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CLOSURE ANALYSIS AND TEST STUDY VOLUME IV

SUBSCALE STATIC TEST REPORT

APPENDIX 2

TECHNICAL REPORT NO. 4171-1 31 JULY 1969

Prepared for

DEPARTMENT OF THE AIR FORCE SPACE AND MISSILE SYSTEMS ORGANIZATION AIR FORCE SYSTEMS COMMAND NORTON AIR FORCE BASE, CALIFORNIA

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FOREWORD

This final report, which is composed of Volumes I through VIII, contains the results of a research study conducted by The Ralph M. Parsons Company under the direction of John E. McCarney. The Ralph M. Parsons Company personnel making significant technical contribution to this effort include David M. Hopper, Philip R. Sands, Richard C. Mayer, and Philip Mannes.

The study was performed from 1 June 1967 to 31 July 1969 under Contract No. F04694-67-C-0105 for the Department of the Air Force, Space and Missile Systems Organization (AFSC), Norton Air Force Base, California 92409. The SAMSO project officers were Maj G. W. Barnes, Capt F. G. Harms, and 1st Lt H. S. Yoshioka. The Aprospace Corporation provided systems engineering and technical direction, with Warren Pfefferle acting as Technical Director.

This technical report has been reviewed and is approved.

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CLOSURE ANALYSIS AND TEST STUDY

INDEX OF VOLUMES

- VOLUME I PROJECT SUMMARY AND ABSTRACT OF FINDINGS
- VOLUME II TECHNICAL REPORT

1

I

I

- VOLUME III RADIATION ANALYSIS (Appendix 1) [CLASSIFIED]
- VOLUME IV SUBSCALE STATIC TEST REPORT (Appendix 2)
- VOLUME V SUBSCALE OPERATING TEST REPORT (Appendix 3)
- VOLUME VI SUBSCALE DYNAMIC TEST REPORT (Appendix 4)
- VOLUME VII TEST HARDWARE DESCRIPTION AND DATA (Appendix 5)
- VOLUME VIII CLOSURE ANALYSIS AND TEST STUDY CRITERIA (Appendix 6) [CLASSIFIED]

APPENDIX 2

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SUBSCALE STATIC TEST REPORT

ABSTRACT

This report describes subscale static tests of composite silo closure and closure support models performed by The Ralph M. Parsons Company as a part of the Closure Analysis and Test Study. A total of 41 tests were performed in two phases. The effect on total load capacity caused by varying a broad range of design parameters including strength and relative thicknesses of steel shell material, span-to-depth ratio, support configuration, and concrete strength were investigated. Closure/closure support combinations with ultimate load capacities ranging from 1000 psi to 23,000 psi were tested.

Manual, empirical and finite-element design and analytical techniques are presented which will yield accurate predictions of the ultimate load capacities of full-scale composite closures.

TABLE OF CONTENTS

1

-

I

I

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IJ

1

Ι

I

I

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I

J

()

10	
1.0	INTRODUCTION
	1.1 Background
	1.2 Scope of Analysis and Tests
2.0	DESCRIPTION OF TEST HARDWARE AND INSTRUMENTATION
	2.1 Instrumentation
	2.1.1 Phase I Instrumentation
	2.1.2 Phase II Instrumentation
	2.2 Model Construction
3.0	SUBSCALE STATIC TESTS
	3.1 Phase T
	3.2 Phase I
1.0	5.2 Phase 11
4.0	SUMMARY AND CONCLUSIONS A2-205
	4.1 Effects of Various Structural Parameters
	4.1.1 Steel Plate Thickness A2-205
	4.1.2 Steel Strength
	4.1.3 Concrete Strength
	4.1.4 Span-to-Depth Ratio
	4.1.5 Bearing Width
	4.1.6 Interaction Phenomena
	4.1.7 Support Structure
2	4.2 Best Representation of Closure Det
	4.3 Regression Analysis of Mart D.
	4.4 Conclusion

A2-ix

LIST OF ILLUSTRATIONS

*** ***

-

...

11

I margaret

11

1

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t manual 1

a meriliana a

Į

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IJ

i

2-1	Proposed Fixture for Subscale Static Tests
2-2	Subscale Static Test Fixture Phase I Assembly - Front View
2-3	Schematic of Test Assembly
2-4	Closure Instrumentation Array
2-5	Silo Instrumentation Array.
2-6	Low-Friction Support Test Setup
2-7	Silo Deflection Gauge Mountings
2-8	Instrumentation Array for Closures - Tests 16 Through 00
2-9	Instrumentation Array for Closures and Silos - Tests 29, 30, 31 and 34
2-10	Instrumentation Array for Closure M130-9 and Silo M129-0 - Test 32 (Hercules).
2-11	Instrumentation Array for Closure M130-8 and Silo M128-0 - Test 33 (Rock Test I)
2-12	Design Geometry and Instrumentation Gauge Array for Silo M127-05 - Test 35 (Rise and Rotate).
2-13	Deflection Gauge Mounts
2-14	Closure Shell Design
2-15	Silo Design
2-16	Steel Shell Design Specifications for Closures M120-2E and M130-1 Through M130-14
2-1.7	Design Geometries of Silos M131-1 Through -4 and M131-6, -7, -9, and -10
2-18	Design Geometry of Silo M129-0 - Test 32 (Hercules)
2-19	Design Geometry of Silo M128-0 - Test 33 (Rock Test I)
2-20	Design Geometry and Instrumentation Gauge Array for Closure M127-0 - Test 35 (Rise and Rotate)

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A Contract of

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1

3-1	Closure Geometry for Tests 1 and 2 (University of Illinois) A2-39
3-2	Test 1 (Test Photograph)
3-3	Test 2 (Test Photograph)
3-4	Test 2 (Test Photograph)
3-5	Closure M120-1A Test 1 - Pressure vs. Deflection
3-6	Closure M120-1B Test 2 - Pressure vs. Deflection
3-7	Closure Geometry for Tests 3 and 4
3- 8	Test 3 (Test Photograph).
3-9	Test 3 (Test Photograph)
3-10	Test 3 (Test Photograph)
3-11	Test 4 (Test Photograph)
3-12	Closure M120-2A Test 3 - Pressure vs. Deflection
3-13	Closure M120-2B Test 4 - Pressure vs. Deflection
3-14	Support Geometry for Tests 5 and 6.
3-15	Support M121-1A Test 5 - Pressure vs. Deflection
3-16	Support M121-1B Test 6 - Pressure vs. Deflection
3-17	Closure M120-2C Test 5 - Pressure vs. Deflection
3-18	Closure M120-2C Test 6 - Pressure vs. Deflection
3-19	Test 5 (Test Photograph).
3-20	Test 5 (Test Photograph).
3-21	Test 6 (Test Photograph)
3-22	Test 6 (Test Photograph)
3-23	Support Geometry for Test 7
3-24	Support M121-2A Test 7 - Pressure vs. Deflection
3-25	Closure M120-2C Test 7 - Pressure vs. Deflection
	$\mathbf{A2-58}$

A2-xi

.

The

. .

9. A

. .

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•••

* 4 5 #

....

Br 10

1 ------ 1

11

[]

3-26 Test 7 (Test Photograph)	
3-27 Closure Geometry for Test 8	A2-59
3-28 Support M121-2A Test 8 - Pressure vs. Deflection	A2-60
3-29 Closure NI20-3A Test 8 - Pressure vs. Deflection	A2-62
3-30 Test 8 (Test Photograph)	A2-63
3-31 Test 6 (Test Photograph)	A2-64
3-32 Support Geometry for Test Q	A2-64
3-33 Support M121-3A Test 9 - Pressure	A2-65
3-34 Closure M120 3P meet 0	A2-67
3-35 Test Q (Test D)	A2-67
3-36 Test 9 (lest Photograph)	A2-68
3.37 met o (met Photograph)	A2-69
5-57 Test 9 (Test Photograph)	A2-69
3-38 Test 9 (Test Photograph)	A2-70
3-39 Closure Geometry for Test 10	42-71
3-40 Support M121-2B Test 10 - Pressure vs. Deflection	AQ 70
3-41 Closure M120-5A Test 10 - Pressure vs. Deflection.	A2-(3
3-42 Test 10 (Test Photograph)	A2-74
3-43 Test 10 (Test Photograph)	A2-75
3-44 Test 10 (Test Photograph)	A2-75
3-45 Support M121-3B Test 11 - Pressure ve Declarte	A2-76
3-46 Closure M120-5B Test 11 - Pressure ve Declart	A2-78
3-47 Test 11 (Test Photograph)	12-78
3-48 Test 11 (Test Photograph)	2-79
3-49 Closure Geometry for Tests 12 and 15	2-79
3-50 Closure M120-44 Test 10 .	2-80
A result - Pressure vs. Deflection	2-81

A2-x11

...

8-10 6 p

-

1

1

I

18

3-51	Closure M120-4B Test 15 - Pressure vs. Deflection A	2-81
3-52	P Test 12 (Test Photograph)	2-82
3-53	Test 15 (Test Photograph)	2-82
3-54	Closure Geometry for Test 13	2-83
3-55	Closure M120-5B Test 13 - Pressure vs. Deflection	2-85
3-56	Test 13 (Test Photograph)	- 86
3-57	Closure Geometry for Test 14	-87
3-58	Test 14 (Test Photograph)	-89
3-59	Closure M120-2D Test 14 - Pressure vs. Deflection	-90
3–60	Closure Geometry for Test 16	-93
3-61	Closure M120-2E Test 16 - Pressure vs. Deflection	-95
3-62	Data Plot - Test 16	-96
3-63	Test 16 (Test Photograph)	-97
3-64	Test 16 (Test Photograph)	-07
3-65	Test 16 (Test Photograph)	-08
3-66	Test 16 (Test Photograph)	_08
3-67	Closure Geometry for Test 17	_00
3-68	Closure M130-1-7 Test 17 - Pressure vs. Deflection	.00
3-69	Data Plot - Test 17 A2-	100
3-70	Test 17 (Test Photograph)	.02
3-71	Closure Geometry for Test 18	.02
3-72	Closure M130-1-6 Test 18 - Pressure vs. Deflection	.05
3-73	Data Plot - Test 18	.04
3-74	Test 18 (Test Photograph)	.07
3-75	Closure Geometry for Test 19	.00
		U I

A2-xiii

I

Ī

12.5

-

1

]]

1

ĪĨ

[]

[]

[]

3-76	Closure M130-2-2 Test 19 - Pressure vs. Deflection
3-77	Data Plot - Test 19
3-78	Test 19 (Test Photograph)
3-79	Closure Geometry for Test 20
3-80	Closure M130-2-1 Test 20 - Pressure vs. Deflection
3-81	Data Plot - Test 20
3-82	Test 20 (Test Photograph)
3-83	Closure Geometry for Test 21
3-84	Closure M130-3 Test 21 - Pressure vs. Deflection
3-85	Data Plot - Test 21
3-86	Test Photograph (Test 21)
3-87	Test Photograph (Test 21)
3-88	Closure Geometry for Test 22
3-89	Closure M130-4 Test 22 - Pressure vs. Deflection
3-90	Data Plot - Test 22
3-91	Test 22 (Test Photograph)
3-92	Closure Geometry for Test 23
3-93	Closure M130-7 Test 23 - Pressure vs. Deflection
3-94	Data Plot - Test 23
3-95	Test 23 (Test Photograph)
3-96	Closure Geometry for Test 24
3-97	Closure M130-1-3 Test 24 - Pressure vs. Deflection
3-98	Data Plot - Test 24
3-99	Test 24 (Test Photograph)

A2-xiv

,

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Π

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3-100	Closure Comptant for Mark or
2 101	Closure Geometry for Test 25
3-101	Closure M130-5 Test 25 - Pressure vs. Deflection
3-102	Data Plot - Test 25
3-103	Test 25 (Test Photograph)
3-104	Closure M133-1 Test 26 - Pressure vs. Deflection
3-105	Data Plot - Test 26
3-106	Test 26 (Test Photograph)
3-107	Test 26 (Test Photograph)
3-108	Closure M133-2 Test 27 - Pressure vs. Deflection
3-109	Data Plot - Test 27
3-110	Test 27 (Test Photograph)
3-111	Test 27 (Test Photograph)
3-112	Closure Geometry for Test 28
3-113	Closure M130-6 Test 28 - Pressure vs. Deflection.
3-114	Data Plot - Test 28
3 -1 15	Test 28 (Test Photograph)
3-116	Closure Geometry for Test 29.
3-117	Closure M130-14 Test 29 - Pressure vs. Deflection
3-118	Data Plot - Test 29
3-119	Test 29 (Test Photograph)
3-120	Closure Geometry for Test 30.
3-121	Closure M130-13 Test 30 - Pressure ve Declastica
3-122	Data Plot - Test 30
3-123	Model Geometry for Test 21
3-124	Closure M130 7 1 and Guard
J 467	Deflection

A2-XV

Ī,

Hen Vi

1

1

1

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I

I

I

T

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A

IJ

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3-125	Data Plot - Test 31
3-126	Test 31 (Test Photograph)
3-127	Test 31 (Test Photograph)
3-128	Test 31 (Test Photograph)
3-129	Test 31 (Test Photograph)
3-130	Model Geometry for Test 32 (Hercules)
3-131	Closure M130-9 and Support M129-0 Test 32 - Pressure vs. Deflection
3-132	Data Plot - Test 32
3-133	Test Photograph - Test 32
3-134	Model Geometry for Test 33 (Rock Test I)
3-135	Closure M130-8 and Support M128-0 Test 33 - Pressure vs. Deflection
3-136	Data Plot - Test 33
3-137	Test 33 (Test Photograph)
3-138	Test 33 (Test Photograph)
3-139	Test 33 (Test Photograph)
3-140	Test 33 (Test Photograph)
3-141	Model Geometry for Test 34
3-142	Closure M130-7-2 and Support M131-2 Test 3 ¹ - Pressure vs. Deflection
3-143	Data Plot ~ Test 34
3-144	Test 34 (Test Photograph)
3-145	Model Geometry for Test 35 (Rise and Rotate)
3-146	Closure and Support M127-0 Test 35 - Pressure vs. Deflection A2-177
3-147	Data Plot - Test 35
3-148	Model Geometry for Test 36

,

 $\left(\right)$

1

ſ

[]

[]

[]

1

Π

Ī

3-149 Closure M130-10 and Support M131-4 Test 36 - Pressure vs. Deflection.	
3-150 Test 36 (Test Photograph)	31
3-151 Test 36 (Test Photograph)	12
3-152 Model Geometry for Test 37	2
3-153 Closure M130-7-3 and Support Mar and	3
Deflection.	5
3-154 Test 37 (Test Photograph)	, ,
3-155 Test 37 (Test Photograph)	2
3-156 Model Geometry for Test 38.	2
3-157 Closure M130-11 and Support M131-7 Test 38 - Pressure	ľ
Deflection.)
3-158 Test 38 (Test Photograph)	
3-159 Test 38 (Test Photograph)	
3-160 Test 38 (Test Photograph)	
3-161 Test 38 (Test Photograph)	
3-162 Test 38 (Test Photograph)	
3-163 Model Geometry for Test 39.	
3-164 Closure M130-12 and Support M131-6 Test 30 Press	
Deflection.	
3-165 Test 39 (Test Photograph)	
3-166 Test 39 (Test Photograph)	
3-167 Test 39 (Test Photograph)	
3-168 Model Geometry for Test 40.	
3-169 Closure M130-7-6 and Support M131-10 Test 40 - Pressure vs.	
2 170	
3-110 Test 40 (Test Photograph)	
3-171 Test 40 (Test Photograph)	

A2-xvii

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3-172	Model Geometry for Test 41
3-173	Closure M130-7-5 and Support M131-9 Test 41 - Pressure vs.
	Deflection
3-174	Test 41 (Test Photograph)
4-1	Closure Test Data - Load Capacity vs. Plate Thickness/
	Span Ratio
4-2	Closure Test Data - Load Capacity vs. Steel Strength A2-208
4-3	Closure Test Data - Load Capacity vs. Concrete Strength A2-209
4-4	Closure Test Data - Load Capacity vs. Span/Depth Ratio A2-211
4-5	Closure Support Test Data - Load Capacity vs. Outer Shell Thickness/Outer Shell Diameter Ratio
4-6	Closure Support Test Data - Load Capacity vs. Bearing Plate Thickness/Closure Span Ratio
4-7	Static Test Data - Load Capacity vs. Nondimensional Shell
	Inickness

LIST OF TABLES

2-1	Number and Type of Instrumentation Channels (Phase I)	
2-2	Number and Type of Instrumentation Channels (Phase II) A2-17	
2-3	Phase II, Closure and Silo Design Parameters	
3-1	Static Test Specimen Geometry and Load Capacities	

A2-xviii

1.0 INTRODUCTION

This report describes subscale static tests of composite silo closure and closure support models performed as a part of the Closure Analysis and Test (CAT) Study by The Ralph M. Parsons Company. A total of 41 tests were performed in two phases. The tests encompassed a broad range of test parameters and variables, at ultimate load capacities ranging from 1,000 to 23,000 psi. The test variables included strength and relative thicknesses of materials; span-to-depth ratios; support stiffness and friction; symmetry; and silo configurations. These variables were applied to both closure and silo support models. The predominant model scale was approximately one to six.

