)
500 00000763 WRITE(9,9010) AT,TT,PT,MT
501 00000764 WRITE(9,9006)
502 00000765 9006 FORMAT(//,7X,2HWN,12X,4KNAB,12X,3HNDT,11X,4HTMDT,12X,2HTM)
503 00000766 WRITE(9,9010) WKNAB,NDT,TMDT,TM
504 00000767 9010 FORMAT(2X,E12.5,7(3X,E12.5))
505 00000770 WRITE(9,8061)
506 00000771 WRITE(9,8060)(DX(I),I=1,NX)
507 00000772 WRITE(9,8061)
508 00000773 WRITE(9,8060)(DX(I),I=NX1,NXR)
509 00000774 DO 8031 I=1,NXR
510 00000775 8031 DX(I)=DX(I)
511 00000776 C ANF
512 00000777 C TWX
513 00000780 DO 8040 J=1,NXU
514 00000781 8032 SIGN=-1.*SIGN
515 00000782 IF(X(J).NE..0) GO TO 3703
516 00000783 IF(SIGN.LT..0) GO TO 8032
517 00000784 PERT=DPERT
518 00000785 GO TO 3704
519 00000786 3703 CONTINUE
520 00000787 PERT=SIGN*X(J)*DPERT
521 00000788 3704 CONTINUE
522 00000789 X(J)=X(J)+PERT
523 00000790 CALL DY,AK(A,PV,TV,KJAL,KVBL,KNR,RAD,KRAD,KA,TW,WD,WV,KBLD,DRB,2
524 00000791 1)
525 00000792 VOL3=WBL3S
526 00000793 WBL4=WBL4S
527 00000794 WBL5=WBL5S
528 00000795 AT=AT5
529 00000796 HT=HT5
530 00000797 X(J)=X(J)+PERT
531 00000798 IF(X(J).EQ..0) GO TO 3705
532 00000799 IF(SIGN.LT..0) GO TO 8033,8034,8035
533 00000800 8033 CONTINUE
534 00000801 IF(SIGN.LT..0) GO TO 8034,8035
535 00000802 DO 8034 I=1,NXR
536 00000803 DX(I)=DX(I)
537 00000804 C THREE
538 00000805 C FOUR
539 00000806 8034 DX(I)=DX(I)
540 00000807 GO TO 8032
541 00000808 3701 CONTINUE
542 00000809 SIGN=-1.*SIGN
543 00000810 GO 3707 I=1,NXR
544 00000811 DX(I)=DX(I)
545 00000812 3707 X(I)=DX(I)
546 00000813 PERT=DPERT
547 00000814 GO TO 8032
548 00000815 3705 CONTINUE

```

Table A-2. Reduced-Order Component Model (Continued)

```

549 00001044      DO 3702 I=1,NXR
550 00001045      3702 DX1(I)=DXN(I)
551 00001046      PERT=.5*PERT
552 00001047      C FIVE
553 00001050      C SIX
554 00001051      GO TO 8035
555 00001052      8035 CONTINUE
556 00001053      3074 FORMAT(1H1/7X,8H COLUMN 13/)
557 00001054      DO 8036 I=1,NXR
558 00001055      CLM(I)=(DX(I)+DX1(I))/(2.*ABS(PERT))
559 00001056      3076 FORMAT(13,4E20.10)
560 00001057      8036 DX(I)=DXN(I)
561 00001060      WRITE(3) (CLM(I),I=1,NX)
562 00001061      WRITE(3) (CLM(I),I=NX1,NXR)
563 00001062      8040 CONTINUE
564 00001063      GO TO 8099
565 00001064      8092 CONTINUE
566 00001065      PAUSE
567 00001066      GO TO 8099
568 00001067      END

```

```

1 00000000      FUNCTION INTGRL(IC,DXDT)
2 00000001      COMMON/CDATA,TIME,DTM
3 00000002      DIMENSION DN1(50)
4 00000003      REAL IC,INTGRL
5 00000004      INTGRL=IC
6 00000005      RETURN
7 00000006      END

```

```

1 00000000      FUNCTION IDX(I,J,NX,NY)
2 00000001      DIMENSION MX(5),MY(5)
3 00000002      KSUM=0
4 00000003      IF(N.EQ.1) GO TO 1
5 00000004      NN=N-1
6 00000005      DO 2 L=1,NN
7 00000006      2 KSUM=KSUM+MX(L)*NY(L)
8 00000007      1 CONTINUE
9 00000010      1 IDX=KSUM+1+(J-1)*NX(1)
10 00000011      RETURN
11 00000012      END

```

```

1 00000000      FUNCTION ABFN(NRAT)
2 00000001      REAL NRAT
3 00000002      IF(NRAT.GT..45) GO TO 1
4 00000003      ABFN=.162*01
5 00000004      RETURN
6 00000005      1 A=.7051/.15+.15
7 00000006      B=.7051/.15
8 00000007      ABFN=A+B*NRAT
9 00000010      RETURN
10 00000011      END

```

Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000      FUNCTION FN1SET(N,ZX,NP,N1,N2)
2 00000001      COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZ(593),YDEL(10),II(10),JJ(1
3 00000002      10),SLP1(10),SLP2(10),ZPT1(10),ZPT2(10)
4 00000003      COMMON X1(21,21),Z1(21,21),KK(40),MX(23),XDIF(40),SLP(40),ZPT(40)
5 00000004      DIMENSION ZX(1)
6 00000005      MX(N) = NP
7 00000006      DO 10 J=1,NP
8 00000007      K=2*J
9 00000010      X1(J,N) = ZX(K-1)
10 00000011     10 Z1(J,N) = ZX(K)
11 00000012      FN1SET = 1.0
12 00000013      DO 30 NR=N1,N2
13 00000014      KK(NR) = 2
14 00000015      XDIF(NR) = X1(2,N)-X1(1,N)
15 00000016      ZPT(NR) = Z1(1,N)
16 00000017      SLP(NR) = (Z1(2,N)-Z1(1,N))/XDIF(NR)
17 00000020     30 CONTINUE
18 00000021      RETURN
19 00000022      END

```

```

1 00000000      FUNCTION FN2SET(N,X,Y,Z,NXP,NYP,N1,N2)
2 00000001      COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZ(593),YDEL(10),II(10),JJ(1
3 00000002      10),SLP1(10),SLP2(10),ZPT1(10),ZPT2(10)
4 00000003      COMMON X1(21,21),Z1(21,21),KK(40),MX(23),XDIF(40),SLP(40),ZPT(40)
5 00000004      DIMENSION X(1),Y(1),Z(1)
6 00000005     10 NX(N) = NXP
7 00000006      NY(N) = NYP
8 00000007      DO 15 J=1,NYP
9 00000010      YY(J,N) = Y(J)
10 00000011      DO 15 I=1,NXP
11 00000012      K = 1+(J-1)*NXP
12 00000013      LL=IDX(I,J,N,NX,NY)
13 00000014     15 ZZZ(LL)=Z(K)
14 00000015      DO 20 I=1,NXP
15 00000016     20 XX(I,N) = X(I)
16 00000017      FN2SET = 1.0
17 00000020      DO 30 NR=N1,N2
18 00000021      II(NR) = 2
19 00000022      JJ(NR) = 2
20 00000023      XDEL = XX(2,N)-XX(1,N)
21 00000024      YDEL(NR) = YY(2,N)-YY(1,N)
22 00000025      LL=IDX(1,1,N,NX,NY)
23 00000026      ZPT1(NR)=ZZZ(LL)
24 00000027      LL=IDX(1,2,N,NX,NY)
25 00000030      ZPT2(NR)=ZZZ(LL)
26 00000031      LL=IDX(2,1,N,NX,NY)
27 00000032      SLP1(NR)=(ZZZ(LL)-ZPT1(NR))/XDEL
28 00000033      LL=IDX(2,2,N,NX,NY)
29 00000034      SLP2(NR)=(ZZZ(LL)-ZPT2(NR))/XDEL
30 00000035     30 CONTINUE
31 00000036      RETURN
32 00000037      END

```

Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000      FUNCTION RNFH(C)
2 00000001      XK=C
3 00000002      IF(C.LT..5R) GOTO 1
4 00000003      RNFH=1.
5 00000004      RETURN
6 00000005      1 XKS=XK*XK
7 00000006      A=(1.+2*XKS)
8 00000007      AS=A*A
9 00000010      AC=AS*A
10 00000011      UP=C*AC*XK
11 00000012      DN=1.2*C*XK*AS-1.
12 00000013      XKP1=XK-(UP/DN)
13 00000014      RAT=ABS(XKP1/XK)
14 00000015      RAT=ABS(RAT-1.)
15 00000016      IF(RAT.GT..001) GOTO 10
16 00000017      RNFH=XKP1
17 00000020      RETURN
18 00000021      10 XK=XKP1
19 00000022      GOTO 1
20 00000023      END

```

```

1 00000000      FUNCTION TFNH(NX, FAX, HX, TV)
2 00000001      DIMENSION TV(20)
3 00000002      DTX=50.
4 00000003      TX=TV(NX)
5 00000004      51 CALL PRBCOM(FAX, TX, CPX, GMX, GMXX, HX1, IFA)
6 00000005      IF(IFA.GT.1) GOTO 70
7 00000006      TXERR=HX*HX1
8 00000007      IF(TXERR.GT..001) GOTO 52
9 00000010      IF(TXERR.LT..001) GOTO 53
10 00000011      GOTO 60
11 00000012      52 IF(DTX)53,53,54
12 00000013      53 DTX=-DTX*.5
13 00000014      54 TX=TX+DTX
14 00000015      GOTO 51
15 00000016      55 IF(DTX)54,54,53
16 00000017      60 CONTINUE
17 00000020      70 CONTINUE
18 00000021      TFNH=TX
19 00000022      TV(NX)=TX
20 00000023      RETURN
21 00000024      END

```

Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000  SUBROUTINE PRBCRM(FARX,TEX,CP,GF,GMX,H,IFA)
2 00000001  IFA=1
3 00000002  IF(FARX.GT.0.) GOTO 2
4 00000003  FARX=0.
5 00000004  IFA=1
6 00000005  GOTO 3
7 00000006  IF(FARX.LT..067623) GOTO 3
8 00000007  FARX=.067623
9 00000010  IFA=1
10 00000011  IF(TEX=1500.) GOTO 10A
11 00000012  IF(TEX.LT.4000.) GOTO 7
12 00000013  TEX=4000.
13 00000014  IF(IFA.EQ.1) GOTO 50
14 00000015  IFA=2
15 00000016  GOTO 1A
16 00000017  IFA=3
17 00000020  GOTO 1A
18 00000021  IF(TEX=2300.) GOTO 14A
19 00000022  IF(TEX=2500.) GOTO 16A
20 00000023  IF(TEX=2000.) GOTO 12A
21 00000024  1 CPA = .2644+2.*E=5.*(TEX=1500.)
22 00000025  HA = (.22E19+1.*22E=5.*TEX)+TEX+2.*3733
23 00000026  GOTO 40
24 00000027  12 CPA = .27748+1.*2E=5.*(TEX=2000.)
25 00000028  HA = (.22E19+1.*22E=5.*TEX)+TEX+2.*3733
26 00000031  GOTO 40
27 00000032  1A CPA = .27748+1.*2E=5.*(TEX=2000.)
28 00000033  HA = (.25947+5.*36E=6.*TEX)+TEX+.7404
29 00000034  GOTO 40
30 00000035  17 CPA = .2864+1.*17E=5.*(TEX=2500.)
31 00000036  HA = (.25947+5.*36E=6.*TEX)+TEX+.7404
32 00000037  GOTO 40
33 00000040  21 IF(TEX.GT.3000.) GOTO 21A
34 00000041  TEX=3000.
35 00000042  IF(IFA.EQ.1) GOTO 11
36 00000043  IFA=2
37 00000044  GOTO 24
38 00000045  IFA=3
39 00000046  GOTO 24
40 00000047  21 IF(TEX=900.) GOTO 27A
41 00000050  IF(TEX=1200.) GOTO 28A
42 00000051  IF(TEX=700.) GOTO 24A
43 00000052  24 CPA = .2392+1.*1E=5.*(TEX=500.)
44 00000053  HA = (.22623+1.*26E=5.*TEX)+TEX+3.*5214
45 00000054  GOTO 40
46 00000055  2A CPA = .2414+3.*4E=5.*(TEX=700.)
47 00000056  HA = (.22623+1.*26E=5.*TEX)+TEX+3.*5214
48 00000057  GOTO 40
49 00000060  2A CPA = .2457+3.*1E=5.*(TEX=900.)
50 00000061  HA = (.22623+1.*26E=5.*TEX)+TEX+3.*5214
51 00000062  GOTO 40
52 00000063  3A CPA = .2457+3.*1E=5.*(TEX=900.)
53 00000064  HA = (.22579+1.*22E=5.*TEX)+TEX+2.*3733
54 00000065  41 CPF = (.9337+(5.*7E=5+3.*27E=8.*(3500.-TEX)))/(3500.-TEX)
55 00000066  HF = (.50800+6.*10E=5.*TEX)+TEX+132.20
56 00000067  CP = (CPA+FARX*CPF)/(1.+FARX)
57 00000070  H = (HA+FARX*HF)/(1.+FARX)
58 00000071  AMV = 28.97-.946184*FARX
59 00000072  REX = 1.98437/AMV
60 00000073  GM = CP/(CP-REX)
61 00000074  GMX = (GM-1.)/GM
62 00000075  RETURN
63 00000076  END

```

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Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000          FUNCTI@N FUN1(N,XIN,NR)
2 00000001          C@MM@N XX(17,5),YY(17,5),NX(5),NY(5),ZZZ(593),YDEL(10),I1(10),JJ(1
3 00000002          10),SLP1(10),SLP2(10),ZPT1(10),ZPT2(10)
4 00000003          C@MM@N X1(21,21),Z1(21,21),KK(40),MX(23),XDIF(40),SLP(40),ZPT(40)
5 00000004          C
6 00000005          I@LD = KK(NR)
7 00000006          NXP = MX(N)
8 00000007          IF(XIN-X1(I@LD,N)) 105,105,120
9 00000010          105 IF(XIN-X1(I@LD-1,N)) 140,140,110
10 00000011          110 I = I@LD
11 00000012          G@ T@ 250
12 00000013          C@UNT UP
13 00000014          120 IF(XIN-X1(NXP,N)) 125,180,300
14 00000015          125 NF = I@LD + 1
15 00000016          D@ 130 I = NF/NXP
16 00000017          IF(XIN-X1(I,N)) 200,200,130
17 00000020          130 C@NTINUE
18 00000021          G@ T@ 200
19 00000022          C@UNT D@WN
20 00000023          140 IF(XIN-X1(1,N)) 300, 190,145
21 00000024          145 NL = I@LD - 2
22 00000025          D@ 150 K = 1/NL
23 00000026          I = I@LD - K
24 00000027          IF(XIN-X1(I-1,N)) 150,150, 200
25 00000030          150 C@NTINUE
26 00000031          G@ T@ 200
27 00000032          180 I = NXP
28 00000033          G@ T@ 200
29 00000034          190 I = 2
30 00000035          200 XDIF(NR) = X1(I,N)-X1(I-1,N)
31 00000036          ZPT(NR) = Z1(I-1,N)
32 00000037          SLP(NR) = (Z1(I,N)-ZPT(NR))/XDIF(NR)
33 00000040          250 XINC = XIN-X1(I-1,N)
34 00000041          FUN1 = ZPT(NR)+XINC*SLP(NR)
35 00000042          KK(NR) = I
36 00000043          R@TURN
37 00000044          300 C@NTINUE
38 00000045          IF(XIN.LT.X1(1,N))FUN1=Z1(1,N)
39 00000046          IF(XIN.GT.X1(NXP,N))FUN1=Z1(NXP,N)
40 00000047          R@TURN
41 00000050          E@ND

```

Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000          FUNCTION FUN2(N,XIN,YIN,NR)
2 00000001          COMMON XX(17,5),YY(17,5),NX(5),NY(5),ZZZ(593),YDEL(10),I1(10),JJ(1
3 00000002          10),SLP1(10),SLP2(10),ZPT1(10),ZPT2(10)
4 00000003          COMMON X1(21,21),Z1(21,21),KK(40),MX(23),XDIF(40),SLP(40),ZPT(40)
5 00000004          C      TEST FOR X IN PREVIOUS INTERVAL
6 00000005          XP = NX(N)
7 00000006          IOLD = I1(NR)
8 00000007          IF(XIN = XX(IOLD,N)) 105,105,120
9 00000010          105 IF(XIN = XX(IOLD-1,N)) 140,140,110
10 00000011          110 I = IOLD
11 00000012          GO TO 200
12 00000013          C      COUNT UP
13 00000014          120 IF(XIN = XX(NXP,N)) 125,180,180
14 00000015          125 NF = IOLD + 1
15 00000016          DO 130 I = NF,NXP
16 00000017          IF(XIN = XX(I,N))200,200,130
17 00000020          130 CONTINUE
18 00000021          GO TO 200
19 00000022          C      COUNT DOWN
20 00000023          140 IF(XIN=XX(1,N)) 190,190,145
21 00000024          145 NL = IOLD - 2
22 00000025          DO 150 K = 1,NL
23 00000026          I = IOLD - K
24 00000027          IF(XIN=XX(I-1,N))150,150,200
25 00000030          150 CONTINUE
26 00000031          GO TO 200
27 00000032          180 I=NXP
28 00000033          XIN=XX(NXP,N)
29 00000034          GO TO 200
30 00000035          130 I = 2
31 00000036          XIN=XX(1,N)
32 00000037          C      TEST FOR Y IN PREVIOUS INTERVAL
33 00000040          200 NYP = NY(N)
34 00000041          JOLD = JJ(NR)
35 00000042          IF(YIN = YY(JOLD,N)) 205, 205, 220
36 00000043          205 IF(YIN = YY(JOLD-1,N)) 240,240,210
37 00000044          210 J = JOLD
38 00000045          IF(I=IOLD) 300,400,300
39 00000046          C      COUNT UP
40 00000047          220 IF(YIN = YY(NYP,N)) 225, 280,280
41 00000050          225 NF = JOLD + 1
42 00000051          DO 230 J = NF,NYP
43 00000052          IF(YIN = YY(J,N)) 300,300,230
44 00000053          230 CONTINUE
45 00000054          GO TO 300
46 00000055          C      COUNT DOWN
47 00000056          240 IF(YIN = YY(1,N))290, 290,245
48 00000057          245 NL = JOLD - 2
49 00000060          DO 250 K = 1,NL
50 00000061          J = JOLD - K
51 00000062          IF(YIN =YY(J-1,N)) 280,280,300
52 00000063          250 CONTINUE
53 00000064          GO TO 300
54 00000065          280 J = NYP
55 00000066          YIN=YY(NYP,N)
56 00000067          GO TO 300
57 00000070          290 J = 1
58 00000071          YIN=YY(1,N)
59 00000072          C      COMPUTE Z(Y) INTERCEPTS AND SLOPES
60 00000073          300 XDEL = XX(1,N)-XX(I-1,N)

```


Table A-2. Reduced-Order Component Model (Continued)

```

61 00000074 YDEL(NR) = YY(J,N)-YY(J-1,N)
62 00000075 LL=IDX(I=1,J=1,N,NX,NY)
63 00000076 ZPT1(NR)=ZZZ(LL)
64 00000077 LL=IDX(I=1,J,N,NX,NY)
65 00000100 ZPT2(NR)=ZZZ(LL)
66 00000101 LL=IDX(I=1,J=1,N,NX,NY)
67 00000102 SLP1(NR)=(ZZZ(LL)-ZPT1(NR))/XDEL
68 00000103 LL=IDX(I=1,J,N,NX,NY)
69 00000104 SLP2(NR)=(ZZZ(LL)-ZPT2(NR))/XDEL
70 00000105 C INTERPOLATE FOR ANSWER
71 00000106 *00 II(NR) = I
72 00000107 JJ(NR) = J
73 00000110 XINC = XIN-XX(I=1,N)
74 00000111 P1ZZ = ZPT1(NR)+XINC*SLP1(NR)
75 00000112 P2ZZ = ZPT2(NR)+XINC*SLP2(NR)
76 00000113 YFRAC = (YIN-YY(J=1,N))/YDEL(NR)
77 00000114 FUN2 = P1ZZ + YFRAC*(P2ZZ-P1ZZ)
78 00000115 RETURN
79 00000116 END

```

Table A-2. Reduced-Order Component Model (Continued)

```

1 00000000 SUBROUTINE DYNAM(A,PV,TV,KGAL,KVBL,KR,RAD,KRAD,KA,TWV,ND,KV,KBL,
2 00000001 ISP5,ICIT)
3 00000002 DIMENSION A(20),PV(20),TV(20),KGAL(20),KVBL(20),KNR(8),RAD(8)
4 00000003 DIMENSION KRAD(8),KA(30),TAV(20),ND(20),KV(20),KBLD(8),DPR(14)
5 00000004 COMMON/TDATA/TIME,DTM
6 00000005 COMMON/DATA/X(2),L(4),ETA(3),DX(6),DX(6),DX(6),DX(6),CLM(6),KVBL(
7 00000006 10),RTH,KVFLG,KALBG,KVGV,TC,KVBLCD,KGALCD,KAB,KGALB,
8 00000007 PHT,KK2,THCD,KVBLP,PO,KNA,KAB,KSPED,KGALIG,KFIGV,K3,
9 00000010 KBL3,KBL4,KBL5,NC0,NC0D,TWCD,ETA,LHT
10 00000011 *TCD,KB,HB,PB,ERROR
11 00000012 REAL KB,KR,NC1,ICV,NC1',IGVPR,IGV,K1,K3,K4,K5,K7,L,KGAL,KVBL
12 00000013 REAL KVBLB,KGALB,KVBLCD,KGALCD,KGALB,KAB,KVBLB,KVBLT,KSPED
13 00000014 REAL KFIGV,NCX,KNR,KRAD,KA,KGV,KNA,KRTTB,ICPVO,ICD,IC,IC,1
14 00000015 REAL IC,D2, IC,D3, IC,D4
15 00000016 REAL IC,D5, IC,D6, IC,D7
16 00000017 REAL IC,DR, IC,DBG,ICT,CD,IC,CD
17 00000020 REAL IC,DCD,IC,B,ICPB,ICHB,ICRT,ICRT,IC,F,P,NC2,NC3,NC4,NC5,NC6
18 00000021 REAL NC7,NC8,NC9D,ITER,IMPL,INTGRL,KIC
19 00000022 REAL KAB,IC,DBG,N,DT
20 00000023 REAL KALD,K2,NPAT,KGALIG,KVBLI
21 00000024 EQUIVALENCE(ICN,N,X(1))
22 00000025 EQUIVALENCE
23 00000026 2 (NDT,DX(1)),(WF,U(1)),(BV8,U(2)),(AR,U(3)),(P2,ETA(1
24 00000027 3)),(TV,T2,ETA(2)),(PB,ETA(3)),(U(4),A:3L)
25 00000030 EQUIVALENCE(TM,IC,TMX(2)),(TM,T,DX(2))
26 00000031 EQUIVALENCE(PCD,DX(3)),(PT,DX(4)),(TB,DX(5)),(TT,DX(6))
27 00000032 K=NT*AF
28 00000033 KCD=K+TC+KBL3+KBL4+KBL5
29 00000034 90 CONTINUE
30 00000035 DELPT=0.1
31 00000036 DELW0=C*0.1
32 00000037 ITER1=0
33 00000040 ITERP=0
34 00000041 99 CONTINUE
35 00000042 RTH=SQRT(TV0/518.7)
36 00000043 DB 299 I=1,5
37 00000044 299 KNR(I)=(NRAD(I))*2/K2
38 00000045 C
39 00000046 C DYNAMICS
40 00000047 C
41 00000050 C INLET AND STAGE B/E
42 00000051 C
43 00000052 NC1 = V/RTH0
44 00000053 NC1N = NC1/16500
45 00000054 IGVPR = FUN1(2,NC1,2)
46 00000055 PVO = P2+IGVPR + .005*P2
47 00000056 W01 = W00
48 00000057 DELO = PVO/14.7
49 00000060 FP1 = W01*RTH0/(DELO+A(11))
50 00000061 VZT1 = KA(1) + KA(11)*FP1 + KA(21)*FP1*FP1
51 00000062 PH11 = VZT1/(KRAD(1)*NC1)
52 00000063 PSIP1 = FUN2(2,PH11,BV8,4)
53 00000064 PD1 = PVO*(1+PSIP1)*KNR(1)/TV0)**3.5
54 00000065 PV1 = PD1
55 00000066 PSIT1 = FUN2(5,PH11,BV8,10)
56 00000067 TD1 = TV0*KNR(1)*PSIT1
57 00000070 TV1 = TD1
58 00000071 C
59 00000072 C STAGE T/W
60 00000073 C

```

Table A-2. Reduced-Order Component Model (Continued)

61	00000074	RTH1 = SQRT(TV1/518.7)
62	00000075	NC2 = N/RTH1
63	00000076	DEL1 = PV1/14.7
64	00000077	WD2 = WD1
65	00000100	FP2 = WD2*RTH1/(DEL1*A(12))
66	00000101	VZT2 = KA(2) + KA(12)*FP2 + KA(22)*FP2*FP2
67	00000102	PHI2 = VZT2/(KRAD(2)*NC2)
68	00000103	PSIP2 = FUN1(5, PHI2, 7)
69	00000104	PD2 = PV1*(1.0+PSIP2*KNR(2)/TV1)**3.5
70	00000105	PV2 = PD2
71	00000106	PSIT2 = FUN1(6, PHI2, 9)
72	00000107	TD2 = TV1*KNR(2)*PSIT2
73	00000110	TV2 = TD2
74	00000111	C
75	00000112	C STAGE THREE
76	00000113	C
77	00000114	RTH2 = SQRT(TV2/518.7)
78	00000115	NC3 = N/RTH2
79	00000116	DEL2 = PV2/14.7
80	00000117	WD3 = WD2
81	00000120	FP3 = WD3*RTH2/(DEL2*A(13))
82	00000121	VZT3 = KA(3) + KA(13)*FP3 + KA(23)*FP3*FP3
83	00000122	PHI3 = VZT3/(KRAD(3)*NC3)
84	00000123	PSIP3 = FUN1(7, PHI3, 11)
85	00000124	PD3 = PV2*(1.0+PSIP3*KNR(3)/TV2)**3.5
86	00000125	PV3 = PD3
87	00000126	PSIT3 = FUN1(8, PHI3, 13)
88	00000127	TD3 = TV2*KNR(3)*PSIT3
89	00000130	TV3 = TD3
90	00000131	RTH3 = SQRT(TV3/518.7)
91	00000132	WBL3 = KBLD(3)*ABL*PV3/(K3*RTH3)
92	00000133	C
93	00000134	C STAGE FOUR
94	00000135	C
95	00000136	WD4 = WD3-WBL3
96	00000137	NC4 = N/RTH3
97	00000140	DEL3 = PV3/14.7
98	00000141	FP4 = WD4*RTH3/(DEL3*A(14))
99	00000142	VZT4 = KA(4) + KA(14)*FP4 + KA(24)*FP4*FP4
100	00000143	PHI4 = VZT4/(KRAD(4)*NC4)
101	00000144	PSIP4 = FUN1(9, PHI4, 15)
102	00000145	PD4 = PV3*(1.0+PSIP4*KNR(4)/TV3)**3.5
103	00000146	PV4 = PD4
104	00000147	PSIT4 = FUN1(10, PHI4, 17)
105	00000150	TD4 = TV3*KNR(4)*PSIT4
106	00000151	TV4 = TD4
107	00000152	RTH4 = SQRT(TV4/518.7)
108	00000153	WBL4 = KBLD(4)*ABL*PV4/(K3*RTH4)
109	00000154	C
110	00000155	C STAGE FIVE
111	00000156	C
112	00000157	WD5 = WD4-WBL4
113	00000160	NC5 = N/RTH4
114	00000161	DEL4 = PV4/14.7
115	00000162	FP5 = WD5*RTH4/(DEL4*A(15))
116	00000163	VZT5 = KA(5) + KA(15)*FP5 + KA(25)*FP5*FP5
117	00000164	PHI5 = VZT5/(KRAD(5)*NC5)
118	00000165	PSIP5 = FUN1(11, PHI5, 19)
119	00000166	PD5 = PV4*(1.0+PSIP5*KNR(5)/TV4)**3.5
120	00000167	PV5 = PD5
121	00000170	PSIT5 = FUN1(12, PHI5, 21)

Table A-2. Reduced-Order Component Model (Continued)

```

122 00000171      TD6 = TV6*KNR(6)*PSIT6
123 00000172      TV6 = TDS
124 00000173      RTH6 = SQRT(TV6/518.7)
125 00000174      WDL6 = KBLD(6)*ABL*PVS/(K3*RTM6)
126 00000175
127 00000176      C
128 00000177      C
129 00000200      C
130 00000201      C
131 00000202      C
132 00000203      STAGE SIX
133 00000204      WDL6 = WDL6*WDL6
134 00000205      NCL6 = N/RTH6
135 00000206      DEL6 = PVS/14.7
136 00000207      FP6 = WDL6*RTH6/(DEL6*A(16))
137 00000210      VZT6 = KA(6) + KA(16)*FP6 + KA(26)*FP6*FP6
138 00000211      PH16 = VZT6/(KRAD(6)*NCL6)
139 00000212      PSIP6 = FUN1(13,PH16,23)
140 00000213      PD6 = PVS*(1.+PSIP6*KNR(6)/TV6)**3.5
141 00000214      PV6 = PD6
142 00000215      PSIT6 = FUN1(14,PH16,25)
143 00000216      TD6 = TV6*KNR(6)*PSIT6
144 00000217      TV6 = TD6
145 00000220      C
146 00000221      C
147 00000222      C
148 00000223      STAGE SEVEN
149 00000224      WDL7 = WDL6
150 00000225      RTH7 = SQRT(TV6/518.7)
151 00000226      NCL7 = N/RTH7
152 00000227      DEL7 = PVS/14.7
153 00000230      FP7 = WDL7*RTH7/(DEL7*A(17))
154 00000231      VZT7 = KA(7) + KA(17)*FP7 + KA(27)*FP7*FP7
155 00000232      PH17 = VZT7/(KRAD(7)*NCL7)
156 00000233      PSIP7 = FUN1(15,PH17,27)
157 00000234      PD7 = PVS*(1.+PSIP7*KNR(7)/TV6)**3.5
158 00000235      PV7 = PD7
159 00000236      PSIT7 = FUN1(16,PH17,29)
160 00000237      TD7 = TV6*KNR(7)*PSIT7
161 00000240      TV7 = TD7
162 00000241      C
163 00000242      C
164 00000243      C
165 00000244      STAGE EIGHT
166 00000245      WDL8 = WDL7
167 00000246      RTH8 = SQRT(TV7/518.7)
168 00000247      NCL8 = N/RTH8
169 00000250      DEL8 = PVS/14.7
170 00000251      FP8 = WDL8*RTH8/(DEL8*A(18))
171 00000252      VZT8 = KA(8) + KA(18)*FP8 + KA(28)*FP8*FP8
172 00000253      PH18 = VZT8/(KRAD(8)*NCL8)
173 00000254      PSIP8 = FUN1(17,PH18,31)
174 00000255      PD8 = PVS*(1.+PSIP8*KNR(8)/TV7)**3.5
175 00000256      PV8 = PD8
176 00000257      PSIT8 = FUN1(18,PH18,33)
177 00000260      TD8 = TV7*KNR(8)*PSIT8
178 00000261      TV8 = TD8
179 00000262      C
180 00000263      C
181 00000264      C
182 00000265      C
183 00000266      C
184 00000267      C
185 00000268      C
186 00000269      C
187 00000270      C
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617 00000700      C
618 00000701      C
619 00000702      C
620 00000703      C
621 00000704      C
622 00000705      C
623 00000706      C
624 00000707      C
625 00000708      C
626 00000709      C
627 00000710      C
628 00000711      C
629 00000712      C
630 00000713      C
631 00000714      C
632 00000715      C
633 00000716      C
634 00000717      C
635 00000718      C
636 00000719      C
637 00000720      C
638 00000721      C
639 00000722      C
640 00000723      C
641 00000724      C
642 00000725      C
643 00000726      C
644 00000727      C
645 00000728      C
646 00000729      C
647 00000730      C
648 00000731      C
649 00000732      C
650 00000733      C
651 00000734      C
652 00000735      C
653 00000736      C
654 00000737      C
655 00000738      C
656 00000739      C
657 00000740      C
658 00000741      C
659 00000742      C
660 00000743      C
661 00000744      C
662 00000745      C
663 00000746      C
664 00000747      C
665 00000748      C
666 00000749      C
667 00000750      C
668 00000751      C
669 00000752      C
670 00000753      C
671 00000754      C
672 00000755      C
673 00000756      C
674 00000757      C
675 00000758      C
676 00000759      C
677 00000760      C
678 00000761      C
679 00000762      C
680 00000763      C
681 00000764      C
682 00000765      C
683 00000766      C
684 00000767      C
685 00000768      C
686 00000769      C
687 00000770      C
688 00000771      C
689 00000772      C
690 00000773      C
691 00000774      C
692 00000775      C
693 00000776      C
694 00000777      C
695 00000778      C
696 00000779      C
697 00000780      C
698 00000781      C
699 00000782      C
700 00000783      C
701 00000784      C
702 00000785      C
703 00000786      C
704 00000787      C
705 00000788      C
706 00000789      C
707 00000790      C
708 00000791      C
709 00000792      C
710 00000793      C
711 00000794      C
712 00000795      C
713 00000796      C
714 00000797      C
715 00000798      C
716 00000799      C
717 00000800      C
718 00000801      C
719 00000802      C
720 00000803      C
721 00000804      C
722 00000805      C
723 00000806      C
724 00000807      C
725 00000808      C
726 00000809      C
727 00000810      C
728 00000811      C
729 00000812      C
730 00000813      C
731 00000814      C
732 00000815      C
733 00000816      C
734 00000817      C
735 00000818      C
736 00000819      C
737 00000820      C
738 00000821      C
739 00000822      C
740 00000823      C
741 00000824      C
742 00000825      C
743 00000826      C
744 00000827      C
745 00000828      C
746 00000829      C
747 00000830      C
748 00000831      C
749 00000832      C
750 00000833      C
751 00000834      C
752 00000835      C
753 00000836      C
754 00000837      C
755 00000838      C
756 00000839      C
757 00000840      C
758 00000841      C
759 00000842      C
760 00000843      C
761 00000844      C
762 00000845      C
763 00000846      C
764 00000847      C
765 00000848      C
766 00000849      C
767 00000850      C
768 00000851      C
769 00000852      C
770 00000853      C
771 00000854      C
772 00000855      C
773 00000856      C
774 00000857      C
775 00000858      C
776 00000859      C
777 00000860      C
778 00000861      C
779 00000862      C
780 00000863      C
781 00000864      C
782 00000865      C
783 00000866      C
784 00000867      C
785 00000868      C
786 00000869      C
787 00000870      C
788 00000871      C
789 00000872      C
790 00000873      C
791 00000874      C
792 00000875      C
793 00000876      C
794 00000877      C
795 00000878      C
796 00000879      C
797 00000880      C
798 00000881      C
799 00000882      C
800 00000883      C
801 00000884      C
802 00000885      C
803 00000886      C
804 00000887      C
805 00000888      C
806 00000889      C
807 00000890      C
808 00000891      C
809 00000892      C
810 00000893      C
811 00000894      C
812 00000895      C
813 00000896      C
814 00000897      C
815 00000898      C
816 00000899      C
817 00000900      C
818 00000901      C
819 00000902      C
820 00000903      C
821 00000904      C
822 00000905      C
823 00000906      C
824 00000907      C
825 00000908      C
826 00000909      C
827 00000910      C
828 00000911      C
829 00000912      C
830 00000913      C
831 00000914      C
832 00000915      C
833 00000916      C
834 00000917      C
835 00000918      C
836 00000919      C
837 00000920      C
838 00000921      C
839 00000922      C
840 00000923      C
841 00000924      C
842 00000925      C
843 00000926      C
844 00000927      C
845 00000928      C
846 00000929      C
847 00000930      C
848 00000931      C
849 00000932      C
850 00000933      C
851 00000934      C
852 00000935      C
853 00000936      C
854 00000937      C
855 00000938      C
856 00000939      C
857 00000940      C
858 00000941      C
859 00000942      C
860 00000943      C
861 00000944      C
862 00000945      C
863 00000946      C
864 00000947      C
865 00000948      C
866 00000949      C
867 00000950      C
868 00000951      C
869 00000952      C
870 00000953      C
871 00000954      C
872 00000955      C
873 00000956      C
874 00000957      C
875 00000958      C
876 00000959      C
877 00000960      C
878 00000961      C
879 00000962      C
880 00000963      C
881 00000964      C
882 00000965      C
883 00000966      C
884 00000967      C
885 00000968      C
886 00000969      C
887 00000970      C
888 00000971      C
889 00000972      C
890 00000973      C
891 00000974      C
892 00000975      C
893 00000976      C
894 00000977      C
895 00000978      C
896 00000979      C
897 00000980      C
898 00000981      C
899 00000982      C
900 00000983      C
901 00000984      C
902 00000985      C
903 00000986      C
904 00000987     
```

Table A-2. Reduced-Order Component Model (Continued)

```

183 0000266      WTC = .03*WDO
184 0000267      TCD = TGGV
185 0000270      WCD = WGGV
186 0000271      PCD = PGGV
187 0000272      TWCD = PCD/KVGLCD
188 0000273      WDCD = TWCD/TCD
189 0000274
190 0000275      C
191 0000276      C BURNER
192 0000277      C
193 0000300      CALL PRBCOM(0.,TCD,CPCD,GMCD,GMCDX,MCD,IFA)
194 0000301      WB=WCD-WTC
195 0000302      FAB=WF/WB
196 0000303      NRAT=N/16500.
197 0000304      KWB=FUN1(20,NRAT,37)
198 0000305      140 ETAB0=ETAB
199 0000306      MB=MCD*.1650.*ETAB*FAB
200 0000307      TEB=TFNH(2,FAB,MB,TV)
201 0000310      IF(INIT.EQ.1) TM=TEB
202 0000311      TMDT=.5*0.248*(TEB-TM)/(15.*0.12)
203 0000312      TB=(0.24*WB*TEB-15.*0.12*TMDT)/(0.24*WB)
204 0000313      DELPB = KWB*WB*.2/PCD*(.771*TCD+.085*TB)
205 0000314      PB=PCD*DELPB
206 0000315      PBDLTB=PB*(TB-TCD)
207 0000316      ETAB=FUN1(19,PBDLTB,35)
208 0000317      IF(ABS(ETAB-ETAB0).GT.1.E-10) GO TO 140
209 0000320      NRTTB = N/SQRT(TB)
210 0000321      WT=WB*WF
211 0000322      FAT=WF/(WT+WTC)
212 0000323      PTPB = PT/PB
213 0000324      WTTNPB = FUN2(3,PTPB,NRTTB,6)
214 0000325      WTCAL = WTTNPB*N*PB/TB
215 0000326      PTERR=WTCAL-WT
216 0000327      WN=WT+WTC
217 0000330      DHTNTB = FUN2(4,PTPB,NRTTB,8)
218 0000331      DHT=N/1000.*DHTNTB*SQRT(TB)
219 0000332      HT=MB*DHT
220 0000333      HT=(WT*HT+WTC*MCD)/(WT+WTC)
221 0000334      TT=TFNH(3,FAT,HT,TV)
222 0000335      POPT=PB/PT
223 0000336      WNTKNP=HKEY(POPT)
224 0000337      KNAB=5.0405-.0429772*AB+.000126664*AB**2
225 0000340      WNCAL = WNTKNP*AB*KNAB*PT/SQRT(TT)
226 0000341      WNERR=WNCAL-WN
227 0000342      WBLTBL = WBL3*TV3+WBL4*TV4+WBL5*TV5
228 0000343      DLWMC=MCD*WCD+.2*(WBLTBL-TV0*WDO)
229 0000344      DLWHT=WT*DHT
230 0000345      NDT=KSPEED/N*(DLWHT-DLWMC)
231 0000346      IF(ABS(PTERR).LT.ERROR.AND.ABS(WNERR).LT.ERROR) GO TO 100
232 0000347      C GRADIENT CALCULATION
233 0000350      ITER1=ITER1+1
234 0000351      GO TO (10,20,30,40,50) ITER1
235 0000352      10 F=PTERR
236 0000353      G=WNERR
237 0000354      IF(SENSE SWITCH 5) 11,12
238 0000355      11 OUTPUT(9) WDO,PVO,TVO,W01,PD1,TD1,W02,PD2,TD2,W03,PD3,TD3,WBL3,W04
239 0000356      1,PD4,TD4,WBL4,W05,PD5,TD5,WBL5,W06,PD6,TD6,W07,PD7,TD7,W08,PD8,
240 0000357      2,TO8,W09,PD9,TD9,WTC,PCD,TCD,WCD,MCD,WF,WB,PB,TB,MB,WT,PT,TT,HT,
241 0000360      3,WN,FAB,FAT,ETAB,KNAB,DELPB,NRTTB,WTTNPB,PBDLTB,DHTNTB,DHT,WBLTBL,
242 0000361      4,DLWMC,DLWHT,NDT,N,PT,WDO,PTERR,WNERR,WTCAL,WT,WNCAL,WN,ITER1,ITER2
243 0000362      OUTPUT(9) TEB,TM,TMDT
IF(SENSE SWITCH 6) 13,12

```

Table A-2. Reduced-Order Component Model (Concluded)

```

244 00000363      13 CONTINUE
245 00000364      PAUSE
246 00000365      READ(5,300) IDUM
247 00000366      READ(5,300) PT,WDO
248 00000367      300 FORMAT(2E12.5)
249 00000370      GO TO 95
250 00000371      12 ITER2=ITER2+1
251 00000372      PT=PT+DELPT
252 00000373      GO TO 99
253 00000374      20 FY=PTERR
254 00000375      GX=WNERR
255 00000376      PT=PT-2.*DELPT
256 00000377      GO TO 99
257 00000400      30 FX=(FX-PTERR)/(2.*DELPT)
258 00000401      GX=(GX-WNERR)/(2.*DELPT)
259 00000402      PT=PT+DELPT
260 00000403      WDO=WDO+DELWDO
261 00000404      GO TO 99
262 00000405      40 FY=PTERR
263 00000406      GY=WNERR
264 00000407      WDO=WDO-2.*DELWDO
265 00000410      GO TO 99
266 00000411      50 FY=(FY-PTERR)/(2.*DELWDO)
267 00000412      GY=(GY-WNERR)/(2.*DELWDO)
268 00000413      WDO=WDO+DELWDO
269 00000414      D=FX-GY-GX*FY
270 00000415      IF(ABS(D).LT.0.000001) STOP 77
271 00000416      DX=(-F*GY+G*FY)/D
272 00000417      DYY=(-G*FX+F*GX)/D
273 00000420      IF(ABS(DXX).LT.(2.*DELPT)) GO TO 60
274 00000421      FACTOR=2.*DELPT/ABS(DXX)
275 00000422      DXX=FACTOR*DXX
276 00000423      DYY=FACTOR*DYY
277 00000424      60 IF(ABS(DYY).LT.(2.*DELWDO)) GO TO 70
278 00000425      FACTOR=2.*DELWDO/ABS(DYY)
279 00000426      DYY=FACTOR*DYY
280 00000427      DXX=FACTOR*DXX
281 00000430      70 PT=PT+DXX
282 00000431      WDO=WDO+DYY
283 00000432      ITER1=0
284 00000433      GO TO 99
285 00000434      100 CONTINUE
286 00000435      N = INTGR(L,ICN,NDT)
287 00000436      IF(SENSE SWITCH 5) 110,120
288 00000437      110 WRITE(9,511) ITER2
289 00000440      511 FORMAT(1H1,5X,12HCONVERGED IN,110,12H ITERATIONS)
290 00000441      OUTPUT(9) N,W,F,AB,BV0,ABL,NDT,PTERR,WNERR,PT,WDO
291 00000442      WRITE(9,512)
292 00000443      512 FORMAT(1H1)
293 00000444      12 CONTINUE
294 00000445      RETURN
295 00000446      END

```

```

1 00000000      FUNCTION H0KEY(P0PT)
2 00000001      IF(P0PT.GE.1.) GO TO 1
3 00000002      IF(P0PT.GE..53) H0KEY=P0PT*((1./1.4)*SQRT(1.-P0PT*(.4/1.4)))
4 00000003      IF(P0PT.GE.0..AND.P0PT.LE..53) H0KEY=.2588
5 00000004      RETURN
6 00000005      1 H0KEY=0.
7 00000006      RETURN
8 00000007      END

```

Table A-3. Engine Component Characteristics

FUNCTION F11: ABLB = f [BVOB]

<u>BVOB</u>	<u>ABLB</u>
•00000E 00	•00000E 00
•10000E 00	•18000E 00
•20000E 00	•33000E 00
•25000E 00	•39500E 00
•30000E 00	•45500E 00
•40000E 00	•54500E 00
•50000E 00	•63000E 00
•70000E 00	•78800E 00
•10000E 01	•10000E 01

FUNCTION F12: IGVPR = f [N/N_{MAX}]

<u>N/N_{MAX}</u>	<u>IGVPR</u>
•00000E 00	•99800E 00
•60000E 00	•99800E 00
•65000E 00	•99750E 00
•70000E 00	•99680E 00
•75000E 00	•99570E 00
•80000E 00	•99400E 00
•85000E 00	•99200E 00
•90000E 00	•98980E 00
•95000E 00	•98750E 00
•10000E 01	•98500E 00

FUNCTION F13: OGVPR = f [N/N_{MAX}]

<u>N/N_{MAX}</u>	<u>OGVPR</u>
•00000E 00	•99800E 00
•60000E 00	•99800E 00
•65000E 00	•99750E 00
•70000E 00	•99680E 00
•75000E 00	•99620E 00
•80000E 00	•99570E 00
•85000E 00	•99530E 00
•90000E 00	•99500E 00
•10000E 01	•99500E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F15: $\psi_2^P = f[\phi_2]$

ϕ_2	ψ_2^P
•00000E 00	•25000E 00
•45000E 00	•69500E 00
•50000E 00	•74550E 00
•55000E 00	•79200E 00
•56800E 00	•80700E 00
•58000E 00	•81600E 00
•60000E 00	•83100E 00
•62000E 00	•84400E 00
•64000E 00	•85600E 00
•66000E 00	•86600E 00
•68000E 00	•87300E 00
•70000E 00	•87800E 00
•72000E 00	•87900E 00
•73000E 00	•87800E 00
•74000E 00	•86800E 00
•75000E 00	•85300E 00
•76000E 00	•82700E 00
•76700E 00	•78000E 00
•79000E 00	•50000E 00
•80500E 00	•25000E-01

FUNCTION F16: $\psi_2^T = f[\phi_2]$

ϕ_2	ψ_2^T
•00000E 00	•29500E 01
•45000E 00	•10000E 01
•50000E 00	•97000E 00
•55000E 00	•95500E 00
•56800E 00	•95300E 00
•58000E 00	•95300E 00
•60000E 00	•95200E 00
•62000E 00	•95600E 00
•64000E 00	•96000E 00
•66000E 00	•97000E 00
•68000E 00	•97500E 00
•70000E 00	•98000E 00
•72000E 00	•98000E 00
•73000E 00	•97800E 00
•74000E 00	•97200E 00
•75000E 00	•96200E 00
•76000E 00	•93500E 00
•76700E 00	•90300E 00
•79000E 00	•62000E 00
•80500E 00	•22000E 00
•82000E 00	•26000E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F17: $\psi_3^P = f[\phi_3]$

ϕ_3	ψ_3^P
•00000E 00	•55000E 00
•50000E 00	•69300E 00
•53000E 00	•70200E 00
•57000E 00	•71200E 00
•58000E 00	•71500E 00
•60000E 00	•71900E 00
•62000E 00	•72400E 00
•64000E 00	•72800E 00
•65000E 00	•73000E 00
•66000E 00	•73300E 00
•67000E 00	•73400E 00
•68000E 00	•72900E 00
•69000E 00	•71800E 00
•69500E 00	•70400E 00
•69800E 00	•67000E 00
•69900E 00	•62400E 00
•70400E 00	•39400E 00

FUNCTION F18: $\psi_3^T = f[\phi_3]$

ϕ_3	ψ_3^T
•00000E 00	•10600E 01
•50000E 00	•83500E 00
•53000E 00	•82500E 00
•57000E 00	•82000E 00
•58000E 00	•81800E 00
•60000E 00	•82000E 00
•62000E 00	•82200E 00
•64000E 00	•82500E 00
•65000E 00	•82800E 00
•66000E 00	•83100E 00
•67000E 00	•83200E 00
•68000E 00	•83000E 00
•69000E 00	•81800E 00
•69500E 00	•80200E 00
•69800E 00	•71600E 00
•69900E 00	•68500E 00
•70400E 00	•62000E 00
•72000E 00	•40000E 00
•74000E 00	•12300E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F19: $\psi_4^P = f[\phi_4]$

ϕ_4	ψ_4^P
•00000E 00	•88000E 00
•53000E 00	•84200E 00
•55000E 00	•84100E 00
•57000E 00	•83600E 00
•58000E 00	•83000E 00
•60000E 00	•81900E 00
•61000E 00	•81300E 00
•62000E 00	•80700E 00
•63000E 00	•79900E 00
•64000E 00	•79200E 00
•65000E 00	•78300E 00
•65700E 00	•77700E 00
•66000E 00	•77300E 00
•66300E 00	•76600E 00
•66900E 00	•75200E 00
•67500E 00	•73800E 00
•72500E 00	•00000E 00
•77500E 00	•73800E 00

FUNCTION F110: $\psi_4^T = f[\phi_4]$

ϕ_4	ψ_4^T
•00000E 00	•14650E 01
•53000E 00	•98500E 00
•55000E 00	•96300E 00
•57000E 00	•94800E 00
•58000E 00	•93200E 00
•60000E 00	•92300E 00
•61000E 00	•91400E 00
•62000E 00	•90400E 00
•63000E 00	•89300E 00
•64000E 00	•88400E 00
•65000E 00	•87700E 00
•65700E 00	•87500E 00
•66000E 00	•87500E 00
•66300E 00	•87500E 00
•66900E 00	•88000E 00
•67500E 00	•88500E 00
•68500E 00	•86200E 00
•70000E 00	•62000E 00
•72500E 00	•13000E 00
•77500E 00	•87000E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F111: $\psi_5^P = f[\phi_5]$

ϕ_5	ψ_4^P
.00000E 00	.70000E 00
.52000E 00	.70000E 00
.54000E 00	.69700E 00
.55800E 00	.69100E 00
.57000E 00	.68500E 00
.58000E 00	.67800E 00
.59000E 00	.67200E 00
.59500E 00	.66700E 00
.60000E 00	.66300E 00
.61000E 00	.64800E 00
.61500E 00	.63600E 00
.62000E 00	.61700E 00
.62500E 00	.57900E 00
.64000E 00	.37200E 00
.66250E 00	.00000E 00
.68500E 00	-.37200E 00

FUNCTION F112: $\psi_5^T = f[\phi_5]$

ϕ_5	ψ_5^T
.00000E 00	.35800E 01
.42000E 00	.10070E 01
.47500E 00	.91200E 00
.52000E 00	.85300E 00
.54000E 00	.82600E 00
.55800E 00	.80500E 00
.57000E 00	.78700E 00
.58000E 00	.77500E 00
.59000E 00	.76600E 00
.59500E 00	.76000E 00
.60000E 00	.74500E 00
.61000E 00	.73000E 00
.61500E 00	.71500E 00
.62000E 00	.70000E 00
.62500E 00	.67500E 00
.64000E 00	.51000E 00
.66250E 00	.79000E -01
.68500E 00	-.40200E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F113: $\psi_6^P = f[\phi_6]$

ϕ_6	ψ_6^P
.00000E 00	.61600E 00
.50000E 00	.61600E 00
.52000E 00	.61200E 00
.53500E 00	.60500E 00
.55000E 00	.58500E 00
.57000E 00	.55000E 00
.58000E 00	.52500E 00
.60000E 00	.47800E 00
.61000E 00	.45000E 00
.62500E 00	.40000E 00
.67500E 00	.20000E 00
.72500E 00	.00000E 00
.77500E 00	-.20000E 00

FUNCTION F114: $\psi_6^T = f[\phi_6]$

ϕ_6	ψ_6^T
.00000E 00	.82000E 00
.50000E 00	.70500E 00
.52000E 00	.69500E 00
.53500E 00	.68300E 00
.55000E 00	.66600E 00
.57000E 00	.63200E 00
.58000E 00	.61200E 00
.60000E 00	.56200E 00
.61000E 00	.53000E 00
.62500E 00	.48200E 00
.67500E 00	.29700E 00
.72500E 00	.75000E -01
.77500E 00	-.14700E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F115: $\psi_7^P = f[\phi_7]$

ϕ_7	ψ_7^P
•00000E 00	•48600E 00
•47500E 00	•48600E 00
•48500E 00	•48600E 00
•50000E 00	•48400E 00
•51500E 00	•48000E 00
•52500E 00	•46500E 00
•55000E 00	•41500E 00
•56500E 00	•37500E 00
•57500E 00	•34500E 00
•59000E 00	•29500E 00
•59500E 00	•27500E 00
•60000E 00	•25500E 00
•62500E 00	•15500E 00
•66000E 00	•00000E 00
•69500E 00	•15500E 00

FUNCTION F116: $\psi_7^T = f[\phi_7]$

ϕ_7	ψ_7^T
•00000E 00	•56200E 00
•30000E 00	•56000E 00
•51500E 00	•55200E 00
•52500E 00	•53700E 00
•55000E 00	•48700E 00
•56500E 00	•44300E 00
•57500E 00	•41200E 00
•59000E 00	•36200E 00
•59500E 00	•34200E 00
•60000E 00	•32300E 00
•62500E 00	•22000E 00
•66000E 00	•00000E 00
•69500E 00	•25200E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F117: $\psi_8^P = f[\phi_8]$

ϕ_8	ψ_8^P
•45000E 00	•40000E 00
•46000E 00	•48400E 00
•46500E 00	•48200E 00
•47500E 00	•47800E 00
•49000E 00	•46000E 00
•50000E 00	•43500E 00
•51000E 00	•39400E 00
•52500E 00	•33000E 00
•54000E 00	•26500E 00
•55000E 00	•22500E 00
•56500E 00	•16300E 00
•57500E 00	•12200E 00
•60000E 00	•25000E-01
•60550E 00	•00000E 00
•66100E 00	•22500E 00

FUNCTION F118: $\psi_8^T = f[\phi_8]$

ϕ_8	ψ_8^T
•00000E 00	•56000E 00
•45000E 00	•56000E 00
•46000E 00	•56000E 00
•46500E 00	•56000E 00
•47500E 00	•56000E 00
•49000E 00	•56000E 00
•50000E 00	•53500E 00
•51000E 00	•50700E 00
•52500E 00	•46500E 00
•54000E 00	•40000E 00
•55000E 00	•32000E 00
•56500E 00	•27000E 00
•57500E 00	•20500E 00
•60000E 00	•16000E 00
•60550E 00	•45000E-01
•66100E 00	•00000E 00
	•45200E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F119: $\eta_B = f [PB(TB-TCD)]$

<u>PB(TB-TCD)</u>	<u>η_B</u>
.00000E 00	.79450E 00
.20000E 04	.88000E 00
.15000E 05	.93100E 00
.19250E 05	.95500E 00
.24000E 05	.97100E 00
.30000E 05	.98100E 00
.36500E 05	.98700E 00
.47500E 05	.99000E 00
.55000E 05	.99000E 00
.72500E 05	.98620E 00
.95000E 05	.98320E 00
.12500E 06	.98100E 00
.14000E 06	.98050E 00
.16000E 06	.98000E 00

FUNCTION F120: $KWB = f [N/N_{MAX}]$

<u>N/N_{MAX}</u>	<u>KWB</u>
.60000E 00	.72600E-03
.70000E 00	.70700E-03
.80000E 00	.69800E-03
.85000E 00	.69000E-03
.90000E 00	.69600E-03
.97000E 00	.69600E-03
.10000E 01	.73800E-03

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F1: BV08 = f [N/M_{MAX}, T₀]

T_0	N/M _{MAX}	.4800E 03	.5250E 03	.5030E 03	.5185E 03	.8300E 03
.0000E 00	.1000E 01	.1000E 01	.1000E 01	.1000E 01	.1000E 01	.1000E 01
.8000E 00	.1000E 01	.1000E 01	.1000E 01	.1000E 01	.1000E 01	.1000E 01
.8200E 00	.8350E 00	.8350E 00	.8350E 00	.8350E 00	.8350E 00	.8350E 00
.8400E 00	.7050E 00	.6970E 00	.6900E 00	.6840E 00	.6780E 00	.6780E 00
.8600E 00	.5920E 00	.5800E 00	.5680E 00	.5550E 00	.5350E 00	.5350E 00
.8800E 00	.5200E 00	.5000E 00	.4700E 00	.4350E 00	.3950E 00	.3950E 00
.9000E 00	.4800E 00	.4320E 00	.3900E 00	.3250E 00	.2650E 00	.2650E 00
.9200E 00	.4150E 00	.3650E 00	.3000E 00	.2200E 00	.1300E 00	.1300E 00
.9400E 00	.3450E 00	.2750E 00	.1900E 00	.1050E 00	.0000E 00	.0000E 00
.9550E 00	.2750E 00	.1900E 00	.1000E 00	.0000E 00	.0000E 00	.0000E 00
.9700E 00	.2000E 00	.1300E 00	.0000E 00	.0000E 00	.0000E 00	.0000E 00
.9850E 00	.1000E 00	.0000E 00	.0000E 00	.0000E 00	.0000E 00	.0000E 00
.1000E 01	.0000E 00	.0000E 00	.0000E 00	.0000E 00	.0000E 00	.0000E 00

Table A-3. Engine Component Characteristics (Continued)

FUNCTION F2: $\psi_2^P = f(\psi_2, IGV)$

ψ_2	IGV	.0000E 00	.5000E 00	.7500E 00	.1000E 01
.0000E 00		.4200E 00	.3000E 00	.2800E 00	.2600E 00
.4500E 00		.8450E 00	.8640E 00	.8300E 00	.7920E 00
.4750E 00		.8700E 00	.8920E 00	.8610E 00	.8220E 00
.5000E 00		.8930E 00	.9100E 00	.8820E 00	.8430E 00
.5250E 00		.9190E 00	.9200E 00	.8890E 00	.8500E 00
.5500E 00		.9250E 00	.9150E 00	.8850E 00	.8490E 00
.5750E 00		.9300E 00	.9030E 00	.8700E 00	.8400E 00
.6000E 00		.9300E 00	.8750E 00	.8480E 00	.8130E 00
.6250E 00		.9200E 00	.8400E 00	.8200E 00	.7820E 00
.6500E 00		.9050E 00	.7950E 00	.7870E 00	.7470E 00
.6750E 00		.8830E 00	.7460E 00	.7460E 00	.7050E 00
.7000E 00		.8590E 00	.6750E 00	.6750E 00	.6560E 00
.7250E 00		.8270E 00	.6150E 00	.6150E 00	.6000E 00
.7500E 00		.7700E 00	.5100E 00	.5100E 00	.5100E 00
.7750E 00		.6150E 00	.3620E 00	.3620E 00	.3620E 00
.7900E 00		.4000E 00	.2650E 00	.2650E 00	.2650E 00
.8350E 00		.2500E-01	.2500E-01	.2500E-01	.2500E-01

Table A-3. Engine Components

FUNCTION F3: $\frac{WT-TB}{N \cdot PB} = f \left[\frac{PT}{PB}, \sqrt{\frac{N}{TB}} \right]$

PT/PB	$\frac{N}{\sqrt{TB}}$						
	.10000E 03	.15000E 03	.20000E 03	.24000E 03	.26000E 03	.28000E 03	.30000E 03
.00000E 00	.22480E 00	.14470E 00	.11120E 00	.92500E-01	.83300E-01	.77300E-01	.72000E-01
.10000E 00	.22480E 00	.14470E 00	.11120E 00	.92500E-01	.83300E-01	.77300E-01	.72000E-01
.20000E 00	.22480E 00	.14470E 00	.11100E 00	.92500E-01	.83300E-01	.77300E-01	.72000E-01
.30000E 00	.22390E 00	.14400E 00	.11000E 00	.91500E-01	.83300E-01	.77300E-01	.72000E-01
.35000E 00	.22380E 00	.14300E 00	.10910E 00	.90600E-01	.83200E-01	.77200E-01	.71900E-01
.40000E 00	.22270E 00	.14200E 00	.10830E 00	.89800E-01	.82800E-01	.76900E-01	.71600E-01
.45000E 00	.22110E 00	.14010E 00	.10750E 00	.89100E-01	.81900E-01	.76000E-01	.70800E-01
.50000E 00	.21920E 00	.13900E 00	.10630E 00	.88500E-01	.80800E-01	.74900E-01	.69700E-01
.55000E 00	.21690E 00	.13730E 00	.10470E 00	.88000E-01	.79500E-01	.73500E-01	.68300E-01
.60000E 00	.21430E 00	.13560E 00	.10240E 00	.87600E-01	.77100E-01	.71300E-01	.66400E-01
.70000E 00	.20410E 00	.12470E 00	.93000E-01	.74900E-01	.68600E-01	.62900E-01	.58000E-01
.80000E 00	.16310E 00	.99000E-01	.68700E-01	.65000E-01	.60500E-01	.44000E-01	.41000E-01
.90000E 00	.88700E-01	.53200E-01	.37000E-01	.28500E-01	.26500E-01	.22400E-01	.21300E-01
.10000E 01	.40000E 00	.30000E 00	.20000E 00	.10000E 00	.10000E 00	.10000E 00	.10000E 00

FUNCTION F4: $\frac{\Delta HT}{N \sqrt{TB}} = f \left[\frac{PT}{PB}, \sqrt{\frac{N}{TB}} \right]$

PT/PB	$\frac{N}{\sqrt{TB}}$						
	.10000E 03	.15000E 03	.20000E 03	.24000E 03	.26000E 03	.28000E 03	.30000E 03
.00000E 00	.26000E 00	.24000E 00	.23400E 00	.22400E 00	.21800E 00	.21100E 00	.20500E 00
.10000E 00	.26000E 00	.24000E 00	.23600E 00	.22400E 00	.21800E 00	.21100E 00	.20500E 00
.20000E 00	.26000E 00	.24000E 00	.23600E 00	.22400E 00	.21800E 00	.21100E 00	.20500E 00
.30000E 00	.26000E 00	.24000E 00	.22900E 00	.22900E 00	.20300E 00	.20000E 00	.19600E 00
.35000E 00	.26000E 00	.24000E 00	.20700E 00	.19700E 00	.18500E 00	.18000E 00	.17500E 00
.40000E 00	.26000E 00	.23900E 00	.18800E 00	.17100E 00	.16600E 00	.16100E 00	.15600E 00
.45000E 00	.26000E 00	.20900E 00	.17100E 00	.15400E 00	.14800E 00	.14200E 00	.13700E 00
.50000E 00	.25000E 00	.18400E 00	.15400E 00	.13800E 00	.13100E 00	.12600E 00	.12100E 00
.55000E 00	.22300E 00	.16700E 00	.13700E 00	.12200E 00	.11500E 00	.11000E 00	.10500E 00
.60000E 00	.19700E 00	.14700E 00	.12000E 00	.10600E 00	.10000E 00	.94900E-01	.90100E-01
.70000E 00	.14700E 00	.10900E 00	.87700E-01	.76200E-01	.70800E-01	.66000E-01	.61500E-01
.80000E 00	.96400E-01	.70100E-01	.56100E-01	.47700E-01	.44100E-01	.40600E-01	.37500E-01
.90000E 00	.47600E-01	.33400E-01	.25000E-01	.21700E-01	.19400E-01	.17500E-01	.16300E-01
.10000E 01	.00000E 00	.00000E 00	.00000E 00	.00000E 00	.00000E 00	.00000E 00	.00000E 00

a

Component Characteristics (Continued)

$$\frac{WT-TB}{N-PB} = f \left[\frac{PT}{PB}, \frac{N}{\sqrt{TB}} \right]$$

.3000E-03	.3200E-03	.3400E-03	.3600E-03	.3800E-03	.4000E-03	.4200E-03	.4400E-03
.7200E-01	.6740E-01	.6330E-01	.5970E-01	.5640E-01	.5350E-01	.5080E-01	.4840E-01
.7200E-01	.6740E-01	.6330E-01	.5970E-01	.5640E-01	.5350E-01	.5080E-01	.4840E-01
.7200E-01	.6740E-01	.6330E-01	.5970E-01	.5640E-01	.5350E-01	.5080E-01	.4840E-01
.7190E-01	.6730E-01	.6320E-01	.5960E-01	.5630E-01	.5340E-01	.5070E-01	.4830E-01
.7150E-01	.6700E-01	.6300E-01	.5930E-01	.5600E-01	.5310E-01	.5050E-01	.4810E-01
.7080E-01	.6630E-01	.6230E-01	.5870E-01	.5540E-01	.5240E-01	.4970E-01	.4730E-01
.6970E-01	.6510E-01	.6120E-01	.5760E-01	.5450E-01	.5160E-01	.4900E-01	.4650E-01
.6830E-01	.6380E-01	.6000E-01	.5640E-01	.5350E-01	.5070E-01	.4790E-01	.4520E-01
.6640E-01	.6220E-01	.5850E-01	.5520E-01	.5200E-01	.4890E-01	.4600E-01	.4340E-01
.6400E-01	.5970E-01	.5610E-01	.5280E-01	.4980E-01	.4680E-01	.4400E-01	.4140E-01
.6100E-01	.5670E-01	.5320E-01	.5000E-01	.4700E-01	.4400E-01	.4120E-01	.3860E-01
.5700E-01	.5270E-01	.4920E-01	.4600E-01	.4300E-01	.4000E-01	.3720E-01	.3460E-01
.5200E-01	.4770E-01	.4420E-01	.4100E-01	.3800E-01	.3500E-01	.3220E-01	.2960E-01
.4600E-01	.4170E-01	.3820E-01	.3500E-01	.3200E-01	.2900E-01	.2620E-01	.2360E-01
.3900E-01	.3470E-01	.3120E-01	.2800E-01	.2500E-01	.2200E-01	.1920E-01	.1660E-01
.3100E-01	.2670E-01	.2320E-01	.2000E-01	.1700E-01	.1400E-01	.1120E-01	.8600E-02

$$\frac{\Delta HT}{N\sqrt{TB}} = f \left[\frac{PT}{PB}, \frac{N}{\sqrt{TB}} \right]$$

