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THESIS

PERSISTENT PERIMETER SURVEILLANCE USING MULTIPLE SWAPPING MULTI-ROTOR UAS

by

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September 2018

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PERSISTENT PERIMETER SURVEILLANCE USING MULTIPLE SWAPPING MULTI-ROTOR UAS

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ABSTRACT

This study evaluated the feasibility of swapping a system of small multi-rotor unmanned aerial systems during flight for persistent perimeter surveillance. The systems engineering processes, such as requirement analysis and functional analysis, were followed by designing the overall concept of system and identifying high-level requirements for it. The research also included system prototyping. The developed system prototype consisted of three COTS products: multiple quadcopters, a single router, and a laptop running Python code. Quadcopters were programmed to fly in different predefined patterns over the Military Operation at Urban Environment Training site at Impossible City, CA, and McMillan Field at Camp Roberts, CA. Videos were recorded during the tests. While one quadcopter was flying, the remaining two stayed on stand-by. Once the airborne quadcopter depleted its battery life to the predetermined level, one of the standby quadcopters was activated as a replacement. The process can be repeated continuously, assuring an uninterrupted video stream. By monitoring the battery level and autonomously swapping quadcopters, the system showed that endurance can exceed the capability of a single quadcopter and possibly perform 24/7 surveillance, or until the system fails mechanically.

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LIST OF ACRONYMS AND ABBREVIATIONS

API	Application Programming Interface
CCTV	Closed Circuit Television
COTS	commercial off-the-shelf
DARPA	Defense Advanced Research Projects Agency
DoD	Department of Defense
DODAF	Department of Defense Architecture Framework
FAA	Federal Aviation Administration
FFBD	Functional Flow Block Diagram
FY	fiscal year
FV	field of view
GCS	ground control system
GHz	Gigahertz
GPS	global positioning system
ISR	Intelligence, Surveillance, and Reconnaissance
MAVLink	micro air vehicle link
MOE	Measure of Effectiveness
MOP	Measures of Performance
NPS	Naval Postgraduate School
SV	System View
SUAS	small multi-rotor UAS
TCP/IP	Transmission Control Protocol/Internet Protocol
UAS	unmanned aerial systems
UAV	unmanned aerial vehicle
UDP	User Datagram Protocol
USB	Universal Serial Bus

EXECUTIVE SUMMARY

Unmanned systems are an effective tool in military and commercial applications, such as surveying. As unmanned systems get more autonomous, they promise to offer distinct advantages for persistent perimeter patrol, as these systems can be deployed more rapidly and extensively than humans. Thus, the focus of this thesis is to explore the feasibility and limitations of using a small multi-rotor unmanned aerial system (SUAS) to provide a persistent perimeter patrol by swapping SUAS autonomously.

The research involves the prototyping of a system composed of commercial offthe-shelf (COTS) products, including 3DR Solo quadcopters, a single router, and a laptop running Python software. The objective of the proof-of-concept (POC) is to show that an SUAS on patrol can be swapped seamlessly with another SUAS while maintaining a consistent field of view (FOV) during the swap. Figure 1 shows the swapping process in this POC. The first SUAS, shown in red, is replaced by the green SUAS, while the yellow SUAS is replaced by the green SUAS when its battery is depleted below the threshold.



battery level UAS fly back to launch point

Figure 1. SUAS Swapping Process

Low battery level UAS charges for next launch.

The results show that nine out of ten trials at Impossible City, California, and at Camp Roberts, California, were successful, with one failure due to inconsistent FOV. This occurred because the SUAS being replaced was performing a 180-degree turn at the time of the swap. The FOV, however, recovered within half a second.

The system of UAS proved to be a feasible solution to provide persistent surveillance. By monitoring the battery level and autonomously replacing it with a ground control station, the system showed that its endurance can exceed the capability of a single UAS, and the SUAS can possibly perform 24/7 persistent surveillance, or until the system mechanically fails.

The solution proposed can be extended to other surveillance or target tracking tasks where persistent surveillance or targeting is required. The autonomous operations would also relieve security personnel from mundane tasks and so they could focus on more high value work. With further development on the operation effectiveness and suitability of this system, the military forces would benefit in terms of mission effectiveness and manpower savings from this technology.

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I. INTRODUCTION

A. BACKGROUND

The number of applications utilizing unmanned aerial systems (UAS) has been growing rapidly in the past few years (Joshi 2017). With Federal Aviation Administration (FAA) help in granting exemptions for industries like agriculture, insurance, and construction to operate UAS, a wide range of applications for these industries has opened. The FAA reported that there were 110,000 registered commercial, non-model¹ UAS in 2017 and expects the average annual growth rate to be 32.5 percent (Office of Aviation Policy and Plans 2017). The usage of UAS for military applications has also increased at a tremendous rate. According to a study conducted by Bard College, the Department of Defense (DoD) has requested \$9.39 billion in UAS-related funding for fiscal year (FY) 2019, which is 26 percent more than for FY 2018 (Gettinger 2017).

From facilitating the delivery of goods during traffic congestion to surveying the most remote areas, UAS have proven to be an effective tool to support operations, with little to no manpower. In its unmanned system integration roadmap report, the DoD explained that UAS have not only provided improved situational awareness, they have proven capable of reducing human workload, as well as risk to personnel and cost (Undersecretary of Defense Acquisition 2014).

Dr. Graham Drozeski, program manager in the Tactical Technology Office of the Defense Advanced Research Projects Agency (DARPA), has shared that effective 21st century warfare requires persistent collection of airborne intelligence, surveillance, and reconnaissance not possible for current technologies like helicopters, which are limited by distance and flight time, nor fixed wing aircraft, which require runways (DARPA 2017).

There has been new interest in many organizations to explore the use of small and inexpensive electric propulsion UAS. A small UAS provides lower capital cost, reduced carbon footprint, is quieter to operate and faster to deploy as compared to a larger UAS

¹ Non-Model refers to commercial, government, or other non-hobby purpose UAS.

like the Predator. While the Predator class has a longer endurance period than a small UAS, it is expensive to both procure and operate. Furthermore, a small UAS can operate at a lower altitude, which eliminates possible airspace sharing conflicts in the controlled airspace. LTC Brett Clark of the United States Army discusses the advantages of the Department of Homeland Security using a small UAS over the Predator UAS for southern border Intelligence, Surveillance, and Reconnaissance (ISR) requirements (Clark 2013).

The U.S. Marine Corps has also begun fielding small multi-rotor UAS (SUAS), by InstantEye Robotics, to its infantry units and will be ordering 800 InstantEye systems (Fuentes 2018; Kohlhepp 2018). The SUAS could help squads to see over buildings or hills to assess the risks and threats ahead (Fuentes 2018).

In the area of perimeter surveillance, Lee et al. (2015) discussed the potential of using SUAS for perimeter surveillance, which would address the disadvantages of human patrols that are limited in numbers available to patrol and in duration due to human fatigue. Apart from this, SUAS can be deployed more rapidly and extensively than humans, and are not restricted by any obstacles on the ground. Lee et al. also shared how using SUAS is a better option compared to installing Closed Circuit Television (CCTV) cameras that are typically mounted on fixed infrastructure with limited view. It is also cheaper and faster to implement an SUAS solution as compared to installing CCTVs for perimeter security. This is because CCTVs require the installation of long power and network cables as well as ground trenching for the cables, which would take more labor, money, and time. SUAS can overcome these constraints, which translates to a substantial cost savings. SUAS could also provide flexibility in surveying any area of interest and have the ability to respond to any incident site faster than a security patrol could to assess the risk before deploying any security personnel. SUAS could also provide more points of view of the situation so that more vulnerabilities could be identified than would normally be seen.

While there are many advantages in using a small SUAS, there is still no wide adoption of SUAS for perimeter surveillance. This could be due to the endurance of SUAS, which is typically around 30 minutes. This limits the type of mission that the UAS could carry out. Therefore, many research efforts have been done to extend the UAS's duration and range (Lance 2011; Gudmundsson 2016). The advantages of using small UAS were discussed in the preceding paragraphs. Nonetheless, small UAS are still limited by battery life which is typically around 15 to 30 minutes. This limits the type of mission that the UAS could carry out. Therefore, many research efforts have attempted to extend the UAS's duration and range (Zou et al. 2015; Chin 2011; Koumadi et al. 2017; Gudmundsson 2016).

