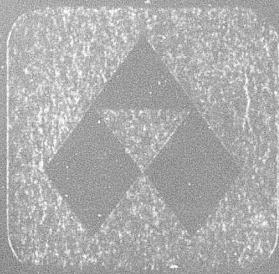


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## MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, AND RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS- RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

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Contract No. AF33(615)-3580  
BPSN: 66 (687381-738106-62405514)

## Fourth Quarterly Report

Dec. 15, 1966 - March 15, 1967

New Kensington, Pa. March 15, 1967

New Kensington, Pa. March 15, 1967

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### ABSTRACT

The tensile and some compressive, shear, bearing, fatigue and fracture-toughness properties of a total of 130 samples of 2014, 2024, 6061, 7075 and 7178 aluminum alloy extrusions in the TX51X and "heat-treated-by-user" tempers have been determined. The extrusions ranged in thickness from 0.050 to 6.500 in. Ratios among the tensile, compressive, shear and bearing properties have also been computed.

Stress-corrosion tests of 18 samples of extrusions have been completed.

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## FOURTH QUARTERLY REPORT

### MECHANICAL PROPERTIES, INCLUDING FRACTURE TOUGHNESS AND FATIGUE, AND RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS

#### I. Introduction.

The tests being made under this contract are for use in establishing design mechanical properties in MIL-HDBK-5A, including stress-strain and tangent-modulus curves, for 2014, 2024, 6061, 7075, 7079 and 7178 aluminum alloy extrusions in the TX51X tempers. For comparison, a limited number of similar tests are being made of extrusions in the "heat-treated-by-user" temper. Also, some fracture-toughness, axial-stress fatigue and stress-corrosion tests are being made.

This Fourth Quarterly Report summarizes the results of tensile, compressive, shear, bearing, fatigue, fracture toughness and stress-corrosion tests made to date on 102 samples in the TX51X temper and on 28 samples in the "heat-treated-by-user" tempers. The samples ranged in thickness from 0.050 to 6.500 in.

#### II. Material.

A total of 110 samples of commercially-produced extrusions in the TX51X temper and 18 samples in the O temper have been received from two producers. The section thickness and identification of each sample is shown in Table I. Eighteen of the as-received samples in the O temper have been heat treated, or heat treated and aged, in accordance with applicable

conditions in MIL-H-6088D. Five samples each of 2024-O and 7075-O were tested in two "heat-treated-by-user" tempers, so that the total number of samples tested in those tempers is 28.

### III. Procedure.

#### Mechanical Properties

Tensile, compressive, shear and bearing specimens were taken at locations described in the Second Quarterly Report, dated September 15, 1966. The general dimensions of these specimens were shown in Figs. 1 to 3, inclusive, of that report. The testing procedures were as outlined in the First Quarterly Report, dated June 15, 1966.

Tensile and compressive modulus and stress-strain tests have been made of 27 samples of extrusions as described in the Third Quarterly Report, dated December 15, 1966.

Fracture-toughness tests were made in accordance with the methods described in ASTM STP 411\* on fatigue-cracked single-edge-notched specimens from the longitudinal and long-transverse directions. The types of specimens are shown in Fig. 1; the proportions of these specimens are the same as those of specimens used by NASA, Lewis Research Center. The fracture parameters were calculated from relationships developed from the NASA calibration.

Values of the critical plane-strain stress intensity factor,  $K_{Ic}$ , were calculated using two values of load from the

\* W. F. Brown and J. E. Srawley, "Plane Strain Fracture Toughness Testing of High Strength Metals", ASTM STP 411, February, 1967.

fracture-toughness tests. The first value was calculated using the load at the initial burst of unstable crack growth, as indicated by the initial significant deviation from linearity in the load deformation curve. In reporting the data, the degree of clarity of the initial deviation has been indicated by the use of the letter "P" to indicate a clear instability or pop-in, the letter "I" to indicate a less pronounced but yet abrupt initial deviation from linearity which is believed to be a suppressed pop-in, and "M" to indicate that the initial deviation was at the maximum or fracture load. The second value was calculated using the load at a 5 per cent secant offset, equivalent to about 2 per cent of crack extension; this was done as a result of recent recommendations of ASTM Committee E-24\* that the secant-offset method be considered for establishing  $K_{Ic}$ , and values thus determined are indicated in the table with the letter "S", or the letters "SM" when the load at the secant offset was the maximum or fracture load. Additional analysis of the results will be necessary to establish the permanent validity of the numbers. No values of  $K_{Ic}$  are reported at this time for those materials (principally 2024-T351X and 6061-T6510) which exhibited no initial instability and/or general yielding.

