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MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, CORROSION CHARACTERISTICS AND FATIGUE-CRACK PROPAGATION RATES OF STRESS-RELIEVED ALUMINUM ALLOY HAND FORGINGS

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Contract No. F33615-68-C-1385
Project No. 7381

Third Technical Management Report
August 15, 1968 - November 15, 1968

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ABSTRACT

Except for a few tests, the tensile, compressive, shear and bearing properties have been determined for all but three of a total of forty 2014-T652, 2024-T852, 7075-T7352 and 7079-T652 hand forgings scheduled for test. Ratios among these properties have been calculated. Plane-strain fracture-toughness values have been determined for two of the hand forgings. The results of axial-stress fatigue tests ($R=0.0$) of smooth specimens are presented.

The preparation of specimens for the stress-corrosion and exfoliation tests has been initiated, and exposure of the specimens will be started soon.

Tests of the 6x24-in. 2014-T652 hand forging are in progress to study the effects of notch geometry, specimen length and stress on the rate of fatigue crack propagation.

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Third Technical Management Report

MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, CORROSION CHARACTERISTICS AND FATIGUE-CRACK-PROPAGATION RATES OF STRESS-RELIEVED ALUMINUM ALLOY HAND FORGINGS

I. Introduction.

The design mechanical properties, fracture toughness, corrosion characteristics and fatigue-crack propagation rates are four of the most important factors involved in the selection and efficient design of aircraft structures. Such data are needed for aluminum alloy hand forgings for several reasons: (1) much of the published design data has become obsolete by a change in the basis of specifying minimum properties, from one in which the length, width and thickness were considered, to one where only the thickness is involved; (2) the development of a technique of stress relief by cold work in compression has resulted in relatively new tempers (TX52) for many of the alloys; and (3) there have been some significant problems with forged parts in recent years that were related to fracture and stress-corrosion characteristics.

Accordingly, the properties of hand forgings of several aluminum alloys currently being used in aircraft structures are being determined under this contract. The tests are intended to provide statistically reliable data for deriving design mechanical properties for MIL-HDBK-5A, including stress-strain and compressive tangent-modulus curves. In addition, data concerning the fracture toughness, axial-stress fatigue, stress-corrosion, exfoliation and fatigue-crack propagation rates are being obtained.

This Third Technical Management Report summarizes the results of tests carried out during the third quarter of the contract, and the general status of the program at this time.

II. Material.

All but one of the hand forging samples to be investigated have now been obtained. The 6x24-in. 7075-T7352 forging which had been received did not meet the tentative minimum tensile properties and was returned to the plant for reheat treatment. The reheat-treated sample is expected within the next month.

III. Procedure.

All the specimens and test procedures were described in the First Technical Management Report, dated May 15, 1968.

IV. Progress During Quarter.

A. Mechanical Properties

A.1. Tensile, Compressive, Shear and Bearing

Tensile, compressive, shear and bearing tests have been made of 37 forgings, the results of which are shown in Tables I through IV. The ratios showing the relationships among these mechanical properties are shown in Table V.

Since the data for the various sizes of hand forgings of any one alloy are not complete, no detailed analysis has been made of the test results.

A.2. Fracture Toughness

Notch-bend fracture tests have been made of two of the hand forgings. The average values obtained are as follows:

<u>Alloy and Temper</u>	<u>Sample</u>			<u>K_{Ic}, psi√in.</u>
	<u>Cross-Sect. Size, in.</u>	<u>Number</u>	<u>Direction*</u>	
2014-T652	5x20	341013	LW	29 200
			WL	19 600
7075-T7352	4x16	341030	LW	32 700
			WL	26 500

The above values are considered to be valid, although for some individual specimens the stress intensity used in fatigue cracking was slightly in excess of 50 per cent of the K_{Ic} value, and the fatigue crack deviated from straightness by slightly more than 5 per cent of the thickness.

Specimens from most of the other hand forging samples have been prepared and are in the process of being fatigue cracked.

A.3. Axial Stress Fatigue

The axial-stress fatigue (R=0.0) tests of specimens from all except the 6x12-in. 7075-T7352 hand forging have been started, and approximately 80 per cent of the scheduled tests have been completed. The results of the tests are plotted in Figs. 1 through 4.

* The first letter indicates the direction of a line normal to the crack plane in the specimens; the second letter indicates the direction of crack growth. L - Longitudinal (major axis of forging); W - Width.

B. Corrosion Characteristics

Specimens have been obtained from sixteen of the twenty-three hand forgings scheduled for corrosion testing.