A lightweight SUAS capable of hovering for 40 minutes was reported by Zou et al. (2015). Their research focused on extending the duration and target position accuracy of the UAS for a search and rescue mission. Carbon fiber materials were used to keep the weight to around five pounds. This may not be a sustainable solution as there is a limit to how much the weight can be reduced using carbon fiber material. Furthermore, the main challenge in extending the flight duration, as highlighted by Zou et al., is still limited by the current battery technology.

Another method to extend UAS duration and range is to use atmospheric wind as an energy source, which was proposed by Gudmundsson (2016). Gudmundsson showed that almost 50 percent of the energy lost could be recovered in a complex, multi-heading, multi-altitude mission. The proposed method, however, may only be applicable to areas with windy conditions, which may not be practical in actual deployment. A similar method of harvesting natural energy is by using solar cells, as discussed by Chin (2011). This technology also faces a similar limitation in that it is dependent on the availability of an energy source, in this case, the sun.

An alternative to Gudmundsson's and Chin's methods is to use a rotational energy harvesting device installed on the UAS. In the research of Koumadi et al. (2017), a brushless DC generator is used as a rotor to fly the UAS and to recharge the batteries on the UAS. The Koumadi report claims that the generator was able to extend flight duration by at least 42 percent. While this solution has greater potential than the previous methods, the improvement in the flight duration still poses a limitation to longer missions like perimeter surveillance.

More recent methods have included hybrid fuel cell technology and laser beam charging technology. The hybrid fuel cell technology uses fuel to convert electricity, which

is used as the energy source for the UAS. In an experiment conducted by Rees (2017), the UAS was able to achieve 4.5 hours of flight duration and was claimed that it would continue to be improved. Laser-beam charging uses a laser beam to wirelessly charge the UAS while it is flying. But this approach is still in its infancy and could only transfer 20 percent of electricity from the ground to the UAS. Furthermore, it may also require the UAS to be stationary or to move at a slower speed to charge (Whittle 2012).

There are many companies like Skyfront and Quaternium that use hybrid-electric technology, which uses fuel to produce electricity to extend the endurance of the UAS to four hours. Even so, the UAS will still need to be recharged, leading to surveillance gaps. This may be unacceptable in a situation where the UAS is tracking an intruder. Another company, Elistair, uses tethering of the UAS to extend the power source from the ground to the UAS, which could provide 24/7 continuous surveillance. This is, however, limited in range and mobility. In one case, the U.S. military has also used an unmanned robotic vehicle, shown in Figure 1, to patrol its camp in Horn of Africa to overcome the limitation of flight endurance in UAS (Reid 2016). Nevertheless, the unmanned vehicle is limited in its maneuverability, especially in built-up areas or in challenging terrain. These challenges also limit the response time of the vehicle to incident areas.



Figure 1. Unmanned Surveillance Vehicle. Source: Reid (2016).

In experiment by Bethke (2007), multiple UAS were used to track a single moving target. The replacement of UAS is done sequentially, which means that one UAS would

fly back before the swapping UAS is launched. Since there are multiple UAS, the swapping of one UAS would not impact the current tracking as the tracking is still being done by the remaining UAS. While this method would ensure the tracking of target is not lost, it may not be a cost effective solution to have multiple UAS track a target. Furthermore, it would add complexity to UAS management as there is a need to ensure all UAS do not fully deplete their battery life at the same time.

B. THE NEED FOR PERSISTENT PERIMETER SURVEILLANCE

The intent of the system being developed and described in this thesis is to provide persistent surveillance for perimeter security using a commercial off-the-shelf (COTS) system. Persistent surveillance is achieved by swapping small multi-rotor UAS autonomously and continuously without having any surveillance gaps or interruption. The monitoring and management of the SUAS is done by a ground control system (GCS). The GCS is used to monitor the system health and perform air traffic control during the SUAS swapping.

The focus of this thesis is not to prove that a specific SUAS used in this thesis can be deployed for perimeter surveillance, but rather to show the feasibility of using an aerial platform for perimeter surveillance. Furthermore, the research does not focus on changing the battery automatically or having wireless charging, but rather looks at what is currently available on the market.

The proposed system is intended to show how SUAS could complement the existing perimeter patrol tasks in a military installation. The system would be launched by a user to aid in 24/7 patrolling tasks. This is done by replacing the patrolling SUAS with a new one when its battery level is low. The solution could also be used as a first responder to any incident area within the perimeter and provide persistent tracking of the intruder.

C. RESEARCH APPLICATION

The thesis focuses on determining the feasibility of using a system of SUAS for persistent surveillance of a given perimeter. Using low-cost COTS quadcopters as a proof of concept, this thesis evaluates the capabilities and challenges of swapping SUAS while SUAS is moving. Exploring such persistent technologies would allow the military to minimize the number of patrollers and the installation of new surveillance infrastructure while also minimizing any coverage gaps. The solution would therefore be very applicable to DoD needs as an economical and effective method to provide persistent detection, identification, and monitoring of potential threats capability while protecting DoD assets such as forward operating bases and petroleum facilities. The solution also gives the military the capability to do early risk assessment and provides more reaction time by flying the SUAS to the incident site faster than any patroller likely could without any harm.

D. THESIS OBJECTIVES AND RESEARCH APPROACH

The objective of this thesis is to determine whether a system of SUAS can be operated autonomously providing persistent surveillance of a given perimeter, and what the limitations, if any, there are to the proposed solution. The system engineering process is followed in designing the overall concept of operations and identifying high-level requirements for the system.

The research involves prototyping of a system composed of COTS products, including 3DR Solo quadcopters, a single router, a laptop, and Python software. A quadcopter was selected because of its ability to hover and do abrupt maneuvers, which were deemed essential for perimeter surveillance. Quadcopters are programmed to fly in different predefined patterns over the Military Operation at Urban Environment Training site at Impossible City and McMillan Field, located at Camp Roberts in Monterey, California, with recording videos. One quadcopter is in flight while the remaining two are on standby. Once the airborne quadcopter has depleted its battery life to 30 percent, one of the standby quadcopters is activated as a replacement. The process is repeated a couple of times to verify that video consistency can be achieved during the swapping process. The battery is changed manually as battery swapping is not part of the scope of this thesis.

The system developed is not deployable in the field as it was developed only as a proof of concept. In order for the system to be used in the field, further analysis of the user's maintenance and other suitability requirements would be required.

E. THESIS ORGANIZATION

This thesis is organized as follows:

Chapter II describes the perceived operational concept of the system by looking at a scenario and vignettes of the system. It then follows a system engineering approach in order to derive the system architecture of the system.

Chapter III describes the developed prototype. It introduces the components that were used and describes the developed software.

Chapter IV discusses the test and evaluation approach of the system. The chapter includes a description of the different phases of the test conducted and the environmental conditions of the test scenario. It highlights the results and analysis.

Chapter V describes two examples on how the system can be implemented and discusses the survivability of the system.

Chapter VI provides conclusions and recommendations.

II. CONCEPTUAL DESIGN

A. OPERATIONAL CONCEPT DESIGN

This section describes the perceived operational concept of operating the SUAS for camp surveillance. On a daily basis, the operator would send out the SUAS to do routine patrols within the camp perimeters. The capability of pursuing a suspect outside the camp perimeter is outside the scope of this thesis. In the next section, the operational scenario and vignettes are discussed. These would help to identify and provide insights about the type of responses or output required by the system.

1. Operational Scenario

Based on the problem statement discussed in Chapter I, a scenario has been generated to describe the process from the operator starting the system until the operation is terminated. In this scenario, the operator first starts the system and the system performs health checks on the SUAS and reports the number of SUAS available for the operation. The operator then decides whether there are sufficient SUAS for the operation. If the operation is started, the system then monitors the battery level of the SUAS that was launched. The system will swap a UAS that has a low battery level with a new one autonomously. The system informs the operator on the SUAS's status and how many SUAS are available for swapping. During the operation, live video feeds are sent to the operator for situation awareness. The operator can also choose to terminate the patrol at any time and the system will provide a summary of the patrol route and battery usage patterns to detect any anomalies in the system. The collected data could be used for maintenance planning.

2. Vignette 1: No SUAS Available for Swapping

A possible vignette would be that there is no available SUAS to take over the patrolling duties. In this scenario, the operator has started the operation and the system has detected that the available SUAS are not capable of flying. The system informs the operator of the issue and brings back the SUAS that was patrolling when its battery has reached a critical level.