Some axial-stress fatigue tests have been made of fourteen selected samples of extrusions. Procedures for making these tests were described in the First Quarterly Report.

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\* "Draft Recommended Practice for Notch-Bend Fracture Toughness Testing", ASTM Committee E24, February, 1967.

### Resistance to Stress Corrosion

Stress-corrosion tests are being made of 43 selected samples of extrusions; of these, 31 are in the TX51X temper and 12 are in the "heat-treated-by-user" temper. Procedures for making these tests were described in the First Quarterly Report.

### IV. Summary.

The results of tensile, compressive, shear and bearing tests of 102 samples of extrusions in the TX51X temper are shown, according to alloy and temper, in Tables II through IX; the corresponding properties for extrusions in the "heat-treated-by-user" tempers are shown in Table X. The tensile properties of all samples exceed the values in applicable Federal Specifications. Specified minimum tensile properties for extrusions are shown in Table XI; the values shown in this table are those published in the Aluminum Association Booklet, "Standards for Aluminum Mill Products", 1967. Some of these values, as indicated in the table, are lower than those shown presently in Federal Specifications. It is understood, however, that the values in "SAMP" will be in the next revisions of the Federal Specifications.

The ratios among the tensile, compressive and shear properties of the extrusions in the TX51X tempers and the "heat-treated-by-user" tempers are shown in Tables XII and XIII, respectively, and the ratios among the bearing and tensile properties of the corresponding extrusions are shown in Tables XIV and XV, respectively. The ratios among the properties at

different locations with regard to width and thickness are shown in Table XVI. The ratios among bearing properties obtained using edgewise specimens to those obtained using flatwise specimens are shown in Table XVII.

The results of the fracture-toughness tests are shown in Table XVIII and those of the axial-stress fatigue tests are shown in Fig. . 2 through 6.

The current status of the stress-corrosion tests is given in Table XIX. The tests of specimens of 18 samples were terminated after 84 days exposure, and tensile tests were made of these specimens to determine losses in tensile strength as a result of corrosion. The tests thus far indicate typical performances for the various materials.

Because it is expected that more samples of each alloy and temper will be tested, no analysis of the data already obtained is being presented at this time.

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V. Tables and Figures.

**TABLE I**  
**EXAMPLES OF EDITIONS RECEIVED AS OF MARCH 15, 1967**  
**[F33(615)-3580]**

\* Producer B; all others from Producer A

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TABLE II  
MECHANICAL PROPERTIES OF STRESS-RELIEVED, STRETCHED 2014-T6510 ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3560]

Sample	Section Cross- Sections, in. <sup>1</sup>	Loca- tion*	Dir- ection†	Tensile		Elongation in. or 4D, %	Comp. Yield Stress,‡ psi	Shear Ultimate Stress, psi	Bearing**	
				Tensile Ultimate Stress, psi	Tensile Yield Stress,‡ psi				Flatwise	
				e/D=1.5	e/D=2.0				e/D=1.5	e/D=2.0
0.061	0.30	317950	T/2	L	67 100	62 200	9.0	64 600	--	105 600
0.070	0.24	318017#	T/2	LT	74 700	68 600	5.0‡	62 800	--	103 900
0.073	0.16	317951	T/2	L	64 900	61 900	9.5	59 900	--	--
0.246	0.45	318130#	T/2	L	62 800	59 100	11.5	61 000	--	--
0.250	3.7	340154	T/2	L	64 400	62 400	10.0	44 300	110 100	107 500
0.271	0.40	317994	T/2	LT	64 900	62 000	12.5	63 800	109 200	113 700
				L	68 800	63 700	5.0‡	69 300	93 200	--
				L	67 500	63 600	13.5	63 400	45 700	--
				L	71 700	64 600	6.2	--	47 200	--
0.625	0.50	317952	T/2	L	65 900	61 900	14.3	60 300	44 700	--
0.625	0.55	340291	T/2	L	77 500	71 900	10.7	68 000	42 100	--
0.750	1.4	317924	T/2	L	76 700	71 100	11.5	68 400	42 800	110 000
1.657	2.2	318046	T/4	LT	70 300	63 500	7.8	73 100	142 800	93 600
				L	72 600	66 100	10.5	67 800	41 500	106 200
				L	73 200	66 500	10.5	68 200	105 700	--
				L	67 400	60 200	8.8	--	104 800	88 000
				L	--	--	--	--	132 500	103 100
				L	--	--	--	--	--	99 700