Specimens from the 2x8-in., 3x12-in. and 5x20-in. forgings have been machined, and it is expected that the stress-corrosion specimens will be exposed in late November, 1968. Specimens have also been prepared for exposure to the accelerated exfoliation test; however, due to the extensive backlog of specimens for this exposure, there will be some delay before the tests are started. Some of the specimens from the 4 and 6-in. thick samples have been prepared, but the testing is being delayed until all of the specimens from a given size of forging can be exposed concurrently.

Machined slices of the 6x6-in. 7075-T7352 and 7079-T652 forgings have been obtained for macroetching to determine the grain orientation. Preparation of specimens from these samples will be initiated shortly.

C. Fatigue Crack Propagation

Twenty long-transverse crack propagation specimens were prepared from the 6x24-in. 2014-T652 hand forging. Tests of these specimens have been initiated to study the effects of notch geometry, specimen length and periodic changes in stress on the rate of fatigue-crack propagation. The 6x24-in. 7079-T652 hand forging has been submitted to the Machine Shop for preparation of six additional specimens.

The program calls for evaluation of the effects of specimen orientation and humidity on the rate of crack propagation for two of the four alloys. It was originally indicated that the selection of the two alloys might be made when data comparing the propagation behavior of all four alloys was available. However, with the delay in receiving the forgings, this is not feasible. Thus, it is proposed that these tests be made on the alloys expected to show the greatest difference in resistance to stress-corrosion cracking; i.e., 7075-T7352 and 7079-T652.

V. Summary.

The tensile, compressive, shear and bearing properties determined for 37 of the 40 hand forgings scheduled for test are shown in Tables I through IV. The tensile properties of the hand forgings meet the applicable specified minimum-property requirements shown in Table VI. Ratios among the properties are as shown in Table V.

Notch-bend fracture toughness tests were made of two hand forgings, the average K_{Ic} values are shown in Section IV, A.2 of the text.

Approximately 80 per cent of the axial-stress fatigue tests have been completed. The results of the tests are plotted in Figs. 1 through 4.

Specimens from the 2, 3 and 5-in. thick forgings have been prepared and the stress-corrosion and accelerated exfoliation tests are to be started soon. Although some of the

specimens from the 4 and 6-in. thick samples have already been prepared, testing is being delayed so that all of the specimens from a given size of forging can be exposed concurrently. Machined slices from the 6x6-in. 7075-T7352 and 7079-T652 forgings have been obtained to determine the grain orientation.

Tests of the 6x24-in. 2014-T652 hand forging have been initiated to study the effects of notch geometry, specimen length and stress on the rate of fatigue-crack propagation. It is planned to evaluate the effects of specimen orientation and humidity for alloys 7075-T7352 and 7079-T652; these two alloys would be expected to show the greatest difference in resistance to stress-corrosion cracking.

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VI. Tables and Figures.

TABLE I

TABLE I

Mechanical Properties of Stress-Relieved 2014-T452 Aluminum Alloy Hand Forgings
(F33615-5R-C-13R5)

Sample Process- Sectional Size, In.	TENSILE			FLANG. IN 2 IN. OF AN.	RED. OF AREA, %	COMP. YIELD STRESS,* PSI	SHEAR ULT. STRESS, PSI	BEARING [†]		YIELD STRESS,† PSI e/D=1.5 e/D=2.0
	ULT. STRESS, PSI	YIELD STRESS,* PSI	ULT. STRESS, PSI e/D=1.5					ULT. STRESS, PSI e/D=2.0		
2x 9 341007	L	71 400	44 500	11.5	30	69 200	44 200	101 000	122 500	87 200
	LT	71 700	44 300	6.0	9	70 300	43 500	101 000	130 100	89 300
3x12 341008	L	71 400	44 200	10.5	28	68 400	42 200	102 300	132 500	89 400
	LT	71 000	45 100	7.5	12	69 400	41 400	97 100	126 400	88 500
4x 9 341009	L	69 700	42 200	5.0	7	69 700	41 300	---	---	---
	LT	70 300	44 200	12.5	29	66 400	40 400	89 300	123 700	85 500
4x14 341010	L	69 900	43 000	7.5	12	65 100	40 600	90 700	121 700	87 200
	LT	66 900	59 500	24.5	4	69 300	39 900	---	---	---
5x 5 341011	L	68 600	43 200	12.0	28	65 300	41 400	---	---	---
	LT	67 500	41 200	4.0	5	62 500	40 700	87 200	117 300	85 900
5x10 341012	L	65 200	54 200	2.0	4	66 500	41 200	---	---	---
	LT	68 800	41 600	11.5	27	63 000	40 600	93 400	117 400	82 700
5x20 341013	L	67 300	40 200	5.5	9	61 700	40 300	88 700	123 400	83 600
	LT	64 400	57 400	3.0	6	65 300	38 700	---	---	---
6x 6 341014	L	68 500	40 700	11.5	24	61 200	38 400	90 100	113 500	79 000
	LT	64 700	57 300	5.0	7	63 500	38 400	86 600	117 500	79 000
6x12 341015	L	63 900	56 100	3.7	7	62 400	37 300	---	---	---
	LT	67 700	42 000	12.0	31	64 000	42 400	97 400	114 200	86 700
6x24 341016	L	64 500	59 500	3.5	5	60 400	40 700	89 300	121 100	83 900
	LT	64 200	59 500	2.8	3	65 700	40 500	---	---	---
6x24 341016	L	64 200	58 400	11.0	27	60 300	40 200	91 100	120 100	81 600
	LT	64 200	58 400	3.5	6	61 900	38 800	87 700	119 000	80 400
6x24 341016	L	63 900	55 200	3.5	2	61 900	38 700	---	---	---
	LT	63 000	55 900	9.5	19	57 900	42 500	89 500	118 100	81 200
6x24 341016	L	64 400	57 700	6.0	6	62 400	38 400	86 300	117 900	80 000
	LT	62 600	54 000	6.0	14	59 300	39 000	---	---	---