1.1 Background

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Prior to initiation of the CAT project, only a limited amount of research had been performed in the study of response characteristics of composite concrete/steel plates. While studies exploring the strength of plain concrete under triaxial load conditions have been performed for many decades, structures had not been tested for the specific purpose of examining these phenomena until very recently. The significant efforts in this area were by the Illinois Institute of Technology Research Institute (IITRI) in 1964 and by the University of Illinois in 1967-1968. These two organizations studied the problems associated with composite closure design and explored the effects of various types of shear reinforcing on the load capacity of small-scale models.

The conclusions drawn by both groups of researchers were similar to the extent that essential agreement existed concerning the influence of the confining steel upon the apparent strength of the composite models. The extent of this influence however was not clearly defined by either effort.

In an earlier related effort, the Armour Research Foundation (ARF) performed a similar series of tests on small-scale composite support structures. Although this was a very limited test effort (three specimens), results indicated clearly that significant increases in load support capacity could be attained by laterally restraining the concrete subjected to high bearing stresses. As in the other two efforts, the conclusions were somewhat qualitative.

1.2 Scope of Analysis and Tests

Recognizing the need for definitive research to advance the technology of hardened construction, and fully aware of the interesting research into composite construction described above, SAMSO/Aerospace outlined a comprehensive and parametric plan to develop empirical design data applicable to advanced missile facility design and construction. This approach was structured (in the contract work statement) to identify the parametric nature of the necessary analytical and experimental work and specified a minimum level of effort. Implicit however was the latitude to restructure the effort to pursue significant findings. That this latitude was exercised on several occasions can be demonstrated by the fact that the proposed level of effort in the area of static tests and analysis was increased threefold following the award of the contract.

The scope of the analysis and test efforts, as amended and redirected, included the following:

a. Perform analyses of all influencing factors and design both silo closure and silo support structure models to fail at assigned loads. These loads were preliminarily established at 4000, 8000, and 12,000 psi and were later expanded to include 20,000 and 30,000 psi.

b. Examine and establish the effects of variations of material strengths on the load capacities of the various structures. Concrete strengths (f_c^{\prime}) were established at 3000, 6000 and 12,000 psi. Steel strengths (f_y) were established at 36,000, 70,000, 150,000 and 180,000 psi.

c. Examine and establish the influence of steel stiffness (as opposed to strength) upon the load capacity and ductility of the structures.

d. Determine the effects of friction at the closure/support interface; μ values were arbitrarily established at < 0.1, ~ 0.02 and > 0.5. In an effort to separate the effects of friction from the interaction of the closure and the silo, the friction tests were to be performed on a nonyielding (nonsilo) support.

e. Determine the effects of closure/silo support interaction and, if favorable to the gross load capacity, develop designs to ensure this interaction under shock and vibration loads.

f. Examine the effects of structural symmetry in an effort to identify variations in load capacity, response and predictability.

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g. Examine thick plate versus thin plate relationships by varying span-to-depth ratios.

h. Examine the influence of relative bearing widths, with the aim of establishing a practical minimum.

i. Examine static and dynamic test results in an effort to establish correlation and to identify (a range of) dynamic load factors.

j. From analysis of the static test results, develop a semi-empirical design technique applicable to full-scale composite structures.
2.0 DESCRIPTION OF TEST HARDWARE AND INSTRUMENTATION

As originally proposed, a maximum of 6000 psi was to have been sufficient for all static tests. At model scales of up to 1/5, this maximum allowed a rather straightforward design solution. During the proposal effort, a test fixture was designed that tested two specimens at a time (Figure 2-1). The specimens were placed back to back and the top surfaces of both were loaded with a flat jack. Reaction loads were taken, through the supports, by opposing platens connected by tie-bolts. While somewhat primitive, this was an adequate and inexpensive solution to the proposed maximum test pressures.

When the maximum test pressure was increased to 12,000 psi following contract award, this design could not be extrapolated, and parametric studies were made to develop an adequate technique. These studies are covered in Appendix 5, Test Hardware Description and Data, and require little elaboration here. As a result of these studies, the test fixture pictured in Figure 2-2



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PROPOSED FIXTURE FOR SUBSCALE STATIC TESTS

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and graphically presented in Figure 2-3 was proposed. This fixture develops 12 million pounds of force at 15,000 psi and, with the adapters available, can hydrostatically load a uniaxial test envelope of up to 55 inches diameter by 111 inches high. This diameter could be increased to 65 inches by providing one additional adapter. The fixture was designed and fabricated by Parsons and assembled by Nuclear Weapons Effects Division (NWED) personnel of the Waterways Experiment Station (WES) at Vicksburg, Mississippi. The fixture was installed in the Central Firing System at WES, which has a reaction capacity of approximately 60 million pounds.

As can be seen in Figure 2-3, the test structures were loaded by means of a contained elastomer, pressurized by the large (32-1/4-inch diameter) hydraulic cylinder. Radial forces exerted by the elastomer were resisted by a heavy "T" section ring. The elastomer was sealed against extrusion by means of 3/16-inch-diameter aluminum wire seals placed on each side of the leg of the section. Since the elastomer behaved as a fluid at pressures greater than approximately 100 psi, and was less than 6 percent compressible at 15 ksi, approximately three-eighths of the volume of the 1-inch-thick pad was available to deflect the test structures. This resulted in a centerline deflection capability of approximately 1 inch on the 32-1/4-inch-diameter models, the limiting factor being the complete compression of the seal rings. Naturally, on larger (52-inch diameter) or smaller (12-inch diameter) modèls, the deflection capability varied proportionally since the seal diameter was 3/16 inch for all models. On the tests with an anticipated ultimate unit





SCHEMATIC OF TEST ASSEMBLY

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FIGURE 2-3

load capacity of less than 12,000 psi, full annealed (1100-0) aluminum wire was used for the seals. This wire has a yield strength of 5000 psi and made an excellent seal. At test pressures above 12,000 psi, 1100-H14 wire was used, providing a yield strength of 17,000 psi. Both types of wire were heli-arc welded into continuous rings.

2.1 Instrumentation

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Prior to the start of the tests, rather ambitious plans were made concerning the instrumentation to be used. It was hoped that by accumulating masses of strain and deflection data the failure mechanisms of the various models could be pinpointed, aiding in the correlation with the analytical data. Unfortunately, this correlation proved to be no simple task. Three basic factors contribute to the problem: (1) throughout the tests, the quality of the strain data was somewhat suspect, being erratic and in some cases contrary to simple observations; (2) the responses of the various structures were relatively unknown and therefore it was difficult to range the instruments properly; and (3) receipt of the data rather late in the project lift insufficient opportunity for a comprehensive evaluation.

However, so that others may benefit from our experience, the instrumentation schemes used in the test project are described following. In the test descriptions (Section 3.0) it will be noted that primary emphasis is placed on the centerline load-deflection curves which proved to be the most useful data collected.

2.1.1 Phase I Instrumentation

Strain Measurements

Table 2-1 gives the number of strain measurements made for each of the first 15 tests (Phase I). The arrangement for the closure models is shown in Figure 2-4. The arrangement for the silo models is shown in Figure 2-5. As shown in these two figures, strain measurements were recorded on the exterior of the steel shell and on 1/4-inch-diameter deformed instrumentation bars placed on the concrete fill.

The strain gauges used in all tests were 120-ohm, epoxy foil backed gauges made by Micro-Measurements, Model No. EA-06-250BG120. These had a length of 1/4 inch and a gauge factor of 2.095. Figure 2-6 shows a typically instrumented closure model being set up for testing.

Deflection Measurements

Model deflections were measured across the closure bottom, vertically at the support bearing, and laterally on the inner silo wall (Figure 2-7). The large gauges shown in the lower photograph are Bourns Model 156 with 2-1/2 inches of travel, and the smaller gauges are Bourns Model 141 with 1/2 inch of travel. Both are linear potentiometers, with the smaller gauge of 5000-ohm resistance and the larger at 10,000 ohm. The gauge shafts were spring-loaded against the test specimen.

Pressure Measurements

Test pressures at the hydraulic cylinder were measured with a 20,000psi Statham transducer, Model Pa-822-20M. This proved to be an excellent instrument, almost perfectly linear to 10,000 psi, the capacity of the WES

TABLE 2-1 NUMBER AND TYPE OF INSTRUMENTATION CHANNELS

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Internal Internal External Deflection Stratin Pressure Channel M120-1A Low friction 11 14 5 3 -ritin 2 -ritin 2 -ritin 11 31 M120-1B Low friction 22 14 5 -1 -1 31 31 M120-1B Low friction 22 16 5 -1 -1 22 21 M120-28 Low friction 22 16 5 -1 2 21 M120-28 Low friction 22 16 5 -1 2 2 M120-28 Low friction 22 16 5 16 11 8 1 60 M120-28 M121-28 -2 16 11 8 1 60 M120-31 M121-28 -2 16 2 16 11 8 1 60 M120-31		Closure	Support			Instrum	sentation Ch	annels			Total
Strain Deflection $\exists Train Deflection \exists Train di Train $					Closure			Support		Pressure	Channels
MI20-1A Internals Externals Internals Externals Externals MI20-1B Low friction 11 14 5 1 31 MI20-1B Low friction 22 16 5 2 22 MI20-2B Low friction 22 16 5 2 2 2 MI20-2B Low friction 22 16 5 16 1 8 1 2 45 MI20-2B Low friction 22 16 5 16 1 8 1 2 45 MI20-2B M21-1B 16 5 16 1 8 1 6 MI20-2B M21-2B 22 16 1 8 1 6 4 MI20-3B M21-2B 22 16 1 8 1 6 6 MI20-3B M21-2B 22				Str	ain	Deflection	Str	ain	Deflection		
W120-1A Iow Friction 11 14 5 1 31 M120-1B Low Friction 14 5 1 31 M120-1B Low Friction 22 16 5 2 2 2 M120-2B Low Friction 22 16 5 16 5 2 45 M120-2B Low Friction 22 16 5 16 1 8 1 82 M120-2B M21-1B 16 5 16 1 8 1 82 M120-2B M21-1B 16 5 16 1 8 1 60 M120-2B M21-2B 16 1 8 1 60 M120-3B M21-2B 16 1 8 1 60 M120-4B M21-2B 22	_			Internals	Externals		Internals	Externals			
W120-L1b Low friction 14 5 2 2 N120-L3k Low friction 22 16 5 2 2 2 N120-L3k Low friction 22 16 5 16 5 2 4 M120-L2k Low friction 22 16 5 16 11 8 1 8 4 M120-2c M121-L3b 16 5 16 14 8 1 8 4 M120-2c M121-L3b 16 5 16 14 8 1 8 M120-2s M121-L3b 16 5 16 14 8 1 8 M120-2s M121-L3b 16 14 8 1 6 4 M120-2s M121-2s 22 16 14 8 1 6	2	AL-051	Low friction	п	14	5	;	1	;	1	11
Mi20-28 Low Friction 22 16 5 - - - 2 4/5 Mi20-28 Low friction 22 16 5 - - - 2 4/5 Mi20-28 Low friction 22 16 5 16 1 8 1 2 4/5 Mi20-28 Mi21-18 16 5 16 14 8 1 8 1 8 Mi20-28 Mi21-28 16 5 16 14 8 1 8 1 8 Mi20-28 Mi21-28 22 16 5 16 14 8 1 6 Mi20-28 Mi21-28 22 16 5 16 14 8 1 6 Mi20-28 Mi21-28 22 16 5 16 14 8 1 6 Mi20-38 Mi21-28 22 16 5 1	x	0120-1B	Low friction	•	14	5	1	1	1	0	10
M20-28 Low friction 22 16 5 2 4,5 M20-28 M21-1A 22 16 5 16 14 8 1 1 82 M20-28 M21-1A 22 16 5 16 14 8 1 1 82 M120-28 M21-2A 16 5 16 14 8 1 1 82 M120-38 M21-2A 16 5 16 14 8 1 1 82 M120-38 M21-2A 16 5 16 14 8 1 6 6 M120-38 M21-2B 22 16 5 16 14 8 1 6 6 M20-58 M21-2B 22 16 5 16 14 8 1 6 6 M20-58 M21-2B 22 16 <td< td=""><td>x</td><td>120-2A</td><td>Low friction</td><td>55</td><td>16</td><td>5</td><td>1</td><td>1</td><td>;</td><td>Q</td><td>54</td></td<>	x	120-2A	Low friction	55	16	5	1	1	;	Q	54
MI20-3C MI21-1A 22 16 1 8 1 82 MI20-3C MI21-1B 16 5 16 14 8 1 1 82 MI20-3C MI21-2A 16 5 16 14 8 1 1 82 MI20-3A MI21-2A 16 5 16 14 8 1 60 MI20-3B MI21-2A 22 16 5 16 14 8 1 60 MI20-3B MI21-2B 22 16 5 16 14 8 1 60 MI20-5B MI21-2B 22 16 5 16 1 8 1 60 MI20-5B MI21-2B 22 16 5 16 1 1 60 MI20-5B MI21-3B 16 1 8 1 66 1 MI20-5B Low Frit	2	120-28	Low friction	22	16	5	1	;	1	Q	45
M120-26 M120-26 M121-1B 16 5 14 B 1 44 M120-23 M121-2A 16 5 16 14 8 1 60 M120-34 M121-2A 16 5 16 14 8 1 60 M120-35 M121-2A 16 5 16 14 8 1 60 M120-35 M121-2B 22 16 5 16 14 8 1 60 M120-5B M21-2B 22 16 5 16 1 8 1 60 M120-5B M21-2B 22 16 5 16 1 6 1 6 M120-5B M21-3B 14 8 1 6 1 6 1 6 1 6 1 6 1 1 6 1 1 1	z	120-20	VI-12TM	83	16	5	16	14	8	1	68
M120-2c M21-2d 16 1 8 1 60 M120-3b M21-2d 22 16 1 8 1 8 1 82 M120-3b M21-2d 22 16 5 16 14 8 1 82 M120-3b M21-2d 22 16 5 16 14 8 1 82 M120-3b M21-2b 22 16 5 16 14 8 1 82 M120-5b M21-2b 22 16 5 16 1 8 1 60 M120-5b M21-2b 22 16 5 1 1 8 1 60 M120-5b M120-5b M120-5b 1 1 8 1 5 5 M120-5b Low frietion 22 16 5 1 5 6 M120-5b Low frietion 22 16 5 1 7 6 6 M120-5b High	Z	120-20	BI-14TM	1	16	5	1	14	8	1	1
M120-34 M121-24 22 16 1 8 1 82 M120-38 M121-34 16 5 16 1 8 1 82 M120-38 M121-34 16 5 16 1 8 1 60 M120-58 M121-29 22 16 5 1 8 1 60 M120-58 M121-39 5 10 8 1 60 M120-58 M121-39 1 1 8 1 60 M120-58 M121-39 1 10 8 1 60 M120-58 M120-39 1 10 8 1 65 M120-58 Low frietion 22 16 5 1 5 1 1 61 M120-58 Low frietion 22 16 5 1 <	N	120-21	M21-2A	1	16	5	16	14	8	T	03
Mi20-38 M21-3A 16 14 8 1 60 Mi20-5A M021-2B 22 16 5 16 14 8 1 60 Mi20-5B M021-2B 22 16 5 14 8 1 60 Mi20-5B M121-3B 5 10 8 1 60 M120-5B M121-3B 5 10 8 1 60 M120-5B M121-3B 5 10 8 1 60 M120-1A Low friction 29 16 5 1 2 1 60 M120-5B Low friction 22 16 5 1 1 61 61 M120-5B High friction 22 16 5 1 1 44 M120-4B High friction 29 1 6 1 1 46 1 67	R	AE-051	M21-2A	52	16	5	16	14	80	1	82
ML20-5A ML21-2B 22 16 5 14 B 1 66 ML20-5B ML21-3B 5 10 8 1 66 ML20-5B ML20-5B ML21-3B 5 1 67 ML20-5B Low friction 29 16 5 10 8 1 24 ML20-5B Low friction 22 16 5 1 21 24 ML20-5B Low friction 22 16 5 1 1 44 ML20-2D High friction 29 16 5 1 44 ML20-4B High frietion 29 16 5 1 1 44	Ŧ	120-38	M21-3A	;	316	5	16	14	60	1	60
M120-58 M121-38 5 10 8 1 24 M120-14 Low Friction 29 16 5 11 51 M120-58 Low Friction 22 16 5 1 1 1 14 M120-20 High Friction 22 16 5 1 1 14 M120-20 High Friction 29 16 5 1 5 1 57	R	NG-02	M121-28	55	16	5	;	14	æ	1	8
M120-M Low fraction 29 16 5 1 51 51 M120-5B Low fraction 22 16 5 1 44 M120-2B High fraction 22 16 5 1 1 44 M120-2B High fraction 22 16 5 1 1 44 M120-2B High fraction 22 16 5 1 1 44 M120-4B High fraction 29 16 5 1 1 44	딮	20-5B	M121-3B	,	1	5	1	10	8	1	17
M120-5B Low Friction 22 16 5 1 14, 4120-2D High Friction 22 16 5 1 14, 0 20-4B High Friction 29 16 5 6 1 57	Į.	A1-05	Low friction	8	36	5	:	1	1	1	51
4120-20 High friction 22 16 5 1 14 020-48 High friction 29 16 5 6 1 57	5	85-02	Low friction	22	36	5	1	;	1	1	14
020-4B ligh friction 29 16 5 - 6 - 1 57	4	20-20	High friction	55	16	5	;	;	,	1	44
	ę.	20-4B	ligh friction	8	16	5	;	9	1	7	57