.3000E-03	.3200E-03	.3400E-03	.3600E-03	.3800E-03	.4000E-03	.4200E-03	.4400E-03
.2050E-00	.1980E-00	.1900E-00	.1830E-00	.1760E-00	.1680E-00	.1610E-00	.1530E-00
.2050E-00	.1980E-00	.1900E-00	.1830E-00	.1760E-00	.1680E-00	.1610E-00	.1530E-00
.2050E-00	.1980E-00	.1900E-00	.1830E-00	.1760E-00	.1680E-00	.1610E-00	.1530E-00
.1960E-00	.1910E-00	.1840E-00	.1750E-00	.1670E-00	.1590E-00	.1530E-00	.1460E-00
.1750E-00	.1700E-00	.1640E-00	.1560E-00	.1490E-00	.1420E-00	.1360E-00	.1290E-00
.1560E-00	.1510E-00	.1450E-00	.1380E-00	.1320E-00	.1260E-00	.1200E-00	.1130E-00
.1370E-00	.1330E-00	.1270E-00	.1210E-00	.1160E-00	.1100E-00	.1030E-00	.9550E-01
.1210E-00	.1160E-00	.1110E-00	.1050E-00	.9950E-01	.9300E-01	.8570E-01	.7820E-01
.1050E-00	.1000E-00	.9530E-01	.9220E-01	.8410E-01	.7680E-01	.6880E-01	.6260E-01
.9010E-01	.8560E-01	.8100E-01	.7600E-01	.6940E-01	.6180E-01	.5460E-01	.4980E-01
.6150E-01	.5700E-01	.5550E-01	.51300E-01	.4380E-01	.3920E-01	.3510E-01	.3180E-01
.3760E-01	.3470E-01	.3200E-01	.2960E-01	.2520E-01	.2170E-01	.1830E-01	.1630E-01
.1630E-01	.1470E-01	.1320E-01	.1200E-01	.1110E-01	.8200E-02	.6000E-02	.4800E-02
.0050E-00	.0000E-00	.0000E-00	.0000E-00	.0000E-00	.0000E-00	.0000E-00	.0000E-00

b

Table A-3. Engine Component Characteristics (Concluded)

$$\text{FUNCTION F5: } \psi_2^T = f(\psi_2, \text{IGV})$$

ψ_2	ICV	.0000E 00	.5000E 00	.7500E 00	.1000E 01
.0000E 00		.6000E 01	.6000E 01	.5600E 01	.5200E 01
.4500E 00		.1152E 01	.1177E 01	.1132E 01	.1080E 01
.4750E 00		.1125E 01	.1155E 01	.1113E 01	.1063E 01
.5000E 00		.1100E 01	.1130E 01	.1095E 01	.1047E 01
.5250E 00		.1095E 01	.1110E 01	.1072E 01	.1026E 01
.5500E 00		.1085E 01	.1073E 01	.1038E 01	.9950E 00
.5750E 00		.1072E 01	.1039E 01	.1000E 01	.9670E 00
.6000E 00		.1058E 01	.9950E 00	.9450E 00	.9250E 00
.6250E 00		.1038E 01	.9480E 00	.9250E 00	.8820E 00
.6500E 00		.1017E 01	.8930E 00	.8840E 00	.8390E 00
.6750E 00		.9910E 00	.8380E 00	.8380E 00	.7920E 00
.7000E 00		.9650E 00	.7580E 00	.7580E 00	.7370E 00
.7250E 00		.9340E 00	.6950E 00	.6950E 00	.6770E 00
.7500E 00		.8750E 00	.5800E 00	.5800E 00	.5800E 00
.7750E 00		.7250E 00	.4260E 00	.4260E 00	.4260E 00
.7900E 00		.5510E 00	.3655E 00	.3655E 00	.3655E 00
.8050E 00		.4800E 01	.4800E 01	.4800E 01	.4800E 01

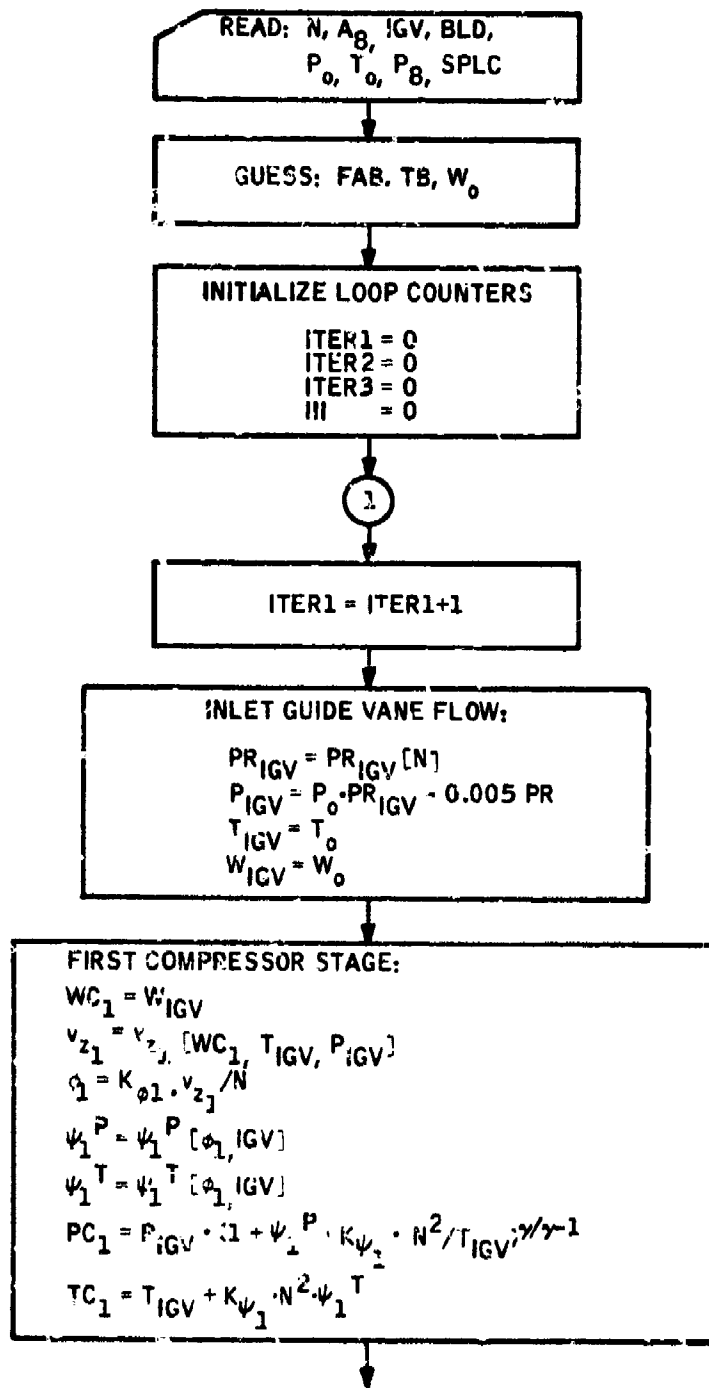


Figure A-1. Trim Routine Flow Chart (Steady-State Trim)

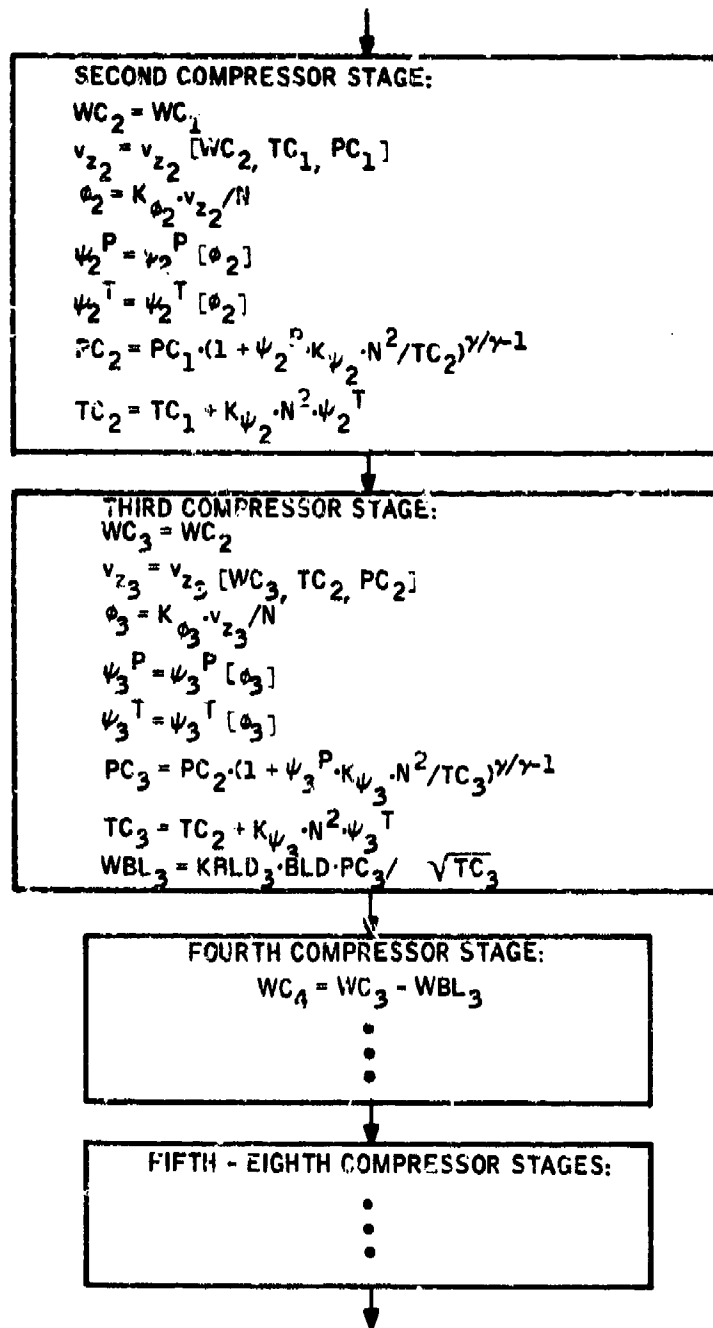


Figure A-1b. Trim Routine Flow Chart (Steady-State Trim)

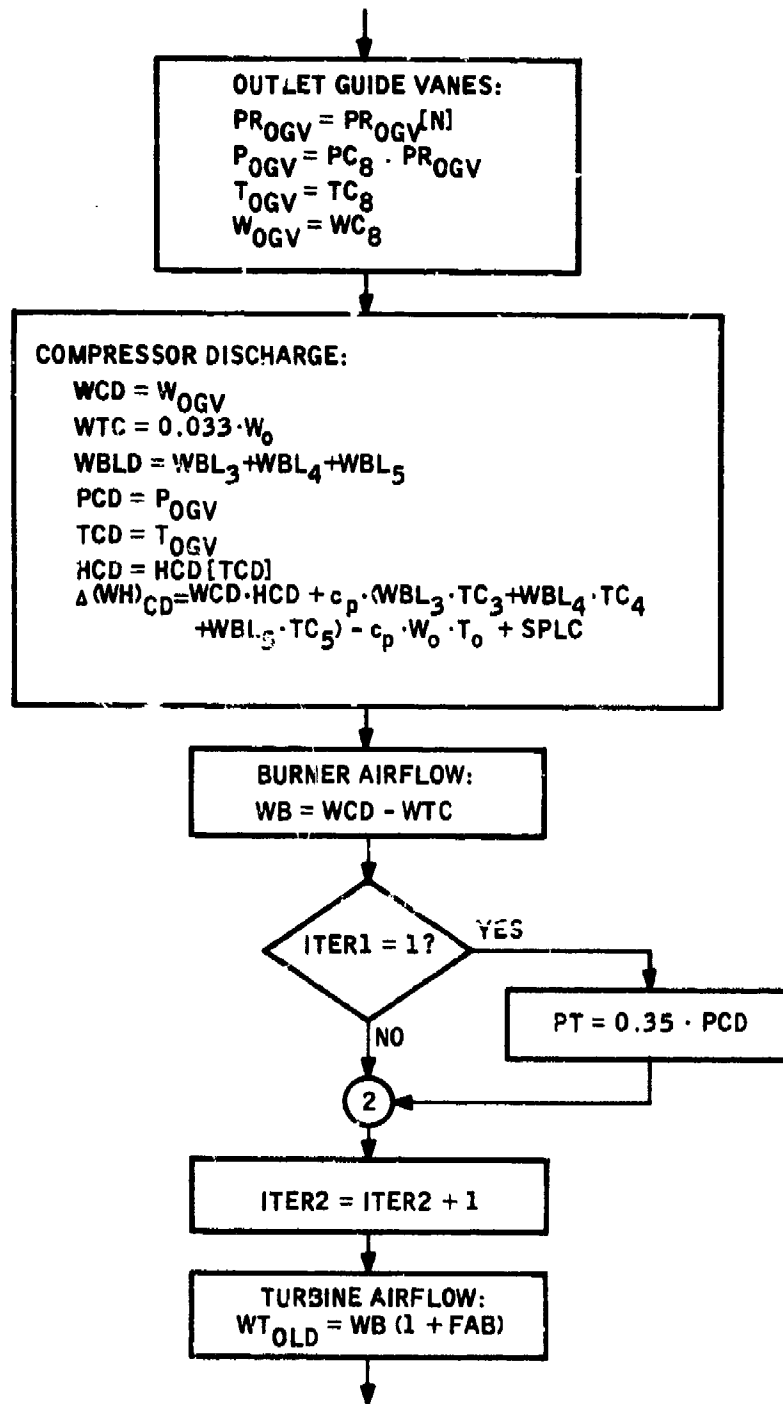


Figure A-1c. Trim Routine Flow Chart (Steady-State Trim)

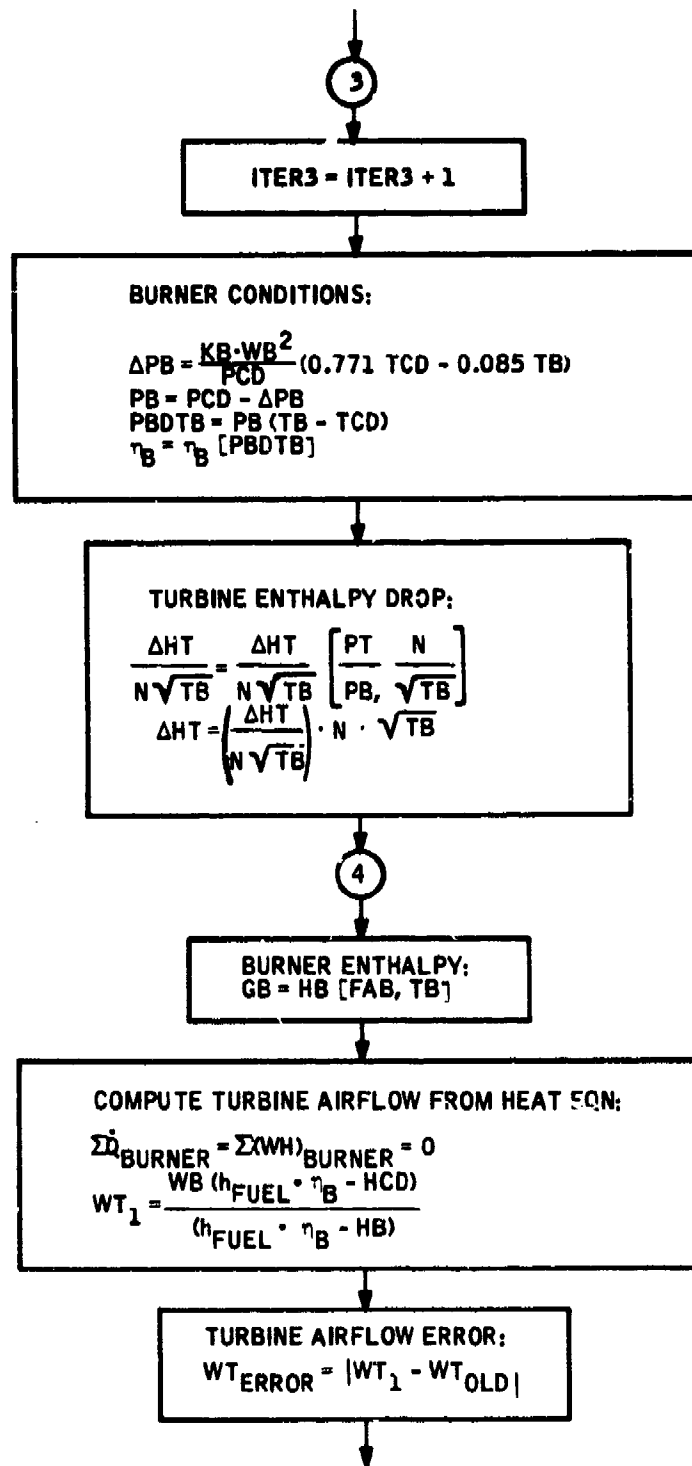


Figure A-1d. Trim Routine Flow Chart (Steady-State Trim)

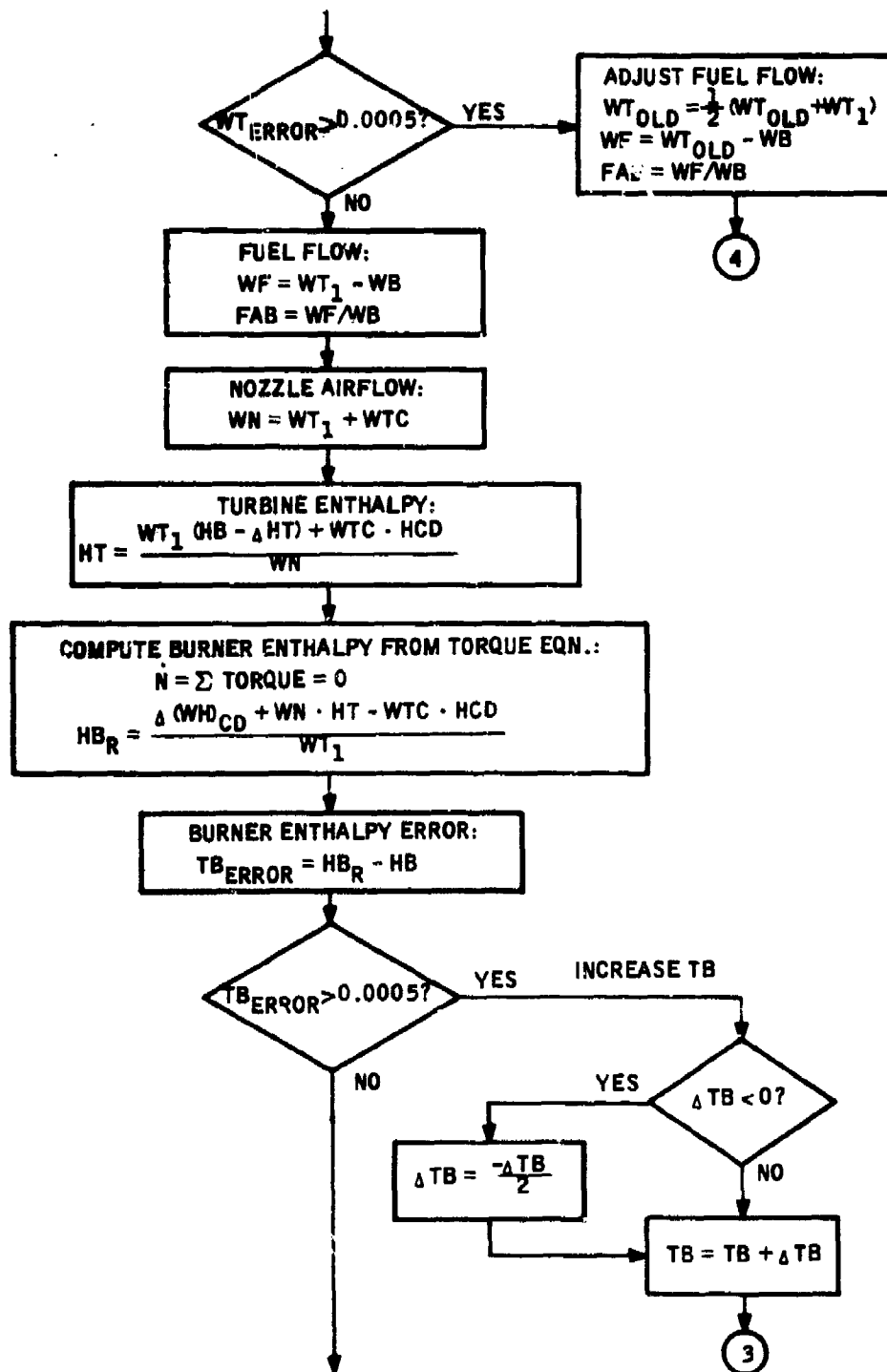


Figure A-1e. Trim Routine Flow Chart (Steady-State Trim)

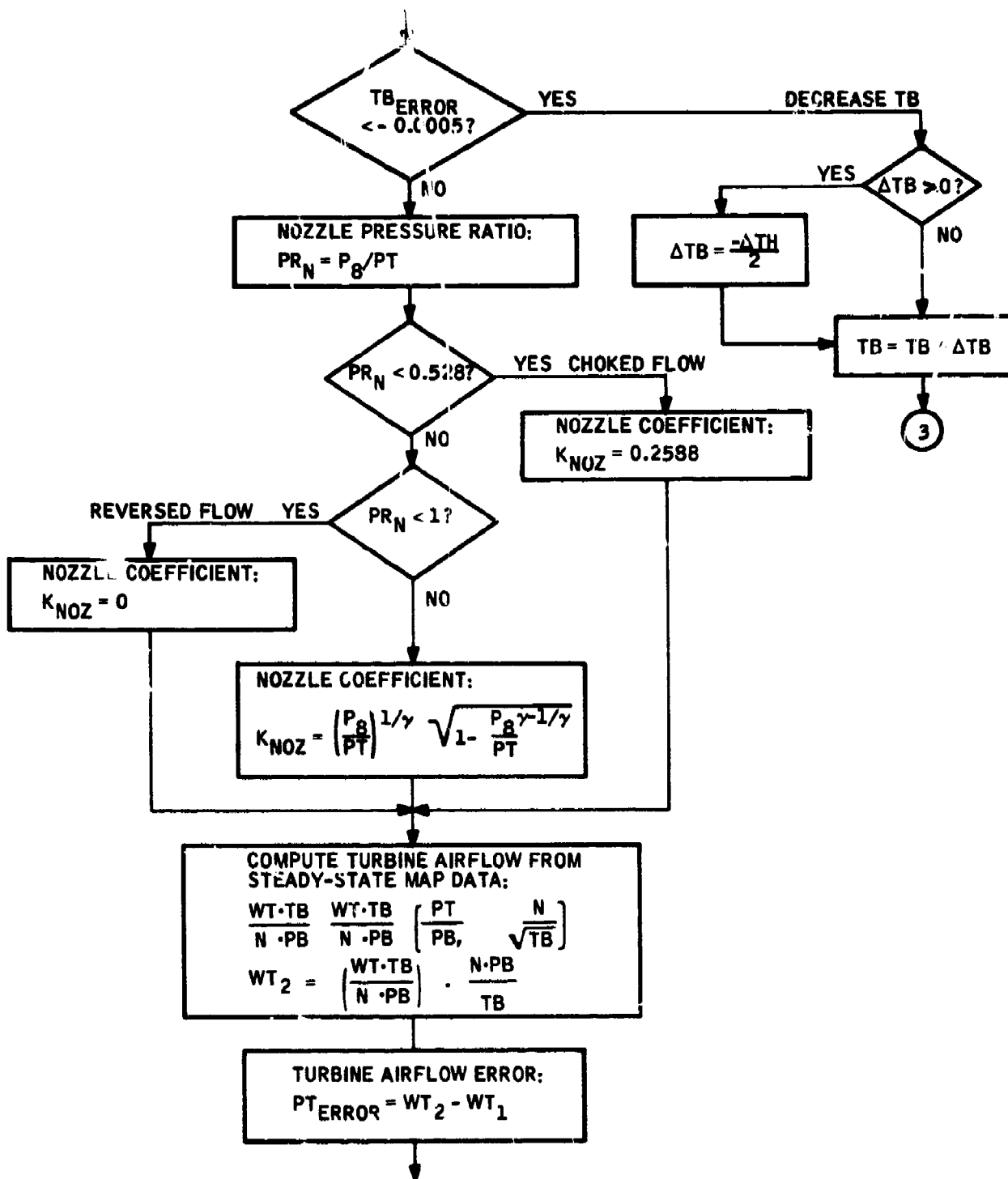


Figure A-1f. Trim Routine Flow Chart (Steady-State Trim)

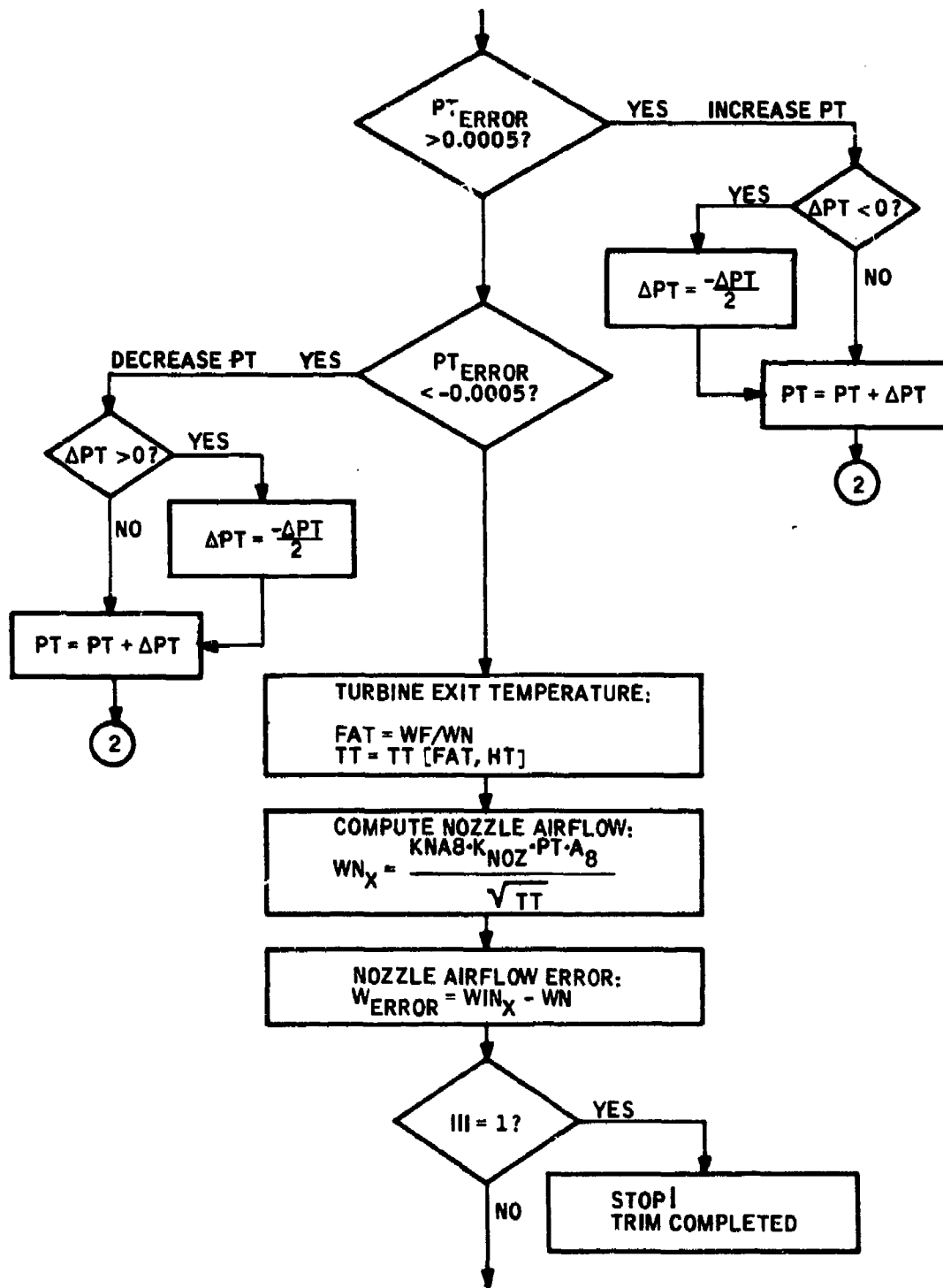


Figure A-1g. Trim Routine Flow Chart (Steady-State Trim)

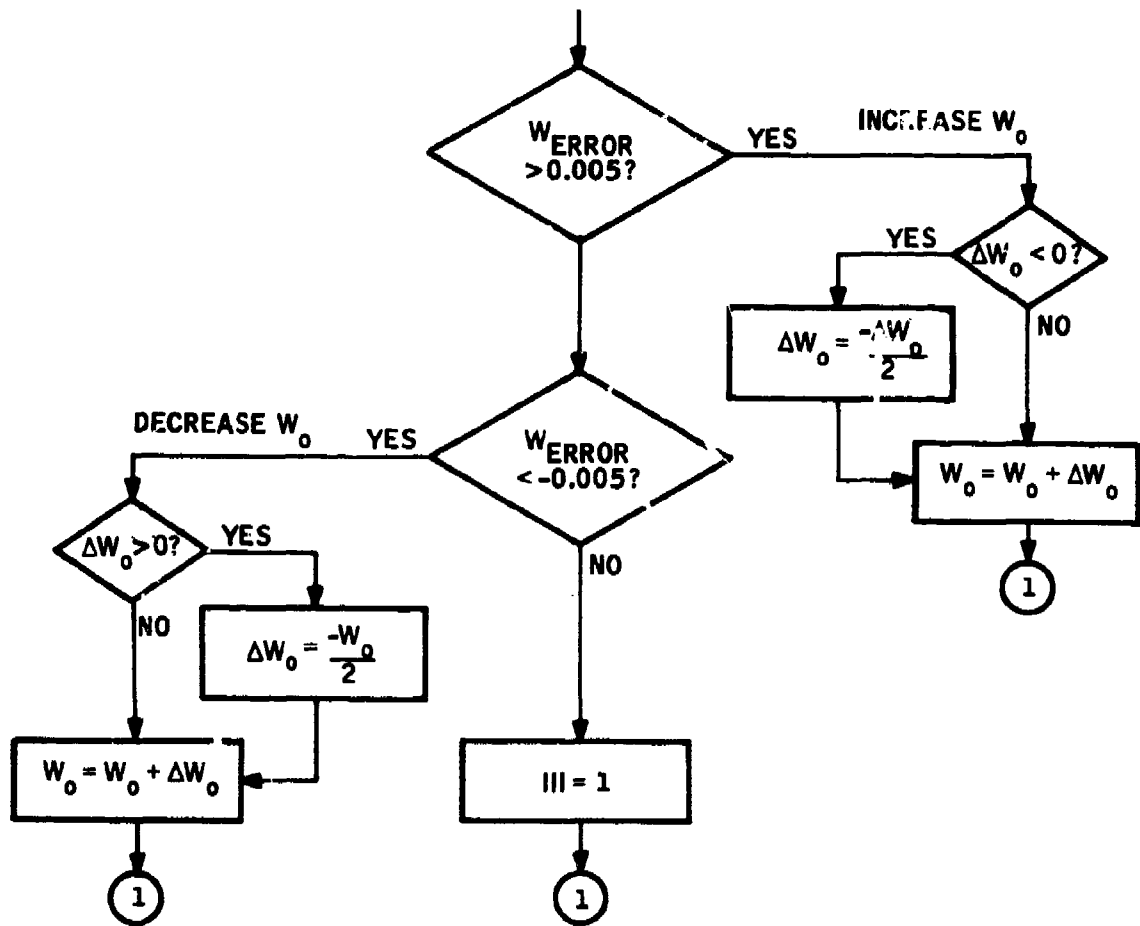


Figure A-1h. Trim Routine Flow Chart (Steady-State Trim)

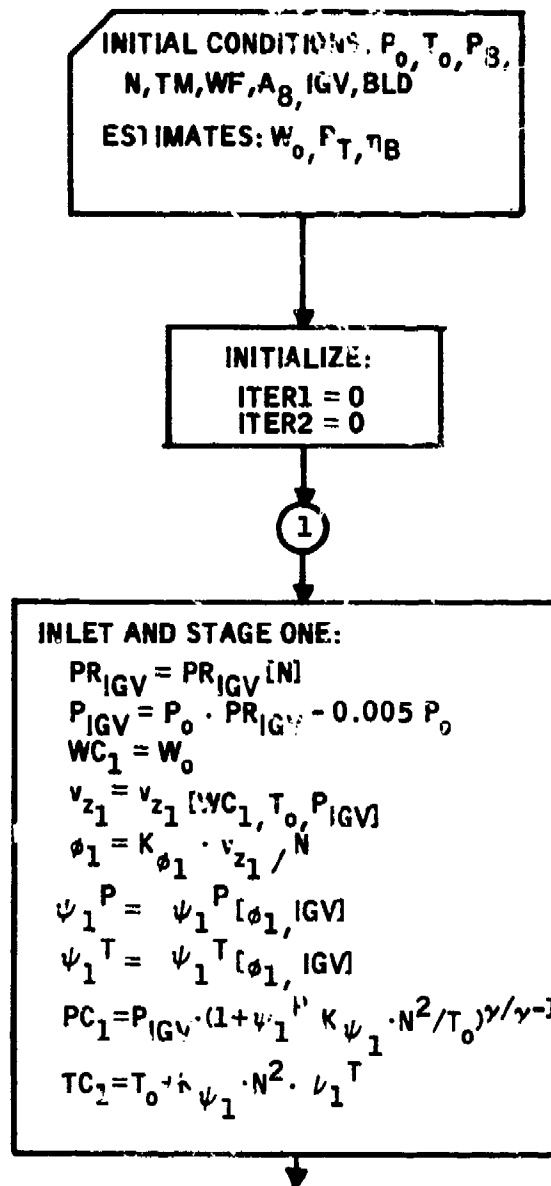


Figure A-2. Subroutine Dynamic Flow Chart

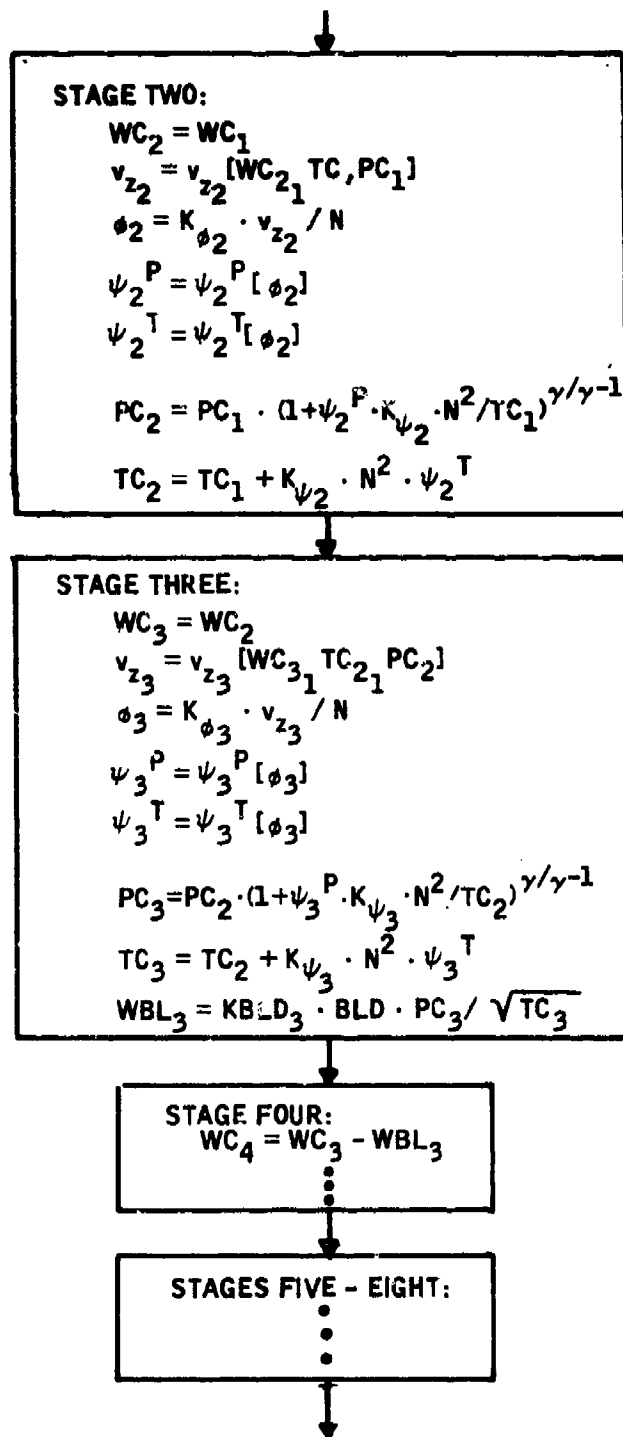


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

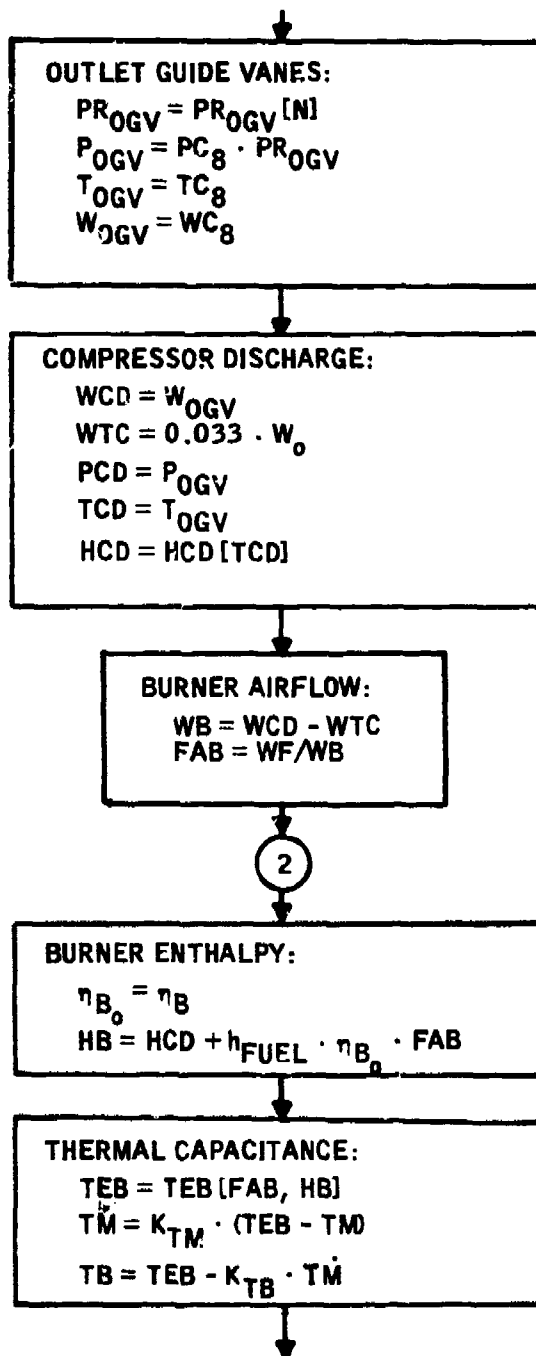


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

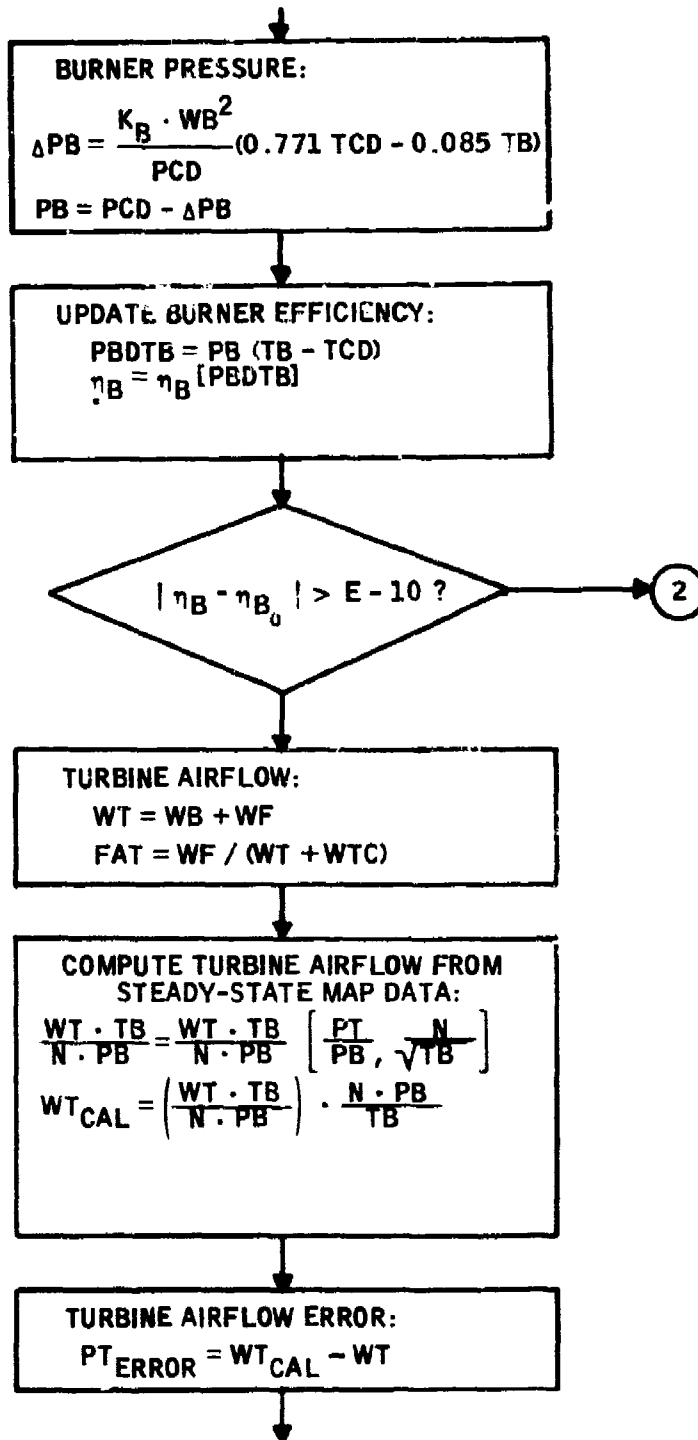


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

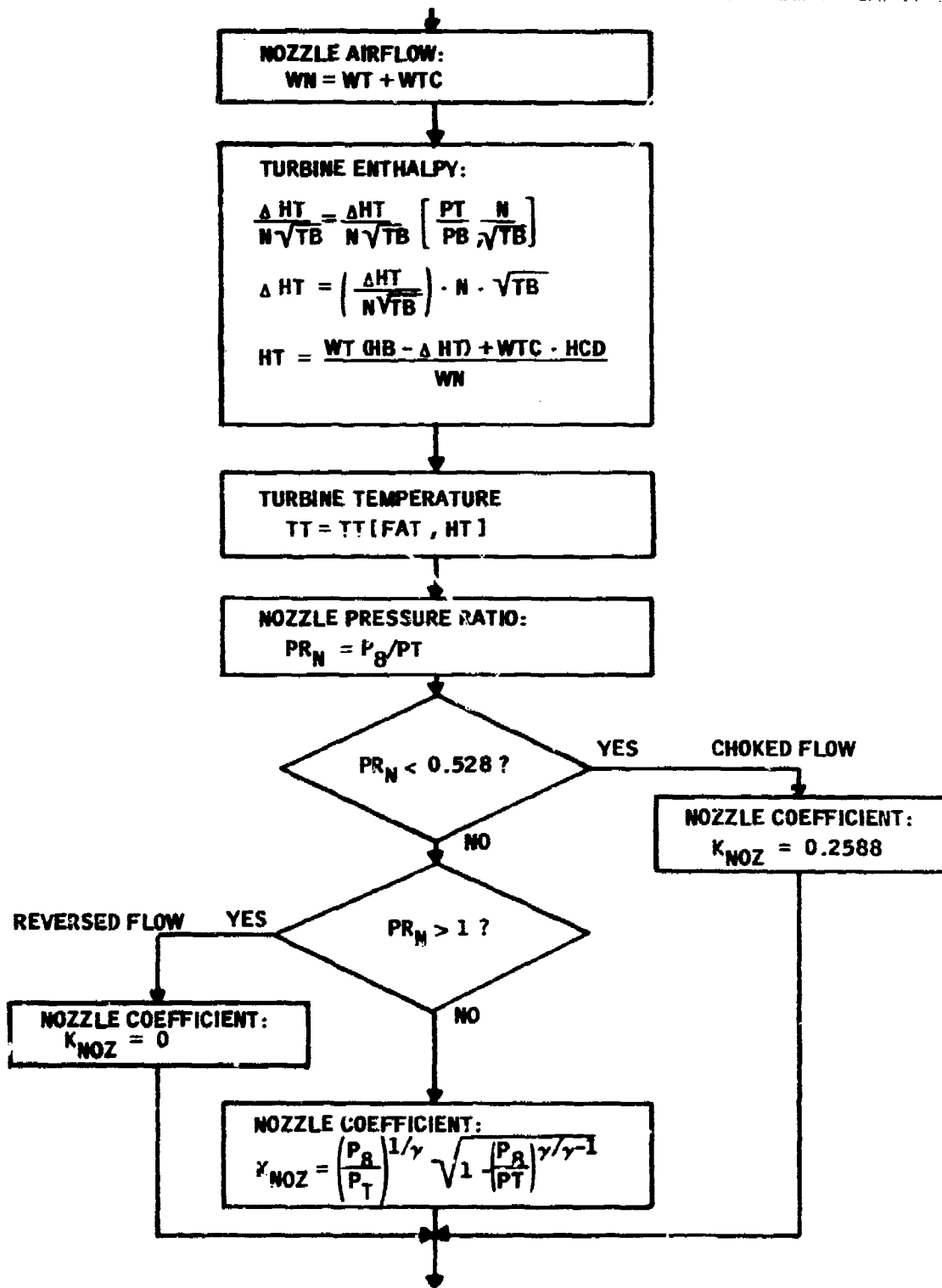


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

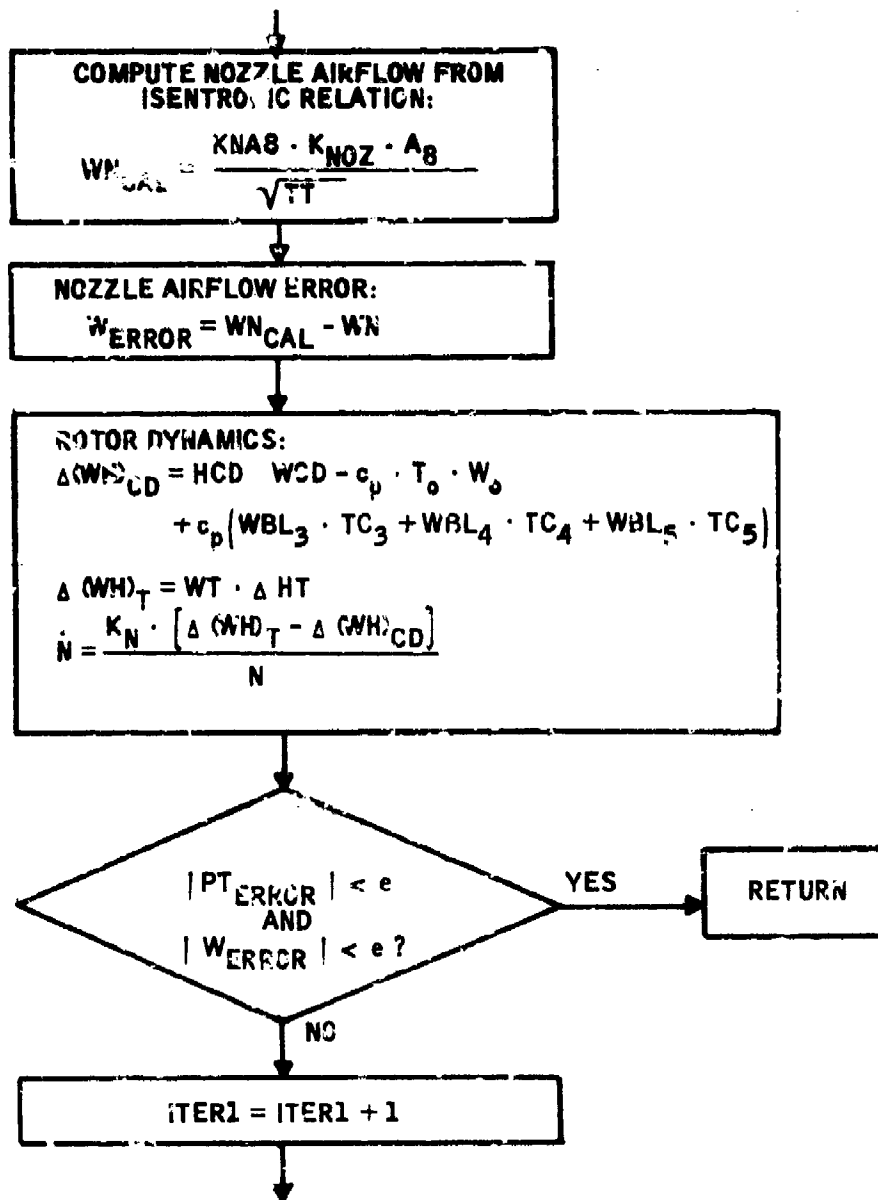


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

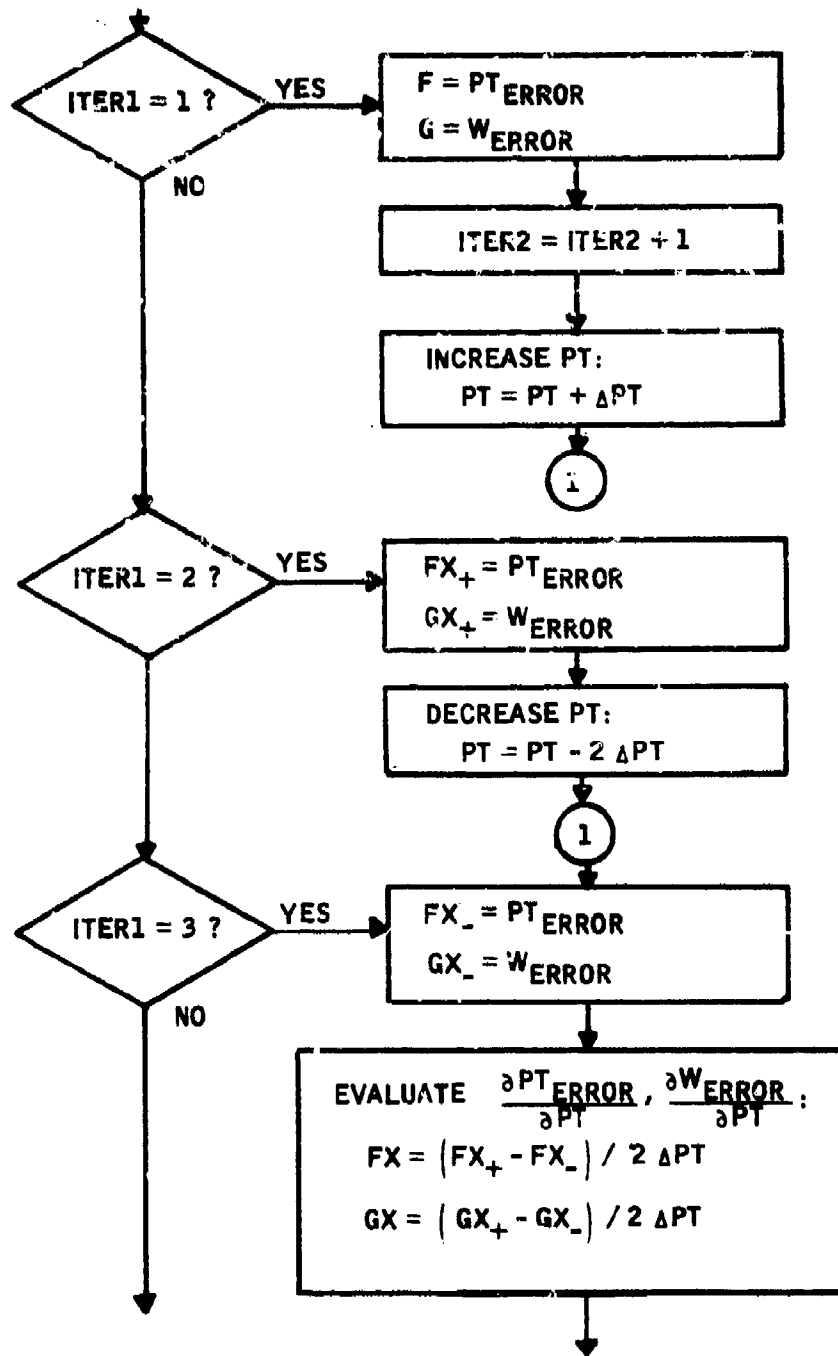


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

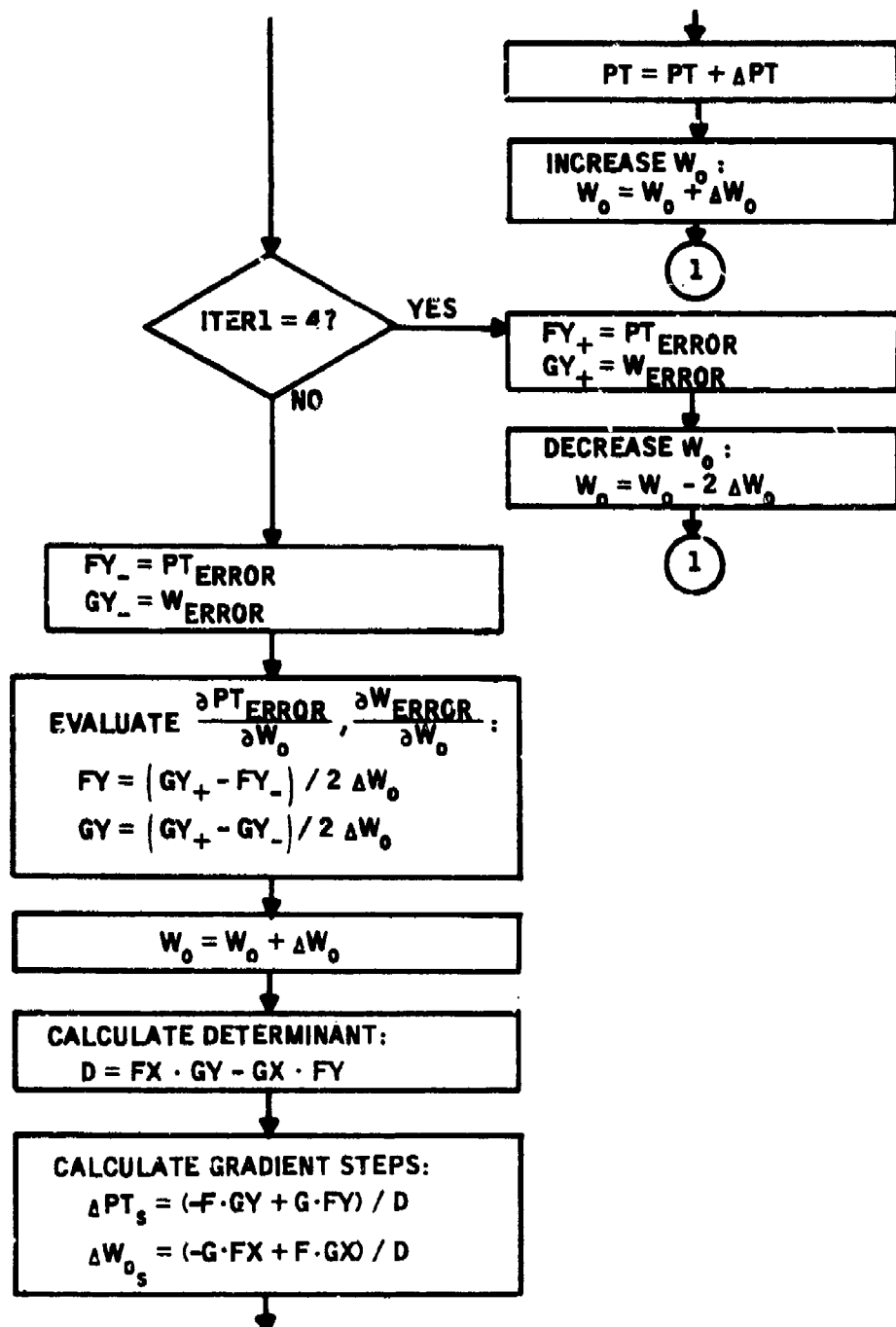


Figure A-2. Subroutine Dynamic Flow Chart (Continued)

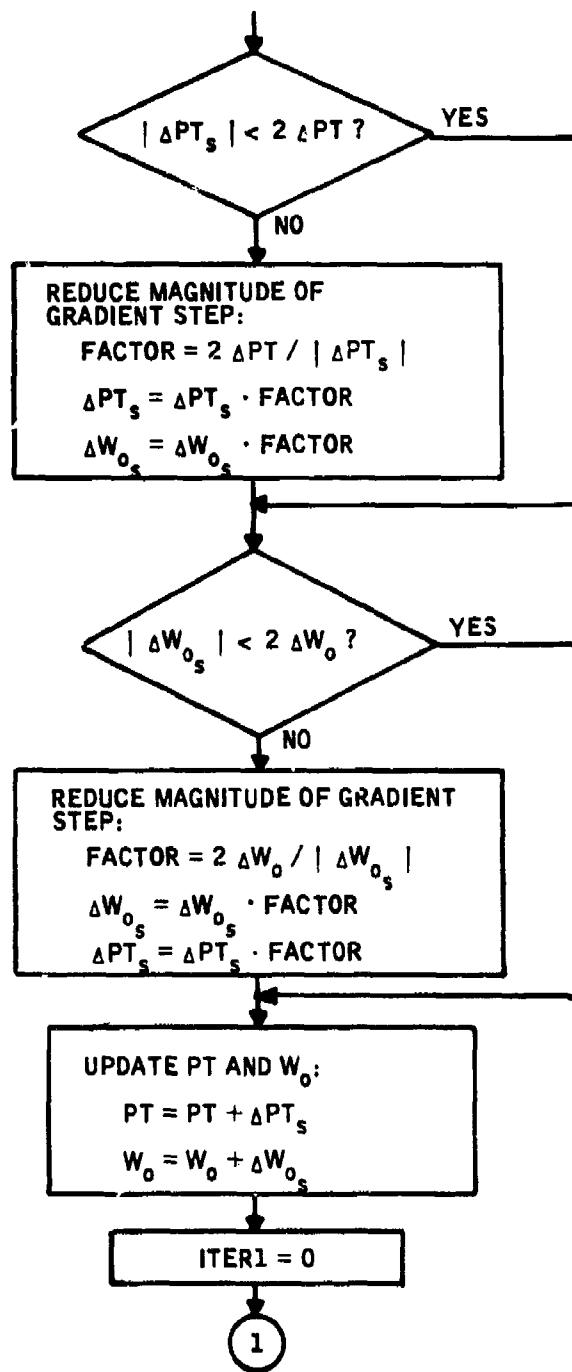


Figure A-2. Subroutine Dynamic Flow Chart (Concluded)

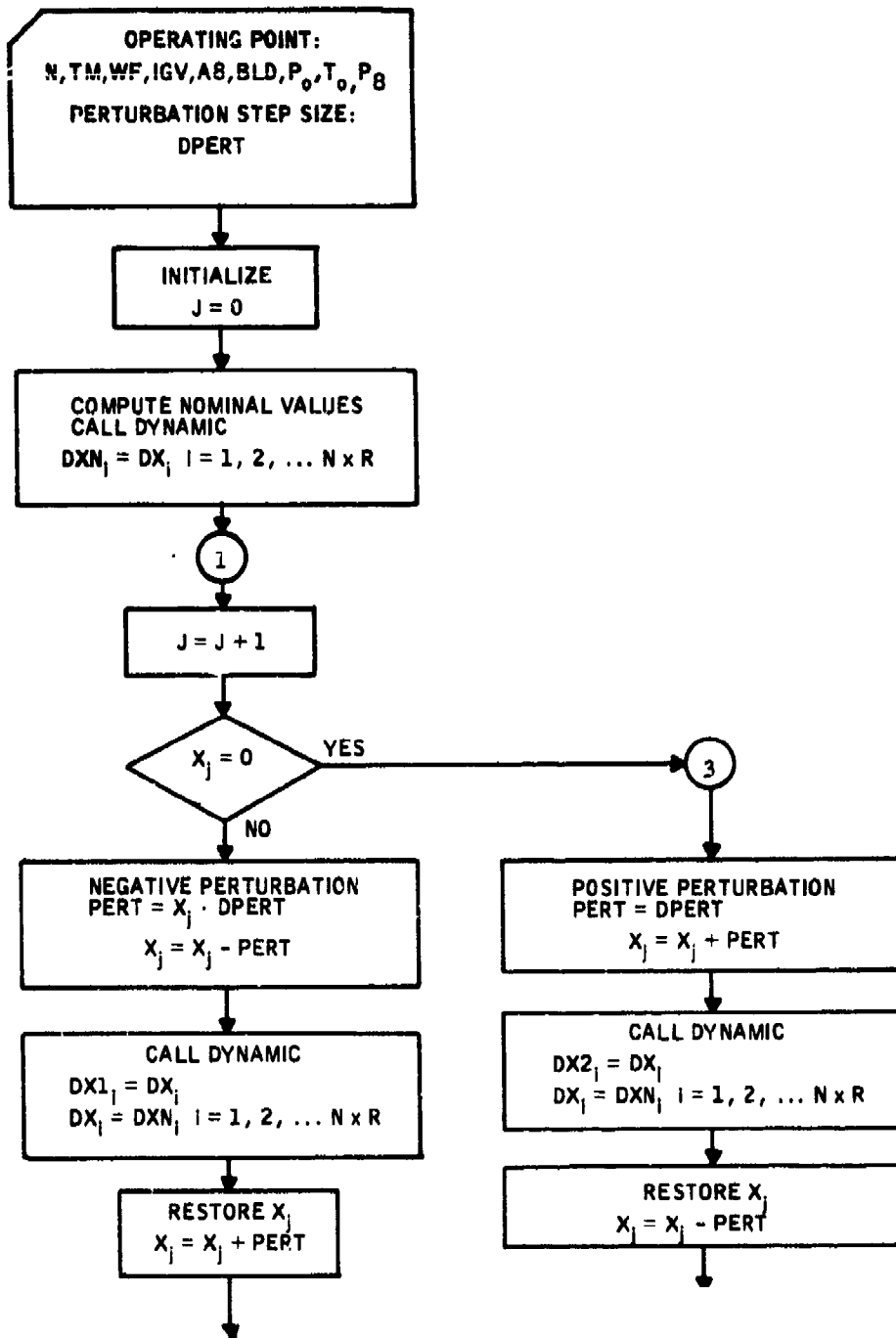


Figure A-3. Linearization Flow Chart

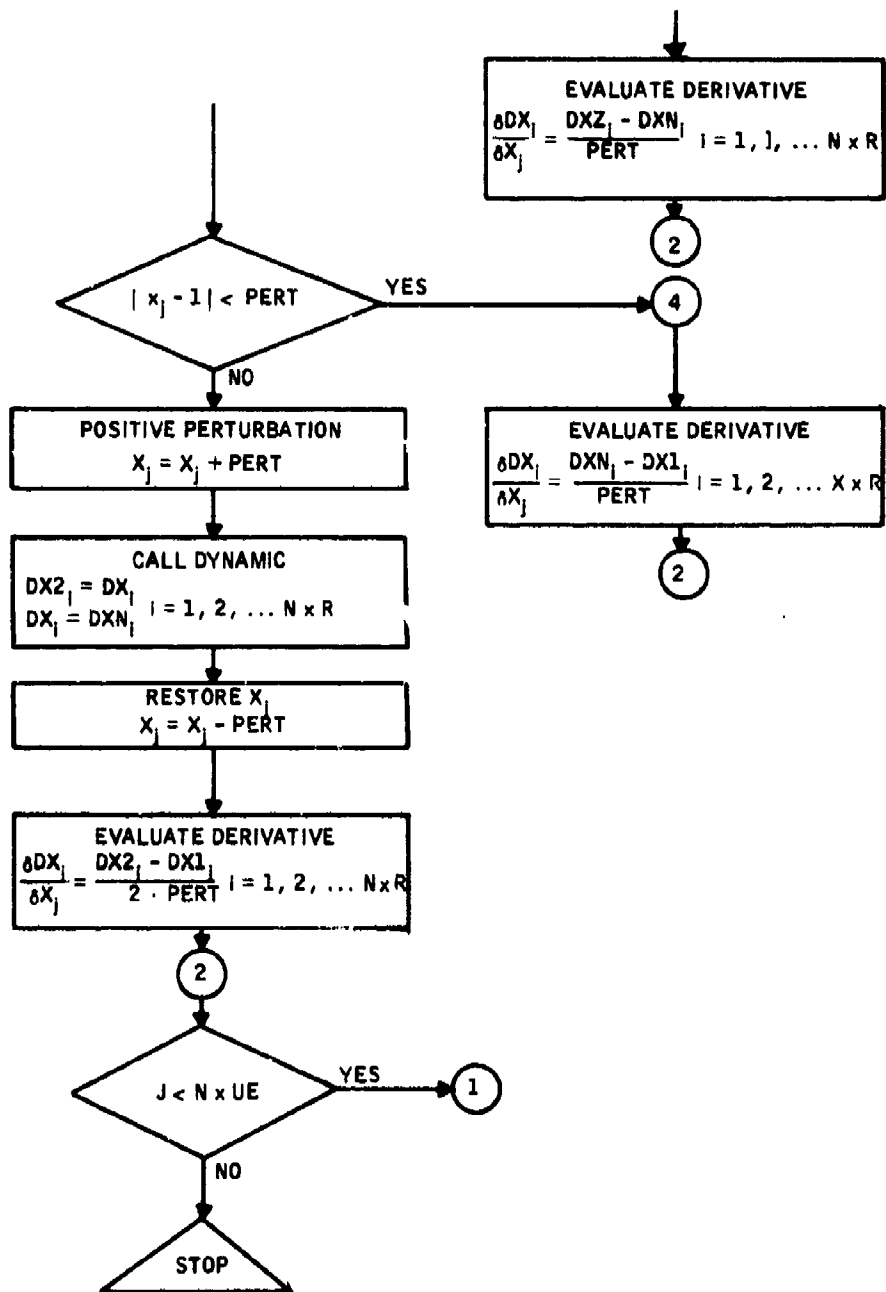


Figure A-3. Linearization Flow Chart (Concluded)

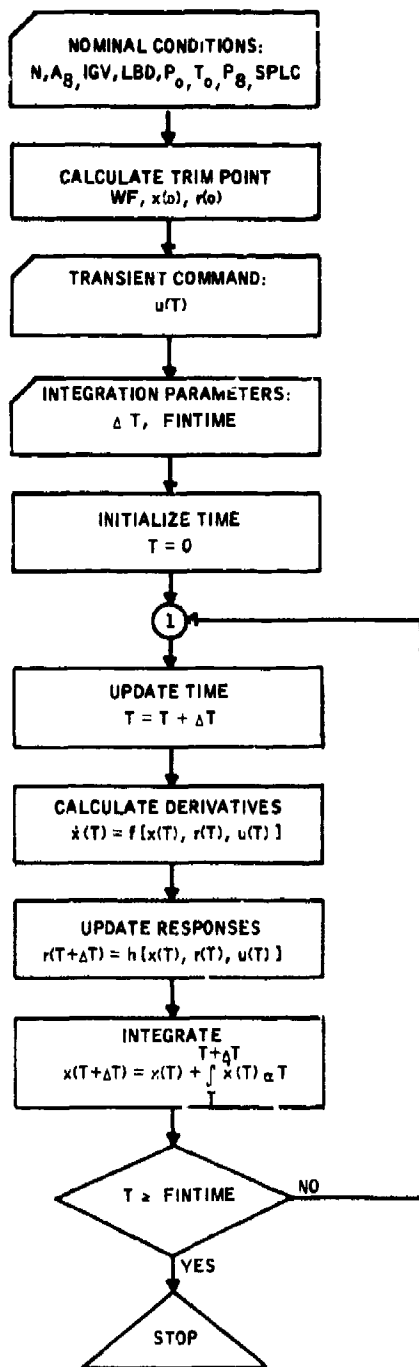


Figure A-4. Nonlinear Engine Simulation Flow Chart

REFERENCES

- A-1. Seldner, Kurt, Mihalow, James R., and Blaha, Ronald J., "Generalized Simulation Technique for Turbojet Engine System Analysis," NASA TN D-6610, Lewis Research Center, National Aeronautics and Space Administration, Cleveland, Ohio, February 1972.
- A-2. Hsu, Jay C. and Meyer, Andrew V., Modern Control Principles and Applications, McGraw-Hill Book Company, New York, 1960.