* OFFSET EQUALS 0.2 PER CENT
 † OFFSET EQUALS 2.0 PER CENT OF PIN DIAMETER
 ‡ SOFTENING AND FIXTURES CLEARER ULTRASONICALLY IN TORSION
 § L - LONGITUDINAL; LT - LONG TRANSVERSE; ST - SHORT TRANSVERSE

TABLE II

MECHANICAL PROPERTIES OF STRESS-RFLEIVED 2024-T852 ALUMINUM ALLOY MAND FORGINGS
(F33615-6A-C-1384)

SAMPLE CONSEC- TIONAL SIZE, IN.	SECTIONAL NUMBER DESIGNATION	TENSILE			RED. OF AREA, %	COMP.	SHEAR	BEARING EDGEWISE			
		ULT. STRESS, PSI	YIELD STRESS, PSI	ELONG. IN 2 IN. OR 4D, %				ULT. STRESS, PSI e/D=1.5	YIELD STRESS, PSI e/D=2.0		
2x 8	341017	L	70 600	44 400	7.0	70 200	42 700	97 700	133 100	95 500	116 300
		LT	72 300	63 400	9.0	72 700	41 800	94 500	125 900	89 500	114 200
		ST	67 400	64 900	1.6	74 600	---	---	---	---	---
3x12	341018	L	72 200	64 700	5.5	70 000	42 400	94 900	123 400	93 700	109 300
		LT	73 700	69 000	3.0	67 900	42 000	94 500	126 100	93 500	113 300
		ST	68 100	64 400	1.0	72 200	40 200	---	---	---	---
4x 8	341019	L	68 900	41 100	9.0	62 200	40 500	91 900	117 900	82 500	100 500
		LT	70 400	63 200	5.0	62 500	39 500	88 800	119 400	82 800	101 000
		ST	65 700	57 200	3.2	65 500	38 600	---	---	---	---
4x14	341020	L	71 400	65 400	4.5	63 500	41 100	92 100	124 000	87 400	104 800
		LT	71 000	65 200	5.0	71 500	40 200	91 500	127 100	90 500	108 200
		ST	70 100	60 600	2.4	70 200	39 900	---	---	---	---
5x 5	341021	L	69 000	42 000	4.5	63 400	40 800	93 500	125 200	89 900	105 100
		LT	68 400	42 100	3.0	63 100	40 700	89 100	121 600	84 400	101 100
		ST	64 500	56 000	2.8	64 700	39 600	---	---	---	---
5x10	341022	L	68 400	41 000	4.5	63 000	40 300	89 100	114 300	87 700	96 800
		LT	69 100	41 500	6.0	64 800	39 700	89 500	120 200	85 100	99 600
		ST	64 100	55 900	1.5	64 400	38 400	---	---	---	---
5x20	341023	L	65 200	55 100	9.0	55 000	38 800	83 400	112 600	79 300	94 400
		LT	62 800	56 700	3.0	60 700	38 000	84 900	114 600	82 500	98 000
		ST	63 200	54 500	3.0	59 400	37 000	---	---	---	---
6x 6	341024	L	69 100	61 600	9.0	63 700	41 500	95 300	123 900	89 800	102 200
		LT	68 800	60 600	6.5	61 500	40 600	92 000	123 200	86 900	102 700
		ST	69 400	58 500	2.3	67 500	39 400	---	---	---	---
6x12	341025	L	67 000	58 700	8.0	59 700	39 600	84 700	117 100	82 400	100 500
		LT	67 400	60 200	3.2	63 500	38 400	85 700	113 400	81 400	95 700
		ST	65 300	55 100	2.9	63 000	37 400	---	---	---	---
6x24	341026	L	64 300	56 100	7.5	53 800	37 500	---	---	---	---
		LT	65 400	57 800	5.0	57 500	---	---	---	---	---
		ST	58 000	53 900	1.0	58 000	---	---	---	---	---