\* T - Thickness; W - Width  
† L - Longitudinal; LT - Long-Transverse;  
‡ Offset equals 0.2 per cent of pin diameter  
# Sub-size sheet-type specimen; 1/6-in. wide; 1/2-in. gage length

\*\* Specimens and fixtures cleaned ultrasonically in Toson 3 solvent  
†† Offset equals 2 per cent of pin diameter  
# Sub-size sheet-type specimen; 1/6-in. wide; 1/2-in. gage length

**TABLE III**  
MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 2024-T3510 ALUMINUM ALLOY EXTENSIONS  
[AF33(615)-3580]

Sample Number	Section Cross-Sectional Area, in. <sup>2</sup>	Thickness, in.	Loca-tion	Dir-ection-tion	Tensile Ultimate Stress, psi	Elongation in 2 in. or 40. s	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Ultimate Plastic		Bearing*		
									Stress, psi		Stress, psi		
									Offset 1/16 in. or 2.0				
0.075	0.70	318132#	T/2	L	65 900	53 000	18.0	45 200	--	96 800	119 200	78 300	94 300
0.094	0.30	318019#	T/2	L	63 900	52 800	15.0***	57 800	--	97 200	127 800	77 900	92 300
0.101	0.33	317985	T/2	L	63 900	52 800	16.5	58 000	--	102 200	126 900	77 200	90 300
0.106	0.31	317904	T/2	L	64 400	51 700	20.5	42 100	--	102 200	125 300	77 200	91 100
0.120	0.27	318028***	T/2	L	65 400	54 400	17.5	45 000	--	101 900	129 500	78 900	92 600
0.151	0.82	317986	T/2	L	67 700	49 900	11.0***	55 200	--	102 300	123 600	78 300	91 200
0.255	2.8	317942	T/2	L	79 400	61 800	16.0	52 200	43 200	99 100	124 000	74 700	85 000
~.258	4.2	318047	T/2	L	79 800	63 900	15.0***	59 200	41 200	102 300	132 100	81 200	99 400
0.375	0.62	317943	T/2	L	63 900	49 300	20.0***	58 100	42 200	107 200	134 400	79 500	97 700
0.510	10.1	317926	T/2, W/4	L	57 600	38 900	6.5	37 400#	--	106 200	133 400	79 500	101 400
0.525	1.9	318020#	T/2, W/4	L	64 900	48 500	15.0	50 000	40 600	100 600	130 600	77 400	94 800
0.550	1.9	317955	T/2, W/4	L	63 000	48 300	23.5	52 500	39 800	100 600	121 200	--	--
0.642	5.8	317945	T/2, W/4	L	62 600	42 600	21.0	40 300	41 200	102 000	118 900	74 400	87 900
0.950	4.6	317944	T/2, W/4	L	62 600	43 500	12.0	47 500	42 200	105 100	130 700	79 500	97 300
1.150	5.6	318077	T/2, W/4	L	78 700	62 100	17.0	55 500	39 600	108 000	126 900	78 200	93 900
1.200	3.9	317946	T/2, W/4	L	69 600	59 200	20.5	51 200	40 100	102 900	129 200	75 400	93 800
1.450	7.3	318021#	T/2, W/4	L	60 300	63 700	16.4	58 000	38 800	105 500	134 300	75 400	101 900
1.705	4.8	3400213	T/4, W/4	L	82 500	62 300	15.0	59 400	40 600	110 800	136 400	76 700	95 900
2.520	8.8	318132#	T/4, W/4	L	69 600	49 500	14.0	60 700	32 000	108 600	133 200	76 400	106 500
4.000	24.0	340214	T/4, W/4	L	65 800	44 100	12.5	49 200	37 800	102 300	130 800	71 600	94 000
4.000	24.0	340214	T/4, W/4	L	79 800	52 800	14.0	57 700	38 800	97 000	122 300	80 700	95 700
4.000	24.0	340214	T/4, W/4	L	64 400	47 700	8.0	51 400	36 700	95 000	120 300	90 700	96 600
2.760	29.6	318048	T/4, W/4	L	62 900	40 200	14.1	49 800	35 700	96 000	123 500	76 600	93 900
2.760	29.6	318048	T/4, W/4	L	60 500	42 800	13.5	54 900	40 900	94 000	118 100	73 400	88 200
				ST	58 900	47 800	6.0	49 700	37 700	91 600	115 400	72 900	90 700

\* T - Thickness; W - Width  
\*\* Offset equals 0.2 per cent  
# Producer B; all others from Producer A

\*\* Specimens and fixtures cleaned ultrasonically in Teson 3 solvent.  
†† Offset equals 2 per cent of pin diameter.  
# Average of two tests; all others, single tests.