APPENDIX B
CONTROLLER SOFTWARE FOR THE
APL WIND TUNNEL TEST FACILITY

Software for the optimal command controller synthesized in Section IV (Volume I) is presented. The software is for the IBM 1800 computer at APL. This software inserts the Honeywell optimal controller within the Bendix Bounds program (Reference B-1). The reader is assumed to be familiar with the IBM 1800 (Reference B-2) and with the Bendix Bounds program.

This appendix is divided into three major parts:

- Controller data
- Equilibrium-pressure software
- Equilibrium-temperature software

In the first part of this Appendix, controller data for deceleration-equilibrium-pressure-temperature modes are combined. This system will provide precise speed control and rapid spool speed responses without surge-stall, excessive temperatures, or flamecuts. This is close to a control system that we recommend. The adjective close would be deleted by applying standard correction procedures to permit operation at other than the sea level standard design condition.

For expediency, in testing in the APL wind tunnel, the system was divided into two parts:

- Deceleration-equilibrium-pressure
- Deceleration-equilibrium-temperature

The first part does not explicitly provide over-temperature protection while the second does not explicitly provide surge-stall protection. Protection is obtained, however, by setting the pressure limit low enough to prevent over-temperature and the temperature limit low enough to prevent surge-stall.

CONTROLLER DATA

The inlet guide vanes (IGV), bleed (BLD), and exhaust actuator (A8) are operated on open-loop schedules (for reasons discussed in Section IV). Closed-loop control is used on the fuel valve.

For control synthesis the IGV and BLD were set on the G. E. schedule. As Bendix employs the same schedule in the Bounds program, the Bounds schedule for IGV and BLD are used with the Honeywell controllers.

The A8 schedule is the same as that used on a previous Honeywell contract to APL; it is not the bill of materials schedule.

Table B-1 summarizes the open-loop schedules for IGV, BLD, and A8.

Fuel valve command data are presented in Tables B-2 through B-5. Table B-2 presents the generic form for the complete control law.

For deceleration-equilibrium-pressure control u_1 is deleted from u_2 in Equation (3) of Table B-2. For deceleration-equilibrium-temperature control, u_2 is deleted from u_2 . Feedback gains, open-loop fuel flows, and "equilibrium" data are presented in Tables B-3, B-4, and B-5, respectively.

The equations and data of Tables B-2 through B-5 could have been programmed; a simplification is made before programming. The simplification permits either variable limits (ENL, EPL, or ETL) to be achieved by constants or variable integration parameters to be made constant. For

example, in the $E\dot{P}$ equation (Table B-2) the parameter PC is variable. It can be made constant without changing the resulting control. This is demonstrated by Table B-6. The generic form of the modified state equations and controllers is presented above the dashed line. The integration parameter (d) can be made to take an arbitrary non-zero value by dividing d by μ and by multiplying the integral gain λ by μ ; this is shown by the equations below the dashed line.

EQUILIBRIUM-PRESSURE SOFTWARE

Flow charts are presented in Figures B-1 through B-11. Table B-7 contains a glossary of terms. The program is presented in Table B-8.

The main computational blocks of the speed and pressure control program are shown in Figure B-1. A detailed flow chart for each block is subsequently presented.

Initialization

In this section of the program (Figure B-2), all of the gains and open-loop information (i. e., fuels and pressures as a function of speed) are transferred from variable-trim locations. (The variable-trim locations are the sole means of communication between the Honeywell control program and the Bendix Bounds program to the proper locations in the control program.) The labels associated with the variable-trim locations have the prefix VT followed by three digits. There are 254 VT locations. The contents of the first 70 variable-trim locations VT001 - VT070 can be monitored and manually changed from the Bendix interface console. Nominal values of these variables are stored in the Bendix Bounds program in the standard trim locations ST001 - ST070. The section of the Bounds program in which ST001 - ST070 are defined is presented in Table B-9. The contents of locations VT071 - VT254

can only be monitored from console. VT039 acts as a logical switch for the initialization section of the program. If VT039 = 16 this portion of the program will be executed and VT039 will be set to zero. If VT039 \neq 16 the initialization section will be bypassed. Since all the VT numbers encountered in this portion of the program are in the range VT001 - VT070, they can all be manually changed from Bendix interface console.

Interpolation Interval Determination

The gains (associated with the feedback quantities) and the open-loop information (fuel and pressure values) for both the speed controller and the pressure controller are given at four values of speed. To obtain values for the gains and open-loop information over the whole speed regime linear interpolation is used (Figure B-3). Since the quantities to be interpolated are given at four values of speed N , there are three possible intervals of interpolation. The four values of speed are $N_1 = 8250$ rpm, $N_2 = 11,550$ rpm, $N_3 = 14,025$ rpm, and $N_4 = 16,500$ rpm. Thus, the three intervals are $[N_1, N_2]$, $[N_2, N_3]$, $[N_3, N_4]$. The sensed speed N (in the program sensed speed is VT157) is tested to determine into which interval it falls. Then any quantity, call it f , given at the four values of N can be written as a linear function of N as follows:

$$f(N) = f(N_i) C_1 + f(N_{i+1}) C_2 \quad (B-1)$$

where

$$C_1 = \frac{(N_{i+1} - N)}{(N_{i+1} - N_i)} \quad \text{and} \quad C_2 = \frac{(N - N_i)}{(N_{i+1} - N_i)} \quad \text{for } i = 1, 2, 3.$$

The interval and the quantities C_1 and C_2 are calculated in this portion of the program.

Exits from this section of the program are given the labels IN1F, IN2F or IN3F, depending on whether the sensed speed N satisfies $N_1 \leq N \leq N_2$, $N_2 \leq N \leq N_3$, or $N_3 \leq N \leq N_4$.

Interpolation Logic

The three sections in this portion of the program (Figure B-4) all evaluate an equation like Equation (B-1). Therefore, the logic in each section is the same. The difference is in the label used for $f(N_i)$ and $f(N_{i+1})$. The different labels represent the initial address in a sequence of addresses of quantities associated with the same speed. In each case the label is influenced by index register one (XR1). Initially (XR1) is set to zero and an equation similar to (B-1) is evaluated in double precision. XR1 is then incremented by one and tested against label NGFT (NGFT = 18). If $XR1 < NGFT$ the interpolation continues, if $XR1 \geq NGFT$, the interpolation is done and we are transferred to label FUELM.

Interpolation Scaling

Both C_1 and C_2 are numbers such that $0 \leq C_1, C_2 \leq 1$ and $C_1 + C_2 = 1$. In the IBM 1800, fractional numbers cannot be represented except as the ratio of two integer numbers. Therefore, the computation of C_1 and C_2 has to be scaled. The scale factor used in the program is $2^7 = 128$. The scale factor of 2^7 is removed after the interpolation by a shift right seven.

Integral Speed and Integral Pressure

The integral speed and integral pressure portion of the program (Figure B-5) consists of logic to initialize, integrate, and limit two simple differential equations in time. The integral speed differential equation is

$$\dot{EN} = -5.3333 (N - N_{PLA}) \quad (B-2)$$

where EN is the integral of the error between sensed speed N(VT157) and requested speed N_{PLA} (VT128).

The integral pressure differential equation is

$$\dot{EP} = -5.3333 (PT3 - PT3\emptyset) F(N) \quad (B-3)$$

where EP is the integral of the error between sensed PT3 (VT102) and a boundary value PT3 \emptyset (PT3NB) and F(N) is a function of sensed speed (i. e. , the coefficient in the differential equation is not constant; c.f. Table B-6 and the related discussion).

Initialization of the Differential Equations

The initial values of EN and EP are in VT-36 and VT037, respectively. The limiting values of EN and EP are taken to be the absolute values of VT036 and VT037, respectively. The initial value and the limiting value are changed whenever VT039 contains a sixty-four (64) or a sixteen (16).

Integration of the Differential Equations

The differential equations are integrated numerically using the trapezoidal rule

$$X_{n+1} = X_n + \frac{\Delta t}{2} (\dot{X}_n + \dot{X}_{n-1}) \quad (B-4)$$

where Δt is 0.015 second, X_n is the current value of the integral, \dot{X}_n is the current value of the derivative, and \dot{X}_{n-1} is the previous value of the derivative.

Interpolation as a Function of Power Lever

Early controllers (not documented) used PT5 and PT3 as well as an open-loop fuel as a function of the power lever (Figure B-7). The speed controllers used in engine tests require only open-loop fuel as a function of power lever. The power lever position is given in terms of a speed request in rpms in VT128. The method of interpolation is the same as it was for sensed speed. However, since only three quantities are being interpolated, no index registers are used.

Fuel Request Calculation

Three fuel requests are calculated: a speed fuel request, a pressure fuel request and a minimum fuel request (Figure B-8). The minimum fuel request is calculated in the interpolation logic as a function of sensed speed and is stored in WFMNN. The speed control fuel request is calculated as the sum of an open-loop fuel scheduled as a linear function of power lever and the following feedback quantities:

- The error between sensed and requested speed
- An integral of the error between sensed and requested speed

The pressure control fuel request is calculated as the sum of an open-loop fuel scheduled as a linear function of sensed speed and the following feedback quantities:

- The error between PT5 sensed and a given PT5 scheduled as a linear function of speed
- The error between P3 sensed and a given P3 scheduled as a linear function of speed
- The integral of the error between P3 sensed and a given P3 as a linear function of speed.

Starting at label MDW6, all of the ingredients used in calculating the fuel request for the speed and pressure controllers are stored in VT162 - VT176 for checking purposes. Beginning at label MEPT, the five feedback quantities mentioned previously are calculated and stored in VT196 - VT200. The speed control fuel request starts at label FREQE and each of the products involved in the sum is stored in VT201 - VT204. Finally, the fuel request for the speed controller is stored in SUMEF and VT071. The pressure fuel request calculation starts at label FREQP and each of the products involved in the sum is stored in VT205 - VT207. The fuel request for the pressure controller is stored in SUMPF and VT072.

Mode Select Logic

In Figure B-9 the mode select logic starts at level MDSWT. The minimum between the speed fuel request VT071 and the pressure fuel request VT072 is stored in VT180. The maximum between VT180 and WFMNN (minimum fuel) is stored in VT180. At this point a mode number is stored in VT074, depending on which controller is used. The mode numbers are: 3276 for the speed controller, 6552 for the pressure controller, and 9828 for the minimum fuel request.

Fuel Request Filter Logic

The fuel request in VT180 is put through a first-order lag $[30/(S+30)]$. The lag is digitized using Tustin's method with the resulting difference equation

$$y_n = \left(\frac{31}{49}\right) y_{n-1} + \left(\frac{9}{49}\right) U_n + \left(\frac{9}{49}\right) U_{n-1} \quad (\text{B-5})$$

where y_n is the current output (i. e., filtered fuel request), y_{n-1} is the previous output, U_n is the current input (unfiltered fuel request) and U_{n-1} is

the previous input. The coefficients in the difference equation are a function of the sample time Δt which is taken to be 0.015 second.

Exhaust Nozzle Request Calculation

Figure B-11 presents the flow chart.

The nozzle is open for speeds less than or equal to 14,025 rpm. The nozzle request representing "open" is stored in VT034. The nozzle is closed for speeds greater than or equal to 16,500 rpm. The nozzle request representing "closed" is stored in VT035. For speeds between 14,025 rpm and 16,500 rpm the nozzle request decreases linearly from "open" to "closed." The "speed" used in the nozzle request calculation is sensed speed (VT157) if the control mode is not speed control. If the control mode is speed control the speed used is that requested by the power lever (VT128). The nozzle request is stored in VT081. After this calculation has been completed and index register one has been restored, one control cycle update has been completed and control is passed to the Bendix program.

EQUILIBRIUM-TEMPERATURE SOFTWARE

Flow charts for the main computational blocks and for each block are presented in Figures B-12 through B-23. Table B-10 is a glossary of terms. The program is presented in Tables B-11 through B-14. A listing of the Bendix Bounds program corresponding to the Equilibrium-Temperature Program is presented in Table B-13.

The main computational blocks for the speed temperature control program are shown in Figure B-12. The major differences between this program and the speed pressure control program are the filtering logic for T4 whistle,

the number of feedbacks, and the names given to the gains and open-loop information. Consequently, a description of each of the blocks in Figure B-12 will be given in comparative terms of the description given for the speed and pressure controller.

Initialization

It is clear from looking at the detached flow chart (Figure B-13) that more items are transferred from variable trim locations to locations in the control program. This is true, because the temperature controller has more feedbacks than the pressure controller. Consequently, the VT numbers encountered in this section of the program are in the range VT001 - VT090 (rather than the previous VT001 - VT070). The Bendix Bounds program has been modified to allow the first 90 VT numbers to be changed at the interface console. Nominal values of these VT variables are stored in the standard trim locations ST001 - ST090 in the beginning of the Bounds program (Table B-13). The logic to get into this section of the program is the same as previously described. In addition to the increased number gains and open-loop information to be transferred, an added logic switch, ISW, is initialized. This switch is used to initialize the filtering logic for T4 whistle.

Interpolation Interval Determination

This section of the program (Figure B-14) is exactly the same as for the speed and pressure control program.

Interpolation Logic

The only differences in this section of the program are the labels (names) given to the gains and open-loop information (fuel, pressures, and temperature) associated with temperature controller as opposed to the pressure controller; cf Figure B-15.

Filtering Logic for T4 Whistle

The temperature sensed by the whistle (VT097) goes through a lead-lag filter and the output of the filter is stored in T4WF, Figure B-16. The transfer function for the filter with VT097 as input and T4WF as output is

$$\frac{T4WF}{VT097} = \frac{\tau_2 S + 1}{(K_1)(\tau_2)S + 1} \quad (B-6)$$

where τ_2 and K_1 are piecewise linear functions of PT3.

The table below gives τ_2 and K_1 versus PT3.

PT3 (psi)	K_1	τ_2
24.5	0.50	30.0
39.0	0.53	17.0
58.5	0.56	10.0
102.0	0.60	8.0

The filter is implemented digitally by the following two equations.

$$\dot{X}T4 = \frac{1}{K_1 \tau_2} (VT097 - XT4) \quad (B-7)$$
$$T4WF = \tau_2 \dot{X}T4 + XT4$$

In the program, K_1 is scaled up by 100 and τ_2 is scaled up by 10. The label for $K_1 \cdot 100$ is K1THD and the label for $\tau_2 \cdot 10$ is TAU2T. The coding starts at label FUELM with the calculation of K1THD and TAU2T as a function of PG3. At label STP1 the logical switch ISW is tested. If ISW is unequal to 1234, initialization of the filter equations takes place. Otherwise branch to STP2. In the initialization logic XT4 is set equal to VT097, XT4 is set equal to zero and ISW is set equal to 1234 followed by a branch to STP3. Starting at label STP2, the derivative $\dot{X}T4$ is calculated double precision and stored in XT4D. At label STP3 the differential equation is integrated one step forward in time using the trapezoidal rule (Δt taken to be 0.015 second). The updated value of XT4 is stored in XT4 in double precision. At this point the filtered T4 whistle is computed and stored in T4WF.

Integral Speed and Integral Temperature

In this portion of the program (Figure B-17) the integral pressure differential equation has been replaced with an integral temperature differential equation

$$\dot{ET} = -13.3333 (T4WF - T4\partial) \quad (B-8)$$

where ET is the integral of the error between sensed T4 whistle filtered and a boundary value T4 as a function of sensed speed. The initial value of ET is stored in VT038 and the limiting value of ET is taken to be the absolute value of VT033.

Additional logic was added to the integration routine in this section to reset the values of EN and ET to zero under the following conditions:

$$EN = 0 \quad \text{if } VT074 \text{ (mode switch)} \neq 3276$$

$$ET = 0 \quad \text{if } VT074 \neq 6552$$

This logic is inserted in the program immediately after the integrals have been updated (section of the program beginning with statement number HWS03300). The parameters EN and ET are updated only if the program is in the right mode; EN is updated if the speed control loop is regulating the engine and ET is updated if the temperature control loop is regulating the engine.

The rest of this section of the program is the same as described previously.

Interpolation as a Function of Power Lever

This portion of the program (Figure B-19) is exactly the same as previously described for the pressure controller.

Fuel Request Calculation

The pressure fuel request calculation is replaced with a temperature fuel request (Figure B-20). The temperature fuel request is calculated as the sum of an open-loop fuel scheduled as a linear function of sensed speed and the following feedback quantities:

- The error between PT5 sensed and a given PT5 scheduled as a linear function of speed
- The error between PT3 sensed and a given PT3 scheduled as a linear function of speed
- The error between T4 whistle filtered and a T4 given scheduled as a linear function of speed
- The integral of the error between PT3 sensed and a given PT3 scheduled as a linear function of speed.

The temperature fuel request calculation starts at label FREQT and each of the products involved in the sum is stored in VT205 - VT207. The fuel request for the temperature controller is stored in SUMTF and VT073.

Mode Select Logic

The only difference in this section of the program is that the minimum between the speed fuel request VT071 and the temperature fuel request VT073 (rather than the pressure fuel request VT072) is stored in VT180; cf Figure B-21.

Fuel Request Filter Logic

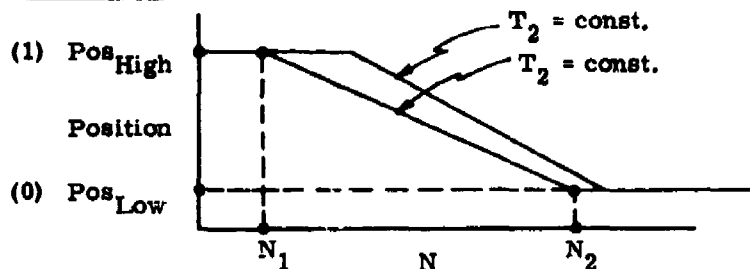
The same as previously described for pressure; Figure B-22.

Exhaust Nozzle Request Calculation

This is the same as previously described for pressure (Figure B-23).

Table B-1. IGV, BLD, and A8 Schedules

IGV and BLD



$$\text{Position} = \text{Pos}_{\text{Low}} + (\text{Pos}_{\text{High}} - \text{Pos}_{\text{Low}}) \times \frac{(N - N_1)}{(N_2 - N_1)}$$

where N is spool speed.

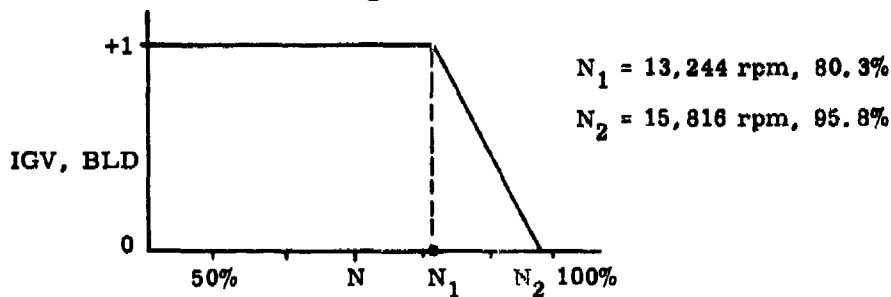
$$N_1 \text{ (rpm)} = 11,800 + (T_2 \text{ } ^\circ\text{R} - 420 \text{ } ^\circ\text{R}) \times \frac{2100}{160}$$

$$N_2 \text{ (rpm)} = 14,900 + (T_2 \text{ } ^\circ\text{R} - 428 \text{ } ^\circ\text{R}) \times \frac{1100}{64} \text{ if } T_2 \text{ } ^\circ\text{F} \leq 25 \text{ } ^\circ\text{F}$$

$$= 16,000 - (T_2 \text{ } ^\circ\text{R} - 484 \text{ } ^\circ\text{R}) \times \frac{200}{50} \text{ if } 25 \text{ } ^\circ\text{F} < T_2 \text{ } ^\circ\text{F} < 75 \text{ } ^\circ\text{F}$$

$$= 15,800 + (T_2 \text{ } ^\circ\text{R} - 534 \text{ } ^\circ\text{R}) \times \frac{500}{32} \text{ if } T_2 \text{ } ^\circ\text{F} \geq 75 \text{ } ^\circ\text{F}$$

∴ on a normal day ($T_2 = 70 \text{ } ^\circ\text{F}$) the schedules are:



A8

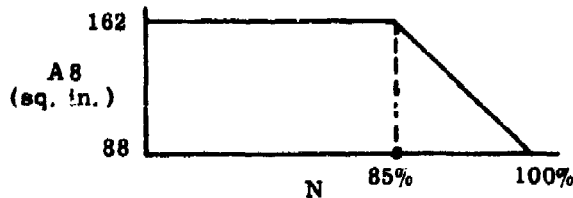


Table B-2. Generic Fuel Control Law

$$u_f = \frac{30.0 u_i}{s + 30.0}$$

$$u_1 = \text{Max} \begin{cases} u_2 \\ u_d \end{cases}$$

$$u_2 = \text{Min} \begin{cases} u_e \\ u_p \\ u_t \end{cases}$$

$$u_d = u_d [N]$$

$$u_e = k_N [N - N_o (pla)] + k_{EN} EN + k_{P3} [P3 - P3_o (pla)] \\ + k_{PT5} [PT5 - PT5_o (pla)] + u_{e_o} (pla)$$

$$u_p = k_{EP} + k_{P3} [P3 - P3_o (N)] + k_{PT5} [PT5 - PT5_o (N)] + u_{p_o} (N)$$

$$u_t = k_{ET} ET + k_{TT4} [PT5 - PT5_o (N)] + u_{t_o} (N)$$

$$\dot{EN} = \begin{cases} 0 & \text{If } EN \geq ENL \text{ \& } -5.3333 (N - N_o) \geq 0 \\ 0 & \text{If } EN \leq ENL \text{ \& } -5.3333 (N - N_o) \leq 0 \\ -5.3333 [N - N_o (pla)] & \text{otherwise} \end{cases}$$

$$\dot{EP} = \begin{cases} 0 & \text{If } EP \geq EPL \text{ \& } -PC (P3 - P3_o) \geq 0 \\ 0 & \text{If } EP \leq EPL \text{ \& } -PC (P3 - P3_o) \leq 0 \\ -PC(N) [P3 - P3_o (N)] & \text{otherwise} \end{cases}$$

Table B-2. Generic Fuel Control Law (Concluded)

where

$$PC = \begin{cases} 5.333 & \text{If } N < 14,025 \text{ rpm} \\ 13.333 & \text{If } N \geq 14,025 \text{ rpm} \end{cases}$$

$$\dot{ET} = \begin{cases} 0 & \text{If } ET \geq ETL \text{ \& } -13.333 (T4WF - TT4_o) \geq 0 \\ 0 & \text{If } ET \leq ETL \text{ \& } -13.333 (T4WF - TT4_o) \leq 0 \\ -13.333 [T4WF - TT4_o (N)] & \text{otherwise} \end{cases}$$

N (rpm), P3 (psi), and PT5 (psi) are taken to be the outputs of engine sensors, pla is throttle in part of full; e. g., 0.75 pla commands 75 percent rpm, T4W (°R) is the output of the Honeywell fluidic (whistle) T4 sensor

$$T4WF = \left[\frac{1}{(KI)\tau_2} \left(1 - \frac{1}{KI} \right) (P3) \right] T4DUM + \frac{1}{KI} (P3) TT4W$$

$$T4DUM = \left[\frac{1}{(KI)(T2)} (P3) \right] T4DUM + TT4W$$

$$T4WF \approx \frac{50.0}{S + 50} TT4$$

$$ENL = 200.0$$

$$EPL = 1.0$$

$$ETL = 100.0$$

Table B-3. Perturbation Gains

% N	k_N (lb/sec)/rpm	$k_{E \begin{smallmatrix} N \\ P \\ T \end{smallmatrix}}$	k_{P3} (lb/sec)/psi	k_{PT5} (lb/sec)/psi	k_{TT4} (lb/sec)/(deg F)
50E	-0.46718-3	+0.58461-4	---	---	---
50P	---	+0.45650-1	-0.18636+0	+0.15861+0	---
50T	---	+0.11304-3	-0.15966-2	+0.12096-2	-0.22757-3
70E	-0.2736-3	+0.53844-4	---	---	---
70P	---	+0.20271-1	-0.62334-1	-0.44936-1	---
70T	---	+0.14074-3	+0.53354-1	-0.20311-2	-0.28462-3
85E	-0.26479-3	+0.12239-3	---	---	---
85P	---	+0.15561-2	-0.71783-1	+0.51485-1	---
85T	---	+0.16155-3	+0.91896-2	-0.74486-3	-0.18812-3
100E	-0.53363-3	+0.31975-3	---	---	---
100P	---	+0.12779-1	-0.49166-1	+0.43431-1	---
100T	---	+0.23413-3	-0.18094-1	+0.13297-1	-0.78669-5

Table B-4. Open-Loop Fuel Flows* (lb/hr)

% N	ue_o [pla]	up_o [N]	ut_o [N]	ud [N]
50.0	519.0	779.0	651.0	200.0
70.0	693.0	1740.0	1000.0	350.0
85.0	934.0	2573.0	2000.0	500.0
100.0	1648.0	3478.0	2400.0	1000.0

* These are for the APL engine at 29.55 inches of Hg and 82°F. They should be corrected with ambient conditions.

Table B-5. Equilibrium and Boundary States*

%N	50	70	85	100
Equilibrium				
N_o [plu]rpm	8,250.0	11,550.0	14,025.0	16,500.0
Pressure				
$P3_o$ [N]psi	22.0	35.5	55.0	80.0
$PT5_o$ [N]psi	14.8	16.4	20.5	25.6
Temperature				
$TT4_o$ [N]°F	1,020.0	900.0	1,050.0	1,160.0
$P3_o$ [N]psi	23.5	35.5	55.0	80.0
$PT5_o$ [N]psi	14.8	16.4	20.5	26.5

*These are for the APL engine at 99.99 inches of Hg and 99°F. They should be corrected with ambient conditions.

Table B-6. An Integral Transformation

\dot{x}	=	+ Fx	
\dot{E}	=	- dx	
\dot{w}_f	=	+ Fx	- a w _f + gu
u	=	+ kx	

\dot{x}	=	+ Fx	
$\frac{\dot{E}}{\mu}$	=	$-\frac{d}{\mu} x$	
\dot{w}_f	=		- a w _f + gu
u	=	+ kx + $(\lambda\mu) \frac{E}{\mu}$	

Table B-7. Glossary for Equilibrium - Pressure Control

VT Number	Transferred To (Program Label)	Description	Standard Value (Defined in the Bendix Program)
009	---	Logic switch: If VT009 > 123 the Honeywell controller is in; otherwise not	0
012	DEF11	Speed control gain associated with $(N-N_{pla})$ at 8250 rpm	$(-215) \times 16$
013	WEF1	Open-loop fuel-speed control at 2250 rpm	519 lb/hr
014	P3P1	Open-loop PT3-speed control at 8250 rpm	2200 (psi x 100)
015	KEF14	Speed control gain associated with EN at 8250 rpm	$(27) \times 16$
016	KEF21	Speed control gain associated with $(N-N_{pla})$ at 11,550 rpm	$(-126) \times 16$
017	WEF2	Open-loop fuel-speed control at 11,550 rpm	693 lb/hr
018	P3P2	Open-loop PT3 - speed control at 11,550 rpm	3550 (psi x 100)
019	KEF24	Speed control gain associated with EN at 11,550 rpm	$(25) \times 16$
020	KEF31	Speed control gain associated with $(N-N_{pla})$ at 14,025 rpm	$(-122) \times 16$
021	WEF3	Open-loop fuel-speed control at 14,025 rpm	934 lb/hr
022	P3P3	Open-loop PT ₃ - speed control at 14,025 rpm	5500 (psi) x 100
023	KEF34	Speed control gain associated with EN at 14,025 rpm	$(56) \times 16$
026	---	If this number is made large, Bendix bound on fuel will not be in effect	2^{14}
028	---	Logical switch: if VT028 = 64 Honeywell nozzle is used; otherwise not	0

Table B-7. Glossary for Equilibrium - Pressure Control (Continued)

VT Number	Transferred To (Program Label)	Description	Standard Value (Defined in the Bendix Program)
034	---	Exhaust request open	9640
035	---	Exhaust request closed	2650
036	ENK, ENKL	Initial value of integral speed limits value	1600
037	EPK, EPKL	Initial value of integral pressure limits value	1600
039	---	Logic switch: VT039 = 16 initializes everything. VT039=64 initializes EN, EP only	16
040	KEF41	Speed control gain associated with $(N-N_{pla})$ at 16,500 rpm	$(-246) \times 16$
041	WEF4	Open-loop fuel-speed control at 16,500 rpm	1648 lb/hr
042	P3P4	Open-loop PT3 - speed control at 16,500 rpm	3090 (psi x 100)
043	KEF44	Speed control gain associated with EN at 16,500 rpm	$(147) \times 16$
044	KPF11	Fudge factor used in EP at 8250 rpm	3089
045	KPF12	Pressure control gain - (PT5 - PT5d) at 8250 rpm	$(571) \times 16$
046	KPF13	Pressure control gain - (PT3 - PT3d) at 8250 rpm	$(-671) \times 16$
047	WPF1	Open-loop fuel-pressure control at 8250 rpm	779 lb/hr
048	KPF21	Fudge factor used in EP at 11,550 rpm	$(576) \times 16$
049	KPF22	Pressure control gain - (PT5 - PT5d) at 11,550 rpm	$(-162) \times 16$
050	KPF23	Pressure control gain - (PT3 - PT3d) at 11,550 rpm	$(-224) \times 16$

Table B-7. Glossary for Equilibrium - Pressure Control (Concluded)

VT Number	Transferred To (Program Label)	Description	Standard Value (Defined in the Bendix Program)
061	WPF2	Open-loop fuel-pressure control at 11,550 rpm	1740 lb/hr
062	KPF31	Fudge factor used in EP at 14,025 rpm	(59) x 16
063	KPF32	Pressure control gain - (PT5 - PT50) at 14,025 rpm	(185) x 16
064	KPF33	Pressure control gain - (PT3 - PT30) at 14,025 rpm	(-258) x 16
065	WPF3	Open-loop fuel-pressure control at 14,025 rpm	2573 lb/hr
066	KPF41	Fudge factor used in EP at 16,500 rpm	(364) x 16
067	KPF42	Pressure control gain - (PT5 - PT50) at 16,500 rpm	(156) x 16
068	KPF43	Pressure control gain (PT3 - PT30) at 16,500 rpm	(-177) x 16
069	WPF4	Open-loop fuel-pressure control at 16,500 rpm	3478 lb/hr
071	---	Speed fuel request 1 count = 4 lb/hr	0
072	---	Pressure fuel request 1 count = 4 lb/hr	0
074	---	Mode number: 3276 = speed control 6552 = pressure control 9228 = minimum fuel	0
081	--	Exhaust actuator request	0
180	---	Fuel request calculated by control program 3.25 counts = 1 lb/hr	---

Table B-8. Equilibrium - Pressure Subprogram

```

// JOB          VDISK
// DMP
*DELETE          HNECT
// ASM
*OVERFLOW SECTORS ,,,9
*LIST
*XREF
*ONEWORDINTEGERS
*COMMON (DUMY(127),IVT00,JDUMY(127),IBT0,HEAST(64),IABCH(2)
          ENT          HNECT
HNECT DC          ***
          STX          L1 XR1+1
*
          LDX          L1 0
          M            L TESTN
*
*   INITIALIZATION
*
TESTN EQU          *
LD          2 VT039
S            L          *16
BNZ          MICK
STO          2 VT039
LD          2 VT012
SRT          *
STO          L KEF11
LD          2 VT013
STO          L WCF1
LD          2 VT014
STO          L P3P1
LD          2 VT015
SRT          *
STO          L KEF14
LD          2 VT016
SRT          *
STO          L KEF21
LD          2 VT017
STO          L WCF2
LD          2 VT018
STO          L P3P2
LD          2 VT019
SRT          *
STO          L KEF24
LD          2 VT020
SRT          *
STO          L KEF31
LD          2 VT021
STO          L WCF3
LD          2 VT022
STO          L P3P3
LD          2 VT023
SRT          *
STO          L KEF34
LD          2 VT040
SRT          *
STO          L KEF41
LD          2 VT041

```

HWS00010
HWS00020
HWS00030
HWS00040
HWS00050
HWS00060
HWS00070
HWS00080
HWS00090
HWS00100
HWS00110
HWS00120

HWS00190

Table B-8. Equilibrium - Pressure Subprogram (Continued)

```

STO L WPF4
LD 2 VT042
STO L P3P4
LD 2 VT043
SRT 4
STO L KEF44
LD 2 VT044
STO L KPF11
LD 2 VT045
SRT 4
STO L KPF12
LD 2 VT046
SRT 4
STO L KPF13
LD 2 VT047
STO L WPF1
LD 2 VT048
SRT 4
STO L KPF21
LD 2 VT049
SRT 4
STO L KPF22
LD 2 VT050
SRT 4
STO L KPF23
LD 2 VT061
STO L WPF2
LD 2 VT062
SRT 4
STO L KPF31
LD 2 VT063
SRT 4
STO L KPF32
LD 2 VT064
SRT 4
STO L KPF33
LD 2 VT065
STO L WPF3
LD 2 VT066
SRT 4
STO L KPF41
LD 2 VT067
SRT 4
STO L KPF42
LD 2 VT068
SRT 4
STO L KPF43
LD 2 VT069
STO L WPF4
LD L 00
STO L TIME
STO L SWLAG

```

• INTERVAL DETERMINATION

```

•
MICK EQU 0
LD 2 VT157
S 06250

```

MWS00160
MWS00170
MWS00180

MWS00200
MWS00210

Table B-8. Equilibrium - Pressure Subprogram (Continued)

	BP		TMAX		HWS00220
	LD		#1		HWS00230
	ST0		NIN		HWS00240
	LD		#128		
	ST0		C1		HWS00260
	LD		#0		HWS00270
	ST0		C2		HWS00280
	B	L	INIF		HWS00290
C1	OC		***		HWS00130
C2	OC		***		HWS00140
NIN	OC		***		HWS00150
	LORG				
TMAX	LD		#16500		HWS00300
	S	2	VT157		HWS00310
	BP		TIN1		HWS00320
	LD		#3		HWS00330
	ST0		NIN		HWS00340
	LD		#0		HWS00350
	ST0		C1		HWS00360
	LD		#128		
	ST0		C2		HWS00380
	B	L	IN3F		HWS00390
TIN1	LD		#11550		HWS00400
	S	2	VT157		HWS00410
	BN		TIN2		HWS00420
	SRT		9		
	D		#3300		HWS00440
	ST0		C1		HWS00450
	LD		#128		
	S		C1		HWS00470
	ST0		C2		HWS00480
	LD		#1		HWS00490
	ST0		NIN		HWS00500
	B	L	INIF		HWS00510
TIN2	LD		#14025		HWS00520
	S	2	VT157		HWS00530
	BN		TIN3		HWS00540
	SRT		9		
	D		#2475		HWS00560
	ST0		C1		HWS00570
	LD		#128		
	S		C1		HWS00590
	ST0		C2		HWS00600
	LD		#2		HWS00610
	ST0		NIN		HWS00620
	B	L	IN3F		HWS00630
TIN3	LD		#16500		HWS00640
	S	2	VT157		HWS00650
	SRT		9		
	D		#2475		HWS00670
	ST0		C1		HWS00680
	LD		#128		
	S		C1		HWS00700
	ST0		C2		HWS00710
	LD		#3		HWS00720
	ST0		NIN		HWS00730
	B	L	IN3F		HWS00740
	LORG				HWS00750

Table B-6. Equilibrium - Pressure Subprogram (Continued)

* EQUILIBRIUM FUEL FLOW 50 GAINS		MWS00760
KEF11	DC	***
KEF12	DC	0
KEF13	DC	0
KEF14	DC	***
WPF1	DC	***
P3E1	DC	2705
P5E1	DC	1633
* PRESSURE FUEL FLOW 50 GAINS		MWS00840
KPF11	DC	***
KPF12	DC	***
KPF13	DC	***
KPF14	DC	53
WPF1	DC	***
P3P1	DC	***
P5P1	DC	1480
WTF1	DC	1118
A61	DC	162
WFMN1	DC	650
T61	DC	2602
BUMP1	DC	1
SETX1	DC	0
NGFT	DC	18
C11	DC	***
C21	DC	***
TST1	DC	***
SUM1	USS E	0
	DC	0
	DC	0
* INTERPOLATE INTERVAL 1		
* INIF EQU *		MWS01050
LD	L C1	MWS01060
ST0	L C11	MWS01070
LD	L C2	MWS01080
ST0	L C21	MWS01090
LD	L SETX1	MWS01100
ST0	L TST1	MWS01110
LUP1	L1 TST1	MWS01120
LD	L1 KEF11	MWS01130
M	L C11	MWS01140
ST0	L SUM1	
LD	L1 KEF21	MWS01170
M	L C21	MWS01180
AD	L SUM1	
SRT	L 7	
SET	L 16	
ST0	L1 KEFN1	MWS01220
LD	L TST1	MWS01230
A	L BUMP1	MWS01240
ST0	L TST1	MWS01250
S	L NGFT	MWS01260
BN	L LUP1	MWS01270
B	L FUFLM	MWS01280
* EQUILIBRIUM FUEL FLOW 70 GAINS		MWS01290
KEF21	DC	***
KEF22	DC	0

Table B-8. Equilibrium - Pressure Subprogram (Continued)

```

KEF23 DC      0
KEF24 DC      0
WPF2  DC      0
P3E2  DC     4361
P5E2  DC     1893
*
*   PRESSURE      FUEL FLOW 70 GAINS      HWS01370
KPF21 DC      0
KPF22 DC      0
KPF23 DC      0
KPF24 DC     127
WPF2  DC      0
P3P2  DC      0
P5P2  DC     1640
WTF2  DC     1877
AB2   DC     162
WFMN2 DC     1138
TB2   DC     2563
C12   DC      0
C22   DC      0
TST2  DC      0
SUM2  BSS E    0
      DC      0
      DC      0
*
*   INTERPOLATE INTERVAL 8
*
IN2F  EQU      0
      LD      L  C1
      ST0     C12
      LD      L  C2
      ST0     C22
      LD      L  SETX1
      ST0     TST2
LUP2  LDX     I1 TST2
      LD      L1 KEF21
      M       C12
      ST0     SUM2
      LD      L1 KEF31
      M       C22
      AD      SUM2
      SRT     7
      SLT     16
      ST0     L1 KEF21
      LD      TST2
      A       L  BUMP1
      ST0     TST2
      B       L  NGFT
      BN      LUP2
      B       L  FUELM
*
*   EQUILIBRIUM FUEL FLOW 85 GAINS
KEF31 DC      0
KEF32 DC      0
KEF33 DC      0
KEF34 DC      0
WPF3  DC      0
P3E3  DC     6161
P5E3  DC     2243
*
*   PRESSURE      FUEL FLOW 85 GAINS      HWS01870

```

Table B-8. Equilibrium - Pressure Subprogram (Continued)

KPF31	DC	...	
KPF32	DC	...	
KPF33	DC	...	
KPF34	DC	236	
WPF3	DC	...	
P3P3	DC	...	
P5P3	DC	2050	
WTF3	DC	3102	HWS01960
A83	DC	162	HWS01970
WFMN3	DC	1625	
Y83	DC	2743	HWS02000
C13	DC	...	HWS02010
C23	DC	...	HWS02020
TST3	DC	...	HWS02040
SUM3	WSB	E	0
	DC		0
	DC		0
* INTERPOLATE INTERVAL 3			
IN3F	EGU	*	HWS02050
	LD	L	C1
	ST0		C13
	LD	L	C2
	ST0		C23
	LD	L	SETX1
	ST0		TST3
LUP3	L0X	I1	TST3
	LD	L1	KEF31
	M		C13
	ST0		SUM3
	LD	L1	KEF41
	M		C23
	AD		SUM3
	SRT		7
	SLT		16
	ST0	L1	KEFN1
	LD		TST3
	A	L	BUMPI
	ST0		TST3
	S	L	NGFT
	BN		LUP3
	M	L	FUELM
* EQUILIBRIUM FUEL FLOW 100 GAINS			
KEF41	DC	...	
KEF42	DC	0	
KEF43	DC	0	
KEF44	DC	...	
WEF4	DC	...	
P3E4	DC	10110	
P5E4	DC	3988	
* PRESSURE FUEL FLOW 100 GAINS			
KPF41	DC	...	HWS02370
KPF42	DC	...	
KPF43	DC	...	
KPF44	DC	317	
WPF4	DC	...	
P3P4	DC	...	
P5P4	DC	2650	

Table B-8. Equilibrium - Pressure Subprogram (Continued)

AFM14	DC	91	11 247.1
T84	DC	3251	
FUELN	EQU	2684	MWS02500
		.	MWS02510
* INITIALIZE INTEGRALS AND LIMITS ON INTEGRALS			
	LD	2 VT039	MWS02520
	S	064	MWS02530
	BNZ	MDW9	MWS02540
	STO	2 VT039	
	LD	2 VT036	MWS02550
	STO	ENK	
	BNN	SENL	
	LD	00	
	S	ENK	
SENL	STO	ENKL	
	LD	2 VT037	MWS02570
	SRT	4	
	STO	EPK	
	BNN	SEPL	
	LD	00	
	S	EPK	
SEPL	STO	EPKL	
	LD	2 VT038	MWS02590
	STO	ETKL	MWS02600
	STO	ETK	
MDW9	EQU	.	MWS02610
			MWS02620
			MWS02630
			MWS02720
* CALCULATE DERIVATIVES FOR EN EP ET			
	LD	2 VT128	MWS02730
	S	2 VT157	MWS02740
	STO	ENDK	MWS02750
	LD	L PT3NB	MWS02760
	S	2 VT102	MWS02770
	STO	EPDK	MWS02780
	LD	2 VT097	MWS02840
	SRT	16	MWS02850
	D	010	MWS02860
	S	L TBBN	MWS02870
	STO	ETDK	MWS02880
	LD	TIME	MWS02890
	BNZ	INTEG	MWS02900
	LD	ENDK	MWS02910
	STO	ENDK1	MWS02920
	LD	EPDK	MWS02930
	STO	EPDK1	MWS02940
	LD	ETDK	MWS02950
	STO	ETDK1	MWS02960
	LD	2 VT036	MWS02970
	STO	ENK	MWS02980
	BNN	STENL	MWS02990
	LD	00	
	S	ENK	
STENL	STO	ENKL	
	LD	2 VT037	MWS03010
	SRT	4	
	STO	EPK	
	BNN	STEPL	
	LD	00	
	S	EPK	
			MWS03030

Table B-8. Equilibrium - Pressure Subprogram (Continued)

STEPL	STO	EPKL			
	LD	VT038			HWS03050
	STO	ETK			HWS03060
	STO	ETKL			HWS03070
	LD	TIME			HWS03080
	STO	TIME			HWS03090
	B	L	INTEG		
	LORG				HWS03110
	*GENERATE	EN	EP	ET	HWS03120
	TIME	DC	0		
	ENDK	DC	***		HWS03140
	ENDK1	DC	***		HWS03150
	EPDK	DC	***		HWS03160
	EPDK1	DC	***		HWS03170
	ETDK	DC	***		HWS03180
	ETDK1	DC	***		HWS03190
	DT	DC	15		HWS03200
	ENK	DC	***		HWS03210
	ENKL	DC	***		HWS03220
	EPK	DC	***		HWS03230
	EPKL	DC	***		HWS03240
	ETK	DC	***		HWS03250
	ETKL	DC	***		HWS03260
	.				HWS03280
	INTEG	EQU	.		HWS03290
	.			CALCULATE EN	HWS03300
	LD	ENDK			
	A	ENDK1			
	M	DT			
	SLT	3		EN SCALED UP BY 8	
	D	*375			
	A	ENK			HWS03400
	STO	ENK			HWS03410
	.			CALCULATE EP	HWS03420
	LD	EPDK			
	A	EPDK1			
	M	DT			
	D	*375			
	M	L	KPFN1		
	D	*1000			
	A	EPK			HWS03490
	STO	EPK			HWS03500
	.			CALCULATE ET	HWS03510
	LD	ETDK			HWS03520
	M	*3			HWS03530
	SLT	16			HWS03540
	S	ETDK1			HWS03550
	M	DT			HWS03560
	D	*2000			HWS03570
	A	ETK			HWS03580
	STO	ETK			HWS03590
	.			LIMITS ON EN EP ET	HWS03600
	LD	ENK			HWS03610
	BN	MW1			HWS03620
	S	ENKL			HWS03630
	BNP	MW2			HWS03640
	LD	ENKL			HWS03650
	STO	ENK			HWS03660

Table E-8. Equilibrium - Pressure Subprogram (Continued)

	ST0	L	P5PL	HWS04240
	LD	L	WEF1	HWS04250
	ST0	L	WEFN	HWS04260
	B	L	MDW6	HWS04270
MDW1	LD	L	NPL4	HWS04280
	S	2	VT128	HWS04290
	BP		MDW2	HWS04300
	LD	L	P3E4	HWS04310
	ST0	L	P3PL	HWS04320
	LD	L	P5E4	HWS04330
	ST0	L	P5FL	HWS04340
	LD	L	WEF4	HWS04350
	ST0	L	WEFN	HWS04360
	B	L	MDW6	HWS04370
MDW2	LD	L	NPL2	HWS04380
	S	2	VT128	HWS04390
	BN		MDW3	HWS04400
	SRT	9		
	D		*3300	HWS04420
	ST0		CX1	HWS04430
	LD		*128	
	S		CX1	HWS04450
	ST0		CX2	HWS04460
	LD	L	P3E1	HWS04470
	ST0		P3L	HWS04480
	LD	L	P3E2	HWS04490
	ST0		P3M	HWS04500
	LD	L	P5E1	HWS04510
	ST0		P5L	HWS04520
	LD	L	P5E2	HWS04530
	ST0		P5M	HWS04540
	LD	L	WEF1	HWS04550
	ST0	L	WEFL	HWS04560
	LD	L	WEF2	HWS04570
	ST0	L	WEFM	HWS04580
	B	L	MDW5	HWS04590
	LORG			HWS04600
MDW3	LD		NPL3	HWS04610
	S	2	VT128	HWS04620
	BN		MDW4	HWS04630
	SRT	9		
	D		*2475	HWS04650
	ST0		CX1	HWS04660
	LD		*128	
	S		CX1	HWS04680
	ST0		CX2	HWS04690
	LD	L	P3E2	HWS04700
	ST0		P3L	HWS04710
	LD	L	P3E3	HWS04720
	ST0		P3M	HWS04730
	LD	L	P5E2	HWS04740
	ST0		P5L	HWS04750
	LD	L	P5E3	HWS04760
	ST0		P5M	HWS04770
	LD	L	WEF2	HWS04780
	ST0	L	WEFL	HWS04790
	LD	L	WEF3	HWS04800
	ST0	L	WEFM	HWS04810

Table B-8. Equilibrium - Pressure Subprogram (Continued)

	B	L	MDWS	HWS04820
NPL1	DC		8250	HWS04830
NPL2	DC		11850	HWS04840
NPL3	DC		14025	HWS04850
NPL4	DC		16500	HWS04860
CX1	DC		...	HWS04870
CX2	DC		...	HWS04880
P3L	DC		...	HWS04890
P3M	DC		...	HWS04900
P5L	DC		...	HWS04910
P5M	DC		...	HWS04920
WEFL	DC		...	HWS04930
WEFM	DC		...	HWS04950
SUMX	B58	E	0	
	DC		0	
	DC		0	
MDWS	LD		NPL4	HWS04960
	S	P	VT128	HWS04970
	SRT		9	
	D		02475	HWS04990
	ST0		CX1	HWS05000
	LD		0128	
	S		CX1	HWS05020
	ST0		CX2	HWS05030
	LD	L	P3E3	HWS05040
	ST0		P3L	HWS05050
	LD	L	P3E4	HWS05060
	ST0		P3M	HWS05070
	LD	L	P5E3	HWS05080
	ST0		P5L	HWS05090
	LD	L	P5E4	HWS05100
	ST0		P5M	HWS05110
	LD	L	WEF3	HWS05120
	ST0	L	WEFL	HWS05130
	LD	L	WEF4	HWS05140
	ST0	L	WEFM	HWS05150
MDWS	LD		P3L	HWS05160
	M		CX1	HWS05170
	ST0		SUMX	
	LD		P3M	HWS05200
	M		CX2	HWS05210
	AD		SUMX	
	SRT		7	
	SLT		16	
	ST0		P3PL	HWS05250
	LD		P5L	HWS05260
	M		CX1	HWS05270
	ST0		SUMX	
	LD		P5M	HWS05300
	M		CX2	HWS05310
	AD		SUMX	
	SRT		7	
	SLT		16	
	ST0		P5PL	HWS05350
	LD	L	WEFL	HWS05360
	M		CX1	HWS05370
	ST0		SUMX	
	LD		WEFM	HWS05400

Table B-8. Equilibrium - Pressure Subprogram (Continued)

M	CK2		MWS05410
AD	SUMX		
SRT	7		
SLT	16		
STO	WEFN		MWS05460
S	L MDW4		MWS05460
LOGR			MWS05470
MDW6	EQU	*	MWS05510
LD	L PSPL		MWS05520
STO	2 VT162		MWS05530
LD	L P3PL		MWS05540
STO	2 VT163		MWS05550
LD	L ENK		MWS05560
STO	2 VT164		MWS05570
LD	L WEFN		MWS05580
STO	2 VT165		MWS05590
LD	L KEFN1		MWS05600
STO	2 VT166		MWS05610
LD	L KEFN2		MWS05620
STO	2 VT167		MWS05630
LD	L KEFN3		MWS05640
STO	2 VT168		MWS05650
LD	L KEFN4		MWS05660
STO	2 VT169		MWS05670
LD	L PT3NB		MWS05680
STO	2 VT170		MWS05690
LD	L PT3NB		MWS05700
STO	2 VT171		MWS05710
LD	L EPK		MWS05720
STO	2 VT172		MWS05730
LD	L WPFN		MWS05740
STO	2 VT173		MWS05750
LD	L KPFN2		MWS05760
STO	2 VT174		MWS05770
LD	L KPFN3		MWS05780
STO	2 VT175		MWS05790
LD	L KPFN4		MWS05800
STO	2 VT176		MWS05810
.			MWS05820
.	CALCULATE X-X0	FOR EQUILIBRIUM PRESSURE	MWS05830
.			MWS05840
MEPT	EQU	*	
LD	2 VT157		MWS05820
S	2 VT128		MWS05830
STO	ME1		MWS05840
STO	2 VT196		
LD	2 VT108		MWS05850
S	L PSPL		MWS05860
STO	ME2		MWS05890
S	2 VT073 *		MWS07460
STO	2 VT197		
LD	2 VT102		MWS05900
S	L P3PL		MWS05930
STO	ME3		MWS05940
STO	2 VT198		
LD	L ENK		MWS05950
STO	ME4		MWS05960
			MWS05970

Table B-8. Equilibrium - Pressure Subprogram (Continued)

	LD	2	VT106	HWS06090
	M	L	PTSNB	HWS06010
	STO		MP2	HWS06020
	STO	2	VT199	
	LD	2	VT102	HWS06030
	S	L	PT3NB	HWS06060
	STO		MP3	HWS06070
	STO	2	VT200	
	LD	L	EPK	HWS06080
	STO		MP4	HWS06090
	M	L	FREQE	HWS06100
ME1	DC		***	HWS06110
ME2	DC		***	HWS06120
ME3	DC		***	HWS06130
ME4	DC		***	HWS06140
MP1	DC		***	HWS06150
MP2	DC		***	HWS06160
MP3	DC		***	HWS06170
MP4	DC		***	HWS06180
KEFN1	DC		***	HWS06190
KEFN2	DC		***	HWS06200
KEFN3	DC		***	HWS06210
KEFN4	DC		***	HWS06220
WEPN	DC		***	HWS06230
P3PL	DC		***	HWS06240
P5PL	DC		***	HWS06250
KPFN1	DC		***	HWS06260
KPFN2	DC		***	HWS06270
KPFN3	DC		***	HWS06280
KPFN4	DC		***	HWS06290
WPFN	DC		***	HWS06300
PT3NB	DC		***	HWS06310
PTSNB	DC		***	HWS06320
WTFN	DC		***	HWS06330
ASN	DC		***	HWS06340
WFMNN	DC		***	
TBBN	DC		***	HWS06370
SUMEF	DC		***	HWS06380
* CALCULATE FUEL REQ. (S) FOR SPEED AND PRESSURE				
* FREQE EQU				
	LD	L	KEFN1	HWS06450
	M		ME1	HWS06460
	SLY		7	HWS06470
	STO	2	VT201	
	STO		SUMFF	HWS06490
	LD	L	KEFN2	HWS06500
	M		ME2	HWS06510
	D		*100	
	SRT		7	
	STO	2	VT202	
	A		SUMEF	HWS06510
	STO		SUMFF	HWS06520
	LD	L	KEFN3	HWS06530
	M		ME3	HWS06540
	D		*100	
	SRT		9	

Table B-8. Equilibrium - Pressure Subprogram (Continued)

STO	2	VT071	HWS07260
LD	L	SUMPF	HWS07270
STO	2	VT072	HWS07280
LD	L	SUMTF	HWS07290
STO	2	VT073	HWS07300
CHP	2	VT072	HWS07310
LD	2	VT072	HWS07320
NOP			HWS07330
STO		WFM0D	HWS07340
CHP	2	VT071	HWS07350
LD	2	VT071	HWS07360
NOP			HWS07370
STO	2	VT180	HWS07380
BN		MINFL	
M	L	=13	HWS06990
SLT		16	HWS07000
STO	2	VT180	
LD	2	VT180	
S	L	WFMNN	
BNN		MINSB	
MINFL	LD	L	WFMNN
STO	2	VT180	
MINSB	EQU	*	
LD	2	VT180	HWS07390
SRT		16	
D		=13	
S	2	VT071	HWS07400
BNZ		MIKE1	HWS07410
LD		ONE	HWS07420
STO	2	VT074	HWS07430
B	L	RQAIB	HWS07440
MIKE1	LD	2	VT180
SRT		16	HWS07450
D		=13	
S	2	VT072	HWS07460
BNZ		MIKE2	HWS07470
LD		TWO	HWS07480
STO	2	VT074	HWS07490
B	L	RQAIB	HWS07500
MIKE2	LD	THREE	HWS07510
STO	2	VT074	HWS07520
B	L	RQAIB	HWS07530
KLAGD	DC	49	
K1NUM	DC	31	
K2NUM	DC	9	
YNM1	DC	***	
UNM1	DC	***	
TEMP	HSS	E	0
	DC		0
	DC		0
SWLAG	DC	***	
*			
* FINAL FUEL REQUEST IS LAGGED HERE			
*			
RQAIB	EQU	*	HWS07540
LD	L	SWLAG	
BNZ		FILT	
LD	L	=123	

Table B-3. Equilibrium - Pressure Subprogram (Continued)

```

      STP L SWLAB
      LD  2 VT180
      STS L YNM1
      STS L UNM1
      S   L D0N0Z
FILT  LD  L UNM1
      M   L K2NUM
      STD L TEMF
      LD  2 VT180
      STS L UNM1
      M   L K2NUM
      AD  L TEMF
      STD L TEMF
      LD  L YNM1
      M   L K1NUM
      AD  L TEMF
      D   L KLAGD
      STS L YNM1
      STS 2 VT180
  
```

• EXHAUST NOZZLE REQUEST CALCULATION

```

D0N0Z EQU *
      LD  2 VT074
      S   L ONE
      BNZ GT10
      LD  2 VT128
      STS L NAB
      S   L CALAB
GT10  LD  2 VT157
      STS L NAB
CALAB LD  L NAB
      S   L *14025
      BNN GT11
      LD  2 VT034
      STS 2 VT081
      S   L CONT
GT11  S   L *2475
      BN  GT12
      LD  2 VT035
      STS 2 VT081
      S   L CONT
GT12  LD  2 VT034
      S   2 VT035
      STS L AN0ZN
      LD  L *16500
      S   L NAB
      M   L AN0ZN
      D   L *2475
      A   2 VT035
      STS 2 VT081
      S   L CONT
      LORG
NAB   DC   ***
AN0ZN DC   ***
CONT  EQU *
XN1   LDX L1 ***
      HSC I HAFCT
  
```

HW807700
 HW807710
 HW807720

Table B-8. Equilibrium - Pressure Subprogram (Continued)

VT071 EQU	+71	HWS07730
VT072 EQU	+72	HWS07740
VT073 EQU	+73	HWS07750
VT074 EQU	+74	HWS07760
VT081 EQU	+81	HWS07770
VT082 EQU	+82	HWS07780
VT083 EQU	+83	HWS07790
VT157 EQU	+30	HWS07800
VT180 EQU	+53	HWS07810
VT128 EQU	+1	HWS07820
VT102 EQU	+102	HWS07830
VT108 EQU	+108	HWS07840
VT097 EQU	+97	HWS07850
VT036 EQU	+36	HWS07860
VT037 EQU	+37	HWS07870
VT038 EQU	+38	HWS07880
VT039 EQU	+39	HWS07890
VT162 EQU	+3E	HWS07900
VT163 EQU	+36	HWS07910
VT164 EQU	+37	HWS07920
VT163 EQU	+3A	HWS07930
VT166 EQU	+39	HWS07940
VT167 EQU	+40	HWS07950
VT168 EQU	+41	HWS07960
VT169 EQU	+42	HWS07970
VT170 EQU	+43	HWS07980
VT171 EQU	+44	HWS07990
VT172 EQU	+45	HWS08000
VT173 EQU	+46	HWS08010
VT174 EQU	+47	HWS08020
VT175 EQU	+48	HWS08030
VT176 EQU	+49	HWS08040
VT206 EQU	+79	
VT207 EQU	+80	
VT196 EQU	+69	
VT197 EQU	+70	
VT198 EQU	+71	
VT199 EQU	+72	
VT200 EQU	+73	
VT201 EQU	+74	
VT202 EQU	+75	
VT203 EQU	+76	
VT204 EQU	+77	
VT205 EQU	+78	
VT012 EQU	+12	
VT013 EQU	+13	
VT014 EQU	+14	
VT015 EQU	+15	
VT016 EQU	+16	
VT017 EQU	+17	
VT018 EQU	+18	
VT019 EQU	+19	
VT020 EQU	+20	
VT021 EQU	+21	
VT022 EQU	+22	
VT023 EQU	+23	
VT040 EQU	+40	
VT041 EQU	+41	

Table B-8. Equilibrium - Pressure Subprogram (Concluded)

VT042 EQU	042
VT043 EQU	043
VT044 EQU	044
VT045 EQU	045
VT046 EQU	046
VT047 EQU	047
VT048 EQU	048
VT049 EQU	049
VT050 EQU	050
VT061 EQU	061
VT062 EQU	062
VT063 EQU	063
VT064 EQU	064
VT065 EQU	065
VT066 EQU	066
VT067 EQU	067
VT068 EQU	068
VT069 EQU	069
VT034 EQU	034
VT035 EQU	035
END	

Table B-9. Standard Trim Adjustments in Bounds Program
(Equilibrium Pressure)

```

// JOB          DISK 12 JUN 74 08.612 HRS
// DMP 12 JUN 74 08.612 HRS
*DELETE          GTECT
DMP FUNCTION COMPLETED
// ASM GTECT 12 JUN 74 08.613 HRS
*OVERFLOW SECTORS ,,,9
*LIST
*XXEF
*ONE WORD INTEGERS
*COMMON IDUMY(127),IVT00,JDUMY(127),IMT0,MEAST(64),IASCM(2)
0000 078C50E3      1      ENT      GTECT
0000 0 0000        2      GTECT DC      **
0001 01 6D000512   3      STX L1 XR1+1
0003 01 6E000514   4      STX L2 XR2+1
0005 01 6F000516   5      STX L3 XR3+1
                                6      *
0007 03 6700FEC0   7      LDX L3 MEAST-63
0009 03 6600FF80   8      LDX L2 IVT00
000B 00 65000000   9      LDX L1 0
000D 0 C03F        10     LD      =0
                                11     *
000E 01 4C00C0F7   12     B      L      START
0010                                13     KSTAL EQU      *      RESET ALL DIGITAL ADJUST
0010 0 C03E        14     LD      ST001
0011 0 D2FF        15     STO 2 VT001
0012 0 C03D        16     LD      ST002
0013 0 D2FE        17     STO 2 VT002
0014 0 C03C        18     LD      ST003
0015 0 D2FD        19     STO 2 VT003
0016 0 C03B        20     LD      ST004
0017 0 D2FC        21     STO 2 VT004
0018 0 C03A        22     LD      ST005
0019 0 D2FB        23     STO 2 VT005
001A 0 C039        24     LD      ST006
001B 0 D2FA        25     STO 2 VT006
001C 0 C038        26     LD      ST007
001D 0 D2F9        27     STO 2 VT007
001E 0 C037        28     LD      ST008
001F 0 D2F8        29     STO 2 VT008
0020 0 C036        30     LD      ST009
0021 0 D2F7        31     STO 2 VT009
0022 0 C035        32     LD      ST010
0023 0 D2F6        33     STO 2 VT010
0024 0 C034        34     LD      ST011
0025 0 D2F5        35     STO 2 VT011
0026 0 C033        36     LD      ST012
0027 0 D2F4        37     STO 2 VT012
0028 0 C032        38     LD      ST013
0029 0 D2F3        39     STO 2 VT013
002A 0 C031        40     LD      ST014
002B 0 D2F2        41     STO 2 VT014
002C 0 C030        42     LD      ST015
002D 0 D2F1        43     STO 2 VT015
002E 0 C02F        44     LD      ST016
002F 0 D2F0        45     STO 2 VT016
0030 0 C02E        46     LD      ST017
0031 0 D2EF        47     STO 2 VT017
0032 0 C02D        48     LD      ST018
0033 0 D2EE        49     STO 2 VT018
0034 0 C02C        50     LD      ST019

```

Table B-9. Standard Trim Adjustments in Bounds Program
(Equilibrium Pressure) (Continued)

12 JUN 74 PAGE 002

0035	0	D2ED	51	STO	2	VT019		HWE00570
0036	0	C02B	52	LC		ST020		HWE00580
0037	0	D2EC	53	STO	2	VT020		HWE00590
0038	0	C02A	54	LD		ST021		HWE00600
0039	0	D2EB	55	STO	2	VT021		HWF00610
003A	0	C029	56	LD		ST022		HWE00620
003B	0	D2EA	57	STO	2	VT022		HWE00630
003C	0	C02B	58	LD		ST023		HWE00640
003D	0	D2E9	59	STO	2	VT023		HWE00650
003E	0	C027	60	LD		ST024		HWE00660
003F	0	D2E8	61	STO	2	VT024		HWE00670
0040	0	C026	62	LD		ST025		HWE00680
0041	0	D2E7	63	STO	2	VT025		HWE00690
0042	0	C025	64	LD		ST026		HWE00700
0043	0	D2E6	65	STO	2	VT026		HWE00710
0044	0	C024	66	LD		ST027		HWE00720
0045	0	D2E5	67	STO	2	VT027		HWE00730
0046	0	C023	68	LD		ST028		HWE00740
0047	0	D2E4	69	STO	2	VT028		HWE00750
0048	0	C022	70	LD		ST029		HWE00760
0049	0	D2E3	71	STO	2	VT029		HWE00770
004A	0	C021	72	LD		ST030		HWE00780
004B	0	D2E2	73	STO	2	VT030		HWE00790
004C	0	7048	74	B		STTVT		HWE00800
			75			LORG		HWE00810
004D	0	0000	76	+		DC	0	
			77	*				
004E	0	0000	78	ST000	DC	0	SPEED CONTROL FIG10-3&4	HWE00820
004F	0	0000	79	ST001	DC	0	IDLE SPEED TRIM	HWE00840
0050	0	0000	80	ST002	DC	0	MAX SPEED TRIM	HWE00850
0051	0	4E20	91	ST003	DC	20000		HWE00860
0052	0	0000	82	ST004	DC	0	BRANCH COMMAND 64+	HWE00870
0053	0	1000	83	ST005	DC	4096	N INTEGRATION INC	HWE00880
0054	0	1388	84	ST006	DC	5000	N INT PRESS GAIN	HWE00890
0055	0	F00C	85	ST007	DC	-4096	N INT DECREASE	HWE00900
0056	0	EC78	86	ST008	DC	-5000	N INT DEC PRESS GAIN	HWE00910
			87	*				HWE00920
			88	*		FIG10-5	PROP. TEMPERATURE CONTROL	HWE00930
0057	0	0000	89	ST009	DC	0	SPEED CONTROL SELECTION	HWE00940
0058	0	32C8	90	ST010	DC	11000		HWE00950
0059	0	0000	91	ST011	DC	0	ZERO FLOW ADJUST	HWE00960
			92	*				
			93	*			HONEYWELL ST VALUES	
005A	0	F290	94	ST012	DC	-3440	N GAIN (50 ,E)	
005B	0	0177	95	ST013	DC	519	WF (50 ,E)	
005C	0	0000	96	ST014	DC	2200	PT3 (50 ,E)	
005D	0	0180	97	ST015	DC	432	EN GAIN (50 ,E)	
005E	0	F820	98	ST016	DC	-2016	N GAIN (70 ,E)	
005F	0	0315	99	ST017	DC	693	WF (70 ,E)	
0060	0	0000	100	ST018	DC	3550	PT3 (70 ,E)	
0061	0	0190	101	ST019	DC	400	EN GAIN (70 ,E)	
0062	0	F860	102	ST020	DC	-1952	N GAIN (85 ,E)	
0063	0	04A6	103	ST021	DC	934	WF (85 ,E)	
0064	0	0000	104	ST022	DC	5500	PT3 (85 ,E)	
0065	0	0380	105	ST023	DC	896	EN GAIN (85 ,E)	
			106	*				
			107	*			END HONEYWELL ST VALUES	