* OFFSET FOUJALS 0.2 PPD CFMT
 † OFFSET FOUJALS 2.0 PPD CFMT OF PIN DIAMETER
 ‡ SPECIMENS AND FIXTURES CLEANED ULTRASONICALLY IN TOSON
 § L, LONGITUDINAL; LT, LONG TRANSVERSE; ST, SHORT TRANSVERSE

TABLE III

MECHANICAL PROPERTIES OF STRESS-RELIEVED 7075-T7352 ALUMINUM ALLOY HAND FORGINGS

Q33615-6A-C-13851

SAMPLE CROSS- SECTIONAL DIMENSIONS SIZE, IN.	TENSILE			RED. OF AREA, %	COMP. YIELD STRESS, PSI	SHEAR ULT. STRESS, PSI	BEARING [‡]		
	ULT. STRESS, PSI	YIELD STRESS,* PSI	ELONG. IN 2 IN. OR 40. %				ULT. STRESS, PSI e/D=1.5	YIELD STRESS, [†] PSI e/D=2.0	
2X 8 341027	L	73 700	65 300	13.5	69 300	46 900	111 900	93 700	111 200
	LT	74 900	65 300	13.5	69 300	44 500	110 600	92 700	106 500
	ST	73 100	61 600	6.3	69 300	---	---	---	---
3X12 341028	L	76 400	66 200	11.5	66 900	42 400	103 100	89 000	103 800
	LT	71 400	59 300	8.0	65 300	47 600	98 300	89 800	110 300
	ST	73 000	60 800	4.2	69 300	47 900	---	---	---
4X 8 341029	L	68 400	57 300	15.0	60 200	39 800	95 100	83 500	98 600
	LT	65 100	53 000	10.0	57 600	38 400	98 500	81 400	99 000
	ST	64 500	50 400	6.4	57 500	38 200	---	---	---
4X16 341030	L	70 000	59 500	13.0	57 400	40 600	95 300	82 900	95 800
	LT	67 600	55 200	12.0	59 700	40 700	94 200	82 600	99 200
	ST	64 800	52 500	6.4	58 600	39 100	---	---	---
5X 5 341031	L	68 400	54 700	14.0	59 400	41 500	104 400	84 300	99 000
	LT	67 200	55 100	10.5	56 600	40 600	98 000	83 500	100 700
	ST	63 800	51 700	4.0	59 500	41 500	---	---	---
5X10 341032	L	65 200	52 700	14.0	53 400	39 600	95 900	82 300	91 900
	LT	64 000	51 400	9.0	53 800	38 500	97 700	80 100	97 000
	ST	64 200	49 500	7.0	58 000	39 400	---	---	---
5X20 341033	L	64 800	52 500	14.5	52 200	38 800	94 100	76 800	89 100
	LT	64 000	50 700	11.0	54 400	38 300	91 500	77 100	92 600
	ST	63 700	49 300	6.5	54 900	38 000	---	---	---
6X 6 341034	L	62 400	51 100	15.0	54 000	41 300	99 300	82 100	94 400
	LT	63 800	52 100	10.0	53 000	40 100	97 400	81 600	96 100
	ST	63 400	49 700	8.0	55 300	39 000	---	---	---
6X12 341035	L	---	---	---	---	---	---	---	---
	LT	---	---	---	---	---	---	---	---
	ST	---	---	---	---	---	---	---	---
6X24 341036	L	---	---	---	---	---	---	---	---
	LT	---	---	---	---	---	---	---	---
	ST	---	---	---	---	---	---	---	---

* OFFSET EQUALS 0.2 PER CENT
 † OFFSET EQUALS 2.0 PER CENT OF PIN DIAMETER
 ‡ SPECIMENS AND FIXTURES CLEANED ULTRASONICALLY IN TOSOL
 § L - LONGITUDINAL; LT - LONG TRANSVERSE; ST - SHORT TRANSVERSE

