

AD/A-002 855

A STRUCTURAL WEIGHT ESTIMATION PROGRAM (SWEEP) FOR AIRCRAFT. VOLUME III - AIRLOADS ESTIMATION MODULE. APPENDIX A: MODULE FLOW CHARTS AND FORTRAN LISTS. APPENDIX B: SAMPLE OUTPUT

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Rockwell International Corporation

Prepared for:

Aeronautical Systems Division

June 1974

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REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER ASD/XR 74-10	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) A Structural Weight Estimation Program (SWEEP) for Aircraft Volume III - Airloads Estimation Module		5. TYPE OF REPORT & PERIOD COVERED
7. AUTHOR(s) P. Wildermuth, G. Rothhammer, R. Hiyama		6. PERFORMING ORG. REPORT NUMBER
9. PERFORMING ORGANIZATION NAME AND ADDRESS Rockwell International Corp, L.A. Aircraft Div Los Angeles International Airport Los Angeles, California 90009		8. CONTRACT OR GRANT NUMBER(s) F33615-71-C-1922
11. CONTROLLING OFFICE NAME AND ADDRESS Deputy for Development Planning Air Force Systems Command Wright-Patterson Air Force Base, Ohio		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS FX2826-71-01876/C093
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		12. REPORT DATE June 1974
		13. NUMBER OF PAGES 464
		15. SECURITY CLASS. (of this report) Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) Approved for public release; distribution unlimited.		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Reproduced by NATIONAL TECHNICAL INFORMATION SERVICE U.S. Department of Commerce Springfield, VA 22151		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) weight estimation, structural weights, integrated computer programs, preliminary weight estimation, first-order weight estimations, aircraft structure weights, aircraft structural weight optimization, flutter optimization program, structural synthesis		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Three computer programs were written with the objective of predicting the structural weight of aircraft through analytical methods. The first program, the structural weight estimation program (SWEEP), is a completely integrated program including routines for airloads, loads spectra, skin tem- peratures, material properties, flutter stiffness requirements, fatigue life, structural sizing, and for weight estimation of each of the major aircraft structural components. The program produces first-order weight estimates		

and indicates trends when parameters are varied. Fighters, bombers, and cargo aircraft can be analyzed by the program. The program operates within 100,000 octal units on the Control Data Corporation 6600 computer. Two stand-alone programs operating within 100,000 octal units were also developed to provide optional data sources for SWEEP. These include (1) the flexible airloads program to assess the effects of flexibility on lifting surface airloads, and (2) the flutter optimization program to optimize the stiffness distribution required for lifting surface flutter prevention.

The final report is composed of 11 volumes. This volume (Volume III) contains the methodology, program description, and user's information for the airload module of SWEEP.

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APPENDIX A

MODULE FLOW CHARTS AND FORTRAN LISTS

FLOW CHART USAGE

The automatically generated computer program flow charts (AUTOFLOW) presented in this document include a table of contents, flow charts, and FORTRAN lists of all routines in the module. The 80-column card lists are sequenced and grouped by routine.

Because the AUTOFLOW system used is IBM-oriented, the functions of the BUFFERIN and BUFFEROUT statements are not recognized, but these statements appear in proper order in note boxes. Also, the PROGRAM name does not appear on the main program, and library routines READMS and WRITMS are listed as undefined external references.

CROSS-REFERENCE LIST

The AUTOFLOW table of contents which precedes the flow charts and FORTRAN lists serves to cross reference the latter two. This table lists the following from left to right:

- The card identification from columns 73 through 80 of this card, or card sequence number. When sequence number is used in place of card identification, it is enclosed in parentheses.
- The page and box number where this card is displayed in a flow chart.
- The FORTRAN statement number from columns 1 through 5 of this card.
- The card identification(s) or sequence number(s) of the card(s) referring to this card (repeated as required).
- The pages and box numbers where the cards referring to this card are displayed in a flow chart (repeated as required).

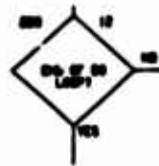
FLOW CHARTS

The flow charts produced by AUTOFLOW use USASI conventional symbols. Since the flow charts are mechanically drawn from the program source deck, there are no omissions or vague generalizations about the processing within the boxes.

Every box on each page is uniquely numbered and may be referred to from elsewhere in the program. The source of a reference to a box will be indicated by showing the page and box number. If the number is followed by an asterisk, there are multiple references to this point, and the others may be found by using the cross-reference list.



The most-often-used symbol is the decision box. Like all boxes, its box number is above and to the right of the box. Its FORTRAN statement number is above and to the left of the box. The decision choices for the paths are printed.



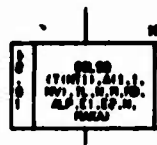
The unconditional transfer connector has its page number destination printed above or to the left of the box number destination within the connector. If there is a FORTRAN statement number at the destination, it is printed below the connector.



The exit box example shows a connector from page 9, box 15.



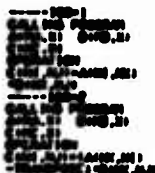
The subroutine call box includes the calling sequence. The page and box numbers of the flow chart of the called subroutine are shown on the left-hand side of the box. The page number is above the box number.



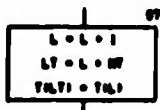
The note box encloses comments of a functional nature,



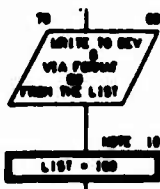
as differentiated from the 21 column comments, which are left justified without a box, that show the comment cards included in the FORTRAN deck.



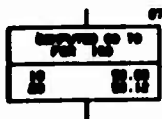
The process box is used to enclose FORTRAN arithmetic statements.



Input and output are shown as communicating with a device. The list used follows, if appropriate:



The computed $G\emptyset T\emptyset$ becomes a branch table showing the page and box number of each of the ordered branches.



The column connectors and initial connectors are the only boxes without external box numbers. The function of the initial connector is always clear,

but the label given is the symbol in the next FORTRAN card, which is often blank.



The column connector identifies the page and box number to which it connects.



MODULE TABLE OF CONTENTS AND REFERENCES

CARD NO	PAGE/FORM	NAME	REFERENCE	SOURCE	SEQUENCE NO.	AND PAGE/FORM
(000000)	0 04 00		(000100)	0 02		
(000001)	0 05 01		(000201)	0 03	(000210)	7 01
(000010)	0 07 02					
(000012)	0 09 03		(000200)	0 05		
(000014)	7 01 04		(000170)	0 13		
(000019)	7 02 05		(000170)	0 13	(000170)	0 13
(000020)	7 03 06		(000170)	0 13	(000170)	0 13
(000025)	7 04 07		(000100)	0 14	(000100)	0 15
(000026)	7 05 08		(000011)	7 02	(000100)	0 16
(000027)	7 06 09				(000100)	0 17
(000028)	7 07 10				(000100)	0 18
(000029)	0 01 00		(000100)	0 10	(000024)	7 03
(000037)	0 02 02		(000020)	7 00	(000020)	7 07
(000040)	0 04 03					
(000041)	0 05 04					
(000045)	0 06 05		(000020)	0 03		
(000046)	0 07 07					
(000047)	0 08 08		(000040)	0 04		
(000048)	0 09 09		(000045)	0 05		
(000049)	0 10 10		(000051)	0 11		
(000050)	0 11 100		(000051)	0 02		
(000052)	0 12 101					
(000053)	0 13 102					
(000054)	0 01 103		(000052)	0 12		
(000055)	0 02 104		(000042)	0 06	(000047)	0 07
(000056)	0 04 105		(000051)	0 05	(000044)	0 08
(000058)	0 05 106		(000050)	0 03	(000050)	0 10
(000059)	0 06 107				(000059)	0 13
(000060)	0 07 108		(000011)	0 07	(000013)	0 09
(000070)	0 10 111				(000020)	0 03
(000072)	0 12 113		(000000)	0 00	(000000)	0 04
(000076)	0 15 114		(000077)	0 13	(000077)	0 13
(000077)	0 16 115		(000077)	0 13	(000077)	0 13
(000078)	0 17 116		(000077)	0 13	(000077)	0 13
(000081)	0 18 117		(000077)	0 13	(000077)	0 13
(000083)	0 19 118		(000077)	0 13	(000077)	0 13
(000085)	0 20 119		(000077)	0 13	(000077)	0 13
(000087)	0 21 120		(000077)	0 13	(000077)	0 13
(000088)	10 01 121		(000077)	0 13	(000077)	0 13
(000089)	10 02 122		(000077)	0 13	(000077)	0 13
(000090)	10 03 123		(000077)	0 13	(000077)	0 13
(000091)	10 04 124		(000077)	0 13	(000077)	0 13
(000092)	10 05 125		(000077)	0 13	(000077)	0 13
(000093)	10 06 126		(000077)	0 13	(000077)	0 13
(000094)	10 07 0		(000077)	0 13	(000077)	0 13
(000095)	10 08 7		(000077)	0 13	(000077)	0 13
(000096)	10 09 8		(000077)	0 13	(000077)	0 13
(000097)	10 10 9		(000077)	0 13	(000077)	0 13
(000098)	10 11 10		(000077)	0 13	(000077)	0 13
(000099)	10 12 11		(000077)	0 13	(000077)	0 13
(000100)	10 13 12		(000077)	0 13	(000077)	0 13
(000101)	10 14 13		(000077)	0 13	(000077)	0 13
(000102)	10 15 14		(000077)	0 13	(000077)	0 13
(000103)	10 16 15		(000077)	0 13	(000077)	0 13
(000104)	10 17 16		(000077)	0 13	(000077)	0 13
(000105)	10 18 17		(000077)	0 13	(000077)	0 13
(000106)	10 19 18		(000077)	0 13	(000077)	0 13
(000107)	10 20 19		(000077)	0 13	(000077)	0 13
(000108)	10 21 20		(000077)	0 13	(000077)	0 13
(000109)	10 22 21		(000077)	0 13	(000077)	0 13
(000110)	10 23 22		(000077)	0 13	(000077)	0 13
(000111)	10 24 23		(000077)	0 13	(000077)	0 13
(000112)	10 25 24		(000077)	0 13	(000077)	0 13
(000113)	10 26 25		(000077)	0 13	(000077)	0 13
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(000132)	10 45 44		(000077)	0 13	(000077)	0 13
(000133)	10 46 45		(000077)	0 13	(000077)	0 13
(000134)	10 47 46		(000077)	0 13	(000077)	0 13
(000135)	10 48 47		(000077)	0 13	(000077)	0 13
(000136)	10 49 48		(000077)	0 13	(000077)	0 13
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(000169)	10 82 81		(000077)	0 13	(000077)	0 13
(000170)	10 83 82		(000077)	0 13	(000077)	0 13
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(000172)	10 85 84		(000077)	0 13	(000077)	0 13
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CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE ATYDS(H, RCH, PH, AN)

(000400) 10 01 ATYDS	(000500) 10 01-2	(000500) 10 10-2
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(000546) 10 06 120	(000530) 10 09	

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE SHLOS

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(000586) 10 04 12				
(000588) 10 05 13				
(000593) 10 06 00				
(000594) 10 06 01	(000587) 10 07			
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(000602) 10 13 17	(000595) 10 11			
(000603) 10 14 18	(000601) 10 12			
(000605) 10 15 19	(000605) 10 15	(000605) 10 15	(000605) 10 15	(000605) 10 15
	(000613) 10 16			
(000608) 10 17 20	(000606) 10 16	(000605) 10 15		
(000611) 10 18 21	(000608) 10 18			
(000614) 20 01 22	(000607) 10 16	(000610) 10 17		
(000623) 20 02 23				
(000627) 20 03 24	(000602) 20 05			
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(000659) 20 27 26	(000653) 20 26	(000653) 20 26	(000653) 20 26	(000653) 20 26
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CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - FUNCTION CODING (H, S, I, V, N, M)

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(000700)	00.10	200				
(000700)	00.13	310	(000707)	00.11		
(000707)	00.15	310	(000700)	00.10		
(000700)	00.16	300	(000707)	00.10		
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(000010)	07.07	000	(000707)	00.00	(000700)	00.00
					(000700)	00.01
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CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE PATHS

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100070	31 05	100070	31 05					100007	31 02
100073	31 07 10								
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100002	31 12	100070	31 21						
100000	31 15 21								
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(001008)	35 33	76				
(001009)	35 31	75	(001007)	35 32		
(001010)	35 32	77	(001008)	35 33		
(001011)	35 33	78	(001009)	35 31		
(001012)	35 37	90-92	(001017)	35 36	(001008)	35 33
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(001077)	35 14	88				
(001078)	35 15	112	(001077)	35 14		
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CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - FUNCTION FORMS (R,V,H1,Z1,H1,H2,H3)

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CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE FUNCT

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(001178)	44 05	10				
(001179)	44 06	700	(001172)	44 01		
(001179)	44 07	701				
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(001186)	44 13	825	(001185)	44 11		
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CARD NO PAGE/FORM NAME REFERENCES (SOURCE SEQUENCE NO. AND PAGE/FORM)

(001200) 45 00 020 (001205) 45 07
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CHART TITLE - NEW-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE PARLODS

(001250) 45 01 PARLODS (000351) 11 00-5 (000370) 12 01-2
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CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE USPAN

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300

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REFERENCES (SOURCE SEQUENCE NO. AND PAGE/BOOK)

(002067)	75.13	04																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																															
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07/03/74	TABLE OF CONTENTS AND REFERENCES		AUTOFLOW CHART SET - SHEEP			PAGE 18
CARD NO	PAGE/BOX	NAME	REFERENCES (SOURCE SEQUENCE NO. AND PAGE/BOX)			
(000000)	01.04	125	(000277)	00.17	(000277)	00.17
(000270)	01.06	127	(000270)	00.10	(000270)	00.10
(000271)	01.07	128			(000270)	00.10
(000273)	01.08	129	(000270)	01.00		
(000276)	01.09	130	(000275)	00.15	(000270)	00.10
(000278)	01.10	95.17			(000277)	01.05
(000277)	01.11	131				
(000278)	01.12	132	(002330)	02.00		
(000278)	01.13	133	(002343)	02.14		
(000280)	01.14	134	(000270)	01.10	(000277)	01.11
(000210)	01.16		(002313)	01.01		
(000213)	01.18	135				
(000216)	01.23	136				
(000217)	01.24	138				
(000220)	02.02		(002290)	02.00		
(000220)	02.03	141				
(000230)	02.05	142	(000216)	01.23		
(000231)	02.07	143	(000270)	01.12		
(000232)	02.10		(002342)	02.13		
(000242)	02.11	145				
(000244)	02.15	146	(000270)	01.13		
(000250)	02.18		(002300)	02.21		
(000250)	02.19	148				
(000250)	02.22	95.18	(000270)	01.00		
(000257)	02.23	149	(000270)	01.13	(000215)	01.02

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE NAME

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(000200)	05.01	4001	(000200)	05.03		
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(000412)	05.06	4012				
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(000423)	05.11	4013				
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(000434)	05.16	4014				
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(000443)	05.23	4020	(000420)	05.06	(000411)	05.22
(000454)	07.01	400	(000452)	05.25	(000411)	05.22
(000450)	07.02	401	(000454)	07.01		
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(000543)	08.06	208	(000530)	08.04		
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(000541)	08.01	209	(000530)	08.05		

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(002571)	09.06	215				
(002572)	09.07	221				
(002573)	09.08	220	(002570)	09.09	(002571)	09.06
(002575)	09.09	222	(002572)	09.07		
(002576)	09.10	223	(002574)	09.08		
(002577)	09.11	409				
(002581)	09.14		(002582)	09.16		
(002582)	09.15	4000				
(002583)	09.17	410	(002581)	09.16		
(002585)	09.18	412	(002586)	09.17		
(002594)	09.20		(002582)	09.04		
(002595)	09.22	216				
(002597)	09.23	231				
(002598)	09.24	230	(002596)	09.21	(002598)	09.22
(002600)	09.25	232	(002597)	09.23		
(002601)	09.26	233	(002599)	09.24		
(002602)	09.28	414				
(002605)	09.28		(002607)	09.28		
(002607)	09.27	4140				
(002610)	09.29	420	(002605)	09.17		
(002613)	09.31	9501				
(002620)	09.34	452	(002610)	09.13		
(002657)	09.20	456	(002610)	09.13		
(002690)	01.01	462	(002657)	09.20		
(002694)	01.15	464				
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(002703)	02.01	472	(002701)	01.17		
(002706)	02.15	9562	(002702)	09.10	(002701)	01.17
(002735)	02.17	3001				
(002736)	02.18	3002				
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					(002737)	02.19

CHART TITLE - NON-PROCEDURAL STATEMENTS

LOCATION		DIAGNOSTIC
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(000030)	2. 02	UNRECOGNIZED SYNTAX
(000005)	2. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000005)	2. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000007)	2. 20	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000008)	2. 21	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 30	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 31	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 32	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000401)	3. 10	UNRECOGNIZED SYNTAX
(000535)	30. 01	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000537)	30. 02	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000538)	30. 03	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000538)	30. 04	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000540)	30. 05	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(001005)	37. 14	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001245)	40. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001303)	40. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001305)	40. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(001705)	01. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(000700)	05. 01	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000701)	05. 02	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000740)	05. 20	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(000741)	05. 21	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE

MODULE FLOW CHARTS

CHART TITLE - INTRODUCTORY COMMENTS

PROGRAM BLCHIL IS LINE 4 OF SHEEP

DETERMINES BASIC AIRLOAD CONDITIONS TO BE COMPUTED ON AND WING
BENDING MOMENT FATIGUE SPECTRA
PROVIDES LOGIC AND CONTROL FOR THE AIRLOAD SUBROUTINES

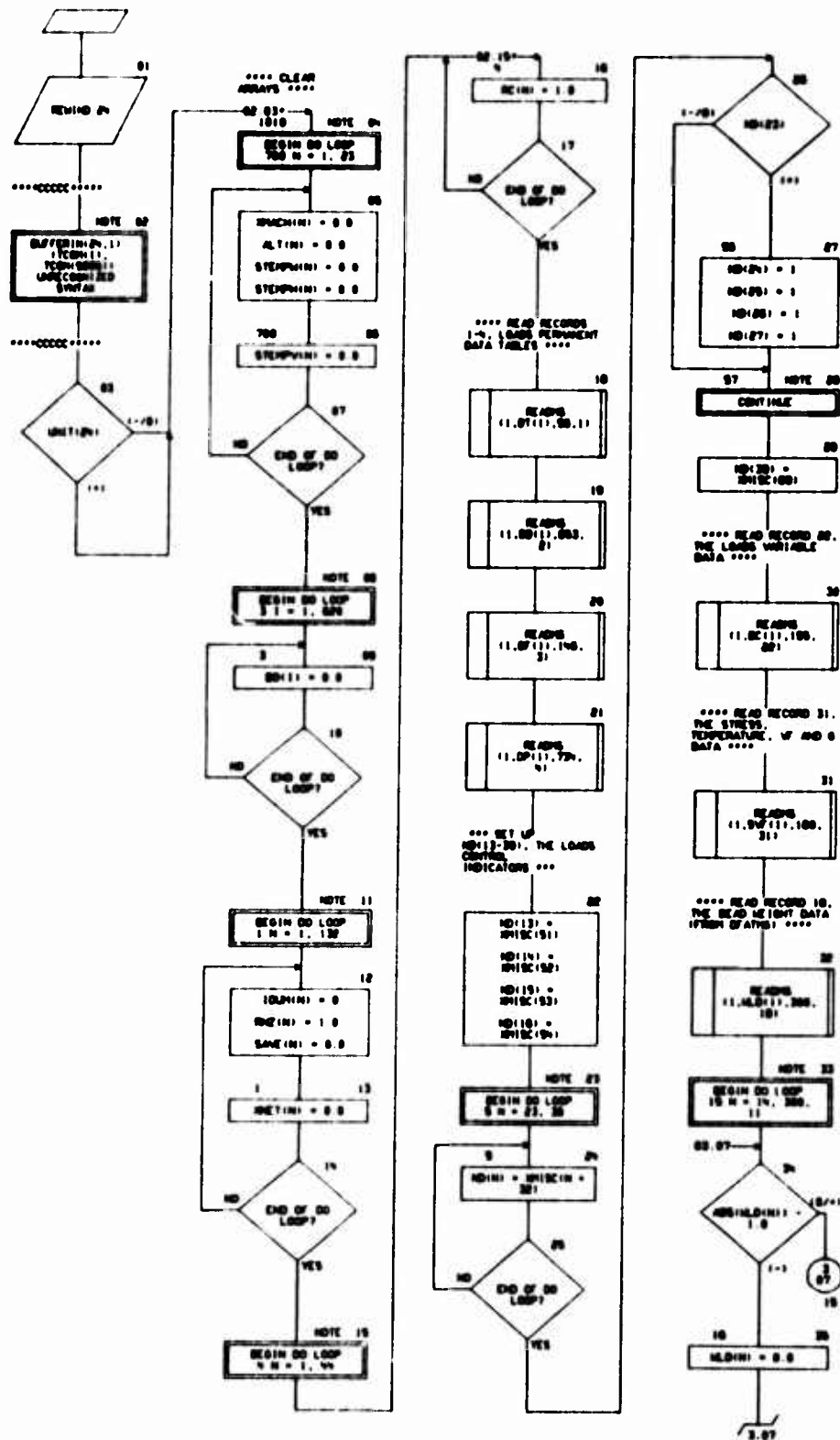
CHART TITLE - PROCESSES

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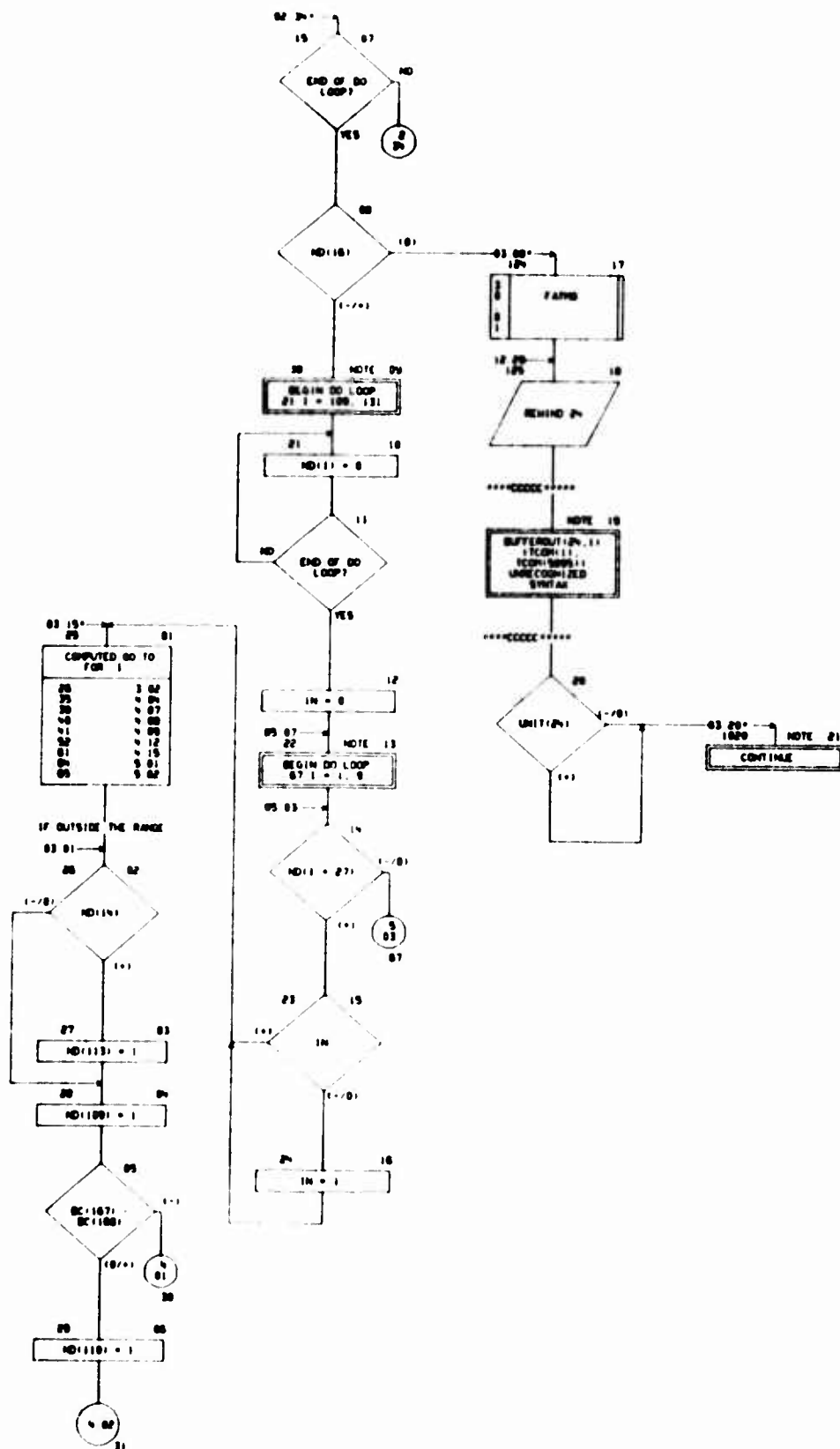


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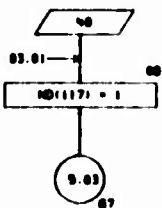
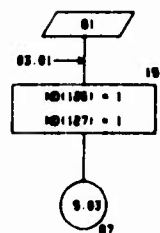
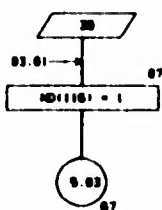
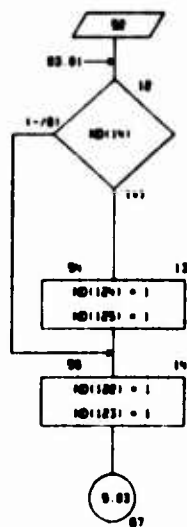
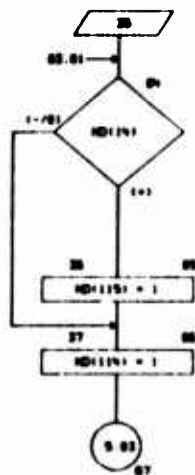
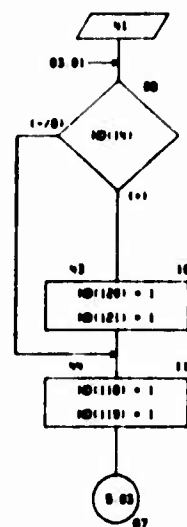
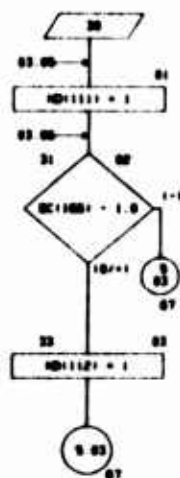


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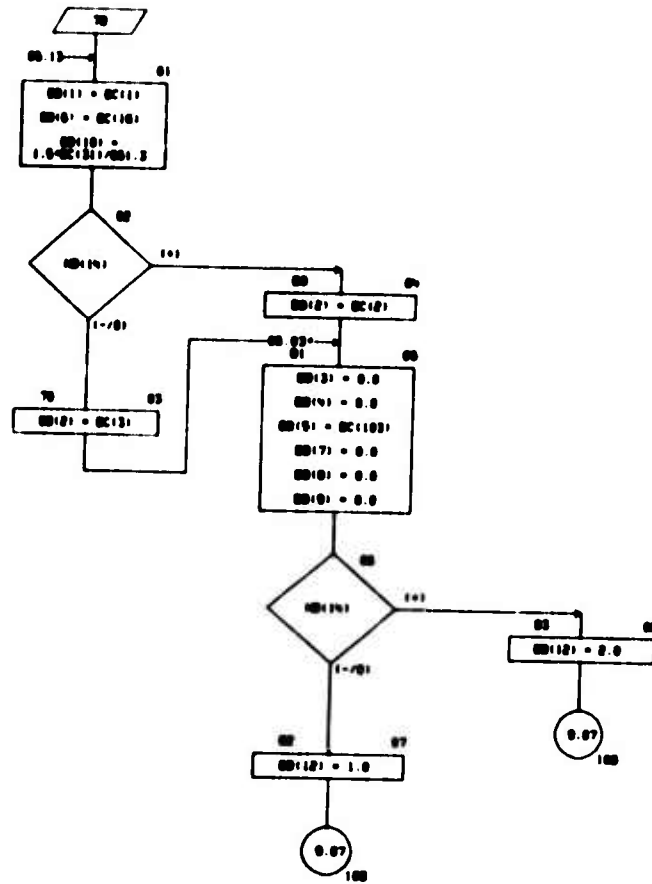


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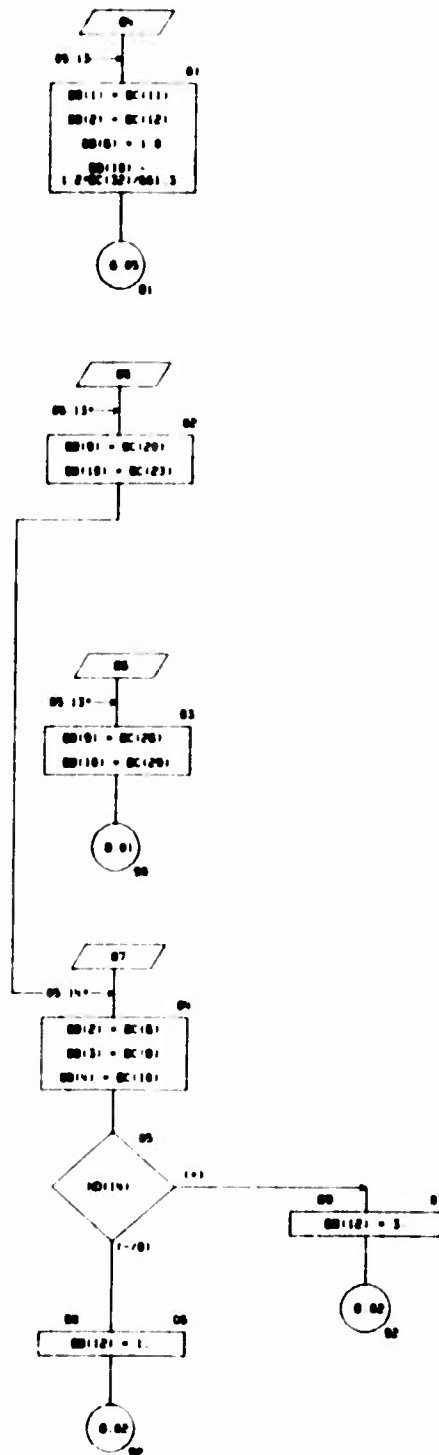


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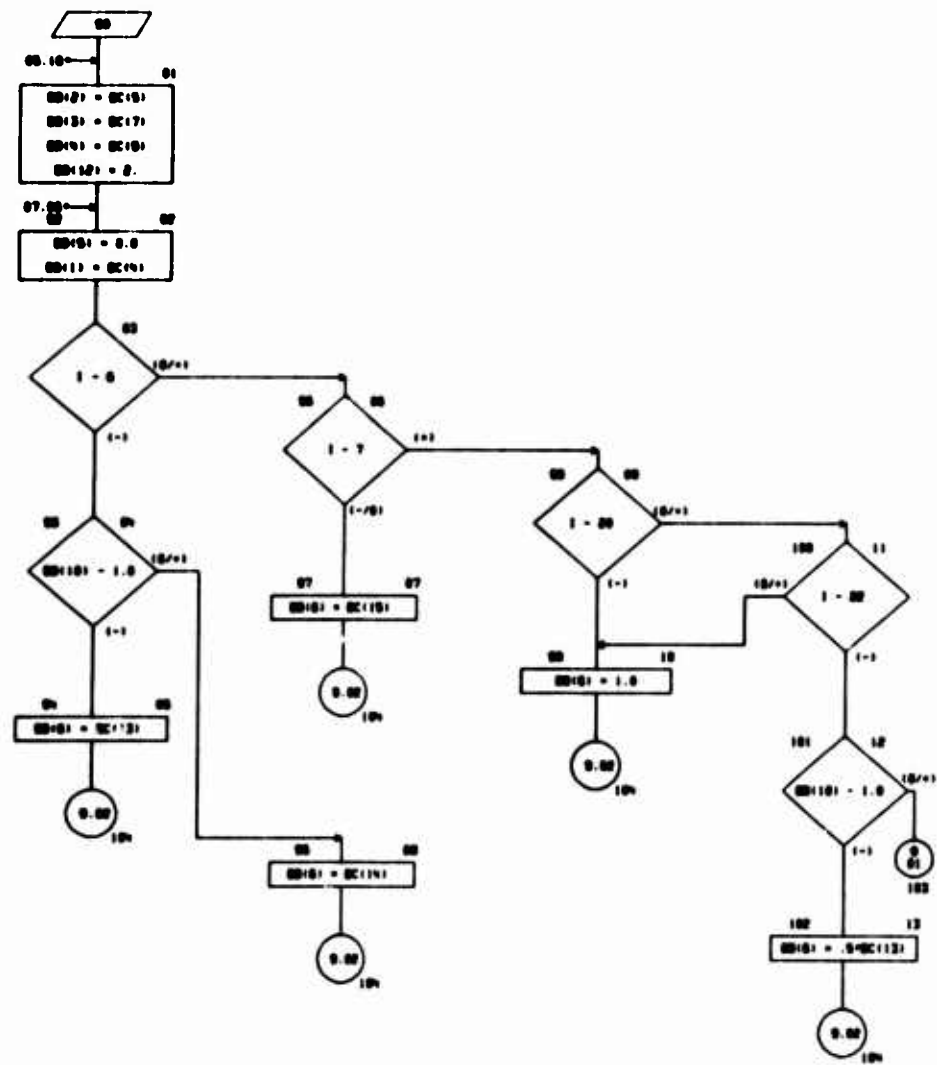


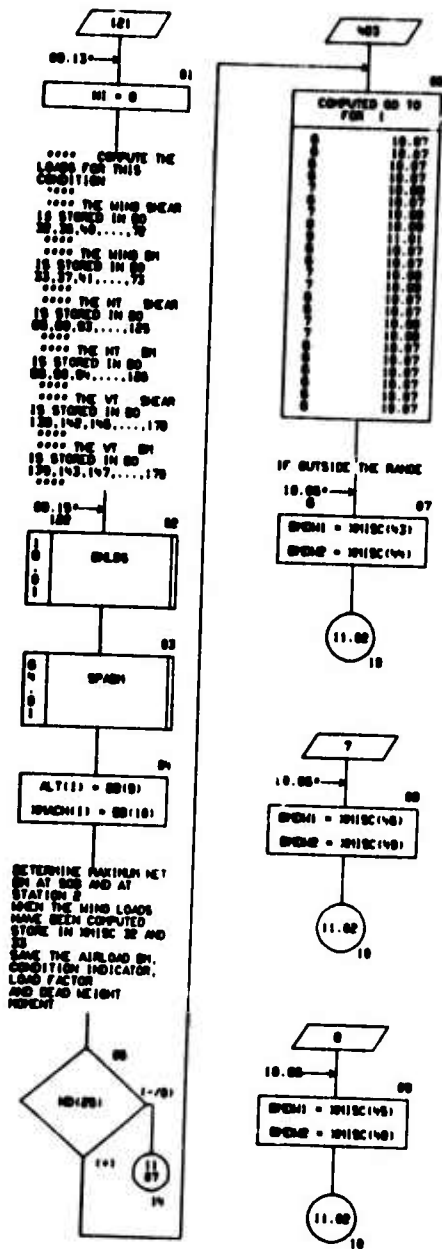
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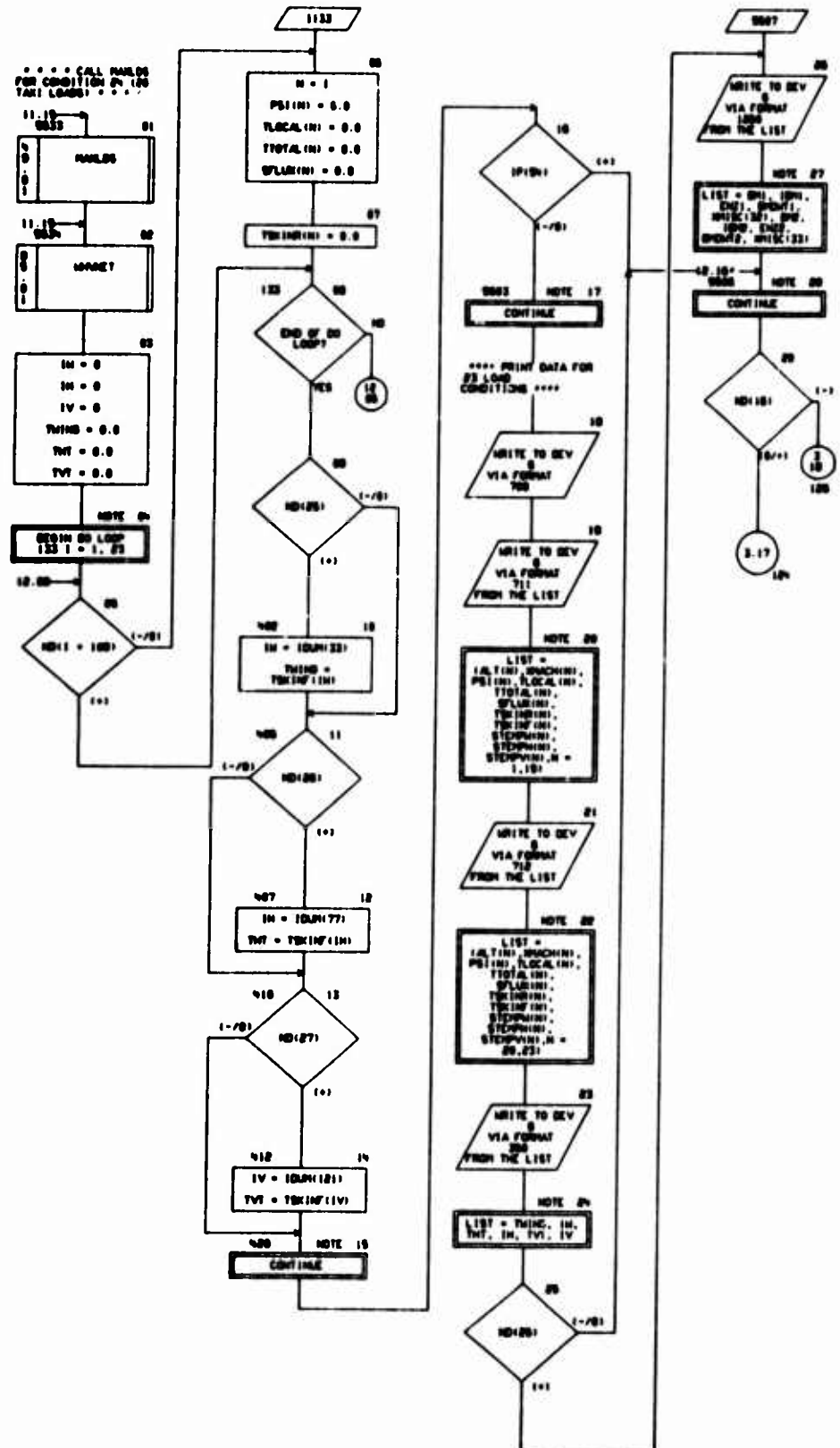