3. Vignette 2: Error in the System

Another possible vignette would be the occurrence of an error in the system. When an error occurs in the system, the system issues return to launch point commands to the any SUAS that is flying and reports that the SUAS has landed safely to the operator.

B. FUNCTIONAL ANALYSIS

This section covers the functional decomposition for the system. The purpose of the functional decomposition is to identify functions and sub-functions of the system by decomposing them in a logical and well-defined manner. Figure 2 illustrates the functional hierarchy diagram, which shows the breakdown of key functions and sub-functions that the system must perform. Table 1 provides the description of each function.



Figure 2. Functional Hierarchy Diagram.

Key Function	Sub-functions	Description
1.0 <u>Patrol Perimeter</u>	1.1 Navigate UAS	Navigates the UAS to designated spots around the perimeter.
of the UAS around the perimeter.	1.2 Control UAS	Controls the UAS to designated spots around the perimeter.
	1.3 Stream Video	Streams the video back to a central system for viewing and recording.
2.0 <u>Monitor UAS</u>	2.1Monitor Battery Level	Monitors the battery level of the UAS while it is flying.
monitoring of various parameters of the UAS to ensure that the system is	2.2 Monitor UAS Location	Monitors the UAS location while UAS is flying.
operating normally.	2.3 Monitor UAS system status	Monitors the UAS response while UAS is flying.
3.0 <u>Replace UAS</u>	3.1 Launch New UAS	Launches a replacement UAS to appropriate height.
replacement of the airborne UAS that has depleted its battery level to the	3.2 Fly to meet point	Flies replacement UAS to existing UAS location.
critical limit or lower.	3.3 Swap UAS	Directs swapping UAS to take over existing UAS patrol tasks.
	3.4 Fly back existing UAS	Flies existing UAS back to launch location.

 Table 1.
 Description of System Functions and Sub-functions.

4.0 Replace / Recharge Battery

This function addresses the replacement or charging of the UAS battery that has been depleted to or below the critical limit.

C. FUNCTIONAL FLOW BLOCK DIAGRAM

The functional flow of the SUAS system is documented in the Functional Flow Block Diagram (FFBD) shown in Figure 3. It is used to determine the required actions and their required sequence. The FFBD takes into account the operational scenario that was discussed previously, which starts from checking the SUAS until swapping the SUAS.



Figure 3. Functional Flow Block Diagram of the UAS.

D. FUNCTIONAL ALLOCATION

This section covers the mapping of functions described in the previous section into physical elements. The purpose of the functional allocation is to ensure that all required functions are addressed by the system and its components. Table 2 shows the components allocated to the functions.

System / Component	Sub-functions	Description
Ground Control Station (GCS), UAS and Global Positioning System (GPS)	1.1 Navigate UAS	Navigates the UAS to designated spots around the perimeter.
UAS	1.2 Control UAS	Controls the UAS to designated spots around the perimeter.
Camera and Wireless System	1.3 Stream Video	Streams the video back to a central system for viewing and recording.
GCS	2.1Monitor Battery Level	Monitors the battery level of the UAS while it is flying.
GCS	2.2 Monitor UAS Location	Monitors the UAS location while the UAS is flying.
GCS	2.3 Monitor UAS system status	Monitors the UAS response while the UAS is flying.
Ground Control System and UAS	3.1 Launch New UAS	Launches replacement UAS to appropriate height.
GCS, UAS, and GPS	3.2 Fly to meet point	Flies replacement UAS to existing UAS location.

Table 2.Functional Allocation.

System / Component	Sub-functions	Description
UAS	3.3 Swap UAS	Directs swapping UAS to take over existing UAS patrol tasks.
Ground Control System and UAS	3.4 Fly back existing UAS	Flies existing UAS back to launch location.
Charging system	4. Replace/ Recharge Battery	Replaces or charges the UAS battery that has been depleted to the critical limit or lower.

E. SYSTEM ARCHITECTURE

The Department of Defense Architecture Framework's (DODAF) System View-1 (SV-1) diagram shows the interfaces between the different system components. Figure 4 shows the SV-1 diagram for the system. The GCS consists of three software modules, namely the main controller module, the Patrol module, and the Replace module. The GCS sends and receives information through a wireless system to the SUAS. The SUAS have a camera gimbal attached with a camera on board the SUAS. Each SUAS is also connected to a charging system when it is not flying.



Figure 4. SV-1 Diagram.

The System View-2 (SV-2) diagram is shown in Figure 5, where the means of communication between the components or system are depicted. The wireless exchanges are based on Transmission Control Protocol and Internet Protocol (TCP/IP) network communications. The gimbal control is communicated through a micro air vehicle Link (MAVLink) between the SUAS controller and gimbal by a serial connection. The gimbal is connected to the camera through a Universal Serial Bus (USB). The charging system can be based on electromagnetic wave to wireless charging technology, which is not within the scope of this thesis.



Figure 5. SV-2 Diagram.

The next chapter discusses the configuration and development process of the system prototype. The system prototype is based on the system architecture developed in this chapter. It is important to note that the prototype is developed as a proof of concept for this thesis and is not representative of an operational system.

III. SYSTEM PROTOTYPING

This chapter describes the process and the system used for the proof-of-concept of the persistent tracking system.

A. SYSTEM OVERVIEW

The system under development builds upon the static persistent surveillance system developed by Williams (2017). Figure 6 shows the overall setup of the system. It consists of a laptop running on the Linux operating system, a WIFI router, and three SUAS (or unmanned aerial vehicles, UAV) with cameras attached.



Figure 6. Overall System Setup.

The laptop is running Python software to control and manage the SUAS. The WIFI router is used as the communication medium between the laptop and the SUAS to send control commands from the laptop and receive status from the SUAS. The SUAS controllers can also be used to control the SUAS manually in case of emergency.

The objective of the proof-of-concept is to show that one SUAS on patrol can be swapped seamlessly with another SUAS. In this setup, the criterion on which to swap a SUAS is based on battery level of the SUAS. The system will first conduct an initialization to determine how many SUAS are available for the mission. Once it is determined, the first SUAS will take off to perform a patrol on a predefined route. The system will monitor the battery level of the airborne SUAS, and when the battery level is depleted to a user-defined threshold, a replacement SUAS will be launched. The system will then monitor the distance between the existing and replacement SUAS. Once the replacement SUAS has reached within one meter of the existing SUAS, the replacement SUAS will take over the patrolling duties while the existing SUAS returns to where it was launched for battery replacement or charging. The system will cycle through all the SUAS until there are no SUAS capable of replacement or the user terminates the mission. Figure 7 shows the swapping process of the SUAS.

2



System monitors Patrol UAS battery level



Replacement UAS takes over patrol duties & low battery level UAS fly back to launch point

n ۳ System launches Replacement UAS



Low battery level UAS charges for next launch.

Figure 7. SUAS Swapping Process. Source: Herzog (2017).
B. HARDWARE SOLUTION

The hardware consists of a laptop, WIFI router, and multiple SUAS. Configurations were done on each of the hardware components to ensure that information was send and received at the appropriate time to achieve the mission objective.

1. Laptop

The laptop runs the Ubuntu 16.04 operating system with the technical specifications shown in Table 3.

Components	Description
Operating System	Ubuntu 16.04 LTS
Processor	Intel Core i7-7700HQ-CPU @ 2.8GHz x 8
Random Access Memory	32 GB DDR4
Graphic Card	Gefore GTX 1060/PCIe/SSE2
Memory Disk	1 TB 2.5" Drive

 Table 3.
 Ground Control Station Specifications of Laptop.

The development software used was Python, which is described later under "Software Solution." In addition to the development software, third-party software was used to test and communicate with the SUAS. The list of required software is shown in Table 4.

Table 4.	Software	Required.
----------	----------	-----------

Software Name	Description
Solo CLI	A command-line tool to control, update, or connect with Solo for development
Virtualenv	A tool to create isolated Python environments
Dronekit	The Application Programming Interface used to communicate with UAS over MAVLink

Software Name	Description
Dronekit-Software In The Loop	The simulator used to test Dronekit applications without using real UAS
APM Planner 2.0	An open-source ground station application for MAVLink based UAS
Geany	A text editor used to debug the software program
VLC Media Player	An open-source cross-platform multimedia player used to record video streams from UAS

2. Unmanned Aerial Vehicle

The SUAS used for this thesis is from 3D Robotics, which is a commercially available low-cost quadcopter SUAS. The SUAS can be easily programmed to fly autonomously by either downloading a script to the SUAS or sending commands wirelessly through a WIFI to a ground control station like a laptop. The SUAS is connected to a three-axis gimbal with a GoPro Hero 4 Silver camera to provide stable shots automatically. Figure 8 shows the picture of the SUAS.