†† Sub-size sheet-type specimen; 1/4-in. wide; 1-in. gauge length.  
\*\*\* Sub-size sheet-type specimen; 1/8-in. wide; 1/2-in. gauge length.  
\*\* Samples were in the machined condition.

TABLE IV  
MECHANICAL PROPERTIES OF STRESS-RELIEVED STRENGTHENED 2024-T8510 ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-358Q]

Section Thickness, in.	Sample Cross- Sectional Area, in. <sup>2</sup>	Loca- tion*	Direc- tion†	Tensile Ultimate Stress, psi	Tensile Yield Stress,* psi	Elongation 1 in. or 4D. %	Shear Ultimate Stress,* psi	Comp. Stress,* psi	Ultimate Flattening		Bearing** Ultimate Stress, psi	Field Stress, psi at 1.5% Offset														
									Offset psi	Offset psi																
0.075	0.70	318022#	T/2	72	400	68	400	6.0	70	300	--	110	200	141	400	98	900	114	800	--	--	--	--			
0.094	0.30	318134#	T/2	72	500	69	500	3.0***	73	300	--	--	109	000	139	500	100	000	118	000	--	--	--	--		
0.101	0.33	317887	T/2	73	200	69	800	7.0	65	700	--	106	800	141	300	95	700	122	400	--	--	--	--			
0.106	0.31	317896	T/2	70	200	66	200	6.0	66	800	--	107	600	141	300	95	500	117	600	--	--	--	--			
0.120	0.27	318023#	T/2	71	300	67	700	8.5	67	800	--	109	600	142	300	97	800	110	300	--	--	--	--			
0.151	0.82	317889	T/2	71	400	67	900	8.0***	69	600	--	112	400	144	800	101	200	117	800	--	--	--	--			
0.255	2.8	317890	T/2	77	600	70	900	8.5	75	500	44	200	113	100	146	600	99	400	117	800	--	--	--	--		
0.258	4.2	318082	T/2	77	600	76	400	10.0***††	73	300	42	900	114	800	118	000	143	600	101	000	119	400	--	--	--	--
0.375	0.62	317891	T/2	74	800	70	000	9.5***††	71	800	42	800	109	700	141	400	98	700	108	300	--	--	--	--		
0.510	10.1	317892	T/2,W/2	73	600	66	900	9.5	70	500	41	600	108	200	139	800	98	000	109	400	--	--	--	--		
0.525	1.9	318024#	T/2,W/4	72	700	65	900	10.0	69	800	41	500	--	--	--	--	--	--	--	--	--	--	--	--		
0.550	1.9	317922	T/2,W/4	74	100	67	900	10.5	67	000	40	900	104	500	135	300	91	600	106	800	--	--	--	--		
0.642	5.8	317894	T/2,W/4	75	300	69	000	8.0	70	200	41	300	108	600	139	600	94	800	112	400	--	--	--	--		
0.950	4.6	317893	T/2,W/4	71	000	65	200	5.0	67	700	41	200	105	700	135	800	96	600	112	200	--	--	--	--		
1.150	5.6	318078	T/2,W/4	70	500	65	700	4.5	67	900	40	900	106	900	136	700	94	800	111	400	--	--	--	--		
1.200	3.9	317895	T/2,W/4	74	600	68	500	7.5	70	300	41	200	105	700	135	800	96	600	112	900	--	--	--	--		
1.450	7.3	318024#	T/2,W/4	74	800	67	200	4.5	70	700	41	200	106	200	135	200	93	800	110	100	101	300	131	000		
1.705	4.8	340169	T/4,W/4	71	400	63	200	10.0	65	700	40	000	106	600	135	500	95	200	112	900	101	900	130	700		
4.000	24.0	340225	T/4,W/4	71	600	63	100	6.0	65	600	39	000	105	700	135	800	90	900	106	000	104	300	129	400		
2.760	29.6	318079	T/4,W/4	70	500	63	200	10.0	65	600	38	300	96	900	128	100	88	000	104	400	95	600	121	900		
		T/2,W/2	T/2,W/2	64	900	67	700	10.0	62	800	39	000	106	600	131	400	89	300	124	900	93	700	122	000		
		T/2,W/2	T/2,W/2	65	600	68	600	6.0	62	900	37	200	100	600	131	400	87	600	105	400	84	000	116	900		
		T/2,W/2	T/2,W/2	64	100	66	200	9.5	61	600	38	100	95	200	126	800	85	900	105	900	87	600	127	400		
		T/2,W/2	T/2,W/2	64	100	63	600	2.5	62	800	36	200	--	--	--	--	--	--	--	--	--	--	--	--		
		T/2,W/2	T/2,W/2	64	200	63	200	9.0	64	200	40	800	89	200	121	600	107	300	104	400	90	300	105	700		
		T/2,W/2	T/2,W/2	64	200	61	300	10.0	62	800	39	000	95	100	124	100	90	500	104	600	87	600	105	200		
		T/2,W/2	T/2,W/2	64	200	61	300	2.4	64	200	29	100	95	700	122	800	89	600	105	400	87	600	126	900		
		T/2,W/2	T/2,W/2	64	200	61	800	8.5	63	700	40	600	40	100	124	100	89	400	118	400	87	600	127	400		
		T/2,W/2	T/2,W/2	64	200	59	600	1.8	64	200	39	200	--	--	--	--	--	--	--	--	--	--	--	--		