CHART TITLE - INTRODUCTORY COMMENTS

SUBMIT THE ATYDS

CHART TITLE - SUBROUTINE ATNDSIM, REND, PH, OH

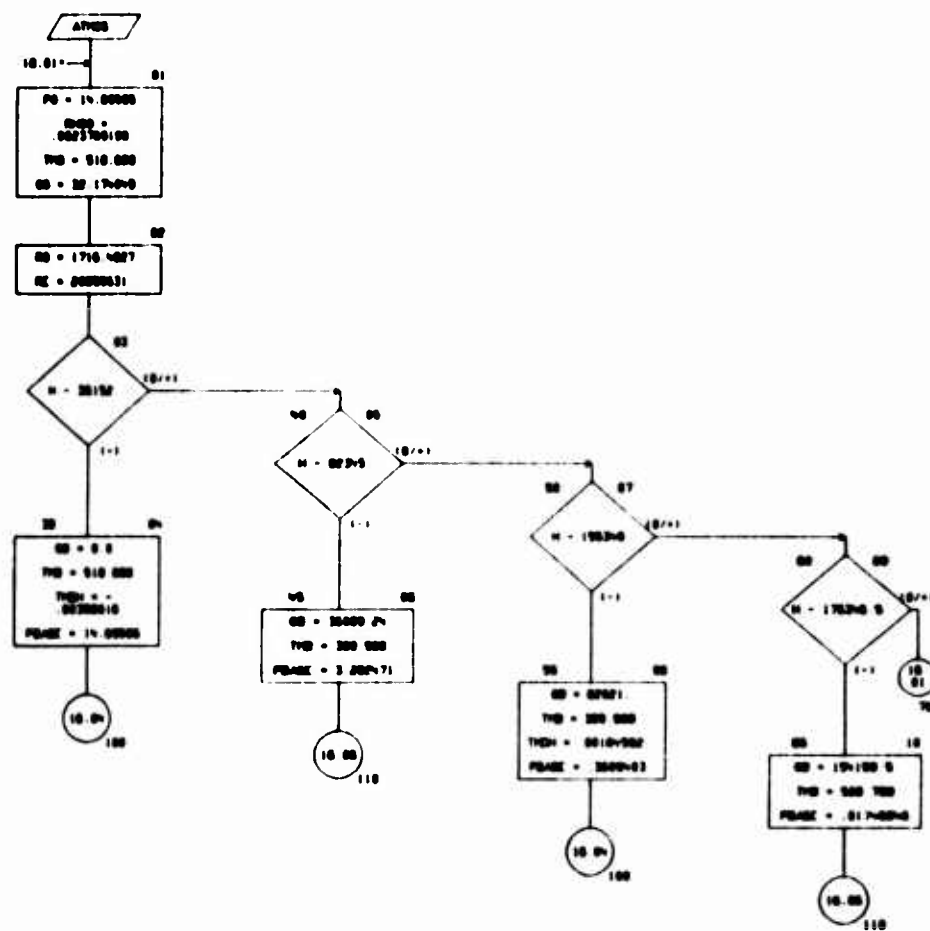


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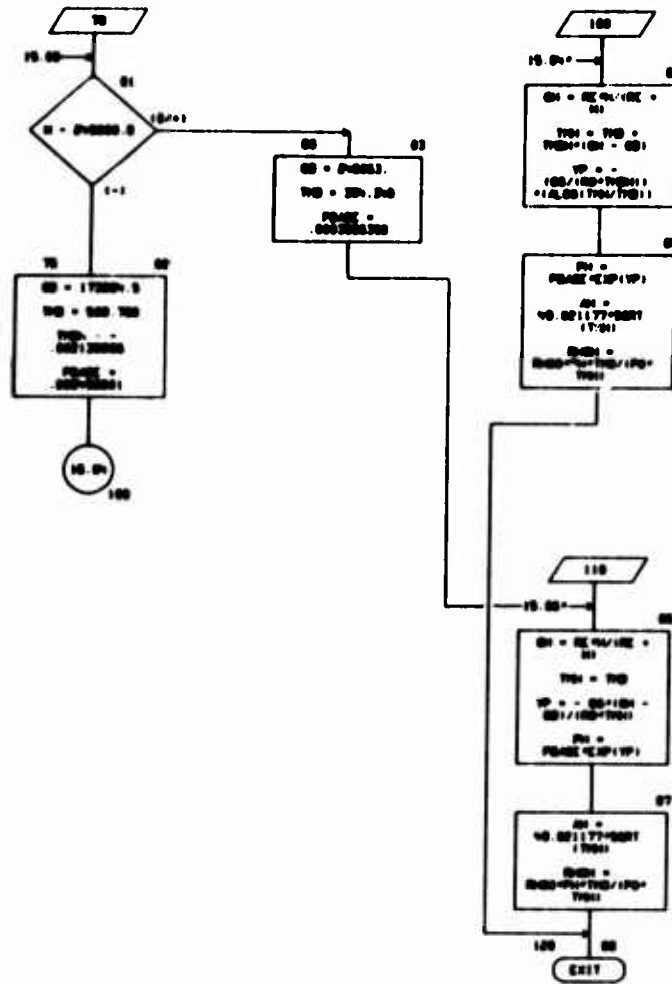


CHART TITLE - INTRODUCTORY COMMENTS

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SUBROUTINE END
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CHART TITLE - SUBROUTINE DLDS

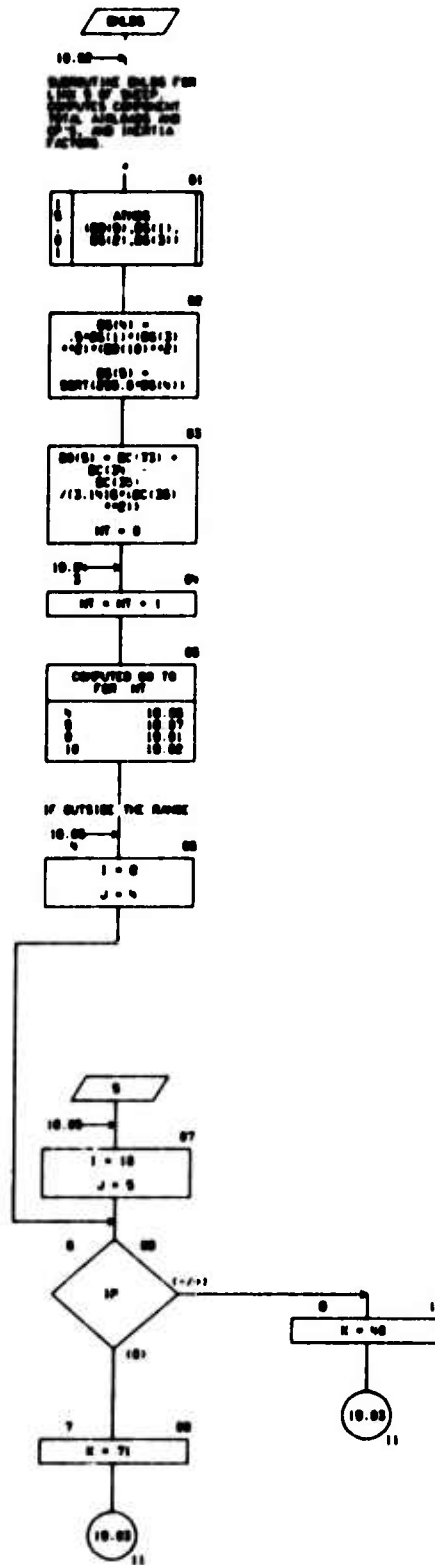
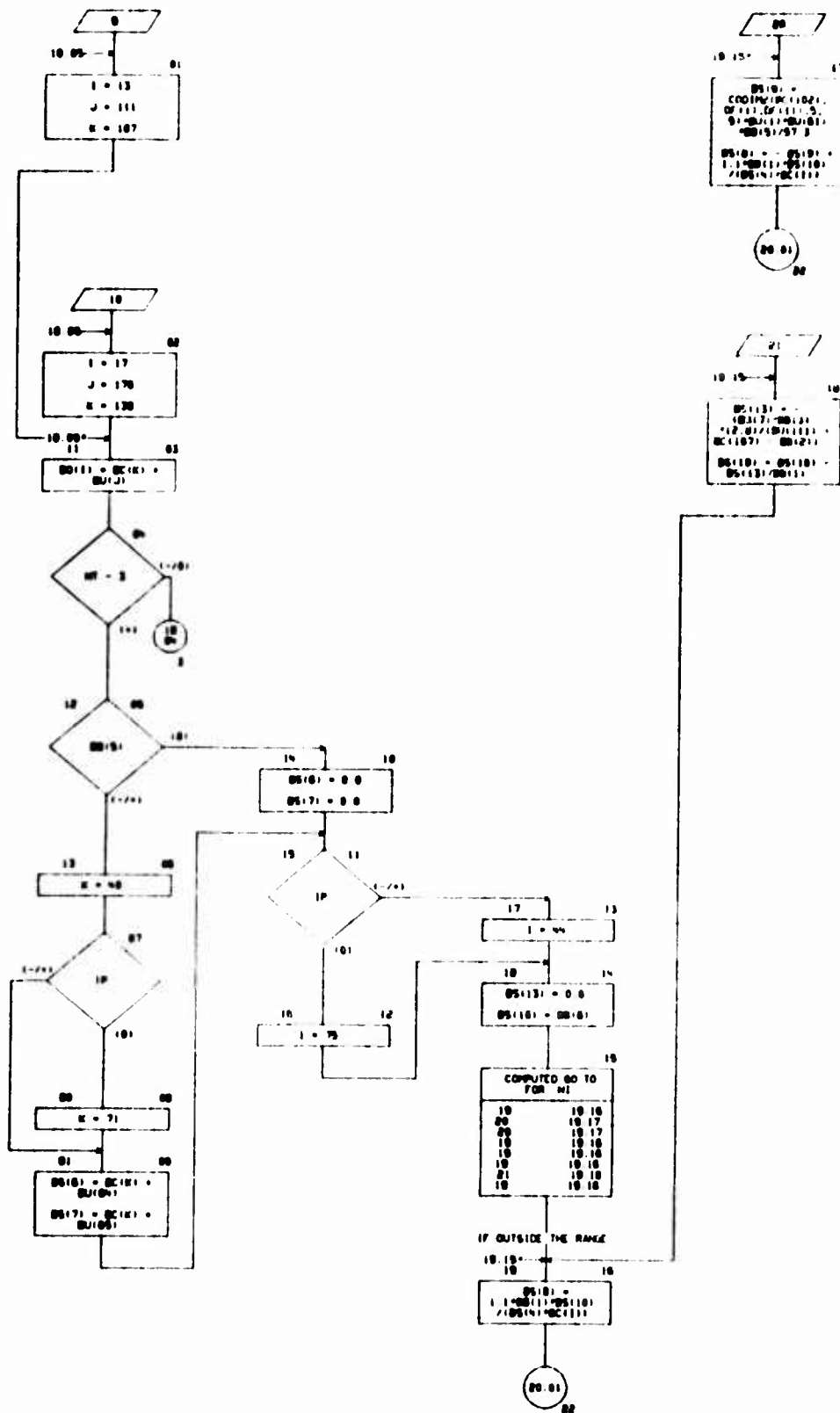


CHART TITLE - SUBROUTINE SWEEP





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AUTOFLEX CHART SET - SHEEP AIRLADS OVERLAY

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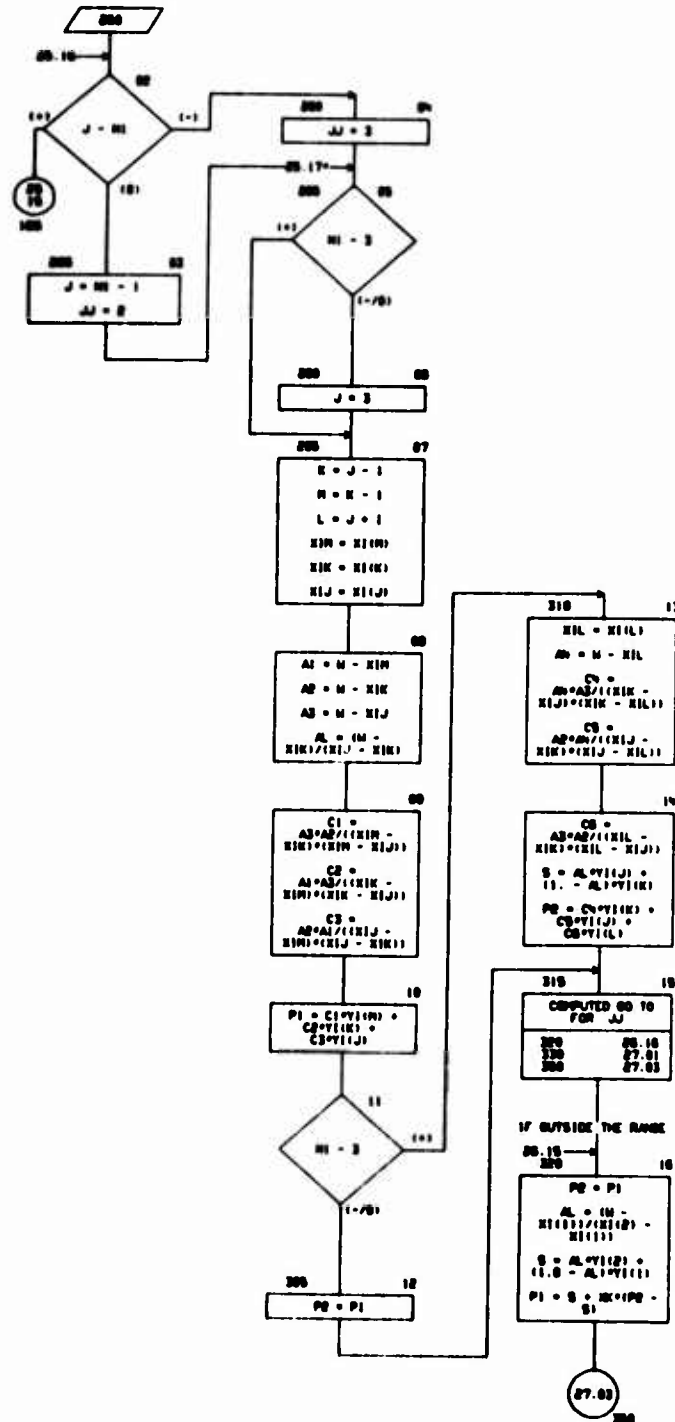
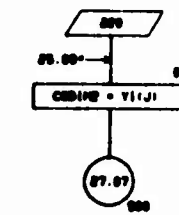
CHART TITLE - INTRODUCTORY COMMENTS

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FUNCTION CODING
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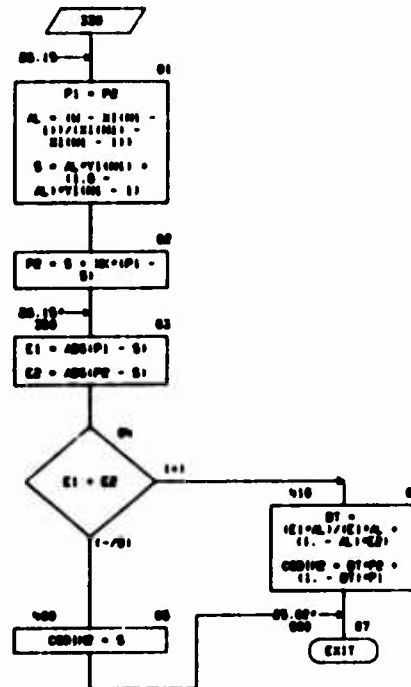

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graph TD
    Start([Start]) --> 05[M = N  
NI = N]
    05 --> 06{NI = 2}
    06 -- (+) --> 07[CODIME = V(NI)  
V(NI) = V(NI) - 1  
NI = NI - 1  
NI = NI - 1  
NI = NI - 1  
NI = NI - 1]
    07 --> 07.07((27.07))
    06 -- (-) --> 08[CODIME = V(NI)]
    08 --> 07.07
    06 -- (+) --> 09{NI11 = NI(NI)}
    09 -- (+) --> 100[NOTE 07  
BEGIN DO LOOP  
170 I = 1, NI]
    09 -- (-) --> 101((27.07))
    100 --> 102[J = 1]
    102 --> 103{M = NI1J1}
    103 -- (+) --> 104((27.07))
    103 -- (-) --> 105{END OF DO LOOP?}
    105 -- YES --> 106[NI = NI]
    106 --> 107((27.07))
    105 -- NO --> 108[170]
    108 --> 109[100]
    109 --> 110{J = 2}
    110 -- (+) --> 111((27.07))
    110 -- (-) --> 112[100]
    112 --> 113[J = 3  
JJ = 1]
    113 --> 114((27.07))
    110 -- (+) --> 115[100]
    115 --> 116{J = 2}
    116 -- (+) --> 117((27.07))
    116 -- (-) --> 118[100]
    118 --> 119[J = 3  
JJ = 1]
    119 --> 120((27.07))
    116 -- (+) --> 121[100]
    121 --> 122{J = 2}
    122 -- (+) --> 123((27.07))
    122 -- (-) --> 124[100]
    124 --> 125[J = 3  
JJ = 1]
    125 --> 126((27.07))
    122 -- (+) --> 127[100]
    127 --> 128{J = 2}
    128 -- (+) --> 129((27.07))
    128 -- (-) --> 130[100]
    130 --> 131[J = 3  
JJ = 1]
    131 --> 132((27.07))
    128 -- (+) --> 133[100]
    133 --> 134{J = 2}
    134 -- (+) --> 135((27.07))
    134 -- (-) --> 136[100]
    136 --> 137[J = 3  
JJ = 1]
    137 --> 138((27.07))
    134 -- (+) --> 139[100]
    139 --> 140{J = 2}
    140 -- (+) --> 141((27.07))
    140 -- (-) --> 142[100]
    142 --> 143[J = 3  
JJ = 1]
    143 --> 144((27.07))
    140 -- (+) --> 145[100]
    145 --> 146{J = 2}
    146 -- (+) --> 147((27.07))
    146 -- (-) --> 148[100]
    148 --> 149[J = 3  
JJ = 1]
    149 --> 150((27.07))
    146 -- (+) --> 151[100]
    151 --> 152{J = 2}
    152 -- (+) --> 153((27.07))
    152 -- (-) --> 154[100]
    154 --> 155[J = 3  
JJ = 1]
    155 --> 156((27.07))
    152 -- (+) --> 157[100]
    157 --> 158{J = 2}
    158 -- (+) --> 159((27.07))
    158 -- (-) --> 160[100]
    160 --> 161[J = 3  
JJ = 1]
    161 --> 162((27.07))
    158 -- (+) --> 163[100]
    163 --> 164{J = 2}
    164 -- (+) --> 165((27.07))
    164 -- (-) --> 166[100]
    166 --> 167[J = 3  
JJ = 1]
    167 --> 168((27.07))
    164 -- (+) --> 169[100]
    169 --> 170{J = 2}
    170 -- (+) --> 171((27.07))
    170 -- (-) --> 172[100]
    172 --> 173[J = 3  
JJ = 1]
    173 --> 174((27.07))
    170 -- (+) --> 175[100]
    175 --> 176{J = 2}
    176 -- (+) --> 177((27.07))
    176 -- (-) --> 178[100]
    178 --> 179[J = 3  
JJ = 1]
    179 --> 180((27.07))
    176 -- (+) --> 181[100]
    181 --> 182{J = 2}
    182 -- (+) --> 183((27.07))
    182 -- (-) --> 184[100]
    184 --> 185[J = 3  
JJ = 1]
    185 --> 186((27.07))
    182 -- (+) --> 187[100]
    187 --> 188{J = 2}
    188 -- (+) --> 189((27.07))
    188 -- (-) --> 190[100]
    190 --> 191[J = 3  
JJ = 1]
    191 --> 192((27.07))
    188 -- (+) --> 193[100]
    193 --> 194{J = 2}
    194 -- (+) --> 195((27.07))
    194 -- (-) --> 196[100]
    196 --> 197[J = 3  
JJ = 1]
    197 --> 198((27.07))
    194 -- (+) --> 199[100]
    199 --> 200{J = 2}
    200 -- (+) --> 201((27.07))
    200 -- (-) --> 202[100]
    202 --> 203[J = 3  
JJ = 1]
    203 --> 204((27.07))
    200 -- (+) --> 205[100]
    205 --> 206{J = 2}
    206 -- (+) --> 207((27.07))
    206 -- (-) --> 208[100]
    208 --> 209[J = 3  
JJ = 1]
    209 --> 210((27.07))
    206 -- (+) --> 211[100]
    211 --> 212{J = 2}
    212 -- (+) --> 213((27.07))
    212 -- (-) --> 214[100]
    214 --> 215[J = 3  
JJ = 1]
    215 --> 216((27.07))
    212 -- (+) --> 217[100]
    217 --> 218{J = 2}
    218 -- (+) --> 219((27.07))
    218 -- (-) --> 220[100]
    220 --> 221[J = 3  
JJ = 1]
    221 --> 222((27.07))
    218 -- (+) --> 223[100]
    223 --> 224{J = 2}
    224 -- (+) --> 225((27.07))
    224 -- (-) --> 226[100]
    226 --> 227[J = 3  
JJ = 1]
    227 --> 228((27.07))
    224 -- (+) --> 229[100]
    229 --> 230{J = 2}
    230 -- (+) --> 231((27.07))
    230 -- (-) --> 232[100]
    232 --> 233[J = 3  
JJ = 1]
    233 --> 234((27.07))
    230 -- (+) --> 235[100]
    235 --> 236{J = 2}
    236 -- (+) --> 237((27.07))
    236 -- (-) --> 238[100]
    238 --> 239[J = 3  
JJ = 1]
    239 --> 240((27.07))
    236 -- (+) --> 241[100]
    241 --> 242{J = 2}
    242 -- (+) --> 243((27.07))
    242 -- (-) --> 244[100]
    244 --> 245[J = 3  
JJ = 1]
    245 --> 246((27.07))
    242 -- (+) --> 247[100]
    247 --> 248{J = 2}
    248 -- (+) --> 249((27.07))
    248 -- (-) --> 250[100]
    250 --> 251[J = 3  
JJ = 1]
    251 --> 252((27.07))
    248 -- (+) --> 253[100]
    253 --> 254{J = 2}
    254 -- (+) --> 255((27.07))
    254 -- (-) --> 256[100]
    256 --> 257[J = 3  
JJ = 1]
    257 --> 258((27.07))
    254 -- (+) --> 259[100]
    259 --> 260{J = 2}
    260 -- (+) --> 261((27.07))
    260 -- (-) --> 262[100]
    262 --> 263[J = 3  
JJ = 1]
    263 --> 264((27.07))
    260 -- (+) --> 265[100]
    265 --> 266{J = 2}
    266 -- (+) --> 267((27.07))
    266 -- (-) --> 268[100]
    268 --> 269[J = 3  
JJ = 1]
    269 --> 270((27.07))
    266 -- (+) --> 271[100]
    271 --> 272{J = 2}
    27
```

CHART TITLE - FUNCTION CODES: K, N, I, Y, H, N, K, I



SUBRT TITLE - FUNCTION CODE1012,01,01,N,001



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AUTOLIN CHART SET - SHEEP AIRLINES OVERLAY

PAGE 00

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

07:03 PM

AUTORELON CHART SET - SHEEP AIRMOOR OVERLAY

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CHART TITLE - INTERPOLATE CURRENTS

.....
SUBROUTINE PATH
.....

.. 2.

CHART TITLE - SUBROUTINE - 1140

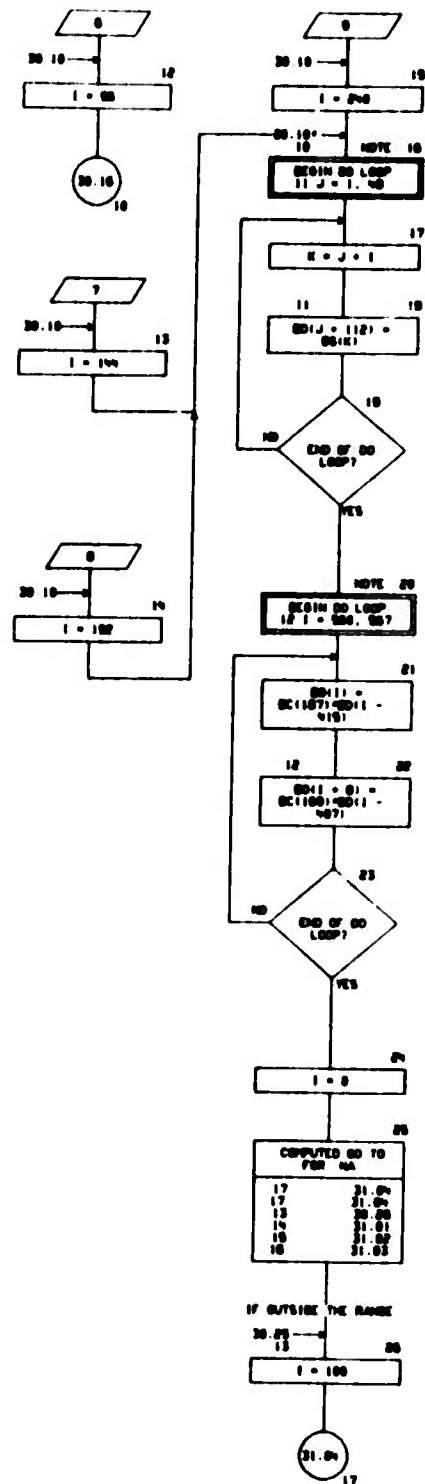
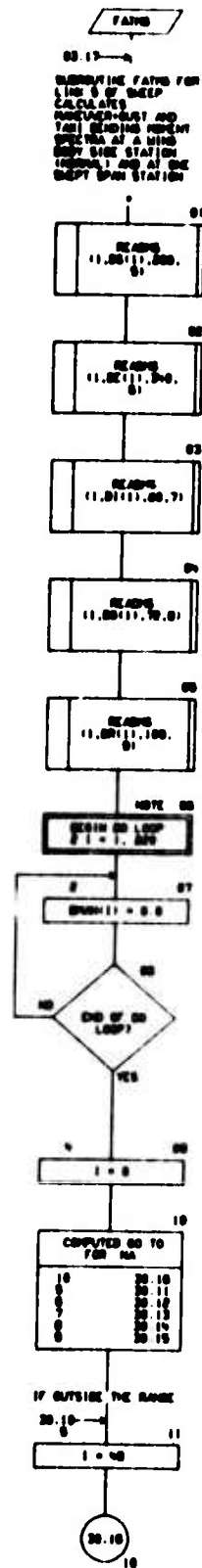
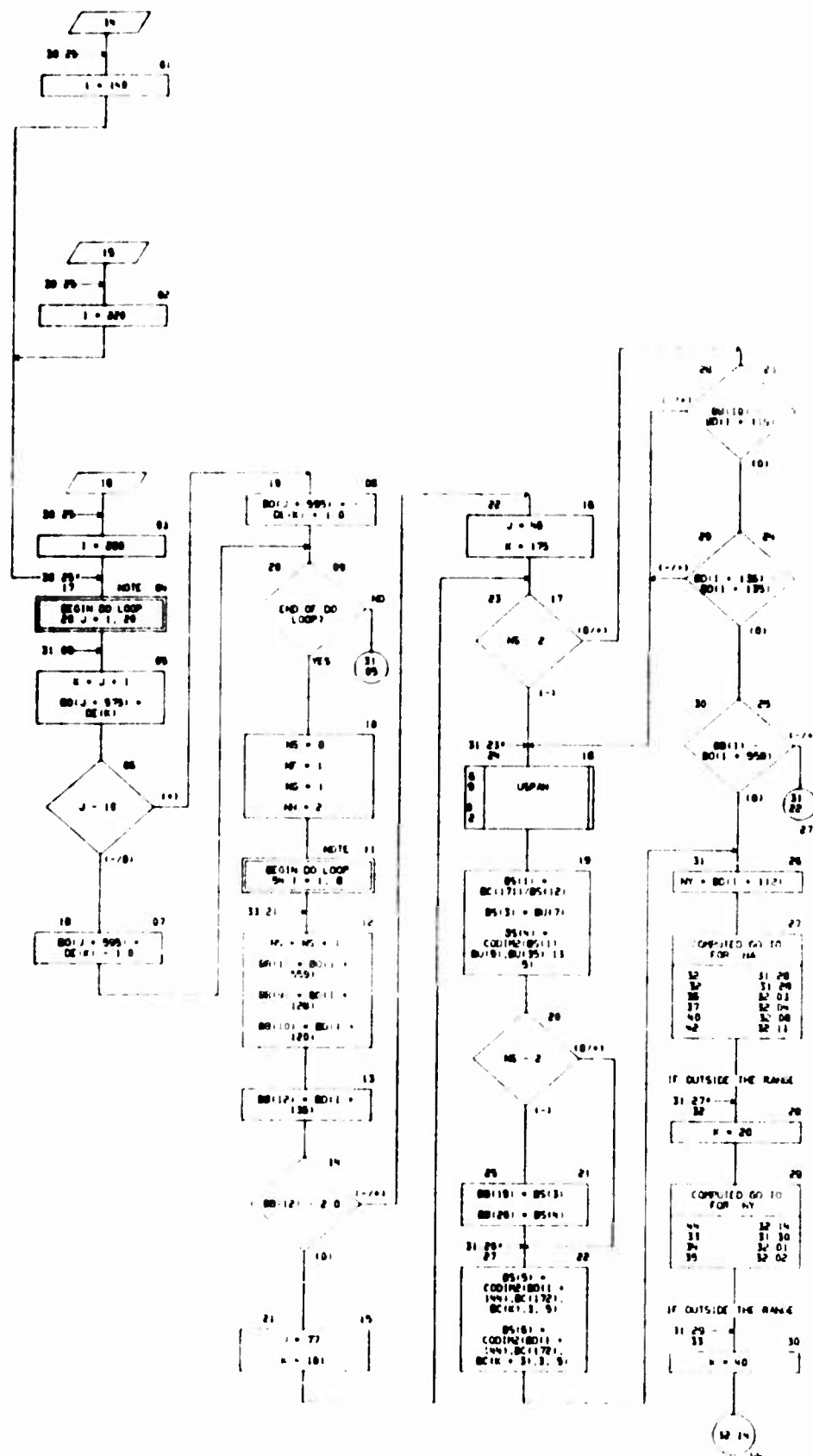


CHART TITLE - SUBROUTINE NAME



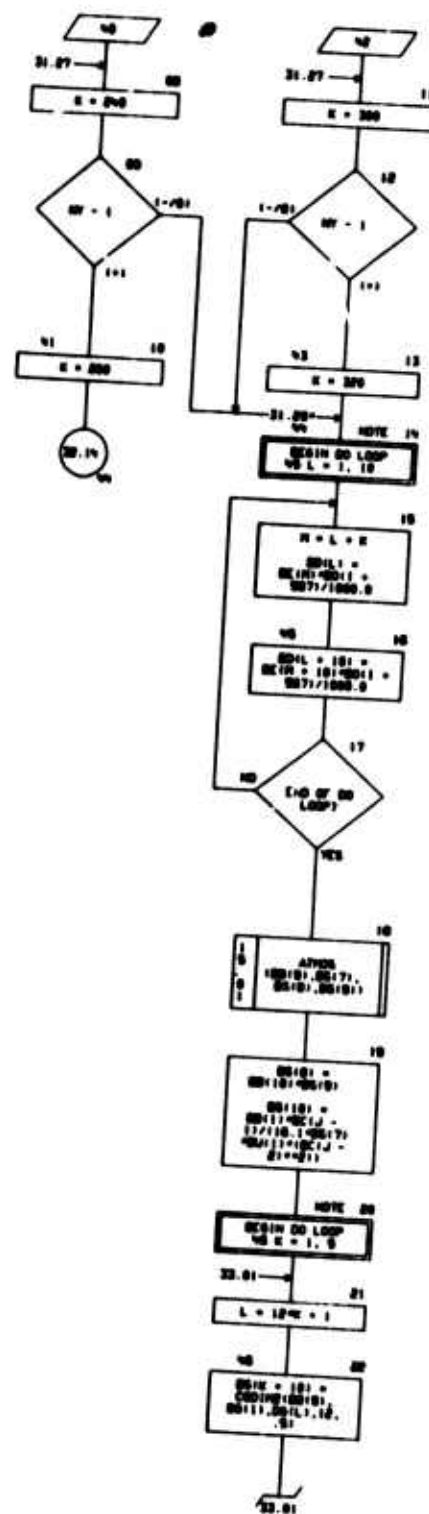


CHART TITLE - SUBROUTINE FATHG

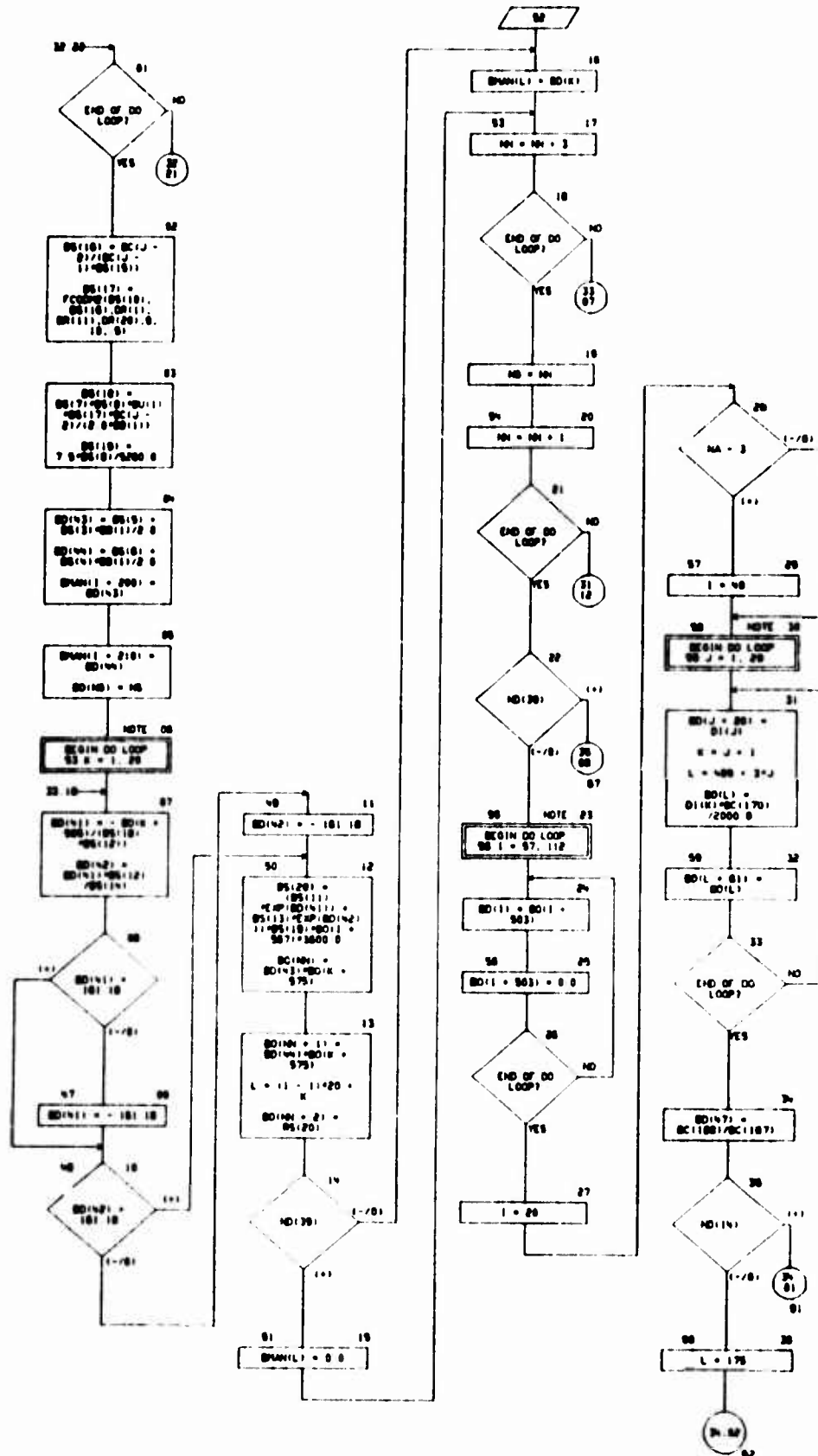
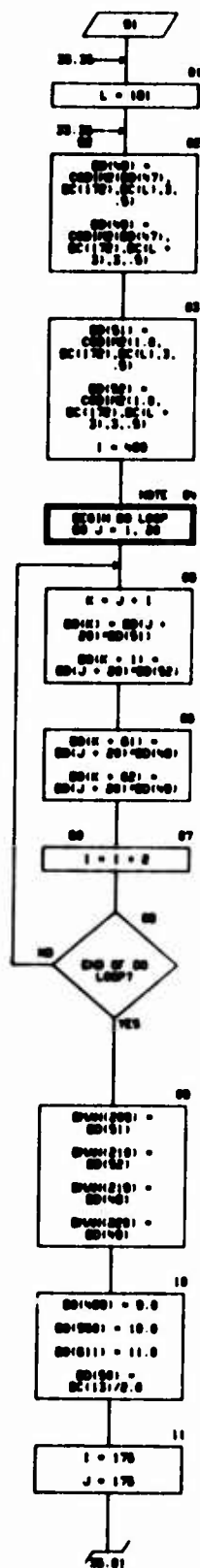


CHART TITLE - SUBROUTINE FATHS



340

CHART TITLE - SUBROUTINE FAIRING

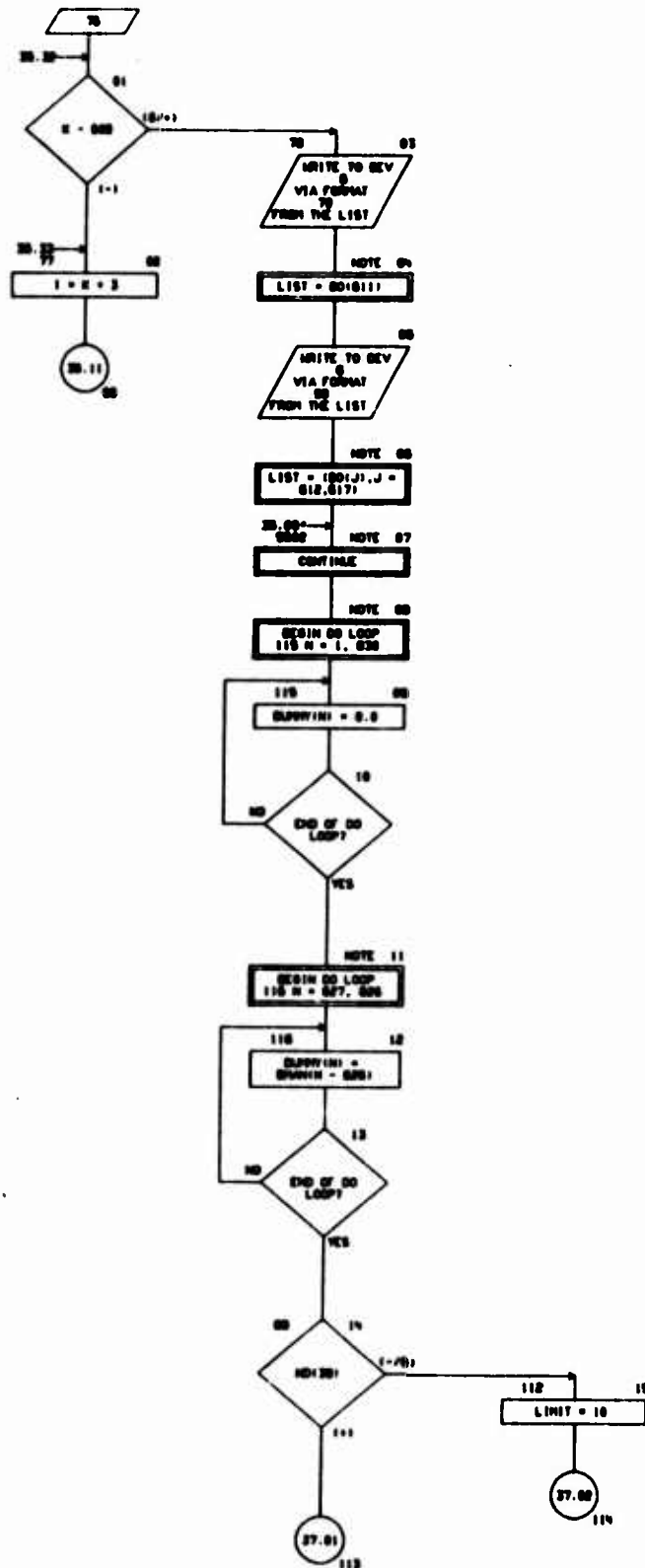


CHART TITLE - SUBROUTINE FATHS

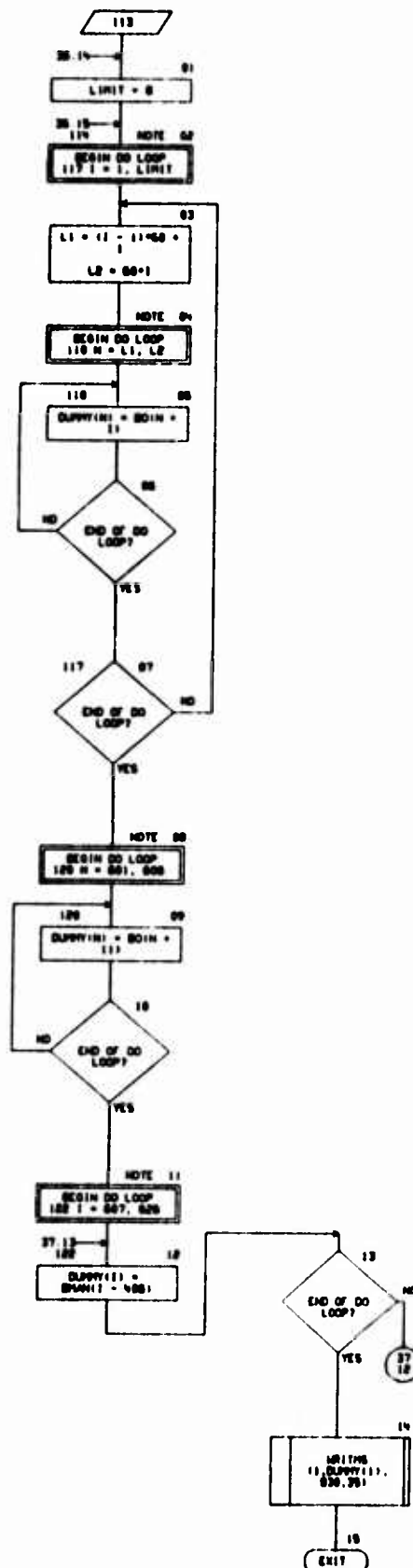


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCOM(400)
COMMON /IPRINT/10(00)
DIMENSION DS(200),DE(240),DI(60),DO(70),DR(100),ND(200),DC(100),DS
(00),DD(100),DU(200),DD(60),DD(20),DMM(200)
DIMENSION SAMP(600)
EQUIVALENCE (DS(1),TCOM(1000)),(DE(1),TCOM(2177)),(DI(1),TCOM(2517
)),(DD(1),TCOM(2577)),(DR(1),TCOM(2640)),(ND(1),TCOM(4201)),(DC(1)
,TCOM(2700)),(DD(1),TCOM(2003)),(DS(1),TCOM(2073)),(DD(1),TCOM(200
3)),(DU(1),TCOM(3103)),(DD(1),TCOM(3201)),(NA,ND(13)),(NF,ND(100))
,(NS,ND(100)),(NS,ND(101)),(NN,ND(102)),(NV,ND(103)),(I,ND(104)),(
J,ND(105)),(K,ND(106)),(L,ND(107)),(M,ND(108))
64 FORMAT(1H,00H,20H** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//7X,
      10H00F SEND MDM,3X,10H40S SEND MDM,5X,10HCEEDANCES-TAKI)
66 FORMAT(1H,00H,20H** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//7X,
      10H00F SEND MDM,3X,10H40S SEND MDM,5X,10HCEEDANCES-GUST,3X,
      10HCEEDANCES-VAN)
71 FORMAT(1H0)
100 FORMAT(1H 4HF10.0,5E12.0)
73 FORMAT(1H 4HF10.0,5E12.0,5E12.0)
65 FORMAT(//11H010 00F M =,F10.0,3X,10H10 MDS M =,F10.0)
70 FORMAT(1H,00H,20H** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//5X,24HROUND-AIR-ORD
      UND CYCLES//7X,27H00F SEND MDM 100S SEND MDM,7X,11H00CURRENCES)
60 FORMAT(1H ,4X,0F10.0,5X,1E12.0 )