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SILO INSTRUMENTATION ARRAY

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FIGURE 2-5



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(a) Low-friction segmented support rings





(b) Placement of closure on support



Completed closure setup prior to placement in static test fixture (P)

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FIGURE 2-6

LOW-FRICTION SUPPORT TEST SETUP

- (c) Positioned closure with clamp



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(a) Silo side wall and inner bearing ring



(b) Closure center line FIGURE 2-7 SILO DEFLECTION GAUGE MOUNTINGS A2-15 dead-weight tester, and in good agreement with a 20,000-psi bourdon gauge of test quality above 10,000 psi. The bourdon gauge was used in most of the tests to verify the pressure traces read out on an X-Y plotter.

Recording Equipment

Analog test data was recorded on 36-channel CEC oscillographs and/or 14-channel FM tape recorders. Signal conditioning was through SAM and DANA dc amplifiers, coupled with galvo drivers. Quick-look data was recorded on X-Y or X-Y-Y plotters.

2.1.2 Phase II Instrumentation

The pause between the first and second phase tests gave us time to redesign the model instrumentation in an attempt to provide more useful data. The number of internal strain gauges was reduced drastically and the number of deflection measurements was increased (see Table 2-2). Figures 2-8 and 2-9 depict the instrument arrays used in Tests 16 through 31 and Test 34. The arrays for Tests 32, 33, and 35 are shown on Figures 2-10, 2-11 and 2-12, respectively. Closure bottom deflections only were recorded in Tests 36 through 41. All Phase II instrumentation was of the same type as thet used for Phase I; the only revision was the redesign of the deflection gauge mounts to facilitate instrumentation setup (Figure 2-13).

In the test descriptions (Section 3), it will be noted that selected data plots have been included for Tests 16 through 35. These plots result directly from the Phase II instrumentation plan and were received from WES in May 1969. Examination of these plots, as time permitted, gave us some

Test No.	t Closure	Support	Instrumentation Channels							Total	_
			Closure				Support	Pressure		Channels	
			Etrain		Deflection	Strain		Deflection			
			Internals	Externals		Internals	Externals				
16	ML 20-22	Low friction		14	4				<u> </u>		_
17	M130-1-1	7 Low friction	3	14	<u>ц</u>					19	
18	MI 30-1-6	Lew friction	3	14	4					22	
19	M1 30-2-2	2 Low friction	3	14	L.					22	
20	MC.30-2-1	Low friction	3	14	14				1	22	
21	M130-3	Low friction	3	14	4				1	22	
22	MI 30-4	Low friction	3	14	L.				1	22	
23	M130-7	Low friction	3	14	Ŀ				1	\$5	
24	M130-1-3	Low friction	3	14	h				1	22	
25	M130-5	Low friction	3	14	1.			•-	1	22	
2 6	M133-1	Low friction	3	14	h				1	22	
27	MI 33-2	Low friction	3	14	,				1	22	
28	M130-6	Low friction	3	14	•				1	22	1
29	MI30-14	Low friction	2				••	*-	1	22	
30	M130-13	Low friction							1	11	
31	M130-7-1	M131-1	2		4				1	11	
32	M130-9	M129-0				2	8	3	1	24	
33	M130-8	M122-0	2	4	- 1	2	8	3	1	25	
34	ML30-7-2	M 31-2	3		1	2	2	3	1	16	
35	MI 27-0	M127=05		4	4	2	8	3	1	24	
36	MI 30-10	Mialah	٤	4	1	2	8	3	1	21	
37	MI 30-7-3	V121-2			1			1	1	3	
8	M130-11	M 21-7			1			1	1	3	
	MI 20-12	M 21 6			1			1	1	3	
	M 20-7-6	-10 10			1			1	1	3	
$\tilde{,}$	-u_3(ли 31-10 Марало			1			1	1	3	
		M.31-9			1			1	1	3	
	-140-1	High friction			1				1	2	
<u> </u>	u40-2	High friction			1				1		

NUMBER AND TYPE OF INSTRUMENTATION CHANNELS

TABLE 2-2

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FIGURE 2-9 INSTRUMENTATION ARRAY FOR CLOSURES AND SILOS - TESTS 29,30,31 AND 34





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FIGURE 2-13 DEFLECTION GAUGE MOUNTS

assurance that the conclusions drawn throughout this report are correct. Of particular interest are the deflections of the side shell of the closure. It will be seen that the applied load increased steadily until relatively minor deflections occurred in the side shell, and as the deflections increased the primary load/deflection curve bent over sharply.

2.2 Model Construction

All test specimens were designed and fabricated by Parsons and were then turned over to WES for instrumentation and concrete placement. Welding of the models was in accordance with Section VIII of the ASME code. It is significant that only two weld failures were experienced and these occurred in two high strength steel models constructed of 4130 chrome molybdenum steel of 69-ksi and 180-ksi yield strengths and at test pressures of 16+ ksi and 23 ksi, respectively.

The Phase I (Tests 1 through 15) closure models are shown in Figure 2-14; the support models are shown in Figure 2-15. The Phase II (Tests 16 through 41) nominal closure and nominal silo models are shown in Figures 2-16 and 2-17, respectively. Table 2-3 gives a composite view of both closure and silo design parameters, by test.

In an effort to correlate with previous HEST results, two specimens were constructed to model structures tested in the Hercules (Test 32) test and in Rock Test I (Test 33). The silo supports for these tests are shown in Figures 2-18 and 2-19, respectively.

Test 35 represented the test of a prototype configuration of the rise and rotate closure concept and is shown on Figure 2-20.



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SECTION A-A

		SILO D	ESIGN	i i	
MODEL	A	8	C	0	1
MIZI-I	1/2"	6-1/2*	4"	1.	12 GAGE
M121-2	1-1/2"	7-1/2*	4"	1*	5/8"
M121-3	1-1/2"	7-1/2"	4*	1"	12 GAGE

FIGURE 2-15

SILO DESIGN A2-26

STEEL SHELL DESIGN SPECIFICATIONS FOR CLOSURES MI20-2E AND MI30-1 THROUGH MI30-14 FIGURE 2-16

CLOBURE THICKNELS DEPTH MARETER MARETER MARETER MARETER MARETER MARETER MARETER MARETIR MARETIR MARETIR MARETIR MARETIR MARETIR MARETIR MARETIRE MARETIRE <th></th> <th>STEEL</th> <th><</th> <th></th> <th>v</th> <th></th>		STEEL	<		v	
M130-2E. 1/4 14.00 32.25 2-1/4 W130-2 1/2 14.00 32.25 2-1/4 M130-3 1/2 14.00 32.25 2-1/4 M130-4 1/2 14.00 32.25 2-1/4 M130-5 1/4 0.00 32.25 2-1/4 M130-6 1/4 0.00 32.25 2-1/4 M130-6 1/4 0.00 32.25 2-1/4 M130-6 1/4 0.00 32.25 2-1/4 M130-1 1/1 14.00 32.25 2-1/4 M130-13 1/4 14.00 32.25 2-1/4 M130-13 1/4 14.00 32.25 2-1/4 M130-13 1/4 14.00 32.25 2-1/4 M130-13 1/	CLOSURE	THICKNESS	DEPTH	DIAMETER	MACHINED BEARING AREA, IN	STEEL STRENGTH
w130-1 1/4 1400 22.33 2/1/4 w130-2 1/4 14.00 22.33 2/1/4 w130-3 1/2 14.00 22.33 2/1/4 w130-3 1/2 14.00 22.33 2/1/4 w130-3 1/2 14.00 22.33 2/1/4 w130-4 1/2 14.00 22.33 2/1/4 w130-5 1/4 8.00 22.33 2/1/4 w130-6 1/2 14.00 22.33 2/1/4 w130-6 1/2 14.00 22.35 2/1/4 w130-6 1/2 14.00 22.35 2/1/4 w130-10 1/2 14.00 22.30 1/1/4 w130-13 1/2 14.00 22.30 1/1/4 w130-13 1/2 14.30 22.30 1/1/4 w130-13 1/2 14.30 22.30 1/1/4 w130-13 1/2 14.30 22.30 1/1/4 w130-14	M120-2E1	1/4	14.00	37 36		
WT302 1/4 14.00 32.33 2-1/4 WT302 1/4 14.00 32.33 2-1/4 MT304 1/2 14.00 32.35 2-1/4 MT304 1/2 <td< td=""><td>with.</td><td></td><td></td><td>C7:36</td><td></td><td>36,000</td></td<>	with.			C7:36		36,000
Million 1/4 1400 32.35 2-1/4 Million 1/2 14.00 32.35 2-1/4 Million 1/2 14.00 32.35 2-1/4 Million 1/2 14.00 32.35 2-1/4 Million 1/2 0.00 32.35 2-1/4 Million 1/2 0.00 32.35 2-1/4 Million 1/2 1.00 32.35 2-1/4 Million 1/2 1.00 32.35 2-1/4 Million 1/2 1.00 32.35 2-1/4 Million 1/2 1.4.00 32.35 2-1/4 Million			8	32.25	2-1/4	36.000
M130-3 1/4 14.00 22.35 21/4 M130-4 1/2 BOTTOM 14.00 22.35 21/4 M130-4 1/2 BOTTOM 14.00 22.35 21/4 M130-4 1/2 BOTTOM 14.00 22.35 21/4 M130-4 1/4 BOTTOM 14.00 22.35 21/4 M130-5 1/7 14.00 22.35 21/4 M130-6 1/7 14.00 22.35 21/4 M130-10 1/7 14.00 22.35 21/4 M130-10 1/7 14.00 22.35 21/4 M130-13 1/7 14.00 22.30 11/4 M130-13 1/7 14.30 22.30 11/4 M130-14 1/7 14.33 22.30 11/4 M130-13 3/4 8.00 22.35 21/4 M130-14 1/7 14.33 22.30 11/4 M130-13 3/4 8.00 22.35 21/4 M130-14 </td <td>2-05</td> <td>1/4</td> <td>14.00</td> <td>32.25</td> <td>2-1/4</td> <td></td>	2-05	1/4	14.00	32.25	2-1/4	
Image 1/2 SIOE 2/2 SIOE <t< td=""><td>E-021M</td><td>1/4</td><td>14.00</td><td>32.25</td><td>2-1-2</td><td></td></t<>	E-021M	1/4	14.00	32.25	2-1-2	
MIX55 1/4 0.00 32.25 7/4 MIX07 1/4 21.00 32.25 7/4 MIX07 1/2 21.00 32.25 7/4 MIX07 1/2 14.00 32.25 7/4 MIX07 1/2 14.00 32.25 7/14 MIX07 1/2 14.00 32.25 7/14 MIX07 1/3 14.00 32.25 7/14 MIX07 1/3 14.00 32.25 7/14 MIX07 1/3 14.00 32.25 7/14 MIX010 1/2 14.00 32.25 7/14 MIX013 1/3 14.00 32.35 7/14 MIX013 1/4 14.00 32.35 7/14 MIX014 7/2 14.35 22.00 19/16 MIX013 3/4 0.00 32.35 7/14 MIX014 7/2 14.35 22.00 19/16 MIX014 21.00 32.35 <td>7-0E IM</td> <td>1/2 SIDE</td> <td>14.00</td> <td>32.25</td> <td>2-1/4</td> <td>000 %</td>	7-0E IM	1/2 SIDE	14.00	32.25	2-1/4	000 %
MI304 1/4 21.00 22.25 21.44 MI304 1/2 21.00 22.25 21.44 MI304 1/2 14.00 22.25 21.44 MI304 1/2 14.00 22.25 21.44 MI304 1/2 14.00 22.25 21.44 MI3041 1/2 14.00 22.25 21.44 MI3041 1/2 14.00 22.25 21.44 MI3041 1/2 14.25 22.20 197.16 MI3041 1/2 14.25 22.20 197.16 MI3041 1/2 14.25 22.00 197.16 MI3043 3/4 6.00 22.25 21/4 MI3043 3/4 6.00 22.26 197.16 MI3043 3/4 6.00 22.26 197.16 MI3044 21.00 22.20 21.74 21.74 MI3045 21.00 22.20 21.74 21.74 MI3044 21.0	M130-5	1/4	00.0	32.25	2-1/4	
MI30-7 1/2 1/2 14.00 22.35 21/4 MI30-6 1/4 14.00 22.35 21/4 MI30-10 1/2 14.00 22.35 21/4 MI30-10 1/2 14.00 22.35 21/4 MI30-10 1/2 14.00 22.35 21/4 MI30-11 1/2 14.00 22.35 21/4 MI30-13 1/2 14.35 22.00 19/16 MI30-13 1/2 9.50 22.00 19/16 MI30-13 3'4 8.00 22.30 19/16 MI30-14 3'4 21.00 22.35 21/4 MI30-13 3'4 21.00 22.35 21/4 MI30-14 3'4 21.00 22.35 21/4 MI30-13 3'4 21.00 22.35 21/4 MI30-14 3'4 21.00 22.35 21/4 MI30-13 3'4 21.00 22.35 21/4 MI30-14<	*9 IN	1/4	21.00	22.25	2-1/4	
MI30-6 \$.16 14.00 22.25 21/4 MI30-6 1/4 14.00 22.35 21/4 MI30-10 1/2 14.00 22.35 21/4 MI30-11 1/2 14.00 22.35 21/4 MI30-12 1/2 14.35 22.00 19/16 MI30-13 1/2 14.35 22.00 19/16 MI30-13 1/2 9.30 22.00 19/16 MI30-13 3/4 8.00 22.30 19/16 MI30-13 3/4 8.00 22.35 21/4 MI30-13 3/4 21.00 22.35 21/4 MI30-13 3/4 21.00 22.35 21/4 MI30-13 3/4 21.00 22.35 21/4 MI30-13 PLAIN CONC 21.00 22.35 21/4 MI30-1 Y.10 21.00 22.35 21/4 MI30-1 Y.10 21.00 22.35 21/4 MI30-13	M130-7	2/1	14.00	2.25	2-1/1	
MI304 1/1 14.00 22.35 4-1/4 MI30-10 1/2 14.00 22.35 4-1/4 MI30-12 1/2 14.00 22.35 4-1/4 MI30-13 1/2 14.35 22.00 19/16 MI30-13 1/2 14.35 22.00 19/16 MI30-13 3/4 8.00 22.20 19/16 MI30-13 3/4 8.00 22.30 19/16 MI30-14 7/2 14.25 22.00 19/16 MI30-13 3/4 8.00 22.30 19/16 MI30-14 7/4 8.00 22.35 21/14 MI30-13 3/4 8.00 22.35 21/14 MI30-14 7.00 22.35 21/14 21/24 MI30-15 7.10 22.35 21/24 21/24 MI30-14 7.10 21.00 22.35 21/24 MI30-15 7.10 22.35 21.00 22.35 MI30-14	POR IN	5/16	14.00	32.25	2-1/4	000 0
Million 1/2 14.00 22.35 2-1/4 Million 1/2 14.00 22.30 19/16 Million 1/2 14.33 22.00 19/16 Million 1/2 14.33 22.00 19/16 Million 1/2 14.33 22.00 19/16 Million 1/2 3.4 0.00 22.30 19/16 Million 3'4 0.00 22.30 19/16 11/16 Million 2'1.00 22.30 22.30 21/14 21/14 Million 2'1.00 22.33 22.35 21/14 21/14 Million 2'1.00 22.33 22.35 21/14 21/14 Million 2'1.00 22.33 22.35	+0EIN	1/1	14.00	32.25	8/1-8	000
M130-12 1/2 14.25 22.00 1-9/16 M130-12 1/2 9.50 22.00 1-9/16 M130-13 3'4 8.00 32.25 2-1/4 M130-1 3'4 21.00 32.25 2-1/4 M130-1 3'1 8.00 22.25 2-1/4 M130-1 3'16 5.23 11.50 1	01-06 IM	1/2	14.00	32.25	2-1/4	20,000
M130-13 1/2 9.50 1-9/16 M130-13 3'4 8.00 32.20 1-9/16 M130-14 3/4 8.00 32.25 2-1/4 M139-2 PLAIN CONC 8.00 32.25 2-1/4 M139-2 PLAIN CONC 8.00 32.25 2-1/4 M139-1 3'16 9.00 32.25 2-1/4		2/1	14,25	22.00	1-9/16	180.000
Nilocia 3'4 0.00 32.25 21/4 Millocia 31.00 32.25 21/4 Millocia 21.00 32.25 21/4		2/3	9:20	22.00	91/6-1	60.000
M1392 PLAIN CONC 21.00 32.25 241/4 M1392 PLAIN CONC 21.00 32.25 241/4 M1392 PLAIN CONC 21.00 32.25 241/4 M1392 PLAIN CONC 21.00 32.25 241/4			0.0	32.25	2-1/4	34.000
M139-2 PLAIN CONC 8.00 32.25 M139-2 ML 40-1 3.16 5.23 11.50 22.25 ML 40-1 3.16 5.23 11.50			21.00	32.25	2-1.4	34.000
M1372 PLAIN CONC 21.00 32.23		LAIN CONC	8.8	32.25	ł	. 1
miapri 3.16 5.25 11.50	2-66 1	FLAIN CONC	21.00	32.25	1	i
	I-10-1	91,12	5.25	11.50	I	