TABLE IV
MECHANICAL PROPERTIES OF STRESS-RELIEVED 7079-T652 ALUMINUM ALLOY HAND FORGINGS
(F33615-6R-C-1385)

SAMPLE CROSS- SECTIONAL NUMBER DIRFC- SIZE, IN.	TENSILE			RED. OF AREA, %	COMP. YIELD STRESS,* PSI	SHEAR ULT. STRESS, PSI	READING‡ EDGEWISE			
	ULT. STRESS, PSI	YIELD STRESS,* PSI	ELONG. IN 2 IN. OR 4D, %				ULT. STRESS, PSI e/D=1.5	YIELD STRESS, PSI e/D=2.0		
2x 8	L	79 600	71 000	14.0	73 300	48 700	115 100	154 700	99 100	114 400
	LT	76 100	64 900	12.0	73 200	46 500	114 500	149 100	98 000	113 400
	ST	76 000	63 700	7.8	74 200	---	---	---	---	---
3x12	L	77 500	68 700	13.0	71 300	46 400	113 200	148 800	94 800	112 900
	LT	74 100	65 700	12.0	70 700	46 100	116 600	149 100	97 900	114 600
	ST	73 700	61 400	8.0	71 800	45 400	---	---	---	---
4x 8	L	78 800	69 600	11.0	72 800	48 900	111 600	148 300	99 400	115 200
	LT	77 500	66 500	11.5	72 900	48 200	117 100	148 700	102 300	117 200
	ST	74 300	62 900	5.0	73 200	67 300	---	---	---	---
4x16	L	77 900	68 000	12.0	70 100	66 600	113 000	145 900	95 200	110 300
	LT	74 600	63 000	9.5	66 800	45 700	107 500	146 400	94 000	105 700
	ST	74 000	62 900	7.9	70 600	44 900	---	---	---	---
5x 8	L	75 600	67 600	13.0	69 700	47 900	112 600	149 900	94 400	108 900
	LT	72 900	63 000	8.5	67 000	45 900	105 200	143 600	92 100	107 600
	ST	71 300	59 500	7.0	68 400	46 300	---	---	---	---
5x10	L	74 100	68 000	13.0	68 800	45 700	108 200	140 900	92 800	107 000
	LT	74 100	62 400	10.5	69 300	45 900	108 300	141 300	94 300	109 100
	ST	73 000	61 300	5.5	72 200	44 400	---	---	---	---
5x20	L	74 900	65 600	13.0	67 000	46 200	104 600	135 900	91 800	106 400
	LT	73 300	61 400	11.0	65 700	44 400	103 300	136 900	89 800	105 400
	ST	71 300	58 300	6.0	68 300	44 000	---	---	---	---
6x 6	L	73 600	63 800	15.0	68 900	48 400	112 200	148 100	95 600	105 400
	LT	72 400	61 400	9.0	69 700	47 900	111 000	146 000	96 700	109 400
	ST	71 700	61 800	8.5	67 100	47 300	---	---	---	---
6x12	L	75 200	65 700	11.0	67 500	46 300	109 000	139 100	93 800	107 500
	LT	72 800	62 100	7.5	68 200	45 500	104 000	140 700	92 300	107 600
	ST	72 400	58 800	6.0	69 300	44 700	---	---	---	---
6x24	L	73 900	63 900	12.0	63 300	43 800	---	---	---	---
	LT	69 100	57 500	10.0	62 900	42 000	---	---	---	---
	ST	69 300	58 100	6.5	67 300	42 000	---	---	---	---

* OFFSET FOCALS 0.2 PER CFMT
 † OFFSET FOCALS 2.0 PER CFMT OF PIN DIAMETER
 ‡ SPECIMENS AND FIXTURES CLEANED ULTRASONICALLY IN TOSOL
 § L, LONGITUDINAL; LT, LONG. TRANSVERSE; ST, SHORT TRANSVERSE

TABLE VI

SPECIFIED MINIMUM VALUES FOR ALUMINUM ALLOY HAND FORGINGS
(F33015-08-C-1985)