```

CHART TITLE - INTRODUCTORY COMMENTS

FUNCTION ROOMS

TWO DIMENSION CURVE
 SUBPROGRAM -- FCONE
 CALLING SEQUENCE
 FCONEIN,V,II,V1,ZI,
 N,NO,NI
 N = NUMBER - 1ST
 VARIABLE
 NI = NUMBER - 2ND
 VARIABLE
 ZI = ARRAY OF 1ST
 VARIABLE
 V1 = ARRAY OF 2ND
 VARIABLE
 ZI = ARRAY OF THE
 INDEPENDENT VARIABLE
 NI = NO. OF POINTS -
 NO. = NO. OF POINTS -
 NI
 NI = END INTERVAL
 INTERPOLATION CONTROL
 CONSTANT

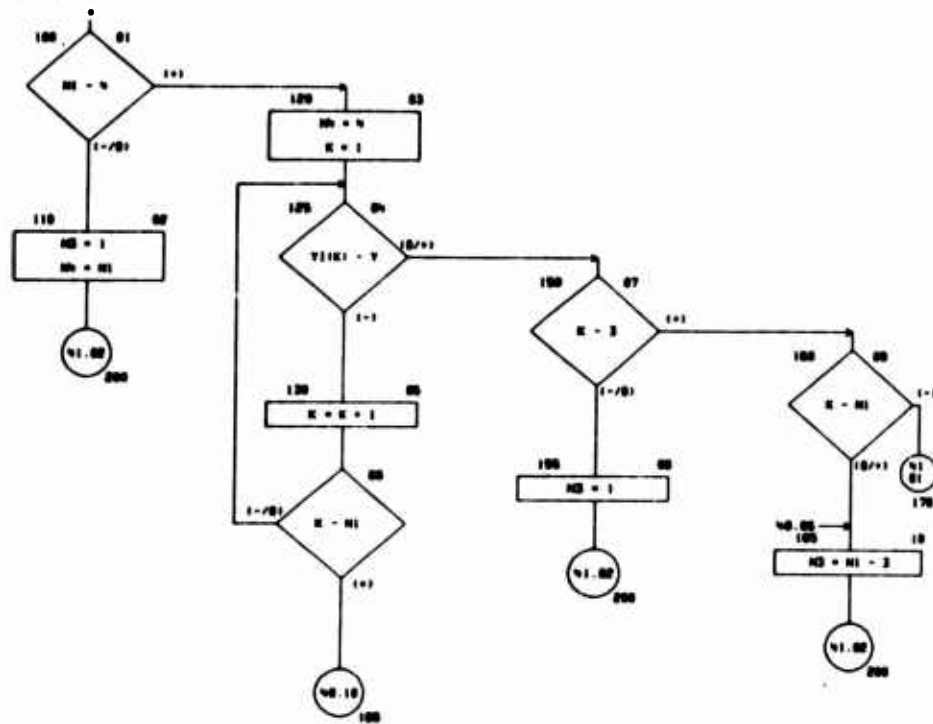
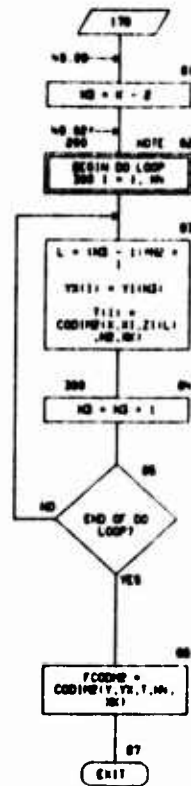


CHART TITLE - FUNCTION FCODEP(X,V,XI,YI,ZI,NI,NO,NOI)



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```

COMMON TCOM(400)
DIMENSION MD(200),M1(1),V(11),Z(11),V4(1),VX(4)
EQUIVALENCE MD(1),TCOM(401),
              (MD,MD(100),MD,MD(100),1,MD(140),1,MD(140),1,
MD(140))

```

07:03:24

AUTOFLOW CHART SET SWEEP AIRLOADS OVERLAY

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CHART TITLE INTERLUCKTORY CURRENTS

SUBROUTINE ROUTINE

CHART TITLE - SUBROUTINE FUSNET

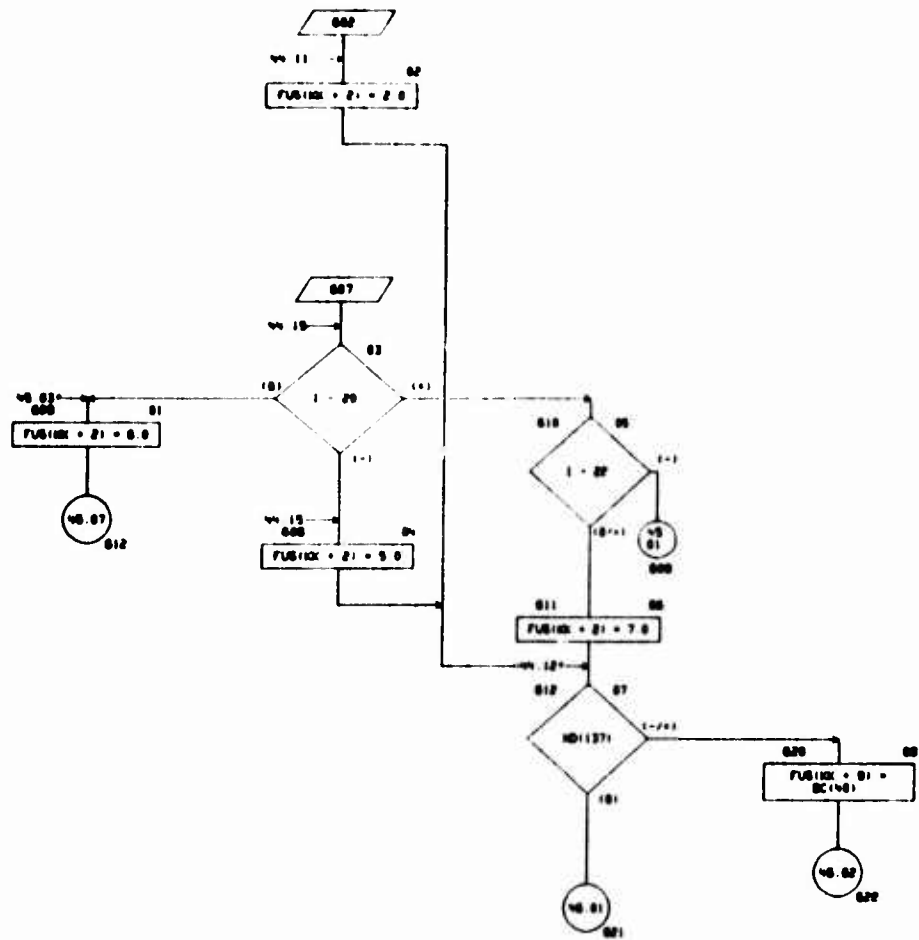


CHART TITLE - SUBROUTINE FUSHE1

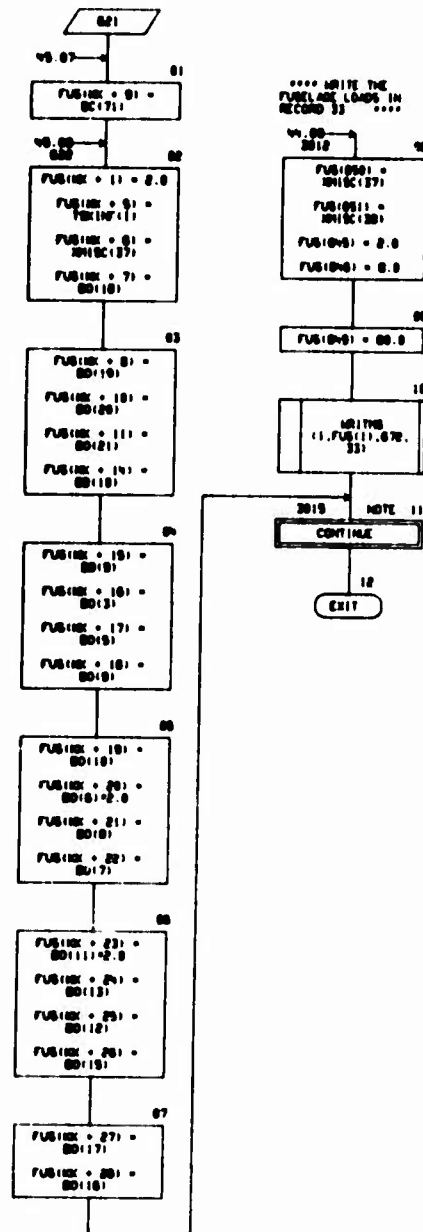


CHART TITLE - NON-PROCEDURAL STATEMENTS

```
COMMON /COM1400/,MD1300,LDUR1132,SA4E1132,ME11132,
      RN21132,RC144,SW1100,STEMPH123,STEMPH123,STEMPH123,
      DUM1100,
COMMON /MISC/,MISC1100,
DIMENSION MD1200,BO1620,BC1105,
DIMENSION TSWINF123,
DIMENSION PUS1070,OB120,
EQUIVALENCE (MD11,TCOM140011,10011,TCOM130011,10011,TCOM12700
11,11,MD11011,10011,TCOM120511,
EQUIVALENCE (SW1101,TSWINF111)
```

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AUTOFLEX CHART SET - SHEEP AIRLOADS OVERLAY

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CHART TITLE - INTERDUCTORY CONTENTS

.....
SUBROUTINE MAULDS
.....


```

graph TD
    Start(( )) --> Box[COMPUTED SQ RT FOR X  
SQ1 40.25  
SQ2 50.50  
SQ3 50.50]
    Box --> Decision{WH'EC(1)}
    Decision -- Y --> Circle1((SQ SQ))
    Decision -- N --> Circle2((SQ 01))
    Circle1 --> End1(( ))
    Circle2 --> End2(( ))
  
```

CHART TITLE - SUBROUTINE MAULD

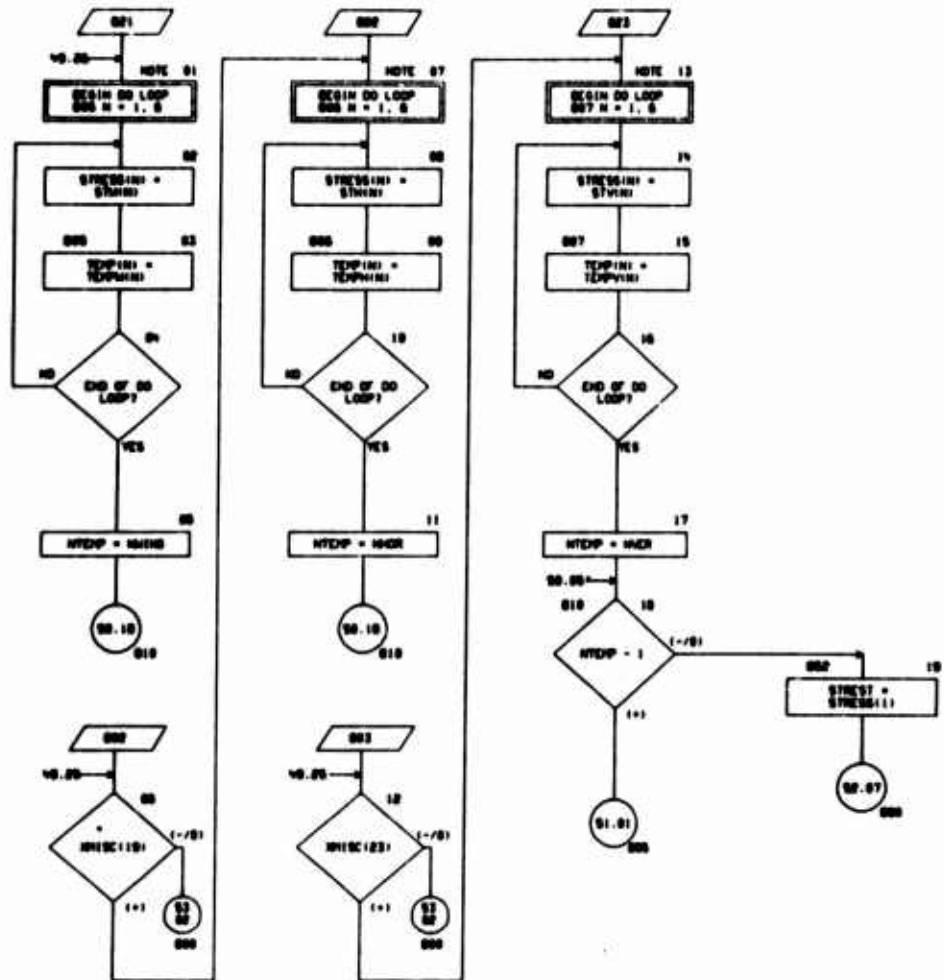


CHART TITLE - SUBROUTINE HARDS

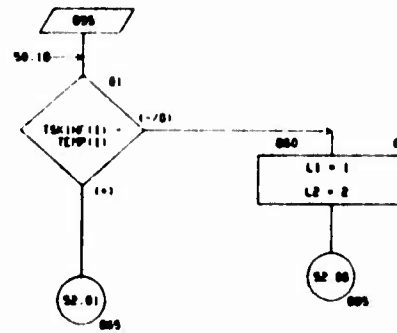




CHART TITLE - SUBROUTINE HANDS

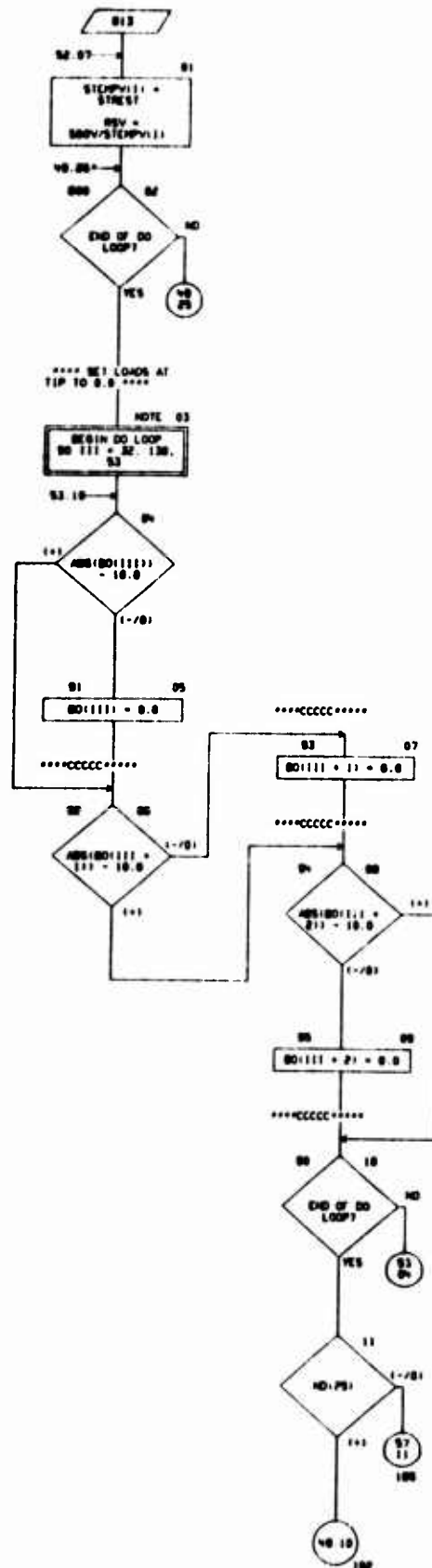


CHART TITLE - SUBROUTINE HAILDS

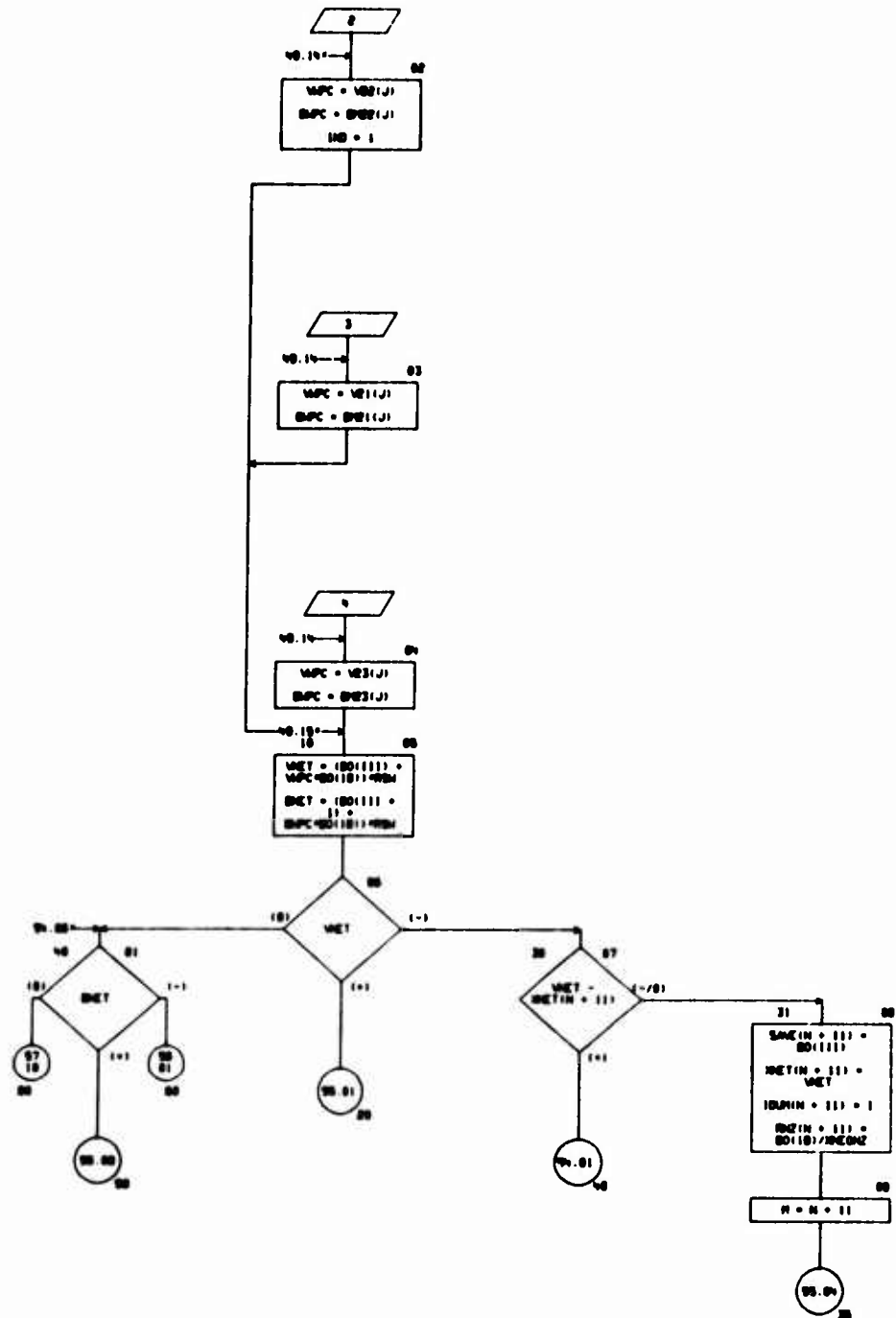


CHART TITLE - SUBROUTINE PARLOS

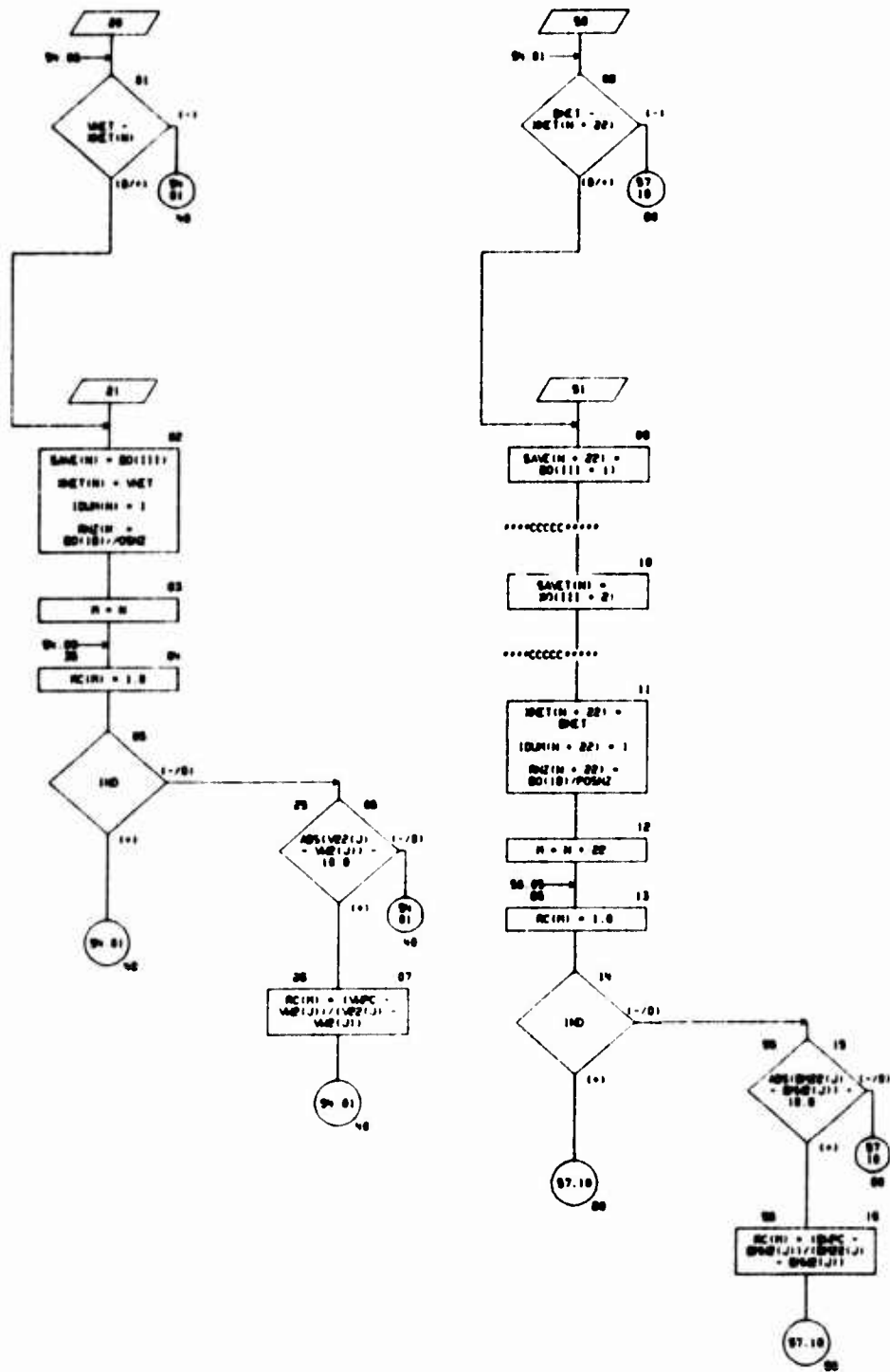


CHART TITLE - SUBROUTINE PARLO6

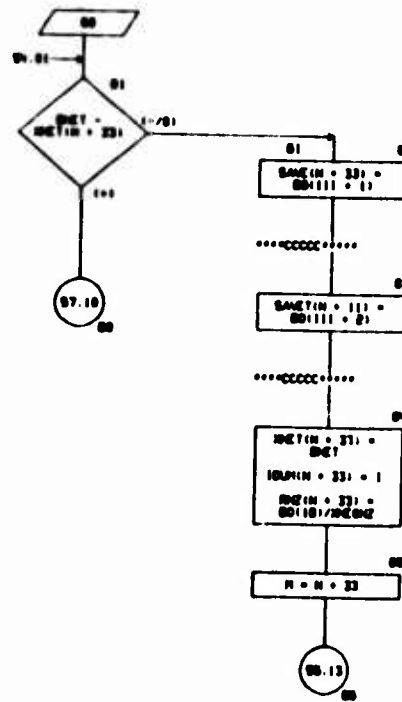


CHART TITLE - SUBROUTINE NAME DS

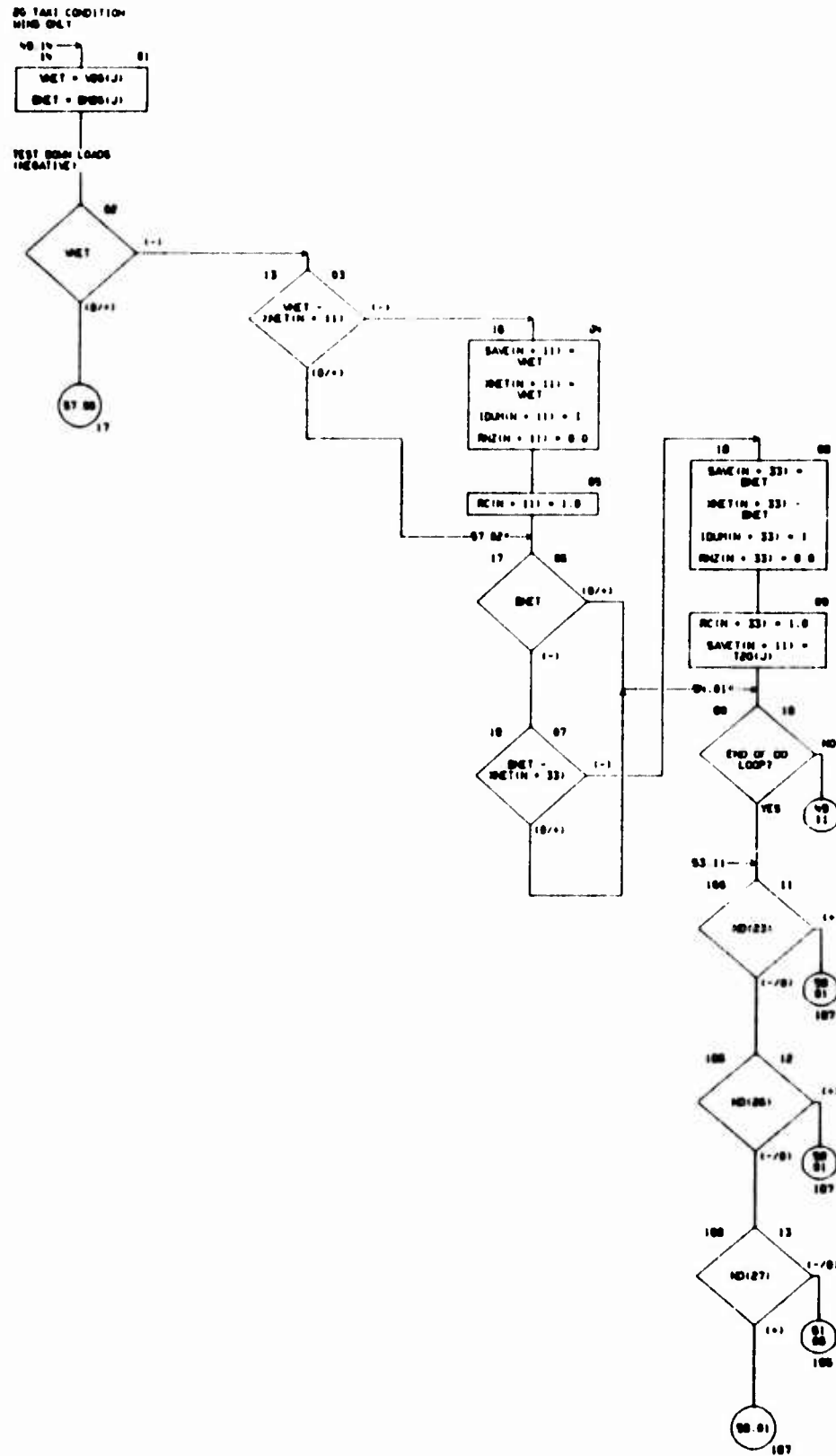
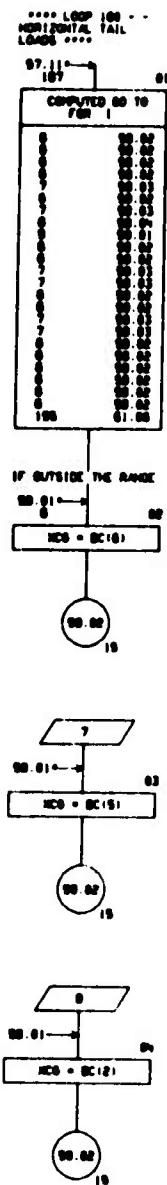


CHART TITLE - SUBROUTINE NAME(S)



[illegible]

CHART TITLE - SUBROUTINE MAIL00

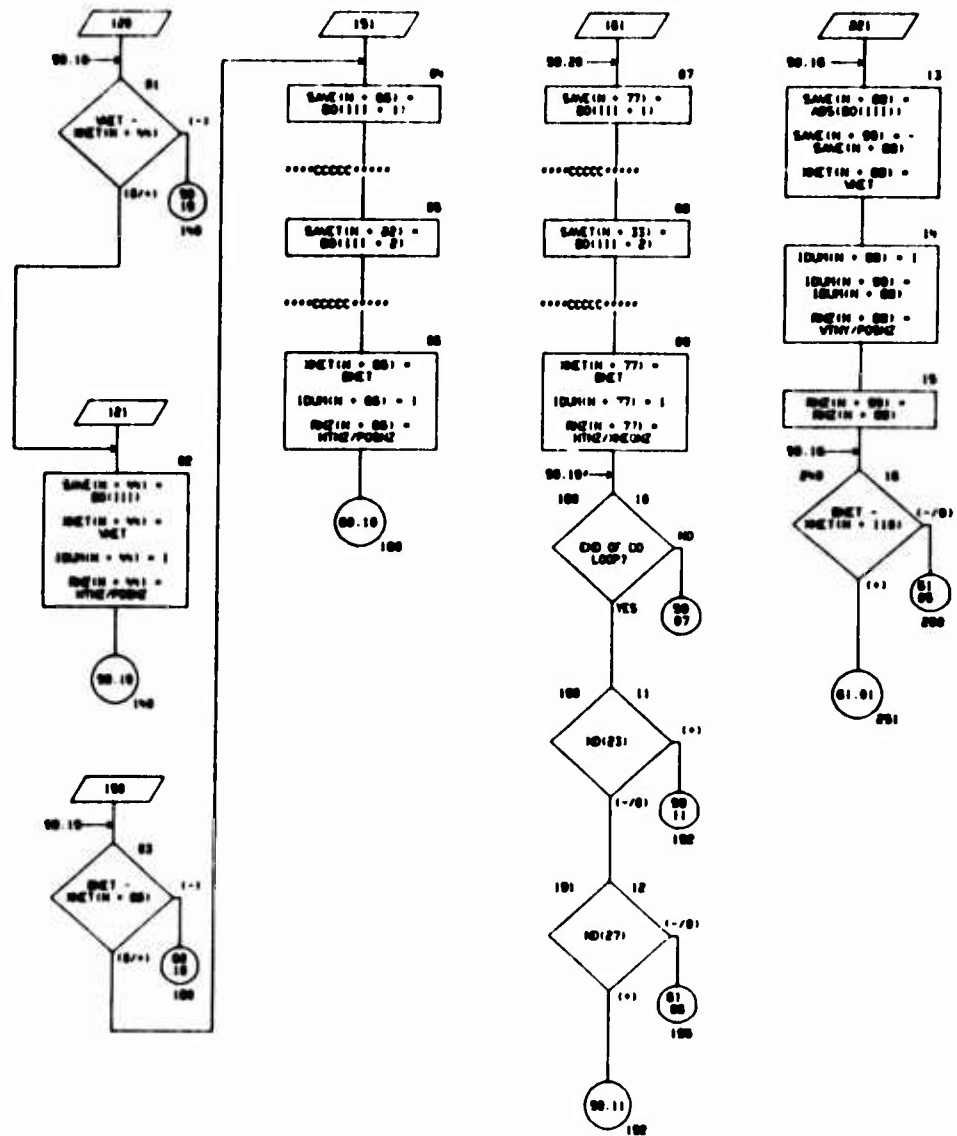


CHART TITLE - SUBROUTINE PARLOD

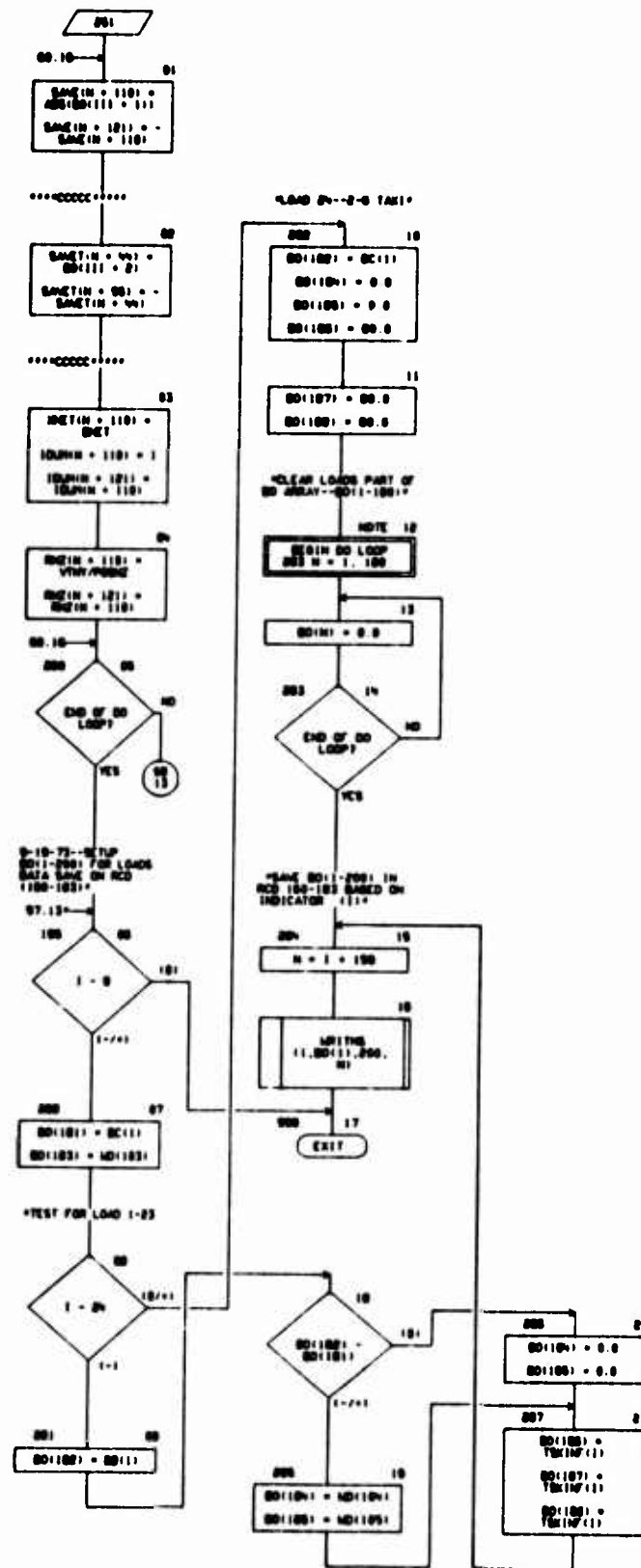


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

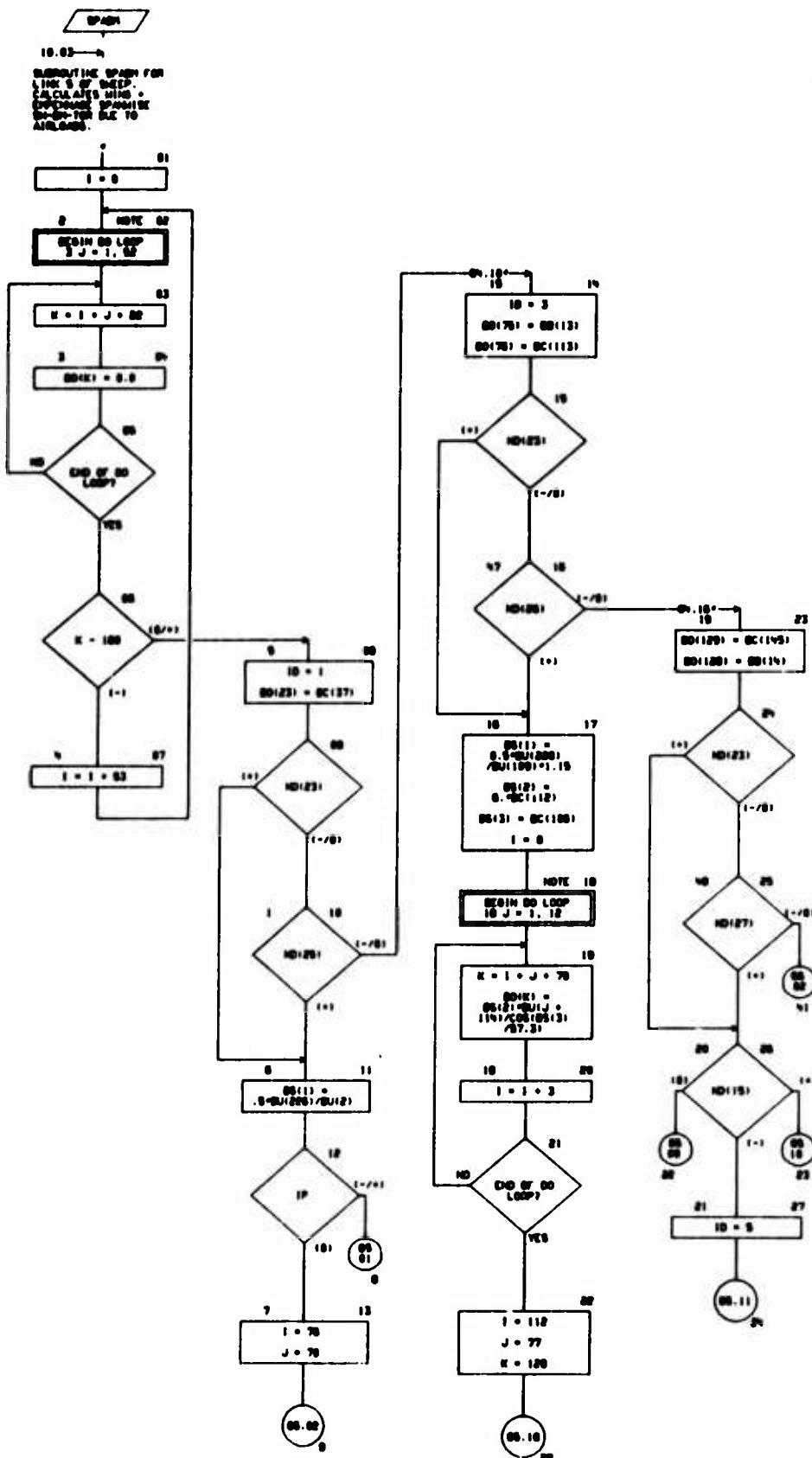
COMMON TCON(400),MLD(300),TDPM(132),SAC(132),HET(132),
      RHE(132),RC(44),SW(100),STEPH(23),STEPV(23),STEPV(23),
      ,RPH(80),SAC(100)
COMMON /MISC/ MISC(100)
DIMENSION MD(200),BD(620),TKINF(23),STRES(6),TEMP(6),
      ,STH(6),STV(6),STV(6),TEMPH(6),TEMPH(6),TEMPV(6)
DIMENSION BC(100)
DIMENSION BB(20)
DIMENSION MML(1024),MD(200)
DIMENSION VDS(11),DPS(11),TDS(11),WM(11),DPM(11),
      ,VDS(11),DPS(11),VDS(11),DPS(11),
      ,VDS(11),DPS(11),VDS(11),DPS(11),
      ,VDS(11),DPS(11),WM(11),DPM(11),
      ,WM(11),DPM(11)
EQUIVALENCE (MD(1),TCON(420)),(BD(1),TCON(330)),(1,MD(10))
EQUIVALENCE
      (SW(10),TKINF(1)),
      ,
      (SW(130),STH (1)),
      (SW(140),TEMP (1)),(SW(153),STH (1)),(SW(160),TEMP (1)),
      (SW(167),STV (1)),(SW(173),TEMPV (1)),(SW(181),MINS (1)),
      (SW(185),MGR (1)),(SW(170),MGR (1)),(SW(152),BDSH (1)),
      (SW(180),BDSH (1)),(SW(180),BDSH (1))
EQUIVALENCE (BDSH,MLD(1)), (PDH2,MLD(2)), (HDSH2,MLD(3)),
      , (VDS(1),MLD( 4)), (DPS(1),MLD( 5)), (TDS(1),MLD( 20)),
      , (WM(1),MLD( 27)), (DPM(1),MLD( 40)),
      , (VDS(1),MLD( 70)), (DPS(1),MLD( 81)),
      , (VDS(1),MLD(103)), (DPS(1),MLD(114)),
      , (VDS(1),MLD(130)), (DPS(1),MLD(147)),
      , (VDS(1),MLD(160)), (DPS(1),MLD(160)),
      , (VDS(1),MLD(202)), (DPS(1),MLD(213)),
      , (WM(1),MLD(270)), (DPM(1),MLD(240)),
      , (WM(1),MLD(280)), (DPM(1),MLD(270))
EQUIVALENCE (BC(1),TCON(270))
EQUIVALENCE (BB(1),TCON(2953))