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GENERAL NOTES:

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MIL-570-00

SECTION VIII ASME CODE. NO CODE STAMP REQUIRED. ONE PIECE

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INSUE SUPPACES TO BE SANDELASTED TO INVITE METAL AFTER COL MACHINED SUPPACES ONLY. 3. ALL TOLENANCES TO APPLY IN THE UNRESTRAINED CONDITION.

A2-27



A. CLOSURES MI20-2E, MI30-1 - MI30-14, AND MI40-1





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TABLE 2-3 PHASE II, CLOSURE AND SILO DESIGN PARAMETERS

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				Clos	-					Frietlon 1	Mings and	Sile Sur				
	Tient I Toxuis	5	-	•	NIL:	Concret-	Steel r	Thickness	Identification No.	Createret	strel "	-	Teel Thi	Christian	(1m/4	(50
		Inch	Inch	Inch		pet	pat	Inch		ind	1 ad					
9	#P-OFTM	32.25	8	-	2.0	12,000	¥.,000	1/1	(Low friction segmented rings)			t,				
41	1-1-06TA	32.35	*	-	2.0	Nonshrink	×,000	VI		6				I		
	3-1-0(FM	32.25	R	-	2.0	3,000	¥,000	14			1-	1	-	2		
0	2-2-0(1)	32.2.	R	14	2.0	3,000	70,000	14			,	in the	-	24		
	1-5-04 TK	32.25	*	14	2.0	12,000	70,000	14	•	-			1	_		
	E-0615	32.2	R	14	2.0	6,000	150,000	VI	(Low friction segmented rings with			~	+	-		
-	+-Of Ta	32.25	R	11	5.0	6,000	36,000	1/2 sides			-			-		-
	1-061#	32.25	R	*	2.0	6,000	¥.,000	1/2	•					-		-
	E-1-0(14	32.2	8	-	2.0	6,000	36,000	VI	(Low friction seguented rings with		t	×		-		
	5-0ET.n	10.2	R	m	3.5	6,000	¥, 000	~	(Les friction segmented rings with		-			-		
	1-61 54	32	R	æ	1.5	6,000	Plain	concrete	(Low Priction segmented rings)		-			-		
	2-11 ta	27.22	*	-	1.33	6,000	Plain	concrete			-					
	-	32.35	R	1	1.33	6,000	36,000	~			-			-		
	41-06 LA	8.8	E	1	1.33	6,000	y, 000	V			-			-		
	£1-06EA	10.3	r,	-	3.5	6,000	36,000	VE	•		1	l		-		-
	1-1-0ELA	\$5.23	3	-	1.4	6,000	36,000	1/2	1-1614	0000')	W.000	-	2/2	1 4	-	ź
	6-06 IM	32.25	ST81.45	*	1.73	6,000	¥,000		(Merculas)	6,000	×.000	6.4	-			· inte
-	1-06 TM	19.2	21.375	-	2.02	6,000	000'0%	3/16	M128-0 (Noch Test I)	6,000	000'05	2.2	1/2	2	5	
5	2-1-0612	\$.3	30.25	-	2.16	6,000	36,000	2/2	5-171-S	6,000	36,000	1.875	1/2	1.		3
	0-1254	10.2	05.90	13.25	2.0	6,000	36,000	<	W127-CB (Nise and rotate)	6,000	36,000	•	1	1	-	-
-	01-1411-	32.35	8	-	2.0	0000"9	20,000	2/2		6,000	36,000	3.0		12 1	2	1
	E-1-0614	32.2	8	4	5.0	6,000	36,000	1/2	(-tith	6,000	36,000	3.0		10 10	~	6
-	II-OEIM	22.00	19	14.25	1.33	12,000	100,000	1/2	1-1674	12,000	180,000	2.5	1.0	.0		1/2
	21-061%	8.22	19	9.30	5.0	12,000	000'00		9-17.h	12,000	60,000	2.5	1.0	.0	5	2/2
	3-1-OLIN	12.2	8	1	5.0	6,000	X,000	4	01-1614	6,000	30.000	3.0	1/2	12 1	-	8
-	6-1-06TM	8.3	8	1	2.0	6,000	36,000	12	6-181M	6,000	¥,000	3.0		12 2/		1
-	t t-oyta	05.11	10	8.8	1.91	:	1	100	(Wigh friction solid ring)		Closure pr	vilounity	tested	dynamics		
-	5-04Em	1.50	0	5.8	1.91	1	1	126	(Wigh friction solid rise)							









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FIGURE 2-20 DESIGN GEOMETRY AND INSTRUMENTATION GAUGE ARRAY FOR CLOSURE M127-0 - TEST 35 (RISE AND ROTATE) A2-32

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3.0 SUBSCALE STATIC TESTS

Table 3-1 defines the grometry, material strength, and load capacity of the models tested. Low friction and high friction tests are also indicated. For the convenience of the reader, a typical cross section of a test model is included as part of this table.

3.1 Phase I

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The first fifteen static tests, designated as Phase I, were conducted to provide a baseline for future tests, to validate the test hardware, and to allow us to check our design approach.

Our initial plan was to duplicate all tests in the series. However, after the first four tests, where two pairs of identical models were tested with nearly equivalent load capacities and failure modes, it was decided to expend test effort in examination of closure/support variables rather than in duplication of test results. Thus, a great variety of variables were examined in Phase II of the series.

Full-size structures were designed and scaled down for testing purposes in the preliminary effort. It was necessary to consider the closure and the closure support as uncoupled since little was known about interaction phenomena. The variables examined were closure plate thickness, closure geometry, closure support plate thicknesses, and bearing restraint. Two models containing conical sections across the predicted shear plane were also included to determined if additional strength could be achieved with this type of section.

The Phase I tests are described in the following paragraphs.

LEGEND - TABLE 3-1

CLOSURE

- d DIAMETER
- A DEPTH
- IS SIDE SHELL THICKNESS
- & BOTTOM PLATE THICKNESS

CLOSURE SUPPORT

FOUNDATION

- dy CLEAR INSIDE DIAMETER OF SUPPORT UNDER CLOSURE
- / TOTAL DEPTH
- 100 OUTSIDE DIAMETER
- CLEAR INSIDE DIAMETER

BEARING ASSEMBLY

- 18 BEARING PLATE THICKNESS
- IF VERTICAL DIMENSION OF BEARING SHELL
- I BEARING SHELL THICKNESS

LINER AND CONFINING SHELL

- 1/ LINER PLATE THICKNESS
- 1 VERTICAL DIMENSION OF LINER PLATE
- COUTER SHELL THICKNESS



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TABLE

STATIC TEST SPECIMEN GEOME

	Model Tde	ntification						Spe	ecimen	Dimens	ion, I	nel
No.		atilication		C	losure		Con	anata	Pour de la	Clo	sure S	Sup
	Closure	Support	d	h	ts	tb	d	l	dop	d	Bea	T
1	M120-1A	LF ^b /	32.3	8	0.13	0.13	28	-		10	В	+
2	M120-1B	LF	32.3	8	0.13	0.13	28					
3	M120-2A	LF	32.3	14	0.25	0.25	28					
4	M120-2B	LF	32.3	14	0.25	0.25	28					
5	M120-20	M121-1A	32.3	14	0.25	0.25	28	18	50	~	~	L
6	M120-20	MI.21-1B	32.3	14	0.25	0.25	28	18	52	20	36	1
7	M120-20	M121-2A	32.3	14	0.25	0.25	28	18	52	28	36	
8	M120-3A	M121-2A	32.3	14	0.50	0.50	20	18	52	28	36	
9	ML20-3B	M121-3A	32.3	14	0.50	0.50	20	18	52	20	36	
10	ML20-5A	M121-2B	32.3	14	0.75	0.75	20	10	52	28	36	1
11	M120-5B	M121-3B	32.3	14	0.75	0.75	20	10	52	28	36	1
12	MI20-4A	LF	32.3	14	0.25	0.25	28	10	52	28	36	1
13	M120-5B	LF	32.3	14	0.75	0.75	28					
14	MI.20-2D	HF C	32.3	14	0.25	0.25	28					
15	M120-4B	HF	32.3	14	0.25	0.25	28					
16	M120-2E	LF	32.3	14	0.25	0.25	28					
17	M130-1-7	LF	32.3	14	0.25	0.25	28		1			
18	M130-1-6	LF	32.3	14	0.25	0.25	28					
19	M130-2-2	LF	32.3	14	0,25	0.25	28					
20	M130-2-1	LF	32.3	14	0.25	0.25	28					
21	M130-3	LF	32.3	14	0.25	0.25	28					
22	M130-4	LF	32.3	14	0.50	0.25	28					
23	M130-7	LF	32.3	14	0.50	0.50	28					
24	M130-1-3	LF	32.3	14	0.25	0.25	28					
25	ML30-5	LF	32.3	8	0.25	0.25	28					
26	ML33-1	LF	32.3	8			28					

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a/ For definition of terms, see page A2-34.