Alloy and Temper	Thickness, in.	Longitudinal		Long-Transverse		Short-Transverse		Federal Specification
		Tensile Strength, psi	Yield Strength, in 4D, psi	Tensile Strength, psi	Yield Strength, in 4D, psi	Tensile Strength, psi	Yield Strength, in 4D, psi	
2014-T52	Up thru 2.000	65 000	56 000	65 000	56 000	62 000	52 000	SQ-A-3678
	2.001-3.000	64 000	56 000	64 000	55 000	61 000	51 000	
	3.001-4.000	63 000	55 000	63 000	54 000	60 000	50 000	
	4.001-5.000	62 000	54 000	62 000	53 000	59 000	50 000	
	5.001-6.000	61 000	53 000	61 000	53 000	59 000	50 000	
2024-T52	All	---	---	---	---	---	---	None
7075-T752	Up thru 3.000	66 000	54 000	64 000	52 000	61 000	50 000	None**
	3.001-4.000	64 000	53 000	63 000	50 000	60 000	49 000	
	4.001-5.000	62 000	51 000	61 000	48 000	58 000	48 000	
	5.001-6.000	61 000	49 000	59 000	45 000	57 000	47 000	
	Up thru 2.000	72 000	61 000	71 000	61 000	67 000	55 000	
7075-T62	2.001-3.000	72 000	62 000	70 000	60 000	67 000	55 000	SQ-A-3678
	3.001-4.000	71 000	61 000	70 000	59 000	66 000	54 000	
	4.001-5.000	70 000	60 000	69 000	58 000	66 000	53 000	
	5.001-6.000	69 000	59 000	68 000	56 000	66 000	53 000	

* Official equals 0.2 per cent.
** The Aluminum Association, "Aluminum Standards and Data", April 1968.