\* T - Thickness; W - Width  
† L - Longitudinal; LF - Long-Transverse; ST - Short-Transverse  
‡ Offset equals 0.2 per cent.  
# Producer B: all others from Producer A

\*\* Specimens and fixtures cleaned ultrasonically in toson 3 solvent.

†† Offset equals 2 per cent of pin diameter.

\*\*\* Bearing specimen failed before reaching yield stress (2 per cent of offset).

TABLE V

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 6061-T6510 ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580]

Section Thickness, in.	Sample Number	Cross-Sectional Area, in. <sup>2</sup>	Loca-tion*	Direc-tion†	Tensile Properties			Shear Ultimate Stress, psi	Com-p. Yield Stress, psi	Ultimate Flatwise Stress, psi		Bearing**		
					Ultimate Tensile Stress, psi		Elongation in 2 in. or 4D, %			Ultimate Yield Stress, psi		Elongation in 2 in. or 4D, %	Ultimate Yield Stress, psi	
					psi	psi	%			psi	psi	%	psi	psi
0.050	0.42	318136#	T/2	L	45,000	42,400	11.0	41,500	--	79,400	107,300	66,900	75,800	--
0.075	0.59	317857	T/2	L	45,900	39,300	13.0	40,100	--	74,200	91,700	59,400	66,800	--
0.090	0.27	318037#	T/2	L	44,700	37,200	15.0††	38,800	--	89,100	105,200	67,400	77,600	--
0.125	0.61	317846	T/2	L	44,700	42,100	13.5	41,300	--	77,600	103,500	66,200	75,600	--
0.126	0.30	317847	T/2	L	43,900	38,700	10.5	38,700	--	70,900	90,700	58,000	64,300	--
0.250	0.36	317848	T/2	L	43,200	43,200	12.0	39,800	--	31,400	--	--	--	--
0.310	6.3	317905	T/2	L	47,700	47,700	20.0	42,300	--	38,800	30,800	31,100	79,500	--
0.315	5.8	317953	T/2	L	49,100	43,400	16.5	44,600	--	39,700	103,400	80,700	73,200	--
0.375	8.6	317927	T/2	L	45,500	40,200	18.5	40,100	--	34,400	78,000	100,600	65,700	--
0.375	7.7	318083	T/2	L	45,900	44,700	18.5	41,600	--	41,600	56,500	80,000	65,300	--
0.918	1.3	317906	T/2, W/2	L	45,900	42,500	16.0	42,000	--	42,000	56,000	80,600	67,400	--
1.240	2.7	317907	T/2, W/4	L	45,500	40,800	20.0	41,700	--	42,000	35,400	81,800	72,600	--
1.960	4.4	317896	T/2, W/4	L	48,500	43,500	18.5	44,300	--	41,700	33,000	81,400	69,900	--
3.000	15.0	340226	T/4, W/4	L	43,300	37,600	18.5	46,800	--	29,600	76,000	96,300	75,700	--
6.500	33.2	317897	T/4	L	51,600	46,800	12.5	47,200	--	75,100	96,300	60,000	68,900	--
			T/2, W/2	L	52,000	47,200	18.5	47,800	--	27,800	76,100	98,600	73,200	--
			T/2, W/2	L	46,900	43,800	14.0	45,500	--	27,800	74,500	95,900	74,100	--
			ST	L	47,100	40,600	13.0	44,700	--	28,700	72,600	93,800	71,000	--
			T/4	L	51,500	46,100	13.5	46,000	--	27,000	57,200	71,600	57,200	--
			T/2	L	44,600	38,000	12.5	40,700	--	27,000	89,500	68,200	86,600	--
			T/2	L	51,200	45,700	13.5	45,600	--	26,800	71,500	56,500	67,500	--
			T/2	L	44,600	37,000	12.5	39,600	--	26,600	56,500	67,500	68,500	--