LF = low friction support as discussed in Appendix 2, Section 2. <u>b</u>/

HF = high friction support as discussed in Appendix 2, Section 2. <u>c</u>/

Expansive concrete. d/

TABLE 3-1

N GEOMETRY AND LOAD CAPACITIES a

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on, In	ches									
ire Su	pport						Stre	ngth,	Load	
Bear	ing Ass	sembly	Line	r and Co	onfining	Shell			Capacity,	Remarks
ЪB	t _B	l _r	tr	^t l	LE	t _c	f'c	fs	ALU L	
							8	36	2.6	
							8	36	2.6	
							7	36	3.4	
							7	36	3.6	
36	0.50	6	1.0	1.0		0.10	7	36	3.2	
36	0.50	6	1.0	1.0		0.10	7	36	3.0	
36	J.,50	6	1.0	1.0		0.62	8	36	4.7	
36	1.50	6	1.0	1.0		0.62	7	36	6.4	
36	1.50	6	1.0	1.0		0.10	8	36	3.8	
36	1.50	6	1.0	1.0		0.62	7	36	б.8	
36	1.50	6	1.0	1.0		0.10	8	36	4.1	
							8	36	2.9	<pre>1/4-in. shear cone in closure model</pre>
				1			7	36	8.8	
							7	36	4.2	
							8	36	2.8	<pre>1/4-in. shear cone in closure model</pre>
							13	36	5.5	
							7₫/	36	3.7	
		ĺ					3	36	2.8	
							3	70	5.5	
							11	70	6.6	
							6	150	7.3	Shims
							8	36	4.9	Shims
							8	36	6.4	Shims
							6	36	3.4	Teflon on supports
	1						7	36	3.3	Shims
							7		1.0	

TABLE 3-1 A2-35

TABLE 3-1 (CONTI

			T									
	Madal Ta							Speci	men Di	mension	, Inch	es
Test	Model Iden	tilication		Cl	SELLTO					Closu:	re Sup	port
NO.					Joure		Con	crete Fo	oundati	lon	Bea	ring
	Closure	Support.	d	h	t _s	t _b	dı	1	d _{OD}	d_ID	d_B	t
27	M133-2	LF	32.3	21			28					
28	ML30-6	LF	32.3	21	0.25	0.25	28			ì	0	
29	M130-14	LF	32.3	21	0.75	0.75	28					
30	MI30-13	LF	32.3	8	0.75	0.75	28					
31	ML30-7-1	M131-1	32.3	14	0.50	0.50	23	8	52	31	34	0.5
32	ML30-9	M129-0	32.3	13.4	0.12	0.12	23.5	6.38	54	24.2	37	0.1
33	ML30-8	M128-0	32.3	14.2	0.31	0.31	28	12.4	51.9	28.4	32 0	0.5
34	ML30-7-2	M131-2	32.3	14	0.50	0.50	30.2	8	52	38	34	0.5
35	ML27-0	M127-0	31.0	13.2	0.25	0.25	26.5	20.9	52	26.5	52	0.5
36	M130-10	MI31-4	32.3	14	0.50	0.50	28	8	52	36	34	0.7
37	M130-7-3	ML31-3	32.3	14	0.50	0.50	28	8	52	36	34	0.5
38	M130-11	M131-7	22	14.3	0.50	0.50	19	45	52	22	24	1.0
39	ML30-12	ML31-6	22	9.5	0.50	0.50	19	49.8	52	22	24	1 0
40	M130-7-6	M131-10	32.3	14	0.50	0.50	28	50	52	36	34	0.5
41	ML30-7-5	M131-9	32.3	14	0.50	0.50	28	50	52	36	34	No Pla

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3-1 (CONTINUED)

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Bea	port ring As	sembly	Line	er and Co	onfini	ng Shell	St St	ksi	Load Capacity,	Remarks
đв	tB	l _r	tr	^t l	l	tc	f'c	fg	ksi	
							7		1.1	
							7	36	4.1	
							8	36	8.3	
-1	1.000		1.1.1	1.70			8	36	6.8	
34	0.50	8	0.50	No Plate	7	0.50	7	36	6.6	Wide bearing
37	0.12	6.38	0.12	0.12		0.12	8	36	5.3	Hercules
32.9	0.50	12.38	0.50	0.12	1.1	Re-bar	8	50	4.5	Rock Test T
34	0.50	8	0.50	No Plate	7	0.50	8	36	3.9	Narrow bearing
52	0.50	20.88	0.50	0.50		0.50	8	36	5.8	Rise and motate
34	0.75	8	0.50	No Plate	7	0.50	8	50	5.3	Notched bearing
34	0.50	8	0.50	No Plate	7	0.50	9	36	5.5	Inclined bearing
24	1.0	8	1.0	0.50	7	0.50	12	180	22.0	
24	1.0	8	1.0	0.50	7	0.50	12	70	15.8	
34	0.50	8	0.50	No Plate	7	0.25	8	36	4.2	
34	No Plate	8	0.50	0.25	7	0.50	8	36	5.0	No bearing ring

TABLE 3-1 (Cont'd) A2-37

TESTS 1 AND 2

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These tests were performed to determine if models used in past closure test activities could be scaled up, and to determine the repeatability of the failure load. The models selected for Tests 1 and 2 were scaled-up versions of static test models previously tested by University of Illinois researchers. These tests essentially doubled the 1/14 scale used by the University of Illinois. Except for the bearing width which was somewhat narrower, all test variables were scaled up by a factor of two, as can be seen in Figure 3-1. The tests were performed on a low friction support.



⁽UNIVERSITY OF ILLINOIS)

This section proved to be quite well proportioned since shear, bearing, and bending distress were all evident at the time of the post-test inspection, as can be seen in Figures 3-2, 3-3, and 3-4. Note the cracks starting at the bearing in both tests. The dark areas in the photographs are caused by moisture resulting from the cutting operation and are not related to the tests. The bulge that appears above the area in contact with the bearing

(Figure 3-4) is due to the yielding of the steel from the load imposed by the horizontal straining of the concrete; i.e., the outer steel ring is acting as a confinement vessel.

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Originally, the University of Illinois models were designed for 1000 psi; however, failure occurred at 2150 psi. In the present tests, failure occurred at 2600 psi in both Test 1 and Test 2. The ductility of the model, as well as its ability to continue to carry a load even though high strain rates were being encountered, is evidenced in the load deflection diagrams (Figures 3-5 and 3-6).

The results of these tests established the fact that the failure load for the scaled-up models was essentially the same as that found by the University of Illinois for the smaller model. The ability to duplicate test results accurately was also established.





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FIGURE 3-6 A2-43

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TESTS 3 AND 4

The models in Tests 3 and 4 were designed for an ultimate load capacity of 4000 psi. These tests were performed to check the validity of the design approach, to verify the repeatibility of the test results, and to provide a point of departure for further model designs. The model geometry for Tests 3 and 4 is shown in Figure 3-7. The tests were performed on low friction supports.



CLOSURE GEOMETRY FOR TESTS 3 AND 4 FIGURE 3-7

The section behaved as might be predicted for a deep section; steep shear cracks as well as bearing distress were observed at the time of post-test inspection. The punching of the 1/4-inch plate which can be seen in Figure 3-8 was caused by the sharp edge of the low friction bearing and is not indicative of a general-type failure mode for these models. This proved to be a test anomaly and was not evident in the models with heavier bottom plates. The maximum load capacity, however, was undoubtedly achieved before the punching action took place. In the later tests steps were taken to minimize this punching action by placing mild steel shims between the hardened bearing and the model.

Figure 3-9 shows the shell which has been cut by a cutting torch (flume cut) to expose the concrete and to allow the models to be cut in two with a diamond saw. Distress in the concrete in the area of the Learing and above the bearing is also evident. Figures 3-10 and 3-11 show close-up views of the failed area in the section.

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The evident distress in the side shell supports the hypothesis that this shell acts as a confining ring and carries a sizable portion of the vertical load as well. Failure occurred at 3450 psi in Test 3 and at 3600 psi in Test 4 (see Figures 3-12 and 3-13). It should be pointed out that when the pressure deflection curves show sharp "mop-off in the vertical load capacity this does not indicate a sudden failure, but is due to actually releasing the load on the model. More ductility would have been achieved had the load been carried on for a longer period of time.

The ability to duplicate test results was established as well as the ability to provide resistance to pressures as high as 3500 psi. Although the models failed below their predicted load capacity, the results were still satisfying in view of the overall structural behavior.



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TEST 3 FIGURE 3-8



TEST 3 FIGURE 3-9 A2-46





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TESTS 5 AND 6

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Tests 5 and 6 represented the first attempt to test a closure model on a silo support. The silo support was proportioned to fail at 5000 psi, with the expectation that a closure could be failed on this silo and the increase in closure load capacity due to support friction could be determined.

The closure tested was identical to those used for Tests 3 and 4. The model support geometry for Tests 5 and 6 is shown in Figure 3-14.



SUPPORT GEOMETRY FOR TESTS 5 AND 6 FIGURE 3-14

The radial load introduced into the support due to the expansion of the base of the closure was underestimated in initial calculations. The test failure was in the support structure which yielded an applied load of approximately 3000 psi, as is shown in the deflection curves (Figures 3-15 through 3-18). Radial cracks were evident in the concrete as well as conical cracks progressing from the bearing plate (Figures 3-19 through 3-22). The failure appeared to occur after yielding in the outer steel confining shell.

The distressed area of concrete directly under the inner confinement ring (Figure 3-20) was predicted and does not lead to failure or decrease in load capacity, but is caused by the vertical bearing of the inner confinement ring on the unconfined concrete. The spalling was expected to occur, and it did so on all the silo models tested. As can also be seen in this figure, a clip is welded to the inner confinement ring to support the displacement gauge.

Figure 3-19 shows the radial cracks in the top of the support and another view of the spalled area directly below the inner confinement ring, as well . as the clip angles supporting the displacement gauge for the silo support.

It should be noted on the load displacement curve for Test 6 (Figure 3-18) that the closure did not fail but had a higher load capacity than the silo, which explains the sharp rise and return to zero on the displacement curve. The closure model remained elastic. Radial cracking was extensive in the failed support models (Figures 3-21 and 3-22). When the outer confinement ring was removed, the model fell apart.

The tests demonstrated that the assumed support failure mode was essentially correct. The contribution of the outer confining shell was vividly portrayed. However, we did not have enough outer confinement to mobilize the strength of the model and therefore did not get enough load on the closure support to load the inner confinement ring.



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SUPPORT M121-1A TEST 5 - PRESSURE VS. DEFLECTION

FIGURE 3-15











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TEST 6 FIGURE 3-21



TEST 6 FIGURE 3-22 A2-54

TEST 7

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The closure that failed the supporting structure in Tests 5 and 6 was next tested on a support designed for a ultimate load capacity of 10,000 psi. This 10,000-psi level represents the load to be imposed on the total area of the closure; i.e., the unit load on the closure model, and not bearing stress. A 10,000-psi load on the closure would result in a load of approximately 40,000 psi on the bearings. The model support geometry for Test 7 is shown in Figure 3-23.



SUPPORT GEOMETRY FOR TEST 7 FIGURE 3-23

The test was undertaken to determine the increase in closure strength from high bearing friction, which could not be achieved in the earlier tests. The support had enough inherent load capacity to allow the closure model to be broken at a load of 4660 psi without visible damage to the support structure (an apparent unit bearing stress of 18,640 psi).

The pressure deflection curve for the silo (Figure 3-24) indicates essentially total recovery of the observed deflection and supports the contention that

the silo did not fail. The ductility exhibited by the closure is evident in Figure 3-25. Examination of Figure 3-26 shows that no punching action took place between the support and the bottom plate of the closure, which further substantiates the belief that the shearing was caused by the hardened low friction bearing blocks. Π

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This test supported the contention that bearing friction could add considerable additional strength to a closure structure. In this particular situation the increased capacity was approximately 20 percent over that observed on the low friction support.




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CLOSURE M120-2C TEST 7 - PRESSURE VS. DEFLECTION

FIGURE 3-25 A2-58



The support used in Test 7 was next used to support a stronger closure model than that used in the preceding test. The closure was proportioned to fail at the 8000-psi load level. This test was intended to show the behavior of a well-matched closure and support. 1

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The closure geometry for Test 8 is indicated in Figure 3-27. The closure support structure is shown in Figure 3-24 (Test 7).





Failure was observed simultaneously in the closure and the silo at a load level of 6375 psi (an approximate bearing stress of 25,500 psi). However, examination of the load deflection curves (Figures 3-28 and 3-29) indicates that the closure could probably have taken more load, as the curve was still rising Figure 3-30 shows only relatively minor distress in the closure, although some distress in the bearing is evident. The support shows diagonal cracking from the bearing and spalling of concrete immediately under the confinement ring, as can be seen in Figure 3-31. This appears to be a characteristic failure pattern for this support configuration. The vertical crack under the bearing of the silo (Figure 3-31) is very similar to the crack in the closure (Figure 3-30), and points out the fact that the inner confinement ring is serving the same function for the silo as that provided by the outer shell for the closure. The mode of support response was the same as previously observed. There was no shear crack in the closure model after testing. Failure appeared to be local and to have been caused by yielding of the steel shell. The silo and the closure appear to be rather closely matched in total load capacity.



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SUPPORT M121-2A TEST 8 - PRESSURE VS. DEFLECTION

FIGURE 3-28 A2-62





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TEST 8 FIGURE 3-30



TEST 8 FIGURE 3-31 A2-64

A closure model with 1/2-inch plates was next tested on a silo which was similar to those used in Tests 5 and 6, but with a much thicker bearing plate (1-1/2 inch, compared with 1/2 inch for Tests 5 and 6). This test was performed to determine the increase in support strength from a thicker bearing plate and a stronger closure.

The closure tested was identical to that of Test 8 (see Figure 3-27). The support geometry is shown in Figure 3-32.



SUPPORT GEOMETRY FOR TEST 9 FIGURE 3-32

The failure occurred in the support structure, as had been predicted, at a load level of 3846 psi applied to the closure (see Figures 3-33 and 34). It should be pointed out that the slope in the load deflection curve for the closure does not represent a yielding or a reduction in load capacity, but merely models the capacity of the silc. The closure was not damaged, but the silo failed.

After the test event, the outer shell was removed from the support structure and the tendency of the structure to grow could be observed from the presence of a bend in the removed plate at its point of contact with the base of the test fixture. Note the yielded area in the lower section of the outer confinement shell of the silo (Figure 3-35). The lowest edge of the outer confinement ring was in contact with the steel base plate and could not move outward due to the friction generated between the two steel surfaces. Immediately above that a yield area can be observed - a curved section which was caused by the horizontal loading from the concrete. This is confirmed by the crack pattern in Figure 3-36. The apparent formation of an arch (Figure 3-36) is essentially the continuation of the closure structure; the closure and the silo support actually function as a single structure.

In the crack pattern shown on the right side of Figure 3-36 there is evidence of continuation of an arch forming, which throws both vertical and horizontal loading into the outer confinement ring provided on the silo. The strength of the silo influences the apparent load capacity of the model, and vice versa to some extent, emphasizing again that the silo support and the closure are really one structure, and should be analyzed as such.

Figure 3-37 shows the closure specimen used to load the silo support in Test 9; note that it was essentially undamaged. Figure 3-38 illustrates the failed silo support.

Comparison of Test 9 with Tests 5 and 6 demonstrates that increasing the bearing plate thickness apparently accounted for a 2[°]-percent increase in load capacity, although some increase is probably due to the heavier closure.

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FIGURE 3-33



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This test was designed specifically to examine the interaction phenomenon. A support equivalent to that of Tests 7 and 8 was loaded by a closure specimen with 3/4-inch plates. The increased load capacity of the support structure when loaded with a stiff closure was to be examined. The closure geometry is shown in Figure 3-39.



FIGURE 3-39

There was an anomaly in the data recovery from this test which tends to discredit the apparent results. This anomaly is seen in the silo load deflection curve (Figure 3-4C), where there is a sharp rise in load capacity with no apparent displacement. This was caused by the bottoming out of the displacement gauge; therefore the vertical portion of the curve should be ignored. In the curve for the closure (Figure 3-41), none of the unloading portion of the test sequence is shown. Evidence of silo failure, which occurred at a load level of 6800 psi, can be seen in Figures 3-42 and 3-43. The concrete in the closure does not appear to be distressed (Figure 3-44); however, significant yielding had occurred in the bottom plate. The heavy confinement ring of 3/4-inch steel has prevented a bearing failure on the closure. It appears that the strength of the combination was controlled by the silo.

Confirming the assumption that the closure probably had more load capacity is the fact that, in Test 13, a 3/4-inch closure was tested on a low friction support and attained an 8815-psi level. We are confident that this should be possible on a silo or rigid support.

Inspecting the model after the test, it was evident that the support collapsed as in previous tests and the closure distress was local due to yielding in the steel plates.





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CLOSURE M120-5A TEST 10 - PRESSURE VS. DEFLECTION FIGURE 3-41 A2-74



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TEST 10 FIGURE 3-43 A2-75



To further explore the interaction phenomena, a 3/4-inch closure was next used to test a support identical to that used with a 1/2-inch closure in Test 9. The closure and the support configuration are shown in Figures 3-39 (Test 10) and Figures 3-32 (Test 9), respectively.