Table B-9. Standard Trim Adjustments in Bounds Program
(Equilibrium Pressure) (Concluded)

12 JUN 74 PAGE 003

		108	*				
0066	0	1770	109	ST024 DC	6000	ZERU N RATIOS INTERCEPT	HWE01140
0067	0	A240	110	ST025 DC	-24000	BACK SLOPE SPEED BREAK PT	HWE01150
0068	0	4000	111	ST026 DC	16384		
		112	*				HWE01170
		113	*			FIGURE10-8 RATIOS INTEGRATION	HWE01180
0069	0	7FF8	114	ST027 DC	32760		
006A	0	0000	115	ST028 DC	0		
006B	0	0000	116	ST029 DC	0	MINIMUM RATIOS SLOPE	HWE01210
006C	0	5014	117	ST030 DC	20500	MINIMUM RATIOS LEVEL	HWE01220
006D	0	0000	118	ST031 DC	0		HWE01230
006E	0	7FF8	119	ST032 DC	32760	VALVE MAXIMUM POSITION	HWE01240
006F	0	0000	120	ST033 DC	0	VALVE MINIMUM POSITION	HWE01250
0070	0	25A8	121	ST034 DC	9640		
0071	0	0A5A	122	ST035 DC	2650		
		123	*				
		124	*			HONEYWELL ST VALUES	
		125	*				
0072	0	0640	126	ST036 DC	1600		
0073	0	0640	127	ST037 DC	1600		
0074	0	000A	128	ST038 DC	10		
0075	0	0010	129	ST039 DC	16		
0076	0	F0A0	130	ST040 DC	-3936	N GAIN (100,E)	
0077	0	0960	131	ST041 DC	1648	WF (100,E)	
0078	0	0000	132	ST042 DC	8000	PT3 (100,E)	
0079	0	0930	133	ST043 DC	2352	EN GAIN (100,E)	
007A	0	0C11	134	ST044 DC	3089	FUG GAIN (50 ,P)	
007B	0	23B0	135	ST045 DC	9136	PT5 GAIN (50 ,P)	
007C	0	0610	136	ST046 DC	-10736	PT3 GAIN (50 ,P)	
007D	0	030B	137	ST047 DC	651	EP GAIN (50 ,P)	
007E	0	2400	138	ST048 DC	9216	FUG GAIN (70 ,P)	
007F	0	F5E0	139	ST049 DC	-2592	PT5 GAIN (70 ,P)	
0080	0	F200	140	ST050 DC	-3584	PT3 GAIN (70 ,P)	
		141	*				
		142	*			END HONEYWELL ST VALUES	
		143	*				
		144	*				
		145	*			FIGURE10-12 IGV & BLEED CONTR	HWE01460
0081	0	0000	146	ST051 DC	0	LOW N TRIM OF IGV	HWE01480
0082	0	3E80	147	ST052 DC	16000	HIGH N TRIM OF IGV	HWE01490
0083	0	0000	148	ST053 DC	0	LOW N TRIM OF BLEEDS	HWE01500
0084	0	3E80	149	ST054 DC	16000	HIGH N TRIM OF BLEEDS	HWE01510
		150	*				HWE01520
		151	*			FIGURE10-14 NOZZLE CONTROL	HWE01530
0085	0	105E	152	ST055 DC	4190	NOZZLE FLAT	BEN01530
0086	0	40D8	153	ST056 DC	16600	T5 REQUEST	HWE01550
0087	0	4000	154	ST057 DC	16384	T5 CONTROL GAIN	HWE01560
0088	0	0000	155	ST058 DC	0		HWE01570
0089	0	0000	156	ST059 DC	0		HWE01580
008A	0	0000	157	ST060 DC	0		HWE01590
008B	0	06CC	158	ST061 DC	1000	EP GAIN (70 ,P)	
008C	0	0380	159	ST062 DC	944	FUG GAIN (85 ,P)	
008D	0	0B90	160	ST063 DC	2960	PT5 GAIN (85 ,P)	
008E	0	EFEO	161	ST064 DC	-4128	PT3 GAIN (85 ,P)	
008F	0	0A0D	162	ST065 DC	2000	EP GAIN (85 ,P)	
0090	0	16C0	163	ST066 DC	5824	FUG GAIN (100,P)	
0091	0	09C0	164	ST067 DC	2496	PT5 GAIN (100,P)	
0092	0	F4F0	165	ST068 DC	-2832	PT3 GAIN (100,P)	
0093	0	0D96	166	ST069 DC	2400	EP GAIN (100,P)	
0094	0	0000	167	ST070 DC	0		

Table B-10. Glossary for Equilibrium-Temperature Control

VT Number	Transferred to (Program Label)	Description	Standard Value (Defined in the Bendix Program)
009	---	Logic switch: If VT009 > 123 the Honeywell controller is in; otherwise not	0
012	KEF11	Speed control gain associated with $(N-N_{pla})$ at 8250 rpm	(-215 x 16)
013	WEF1	Open-loop fuel-speed control at 8250 rpm	519 lb/hr
014	P3P1	Open-loop PT3-speed control at 8250 rpm	2200 psi x 100
015	KEF14	Speed control gain associated with EN at 825 rpm	(27) x 16
016	KEF21	Speed control gain associated with $(N-N_{pla})$ at 11,550 rpm	(-126) x 16
017	WEF2	Open-loop fuel-speed control at 11,550 rpm	693 lb/hr
018	P3P2	Open-loop PT3 - speed control at 11,550 rpm	3550 psi x 100
019	KEF24	Speed control gain associated with EN at 11,550 rpm	(25) x 16
020	KEF31	Speed control gain associated with $(N-N_{pla})$ at 14,025 rpm	(-122) x 16
021	WEF3	Open-loop fuel-speed control at 14,025 rpm	934 lb/hr
022	P3P3	Open-loop PT3-speed control at 14,025 rpm	5500 psi x 100
023	KEF34	Speed control gain associated with EN at 14,025 rpm	(56) x 16
026	---	If this number is made large, Bendix bound on fuel will not be in effect	2^{14}
028	---	Logical switch: if VTO28 = 64 Honeywell nozzle is used; otherwise not	0

Table B-16. Glossary for Equilibrium-Temperature Control (Continued)

VT Number	Transferred to (Program Label)	Description	Standard Value (Defined in the Bendix Program)
034	---	Exhaust request: open	9640
035	---	Exhaust request: closed	2650
036	ENK, ENKL	Initial value of integral speed and limits valve	1600
038	ETK, ETKL	Initial value of integral temperature and limiting valve	25,600
039	---	Logical switch: VT089 = 16 initialize everything. VT039 = 64 initializes EN and ET only	16
040	KEF41	Speed control gain associated with $(N-N_{pla})$ at 16,500 rpm	$(-246) \times 16$
041	WEF4	Open-loop fuel-speed control at 16,500 rpm	1648 lb/hr
042	P3P4	Open-loop PT3-speed control at 16,500 rpm	8000 psi x 100
043	KEF44	Speed control gain associated with EN at 16,500 rpm	$(147) \times 16$
044	KTF11	Temperature control gain - (PT5-PT5b) - at 8250 rpm	$(557) \times 16$
045	KTF12	Temperature control gain - (PT3-PT3b) - at 8250 rpm	$(-736) \times 16$
046	KTF13	Temperature control gain - (T4WF - T4b) at 8250 rpm	$(-105) \times 16$
047	KTF14	Temperature control gain - ET at 8250 rpm	$(52) \times 16$
048	WTF1	Open-loop fuel-temperature control - at 8250 rpm	651 lb/hr
049	KTF21	Temperature control gain - (PT5-PT5b) at 11,550 rpm	$(938) \times 16$
050	KTF22	Temperature control gain - (PT3-PT3b) at 11,550 rpm	24585

Table B-10. Glossary for Equilibrium-Temperature Control (Continued)

VT Number	Transferred to (Program Label)	Description	Standard Value (Defined in the Bendix Program)
061	KTF23	Temperature control gain - (T4WF-T4b) at 11,550 rpm	(-131) x 16
062	KTF24	Temperature control gain - ET at 11,550 rpm	(65) x 16
063	WTF2	Open-loop fuel - temperature control at 11,550 rpm	1000 lb/hr
064	KTF31	Temperature control gain - (PT5-PT5b) at 14,025 rpm	(-343) x 16
065	KTF32	Temperature control gain - (PT3-PT3b) at 14,025 rpm	4235
066	KTF33	Temperature control gain - (T4WF - T4b) at 14,025 rpm	(-87) x 16
067	KTF34	Temperature control gain - ET at 14,025 rpm	(74) x 16
068	WTF3	Open-loop fuel - temperature control at 14,025 rpm	2000 lb/hr
069	KTF41	Temperature control gain - (PT5-PT5b) at 16,500 rpm	6127
071	---	Fuel request - speed control 1 count = 4 lb/hr	0
073	---	Fuel request - temperature control, 1 count = 4 lb/hr	0
074	---	Mode number 3276 = speed control 6552 = temperature control 9828 = minimum fuel control	0
075	KTF42	Temperature control gain - (PT3-PT3b) at 16,500 rpm	-8338
076	KTF43	Temperature control gain - (T4WF-T4b)	(-4) x 16
077	KTF44	---	(108) x 16

Table B-10. Glossary for Equilibrium-Temperature Control (Concluded)

VT Number	Transferred to (Program Label)	Description	Standard Value (Defined in the Bendix Program)
078	WTF4	---	2400 lb/hr
081	---	Nozzle fuel request	---
082	TB1	T40 at 8250 rpm	10,200 °F x 10
083	TB2	T40 at 11,550 rpm	9000 °F x 10
084	TB3	T40 at 14,025 rpm	10,500 °F x 10
085	TB4	T40 at 16,500 rpm	11,600 °F x 10
086	P5T1	PT50 at 8250 rpm	1480 psi x 100
087	P5T2	PT50 at 11,550 rpm	1640 psi x 100
088	P5T3	PT50 at 14,025 rpm	2050 psi x 100
089	P5T4	PT50 at 16,500 rpm	2550 psi x 100
180	---	Fuel request 3.25 counts = 1 lb/hr	---

Table B-11. Honeywell Control Program

```
// JOB          VDISK   17 JUL 74 15.584 HRS
// DMP          17 JUL 74 15.585 HRS
*DELETE        HWECT
DMP FUNCTION COMPLETED
// ASM          17 JUL 74 15.586 HRS
*OVERFLOW SECTORS ,,,9
*LIST
*XREF
```

*ONEWORD INTEGERS

*COMMON IDUMY(127),IVT00,JDUMY(127),IBTO,NEAST(64),IASCW(2)

0000	089850E3	1	ENT	HWECT		HWS00070
0000	0 0000	2	HWECT	DC	**	HWS00080
0001	01 6000050A	3		STX	L1 XR1+1	HWS00090
		4	*			HWS00100
0003	00 65000000	5		LDX	L1 0	HWS00110
0005	01 4C000007	6		B	L TESTN	HWS00120
		7	*			HWS00160
		8	*INTERVAL DETERMINATION			HWS00170
		9	*			HWS00180
0007		10	TESTN	EQU	*	HWS00190
0007	0 C2D9	11		LD	2 VT039	
0008	01 940000BC	12		S	L =16	
000A	01 4C2000AD	13		BNZ	MICK	
000C	0 D2D9	14		STO	2 VT039	
000D	0 C2F4	15		LD	2 VT012	
000E	0 1884	16		SRT	4	
000F	01 D40000FC	17		STO	L KEF11	
0011	0 C2F3	18		LD	2 VT013	
0012	01 D4000100	19		STO	L WEF1	
0014	0 C2F2	20		LD	2 VT014	
0015	01 D4000108	21		STO	L P3T1	
0017	0 C2F1	22		LD	2 VT015	
0018	0 1884	23		SRT	4	
0019	01 D40000FF	24		STO	L KEF14	
001B	0 C2E9	25		LD	2 VT016	
001C	0 1884	26		SRT	4	
001D	01 D4000135	27		STO	L KEF21	
001F	0 C2EF	28		LD	2 VT017	
0020	01 D4000139	29		STO	L WEF2	
0022	0 C2EE	30		LD	2 VT018	
0023	01 D4000141	31		STO	L P3T2	
0025	0 C2ED	32		LD	2 VT019	
0026	0 1884	33		SRT	4	
0027	01 D4000138	34		STO	L KEF24	
0029	0 C2EC	35		LD	2 VT020	
002A	0 1884	36		SRT	4	
002B	01 D400016B	37		STO	L KEF31	
002D	0 C2EB	38		LD	2 VT021	
002E	01 D400016F	39		STO	L WEF3	
0030	0 C2EA	40		LD	2 VT022	
0031	01 D4000177	41		STO	L P3T3	
0033	0 C2E9	42		LD	2 VT023	
0034	0 1884	43		SRT	4	
0035	01 D400016E	44		STO	L KEF34	
0037	0 C2D8	45		LD	2 VT040	
0038	0 1884	46		SRT	4	
0039	01 D40001A1	47		STO	L KEF41	
003B	0 C2D7	48		LD	2 VT041	
003C	01 D40001A5	49		STO	L WEF4	
003E	0 C2D6	50		LD	2 VT042	

Table B-11. Honeywell Control Program (Continued)

17 JUL 74 PAGE 002

003F	01	D40001AD	51	STD	L	P3T4
0041	0	C2D5	52	LD	2	VT043
0042	0	1884	53	SRT		4
0043	01	D40001A4	54	STD	L	K5F44
0045	0	C2D4	55	LD	2	VT044
0046	0	1884	56	SRT		4
0047	01	D4000103	57	STD	L	KTF11
0049	0	C2D3	58	LD	2	VT045
004A	0	1884	59	SRT		4
004B	01	D4000104	60	STD	L	KTF12
004D	0	C2D2	61	LD	2	VT046
004E	0	1884	62	SRT		4
004F	01	D4000105	63	STD	L	KTF13
0051	0	C2D1	64	LD	2	VT047
0052	0	1884	65	SRT		4
0053	01	D4000106	66	STD	L	KTF14
0055	0	C2D0	67	LD	2	VT048
0056	01	D4000107	68	STD	L	KTF1
0058	0	C2C	69	LD	2	VT049
0059	0	1884	70	SRT		4
005A	01	D400013C	71	STD	L	KTF21
005C	0	C2CE	72	LD	2	VT050
005D	01	D400013D	73	STD	L	KTF22
005F	0	C2C3	74	LD	2	VT061
0060	0	1884	75	SRT		4
0061	01	D400013E	76	STD	L	KTF23
0063	0	C2C2	77	LD	2	VT062
0064	0	1884	78	SRT		4
0065	01	D400013F	79	STD	L	KTF24
0067	0	C2C1	80	LD	2	VT063
0068	01	D4000140	81	STD	L	KTF2
006A	0	C2C0	82	LD	2	VT064
006B	0	1884	83	SRT		4
006C	01	D4000172	84	STD	L	KTF31
006E	0	C2BF	85	LD	2	VT065
006F	01	D4000173	86	STD	L	KTF32
0071	0	C2BE	87	LD	2	VT066
0072	0	1884	88	SRT		4
0073	01	D4000174	89	STD	L	KTF33
0075	0	C2BD	90	LD	2	VT067
0076	0	1884	91	SRT		4
0077	01	D4000175	92	STD	L	KTF34
0079	0	C2BC	93	LD	2	VT068
007A	01	D4000176	94	STD	L	KTF3
007C	0	C2BB	95	LD	2	VT069
007D	01	D40001A8	96	STD	L	KTF41
007F	0	C2B5	97	LD	2	VT075
0080	01	D40001A9	98	STD	L	KTF42
0082	0	C2B7	99	LD	2	VT076
0083	0	1884	100	SRT		4
0084	01	D40001AA	101	STD	L	KTF43
0086	0	C2B5	102	LD	2	VT077
0087	0	1884	103	SRT		4
0088	01	D40001AB	104	STD	L	KTF44
008A	0	C2B2	105	LD	2	VT078
008H	01	D40001AC	106	STD	L	KTF4
008D	01	D40001AD	107	LD	L	=0

Table B-11. Honeywell Control Program (Continued)

17 JUL 74 PAGE 003

008F	01	D40004AC	108		STO	L	SMLAG		
0091	01	D4000287	109		STO	L	TIME		
0093	01	D40001FD	110		STO	L	ISW		
0095	0	C2AE	111		LD	2	VT082		
0096	01	D400010A	112		STO	L	TB1		
0098	0	C2AD	113		LD	2	VT083		
0099	01	D4000143	114		STO	L	TB2		
009B	0	C2AC	115		LD	2	VT084		
009C	01	D4000179	116		STO	L	TB3		
009E	0	C2AB	117		LD	2	VT085		
009F	01	D40001AF	118		STO	L	TB4		
00A1	0	C2AA	119		LD	2	VT086		
00A2	01	D4000109	120		STO	L	P5T1		
00A4	0	C2A9	121		LD	2	VT087		
00A5	01	D4000142	122		STO	L	P5T2		
00A7	0	C2A8	123		LD	2	VT088		
00A8	01	D4000178	124		STO	L	P5T3		
00AA	0	C2A7	125		LD	2	VT089		
00AB	01	D40001AE	126		STO	L	P5T4		
00AD			127	MICK	EQU		*		
00AD	0	C21E	128		LD	2	VT157		HWS00 200
00AE	0	900F	129		S		=8250		HWS00 210
00AF	01	4C3000C1	130		BP		TMAX		HWS00 220
00B1	0	C00D	131		LD		=1		HWS00 230
00B2	0	D008	132		STO		NIN		HWS00 240
00B3	0	C00C	133		LD		=128		
00B4	0	D004	134		STO		C1		HWS00 260
00B5	0	C007	135		LD		=0		HWS00 270
00B6	0	D003	136		STO		C2		HWS00 280
00B7	01	4C000114	137		B	L	IN1F		HWS00 290
00B9	0	0000	138	C1	DC		---		HWS00 130
00BA	0	0000	139	C2	DC		---		HWS00 140
00BB	0	0000	140	NIN	DC		---		HWS00 150
			141		LORG				
00BC	0	0010	142	+	DC		16		
00BD	0	0000	143	+	DC		0		
00BE	0	203A	144	+	DC		8250		
00BF	0	0001	145	+	DC		1		
00C0	0	0080	146	+	DC		128		
00C1	0	C033	147	TMAX	LD		=16500		HWS00 300
00C2	0	921E	148		S	2	VT157		HWS00 310
00C3	01	4C3000CD	149		BP		TIN1		HWS00 320
00C5	0	C030	150		LD		=3		HWS00 330
00C6	0	D0F4	151		STO		NIN		HWS00 340
00C7	0	C0F5	152		LD		=0		HWS00 350
00C8	0	D0F0	153		STO		C1		HWS00 360
00C9	0	C0F6	154		LD		=128		
00CA	0	D0EF	155		STO		C2		HWS00 380
00CB	01	4C000180	156		B	L	IN3F		HWS00 390
00CD	0	C029	157	TIN1	LD		=11550		HWS00 400
00CE	0	921E	158		S	2	VT157		HWS00 410
00CF	01	4C2800DB	159		BN		TIN2		HWS00 420
00D1	0	1889	160		SRT		9		
00D2	0	AB25	161		D		=3300		HWS00 440
00D3	0	D0E5	162		STO		C1		HWS00 450
00D4	0	C0E8	163		LD		=128		
00D5	0	90E3	164		S		C1		HWS00 470

Table B-11. Honeywell Control Program (Continued)

17 JUL 74 PAGE 004

00D6	0	DOE3	165		STO	C2			HWS00480
00D7	0	COE7	166		LD	=1			HWS00490
00D8	0	DOE2	167		STO	NIN			HWS00500
00D9	01	4C000114	168		B	L	IN1F		HWS00510
0G0B	0	C01D	169	TIN2	LD	=14025			HWS00520
00DC	0	921E	170		S	2	VT157		HWS00530
00DD	01	4C2800E9	171		BN	TIN3			HWS00540
00DF	0	1889	172		SRT	9			
00E0	0	A819	173		D	=2475			HWS00560
00E1	0	D0D7	174		STO	C1			HWS00570
00E2	0	C0DD	175		LD	=128			
00E3	0	90D5	176		S	C1			HWS00590
00E4	0	D0D5	177		STO	C2			HWS00600
00E5	0	C015	178		LD	=2			HWS00610
00E6	0	D0D4	179		STO	NIN			HWS00620
00E7	01	4C00014A	180		B	L	IN2F		HWS00630
00E9	0	C00B	181	TIN3	LD	=16500			HWS00640
00EA	0	921E	182		S	2	VT157		HWS00650
00EB	0	1889	183		SRT	9			
00EC	0	A80D	184		D	=2475			HWS00670
00ED	0	D0CB	185		STO	C1			HWS00680
00EE	0	C0D1	186		LD	=128			
00EF	0	90C9	187		S	C1			HWS00700
00F0	0	D0C9	188		STO	C2			HWS00710
00F1	0	C004	189		LD	=3			HWS00720
00F2	0	D0C8	190		STO	NIN			HWS00730
00F3	01	4C000180	191		B	L	IN3F		HWS00740
			192		LORG				HWS00750
00F5	0	4074	193	+	DC	16500			
00F6	0	0003	194	+	DC	3			
00F7	0	2D1E	195	+	DC	11550			
00F8	0	0CE4	196	+	DC	3300			
00F9	0	36C9	197	+	DC	14025			
00FA	0	09AB	198	+	DC	2475			
00FB	0	0002	199	+	DC	2			
			200	*					
00FC	0	0000	201		REF11	DC	**		HWS00760
00FD	0	0000	202		KEF12	DC	0		
00FE	0	0000	203		KEF13	DC	0		
00FF	0	0000	204		KEF14	DC	**		
0100	0	0000	205		WEF1	DC	**		
0101	0	0A91	206		P3E1	DC	2705		
0102	0	0661	207		P5E1	DC	1633		
			208	*					
0103	0	0000	209		KTF11	DC	**		HWS00920
0104	0	0000	210		KTF12	DC	**		
0105	0	0000	211		KTF13	DC	**		
0106	0	00J0	212		KTF14	DC	**		
0107	0	0000	213		HTF1	DC	**		
0108	0	0000	214		P3T1	DC	**		
0109	0	0000	215		P5T1	DC	**		
010A	0	0000	216		TB1	DC	**		
010B	0	028A	217		WFMN1	DC	650		
010C	0	0001	218		BUMP1	DC	1		HWS00980
010D	0	0000	219		SETX1	DC	0		HWS00990
010E	0	0010	220		NGFT	DC	16		
010F	0	0000	221		C11	DC	**		HWS01010

Table B-11. Honeywell Control Program (Continued)

17 JUL 74 PAGE 005

0110	0	0000	222	C21	DC	*--*		HWS01020
0111	0	0000	223	TST1	DC	*--*		HWS01040
0112	0	0000	224	SUM1	BSS	E	0	
0112	0	0000	225		DC		0	
0113	0	0000	226		DC		0	
0114			227	IN1F	EQU	*		HWS01050
0114	01	C40000B9	228		LD	L	C1	HWS01060
0116	0	D0F8	229		STO		C11	HWS01070
0117	01	C40000BA	230		LD	L	C2	HWS01080
0119	0	D0F6	231		STO		C21	HWS01090
011A	01	C400010D	232		LD	L	SETX1	HWS01100
011C	0	D0F4	233		STO		TST1	HWS01110
011D	01	65800111	234	LUP1	LDX	I1	TST1	HWS01120
011F	01	C50000FC	235		LD	L1	KEF11	HWS01130
0121	0	A0ED	236		M		C11	HWS01140
0122	0	D8EF	237		STD		SUM1	
0123	01	C5000135	238		LD	L1	KEF21	HWS01170
0125	0	A0EA	239		M		C21	HWS01180
0126	0	88E8	240		AD		SUM1	
0127	0	1887	241		SRT		7	
0128	0	1090	242		SLT		16	
0129	01	D5000406	243		STO	L1	KEFN1	HWS01220
012B	0	C0E5	244		LD		TST1	HWS01230
012C	01	8400010C	245		A	L	BUMP1	HWS01240
012E	0	D0E2	246		STD		TST1	HWS01250
012F	01	9400010E	247		S	L	NGFT	HWS01260
0131	01	4C28011D	248		BN		LUP1	HWS01270
0133	01	4C0001B1	249		B	L	FUELM	HWS01280
			250	*			EQUILIBRIUM FUEL FLOW 70 GAINS	HWS01290
0135	0	0000	251	KEF21	DC	*--*		
0136	0	0000	252	KEF22	DC		0	
0137	0	0000	253	KEF23	DC		0	
0138	0	0000	254	KEF24	DC	*--*		
0139	0	0000	255	WEF2	DC	*--*		
013A	0	1109	256	P3E2	DC		4361	
013B	0	0765	257	P5E2	DC		1893	
			258	*			TEMPERATURE FUEL FLOW 70 GAINS	HWS01450
013C	0	0000	259	KTF21	DC	*--*		
013D	0	0000	260	KTF22	DC	*--*		
013E	0	0000	261	KTF23	DC	*--*		
013F	0	0000	262	KTF24	DC	*--*		
0140	0	0000	263	WTF2	DC	*--*		
0141	0	0000	264	P3T2	DC	*--*		
0142	0	0000	265	P5T2	DC	*--*		
0143	0	0000	266	T82	DC	*--*		
0144	0	0472	267	WFMN2	DC		1138	
0145	0	0000	268	C12	DC	*--*		HWS01510
0146	0	0000	269	C22	DC	*--*		HWS01520
0147	0	0000	270	TST2	DC	*--*		HWS01540
0148	0	0000	271	SUM2	BSS	E	0	
0148	0	0000	272		DC		0	
0149	0	0000	273		DC		0	
014A			274	IN2F	EQU	*		HWS01550
014A	01	C40000B9	275		LD	L	C1	HWS01560
014C	0	D0F8	276		STO		C12	HWS01570
014D	01	C40000BA	277		LD	L	C2	HWS01580
014F	0	D0F6	278		STO		C22	HWS01590

Table B-11. Honeywell Control Program (Continued)

17 JUL 74 PAGE 006

0150	01	C400010D	279		LD	L	SETX1		HWS01600
0152	0	DOF4	280		STO		TST2		HWS01610
0153	01	65800147	281	LUP2	LDX	I1	TST2		HWS01620
0155	01	C5000135	282		LD	L1	KEF21		HWS01630
0157	0	AOED	283		M		C12		HWS01640
0158	0	D8EF	284		STD		SUM2		
0159	01	C5000168	285		LD	L1	KEF31		HWS01670
015B	0	AOEA	286		M		C22		HWS01680
015C	0	88EB	287		AD		SUM2		
015D	0	1887	288		SRT		7		
015E	0	1090	289		SLT		16		
015F	01	D5000406	290		STO	L1	KEFN1		HWS01720
0161	0	COE5	291		LD		TST2		HWS01730
0162	01	8400010C	292		A	L	BUMP1		HWS01740
0164	0	DOE2	293		STO		TST2		HWS01750
0165	01	9400010E	294		S	L	NGFT		HWS01760
0167	01	4C280153	295		BN		LUP2		HWS01770
0169	01	4C0001B1	296		B	L	FUELM		HWS01780
			297	*			EQUILIBRIUM FUEL FLOW 85 GAINS		HWS01790
C 68	0	0000	298		KEF31	DC	**		
016C	0	0000	299		KEF32	DC	0		
016D	0	0000	300		KEF33	DC	0		
016E	0	0000	301		KEF34	DC	**		
016F	0	0000	302		WEF3	DC	**		
0170	0	1811	303		P3E3	DC	6161		
0171	0	08C3	304		P5E3	DC	2243		
			305	*			TEMPERATURE FUEL FLOW 85 GAINS		HWS01950
0172	0	0000	306		KTF31	DC	**		
0173	0	0000	307		KTF32	DC	**		
0174	0	0000	308		KTF33	DC	**		
0175	0	0000	309		KTF34	DC	**		
0176	0	0000	310		WTF3	DC	**		
0177	0	0000	311		P3T3	DC	**		
0178	0	0000	312		P5T3	DC	**		
0179	0	0000	313		T83	DC	**		
017A	0	0659	314		WFMN3	DC	1625		
017B	0	0000	315		C13	DC	**		HWS02010
017C	0	0000	316		C23	DC	**		HWS02020
017D	0	0000	317		TST3	DC	**		HWS02040
017E	0	0000	318		SUM3	BSS	E	0	
017F	0	0000	319			DC		0	
			320			DC		0	
0180			321		IN3F	EQU	*		HWS02050
0180	01	C4000089	322		LD	L	C1		HWS02060
0182	0	DOF8	323		STO		C13		HWS02070
0183	01	C400008A	324		LD	L	C2		HWS02080
0185	0	DOF6	325		STO		C23		HWS02090
0186	01	C400010D	326		LD	L	SETX1		HWS02100
0188	0	DOF4	327		STO		TST3		HWS02110
0189	01	6580017D	328	LUP3	LDX	I1	TST3		HWS02120
018B	01	C500016B	329		LD	L1	KEF31		HWS02130
018D	0	AOED	330		M		C13		HWS02140
018E	0	D8EF	331		STD		SUM3		
018F	01	C50001A1	332		LD	L1	KEF41		HWS02170
0191	0	AOEA	333		M		C23		HWS02180
0192	0	88EB	334		AD		SUM3		
0193	0	1887	335		SRT		7		

Table B-11. Honeywell Control Program (Continued)

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0194	0	1090	336	SLT		15		
0195	01	D5000406	337	STD	L1	KEFN1		HWS02220
0197	0	C0E5	338	LD		TST3		HWS02230
C198	01	8400010C	339	A	L	BUMPL		HWS02240
019A	0	D0E2	340	STD		TST3		HWS02250
019B	01	9400010E	341	S	L	NGFT		HWS02260
019D	01	4C280189	342	BN		LUP3		HWS02270
019F	01	4C0001B1	343	B	L	FUELM		HWS02280
			344	*		EQUILIBRIUM FUEL FLOW 100 GAINS		HWS02290
01A1	0	0000	345	KEF41	DC	**		
01A2	0	0000	346	KEF42	DC	0		
01A3	0	0000	347	KEF43	DC	0		
01A4	0	0000	348	KEF44	DC	**		
01A5	0	0000	349	MEF4	DC	**		
01A6	0	277E	350	P3E4	DC	10110		
01A7	0	0F94	351	P5E4	DC	3988		
			352	*		TEMPERATURE FUEL FLOW 100 GAINS		HWS02450
01A8	0	0000	353	KTF41	DC	**		
01A9	0	0000	354	KTF42	DC	**		
01AA	0	0000	355	KTF43	DC	**		
01AB	0	0000	356	KTF44	DC	**		
01AC	0	0000	357	WTF4	DC	**		
01AD	0	0000	358	P3T4	DC	**		
01AE	0	0000	359	P5T4	DC	**		
01AF	0	0000	360	T84	DC	**		
01B0	0	0CB2	361	WFMN4	DC	3250		
01B1			362	FUELM	EQU	*		HWS02510
01B1			363	FT4W	EQU	*		
01B1	0	C29A	364	LD	2	VT102	LOAD PBX100	
01B2	0	9032	365	S		=5850		
01B3	01	4C3001CD	366	BP		NEXT		
01B5	0	C030	367	LD		=1553		
01B6	0	A030	368	M		=100		
01B7	01	DC0001F2	369	STD	L	TMPE		
01B9	0	C29A	370	LD	2	VT102		
01BA	0	A02D	371	M		=6		
01BB	01	8C0001F2	372	AD	L	TMPE		
01BD	0	A82B	373	D		=3400		
01BF	01	D40001FB	374	STD	L	K1THD		
01C0	0	C29A	375	LD	2	VT102		
01C1	0	A02B	376	M		=200		
01C2	01	DC0001F2	377	STD	L	TMPE		
01C4	0	C026	378	LD		=15100		
01C5	0	A021	379	M		=100		
01C6	01	9C0001F2	380	SD	L	TMPE		
01C8	0	A820	381	D		=3400		
01C9	01	D40001FC	382	STD	L	TAU2T		
01CB	01	4C0001E3	383	B	L	FT4WC		
01CD	0	C01E	384	NEXT	LD	=2202		
01CE	0	A01B	385	M		=100		
01CF	01	DC0001F2	386	STD	L	TMPE		
01D1	0	C29A	387	LD	2	VT102		
01D2	0	A01A	388	M		=4		
01D3	01	8C0001F2	389	AD	L	TMPE		
01D5	0	A81B	390	D		=4350		
01D6	01	D40001FB	391	STD	L	K1THD		
01D8	0	C29A	392	LD	2	VT102		

Table B-11. Honeywell Control Program (Continued)

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01D9	0	A015	393		M		=20
01DA	01	DC0001E2	394		STD	L	TMPP
01DC	0	C013	395		LD		=5520
01DD	0	A009	396		M		=100
01DE	01	9C0001F2	397		SD	L	TMPP
01EO	0	A80D	398		D		=4350
01E1	01	D40001FC	399		STD	L	TAU2T
01E3	01	4C0001FE	400	FT4WC	B	L	STP1
			401		LORG		
01E5	0	16DA	402	+	DC		5850
01E6	0	0611	403	+	DC		1553
01E7	0	0064	404	+	DC		100
01E8	0	0006	405	+	DC		6
01E9	0	0048	406	+	DC		3400
01EA	0	00C8	407	+	DC		200
01EB	0	3AFC	408	+	DC		15100
01EC	0	089A	409	+	DC		2202
01ED	0	0004	410	+	DC		4
01EE	0	10FE	411	+	DC		4350
01EF	0	0014	412	+	DC		20
01FO	0	1590	413	+	DC		5520
01F2	0	0000	414		TMPP	BSS	E 0
01F2	0	0000	415		DC		0
01F3	0	0000	416		DC		0
01F4	0	0000	417	XT4	BSS	E	0
01F4	0	0000	418		DC		0
01F5	0	0000	419		DC		0
01F6	0	0000	420	XT4D	BSS	E	0
01F6	0	0000	421		DC		0
01F7	0	0000	422		DC		0
01F8	0	0000	423	XT4D1	BSS	F	0
01F8	0	0000	424		DC		0
01F9	0	0000	425		DC		0
01FA	0	0000	426	T4WF	DC		**
01FB	0	0000	427	R1THD	DC		**
01FC	0	0000	428	TAU2T	DC		**
01FD	0	0000	429	ISW	DC		0
01FE	01	C4000280	430	STP1	LD	L	=1234
0200	0	90FC	431		S		ISW
0201	01	4C18020E	432		BZ		STP2
0203	0	D0F9	433		STD		ISW
0204	01	C40000BD	434		LD	L	=0
0206	0	1890	435		SRT		16
0207	0	D8EE	436		STD		XT4D
0208	0	D8EF	437		STD		XT4D1
0209	0	C29F	438		LD	2	VT097
020A	0	1890	439		SRT		16
020B	0	D8E8	440		STD		XT4
020C	01	4C000219	441		B	L	STP3
020E	01	C40000BD	442	STP2	LD	L	=0
0210	0	1890	443		SRT		16
0211	0	C29F	444		LD	2	VT097
0212	0	1890	445		SRT		16
0213	0	98E0	446		SD		XT4
0214	0	A3E6	447		D		K1THD
0215	0	A0D1	448		M		=100
0216	0	A8E5	449		D		TAU2T

Table B-11. Honeywell Control Program (Continued)

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0217	0	A069	450	M	=10		
0218	0	D8DD	451	STD	XT4D		
0219	0	C8DC	452	STP3 LDD	XT4D	*** UPDATE D E ***	
021A	0	88DD	453	AD	XT4D1		
021B	0	A866	454	D	=50		
021C	01	A400028E	455	M	L DT		
021E	0	A864	456	D	=40		
021F	0	1890	457	SRT	16		
0220	0	88D3	458	AD	XT4		
0221	0	D8D2	459	STD	XT4		
0222	0	D2B0	460	STO	2 VT080	*** MOST SIGNIFICANT PART	
0223	0	1090	461	SLT	16	OF XT4	
0224	0	D2A6	462	STO	2 VT090	LEAST SIGNIFICANT PART	
0225	0	C8D0	463	LDD	XT4D		
0226	0	D8D1	464	STD	XT4D1	*** CALCULATE T4WF ***	
0227	0	A859	465	D	=10		
0228	0	A0D3	466	M	TAU2T		
0229	0	88CA	467	AD	XT4		
022A	0	1090	468	SLT	16		
022B	0	DOCE	469	STO	T4WF		
022C	0	D24C	470	STO	2 VT203		
022D	0	C2D9	471	LD	2 VT039		HWS02520
022E	0	9055	472	S	=64		HWS02530
022F	01	4C200248	473	BNZ	MDW9		HWS02540
0231	0	D2D9	474	STO	2 VT039		
0232	0	C2DC	475	LD	2 VT036		HWS02550
0233	0	D05B	476	STO	ENK		
0234	01	4C100238	477	BNZ	SENL		
0236	0	C04E	478	LD	=0		
0237	0	9057	479	S	ENK		
0238	0	D057	480	SENL STD	ENKL		
0239	0	C2DB	481	LD	2 VT037		HWS02570
023A	0	1884	482	SRT	4		
023B	0	D055	483	STD	ENK		
023C	01	4C100240	484	BNN	SEPL		
023E	0	C046	485	LD	=0		
023F	0	9051	486	S	EPK		
0240	0	D051	487	SEPL STD	EPKL		
0241	0	C2DA	488	LD	2 VT038		HWS02590
0242	01	D4000293	489	STO	L ETK		
0244	01	4C100249	490	BNN	SETL		
0246	0	C03E	491	LD	=0		
0247	01	94000293	492	S	L ETK		
0249	01	D4000294	493	SETL STD	L ETKL		
024B			494	MDW9	EQU	*	HWS02610
			495	*			HWS02620
			496	*			HWS02630
			497	*			HWS02640
			498	*			HWS02650
			499	*	THIS SECTION OF THE PROGRAM		HWS02660
			500	*	WILL SET UP THE MEASUREMENT FEED-BACK		HWS02670
			501	*	VECTOR FOR THE THREE CONTROLLERS		HWS02680
			502	*	EQUILIBRIUM, PRESSURE, TEMPERATURE		HWS02690
			503	*			HWS02700
			504	*			HWS02710
			505	*			HWS02720
			506	*	CALCULATE DERIVATIVES FOR EN EP ET		HWS02730

Table B-11. Honeywell Control Program (Continued)

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0248	0	C201	507	*	LD	2	VT128		HWS02740
024C	0	921E	508		S	2	VT157		HWS02750
024D	0	D03A	509		STO		ENDK		HWS02760
024E	01	C4000412	510		LD	L	P3TNB		HWS02770
0250	0	929A	511		S	2	VT102		
0251	0	D038	512		STO		EPDK		HWS02840
0252	01	C4000414	513		LD	L	TBBN	*** CALCULATE ETDOT ***	
0254	0	90A5	514		S		T4WF		
0255	01	D400028C	515		STO	L	ETDK		
0257	0	C02F	516		LD		TIME		HWS02900
0258	01	4C200295	517		BNZ		INTEG		HWS02910
025A	01	C400046A	518		LD	L	JNE		
025C	0	D286	519		STO	2	VT074		
025D	0	C02A	520		LD		ENDK		HWS02920
025E	0	D02A	521		STO		ENDK1		HWS02930
025F	0	C02A	522		LD		EPDK		HWS02940
0260	0	D02A	523		STO		EPDK1		HWS02950
0261	0	C02A	524		LD		ETDK		HWS02960
0262	0	D02A	525		STO		ETDK1		HWS02970
0263	0	C20C	526		LD	2	VT036		HWS02980
0264	0	D02A	527		STO		ENK		HWS02990
0265	01	4C100269	528		BNN		STENL		
0267	0	C01D	529		LD		=0		
0268	0	9026	530		S		ENK		
0269	0	D026	531		STENL	STO	ENKL		
026A	0	C20D	532		LD	2	VT037		HWS03010
026B	0	1884	533		SRT		4		
026C	0	D024	534		STO		EPK		HWS03030
026D	01	4C100271	535		BNN		STEPL		
026F	0	C015	536		LD		=0		
0270	0	9020	537		S		EPK		
0271	0	D020	538		STEPL	STO	EPKL		
0272	0	C2DA	539		LD	2	VT038		HWS03050
0273	01	D4000293	540		STO	L	ETK		
0275	01	4C10027A	541		BNN		STETL		
0277	0	C00D	542		LD		=0		
0278	01	94000293	543		S	L	ETK		
027A	01	D4000294	544		STETL	STO	L	ETKL	
027C	0	C009	545		LD		=1		HWS03080
027D	0	D009	546		STO		TIME		HWS03090
027E	01	4C000295	547		B	L	INTEG		
			548		LORG				HWS03110
0280	0	0402	549		DC		1234		
0281	0	000A	550	+	DC		10		
0282	0	0032	551	+	DC		50		
0283	0	0028	552	+	DC		40		
0284	0	0040	553	+	DC		64		
0285	0	0000	554	+	DC		0		
0286	0	0001	555	+	DC		1		
			556		*GENERATE EN ET				HWS03120
0287	0	0000	557		TIME	DC	0		
0288	0	0000	558		ENDK	DC	**		HWS03140
0289	0	0000	559		ENDK1	DC	**		HWS03150
028A	0	0000	560		EPDK	DC	**		HWS03160
028B	0	0000	561		EPDK1	DC	**		HWS03170
028C	0	0000	562		ETDK	DC	**		HWS03180
			563				**		

Table B-11. Honeywell Control Program (Continued)

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028D	0	0000	564	ETDK1	DC	*--*		HWS03190
028E	0	000F	565	DT	DC	15		HWS03200
028F	0	0000	566	ENK	DC	*--*		HWS03210
0290	0	0000	567	ENKL	DC	*--*		HWS03220
0291	0	0000	568	EPK	DC	*--*		HWS03230
0292	0	0000	569	EPKL	DC	*--*		HWS03240
0293	0	0000	570	ETK	DC	*--*		HWS03250
0294	0	0000	571	ETKL	DC	*--*		HWS03260
			572	*				HWS03280
0295			573	INTEG	EQU	*		HWS03290
			574	*			CALCULATE EN	HWS03300
0295	0	COF2	575		LD	ENDK		
0296	0	80F2	576		A	ENDK1		
0297	0	A0F6	577		M	DT		
0298	0	1083	578		SLT	3		
0299	0	A84F	579		D	=375		
029A	0	80r4	580		A	ENK		HWS03400
029B	0	DOF3	581		STO	ENK		HWS03410
029C	0	C286	582		LD	2 VT074		
029D	01	9400046A	583		S	L ONE		
029F	01	4C1802A3	584		BZ	NM1		
02A1	0	COE3	585		LD	=0		
02A2	0	DOEC	586		STO	ENK		
			587	*			CALCULATE EP	HWS03420
			588	*			CALCULATE ET	HWS03510
02A3	01	C400028C	589	NM1	LD	L ETDK		
02A5	01	8400028D	590		A	L ETDK1		
02A7	01	A400028E	591		M	L DT		
02A9	0	1084	592		SLT	4	*** SCALE FACTOR 16 ***	
02AA	0	A83F	593		D	=1500		
02AB	0	80E7	594		A	ETK		
02AC	0	DOE6	595		STO	ETK		
02AD	0	C286	596		LD	2 VT074		
02AE	01	9400046B	597		S	L TWO		
02B0	01	4C1802B4	598		BZ	NM2		
02B2	0	COE2	599		LD	=0		
02B3	0	DODF	600		STO	ETK		
			601	*			LIMITS ON EN EP ET	HWS03600
02B4	0	CODA	602	NM2	LD	ENK		
02B5	01	4C2802BE	603		BN	MW1		HWS03620
02B7	0	90DB	604		S	ENKL		HWS03630
02B8	01	4C0802C5	605		BNP	MW2		HWS03640
02BA	0	COD5	606		LD	ENKL		HWS03650
02BB	0	D0D3	607		STO	ENK		HWS03660
02BC	01	4C0002C5	608		B	L MW2		HWS03670
02BE			609	MW1	EQU	*		HWS03680
02BE	0	COD0	610		LD	ENK		HWS03690
02BF	0	80D0	611		A	ENKL		HWS03700
02C0	01	4C3002C5	612		BP	MW2		HWS03710
02C2	0	COC2	613		LD	=0		HWS03720
02C3	0	90CC	614		S	ENKL		HWS03730
02C4	0	DOCA	615		STO	ENK		HWS03740
02C5			616	MW2	EQU	*		HWS03750
02C5	0	COCB	617		LD	EPK		HWS03760
02C6	01	4C2802CF	618		BN	MW3		HWS03770
02C8	0	90C9	619		S	EPKL		HWS03780
02C9	01	4C0802D6	620		BNP	MW4		HWS03790

Table B-11. Honeywell Control Program (Continued)

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02CB	0	C0C6	621		LD	EPKL	HWS03800
02CC	0	00C4	622		STO	EPK	HWS03810
02CD	01	4C0002D6	623		B	L MW4	HWS03820
02CF			624	MW3	EQU	*	HWS03830
02CF	0	C0C1	625		LD	EPK	HWS03840
02D0	0	80C1	626		A	EPKL	HWS03850
02D1	01	4C3002D6	627		BP	MW4	HWS03860
02D3	0	C0B1	628		LD	=0	HWS03870
02D4	0	90B0	629		S	EPKL	HWS03880
02D5	0	D0BB	630		STO	EPK	HWS03890
02D6			631	MW4	EQU	*	HWS03900
02D6	0	C0BC	632		LD	ETK	HWS03910
02D7	01	4C2802E0	633		BN	MW5	HWS03920
02D9	0	90BA	634		S	ETKL	HWS03930
02DA	01	4C0802EB	635		BNP	MW6	HWS03940
02DC	0	C0B7	636		LD	ETKL	HWS03950
02DD	0	D0B5	637		STO	ETK	HWS03960
02DE	01	4C0002EB	638		B	L MW6	HWS03970
02E0			639	MW5	EQU	*	HWS03980
02E0	0	C0B2	640		LD	ETK	HWS03990
02E1	0	80B2	641		A	ETKL	HWS04000
02E2	01	4C3002EB	642		BP	MW6	HWS04010
02E4	0	COA0	643		LD	=0	HWS04020
02E5	0	90AE	644		S	ETKL	HWS04030
02E6	0	D0AC	645		STO	ETK	HWS04040
02E7	01	4C0002EB	646		B	L MW6	HWS04050
			647		LORG		HWS04060
02E9	0	0177	648	+	DC	375	
02EA	0	05D5	649	+	DC	1500	
			650	*			
						AGE DERIVATIVES	HWS04070
02EB			651	MW6	EQU	*	HWS04080
02EB	01	C4000288	652		LD	L ENDK	HWS04090
02ED	01	D4000289	653		STO	L ENDK1	HWS04100
02EF	01	C400028A	654		LD	L EPDK	HWS04110
02F1	01	D400028B	655		STO	L EPDK1	HWS04120
02F3	01	C400028C	656		LD	L ETDK	HWS04130
02F5	01	D400028D	657		STO	L ETDK1	HWS04140
			658	*			HWS04150
			659	*		INTERPOLATE FOR PT3 AND PT5	HWS04160
			660	*		AS A FUNCTION OF PLA	HWS04170
02F7			661	PLA	EQU	*	
02F7	0	C066	662		LD	NPL1	HWS04180
02F8	0	9201	663		S	2 VT128	HWS04190
02F9	01	4C280309	664		BN	MDW1	HWS04200
02F8	01	C4000101	665		LD	L P3E1	HWS04210
02FD	01	D400040B	666		STO	L P3PL	HWS04220
02FF	01	C4000102	667		LD	L P5E1	HWS04230
0301	01	D400040C	668		STO	L P5PL	HWS04240
0303	01	C4000100	669		LD	L WEF1	HWS04250
0305	01	D400040A	670		STO	L WEFN	HWS04260
0307	01	4C0003A7	671		B	L MDW6	HWS04270
0309	0	C057	672	MDW1	LD	NPL4	HWS04280
030A	0	9201	673		S	2 VT128	HWS04290
030B	01	4C300318	674		BP	MDW2	HWS04300
030D	01	C40001A6	675		LD	L P3E4	HWS04310
030F	01	D400040B	676		STO	L P3PL	HWS04320
0311	01	C40001A7	677		LD	L P5E4	HWS04330

Table B-11. Honeywell Control Program (Continued)

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0313	01	D400040C	678		STO	L	P5PL	HWSO 4340
0315	01	C40001A5	679		LD	L	MEF4	HWSO 4350
0317	01	D400040A	680		STO	L	MEFM	HWSO 4360
0319	01	4C0003A7	681		B	L	MDW6	HWSO 4370
0318	01	C400035F	682	MDW2	LD	L	NPL2	HWSO 4380
031D	0	9201	683		S	2	VT128	HWSO 4390
031E	01	4C28033E	684		BN		MDW3	HWSO 4400
0320	0	1889	685		SRT		9	
0321	0	A81A	686		D		=3300	HWSO 4420
0322	0	D03F	687		STO		CX1	HWSO 4430
0323	0	C019	688		LD		=128	
0324	0	903D	689		S		CX1	HWSO 4450
0325	0	D03D	690		STO		CX2	HWSO 4460
0326	01	C4000101	691		LD	L	P3E1	HWSO 4470
0328	0	D03B	692		STO		P3L	HWSO 4480
0329	01	C400013A	693		LD	L	P3E2	HWSO 4490
032B	0	D039	694		STO		P3M	HWSO 4500
032C	01	C4000102	695		LD	L	P5E1	HWSO 4510
032E	0	D037	696		STO		P5L	HWSO 4520
032F	01	C4000138	697		LD	L	P5E2	HWSO 4530
0331	0	D035	698		STO		P5M	HWSO 4540
0332	01	C4000100	699		LD	L	MEF1	HWSO 4550
0334	01	D4000368	700		STO	L	WEFL	HWSO 4560
0336	01	C4000139	701		LD	L	MEF2	HWSO 4570
0338	01	D4000369	702		STO	L	WEFM	HWSO 4580
033A	01	4C000388	703		B	L	MDW5	HWSO 4590
			704		LORG			HWSO 4600
033C	0	0CE4	705	+	DC		3300	
033D	0	0080	706	+	DC		128	
033E	0	C021	707	MDW3	LD		NPL3	HWSO 4610
033F	0	9201	708		S	2	VT128	HWSO 4620
0340	01	4C28036C	709		BN		MDW4	HWSO 4630
0342	0	1889	710		SRT		9	
0343	0	A862	711		D		=2475	HWSO 4650
0344	0	D01D	712		STO		CX1	HWSO 4660
0345	0	C0F7	713		LD		=128	
0346	0	9018	714		S		CX1	HWSO 4680
0347	0	D018	715		STO		CX2	HWSO 4690
0348	01	C400013A	716		LD	L	P3E2	HWSO 4700
034A	0	D019	717		STO		P3L	HWSO 4710
034B	01	C4000170	718		LD	L	P3E3	HWSO 4720
034D	0	D017	719		STO		P3M	HWSO 4730
034E	01	C4000138	720		LD	L	P5E2	HWSO 4740
0350	0	D015	721		STO		P5L	HWSO 4750
0351	01	C4000171	722		LD	L	P5E3	HWSO 4760
0353	0	D013	723		STO		P5M	HWSO 4770
0354	01	C4000139	724		LD	L	WEF2	HWSO 4780
0356	01	D4000368	725		STO	L	WEFL	HWSO 4790
0358	01	C400016F	726		LD	L	WEF3	HWSO 4800
035A	01	D4000369	727		STO	L	WEFM	HWSO 4810
035C	01	4C000388	728		B	L	MDW5	HWSO 4820
035E	0	203A	729	NPL1	DC		8250	HWSO 4830
035F	0	2D1E	730	NPL2	DC		11550	HWSO 4840
0360	0	36C9	731	NPL3	DC		14025	HWSO 4850
0361	0	4074	732	NPL4	DC		16500	HWSO 4860
0362	0	0000	733	CX1	DC		*--	HWSO 4870
0363	0	0000	734	CX2	DC		*--	HWSO 4880

Table B-11. Honeywell Control Program (Continued)

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0364	0	0000	735	P3L	DC	**	HWS04890
0365	0	0000	736	P3M	DC	**	HWS04900
0366	0	0000	737	P5L	DC	**	HWS04910
0367	0	0000	738	P5M	DC	**	HWS04920
0368	0	0000	739	WEFL	DC	**	HWS04940
0369	0	0000	740	WEFM	DC	**	HWS04950
036A	0	0000	741	SUMX	BSS	E 0	
036A	0	0000	742		DC	0	
036B	0	0000	743		DC	0	
036C	0	C0F4	744	MDW4	LD	NPL4	HWS04960
036D	0	9201	745		S	2 VT128	HWS04970
036E	0	1889	746		SRT	9	
036F	0	A836	747		D	=2475	HWS04990
0370	0	D0F1	748		STO	CX1	HWS05000
0371	0	C0C8	749		LD	=128	
0372	0	90FF	750		S	CX1	HWS05020
0373	0	D0EF	751		STO	CX2	HWS05030
0374	01	C4000170	752		LD	L P3E3	HWS05040
0376	0	D0ED	753		STO	P3L	HWS05050
0377	01	C40001A6	754		LD	L P3E4	HWS05060
0379	0	D0E8	755		STO	P3M	HWS05070
037A	01	C4000171	756		LD	L P5E3	HWS05080
037C	0	D0E9	757		STO	P5L	HWS05090
037D	01	C40001A7	758		LD	L P5E4	HWS05100
037F	0	D0E7	759		STO	P5M	HWS05110
0380	01	C400016F	760		LD	L WEF3	HWS05120
0382	01	D4000368	761		STO	L WEFL	HWS05130
0384	01	C40001A5	762		LD	L WEF4	HWS05140
0386	01	D4000369	763		STO	L WEFM	HWS05150
0388	0	C0DB	764	MDW5	LD	P3L	HWS05160
0389	0	A0D8	765		M	CX1	HWS05170
038A	0	D8DF	766		STO	SUMX	
038E	0	C0D9	767		LD	P3M	HWS05200
038C	0	A0D6	768		M	CX2	HWS05210
038D	0	88DC	769		AD	SUMX	
038E	0	1887	770		SRT	7	
038F	0	1090	771		SLT	16	
0390	0	D07A	772		STO	P3PL	HWS05250
0391	0	C0D4	773		LD	P5L	HWS05260
0392	0	A0CF	774		M	CX1	HWS05270
0393	0	D8D6	775		STO	SUMX	
0394	0	C0D2	776		LD	P5M	HWS05300
0395	0	A0CD	777		M	CX2	HWS05310
0396	0	88D3	778		AD	SUMX	
0397	0	1887	779		SRT	7	
0398	0	1090	780		SLT	16	
0399	0	D072	781		STO	P5PL	HWS05350
039A	01	C4000368	782		LD	L WEFL	HWS05360
039C	0	A0C5	783		M	CX1	HWS05370
039D	0	D8CC	784		STO	SUMX	
039E	0	C0CA	785		LD	WEFM	HWS05400
039F	0	A0C3	786		M	CX2	HWS05410
03A0	0	88C9	787		AD	SUMX	
03A1	0	1887	788		SRT	7	
03A2	0	1090	789		SLT	16	
03A3	0	D066	790		STO	WEFN	HWS05450
03A4	01	C40003A7	791		B	L MDW6	HWS05460

Table B-11. Honeywell Control Program (Continued)

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03A6	0	09AB	792		LORG		HWS05470
			793	+	DC	2475	
03A7			794	MDW6	EQU	*	HWS05510
03A7	01	C400040C	795		LD	L P5PL	HWS05520
03A9	0	D223	796		STO	2 VT162	HWS05530
03AA	01	C400040B	797		LD	L P3PL	HWS05540
03AC	0	D224	798		STO	2 VT153	HWS05550
03AD	01	C400028F	799		LD	L ENK	HWS05560
03AF	0	D225	800		STO	2 VT164	HWS05570
03B0	01	C400040A	801		LD	L MEFN	HWS05580
03B2	0	D226	802		STO	2 VT165	HWS05590
03B3	01	C4000406	803		LD	L KEFN1	HWS05600
03B5	0	D227	804		STO	2 VT166	HWS05610
03B6	01	C4000407	805		LD	L KEFN2	HWS05620
03B8	0	D228	806		STO	2 VT167	HWS05630
03B9	01	C4000411	807		LD	L MTFN	
03B8	0	D229	808		STO	2 VT168	HWS05650
03BC	01	C4000409	809		LD	L KEFN4	HWS05660
03BE	0	D22A	810		STO	2 VT169	HWS05670
03BF	01	C4000413	811		LD	L P5TNB	
03C1	0	D22B	812		STO	2 VT170	HWS05690
03C2	01	C4000412	813		LD	L P3TNB	
03C4	0	D22C	814		STO	2 VT171	HWS05710
03C5	01	C4000293	815		LD	L ETK	
03C7	0	D22D	816		STO	2 VT172	HWS05730
03C8	01	C400040D	817		LD	L KTFN1	
03CA	0	D22E	818		STO	2 VT173	HWS05750
03CB	01	C400040E	819		LD	L KTFN2	
03CD	0	D22F	820		STO	2 VT174	HWS05770
03CE	01	C400040F	821		LD	L KTFN3	
03D0	0	D230	822		STO	2 VT175	HWS05790
03D1	01	C4000410	823		LD	L KTFN4	
03D3	0	D231	824		STO	2 VT176	HWS05810
03D4	01	C4000414	825		LD	L TBBN	
03D6	0	D248	826		STO	2 VT202	
			827	*			HWS05480
			828	*	CALCULATE X-X0	FOR EQUILIBRIUM PRESSURE	HWS05490
			829	*			HWS05500
0307			830	MEPT	EQU	*	
03D7	0	C21E	831		LD	2 VT157	HWS05820
03D8	0	9201	832		S	2 VT128	HWS05830
03D9	0	D024	833		STO	ME1	HWS05840
03DA	0	D245	834		STO	2 VT196	
03DB	0	C294	835		LD	2 VT108	HWS05850
03DC	01	9400040C	836		S	L P5PL	HWS05880
03DE	0	D020	837		STO	ME2	HWS05890
03DF	0	D246	838		STO	2 VT197	
03E0	0	C29A	839		LD	2 VT102	HWS05900
03E1	01	9400040B	840		S	L P3PL	HWS05930
03E3	0	D01L	841		STO	ME3	HWS05940
03E4	0	D247	842		STO	2 VT198	
03E5	01	C400028F	843		LD	L ENK	HWS05950
03E7	0	D019	844		STO	ME4	HWS05960
			845	*			HWS05970
03E8	0	C294	846		LD	2 VT108	
03E9	01	94000413	847		S	L P5TNB	
03EB	01	D4000402	848		STO	L MT1	

Table B-11. Honeywell Control Program (Continued)

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03ED 0	C29A	849	LD	2	VT102	
03EE 01	94000412	850	S	L	P3TNB	
03FO 01	D4000403	851	STD	L	MT2	
03F2 0	D248	852	STD	2	VT199	
03F3 01	C40001FA	853	LD	L	T4WF	
03F5 01	94000414	854	S	L	T8BN	
03F7 0	D00C	855	STD		MT3	
03F8 0	D249	856	STD	2	VT200	
03F9 01	C4000293	857	LD	L	ETK	
03FB 0	D009	858	STD		MT4	
03FC 01	4C000417	859	B	L	FREQE	HWS06100
03FE 0	0000	860	ME1	DC	**	HWS06110
03FF 0	0000	861	ME2	DC	**	HWS06120
0400 0	0000	862	ME3	DC	**	HWS06130
0401 0	0000	863	ME4	DC	**	HWS06140
0402 0	0000	864	MT1	DC	**	
0403 0	0000	865	MT2	DC	**	
0404 0	0000	866	MT3	DC	**	
0405 0	0000	867	MT4	DC	**	
0406 0	0000	868	KEFN1	DC	**	HWS06190
0407 0	0000	869	KEFN2	DC	**	HWS06200
0408 0	0000	870	KEFN3	DC	**	HWS06210
0409 0	0000	871	KEFN4	DC	**	HWS06220
040A 0	0000	872	WEFN	DC	**	HWS06230
040B 0	0000	873	P3PL	DC	**	HWS06240
040C 0	0000	874	P5PL	DC	**	HWS06250
040D 0	0000	875	KTFN1	DC	**	
040E 0	0000	876	KTFN2	DC	**	
040F 0	0000	877	KTFN3	DC	**	
0410 0	0000	878	KTFN4	DC	**	
0411 0	0000	879	WTFN	DC	**	HWS06330
0412 0	0000	880	P3TNB	DC	**	
0413 0	0000	881	P5TNB	DC	**	
0414 0	0000	882	T1BN	DC	**	HWS06370
0415 0	0000	883	VFMNN	DC	**	
0416 0	0000	884	SUMEF	DC	**	HWS06380
		885	*			HWS06390
		886	* AT THIS POINT THE FUEL FLOW REQUEST IS			HWS06400
		887	* COMPUTED FOR THE THREE CONTROLLERS			HWS06410
		888	* %IE EQUILIBRIUM,PRESSURE,AND TEMPERATURE			HWS06420
		889	* AND A SELECT LOW DETERMINS WHICH CONTROLLER			HWS06430
		890	* WILL BE USED			HWS06440
0417		891	FREQE	EQU	*	HWS06450
0417 01	C4000406	892	LD	L	KEFN1	HWS06460
0419 0	A0E4	893	M		ME1	HWS06470
041A 0	1087	894	SLT		7	
041B 0	D24A	895	STD	2	VT201	
041C 0	D0F9	896	STD		SUMEF	HWS06490
041D 01	C4000407	897	LD	L	KEFN2	HWS06500
041F 0	A0DF	898	M		ME2	HWS06510
0420 0	AP18	899	D		=100	
0421 0	1989	900	SRT		9	
0422 0	8CF3	901	A		SUMEF	HWS06530
0423 0	D0F2	902	STD		SUMEF	HWS06540
0424 01	C4000408	903	LD	L	KEFN3	HWS06550
0426 0	A0D9	904	M		ME3	HWS06560
0427 0	A811	905	D		=100	

Table B-11. Honeywell Control Program (Continued)

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0428	0	1889	906	SRT	9		
0429	0	80EC	907	A	SUMEF		HWS06580
042A	0	D0EB	908	STO	SUMEF		HWS06590
042B	01	C4000409	909	LD	L	KEFN4	HWS06600
042D	0	A0D3	910	M		ME4	HWS06610
042E	0	1084	911	SLT	4		
042F	0	D24D	912	STO	2	VT204	
0430	0	80E5	913	A		SUMEF	HWS06630
0431	0	D0E4	914	STO		SUMEF	HWS06650
0432	01	C400040A	915	LD	L	MEFN	HWS06670
0434	0	1882	916	SRT	2		
0435	0	80E0	917	A		SUMEF	HWS06700
0436	0	D0DF	918	STO		SUMEF	HWS06720
0437	01	4C00043B	919	B	L	FREQP	HWS06770
			920	LORG			HWS06780
0439	0	0064	921	+	DC	100	
043A	0	0000	922	SUMPF	DC	*--	HWS06790
043B			923	FREQP	EQU	*	HWS06800
043B	0	C003	924	LD		32700	HWS07160
043C	0	D0FD	925	STO		SUMPF	HWS07070
043D	01	4C000441	926	B	L	FREQT	HWS07120
			927	LORG			HWS07130
043F	0	7FBC	928	+	DC	32700	
0440	0	0000	929	SUMTF	DC	*--	HWS07140
0441			930	FREQT	EQU	*	HWS07150
0441	01	C400040D	931	LD	L	KTFN1	
0443	01	A4000402	932	M	L	MT1	
0445	0	A8F3	933	D		=100	
0446	0	1889	934	SRT	9		
0447	0	D0F8	935	STO		SUMTF	
0448	01	C400040E	936	LD	L	KTFN2	
044A	01	A4000403	937	M	L	MT2	
044C	0	A8EC	938	D		=100	
044D	0	1889	939	SRT	9		
044E	0	D24E	940	STO	2	VT205	
044F	0	80F0	941	A		SUMTF	
0450	0	D0EF	942	STO		SUMTF	
0451	01	C400040F	943	LD	L	KTFN3	
0453	01	A4000404	944	M	L	MT3	
0455	0	A813	945	D		=10	
0456	0	1889	946	SRT	9		
0457	0	D24F	947	STO	2	VT206	
0458	0	80E7	948	A		SUMTF	
0459	0	D0E6	949	STO		SUMTF	
045A	01	C4000410	950	LD	L	KTFN4	
045C	01	A4000405	951	M	L	MT4	
045E	0	1083	952	SLT	3		
045F	0	D250	953	STO	2	VT207	
0460	0	80DF	954	A		SUMTF	
0461	0	D0DE	955	STO		SUMTF	
0462	01	C4000411	956	LD	L	WTFN	
0464	0	1882	957	SRT	2		
0465	0	80DA	958	A		SUMTF	
0466	0	D0D9	959	STO		SUMTF	
0467	01	4C00046E	960	B	L	MDSWT	HWS07180
			961	LORG			HWS07190
0469	0	000A	962	+	DC	10	

Table B-11. Honeywell Control Program (Continued)