Figure 8. 3DR Solo Drone. 18

The specifications of the SUAS are shown in Table 5. The SUAS is powered by two 1-gigahertz (GHz) processor flight control systems. The flight controller uses a Global Positioning System (GPS) to navigate and uses gyros, accelerators, and other sensors as safety features for flying. The SUAS comes in a ready-to-fly bundle with a radio controller and mobile applications to control both the SUAS and the camera.

Components	Description
Dimensions	18 x 18 x 10 inches
Weight	3.9 lbs. with GoPro camera and Solo Gimbal
Maximum Speed	55 miles per hour
Flight time	20–25 minutes
Battery	Lithium Polymer 5200mAh@14.8Vdc
Communication	2.4GHz WIFI network
Motors	880 kilovolts (kV)

Table 5.3DR Solo Drone Specifications.

3. WIFI ROUTER

The WIFI Router is used to connect the ground control system with the SUAS in the same network. In order to achieve this, the SUAS are each configured within the same Internet Protocol (IP) address class and operate on the 2.4GHz frequency bandwidth. Unique User Datagram Protocol (UDP) ports were also assigned to each SUAS that uses the MAVLink protocol to communicate with the laptop. Figure 9 shows the wireless network configuration diagram of the SUAS and the laptop.



Figure 9. System Network Configuration Diagram

C. SOFTWARE SOLUTION

The developed software is written in the Python programming language. The Dronekit-Python Application Programming Interface (API) was used to communicate with the ArduPilot Flight Controller using a low latency link. The API provides a library of ready-to-use scripts to connect, control, and monitor the SUAS. Various functions could be called simply by importing the Dronekit-Python Library and referencing the function name. These functions were used to develop the autonomous functions of the system.

Software was based on having the GCS as a central control. This means that all SUAS will take commands from the GCS. Three software functions were run in parallel in order to maintain persistent tracking, air traffic control, and flight of the replacement SUAS when the airborne SUAS's battery level is depleted. Figure 10 shows the flow diagram for the developed software.



Figure 10. Software Flow Diagram.

In the next chapter, the test and evaluation process and environment for the prototype described in this chapter are discussed. This discussion includes the different phases of test that were done in order to arrive at the final phase of the test to validate the feasibility of the solution.

IV. SYSTEM TEST AND EVALUATION

This chapter describes the test and evaluation procedures and test environment to evaluate the feasibility of having 24/7 persistent surveillance while the SUAS is moving. The main focus of this experiment is to determine whether a similar FOV could be achieved during SUAS swapping. The results of the test and evaluation are then presented and discussed in the next chapter.

A. TEST PROCEDURES

The test was broken into many phases to ensure that the SUAS would operate in a safe manner as intended. Figure 11 shows the different phases of testing. The test began in March 2018 and ended in July 2018.



Figure 11. Different Phases of Testing.

Apart from using software debugging tools to check for syntax errors, a softwarein-the-loop simulator was used to ensure the logic of the code was correct. As multithreading was used as part of the solution, simple commands to fly two SUAS simultaneously were tested to observe the behavior of the SUAS before performing more complex commands. Different software functionality tests with the SUAS were done to facilitate faster troubleshooting and debugging of the system. The software codes contain a safety feature to ensure that the SUAS is not damaged if there is an error. This safety feature includes a command for the SUAS to return to the launching point when there is an error and also to have the software monitor the distances between two SUAS to prevent collision.

B. TEST ENVIRONMENT

The experiment was conducted at the Military Urban Environment Training site at Impossible City, California, and McMillan Field, at Camp Roberts, California. The test was conducted during the day and at about 30 meters and 15 meters above ground at Impossible City and Camp Roberts, respectively. The flying height was above the urban buildup in that area as maneuvering within buildings was not part of the scope of this experiment. Figure 12 shows a picture of the test sites.

Impossible City



Camp Roberts



Figure 12. Experiment Conducted at Impossible City and Camp Roberts. Source: Google Maps (2018a, 2018b).

C. SCENARIO

A perimeter patrol scenario was selected to evaluate the feasibility of the system. The SUAS flies in a square loop and on a straight-line through and flow at an altitude of 15 and 30 meters. Figure 13 shows the location of the launch and flight patterns at the respective test sites. The camera on board the SUAS will record the whole duration of the flight until the SUAS has landed. The SUAS will continue to provide perimeter surveillance until the user terminates the operation or battery level is low. This simulates the real-world environment where the SUAS operates around a perimeter providing airborne surveillance.

Impossible City



Impossible city

Figure 13. Launch Site and Flight Patterns at Impossible City and Camp Roberts. Source: Google Maps (2018a, 2018b).

The test scenario is deemed successful when coherent video can be seen during the SUAS airborne swapping and the developed SUAS can provide a persistent surveillance that is longer than the capability of a single SUAS.

D. SCOPE

The focus of the development and testing is on the SUAS' ability to perform onthe-move replacement when flying autonomously. The autonomous charging and target tracking functions were not included but should be developed in the future.

E. OBJECTIVE

The Measure of Effectiveness (MOE) and Measures of Performance (MOP) were used to quantify the results of the test and evaluation. This is shown in Figure 14. The MOE used for the SUAS is the persistent surveillance capability of the system. The MOPs used to evaluate the system were success rate of swapping and success rate of having a consistent FOV during the swap. A consistent FOV is achieved when the swapping SUAS field of view could cover at least 80 percent of the original SUAS field of view.



Figure 14. System MOE and MOPs.

F. SETUP

The experiment setup consists of the GCS, which is a laptop, a router, 3x SUAS controllers and 3x SUAS, as shown in Figure 15. The SUAS controllers were used to take over control from the GCS when any error occurs.



Figure 15. Experimental Setup.

Due to the inaccuracies of the GPS sensor, the replacement distance between the patrolling SUAS and replacing SUAS was set to be 10 meters. The update rate of the patrolling SUAS location to the replacing SUAS was every second.

G. TEST RESULTS

A total of ten samples were calculated during the test. Due to communication and GPS errors in the SUAS, most of the test sample consists of only two SUAS. Nevertheless, this does not affect the results of the test since the focus is on the successful swapping of the SUAS. Figure 16 shows the photos of the swapping process during the test.



Figure 16. SUAS Swap.

Figure 17 shows the three-dimensional (3D) plot of the swapping process. From the results, it can be seen that the system successfully operated multiple SUAS to patrol in a square pattern and swapped the vehicles autonomously.



Figure 17. SUAS Flight Path.

The two-vehicle swap occurred during the field testing. From the results, it can be seen that the system is more capable of supporting a single SUAS that can only last for an average duration of 11–15 minutes in the test. With additional batteries, the system could provide a 24/7 perimeter surveillance. Due to limited resources, the 24/7 operation test was not conducted but should be done in the future to determine its feasibility.

Figures 18 and 19 show some screenshots of the camera videos from both SUAS during patrolling and swapping. All of the test screenshots can be found in Appendix B.

 Patrolling UAS
 Swapping UAS

 Screenshot flying in a straight line (Test 4)

 Patrolling UAS

 Screenshot flying in a straight line (Test 4)

Screenshot flying in a square loop (Test 1)

Figure 18. Screenshots of Videos during SUAS Swapping at Impossible City.

Screenshot flying in a square loop (Test 10)



Figure 19. Screenshots of Videos during SUAS Swapping at Camp Roberts.

Table 6 shows the summary of the results for the ten test trials.

Test	Location	Patrol	Successful	FOV
Number		Pattern	Swapping	Consistency
			(Pass/Fail)	(Pass/ Fail)
Test 1	Camp	Square Loop	Pass	Pass
Test 2	Roberts	Square Loop	Pass	Pass
Test 3		Square Loop	Pass	Pass
Test 4		Straight Line	Pass	Pass
Test 5		Straight Line	Pass	Pass
Test 6	Impossible	Straight Line	Pass	Fail
Test 7	City	Straight Line	Pass	Pass
Test 8		Straight Line	Pass	Pass

Table 6.Results Summary.