```

CHART TITLE - INTRODUCTORY COMMENTS

SUBROUTINE SPAN
#####

CHART TITLE - SUBROUTINE SPASH



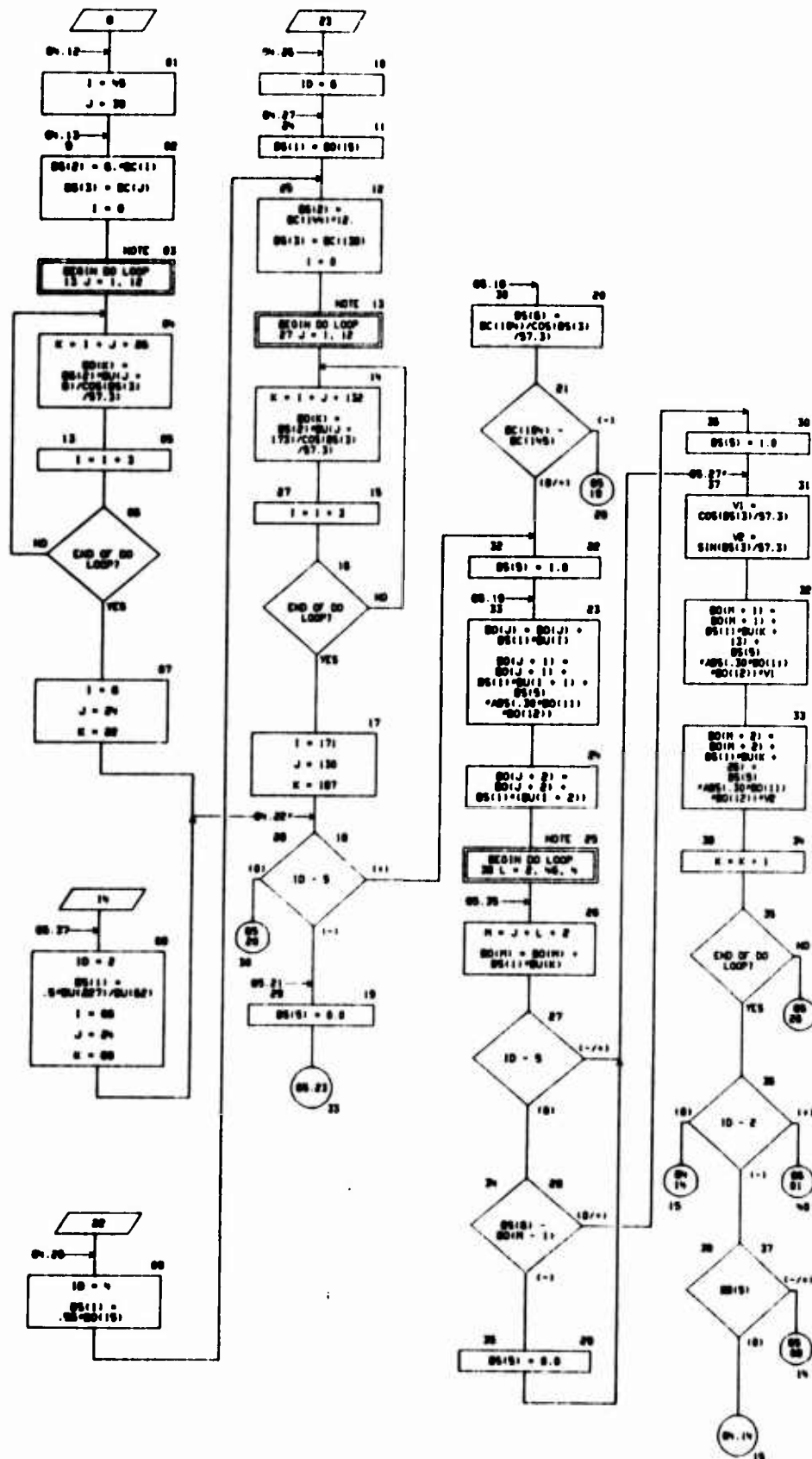


CHART TITLE - SUBROUTINE SPASH

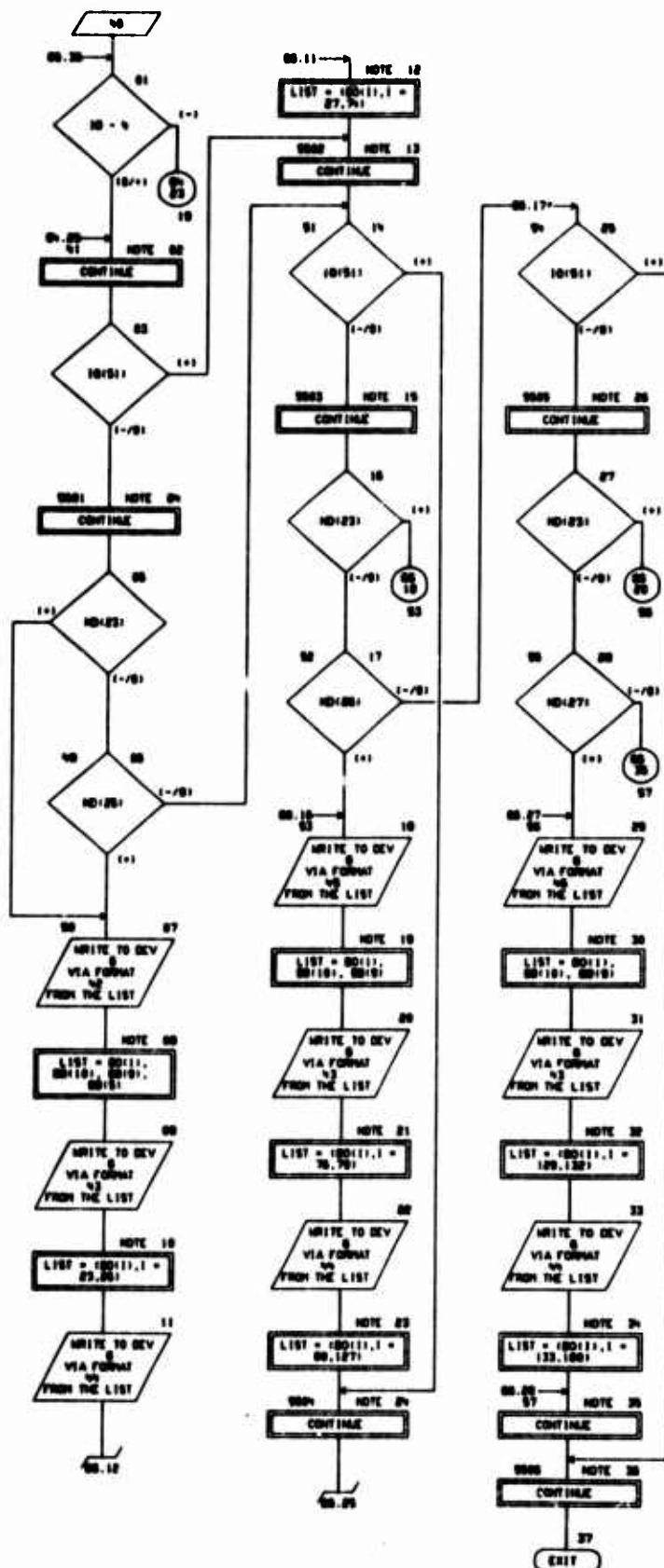


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(400)
COMMON /PRINT/IO(80)
DIMENSION BD(20),BC(100),BD(20),BS(20),BU(20),MD(200)
EQUIVALENCE (BC(1),TCON(275)),(BU(1),TCON(315)),(BD(1),TCON(330)
), (BS(1),TCON(297)),(BD(11),TCON(295)),(MD(1),TCON(420)),IP,MD
(137),110,MD(104),11,MD(100),12,MD(105),14,MD(107),15,MD(100)
),16,MD(100))
42  FORMAT(1H,3X,10HINS LOADS,3X,8COND NO=,F7.0,3X,300H,7B.3,
    3X,4HMT=,F7.0,3X,300H,7B.3,10H,20H** SPAN = (P(5)) **)
43  FORMAT(1H,3X,8COND DIST=,F7.0,3HIN,4X,10H AT SOB=,F7.0,2HLS,4X,
    10H AT SOB=,F12.0,3HIN-LS,4X,10H AT SOB=,F12.0,3HIN-LS///
    1X,10HSTATION,10H,10HSEAR,10H,10HEND ROM,21X,10HORS ROM/
    1X,10HIN,21X,10HLS),10X,10HIN-LS),20X,10HIN-LS//)
44  FORMAT(1H,3X,2H,0,F20.0,F20.0)
45  FORMAT(1H,3X,2HORIZONTAL TAIL LOADS,3X,8COND NO=,F7.0,3X,
    300H,7B.3,3X,4HMT=,F7.0,10H,20H** SPAN = (P(5)) **)
46  FORMAT(1H,3X,2HVERTICAL TAIL LOADS,3X,8COND NO=,F7.0,3X,
    300H,7B.3,3X,4HMT=,F7.0,21X,20H** SPAN = (P(5)) **)

```

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AUTOFLEX CHART SET - SHEEP AIRLOADS OVERLAY

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CHART TITLE - INTRODUCTORY COMMENTS

[illegible]

00.110-2

DISTRIBUTING JAPAN FOR
LION & SHEEP.
SERVES MIND AND
EXPOSURE UNIT
AIRBORNE DISTRIBUTION
AND FACTORS.

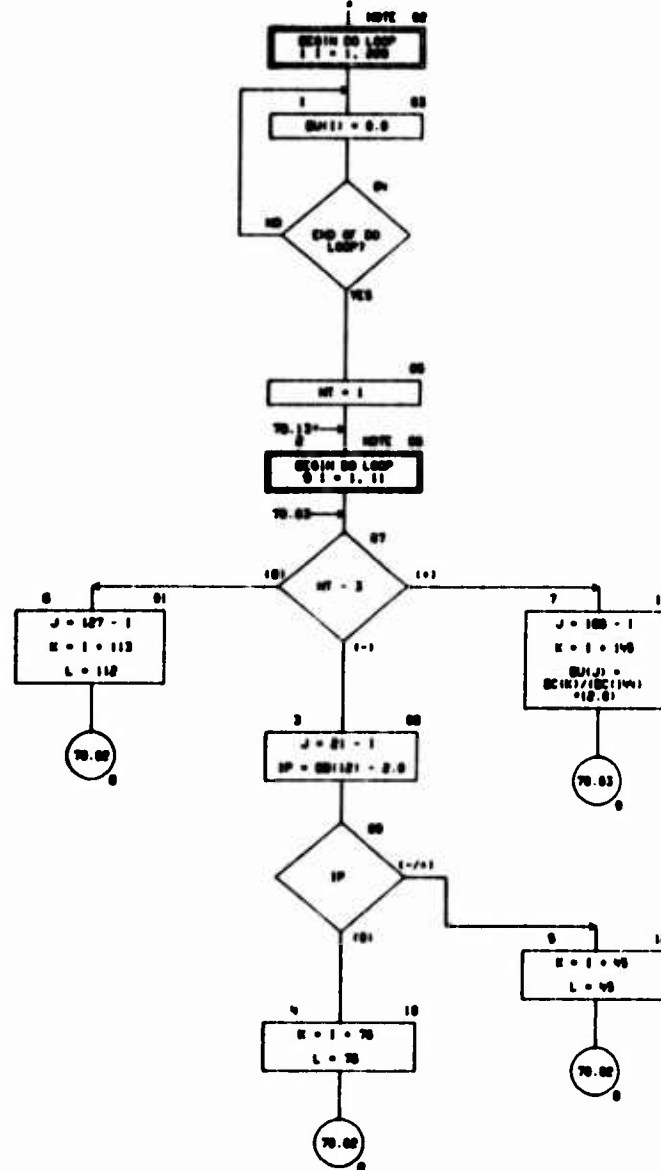


CHART TITLE - SUBROUTINE UPGRADE

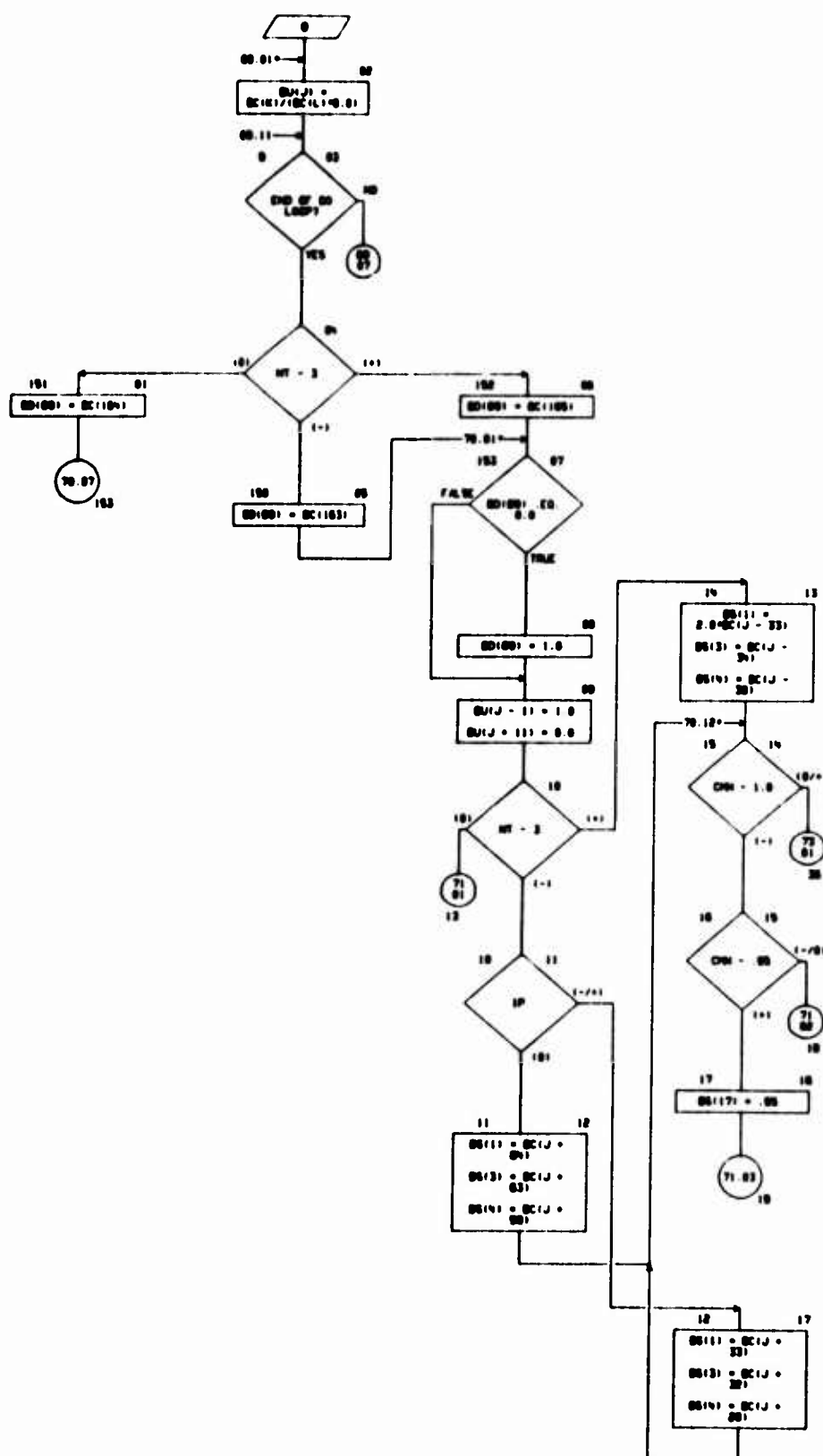


CHART TITLE - SUBMITTING AGENCY

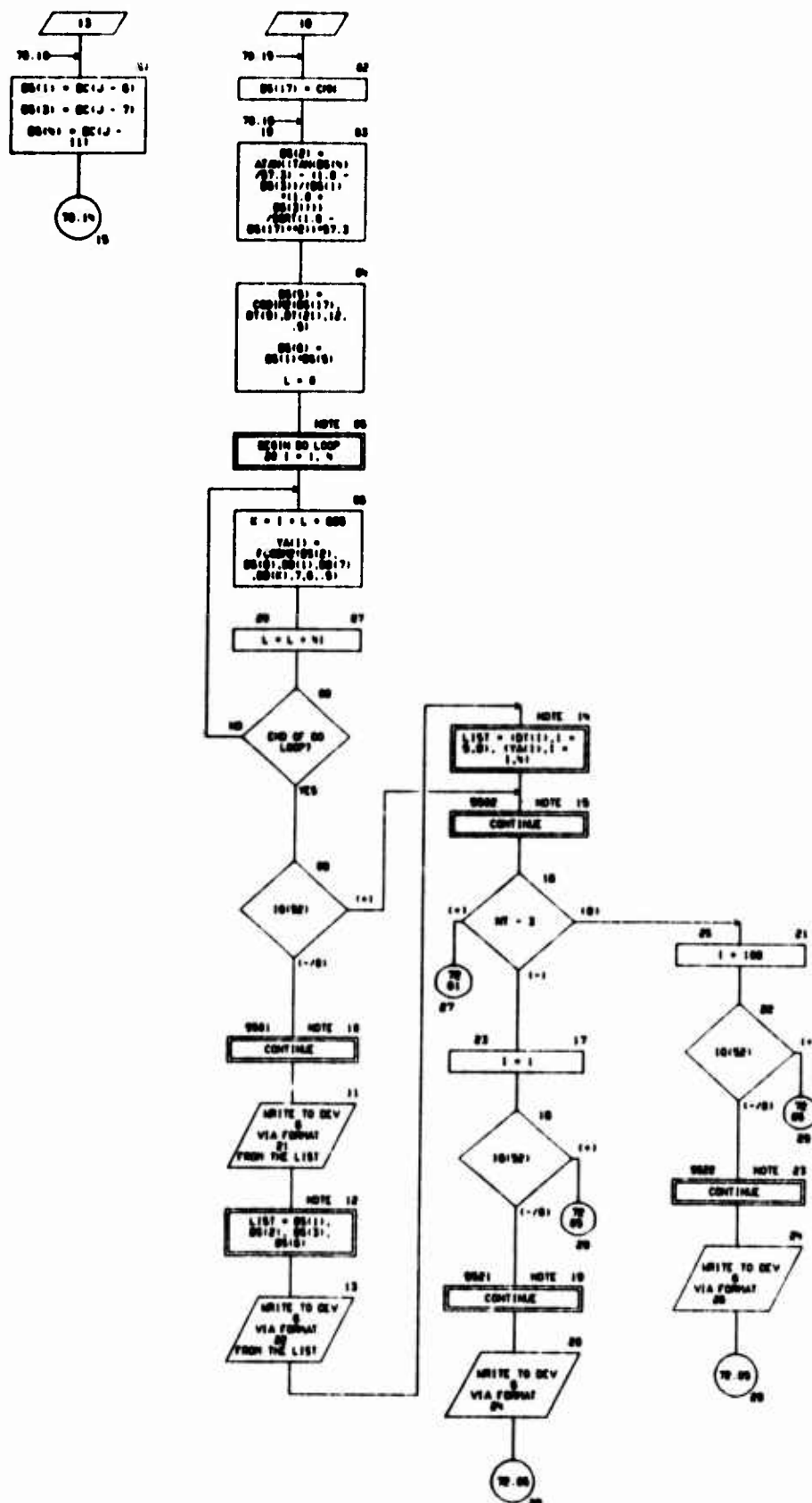


CHART TITLE - SUBROUTINE UPGRA



CHART TITLE - SUBROUTINE USPHN

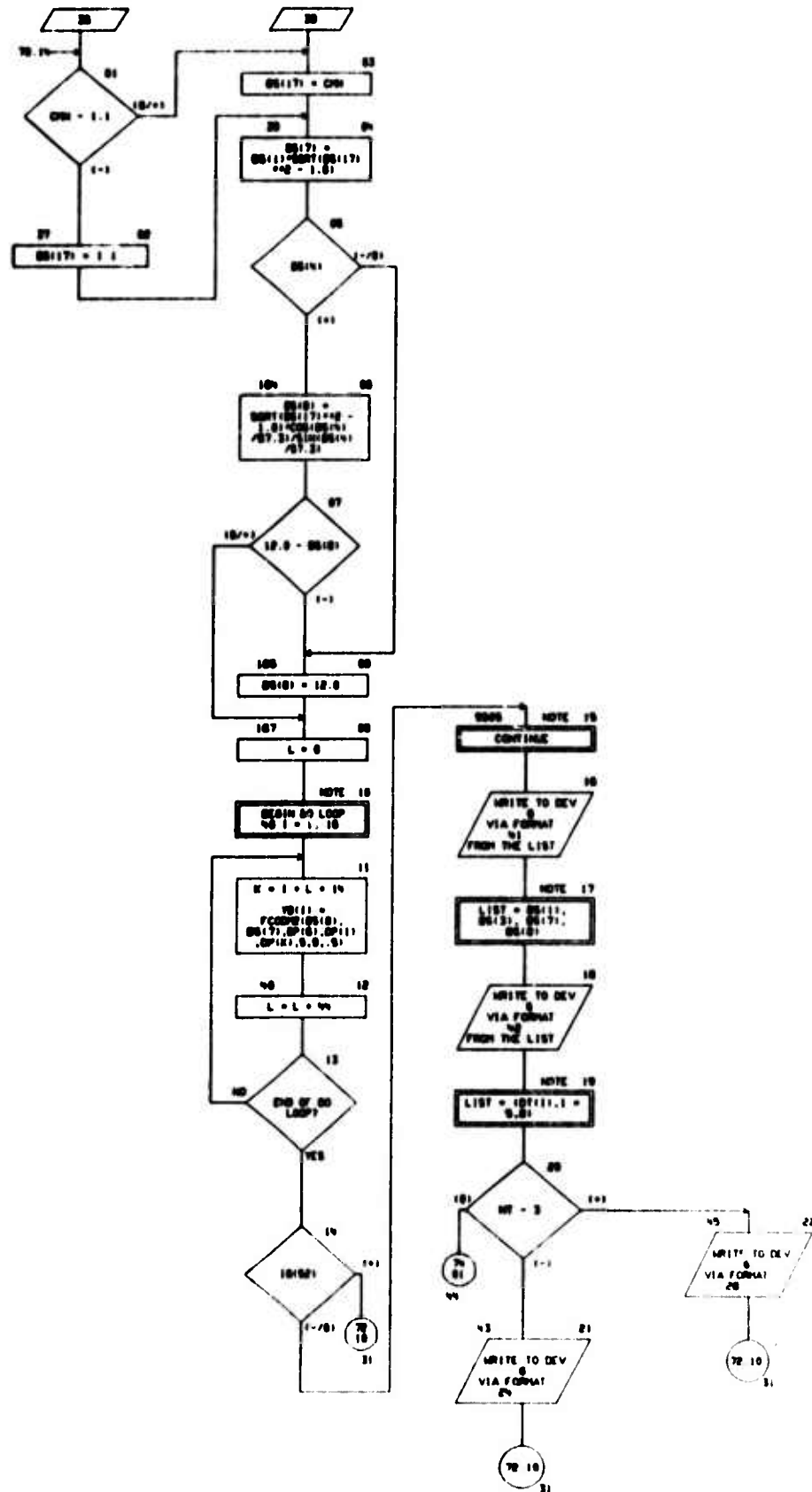


CHART TITLE - SUBROUTINE USPM

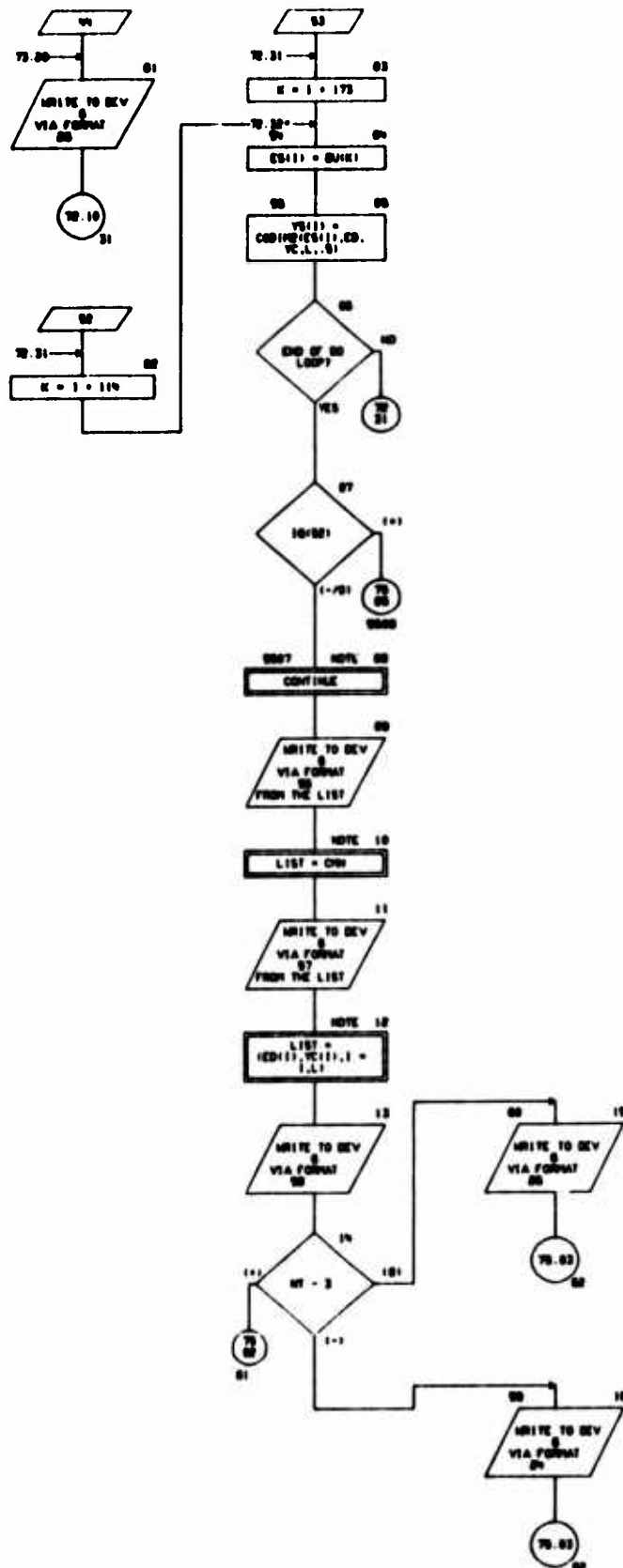


CHART TITLE - SUBROUTINE USPM

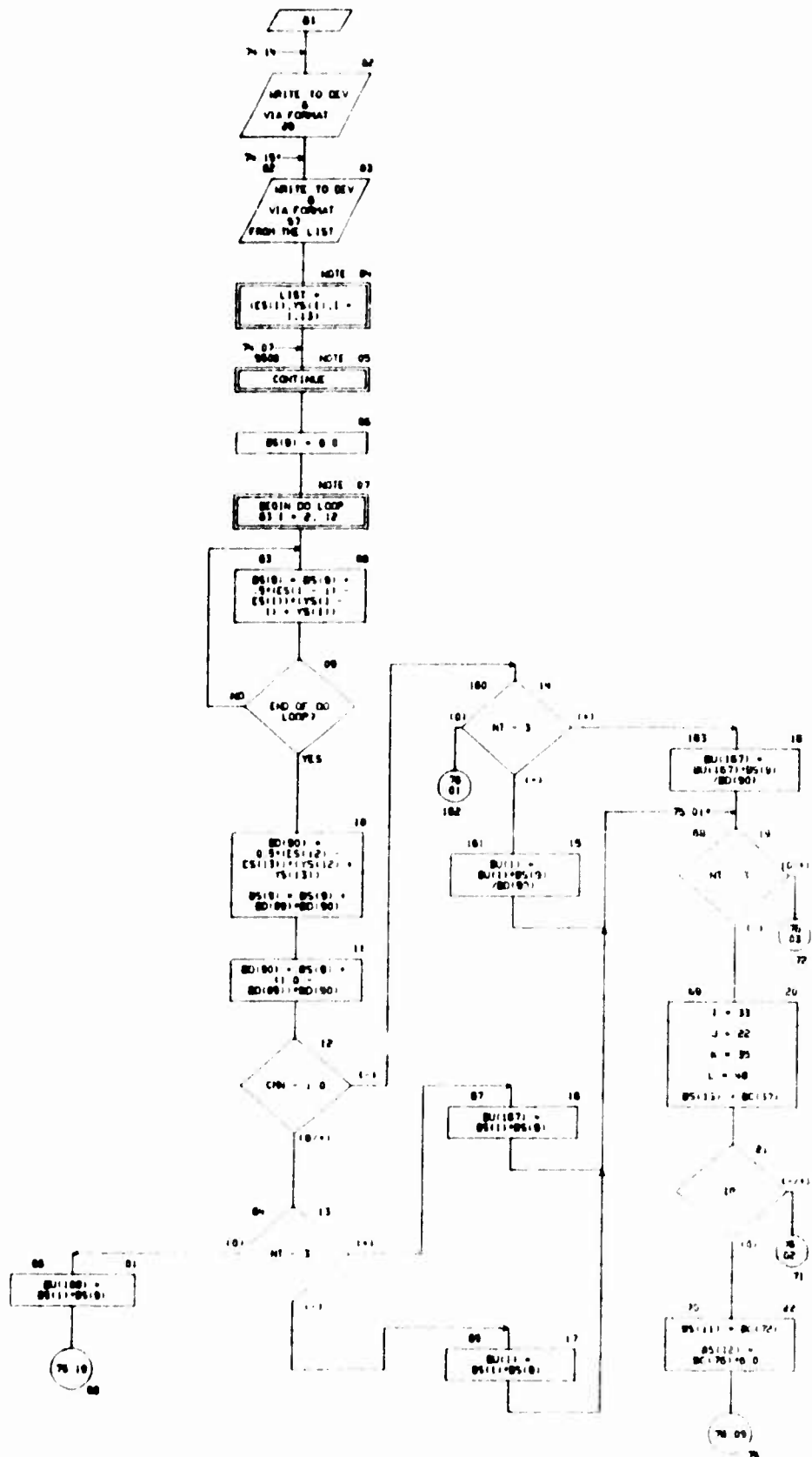


CHART TITLE - SUBSISTING UPRON

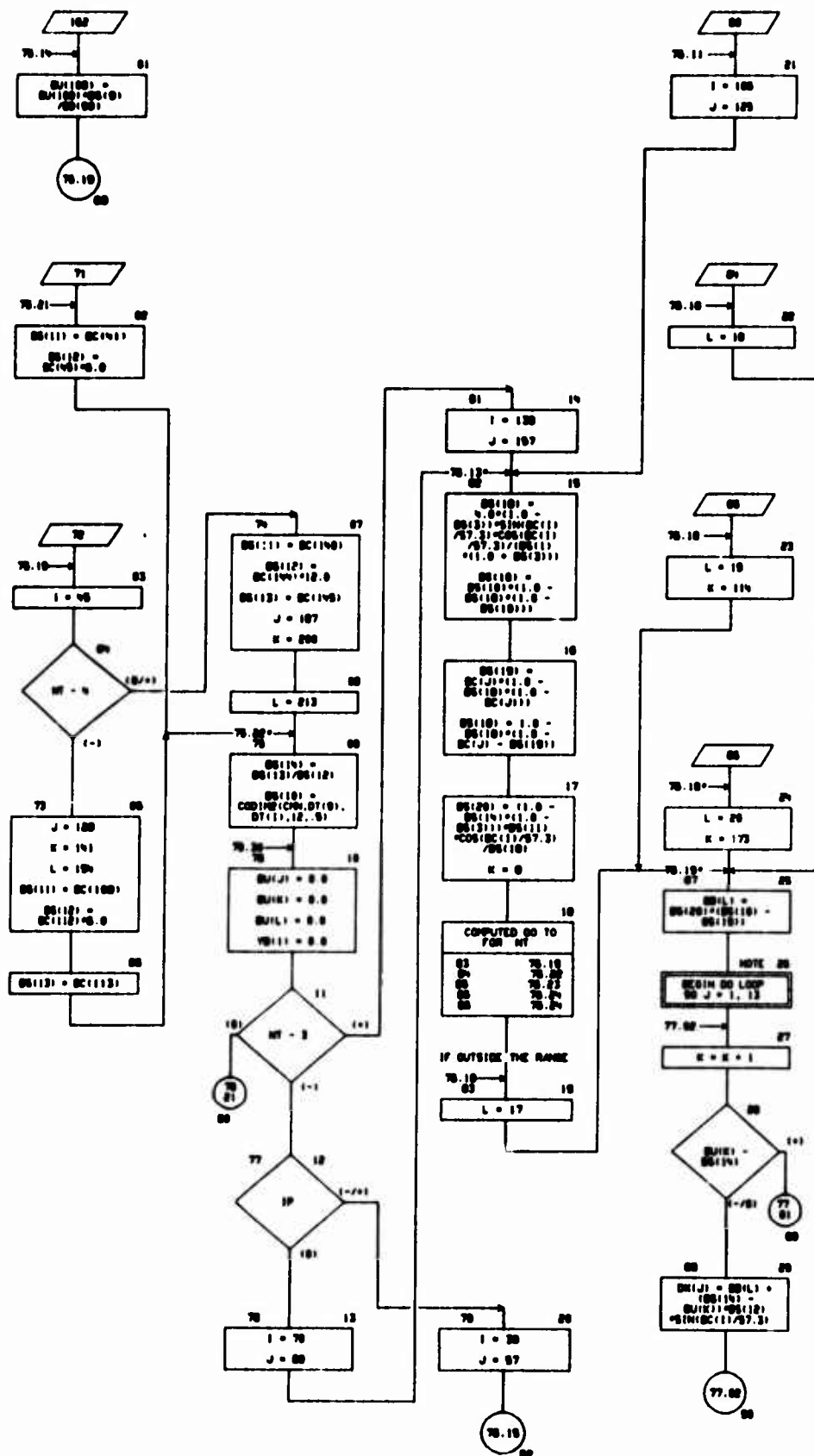


CHART TITLE - SUBROUTINE USPAH

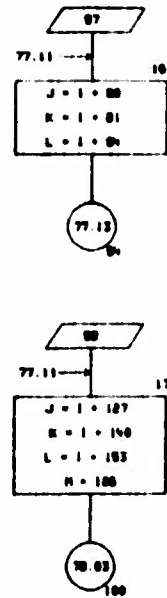
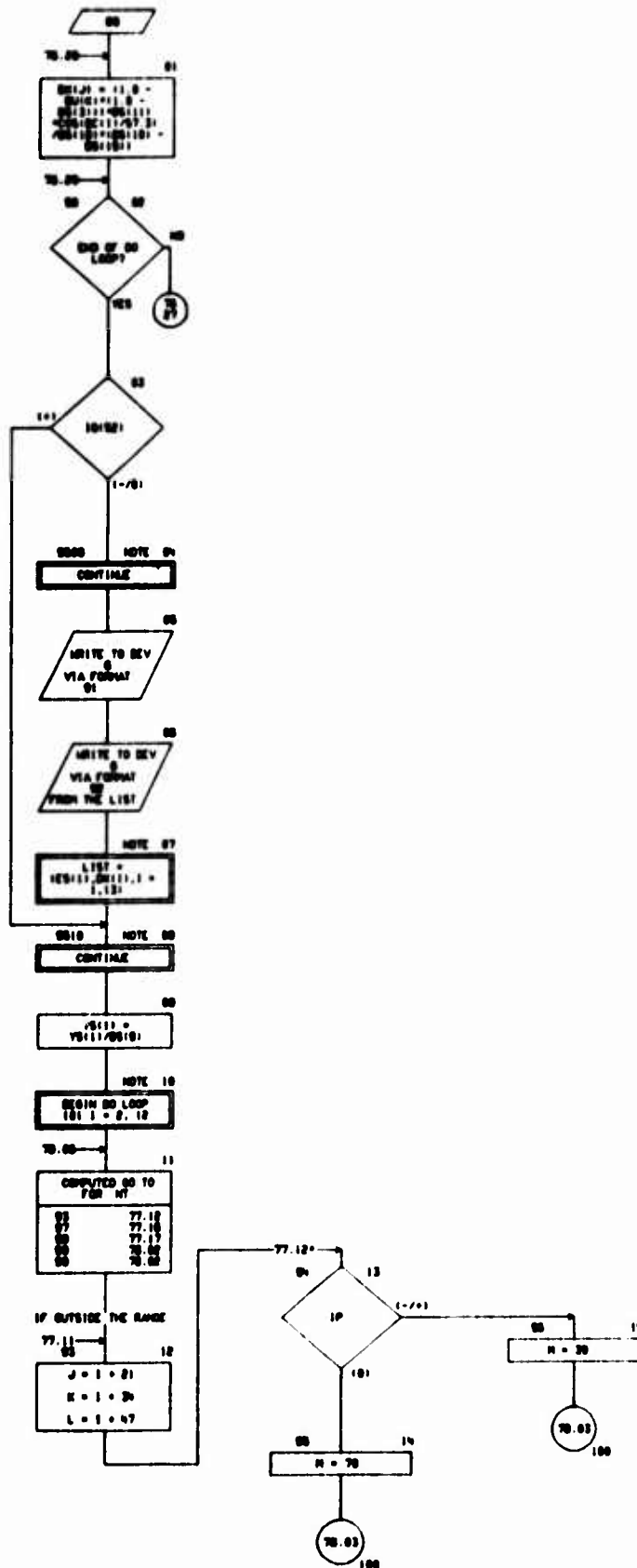


CHART TITLE - SUBROUTINE USPAN

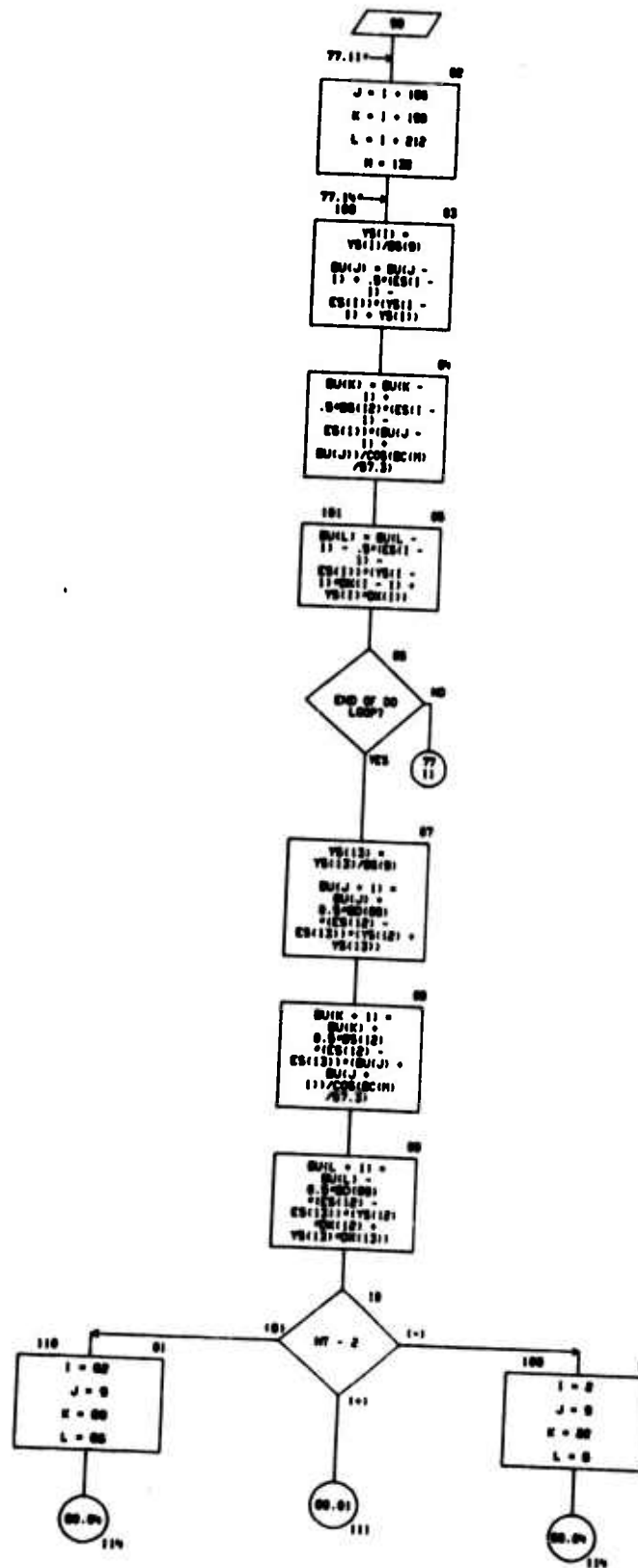


CHART TITLE - SUBROUTINE USPAN

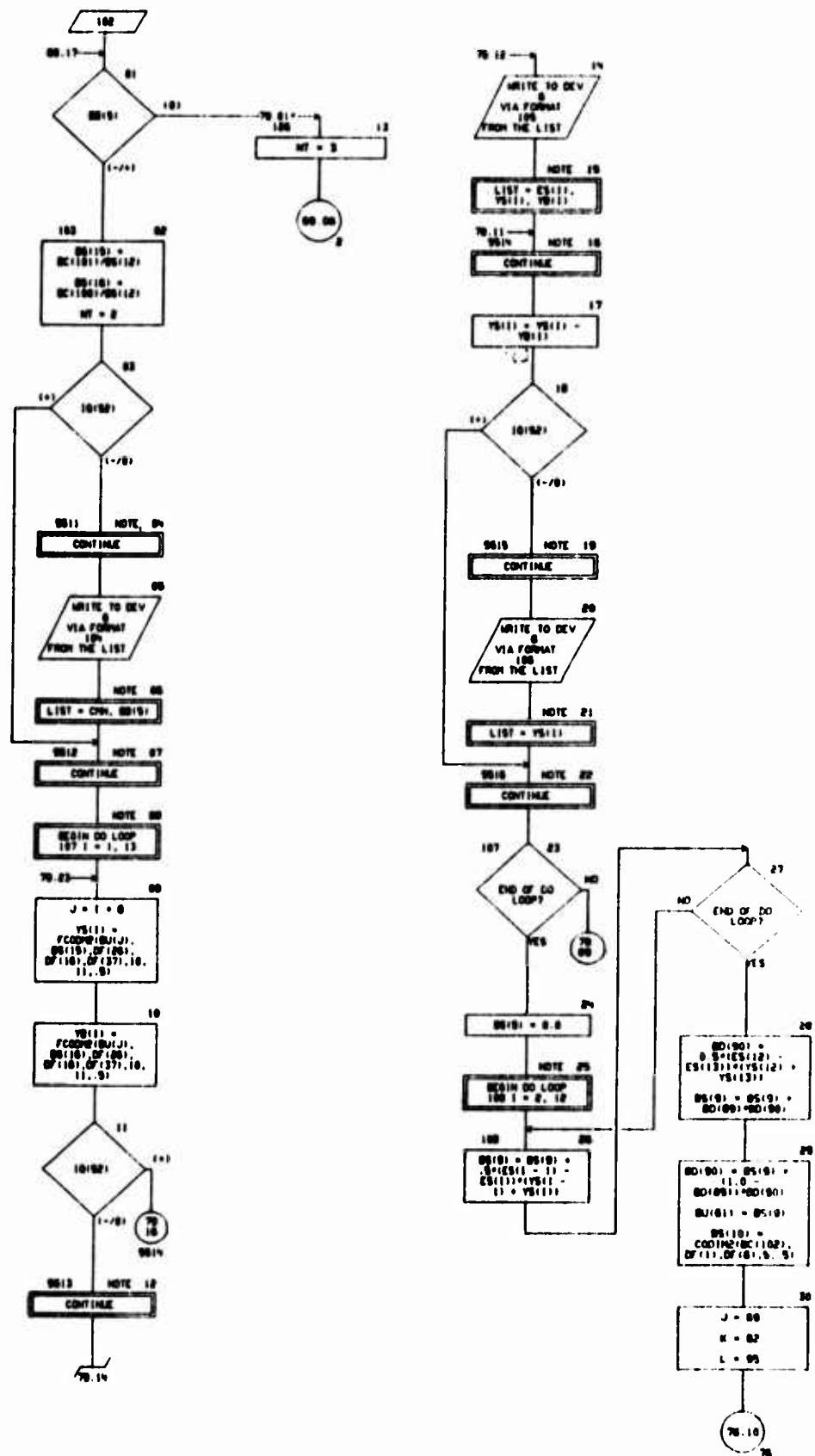


CHART TITLE - SUBROUTINE URPAN

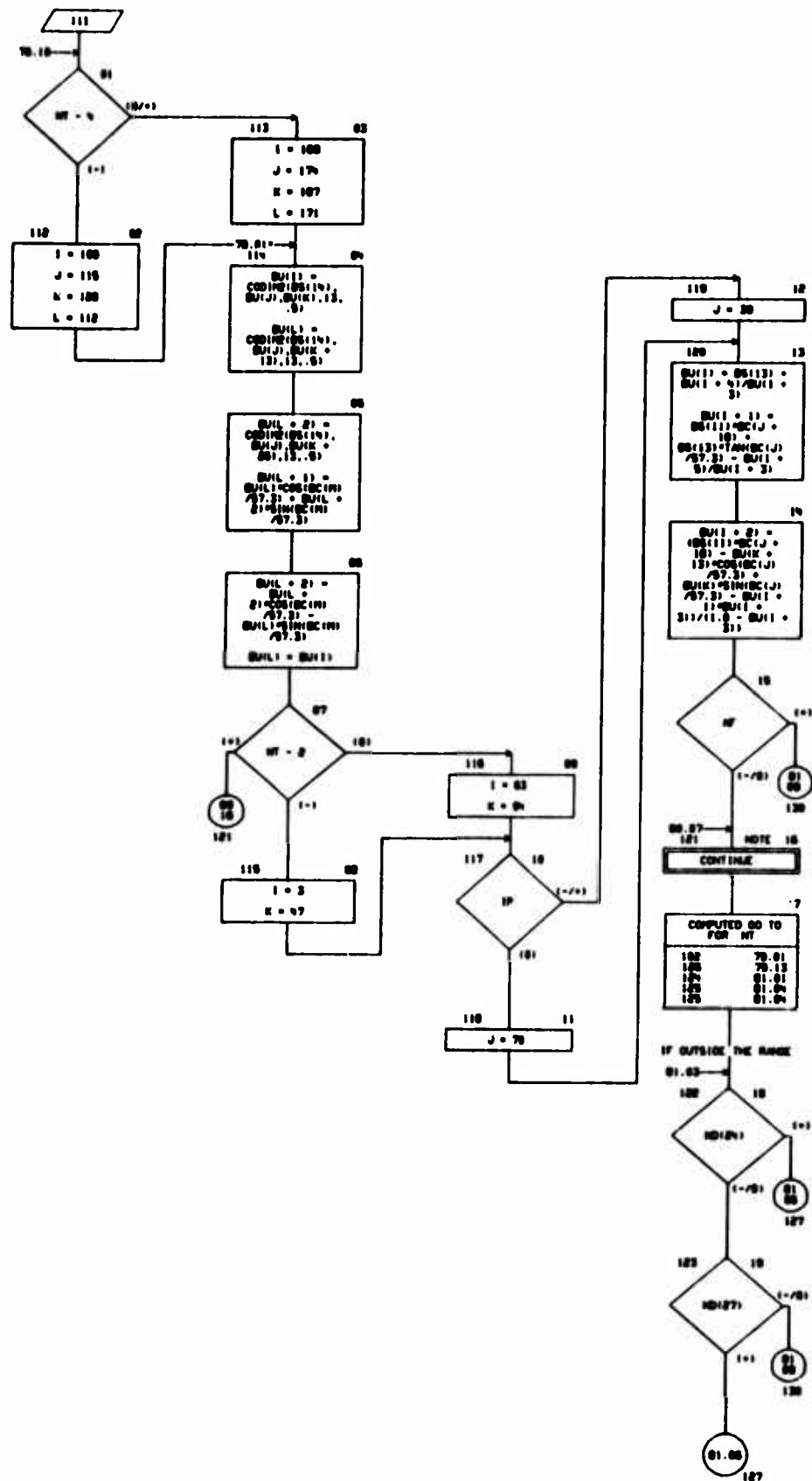


CHART TITLE - SUBROUTINE JAPAN

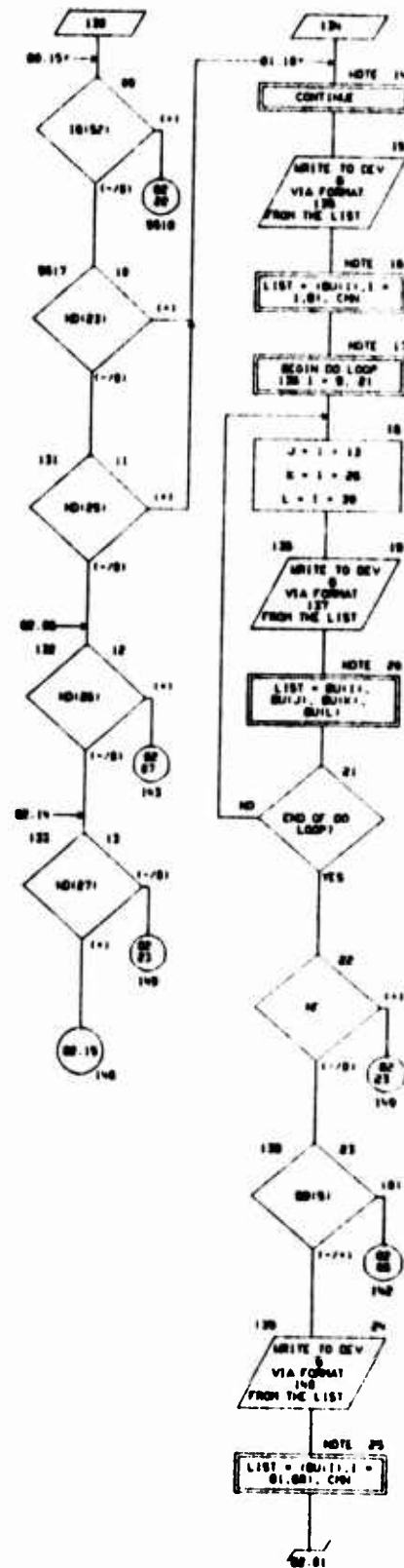
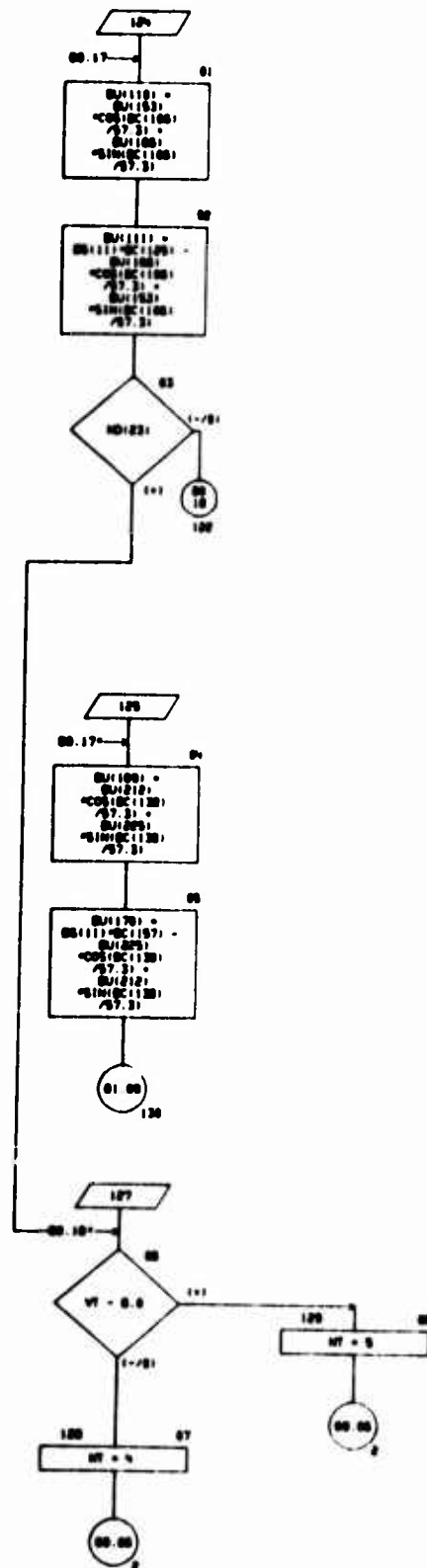


CHART TITLE - SUBROUTINE WSPAN

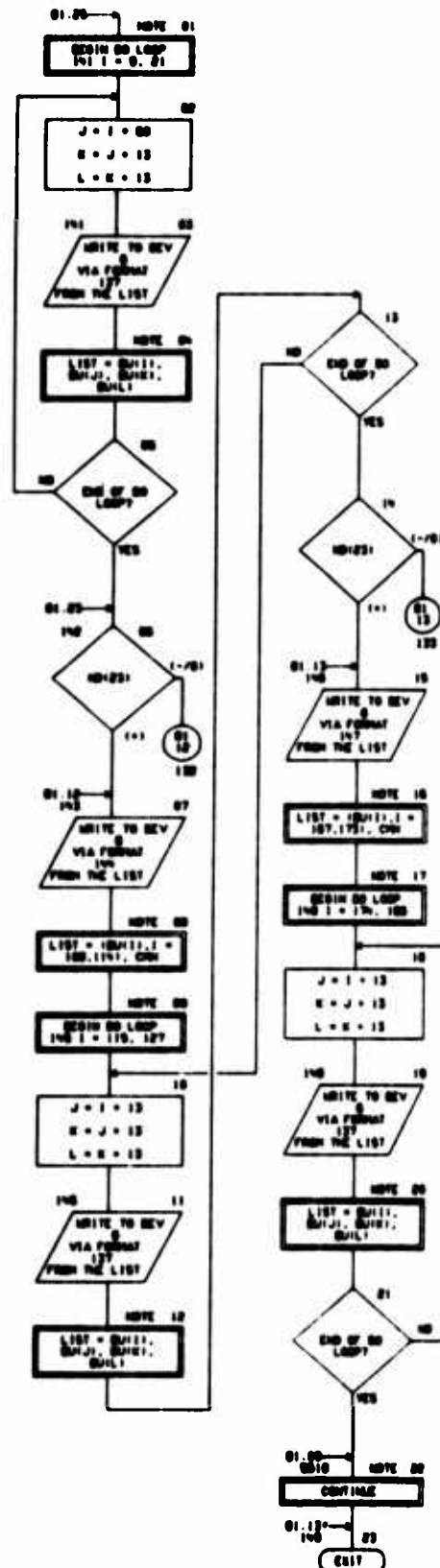


CHART TITLE - INTRODUCTORY COMMENTS

.....
SUBROUTINE WMMET
.....

CHART TITLE - SUBROUTINE WAPNE1

WAPNE1

10 00 - n

SUBROUTINE TO ARRANGE
SURFACE AIRLOADS DATA
AND
NORMALIZING FACTORS

**** READ RECORD 37
THE WIND, HORIZONTAL
AND VERTICAL
LOADS ****

01

READS
(1, 01, 17, 100,
37)

**** READ RECORD 17
THE DEAD WEIGHT
RATIOS ****

(INITIALIZED AT 1.0
IN READ) ****

02

READS
(1, 02, 17, 100,
17)

LEARN LOADS WHICH ARE
TO BE COMPUTED

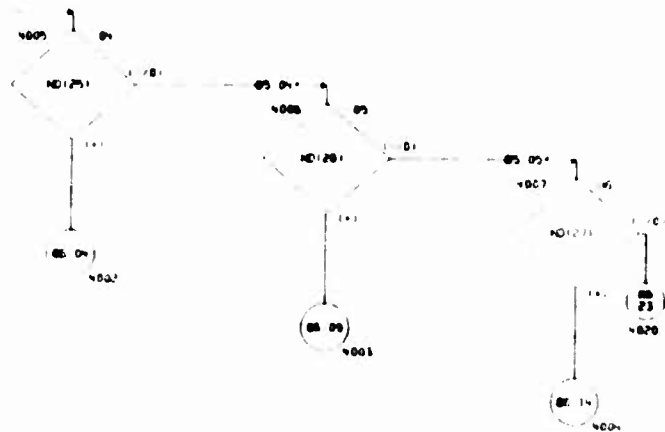
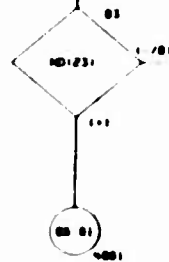
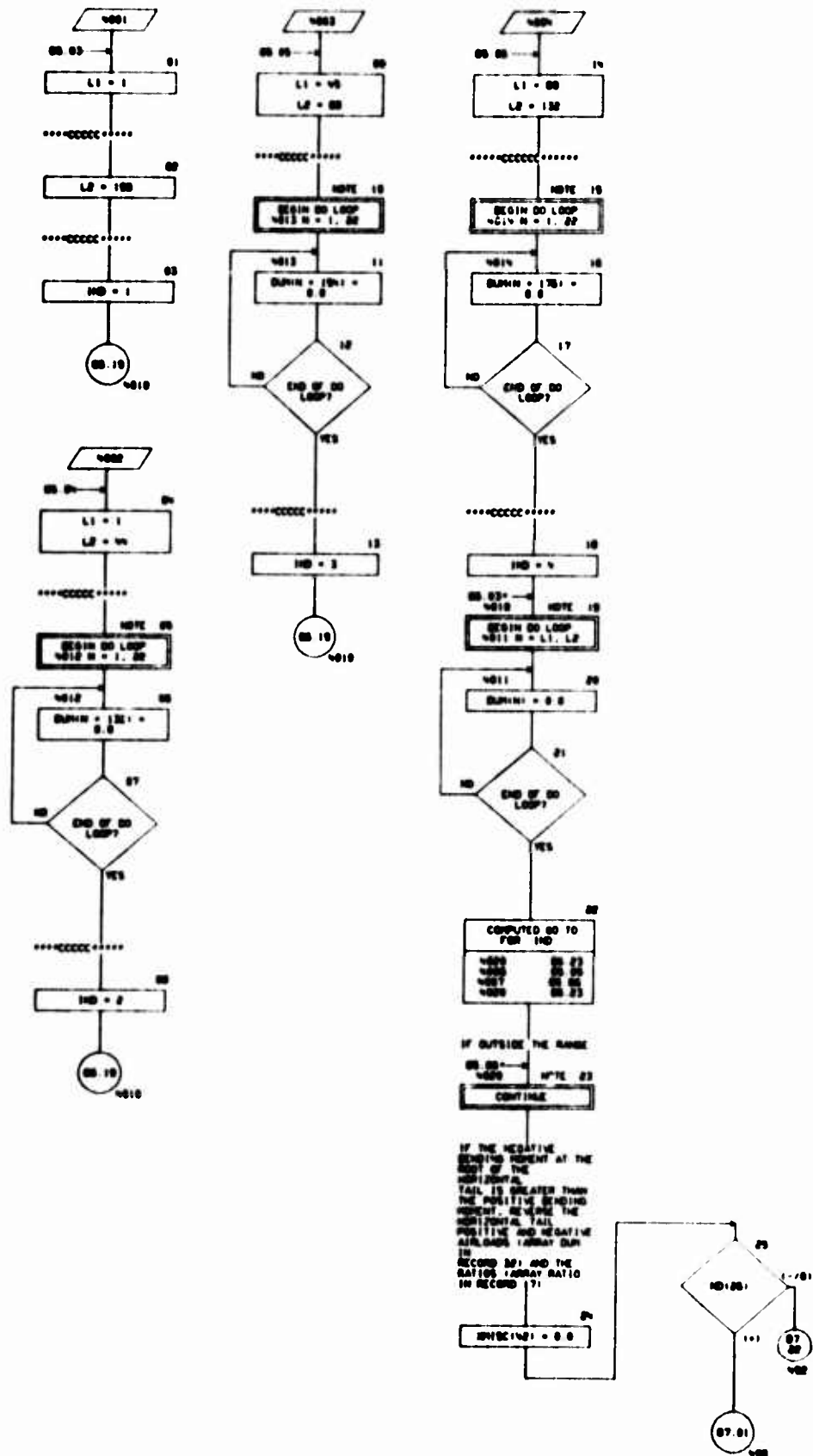


CHART TITLE - SUBROUTINE WANDT



[illegible]

CHART TITLE - SUBROUTINE 000001

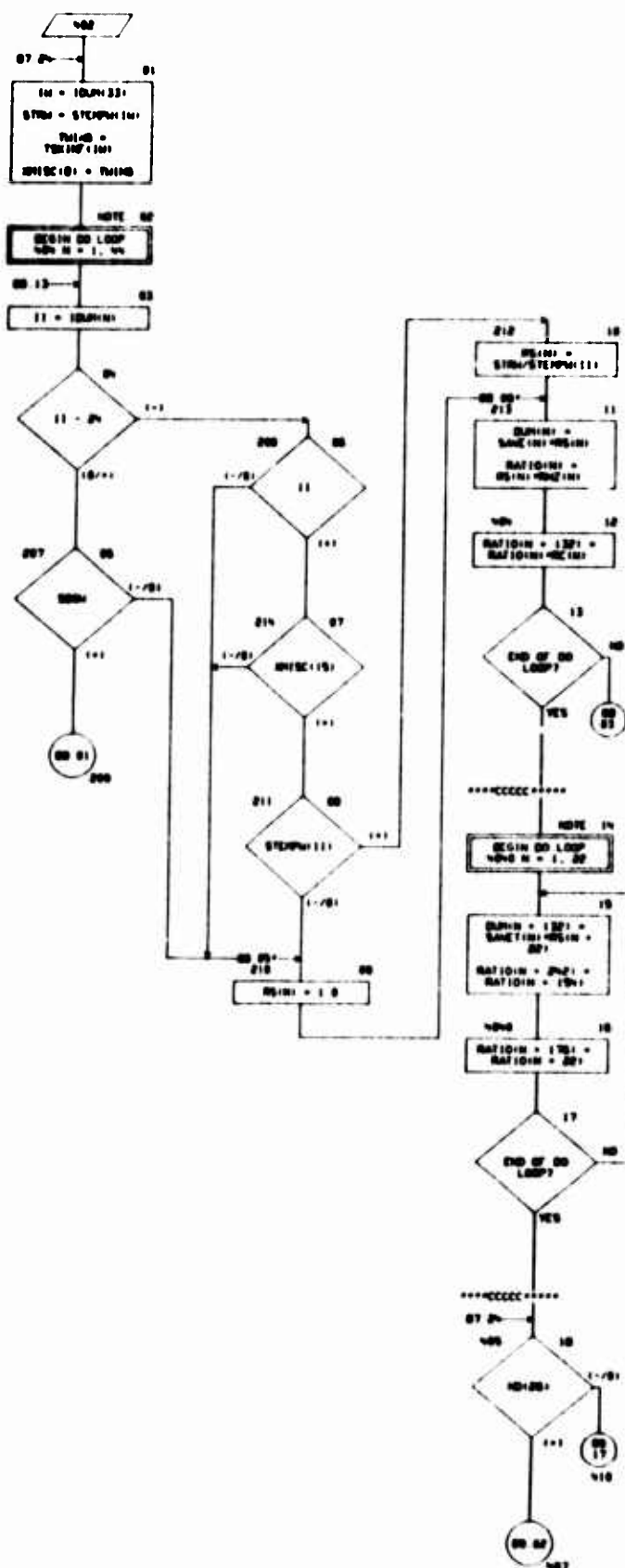


CHART TITLE - SUBROUTINE NAME 1

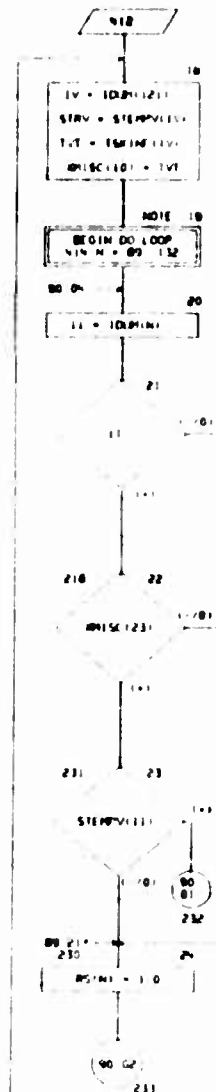
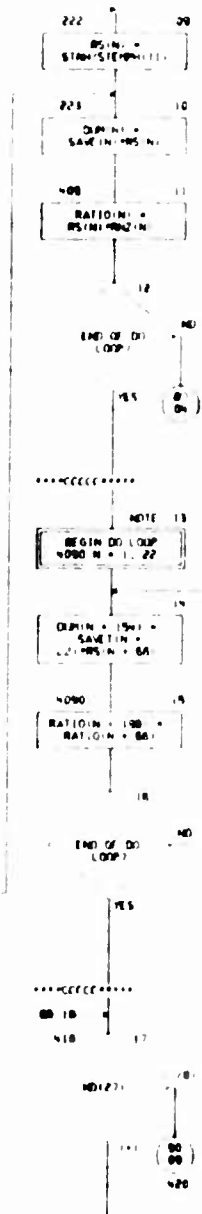
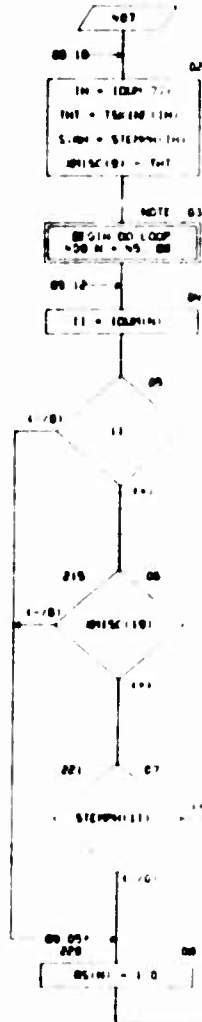


CHART TITLE - SUBROUTINE 100A01

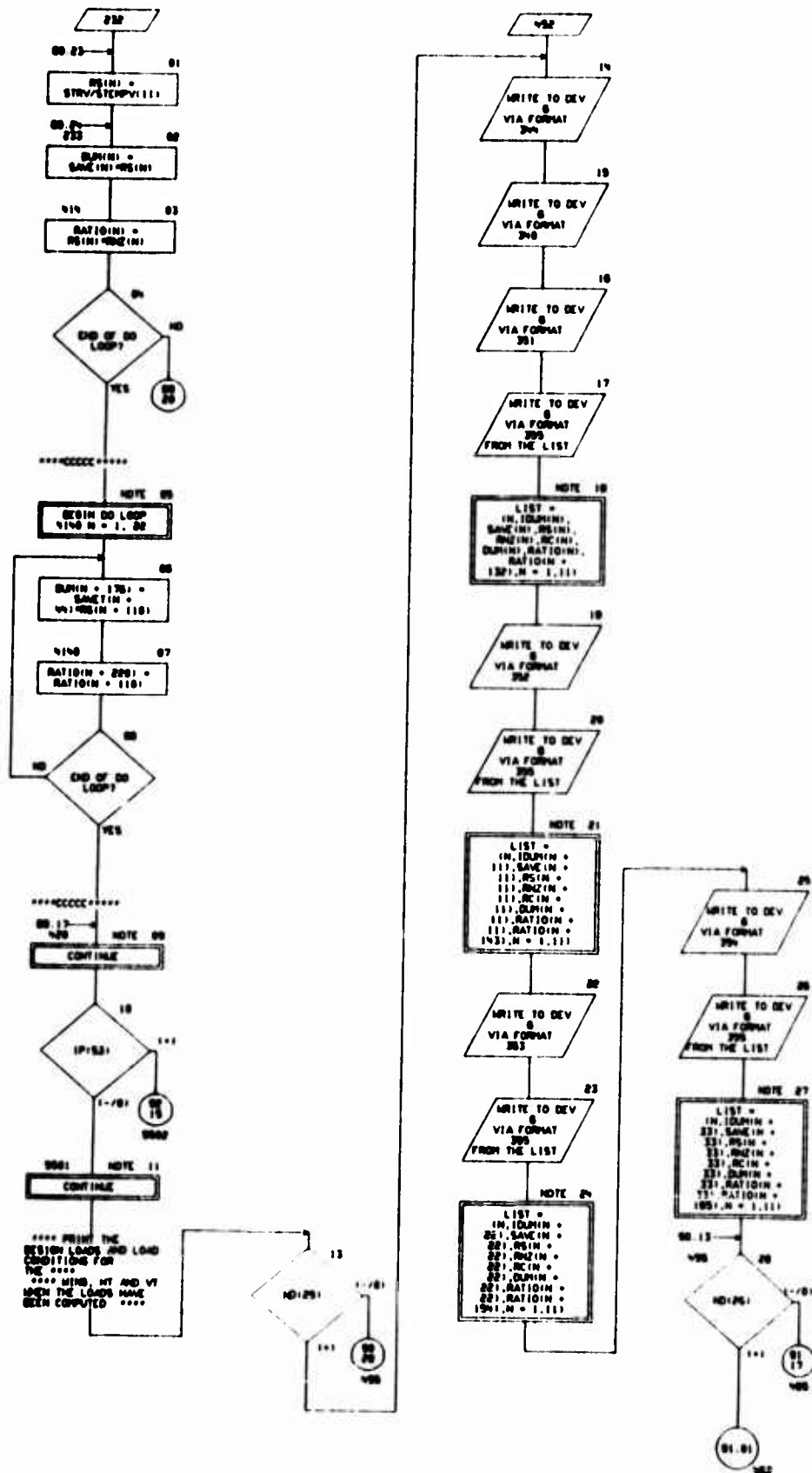


CHART TITLE - SUBROUTINE M0001

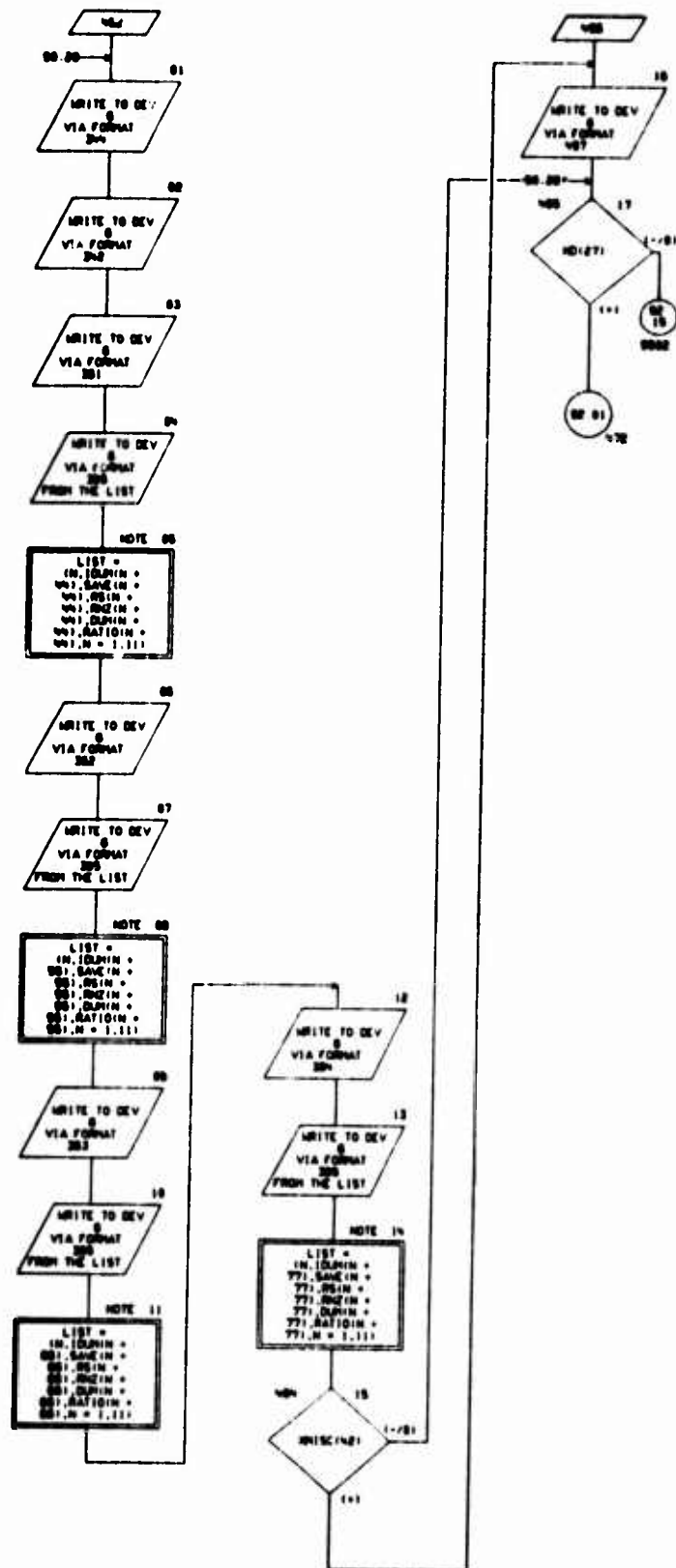


CHART TITLE - SUBROUTINE WMMET

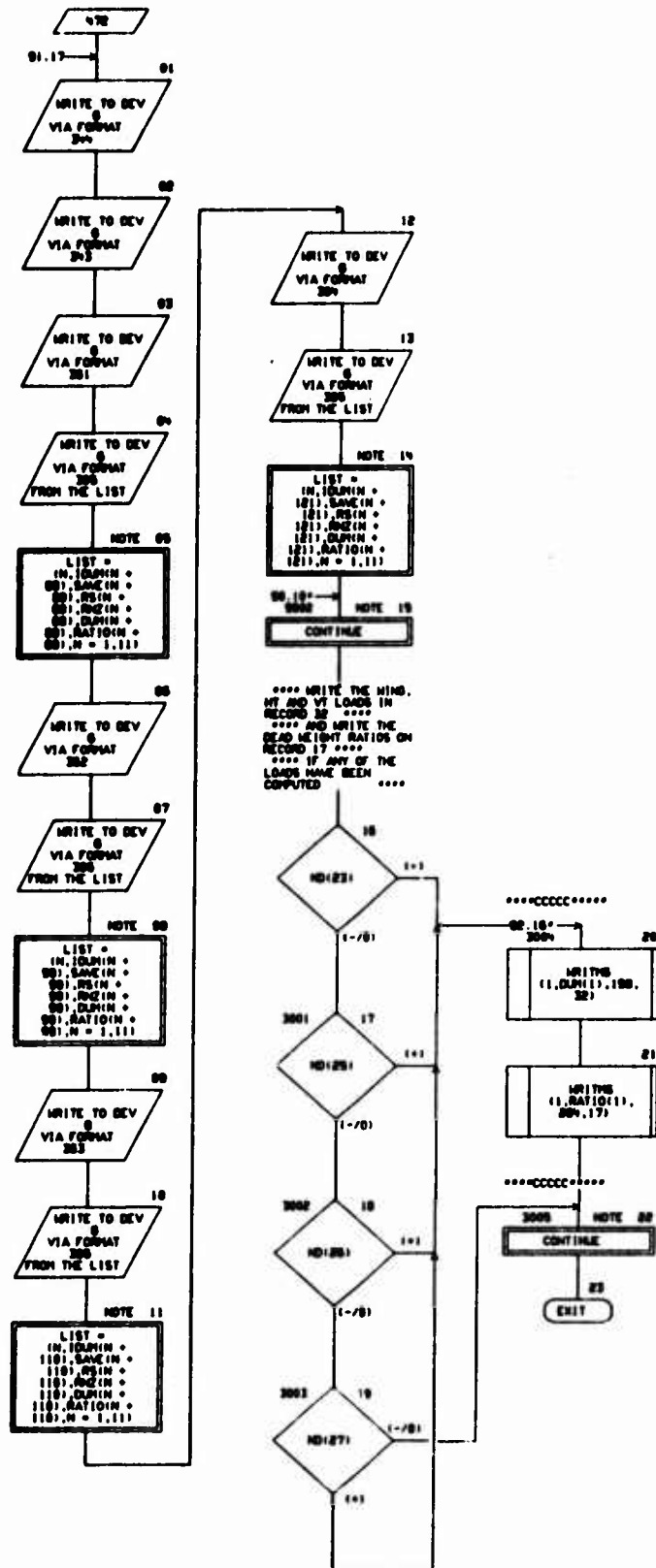


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(400),L.D(300),LOUP(132),SAC(132),DET(132),
RUE(132),RC(144),SW(100),STEPH(23),STEPV(23),STEPV(23),
LUN(100),LANE(100)
COMMON /IPRINT/ IP(80)
COMMON /MISC/ MISC(100)
DIMENSION RATIO(204),RS(132)
DIMENSION MD(200)
DIMENSION TSKINF(23)
EQUIVALENCE (MD(1),TCON(420))
EQUIVALENCE (SW(116),TSKINF(1)),(SW(152),SDBM)
344 FORMAT(1H,BSX,BDM*** DESIGN LOADS (RECORD 32) AND RATIO (RECORD
17) ***,BX,BDM** MPMET - (P(63) **))
345 FORMAT(4BX,1BX* * * * M I N U S * * *)
351 FORMAT(//BSX,BDMSTATION LC +V RS RUE RC
RS(1+V) RS+RUE RS+RUE*RC/)
355 FORMAT(4X,12,5X,12,F13.0,F8.3,F8.3,F13.0,F8.3,F8.3)
352 FORMAT(//BSX,BDMSTATION LC -V RS RUE RC
RS(1-V) RS+RUE RS+RUE*RC/)
353 FORMAT(//BSX,BDMSTATION LC +BM RS RUE RC
RS(1+BM) RS+RUE RS+RUE*RC/)
354 FORMAT(//BSX,BDMSTATION LC -BM RS RUE RC
RS(1-BM) RS+RUE RS+RUE*RC/)
342 FORMAT(2BX,2BX*** HORIZONTAL TAIL ***)
351 FORMAT(//BSX,BDMSTATION LC +V RS RUE RS(1+
V) RS+RUE/)
355 FORMAT(4X,12,5X,12,F13.0,F8.3,F8.3,F13.0,F8.3)
352 FORMAT(//BSX,BDMSTATION LC -V RS RUE RS(1-
V) RS+RUE/)
353 FORMAT(//BSX,BDMSTATION LC +BM RS RUE RS(1+
BM) RS+RUE/)
354 FORMAT(//BSX,BDMSTATION LC -BM RS RUE RS(1-
BM) RS+RUE/)
407 FORMAT(//17X,7BTHE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL T
AIL HAVE BEEN REVERSED//14X,8BDBECAUSE THE NEGATIVE BM AT THE ROOT
WAS GREATER THAN THE POSITIVE BM AT THE ROOT)
343 FORMAT(2BX,2BX*** VERTICAL TAIL ***)

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MODULE FORTRAN LISTING

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07/03/74      INPUT LISTING      AUTOFLON CHART SET - SHEEP      AIRLOADS OVERLAY
FORTRAN MEDIA  (LIST, AUTOGEN)

CARD NO      ****      CONTENTS      ****

1              PROGRAM BLCHTL
2              C
3              C      PROGRAM BLCHTL IS LINK 4 OF SHEEP.      BLCH0001
4              C
5              C      DETERMINES BASIC AIRLOAD CONDITIONS TO BE COMPUTED ON/AND WIND      BLCH0002
6              C      DETERMINES PRESENT FATIGUE SPECTRA      BLCH0003
7              C      PROVIDES LOGIC AND CONTROL FOR THE AIRLOAD SUBROUTINES      BLCH0004
8              C
9              C      COMMON TCON(400),ALD(300),IDUM(132),SAME(132),INET(132),
10             *   SUE(132),RC(44),SWF(100),STEPW(23),STEPW(23),STEPV(23),
11             *   BLN(100)
12             *   SANE(100)
13             C*****C*****
14             C
15             C      COMMON /IPRINT/ IP(00)
16             C      COMMON /MISC/ MISC(100)
17             C
18             C      DIMENSION MD(200),BD(670),BC(100),BD(20),DT(50),DD(50),DF(140),DP(140),DP(140)
19             C      (170)      BLCH0011
20             C
21             C      DIMENSION PS(23),TLOCAL(23),TTOTAL(23),WFLUX(23),TSKINH(23)
22             C
23             C      DIMENSION IMACH(23),ALT(23),TSKINH(23)
24             C
25             C      EQUIVALENCE (MD(1),TCON(400)),(BD(1),TCON(330)),(BC(1),TCON(270)),(BLCH0020
26             *   01),((BD(1),TCON(200)),(1,MD(10))),
27             *   2   (IN,MD(100)),(IN,MD(100)),(IN,MD(100)),(DT(1),TCON(100)),(DD(1)
28             *   3),TCON(100)),(DF(1),TCON(100)),(DP(1),TCON(110))      BLCH0023
29             C
30             C      EQUIVALENCE (ALT(1),BD(201)),(IMACH(1),BD(241))
31             C
32             C      EQUIVALENCE (SWF(10),TSKINH(1))
33             C      EQUIVALENCE (SWF( 1),PS( 1)),(SWF( 2),TLOCAL(1)),
34             *   (SWF( 4),TTOTAL(1)),(SWF( 70),WFLUX(1)),(SWF( 83),TSKINH(1))
35             C
36             C      REMIND 24
37             C
38             C*****C*****
39             C      BUFFER IN(24),((TCON(1),TCON(500)))
40             C*****C*****
41             C
42             C      IF (UNIT(24)=1010,1010,1010)
43             C
44             C      **** CLEAR ARRAYS ****
45             C
46             C      1010 DO 700 N=1,23
47             C          IMACH(N) = 0.0
48             C          ALT(N) = 0.0
49             C          STEPW(N) = 0.0
50             C          STEPV(N) = 0.0