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046A	0	0000	963	ONE	DC	3276	HWS07200
046B	0	1998	964	TWO	DC	6552	HWS07210
046C	0	2664	965	THREE	DC	9828	HWS07220
046D	0	0000	966	WFMOD	DC	*--	HWS07230
046E			967	MDSWT	EQU	*	HWS07240
046E	01	C4000416	968		LD	L SUMEF	HWS07250
0470	0	D289	969		STO	2 VT071	HWS07260
0471	01	C400043A	970		LD	L SUMPFF	HWS07270
0473	0	D288	971		STO	2 VT072	HWS07280
0474	01	C400044U	972		LD	L SUMTF	HWS07290
0476	0	D237	973		STO	2 VT073	HWS07300
0477	0	B288	974		CMP	2 VT072	HWS07310
0478	0	C288	975		LD	2 VT072	HWS07320
0479	0	1000	976		NOP		HWS07330
047A	0	D0F2	977		STO	WFMOD	HWS07340
047B	0	B289	978		CMP	2 VT071	HWS07350
047C	0	C289	979		LD	2 VT071	HWS07360
047D	0	1000	980		NOP		HWS07370
047E	0	D235	981		STO	2 VT180	HWS07380
047F	01	4C28048A	982		BN	MINFL	
0481	01	A4000505	983		M	L =13	HWS06990
0483	0	1090	984		SLT	16	HWS07000
0484	0	D235	985		STO	2 VT180	
0485	0	C235	986		LD	2 VT180	
0486	01	94000415	987		S	L WFMNN	
0488	01	4C10048D	988		BNN	MINSS	
048A	01	4C000415	989	MINFL	LD	L WFMNN	
048C	0	D235	990		STO	2 VT180	
048D			991	MINSS	EQU	*	
048D	0	C235	992		LD	2 VT180	HWS07390
048E	0	1890	993		SRT	16	
048F	0	A875	994		D	=13	
0490	0	9289	995		S	2 VT071	HWS07400
0491	01	4C200497	996		BNZ	MIKE1	HWS07410
0493	0	C0D6	997		LD	ONE	HWS07420
0494	0	D286	998		STO	2 VT074	HWS07430
0495	01	4C0004AD	999		B	L RQA1B	HWS07440
0497	0	C235	1000	MIKE1	LD	2 VT180	HWS07450
0498	0	1190	1001		SRT	16	
0499	0	A868	1002		D	=13	
049A	0	9287	1003		S	2 VT073	HWS07460
049B	01	4C2004A1	1004		BNZ	MIKE2	HWS07470
049D	0	C0C0	1005		LD	TWO	HWS07480
049E	0	D286	1006		STO	2 VT074	HWS07490
049F	01	4C0004AD	1007		B	L RQA1B	HWS07500
04A1	0	C0CA	1008	MIKE2	LD	THREE	HWS07510
04A2	0	D286	1009		STO	2 VT074	HWS07520
04A3	01	4C0004AD	1010		B	L RQA1B	HWS07530
04A5	0	0031	1011	KLGD	DC	49	
04A6	0	001F	1012	K1NUM	DC	31	
04A7	0	0009	1013	K2NUM	DC	9	
04A8	0	0000	1014	YNM1	DC	*--	
04A9	0	0000	1015	UNM1	DC	*--	
04AA	0	0000	1016	TEMF	BSS	E 0	
04AA	0	0000	1017		DC	0	
04AB	0	0000	1018		DC	0	
04AC	0	0000	1019	SWLAG	DC	*--	

Table B-11. Honeywell Control Program (Continued)

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Address	Op	Operand	Label	Code	Value	Comments
04AD					1020	RQATB EQU *
04AD	01	C40004AC		LD	1021	L SWLAG
04AF	01	4C2004BC		BNZ	1022	FILT
04B1	01	C4000506		LD	1023	L =123
04B3	01	D40004AC		STO	1024	L SWLAG
04B5	0	C235		LD	1025	2 VT180
04B6	01	D40004A8		STO	1026	L YNMI
04B8	01	D40004A9		STO	1027	L UNMI
04BA	01	4C0004D6		B	1028	L DONOZ
04BC	01	C40004A9		LD	1029	FILT L UNMI
04BE	01	A40004A7		M	1030	L K2NUM
04C0	01	DC0004AA		STO	1031	L TEMP
04C2	0	C235		LD	1032	2 VT180
04C3	01	D40004A9		STO	1033	L UNMI
04C5	01	A40004A7		M	1034	L K2NUM
04C7	01	8C0004AA		AD	1035	L TEMP
04C9	01	DC0004AA		STO	1036	L TEMP
04CB	01	C40004A8		LD	1037	L YNMI
04CD	01	A40004A6		M	1038	L K1NUM
04CF	01	8C0004AA		AD	1039	L TEMP
04D1	01	AC0004A5		D	1040	L KLAGD
04D3	01	D40004A8		STO	1041	L YNMI
04D5	0	D235		STO	1042	2 VT180
04D6			043	DONOZ EQU *		
04D6	0	C2B6		LD	1044	2 VT074
04D7	01	9400046A		S	1045	L ONE
04D9	01	4C2004E0		BNZ	1046	GT10
04DB	0	C201		LD	1047	2 VT128
04DC	01	D4000507		STO	1048	L NAB
04DE	01	4C0004E3		B	1049	L CALA8
04E0	0	C21E		LD	1050	GT10 2 VT157
04E1	01	D4000507		STO	1051	L NAB
04E3	01	C4000507		LD	1052	CALA8 L NAB
04E5	01	940000F9		S	1053	L =14025
04E7	01	4C1004E0		BNN	1054	GT11
04E9	0	C2DE		LD	1055	2 VT034
04EA	0	D2AF		STO	1056	2 VT081
04EB	01	4C000509		B	1057	L CONT
04ED	01	940000FA		S	1058	GT11 L =2475
04EF	01	4C2804F5		BN	1059	GT12
04F1	0	C2DD		LD	1060	2 VT035
04F2	0	D2AF		STO	1061	2 VT081
04F3	01	4C000509		B	1062	L CONT
04F5	0	C2DE		LD	1063	GT12 2 VT034
04F6	0	92DD		S	1064	2 VT035
04F7	01	D4000503		STO	1065	L ANOZN
04F9	01	C40000F5		LD	1066	L =16500
04FB	01	94000507		S	1067	L NAB
04FD	01	A4000508		M	1068	L ANOZN
04FF	01	AC0000FA		D	1069	L =2-75
0501	0	B2DD		A	1070	2 VT035
0502	0	D2AF		STO	1071	2 VT081
0503	01	4C000509		B	1072	L CONT
					1073	LORG
0505	0	000D		+	1074	DC 13
0506	0	007B		+	1075	DC 123
0507	0	0000		NAB	1076	DC **

Table B-11. Honeywell Control Program (Continued)

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0508 0 0000	1077	ANOZN DC	**	
0509	1078	CONT EQU	*	HWS0 7700
0509 00 65000000	1079	XRI LDX L1	**	HWS0 7710
0508 01 4C800000	1080	SSC I	HWECT	HWS0 7720
FFB9	1081	VT071 EQU	-71	HWS0 7730
FFB8	1082	VT072 EQU	-72	HWS0 7740
FFB7	1083	VT073 EQU	-73	HWS0 7750
FFB6	1084	VT074 EQU	-74	HWS0 7760
FFAF	1085	VT081 EQU	-81	HWS0 7770
FFAE	1086	VT082 EQU	-82	HWS0 7780
FFAD	1087	VT083 EQU	-83	HWS0 7790
001E	1088	VT157 EQU	+30	HWS0 7800
0035	1089	VT180 EQU	+53	HWS0 7810
0001	1090	VT128 EQU	+1	HWS0 7820
FF9A	1091	VT107 EQU	-102	HWS0 7830
FF94	1092	VT108 EQU	-107	HWS0 7840
FF9F	1093	VT097 EQU	-97	HWS0 7850
FFDC	1094	VT036 EQU	-36	HWS0 7860
FFDB	1095	VT037 EQU	-37	HWS0 7870
FFDA	1096	VT038 EQU	-38	HWS0 7880
FFD9	1097	VT039 EQU	-39	HWS0 7890
0023	1098	VT162 EQU	+35	HWS0 7900
0024	1099	VT163 EQU	+36	HWS0 7910
0025	1100	VT164 EQU	+37	HWS0 7920
0026	1101	VT165 EQU	+38	HWS0 7930
0027	1102	VT166 EQU	+39	HWS0 7940
0028	1103	VT167 EQU	+40	HWS0 7950
0029	1104	VT168 EQU	+41	HWS0 7960
002A	1105	VT169 EQU	+42	HWS0 7970
002B	1106	VT170 EQU	+43	HWS0 7980
002C	1107	VT171 EQU	+44	HWS0 7990
002D	1108	VT172 EQU	+45	HWS0 8000
002E	1109	VT173 EQU	+46	HWS0 8010
002F	1110	VT174 EQU	+47	HWS0 8020
0030	1111	VT175 EQU	+48	HWS0 8030
0031	1112	VT176 EQU	+49	HWS0 8040
004F	1113	VT206 EQU	+79	
0050	1114	VT207 EQU	+80	
0045	1115	VT196 EQU	+69	
0046	1116	VT197 EQU	+70	
0047	1117	VT198 EQU	+71	
0048	1118	VT199 EQU	+72	
0049	1119	VT200 EQU	+73	
004A	1120	VT201 EQU	+74	
004B	1121	VT202 EQU	+75	
004C	1122	VT203 EQU	+76	
004D	1123	VT204 EQU	+77	
004E	1124	VT205 EQU	+78	
FFF4	1125	VT012 EQU	-12	
FFF3	1126	VT013 EQU	-13	
FFF2	1127	VT014 EQU	-14	
FFF1	1128	VT015 EQU	-15	
FFF0	1129	VT016 EQU	-16	
FFEF	1130	VT017 EQU	-17	
FFEE	1131	VT018 EQU	-18	
FFED	1132	VT019 EQU	-19	
FFLC	1133	VT020 EQU	-20	

Table B-11. Honeywell Control Program (Concluded)

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FFEB	1134	VT021	EQU	-21
FFEA	1135	VT022	EQU	-22
FFE9	1136	VT023	EQU	-23
FFD8	1137	VT040	EQU	-40
FFD7	1138	VT041	EQU	-41
FFD6	1139	VT042	EQU	-42
FFD5	1140	VT043	EQU	-43
FFD4	1141	VT044	EQU	-44
FFD3	1142	VT045	EQU	-45
FFD2	1143	VT046	EQU	-46
FFD1	1144	VT047	EQU	-47
FFD0	1145	VT048	EQU	-48
FFCF	1146	VT049	EQU	-49
FFCE	1147	VT050	EQU	-50
FFC3	1148	VT061	EQU	-61
FFC2	1149	VT062	EQU	-62
FFC1	1150	VT063	EQU	-63
FFC0	1151	VT064	EQU	-64
FFBF	1152	VT065	EQU	-65
FFBE	1153	VT066	EQU	-66
FFBD	1154	VT067	EQU	-67
FFBC	1155	VT068	EQU	-68
FFBB	1156	VT069	EQU	-69
FFDE	1157	VT034	EQU	-34
FFDD	1158	VT035	EQU	-35
FFB5	1159	VT075	EQU	-75
FFB4	1160	VT076	EQU	-76
FFB3	1161	VT077	EQU	-77
FFB2	1162	VT078	EQU	-78
FFAC	1163	VT084	EQU	-84
FFAB	1164	VT085	EQU	-85
FFAA	1165	VT086	EQU	-86
FFA9	1166	VT087	EQU	-87
FFA8	1167	VT088	EQU	-88
FFA7	1168	VT089	EQU	-89
FFB1	1169	VT079	EQU	-79
FFB0	1170	VT080	EQU	-80
FFA6	1171	VT090	EQU	-90
050E	1172	END		

000 ERROR(S) AND 000 WARNING(S) IN ABOVE ASSEMBLY.

Table B-12. Honeywell Control Program Cross Reference

SYMBOL	VALUE	REL	DEFN	REFERENCES-																
AMPZN	0508	1	1077	1065M	1068R															
HUMP1	010C	1	218	245R	292R	339R														
CALAB	04E3	1	1052	1049R																
CUNT	0509	1	1078	1057R	1C62R	1072R														
CX3	0362	1	733	687M	689R	712M	714R	748M	750R	765R	774R	783R								
CX2	0363	1	734	690M	715M	751M	768R	777R	786R											
C1	00B9	1	138	134M	153M	162M	164R	174M	176R	185M	187R	228R	275R	322R						
C11	010F	1	221	229M	236R															
C12	0145	1	268	276M	283R															
C13	0178	1	315	323M	330R															
C2	00BA	1	139	136M	155M	165M	177M	188M	230R	277K	324R									
C21	0110	1	222	231M	239R															
C22	0146	1	269	278M	286R															
C23	017C	1	314	325M	333R															
DOMM2	04D6	1	1043	1028R																
LT	028E	1	565	455R	577R	591R														
ENDK	0288	1	559	510M	521R	575R	652R													
ENK1	0289	1	560	527M	576R	653M														
ENK	028F	1	566	476M	479R	528M	531R	580R	581M	586M	602R	607M	610R	615M	799R					
ENKL	0290	1	567	480M	532M	604R	606R	611R	614R											
EPDK	028A	1	561	513M	523R	654R														
EPDK1	028B	1	562	524M	655M															
FPK	0291	1	568	483M	486R	535M	538R	617R	622M	625R	630M									
EPKL	0292	1	569	487M	539M	619R	621R	626R	629R											
FTDK	028C	1	565	516M	525R	589R	656R													
ETDK1	028D	1	564	526M	590R	657M														
ETK	0293	1	570	489M	492R	544R	594R	595M	600M	632R	637M	640R	645M	815R						
ETKL	0294	1	571	493M	545M	634R	636R	641R	644R											
FILT	04BC	1	1029	1022R																
FREQE	0417	1	891	859R																
FREQP	043B	1	923	919R																
FREQT	0441	1	920	926R																
FT4W	0181	1	363																	
FT4WC	01E3	1	400	383R																
FUELM	0181	1	362	249R	296R	343R														
GT10	04E0	1	1050	1046R																
GT11	04ED	1	1058	1054R																
GT12	04F5	1	1063	1059M																
HRECT	0000	1	2	1R	1030R															
IASCW	FEBF			C-COMMON																
IBTO	FF00			C-COMMON																
IDUMY	FFFF			C-COMMON																
INTEG	0295	1	573	518R	548R															
IN1F	0114	1	227	137R	168R															
IN2F	014A	1	274	180R																
IN3F	0180	1	321	156R	191R															
ISW	01FD	1	429	116M	431R	433M														
IVT00	FF80			C-COMMON																
J0UMY	FF7F			C-COMMON																
KEFN1	0406	1	868	243M	290M	337M	803R	892R												
KEFN2	0407	1	869	805R	897R															
KEFN3	0408	1	870	903R																
KEFN4	0409	1	871	809R	909R															
KEF11	00FC	1	201	17M	235R															
KEF12	00FD	1	202																	
KEF13	00FE	1	203																	
KEF14	00FF	1	204	24M																

Table B-12. Honeywell Control Program Cross Reference
(Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-		
KEF21	0135	1	251	27M	238R	282R
KEF22	0136	1	252			
KEF23	0137	1	253			
KEF24	0138	1	254	34M		
KEF31	016B	1	298	37M	285R	329R
KEF32	016C	1	299			
KEF33	016D	1	300			
KEF34	016E	1	301	44M		
KEF41	01A1	1	345	47M	332R	
KEF42	01A2	1	346			
KEF43	01A3	1	347			
KEF44	01A4	1	348	54M		
KLGD	04A5	1	1011	1040R		
KTFN1	040D	1	875	817R	931R	
KTFN2	040E	1	876	819R	936R	
KTFN3	040F	1	877	821R	943R	
KTFN4	0410	1	878	823R	950R	
KTF11	0103	1	209	57M		
KTF12	0104	1	210	60M		
KTF13	0105	1	211	63M		
KTF14	0106	1	212	66M		
KTF21	013C	1	259	71M		
KTF22	013D	1	260	73M		
KTF23	013E	1	261	76M		
KTF24	013F	1	262	79M		
KTF31	0172	1	305	84M		
KTF32	0173	1	307	86M		
KTF33	0174	1	308	89M		
KTF34	0175	1	309	92M		
KTF41	01A8	1	353	96M		
KTF42	01A9	1	35	98M		
KTF43	01AA	1	355	101M		
KTF44	01AB	1	356	104M		
K1NUM	04A6	1	1012	1038R		
K1THD	01FB	1	427	374M	391M	447R
K2NUM	04A7	1	1013	1030R	1034R	
LUP1	011D	1	234	248M		
LUP2	0153	1	281	295M		
LUP3	0189	1	328	342M		
MDSWT	046E	1	967	960R		
MDW1	0309	1	672	664M		
MDW2	031B	1	682	674R		
MDW3	033E	1	707	684M		
MDW4	036C	1	744	709M		
MDW5	0388	1	764	703R	728R	
MDW6	03A7	1	794	671R	681R	791R
MDW9	024B	1	494	473R		
MEAST	FEFF		C-COMMON			
MEPT	03D7	1	830			
ME1	03FE	1	860	833M	893R	
ME2	03FF	1	861	837M	898R	
ME3	0400	1	862	841M	904R	
ME4	0401	1	863	844M	910R	
MICK	03AD	1	127	13R		
MIKE1	0497	1	1000	996R		
MIKE2	04A1	1	1008	1004R		
MINFL	048A	1	989	982M		
MINSS	048D	1	991	988R		
MT1	0402	1	854	848M	932R	

**Table B-12. Honeywell Control Program Cross Reference
(Continued)**

SYMBOL	VALUE	REL	DEFN	REFERENCES-															
MT2	0403	1	865	851M	937R														
MT3	0404	1	866	855M	944R														
MT4	0405	1	867	858M	951R														
MW1	02BE	1	609	603M															
MW2	02C5	1	616	605R	608R	612R													
MW3	02CF	1	624	618M															
MW4	02D6	1	631	620R	623R	627R													
MW5	02E0	1	639	633M															
MW6	02E3	1	651	635R	638R	642R	646R												
NAR	Q507	1	1076	1048M	1051M	1052R	1067R												
NE XT	01C0	1	384	366R															
NGFT	010E	1	220	247R	294R	341R													
NIN	00BB	1	140	132M	151M	167M	179M	190M											
NML	02A3	1	589	584R															
NM2	02B4	1	602	598R															
NPL1	035E	1	729	662R															
NPL2	035F	1	730	682R															
NPL3	0360	1	731	707R															
NPL4	0361	1	732	672R	744R														
ONE	046A	1	963	519R	583R	997R	1045R												
PLA	U2F7	1	661																
P3E1	0101	1	206	665R	691R														
P3E2	013A	1	256	693R	716R														
P3E3	0170	1	303	718R	752R														
P3E4	01A6	1	350	675R	754R														
P3L	0364	1	735	692M	717M	753M	764R												
P3M	0365	1	736	694M	719M	735M	767R												
P3PL	040B	1	873	666M	676M	772M	797R	840R											
P3TNB	0412	1	880	511R	813R	850R													
P3T1	0108	1	214	21M															
P3T2	0141	1	264	31M															
P3T3	0177	1	311	41M															
P3T4	01AD	1	358	51M															
P5E1	0102	1	207	667R	695R														
P5E2	013B	1	257	697R	720R														
P5E3	0171	1	304	722R	756R														
P5E4	01A7	1	351	677R	758R														
P5L	0366	1	737	696M	721M	757M	773R												
P5M	0367	1	738	698M	723M	759M	776R												
P5PL	040C	1	874	668M	678M	781M	795R	836R											
P5TNB	0413	1	881	811R	847R														
P5T1	0109	1	215	120M															
P5T2	0142	1	265	122M															
P5T3	0178	1	312	124M															
P5T4	01AE	1	359	126M															
RQA1B	04AD	1	1020	999R	1007R	1010R													
SENL	0238	1	480	477R															
SEPL	0240	1	487	484R															
SETL	0249	1	493	490R															
SETX1	010D	1	219	232R	279R	326R													
STENL	0269	1	532	529R															
STEPL	0271	1	539	536R															
STETL	027A	1	545	542R															
STP1	01FE	1	430	400R															
STP2	020E	1	442	432R															
STP3	0219	1	452	441R															
SUMEF	0415	1	884	896M	901R	902M	907R	908M	913P	914M	917R	918M	968R						
SUMPF	043A	1	922	925M	970R														
SUMTF	0440	1	923	935M	941R	942M	948R	949M	954R	955M	958R	954M	972R						

Table B-12. Honeywell Control Program Cross Reference
(Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-							
SUMX	036A	1	741	766M 769R	775M	778R	784M	787R			
SUM1	0112	1	224	237M 240R							
SUM2	0148	1	271	284M 287R							
SUM3	017E	1	318	331M 334R							
SWLAG	04AC	1	1019	108M 1021R	1024M						
TAU2T	01FC	1	428	382M 399M		466R					
TBBN	0414	1	802	514R 825R		854R					
T81	010A	1	216	112M							
T82	0143	1	265	114M							
T83	0179	1	313	116M							
T84	01AF	1	360	118M							
TEMP	04AA	1	1016	1031M	1035R	1036M	1039R				
TESTN	0007	1	10	6R							
THREE	046C	1	965	1008R							
TIME	0287	1	558	109M	517R	547M					
TIN1	00CD	1	157	149R							
TIN2	00DB	1	169	159M							
TIN3	00E9	1	181	171M							
THAX	00C1	1	147	130R							
TMPP	01F2	1	414	369M	372R	377M	380R	386M	389R	394M	397R
TST1	0111	1	223	233M	234R	244R	246M				
TST2	0147	1	270	250M	281R	291R	293M				
TST3	017D	1	317	327M	328R	338R	340M				
TWD	046B	1	964	597R	1005R						
T4WF	01FA	1	426	469M	515R	853R					
UNM1	04A9	1	1015	1027M	1029R	1033M					
VT012	FFF4	0	1125	15R							
VT013	FFF3	0	1126	18R							
VT014	FFF2	0	1127	20R							
VT015	FFF1	0	1128	22R							
VT016	FFF0	0	1129	25R							
VT017	FFEF	0	1130	28R							
VT018	FFEE	0	1131	30R							
VT019	FFED	0	1132	32R							
VT020	FFEC	0	1133	35R							
VT021	FFEB	0	1134	38R							
VT022	FFEA	0	1135	40R							
VT023	FFE9	0	1136	42R							
VT034	FFDE	0	1157	1055R	1063R						
VT035	FFDD	0	1158	1060R	1064R	1070R					
VT036	FFDC	0	1094	475R	527R						
VT037	FFDB	0	1095	481R	533R						
VT038	FFDA	0	1096	488R	540R						
VT039	FFD9	0	1097	11R	14M	471R	474M				
VT040	FFD8	0	1137	45R							
VT041	FFD7	0	1138	48R							
VT042	FFD6	0	1139	50R							
VT043	FFD5	0	1140	52R							
VT044	FFD4	0	1141	55R							
VT045	FFD3	0	1142	58R							
VT046	FFD2	0	1143	61R							
VT047	FFD1	0	1144	64R							
VT048	FFD0	0	1145	67R							
VT049	FFCF	0	1146	69R							
VT050	FFCE	0	1147	72R							
VT061	FFC3	0	1148	74R							
VT062	FFC2	0	1149	77R							
VT063	FFC1	0	1150	80R							
VT064	FFC0	0	1151	82R							

Table B-12. Honeywell Control Program Cross Reference
(Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-									
VT065	FFBF	0	1152	65R									
VT066	FFBF	0	1153	87R									
VT067	FFBD	0	1154	90R									
VT068	FFBC	0	1155	93R									
VT069	FFBB	0	1156	95R									
VT071	FFB9	0	1081	969M	978R	979R	995R						
VT072	FFB8	0	1082	971M	974R	975R							
VT073	FFB7	0	1083	973M	1003R								
VT074	FFB6	0	1084	520M	582R	596R	998M	1006M	1009M	1044R			
VT075	FFB5	0	1159	97R									
VT076	FFB4	0	1160	99R									
VT077	FFB3	0	1161	102R									
VT078	FFB2	0	1162	105R									
VT079	FFB1	0	1169										
VT080	FFB0	0	1170	460M									
VT081	FFAF	0	1085	1056M	1061M	1071M							
VT082	FFAE	0	1086	111R									
VT083	FFAD	0	1087	113R									
VT084	FFAC	0	1163	115R									
VT085	FFAB	0	1164	117R									
VT086	FFAA	0	1165	119R									
VT087	FFA9	0	1166	121R									
VT088	FFAB	0	1167	123R									
VT089	FFA7	0	1168	125R									
VT090	FFA6	0	1171	462M									
VT097	FF9F	0	1093	438R	444R								
VT102	FF9A	0	1091	364R	370R	375R	387R	392R	512R	839R	849R		
VT108	FF94	0	1092	835R	846R								
VT128	0001	0	1090	508R	663R	673R	683R	708R	745R	832R	1047R		
VT157	001E	0	1088	128R	148R	158R	170R	182R	509R	831R	1050R		
VT162	0023	0	1098	796M									
VT163	0024	0	1099	798M									
VT164	0025	0	1100	800M									
VT165	0026	0	1101	802M									
VT166	0027	0	1102	804M									
VT167	0028	0	1103	806M									
VT168	0029	0	1104	808M									
VT169	002A	0	1105	810M									
VT170	002B	0	1106	812M									
VT171	002C	0	1107	814M									
VT172	002D	0	1108	816M									
VT173	002E	0	1109	818M									
VT174	002F	0	1110	820M									
VT175	0030	0	1111	822M									
VT176	0031	0	1112	824M									
VT180	0035	0	1089	981M	985M	986R	990M	992R	1000R	1025R	1032R	1042M	
VT196	0045	0	1115	834M									
VT197	0046	0	1116	838M									
VT198	0047	0	1117	842M									
VT199	0048	0	1118	852M									
VT200	0049	0	1119	856M									
VT201	004A	0	1120	859M									
VT202	004B	0	1121	826M									
VT203	004C	0	1122	470M									
VT204	004D	0	1123	912M									
VT205	004E	0	1124	940M									
VT206	004F	0	1113	947M									
VT207	0050	0	1114	953M									
WEFL	0368	1	739	700M	725M	761M	782R						

Table B-12. Honeywell Control Program Cross Reference
(Concluded)

SYMBOL	VALUE	REL	DEFN	REFERENCES-				
WEFM	0369	1	740	702M	727M	762M	785R	
WEFN	040A	1	872	670M	680M	790M	801R	915R
WEF1	0100	1	205	19M	669R	699R		
WEF2	0139	1	255	29M	701R	724R		
WEF3	016F	1	302	39M	724R	760R		
WEF4	01A5	1	349	49M	679R	702R		
WFNN	0415	1	883	987R	989R			
WFNN1	010B	1	217					
WFNN2	0144	1	267					
WFNN3	017A	1	314					
WFNN4	0180	1	361					
WFNO0	046D	1	966	977M				
WTFN	0411	1	879	807R	956R			
WTF-1	0107	1	213	68M				
WTF2	0140	1	263	81M				
WTF3	0176	1	310	94M				
WTF4	01AC	1	357	106M				
XRI	0509	1	1079	3M				
XT4	01F4	1	417	440M	446R	458R	459M	467R
XT4D	01F6	1	420	436M	451M	452R	463R	
XT4D1	01F8	1	423	437M	453R	464M		
YNM1	04A8	1	1014	1026M	1037R	1041M		
HWECT								
DMP FUNCTION COMPLETED								
*SYORE			HWECT					
HWECT								
DMP FUNCTION COMPLETED								

Table B-13. Bendix Bounds Program

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// JOB          VDIEX    17 JUL 74 15.768 HRS
// DMP          17 JUL 74 15.766 HRS
*DELETE        GTECT
DMP FUNCTION COMPLETED
// ASM GTECT   17 JUL 74 15.769 HRS
*OVERFLOW SECTORS ,,,9
*LIST
*YREF
*ONE WORD INTEGERS
*COMMON IDUMY(127),IVT00,JDUMY(127),IBFO,MEAST(64),IASCH(2)
0000 0 078C50E3      1      ENT      GTECT      HWE00070
0000 0 0000          2      GTECT  DC      *--*      HWE00080
0001 01 6D000568     3      STX      L1 XR1+1      HWE00090
0003 01 6E00056A     4      STX      L2 XR2+1      HWE00100
0005 01 6F00056C     5      STX      L3 XR3+1      HWE00110
0007 03 67C0FE00     6      *
0007 03 67C0FE00     7      LDX      L3 MEAST-63      HWE00120
0009 03 6600FF80     8      LDX      L2 IVT00      HWE00130
000B 00 65000000     9      LDX      L1 0      HWE00140
000D 0  C03F         10     LD       =0      HWE00160
000D 0  C03F         11     *
000E 01 4C000147     12     B       L START      HWE00180
000E 01 4C000147     13     RSTAL  EQU      *      RESEI. ALL DIGITAL ADJUST HWE00190
0010 0  C03E         14     LD       ST001      HWE00200
0011 0  D2FF         15     STO      2 VT001      HWE00210
0012 0  C03D         16     LD       ST002      HWE00220
0013 0  D2FE         17     STO      2 VT002      HWE00230
0014 0  C03C         18     LD       ST003      HWE00240
0015 0  D2FD         19     STO      2 VT003      HWE00250
0016 0  C03B         20     LD       ST004      HWE00260
0017 0  D2FC         21     STO      2 VT004      HWE00270
0018 0  C03A         22     LD       ST005      HWE00280
0019 0  D2EB         23     STO      2 VT005      HWE00290
001A 0  C039         24     LD       ST006      HWE00300
001B 0  D2EA         25     STO      2 VT006      HWE00310
001C 0  C038         26     LD       ST007      HWE00320
001D 0  D2E9         27     STO      2 VT007      HWE00330
001E 0  C037         28     LD       ST008      HWE00340
001F 0  D2E8         29     STO      2 VT008      HWE00350
0020 0  C036         30     LD       ST009      HWE00360
0021 0  D2E7         31     STO      2 VT009      HWE00370
0022 0  C035         32     LD       ST010      HWE00380
0023 0  D2E6         33     STO      2 VT010      HWE00390
0024 0  C034         34     LD       ST011      HWE00400
0025 0  D2E5         35     STO      2 VT011      HWE00410
0026 0  C033         36     LD       ST012      HWE00420
0027 0  D2E4         37     STO      2 VT012      HWE00430
0028 0  C032         38     LD       ST013      HWE00440
0029 0  D2E3         39     STO      2 VT013      HWE00450
002A 0  C031         40     LD       ST014      HWE00460
0029 0  D2F2         41     STO      2 VT014      HWE00470
002C 0  C030         42     LD       ST015      HWE00480
002D 0  D2F1         43     STO      2 VT015      HWE00490
002E 0  C02F         44     LD       ST016      HWE00500
002F 0  D2F0         45     STO      2 VT016      HWE00510
0030 0  C02E         46     LD       ST017      HWE00520
0031 0  D2EF         47     STO      2 VT017      HWE00530
0032 0  C02C         48     LD       ST018      HWE00540
0033 0  D2EE         49     STO      2 VT018      HWE00550
0034 0  C02C         50     LD       ST019      HWE00560

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Table B-13. Bendix Bounds Program (Continued)

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0035	0	D2ED	51	STO	2	VT019	HWE00570
0036	0	C02B	52	LD		ST020	HWE00580
0037	0	D2EC	53	STO	2	VT020	HWE00590
0038	0	C02A	54	LD		ST021	HWE00600
0039	0	D2EB	55	STO	2	VT021	HWE00610
003A	0	C029	56	LD		ST022	HWE00620
003B	0	D2EA	57	STO	2	VT022	HWE00630
003C	0	C028	58	LD		ST023	HWE00640
003D	0	D2E9	59	STO	2	VT023	HWE00650
003E	0	C027	60	LD		ST024	HWE00660
003F	0	D2E8	61	STO	2	VT024	HWE00670
0040	0	C026	62	LD		ST025	HWE00680
0041	0	D2E7	63	STO	2	VT025	HWE00690
0042	0	C025	64	LD		ST026	HWE00700
0043	0	D2E6	65	STO	2	VT026	HWE00710
0044	0	C024	66	LD		ST027	HWE00720
0045	0	D2E5	67	STO	2	VT027	HWE00730
0046	0	C023	68	LD		ST028	HWE00740
0047	0	D2E4	69	STO	2	VT028	HWE00750
0048	0	C022	70	LD		ST029	HWE00760
0049	0	D2E3	71	STO	2	VT029	HWE00770
004A	0	C021	72	LD		ST030	HWE00780
004B	0	D2E2	73	STO	2	VT030	HWE00790
004C	0	705C	74	B		STTVT	HWE00800
			75	LDRG			HWE00810
004D	0	0000	76	+	DC	0	
			77	*			
004E	0	0000	78	ST000	DC	0	SPEED CONTROL FIG10-364 HWE00820
004F	0	0000	79	ST001	DC	0	HWE00830
0050	0	0000	80	ST002	DC	0	IDLE SPEED TRIM HWE00840
							MAX SPEED TRIM HWE00850
0051	0	4E20	81	ST003	DC	20000	HWE00860
0052	0	0000	82	ST004	DC	0	BRANCH COMMAND 64+ HWE00870
0053	0	1000	83	ST005	DC	4096	N INTEGRATION INC HWE00880
0054	0	1388	84	ST006	DC	3000	N INT PRESS GAIN HWE00890
0055	0	F000	85	ST007	DC	-4096	N INT DECREASE HWE00900
0056	0	EC78	86	ST008	DC	-5000	N INT DEC PRESS GAIN HWE00910
			87	*			HWE00920
			88	+			
0057	0	0000	89	ST009	DC	0	FIG10-5 PROP. TEMPERATURE CONTROL HWE00930
0058	0	2AF8	90	ST010	DC	11000	SPEED CONTROL SELECTION HWE00940
0059	0	0000	91	ST011	DC	0	ZERO FLOW ADJUST HWE00950
			92	*			HWE00960
			93	*			HONEYWELL ST VALUES
005A	0	F290	94	ST012	DC	-3440	N GAIN (50,E)
005B	0	0207	95	ST013	DC	519	WF (50,E)
005C	0	0992	96	ST014	DC	2450	PT3 BOND (50,P)
005D	0	01B0	97	ST015	DC	432	EN GAIN (50,E)
005E	0	F820	98	ST016	DC	-2016	N GAIN (70,E)
005F	0	02B5	99	ST017	DC	693	WF (70,E)
0060	0	0FA0	100	ST018	DC	4000	PT3 BOND (70,P)
0061	0	0190	101	ST019	DC	400	EN GAIN (70,E)
0062	0	F860	102	ST020	DC	-1952	N GAIN (85,E)
0063	0	03A6	103	ST021	DC	934	WF (85,E)
0064	0	1090	104	ST022	DC	6300	PT3 BOND (85,P)
0065	0	0380	105	ST023	DC	896	EN GAIN (85,E)
			106	*			
			107	*			END HONEYWELL ST VALUES

Table B-13. Bendix Bounds Program (Continued)

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		108	*				
0066	0	1770	109	ST024	DC	6000	ZERO N RATIOS INTERCEPT HWE01140
0067	0	A240	110	ST025	DC	-24000	BACK SLOPE SPEED BREAK PT HWE01150
0068	0	4000	111	ST026	DC	16384	
		112	*				HWE01170
		113	*				HWE01180
							FIGURE10-8 RATIOS INTEGRATION
0069	0	7FF8	114	ST027	DC	32760	
006A	0	0000	115	ST028	DC	0	
006B	0	0000	116	ST029	DC	0	MINIMUM RATIOS SLOPE HWE01210
006C	0	5014	117	ST030	DC	20500	MINIMUM RATIOS LEVEL HWE01220
006D	0	0000	118	ST031	DC	0	HWE01230
006E	0	7FF8	119	ST032	DC	32760	VALVE MAXIMUM POSITION HWE01240
006F	0	0000	120	ST033	DC	0	VALVE MINIMUM POSITION HWE01250
0070	0	25A8	121	ST034	DC	9640	
0071	0	0A5A	122	ST035	DC	2650	
		123	*				
		124	*				HONEYWELL ST VALUES
		125	*				
0072	0	0640	126	ST036	DC	1600	
0073	0	0640	127	ST037	DC	1600	
0074	0	567D	128	ST038	DC	22141	
0075	0	0010	129	ST039	DC	16	
0076	0	F0A0	130	ST040	DC	-3936	N GAIN (100,E)
0077	0	0670	131	ST041	DC	1648	MF (100,E)
0078	0	1FA4	132	ST042	DC	8100	PT3 BOND (100,P)
0079	0	0930	133	ST043	DC	2352	EN GAIN (100,E)
007A	0	22D0	134	ST044	DC	8912	PT5 GAIN (50 T)
007B	0	FB66	135	ST045	DC	-1178	PT3 GAIN (50 T)
007C	0	F970	136	ST046	DC	-1680	T4W GAIN (50 T)
007D	0	0340	137	ST047	DC	832	ET GAIN (50 T)
007E	0	028B	138	ST048	DC	651	WTF (50 T)
007F	0	3A80	139	ST049	DC	14976	PT5 GAIN (70 T)
0080	0	6009	140	ST050	DC	24585	PT3 GAIN (70 T)
		141	*				
		142	*				END HONEYWELL ST VALUES
		143	*				
		144	*				
		145	*				FIGURE10-12 1GV & BLEED CONTR HWE01460
0081	0	0000	146	ST051	DC	0	LOW N TRIM OF 1GV HWE01470
0082	0	3E80	147	ST052	DC	16000	HIGH N TRIM OF 1GV HWE01480
0083	0	0000	148	ST053	DC	0	LOW N TRIM OF BLEEDS HWE01490
0084	0	3E80	149	ST054	DC	16000	HIGH N TRIM OF BLEEDS HWE01500
		150	*				HWE01510
		151	*				FIGURE10-14 NOZZLE CONTROL HWE01520
0085	0	105E	152	ST055	DC	4190	NOZZLE FLAT HWE01530
0086	0	40D8	153	ST056	DC	16600	T5 REQUEST HWE01550
0087	0	4000	154	ST057	DC	16384	T5 CONTROL GAIN HWE01560
0088	0	0000	155	ST058	DC	0	HWE01570
0089	0	0000	156	ST059	DC	0	HWE01580
008A	0	0000	157	ST060	DC	0	HWE01590
008B	0	F7CE	158	ST061	DC	-2098	T4W GAIN (70 T)
008C	0	0410	159	ST062	DC	1040	ET GAIN (70 T)
008D	0	03E8	160	ST063	DC	1000	WTF (70 T)
008E	0	EA90	161	ST064	DC	-5488	PT5 GAIN (85 T)
008F	0	108B	162	ST065	DC	4235	PT3 GAIN (85 T)
0090	0	FA95	163	ST066	DC	-1387	T4W GAIN (85 T)
0091	0	04A0	164	ST067	DC	1184	ET GAIN (85 T)

Table B-13. Bendix Bounds Program (Continued)

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0092	0	0898	165	ST068	DC	2200	WTF (85 T)
0093	0	17EF	166	ST069	DC	6127	PT5 GAIN (100 T)
0094	0	C000	167	ST070	DC	0	
0095	0	0000	168	ST071	DC	0	
0096	0	0000	169	ST072	DC	0	
0097	0	0000	170	ST073	DC	0	
0098	0	0000	171	ST074	DC	0	
0099	0	DF6E	172	ST075	DC	-8338	PT3 GAIN (100 T)
009A	0	FFC0	173	ST076	DC	-64	T4W GAIN (100 T)
009B	0	06C0	174	ST077	DC	1728	ET GAIN (100 T)
009C	0	C8B8	175	ST078	DC	3000	WTF (100 T)
009D	0	0000	176	ST079	DC	0	
009E	0	0000	177	ST080	DC	0	
009F	0	0000	178	ST081	DC	0	
00A0	0	299A	179	ST082	DC	10650	
00A1	0	2666	180	ST083	DC	9830	
00A2	0	30AC	181	ST084	DC	12460	
00A3	0	2EE0	182	ST085	DC	12000	
00A4	0	05DC	183	ST086	DC	1500	
00A5	0	C690	184	ST087	DC	1680	
00A6	0	0898	185	ST088	DC	2200	
00A7	0	0B54	186	ST089	DC	2900	
00A8	0	0000	187	ST090	DC	0	
00A9			188	STTVT	EQU	*	HWE01600
00A9	0	C0C3	189	LD		ST031	HWE01610
00AA	0	D2E1	190	STO	2	VT031	HWE01620
00AB	0	C0C2	191	LD		ST032	HWE01630
00AC	0	D2E0	192	STO	2	VT032	HWE01640
00AD	0	C0C1	193	LD		ST033	HWE01650
00AE	0	D2DF	194	STO	2	VT033	HWE01660
00AF	0	C0C0	195	LD		ST034	HWE01670
00B0	0	D2DE	196	STO	2	VT034	HWE01680
00B1	0	C08F	197	LD		ST035	HWE01690
00B2	0	D2DD	198	STO	2	VT035	HWE01700
00B3	0	C08E	199	LD		ST036	HWE01710
00B4	0	D2DC	200	STO	2	VT036	HWE01720
00B5	0	C08D	201	LD		ST037	HWE01730
00B6	0	D2DB	202	STO	2	VT037	HWE01740
00B7	0	C08C	203	LD		ST038	HWE01750
00B8	0	D2DA	204	STO	2	VT038	HWE01760
00B9	0	C08B	205	LD		ST039	HWE01770
00BA	0	D2D9	206	STO	2	VT039	HWE01780
00BB	0	C08A	207	LD		ST040	HWE01790
00BC	0	D2D8	208	STO	2	VT040	HWE01800
00BD	0	C089	209	LD		ST041	HWE01810
00BE	0	D2D7	210	STO	2	VT041	HWE01820
00BF	0	C088	211	LD		ST042	HWE01830
00C0	0	D2D6	212	STO	2	VT042	HWE01840
00C1	0	C087	213	LD		ST043	HWE01850
00C2	0	D2D5	214	STO	2	VT043	HWE01860
00C3	0	C086	215	LD		ST044	HWE01870
00C4	0	D2D4	216	STO	2	VT044	HWE01880
00C5	0	C085	217	LD		ST045	HWE01890
00C6	0	D2D3	218	STO	2	VT045	HWE01900
00C7	0	C084	219	LD		ST046	HWE01910
00C8	0	D2D2	220	STO	2	VT046	HWE01920
00C9	0	C083	221	LD		ST047	HWE01930

Table B-13. Bendix Bounds Program (Continued)

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00CA	0	D2D1	222	STO	2	VT047	HWE01940
00CB	0	C0B2	223	LD		ST048	HWE01950
00CC	0	D2D0	224	STO	2	VT048	HWE01960
00CD	0	C0B1	225	LD		ST049	HWE01970
00CE	0	D2CF	226	STO	2	VT049	HWE01980
00CF	0	C0B0	227	LD		ST050	HWE01990
00D0	0	D2CE	228	STO	2	VT050	HWE02000
00D1	0	C0AF	229	LD		ST051	HWE02010
00D2	0	D2CD	230	STO	2	VT051	HWE02020
00D3	0	C0AE	231	LD		ST052	HWE02030
00D4	0	D2CC	232	STO	2	VT052	HWE02040
00D5	0	C0AD	233	LD		ST053	HWE02050
00D6	0	D2CB	234	STO	2	VT053	HWE02060
00D7	0	C0AC	235	LD		ST054	HWE02070
00D8	0	D2CA	236	STO	2	VT054	HWE02080
00D9	0	C0AB	237	LD		ST055	HWE02090
00DA	0	D2C9	238	STO	2	VT055	HWE02100
00DB	0	C0AA	239	LD		ST056	HWE02110
00DC	0	D2C8	240	STO	2	VT056	HWE02120
00DD	0	C0A9	241	LD		ST057	HWE02130
00DE	0	D2C7	242	STO	2	VT057	HWE02140
00DF	0	C0A8	243	LD		ST058	HWE02150
00E0	0	D2C6	244	STO	2	VT058	HWE02160
00E1	0	C0A7	245	LD		ST059	HWE02170
00E2	0	D2C5	246	STO	2	VT059	HWE02180
00E3	0	C0A6	247	LD		ST060	HWE02190
00E4	0	D2C4	248	STO	2	VT060	HWE02200
00E5	0	C0A5	249	LD		ST061	
00E6	0	D2C3	250	STO	2	VT061	
00E7	0	C0A4	251	LD		ST062	
00E8	0	D2C2	252	STO	2	VT062	
00E9	0	C0A3	253	LD		ST063	
00EA	0	D2C1	254	STO	2	VT063	
00EB	0	C0A2	255	LD		ST064	
00EC	0	D2C0	256	STO	2	VT064	
00ED	0	C0A1	257	LD		ST065	
00EE	0	D2BF	258	STO	2	VT065	
00EF	0	C0A0	259	LD		ST066	
00F0	0	D2BE	260	STO	2	VT066	
00F1	0	C09F	261	LD		ST067	
00F2	0	D2BD	262	STO	2	VT067	
00F3	0	C09E	263	LD		ST068	
00F4	0	D2BC	264	STO	2	VT068	
00F5	0	C09D	265	LD		ST069	
00F6	0	D2BB	266	STO	2	VT069	
00F7	0	C09C	267	LD		ST070	
00F8	0	D2BA	268	STO	2	VT070	
00F9	01	C4000095	269	LD	L	ST071	
00FB	0	D2B9	270	STO	2	VT071	
00FC	01	C4000096	271	LD	L	ST072	
00FE	0	D2BB	272	STO	2	VT072	
00FF	01	C4000097	273	LD	L	ST073	
0101	0	D2B7	274	STO	2	VT073	
0102	01	C4000098	275	LD	L	ST074	
0104	0	D2B6	276	STO	2	VT074	
0105	01	C4000099	277	LD	L	ST075	
0107	0	D2B5	278	STO	2	VT075	

Table B-13. Bendix Bounds Program (Continued)

0108	01	C400009A	279	LD	L	ST076		
010A	0	D2B4	280	STO	2	VT076		
010F	01	C400009B	281	LD	L	ST077		
010D	0	D2B3	282	STO	2	VT077		
010E	01	C400009C	283	LD	L	ST078		
0110	0	D2B2	284	STO	2	VT078		
0111	01	C400009D	285	LD	L	ST079		
0113	0	D2B1	286	STO	2	VT079		
0114	01	C400009E	287	LD	L	ST080		
0116	0	D2B0	288	STO	2	VT080		
0117	01	C400009F	289	LD	L	ST081		
0119	0	D2AF	290	STO	2	VT081		
011A	01	C40000A0	291	LD	L	ST082		
011C	0	D2AE	292	STO	2	VT082		
011D	01	C40000A1	293	LD	L	ST083		
011F	0	D2AD	294	STO	2	VT083		
0120	01	C40000A2	295	LD	L	ST084		
0122	0	D2AC	296	STO	2	VT084		
0123	01	C40000A3	297	LD	L	ST085		
0125	0	D2AB	298	STO	2	VT085		
0126	01	C40000A4	299	LD	L	ST086		
0128	0	D2AA	300	STO	2	VT086		
0129	01	C40000A5	301	LD	L	ST087		
012B	0	D2A9	302	STO	2	VT087		
012C	01	C40000A6	303	LD	L	ST088		
012E	0	D2A8	304	STO	2	VT088		
012F	01	C40000A7	305	LD	L	ST089		
0131	0	D2A7	306	STO	2	VT089		
0132	01	C40000A8	307	LD	L	ST090		
0134	0	D2A6	308	STO	2	VT090		
0135	0	7054	309	B		DAC4L	BRANCH TO DAC4 OUTPUT LOOP	HWE02210
			310	*				HWE02220
0136		0000	311	CEON	BSS	E	0	HWE02230
0136	0	0000	312		DC		0	HWE02240
0137	0	E401	313		DC		/E401	HWE02250
			314	*				HWE02260
0138		0000	315	DIV64	BSS	E	0	HWE02270
0138	0	0000	316		DC		0	HWE02280
0139	0	5F40	317		DC		/5F40	HWE02290
			318	*				HWE02300
013A		0000	319	DIV40	BSS	E	0	HWE02310
013A	0	0000	320		DC		0	HWE02320
013B	0	DF40	321		DC		/DF40	HWE02330
			322	*				HWE02340
013C		0000	323	DO7E	BSS	E	0	HWE02350
013C	1	013E	324		DC		VALUE	HWE02360
013D	0	617E	325		DC		/617E	HWE02370
			326	*				HWE02380
013E	0	0000	327	VALUE	DC		---	HWE02390
013F	0	0000	328	NUM	DC		---	HWE02400
0140	0	0000	329	TMNR	DC		---	HWE02410
0141	0	0032	330	TRIMS	DC		50	HWE02420
0142	0	0000	331	TEMP3	DC		---	HWE02430
0143	0	0000	332	TEMP4	DC		---	HWE02440
0144	0	0000	333	TEMP5	DC		---	HWE02450
0145	0	0000	334	DKOUT	DC		---	HWE02460
0146	0	0000	335	DK20T	DC		---	HWE02470

Table B-13. Bendix Bounds Program (Continued)

0147		336	*		OBTAIN INPUT DATA	HWE02480
0147 0	08EE	337	START EQU	*		HWE02490
		338	XIO	CEON		HWE02500
		339	*		DIGITAL ADJUSTMENTS	HWE02510
0148 0	08EF	340	XIO	DIV64		HWE02520
0149 01	E40001E4	341	AND L	=/7F80		HWE02540
0148 0	D2A2	342	STO 2	VT094	DIGITAL ADJUSTMENT	HWE02530
014C 0	1807	343	SRA	7		HWE02550
014D 0	DOF1	344	STO	NUM		HWE02560
014E 01	940001E5	345	S L	=127		HWE02570
0150 0	DOEF	346	STO	TMNR		HWE02580
0151 01	4C300157	347	BP	PLUS		HWE02590
0153 01	C400004D	348	LD L	=0		HWE02600
0155 0	90E9	349	S	NUM		HWE02610
0156 0	DOE9	350	STO	TMNR		HWE02620
0157 01	65800140	351	PLUS LDX 11	TMNR		HWE02630
0159 03	C500FF80	352	LD L1	IVT00		HWE02640
015B 0	DOE2	353	STO	VALUE		HWE02650
015C 0	DOE6	354	STO	TEMP4		HWE02660
015D 01	4C30015	355	BP	RDOUT		HWE02670
015F 01	C40001D	356	LD L	=0		HWE02680
0161 0	90DC	357	S	VALUE		HWE02690
0162 01	EC0001E6	358	OR L	=/8000		HWE02700
0164 0	D0D9	359	STO	VALUE		HWE02710
0165		360	RDOUT EQU			HWE02720
0165 0	08D6	361	XIO	D07E		HWE02730
		362	*		VTXXX VALUE OUTPUT	HWE02740
		363	*		RESET AND SAFETY	HWE02750
0166 0	08D3	364	XIO	DIV40		HWE02760
0167 0	E07F	365	AND	=/7FFF		HWE02770
0168 0	D0D9	366	STO	TEMP3		HWE02780
0169 0	D29B	367	STO 2	VT101		HWE02790
016A 0	E07D	368	AND	=/4000		HWE02800
016B 01	4C300010	369	BP	RSTAL	RESET ALL ADJUSTMENTS	HWE02810
		370	*		SINGLE ADJUSTMENT RESET ROUTINE	HWE02820
016D 0	C29D	371	LD 2	VT099		HWE02830
016E 0	90D0	372	S	NUM		HWE02840
016F 01	4C18017	373	BZ	RSTSA		HWE02850
0171 0	C0CD	374	LD	NUM		HWE02860
0172 0	D29D	375	STO 2	VT099		HWE02870
0173 0	C0CF	376	LD	TEMP4		HWE02880
0174 0	D29C	377	STO 2	VT100		HWE02890
0175 0	C29B	378	RSTSA LD 2	VT101		HWE02900
0176 0	E072	379	AND	=/2000		HWE02910
0177 01	4C18018A	380	BZ	DAC4L		HWE02920
0179 0	C0C5	381	LD	NUM		HWE02930
017A 0	906F	382	S	=90	NUMBER OF ADJUSTMENTS	HWE02940
017B 01	4C30018A	383	BP	DAC4L		HWE02950
		384	*RESET ONLY ONE TRIM			HWE02960
017D 0	C06D	385	LD	=C		HWE02970
017E 0	90C0	386	S	NUM		HWE02980
017F 0	D0C4	387	STO	TEMP5		HWE02990
0180 01	6500004E	388	LDX L1	ST000		HWE03000
0182 01	7580013F	389	MDX 11	NUM		HWE03010
0184 0	C100	390	LD	1 0		HWE03020
0185 03	6500FF80	391	LDX L1	IVT00		HWE03030
0187 01	75800144	392	MDX 11	TEMP5		HWE03040

Table B-13. Bendix Bounds Program (Continued)

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0189	0	D100	393	STO	1	0	HWE03050
			394	* END RESET ONLY ONE TRIM			HWE03060
			395	* DAC4 OUTPUT			HWE03070
018A			396	DAC4L	EQU	*	HWE03080
018A	0	C298	397	LD	2	VT101	HWE03090
018B	0	E060	398	AND		=/1000	HWE03100
018C	01	4C180190	399	BZ		DAC40	HWE03110
018E	0	C081	400	LD		TMR	HWE03120
018F	0	D216	401	STO	2	VT149	HWE03130
			402	* DAC4 OUTPUT ROUTINE			HWE03140
0190			403	DAC40	EQU	*	HWE03150
0190	0	C216	404	LD	2	VT149	HWE03160
0191	0	D083	405	STO		DKOUT	HWE03170
0192	01	6580C145	406	LDX	11	DKOUT	HWE03180
0194	03	C500FF80	407	LD	L1	IVT00	HWE03190
0196	0	1881	408	SRT		1	HWE03200
0197	01	D4000589	409	STO	L	ALOG4	HWE03210
0199	0	D250	410	STO	2	VT220	HWE03220
			411	* OUTPUT VTXXX TO DAC 2			HWE03230
			412	* HWE03240			HWE03240
019A	0	C298	413	LD	2	VT101	HWE03250
019B	0	E051	414	AND		=/0800	HWE03260
019C	01	4C1801A0	415	BZ		DAC20	HWE03270
019E	0	C0A1	416	LD		TMR	HWE03280
019F	0	D261	417	STO	2	VT224	HWE03290
			418	* HWE03300			HWE03300
01A0			419	DAC20	EQU	*	HWE03310
01A0	0	C261	420	LD	2	VT224	HWE03320
01A1	0	D0A4	421	STO		DK20T	HWE03330
01A2	01	65800146	422	LDX	11	DK20T	HWE03340
01A4	03	C500FF80	423	LD	L1	IVT00	HWE03350
01A6	0	1881	424	SRT		1	HWE03360
01A7	01	D4000587	425	STO	L	BLEED	HWE03370
01A9	0	D262	426	STO	2	VT225	HWE03380
			427	* HWE03390			HWE03390
			428	* VALVE POSITION SIMULATION			HWE03400
01AA	0	C298	429	LD	2	VT101	HWE03410
01AB	0	E042	430	AND		=/0400	HWE03420
01AC	01	4C300188	431	BP		VLVEG	VALVE POSITION ENGINE
01AE	0	C298	432	LD	2	VT101	HWE03440
01AF	0	E03F	433	AND		=/00FF	HWE03450
01B0	0	903F	434	S		=/00AA	HWE03460
01B1	01	4C1801C6	435	BZ		SAFND	END OF SAFETY ROUTINE
01B3	0	C03D	436	LD		=-5000	HWE03480
01B4	01	D4000585	437	STO	L	FUEL	HWE03490
01B6	01	4C000551	438	B	L	DONE	HWE03500
			439	* VALVE POSITION ENGINE RUN			HWE03510
01B8			440	VLVEG	EQU	*	HWE03520
01B8	0	C21E	441	LD	2	VT157	HWE03530
01B9	0	9038	442	S		=7000	HWE03540
01BA	01	4C3001C6	443	BP		SAFND	HWE03550
01BC	0	C298	444	LD	2	VT101	HWE03560
01BD	0	E031	445	AND		=/00FF	HWE03570
01BE	0	9034	446	S		=/0055	HWE03580
01BF	01	4C1801C6	447	BZ		SAFND	HWE03590
01C1	0	C02F	448	LD		=-5000	HWE03600
01C2	01	D4000585	449	STO	L	FUEL	HWE03610

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01F6 0	D292	507	STO	2	VT110			HWE04020
01F7 0	C313	508	LD	3	P19			HWE04030
01F8 0	D291	509	STO	2	VT111			HWE04040
01F9 0	C314	510	LD	3	P20			HWE04050
01FA 0	D290	511	STO	2	VT112			HWE04060
01FB 0	C315	512	LD	3	P21			HWE04070
01FC 0	D28F	513	STO	2	VT113			HWE04080
01FD 0	C316	514	LD	3	P22			HWE04090
01FE 0	D28E	515	STO	2	VT114			HWE04100
01FF 0	C317	516	LD	3	P23			HWE04110
0200 0	D28D	517	STO	2	VT115			HWE04120
0201 0	C318	518	LD	3	P24			HWE04130
0202 0	D28C	519	STO	2	VT116			HWE04140
0203 0	C319	520	LD	3	P25			HWE04150
0204 0	D28B	521	STO	2	VT117			HWE04160
0205 0	C31A	522	LD	3	P26			HWE04170
0206 0	D28A	523	STO	2	VT118			HWE04180
		524	*			STRIP 4		HWE04190
0207 0	C31B	525	LD	3	P27			HWE04200
0208 0	D289	526	STO	2	VT119			HWE04210
0209 0	C31C	527	LD	3	P28			HWE04220
020A 0	D288	528	STO	2	VT120			HWE04230
020B 0	C31D	529	LD	3	P29			HWE04240
020C 0	D287	530	STO	2	VT121			HWE04250
020D 0	C31E	531	LD	3	P30			HWE04260
020E 0	D286	532	STO	2	VT122			HWE04270
020F 0	C31F	533	LD	3	P31			HWE04280
0210 0	D285	534	STO	2	VT123			HWE04290
0211 0	C320	535	LD	3	P32			HWE04300
0212 0	D284	536	STO	2	VT124			HWE04310
0213 0	C321	537	LD	3	P33			HWE04320
0214 0	D283	538	STO	2	VT125			HWE04330
0215 0	C322	539	LD	3	P34			HWE04340
0216 0	D282	540	STO	2	VT126			HWE04350
0217 0	C323	541	LD	3	P35			HWE04360
0218 0	D281	542	STO	2	VT127			HWE04370
0219 0	C090	543	LD		=50			HWE04380
021A 01	D4000141	544	STO	L	TRIMS			HWE04390
021C		545	ENDTM	EQU	*			HWE04400
		546	*			STRIP 5		HWE04410
021C 0	C324	547	LD	3	P36	DP/P	EK14	HWE04420
021D 0	D264	548	STO	2	VT227			HWE04430
021E 0	C325	549	LD	3	P37			HWE04440
021F 0	D265	550	STO	2	VT228	SPARE		HWE04450
0220 0	C326	551	LD	3	P38			HWE04460
0221 0	D266	552	STO	2	VT229	SPARE		HWE04470
0222 0	C327	553	LD	3	P39	P3	EK14	HWE04480
0223 0	D267	554	STO	2	VT230			HWE04490
0224 0	C328	555	LD	3	P40	PB	EK15	HWE04500
0225 0	D268	556	STO	2	VT231			HWE04510
0226 0	A054	557	M		=20000			HWE04520
0227 0	1081	558	SLT		1			HWE04530
0228 0	D29A	559	STO	2	VT102	PB=100XPS1		HWE04540
0229 0	C329	560	LD	3	P41	DP	EK15	HWE04550
022A 0	D269	561	STO	2	VT232			HWE04560
022B 0	A050	562	M		=30000			HWE04570
022C 0	1081	563	SLT		1			HWE04580

Table B-13. Bendix Bounds Program (Continued)

0267	0	D29F	621		STO	2	VT097	T4=10XF			HWE05160
0268	0	C339	622		LD	3	P57	T5	EK18		HWE05170
0269	0	801C	623		A		=6100				HWE05180
026A	0	D279	624		STO	2	VT248				HWE05190
026B	0	9016	625		S		=4600				HWE05200
026C	0	D29E	626		STO	2	VT098	T5=10XF			HWE05210
026D	0	C33A	627		LD	3	P58	PLA1	EK18		HWE05220
026E	0	D27A	628		STO	2	VT249				HWE05230
026F	0	C33B	629		LD	3	P59	PLA2	EK18		HWE05240
0270	0	D27B	630		STO	2	VT250				HWE05250
0271	0	C33C	631		LD	3	P60	LFAD-LAG SIGNAL	EK18		HWE05260
0272	0	D27C	632		STO	2	VT251				HWE05270
0273	0	C33D	633		LD	3	P61		EK18		HWE05280
0274	0	D27D	634		STO	2	VT252				HWE05290
0275	0	C33E	635		LD	3	P62		EK19		HWE05300
0276	0	D27E	636		STO	2	VT253				HWE05310
			637	*				64TH POINT			HWE05320
0277	0	C33F	638		LD	3	P63				HWE05330
0278	0	D27F	639		STO	2	VT254				HWE05340
0279	0	700D	640		B		GOTO1				HWE05350
			641		LORG						HWE05360
027A	0	0032	642	+	DC		50				
027B	0	4E20	643	+	DC		20000				
027C	0	7530	644	+	DC		30000				
027D	0	61A8	645	+	DC		25000				
027E	0	3A98	646	+	DC		15000				
027F	0	2710	647	+	DC		10000				
0280	0	4C40	648	+	DC		19520				
0281	0	5000	649	+	DC		20480				
0282	0	11F8	650	+	DC		4600				
0283	0	0FA0	651	+	DC		4000				
0284	0	2A3D	652	+	DC		10813				
0285	0	066D	653	+	DC		1645				
0286	0	17D4	654	+	DC		6100				
0287			655		GOTO1 EQU		*				HWE05370
			656	*				POWER REQUEST			HWE05380
0287	0	C2FF	657		LD	2	VT001	IDLE SPEED TRIM			HWE05390
0288	0	1883	658		SRT		3				HWE05400
0289	0	806E	659		A		=7950				HWE05410
028A	0	D218	660		STO	2	VT151				HWE05420
028B	0	C2FE	661		LD	2	VT002	MAX SPEED TRIM			HWE05430
028C	0	1883	662		SRT		3				HWE05440
028D	0	8068	663		A		=16542				HWE05450
028E	0	D219	664		STO	2	VT152				HWE05460
028F	0	C27A	665		LD	2	VT249	POWER LEVER			HWE05470
0290	0	1881	666		SRT		1				HWE05480
0291	0	8068	667		A		=7212				HWE05490
0292	0	D217	668		STO	2	VT150				HWE05500
			669	*	SELECT HIGH						HWE05510
0293	0	B218	670		CMP	2	VT151				HWE05520
0294	0	7002	671		MDX		**2				HWE05530
0295	0	1000	672		NOP						HWE05540
0296	0	C218	673		LD	2	VT151				HWE05550
			674	*	SELECT LOW						HWE05560
0297	0	B219	675		CMP	2	VT152				HWE05570
0298	0	C219	676		LD	2	VT152				HWE05580
0299	0	1000	677		NOP						HWE05590

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029A	0	D201	678	STO	2	VT126	HWE05600	
029B	Q	C2FC	679	LD	2	VT004	HWE05610	
029C	0	905E	680	S		=64	HWE05620	
029D	01	4C2802C2	681	BN		SAM1	HWE05630	
029F	Q	C2FB	682	LD	2	VT005	HWE05640	
02A0	Q	1889	683	SRT		9	HWE05650	
02A1	0	D206	684	STO	2	VT133	HWE05660	
02A2	0	C2FA	685	LD	2	VT006	HWE05670	
02A3	0	A268	686	M	2	VT231	HWE05680	
02A4	0	1885	687	SRT		6	HWE05690	
			688	* SELECT HIGH			HWE05700	
02A5	0	B205	689	CMP	2	VT133	HWE05710	
02A6	0	7002	690	MDX		**2	HWE05720	
02A7	0	1000	691	NOP			HWE05730	
02A8	Q	C206	692	LD	2	VT133	HWE05740	
02A9	0	D21A	693	STO	2	VT153	HWE05750	
02AA	0	C2F9	694	LD	2	VT007	HWE05760	
02AB	0	1889	695	SRT		9	HWE05770	
02AC	Q	D207	696	STO	2	VT134	HWE05780	
02AD	0	C2F8	697	LD	2	VT008	HWE05790	
02AE	0	A268	698	M	2	VT231	HWE05800	
02AF	0	1886	699	SRT		5	HWE05810	
			700	* SELECT LOW			HWE05820	
02B0	0	B207	701	CMP	2	VT134	HWE05830	
02B1	0	C207	702	LD	2	VT134	HWE05840	
02B2	0	1000	703	NOP			HWE05850	
02B3	0	D21B	704	STO	2	VT154	HWE05860	
02B4	0	C201	705	LD	2	VT128	HWE05870	
02B5	0	9205	706	S	2	VT132	HWE05880	
02B6	0	D202	707	STO	2	VT129	HWE05890	
			708	* SELECT LOW			HWE05900	
02B7	0	B21A	709	CMP	2	VT153	HWE05910	
02B8	0	C21A	710	LD	2	VT153	HWE05920	
02B9	0	1000	711	NOP			HWE05930	
02BA	0	D203	712	STO	2	VT130	HWE05940	
			713	* SELECT HIGH			HWE05950	
02BB	0	B21B	714	CMP	2	VT154	HWE05960	
02BC	0	7002	715	MDX		**2	HWE05970	
02BD	0	1000	716	NOP			HWE05980	
02BE	0	C21B	717	LD	2	VT154	HWE05990	
02BF	0	D204	718	STO	2	VT131	HWE06000	
02C0	0	8205	719	A	2	VT132	HWE06010	
02C1	0	7001	720	B		SAM2	HWE06020	
02C2	0	C201	721	SAM1	LD	2	VT128	HWE06030
02C3	0	D205	722	SAM2	STO	2	VT132	HWE06040
02C4	0	C276	723	LD	2	VT245	HWE06050	
02C5	0	9036	724	S		=12640	HWE06060	
02C6	0	A0BA	725	M		=20480	HWE06070	
02C7	0	1081	726	SLT		1	HWE06080	
02C8	0	8034	727	A		=15542	HWE06090	
02C9	0	D21C	728	STO	2	VT155	HWE06100	
			729	* SELECT LOW			HWE06110	
02CA	0	B205	730	CMP	2	VT132	HWE06120	
02CB	0	C205	731	LD	2	VT132	HWE06130	
02CC	0	1000	732	NCP			HWE06140	
02CD	0	D21D	733	STO	2	VT156	HWE06150	
			734	* SPEED CONTROL			HWE06160	

Table B-13. Bendix Rounds Program (Continued)