Test	Location	Patrol	Successful	FOV
Number		Pattern	Swapping	Consistency
			(Pass/Fail)	(Pass/ Fail)
Test 9		Straight Line	Pass	Pass
Test 10		Square Loop	Pass	Pass
Success ra	te		100%	90%

From the results, it can be seen that the SUAS was able to take over the patrol task with consistent FOV at Impossible City. For test 6 that was conducted at Camp Roberts, the swapping was done when the patrolling SUAS was about to turn back and therefore a significant portion of the FOV was lost. Nonetheless, the replacing SUAS was able to cover the surveillance gap in half a second. The SUAS could also be configured to fly forward and backward without turning; this would prevent such inconsistency in the field of view.

V. FEASIBILITY ANALYSIS

In the experiment conducted, only three SUAS were used for the perimeter patrol tasks. Depending on the size of the surveillance area and revisit rates requirement, more SUAS could be added to expand the surveillance capability and survivability of the system. The following sections give two examples of the quantities of SUAS needed given the distance of the perimeter and revisit rates requirements.

A. NAVAL POSTGRADUATE SCHOOL

The Naval Postgraduate School (NPS) has an approximate perimeter distance of three kilometers, as shown in Figure 20. Thus, a total of at least 21 SUAS would be required given that the SUAS is flying at 16 Kilometers per hour, the revisit rate is every five minutes, and charging takes 90 minutes.



Figure 20. Map of Naval Postgraduate School. Source: Google Maps (2018c).

In order to meet the revisit rate of five minutes, three SUAS would have to patrol simultaneously at any given time. The total number of SUAS can then be calculated using the following formula.

Number of UAS required = (1)

where is the number of UAS patrolling at any given time.

Specifically, for = 3

Number of UAS required = 21

Figure 21 shows the time schedule of the 21 SUAS for one continuous cycle. From the figure, it can be seen that a total of 18 SUAS are required to patrol the perimeter before the first three batteries are charged and can be used again.

	15 mins	30 mins	45 mins	60 mins	75 mins	90 mins	105 mins	120 mins	135 mins	150 mins	165 mins	180 mins	195 mins
UAS (1-3)	Patrol			Cha	rging								
UAS (4-6)		Patrol		Charging									
UAS (7-9)			Patrol		Charging								
UAS (10-12)				Patrol			Cha	rging					
UAS (13-15)					Patrol			Cha	rging				
UAS (16-18)						Patrol Charging							
UAS (19-21)							Patrol Charging						

Figure 21. Time Schedule to Patrol the Naval Postgraduate School Perimeter.

Furthermore, the number of SUAS could be reduced if new batteries could be swapped with old ones. This would mean that a minimum of six SUAS with 21 batteries would be able to cycle through until the first set of batteries is fully charged. Figure 22 shows the time schedule of using swappable batteries.

	15 mins	30 mins	45 mins	60 mins	75 mins	90 mins	105 mins	120 mins
			Battery (1-3) Charging					
UAS (1-3)	Battery (1-3)		Battery (7-9)		Battery (13-15)		Battery (19-21)	
UAS (4-6)		Battery (4-6)		Battery (10-12)		Battery (16-18)		Battery (1-3)

Figure 22. Time Schedule when Using Swappable Batteries (NPS).

B. CORRECTIONAL TRAINING FACILITIES IN SOLEDAD

The Correctional Training Facilities in Soledad, California, provides another example of how the SUAS can be used to patrol the perimeter. Figure 23 shows the map of the training facility.



Figure 23. Map of Correctional Training Facilities at Soledad. Source: Google Maps (2018d).

The training facility has an approximate perimeter of 2 kilometers. It is assumed that the revisit rate for such facility would be 1 minute. This would require at least eight SUAS patrolling simultaneously. The total number of SUAS can be calculated using Equation (1):

Number of UAS required = 56

Given = 8, time to charge battery = 90 and time to deplete battery = 15

Similar to the time schedule for patrolling the Naval Postgraduate School in Figure 21, the time schedule for patrolling the Correctional Training Facilities is shown in Figure 24. A total of 46 SUAS (from UAS 9 to 56) are required before the first eight SUAS are fully charged and ready to patrol again.

	15 mins	30 mins	45 mins	60 mins	75 mins	90 mins	105 mins	120 mins	135 mins	150 mins	165 mins	180 mins	195 mins
UAS (1-8)	Patrol			Cha	rging								
UAS (9-16)		Patrol		Charging									
UAS (17-24)			Patrol		Charging								
UAS (25-32)				Patrol			Cha	rging					
UAS (33-40)					Patrol	Charging							
UAS (41-48)						Patrol Charging							
UAS (49-56)							Patrol Charging						

Figure 24. Time Schedule to Patrol Correctional Training Facilities Perimeter.

The number of SUAS can be reduced to 16 if batteries are swapped with new ones when the batteries are depleted. This would mean that a total of 56 batteries is required to cycle through until the first sets of batteries were recharged. Figure 25 shows the time schedule using swappable batteries.

	15 mins	30 mins	45 mins	60 mins	75 mins	90 mins	105 mins	120 mins
			Battery (1-8) Charging					
UAS (1-8)	Battery (1-8)		Battery (17-24)		Battery (33-40)		Battery (49-56)	
UAS (9-16)		Battery (9-16)		Battery (25-32)		Battery (41-48)		Battery (1-8)

Figure 25. Time Schedule When Using Swappable Batteries (Correctional Training Facilities).

It can be observed that the number of SUAS or batteries requirement increases significantly when the revisit rates are reduced to a minute.

C. SUAS QUANTITY IMPLICATIONS

The previous calculations are based on the minimum number of UAS required for perimeter surveillance, but these calculations do not take into consideration of the operation suitability of the system. Operation suitability like reliability, availability, and maintainability requirements would have an impact on the number of SUAS required for the mission.

Increasing the number of SUAS in the system also increases the complexity of the software. The GCS needs to manage multiple UAS and the battery depletion rate in such cases may not be uniform. This also increases the bandwidth needed for control and to do video streaming.

An alternative solution is to have onboard computers on the SUAS such that the SUAS control is distributed to individual SUAS rather being performed by a central GCS. The SUAS could also communicate among themselves to request a replacement when their battery levels are low. This would simplify the software and reduce the bandwidth requirement by not having all communication data going through a central GCS. The bandwidth requirement could also be further reduced by having local recording on the SUAS and only streaming on demand. The videos could then be downloaded to a central storage location during charging. The limitation to this solution is that the endurance of the SUAS may be reduced due to added load and increased processing power that draws more electrical power.

D. SUAS SURVIVABILITY

In terms of system survivability, the SUAS is susceptible to hostile attacks either by kinetic weapons like small arms or by electronic weapons. It is assumed, however, that the intruder does not want to be detected and it may be challenging to attack a small and fast moving SUAS at a distance. Even if it is possible, a replacement drone could be launched to the last known location within minutes and security forces could also be deployed to the location to suppress the threat. While shielding on SUAS could potentially minimize the vulnerability of the SUAS, it may not be as cost effective as having more redundant SUAS.

Furthermore, the SUAS could also be subjected to bad weather conditions, which may impact its ability to fly. As such, weather resistant SUAS should be implemented to ensure that surveillance could be performed in all weather conditions. THIS PAGE INTENTIONALLY LEFT BLANK

VI. CONCLUSION AND RECOMMENDATIONS

A. CONCLUSION

In this thesis, the challenge of persistent surveillance in the area of perimeter security was discussed. A set of multi-rotor UAS to provide persistent surveillance through sequential swapping was proposed and a proof of concept was developed to evaluate the solution feasibility. The proof of concept was conducted at Impossible City and Camp Roberts in California. The results showed that the SUAS could be swapped autonomously and was able to maintain consistent FOV during the swap. The FOV during transition could be further improved if higher accuracy GPS sensors were used.

The system of SUAS proved to be a feasible solution to provide persistent surveillance. By monitoring the battery level and autonomously replacing the depleted SUAS via a GCS, the system showed that the endurance can exceed the capability of a single SUAS and possibly perform 24/7 persistent surveillance or until the system mechanically fails.