The pressure deflection curves (Figures 3-45 and 3-46) indicate that the support structure failed at a load of 4084 psi; no additional strength was observed over Test 9. This shows that for these particular models interaction is no more pronounced than for a less stiff closure tested on an identical support.

The general pattern of the radial cracks in Figure 3-47 is quite similar to that observed in Test 9. Figure 3-48 shows the support at the time of post-test inspection.



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TEST 11 FIGURE 3-47



TEST 11 FIGURE 3-48 A2-79

TESTS 12 AND 15

Before the test series began, there was a great deal of concern about shear failures in closures. In these tests, an attempt was made to increase the shear capacity of the models. A steel cone was placed in two of the 1/4-inch closure models, as shown in Figure 3-49.



CLOSURE GEOMETRY FOR TESTS 12 AND 15 FIGURE 3-49

The models were tested on low friction and high friction (steel to steel) bearings, with identical results. The total load capacity of the models was decreased by the addition of the cone. The model on the low friction bearing failed at 2900 psi (Figure 3-50), while the model on the high friction bearing failed at 2792 psi (Figure 3-51).

The cone introduced a vertical load component directly into the outer shell and caused buckling of this element at a low load level (Figures 3-52 and 3-53). The concept could be used effectively only if the outer shell were stiffened considerably.

Since other tests have shown that a shear-type failure is not critical, this concept should be eliminated from practical consideration.

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The Test 13 model was designed for an ultimate load capacity of 12,000 psi. This test was made to compare with the results of Test 10. The closure geometry is shown in Figure 3-54. The test was performed on a low friction support.



CLOSURE GEOMETRY FOR TEST 13 FIGURE 3-54

The maximum test pressure reached was 8815 psi (Figure 3-55), which lends support to the assumption that the closure in Test 10 had additional load capacity available and that the load capacity of the combination was controlled by the silo.

The test photograph shows a damaged zone of concrete extending all the way from the bearing to the top surface of the model (Figure 3-56). This is misleading since after the maximum pressure was reached vertical load was unintentionally put into the side shell from the testing apparatus. The vertical point on the deflection curve (Figure 3-55) is the point at which

A2-83

the deflection gauge bottomed out. Since there was no more capacity in the gauge, it indicates zero deflection with an increase in load. Because the deflection reading was lost during the test, the test was continued to a point where the piston in the loading machine came in contact with the side-walls of the closure model. The true collapse mode is thus not properly in-dicated, although it is predicted that a local failure occurred as before due to yielding of the steel plates.

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CLOSURE M120-5B TEST 13 - PRESSURE VS. DEFLECTION FIGURE 3-55 A2-85



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A closure model identical to those used in Tests 3, 4, 5, 6, and 7 was tested on a high friction (steel-to-steel) support in order to ascertain the increased load capacity from friction and to compare this with the additional load capacity obtained when testing the model on a model support structure. The high friction support consisted of a 4-foot outside diameter by 28-inch inside diameter 2-inch-thick steel plate, which rested directly on the adjusting plates -- the steel support within the fixture. Because of the detail of the support, the stiffness of the leveling plates was actually mobilized, as was the radial stiffness of the 2-inch plate that supported the model. The closure geometry is shown in Figure 3-57.



CLOSURE GEOMETRY FOR TEST 14 FIGURE 3-57

One of the effects of the increased radial stiffness and friction of the support is to increase the shear crack angle (Figure 3-58). In this test, the shear crack angles were more nearly vertical than in Tests 3 and b. The

A2-87

increased radial stiffness of the support enhances the load capacity of the model, demonstrating again the general phenomenon of the interaction between the closure and the support.

The closure failed at a pressure of 4152 psi (Figure 3-59). This test result can be compared with the results of Test 7 where an identical closure model was tested to failure on a silo support. The closure in Test 7 failed at a load of 4660 psi. After correcting the load capacity of the closure tested in Test 14 to account for the variation in concrete strength between the closure mode's, the apparent load capacities of the closures in Test 14 and Test 7 are 4660 psi and 4450 psi, respectively. These load capacities are in good agreement and support the assumption that support interaction through friction can increase closure load capacity by as much as 20 percent.





CLOSURE M120-2D TEST 14 - PRESSURE VS. DEFLECTION FIGURE 3-59 A2-90 1

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3.2 Phase II

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After the first 15 tests, there was a pause in the test sequence in order to evaluate the information obtained in the first test series and to examine critical parameters identified in Phase I. Evaluation of these tests led to the following conclusions:

- 1. The composite section is extremely efficient and can produce a high load capacity closure.
- Bearing stresses are not as critical as originally imagined. Extremely high stresses can be tolerated merely by providing the effective confinement by the steel shell or the confinement ring on the silo.
- 3. The mode of failure apparently changes as the stiffness of the steel section is increased. As the thickness of the closure shell was increased, shear crack failure changed to a general bearing distress or steel yielding failure. With the 8 inch-deep model and thin gauge confinement, three failure modes were observed: bearing failure, shear failure and bending failure all within one model. Shear and bearing distress were predominant in the 14-inch-deep models. In the 1/2- and 3/4-inch shells, the concrete apparently did not control the strength of the model. The stiffness prevented a bearing failure in the concrete until the steel shell had reached its yield point.

The models for the Phase II tests were modified as indicated by the results of Phase I testing. The following parameters were investigated in the second phase of the testing:

- . Section depth
- . Bearing width

A2--91

- . Closure confining shell thickness
- . Closure bottom plate thickness
- . Closure bearing plate thickness
- . Bearing confinement shell thickness
- . Support outer confining shell thickness
- . Bearing haunch configuration
- . Concrete strength
- . Steel strength
- . Reduced support friction
- . Interaction phenomena

During the preparation of this report, data plots became available showing not only load versus deflection curves, but also strain data. These have been included with the Phase II test descriptions that follow. T

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Test 16 was performed to determine the increase in load capacity with the use of high strength concrete. The model geometry for this test is shown in Figure 3-60. The test was performed on a low friction support.



CLOSURE GEOMETRY FOR TEST 16 FIGURE 3-60

The load deflection curve shows the ductility evidenced by the model despite the relatively high strength of the concrete. The section failed at an applied pressure of 5526 psi (Figure 3-61). This represents a substantial increase in load capacity over that of Test 3 and 4 models which failed at the 3500-psi level.

The circumferential crack that can be seen just outside the bearing area (Figures 3-63 and 3-64) may be an indication that the high concrete strength is changing the failure mode. The bulging below this crack indicates that the concrete in this section has flowed plastically, thus loading and yielding the outer confinement steel.
In Figure 3-65, note the circumferential crack at the edge of the bearing; the gouges in the underside of the concrete were caused during the cutting operation and are not a result of the test loading. The apparent burnishing of the concrete in the bearing area is indicative of the applied stress, which is approximately 20,000 psi for this test. In Figure 3-66 burnishing is very evident on the bottom plate in the bearing area. The shell fell away from the model because of the deletion of the internal instrumentation bars. During Phase I of the test series, the bars had been welded to the outer confinement shell, and therefore held the concrete section and the steel shell together. No horizontal or circumferential bars were welded to the steel side shell in Phase II; the only instrumentation bars welded are the vertical bars that are welded to the bottom plate. The results of this test support the contention that the concrete strength enhances the load capacity in approximately the same ratio as the square root of the unconfined compressive strength of the concrete.



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CLOSURE M120-2E TEST 16 - PRESSURE VS. DEFLECTION FIGURE 3-61 A2-95

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DATA PLOT - TEST 16 FIGURE 3-62 0

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1 1 [] 1 [] [] 1 T TEST 16 FIGURE 3-63 I I 1 1 1 1 N M120-2E 1 ŵ I TEST 16 1

FIGURE 3-64 A2-97

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TEST 16 FIGURE 3-65



TEST 16 FIGURE 3-66 A2-98

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The effect of expansive concrete was examined in Test 17. The mix was designed so that all expected shrinkage would be overcome by the expansive admixture. It was anticipated that the concrete shear strength could be increased significantly with this procedure.

The closure geometry for Test 17 is shown in Figure 3-67. The test was performed on a low friction support.



FIGURE 3-67

The expected increase in load capacity was not observed. The model f-iled at a load of 3710 psi (Figure 3-68) which compares well with the 3500-psi load capacity of the models for Tests 3 and 4. From the tests performed, there was no apparent structural advantage in using expansive concrete; however, because of increased shrinkage in full-scale structures, the use of expansive concrete may be advantageous in order to maintain the concrete in contact with the steel side shell. Figure 3-70 shows the model at the time of post-test inspection.



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The effect of concrete strength was further examined by testing a model with a lower strength concrete. The closure geometry of the model is shown in Figure 3-71. The test was performed on a low friction support.



CLOSURE GEOMETRY FOR TEST 18 FIGURE 3-71

The section failed at an applied pressure of 2752 psi (Figure 3-72). The failure mode was the same as that observed in Tests 3 and 4 with the exception that greater bearing distress was observed with the lower strength concrete (Figure 3-74). This test supports the contention that the load capacity varies approximately as the square root of the unconfined compressive strength of the concrete.



CLOSURE M130-1-6 TEST 18 - PRESSURE V5. DEFLECTION FIGURE 3-72



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In Test 19 the effect of using high strength steel was examined to determine if failure was steel strength or deformation sensitive. The model geometry is shown in Figure 3-75. The model was tested on a low friction support.





The ductility of the model, as shown on the pressure deflection curve (Figure 3-76), is attributed to the high strength steel. It appears that the total steel strength is the critical factor since the model failed at a load of 5497 psi. This compares favorably with the test of the 1/2-inch model on the support structure which failed at 6375 psi (Test 8); much greater load capacity was shown than that observed in Test 18 where an identical model constructed from A-36 steel collapsed at 2570 psi. The failure was attributed to bearing distress.

No shear crack could be observed (Figure 3-78). The increase in load capacity due to the high strength concrete noted in Test 16 was not as great as the increase noted in this test as a result of using high strength steel. It appears that a significant increase in load capacity may be attained by the use of high strength structural materials. 1



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DATA PLOT - TEST 19 FIGURE 3-77 0

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In Test 20 the effect of both high strength steel and concrete was examined. The model geometry is shown in Figure 3-79; the test was performed on a low friction support.



CLOSURE GEOMETRY FOR TEST 20 FIGURE 3-79

The model collapsed at a load level of 6584 psi (Figure 3-80); shear cracks were evident in the collapsed structure. There was also an unusual horizontal crack approximately 2 inches from the top surface of the model (Figure 3-82). There seems to be no explanation for the formation of these cracks, since no bearing distress was evident at the time of post-test inspection. However, these horizontal crack patterns were also noted in Test 21 (Figure 3-86). This phenomenon is undoubtedly associated with the high strength steel shell and the high strain that is being obtained in the model.



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CLOSURE M130-2-1 JEST 20 - PRESSURE VS. DEFLECTION FIGURE 3-80 A2-113



DATA PLOT - TEST 20 FIGURE 3-81 A2-114



The effect of ultrahigh strength steel was examined in Test 21. The model was tested on a low friction support. The closure geometry for Test 21 is shown in Figure 3-83.



CLOSURE GEOMETRY FOR TEST 21 FIGURE 3-83

The model failed at 7300 psi (Figure 3-84), a load lower than would be expected from extrapolation of previous test results. Many horizontal cracks were observed at the time of post-test inspection (Figures 3-86 and 3-87). This indicates that a critical concrete strain situation was evidently reached. Therefore it is not possible to say that the total plate strength is the only critical design parameter.

It has been postulated that the bearing strength of the concrete may be expressed in terms of the unconfined concrete compressive strength and a linear function of the confinement pressure available from the side steel shell. In this particular test, observations of the strain records indicate that the steel remained elastic until failure. Thus the effective vertical stress in the side shell could be determined. The vertical load not carried in the side shell was assumed to be in the concrete. This indicates that the effective concrete bearing stress was approximately 15,000 psi, which implies that in this case the concrete bearing strength could be expressed as $\sigma_{concrete} = \frac{f'_{c}}{c} + \frac{5.4}{c}$



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CLOSURE M130-3 TEST 21 - PRESSURE VS. DEFLECTION

FIGURE 3-84 A2-117



DATA PLOT - TEST 21 FIGURE 3-85 A2-118

------30-3 TEST 21 FIGURE 3-86 TEST 21 FIGURE 3-87

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In Test 22 the relative contributions of the side shell and bottom plate were examined by testing a model with a thick shell and a thinner bottom plate. The model geometry is shown in Figure 3-88; the model was tested on a low friction support.



CLOSURE GEOMETRY FOR TEST 22 FIGURE 3-88

The model collapsed at 4871 psi (Figure 3-89) due to a local-type failure initiated by yielding of the steel plates. No shear crack was apparent at the time of post-test inspection (Figure 3-91). There was a 30-percent increase in load capacity over that for models with a 1/4-inch side shell and bottom plate (Tests 3 and 4), and the failure mode was similar to that observed in the 1/2-inch model tested on a support structure (Test 8).

A thicker outer shell appears advantageous from the standpoint of both failure mode and ultimate load capacity.



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CLOSURE M130-4 TEST 22 - PRESSURE VS. DEFLECTION FIGURE 3-89



DATA PLOT - TEST 22 FIGURE 3-90 A2-122



Test 23 represented the first test of a 1/2-inch model on a low friction support. The results obtained were correlated with those of Tests 8, 19, 20 and 22. The model geometry for Test 23 is shown in Figure 3-92.



CLOSURE GEOMETRY FOR TEST 23 FIGURE 3-92

The model collapsed at a load level of 6361 psi (Figure 3-93), which is close to the 6375-psi collapse load observed on the support structure for Test 8.

Examination of Tests 19 and 20 shows that total steel strength appears to be the critical design parameter since an equivalent strength steel model with 3-ksi concrete failed at 5497 psi and one with 12-ksi concrete failed at 6584 psi.

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The mode of failure was as previously observed, with shear cracks in evidence as well as severe bearing distress and the associated yielding of the steel plates (Figure 3-95). It should be noted that no shear cracks were observed with the 1/2-inch model in Test 8. In the first test series, the 1/2-inch model was tested on a silo, whereas in Test 23 the model was tested on a lowfriction support, which suggests that the support condition has influenced the failure mode.

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CLOSURE M130-7 TEST 23 - PRESSURE VS. DEFLECTION FIGURE 3-93 A2-126 Į

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A model similar to those used in Tests 3 and 4 was tested to investigate the effect of friction in the so-called low friction support. The test was performed on teflon-coated blocks with a tested coefficient of friction of 0.005, as opposed to 0.20 on the standard low friction bearing (Tests 3 and 4). The model geometry is shown in Figure 3-96.





The model failed at 3409 psi (Figure 3-97), which was only a nominal decrease from the previous 3500-psi load level. The failure mode was basically the same as that previously observed (Figure 3-99). This decrease in load capacity does not appear significant in light of other uncertainties involved in the testing.


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DATA PLOT - TEST 24 FIGURE 3-98





In Test 25 a much thinner section than had been used in earlier tests was examined to determine the effect of the closure section depth. It was anticipated that the mode of response at this larger span-to-depth ratio would be of interest; flexural response might influence the load capacity to some degree. Since the response mode observed in Tests 1 and 2 differed significantly from that of Tests 3 and 4 in terms of flexural cracks, this test was intended to further explore the flexural response phenomenon.

The model geometry is shown in Figure 3-100. The model was tested on a low friction support.