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02CE 0	C02F	735	LD	=-1600		HWE06170
		736	*		SELECT LOW	HWE06180
02CF 0	9260	737	CMP	2 VT236		HWE06190
02D0 0	C260	738	LD	2 VT236		HWE06200
02D1 0	1000	739	NOP			HWE06210
02D2 0	821E	740	A	2 VT157		HWE06220
02D3 01	4C2802D9	741	BN	NEG1	SPEED	HWE06230
02D5 0	9029	742	S	=800	COMPARISON	HWE06240
02D6 01	4C3002EC	743	BP	POS1		HWE06250
02D8 0	7007	744	B	SAM3		HWE06260
02D9 0	8025	745	NEG1 A	=800		HWE06270
02DA 01	4C3002E0	746	BP	SAM3		HWE06280
02DC 0	C26A	747	POS1 LD	2 VT233		HWE06290
02DD C	A022	748	M	=8873		HWE06300
02DE 0	1020	749	SLT	0		HWE06310
02DF 0	7001	750	B	SAM4		HWE06320
02E0 0	C020	751	SAM3 LD	=21000		HWE06330
02E1 0	D21F	752	SAM4 STO	2 VT158		HWE06340
		753	*			HWE06350
02E2 0	C2F7	754	LD	2 VT009		HWE06360
02E3 0	801E	755	A	=-123		HWE06370
02E4 01	4C3002F3	756	BP	NOUT		HWE06380
02E6 0	C01A	757	LD	=21000		HWE06390
02E7 0	D235	758	STO	2 VT180	INPUT POINT OF VALVE POS	HWE06400
02E8 0	C219	759	LD	2 VT156		HWE06410
02E9 0	921E	760	S	2 VT157		HWE06420
02EA 0	D220	761	STO	2 VT159		HWE06430
02EB 0	A2FD	762	M	2 VT003		HWE06440
02EC 0	1082	763	SLT	2		HWE06450
02ED 0	D221	764	STO	2 VT160		HWE06460
02EE 0	C2F6	765	LD	2 VT010		HWE06470
02EF 0	1882	766	SRT	2		HWE06480
02F0 0	8221	767	A	2 VT160		HWE06490
02F1 0	D222	768	STO	2 VT161		HWE06500
02F2 0	7004	769	B	WFP3		HWE06510
		770	*			
		771	*			
		772	*			
		773	*		*****	
		774	*		*****	
		775	*			
02F3 0	C097	776	NOUT LD	=20000	CALL HONEYWELL CONTROL PROG	HWE06520
02F4 0	D222	777	STO	2 VT161		HWE06530
02F5 30	089850E3	778	CALL	HWECT		
		779	*			
		780	*		*****	
		781	*		*****	
		782	*			
		783	*			
		784	*			
02F7		785	MFP3 EQU	*		HWE06540
02F7 0	703B	786	B	GO Tu2		HWE06550
		787	LDRG			HWE06560
02F8 0	1F0E	788	+	CC	7950	
02F9 0	409E	789	+	DC	16542	
02FA 0	1C2C	790	+	DC	7212	
02FB 0	0040	791	-	DC	64	

Table E-13. Bendix Bounds Program (Continued)

02FC	0	3160	792	+	DC	12640		
02FD	0	3CB6	793	+	DC	15542		
02FE	0	F9C0	794	+	DC	-1500		
02FF	0	0320	795	+	DC	800		
0300	0	22A9	796	+	DC	8873		
0301	0	5208	797	+	DC	21000		
0302	G	FF85	798	+	DC	-123		
0303			799		GOTO2 EQU	*		
			800	*				HWE06570
			801	*			THIS FOLLOWS BENO6220	HWE06580
			802				TEMPERATURE TRACK COMPUTE	HWE06590
0303	0	C276	802		LD	2 VT245	T2	HWE06600
0304	0	9064	803		S	=18320	112.5 DEGREES F	HWE06610
0305	01	4C300422	804		BP	L T2125		HWE06620
0307	0	80F7	805		A	=800	25 DEG F	HWE06630
0308	01	4C3003E1	806		BP	L T2100		HWE06640
030A	0	80F4	807		A	=800		HWE06650
030B	01	4C3003B6	808		BP	L T275		HWE06660
030D	0	80F1	809		A	=800		HWE06670
030E	G1	4C300381	810		BP	L T250		HWE06680
0310	0	80EE	811		A	=800		HWE06690
0311	01	4C30033E	812		BP	T225		HWE06700
			813	*			ZERO DEGREES F TRACK	HWE06710
0313	0	C2E7	814		LD	2 VT025		HWE06720
0314	0	1885	815		SRK	5		HWE06730
0315	0	8054	816		A	=15715		HWE06740
0316	0	921E	817		S	2 VT157		HWE06750
0317	01	4C30051E	818		BP	PATH4		HWE06760
0319	0	A051	819		M	=24800		HWE06770
031A	0	1084	820		SLT	4		HWE06780
031B	C	8050	821		A	=19500		HWE06790
031C	01	4C00044D	822		B	L MAXWP		HWE06800
031E	0	C04E	823		PATH4 LD	=13234		HWE06810
031F	0	921E	824		S	2 VT157		HWE06820
0320	01	4C300327	825		BP	PATH3		HWE06830
0322	0	A048	826		M	=0		HWE06840
0323	0	1084	827		SLT	4		HWE06850
0324	0	8047	828		A	=19500		HWE06860
0325	01	4C00044D	829		B	L MAXWP		HWE06870
0327	0	9047	830		PATH3 S	=3561		HWE06880
0328	01	4C30032F	831		BP	PATH2		HWE06890
032A	0	A045	832		M	=-6320		HWE06900
032B	0	1084	833		SLT	4		HWE06910
032C	0	8044	834		A	=14000		HWE06920
032D	01	4C00044D	835		B	L MAXWP		HWE06930
032F	0	9042	836		PATH2 S	=4710		HWE06940
0330	01	4C300337	837		BP	PATH1		HWE06950
0332	0	A040	838		M	=1730		HWE06960
0333	C	1084	839		SLT	4		HWE06970
0334	0	803F	840		A	=16000		HWE06980
0335	01	4C00044D	841		B	L MAXWP		HWE06990
0337	0	C03C	842		PATH1 LD	=16000		HWE07000
0338	0	92EB	843		S	2 VT024		HWE07010
0339	0	A21E	844		M	2 VT157		HWE07020
033A	0	A83A	845		D	=4963		HWE07030
033B	0	82EB	846		A	2 VT024		HWE07040
033C	01	4C00044D	847		B	L MAXWP		HWE07050
			848	*			25 DEGREES F TRACK	HWE07060

Table B-13. Bendix Bounds Program (Continued)

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033E	0	C2E7	849	T225	LD	2	VT025	HWE07070
033F	0	1P85	850		SRT		5	HWE07080
0340	0	8035	851		A		=15797	HWE07090
0341	0	921E	852		S	2	VT157	HWE07100
0342	01	4C300349	853		BP		PTH14	HWE07110
0344	0	A026	854		M		=24800	HWE07120
0345	0	1084	855		SLT		4	HWE07130
0346	0	8030	856		A		=19000	HWE07140
0347	01	4C00044D	857		B	L	MAXWP	HWE07150
0349	0	C023	858	PTH14	LD		=13234	HWE07160
034A	0	921E	859		S	2	VT157	HWE07170
034B	01	4C300352	860		BP		PTH13	HWE07180
034D	0	A02A	861		M		=399	HWE07190
034E	0	1084	862		SLT		4	HWE07200
034F	0	8029	863		A		=19250	HWE07210
0350	01	4C00044D	864		B	L	MAXWP	HWE07220
0352	0	9027	865	PTH13	S		=3309	HWE07230
0353	01	4C30035A	866		BP		PTH12	HWE07240
0355	0	A025	867		M		=-5870	HWE07250
0356	0	1084	868		SLT		4	HWE07260
0357	0	8024	869		A		=14250	HWE07270
0358	01	4C00044D	870		B	L	MAXWP	HWE07280
035A	0	9022	871	PTH12	S		=4549	HWE07290
035B	01	4C300362	872		BP		PTH11	HWE07300
035D	0	A020	873		M		=1570	HWE07310
035E	0	1084	874		SLT		4	HWE07320
035F	0	801F	875		A		=16250	HWE07330
0360	01	4C00044D	876		B	L	MAXWP	HWE07340
0362	0	C01C	877	PTH11	LD		=16250	HWE07350
0363	0	92E8	878		S	2	VT024	HWE07360
0364	0	A21E	879		M	2	VT157	HWE07370
0365	0	A81A	880		D		=5376	HWE07380
0366	0	82E8	881		A	2	VT024	HWE07390
0367	01	4C00044D	882		B	L	MAXWP	HWE07400
			883		LOG			HWE07410
0369	0	4790	884	+	DC		18320	
036A	0	3D63	885	+	DC		15715	
036B	0	60E0	886	+	DC		24800	
036C	0	4C2C	887	+	DC		19500	
036D	0	33B2	888	+	DC		13234	
036E	0	0000	889	+	DC		0	
036F	0	0DE9	890	+	DC		3561	
0370	0	E750	891	+	DC		-6320	
0371	0	36B0	892	+	DC		14000	
0372	0	1266	893	+	DC		4710	
0373	0	06C2	894	+	DC		1730	
0374	0	3E80	895	+	DC		16000	
0375	0	1363	896	+	DC		4963	
0376	0	3DB5	897	+	DC		15797	
0377	0	4A38	898	+	DC		19000	
0378	0	018F	899	+	DC		399	
0379	0	4B32	900	+	DC		19250	
037A	0	0CED	901	+	DC		3309	
037B	0	E912	902	+	DC		-5870	
037C	0	37AA	903	+	DC		14250	
037D	0	11C5	904	+	DC		4549	
037E	0	0622	905	+	DC		1570	

Table B-13. Bendix Bounds Program (Continued)

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037F	0	3F7A	906	+	DC	16250		
0380	0	1500	907	+	DC	5376		
			908	*				
0381	0	C2E7	909	T250	LD	2	VT025	50 DEGREE TRACK
0382	0	1885	910		SRT	5		HWE07420
0383	0	8D28	911		A		=15879	HWE07430
0384	0	921E	912		S	2	VT157	HWE07440
0385	01	4C30038C	913		BP		PTH24	HWE07450
0387	0	A0E3	914		M		=24800	HWE07460
0388	0	1084	915		SLT	4		HWE07470
0389	0	8023	916		A		=18500	HWE07480
038A	01	4C00044D	917		R	L	MAXWP	HWE07490
038C	0	C0E0	918	PTH24	LD		=13234	HWE07500
038D	0	921E	919		S	2	VT157	HWE07510
038E	01	4C300395	920		BP		PTH23	HWE07520
0390	0	A01D	921		M		=774	HWE07530
0391	0	1084	922		SLT	4		HWE07540
0392	0	80E4	923		A		=19000	HWE07550
0393	01	4C00044D	924		B	L	MAXWP	HWE07560
0395	0	9019	925	PTH23	S		=3071	HWE07570
0396	01	4C30039D	926		BP		PTH22	HWE07580
0398	0	A017	927		M		=-5330	HWE07590
0399	0	1084	928		SLT	4		HWE07600
039A	0	8016	929		A		=15000	HWE07610
039B	01	4C00044D	930		B	L	MAXWP	HWE07620
039D	0	9014	931	PTH22	S		=4373	HWE07630
039E	01	4C3003A5	932		BP		PTH21	HWE07640
03A0	0	A012	933		M		=1404	HWE07650
03A1	0	1084	934		SLT	4		HWE07660
03A2	0	8011	935		A		=16500	HWE07670
03A3	01	4C00044D	936		B	L	MAXWP	HWE07680
03A5	0	C00E	937	PTH21	LD		=16500	HWE07690
03A6	0	92E8	938		S	2	VT024	HWE07700
03A7	0	A21E	939		M	2	VT157	HWE07710
03A8	0	A80C	940		D		=5790	HWE07720
03A9	0	82E8	941		A	2	VT024	HWE07730
03AA	01	4C00044D	942		B	L	MAXWP	HWE07740
			943		LORG			HWE07750
03AC	0	3E07	944	+	DC		15879	
03AD	0	4844	945	+	DC		18500	
03AE	0	0306	946	+	DC		774	
03AF	0	0BFF	947	+	DC		3071	
03B0	0	EB2E	948	+	DC		-5330	
03B1	0	3A98	949	+	DC		15000	
03B2	0	1115	950	+	DC		4373	
03B3	0	057C	951	+	DC		1404	
03B4	0	4074	952	+	DC		16500	
03B5	0	169E	953	+	DC		5790	
			954	*				
03B6	0	C2E7	955	T275	LD	2	VT025	75 DEGREE F TRACK
03B7	0	1885	956		SRT	5		HWE07780
03B8	0	8054	957		A		=15961	HWE07790
03B9	0	921E	958		S	2	VT157	HWE07800
03BA	01	4C3003C1	959		BP		PTH34	HWE07810
03BC	0	A0AE	960		M		=24800	HWE07820
03BD	0	1084	961		SLT	4		HWE07830
03BE	0	804F	962		A		=18000	HWE07840
								HWE07850
								HWE07860

Table B-13. Bendix Bounds Program (Continued)

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03BF	01	4C00044D	963		B	L	MAXWP		HWE07870
03C1	0	COAB	964	PTH34	LD		=13234		HWE0788C
03C2	0	921E	965		S	2	VT157		HWE07890
03C3	01	4C3003CA	966		BP		PTH33		HWE07900
03C5	0	A049	967		M		=1127		HWE07910
03C6	0	1084	968		SLT		4		HWE07920
03C7	0	8048	969		A		=18750		HWE07930
03C8	01	4C00044D	970		B	L	MAXWP		HWE07940
03CA	0	9046	971	PTH33	S		=2823		HWE07950
03CB	01	4C3003D2	972		BP		PTH32		HWE07960
03CD	0	A044	973		M		=-4710		HWE07970
03CE	0	1084	974		SLT		4		HWE07980
03CF	0	8043	975		A		=15500		HWE07990
03D0	01	4C00044D	976		B	L	MAXWP		HWE08000
03D2	0	9041	977	PTH32	S		=4208		HWE08010
03D3	01	4C3003DA	978		BP		PTH31		HWE08020
03D5	0	A03F	979		M		=1215		HWE08030
03D6	0	1084	980		SLT		4		HWE08040
03D7	0	803E	981		A		=16750		HWE08050
03D8	01	4C00044D	982		B	L	MAXWP		HWE08060
03DA	0	C03B	983	PTH31	LD		=16750		HWE08070
03DB	0	92E8	984		S	2	VT024		HWE08080
03DC	0	A21E	985		M	2	VT157		HWE08090
03DD	0	A839	986		D		=6203		HWE08100
03DE	0	82E8	987		A	2	VT024		HWE08110
03DF	01	4C00044D	988		B	L	MAXWP		HWE08120
			989	*				100 DEGREE F TRACK	
03E1	0	C2E7	990	T2100	LD	2	VT025		HWE08130
03E2	0	1885	991		SRT		5		HWE08140
03E3	0	8034	992		A		=16045		HWE08150
03E4	0	921E	993		S	2	VT157		HWE08170
03E5	01	4C3003EC	994		BP		PTH44		HWE0818C
03E7	0	A083	995		M		=24800		HWE08190
03E8	0	1084	996		SLT		4		HWE08200
03E9	0	802F	997		A		=17500		HWE08210
03EA	01	4C00044D	998		B	L	MAXWP		HWE08220
03EC	01	4C00036D	999	PTH44	LD	L	=13234		HWE08230
03EE	0	921E	1000		S	2	VT157		HWE08240
03EF	01	4C3003F6	1001		BP		PTH43		HWE08250
03F1	0	A028	1002		M		=1455		HWE08260
03F2	0	1084	1003		SLT		4		HWE08270
03F3	0	8089	1004		A		=18500		HWE08280
03F4	01	4C00044D	1005		B	L	MAXWP		HWE08290
03F6	0	9024	1006	PTH43	S		=2573		HWE08300
03F7	01	4C3003FE	1007		BP		PTH42		HWE08310
03F9	0	A022	1008		M		=-3980		HWE08320
03FA	0	1084	1009		SLT		4		HWE08330
03FB	0	8021	1010		A		=16000		HWE08340
03FC	01	4C00044D	1011		B	L	MAXWP		HWE08350
03FE	0	901F	1012	PTH42	S		=4042		HWE08360
03FF	01	4C300406	1013		BP		PTH41		HWE08370
0401	0	A01D	1014		M		=1012		HWE08380
0402	0	1084	1015		SLT		4		HWE08390
0403	0	801C	1016		A		=17000		HWE08400
0404	01	4C00044D	1017		B	L	MAXWP		HWE08410
0406	0	C019	1018	PTH41	LD		=17000		HWE08420
0407	0	92E8	1019		S	2	VT024		HWE08430

Table B-13. Bendix Bounds Program (Continued)

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0408 0	A21E	1020		M	2	VT157		HWE08440
0409 0	A817	1021		D		=6617		HWE08450
040A 0	82E8	1022		A	2	VT024		HWE08460
040B 01	4C00044D	1023		B		L	MAXWP	HWE08470
		1024					LORG	HWE08480
040D 0	3E59	1025	+	DC		15961		
040E 0	4650	1026	+	DC		18000		
040F 0	0467	1027	+	DC		1127		
0410 0	493E	1028	+	DC		18750		
0411 0	0807	1029	+	DC		2823		
0412 0	ED9A	1030	+	DC		-4710		
0413 0	3C8C	1031	+	DC		15500		
0414 0	1070	1032	+	DC		4208		
0415 0	048F	1033	+	DC		1215		
0416 0	416E	1034	+	DC		16750		
0417 0	1838	1035	+	DC		6203		
0418 0	3EAD	1036	+	DC		16045		
0419 0	445C	1037	+	DC		17500		
041A 0	05AF	1038	+	DC		1455		
041B 0	0A0D	1039	+	DC		2573		
041C 0	F074	1040	+	DC		-3980		
041D 0	3E80	1041	+	DC		16000		
041E 0	0FCA	1042	+	DC		4042		
041F 0	03F4	1043	+	DC		1012		
0420 0	4268	1044	+	DC		17000		
0421 0	19D9	1045	+	DC		6617		
		1046	*					
0422 0	C2E7	1047	T2125	LD	2	VT025	125 DEGREE F TRACK	HWE08490
0423 0	1885	1048		SRT		5		HWE08500
0424 0	904C	1049		A		=16128		HWE08510
0425 0	921E	1050		S	2	VT157		HWE08520
0426 01	4C30042D	1051		BP		PTH54		HWE08530
0428 0	A049	1052		M		=24800		HWE08540
0429 0	1084	1053		SLT		4		HWE08550
042A 0	80F5	1054		A		=17000		HWE08560
042B 01	4C00044D	1055		B	L	MAXWP		HWE08570
042D 0	C045	1056	PTH54	LD		=13234		HWE08580
042E 0	921E	1057		S	2	VT157		HWE08590
042F 01	4C300436	1058		BP		PTH53		HWE08600
0431 0	A042	1059		M		=1769		HWE08610
0432 0	1084	1060		SLT		4		HWE08620
0433 0	8041	1061		A		=18250		HWE08630
0434 01	4C00044D	1062		B	L	MAXWP		HWE08640
0436 0	903F	1063	PTH53	S		=2327		HWE08650
0437 01	4C30043E	1064		BP		PTH52		HWE08660
0439 0	A03D	1065		M		=-3080		HWE08670
043A 0	1084	1066		SLT		4		HWE08680
043B 0	803C	1067		A		=16500		HWE08690
043C 01	4C00044D	1068		B	L	MAXWP		HWE08700
043E 0	903A	1069	PTH52	S		=3877		HWE08710
043F 01	4C300446	1070		BP		PTH51		HWE08720
0441 0	A038	1071		M		=792		HWE08730
0442 0	1084	1072		SLT		4		HWE08740
0443 0	8037	1073		A		=17250		HWE08750
0444 01	4C00044D	1074		B	L	MAXWP		HWE08760
0446 0	C034	1075	PTH51	LD		=17250		HWE08770
0447 0	92E8	1076		S	2	VT024		HWE08780
								HWE08790

Table B-13. Bendix Bounds Program (Continued)

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0448	0	A21E	1077		M	2	VT157		HWE08800
0449	0	A832	1078		D		=7030		HWE08810
044A	0	82E8	1079		A	2	VT024		HWE08820
044B	01	4C00044D	1080		B	L	MAXWP		HWE08830
044D	0	1882	1081	MAXWP	SRT	2			HWE08840
044E	0	D236	1082		STO	2	VT181		HWE08850
			1083	*				MINIMUM RATIOS COMPUTATION	HWE08860
044F	0	C2E2	1084		LD	2	VT030		HWE08870
0450	0	1882	1085		SRT	2			HWE08880
0451	0	D20D	1086		STO	2	VT140		HWE08890
0452	0	C21E	1087		LD	2	VT157		HWE08900
0453	0	9029	1088		S		=7100		HWE08910
0454	0	A2E3	1089		M	2	VT029		HWE08920
0455	0	1881	1090		SRT	1			HWE08930
0456	0	820D	1091		A	2	VT140		HWE08940
0457	0	1882	1092		SRT	2			HWE08950
0458	0	D23F	1093		STO	2	VT190		HWE08960
			1094	*				SELECT HIGH	HWE08970
0459	0	C21E	1095		LD	2	VT157		HWE08980
045A	0	B022	1096		CMP		=7100		HWE08990
045B	0	7002	1097		MDX		**+2		HWE09000
045C	0	1000	1098		NOP				HWE09010
045D	0	C01F	1099		LD		=7100		HWE09020
045E	0	D011	1100		STO		TEMP6		HWE09030
045F	0	C01E	1101		LD		=9600		HWE09040
0460	0	900F	1102		S		TEMP6		HWE09050
0461	0	A01D	1103		M		=19650		HWE09060
0462	0	1082	1104		SLT	2			HWE09070
0463	0	D23D	1105		STO	2	VT188		HWE09080
			1106	*				SELECT LOW	HWE09090
0464	0	C01B	1107		LD		=27687		HWE09100
0465	0	A21E	1108		M	2	VT157		HWE09110
0466	0	B23D	1109		CMP	2	VT188		HWE09120
0467	0	C23D	1110		LD	2	VT188		HWE09130
0468	0	1000	1111		NOP				HWE09140
0469	0	D23E	1112		STO	2	VT189		HWE09150
			1113	*				SELECT HIGH	HWE09160
046A	0	B23F	1114		CMP	2	VT190		HWE09170
046B	0	7002	1115		MDX		**+2		HWE09180
046C	0	1000	1116		NOP				HWE09190
046D	0	C23F	1117		LD	2	VT190		HWE09200
046E	0	D240	1118		STO	2	VT191		HWE09210
			1119	*				MINIMUM RATIOS BRANCH TO VALVE CONTROL	HWE09220
046F	0	7011	1120		B		CNTLB		HWE09230
0470	0	0000	1121	TEMP6	DC		**+		HWE09240
			1122		LDRG				HWE09250
0471	0	3F00	1123	+	DC		16128		
0472	0	60E0	1124	+	DC		24800		
0473	0	3382	1125	+	DC		13234		
0474	0	06E9	1126	+	DC		1769		
0475	0	474A	1127	+	DC		18250		
0476	0	0917	1128	+	DC		2327		
0477	0	F3F8	1129	+	DC		-3080		
0478	0	4074	1130	+	DC		16500		
0479	0	0F25	1131	+	DC		3877		
047A	0	0318	1132	+	DC		792		
047B	0	4362	1133	+	DC		17250		

Table B-13. Bendix Bounds Program (Continued)

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047C	0	1876	1134	+	DC	7030		
047D	0	188C	1135	+	DC	7100		
047E	0	2580	1136	+	DC	9600		
047F	0	4CC2	1137	+	DC	19650		
0480	0	6C27	1138	+	DC	27687		
0481			1139		CNTLB EQU	*	ADD. VALVE CONTROL HERE	HWE09260
0481	0	C235	1140		LD	2 VT180	THIS WILL BE COMPUTED VALU	HWE09270
0482	0	D235	1141		STO	2 VT180	FROM ADDED CONTROL LOOP	HWE09280
0483	0	7000	1142		B	VALPO		HWE09290
0484			1143		VALPO EQU	*		HWE09300
0484	0	C2E6	1144		LD	2 VT026		HWE09310
0485	0	A236	1145		M	2 VT181		HWE09320
0486	0	1082	1146		SLT	2		
0487	0	D237	1147		STO	2 VT182		HWE09340
0488	0	C2E1	1148		LD	2 VT031		HWE09350
0489	0	1885	1149		SRT	5		HWE09360
048A	0	804A	1150		A	=5320		HWE09370
048B	0	A268	1151		M	2 VT231		HWE09380
048C	0	1083	1152		SLT	3		HWE09390
048D	0	D243	1153		STO	2 VT194		HWE09400
048E	0	C2E0	1154		LD	2 VT032		HWE09410
048F	0	1080	1155		SRT	0		HWE09420
0490	0	D20F	1156		STO	2 VT142		HWE09430
0491	0	C20F	1157		LD	2 VT033		HWE09440
0492	0	033	1158		SRT	3		HWE09450
0493	0	D210	1159		STO	2 VT143		HWE09460
			1160	*			VALVE ZERO FLOW TRIM	HWE09470
0494	0	C2F5	1161		LD	2 VT011		HWE09480
0495	0	1886	1162		SRT	6		HWE09490
0496	0	D20E	1163		STO	2 VT141		HWE09500
0497	0	803E	1164		A	=5400		HWE09510
0498	0	D214	1165		STO	2 VT147		HWE09520
			1166	*			MINIMUM VALVE	HWE09530
0499	0	C2E5	1167		LD	2 VT027		
049A	0	A240	1168		M	2 VT191		HWE09550
049B	0	1081	1169		SLT	1		HWE09560
049C	0	A243	1170		M	2 VT194		HWE09570
049D	0	1084	1171		SLT	4		HWE09580
049E	0	D241	1172		STO	2 VT192		HWE09590
049F	0	B210	1173		CMP	2 VT143		HWE09600
04A0	0	7002	1174		MDX	**2		HWE09610
04A1	0	1000	1175		NOP			HWE09620
04A2	0	C210	1176		LD	2 VT143		HWE09630
04A3	0	D215	1177		STO	2 VT148		HWE09640
04A4	0	C237	1178		LD	2 VT182		HWE09650
04A5	0	B222	1179		CMP	2 VT161	SELECT LOW WITH SPEED	HWE09660
04A6	0	C222	1180		LD	2 VT161		HWE09670
04A7	0	1000	1181		NOP			HWE09680
			1182	*			SELECT HIGH	HWE09690
04A8	0	B02E	1183		CMP	=-4000		HWE09700
04A9	0	7002	1184		MDX	**2		HWE09710
04AA	0	1000	1185		NOP			HWE09720
04AB	0	C02B	1186		LD	=-4000		HWE09730
04AC	0	D238	1187		STO	2 VT183		HWE09740
04AD	0	A243	1188		M	2 VT194		HWE09750
04AE	0	1084	1189		SLT	4		HWE09760
04AF	0	D239	1190		STO	2 VT184		HWE09770

Table B-13. Bendix Bounds Program (Continued)

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0480	0	B20F	1191	CMP	2	VT142	SELECT LOW VALVE LIMIT	HWE09790
0481	0	C20F	1192	LD	2	VT142		HWE09790
0482	0	1000	1193	NOP				HWE09800
0483	0	D23A	1194	STO	2	VT185		HWE09810
0484	0	B21F	1195	CMP	2	VT158	SELECT LOW WITH N SAFETY	HWE09820
0485	0	C21F	1196	LD	2	VT158		HWE09830
0486	0	1000	1197	NOP				HWE09840
0487	0	D23B	1198	STO	2	VT186		HWE09850
0488	0	B235	1199	CMP	2	VT180	SELECT LOW WITH CON LOOP	HWE09860
0489	0	C235	1200	LD	2	VT180		HWE09870
048A	0	1000	1201	NOP				HWE09880
048B	0	D23C	1202	STO	2	VT187		HWE09890
048C	0	B215	1203	CMP	2	VT148	SELECT HIGH WITH MINIMUM	HWE09900
048D	0	7002	1204	MOX		**2		HWE09910
048E	0	1000	1205	NOP				HWE09920
048F	0	C215	1206	LD	2	VT148		HWE09930
04C0	0	D242	1207	STO	2	VT193		HWE09940
04C1	0	B214	1208	A	2	VT147		HWE09950
04C2	0	D244	1209	STO	2	VT195		HWE09960
04C3	01	D4000585	1210	STO	L	FUEL		HWE09970
			1211					HWE09980
			1212	*			IGV AND BLEED SCHEDULES	HWE09990
04C5	0	C276	1213	LD	2	VT245		HWE10000
04C6	0	9011	1214	S		=13440		HWE10010
04C7	0	A011	1215	M		=210		HWE10020
04C8	0	1087	1216	SLT		7		HWE10030
04C9	0	8010	1217	A		=11800		HWE10040
04CA	0	D251	1218	STO	2	VT208		HWE10050
04CB	0	C276	1219	LD	2	VT245		HWE10060
04CC	0	900E	1220	S		=17088		HWE10070
04CD	01	4C2804DF	1221	BN		SAM6		HWE10080
04CF	0	A00C	1222	M		=500		HWE10090
04D0	0	1086	1223	SLT		6		HWE10100
04D1	0	800B	1224	A		=15800		HWE10110
04D2	0	D253	1225	STO	2	VT210		HWE10120
04D3	0	701C	1226	B		SAM8		HWE10130
04D4	0	700A	1227	B		GOTO5		HWE10140
			1228	LORG				HWE10150
04D5	0	14C8	1229	+	DC	5320		
04D6	0	1518	1230	+	DC	5400		
04D7	0	F060	1231	+	DC	-4000		
04D8	0	3480	1232	+	DC	13440		
04D9	0	00D2	1233	+	DC	210		
04DA	0	2E18	1234	+	DC	11800		
04DB	0	42C0	1235	+	DC	17088		
04DC	0	01F4	1236	+	DC	500		
04DD	0	30B8	1237	+	DC	15800		
04DE	0	0000	1238	TEMPA	DC	**		HWE10160
04DF			1239	GOTO5	EQU	*		HWE10170
			1240	*				HWE10180
04DF	0	C276	1241	SAM6	LD	2	VT245	HWE10190
04E0	0	903A	1242	S		=15488		HWE10200
04E1	01	4C2804EA	1243	BN		SAM7		HWE10210
04E3	0	A038	1244	M		=128		HWE10220
04E4	0	1086	1245	SLT		6		HWE10230
04E5	0	00F8	1246	STO		TEMPA		HWE10240
04E6	0	C036	1247	LD		=16000		HWE10250

Table B-13. Bendix Bounds Program (Continued)

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04E7	0	90F6	1248		S	TEMPA		HWE10260
04E8	0	0253	1249		STO	2 VT210		HWE10270
04E9	0	7006	1250		B	SAMB		HWE10280
			1251	*				HWE10290
04EA	0	C276	1252	SAM7	LD	2 VT245		HWE10300
04EB	0	90EC	1253		S	=13440		HWE10310
04EC	0	A031	1254		M	=1100		HWE10320
04ED	0	1085	1255		SLT	5		HWE10330
04EE	C	8030	1256		A	=14900		HWE10340
04EF	0	D253	1257		STO	2 VT210		HWE10350
			1258	*				HWE10360
04F0	0	C2CD	1259	SAMB	LD	2 VT051		HWE10370
04F1	0	1884	1260		SRT	4		HWE10380
04F2	0	D211	1261		STO	2 VT144		HWE10390
04F3	0	C2CC	1262		LD	2 VT052		HWE10400
04F4	0	1884	1263		SRT	4		HWE10410
04F5	0	8253	1264		A	2 VT210		HWE10420
04F6	0	9211	1265		S	2 VT144		HWE10430
04F7	0	9251	1266		S	2 VT208		HWE10440
04F8	0	D255	1267		STO	2 VT212		HWE10450
04F9	0	C21E	1268		LD	2 VT157		HWE10460
04FA	0	9251	1269		S	2 VT208		HWE10470
04FB	0	9211	1270		S	2 VT144		HWE10480
			1271	*	SELECT HIGH			HWE10490
04FC	0	8023	1272		CMP	=0		HWE10500
04FD	0	7002	1273		MDX	**+2		HWE10510
04FE	0	1000	1274		NOP			HWE10520
04FF	0	C020	1275		LD	=0		HWE10530
0500	J	A020	1276		M	=8340		HWE10540
0501	0	AA55	1277		D	2 VT212		HWE10550
0502	0	801F	1278		A	=5100		HWE10560
0503	0	D256	1279		STO	2 VT213		HWE10570
0504	01	D4000586	1280		STO	L PIGV		HWE10580
0506	0	C2CB	1281		LD	2 VT053		HWE10590
0507	0	1884	1282		SRT	4		HWE10600
0508	0	D212	1283		STO	2 VT145		HWE10610
0509	0	C2CA	1284		LD	2 VT054		HWE10620
050A	0	1884	1285		SRT	4		HWE10630
050B	0	8253	1286		A	2 VT210		HWE10640
050C	0	9212	1287		S	2 VT145		HWE10650
050D	0	9251	1288		S	2 VT208		HWE10660
050E	0	D257	1289		STO	2 VT214		HWE10670
050F	0	C21E	1290		LD	2 VT157		HWE10680
0510	0	9251	1291		S	2 VT208		HWE10690
0511	0	9212	1292		S	2 VT145		HWE10700
			1293	*	SELECT HIGH			HWE10710
0512	0	B00D	1294		CMP	=0		HWE10720
0513	0	7002	1295		MDX	**+2		HWE10730
0514	0	1000	1296		NOP			HWE10740
0515	0	C00A	1297		LD	=0		HWE10750
0516	J	A00C	1298		M	=5050		HWE10760
0517	0	AA57	1299		D	2 VT214		HWE10770
0518	0	800B	1300		A	=3600		HWE10780
0519	0	D258	1301		STO	2 VT215		HWE10790
			1302	*				HWE10800
			1303	*				HWE10810
			1304	*				HWE10820

STORE HERE IN BLEED IF SEPERATE CONTROL OF THE BLEEDS IS DESIRED

Table B-13. Bendix Bounds Program (Continued)

		1305	*		DAC 2 IS NOW USED FOR VTXXX OUTPUT	HWE10830
		1306	*			HWE10840
051A	0 700A	1307		B	GOTO6	BEN10424
		1308		LDG		BEN10425
051B	0 3C80	1309	+	DC	15488	
051C	0 0080	1310	+	DC	128	
051D	0 3E80	1311	+	DC	16000	
051E	0 044C	1312	+	DC	1100	
051F	0 3A34	1313	+	DC	14900	
0520	0 0000	1314	+	DC	0	
0521	0 2094	1315	+	DC	8340	
0522	0 13EC	1316	+	DC	5100	
0523	0 138A	1317	+	DC	5050	
0524	0 0E10	1318	+	DC	3600	
0525		1319		GOTO6 EQU	*	BEN10426
		1320	*			HWE10850
					NOZZLE CONTROL	
0525	0 C049	1321		LD	=9250	BEN10430
0526	0 9210	1322		S	2 VT156	BEN10440
0527	0 A048	1323		M	=14320	BEN10450
0528	0 1983	1324		SLT	3	BEN10460
0529	0 8047	1325		A	=13740	BEN10470
		1326	*	SELECT LOW		BEN10480
052A	0 8047	1327		CMP	=12200	BEN10490
052B	0 C046	1328		LD	=12200	BEN10500
052C	0 1000	1329		NOP		BEN10510
052D	0 D25A	1330		STO	2 VT217	BEN10570
052E	0 C044	1331		LD	=16042	BEN10580
052F	0 9210	1332		S	2 VT156	BEN10590
		1333	*	SELECT LOW		BEN10600
0530	0 80EF	1334		CMP	=0	BEN10610
0531	0 COEE	1335		LD	=0	BEN10620
0532	0 1000	1336		NOP		BEN10630
0533	0 A040	1337		M	=28900	BEN10640
0534	0 1084	1338		SLT	4	BEN10650
0535	0 82C9	1339		A	2 VT055	BEN10660
		1340	*	SELECT HIGH		
0536	0 825A	1341		CMP	2 VT217	BEN10680
0537	0 7002	1342		MOX	**2	
0538	0 1000	1343		NOP		
0539	0 C25A	1344		LD	2 VT217	
053A	0 0258	1345		STO	2 VT218	BEN10710
		1346	*		TEMPERANCE CONTROL	BEN10720
053B	0 C279	1347		LD	2 VT240	BEN10730
053C	0 92C8	1348		S	2 VT056	BEN10740
		1349	*	SELECT HIGH		BEN10750
053D	0 80E2	1350		CMP	=0	BEN10760
053E	0 7002	1351		MOX	**2	BEN10770
053F	0 1000	1352		NOP		BEN10780
0540	0 C0DF	1353		LD	=0	BEN10790
0541	0 A2C7	1354		M	2 VT057	BEN10800
0542	0 1084	1355		SLT	4	BEN10810
0543	0 825B	1356		A	2 VT218	BEN10820
		1357	*	SELECT HIGH		BEN10830
0544	0 825B	1358		CMP	2 VT218	BEN10840
0545	0 7002	1359		MOX	**2	BEN10850
0546	0 1000	1360		NOP		BEN10860
0547	0 C25B	1361		LD	2 VT218	BEN10870

Table B-13. Bendix Bounds Program (Continued)

0548	0	D25C	1362	STO	2	VT219		BEN10880
0549	0	D03E	1363	STO		NOZ		BEN10890
054A	0	C2E4	1364	LD	2	VT028	THIS GOES IN THE BOUNDS	
054B	01	940002FB	1365	S	L	=64	PROGRAM AT ADDRESS DONE	
054D	01	4C200551	1366	BNZ		DONE	IF VT028=64 HW NOZ IS IN	
054F	0	C2AF	1367	LD	2	VT081	IF VT028 NOT 64 BENDX IN	
0550	0	D037	1368	STO		NOZ		
0551			1369	DONE	EQU	*		HWE11220
			1370	*				HWE11230
			1371	*				HWE11240
0551	30	040565C0	1372	CALL		DAOP		HWE11250
0553	1	057A	1373	DC		DALST		HWE11260
			1374	*				HWE11270
			1375	*			FOLLOWS BEN11070	HWE11280
			1376	*			LOOP DETERMINATION	HWE11290
0554	0	C242	1377	LD	2	VT193		HWE11300
0555	0	901F	1378	S		=20		HWE11310
0556	0	9215	1379	S	2	VT148		HWE11320
0557	01	4C280561	1380	BN		NEGA		HWE11330
0559	0	C242	1381	LD	2	VT193		HWE11340
055A	0	801A	1382	A		=20		HWE11350
055B	0	9239	1383	S	2	VT144		HWE11360
055C	01	4C300564	1384	BP		POSA		HWE11370
055E	0	C286	1385	LD	2	VT074		HWE11380
055F	0	D263	1386	STO	2	VT226		HWE11390
0560	0	7005	1387	B		CON1		HWE11400
0561	0	C014	1388	NEGA	LD	=-32000	MIN CONTROL -5V OUT	HWE11410
0562	0	D263	1389	STO	2	VT226		HWE11420
0563	0	7002	1390	B		CON1		HWE11430
0564	0	C012	1391	POSA	LD	=32000	MAX CONTROL 5V OUT	HWE11440
0565	0	D263	1392	STO	2	VT226		HWE11450
0566			1393	CON1	EQU	*		HWE11460
			1394	*				HWE11470
0566	0	0811	1395	X10		CEOFF		HWE11480
0567	00	65000000	1396	XR1	L1	**		HWE11490
0569	00	66000000	1397	XR2	L2	**		HWE11500
056B	00	67000000	1398	XR3	L3	**		HWE11510
056D	01	4C800000	1399	BSC	I	GTECT		HWE11520
			1400	LORG				HWE11530
056F	0	2422	1401	+	DC	9250		
0570	0	37F0	1402	+	DC	14320		
0571	0	35AC	1403	+	DC	13740		
0572	0	2FA8	1404	+	DC	12200		
0573	0	3EAA	1405	+	DC	16042		
0574	0	70E4	1406	+	DC	28900		
0575	0	0014	1407	+	DC	20		
0576	0	8300	1408	+	DC	-32000		
0577	0	7000	1409	+	DC	32000		
0578		0000	1410	CEOFF	BSS	E	0	HWE11540
0578	0	0000	1411	DC			0	HWE11550
0579	0	E400	1412	DC		/E400		HWE11560
			1413	*				HWE11570
057A	0	0000	1414	DALST	DC	0		HWE11580
057B	0	0000	1415	DC		0		
057C		0004	1416	BSS		4		HWE11600
0580	0	0000	1417	DC		**		HWE11610
0581	0	3000	1418	DC		/3000		
0582	1	0583	1419	DC		AOLST		HWE11630
0583	0	0006	1420	AOLST	DC	/0000+6		
0584	0	0000	1421	APZ	DC	0		
0585	0	0000	1422	FUEL	DC	**		
0586	0	0000	1423	PIGV	DC	**		
0587	0	0000	1424	BLEED	DC	**		
0588	0	0000	1425	NOZ	DC	**		
0589	0	0000	1426	ALOG4	DC	**		
			1427	*				HWE11750
			1428	*				HWE11760
			1429	*				HWE11770

Table B-13. Bendix Bounds Program (Continued)

0000	1431	P00	EQU	00	HWE11820
0001	1432	P01	EQU	01	HWE11830
0002	1433	P02	EQU	02	HWE11840
0003	1434	P03	EQU	03	HWE11850
0004	1435	P04	EQU	04	HWE11860
0005	1436	P05	EQU	05	HWE11870
0006	1437	P06	EQU	06	HWE11880
0007	1438	P07	EQU	07	HWE11890
0008	1439	P08	EQU	08	HWE11900
0009	1440	P09	EQU	09	HWE11910
000A	1441	P10	EQU	10	HWE11920
000B	1442	P11	EQU	11	HWE11930
000C	1443	P12	EQU	12	HWE11940
000D	1444	P13	EQU	13	HWE11950
000E	1445	P14	EQU	14	HWE11960
000F	1446	P15	EQU	15	HWE11970
0010	1447	P16	EQU	16	HWE11980
0011	1448	P17	EQU	17	HWE11990
0012	1449	P18	EQU	18	HWE12000
0013	1450	P19	EQU	19	HWE12010
0014	1451	P20	EQU	20	HWE12020
0015	1452	P21	EQU	21	HWE12030
0016	1453	P22	EQU	22	HWE12040
0017	1454	P23	EQU	23	HWE12050
0018	1455	P24	EQU	24	HWE12060
0019	1456	P25	EQU	25	HWE12070
001A	1457	P26	EQU	26	HWE12080
001B	1458	P27	EQU	27	HWE12090
001C	1459	P28	EQU	28	HWE12100
001D	1460	P29	EQU	29	HWE12110
001E	1461	P30	EQU	30	HWE12120
001F	1462	P31	EQU	31	HWE12130
0020	1463	P32	EQU	32	HWE12140
0021	1464	P33	EQU	33	HWE12150
0022	1465	P34	EQU	34	HWE12160
0023	1466	P35	EQU	35	HWE12170
0024	1467	P36	EQU	36	HWE12180
0025	1468	P37	EQU	37	HWE12190
0026	1469	P38	EQU	38	HWE12200
0027	1470	P39	EQU	39	HWE12210
0028	1471	P40	EQU	40	HWE12220
0029	1472	P41	EQU	41	HWE12230
002A	1473	P42	EQU	42	HWE12240
002B	1474	P43	EQU	43	HWE12250
002C	1475	P44	EQU	44	HWE12260
002D	1476	P45	EQU	45	HWE12270
002E	1477	P46	EQU	46	HWE12280
002F	1478	P47	EQU	47	HWE12290
0030	1479	P48	EQU	48	HWE12300
0031	1480	P49	EQU	49	HWE12310
0032	1481	P50	EQU	50	HWE12320
0033	1482	P51	EQU	51	HWE12330
0034	1483	P52	EQU	52	HWE12340
0035	1484	P53	EQU	53	HWE12350
0036	1485	P54	EQU	54	HWE12360
0037	1486	P55	EQU	55	HWE12370
0038	1487	P56	EQU	56	HWE12380
0039	1488	P57	EQU	57	HWE12390
003A	1489	P58	EQU	58	HWE12400
003B	1490	P59	EQU	59	HWE12410
003C	1491	P60	EQU	60	HWE12420
003D	1492	P61	EQU	61	HWE12430
003E	1493	P62	EQU	62	HWE12440
003F	1494	P63	EQU	63	HWE12450
	1495	*			HWE12460

Table B-13. Bendix Bounds Program (Continued)

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	1497	*		TRIM VALUES	HWE12480
	1498	*		STANDARD TRIMS XR1	HWE12490
	1499	*		ANALOG TRIM EQU	HWE12500
	1500	*		COMPUTED VALUES EQU	HWE12510
0001	1501	VT128 EQU	+1	SPEED REQUEST	HWE12520
0002	1502	VT129 EQU	+2	SPEED REQUEST ERROR FOR INTEGRATION	HWE12530
0003	1503	VT130 EQU	+3	SPEED REQUEST INTEGRATION UP	HWE12540
0004	1504	VT131 EQU	+4	SPEED REQUEST INTEGRATION DOWN	HWE12550
0005	1505	VT132 EQU	+5	INTEGRATED SPEED REQUEST	HWE12560
0006	1506	VT133 EQU	+6	LIMIT UP	HWE12570
0007	1507	VT134 EQU	+7	LIMIT DOWN	HWE12580
0008	1508	VT135 EQU	+8	SCALED BASE RATIOS FIG10-5	HWE12590
0009	1509	VT136 EQU	+9	SCALED START INTERCEPT FIG10-7	HWE12600
000A	1510	VT137 EQU	+10	SCALED THIRD RANGE FIG10-7	HWE12610
000B	1511	VT138 EQU	+11	SCALED INC INTEGRATION FIG10-8	HWE12620
000C	1512	VT139 EQU	+12	SCALED DEC INTEGRATION FIG10-8	HWE12630
000D	1513	VT140 EQU	+13	SCALED MINIMUM RATIOS	HWE12640
000E	1514	VT141 EQU	+14	ZERO FLOW ADJUSTMENT	HWE12650
000F	1515	VT142 EQU	+15	MAXIMUM VALVE SETTING	HWE12660
0010	1516	VT143 EQU	+16	MINIMUM VALVE SETTING	HWE12670
0011	1517	VT144 EQU	+17	SCALED LOW N IGV FIG10-12	HWE12680
0012	1518	VT145 EQU	+18	SCALED LOW N BLEEDS FIG10-12	HWE12690
0013	1519	VT146 EQU	+19	TEMPERATURE REQ	HWE12700
0014	1520	VT147 EQU	+20	FUEL RATIOS FINAL FIG10-8	HWE12710
0015	1521	VT148 EQU	+21	COMPUTED FUEL REQUEST FIG10-8	HWE12720
0016	1522	VT149 EQU	+22	SELECTED VARIABLE STORAGE	HWE12730
	1523	*		FIG10-3 RPM REQUEST CONTROL	HWE12740
0017	1524	VT150 EQU	+23	POWER LEVER RPM REQ	HWE12750
0018	1525	VT151 EQU	+24	LOW SPEED SET	HWE12760
0019	1526	VT152 EQU	+25	HIGH SPEED SET	HWE12770
001A	1527	VT153 EQU	+26	POS RPM DN/DT	HWE12780
001B	1528	VT154 EQU	+27	NEG RPM DN/DT	HWE12790
001C	1529	VT155 EQU	+28	SPEED LIMIT TEMP	HWE12800
001D	1530	VT156 EQU	+29	SPEED REQUEST	HWE12810
	1531	*		FIG10-4 COMPUTED DIGITAL RPM	HWE12820
001E	1532	VT157 EQU	+30		HWE12830
001F	1533	VT158 EQU	+31	MAX FUEL REQUEST	HWE12840
0020	1534	VT159 EQU	+32	SPEED ERROR	HWE12850
0021	1535	VT160 EQU	+33	SPEED RATIOS ERROR	HWE12860
	1536	*			HWE12870
	1537	*		FIG10-5 PROPORTIONAL TEMP CON	HWE12880
	1538	*		VT146 TEMP REQ	HWE12890
0022	1539	VT161 EQU	+34	RATIOS SPEED CONTROL	HWE12900
0023	1540	VT162 EQU	+35	TEMP. RATIOS ERROR	HWE12910
0024	1541	VT163 EQU	+36	LOW OF RPM AND TEMP	HWE12920
	1542	*		FIG10-5 PROP. PRESSURE CONTROL	HWE12930
0025	1543	VT164 EQU	+37	PRESS REQUEST	HWE12940
0026	1544	VT165 EQU	+38	PRESS ERROR	HWE12950
0027	1545	VT166 EQU	+39	RATIOS PRESS ERROR	HWE12960
0028	1546	VT167 EQU	+40	LOW OF P, T, AND RPM	HWE12970
	1547	*			HWE12980
0029	1548	VT168 EQU	+41	RESERVED =VT167	HWE12990
	1549	*		FIG10-5 BASE RATIOS INTEGRATE	HWE13000
002A	1550	VT169 EQU	+42	INTEGRATION VALUE	HWE13010
002B	1551	VT170 EQU	+43	BASE RATIOS INT PLUS	HWE13020
002C	1552	VT171 EQU	+44	RATIOS REQUEST	HWE13030
	1553	*			HWE13040

Table B-13. Bendix Bounds Program (Continued)

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	1554	*		FIG10-7	MAX RATIOS SCHEDULE	HWE13050
002D	1555	VT172 EQU	+45		SCHEDULE T2 VALUE	HWE13060
002E	1556	VT173 EQU	+46		START RATIOS	HWE13070
0C2F	1557	VT174 EQU	+47		2ND RANGE START RATIO	HWE13080
0030	1558	VT175 EQU	+48		LOW OF 173 AND 174	HWE13090
0031	1559	VT176 EQU	+49		3RD RANGE VALUE	HWE13100
0032	1560	VT177 EQU	+50		HIGH OF 175 & 176	HWE13110
0033	1561	VT178 EQU	+51		ACC SCHEDULE	HWE13120
0034	1562	VT179 EQU	+52		LOW 178 & 177	HWE13130
0035	1563	VT180 EQU	+53		VALVE CONTROL INPUT POINT	HWE13140
0036	1564	VT181 EQU	+54		MAXIMUM RATIOS	HWE13150
0037	1565	VT182 EQU	+55		RATIOS MODIFIED	HWE13160
0038	1566	VT183 EQU	+56		LOW RATIOS WITH SPEED	HWE13170
0039	1567	VT184 EQU	+57		MAXIMUM VALVE DUE TO RATIO	HWE13180
	1568	*		FIGURE 10A-3 AND 4 VALVE POS		HWE13190
003A	1569	VT185 EQU	+58		MAXIMUM VALVE	HWE13200
003B	1570	VT186 EQU	+59		MAX VALVE AFTER N SAFETY	HWE13210
003C	1571	VT187 EQU	+60		MAX VALVE AFTER OTHER CONT	HWE13220
003D	1572	VT188 EQU	+61		IDLE MINIMUM SCHEDULE	HWE13230
003E	1573	VT189 EQU	+62		IDLE MINIMUM RATIOS	HWE13240
003F	1574	VT190 EQU	+63		MINIMUM RATIOS	HWE13250
0040	1575	VT191 EQU	+64		MINIMUM RATIOS OUT	HWE13260
0041	1576	VT192 EQU	+65		MINIMUM VALVE REQUEST	HWE13270
0042	1577	VT193 EQU	+66		FUEL REQUEST	HWE13280
0043	1578	VT194 EQU	+67		FACTORED BURNER PRESSURE	HWE13290
0044	1579	VT195 EQU	+68		FUEL REQUEST OUTPUT	HWE13300
0045	1580	VT196 EQU	+69		FUEL RATIOS PROP. ADDER	HWE13310
	1581	*				HWE13320
	1582	*		FIG10-9	TEMPERATURE CONTROL	HWE13330
0046	1583	VT197 EQU	+70		TEMPERATURE REQUEST ACC	HWE13340
0047	1584	VT198 EQU	+71		TEMPERATURE ERROR ACC	HWE13350
0048	1585	VT199 EQU	+72		TEMPERATURE RATIO PRGP ACC	HWE13360
0049	1586	VT200 EQU	+73		TEMPERATURE REQUEST DECEL	HWE13370
004A	1587	VT201 EQU	+74		TEMPERATURE ERROR DECEL	HWE13380
004B	1588	VT202 EQU	+75		TEMPERATURE RATIOS DECEL	HWE13390
	1589	*				HWE13400
	1590	*		FIG10-10	PRESSURE RATIO CONT	HWE13410
004C	1591	VT203 EQU	+76		DP/P LOW N SCHEDULE REQ	HWE13420
004D	1592	VT204 EQU	+77		DP/P MID N SCHEDULE REQ	HWE13430
004E	1593	VT205 EQU	+78		DP/P HIGH N SCHEDULE REQ	HWE13440
004F	1594	VT206 EQU	+79		DP/P ERROR	HWE13450
0050	1595	VT207 EQU	+80		DP/P INTEGRATION	HWE13460
	1596	*				HWE13470
	1597	*		FIG10-12	IGV AND BLEED SCHEDULE	HWE13480
0051	1598	VT208 EQU	+81		LOW N SCHEDULE	HWE13490
0052	1599	VT209 EQU	+82			HWE13500
0053	1600	VT210 EQU	+83		HIGH N MID T	HWE13510
0054	1601	VT211 EQU	+84			HWE13520
0055	1602	VT212 EQU	+85		SPEED RANGE IGV	HWE13530
0056	1603	VT213 EQU	+86		IGV REQUEST DAC /	HWE13540
0057	1604	VT214 EQU	+87		SPEED RANGE BLEEDS	HWE13550
0058	1605	VT215 EQU	+88		BLEED REQUEST DAC2	HWE13560
	1606	*				HWE13570
	1607	*		FIG10-14	NOZZLE CONTROL	HWE13580
0059	1608	VT216 EQU	+89			HWE13590
005A	1609	VT217 EQU	+90		NOZZLE MID SPEED	HWE13600
005B	1610	VT218 EQU	+91		NOZZLE HIGH SPEED	HWE13610

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005C	1611	VT219 EQU	+92	NOZZLE REQUEST DAC 3	HWE13620
005D	1612	VT220 EQU	+93	DAC4 OUTPUT VALUE	HWE13630
005E	1613	VT221 EQU	+94		HWE13640
005F	1614	VT222 EQU	+95		HWE13650
0060	1615	VT223 EQU	+96		HWE13660
0061	1616	VT224 EQU	+97	DAC2 OUTPUT ADJUSTMENT NO	HWE13670
0062	1617	VT225 EQU	+98	DAC2 OUTPUT VALUE	HWE13680
0063	1618	VT226 EQU	+99	EFFECTIVE LOOP OUTPUT	HWE13690
	1619	*		ANALOG VARIABLE	HWE13700
	1620	*		FIRST STRIP	HWE13710
0064	1621	VT22 EQU	+100	DP/P EK14	HWE13720
0065	1622	VT228 EQU	+101	POWER LEVER EK14	HWE13730
0066	1623	VT229 EQU	+102	INSTRUMENT VAR EK14T4	HWE13740
0067	1624	VT230 EQU	+103	BURNER PRESS EK14	HWE13750
0068	1625	VT231 EQU	+104	BURNER PRESS EK15P1	HWE13760
0069	1626	VT232 EQU	+105	DP= P3-PS EK15P2	HWE13770
006A	1627	VT233 EQU	+106	P2 COMP INLET EK15P3	HWE13780
006B	1628	VT234 EQU	+107	BLEED PRESS P23EK15P4	HWE13790
006C	1629	VT235 EQU	+108	POSITION INPUT EK15	HWE13800
006D	1630	VT236 EQU	+109	ANALOG SPEED INST	HWE13810
006E	1631	VT237 EQU	+110	BLEED PRESS P2.4 P5	HWE13820
006F	1632	VT238 EQU	+111	BLEED PRESS P2.5 P6	HWE13830
0070	1633	VT239 EQU	+112	TURBINE DISCH PRES P8	HWE13840
0071	1634	VT240 EQU	+113	ENGINE DISCH PRES P9	HWE13850
0072	1635	VT241 EQU	+114	PRESSURE RATIO	HWE13860
0073	1636	VT242 EQU	+115		HWE13870
0074	1637	VT243 EQU	+116		HWE13880
0075	1638	VT244 EQU	+117		HWE13890
	1639	*		THIRD STRIP EK18	HWE13900
0076	1640	VT245 EQU	+118	COMP TEMP INLET TA	HWE13910
0077	1641	VT246 EQU	+119	COMP TEMP DISCH TB	HWE13920
0078	1642	VT247 EQU	+120	TURBINE INLET TC	HWE13930
0079	1643	VT248 EQU	+121	TURBINE DISCH TD	HWE13940
007A	1644	VT249 EQU	+122	POWER LEVER PLA1	HWE13950
007B	1645	VT250 EQU	+123	POWER LEVER PLA2	HWE13960
007C	1646	VT251 EQU	+124	FILTER-LEAD-LAG VAR	HWE13970
007D	1647	VT252 EQU	+125	SPARE	HWE13980
007E	1648	VT253 EQU	+126	SPARE	HWE13990
	1649	*		SPARE POINT	HWE14000
007F	1650	VT254 EQU	+127		HWE14010

Table B-13. Bendix Bounds Program (Continued)

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				TRIMS LOCATION VALVES	
FFFF	1652	*			HWE14030
FFFE	1653	VT001	EQU	-1	HWE14040
FFF0	1654	VT002	EQU	-2	HWE14050
FFFC	1655	VT003	EQU	-3	HWE14060
FFFB	1656	VT004	EQU	-4	HWE14070
FFFA	1657	VT005	EQU	-5	HWE14080
FFF9	1658	VT006	EQU	-6	HWE14090
FFF8	1659	VT007	EQU	-7	HWE14100
FFF7	1660	VT008	EQU	-8	HWE14110
FFF6	1661	VT009	EQU	-9	HWE14120
FFF5	1662	VT010	EQU	-10	HWE14130
FFF4	1663	VT011	EQU	-11	HWE14140
FFF3	1664	VT012	EQU	-12	HWE14150
FFF2	1665	VT013	EQU	-13	HWE14160
FFF1	1666	VT014	EQU	-14	HWE14170
FFF0	1667	VT015	EQU	-15	HWE14180
FFEF	1668	VT016	EQU	-16	HWE14190
FFEE	1669	VT017	EQU	-17	HWE14200
FFED	1670	VT018	EQU	-18	HWE14210
FFEC	1671	VT019	EQU	-19	HWE14220
FFEB	1672	VT020	EQU	-20	HWE14230
FFEA	1673	VT021	EQU	-21	HWE14240
FFE9	1674	VT022	EQU	-22	HWE14250
FFE8	1675	VT023	EQU	-23	HWE14260
FFE7	1676	VT024	EQU	-24	HWE14270
FFE6	1677	VT025	EQU	-25	HWE14280
FFE5	1678	VT026	EQU	-26	HWE14290
FFE4	1679	VT027	EQU	-27	HWE14300
FFE3	1680	VT028	EQU	-28	HWE14310
FFE2	1681	VT029	EQU	-29	HWE14320
FFE1	1682	VT030	EQU	-30	HWE14330
FFE0	1683	VT031	EQU	-31	HWE14340
FFDF	1684	VT032	EQU	-32	HWE14350
FFDE	1685	VT033	EQU	-33	HWE14360
FFDD	1686	VT034	EQU	-34	HWE14370
FFDC	1687	VT035	EQU	-35	HWE14380
FFDB	1688	VT036	EQU	-36	HWE14390
FFDA	1689	VT037	EQU	-37	HWE14400
FFD9	1690	VT038	EQU	-38	HWE14410
FFD8	1691	VT039	EQU	-39	HWE14420
FFD7	1692	VT040	EQU	-40	HWE14430
FFD6	1693	VT041	EQU	-41	HWE14440
FFD5	1694	VT042	EQU	-42	HWE14450
FFD4	1695	VT043	EQU	-43	HWE14460
FFD3	1696	VT044	EQU	-44	HWE14470
FFD2	1697	VT045	EQU	-45	HWE14480
FFD1	1698	VT046	EQU	-46	HWE14490
FFD0	1699	VT047	EQU	-47	HWE14500
FFCF	1700	VT048	EQU	-48	HWE14510
FFCE	1701	VT049	EQU	-49	HWE14520
FFCD	1702	VT050	EQU	-50	HWE14530
FFCC	1703	VT051	EQU	-51	HWE14540
FFCB	1704	VT052	EQU	-52	HWE14550
FFCA	1705	VT053	EQU	-53	HWE14560
FFC9	1706	VT054	EQU	-54	HWE14570
FFC8	1707	VT055	EQU	-55	HWE14580
FFC7	1708	VT056	EQU	-56	HWE14590

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FFC7	1709	VT057 EQU	-57		HWE14600
FFC6	1710	VT058 EQU	-58		HWE14610
FFC5	1711	VT059 EQU	-59		HWE14620
FFC4	1712	VT060 EQU	-60		HWE14630
FFC3	1713	VT061 EQU	-61		HWE14640
FFC2	1714	VT062 EQU	-62		HWE14650
FFC1	1715	VT063 EQU	-63		HWE14660
FFC0	1716	VT064 EQU	-64		HWE14670
FFBF	1717	VT065 EQU	-65		HWE14680
FFBE	1718	VT066 EQU	-66		HWE14690
FFBD	1719	VT067 EQU	-67		HWE14700
FFBC	1720	VT068 EQU	-68		HWE14710
FFBB	1721	VT069 EQU	-69		HWE14720
FFBA	1722	VT070 EQU	-70		HWE14730
FFB9	1723	VT071 EQU	-71		HWE14740
FFB8	1724	VT072 EQU	-72		HWE14750
FFB7	1725	VT073 EQU	-73		HWE14760
FFB6	1726	VT074 EQU	-74		HWE14770
FFB5	1727	VT075 EQU	-75		HWE14780
FFB4	1728	VT076 EQU	-76		HWE14790
FFB3	1729	VT077 EQU	-77		HWE14800
FFB2	1730	VT078 EQU	-78		HWE14810
FFB1	1731	VT079 EQU	-79		HWE14820
FFB0	1732	VT080 EQU	-80		HWE14830
FFAF	1733	VT081 EQU	-81		HWE14840
FFAE	1734	VT082 EQU	-82		HWE14850
FFAD	1735	VT083 EQU	-83		HWE14860
FFAC	1736	VT084 EQU	-84		HWE14870
FFAB	1737	VT085 EQU	-85		HWE14880
FFAA	1738	VT086 EQU	-86		HWE14890
FFA9	1739	VT087 EQU	-87		HWE14900
FFA8	1740	VT088 EQU	-88		HWE14910
FFA7	1741	VT089 EQU	-89		HWE14920
FFA6	1742	VT090 EQU	-90		HWE14930
FFA5	1743	VT091 EQU	-91		HWE14940
FFA4	1744	VT092 EQU	-92		HWE14950
FFA3	1745	VT093 EQU	-93		HWE14960
FFA2	1746	VT094 EQU	-94		HWE14970
FFA1	1747	VT095 EQU	-95	T2=10XF DEG	HWE14980
FFA0	1748	VT096 EQU	-96	T3=10XF DEG	HWE14990
FF9F	1749	VT097 EQU	-97	T4=10XF DEG	HWE15000
FF9E	1750	VT098 EQU	-98	T5=10XF DEG	HWE15010
FF9D	1751	VT099 EQU	-99	ADJUSTMENT NUMBER SELECTED	HWE15020
FF9C	1752	VT100 EQU	-100	ADJUSTMENT REGISTER NUMBER	HWE15030
FF9B	1753	VT101 EQU	-101	SAFETY DIGITAL NUMBER	HWE15040
FF9A	1754	VT102 EQU	-102	PB=100XPSI	HWE15050
FF99	1755	VT103 EQU	-103	DP=1000XPSI	HWE15060
FF98	1756	VT104 EQU	-104	P2=1000XPSI	HWE15070
FF97	1757	VT105 EQU	-105	P23-P2=100XPSI	HWE15080
FF96	1758	VT106 EQU	-106	P24-P2=100XPSI	HWE15090
FF95	1759	VT107 EQU	-107	P25-P2=100XPSI	HWE15100
FF94	1760	VT108 EQU	-108	P5 =100XPSI	HWE15110
FF93	1761	VT109 EQU	-109	P0 =1000XPSI	HWE15120
FF92	1762	VT110 EQU	-110		HWE15130
FF91	1763	VT111 EQU	-111		HWE15140
FF90	1764	VT112 EQU	-112		HWE15150
FF8F	1765	VT113 EQU	-113		HWE15160

Table B-13. Bendix Bounds Program (Concluded)

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FF8E	1766	VT114 EQU	-114	HWE15170
FF8D	1767	VT115 EQU	-115	HWE15180
FF8C	1768	VT116 EQU	-116	HWE15190
FF8B	1769	VT117 EQU	-117	HWE15200
FF8A	1770	VT118 EQU	-118	HWE15210
FF89	1771	VT119 EQU	-119	HWE15220
FF88	1772	VT120 EQU	-120	HWE15230
FF87	1773	VT121 EQU	-121	HWE15240
FF86	1774	VT122 EQU	-122	HWE15250
FF85	1775	VT123 EQU	-123	HWE15260
FF84	1776	VT124 EQU	-124	HWE15270
FF83	1777	VT125 EQU	-125	HWE15280
FF82	1778	VT126 EQU	-126	HWE15290
FF81	1779	VT127 EQU	-127	HWE15300
058A	1780	END		HWE15310

000 ERROR(S) AND 000 WARNING(S) IN ABOVE ASSEMBLY.

Table B-14. Bendix Bounds Program
Cross Reference

SYMBOL	VALUE	RSL	DEFN	REFERENCES-
PTH34	0789	1	964	959R
PTH41	0406	1	1018	1013R
PTH42	03FE	1	1012	1007R
PTH43	03F6	1	1026	1001R
PTH44	03EC	1	999	994R
PTH51	0446	1	1075	1070R
PTH52	043E	1	1069	1064R
PTH53	0436	1	1063	1058R
PTH54	042D	1	1056	1051R
POSA	0564	1	391	1384R
P00	0000	0	1431	
P01	0001	0	1432	
P02	0002	0	1433	
P03	0003	0	1434	
P04	0004	0	1435	
P05	0005	0	1436	
P06	0006	0	1437	
P07	0007	0	1438	
P08	0008	0	1439	
P09	0009	0	1440	
P10	000A	0	1441	
P11	000B	0	1442	
P12	000C	0	1443	
P13	000D	0	1444	
P14	000E	0	1445	
P15	000F	0	1446	
P16	0010	0	1447	
P17	0011	0	1448	
P18	0012	0	1449	506R
P19	0013	0	1450	508R
P20	0014	0	1451	510R
P21	0015	0	1452	512R
P22	0016	0	1453	514R
P23	0017	0	1454	516R
P24	0018	0	1455	518R
P25	0019	0	1456	520R
P26	001A	0	1457	522R
P27	001B	0	1458	525R
P28	001C	0	1459	527R
P29	001D	0	1460	529R
P30	001E	0	1461	531R
P31	001F	0	1462	533R
P32	0020	0	1463	535R
P33	0021	0	1464	537R
P34	0022	0	1465	539R
P35	0023	0	1466	541R
P36	0024	0	1467	547R
P37	0025	0	1468	549R
P38	0026	0	1469	551R
P39	0027	0	1470	553R
P40	0028	0	1471	555R
P41	0029	0	1472	560R
P42	002A	0	1473	565R
P43	002B	0	1474	570R
P44	002C	0	1475	574R
P45	002D	0	1476	577R
P46	002E	0	1477	579R
P47	002F	0	1478	583R
P48	0030	0	1479	587R