51             C          700 STEPV(N) = 0.0
52             C
53             C      DO 3 I=1,620      BLCH0006
54             C          3 BD(I)=0.0      BLCH0005
55             C
56             C      DO 1 N=1,132
57             C          IDUM(N) = 0
58             C          INET(N) = 1.0
59             C          SAME(N) = 0.0
60             C          1 INET(N) = 0.0
61             C
62             C      DO 4 N=1,44
63             C          4 RC(N) = 1.0
64             C      **** READ RECORDS 1-4, LOADS PERMANENT DATA TABLES ****
65             C      CALL READP(1,DT(1),50,1)
66             C      CALL READP(1,DD(1),50,2)
67             C      CALL READP(1,DF(1),140,3)
68             C      CALL READP(1,DP(1),70,4)
69             C
70             C      *** SET UP MD(13-30), THE LOADS CONTROL INDICATORS ***

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07-03-74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADDS OVERLAY
CARD NO	*****	CONTENTS	*****
71	C		
72	ND(13) = ND(BC(19))		
73	ND(14) = ND(BC(19))		
74	ND(15) = ND(BC(19))		
75	ND(16) = ND(BC(19))		
76	DO 9 N=23,30		
77	9 ND(14) = ND(BC(19))		
78	C		
79	IF(ND(23)197.97.98		
80	80 ND(24) = 1		
81	ND(25) = 1		
82	ND(26) = 1		
83	ND(27) = 1		
84	87 CONTINUE		
85	C		
86	ND(30) = ND(BC(100))		
87	C		
88	C **** READ RECORD 27, THE LOADS VARIABLE DATA ****		
89	CALL READ(11,BC(1),100,27)		
90	C		
91	C **** READ RECORD 31, THE STRESS, TEMPERATURE, WF AND G DATA ****		
92	CALL READ(11,94(1),100,31)		
93	C		
94	C **** READ RECORD 18, THE DEAD WEIGHT DATA (FROM OFATHS) ****		
95	CALL READ(11,MD(1),300,18)		
96	C		
97	DO 15 N=14,300,11		
98	IF(ABS(MD(N)) - 1.0)16,15,15		
99	16 MD(N) = 0.0		
100	15 CONTINUE		
101	C		
102	IF(ND(16)130,124,30		ELCH222
103	30 DO 21 I=100,131		ELCH224
104	21 ND(11)=0		ELCH226
105	C		
106	IN=0		ELCH228
107	22 DO 67 I=1,8		ELCH230
108	IF(ND(1027)167.67.73		ELCH232
109	23 IF(11124,24,25		ELCH234
110	24 IN=1		ELCH236
111	25 DO 10 (26,35,36,40,41,52,61,64,65),1		ELCH238
112	26 IF(ND(14)120,20,27		ELCH239
113	27 ND(113)=1		ELCH240
114	28 ND(106)=1		ELCH242
115	IF(BC(167)-BC(100)130,20,29		ELCH244
116	29 ND(118)=1		ELCH246
117	30 TO 31		ELCH248
118	31 ND(111)=1		ELCH249
119	31 IF(BC(100) - 1.0) 67,33,33		ELCH250
120	33 ND(112)=1		ELCH252
121	34 TO 67		ELCH254
122	35 IF(ND(14)137.37.38		ELCH256
123	36 ND(115)=1		ELCH258
124	37 ND(114)=1		ELCH260
125	38 TO 67		ELCH262
126	39 ND(116)=1		ELCH264
127	40 TO 67		ELCH266
128	40 ND(117)=1		ELCH268
129	41 TO 67		ELCH270
130	41 IF(ND(14)144,44,43		ELCH272
131	43 ND(120)=1		ELCH274
132	ND(121)=1		ELCH276
133	44 ND(118)=1		ELCH278
134	ND(119)=1		ELCH280
135	45 TO 67		ELCH282
136	52 IF(ND(14)195.95.94		ELCH284
137	94 ND(124)=1		ELCH286
138	ND(125)=1		ELCH288
139	95 ND(122)=1		ELCH290
140	ND(123)=1		ELCH292
141	96 TO 67		ELCH294

402

07/00/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOAD OVERLAY
CARD NO	*****	CONTENTS	*****
004	GO TO 100		ELCH1200
005	110 MI=0		ELCH1200
006	GO TO 100		ELCH1270
007	100 MI=7		ELCH1275
008	GO TO 100		ELCH1200
009	101 MI=0		ELCH1200
010	C		
011	C **** COMPUTE THE LOADS FOR THIS CONDITION ****		
012	C **** THE WIND SHEAR IS STORED IN 00 30.30.40. .70 ****		
013	C **** THE WIND ON IS STORED IN 00 33.37.41. .75 ****		
014	C **** THE WT SHEAR IS STORED IN 00 00.00.03. .100 ****		
015	C **** THE WT ON IS STORED IN 00 00.00.04. .100 ****		
016	C **** THE VT SHEAR IS STORED IN 00 130.142.146. .170 ****		
017	C **** THE VT ON IS STORED IN 00 130.143.147. .170 ****		
018	C		
019	100 CALL 04000		ELCH1200
020	CALL 04000		ELCH1200
021	C		
022	ALTI(1) = 00(0)		
023	WINDON(1) = 00(10)		
024	C		
025	C		
026	C DETERMINE MAXIMUM MET ON AT 000 AND AT STATION 2		
027	C WHEN THE WIND LOADS HAVE BEEN COMPUTED		
028	C STORE IN WINDC 30 AND 33		
029	C SAVE THE AIRLOAD ON, CONDITION INDICATOR, LOAD FACTOR		
030	C AND DEAD WEIGHT PERCENT		
031	C		
032	WINDC(00)=114.14.400		
033	C		
034	400 00 TO 0.0.0.0.7.0.7.0.0.0.7.7.0.0.7.0.0.0.0.0.1		
035	C		
036	0 WINDC(1) = WINDC(43)		
037	WINDC(1) = WINDC(44)		
038	GO TO 10		
039	C		
040	7 WINDC(1) = WINDC(40)		
041	WINDC(1) = WINDC(40)		
042	GO TO 10		
043	C		
044	0 WINDC(1) = WINDC(40)		
045	WINDC(1) = WINDC(40)		
046	GO TO 10		
047	C		
048	0 WINDC(1) = WINDC(47)		
049	WINDC(1) = WINDC(40)		
050	C		
051	10 WINDC(1) = 00(70) + WINDC(1) * 00(10)		
052	WINDC(2) = 00(00) + WINDC(1) * 00(10)		
053	C		
054	IF(WINDC(32) - ABS(WINDC(1)+11.12.12		
055	C		
056	11 WINDC(32) = WINDC(1)		
057	WINDC(1) = 00(20)		
058	WINDC(1) = 1		
059	WINDC(1) = 00(10)		
060	WINDC(1) = WINDC(1)		
061	C		
062	12 IF(WINDC(32) - ABS(WINDC(2)+11.14.14		
063	C		
064	13 WINDC(32) = WINDC(2)		
065	WINDC(1) = 00(00)		
066	WINDC(1) = 1		
067	WINDC(1) = 00(10)		
068	WINDC(2) = WINDC(2)		
069	14 CONTINUE		
070	C		
071	CALL PAR000		
072	C		
073	WINDC(1)=1125.125.010		
074	C		

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07/03/74      INPUT LISTING      AUTOFLON CHART SET - SHEEP      AIRLOADS OVERLAY

CARD NO      ****      CONTENTS      ****

300      C      **** STORE THE FUELAGE LOADS IN ARRAY FUEL(400) ****
301      C
302      010 CALL FUELRT
303      C
304      C      **** END LOOP FOR 23 LOAD CONDITIONS ****
305      C
306      100 CONTINUE
307      C
308      I = 24
309      IF (IND(24) = 0.00, 0.00, 0.00)
310      C
311      020 CALL FUELRT
312      C
313      030 IF (IND(25) = 0.00, 0.00, 0.00)
314      C
315      C      * * * CALL PARLOS FOR CONDITION 24 120 TAIL LOADS * * *
316      C
317      000 CALL PARLOS
318      C
319      000 CALL WPARCT
320      C
321      IV = 0
322      C
323      IV = 0
324      C
325      TVIB = 0.0
326      C
327      TVI = 0.0
328      C
329      TVI = 0.0
330      C
331      00 100 I=1,03
332      C
333      IF (IND(1+100) = 1.00, 1.00, 1.00)
334      C
335      1100 N = 1
336      C
337      PSI (N) = 0.0
338      C
339      FLOCAL (N) = 0.0
340      C
341      FTOTAL (N) = 0.0
342      C
343      WFLUX (N) = 0.0
344      C
345      TSKIN (N) = 0.0
346      C
347      120 CONTINUE
348      C
349      IF (IND(26) = 400, 400, 400)
350      C
351      400 IM = 1000000
352      C
353      TVIB = TSKIN (IM)
354      C
355      400 IF (IND(26) = 410, 410, 407)
356      C
357      407 IM = 1000000
358      C
359      TVI = TSKIN (IM)
360      C
361      410 IF (IND(27) = 420, 420, 412)
362      C
363      412 IV = 1000000
364      C
365      TVI = TSKIN (IV)
366      C
367      420 CONTINUE
368      C
369      IF (IP(28) = 9000, 9000, 9000)
370      C
371      9000 CONTINUE
372      C
373      C      **** PRINT DATA FOR 23 LOAD CONDITIONS ****
374      C
375      WRITE (6, 700)
376      C
377      700 FORMAT (1H, 2X, 21H ** BLCHL = (P(24)) ****
378      C
379      I
380      3X, 14H LOAD CONDITION, 3X, 04H ALTITUDE, 2X, 40H WIND,
381      C
382      4X, 04H PRESSURE, 1X, 10H LOCAL TEMP, 1X, 10H TOTAL TEMP, 2X,
383      C
384      04H FLUX, 1X, 04H IN TEMP, 1X, 04H IN TEMP, 3X, 04H STRESS,
385      C
386      4X, 04H STRESS, 4X, 04H STRESS / 25X, 40H CFT, 3X, 04H DENSITY,
387      C
388      3X, 04H PSI, 7X, 04H R, 3X, 04H R, 3X, 10H DENSITY, 12,
389      C
390      2X, 04H R, 3X, 04H R, 3X, 04H R, 3X, 10H DENSITY,
391      C
392      1X, 04H VERTICAL /
393      C
394      WRITE (6, 710) PSI (N), WIND (N), PSI (N), FLOCAL (N), FTOTAL (N),
395      C
396      04H FLUX (N), TSKIN (N), TSKIN (N), STEMP (N), STEMP (N),
397      C
398      04H STEMP (N), N=1, 10)
399      C

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ASTROPHYSICAL JOURNAL

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INPUT LISTING

AUTOFLOW CHART SET - SHEEP

AIRLOADS OVERLAY

CARD NO

CONTENTS