CLOSURE GEOMETRY FOR TEST 25 FIGURE 3-100

The model failed at a load of 3307 psi (Figure 3-101), which compares very favorably with the results of Tests 1, 2, 3 and 4. In Tests 1 and 2, identical models 8 inches deep with 1/8-inch plates failed at the 2600 psi level; in Tests 3 and 4, 14-inch-deep models with 1/4-inch plates failed at the 3500-psi level.

Bearing distress was evident in the model. Note the shear crack and the beginning of tensile cracks on the lower concrete surface (Figure 3-103). It is of interest to note that the load capacity of this model was only slightly less than that of the 1^{1} -inch-deep models in Tests 3 and $\frac{1}{4}$. This tends to support the contention that the total load capacity is rather insensitive to section depth but depends rather on the steel plate thicknesses. 1.

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CLOSURE M130-5 TEST 25 - PRESSURE VS. DEFLECTION FIGURE 3-101 A2-134



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DATA PLOT - TEST 25 FIGURE 3-102





TEST 25 FIGURE 3-103 A2-136 0

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Test 26 was a test of a plain concrete model with no plate reinforcing. The model tested had a clear span of 28 inches, an outside diameter of 32-1/4 inches, and a section depth of 8 inches. The model was tested on a low friction support. The concrete had an unconfined compressive strength of 7070 psi.

The model failed at the 1000-psi load level (Figure 3-104). Failure appeared to be basically due to excessive shear load. A crack formed from the inside of the bearing and ran to the outside of the closure model, causing sizable pieces of concrete to spall off the model.

The ring at the bottom of the specimen (Figure 3-106) served as an enclosure for the segmented blocks (which are just visible above the ring). The large circular steel ring at the top of the specimen is the T-ring; immediately below this is the steel which was inserted in the model to provide a seal surface for the T-ring gaskets. Tensile cracks were apparent on the bottom surface of the closure (Figure 3-107), but it is postulated that this did not influence the total load capacity of the closure.



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TEST 26 FIGURE 3-106



TEST 26 FIGURE 3-107 A2-140

Test 27 was another test of a plain concrete model with no plate reinforcing. This test was performed to determine the effect of a deeper section on the total load capacity. The model tested had a clear span of 28 inches, an outside diameter of 32-1/4 inches, and a section dept' of 21 inches. The model was tested on a low friction support. The concrete had an unconfined compressive strength of 6190 psi.

The model failed at 1149-psi applied load (Figure 3-108); failure again appeared to be due to excess local shear load. A crack formed from the inside of the bearing and ran vertically for some distance, causing sizable pieces of concrete to spall off the model (Figures 3-11?) and 3-111). Only one tensile crack was apparent on the bottom surface of the closure, but again it is postulated that this response did not influence the total load capacity of the closure.



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TEST 27 FIGURE 3-110



TEST 27 FIGURE 3-111 A2-144

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In Test 28 the effect of the closure section depth was further examined by testing a much thicker section than tested earlier. The mode of response was again considered to be of critical importance; flexural response might influence the load capacity. The closure geometry is shown in Figure 3-112. The model was tested on a low friction support.



CLOSURE GEOMETRY FOR TEST 28 FIGURE 3-112

Failure was observed at 4076 psi (Figure 3-113). This was only a 12-percent increase in load capacity over the 3500 psi for Tests 3 and 4, which tends to support the assumption that load capacity is not greatly influenced by section depth.

Note the absence of cracks in the model (Figure 3-115). No distress was apparent, except in the bearing area. Results of Tests 3, 4, 25 and 28 indicate that failure is initiated by yielding of the steel plates, but that collapse is a result of the interaction of failure modes that do depend on section depth - flexure, bearing and shear. [

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CLOSURE M130-6 TEST 28 - PRESSURE VS. DEFLECTION FIGURE 3-113 A2-146



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Test 29 was designed to examine the effect of span-to-depth ratio on total load capacity. The results were intended to correlate with Tests 13 and 30 where similar structures were tested. Figure 3-116 shows the closure geometry for Test 29. The test was performed on a low friction support.



CLOSURE GEOMETRY FOR TEST 29 FIGURE 3-116

The maximum test pressure reached was 8270 psi (Figure 3-117); this compares well with the pressures observed in Tests 13 and 30. The effect of a decrease in the span-to-depth ratio is not very significant for models with the amount of confining steel used in this test. The failure mode is the same as that observed in the other similar tests, suggesting that the failure mode depends on the steel plate thickness. Here again, a local failure due to yielding of the steel plates was noted. In Figure 3-119, minor distress is evident in the concrete over the bearing where the plate yielded.



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CLOSURE M130-14 TEST 29 - PRESSURE VS. DEFLECTION FIGURE 3-117 A2-150



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Test 30 was performed to examine the effect of span-to-depth ratio on total load capacity. The results were intended to show the correlation between this test situation and Test 13 where a similar structure with a smaller span-to-depth ratio was used. Figure 3-120 shows the closure geometry for Test 30.



The maximum test pressure reached was 6847 psi (Figure 3-121), which does not compare well with the test pressure of 8800 psi achieved in Test 13. In light of the other test results, the observed load capacity appears too low. The failure mode is apparently somewhat different from that observed in thicker sections with the same steel plates. After the model has been observed, there may be some doubt that the specimen actually collapsed.

A2-153



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CLOSURE M130-13 TEST 30 - PRESSURE VS. DEFLECTION FIGURE 3-121 A2-154

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DATA PLOT - TEST 30 FIGURE 3-122



In Test 31 the effect of a wide bearing on total load capacity was examined. A support structure with a haunch under the bearing was also tested for the first time. It was anticipated that these parameters would influence the design approach. The model geometry for Test 31 is shown in Figure 3-123. Ľ



MODEL GEOMETRY FOR TEST 31 FIGURE 3-123

The collapse load for the system was 6568 psi (Figure 3-124). This load capacity may be compared with that observed in Test 8 (6375 psi) where a similar test was performed on a narrower bearing support without a haunch.

A2-156

No distress was observed in the closure structure at the time of post-test inspection. The support structure failed because of a vertical shear crack propagating from the bearing to the bottom of the haunch inside the silo (Figure 3-126). Radial cracks can be seen in Figure 3-127. In Figure 3-128. which shows the bottom view of the silo looking toward the bearing area, note the circumferential crack at the base of the haunched area. There was no evident bearing distress in the support structure.

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The undamaged appearance of the concrete after it was sawed (Figure 3-129) was due to the increase in bearing area provided and the fact that the shell was 1/2-inch thick (which is not shown in the photograph). No distress was observed in the closure structure at the time of post-test inspection.



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CLOSURE M130-7-1 AND SUPPORT M131-1 TEST 31 - PRESSURE VS. DEFLECTION FIGURE 3-124 A2-158

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DATA PLOT - TEST 31 FIGURE 3-125 A2-159



TEST 31 FIGURE 3-126



TEST 31 FIGURE 3-127 A2-160



In Test 32 a scaled model of one of the Project Hercules test closures was loaded statically. This represented an attempt to correlate static and dynamic test data. The model geometry for Test 32 is shown in Figure 3-130.



MODEL GEOMETRY FOR TEST 32 (HERCULES) FIGURE 3-130

The closure model failed statically at 5330 psi (Figure 3-131) with no evidence of damage in the bearing structure. Pictures of the post-test closure are not available, but it is postulated that the failure was of the shear type and the high load capacity observed was due to the extremely large width of the bearing. Figure 3-133 shows the support at post-test inspection.

It has been hypothesized that the bearing strength of concrete may be expressed in terms of the unconfined concrete compressive strength and a linear function

A2-162

 Imatly 11,000 psi, which implies that the confined bearing strength in this particular instance can be expressed as $\sigma_{concrete} = f_c^* + 9.1 \sigma_{conf.}$

 Imatly 11,000 psi, which implies that the expressed as $\sigma_{concrete} = f_c^* + 9.1 \sigma_{conf.}$

 Imatly 11,000 psi, which implies that significant reserve dynamic load capacity against the Hercules environment was present in the model from which the static specimen was scaled.

 Imatly 11,000 psi, which implies that the model from which the static specimen was scaled.

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of the confinement pressure available from the side shell of the steel. In

this test the effective concrete bearing stress can be computed to be approxi-



CLOSURE M130-9 AND SUPPORT M129-0 TEST 32 - PRESSURE VS. DEFLECTION FIGURE 3-131 A2-164

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DATA PLOT - TEST 32 FIGURE 3-132

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In Test 33 a scaled model of the Rock Test I full size closure and support structure was loaded statically. This represented another attempt to correlate the static and dynamic test data. The model geometry is shown in Figure 3-134.



MODEL GEOMETRY FOR TEST 33 (ROCK TEST I) FIGURE 3-134

The model failed at a static load of 4500 psi (Figure 3-135). The observed failure was in the support structure (see Figures 3-137 through 3-140). The dramatic failure of the silo support element is seen in Figure 3-137. The band of steel at the top of the silo model (on which the letter S is printed) is a 1/2-inch steel bar which provides a gasket seal area and serves no structural purpose. The closure model was recessed into the silo support which probably accounts for the lack of radial cracking in the top support section (Figure 3-140); also, the silo support did not have enough ductility to cause the concrete to crack. As soon as it strained a small amount, the concrete spalled off the outer reinforcing bars and the test was terminated. While photographs of the closure model are not available, it would appear from the load/deflection curve that the closure was essentially undamaged.









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TEST 33 FIGURE 3-137



TEST 33 FIGURE 3-138 A2-170

I I I I I I I Ι I TEST 33 FIGURE 3-139 I 1 1 1 1 1 1 1 2 TEST 33 FIGURE 3-140 A2-171

In Test 34 a support with a narrow bearing was tested with a closure model identical to that used in Test 31. The test was intended to show the effect of bearing width on the total load capacity of the closure support structure. The model geometry for Test 34 is shown in Figure 3-141. II

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The collapse load for the system was 3860 psi (Figure 3-142). The failure occurred in the support structure due to excessive bearing load. Note the radial cracking and the indentation in the steel bearing ring caused by the closure bearing on the support (Figure 3-144). There was no apparent distress in the closure structure after the test.

A2-172

Assuming that all the load is carried into the support structure by the confined concrete, t e apparent bearing stress is computed to be approximately 30,000 psi. This implies that the confined concrete strength may be expressed as $\sigma_{\text{concrete}} = f_{c}^{\prime} + 19.6 \sigma_{\text{conf.}}$

In the other support structures in the second phase of the testing, failure was observed due to pure vertical shear at the haunch. This type of failure was observed in Test 31, thus no direct correlation can be made with this test.









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In Test 35 a scaled model of a typical rise-and-rotate type (nonsymmetric) closure was tested. This test was performed in order to ascertain the increased load capacity of a nonsymmetrical structure. The model tested had a minimum outside diameter of 31 inches. The support structure had an outside diameter of 52 inches, which provided for a load as high as 6000 psi on top of the entire support and closure structure. Figure 3-145 shows the model geometry for Test 35.

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MODEL GEOMETRY FOR TEST 35 (RISE AND ROTATE) FIGURE 3-145

The model failed at a static load of 5800 psi (Figure 3-146), illustrating the additional load capacity of a model with a large bearing area and large nonsymmetical appurtenance. In Tests 3 and 4, similar symmetrical models with 1/4-inch plates were tested to ultimate load capacities on the order of 3500 psi.

A2-176

The model was not sawed in half after the test was completed, and therefore the exact mode of failure is not known. However, in light of the test results, further testing of nonsymmetric closures appears feasible.

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In Test 36 a support with a notched bearing was tested with a closure model identical to that used in several previous tests. The support was identical to that for Test 37 except that a notch, fitting the closure structure, was machined into the bearing ring. The theory was that the locking of the closure and support structure would enhance the load carrying capability of both elements. The model geometry for Test 36 in shown in Figure 3-148.



MODEL GEOMETRY FOR TEST 36 FIGURE 3-148

A2-179

As shown in the pressure deflection curves (Figure 3-149), the collapse load for the system was 5293 psi, which was lower than the 6375 psi observed in Test 8. The collapse load and failure mode were nearly identical to that observed in Test 37 with the inclined bearing. The ring can be seen in Figure 3-150. In Figure 3-151, note the essentially uniform circumferential crack at the beginning of the haunched section.

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Results of this test and Test 37 demonstrate that the locking effect has not been conclusively determined due to the premature failure of the support. The closure model in both tests survived essentially undamaged.

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CLOSURE MI30-10 AND SUPPORT M131-4 TEST 36 - PRESSURE VS. DEFLECTION FIGURE 3-149 A2-181



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TEST 36 FIGURE 3-150



TEST 36 FIGURE 3-151 A2-182

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In Test 37 a support with an inclined bearing was tested with a closure model identical to that used in several previous tests. The test was intended to examine closure/support interaction phenomens. It was proposed that, if the stress concentration caused by the abrupt discontinuity at the inside of the bearing could be eliminated, the total load capacity of the closure structure would be enhanced.

It was anticipated that, by sloping the bearing to allow the model to rotate without concentrated bearing, the stress concentration that occurs at the edge of the bearing would be relieved. It was also expected that this type of support would provide for considerable locking between the closure and the support. The model geometry for Test 37 is shown in Figure 3-152.



A2-183

The ultimate load capacity for this system was 5300 psi, as indicated on the deflection curves (Figure 3-153). This was lower than the 6375 psi observed in Test 8 for a similar structure without the inclined bearing. Therefore, it is doubtful that an inclined bearing would offer any particular advantages.

In Figure 3-154 the machined area defined on the model geometry sketch (Figure 3-152) can be seen. This is not an indentation caused by the closure bearing but the actual machining that sloped the bearing ring. Figure 3-155 is a view looking upward toward the beginning of the haunch. Note the uniform crack where the haunch section was failed.

On the basis of the available shear area and the total load applied, the pure shear strength is approximately 0.21 f', which compares very well with the value of 0.20 f' established in the literature.



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CLOSURE M130-7-3 AND SUPPORT M131-3 TEST 37 - PRESSURE VS. DEFLECTION FIGURE 3-153 A2-185



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TEST 37 FIGURE 3-154



TEST 37 FIGURE 3-155 A2-186

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In Test 38 a high strength closure/closure support combination was tested. The models were designed to provide an indication of the feasibility limit for this type of structure. The model geometry for Test 38 is shown in Figure 3-156.





The test specimen attained a load capacity of 23,000 psi (Figure 3-157) and failed in an interesting manner. A crack initiated at the weld between the bottom plate and the side shell and propagated up the side shell, releasing stored energy rapidly. Figure 3-158 shows the torn section in the side of the shell and the failed weld area. This damage did not occur during the test, but took place during the unloading cycle. The extent of the failed weld area is evident in Figure 3-159, and characteristics indicate a brittle failure. It should be pointed out that during the entire testing sequence this type of failure was apparent in only two models (Test 38 and Test 39).

The burnished area in the concrete that can be seen in Figure 3-160 is indicative of the high bearing stresses. The ratio of the load carried vertically by the 1/2-inch silo liner is insignificant compared with that carried by the concrete. No particular damage was evident in the support structure at the time of post-test inspection (Figures 3-161 and 3-102).

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FIGURE 3-157 A2-189



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TEST 38 FIGURE 3-158



TEST 38 FIGURE 3-159 A2-190



TEST 38 FIGURE 3-161 A2-191

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TEST 38 FIGURE 3-162 A2-192 8

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In Test 39 the models were designed to illustrate the feasibility of relatively high load capacity closure structures. The model geometry for this test is shown in Figure 3-163.



MODEL GEOMETRY FOR TEST 39 FIGURE 3-163

The specimen failed at 15,789 psi (Figure 3-164) due to a weld failure between the closure bottom plate and the side shell which can be attributed to the lack of ductility evidenced in the steel. Weld failure was experienced on only one other occasion (Test 38).