Table B-14. Bendix Bounds Program
Cross Reference (Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-
P49	0031	0	1480	591R
P50	0032	0	1481	596R
P51	0033	0	1482	598R
P52	0034	0	1483	600R
P53	0035	0	1484	602R
P54	0036	0	1485	605R
P55	0037	0	1486	611P
P56	0038	0	1487	617R
P57	0039	0	1488	622R
P58	003A	0	1489	627R
P59	003B	0	1490	629R
P60	003C	0	1491	631R
P61	003D	0	1492	633R
P62	003E	0	1493	635R
P63	003F	0	1494	638R
RAWN	01E3	1	483	461M 465R
R000T	0165	1	360	355R
RPM	01E0	1	479	455R
RSTAL	0010	1	13	369R
RSTSA	0175	1	378	373R
SAFND	01C6	1	451	435R 443R 447R
SAM1	02C2	1	721	681M
SAM2	02C3	1	722	720M
SAM3	02E0	1	751	744M 746R
SAM4	02E1	1	752	750M
SAM6	04DF	1	1241	1221M
SAM7	04EA	1	1252	1243M
SAM8	04F0	1	1259	1226M 1250M
START	0147	1	337	12R
STTVT	00A9	1	188	74M
ST000	004E	1	78	388R
ST001	004F	1	79	14R
ST002	0050	1	80	16R
ST003	0051	1	81	18R
ST004	0052	1	82	20R
ST005	0053	1	83	22R
ST006	0054	1	84	24R
ST007	0055	1	85	26R
ST008	0056	1	86	28R
ST009	0057	1	89	30R
ST010	0058	1	90	32R
ST011	0059	1	91	34R
ST012	005A	1	94	36R
ST013	005B	1	95	38R
ST014	005C	1	96	40R
ST015	005D	1	97	42R
ST016	005E	1	98	44R
ST017	005F	1	99	46R
ST018	0060	1	100	48R
ST019	0061	1	101	50R
ST020	0062	1	102	52R
ST021	0063	1	103	54R
ST022	0064	1	104	56R
ST023	0065	1	105	58R
ST024	0066	1	109	60R
ST025	0067	1	110	62R
ST026	0068	1	111	64R
ST027	0069	1	114	66R
ST028	006A	1	115	68R

Table B-14. Bendix Bounds Program
Cross Reference (Continued)

SYMBOL	VALUE	REL	DSFN	REFERENCES-
ST029	006B	1	116	70R
ST030	006C	1	117	72R
ST031	006D	1	118	189R
ST032	006E	1	119	191R
ST033	006F	1	120	193R
ST034	0070	1	121	195R
ST035	0071	1	122	197R
ST036	0072	1	126	199R
ST037	0073	1	127	201R
ST038	0074	1	128	203R
ST039	0075	1	129	205R
ST040	0076	1	130	207R
ST041	0077	1	131	209R
ST042	0078	1	132	211R
ST043	0079	1	133	213R
ST044	007A	1	134	215R
ST045	007B	1	135	217R
ST046	007C	1	136	219R
ST047	007D	1	137	221R
ST048	007E	1	138	223R
ST049	007F	1	139	225R
ST050	0080	1	140	227R
ST051	0081	1	146	229R
ST052	0082	1	147	231R
ST053	0083	1	148	233R
ST054	0084	1	149	235R
ST055	0085	1	152	237R
ST056	0086	1	153	239R
ST057	0087	1	154	241R
ST058	0088	1	155	243R
ST059	0089	1	156	245R
ST060	008A	1	157	247R
ST061	008B	1	158	249R
ST062	008C	1	159	251R
ST063	008D	1	160	253R
ST064	008E	1	161	255R
ST065	008F	1	162	257R
ST066	0090	1	163	259R
ST067	0091	1	164	261R
ST068	0092	1	165	263R
ST069	0093	1	166	265R
ST070	0094	1	167	267R
ST071	0095	1	168	269R
ST072	0096	1	169	271R
ST073	0097	1	170	273R
ST074	0098	1	171	275R
ST075	0099	1	172	277R
ST076	009A	1	173	279R
ST077	009B	1	174	281R
ST078	009C	1	175	283R
ST079	009D	1	176	285R
ST080	009E	1	177	287R
ST081	009F	1	178	289R
ST082	00A0	1	179	291R
ST083	00A1	1	180	293R
ST084	00A2	1	181	295R
ST085	00A3	1	182	297R
ST086	00A4	1	183	299R
ST087	00A5	1	184	301R

Table B-14. Bendix Bounds Program
Cross Reference (Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-																
ST088	00A6	1	185	303R																
ST089	00A7	1	186	305R																
ST090	00A8	1	187	307R																
TEMPA	04DE	1	1238	1246M	1248R															
TEMP2	01DD	1	474	477R																
TEMP3	0142	1	331	366M																
TEMP4	0143	1	332	354M	376R															
TEMP5	0144	1	333	387M	392M															
TEMP6	0470	1	1121	1100M	1102R															
TESTN	01E2	1	482	454M	457M															
TMNR	0140	!	329	346M	350M	351R	400R	416R												
TRIMS	0141	1	330	467M	544M															
T2100	03E1	1	990	806R																
T2125	0422	1	1047	804R																
T225	033E	1	849	812R																
T250	0381	1	909	810R																
T275	0386	1	955	808R																
VALID	01CF	1	459	456M																
VALPO	0484	1	1143	1142M																
VALUE	013E	1	327	324R	353M	357R	359M													
VLVEL	0188	1	440	431R																
VT001	FFFF	0	1653	15M	657R															
VT002	FFFE	0	1654	17M	661R															
VT003	FFFD	0	1655	19M	762R															
VT004	FFFC	0	1656	21M	674R															
VT005	FFFB	0	1657	23M	687R															
VT006	FFFA	0	1658	25M	689R															
VT007	FFF9	0	1659	27M	694R															
VT008	FFF8	0	1660	29M	697R															
VT009	FFF7	0	1661	31M	754R															
VT010	FFF6	0	1662	33M	765R															
VT011	FFF5	0	1663	35M	1161R															
VT012	FFF4	0	1664	37M																
VT013	FFF3	0	1665	39M																
VT014	FFF2	0	1666	41M																
VT015	FFF1	0	1667	43M																
VT016	FFF0	0	1668	45M																
VT017	FFEF	0	1669	47M																
VT018	FFEE	0	1670	49M																
VT019	FFED	0	1671	51M																
VT020	FFEC	0	1672	53M																
VT021	FFEB	0	1673	55M																
VT022	FFEA	0	1674	57M																
VT023	FFE9	0	1675	59M																
VT024	FFE8	0	1676	61M	843R	846R	878R	881R	938R	941R	984R	987R	1019R	1022R	1076R					
				1079R																
VT025	FFE7	0	1677	63M	814R	849R	909R	955R	990R	1047R										
VT026	FFE6	0	1678	65M	1144R															
VT027	FFE5	0	1679	67M	1167R															
VT028	FFE4	0	1680	69M	1364R															
VT029	FFE3	0	1681	71M	1089R															
VT030	FFE2	0	1682	73M	1084R															
VT031	FFE1	0	1683	190M	1148R															
VT032	FFE0	0	1684	192M	1154R															
VT033	FFDF	0	1685	194M	1157R															
VT034	FFDE	0	1686	196M																
VT035	FFDD	0	1687	198M																
VT036	FFDC	0	1688	200M																
VT037	FFDB	0	1689	202M																

**Table B-14. Bendix Bounds Program
Cross Reference (Continued)**

SYMBOL	VALUE	REL	DEFN	REFERENCES-
VT038	FFDA	0	1690	204M
VT039	FFD9	0	1691	206M
VT040	FFD8	0	1692	208M
VT041	FFD7	0	1693	210M
VT042	FFD6	0	1694	212M
VT043	FFD5	0	1695	214M
VT044	FFD4	0	1696	216M
VT045	FFD3	0	1697	218M
VT046	FFD2	0	1698	220M
VT047	FFD1	0	1699	222M
VT048	FFD0	0	1700	224M
VT049	FFCF	0	1701	226M
VT050	FFCE	0	1702	228M
VT051	FFCD	0	1703	230M 1259R
VT052	FFCC	0	1704	232M 1262R
VT053	FFCB	0	1705	234M 1281R
VT054	FFCA	0	1706	236M 1284R
VT055	FFC9	0	1707	238M 1339R
VT056	FFC8	0	1708	240M 1348R
VT057	FFC7	J	1709	242M 1354R
VT058	FFC6	0	1710	244M
VT059	FFC5	0	1711	246M
VT060	FFC4	0	1712	248M
VT061	FFC3	0	1713	250M
VT062	FFC2	0	1714	252M
VT063	FFC1	0	1715	254M
VT064	FFC0	0	1716	256M
VT065	FFBF	0	1717	258M
VT066	FFBE	0	1718	260M
VT067	FFBD	0	1719	262M
VT068	FFBC	0	1720	264M
VT069	FFBB	0	1721	266M
VT070	FFBA	0	1722	268M
VT071	FFB9	0	1723	270M
VT072	FFB8	0	1724	272M
VT073	FFB7	0	1725	274M
VT074	FFB6	0	1726	276M 1385R
VT075	FFB5	0	1727	278M
VT076	FFB4	0	1728	280M
VT077	FFB3	0	1729	282M
VT078	FFB2	0	1730	284M
VT079	FFB1	0	1731	286M
VT080	FFB0	0	1732	288M
VT081	FFAF	0	1733	290M 1367R
VT082	FFAE	0	1734	292M
VT083	FFAD	0	1735	294M
VT084	FFAC	0	1736	296M
VT085	FFAB	0	1737	298M
VT086	FFAA	0	1738	300M
VT087	FFA9	0	1739	302M
VT088	FFA8	0	1740	304M
VT089	FFA7	0	1741	306M
VT090	FFA6	0	1742	308M
VT091	FFA5	0	1743	
VT092	FFA4	0	1744	
VT093	FFA3	0	1745	462M
VT094	FFA2	0	1746	342M
VT095	FFA1	0	1747	610M
VT096	FFA0	0	1748	616M

Table B-14. Bendix Bounds Program
Cross Reference (Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-
VT097	FF9F	0	1749	621M
VT098	FF9E	0	1750	626M
VT099	FF9D	0	1751	371R 375M
VT100	FF9C	0	1752	377M
VT101	FF9B	0	1753	367M 378R 397R 413R 429R 432R 444R
VT102	FF9A	0	1754	559M
VT103	FF99	0	1755	564M
VT104	FF98	0	1756	569M
VT105	FF97	0	1757	573M
VT106	FF96	0	1758	582M
VT107	FF95	0	1759	586M
VT108	FF94	0	1760	590M
VT109	FF93	0	1761	595M
VT110	FF92	0	1762	507M
VT111	FF91	0	1763	509M
VT112	FF90	0	1764	511M
VT113	FF8F	0	1765	513M
VT114	FF8E	0	1766	515M
VT115	FF8D	0	1767	517M
VT116	FF8C	0	1768	519M
VT117	FF8B	0	1769	521M
VT118	FF8A	0	1770	523M
VT119	FF89	0	1771	526M
VT120	FF88	0	1772	528M
VT121	FF87	0	1773	530M
VT122	FF86	0	1774	532M
VT123	FF85	0	1775	534M
VT124	FF84	0	1776	536M
VT125	FF83	0	1777	538M
VT126	FF82	0	1778	540M
VT127	FF81	0	1779	542M
VT128	0001	0	1501	678M 705R 721R
VT129	0002	0	1502	707M
VT130	0003	0	1503	712M
VT131	0004	0	1504	718M
VT132	0005	0	1505	706R 719R 722M 730R 731R
VT133	0006	0	1506	684M 689R 692R
VT134	0007	0	1507	696M 701R 702R
VT135	0008	0	1508	
VT136	0009	0	1509	
VT137	000A	0	1510	
VT138	000B	0	1511	
VT139	000C	0	1512	
VT140	000D	0	1513	1086M 1091R
VT141	000E	0	1514	1163M
VT142	000F	0	1515	1156M 1191R 1192R
VT143	0010	0	1516	1159M 1173R 1176R
VT144	0011	0	1517	1261M 1265R 1270R
VT145	0012	0	1518	1283M 1287R 1292R
VT146	0013	0	1519	
VT147	0014	0	1520	1165M 1208R
VT148	0015	0	1521	1177M 1203R 1206R 1379R
VT149	0016	0	1522	401M 404R
VT150	0017	0	1524	668M
VT151	0018	0	1525	660M 670R 673R
VT152	0019	0	1526	664M 675R 676R
VT153	001A	0	1527	693M 709R 710R
VT154	001B	0	1528	704M 714R 717R
VT155	001C	0	1529	728M

Table B-14. Bendix Bounds Program
Cross Reference (Continued)

SYMBOL	VALUE	REL	DEFN	REFERENCES-
VT156	001D	0	1530	733M 759R 1322R 1332K
VT157	001E	0	1532	441R 466M 740R 760R 817R 824R 844R 852R 859R 879R 917R 919R 939R 958R 965R 985R 993R 1000R 1020R 1050R 1057R 1077R 1087R 1095R 1108R 1268R 1290R
VT158	001F	0	1533	752M 1195R 1196R
VT159	0020	0	1534	761M
VT160	0021	0	1535	764M 767R
VT161	0022	0	1539	768M 777M 1179R 1180R
VT162	0023	0	1540	
VT163	0024	0	1541	
VT164	0025	0	1543	
VT165	0026	U	1544	
VT166	0027	0	1545	
VT167	0028	U	1546	
VT168	0029	0	1548	
VT169	002A	0	1550	
VT170	002B	0	1551	
VT171	002C	0	1552	
VT172	002D	0	1555	
VT173	002E	0	1556	
VT174	002F	U	1557	
VT175	0030	U	1558	
VT176	0031	0	1559	
VT177	0032	0	1560	
VT178	0033	0	1561	
VT179	0034	U	1562	
VT180	0035	0	1563	758M 1140R 1141M 1199R 1200R
VT181	0036	0	1564	1082M 1145R
VT182	0037	U	1565	1147M 1178R
VT183	0038	0	1566	1187M
VT184	0039	0	1567	1190M 1383R
VT185	003A	U	1569	1194M
VT186	003B	0	1570	1198M
VT187	003C	0	1571	1202M
VT188	003D	0	1572	1105M 1109R 1110R
VT189	003E	0	1573	1112M
VT190	003F	0	1574	1093M 1114R 1117R
VT191	0040	0	1575	1118M 1168R
VT192	0041	U	1576	1172M
VT193	0042	0	1577	1207M 1377R 1381R
VT194	0043	U	1578	1153M 1170R 1168R
VT195	0044	U	1579	1209M
VT196	0045	U	1580	
VT197	0046	U	1583	
VT198	0047	U	1584	
VT199	0048	U	1585	
VT200	0049	U	1586	
VT201	004A	U	1587	
VT202	004B	0	1588	
VT203	004C	0	1591	
VT204	004D	0	1592	
VT205	004E	0	1593	
VT206	004F	0	1594	
VT207	0050	U	1595	
VT208	0051	0	1598	1218M 1256R 1269R 1268R 1291R
VT209	0052	0	1599	
VT210	0053	U	1600	1225M 1249M 1257M 1264R 1286R
VT211	0054	0	1601	
VT212	0055	0	1602	1267M 1277R

**Table B-14. Bendix Bounds Program
Cross Reference (Continued)**

SYMBOL	VALUE	REL	DEFN	REFERENCES-
VT213	0056	0	1603	1279M
VT214	0057	0	1604	1289M 1299R
VT215	0058	0	1605	1301M
VT216	0059	0	1608	
VT217	005A	0	1609	1330M 1341R 1344R
VT218	005B	0	1610	1345M 1356R 1358R 1361R
VT219	005C	0	1611	1362M
VT220	005D	0	1612	410M
VT221	005E	0	1612	
VT222	005F	0	1614	
VT223	0060	0	1615	
VT224	0061	0	1616	417M 420R
VT225	0062	0	1617	426M
VT226	0063	0	1618	1386M 1389M 1392M
VT227	0064	0	1621	548M
VT228	0065	0	1622	550M
VT229	0066	0	1623	552M
VT230	0067	0	1624	554M
VT231	0068	0	1625	556M 686R 698R 1151R
VT232	0069	0	1626	561M
VT233	006A	0	1627	566M 747R
VT234	006B	0	1628	571M
VT235	006C	0	1629	575M
VT236	006D	0	1630	578M 737R 738R
VT237	006E	0	1631	580M
VT238	006F	0	1632	584M
VT239	0070	0	1633	588M
VT240	0071	0	1634	592M
VT241	0072	0	1635	597M
VT242	0073	0	1636	599M
VT243	0074	0	1637	601M
VT244	0075	0	1638	603M
VT245	0076	0	1640	607M 723R 802R 1213R 1219R 1241R 1252R
VT246	0077	0	1641	613M
VT247	0078	0	1642	
VT248	0079	0	1643	624M 1347R
VT249	007A	0	1644	628M 645R
VT250	007B	0	1645	630M
VT251	007C	0	1646	632M
VT252	007D	0	1647	634M
VT253	007E	0	1648	636M
VT254	007F	0	1650	639M
WFP3	02F7	1	785	769M
XR1	0567	1	1396	3M
XR2	0569	1	1397	4M
XR3	056F	1	1398	5M
GTECT				
DMP FUNCTION COMPLETED				
*STORE			GTECT	HWE15320
GTECT				
DMP FUNCTION COMPLETED				

**Table B-14. Bendix Bounds Program
Cross Reference (Concluded)**

```
// JOB          VOISK   17 JUL 74 16.083 HRS
// DMP          17 JUL 74 16.083 HRS
*DELETE      S          GTE85 *****
DMP FUNCTION COMPLETED
*STORECI     S          GTE85                                04

*INCLDGTEIN/0400,GTECT/0604,GETTM/0909
*CCEND
```

MPX, BUILD GTE85

R20 GTEIN LEV.0 NON-REENT PROG

R20 GTECT LEV.0 NON-REENT PROG

R20 GETTM LEV.0 NON-REENT PROG

R20 MHECT LEV.0 NON-REENT PROG

MPX, GTE85 LD XQ

CL WC OF 0080 STORED AT 04FE

DMP FUNCTION COMPLETED

// END 17 JUL 74 16.105 HRS

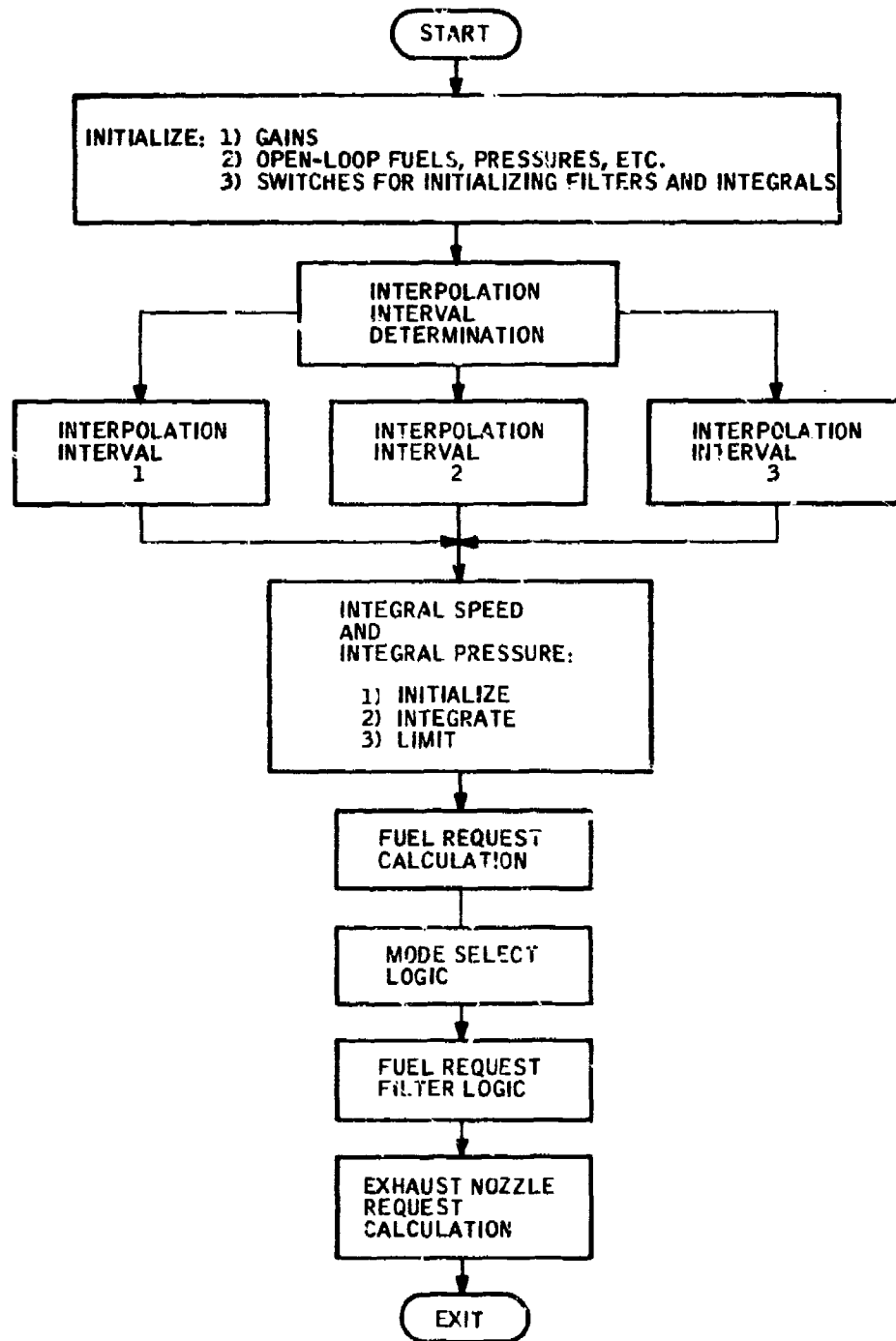


Figure B-1. Functional Flow Diagram Speed and Pressure Control Program

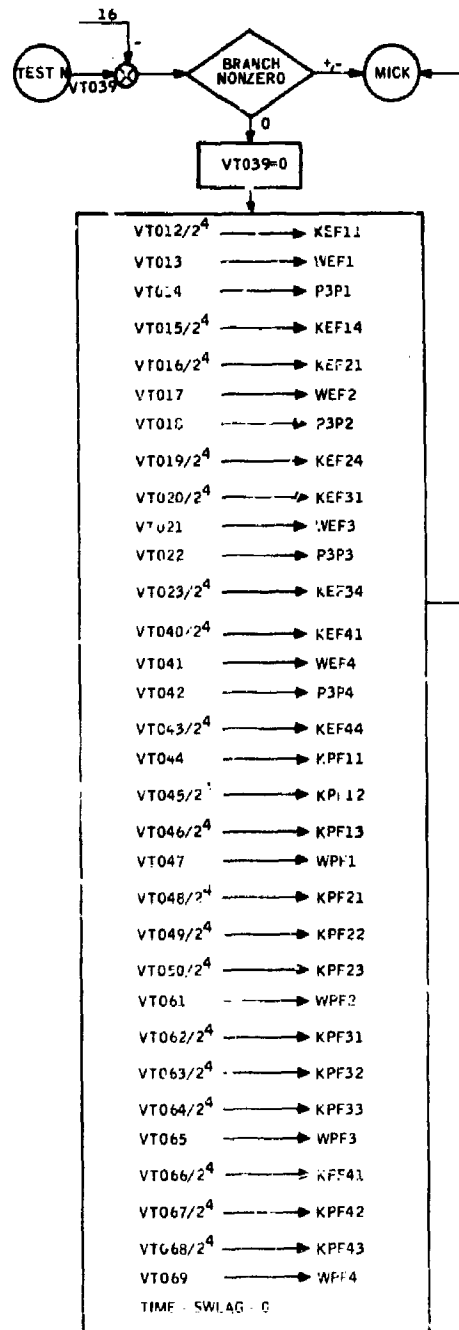


Figure B-2. Initialization Logic for Speed and Pressure Program

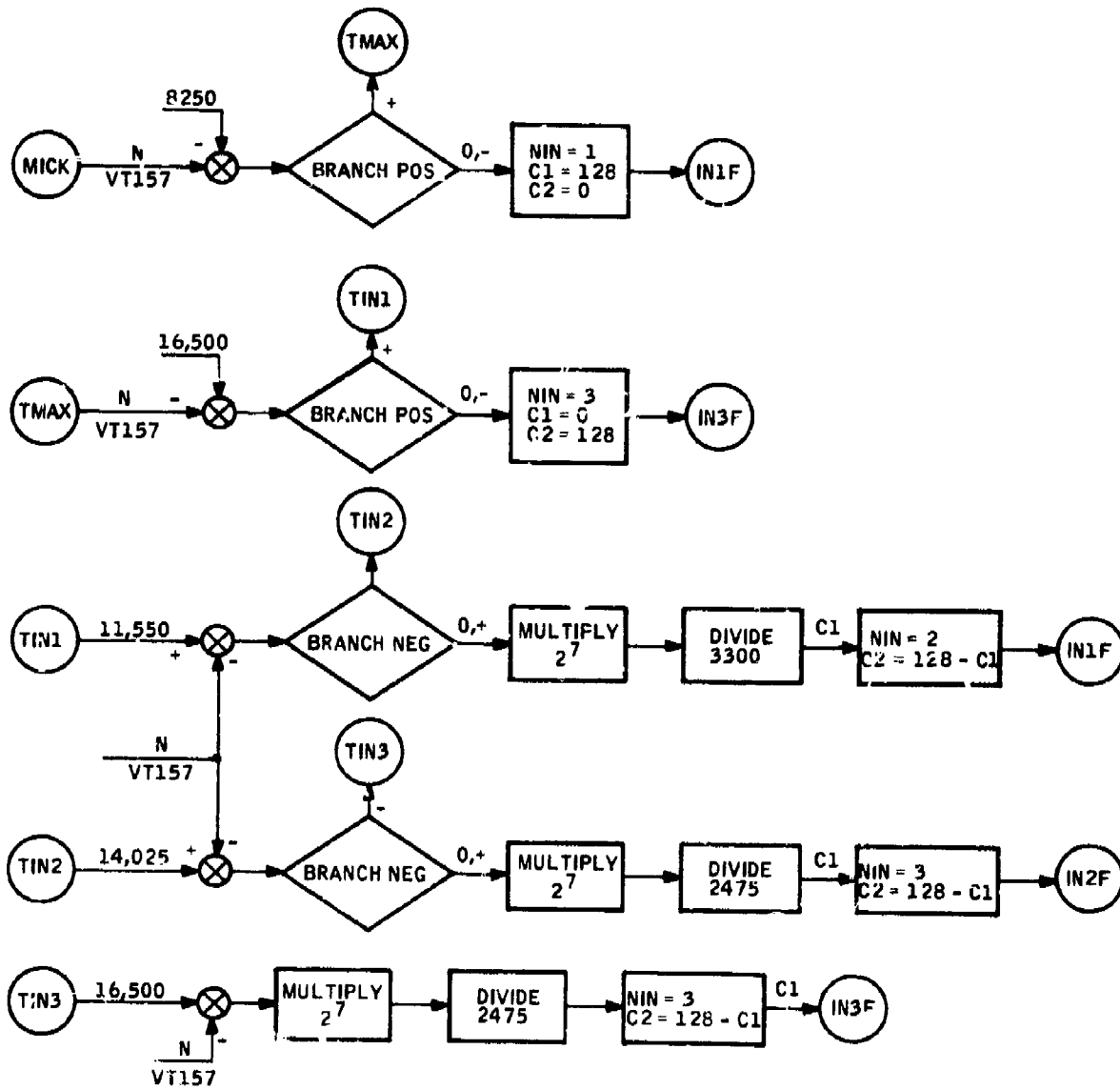


Figure B-3. Interval Determination

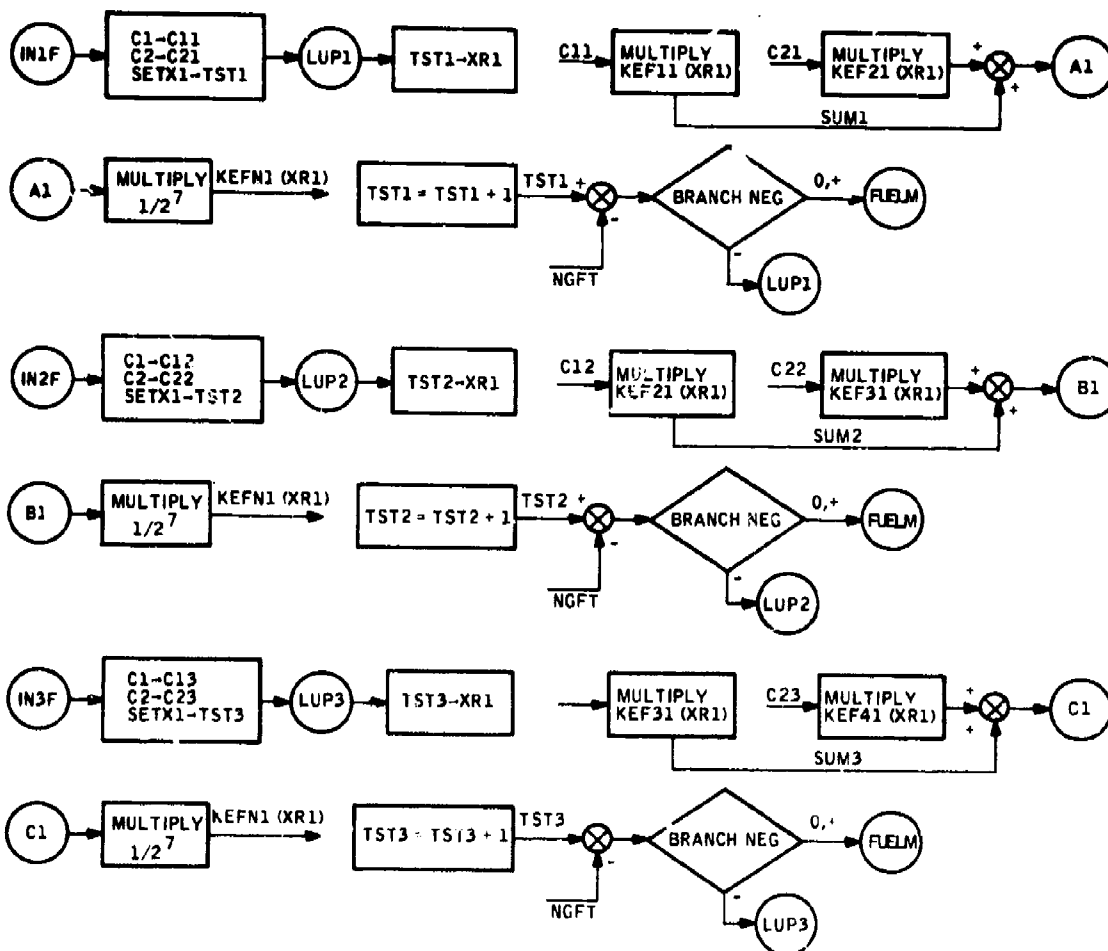


Figure B-4. Interpolation Logic

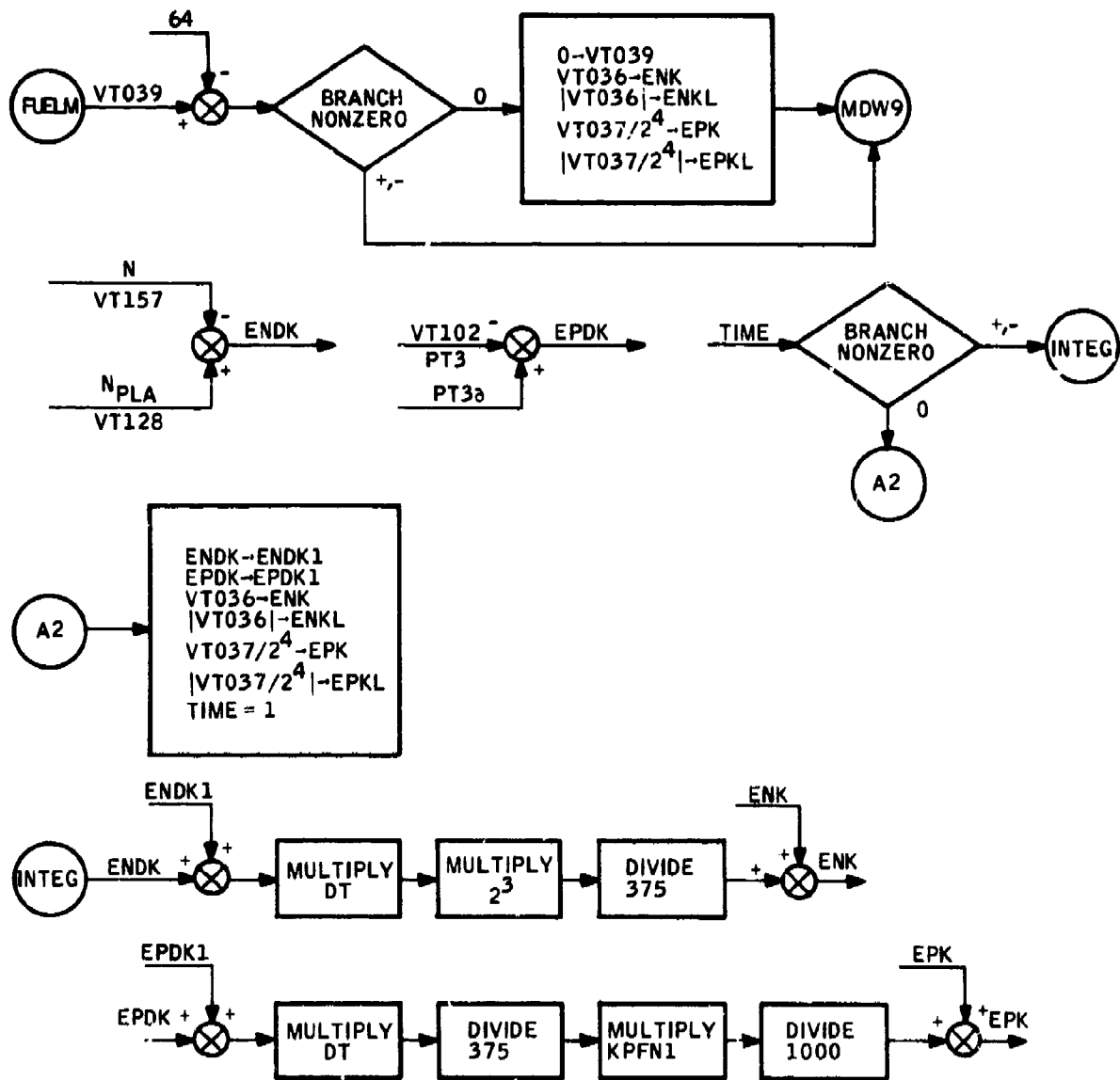


Figure B-5. Integral Speed and Pressure Calculation

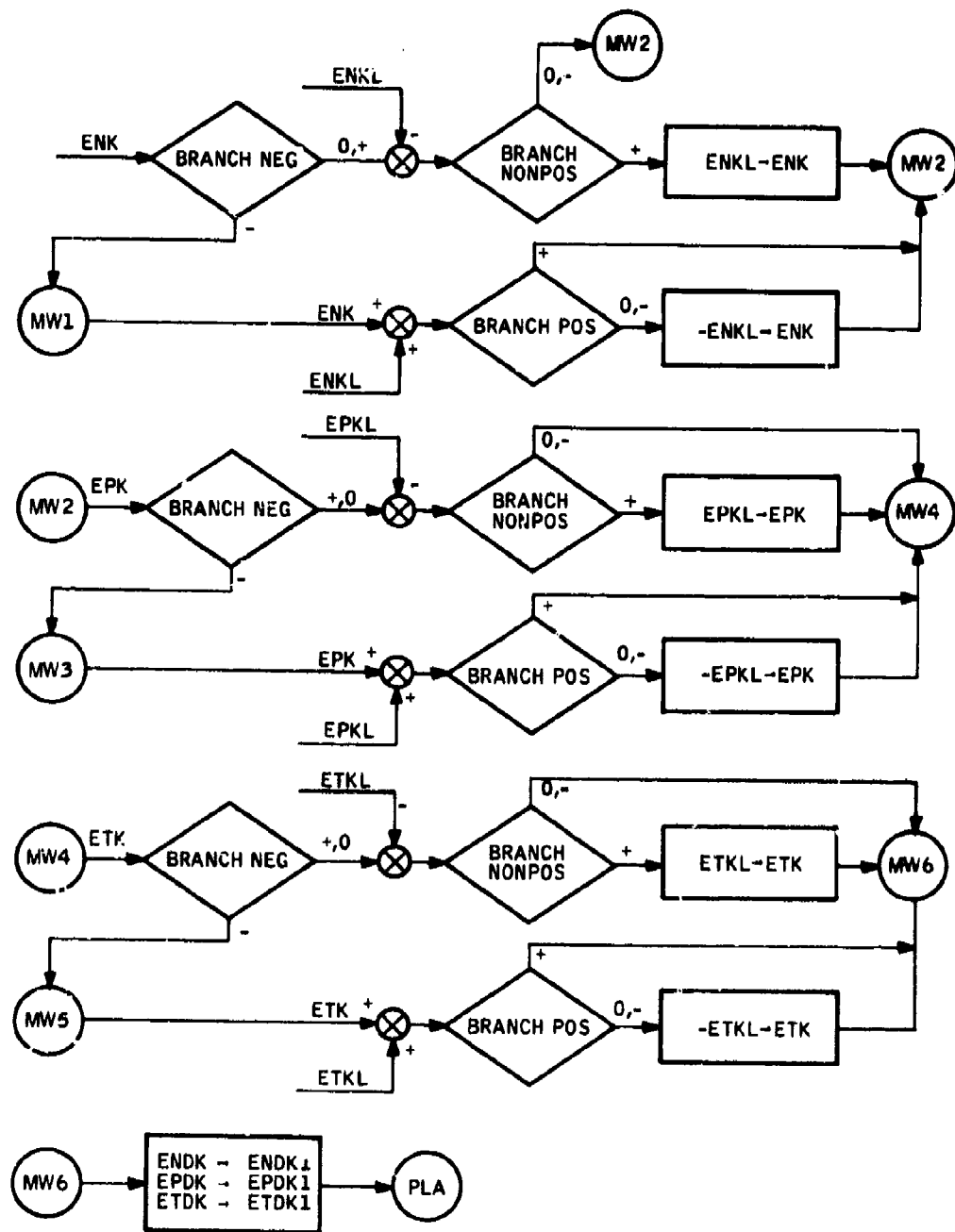


Figure B-6. Limiting Logic for Integral Speed and Pressure

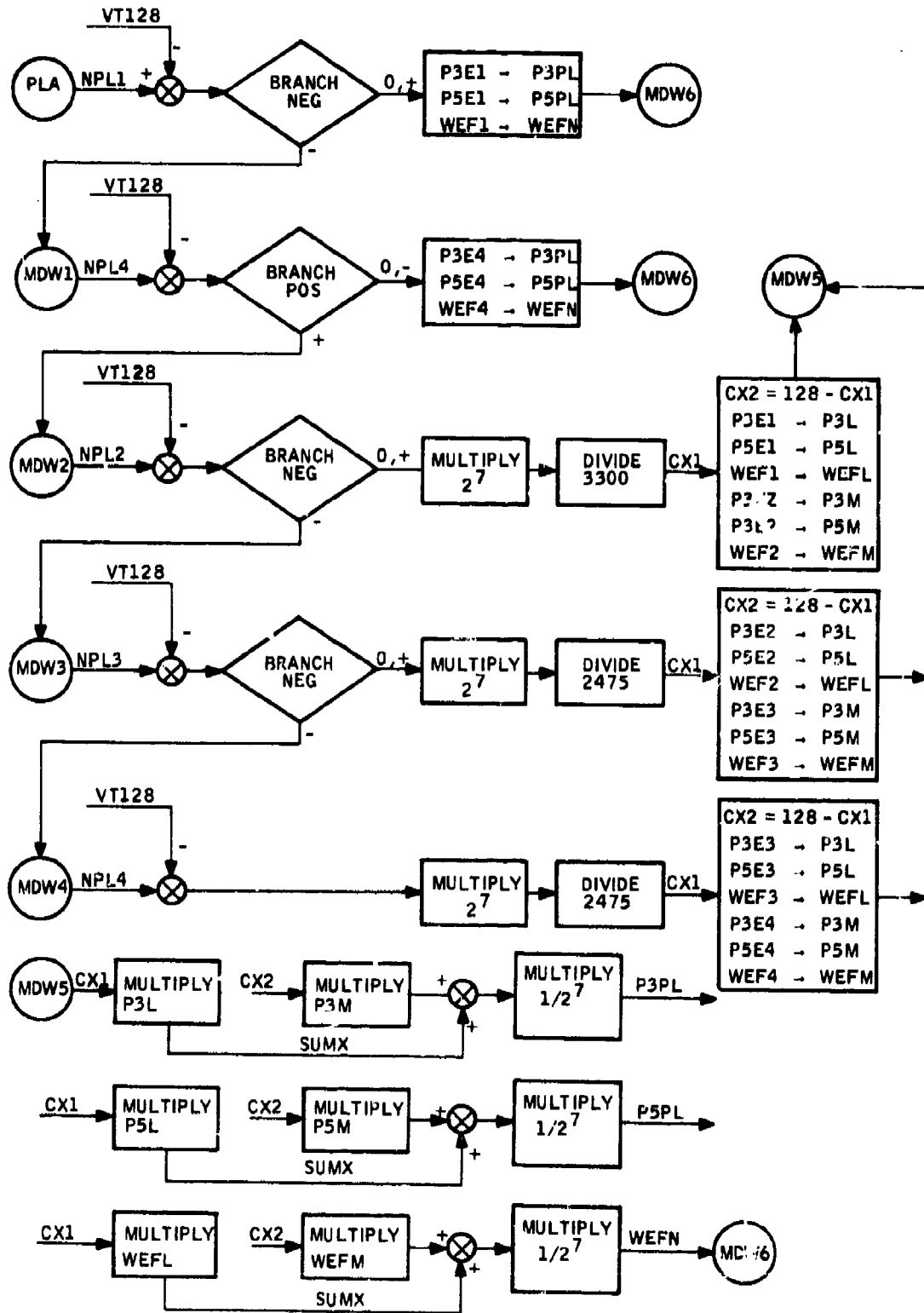


Figure B-7. Interpolation for PT3, PT5 and Fuel Request as a Function of Lower Lever

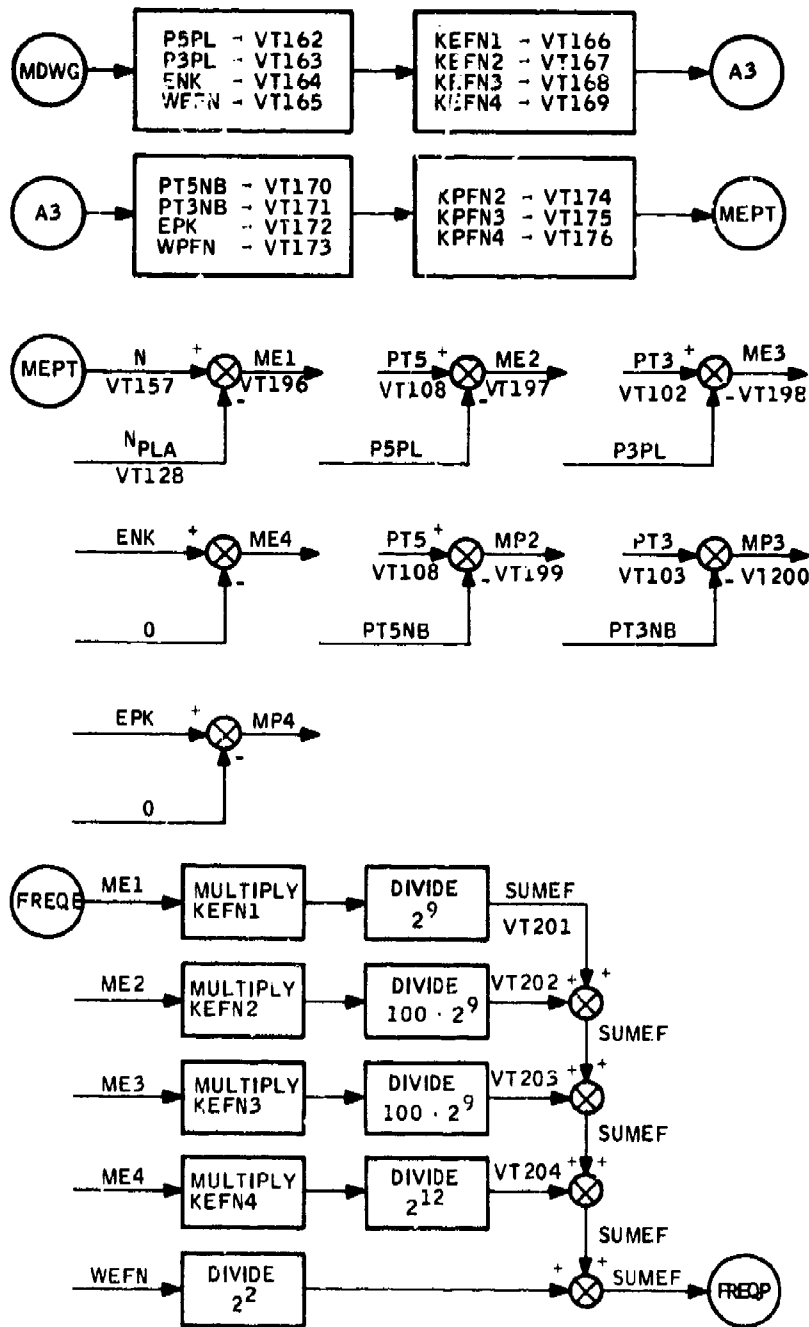


Figure B-8. Fuel Request Calculation

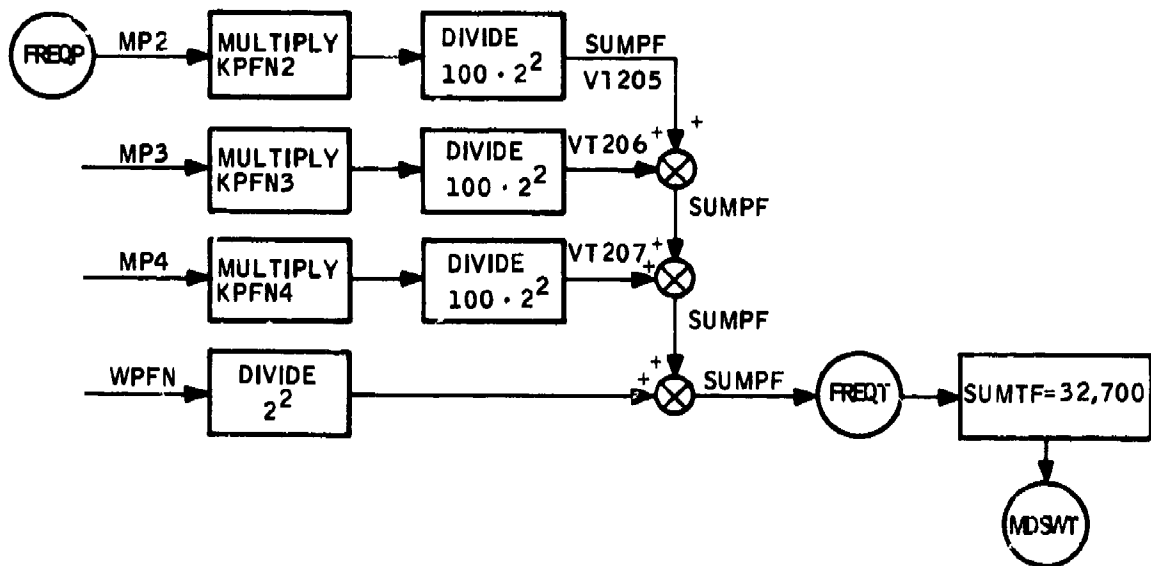


Figure B-8. Fuel Request Calculation
(Concluded)