Certain limitations were identified in the system. One of which was that the number of SUAS increases 3.5 times if depleted batteries are not replaced with new ones. This is due to the 90-minute long charging time for a battery. During SUAS swap, the fields of view can also be inconsistent when the patrolling SUAS is turning, as seen in Figure 19. This can be improved by flying in a straight path (front, back, left, and right) without turning.

Furthermore, the SUAS is susceptible to attacks. This threat could be overcome, however, by having a redundant SUAS available to replace the airborne SUAS when it is compromised. A swarm of SUAS can also be deployed if a surveillance gap during downtime is an issue for the mission. The cost of having a swarm of SUAS versus having countermeasures to reduce the vulnerability of the SUAS should be analyzed to determine the more cost-effective solution.

The solution proposed can also be extended to other surveillance or target tracking tasks where persistent surveillance or targeting is required. The autonomous operations could relieve security personnel from such mundane tasks and allow them to focus on more high value work.

B. FUTURE WORK

In testing of the SUAS, a few potential challenges were identified, and thus, the following future work is recommended:

- (1) Determine whether the SUAS should be launched centrally or distributed based on the flight pattern.
- (2) Determine the feasibility of tracking moving targets during SUAS swapping.
- (3) Develop a mechanism to replace batteries automatically.
- (4) Conduct a 24-hour flight test to determine the feasibility of 24/7 operations.
- (5) Develop an onboard computer for the SUAS to communicate among one another to perform mission task and swapping when required.

With further development on the operational effectiveness and suitability of this system, military forces could benefit in terms of mission effectiveness and manpower savings from this technology. Although the endurance of SUAS continues to improve, SUAS swapping to provide persistent surveillance will be crucial in providing the force deploying the system with a technological edge over its adversaries.

APPENDIX A. PYTHON SCRIPTS

```
# Simplifying threading lock function
 65
           class Interface():
    def init (self):
        self. value = None
        self. lock = threading.Lock()
 66
67
  68
  69
 70
71
72
73
74
75
76
77
78
79
80
81
82
83
84
                   def acquire(self):
    self. lock.acquire()
                   def setandprint(self, newvalue):
    self. value = newvalue
    sleep(random()*0.1)
                   print self. value
def release(self):
    self. lock.release()
           inter = Interface()
           class SimpleLock(object):
    def init (self):
        self. lock = threading.Lock()
 85
86
87
88
                           enter (self):
self. lock.acquire()
return self
                   def
 89
90
91
92
93
94
                           exit (self, type, value, traceback):
self. lock.release()
return
                   def
  95
           inter lock = SimpleLock()
96
97
98
99
100
           # Function for arming and takeoff to set altitude
def arm and takeoff(vehicle,aTargetAltitude,vehiclename):
                   print "Basic pre-arm checks"
# Do not try to arm until autopilot is ready
while not vehicle.is armable:
    print " Waiting for vehicle to initialise..."
    time.sleep(1)
101
102
102
103
104
105
106
107
108
109
110
111
                   print "Arming motors"
                   # Copter should arm in GUIDED mode
vehicle.mode = VehicleMode("GUIDED")
vehicle.armed = True
112
113
114
115
116
117
                   while not vehicle.armed:
    print " Waiting for arming..."
    time.sleep(1)
                   print "Taking off! Heading to ",aTargetAltitude
118
                   takeofftime = datetime.now()
landedalt=vehicle.location.global relative frame.alt
global sortie
120
121
122
123
124
125
126
                   sortie+=1
                   # For data logging
storesortiedata(vehiclename,vehicle)
127
                   # Wait until the vehicle reaches a safe height before
                   # processing the goto (otherwise the command
# after Vehicle.simple_takeoff will execute immediately).
128
129
```

```
while True:
130
                         vehicle.simple takeoff(aTargetAltitude) # Take off to target altitude
timeelapsed=((datetime.now()-takeofftime).total seconds())
print " Altitude: ", vehicle.location.global relative frame.alt, " Time
Elapsed: " timeelapsed. " sec"
131
132
133
                                       Altitude: ", ven
",timeelapsed,
                                                                                                                                                                       J
                             apsed:
                         Elapsed: ",timeetapsed, sec
#Trigger just below target alt
if vehicle.location.global relative frame.alt>=aTargetAltitude*0.95:
    print "Reached target altitude in ",timeelapsed," sec"
    status="success"
134
135
136
137
138
                         break
if timeelapsed>5 and vehicle.location.global relative frame.alt<landedalt+1:
139
140
                                print
                                                                                                                                                                        স
                                print "Land vehicle (%s)" %(vehiclename)
vehicle.mode = VehicleMode("LAND")
141
142
143
                                status="fail"
144
145
146
147
148
149
150
151
152
153
154
155
156
157
158
                                break
                         time.sleep(1)
                 # For data logging
storesortiedata(vehiclename,vehicle)
                  return status
           # Function to calculate the distance in meters between two position
def get distance metres(aLocation1, aLocation2):
                  Returns the ground distance in metres between two LocationGlobal objects.
                  This method is an approximation, and will not be accurate over
large distances and close to the earth's poles. It comes from the
ArduPilot test code:
159
160
161
162
163
164
                  https://github.com/divdrones/ardupilot/blob/master/Tools/autotest/common.pv
                  dlat = aLocation2.lat - aLocation1.lat
dlong = aLocation2.lon - aLocation1.lon
return math.sgrt((dlat*dlat) + (dlong*dlong)) * 1.113195e5
165
166
167
168
169
170
           # Function to store data during testing
def storesortiedata(vehiclename,vehicle):
171
                  timenow = datetime.now()
172
173
174
175
                  global sortie
                 sortiename=vehiclename
sortiedata[sortiename].append(
176
177
178
179
                         (timenow,
                                vehicle.location.global relative frame.lat,
vehicle.location.global relative frame.lon,
vehicle.location.global relative frame.alt,
vehicle.battery.level,
180
                                vehicle.battery.voltage,
181
182
                                sortiename
183
184
185
                  print "Stored data for %s in dictionary"%(vehiclename)
186
187
          # Function to write data to csv file and plot data
def showplot():
    if len(sortiedata)>0:
188
189
190
                         fig=plt.figure()
fig.suptitle('Battery Management Testing')
191
192
```