The cracked weld that led to the failure can be seen in Figure 3-165; in Figure 3-166, the cracking of the concrete inside the steel shall is evident. At the start of the tests, the concrete was level with the steel sides of the

A2-193

surface; at the conclusion of the test, this amount of unrecovered deformation remained. The concrete was permanently compressed and was never recovered. The crack that is visible near the edge of the concrete in the closure model occurred above the failed surface.

It has been postulated that the bearing strength of the concrete may be expressed in terms of the unconfined concrete compressive strength and a linear function of the confinement pressure available from the side steel shell. In this particular test, calculations indicate that the effective concrete bearing stress was approximately 42,000 psi, which implies that the confined bearing strength in this case can be expressed as $\sigma_{\rm conc.} = f_{\rm c}^{i} + 9.2 \sigma_{\rm conf.}$

Figure 3-166 shows the indentation in the concrete surface of the silo support which is caused by high bearing stresses. Although some distress was noted in the concrete of the silo, the support structure was adecuately performing its function. The test illustrated the feasibility of designing a closure structure to withstand extremely high pressures.


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TEST 39 FIGURE 3-166



TEST 39 FIGURE 3-167 A2-197

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In Test 40 a support structure with a 1/4-inch-rhick outer confinement shell was tested with a closure model identical to that used in several previous tests. A support structure that geometrically resembled those of Tests 31, 34, 36, and 37 was selected. It was anticipated that with the reduced thickness of the confinement steel the load capacity would be lower, which would be consistent with the results observed in previous support testing. The model geometry for Test 40 is shown in Figure 3-168.



FIGURE 3-168

As shown in Figure 3-169, the system failed at the 4234-psi load level, which correlates well with all previous support data.

Post-test inspection revealed radial cracks extending down into the silo area, as well as a shear crack at the beginning of the haunch (Figure 3-170). The

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vertical cracks along the sidewalls (Figure 3-171) attest to the reduced stiffness of the outer confinement ring, which allowed the section to grow, thus causing the cracks to form.

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CLOSURE M130-7-6 AND SUPPORT M131-10 TEST 40 - PRESSURE VS. DEFLECTION

FIGURE 3-169 A2-199



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TEST 40 FIGURE 3-170



TEST 40 FIGURE 3-171 A2-200

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In Test 41 a support structure without a bearing plate was tested with a closure model identical to that used in several previous tests. The support structure was selected so that it would resemble those of Tests 31, 34, 36 and 37. The model geometry for Test 41 is shown on Figure 3-172. It was anticipated that the load capacity of this structure would be relatively the same as in the other structures since the confined concrete below the closure bearing had previously exhibited high compressive strength characteristics.



MODEL GEOMETRY FOR TEST 41 FIGURE 3-172

As shown in Figure 3-173, the collapse load for the system was 5000 psi, which compares well with the 5300- and 5500-psi load capacity observed in Tests 36 and 37, respectively. The load was lower than the 6375 psi observed in Test 8
in a similar structure with a bearing plate. This is apparently due to difference in the mode of response. In the Phase II support tests, vertical shear crack was observed at the bearing haunch which was not evident in Phase I.

It is significant that the haunched structures with similar geometric configurations all failed at approximately the same load level. Therefore it may be concluded that the capacity of the silo support is governed by the shear on the haunch section; the bearing plate below the closure structure does not contribute to any extent to the load capacity of the support structure. Pressure seals, or other system requirements, would necessitate a bearing plate in an operational support structure.

On the basis of the available shear area and the total load applied, the pure shear strength of the haunch is approximately 0.24 f_c^i , which compares well with the value of 0.2 f_c^i that is common in the literature.

In Figure 3-174, a local failure is evidenced by the displacement of the concrete in the area directly under the closure bearing.



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CLOSURE M130-7-5 AND SUPPORT M131-9 TEST 41 - PRESSURE VS. DEFLECTION FIGURE 3-173



4.0 SUMMARY AND CONCLUSIONS

4.1 Effects of Various Structural Parameters

The subscale static tests performed as part of the Closure Analysis and Test (CAT) Study covered a broad range of test parameters and variables of both closure and silo support models. In the following paragraphs, the parameters investigated and the effects of each are summarized.

4.1.1 Steel Plate Thickness

For steel of any particular yield strength, the plate thickness provided in a composite closure significantly affects the load capacity and also influences the failure mode. With thin plates, a sudden diagonal tension shear failure was observed; when the plate thickness was increased, the failure mode changed to a gradual or ductile local bearing failure which was initiated by yielding of the side steel shell.

Available strain records indicate that the most severely stressed portion of the closure is the side shell near the bearing. Large vertical strain, presumably introduced by friction load transfer from the concrete was observed, as well as large hoop strain, presumably introduced by the radial expansion of the concrete in the bearing area.

The effect of the variation in plate thickness can be seen in Tests 1 through 4, 7, 8, 13, 14, 17, 22, and 24 through 30. Figure 4-1, which is a plot of load capacity versus plate thickness/span ratio, shows that a change



A2-206

in plate thickness is of far greater significance than a change in any of the other variables tested under this study program.

4.1.2 Steel Strength

Closure models with steel plates of various yield strengths were tested to ascertain if failure could be attributed to steel strength or steel deformation. Tests 3, 4, and 16 through 21 were used to examine this test variable. It appears that, except in the case of the 150-ksi steel, the ultimate strength of the steel plate is far more significant than the thickness as such. This influence is also noted in the observed failure modes which again are a function of total plate strength rather than steel stiffness.

The influence of the steel strength may be observed in Figure 4-2, which is a plot of load capacity versus steel strength.

4.1.3 Concrete Strength

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The effect of concrete strength was examined in Tests 3, 4, 16 through 20, and 24. It was hyp thesized that the influence of the concrete strength would be linear since both the pure shear strength and the bearing strength of concrete vary linearly with the unconfined compressive strength. The influence of this parameter can be seen in Figure 4-3 where the load capacity versus concrete strength is plotted.







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4.1.4 Span-to-Depth Ratio

The effect of the span-to-depth ratio was examined in Tests 3, 4, 13, 17, and 24 through 30. The failure mode was essentially the same for the various ratios tested; however, in the thinner sections bending distress was evident in addition to the shear and bearing yielding more common in the thicker models. The influence of the concrete strength may be examined in Figure 4-4 where the load capacity versus span-to-depth ratio is plotted.

From these test results, it can be concluded that for practical design purposes the design thickness of the closure will be governed by radiation requirements.

4.1.5 Bearing Width

The effect of the bearing width on total load capacity of closure structures was not fully examined in this test series since the program objectives somewhat limited the study to established bearing widths. Tests 32 and 35, in which structures with wide bearings were tested, could not be correlated directly with other testing. Is is postulated that increased bearing width would yield somewhat higher load capacities for a given closure structural geometry. However, as increased bearing width poses system disadvantages because of the increased closure mass and area, and high bearing stresses can be safely resisted by proper design, no justification is seen for further examination of this variable.



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4.1.6 Interaction Phenomenon

Before initiation of the static testing program, the influence of support friction on total closure/load capacity had been widely discussed, but there was actually very little known about this phenomenon. It was postulated that high friction would cause the concrete in the closure to arch across to the bearing in the form of a dome, and would thus greatly enhance the load capacity of the model.

Most of the tests were performed on a low friction support. In some cases, duplicate tests were performed on high friction bearings or on support structures where steel-to-steel bearing was available. Tests 7, 8, and 14 are of interest as they may be correlated directly with low friction tests. The load capacities obtained in Tests 7, 8 and 14 are in good agreement and indicate that support interaction through high friction can increase closure capacity by as much as 20 percent.

For practical design purposes, it is recommended that this phenomenon be ignored, since present test results indicate that it is probably of secondary importance. If future testing is performed, this parameter could then be explored in more detail.

4.1.7 Support Structure

In the first phase of testing, support structures were constructed with different bearing rings and outer shells, but the inner confinement ring thickness was held constant. In all supports collapsed during Phase I, the

A2-212

failure mode was similar. Radial cracks were evident in the concrete as well as the formation of an arch to the outer steel shell. Tests 5, 6, and 8 through 11, which explored the support phenomenon, indicate that the strength of the outer confinement ring significantly influences the load capacity of the support structure. This does not imply that the inner confinement wing is not also of primary importance; however, as the vertical load is distributed at an angle of 45 degrees from the edge of the closure, the vertical stress intensity is rapidly reduced.

Figure 4-5 shows the load capacity versus outer shell thickness/outer shell diameter ratio, and Figure 4-6 shows the load capacity versus the bearing plate thickness/closure span ratio.

In the second phase of testing, support structures were constructed with haunches under the bearing area. The failure mode was essentially the same for all of these models. A pure shear crack formed at the top of the bearing surface and progressed to the haunch interface with the main structure. The value of shear along this plane at collapse compared well with the 0.2 f_c^i common in the literature. This phenomenon is indicated in Tests 31, 36, 37, 40 and 41. In Test 34, where a narrow ocaring was tested, a local failure was observed (as was expected), but could not be correlated with other test data.

It is recommended that the failure mode of closure support structures be thoroughly investigated in future testing efforts.







4.2 Best Representation of Closure Data

Representative closure test data were further analyzed to verify the assumed response modes. To obtain the best representation of the test data, load capacity was plotted versus a nondimensional closure geometry and steel strength parameter (t/D) f_s,

where

- t = plate thickness
- D = clear span
- f = steel yield point strength

This plot is shown in Figure 4-7; here the effects of many of the parameters investigated may be observed. Examination and comparision of these parametric variations suggested that a regression analysis of the test data would be useful. This analysis is discussed in the following paragraph.

4.3 Regression Analysis of Test Data

Since the static tests explored many parameter variations in detail, a statistical model of the variation in collapse load Q with a given set of test parameters was developed to obtain an analytic expression for Q as a function of d_1 , h, t_s , t_b , f'_c , and f_s for the closure structure.

From dimensional studies it became apparent that a number of parameters could be grouped as single variables; then the linear regression analysis



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based on these grouped variables could be used over most of the range of collapse load values. The functional model chosen was

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$$Q = B_0 + B_1 \left(\frac{t_s}{d_1} f_s\right) + B_2 \left(\frac{t_b}{d_1} f_s\right) + B_3 \left(f_c'\right)$$

 B_0, B_1, B_2, B_3 = the linear regression coefficients

Using 23 sets of sampling data available, the regression coefficients B_0 , B_1 , B_2 , and B_3 were determined using a multiple-regression and correlation analysis. Their values were:

Bo	=	0.40
Bl	-	4.9
^B 2		2.3
B 3	=	0.23

The above constants are to be used in conjunction with stresses in kips per square inch. The following limits must be considered in light of the experimental data available:

$$1.35 \leq \frac{h}{d} \leq 3.5$$

$$0.00 \leq \frac{t_b}{d} = \frac{t_s}{d} \leq 0.03$$

$$36 \text{ ksi} \leq f_s \leq 70 \text{ ksi}$$

$$3 \text{ ksi} \leq f_c' \leq 12 \text{ ksi}$$

The multiple-regression coefficient for this model was 0.9, which shows good correlation indicating that the function fits the data well.

4.4 Conclusions

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- . A closure/closure support structure may be designed to resist static loads significantly above the 1-kilobar pressure level. This was demonstrated in Tests 38 and 39 where closure/closure supports were tested to 23,000 psi and 16,000 psi, respectively.
- The most critical closure design parameter is the strength capacity of the closure plates.
- . The total strength capacity is more important than the stiffness of the plates up to the point where the critical strain is reached in the concrete.
- The side shell is of primary design importance since it not only acts a confinement vessel for the concrete in the bearing area but also carries much of the vertical load into the bearing structure. This vertical load component is apparently introduced into the steel shell by friction between the concrete and steel.
- . For a given steel bottom plate and side shell configuration, the load capacity of composite closures will vary as a function of the concrete strength. However, as our ability to define the actual material properties of concrete is still rather primitive, it is recommended that, for full-scale designs, concrete strength requirements be limited to 6000 psi.

- The span-to-depth ratio had minor effect on the total load capacity in the models tested. In the thinner section bending distress was evident, but this did not significantly influence the total load capacity.
- Support/closure interaction phenomenon such as increased friction between the model and support does not appear to increase the load capacity more than 20 percent in any situation. Therefore, it is recommended that for practical design purposes these effects be ignored.
- A closure support structure may be designed to successfully resist static loads significantly above the 1-kilobar pressure level, as was shown in Tests 38 and 39.
- Failure in the structures with haunches below the bearing was due to a pure shear failure in the haunch. In all cases where this failure was observed, the shear stress along the failure plane was computed to be near the 0.2 f_c^{\dagger} referenced in the literature.
- . Failure in the support structures without haunches was due to radial cracks and the arch formations carrying load to the outside shell.
- . In the structures tested, the load capacity was most dependent on the outer shell strength. The bearing plate thickness has only minor influence, and the inner ring does not appear critical if sufficient confinement is provided for the concrete.

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 A regression analysis of the test data for the closure structures yielded a simple functional representation of the failure load level in terms of closure geometric and strength parameters.
 The developed function had the form of

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$$Q = 0.40 + 4.9 \left(\frac{t_s}{d_1} f_s\right) + 2.3 \left(\frac{t_b}{d_1} f_s\right) + 0.23 f_c'$$

where

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¥.		load capacity
ts	=	side shell thickness
ъ		bottom plate thickness
d ₁	=	closure clear span
fs	=	steel yield point
f'c	=	unconfined concrete compressive strength

The above constants are to be used in conjunction with stresses in kips per square inch. The following limits must be considered in light of the experimental data available:

$$1.35 \leq \frac{h}{d} \leq 3.5$$

where

h = section depth $0.00 \leq \frac{t_b}{d} & \frac{t_s}{d} \leq 0.03$ $36 \text{ ksi} \leq f_s \leq 70 \text{ ksi}$ $3 \text{ ksi} \leq f_c' \leq 12 \text{ ksi}$

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Security Classification

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Structural Models, Silo Closure						
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The Ralph M. Parsons Company

Engineers · Constructors

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December 9, 1969

Department of the Air Force Space and Missile Systems Organization (AFSC) Norton Air Force Base, California Vol. 1= AD855784 * Vol. 2 = AD855785 * Vol. 4 = AD855786*

ATTENTION of Lt. H. S. Yoshioks, SMQHF

SUBJECT Job No. 4171-1 - Closure Analysis & Test Study Errata to Final Report Volumes I, II and IV

REFERENCE Contract No. F04694-67-C-0105

Gentlemen:

HD-855 786

It is requested that the changes listed below be made in your copies of Volumes I, II and IV of Closure Analysis and Test Study, dated 31 July 1969.

On Page 4-39 of Volume I, "Project Summary and Abstract of Findings," Page 6-53 of Volume II, "Technical Report," and Page A2-218 of Volume IV, "Subscale Static Test Report," the value for the regression coefficient B_0 is in error and should be corrected to read as $B_0 = -0.40$, as shown below:

Incorrect	Correct				
B ₀ = 0.40	$B_0 = -0.40$				
B1 = 4.9	$B_1 = 4.9$				
B ₂ = 2.3	$B_2 = 2.3$				
B3 = 0.23	Ba = 0.23				

In addition, on Page A2-221 of Volume IV, "Subscale Static Test Report," the function "Q = 0.40 + 4.9 $\left(\frac{t_s}{d_1}f_s\right)^+ 2.3 \left(\frac{t_b}{d_1}f_s\right)^+ 0.23 f_c$ " should be corrected to read as follows:

"Q = -0.40 + 4.9 $\left(\frac{t_0}{d_1} f_0\right)$ + 2.3 $\left(\frac{t_b}{d_1} f_0\right)$ + 0.23 f_0 ".

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