TABLE VI

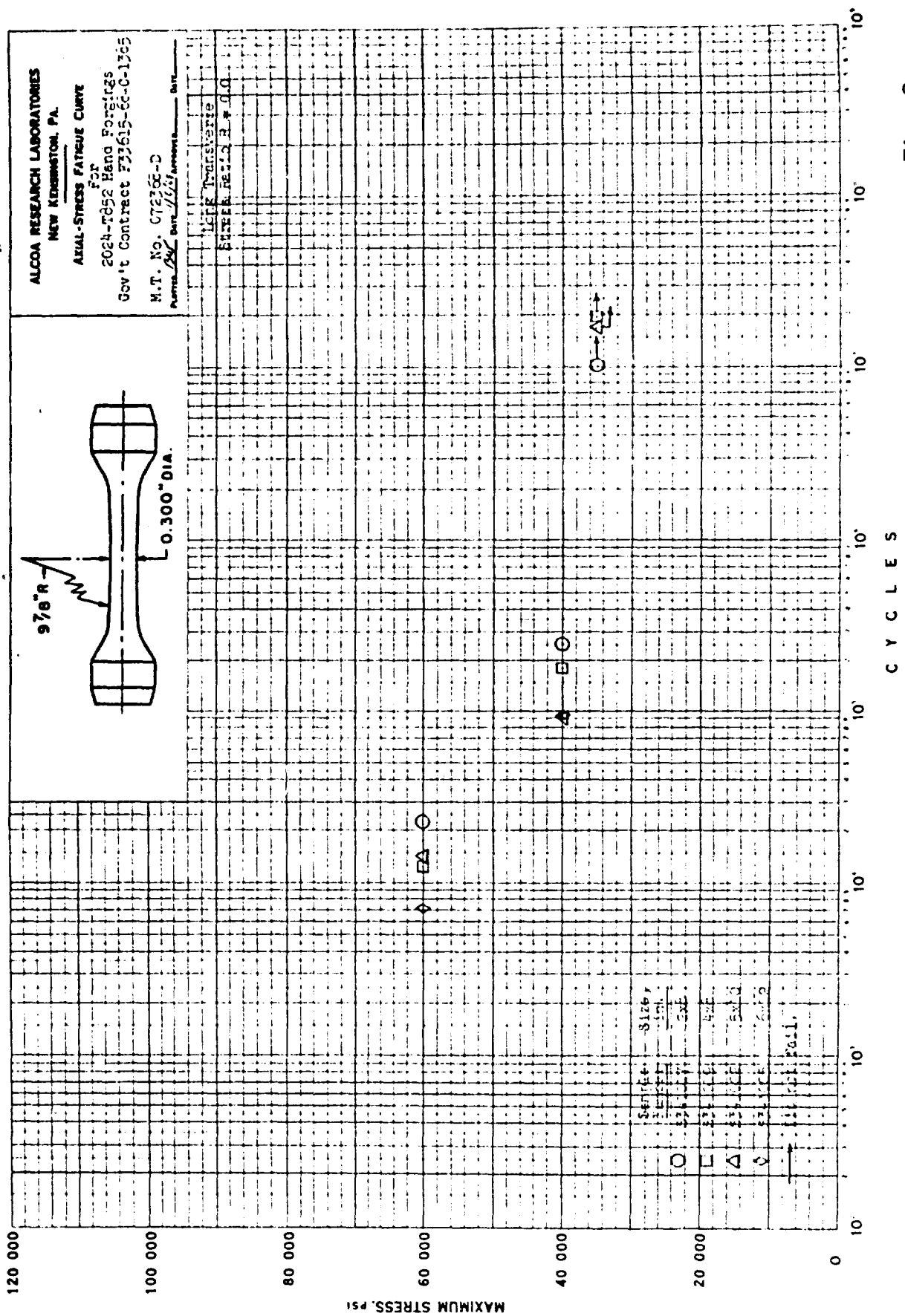


Fig. 2

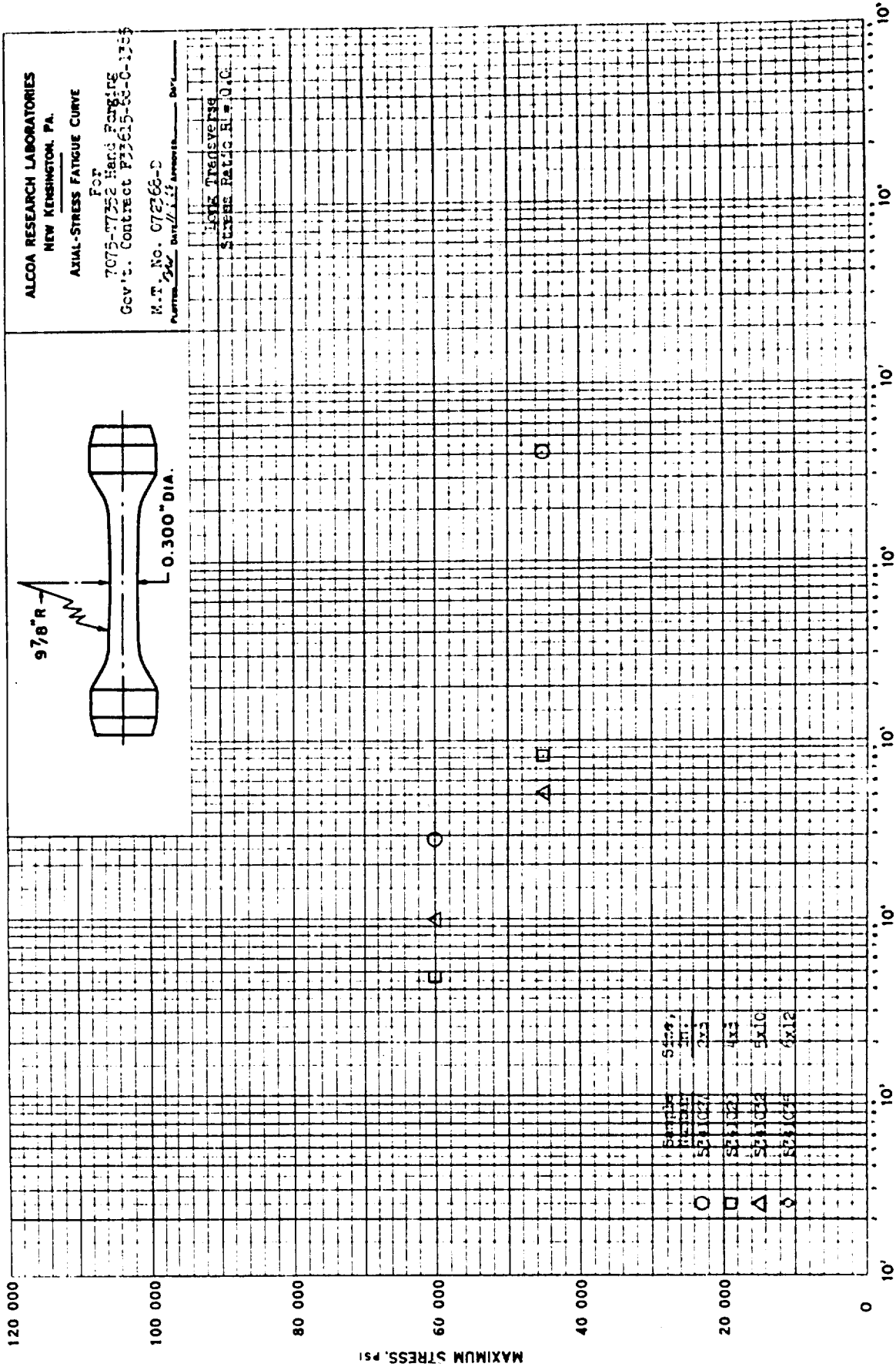


Fig. 3

ALCOA RESEARCH LABORATORIES
 NEW KENNESBETH, PA.
 AXIAL-STRESS FATIGUE CURVE
 PCS
 7079-T652 Hand Forgings
 Gov't. Contract F73615-68-C-1385
 M.T. No. 072766-D
 PART 1-52 DATE 12/2/68 APPROVED DATE