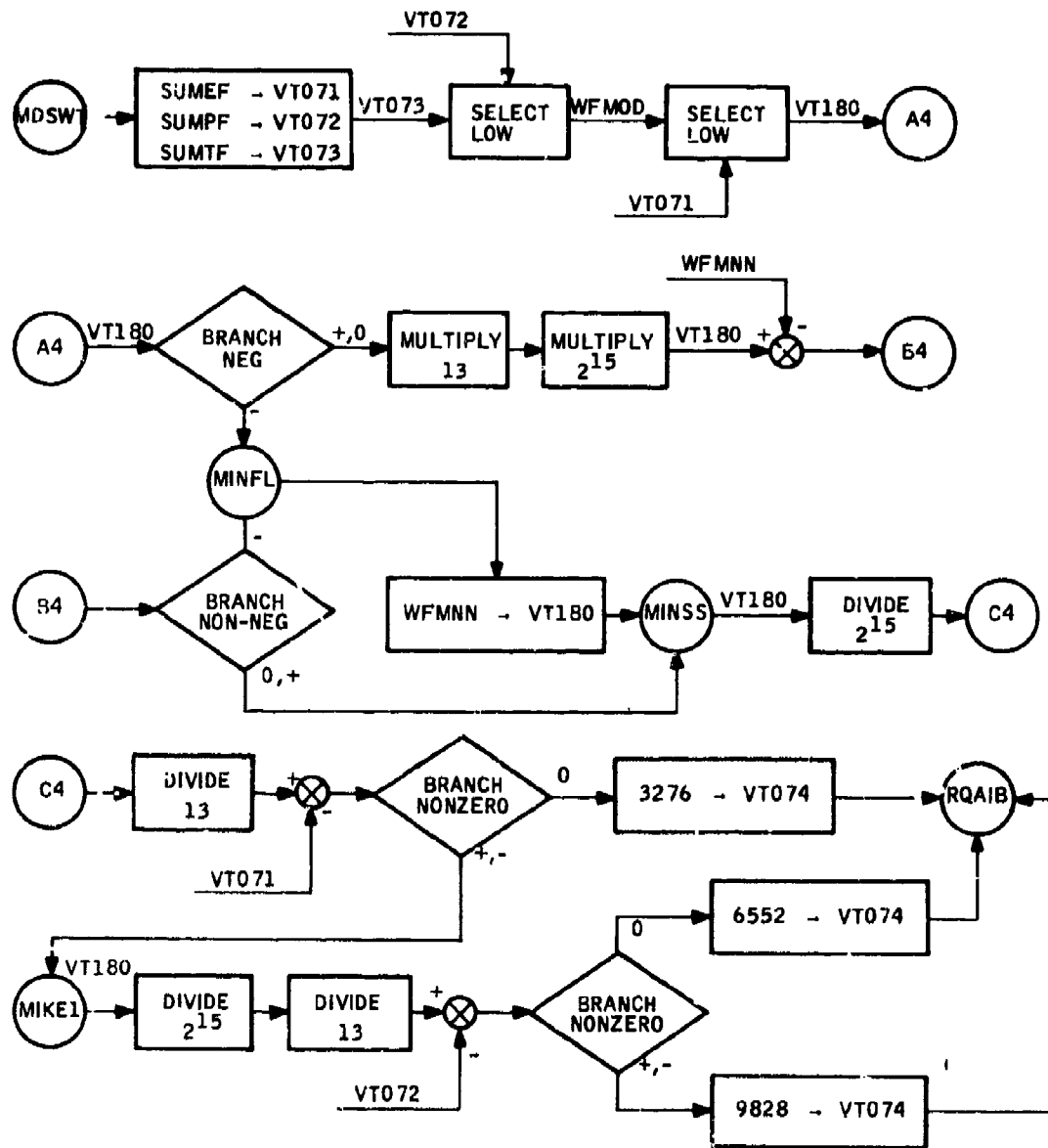


Figure B-9. Mode Select Logic

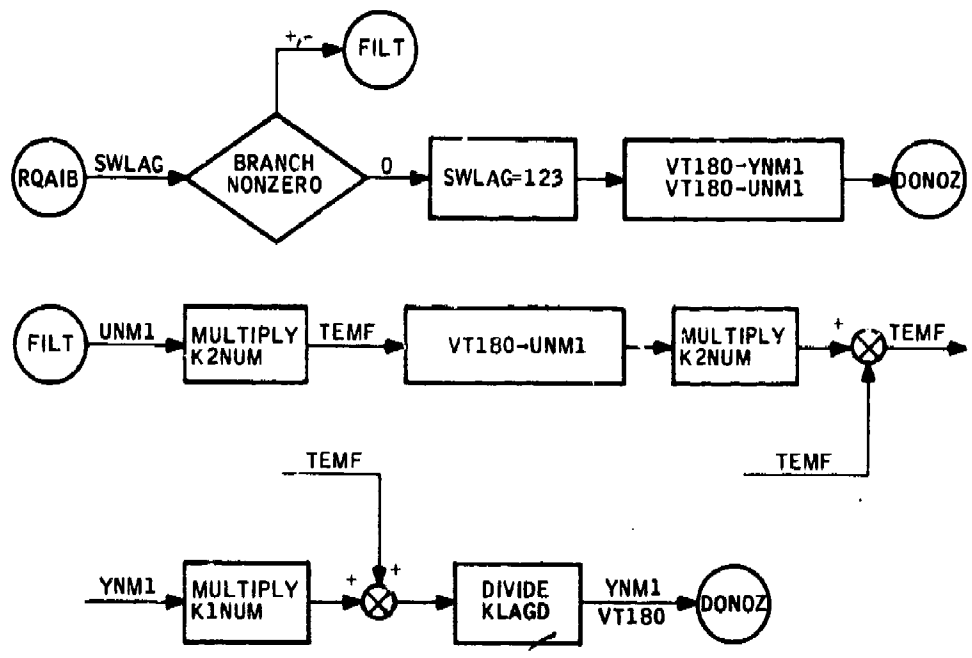


Figure B-10. Fuel Request Filter Logic

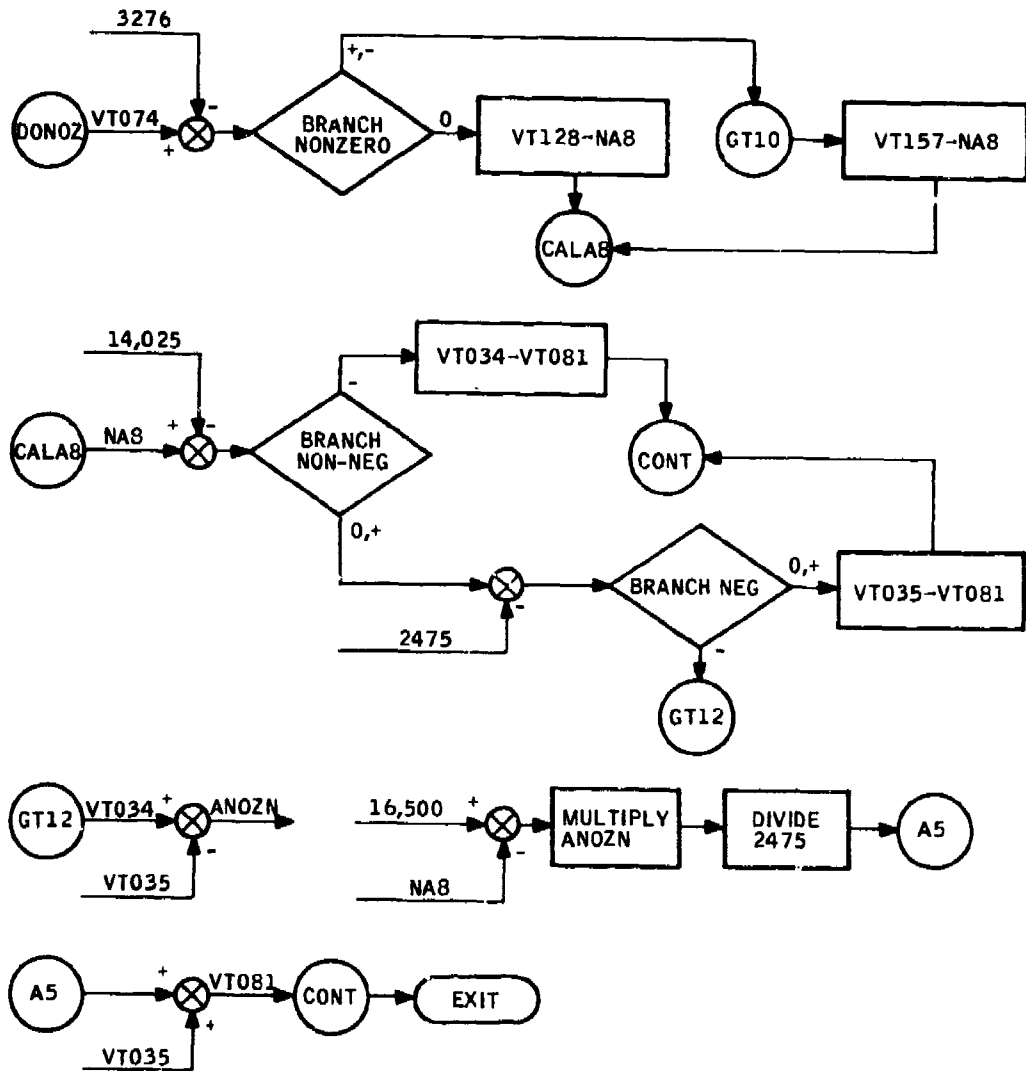


Figure B-11. Exhaust Nozzle Request Calculation

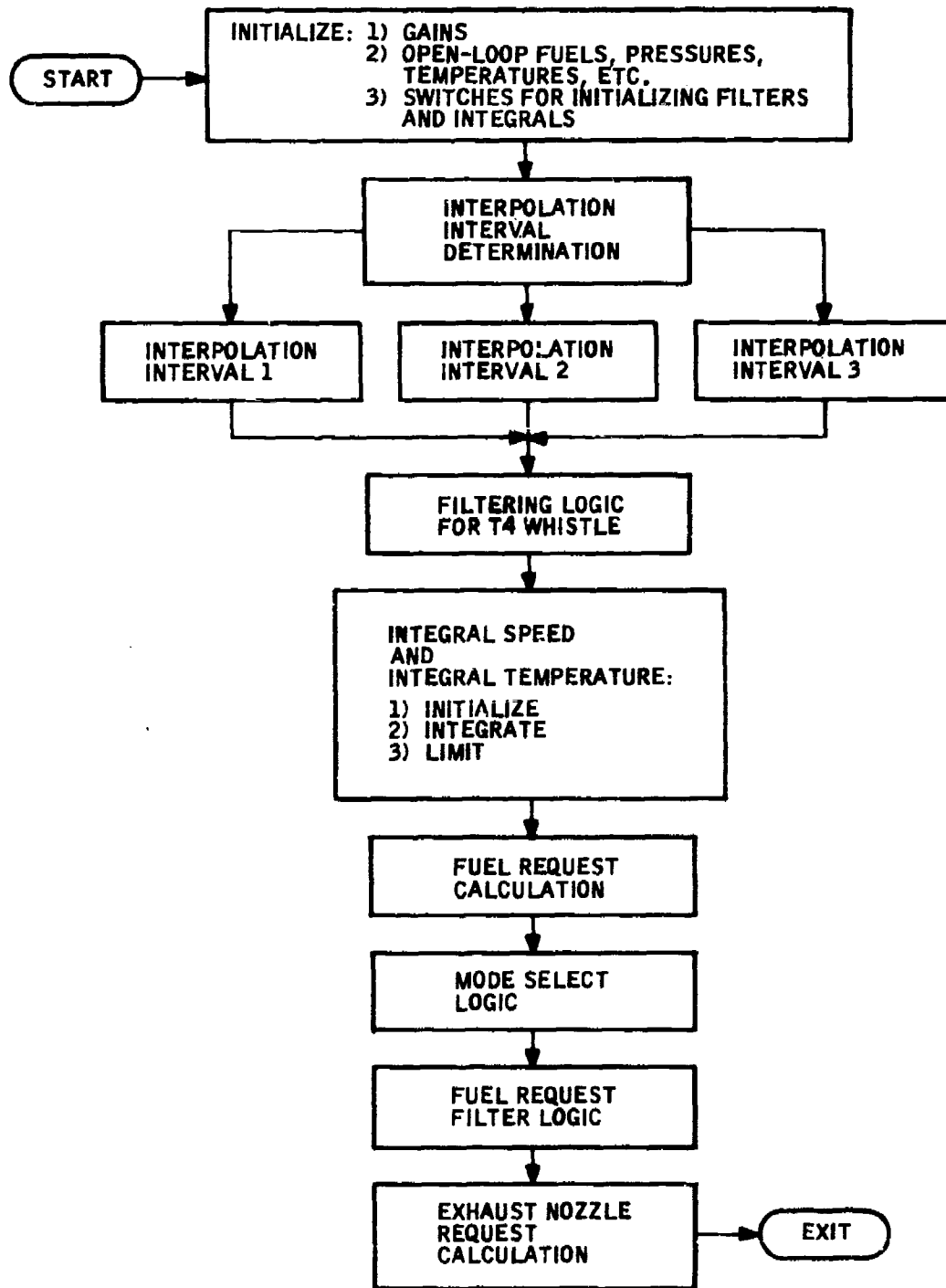


Figure B-12. Functional Flow Diagram Speed and Temperature Control Program

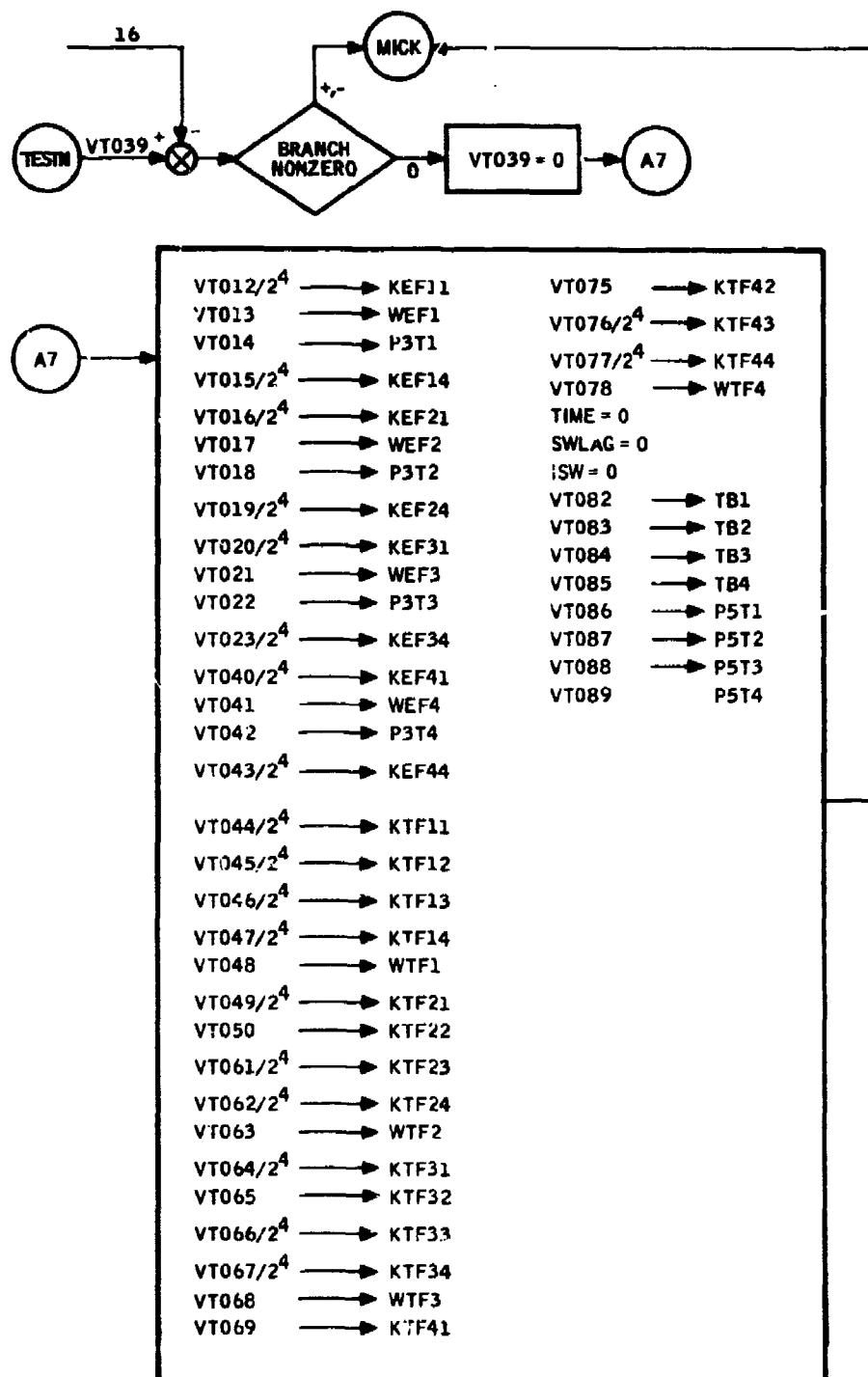


Figure B-13. Initialization Logic for Speed and Temperature Program

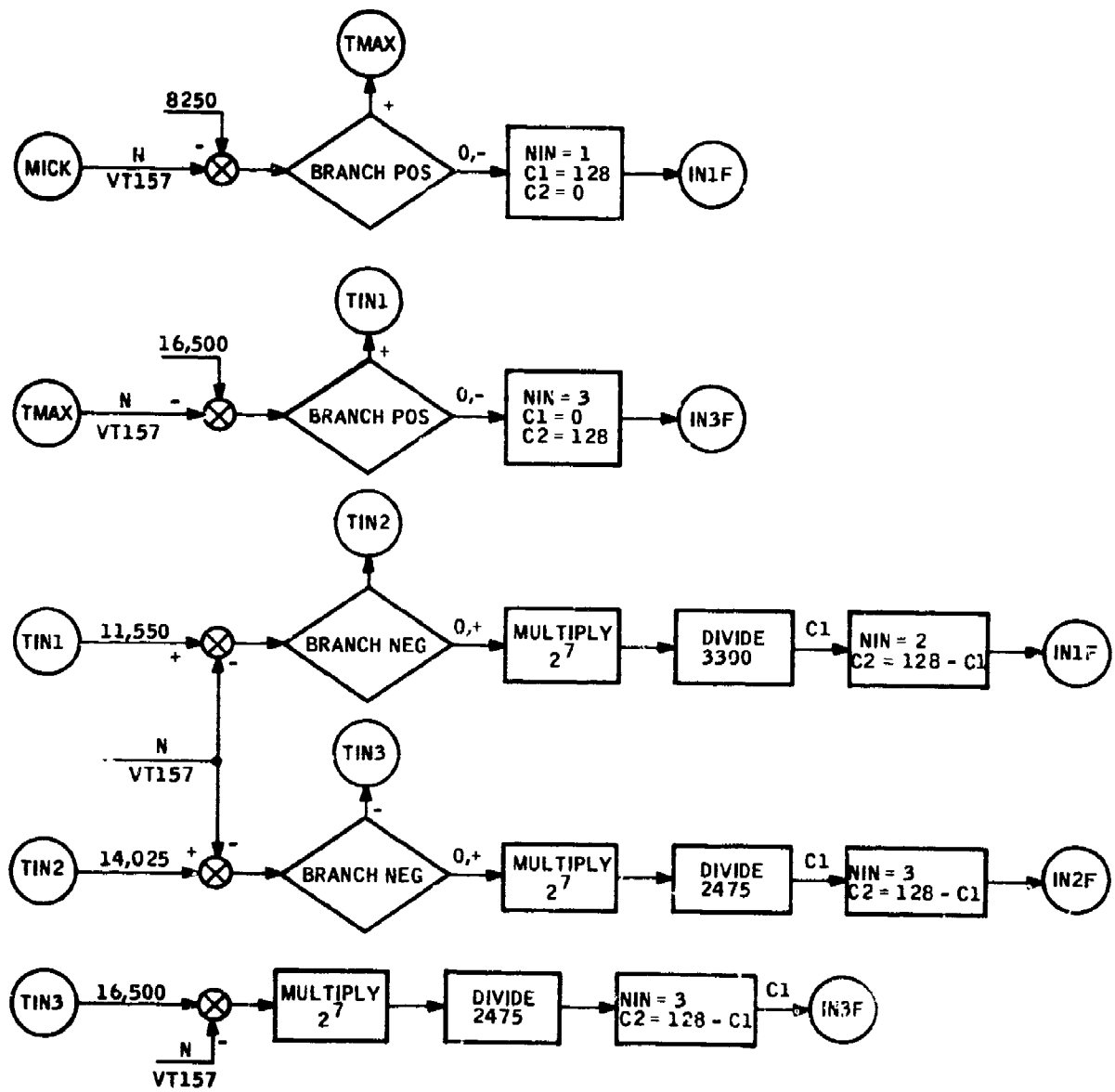


Figure B-14. Interval Determination

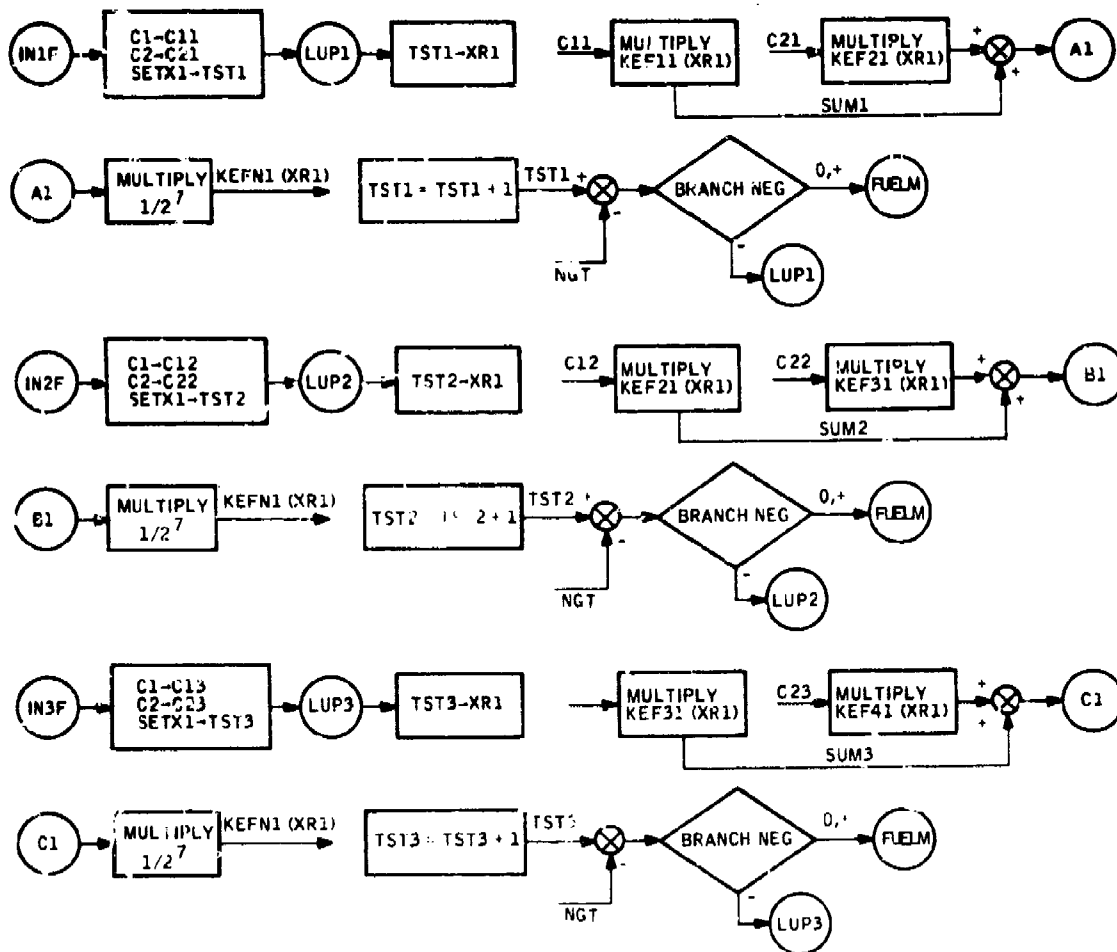


Figure B-15. Interpolation Logic

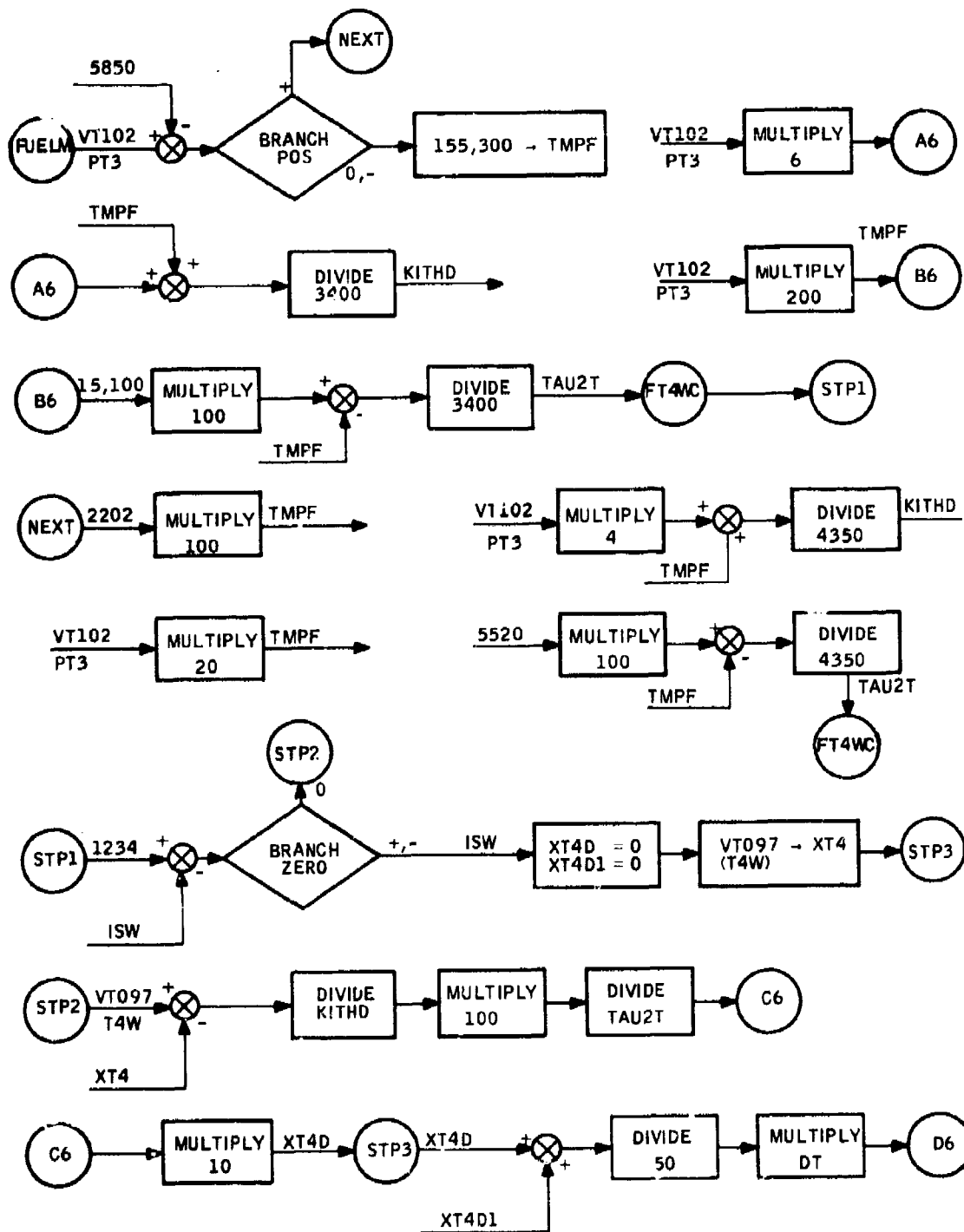


Figure B-16. Filter Logic for T4 Whistle Speed and Temperature Controller

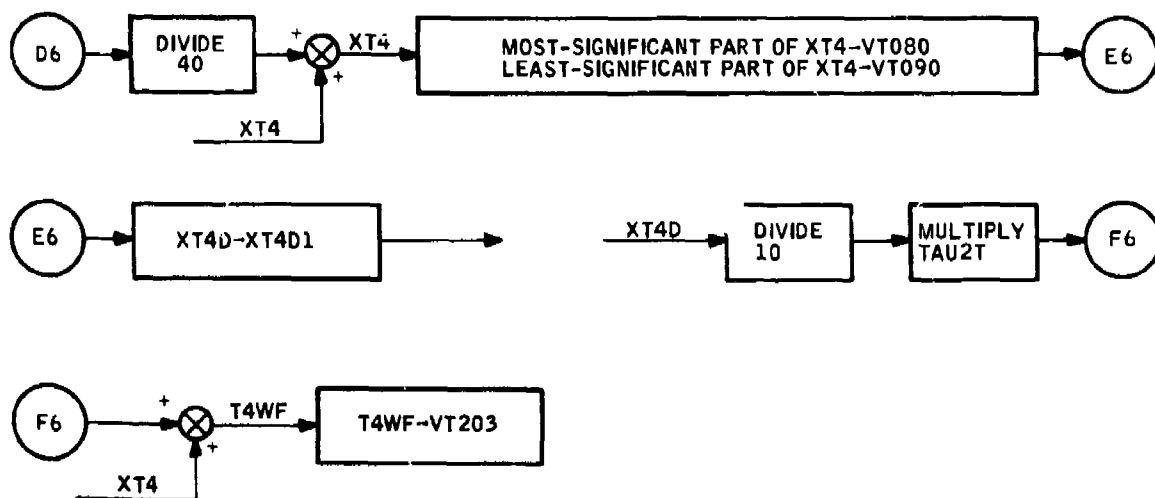


Figure B-16. Filter Logic for T4 Whistle Speed and Temperature Controller (Concluded)

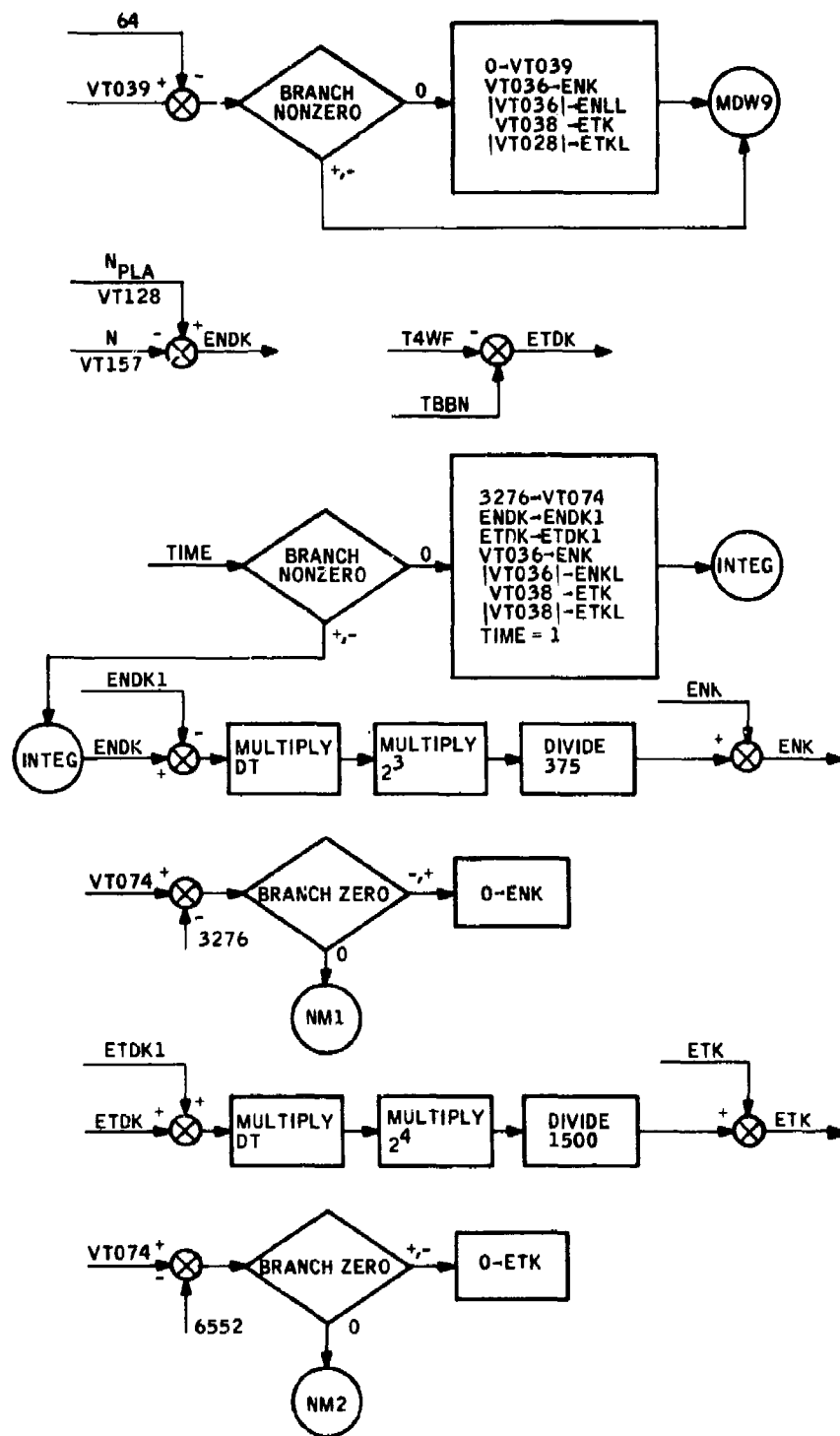


Figure B-17. Integral Speed and Integral Temperature

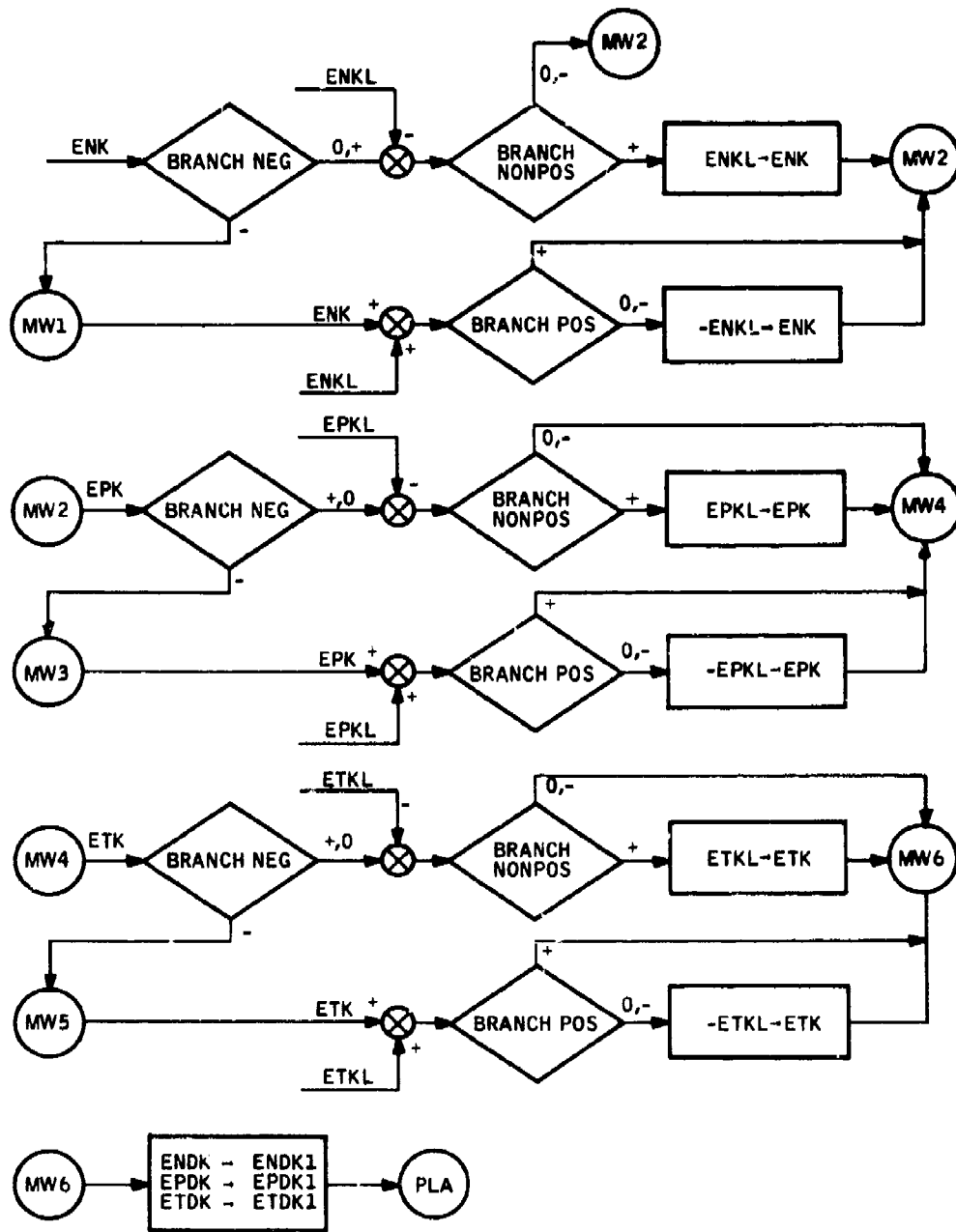


Figure B-18. Limiting Logic for Integral Speed and Integral Temperature

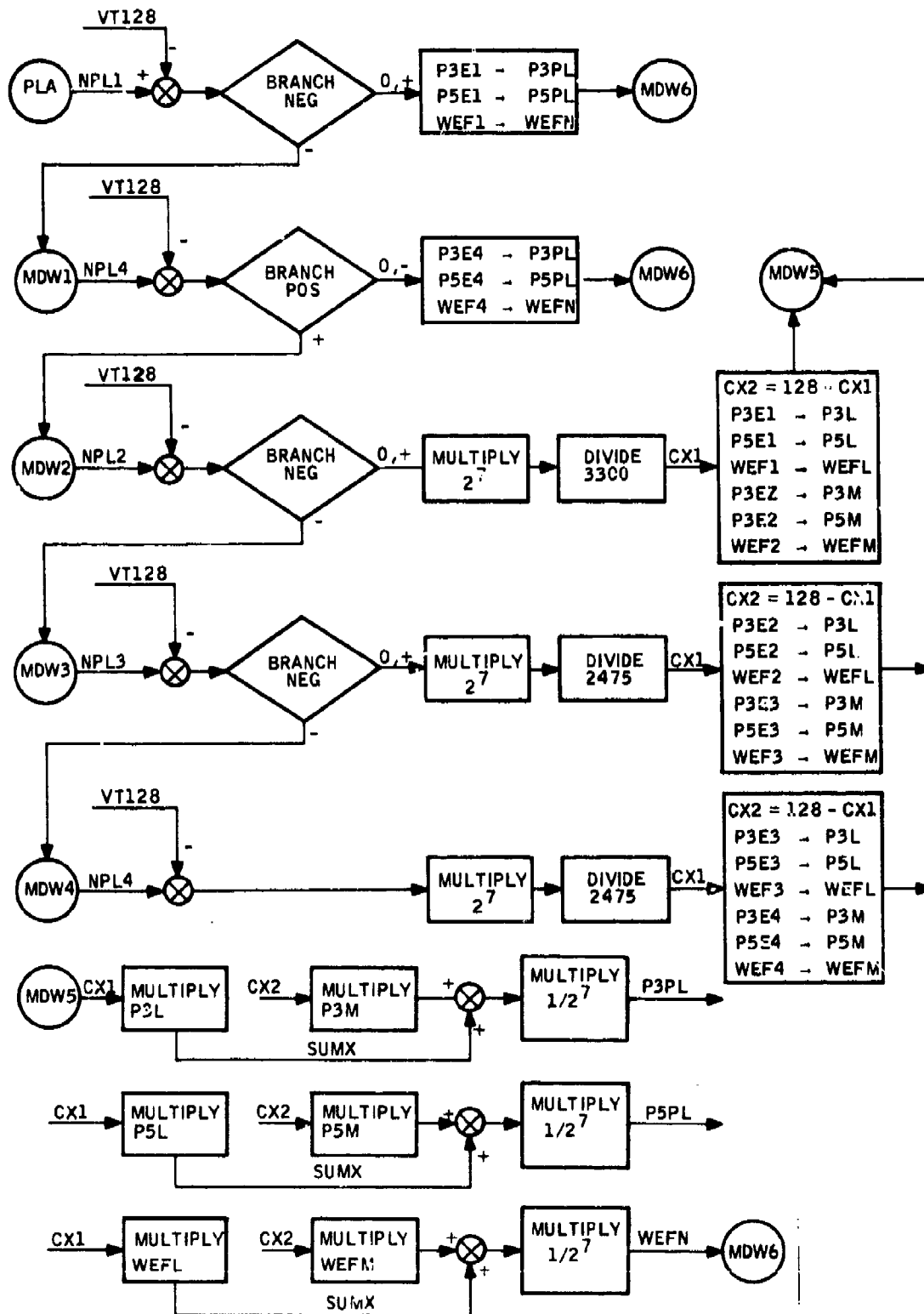


Figure B-19. Interpolation for PT3, PT5 and Fuel Request as a Function of Power Lever

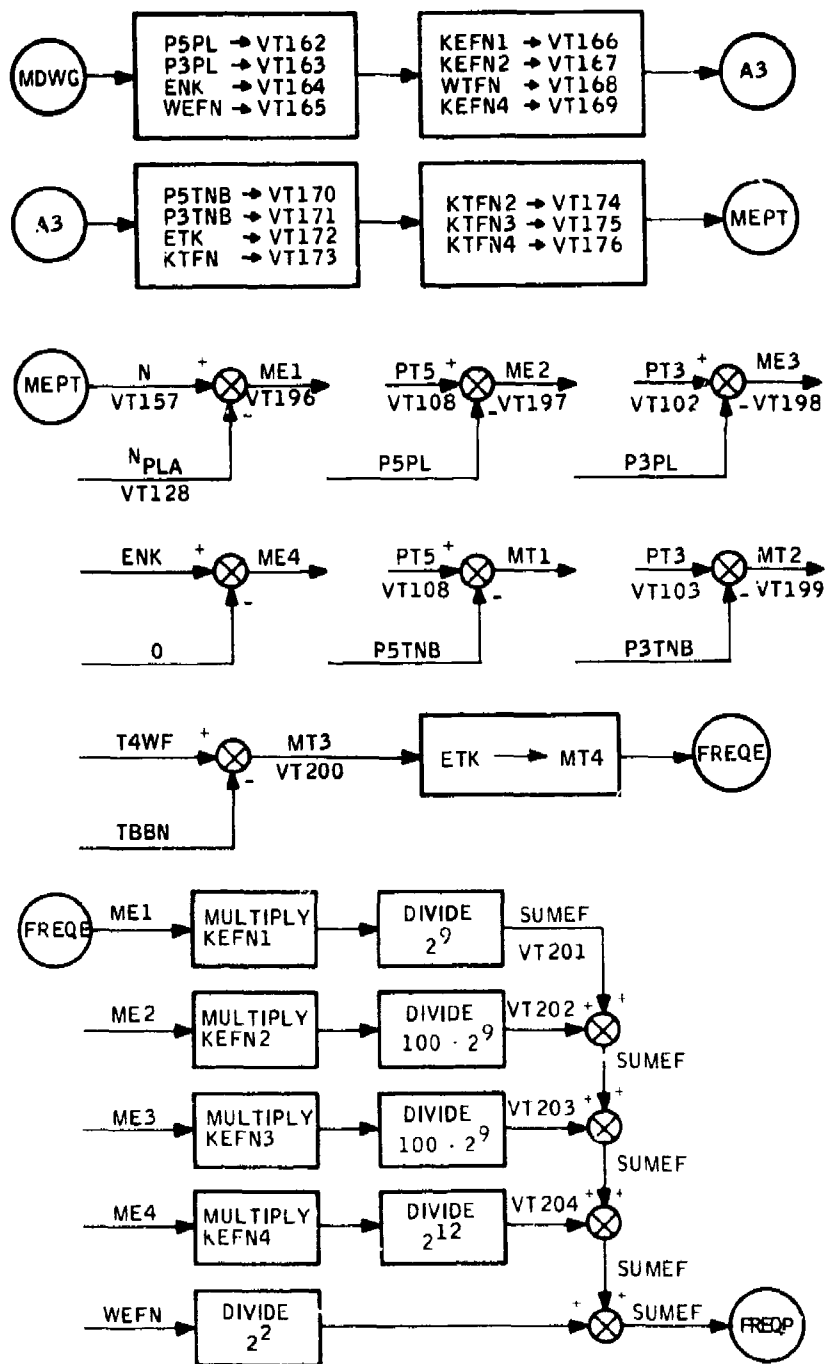


Figure B-20. Fuel Request Calculation

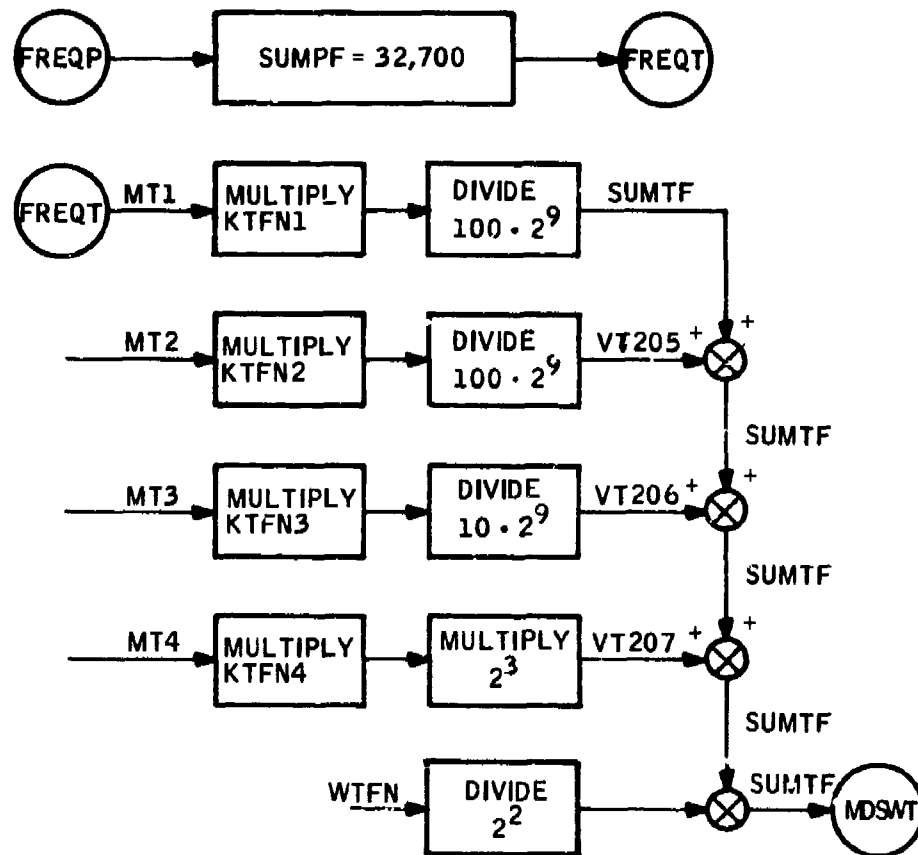


Figure B-20. Fuel Request Calculation
(Concluded)

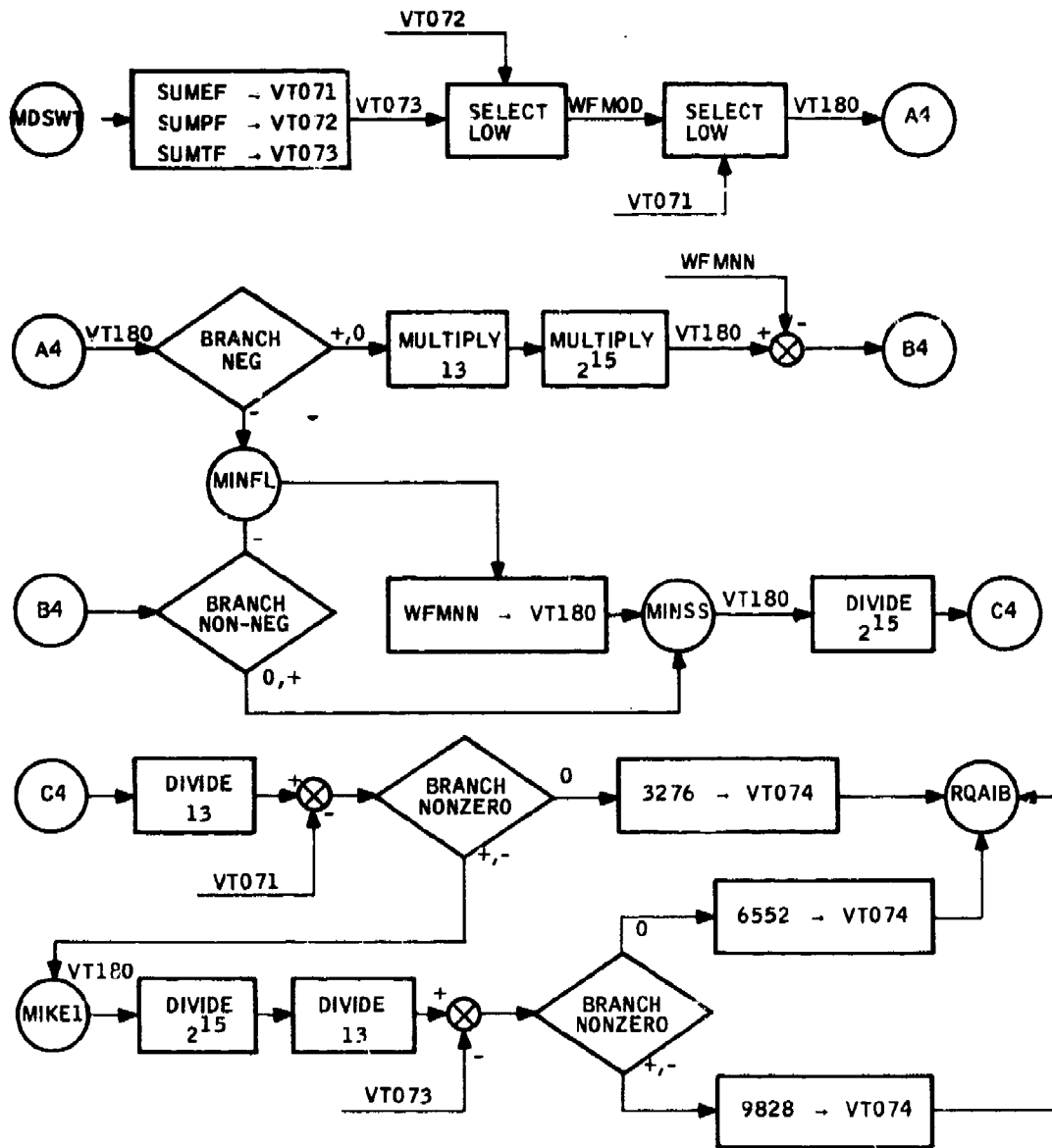


Figure B-21. Mode Select Logic

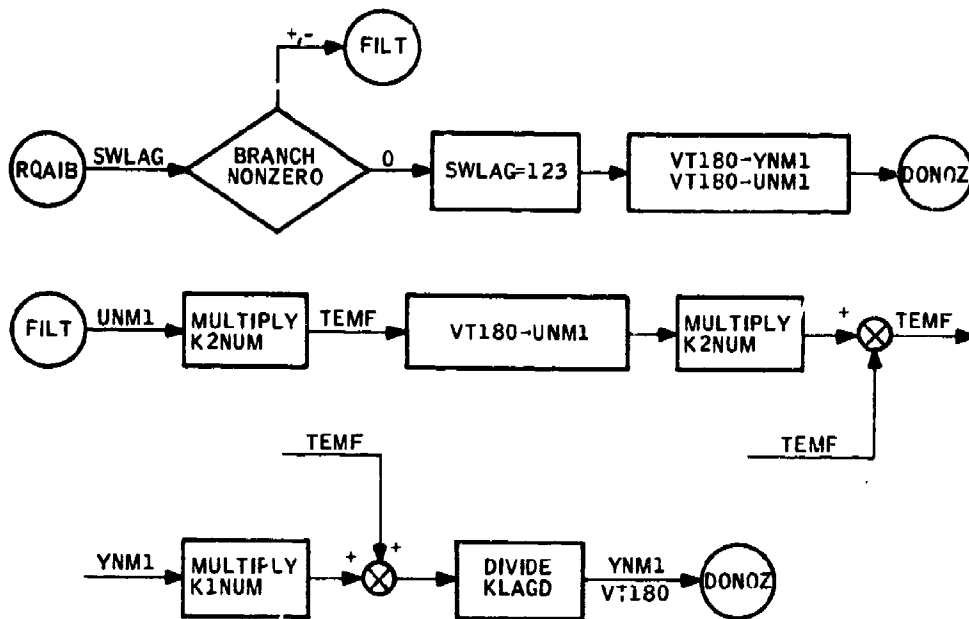


Figure B-22. Fuel Request Filter Logic

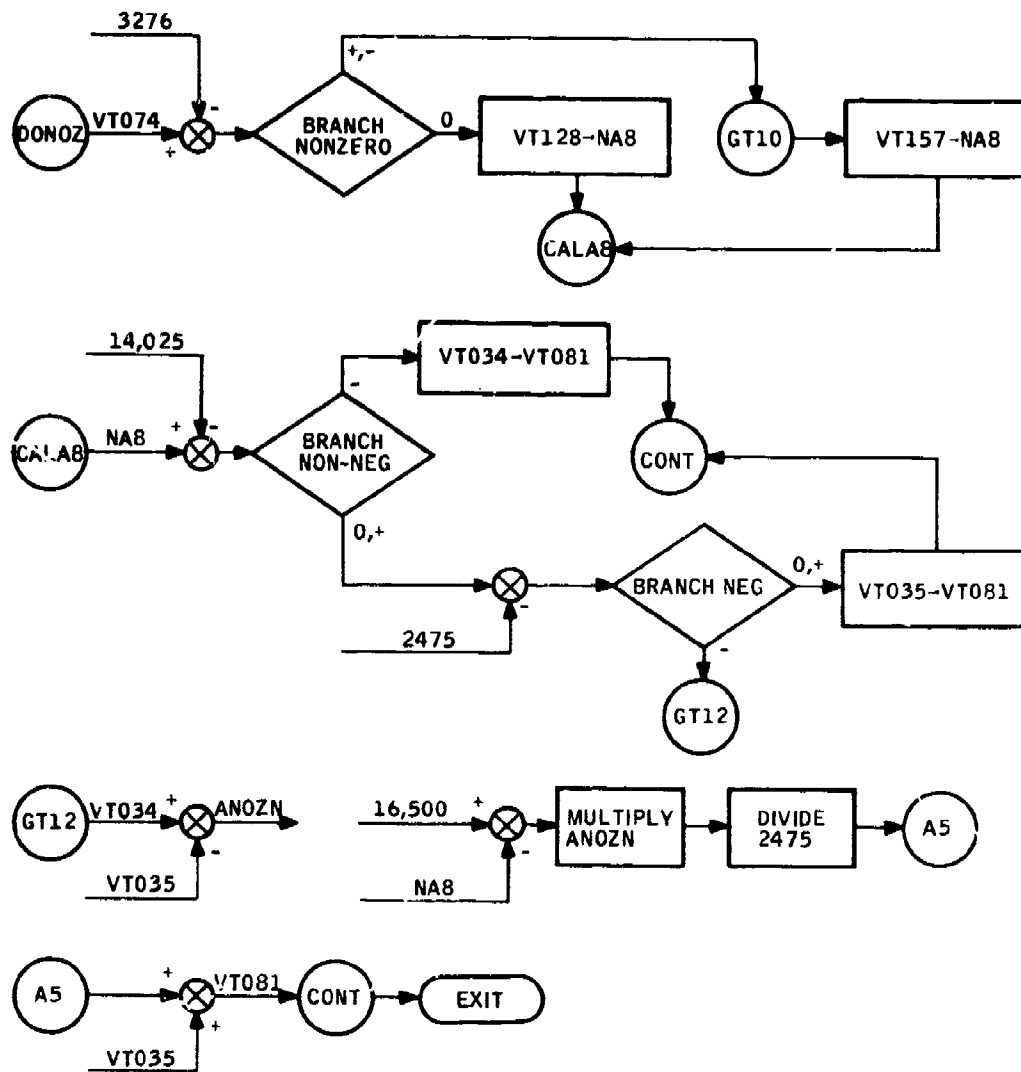


Figure B-23. Exhaust Nozzle Request Calculation

REFERENCES

- B-1. Arnett, Samuel E., "Turbine Engine Control Synthesis," AFAPL-TR-74-113, Air Force Aero Propulsion Laboratory, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio, December 1974.
- B-2. "IBM 1130/1800 Assembler-Language," GN34-0062, IBM Corporation, Systems Publications, Boca Raton, Florida, October 1971.

APPENDIX C
RATE MODELS FOR INTEGRAL CONTROL

In Sections III and IV of Volume I, a rate model (Reference 5) with integral control (Reference 6) is used in the linear quadratic synthesis (see Table 13 of Volume I).

The model is derived here. Spool speed notation is used although the results are applicable to pressure and temperature.

$$\dot{N} = aN + be + ce + \eta \quad (C-1)$$

$$\dot{e} = dN + fP \quad (C-2)$$

where

N = Model spool speed

e = Error

P = Model power lever

η = Disturbance

and a, b, c, d and f are constants to be determined to yield good response characteristics. Good response means that (1) N responds to P like a first-order plant, and (2) there is much integral control (sufficient to hold N against steady load disturbances η).

The model is derived in the following equations.

$$\frac{N}{P} = \frac{fb(s + c/b)}{s^2 - (a + bd)s - cd} \quad (C-3)$$

$$s = \frac{(a + bd) \pm (a + bd) \sqrt{1 + \frac{4cd}{(a + bd)^2}}}{2} \quad (C-4)$$

Choose $(a + bd)/2$ and λ (C-5, C-6)

Take

$$b = 1.0 \quad (C-7)$$

$$c = -b\lambda(a + bd)/2 \quad (C-8)$$

$$d = \frac{(a + bd)/2}{b\lambda} \quad (C-9)$$

$$a = 2\left(\frac{a + bd}{2}\right) - bd \quad (C-10)$$

Then

$$\frac{N}{P} = \frac{f\left\{s - \lambda\left(\frac{a + bd}{2}\right)\right\}}{\left\{s - \left(\frac{a + bd}{2}\right)\right\}^2} \quad (C-11)$$

The transfer function and roots for Equations (C-1) and (C-2) are given by Equations (C-3) and (C-4). If the second term in the radical is equal to -1, two identical roots are obtained. This choice is made.

The quantity $(a + bd)/2$ is chosen equal to the desired pole position.

The value of $\lambda = 0.75$ yields an excellent approximation to first-order response.

Coefficient data are presented below. Equations (C-7), (C-8), (C-9), and (C-10) yield a, b, c and d; λ is then selected by use of Equation (C-2) to yield the correct steady-state relationship between N and P.

Equation (C-11) presents the resulting transfer function. It is seen that λ positions the zero relative to the poles.

Coefficient data

<u>Root</u>	<u>a</u>	<u>b</u>	<u>c</u>	<u>d</u>
-2.0	-1.3333	+1.0	+1.5	-2.6667
-4.0	-2.6667	+1.0	+3.0	-5.3333
-10.0	-6.6667	+1.0	+7.5	-13.5333

APPENDIX D SIMPLE OPTIMIZATION

A derivation is presented of the algorithm used for control simplification (e. g. , paragraph 3, page 131 through paragraph 2, page 134 of Volume I for simple speed control). This derivation is a slight modification of the original (pp. 7 - 19 of Reference D-1). The source program is listed in Appendix I of Reference D-1.

The algorithm has more capability than was used on the Turbine Engine Control Synthesis contract. On this contract, the algorithm was used to find the optimal (simple) gains at each of 12 operating conditions (four each for speed, pressure, and temperature). The algorithm could have been used to determine (say) the best single value of P3 gain (over the 12 operating conditions), while the other gains (N, EN, PT5, etc.) were optimized at each of the 12 operating conditions. In this case, the P3 gain is "fixed" and the N, EN, etc. , gains are variable; hence, the Reference D-1 name for the algorithm: "Fixed-Plus-Variable Gain (FPVG)." For this turbine control synthesis, the fixed-gain feature was suppressed by working each operation condition separately.

BACKGROUND

The fixed-plus-variable (simple optimization) quadratic design procedure helps to solve a technical problem which confronts the major technical issues of engine control system design:

- High dimensionality
- Simplification
- Variability

High dimensionality is the reason the design procedure employs the theory of quadratics. This theory has been used before, on the B-52 LAMS (Ref. D-2), the C-5A LAMS (Ref. D-3), and the YF-12 LAMS (Ref. D-4). All of these programs involved design of flexure control, where the dynamic order of the models could be truncated to no less than 20 to 30 states.

Simplification arises because optimal quadratics, while promising solutions to dimensionality, yield control systems of substantial complexity. They demand feedbacks from all states to all controls. It is necessary to incorporate the constraints of measurement feasibility and control complexity into the fixed-plus-variable design procedure. These constraints were incorporated on the YF-12 LAMS program for single flight conditions and on the F-4 Lateral Axis program (Ref. D-5) for single and multiple flight conditions with fixed gains.

The third problem confronted is that of variability with respect to aerodynamic parameters, vehicle configuration, and mass distribution. Fixed gains were used on the F-4 Lateral-Axis program over an entire flight envelope, but the controller performance suffered because of it, even though the aircraft does not have flexure problems as do the B-52 and YF-12. On this contract (Ref. D-1), we use fixed-plus-variable gains to alleviate the problem of variability.

The formulation of the fixed-plus-variable quadratic design procedure, and the computational techniques used in the procedure, are discussed in this Appendix.

PROBLEM FORMULATION

The aircraft is represented at various points of the flight envelope and for various configurations and mass distributions by a collection of p frozen-point linear plants:

$$\frac{dx_i}{dt} = F_i x_i + G_{1i} u_i + G_{2i} \eta \quad (D-1)$$

$$r_i = H_i x_i + D_i u_i \quad i = 1, \dots, p \quad (D-2)$$

$$y_i = M_i x_i$$

Here x_i is the state vector for plant i which, for flexible aircraft, includes the following dynamics:

- Rigid-body states
- Actuator and servo states
- Significant flexure-mode states
- Low-frequency sensor states
- Model states (if state model-following is used)
- Kúsnér and Wagner states (associated with unsteady aerodynamics)
- Wind states (associated with atmospheric gust models).

The vector u_i represents control variables, η is a unity variance white noise vector, r_i is a vector of responses to be controlled (stresses and stress rates, accelerations at selected fuselage stations, model-following errors, control magnitudes and rates, etc.), and y_i is a vector of measurements (accelerometer outputs, gyro outputs, etc.). The matrices F_i (open-loop stability matrix), G_{1i} (control input matrix), G_{2i} (disturbance input matrix), H_i (response output matrix), D_i (control output matrix) and M_i (measurement matrix) are of appropriate order.

The above enumeration of components, vectors, and matrices is for an airplane for which Reference D-1 was concerned. Tables 40, 41, and 42 of Volume 1 list comparable items for turbine control synthesis.

We now look for a time-invariant controller of the form

$$u_i = K_i y_i \quad (D-3)$$

such that the following performance index is minimized:

$$J = \sum_{i=1}^p \alpha_i J_i \quad (D-4)$$

where

$$J_i = E \left\{ \text{Tr} \left[Q_i r_i r_i^T \right] \right\} \quad i = 1, 2, \dots, p \quad (D-5)$$

Here $E \{ \cdot \}$ denotes expectation, $\text{Tr} [\cdot]$ is the trace operator, and $(\cdot)^T$ denotes transpose of (\cdot) .

The Q_i are quadratic weights for flight condition i which are selected through quadratic equivalence or by means of a few trial design iterations (the art of the design procedure). The α_i are flight-condition weights selected as needed. A few suggestions about how to select them appears later in the discussion of the specific examples. The cost functional J is a generalization of the standard quadratic performance index of a single plant and represents a weighted performance over the flight envelope.

For turbine control synthesis, an operating condition corresponds to a flight condition in aircraft control synthesis. An operating condition for turbine synthesis is given by: (1) equilibrium speed control at (2) sea level static at (3) 70-percent power lever setting.

The gains matrices K_i are in general of the form

$$K_i = K^1 + K_i^5 \quad i = 1, \dots, 1 \quad (D-6)$$

where K^1 is a matrix of fixed gains constant over the flight envelope, and K_i^5 are the matrices of variable gains which vary over the flight envelope. For a fixed-gain design, the K_i^5 are empty.

The necessary conditions for the optimality of the K_i are obtained from the Maximum Principle (Ref. D-6). Let us rewrite the performance index as

$$J = \sum_{i=1}^p \alpha_i \text{Tr} \left\{ \left[H_i + D_i K_i M_i \right]^T Q_i \left[H_i + D_i K_i M_i \right] X_i \right\} \quad (\text{D-7})$$

where the covariance matrices

$$X_i = E \left[x_i x_i^T \right], \quad i = 1, \dots, p \quad (\text{D-8})$$

are solutions of the Lyapunov equations

$$0 = \left[F_i + G_{1i} K_i M_i \right] X_i + X_i \left[F_i + G_{1i} K_i M_i \right]^T, \quad i = 1, \dots, p \quad (\text{D-9})$$

Equations (D-7) and (D-9) are used to define a Hamiltonian:

$$\begin{aligned} H = \sum_{i=1}^p \left\{ \alpha_i \text{Tr} \left[H_i + D_i K_i M_i \right]^T Q_i \left[H_i + D_i K_i M_i \right] X_i \right. \\ \left. + \text{Tr} S_i^T \left[\left(F_i + G_{1i} K_i M_i \right) X_i + X_i \left(F_i + G_{1i} K_i M_i \right)^T \right. \right. \\ \left. \left. + G_{2i} G_{2i}^T \right] \right\} \quad (\text{D-10}) \end{aligned}$$

H is differentiated with respect to the covariance matrices X_i , the adjoint matrices S_i , and with respect to all the nonconstrained gains of the matrices K^1 and K_i^5 . The necessary conditions for optimality for this fixed-plus-variable-gain control are:

$$\bullet \quad \frac{\partial H}{\partial S_i} = \left(F_i + G_{1i} K_i M_i \right) X_i + X_i \left(F_i + G_{1i} K_i M_i \right)^T + G_{2i} G_{2i}^T = 0; \quad i = 1, \dots, p \quad (D-11)$$

$$\bullet \quad \frac{\partial H}{\partial X_i} = \left(F_i + G_{1i} K_i M_i \right)^T S_i + S_i \left(F_i + G_{1i} K_i M_i \right) + \alpha_i \left(H_i + D_i K_i M_i \right)^T Q_i \left(H_i + D_i K_i M_i \right) = 0; \quad (D-12)$$

$$i = 1, \dots, p$$

$$\bullet \quad \frac{\partial H}{\partial K_{lm}^1} = \left\{ \sum_{i=1}^p \left[\alpha_i D_i^T Q_i \left(H_i + D_i K_i M_i \right) + G_{1i}^T S_i \right] X_i M_i^T \right\}_{lm} = 0 \quad (D-13)$$

for all nonconstrained elements K_{lm}^1 of fixed matrix K^1 .
(In the above, $\{A\}_{lm}$ denotes the l th element of matrix A .)

$$\bullet \quad \frac{\partial H}{\partial K_{lmi}^5} = \left\{ \left[\alpha_i D_i^T Q_i \left(H_i + D_i K_i M_i \right) + G_{1i}^T S_i \right] X_i M_i^T \right\}_{lmi} = c; \quad (D-14)$$

$i = 1, \dots, p$, for all nonconstrained elements K_{lmi}^5 of the variable-gain matrices K_i^5 .

$$\bullet \quad K_i = K^1 + K_i^5; \quad i = 1, \dots, p \quad (D-15)$$

COMPUTATIONAL SOLUTION

The solutions of Equations (D-11) through (D-14) obviously do not exist in closed form. Thus, an iterative gradient search is necessary.

Equations (D-11) and (D-12) are solved quite readily for arbitrary gains matrices K_i through the use of computer algorithms that have been available for some time (such as explained in Ref. D-7). The solutions of these equations, the X_i and S_i , are used in the computation of the gradient components of Equations (D-13) and (D-14).

The development of the iterative gradient search algorithm to solve Equations (D-13) and (D-14) was the main effort of this contract.

A Newton-Raphson gradient technique was already developed and used for a fixed-gain design on the F-4 Lateral-Axis program (Ref. D-5); however, for the fixed-plus-variable quadratic design, the number of components in Equation (D-14) can be quite large, causing insurmountable computational difficulties with that technique, because it requires a matrix of second partial derivatives.

Computing a matrix of second partial derivatives requires solving a Lyapunov equation for each fixed gain and for each variable gain for each flight condition.

Other problems encountered with the Newton-Raphson gradient technique can be solved with a variable stepsize.

In view of the problems with this gradient technique, we decided to go with the straight gradient search, computing no second partial derivatives, and using a variable stepsize. We did, however, use some ideas of the predictor corrector scheme in implementing the gradient search. This resulted in what we call the incremental gradient.

INCREMENTAL GRADIENT

Let $K_i(\lambda)$ be the gain matrix for plant i defined as

$$K_i(\lambda) = K_i^1(\lambda) + K_i^5(\lambda) + \lambda K_i^2; 0 \leq \lambda \leq 1; i = 1, \dots, p \quad (D-16)$$

and let

$$K_i(1) = K_i^1(1) + K_i^5(1) + K_i^2 \quad (D-17)$$

be the optimal quadratic gains for plant i on the measurements y_i found through the solution of the Riccati Differential Equation, * and let

$$K_i(0) = K_i^1(0) + K_i^5(0) = K_i^1 + K_i^5 = K_i \quad (D-18)$$

be the final gains matrix for plant i . The expression λ is a scalar parameter; K_i^1 and K_i^5 are found by using the incremental gradient procedure which starts with initial gains $K_i^1(1)$ and $K_i^5(1)$; K_i^2 are simply the difference between the optimal gains $K_i(1)$ and initial gains $K_i^1(1) + K_i^5(1)$.

In terms of Equation (D-16), the necessary conditions for optimality of K_i^1 and K_i^5 are that

$$\left. \frac{\partial J[K_i(\lambda)]}{\partial K_i^1} \right|_{\lambda=0} = 0 \quad (D-19)$$

and

$$\left. \frac{\partial J[K_i(\lambda)]}{\partial K_i^5} \right|_{\lambda=0} = 0 \quad (D-20)$$

*This requires that the M_i be square and nonsingular. They can be made so by adding direct measurements of states not necessarily measurable.

In fact, if we start with $\lambda = 1$ and satisfy Equations (D-19) and (D-20) for all λ in $[0, 1]$, Equations (D-19) and (D-20) are certainly true for $\lambda = 0$. At the same time, we are ensuring with high probability that a global minimum of $J(K^1 + K_i^5)$ is reached because we are starting in the "deepest valley of J " and forcing λ to zero along the trajectory $\{K^1(\lambda), K_i^5(\lambda), K_i^2; 1 \geq \lambda \geq 0\}$. Since we are then "on the walls of the deepest valley," along with the knowledge of $J[K_i^1(1)]$ and $J[K^1(0) + K_i^5(0)]$, we can terminate the search for the global minimum.

Stein and Henke (Ref. D-5) used the Implicit Function Theorem which defined K^1 (in their case it was fixed gains only) from the solution of the differential equation

$$\frac{dK^1(\lambda)}{d\lambda} = - \left[\frac{\partial^2 J(K^1 + \lambda K^2)}{\partial K^1 \partial K^{1T}} \right]^{-1} \frac{\partial^2 J(K^1 + \lambda K^2)}{\partial K^1 \partial \lambda} \quad (D-21)^*$$

by starting with the known terminal condition $K = K^1 + K^2$ for $\lambda = 1$ and integrating it backward toward $\lambda = 0$. The method of numerical integration used was that which used an Adams-Moulton Predictor and a Newton-Raphson Corrector to step λ from 1 to 0.

The main problem with this procedure is that the evaluation of the second partial derivatives is very costly, and gets out of hand when the variable gains are included. Another problem is that the predictor or corrector steps are sometimes too big and can cause one plant or another to go unstable. The incremental gradient procedure alleviates this problem by approximating the second partial derivatives (discussed later), using a simple linear predictor, and a variable step size on the corrector. More than one gradient direction per prediction step and the variable gradient step size more than make up for the approximation and prediction simplification.

* K , K^1 and K^2 must be stacked up as column vectors for this equation to make sense. This is assumed.

The incremental gradient procedure is summarized in Figure D-1 for a single-plant problem. Here, λ is stepped to zero in five steps. There are only two gains, K^1 and K^2 . We wish to eliminate K^2 . However, if we eliminate K^2 without changing K^1 , the system is unstable, and a gradient direction cannot be found. (This frequently happens in real-world problems.)

The first prediction step is in the K^2 direction only. (In practice, this never presented a problem.) A correction is made with a Newton-Raphson gradient search using approximate second partial derivatives and a variable step size determined from a parabolic fit. The subsequent predictions are extrapolations from the initial point through the last correction points. The process continues for each step in λ .

The predicted gains are

$$K_p^1(\lambda_{j+1}) = K_c^1(\lambda_j) + [K_c^1(\lambda_j) - K_c^1(\lambda_{j-1})] \quad (D-22)$$

and

$$K_{ip}^5(\lambda_{j+1}) = K_{ic}^5(\lambda_j) + [K_{ic}^5(\lambda_j) - K_{ic}^5(\lambda_{j-1})]; \quad (D-23)$$

$$i = 1, \dots, p$$

where λ_j is the value of λ on the j^{th} predictor step, and the initial prediction is zero. The "c" and "p" denote "corrected" and "predicted." The predicted gains are the initial gains for the gradient search. The corrected gains result from the gradient search.

For the variable step size for the gradient search, the performance index J is computed for three step sizes -- 0, ϵ_1 , and $2\epsilon_1$ -- and fit to a parabola

$$J(\epsilon) = J(0) + A\epsilon + B\epsilon^2 \quad (D-24)$$

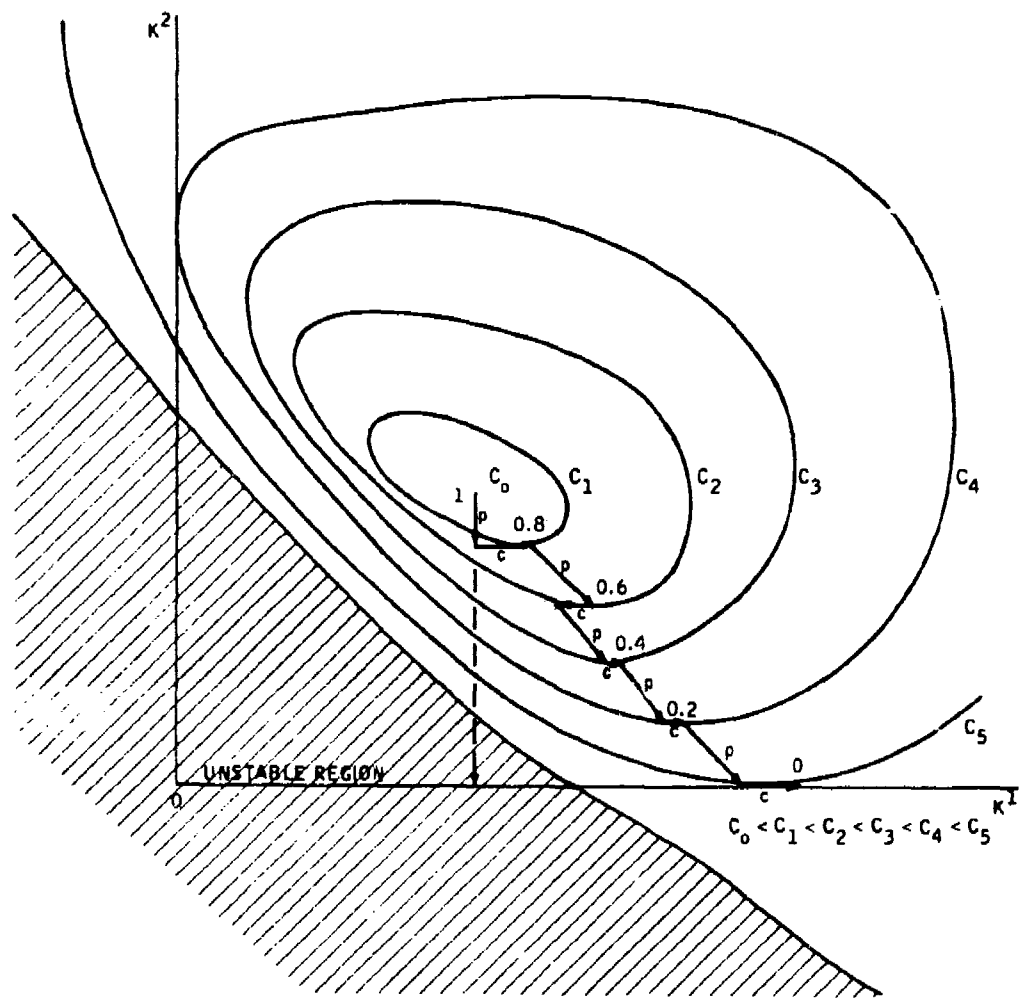


Figure D-1. Incremental Gradient Path

A minimum at

$$\epsilon = -\frac{A}{2B} \quad (D-25)$$

is computed, where A and B are a function of the performances $J(0)$, $J(\epsilon_1)$, $J(2\epsilon_1)$ and ϵ_1 . The logic for halving and doubling the step size for computing these performances is discussed in Appendix I of Reference D-1.

THE GRADIENT TRANSFORMATION

An aircraft example presented a situation that exists on many minimization problems. That is, the performance contours are extremely ellipsoidal. This causes a straight gradient search to converge very slowly or not even noticeably. The ideal situation is to have the performance contours be spheroidal. Then the gradient direction would be right to the center of the spheroid. This is shown in Figure D-2.

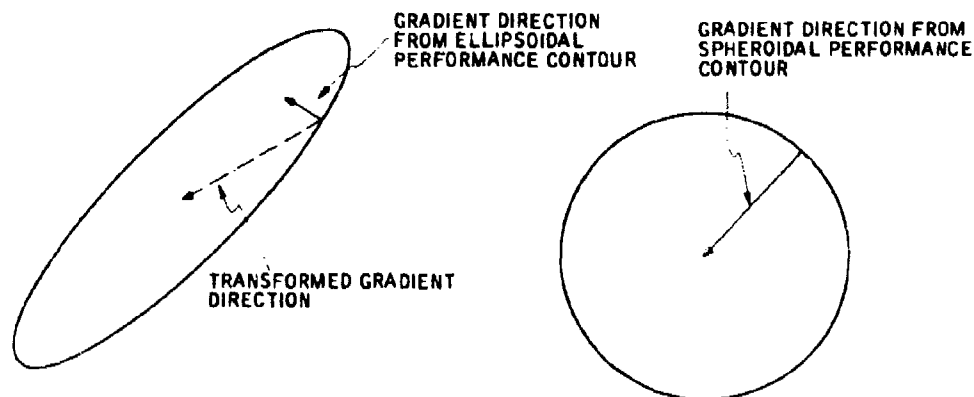


Figure D-2. Comparison of Gradient Directions for Two Performance Contours

If a performance contour is extremely ellipsoidal, the effect of a spheroidal contour can be realized by transforming the gradient vector. This effect is also shown in Figure D-2.

For a problem with a second-order minimum, the ideal transformation is that provided by the Newton-Raphson gradient direction, that is, the inverse of the matrix of second partial derivatives. However, as stated before, the evaluation of the second partial derivatives is very costly. Thus, an approximation was used that works extremely well.

An element in the matrix of second partial derivatives may be written as (assuming for the moment only a fixed-gains matrix for a single flight condition stacked up as vectors):

$$\begin{aligned} \frac{\partial^2 J(K)}{\partial K_{ij}^1 \partial K_{lm}^1} &= 2R_{il} M_j X M_m^T + 2 \sum_{k=1}^n \left[(K^T R)_{ki} + (SG_1)_{ki} \right] M_j \left(\frac{\partial X}{\partial K_{lm}^1} \right)_k \\ &+ 2 \sum_{k=1}^n \left[(K^T R)_{kl} + (SG_1)_{kl} \right] M_m \left(\frac{\partial X}{\partial K_{ij}^1} \right)_k \end{aligned} \quad (D-26)$$

where X is the state covariance matrix, S is the adjoint matrix and

$$R = D^T Q D \quad (D-27)$$

M_k denotes row k of M , and $(\partial X / \partial K_{ij}^1)_k$ denotes the k^{th} column of the partial derivative of X with respect to K_{ij}^1 .

The approximation neglects the last two terms of Equation (D-26) because the partial derivatives $(\partial X / \partial K_{ij}^1)$ require a Lyapunov equation solution for each element in K^1 . This approximation is not a bad one, for the two terms take

care of any warping due to the change in X with respect to K_{ij}^{-1} , and additional gradient directions will take care of this warping.

To extend this transformation to the fixed-plus-variable design, it must include the cross-correlation between measurements with fixed gains and measurements with variable gains. To do this, the gradient vectors for each of r controls must be stacked up end to end to form a vector

$$\frac{\partial J}{\partial K} = \begin{bmatrix} \frac{\partial J^T}{\partial K_1^{-1}} \\ \vdots \\ \frac{\partial J^T}{\partial K_r^{-1}} \\ \vdots \\ \frac{\partial J_1^T}{\partial K_{11}^5} \\ \vdots \\ \frac{\partial J_1^T}{\partial K_{r1}^5} \\ \vdots \\ \frac{\partial J_p^T}{\partial K_{1p}^5} \\ \vdots \\ \frac{\partial J_p^T}{\partial K_{rp}^5} \end{bmatrix} \quad (D-28)$$

where K_j^1 is the j^{th} row of the fixed-gain matrix, K_{ij}^5 is the j^{th} row of the variable-gain matrix for flight condition i , J is the total cost, and J_i is the cost for flight condition i .

The vector $(\partial J / \partial K)$ has $n_f + n_v \cdot p$ elements, where n_f is the number of fixed gains, n_v is the number of variable gains, and p is the number of flight conditions.

The transformation of the gradient for the fixed-plus-variable-gain design is then the inverse of the matrix in Figure D-3. That is

$$\frac{\partial J_T}{\partial K} = \Phi^{-1} \frac{\partial J}{\partial K} \quad (D-29)$$

where

$$\phi_{ijk\ell m} = \alpha_i d_{ji}^T Q_i d_{ki} M_i^{\ell j} X_i M_i^{mkT} \quad (D-30)$$

In Equation (D-30), α_i is the flight condition weight, d_{ji} is column j of D_i , Q_i is the quadratic weighting matrix for flight condition i , and $M_i^{\ell j}$ is the measurement matrix for control j and flight condition i for the fixed gains if $\ell = 1$, or for the variable gains if $\ell = 5$. X_i is the covariance matrix for flight condition i .

Figure D-4 summarizes the incremental gradient scheme using the transformed gradient.

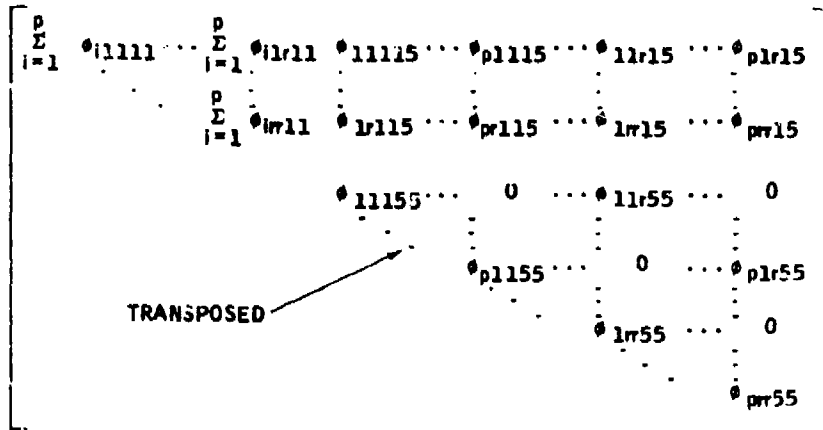


Figure D-3. Transformation Matrix S

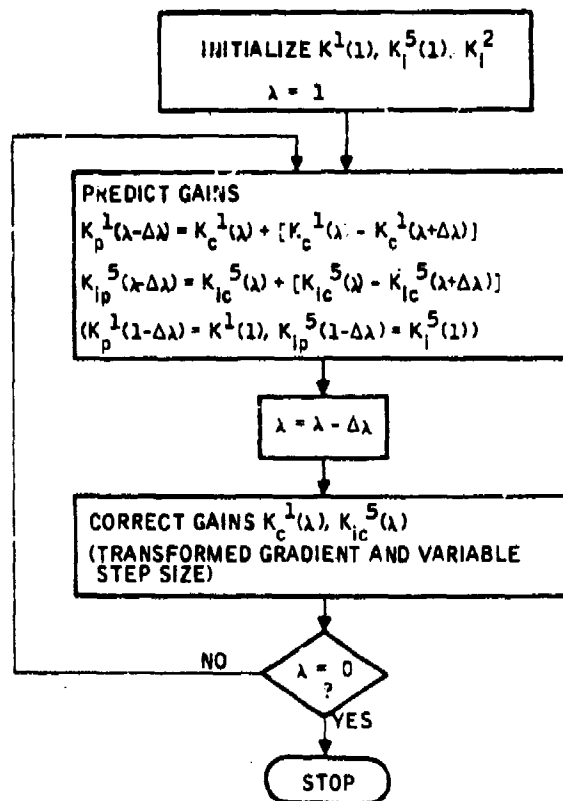


Figure D-4. Incremental Gradient Flow Diagram

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