```

007      TWD=010.000                                A77000+0
008      SD=20 170+0                                A7700000
009      SD=1710.4007                                A7700000
010      RE=00000001                                A7700070
011      IF IN=20152.120.40.40                      00000100
012      SD=0 0                                         00000100
013      TWD=010.000                                000001+0
014      TWD=- 00700010                              00000100
015      PAGE=14.00000                               00000100
016      SD TO 100                                    00000100
017      40 IF IN=20730.140.00.00                    00000170
018      40 SD=20000.7+                              00000100
019      TWD=200.000                                  00000100
020      PAGE=3.000+71                                00000100
021      SD TO 110                                    00000200
022      50 IF IN=10070.100.00.00                    00000210
023      50 SD=00001.                                00000200
024      TWD=200.000                                  00000230
025      TWD=- 0010+002                              000002+0
026      PAGE=- 2000+03                              000002+0
027      SD TO 100                                    00000200
028      60 IF IN=17070.0+00.70.70                  00000200
029      60 SD=10+100.5                               00000270
030      .TWD=000 700                                00000200
031      PAGE=- 017+00+0                              00000200
032      SD TO 110                                    00000200
033      70 IF IN=2+0000 0+70.00.00                  00000200
034      70 SD=1700+0.5                               00000210
035      TWD=000.700                                  00000200
036      TWD=-. 000120000                             00000230
037      PAGE=- 000+0001                              00000200
038      SD TO 100                                    000002+0
039      80 SD=2+0001.                                00000200
040      TWD=20+0.7+0                                00000200
041      PAGE= 0007000.700                           00000200
042      SD TO 110                                    00000270
043      100 SD=RE 4+/(RE 4+))                        00000200
044      TWD=TWD+TWD+/(SD+SD))                        00000200
045      VP=- (SD/(SD+TWD)))/(ALSD/(TWD/TWD))         00000+00
046      P=PAGE < VP / VP)                           00000+10
047      SD=0. 001177+0001/(TWD)                    00000+20
048      SD=SD+SD+VP+TWD/(P0+TWD)                    00000+30
049      SD TO 120                                     00000+40
050      110 SD=RE 4+/(RE 4+))                        00000+00
051      TWD=TWD                                     00000+00
052      VP=-SD/(SD+SD)/(SD+TWD)                      00000+70
053      P=PAGE < VP / VP)                           00000+00
054      SD=0. 001177+0001/(TWD)                    00000+00
055      SD=SD+SD+VP+TWD/(P0+TWD)                    00000+00
056      120 RETURN                                     00000+00
057      END
058
059      C
060      C
061      C SUBROUTINE DLDS
062      C
063      C SUBROUTINE DLDS
064      C SUBROUTINE DLDS FOR LINK S OF SHEEP.
065      C COMPUTES COMPONENT TOTAL AIRLOADS AND CP'S. AND INERTIA FACTORS.
066      COMMON TCDH(400)
067      COMMON /IPRINT/IO(00)
068      DIMENSION DD(20),DD(00),BC(100),DS(20),DU(20),DC(20),DF(10),ND(200000)
069      I(200)
070      EQUIVALENCE (DF(1),TCDH(1000)),(DD(1),TCDH(200)),(BC(1),TCDH(27002000010
071      1), (DU(1),TCDH(3103)), (DS(1),TCDH(3070)), (DD(1),TCDH(3301)), (DD(1),32000011
072      2,TCDH(2003)), (ND(1),TCDH(4001)), (IN,ND(1001)), (INT,ND(1501)), (I,ND(1532000012
073      31)), (J,ND(1021)), (K,ND(1031)), (IP,ND(1371))
074      CALL A77000(0),DS(1),DS(2),DS(3)
075      DS(4)=DS(1)+DS(2)+DS(3)+(DD(10)+0)
076      DS(5)=SQR(200.0 + DS(4))
077      DD(5)=BC(33)+BC(34)+BC(35)/(3.1416*(BC(36)+0))

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07-03-74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADDS OVERLAY
CARD NO	****	CONTENTS	****
000	MT=0		20000120
000	3 MT=MT+1		20000140
070	00 TO 14,0,0,101,MT		20000100
071	4 1=0		20000100
072	J=4		20000170
073	00 TO 0		20000100
074	0 1=10		20000100
075	J=0		20000000
076	0 IF(1P)0,7,0		20000710
077	7 0=71		20000000
078	00 TO 11		20000720
079	0 0=0		20000740
080	00 TO 11		20000750
081	0 1=13		20000000
082	J=11		20000760
083	0=107		20000000
084	00 TO 11		20000700
085	10 1=17		20000700
086	J=170		20000310
087	0=120		20000120
088	11 00(11)=0C(11)+0U(11)		20000120
089	IF(107)-313,3,12		20000740
090	12 IF(00(5)+113,14,13		20000100
091	13 0 = 00		20000702
092	IF(1P) 01,00,01		20000704
093	00 0 = 71		20000705
094	01 00(0) = 0C(0) + 0U(0)		20000700
095	00(7) = 0C(7) + 0U(05)		20000770
096	00 TO 10		20000100
097	14 00(0)=0.0		20000100
098	00(7)=0.0		20000700
099	15 IF(1P)17,10,17		20000710
000	16 1=70		20000730
001	00 TO 10		20000730
002	17 1=4		20000740
003	18 00(13)=0.0		20000700
004	00(10)=00(0)		20000700
005	00 TO 110,20,20,10,10,10,21,101,01		20000770
006	19 00(0)=1.1+00(11)+00(10)/00(4)+0C(11)		20000700
007	0U TO 20		20000700
008	20 00(0)=00(10)+0C(10)+07(11)+07(11)+0.5)+0U(11)+0U(01)+00(5)/07 3		20000700
009	00(0)=00(0)+1.1+00(11)+00(10)/00(4)+0C(11)		20000710
010	00 TO 20		20000720
011	21 00(13)= (00(7)+00(13)+12.0)/0U(11)+0C(107)+00(21)		20000730
012	00(10)+00(10)+00(13)/00(11)		20000740
013	00 TO 10		20000750
014	22 00(11)+00(0)/0U(1)		20000700
015	00(11)= 04 2032+00(11)+00(4)+0C(101)+02		20000770
016	00(21)+0U(21)+00(0)+00(4)+0C(11)		20000700
017	00(31)+00(21)+1.0-0U(21)/0U(2)		20000700
018	00(4)=1+00(21)+00(0)+00(11)+00(10)+00(21)+00(31)+00(0)+00(21)+00(23)+00(00)		20000700
019	111/00(131)+00(21)		20000710
020	00(0)=0.0		20000720
021	00(0)=0.0		20000730
022	IF(00(0)+23,24,23		20000740
023	23 00(0)+0U(02)+00(0)+00(4)+0C(11)		20000700
024	00(0)+00(0)+1.0-0U(02)/0U(02)		20000700
025	00(4)+00(4)+1+00(0)+00(21)+00(0)+00(17)+00(21)+00(0)+1/00(131)+00(23)+00(07)		20000700
026	111		20000700
027	24 00(12)+00(11)+00(21)+00(31)+00(4)+00(0)+00(0)		20000700
028	IF(00(12)) 53,52,53		20000702
029	00 00(10) = 1.0		20000704
030	00 TO 04		20000700
031	03 00(10) = 00(11)+00(10)/00(12)		20000700
032	04 0U(000) = 00(21)+00(10)		20000700
033	0U(007) = 00(0)+00(10)		20000710
034	0U(000) = 00(13) + 00(4)+00(10)		20000700
035	00(3)+00(11)+00(11)+00(10)/00(12)		20000730
036	00(0)+0.0+0U(000)+0U(007)		20000740
037	00(0) + (00(3) + 00(0)+00(10)		20000700
038	00(11) + 0.0+00(13) + 00(4)+00(10)		20000700

07/03/74	INPUT LISTING	AUTOLOAD CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
650	IF (BO(5)) 00,00,00		32000700
656	00 BO(7) = BU(3)		32000770
661	00 TO 57		32000775
662	00 BO(7) = (BU(3)-BU(200) + BU(53)-BU(227)) / (2.0-BO(6))		32000780
663	57 IF (BO(9)) 00,00,00		32000785
664	00 BO(10) = (BO(10)-BO(3) + BO(7)-BO(6)) / (BO(10)-BO(9))		32000790
665	00 BO(12) = BU(110)		32000800
666	BU(200) = 2.0 - BU(100) - BO(111)		32000810
667	BO(14) = ABS(.30 - BO(111) - BO(12))		32000820
668	BO(15) = 0.0		32000825
669	BO(16) = 0.0		32000835
670	00 25 J=10,21		32000839
671	25 BO(21) = 0		32000840
672	BO(14) = 0.0		32000843
673	00 TO (20,20,20,27,27,34,30,30),H1		32000850
674	00 BO(10) = BO(0)		32000860
675	00 TO 37		32000870
676	37 IF (IP) 20,20,20		32000880
677	20 J=70		32000890
678	00 TO 30		32000900
679	30 J=0		32000910
680	30 BO(14) = BO(11) - BO(2) / (10.0 - BO(1)) - BU(11) - (BO(11) - BO(2))		32000920
681	IF (BO(10) - 1.0) 31,32,32		32000930
682	31 BO(15) = (BO - BO(14)) / (0.3 - BO(14))		32000940
683	00 TO 33		32000950
684	32 BO(15) = (BO(14) ** .03) / (0.00 + (BO(14) ** .03))		32000960
685	33 BO(7) = BO(5) + (BO(35) - BO(4)) / ABS(.3753)		32000970
686	BO(0) = (.100304 - BO(15) - BU(11) - BU(2)) - BO(11) - BO(5)		32000980
687	BO(0) = BO(0) + (1.0 - BU(2)) / BU(2)		32000990
688	BO(10) = (.100304 - BO(15) - BU(100) - BO(111) - BO(5))		32001000
689	IF (H1 - 0) 40,1,40		32001002
690	1 00 2 J=7,10		32001004
691	2 BO(21) = BO(21)		32001005
692	40 BO(10) = 1.0 - (BO(7) - BO(0) - BO(0) - BO(10)) / BO(11)		32001010
693	BO(20) = ((BO(2) - BO(0) - BO(7) - (BO(0) - BO(2)) - BO(0) - (BO(10) - BO(2))) - BO(10) - BO(2)) / BO(10)		32001020
694	101 = (BO(11) - BO(2)) - BO(10) / (BO(3) + 12.0)		32001030
695	BO(2) = BO(3) - BO(7)		32001040
696	BO(0) = BO(0) + 0.5 - BO(0)		32001050
697	BO(0) = BO(0) - BO(0)		32001060
698	BO(11) = BO(11) + 0.5 - BO(10)		32001070
699	BO(14) = BO(14) - ABS(.15 - BO(10) - BO(12))		32001080
700	BU(200) = BU(200) - BO(0)		32001090
701	BU(200) = BU(200) - BO(10) - BU(100)		32001100
702	00 TO 37		32001110
703	34 BO(14) = (0.43700 - BO(5)) - (BO(35) - BO(2))		32001120
704	BO(15) = (.100304 - BU(107) - BO(143) - BO(5))		32001130
705	IF (BO(15)) 101,50,51		32001133
706	50 BO(15) = 2.0 - BO(15)		32001135
707	51 BO(10) = BU(100) + BO(150)		1140
708	BO(10) = (BO(14) - BO(15)) / BO(11)		32001150
709	BO(21) = ((BO(2) - BO(5)) - BO(4) - (BO(17) - BO(2)) - BO(15)) / (BO(14) + 12.0)		32001160
710	00 TO 30		32001170
711	30 BO(20) = BO(7)		32001180
712	00 TO 35		32001190
713	35 BO(15) = (BO(0) - BO(14)) / ((BO(17) - BO(2)) + 12.0)		32001210
714	BO(10) = BU(100) + BO(150)		1200
715	BO(10) = BO(10) / BO(11)		32001230
716	BO(21) = BO(0)		32001240
717	00 TO 30		32001250
718	37 IF (1000) 10001,0001,0002		
719	0001 CONTINUE		
720	IF (ND(2)) 147,47,40		
721	47 IF (ND(2)) 140,40,40		32001253
722	40 WRITE(0,40) BO(11),BO(10),BO(0),BO(5)		32001254
723	40 FORMAT(11H,6X,13CONDITON NO=,F7.0,5X,30000,F0.3,5X,4HALLT=,F7.0,		
724	1 5X,30000,F0.2,10X,20H** 04.05 - (P(50) **)		
725	WRITE(0,30) BO(3),BO(4),BO(5)		32001260
726	30 FORMAT(11H,6X,10BODY LOADS/7X,40P2H=,F0.0,4X,40P7H=,F7.0,4X,		32001270
727	1 4000H=,F0.2//		32001271
728	WRITE(0,30) BO(6),BO(7),BO(0)		32001280
729	30 FORMAT(11H,6X,10SHIMS PANEL LOAD/7X,40P2H(0)/2=,F0.0,4X,30P2H(0)=,32001290		

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411

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
002	GO TO 10		FATH0100
003	9 I=240		FATH0105
004	10 DO 11 J=1,40		FATH0110
005	K=J+1		FATH0115
006	11 DO 12 I=1,40		FATH0120
007	DO 12 J=000,007		FATH0125
008	DO 13 I=0C(107)+0D(1+415)		FATH0130
009	12 DO 14 I=0C(100)+0D(1+07)		FATH0135
010	I=0		FATH0140
011	GO TO 117,17,13,14,10,101,NA		FATH0145
012	13 I=100		FATH0150
013	GO TO 17		FATH0155
014	14 I=140		FATH0160
015	GO TO 17		FATH0165
016	15 I=020		FATH0170
017	GO TO 17		FATH0175
018	16 I=000		FATH0180
019	17 DO 20 J=1,20		FATH0185
020	K=J+1		FATH0190
021	DO 21 J=575+0C(K)		FATH0195
022	IF (J-10) 10, 10, 10		FATH0200
023	10 DO 22 J=005+0C(K)-1.0		FATH0205
024	GO TO 20		FATH0210
025	10 DO 23 J=005+0C(K)+1.0		FATH0215
026	20 CONTINUE		FATH0220
027	MS=0		FATH0225
028	MS=1		FATH0230
029	MS=1		FATH0235
030	MS=2		FATH0240
031	DO 34 I=1,0		FATH0245
032	MS=MS+1		FATH0250
033	00(1)=00(1)+000		FATH0255
034	00(9)=00(1)+120		FATH0260
035	00(10)=00(1)+120		FATH0265
036	00(12)=00(1)+130		FATH0270
037	IF (00(12)-2.0) 20,21,22		FATH0275
038	21 J=77		FATH0280
039	K=101		FATH0285
040	GO TO 23		FATH0290
041	22 J=0		FATH0295
042	K=170		FATH0300
043	23 IF (MS-2) 24,20,20		FATH0305
044	24 CALL UEPAN		FATH0310
045	05(1)=0C(171)/05(12)		FATH0315
046	05(3)=0J(7)		FATH0320
047	05(4)=CODIMP(05(1),0J(9),0J(35),13,5)		FATH0325
048	IF (MS - 2) 25,27,27		FATH0330
049	25 00(10)=05(3)		FATH0335
050	00(20)=05(4)		FATH0340
051	27 05(5)=CODIMP(00(1)+144,0C(170),0C(K),3,5)		FATH0345
052	05(6)=CODIMP(00(1)+144,0C(170),0C(K+3),3,5)		FATH0350
053	GO TO 31		FATH0355
054	28 IF (00(10)-00(1)+10) 24,20,24		FATH0360
055	29 IF (00(1)+130-00(1)+120) 24,30,24		FATH0365
056	30 IF (00(1)-00(1)+900) 127,31,27		FATH0370
057	31 NY=00(1)+112		FATH0375
058	GO TO 32,32,30,37,40,42,NA		FATH0380
059	32 K=20		FATH0385
060	GO TO 144,33,34,35,NY		FATH0390
061	33 K=0		FATH0395
062	GO TO 44		FATH0400
063	34 K=00		FATH0405
064	GO TO 44		FATH0410
065	35 K=00		FATH0415
066	GO TO 44		FATH0420
067	36 K=120		FATH0425
068	GO TO 44		FATH0430
069	37 K=100		FATH0435
070	IF (NY-2144,30,30)		FATH0440
071	38 K=100		FATH0445
072	GO TO 44		FATH0450

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
004	00 00 J=1.20		FATH0700
005	K=J+1		FATH0705
006	00(K)=00(J-20)+00(51)		FATH0710
007	00(K+1)=00(J-20)+00(52)		FATH0715
008	00(K+51)=00(J-20)+00(40)		FATH0740
009	00(K+52)=00(J-20)+00(40)		FATH0745
1000	00 I=1+2		FATH0750
1001	00(M+200)=00(51)		FATH0751
1002	00(M+210)=00(52)		FATH0752
1003	00(M+210)=00(40)		FATH0753
1004	00(M+200)=00(40)		FATH0754
1005	00(400)=0.0		FATH0755
1006	00(500)=10.0		FATH0760
1007	00(511)=11.0		FATH0765
1008	00(50)=00(131)/2.0		FATH0770
1009	I=175		FATH0810
1010	J=175		FATH0812
1011	M=00(14)+00.00.03		FATH0813
1012	00 J=101		FATH0814
1013	M=00(137)-2.0+00.00.05		FATH0815
1014	00 I=101		FATH0820
1015	00 00(40)=00(100)+00(140)+00(170)+00(11+3+3+5)		FATH0825
1016	00(40)=00(100)+00(140)+00(170)+00(11+3+3+5)		FATH0830
1017	00(512)=00(50)+00(40)+00(10)+00(57)/2.0		FATH0835
1018	00(513)=00(50)+00(40)+00(20)+00(57)/2.0		FATH0840
1019	00(514)=00(170)		FATH0845
1020	00(515)=00(100)+00(170)+00(11+3+3+5)+1.2		FATH0850
1021	00(516)=00(100)+00(170)+00(11+3+3+5)+1.2		FATH0855
1022	00(517)=00(170)		FATH0860
1023	07 I=1		FATH0862
1024	K=1		FATH0864
1025	L=1		FATH0867
1026	M=1		FATH0868
1027	M=10(50)+00(100)+00(100)+00(2)		
1028	0001 CONTINUE		
1029	00 IF K=00(50)+00.04		FATH0870
1030	04 WRITE(6,04)+00(1)		FATH0872
1031	04 FORMAT(1H) 00H,20H** FATH0 - (P100) ****		FATH0873
1032	I 4H,10H SPECTRA SEGMENT NO,74 0//74,		FATH0874
1033	I 10H00F SEND MEM,34,10H40S SEND MEM,54,10H RECDANCES-TAKI		FATH0875
1034	00 TO 07		FATH0876
1035	00 WRITE(6,00)+00(1)		FATH0878
1036	00 FORMAT(1H) 00H,20H** FATH0 - (P100) ****		FATH0879
1037	I 4H,10H SPECTRA SEGMENT NO,74 0//74,		FATH0880
1038	I 10H00F SEND MEM,34,10H40S SEND MEM,54,10H RECDANCES-OUT,34,		FATH0880
1039	2 10H RECDANCES-RUN)		FATH0884
1040	07 00 74 J=1.20		FATH0891
1041	IF J=1170,70,70		FATH0892
1042	70 WRITE(6,71)		FATH0894
1043	71 FORMAT(1H)		FATH0895
1044	72 K=J+1		FATH0896
1045	IF K=00(101),101.00		FATH0898
1046	00 WRITE(6,100)+00(K),00(K+1),00(K+2)		FATH0899
1047	100 FORMAT(1H) 4H F15.0,5H E12.5)		FATH0903
1048	00 TO 74		FATH0904
1049	101 WRITE(6,73)+00(K),00(K+1),00(K+2),00(M+L)		FATH0905
1050	73 FORMAT(1H) 4H F15.0,5H E12.5,5H E12.5)		FATH0907
1051	L=L+1		FATH0908
1052	74 I=1+2		FATH0909
1053	IF K=00(102),102.103		FATH0911
1054	102 WRITE(6,00)+00(M+200)+00(M+210)		FATH0912
1055	00 FORMAT(1H) 10H 10 F15.0,715 0.34,10H10 MDS M=F15.0)		FATH0913
1056	M=M+1		FATH0914
1057	103 IF 10(20)+70,70,70		FATH0915
1058	70 IF K=00(77),9502,9502		
1059	70 IF K=00(77),70,70		FATH0918
1060	77 I=K+3		FATH0919
1061	00 TO 00		FATH0919
1062	70 WRITE(6,70)+00(511)		FATH0920
1063	70 FORMAT(1H) 10H,20H** FATH0 - (P100) ****		
1064	I 4H,10H SPECTRA SEGMENT NO,74 0//74,2+H0000-AIR-000		

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
1130	100 NS = N1-3		FC000100
1131	00 TO 000		FC000100
1132	170 NS = E-2		FC000170
1133	200 00 200 1 = 1.04		FC000170
1140	L = (N3-1) * 42 + 1		FC000100
1141	TS(1) = T(LNS)		FC000100
1142	T(1) = CDBTIME(1,1),Z(L),N2,N3		FC000100
1143	300 NS = N3-1		FC000100
1144	FC0002 = CDBTIME(T,TS,T,N2,N3)		FC000200
1145	RETURN		FC000200
1146	END		FC000210
1147	C		
1148	C (.....)		
1149	C SUBROUTINE FUELST		
1150	C (.....)		
1151	C		
1152	C SUBROUTINE FUELST		
1153	C		
1154	C		
1155	C SUBROUTINE TO STORE TOTAL AIRLOADS DATA FOR FUELAGE		
1156	C IN RECORD 33		
1157	C		
1158	COMMON TCON(400),MLD(300),IDUM(132),SAGE(132),MDE(132),		
1159	* RHE(132),RC(44),SW(100),STEPH(23),STEPV(23),STEPW(23),		
1160	* BLPH(100)		
1161	C		
1162	COMMON /PHSC/ PHSC(100)		
1163	C		
1164	DIMENSION MD(200),DD(620),BC(100)		
1165	DIMENSION TCON(23)		
1166	DIMENSION FUS(67),DD(20)		
1167	C		
1168	EQUIVALENCE (MD(1),TCON(420)),(DD(1),TCON(330)),(BC(1),TCON(270)		
1169	*(1,1,MD(101)),(DD(1),TCON(205))		
1170	EQUIVALENCE (SW(110),TCON(13))		
1171	C		
1172	IF(1) = 1) 700,700,0		
1173	0 K = 1 - 1		
1174	00 10 J=1,K		
1175	IF(MD(J+100)) 10,10,710		
1176	10 CONTINUE		
1177	C		
1178	700 00 701 N=1,070		
1179	701 FUELIN = 0.0		
1180	C		
1181	710 IF(1) = 23) 019,019,0012		
1182	C		
1183	019 NK = 1 - 1 = 20		
1184	C		
1185	IF(1) = 0)001,002,003		
1186	001 FUS(NK+2) = 1.0		
1187	00 TO 012		
1188	002 FUS(NK+2) = 2.0		
1189	00 TO 012		
1190	003 IF(1-0)004,004,005		
1191	004 FUS(NK+2) = 3.0		
1192	FUS(230) = PHSC(35)		
1193	FUS(237) = PHSC(30)		
1194	00 TO 012		
1195	005 IF(1-10)006,006,007		
1196	006 FUS(NK+2) = 4.0		
1197	00 TO 012		
1198	007 IF(1-20)008,008,010		
1199	008 FUS(NK+2) = 5.0		
1200	00 TO 012		
1201	009 FUS(NK+2) = 6.0		
1202	00 TO 012		
1203	010 IF(1-32)009,011,011		
1204	011 FUS(NK+2) = 7.0		
1205	C		
1206	012 IF(MD(137))020,021,020		

07/05/74	INPUT LISTING	AUTOFLOW CHART SET - MEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
1007	C		
1008	GO FUSION(0) = SC(10)		
1009	GO TO 002		
1010	C		
1011	GO FUSION(0) = SC(71)		
1012	C		
1013	GO FUSION(1) = 2.0		
1014	FUSION(0) = TSC(1071)		
1015	FUSION(0) = HISC(37)		
1016	FUSION(7) = SC(10)		
1017	FUSION(0) = SC(10)		
1018	FUSION(10) = SC(20)		
1019	FUSION(11) = SC(21)		
1020	FUSION(14) = SC(10)		
1021	FUSION(15) = SC(0)		
1022	FUSION(16) = SC(3)		
1023	FUSION(17) = SC(0)		
1024	FUSION(18) = SC(0)		
1025	FUSION(19) = SC(10)		
1026	FUSION(20) = SC(0) = 2.0		
1027	FUSION(21) = SC(0)		
1028	FUSION(22) = SC(7)		
1029	FUSION(23) = SC(11) = 2.0		
1030	FUSION(24) = SC(13)		
1031	FUSION(25) = SC(12)		
1032	FUSION(26) = SC(15)		
1033	FUSION(27) = SC(17)		
1034	FUSION(28) = SC(16)		
1035	GO TO 3015		
1036	C		
1037	C **** WRITE THE FUSELAGE LOADS IN RECORD 33 ****		
1038	C		
1039	3012 FUSION(0) = HISC(37)		
1040	FUSION(1) = HISC(30)		
1041	FUSION(5) = 2.0		
1042	FUSION(6) = 0.0		
1043	FUSION(9) = SC(0)		
1044	C		
1045	CALL WRITE(1,FUSION,070,33)		
1046	C		
1047	3015 CONTINUE		
1048	RETURN		
1049	END		
1050	C		
1051	C		
1052	C		
1053	C		
1054	C		
1055	C		
1056	C		
1057	C		
1058	C		
1059	C		
1060	C		
1061	C		
1062	C		
1063	C		
1064	C		
1065	C		
1066	C		
1067	C		
1068	C		
1069	C		
1070	C		
1071	C		
1072	C		
1073	C		
1074	C		
1075	C		
1076	C		
1077	C		


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07/05/74      INPUT LISTING      AUTOFLIGHT CHART SET - SHEEP      AIRLOADS OVERLAY