\* T = Thickness; W = Width

† L = Longitudinal; LT = Long-Transverse; ST = Short-Transverse

‡ Offset equals 0.2 per cent.

# Producer B; all others from Producer A

\*\* Specimens and fixtures cleaned ultrasonically in Toston 3 solvent.

†† Offset equals 2 per cent of pin diameter.

††† Sub-size sheet-type specimen; 1/4-in. wide; 1-in. gage length.

TABLE VI

MECHANICAL PROPERTIES OF STRESS-RELIEVED STRECHED 7075-T6510 ALUMINUM ALLOY EXTRUSIONS  
[AFS3(615)-3580]

Section Thickness, in.	Sample Cross- Sectional Area, in. <sup>2</sup>	Loca- tion*	Direc- tion†	Tensile		Elongation in. or 4D, %	Shear Ultimate Stress,‡ psi	Comp. Yield Stress,‡ psi	Platwise		Ultimate Stress, psi 67D-1.5 67D-2.0	Bearing** psi 67D-1.5 67D-2.0
				Ultimate Stress,¶ psi	Tensile Yield Stress,¶ psi				Ultimate Stress,¶ psi 67D-1.5 67D-2.0	Yield Stress, psi 67D-1.5 67D-2.0		
0.065	0.18	317899	T/2	L	87 000	79 200	11.0	75 400	—	126 600	158 400	106 900
0.065	0.27	318033#	T/2	L	85 200	77 700	10.5	77 300	—	125 900	158 700	102 400
0.080	0.18	317858	T/2	L	83 000	74 500	12.0***	77 800	—	125 500	159 900	109 600
0.133	0.97	318029#	T/2	L	84 200	81 900	12.0	80 500	—	124 600	156 200	107 000
0.160	0.26	318030#	T/2	L	82 800	75 800	15.0***	76 600	—	126 500	160 100	108 600
0.260	1.2	318028#	T/2	L	87 500	79 400	12.0	82 800	—	126 100	158 000	104 800
0.323	0.51	317908	T/2	L	85 500	77 900	11.5***	73 800	48 500	126 100	158 000	123 400
0.275	2.2	317954	T/2	L	89 400	82 700	11.0	85 300	49 400	—	—	—
0.438	7.2	317859	T/2	L	83 500	80 900	11.0	79 200	48 000	125 000	160 600	104 400
0.453	1.9	318032#	T/2	L	87 600	80 300	13.0	82 100	47 100	—	—	—
0.935	7.2	340155	T/2, W/4	L	83 800	77 700	12.0	86 100	49 000	129 100	160 200	113 000
1.023	1.8	318033#	T/2, W/4	L	88 300	81 700	11.5	80 400	47 500	127 600	161 200	108 500
1.188	27.1	317860	T/2, W/4	L	82 900	76 100	12.0	81 100	46 100	129 400	157 900	107 600
1.500	1.8	317955	T/2	L	82 900	75 300	12.0	79 400	45 600	125 900	157 400	105 600
2.000	3.1	317861	T/4, W/4	L	86 600	82 500	9.0	84 200	47 100	121 000	156 100	112 000
2.190	17.0	318137#	T/4, W/4	L	84 300	79 800	11.3	79 500	51 100	126 900	159 400	108 900
3.040	13.8	318138#	T/4, W/4	L	85 100	78 300	12.0	81 000	47 200	125 300	155 600	107 400
				ST	83 800	78 000	12.0	80 000	47 500	128 700	159 900	110 800
				ST	93 600	74 500	13.5	88 000	48 200	122 400	156 300	102 300
				ST	77 600	67 500	9.2	86 400	49 200	127 300	164 500	108 400
				ST	93 400	86 700	11.0	84 900	47 700	120 700	155 700	102 300
				ST	91 200	84 100	10.0	75 100	46 300	120 600	151 400	105 200
				ST	79 100	68 700	8.6	83 800	76 300	124 600	145 500	111 700
				ST	75 000	69 300	10.0	72 400	45 200	119 900	149 100	106 900
				ST	81 700	73 300	10.0	70 000	44 500	117 900	149 100	105 800
				ST	70 500	61 500	9.0	73 000	43 100	120 300	152 200	103 800
				ST	70 500	61 500	3.9	72 000	—	—	—	—
				ST	86 700	79 100	10.0	80 700	46 200	120 100	152 000	104 500
				ST	76 000	67 000	7.0	73 900	45 400	118 500	149 300	101 800
				ST	84 400	76 700	10.5	77 800	43 700	118 300	151 400	101 800
				ST	75 400	66 700	8.2	72 800	43 600	—	—	—
				ST	73 400	63 200	6.0	70 400	—	—	—	—