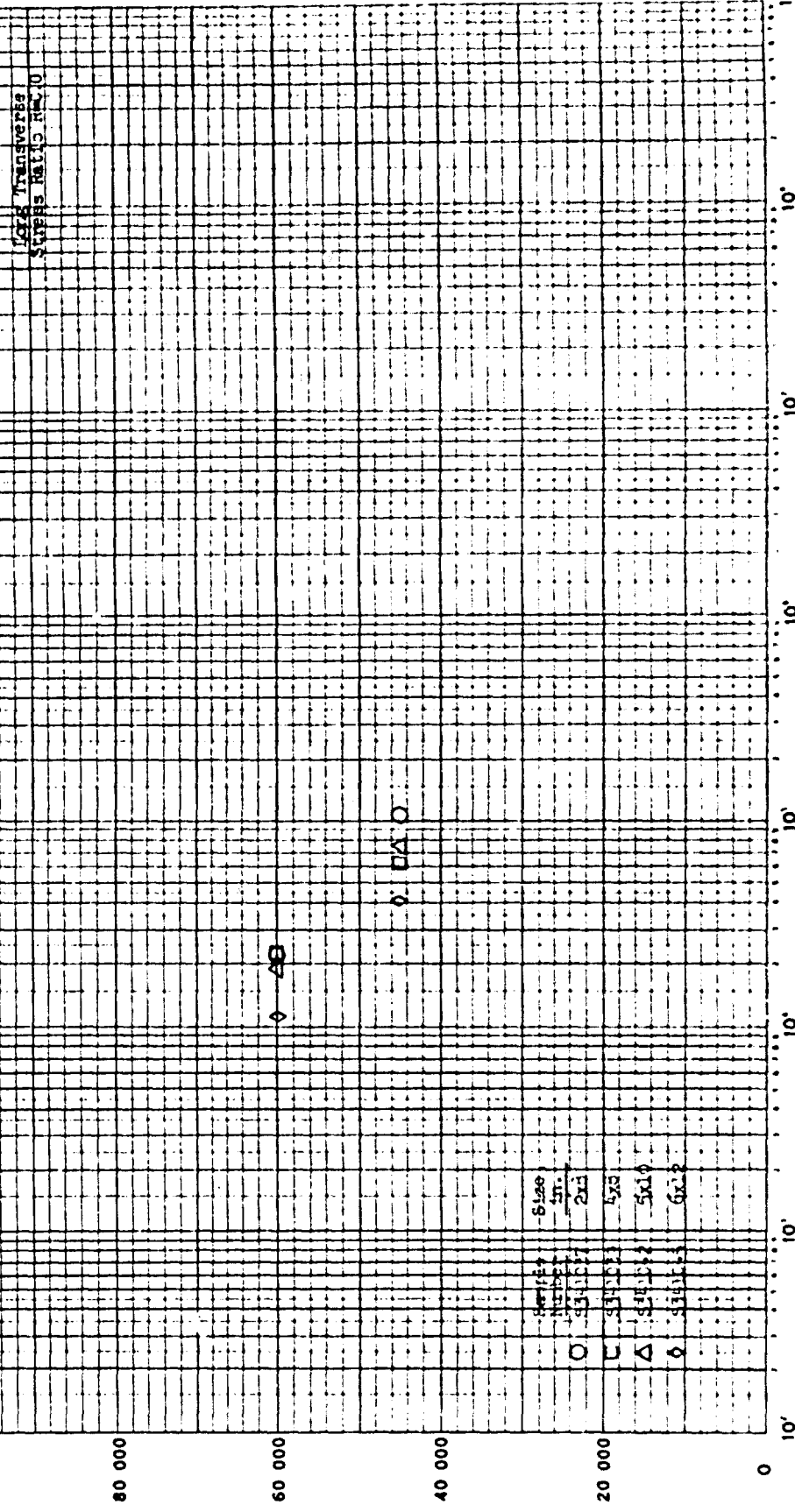
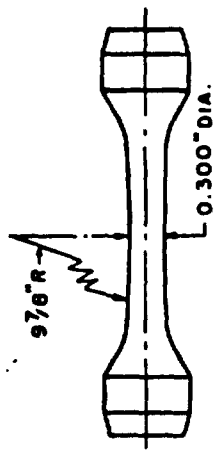


FIG. 4

C Y C L E S

MAXIMUM STRESS, PSI