CARD NO      ****      CONTENTS      ****

1270      EQUIVALENCE (1001,TCOM4001), (1001,TCOM5001), (1,100101)
1270      C
1280      EQUIVALENCE
1281      * (1001101,TEMP111),
1282      *
1283      * (1001101,TEMP111), (1001101,STN111), (1001101,TEMP111),
1284      * (1001101,STV111), (1001101,TEMP111), (1001101,TEMP111),
1285      * (1001101,TEMP111), (1001101,TEMP111), (1001101,TEMP111),
1286      * (1001101,TEMP111), (1001101,TEMP111), (1001101,TEMP111),
1287      C
1288      EQUIVALENCE (1001,ALD111), (1001,ALD111), (1001,ALD111),
1289      * (1001,ALD111), (1001,ALD111), (1001,ALD111), (1001,ALD111),
1290      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1291      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1292      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1293      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1294      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1295      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1296      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1297      * (1001,ALD111), (1001,ALD111), (1001,ALD111),
1298      C
1299      EQUIVALENCE (1001,TCOM2700)
1300      EQUIVALENCE (1001,TCOM2700)
1301      C
1302      C
1303      C      0-10-70--TEST FOR INITIAL CALL--SETUP MPAL10 AND MD ARRAYS
1304      100 IF (1 - 1) 101,104,104
1305      101 IF (1 - 2) 102,102,102
1306      102 J = 1 - 1
1307      DO 103 N=1,J
1308      IF (1001100) 103,101,700
1309      103 CONTINUE
1310      C
1311      C      0-10-70--TEST FOR INITIAL CALL--SETUP MPAL10 AND MD ARRAYS*
1312      C      MPAL10(1:24)=INDICATORS FOR CONDITIONS EVALUATED*
1313      104 DO 105 N=1,25
1314      MPAL10(N) = 1001100
1315      105 CONTINUE
1316      C      10 LANDING NOT NEEDED FOR LIFTING SURFACES
1317      MPAL10(1) = 0.0
1318      C
1319      C      MPAL10(24)=2-0 TAXI 10 = MD(25)*
1320      MPAL10(24) = MD(25)
1321      C
1322      C      NAME MPAL10(10:24) IN RCD 100*
1323      CALL WRITING (1,MPAL10(1),24,100)
1324      C
1325      C      SETUP MD(1:200) FROM RCD 01*
1326      CALL READS (1,MD(1),200,01)
1327      C
1328      C
1329      C      *CLEAR SAKET ARRAY FOR MAX TUMBLE SELECTION*
1330      DO 700 I=1,50
1331      SAKET(I) = 0.0
1332      700 CONTINUE
1333      C
1334      C
1335      C      **** LOOP 000 ****
1336      C      **** DETERMINE STEPM, STEPM AND STEPMV, THE STRESS LEVELS ****
1337      C      **** AT THE SKIN TEMPERATURE OF THIS LOAD CONDITION ****
1338      C      **** AND RSM, RSM AND RSV, THE RATIOS OF THE STRESS AT 00 DEG ****
1339      C      **** TO THESE STRESS IF THE MATERIAL OF THE COMPONENT IS KNOWN****
1340      C
1341      700 RSM = 1.0
1342      RSM = 1.0
1343      RSV = 1.0
1344      C
1345      DO 000 N=1,3
1346      C
1347      DO TO (001,002,003),1
1348      C

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07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AMPLITUDE OVERLAY
CARD NO	CONTENTS		
1340	001 IF IHTIME(10)1000.000,001		
1350	C		
1351	001 DO 005 N=1,5		
1352	STRESS(N) = STMIN		
1353	005 TEMP(N) = TEMPMIN		
1354	NTMP = N*NT		
1355	GO TO 010		
1356	C		
1357	002 IF IHTIME(10)1000.000,002		
1358	C		
1359	002 DO 006 N=1,5		
1360	STRESS(N) = STMIN		
1361	006 TEMP(N) = TEMPMIN		
1362	NTMP = N*NT		
1363	GO TO 010		
1364	C		
1365	003 IF IHTIME(20)1000.000,003		
1366	C		
1367	003 DO 007 N=1,5		
1368	STRESS(N) = STMIN		
1369	007 TEMP(N) = TEMPMIN		
1370	NTMP = N*NT		
1371	C		
1372	010 IF IHTMP = 1000.000,000		
1373	C		
1374	004 STRES1 = STRESS(1)		
1375	GO TO 000		
1376	C		
1377	005 IF IHTMP(1) = TEMP(1)000.000,005		
1378	C		
1379	005 L1 = 1		
1380	L2 = 2		
1381	GO TO 000		
1382	C		
1383	006 IF IHTMP(1) = TEMP(1)075.075,006		
1384	C		
1385	006 L1 = IHTMP - 1		
1386	L2 = IHTMP		
1387	GO TO 000		
1388	C		
1389	075 DO 002 N=2, IHTMP		
1390	IF IHTMP(1) = TEMP(1)075.075,002		
1391	075 L1 = N - 1		
1392	L2 = N		
1393	GO TO 000		
1394	002 CONTINUE		
1395	C		
1396	005 STRES1 = STRESS(L2) - (STRESS(L2) - STRESS(L1)) * = (TEMP(L2) - TEMP(L1)) / (TEMP(L2) - TEMP(L1))		
1397			
1398	C		
1399	C		
1400	000 GO TO (011,012,013),N		
1401	C		
1402	011 STEPM(1) = STRES1		
1403	STN = 5004 / STEPM(1)		
1404	GO TO 000		
1405	C		
1406	012 STEPM(1) = STRES1		
1407	STN = 5004 / STEPM(1)		
1408	GO TO 000		
1409	C		
1410	013 STEPM(1) = STRES1		
1411	STN = 5004 / STEPM(1)		
1412	C		
1413	000 CONTINUE		
1414	C		
1415	C **** SET LOADS AT TIP TO 0.0 ****		
1416	C		
1417	00 00 111 = 32,130,93		
1418	IF IADS(001111) = 10.0191,01,02		
1419	01 001111 = 0.0		

07/03/74	INPUT LISTING	AUTOTON CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
1400	C-----CCCCC-----		
1401	00 IF (ABS(BD(111)-1)) - 10.01 03.03.04		
1402	C-----CCCCC-----		
1403	00 BD(111)+1 = 0.0		
1404	C-----CCCCC-----		
1405	04 IF (ABS(BD(111)-2)) - 10.01 05.05.06		
1406	00 BD(111)-2 = 0.0		
1407	C-----CCCCC-----		
1408	00 CONTINUE		
1409	C		
1410	IF (ABS(BD(110),100,100		
1411	C		
1412	C ---- LOOP 00 - - WIND LOADS ----		
1413	C		
1414	100 00 00 N=1,11		
1415	C		
1416	J = 10 - N		
1417	C		
1418	111 = 10 + (N-1) * 4		
1419	C		
1420	100 = 0		
1421	C		
1422	00 101,1,1,1,0,1,0,1,0,1,1,0,0,1,1,0,0,1,1,1,1,1,1,1,1		
1423	C		
1424	1 WPC = V2(J)		
1425	00PC = 0012(J)		
1426	00 TO 10		
1427	C		
1428	2 WPC = V2(J)		
1429	00PC = 0002(J)		
1430	100 = 1		
1431	00 TO 10		
1432	C		
1433	3 WPC = V2(J)		
1434	00PC = 0011(J)		
1435	00 TO 10		
1436	C		
1437	4 WPC = V2(J)		
1438	00PC = 0003(J)		
1439	C		
1440	10 WET = (BD(111)) + WPC + BD(10) * RM		
1441	00ET = (BD(111)+1) + 00PC + BD(10) * RM		
1442	C		
1443	IF (WET) 20,40,20		
1444	C		
1445	20 IF (WET - 00ET) 100,21,21		
1446	C		
1447	21 SAME(11) = BD(111)		
1448	00ET(11) = WET		
1449	100(11) = 1		
1450	00ET(11) = BD(10) / 100(11)		
1451	C		
1452	N = N		
1453	C		
1454	20 RE(11) = 1.0		
1455	C		
1456	IF (100) 25,25,40		
1457	C		
1458	25 IF (ABS(W2(J) - W2(J)) - 10.0140,40,05		
1459	00 RE(11) = (WPC - W2(J)) / (W2(J) - W2(J))		
1460	00 TO 40		
1461	C		
1462	3 IF (WET - 00ET(11)) 31,31,40		
1463	C		
1464	31 SAME(11) = BD(111)		
1465	00ET(11) = WET		
1466	100(11) = 1		
1467	00ET(11) = BD(10) / 00ET(11)		
1468	C		
1469	N = N + 11		
1470	00 TO 35		

07/03/76	INPUT LISTING	AUTOLON CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1401	C		
1402		40 IF (BNET) 100.00.00	
1403	C		
1404		50 IF (BNET) - BNETIN=20:100.01.01	
1405	C		
1406		51 SAME (N=20) = 00:111:11	
1407	C	*****	
1408		BNETIN = 00:111:21	
1409	C	*****	
1410		BNETIN=20) = BNET	
1411		ISLPHIN=20) = 1	
1412		BNETIN=20) = 00:10) / POUND	
1413	C		
1414		N = N + 20	
1415	C		
1416		05 RCIN) = 1 0	
1417	C		
1418		IF (IND) 50.00.00	
1419	C		
1410		00 IF (ABS(BNETJ) - 0000:J) - 10 0:00.00.00	
1411		00 RCIN) = (BNET - 0000:J) / (BNETJ) - 0000:J)	
1412		00 TO 00	
1413	C		
1414		00 IF (BNET) - BNETIN=33:101.01.00	
1415	C		
1416		01 SAME (N=33) = 00:111:11	
1417	C	*****	
1418		BNETIN=11) = 00:111:21	
1419	C	*****	
1420		BNETIN=33) = BNET	
1421		ISLPHIN=33) = 1	
1422		BNETIN=33) = 00:10) / MEGHZ	
1423	C		
1424		N = N + 33	
1425		00 TO 05	
1426	C		
1427	C	00 TAIL CONDITION MIND ONLY	
1428		14 VNET = VNETJ	
1429		BNET = BNETJ	
1430	C	TEST DOWN LOADS (NEGATIVE)	
1431		IF (VNET) 13.17.17	
1432	C		
1433		13 IF (VNET - BNETIN=11) 10.17.17	
1434	C		
1435		10 SAME (N=11) = VNET	
1436		BNETIN=11) = VNET	
1437		ISLPHIN=11) = 1	
1438		BNETIN=11) = 0 0	
1439		RCIN=11) = 1 0	
1440	C		
1441		17 IF (BNET) 10.00.00	
1442		10 IF (BNET - BNETIN=33) 10.00.00	
1443	C		
1444		10 SAME (N=33) = BNET	
1445		BNETIN=33) = BNET	
1446		ISLPHIN=33) = 1	
1447		BNETIN=33) = 0 0	
1448		RCIN=33) = 1 0	
1449		BNETIN=11) = 100:J)	
1450	C		
1451		00 CONTINUE	
1452	C		
1453		105 IF (IND) 23:105.105.107	
1454		100 IF (IND) 05:100.100.107	
1455		100 IF (IND) 77:105.105.107	
1456	C		
1457	C	**** LOOP 100 - - HORIZONTAL TAIL LOADS ****	
1458	C		
1459		107 00 10:0.0.0.0.7.0.7.0.0.0.0.7.7.0.0.7.7.0.0.0.0.0.0.105:1	
1460	C		
1461		0 JCS = 0C:10)	

07/03/74

HPLV LISTING

AUTOFLOW CHART SET - SHEEP

AIRLOADS OVERLAY

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CARGO NO      ****      COMMENTS      ****

1002          GO TO 10
1003          C
1004          7 HES = EC(15)
1005          GO TO 10
1006          C
1007          8 HES = EC(12)
1008          GO TO 10
1009          C
1010          9 HES = EC(10)
1011          C
1012          10 IF (HES(23)) 11,11,12
1013          11 IF (HES(20)) 100,100,12
1014          12 HES(17) = EC(107) * .057 * EC(100)
1015          C
1016          HTHZ = ES(10) - (ES(20) * (HES(17)-HES) / 200.4)
1017          C
1018          GO 100 H=1,11
1019          C
1020          111 = ES * (H-11)
1021          C
1022          J = 12 - H
1023          C
1024          VDET = (ES(111)) * VHI(J) * HTHZ * RSH
1025          SHET = (ES(111+1)) * SHH(J) * HTHZ * RSH
1026          C
1027          IF (VDET) 120,140,120
1028          C
1029          120 IF (VDET - SHET) (H-4) 1140,121,121
1030          C
1031          121 SAME (H-4) = ES(111)
1032          HDET(H-4) = VDET
1033          ISLPH(H-4) = 1
1034          HTHZ(H-4) = HTHZ / POSH2
1035          GO TO 140
1036          C
1037          130 IF (VDET - SHET) (H-5) 131,131,140
1038          C
1039          131 SAME (H-5) = ES(111)
1040          HDET(H-5) = VDET
1041          ISLPH(H-5) = 1
1042          HTHZ(H-5) = HTHZ / HES(12)
1043          C
1044          140 IF (SHET) 100,100,150
1045          C
1046          150 IF (SHET - HDET) (H-6) 150,151,151
1047          C
1048          151 SAME (H-6) = ES(111+1)
1049          C*****CCCCC*****
1050          SAME (H-6) = ES(111+2)
1051          C*****CCCCC*****
1052          HDET(H-6) = SHET
1053          ISLPH(H-6) = 1
1054          HTHZ(H-6) = HTHZ / POSH2
1055          GO TO 100
1056          C
1057          160 IF (SHET - HDET) (H-7) 161,161,100
1058          C
1059          161 SAME (H-7) = ES(111+1)
1060          C*****CCCCC*****
1061          SAME (H-7) = ES(111+2)
1062          C*****CCCCC*****
1063          HDET(H-7) = SHET
1064          ISLPH(H-7) = 1
1065          HTHZ(H-7) = HTHZ / HES(12)
1066          C
1067          100 CONTINUE
1068          C
1069          100 IF (HES(23)) 101,101,100
1070          101 IF (HES(27)) 100,100,100
1071          C
1072          C ***** LOOP END - - VERTICAL TAIL LOADS *****

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07/03/74      INPUT LISTING      AUTOFLOW CHART SET - SHEEP      AIRLOADS OVERLAY

CARD NO      ****      CONTENTS      ****

1633      C
1634      102 RECVT = BC(130) * .007 * BC(140)
1635      VINY = BD(110) - 100*211 * 100VY - RCO / 200 * 1
1636      DO 200 N=1,11
1637      C
1638      111 = 130 * (N-1) * 1
1639      C
1640      J = 12 - N
1641      C
1642      VNET = ABS(BD(111)) * VY(J) * VINY * RSV
1643      SHET = ABS(BD(111+1)) * SHV(J) * VINY * RSV
1644      C
1645      IF VNET - SHET IN (0,1,2,3,4,5,6,7)
1646      C
1647      201 SAVE(N+00) = ABS(BD(111))
1648      SAVE(N+00) = -SAVE(N+00)
1649      SHET(N+00) = VNET
1650      IURN(N+00) = 1
1651      IURN(N+00) = IURN(N+00)
1652      RQZ(N+00) = VINY / POSHZ
1653      RQZ(N+00) = RQZ(N+00)
1654      C
1655      240 IF SHET - SHET(N+110) 200,200,201
1656      C
1657      201 SAVE(N+110) = ABS(BD(111+1))
1658      SAVE(N+121) = -SAVE(N+110)
1659      C*****CCCCC*****
1660      SAVE(N+44) = BD(111+2)
1661      SAVE(N+50) = -SAVE(N+44)
1662      C*****CCCCC*****
1663      SHET(N+110) = SHET
1664      IURN(N+110) = 1
1665      IURN(N+121) = IURN(N+110)
1666      RQZ(N+110) = VINY / POSHZ
1667      RQZ(N+121) = RQZ(N+110)
1668      C
1669      200 CONTINUE
1670      C
1671      C      9-10-73--SETUP BD(1-200) FOR LOADS DATA SAVE ON RCD (100-103)*
1672      100 IF (1 - 9) 200,200,200
1673      200 BD(101) = BC(11)
1674      BD(103) = MD(103)
1675      C
1676      C      *TEST FOR LOAD 1-23
1677      IF (1 - 24) 201,200,202
1678      201 BD(102) = BD(11)
1679      IF (BD(102) - BD(101)) 205,205,205
1680      205 BD(104) = MD(104)
1681      BD(105) = MD(105)
1682      GO TO 207
1683      205 BD(104) = 0.0
1684      BD(105) = 0.0
1685      207 BD(106) = TSK(NF(1))
1686      BD(107) = TSK(NF(1))
1687      BD(108) = TSK(NF(1))
1688      GO TO 204
1689      C
1690      C      *LOAD 24--2-6 TAXI*
1691      202 BD(102) = BC(11)
1692      BD(104) = 0.0
1693      BD(105) = 0.0
1694      BD(106) = 0.0
1695      BD(107) = 0.0
1696      BD(108) = 0.0
1697      C
1698      C      *CLEAR LOADS PART OF BD ARRAY--BC(1-100)*
1699      DO 203 N=1,100
1700      BD(N) = 0.0
1701      203 CONTINUE
1702      C
1703      C      *SAVE BD(1-200) IN RCD 100-103 BASED ON INDICATOR (1)*

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424

CARD NO	CONTENTS	33000710
1775	K=100	33000710
1776	GO TO 80	33000710
1777	10 BO(120)=BC(145)	33000730
1778	BO(120)=BC(14)	33000730
1779	IF(ND(23)=40,40,20	33000737
1780	40 IF(ND(27)=41,41,20	33000740
1781	20 IF(ND(15)=21,20,23	33000750
1782	21 ID=0	33000760
1783	GO TO 24	33000760
1784	22 ID=4	33000760
1785	BC(1)=90*BO(15)	33000800
1786	GO TO 26	33000810
1787	23 ID=0	33000820
1788	24 BC(1)=BO(15)	33000830
1789	25 BC(2)=BC(144)+12	33000840
1790	BC(3)=BC(130)	33000850
1791	I=0	33000810
1792	DO 27 J=1,12	33000820
1793	K=1+J*120	33000830
1794	BO(K)=BC(2)*BU(J+173)/COS(BC(3)/57.3)	33000840
1795	27 I=I+1	33000850
1796	I=171	33000860
1797	J=130	33000870
1798	K=107	33000880
1799	20 IF(ID=0,100,30,30	33000890
1800	20 BC(5)=0.0	33001000
1801	GO TO 33	33001010
1802	30 BC(6)=BC(104)/COS(BC(3)/57.3)	33001015
1803	IF(BC(104)=BC(145)/20,30,30	33001020
1804	32 BC(5)=1.0	33001040
1805	33 BO(J)=BO(J)+BC(1)*BU(I)	33001050
1806	BO(J+1)=BO(J+1)+BC(1)*BU(I+1)+BC(5)*ABS(.30*BO(11)+BO(12))	33001060
1807	BO(J+2)=BO(J+2)+BC(1)*BU(I+2)	33001070
1808	GO 30 L=2,40,4	33001080
1809	M=J+L+2	33001100
1810	BO(M)=BO(M)+BC(1)*BU(K)	33001110
1811	IF(ID=0,137,34,37	33001120
1812	34 IF(BC(6)=BO(M)-1)35,30,30	33001130
1813	35 BC(5)=0.0	33001140
1814	GO TO 37	33001150
1815	35 BC(5)=1.0	33001160
1816	37 V1=COS(BC(3)/57.3)	33001181
1817	V2=SIN(BC(3)/57.3)	33001182
1818	BO(M+1)=BO(M+1)+BC(1)*BU(K+13)+BC(5)*ABS(.30*BO(11)+BO(12))*V1	33001170
1819	BO(M+2)=BO(M+2)+BC(1)*BU(K+26)+BC(5)*ABS(.30*BO(11)+BO(12))*V2	33001190
1820	30 K=K+1	33001200
1821	IF(ID=2,30,15,40	33001210
1822	30 IF(BC(5)=14,15,14	33001220
1823	40 IF(ID=4,10,41,41	33001230
1824	41 CONTINUE	33001240
1825	IF(10(51)9501,9501,9502	
1826	9501 CONTINUE	
1827	IF(ND(23)=40,40,50	33001250
1828	40 IF(ND(25)=51,51,50	33001260
1829	50 WRITE(6,42)BO(1),BO(10),BO(5),BO(5)	33001270
1830	42 FORMAT(1H,3X,8HEND LOADS,3X,8HEND NO=,F7.0,3X,30H, F6.3,	
1831	+ 3X,4HAT=,F7.0,3X,30H, F6.2,10X,20H SPAN = (P(5)) **)	
1832	WRITE(6,43)BO(1),1=23,20)	33001280
1833	43 FORMAT(1H,3X,8HNO DIST=,F7.2,2HIN,4X,10HIN AT SOB=,F6.0,2HLS,4X,	
1834	+ 10HIN AT SOB=,F12.0,5HIN-LB,4X,10HIN AT SOB=,F12.0,5HIN-LB,///	
1835	+ 13X,7HETATION,10X,5HMEAP,10X,8HEND MOM,21X,5HATORS MOM,	
1836	+ 10X,4H(1),21X,4H(LB),10X,5HIN-LB),20X,7HIN-LB,///	
1837	WRITE(6,44)BO(1),1=27,74)	33001290
1838	44 FORMAT(1H F10.2,F24.0,F20.0,F20.0)	33001301
1839	9502 CONTINUE	
1840	51 IF(10(51)9503,9503,9504	
1841	9503 CONTINUE	
1842	IF(ND(23)=52,52,53	
1843	52 IF(ND(26)=54,54,53	
1844	53 WRITE(6,45)BO(1),BO(10),BO(5)	33001300
1845	45 FORMAT(1H,3X,21HORIZONTAL TAIL LOADS,3X,8HEND NO=,F7.0,3X,	

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07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLORDS OVERLAY
CARD NO	CONTENTS		
1017	11 00(11)=0C(1,0)		31000000
1018	00(31)=0C(1,0)		31000000
1019	00(41)=0C(1,0)		31000000
1020	00 TO 10		31000010
1021	12 00(11)=0C(1,3)		31000020
1022	00(31)=0C(1,3)		31000030
1023	00(41)=0C(1,3)		31000040
1024	00 TO 10		31000050
1025	13 00(11)=0C(1,6)		31000060
1026	00(31)=0C(1,6)		31000070
1027	00(41)=0C(1,6)		31000080
1028	00 TO 10		31000090
1029	14 00(11)=0.0=0C(1,3)		31000100
1030	00(31)=0C(1,3)		31000110
1031	00(41)=0C(1,3)		31000120
1032	15 IF(CM)=1.0:10.20.20		31000130
1033	16 IF(CM)=.00:10.10.17		31000140
1034	17 00(17)=.00		31000150
1035	00 TO 10		31000160
1036	18 00(17)=CM		31000170
1037	19 00(2)=ATAN(17.0/05(4)/07.3)-11.0-00(3)/100(1)+11.0-00(3)/100(1)31000180		31000180
1038	11.0-00(17)+02)+07.3		31000190
1039	00(5)=CODE(00(17),07(0),07(1),10.1)		31000200
1040	00(6)=00(11)+00(5)		31000210
1041	L=0		31000220
1042	00 20 1=1.4		31000230
1043	K=14+000		31000240
1044	VA(1)=CODE(00(2),00(5),00(11),00(17),00(1),7.0,.5)		31000250
1045	00 L=44		31000260
1046	IF(10:02)10001,0001,0002		
1047	0001 CONTINUE		
1048	WRITE(6,21)00(11),00(21),00(31),00(5)		31000270
1049	21 FORMAT(1H,000,20H+ USPAH - (P(52) ****		
1050	1 11X,30VALUES FROM BCLAY TABLES FOR AB,FB,N,3X,40000,FB,2,3X,		31000280
1051	2 3HTR=77.4,3X,00000,FB,N)		31000290
1052	WRITE(6,22)07(1),1=0.0),VA(1),1=1.4)		31000300
1053	22 FORMAT(10,7X,3HTR=,4F10.2//11X,4F10.5)		31000310
1054	0002 CONTINUE		
1055	IF(10:32)20.20.27		31000320
1056	23 1=1		31000330
1057	IF(10:02)10021,0021,20		
1058	0021 CONTINUE		
1059	WRITE(6,24)		31000340
1060	24 FORMAT(1H,03X,0HINVERT)		31000350
1061	00 TO 20		31000360
1062	25 1=100		31000370
1063	IF(10:02)10520,0050,20		
1064	0050 CONTINUE		
1065	WRITE(6,26)		31000380
1066	26 FORMAT(1H,03X,10HINVERT TAIL)		31000390
1067	00 TO 20		31000400
1068	27 1=107		31000410
1069	IF(10:02)10523,0053,20		
1070	0053 CONTINUE		
1071	WRITE(6,28)		31000420
1072	28 FORMAT(1H,03X,10HINVERT TAIL)		31000430
1073	29 00(11)=CODE(00(3),07(5),VA(4),5)+07.3/00(5)		31000440
1074	L=0		31000450
1075	00 30 1=1.10		31000460
1076	K=14+13		31000470
1077	VA(1)=CODE(00(2),00(5),00(11),00(17),00(1),7.0,.5)		31000480
1078	30 L=44		31000490
1079	31 00 32 1=1.4		31000500
1080	L=11+14+1		31000510
1081	32 VA(1)=CODE(00(3),07(5),VA(1),4,.5)		31000520
1082	IF(10:02)10003,0003,40		
1083	0003 CONTINUE		
1084	WRITE(6,33)CM		31000530
1085	33 FORMAT(10,10X,30VALUES FROM LOADING TABLES FOR MM,FB,3)		31000540
1086	J=1		31000550
1087	K=4		31000560

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1000	L=1		31000000
1001	34 WRITE(10,35)YB(1),1-J,K) ,DT(L)		31000070
1002	35 FORMAT(1H0,10X,4F10.0,3X,4D5F4,70 3)		31000071
1003	J=J+4		31000072
1004	K=K+4		31000074
1005	L=L+1		31000075
1006	IF(L-1)34,34,40		31000076
1007	30 IF(C00-1)1137,30,30		31000080
1008	37 DS(17)=1		31000084
1009	GO TO 30		31000085
1010	38 DS(17)=C00		31000086
1011	30 DS(7)=DS(1)+SQRT(DS(17)**2-1 0)		31000087
1012	IF(DS(4)) 105,105,104		31000089
1013	104 DS(8) = SQRT(DS(17)**2-1 0)*COS(DS(4)/57 3)/SIN(DS(4)/57 3)		31000090
1014	IF(12 0 - DS(8)) 105,107,107		31000091
1015	105 DS(8) = 12 0		31000092
1016	107 L = 0		31000700
1017	DO 40 I=1,10		31000710
1018	K=L+14		31000720
1019	YB(1)=C000+DS(8)*DS(7)*DP(5)*DP(1)*DP(1)*0.0 5)		31000730
1020	40 L=L+4		31000740
1021	IF(10.52)9505,9505,3)		
1022	9505 CONTINUE		
1023	WRITE(10,41)DS(1),DS(3),DS(7),DS(8)		31000742
1024	41 FORMAT(1H1,7X,3F40.7 3X,3D5F4,7 3X,3D5F4,70 3X,3D5F4,		31000744
1025	1 F0 4,30X,20H** USPH - (P(52) **)		31000745
1026	WRITE(10,42)DT(1),1-5,0)		31000746
1027	42 FORMAT(1H0,7X,3F40.7 3X,3D5F4,70 3X,3D5F4,		31000748
1028	IF(NT-3)43,44,45		31000749
1029	43 WRITE(10,44)		31000750
1030	GO TO 31		31000761
1031	44 WRITE(10,45)		31000763
1032	GO TO 31		31000764
1033	45 WRITE(10,46)		31000765
1034	GO TO 31		31000767
1035	46 IF(NT-4)47,47,48		31000768
1036	47 L=5		31000770
1037	DO 48 I=1,4		31000780
1038	48 ED(1)=DT(1)		31000780
1039	ED(5)=1 0		31000800
1040	YC(5)=0 0		31000810
1041	GO TO 50		31000820
1042	49 L=3		31000830
1043	ED(1)=0 0		31000840
1044	ED(2)= 9+2		31000850
1045	ED(3)=1 0		31000860
1046	50 DO 55 I=1,13		31000870
1047	IF(NT-5)51,52,53		31000880
1048	51 K=1+0		31000890
1049	GO TO 54		31000900
1050	52 K=1+14		31000910
1051	GO TO 54		31000920
1052	53 K=1+175		31000930
1053	54 ES(1)=DS(K)		31000940
1054	55 YS(1)=C000+ES(1)*ED*YC(L 5)		31000950
1055	IF(10.52)9507,9507,9508		
1056	9507 CONTINUE		
1057	WRITE(10,56)CHN		31000952
1058	56 FORMAT(1H1,10X,70STATION,10X,20HLOADING AT DATA STATIONS,5X,30H**		31000953
1059	1 F0 3,30X,20H** USPH - (P(52) **)		31000954
1060	WRITE(10,57)ED(1),YC(1),1-1,1)		31000955
1061	57 FORMAT(1H1 7 5)		31000956
1062	WRITE(10,58)		31000958
1063	58 FORMAT(1H1,10X,70STATION,10X,20HLOADING-ANALYSIS STATIONS)		31000957
1064	IF(NT-3)59,60,61		31000960
1065	59 WRITE(10,62)		31000960
1066	GO TO 62		31000960
1067	60 WRITE(10,63)		31000961
1068	GO TO 62		31000962
1069	61 WRITE(10,64)		31000963
1070	62 WRITE(10,57)ES(1),YS(1),1-1,1)		31000964

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADS OVERLAY
CARD NO	****	CONTENTS	****
0000	0000 CONT MAX		
0000	05(01)=0.0		31000000
0001	00 03 1-0.12		31000070
0002	03 05(01)=05(01)+ 5*(05(11)-11-05(11))*(05(11)-11)*05(11)		31000000
0003	00(00) = 0 5*(05(12) - 05(13))*(05(12) + 05(13))		31000001
0004	05(01) = 05(01) + 00(00)*00(00)		31000002
0005	00(00) = 05(01) + 11 0 - 00(00)*00(00)		31000003
0006	IF(CM) = 1 01 100.04.04		31000000
0007	04 IF(NT)=300.00.07		31001000
0008	00 04(11)=05(11)*05(01)		31001010
0009	00 70 00		31001020
0070	00 04(100)=05(11)*05(01)		31001030
0071	00 70 00		31001040
0072	07 04(107)=05(11)*05(01)		31001050
0073	00 70 00		31001051
0074	100 IF(NT) = 31 101.102.103		31001052
0075	101 04(11) = 04(11)*05(01)*00(00)		31001053
0076	00 70 00		31001054
0077	102 04(100) = 04(100)*05(01)*00(00)		31001055
0078	00 70 00		31001056
0079	103 04(107) = 04(107)*05(01)*00(00)		31001057
0080	00 IF(NT)=3100.70.70		31001058
0081	00 1-33		31001070
0082	J=02		31001080
0083	K=35		31001090
0084	L=00		31001100
0085	05(13)=00(37)		31001110
0086	IF(IP)71,70,71		31001120
0087	70 05(11)=00(72)		31001130
0088	05(12)=00(70)+0.0		31001140
0089	00 70 75		31001150
0090	71 05(11)=00(141)		31001160
0091	05(12)=00(140)+0.0		31001170
0092	00 70 75		31001180
0093	72 1-45		31001190
0094	IF(NT)=173.74.74		31001200
0095	73 J=100		31001210
0096	K=141		31001220
0097	L=104		31001230
0098	05(11)=00(100)		31001240
0099	05(12)=00(112)+0.0		31001250
0100	05(13)=00(113)		31001260
0101	00 70 75		31001270
0102	74 05(11)=00(114)		31001280
0103	05(12)=00(114)+12.0		31001290
0104	05(13)=00(115)		31001300
0105	J=107		31001310
0106	K=200		31001320
0107	L=013		31001330
0108	75 05(11)=05(13)+05(12)		31001340
0109	05(10)=00(100)*CM,07(01),07(11),12.51		31001350
0110	76 04(1)=0.0		31001360
0111	04(12)=0.0		31001370
0112	04(13)=0.0		31001380
0113	00(11)=0.0		31001390
0114	IF(NT)=3177.00.01		31001400
0115	77 IF(IP)70,70,70		31001402
0116	78 1-70		31001404
0117	J=00		31001405
0118	00 70 00		31001406
0119	79 1-30		31001407
0120	J=07		31001408
0121	00 70 00		31001409
0122	00 1-100		31001410
0123	J=105		31001411
0124	01 1-120		31001412
0125	J=107		31001413
0127	00 05(10)=4.0*(1.0-05(11)+05(100))/07 31*05(00(11)/07 31/05(11)+1 031001410		31001410
0128	1-05(11)		31001410
0129	00(10)=05(10)+11 0-05(10)+11 0-05(10)+11		31001420

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
0001	0011 CONTINUE		
0002	WRITE(6,100)CYN,0010		3100174
0003	100 FORMAT(1H1,00H,20H** UPRN = (P102) **//		
0004	11H,7H1010H,10H,0H010 LEND,10H,0H100 LEND,10H,0H10P INCR,		3100176
0005	2 5H,30H**70 3,5H,30H**70 21		3100177
0006	0012 CONTINUE		
0007	00 107 1-1,13		3100178
0008	J=10		3100179
0009	VS(1)=CODE(0U(1),05(10),07(10),07(10),07(10),10,11,5)		3100180
0010	VS(1)=CODE(0U(1),05(10),07(10),07(10),07(10),10,11,5)		3100180
0011	IF(10:02)0013,0013,0014		
0012	0013 CONTINUE		
0013	WRITE(6,100)VS(1),VS(1),VS(1)		3100177
0014	100 FORMAT(1H17,0,710,0,710,0)		3100177
0015	0014 CONTINUE		
0016	VS(1)=VS(1)-VS(1)		3100180
0017	IF(10:02)0015,0015,0016		
0018	0015 CONTINUE		
0019	WRITE(6,100)VS(1)		3100180
0020	100 FORMAT(1H34H10,0)		3100180
0021	0016 CONTINUE		
0022	107 CONTINUE		3100180
0023	00(0)=0		3100180
0024	00 100 1-2,12		3100180
0025	100 00(0)=05(0)+0*(05(1)-1)-(05(1)+VS(1)-1)+VS(1)		3100181
0026	00(0)=0.5*(05(12)+05(13))+(VS(12)+VS(13))		3100181
0027	00(0)=05(0)+00(00)+00(00)		3100182
0028	00(00)=05(0)+(1.0-00(00)+00(00))		3100183
0029	0U(0)=00(0)		3100183
0030	05(0)=CODE(0C(102),07(11),07(0),0,5)		3100183
0031	J=00		3100184
0032	K=00		3100184
0033	L=00		3100184
0034	00 TO 70		3100184
0035	100 1-2		3100184
0036	J=0		3100184
0037	K=00		3100184
0038	L=0		3100184
0039	00 TO 114		3100184
0040	110 1-02		3100184
0041	J=0		3100184
0042	K=00		3100184
0043	L=00		3100184
0044	00 TO 114		3100184
0045	111 IF(07-0)112,113,113		3100184
0046	112 1-100		3100184
0047	J=115		3100184
0048	K=120		3100184
0049	L=112		3100184
0050	00 TO 114		3100184
0051	113 1-100		3100184
0052	J=170		3100184
0053	K=107		3100184
0054	L=171		3100184
0055	114 0U(1)=CODE(05(14),0U(1),0U(1),13,5)		3100184
0056	0U(1)=CODE(05(14),0U(1),0U(1),13,5)		3100184
0057	0U(1)=CODE(05(14),0U(1),0U(1),20,13,5)		3100184
0058	0U(1)=0U(1)*COS(0C(11)/57.3)+0U(2)*SIN(0C(11)/57.3)		3100210
0059	0U(2)=0U(1)*COS(0C(11)/57.3)-0U(1)*SIN(0C(11)/57.3)		3100210
0060	0U(1)=0U(1)		3100210
0061	IF(07-2)115,116,121		3100210
0062	115 1-3		3100210
0063	K=7		3100210
0064	00 TO 117		3100210
0065	116 1-03		3100210
0066	K=04		3100210
0067	117 IF(07)119,119,119		3100210
0068	118 J=70		3100210
0069	00 TO 120		3100210
0070	119 J=30		3100210
0071	120 0U(1)=05(13)+0U(1)+0U(1)+3)		3100210

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07/03/74	INPUT LISTING	AUTOFLEX CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	*****	CONTENTS	*****
014	INB = 2		
015	00 TO 4010		
016	C		
017	4005 IF INB(20) 14007,4007,4003		
018	C		
019	4003 L1 = 45		
020	L2 = 00		
021	C*****CCCCC*****		
022	00 4013 IN=1.00		
023	4013 BLPH(N=100) = 0.0		
024	C*****CCCCC*****		
025	INB = 3		
026	00 TO 4010		
027	C		
028	4007 IF INB(27) 14009,4009,4004		
029	C		
030	4004 L1 = 00		
031	L2 = 100		
032	C*****CCCCC*****		
033	00 4014 IN=1.00		
034	4014 BLPH(N=170) = 0.0		
035	C*****CCCCC*****		
036	INB = 4		
037	C		
038	4010 00 4011 IN=1.12		
039	4011 BLPH(N) = 0.0		
040	C		
041	00 TO 4005,4005,4007,4009,1400		
042	C		
043	4000 CONTINUE		
044	C		
045	C IF THE NEGATIVE BENDING MOMENT AT THE ROOT OF THE HORIZONTAL		
046	C TAIL IS GREATER THAN THE POSITIVE BENDING MOMENT, REVERSE THE		
047	C HORIZONTAL TAIL POSITIVE AND NEGATIVE AIRLOADS (ARRAY BLN IN		
048	C RECORD 32) AND THE RATIOS (ARRAY RATIO IN RECORD 17)		
049	C		
050	10410C(40) = 0.0		
051	C		
052	IF INB(20) 14007,4007,400		
053	C		
054	400 IF 10410C(77) = 10410C(20) 1401,4007,400		
055	C		
056	401 10410C(40) = 1.0		
057	C		
058	00 403 IN=0.50		
059	C		
060	REBLME = RE(10)		
061	RE(10) = RE(10+1)		
062	RE(10+1) = REBLME		
063	C		
064	REBLME = RE(10+20)		
065	RE(10+20) = RE(10+33)		
066	RE(10+33) = REBLME		
067	C		
068	REBLAY = RE(10)		
069	RE(10) = RE(10+1)		
070	RE(10+1) = REBLAY		
071	C		
072	REBLAY = RE(10+20)		
073	RE(10+20) = RE(10+33)		
074	RE(10+33) = REBLAY		
075	C		
076	RAT100 = RATIO(10)		
077	RATIO(10) = RATIO(10+1)		
078	RATIO(10+1) = RAT100		
079	C		
080	RAT100 = RATIO(10+20)		
081	RATIO(10+20) = RATIO(10+33)		
082	RATIO(10+33) = RAT100		
083	C		
084	BLNE'V = BLME(N)		

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLANS OVERLAY
CARD NO	CONTENTS		
0405	SAVE(10) = -SAVE(10+1)		
0406	SAVE(10+1) = -SAVE(10)		
0407	C		
0408	SAVE(10) = SAVE(10+2)		
0409	SAVE(10+2) = -SAVE(10+3)		
0410	SAVE(10+3) = -SAVE(10)		
0411	C		
0412	SAVE(10) = BLP(10)		
0413	BLP(10) = -BLP(10+1)		
0414	BLP(10+1) = -SAVE(10)		
0415	C		
0416	SAVE(10) = BLP(10+2)		
0417	BLP(10+2) = -BLP(10+3)		
0418	BLP(10+3) = -SAVE(10)		
0419	C		
0420	ISDH = (BLP(10))		
0421	ISDH(1) = (BLP(10+1))		
0422	ISDH(1+1) = ISDH		
0423	C		
0424	ISDH = (BLP(10+2))		
0425	ISDH(10+2) = (BLP(10+3))		
0426	YES ISDH(10+3) = (ISDH)		
0427	C*****CCCCC*****		
0428	C		
0429	C *** T. MEN ***		
0430	DO VDSH = 1,11		
0431	SAVE(10) = SAVE(10+2)		
0432	SAVE(10+2) = SAVE(10+3)		
0433	VDSH SAVE(10+3) = SAVE(10)		
0434	C		
0435	C*****CCCCC*****		
0436	C		
0437	VDSH CONTINUE		
0438	C		
0439	C **** DETERMINE THE DESIGN LOAD CONDITIONS FOR THE WIND. ****		
0440	C **** HORIZONTAL TAIL AND VERTICAL TAIL (IM, IM, IV) ****		
0441	C **** THE DESIGN CONDITION IS THE CONDITION WHICH GIVES ****		
0442	C **** THE MAXIMUM POSITIVE BENDING MOMENT AT THE ROOT ****		
0443	C		
0444	C **** DETERMINE THE STRESS LEVEL AT THE DESIGN CONDITION ****		
0445	C **** (STRE,STRE,STRE) AND THE TEMPERATURE AT THE DESIGN ****		
0446	C **** CONDITION (TWIND,THT,TVT) FOR THE WIND, HORIZONTAL ****		
0447	C **** AND VERTICAL ****		
0448	C		
0449	IF (IND1) 201,405,405,402		
0450	C		
0451	VDSH IM = (BLP(10+3))		
0452	STRE = STRE(10+1)		
0453	TWIND = TWIND(10+1)		
0454	THT(10+1) = TWIND		
0455	C		
0456	DO VDSH = 1,40		
0457	II = (BLP(10))		
0458	IF (II) - 201,200,207,207		
0459	207 IF (ISDH) 210,210,200		
0460	C		
0461	200 RS(10) = STRE / 1000		
0462	GO TO 213		
0463	200 IF (II) 210,210,214		
0464	214 IF (TWIND) 210,210,211		
0465	211 IF (STRE) 210,210,212		
0466	210 RS(10) = 1.0		
0467	GO TO 213		
0468	210 RS(10) = STRE / STRE(10+1)		
0469	213 BLP(10) = SAVE(10) + RS(10)		
0470	RATIO(10) = RS(10) + RS(10)		
0471	VDSH RATIO(10+2) = RATIO(10) + RS(10)		
0472	C*****CCCCC*****		
0473	C		
0474	DO VDSH = 1,22		
0475	BLP(10+2) = SAVE(10+2) + RS(10+2)		