\* T - Thickness; W - Width

† L - Longitudinal; LL - Long-Transverse; ST - Short-Transverse

‡ Offset equals 0.2 per cent.

§ Producer B; all others from Producer A.

\*\* Specimens and fixtures cleaned ultrasonically in Teson 3 solvent.

†† Offset equals 2 per cent of pin diameter.

‡‡ Sub-size sheet-type specimen; 1/8-in. wide; 1/2-in. gauge length.

TABLE VII  
MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7075-T73510 ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580]

Section Thickness, in. <sup>2</sup>	Cross- Sectional Area, <sup>2</sup> in. <sup>2</sup>	Sample Number	Loca- tion*	Dir- ection†	Tensile Ultimate Stress, psi	Tensile Yield Stress, psi	Elongation 2 in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Ultimate Flatwise		Bearing**		
										Offset psi e/D=1.5	Offset psi e/D=2.0			
0.080	0.18	317862	T/2	L	79	400	71 100	9.0	71 300	--	120 600	153 800	103 200	120 500
0.313	0.51	317909	T/2	L	76	100	66 200	11.5	69 400	44 200	--	--	--	--
0.375	2.2	317900	T/2	L	74	900	64 300	11.5	65 800	44 300	113 400	146 300	92 300	109 400
0.438	7.2	317910	T/2	L	72	600	62 300	11.0	66 600	42 800	--	--	--	--
0.935	7.2	340292	T/2,W/4	L	76	200	67 500	12.0	69 800	45 100	118 100	152 800	98 900	117 200
1.500	1.8	317956	T/2	L	79	200	70 300	12.5	71 800	44 600	118 800	152 600	97 600	120 000
2.000	3.1	317948	T/4	L	78	800	71 100	11.5	72 800	43 300	--	--	--	--
			T/2	L	77	800	70 300	12.0	69 800	42 900	--	--	--	--
			1/2	L	70	200	60 500	6.2	73 800	44 700	111 100	145 700	93 800	111 500
										--	--	--	--	--

\* T - Thickness; W - Width.

† L - Longitudinal; LT - Long-Transverse.

‡ Offset equals 0.2 per cent.

\*\* Specimens and Fixtures cleaned ultrasonically in Teson 3 solvent.

†† Offset equals 2 per cent of Pin diameter.

TABLE VIII  
MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7079-T5510 ALUMINUM ALLOY EXTRUSIONS  
[API33(615)-3580]

Sample	Section Cross- Sectional Area, in. <sup>2</sup>	Loca- tion*	Direc- tion†	Tensile Ultimate Stress, psi	Elongation in. 2 in. or 4D, %	Comp. Yield Stress,* psi	Shear Ultimate Stress, psi	Plastic		Bearing**			
								Ultimate Stress, psi	Yield Stress, psi	Ultimate Stress, psi	Yield Stress, psi		
0.161	0.72	340252	T/2	L	86 400	79 300	10.5	78 400	--	124 400	156 500	106 300	123 300
			LT	LT	84 100	75 100	16.0***	84 100	--	--	--	--	--
0.251	0.82	340253	T/2	L	85 700	78 900	12.0	79 700	48 400	125 100	156 100	105 200	118 600
			LT	LT	81 0C0	72 600	14.1	79 600	47 500	--	--	--	--

\* T - Thickness; W - Width

† L - Longitudinal; LT - Long-Transverse

\* Offset equals 0.2 per cent.

\*\* Specimens and fixtures cleaned ultrasonically in Toston 3 solvent.

†† Offset equals 2 per cent of pin diameter.

\*\*\* Sub-size sheet-type specimen; 1/8-in. wide; 1/2-in. ~~wide~~ length.