193	plt1=fig.add subplot(211)
194	plt2=fig.add subplot(212)
195	fig2=plt.figure()
196	plt3=fig2 add subplot(111 projection='3d')
107	preserigziada subprocerin, profeerine sa /
197	the first second s
198	# Upen csv file and append
199	t = open('data.csv', 'a')
200	
201	fulltime=[]
202	fulllatitude-[]
202	fullionatudo-[]
203	
204	Tullaltitude=[]
205	fullbatterypct=[]
206	fullbatteryvolt=[]
207	fullsolo=[]
208	
200	alobal starttime
205	dtobat starttine
210	
211	for key, value in sorted(sortledata.iteritems()):
212	soloid=key
213	time=[]
214	latitude=[]
215	longitude=[]
216	
210	
21/	pattervpct=[]
218	batteryvolt=[]
219	solo=[]
220	
221	for var in value.
222	time annend((var[A]_starttime) total seconds())
222	latitude encond (up [1])
223	(atitude.append(var[1])
224	longitude.append(var[2])
225	altitude.append(var[3])
226	battervpct.append(var[4])
227	hatteryvolt append(var[5])
220	solo append(upr[6])
220	soto.append(var[0])
229	
230	piti.plot(time,altitude,label=soloid)
231	plt1.set xlabel('Time (sec)')
232	plt1.set vlabel('Altitude (m)')
233	plt2.plot(time.battervpct.label=soloid)
234	plt2.set xlabel('Time (sec)')
235	plt2 set vlabel('Battery Life (%)')
226	plt2. slot (labit de logitude altitude)
230	pres. proc (active, tongitude, activude)
231	pits.set xlapel('Latitude (m)')
238	plt3.set xlabel('Longitude (m)')
239	plt3.set xlabel('Altitude (m)')
240	
241	<pre>pltl.scatter(time[0].altitude[0].marker='0'.color='0' s=50)</pre>
242	niti scatter(time[.]] altitude[.]] marker (c) color (c) (c)
242	plt1. scatter(time[1], attructer[1], marken_io, colon_i, s_50)
243	p_{122} , scatter (time[0], batter p_{122} , marker = 0, cotor = 0, s=30)
244	pit2.scatter(time[-1],batterypct[-1],marker= 0,color= r,s=50)
245	
246	fulltime.extend(time)
247	fulllatitude.extend(latitude)
248	fulllongitude extend(longitude)
240	fullaltitude extend(altitude)
250	full hatter under extend (hatter under)
250	full better vpcl.extend (better vpcl)
251	fullbatteryvolt.extend(batteryvolt)
252	fullsolo.extend(solo)
253	
254	duration=math.ceil(fulltime[-11/60)
255	start=starttime.strftime('%Y%m%d'%H&M')
256	stoptime-var[0]
250	stop starting (19)
251	scop=scopcine.scrictine(srand aran)

```
flightdata=['Start: '+start, 'Stop: '+stop, 'Duration: '+`duration`+ '
minutes', 'Sortie Count: '+`len(sortiedata)`]
258
259
                            # Add row titles
fulltime.insert(0, 'Time (sec)')
fullsolo.insert(0, 'Vehicle ID')
fullatitude.insert(0, 'Latitude')
fullatitude.insert(0, 'Longitude')
fullatitude.insert(0, 'Altitude (m)')
fullbattervpct.insert(0, 'Batterv (%)')
fullbattervvolt.insert(0, 'Batterv (V)')
260
261
262
263
264
265
266
267
268
269
270
271
272
273
274
275
277
278
279
280
                           # Write to csv file
w = csv.writer(f, delimiter=',')
w.writerow(flightdata)
w.writerow(fullsolo)
w.writerow(fullatitude)
w.writerow(fullatitude)
w.writerow(fulllongitude)
w.writerow(fullaltitude)
w.writerow(fullaltitude)
w.writerow(fullbatterypct)
w.writerow(fullbatteryvolt)
                            281
282
283
284
285
286
                            # Close csv file
f.close()
                            plt1.autoscale(enable=True, axis='both', tight=False)
plt2.autoscale(enable=True, axis='both', tight=False)
287
288
289
290
291
292
293
294
295
296
297
298
299
300
           # Initialize lists of available Solos (connect and names) set in
# connection (set empty here)
def connect solo():
        global solo, soloname, soloid, i, vehicle
                    i=0
                    solo=[]
                    soloname=[]
                    print "----- CONNECTING TO SOLOS ----
301
302
303
                    #Loop through solo ports to attempt connection with each vehicle
for soloport in soloports:
304
305
306
307
308
                             index=soloports.index(soloport)
soloid=soloids[index]
                            print("Connecting to solo: %s (%s)" % (soloport, soloid))
309
310
                             # Try to connect to the solo for XX heartbeat timeout seconds
311
312
313
314
315
                             try:
                                     vehicle=connect(soloport, wait ready=True, heartbeat timeout=10)
                                     solo.append(vehicle)
                                     soloname.append(soloid)
316
317
318
319
                                     solos[soloid]=vehicle
                                     # Print some vehicle attributes
                                     print "Get some vehicle attribute values:"
print " GPS: %s" % solo[i].gps 0
print " Battery: %s" % solo[i].battery
320
321
                                                                                          - 5 -
```

ਸ

print " Last Heartbeat: %s" % solo[i].last heartbeat
print " Is Armable?: %s" % solo[i].is armable
print " System status: %s" % solo[i].system status.state
print " Mode: %s" % solo[i].mode.name 322 323 324 325 326 327 # Increment 1 to look on the next solo port 328 329 330 331 332 i+=1 except: print (soloid+" vehicle not found.") 333 334 335 336 # Write which solos were found
print "%s Solos available: " %
print soloname 337 338 339 340 341 %len(solo) # Go around perimeter
def loiter():
 global loiterindex, currentaltitude, loiteralt, loiterlat1, loiterlon1,
 vehicleindex, vehicle, loiterpoint, loitervehicle,loitername, shuttingdown,
 loitershuttingdown, battery level limit 342 343 J 344 345 346 loiterpoint = LocationGlobalRelative(loiterlat1[loiterindex],loiterlon1[loiterindex], loiteralt) loitervehicle=vehicle loitervehicle.simple goto(loiterpoint) flv='True' 347 348 349 350 351 352 353 354 355 shuttingdown='n loitershuttingdown='n' try: while True and fly=='True' and shuttingdown=='n': try: currentaltitude=loitervehicle.location.global relative frame.alt
dist = get distance metres(loitervehicle.location.global frame, 356 357 ਕ loiterpoint) int "Loiter: Distance to go: %s Altitude goal: %s Current: "%(dist,loiteralt,currentaltitude) 358 print J viiit Loiter: %s Battery limit: %s Current battery: %s '%(soloname[vehicleindex],max(battery level limit,battery level crit ical limit),loitervehicle.battery.level) time.sleep(1) if current/limit. 359 360 361 if currentaltitude>loiteralt: 362 363 qoalpct=1.02 else: 364 goalpct=0.98 # Check if UAS reached waypoint if dist < 2: print 'Loiter: Within 2m of waypoint' 365 366 367 368 369 370 371 372 373 374 375 376 377 storesortiedata(soloname[vehicleindex],loitervehicle) loiterindex += 1
if loiterindex==4:
 loiterindex = 0
 time.sleep(1) if loitershuttingdown=='y': print"Loiter: shutdown 378 379 break

380	<pre>loiterpoint =</pre>
382 383	loitervehicle.simple goto(loiterpoint)
384	<pre>print "Loiter: moving to Lat:%s, Lon:%s" %(loiterlat1[loiterindex], loiterlon1[loiterindex])</pre>
386 387	if loitershuttingdown=='v': break
388 389	except KeyboardInterrupt: traceback.print exc()
391	print "Exited"
393 394 305	<pre>go home(loitervehicle, vehicleindex) print"Loiter shutting down"</pre>
396 397	<pre>time.sleep(1) # Function to go back launch point and land def me here(usbic)eberge usbic)ebergeindex);</pre>
399	<pre>do home(venictenome, venictenome/ndex):</pre>
400 401 402	<pre>print "Return to launch" currentaltitude=vehiclehome.location.global relative frame.alt</pre>
403	LocationGlobalRelative(homelat[vehiclehomeindex],homelon[vehiclehomeindex], currentaltitude)
404	veniclenome.simple goto(nome)
407	currentaltitude=vehiclehome.location.global relative frame.alt
409 410	if veliclehome.location.global relative frame.alt<=2: print"Home: Vehicle already landed"
411 412	<pre>break print "Home: Distance to go: %s Altitude goal: %s Current: ** ** (dist loiteralt currentaltitude)</pre>
413	print 'Home: %s Battery limit: %s Current battery: %s'%(soloname[vehiclehomeindex],max(battery level limit,battery level critical limit) vehiclehome battery level
414	time.sleep(1) if vehiclehome location global relative frame alt<=2:
416	print "Home: Vehicle already landed"
418	<pre>if dist < 1: print "Home: Land vehicle (%s)" %(soloname[vehiclehomeindex])</pre>
420 421	<pre>vehiclehome.mode = VehicleMode("LAND") while True:</pre>
422	print "Home: Landing %s Altitude: %s" %(soloname[vehiclehomeindex],vehiclehome.location.global relative fram #
423 424	time.sleep(1) if_vehiclehome.location.qlobal_relative_frame.alt<=2: #Trigger just ====================================
425	below target alt. print "Home: Vehicle On Deck"
426	break
428	#Function to tokaoff UAC
430 431 432	def take off(vehicleto, vehicleindexto,takeoffaltto): global takeoffstatus, takeoffalt, soloname

```
433
           try:
434
               cmds = vehicleto.commands
435
               cmds.download()
436
               cmds.wait ready()
437
               # Arm and take off - receive status
takeoffstatus=arm and takeoff(vehicleto,takeoffaltto,soloname[vehicleindexto])
438
439
440
441
442
443
               # If launch fail
if takeoffstatus=="fail":
    print "Launch failed, vehicleindex=%s" %(vehicleindexto)
443
444
445
446
447
448
      449
450
451
                                                                                                        =
           check status
452
453
           try:
454
455
               print "Check index=%s" %(checkindex)
456
               457
458
                                                                                                        म
459
                                                                                                        র
                    %(soloname[checkindex])
                    unavailablevehicles+=1
check status='n'
460
461
462
               else:
463
                    check status='v'
464
                    sortienum+=1
465
                    return checkindex
466
467
468
469
470
471
472
           except KeyboardInterrupt:
                                        upted"
               print
               unavailablevehicles=0
      check status='n'
#Function to fly standby UAS for swap
473
474
475
      def replace():
476
           qlobal vehicle, vehicle2, vehicleindex, vehicleindex2, solo, takeoffalt1,
soloname, homelat, homelon, takeoffstatus, loitervehicle, shuttinqdown, checkindex
global replacementfails, replacestop, vehicle2status, replaceshutdown
477
478
           replaceshutdown='n
479
           try:
               480
481
                                                                                                       T
482
483
                                 print "Replace: No standby available"
shuttingdown='y'
break
484
485
486
487
488
                             vehicleindex2=vehicleindex+1
489
                             if vehicleindex2>=len(solo)and vehicleindex!=0:
490
                                 vehicleindex2=0
```