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CARD NO      ****      CONTENTS      ****

0007      WRITE(6,3011)
0008      301 FORMAT(//2X,01HSTATION LC      +V      RS      RQZ      RC
0009      *      RS*(+V)      RS+RQZ      RS+RQZ+RC/)
0010      C
0011      WRITE(6,3001)(N,IDLPH(N),SAVE(N),RS(N),RQZ(N),RC(N),IDLPH(N),RATIO(N),
0012      *      RATIO(N+100),N=1,11)
0013      C
0014      300 FORMAT(2X,12,5X,12,F13.0,F0.3,F0.3,F13.0,F0.3,F0.3)
0015      C
0016      WRITE(6,3002)
0017      302 FORMAT(//2X,01HSTATION LC      -V      RS      RQZ      RC
0018      *      RS*(+V)      RS+RQZ      RS+RQZ+RC/)
0019      C
0020      WRITE(6,3003)(N,IDLPH(N+11),SAVE(N+11),RS(N+11),RQZ(N+11),RC(N+11),
0021      *      IDLPH(N+11),RATIO(N+11),RATIO(N+100),N=1,11)
0022      C
0023      WRITE(6,3003)
0024      303 FORMAT(//2X,01HSTATION LC      +BN      RS      RQZ      RC
0025      *      RS*(+BN)      RS+RQZ      RS+RQZ+RC/)
0026      C
0027      WRITE(6,3004)(N,IDLPH(N+20),SAVE(N+20),RS(N+20),RQZ(N+20),RC(N+20),
0028      *      IDLPH(N+20),RATIO(N+20),RATIO(N+100),N=1,11)
0029      C
0030      WRITE(6,3004)
0031      304 FORMAT(//2X,01HSTATION LC      -BN      RS      RQZ      RC
0032      *      RS*(+BN)      RS+RQZ      RS+RQZ+RC/)
0033      C
0034      WRITE(6,3005)(N,IDLPH(N+33),SAVE(N+33),RS(N+33),RQZ(N+33),RC(N+33),
0035      *      IDLPH(N+33),RATIO(N+33),RATIO(N+100),N=1,11)
0036      C
0037      400 IF (NOD(00))=000,400,402
0038      C
0039      400 WRITE(6,3041)
0040      C
0041      WRITE(6,3042)
0042      304 FORMAT(3X,23H*** HORIZONTAL TAIL ***)
0043      C
0044      WRITE(6,3011)
0045      301 FORMAT(//2X,01HSTATION LC      +V      RS      RQZ      RS*(+
0046      *      V)      RS+RQZ/)
0047      C
0048      WRITE(6,3006)(N,IDLPH(N+44),SAVE(N+44),RS(N+44),RQZ(N+44),
0049      *      IDLPH(N+44),RATIO(N+44),N=1,11)
0050      C
0051      306 FORMAT(2X,12,5X,12,F13.0,F0.3,F0.3,F13.0,F0.3)
0052      C
0053      WRITE(6,3002)
0054      302 FORMAT(//2X,01HSTATION LC      -V      RS      RQZ      RS*(+
0055      *      V)      RS+RQZ/)
0056      C
0057      WRITE(6,3007)(N,IDLPH(N+50),SAVE(N+50),RS(N+50),RQZ(N+50),
0058      *      IDLPH(N+50),RATIO(N+50),N=1,11)
0059      C
0060      WRITE(6,3003)
0061      303 FORMAT(//2X,01HSTATION LC      +BN      RS      RQZ      RS*(+
0062      *      BN)      RS+RQZ/)
0063      C
0064      WRITE(6,3008)(N,IDLPH(N+60),SAVE(N+60),RS(N+60),RQZ(N+60),
0065      *      IDLPH(N+60),RATIO(N+60),N=1,11)
0066      C
0067      WRITE(6,3004)
0068      304 FORMAT(//2X,01HSTATION LC      -BN      RS      RQZ      RS*(+
0069      *      BN)      RS+RQZ/)
0070      C
0071      WRITE(6,3009)(N,IDLPH(N+77),SAVE(N+77),RS(N+77),RQZ(N+77),
0072      *      IDLPH(N+77),RATIO(N+77),N=1,11)
0073      C
0074      404 IF (NOD(02))=000,400,400
0075      C
0076      400 WRITE(6,407)
0077      407 FORMAT(//7X,72HTHE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL T

```

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INPUT LISTING

AUTOFLOW CHART SET - SHEEP AIRLOADS OVERLAY

```

CAND NO      ****      CONTENTS      ****

2700          *ALL HAVE BEEN REVERSED/INX, BECAUSE THE NEGATIVE ON AT THE ROPT
2700          *WAS GREATER THAN THE POSITIVE ON AT THE ROPT)
2701          C
2701          *MS IF (ND*27) 3002, 3002, 472
2702          C
2703          *72 WRITE(6, 244)
2704          C
2705          *WRITE(6, 243)
2705          *43 FORMAT(30N, 21N*** VERTICAL TAIL ***)
2707          C
2709          *WRITE(6, 261)
2710          C
2710          *WRITE(6, 260) (N, 10LPHN+00), SAME(N+00), RS(N+00), RZ(N+00),
2711          *    * 0LPHN+00), RATIO(N+00), N=1, 11)
2712          C
2713          *WRITE(6, 262)
2714          C
2715          *WRITE(6, 260) (N, 10LPHN+00), SAME(N+00), RS(N+00), RZ(N+00),
2716          *    * 0LPHN+00), RATIO(N+00), N=1, 11)
2717          C
2718          *WRITE(6, 263)
2719          C
2720          *WRITE(6, 260) (N, 10LPHN+10), SAME(N+10), RS(N+10), RZ(N+10),
2721          *    * 0LPHN+10), RATIO(N+10), N=1, 11)
2722          C
2723          *WRITE(6, 264)
2724          C
2725          *WRITE(6, 260) (N, 10LPHN+12), SAME(N+12), RS(N+12), RZ(N+12),
2726          *    * 0LPHN+12), RATIO(N+12), N=1, 11)
2727          C
2728          *0000 CONTINUE
2729          C
2730          C      **** WRITE THE MIND, MY AND VY LOADS IN RECORD 32 ****
2731          C      **** AND WRITE THE HEAD HEIGHT RATIOS ON RECORD 17 ****
2732          C      **** IF ANY OF THE LOADS HAVE BEEN COMPUTED ****
2733          C
2734          IF (ND*25) 3001, 3001, 3004
2735          3001 IF (ND*26) 3000, 3002, 3004
2736          3002 IF (ND*26) 3003, 3003, 3004
2737          3003 IF (ND*27) 3005, 3005, 3004
2738          C
2739          C*****CCCC*****
2740          3004 CALL WRITE(1, 0LPH(1), 100, 32)
2741          CALL WRITE(1, RATIO(1), 204, 17)
2742          C*****CCCC*****
2743          3005 CONTINUE
2744          RETURN
2745          END

```

Appendix B

SAMPLE OUTPUT

Sample airload module output are shown in Figures B-1 through B-25. Print control indicators and originating routine name appear in the output samples.

** USPAN - IP(52) **
 VALUES FROM RCLAK TABLES FOR AR= 9.6000 SWB= 12.94 TR= 0.3500 MA/K= 10.0036
 TR= 0.0 0.25 0.50 1.00
 0.09438 0.09651 0.09526 0.09119 (WING)
 VALUES FROM LOADING TABLES FOR MN= 0.318
 1.53218 1.32994 1.21470 1.04169 STA= 0.0
 1.26499 1.17613 1.14092 1.11271 STA= 0.393
 0.73471 0.86268 0.93744 1.04151 STA= 0.707
 0.24414 0.50330 0.60127 0.69093 STA= 0.924

Figure B-1. Sample output from USPAN of wing lift curve slope parameter and spanwise loading parameter.

```

STATION      0.0      0.38300      0.70700      0.92400      1.09000
LOADING AT DATA STATIONS      MN= 0.318
1.27001
1.15001
0.89577
0.55048
0.0

LOADING-ANALYSIS STATIONS (WING)
0.0
0.17758
0.60454
0.74000
0.86827
0.97262
1.05040
1.11372
1.16937
1.20887
1.23799
1.26030
1.27901

NY SWEEP
-9.852
-10.266
-11.508
-13.164
-14.820
-16.476
-18.132
-19.788
-21.444
-23.100
-24.756
-26.412
-4.870

STATION      0.0      0.38300      0.70700      0.92400      1.09000
LOADING AT DATA STATIONS      MN= 0.318
1.27001
1.15001
0.89577
0.55048
0.0

LOADING-ANALYSIS STATIONS (WING)
0.0
0.17758
0.60454
0.74000
0.86827
0.97262
1.05040
1.11372
1.16937
1.20887
1.23799
1.26030
1.27901

NY SWEEP
-9.852
-10.266
-11.508
-13.164
-14.820
-16.476
-18.132
-19.788
-21.444
-23.100
-24.756
-26.412
-4.870

```

Figure B-2. Sample output from USPAN of wing spanwise loading parameter and local center of pressure at load evaluation stations.

USPAN - IP(52)

STATION	QUTR LDNG	INBD LDNG	FLAP INCR	MN= 0.310	DF= 30.00
1.02000	0.0	0.0	0.0		
0.97737	0.06278	0.01386	0.04892		
0.90949	0.24822	0.05030	0.19791		
0.81897	0.48070	0.07124	0.40946		
0.72846	0.71176	0.09669	0.62071		
0.63795	0.88693	0.13111	0.75582		
0.54744	0.99040	0.17987	0.81952		
0.45693	1.08007	0.23407	0.84600		
0.36642	1.13946	0.29832	0.84114		
0.27590	1.18983	0.41185	0.77798		
0.18539	1.22320	0.61159	0.61161		
0.09488	1.23994	0.81019	0.42975		
0.0	1.23953	0.81140	0.42813		

STATION	OX SWEFF
1.02000	6.560
0.97737	6.836
0.90949	7.663
0.81897	8.766
0.72846	9.869
0.63795	10.971
0.54744	12.074
0.45693	13.177
0.36642	14.280
0.27590	15.383
0.18539	16.485
0.09488	17.588
0.0	19.171

Figure B-3. Sample output from USPAN of deflected flap spanwise loading parameter and local center of pressure at load evaluation stations.

```

** USPAN - IP(52) **

VALUES FROM BCLA/K TABLES FOR AR= 2.00000 SWB= 37.73 TR= 0.5000 RA/K= 2.0041
TR= 0.0 0.25 0.50 1.00
0.04268 0.04315 0.04270 0.03980 (HDR TAIL)

VALUES FROM LOADING TABLES FOR MN= 0.318
1.32855 1.26743 1.23203 1.16170 STA= 0.0
1.20663 1.18044 1.17246 1.16334 STA= 0.383
0.86047 0.90957 0.92676 0.96722 STA= 0.707
0.40948 0.49930 0.51510 0.54820 STA= 0.924

```

Figure B-4. Sample output from USPAN of horizontal tail lift curve slope parameter and spanwise loading parameter.

```

STATION      0.0      0.34300      0.70700      0.92400      1.00000
LOADING AT DATA STATIONS      MN= 0.318
1.23203
1.17246
0.92676
0.51510
0.0

LOADING--ANALYSIS STATIONS (FOR TAIL)
0.0
0.18215
0.60436
0.78914
0.93660
1.04091
1.11003
1.16416
1.20282
1.22204
1.23035
1.23203
1.23203

DX SWEEP
-18.648
-19.155
-20.557
-22.475
-24.294
-26.163
-28.032
-29.900
-31.769
-33.638
-35.507
-37.376
-37.376

STATION      0.0      0.34300      0.70700      0.92400      1.00000
LOADING AT DATA STATIONS      MN= 0.318
1.23203
1.17246
0.92676
0.51510
0.0

LOADING--ANALYSIS STATIONS (FOR TAIL)
0.0
0.18215
0.60436
0.78914
0.93660
1.04091
1.11003
1.16416
1.20282
1.22204
1.23035
1.23203
1.23203

DX SWEEP
-18.648
-19.155
-20.557
-22.475
-24.294
-26.163
-28.032
-29.900
-31.769
-33.638
-35.507
-37.376
-37.376

```

Figure B-5. Sample output from USPAN of horizontal tail spanwise loading parameter and local center of pressure at load evaluation stations.

```

** USPAN - IP(52) **

VALUES FROM RCLA/K TABLES FOR AR= 1.7761  SWB= 57.59  TR= 0.4056  BA/K= 1.8507

TR= 0.0  0.25  0.50  1.00
0.03591  0.03528  0.03400  0.03152  (VERT TAIL)

VALUES FROM LOADING TABLES FOR MN= 0.316

1.30547  1.22059  1.16836  1.06486  STA= 0.0
1.20517  1.17735  1.16165  1.15456  STA= 0.383
0.86897  0.93583  0.96552  1.02152  STA= 0.707
0.43151  0.52829  0.54977  0.58320  STA= 0.924

```

Figure B-6. Sample output from USPAN of vertical tail lift curve slope parameter and spanwise loading parameter.

USPAN - IP(52) **

MN= 0.31A

LOADING AT DATA STATIONS

STATION
0.0
0.3A300
0.70700
0.92400
1.03000

1.1A002
1.16640
0.95452
0.541A
0.0

LOADING-ANALYSIS STATIONS (VERT TAIL)

STATION
1.00000
1.07000
0.94336
0.88672
0.83000
0.77345
0.71681
0.66017
0.60353
0.54690
0.49026
0.43301
0.0

0.0
0.0
0.41943
0.66816
0.77313
0.85962
0.94074
1.00911
1.05612
1.09102
1.12119
1.14662
1.1A002

OK SWEEP

STATION
1.03000
1.09000
0.94336
0.88672
0.83000
0.77345
0.71681
0.66017
0.60353
0.54690
0.49026
0.43301
0.0

-26.81A
-26.81A
-20.043
-31.760
-33.495
-35.721
-37.947
-40.173
-42.300
-44.624
-46.850
-49.100
52.500

Figure B-7. Sample output from USPAN of vertical tail spanwise loading parameter and local center of pressure at load evaluation stations.

```

WING PARAMETERS
CLA= 4.73664    KW(R)A=0.87844    VM(R)A= 592.13    DXW(B)A= 243.58    DXB(W)A= 147.04
                                ** USPAN - IP(52) **

SIDE OF BODY UNITS
USZWR= 0.87844    UMXWR= 417.379    UMYWR= -60.949

SPANWISE UNIT DISTRIBUTIONS
STA      USZW(B)    UMXW(R)    UMYW(R)    MN= 0.318
SWEEP    SWEEP      SWEEP      SWEEP

1.00000    0.0      0.0      0.0
0.0774    0.00703    0.070    0.021
0.0005    0.07002    1.342    0.322
0.0100    0.00067    8.127    1.000
0.7705    0.16457    77.620    2.120
0.6300    0.24964    46.003    3.440
0.5474    0.34102    70.565    5.040
0.4560    0.43505    121.904    6.926
0.3664    0.54406    170.772    9.076
0.2750    0.65262    247.721    11.495
0.1854    0.76436    328.170    14.170
0.0949    0.87844    421.450    17.090
0.0      1.00000    533.267    18.970

```

Figure B-8. Sample output from USPAN of wing lift curve slope, center of pressure, and unit spanwise loads.

```

FLAPS DEFLECTED PARAMETERS                                ** USPAN - IP(52) **
KRF= 0.59234      KW(R)F=0.93120      YW(R)F= 594.52      DXW(R)F= 276.91      DXB(W)F= 191.09
SIDE OF BODY UNITS
UISZFR= 0.93129      UMXFR= 444.712      UMYFR= -95.654
SPANWISE UNIT DISTRIBUTIONS
STA      USZF(R)      UMXF(R)      UMYF(R)      MN= 0.31R
SWEEP
1.0000    0.0      0.0      0.0      0.0
0.9774    0.00003    0.013    -0.006
0.9095    0.01508    0.695    -0.112
0.9190    0.06148    5.042    -0.503
0.7285    0.14019    16.494    -1.245
0.6380    0.24536    38.385    -2.346
0.5474    0.36572    73.083    -3.736
0.4569    0.40296    121.840    -5.344
0.3664    0.62186    185.142    -7.113
0.2759    0.74557    262.786    -8.945
0.1854    0.85173    353.493    -10.630
0.0949    0.93129    454.726    -11.977
7.0      1.00000    569.679    -13.926

```

Figure B-9. Sample output from USPAN of deflected flap lift curve clope, center of pressure, and unit spanwise loads.

USPAN - 1P(52) **

HOR TAIL PARAMETERS
 CLA= 2.32056 KM(R)=1.00000 YH= 100.84 DXH= 135.20
 SIDE OF BODY UNITS
 USZHR= 1.00000 UMXHR= 100.845 UMYHR= -29.067

SPANWISE UNIT DISTRIBUTIONS

STA SWEEP	USZH(R)	UMXH(R) SWEEP	UMYH(R) SWEEP	MN= 0.318
1.0000	0.0	0.0	0.0	
0.9750	0.00229	0.007	0.044	
0.9000	0.03199	0.310	0.645	
0.8000	0.10214	1.890	2.161	
0.7000	0.18902	5.319	4.197	
0.6000	0.28857	10.945	6.714	
0.5000	0.39685	19.018	9.651	
0.4000	0.51134	29.717	12.970	
0.3000	0.63050	43.167	16.646	
0.2000	0.75257	59.458	20.639	
0.1000	0.87604	78.642	24.908	
0.0	1.00000	100.741	29.426	
0.0	1.00000	100.741	29.426	

Fig B-10. Sample output from USPAN of horizontal tail lift curve slope, center of pressure, and unit spanwise loads.

USPAN - IP(52)

VERT TAIL PARAMETERS

CVR= 1.89612 KV(R)=0.62286 NZV= 196.37 DXV= 501.46

TOP OF BODY UNITS

USVVR= 0.52296 UMXVR= 94.472 UMZVR= -89.641

SPANWISE INIT DISTRIBUTIONS

STA SWEEP	USVV(R)	UMXV(R) SWEEP	UMZV(R) SWEEP	MV= 0.318
1.0000	0.0	0.0	0.0	
1.0000	0.0	0.0	0.0	
0.9434	0.01202	0.238	0.349	
0.8967	0.04218	1.321	1.297	
0.8301	0.08448	3.860	2.638	
0.7734	0.13127	8.134	4.259	
0.7168	0.18284	14.355	6.162	
0.6602	0.23873	22.706	8.347	
0.6035	0.29790	33.335	10.791	
0.5469	0.35946	46.356	13.471	
0.4903	0.42287	61.851	16.372	
0.4330	0.48856	80.110	19.524	
0.0	1.00000	305.512	18.168	

Figure B-11. Sample output from USPAN of vertical tail lift curve slope, center of pressure, and unit spanwise loads.

```

CONDITION NO= 60R25.      MN= 0.318      ALT= 0.      NF= 30.00      ** BNLD5 - IP(50) **

BODY LOADS
PZN= 28637.      PYN= 0.      XRN= 652.82

WING PANEL LOAD
PZW(R)/2= 801953.      YRW(B)= 592.98      XRW(B)= 2163.54

WING CARRY-OVER LOAD
PZA(W)= 185100.      XRA(W)= 2077.75

HORIZONTAL TAIL LOADS
PZH/?= -68R26.      YRH=100.84      XRH= 3225.20      DMXH= 2082216.

VERTICAL TAIL LOAD
PVV= 0.      ZRV= 0.0      XRV= 2866.75

AIRPLANE INERTIA FACTORS
NZ= 2.00      NY= 0.0      ODOT= 0.0      RDOT= 0.0

COMPONENT SPANWISE FACTORS
PZW(R)A= 103373A.      PZW(R)F= 570167.      PZW(B)=-137651.

```

Figure B-12. Sample output from BNLD5 of vehicle loads, center of pressure, and inertia factors.

WING LOADS COND NO= 6025. MN= 0.318 ALT= 0. DF= 30.00 ** SPARM - IP(51) **

SNR DIST= 117.00IN SW AT SOB= 801953.LB BM AT SOB=381715200.IN-LB TM AT SOB=-65142720.IN-LB

STATION (IN)	SHEAR (LB)	BEND MCM (IN-LB)	TORS MOM (IN-LB)
1254.67	0.	0.	0.
1226.24	1479.	20994.	10291.
1141.11	21571.	1002612.	154764.
1027.55	72172.	6325479.	487559.
913.98	139747.	19358560.	871371.
800.42	221405.	34865296.	1310488.
686.86	312606.	69187200.	1827445.
573.29	409707.	110201152.	2439199.
459.73	510493.	162450896.	3162884.
346.17	612225.	226199728.	4025576.
232.61	710470.	301304064.	5083688.
119.04	901953.	397181568.	6388989.

Figure B-13. Sample output from SPARM of wing airload shears, bending moments, and torques.

HORIZONTAL TAIL LOADS				COND NO= 60825.	MN= 0.318	ALT= 0.	** SPARM - IP(51) **	
SOR DIST=	0.0 IN	SM AT SOR=	-79149.LR	BM AT SOB=	-7981819.IN-LB	TM AT SOR=	2300645.IN-LR	
STATION	(IN)	SHEAR	(LR)	BEND	MOM	TORS	MOM	
				(IN-LR)		(IN-LB)		
235.59		0.		0.		0.		
229.70		-191.		-534.		-3476.		
212.03		-2532.		-24506.		-51030.		
188.47		-8084.		-149560.		-171048.		
164.91		-14961.		-421019.		-322228.		
141.35		-22840.		-866290.		-531398.		
117.79		-31410.		-1505326.		-763888.		
94.23		-40472.		-2352056.		-1026575.		
70.68		-49904.		-3416626.		-1317536.		
47.12		-59566.		-4706107.		-1633594.		
23.56		-69338.		-6274507.		-1971462.		
0.0		-79149.		-7973588.		-2329016.		

Figure B-14. Sample output from SPARM of horizontal tail airload shears, bending moments, and torques.

VERTICAL TAIL LOADS COND NO= 60825. MN= 0.318 ALT= 0. ** SPARM - IP(51) **
 SOR DIST= 132.17IN SH AT SOB= 0.18 BM AT SOB= 2082216.IN-LB TM AT SOB= 0.IN-LB

STATION (IN)	SHEAR (LB)	BEND MOM (IN-LB)	TORS MOM (IN-LB)
699.44	0.	0.	0.
699.44	0.	0.	0.
659.82	0.	1238860.	1673573.
620.21	0.	1238860.	1673573.
580.50	0.	1238860.	1673573.
540.09	0.	1238860.	1673573.
501.36	0.	1238860.	1673573.
461.75	0.	1238860.	1673573.
422.13	0.	1238860.	1673573.
382.52	0.	1238860.	1673573.
342.90	0.	1238860.	1673573.
302.86	0.	1238860.	1673573.

Figure B-15. Sample output from SPARM of vertical tail airload shears, bending moments, and torques.

*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) *** ** WINGNET - IP(53) **

*** WING ***

STATION	LC	+V	RS	RNZ	RC	RS*(-V)	RSRNZ	RSRNZ*RC
1	3	2131.	1.000	1.000	1.000	2131.	1.000	1.000
2	3	32043.	1.000	1.000	1.000	32043.	1.000	1.000
3	3	101635.	1.000	1.000	1.000	101635.	1.000	1.000
4	3	190435.	1.000	1.000	1.000	190435.	1.000	1.000
5	3	263650.	1.000	1.000	1.000	263650.	1.000	1.000
6	3	350229.	1.000	1.000	1.000	350229.	1.000	1.000
7	3	438337.	1.000	1.000	1.000	438337.	1.000	1.000
8	3	526503.	1.000	1.000	1.000	526503.	1.000	1.000
9	3	614028.	1.000	1.000	1.000	614028.	1.000	1.000
10	5	834107.	0.768	1.000	1.000	640469.	0.768	0.768
11	5	957934.	0.768	1.000	1.000	735550.	0.768	0.768

STATION	LC	-V	RS	RNZ	RC	RS*(-V)	RSRNZ	RSRNZ*RC
1	7	-275.	0.768	1.000	1.000	-671.	0.768	0.768
2	7	-12456.	0.768	1.000	1.000	-9564.	0.768	0.768
3	7	-39316.	0.768	1.000	1.000	-30189.	0.768	0.768
4	7	-71552.	0.768	1.000	1.000	-54941.	0.768	0.768
5	7	-108303.	0.768	1.000	1.000	-83161.	0.768	0.768
6	7	-148739.	0.768	1.000	1.000	-114209.	0.768	0.768
7	7	-192000.	0.768	1.000	1.000	-147428.	0.768	0.768
8	7	-237581.	0.768	1.000	1.000	-182427.	0.768	0.768
9	7	-284980.	0.768	1.000	1.000	-218822.	0.768	0.768
10	7	-333643.	0.768	1.000	1.000	-256198.	0.768	0.768
11	7	-383174.	0.768	1.000	1.000	-294220.	0.768	0.768

Figure B-16. Sample output from WINGNET of wing design airload shear envelope and normalizing factors.

STATION	LC	+BM	RS	RNZ	RC	RS*(+BM)	RS*RNZ	RS*RNZ*RC
1	3	30250.	1.000	1.000	1.000	30250.	1.000	1.000
2	3	1485321.	1.000	1.000	1.001	1485321.	1.000	1.001
3	3	9074139.	1.000	1.000	1.000	9074139.	1.000	1.000
4	3	25087216.	1.000	1.000	1.000	25087216.	1.000	1.000
5	3	50297696.	1.000	1.000	1.000	50297696.	1.000	1.000
6	3	85147456.	1.000	1.000	1.000	85147456.	1.000	1.000
7	3	129913898.	1.000	1.000	1.000	129913898.	1.000	1.000
8	3	184687216.	1.000	1.000	1.000	184687216.	1.000	1.000
9	3	249434528.	1.000	1.000	1.000	249434528.	1.000	1.000
10	3	324058624.	1.000	1.000	1.000	324058624.	1.000	1.000
11	3	408404224.	1.000	1.000	0.992	408404224.	1.000	0.992

STATION	LC	-BM	RS	RNZ	RC	RS*(-BM)	RS*RNZ	RS*RNZ*RC
1	7	-12414.	0.768	1.000	1.000	-9532.	0.768	0.768
2	7	-580099.	0.768	1.000	1.000	-445429.	0.768	0.768
3	7	-3519777.	0.768	1.000	1.000	-2702661.	0.768	0.768
4	7	-9815010.	0.768	1.000	1.000	-7536458.	0.768	0.768
5	7	-20027408.	0.768	1.000	1.000	-15379052.	0.768	0.768
6	7	-34622624.	0.768	1.000	1.000	-26584992.	0.768	0.768
7	7	-53970304.	0.768	1.000	1.000	-41441104.	0.768	0.768
8	7	-78362560.	0.768	1.000	1.000	-60170704.	0.768	0.768
9	7	-108034288.	0.768	1.000	1.000	-82954160.	0.768	0.768
10	7	-143160544.	0.768	1.000	1.000	-109975872.	0.768	0.768
11	7	-183962400.	0.768	1.000	1.000	-141178800.	0.768	0.768

Figure B-17. Sample output from WINGNET of wing design airload bending moment envelope and normalizing factors.

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*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) ***
** WHVNET - 1P(53) **

*** HORIZONTAL TAIL ***

STATION LC      +V      RS      RNZ      RS*(+V)  RS*RNZ
1         3      303.    1.000   -2.500    303.    -2.500
2         3     4142.    1.000   -2.500   4142.    -2.500
3         3    12863.    1.000   -2.500   12863.    -2.500
4         3    23237.    1.000   -2.500   23237.    -2.500
5         3    34889.    1.000   -2.500   34889.    -2.500
6         3    47419.    1.000   -2.500   47419.    -2.500
7         3    60552.    1.000   -2.500   60552.    -2.500
8         3    74108.    1.000   -2.500   74108.    -2.500
9         3    87884.    1.000   -2.500   87884.    -2.500
10        3   101709.    1.000   -2.500  101709.    -2.500
11        3  115483.    1.000   -2.500 115483.    -2.500

STATION LC      -V      RS      RNZ      RS*(-V)  RS*RNZ
1         7     -67.    0.768   -0.400    -52.    -0.307
2         7    -930.    0.768   -0.400   -721.    -0.307
3         7   -3000.    0.768   -0.400  -2303.    -0.307
4         7   -5555.    0.768   -0.400  -4265.    -0.307
5         7   -8484.    0.768   -0.400  -6514.    -0.307
6         7  -11671.    0.768   -0.400  -8962.    -0.307
7         7  -15042.    0.768   -0.400 -11550.    -0.307
8         7  -18551.    0.768   -0.400 -14244.    -0.307
9         7  -22146.    0.768   -0.400 -17005.    -0.307
10        7  -25783.    0.768   -0.400 -19797.    -0.307
11        7  -29434.    0.768   -0.400 -22601.    -0.307

```

Figure B-18. Sample output from WHVNET of horizontal tail design airload shear envelope and normalizing factors.

STATION	LC	+RM	RS	RNZ	RS*(+RM)	RS*RNZ
1	3	893.	1.000	-2.500	893.	-2.500
2	3	40168.	1.000	-2.500	40168.	-2.500
3	3	240479.	1.000	-2.500	240479.	-2.500
4	3	665714.	1.000	-2.500	665714.	-2.500
5	3	1350397.	1.000	-2.500	1350397.	-2.500
6	3	2319931.	1.000	-2.500	2319931.	-2.500
7	3	3591762.	1.000	-2.500	3591762.	-2.500
8	3	5177965.	1.000	-2.500	5177965.	-2.500
9	3	7096125.	1.000	-2.500	7096125.	-2.500
10	3	9319415.	1.000	-2.500	9319415.	-2.500
11	3	11877795.	1.000	-2.500	11877795.	-2.500

STATION	LC	-RM	RS	RNZ	RS*(-RM)	RS*RNZ
1	7	-108.	0.768	-0.400	-152.	-0.307
2	7	-9087.	0.768	-0.400	-6978.	-0.307
3	7	-55484.	0.768	-0.400	-42603.	-0.307
4	7	-156252.	0.768	-0.400	-119978.	-0.307
5	7	-321615.	0.768	-0.400	-246953.	-0.307
6	7	-559075.	0.768	-0.400	-429247.	-0.307
7	7	-873685.	0.768	-0.400	-670859.	-0.307
8	7	-1269386.	0.768	-0.400	-974698.	-0.307
9	7	-1749773.	0.763	-0.400	-1347795.	-0.307
10	7	-2313342.	0.768	-0.400	-1776300.	-0.307
11	7	-2963753.	0.768	-0.400	-2275718.	-0.307

THE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL TAIL HAVE BEEN REVERSED
BECAUSE THE NEGATIVE RM AT THE ROOT WAS GREATER THAN THE POSITIVE RM AT THE ROOT

Figure B-19. Sample output from MHNUT of horizontal tail design airload bending moment envelope and normalizing factors.

*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) ***

*** VERTICAL TAIL ***

STATION LC	+V	RS	RNZ	RS*(+V)	RS*RNZ
1	0.	1.000	1.000	0.	1.000
2	1880.	1.000	0.141	1880.	0.141
3	6663.	1.000	0.141	6663.	0.141
4	12830.	1.000	0.141	12830.	0.141
5	19659.	1.000	0.141	19659.	0.141
6	27074.	1.000	0.141	27074.	0.141
7	35015.	1.000	0.141	35015.	0.141
8	43361.	1.000	0.141	43361.	0.141
9	51991.	1.000	0.141	51991.	0.141
10	60831.	1.000	0.141	60831.	0.141
11	69934.	1.000	0.141	69934.	0.141

STATION LC	-V	RS	RNZ	RS*(-V)	RS*RNZ
1	0.	1.000	1.000	0.	1.000
2	-1880.	1.000	0.141	-1880.	0.141
3	-6663.	1.000	0.141	-6663.	0.141
4	-12830.	1.000	0.141	-12830.	0.141
5	-19659.	1.000	0.141	-19659.	0.141
6	-27074.	1.000	0.141	-27074.	0.141
7	-35015.	1.000	0.141	-35015.	0.141
8	-43361.	1.000	0.141	-43361.	0.141
9	-51991.	1.000	0.141	-51991.	0.141
10	-60831.	1.000	0.141	-60831.	0.141
11	-69934.	1.000	0.141	-69934.	0.141

Figure B-20. Sample output from WHVNET of vertical tail design airload shear envelope and normalizing factors.

STATION	LC	+RM	RS	RN7	RS*(-RM)	RS*RNZ
1	0	0.	1.000	1.000	0.	1.000
2	3	146969A.	1.311	0.0	1926856.	0.0
3	3	146969A.	1.311	0.0	1926856.	0.0
4	3	146969A.	1.311	0.0	1926856.	0.0
5	3	146969A.	1.311	0.0	1926856.	0.0
6	19	2706599.	1.000	0.141	2706599.	0.141
7	19	3936427.	1.000	0.141	3936427.	0.141
8	19	5488839.	1.000	0.141	5488839.	0.141
9	19	737751A.	1.000	0.141	737751A.	0.141
10	19	9612231.	1.000	0.141	9612231.	0.141
11	19	12230438.	1.000	0.141	12230438.	0.141
STATION	LC	-RM	RS	RN7	RS*(-RM)	RS*RNZ
1	0	0.	1.000	1.000	0.	1.000
2	3	-146969A.	1.311	0.0	-1926856.	0.0
3	3	-146969A.	1.311	0.0	-1926856.	0.0
4	3	-146969A.	1.311	0.0	-1926856.	0.0
5	3	-146969A.	1.311	0.0	-1926856.	0.0
6	19	-2706599.	1.000	0.141	-2706599.	0.141
7	19	-3936427.	1.000	0.141	-3936427.	0.141
8	19	-5488839.	1.000	0.141	-5488839.	0.141
9	19	-737751A.	1.000	0.141	-737751A.	0.141
10	19	-9612231.	1.000	0.141	-9612231.	0.141
11	19	-12230438.	1.000	0.141	-12230438.	0.141

Figure B-21. Sample output from WAMNET of vertical tail design airload bending moment envelope and normalizing factors.

** BLCNTL - IP(54) **										
LOAD CONDITION	ALTITUDE FEET	PACH NUMBER	PRESSURE PSI	LOCAL TEMP DEC F	TOTAL TEMP DEC F	SUN FLUX BTU/HR/FT2	SKIN TEMP DEC F	STRESS WING	STRESS HORIZONTAL	STRESS VERTICAL
1 +NZ BALANCED	0	0.6100	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
2 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
3 +NZ BALANCED	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.
4 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
5 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
6 -NZ BALANCED	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
7 -NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
8 FLAP DOWN MANEUVER	0	0.5234	14.70	518.7	530.2	356.6	534.7	66000.	66000.	66000.
9 FLAP DOWN 1 G TRIM	0	0.1809	14.70	518.7	522.2	356.6	531.4	66000.	66000.	66000.
10 + VERTICAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
11 + VERTICAL GUST	20000	0.6100	6.77	447.3	406.6	425.0	507.6	66000.	66000.	66000.
12 + VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
13 + VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
14 - VERTICAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
15 - VERTICAL GUST	20000	0.6100	6.75	447.3	506.6	425.0	507.6	66000.	66000.	66000.
16 - VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
17 - VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
18 LATERAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
19 LATERAL GUST	20000	0.6100	6.75	447.3	406.6	425.0	507.6	66000.	66000.	66000.
20 PITCH ACCELERATION	0	0.6000	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
21 PITCH ACCELERATION	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.
22 YAW ACCELERATION	0	0.6100	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
23 YAW ACCELERATION	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.

DESIGN TEMPERATURE CONDITION				
WING				
HORIZONTAL	80.0		3	
VERTICAL	80.0		15	
	80.0		23	

*** MAXIMUM NET BENDING MOMENTS FOR FATIGUE ***				
FM AT SOP	CONDITION	NZ	INERTIA FM/NZ	NET FM AT SOP
140137616.	3	2.50	-28266704.	69470864.
FM AT STA 2	CONDITION	NZ	INERTIA FM/NZ	NET FM AT STA 2
124261152.	3	2.50	-22037552.	4916720.

DESIGN TEMPERATURE CONDITION

WING	CONDITION	
	HORIZONTAL	VERTICAL
	80.0	80.0
	80.0	80.0
	80.0	80.0

*** MAXIMUM NET BENDING MOMENTS FOR FATIGUE ***

FM AT SOP	CONDITION	NZ	INERTIA BM/NZ	NET BM AT SOP
140127616.	3	2.50	-28266704.	69470864.
FM AT STA 2	CONDITION	NZ	INERTIA FM/NZ	NET FM AT STA 2
124241152.	3	2.50	-22037552.	49167210.

Figure B-22. Sample output from BLCNTL of ambient properties equilibrium skin temperature, structure compression yield strength, design conditions, and maximum net bending moments at wing fatigue evaluation stations.

SPECTRA SEGMENT NO 1.

SNF	RFND	WM	WNS	RFND	WM	EXCEEDANCES-GUST	EXCEEDANCES-MAN
208964320.	252434912.	0.72702E-42	0.0	0.72702E-42	0.0	0.0	0.0
192467120.	232874256.	0.50260E-37	0.0	0.50260E-37	0.0	0.0	0.0
175969920.	212913600.	0.34746E-32	0.0	0.34746E-32	0.75000E-01	0.75000E-01	0.75000E-01
159472736.	192952928.	0.24020E-27	0.0	0.24020E-27	0.42500E+00	0.42500E+00	0.42500E+00
142375600.	179992320.	0.16605E-22	0.0	0.16605E-22	0.22500E+01	0.22500E+01	0.22500E+01
126478400.	153031664.	0.11480E-17	0.0	0.11480E-17	0.10000E+02	0.10000E+02	0.10000E+02
109991216.	133071008.	0.79363E-13	0.0	0.79363E-13	0.47500E+02	0.47500E+02	0.47500E+02
92484016.	117110336.	0.54866E-08	0.0	0.54866E-08	0.26250E+03	0.26250E+03	0.26250E+03
76986816.	93149680.	0.37930E-03	0.0	0.37930E-03	0.36750E+04	0.36750E+04	0.36750E+04
60489680.	73189072.	0.26765E+02	0.0	0.26765E+02	0.17500E+06	0.17500E+06	0.17500E+06
49401536.	59891952.	0.26765E+02	0.0	0.26765E+02	0.20000E+05	0.20000E+05	0.20000E+05
37904352.	39921296.	0.37930E-03	0.0	0.37930E-03	0.34000E+03	0.34000E+03	0.34000E+03
16407184.	19960640.	0.54866E-08	0.0	0.54866E-08	0.75000E+00	0.75000E+00	0.75000E+00
0.	0.	0.79363E-13	0.0	0.79363E-13	0.25000E-02	0.25000E-02	0.25000E-02
-16497184.	-19960640.	0.11480E-17	0.0	0.11480E-17	0.0	0.0	0.0
-32004352.	-39921296.	0.16606E-22	0.0	0.16606E-22	0.0	0.0	0.0
-49401536.	-59891952.	0.24020E-27	0.0	0.24020E-27	0.0	0.0	0.0
-65998704.	-79842592.	0.34746E-32	0.0	0.34746E-32	0.0	0.0	0.0
-82485904.	-99803248.	0.50260E-37	0.0	0.50260E-37	0.0	0.0	0.0
-98383104.	-119763904.	0.72702E-42	0.0	0.72702E-42	0.0	0.0	0.0

1G SF M = 540060A. 1G WS M = 66535504.

Figure B-23. Sample output from FATMG of flight segment wing bending moment spectra.

SPECTRA SEGMENT NC '0.

SOF BEND MOM	WJS BEND MOM	EXCEEDANCES-TAXI
-51436116.	-39048752.	0.10000E+02
-50125256.	-32647000.	0.50000E+02
-48770544.	-37845720.	0.25000E+01
-46061072.	-35743164.	0.75000E+00
-43351000.	-33640056.	0.22500E+02
-40642128.	-31530112.	0.50000E+03
-37932640.	-29435552.	0.11250E+05
-35223168.	-27333024.	1.22500E+06
-32513596.	-25230430.	0.21500E+07
-29804224.	-23127952.	0.62500E+07
-24305264.	-19922064.	0.92500E+07
-21615792.	-16810320.	0.21500E+07
-18906320.	-14711725.	0.22500E+06
-16256651.	-12615245.	0.11250E+05
-13547375.	-10512704.	0.50000E+03
-10837900.	-8410162.	0.22500E+02
-8128425.	-6307622.	0.75000E+00
-5418950.	-4205081.	0.25000E+01
-4064212.	-3155310.	0.50000E+02
-2709475.	-2102541.	0.10000E+02

Figure B-24. Sample output from FATMG of taxi segment wing bending moment spectra.

** FATMG - IP(55) **

SPECTRA SEGMENT NO 11.

GROUND-AIR-GROUND CYCLES

SUF PERD MCM NOS BEND MCM
 80134250. 310470.
 -04502304. -50140500.

OCCURRENCES
 0.25400E+05
 0.25400E+05

Figure B-25. Sample output from FATMG of ground-air-ground segment wing bending moment spectra.