491	check_vehicle(vehicleindex2)
492	if check status'V':
102	
495	vehicle2-cele(vehicleindex2)
494	
495	unavai tab tevenic tes=0
496	Dreak
497	else:
498	if vehicleindex2>=len(solo):
499	print "Replace: No standby available"
500	replaceshutdown='v'
501	break
502	vehicleindev2+=1
502	
504	if valid and valid an
504	in vehicleindex2=cell(solo) and vehicleindex:=0:
505	Vehiccendex2=0
506	if unavailablevehicles>len(solo)-1:
507	print "Replace: No standby available"
508	replaceshutdown='v'
509	break
510	if shuttingdown=='v'or replaceshutdown=='v':
511	break
512	print 'Replace: Launch replacement'
513	print 'Beplace: Launch %s' %soloname[vehicleindex2]
514	print heptater, Edular as associate (verte certaex2)
515	replacester in t
515	tep taces top= n
210	white frue and replacescop== n :
51/	
518	take off(venicle2, venicleindex2, takeoffalt1)
519	takeoffcheckr=0
520	if takeoffstatus=="fail":
521	takeoffcheckr+=1
522	vehicleindex2+=1
523	if vehicleindex2>=len(solo) and vehicleindex!=0:
524	vehicleindex2=0
525	vehicle2=solo[vehicleindex2]
526	elif takeoffcheckr=len(solo):
527	print "Penlace, Ne drones available for takeoff"
520	chuttingdown'y'
520	an home(loiterychicle, ychicleinder)
529	do none (cortervenicte, venicteridex)
530	Dreak
531	else:
532	
533	print "Replace: Set vehicle airspeed to 40"
534	vehicle2.airspeed = 40
535	vehicle2status=1
536	if vehicle2.location.global relative frame.alt<=10:
537	print 'Replace: Altitude too low, return to launch point'
538	<pre>go home(vehicle2 vehicleindex2)</pre>
539	renlaceshutdown='v'
540	hrazk
541	with inter lock.
541	with inter cot:
542	storesortiedata(soloname[venicteridex],venicte)
543	storesortiedata(soloname[venicleindex2],venicle2)
544	
545	if vehicleindex2>len(homelat)-1:
546	print "Adding home point for %s" %(soloname[vehicleindex2])
547	with inter lock:
548	cmds = vehicle2.commands
549	cmds.download()
550	cmds.wait readv()
551	with inter lock:
552	
	homelat[vehic]eindex2]=(vehic]e2_location_global_frame_
I	lat)
552	
555	-



```
replacementfails, replacestop, vehicle2status, replaceshutdown
605
              try:
606
                   shuttingdown='n
607
                   while True and shuttingdown=='n':
608
                         connect solo()
                         # A check to see if the script should continue running in loop to follow
continuecheck = 1
609
610
                         # Ask user whether to continue where 'v' and 'n' are only acceptable
611
                                                                                                                                   ਜ
                         answers
                         while continuecheck != 'y' and continuecheck != 'n':
    continuecheck = raw input('Continue? (v/n):')
if continuecheck =='n':
612
613
614
615
                               shuttingdown='y'
616
                               break
617
                         # Create new dictionary variables for home lat and longs
618
                         homelat=defaultdict(list)
                         homelon=defaultdict(list)
619
620
621
622
                         #initialize all variables
                         vehicle=0
                         sortienum=-1
623
                         unavailablevehicles=0
624
                          loiterindex=0
                         takeoffstatus="true"
battery level limit=0
625
626
                          replacementfails=0
627
628
                         vehicle2status=0
629
630
                          replaceshutdown='n'
                         try:
                               # Initialize time for data collection
starttime = datetime.now()
631
632
633
                               vehicleindex=0
634
                               vehicleindex2=1
635
636
                               while True:
637
                                    check vehicle(vehicleindex)
if check status=='v':
    vehicle=solo[vehicleindex]
    loitername=soloname[vehicleindex]
638
639
640
641
642
                                          unavailablevehicles=0
643
644
645
                                          # If vehicle is not already airborne (likely same vehicle)
if vehicle.location.global relative frame.alt <</pre>
646
                                                                                                                                   ਸ
                                          takeoffalt-1:
647
648
                                                takeoffcheck=0
649
                                                while True:
650
651
                                                     take off(vehicle, vehicleindex, takeoffalt)
if takeoffstatus=="fail":
    takeoffcheck+=1
652
653
654
                                                           vehicleindex+=1
if vehicleindex>=len(solo)-1:
655
656
                                                     vehicleindex=0
vehicleisolo[vehicleindex]
elif takeoffcheck==len(solo):
    print"Main controller: No drones available for
657
658
                                                                                                                                   ন
659
                                                           shuttingdown='v'
660
                                                           break
                                                     else:
661
662
663
                                                           vehicleindex2=vehicleindex+1
                                                           if vehicleindex2>=len(solo)-1:
vehicleindex2=0
664
665
```

666 667	# Increasing vehicle airspeed to max (10)
668 669	<pre>print "Set vehicle airspeed to 10" vehicle.airspeed = 1</pre>
670 671 672	<pre># Set home location as the location above takeoff # (for return later)</pre>
673	स homelat[vehicleindex]=(vehicle.location.global fra स me.lat)
074	homelon[vehicleindex]=(vehicle.location.global fraず me.lon)
675 676 677 678 679 680 681 682 683 684 685	<pre># Print for debug print " Home Latitude: %s" % homelat[vehicleindex] print " Home Longitude: %s" % homelon[vehicleindex] b = threading.Thread(target=loiter) b.start() c = threading.Thread(target=replace) c.start() traceback.print exc() break</pre>
686 687	else: vehicleindex+=1
688 689 690 691 692 693 694 695 696	<pre>if vehicleindex>=len(solo)-1: vehicleindex=0 if unavailablevehicles>len(solo)-1: print"No vehicle available" shuttingdown='v' break</pre>
697 698 699	<pre>while True and shuttingdown=='n': # If battery level below limit or voltage level below limit if loitervehicle.battery.voltage < battery volt limit or if loitervehicle.battery.level < battery level limit or if uptervehicle.battery.level < battery level limit or if uptervehicle.battery.level < battery level limit. </pre>
700 701 702 703	<pre>print 'Battery below limit' time.sleep(1) if replaceshutdown=='Y': shuttingdown='Y'</pre>
704 705 706	<pre>if vehicle2status==1:</pre>
708	currentaltitude=vehicle2.location.qlobal relative frame.alt
710	dist2 = get distance metres(vehicle2.location.global frame, loitervehicle location global frame)
711	print " Home: Distance to replacement drone: %s Altitude a
712 713 714 715 716 717 718 719 720 721	<pre>if dist2 < 10: print 'Within replacement distance' with inter lock: rvehicle=loitervehicle loitervehicle=vehicle2 replacestop='y' vehicleindex=vehicleindex2 rvehicleindex=vehicleindex2</pre>


APPENDIX B. SCREENSHOTS OF REPLACEMENT FIELD OF VIEW

The pictures on the left are the patrolling SUAS field of view while those on the right are the swapping SUAS field of view.

Test 1: Square Loop Swap 1, Impossible City



Test 2: Square Loop Swap 2, Impossible City



Test 3: Square Loop Swap 3, Impossible City



Test 4: Straight Line Swap 1, Impossible City



Test 5: Straight Line Swap 2, Impossible City



Test 6: Straight Line Swap 3, Camp Roberts



Test 7: Straight Line Swap 4, Camp Roberts



Test 8: Straight Line Swap 5, Camp Roberts



Test 9: Straight Line Swap 6, Camp Roberts



Test 10: Square Loop Swap 4, Impossible City



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