**TABLE IX**  
**MECHANICAL PROPERTIES OF STRESS-RELIEVED STRETCHED 7178-T6510 ALUMINUM ALLOY EXTRUSIONS**  
[AF33(615)-3580]

Section Cross- Sectional Area, in. <sup>2</sup>	Sample Number	Loca- tion*	Direc- tion†	Tensile Ultimate Stress, psi	Tensile Yield Stress, psi	Elongation in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Flatwise			Bearing**		
									Ultimate Stress, psi	Yield Stress, psi	Bearing Width in.	Ultimate Stress, psi	Yield Stress, psi	Bearing Width in.
0.063	0.37	317902	T/2	95 400	89 900	9.5	90 400	--	133 500	168 900	117 400	138 900	--	--
0.142	1.0	318016	T/2	94 200	87 500	10.0***	93 800	--	131 600	166 700	114 900	137 300	--	--
0.154	0.42	318035#	T/2	93 300	86 900	10.5***	92 300	--	138 300	172 600	120 300	138 500	--	--
0.152	0.49	317903	T/2	92 200	85 800	9.5	90 800	--	132 600	164 400	114 300	129 800	--	--
0.265	0.88	317996	T/2, W/4	95 400	89 400	10.0	89 200	51 800	132 600	166 600	112 700	125 600	--	--
0.625	6.9	317997	T/2, W/4	93 600	86 900	10.7	88 200	51 800	133 400	164 800	114 300	134 900	--	--
0.780	1.7	340254	T/2, W/2	91 100	84 500	10.2	88 200	50 800	132 400	158 200	113 600	134 200	--	--
1.200	3.9	318139#	T/2, W/4	92 600	85 900	10.7	83 900	51 800	131 000	166 500	113 600	129 300	--	--
1.438	6.4	317957	T/2	89 400	80 700	11.4	87 500	50 200	137 200	171 500	110 800	133 100	--	--
2.180	15.5	318140#	T/4, w/4	93 200	73 900	7.5	86 400	49 500	133 100	167 600	115 700	137 100	--	--
			T/2, W/2	101 300	95 500	8.5	97 200	52 500	133 100	167 600	115 700	137 100	--	--
			ST	85 400	76 900	4.7	--	--	122 400	161 500	116 000	136 900	--	--
			ST	85 400	76 900	8.0	83 700	49 100	124 400	161 700	113 600	134 300	112 700	145 000
			ST	82 600	75 000	5.0	82 600	46 600	128 400	155 400	114 300	134 900	108 800	148 100
			ST	88 600	79 500	2.0	80 100	48 500	122 600	155 100	110 700	129 300	108 800	144 600
			ST	82 400	73 500	6.5	79 200	46 000	123 800	157 600	108 100	128 500	102 800	140 600
			ST	76 600	66 600	2.6	77 300	--	--	--	--	--	--	--

\* T - Thickness; W - Width.

† L - Longitudinal; LF - Long-Transverse; ST - Short-Transverse.

‡ Offset equals 0.2 per cent.

# Producer B; all others from Producer A.

\*\* Specimens and Figures cleaned ultrasonically in Toson 3 solvent.

†† Offset equals 2 per cent of pin diameter.

\*\*\* Sub-size sheet-type specimen; 1/8-in. wide; 1/2-in. gage length.

TABLE X  
MECHANICAL PROPERTIES OF EXCITATIONS IN THE "HEAT-TREATED-BY USER" TEMPER  
AF23(615)-3580

Alloy and Temper	Section thickness, in.	Sectional Area, in. <sup>2</sup>	Loca- Number	Tensile Stress, psi	Tensile Yield Stress, psi	Elongation in 2 in. or 4D, %	Comp. Yield Stress, psi	Shear Ultimate Stress, psi	Flatwise		Bearing**	
									Ultimate Stress, psi		Yield Stress, psi	
									e/D=1.5	e/D=2.0	e/D=1.5	e/D=2.0
2014-T62	0.185	1.0	340248§	T/2	L	68 500	62 800	11.5	—	—	109 200	141 200
	0.300	6.3	318084	T/2	LT	74 200	67 200	9.5	70 100	43 100	112 200	144 600
	0.499	1.4	318085	T/2	L	74 700	63 400	12.5	72 600	41 400	112 500	143 500
					LT	71 700	65 100	13.0	69 100	41 300	110 200	141 800
					L	70 300	63 800	10.2	67 300	—	—	—
2024-T42	0.064	0.27	318088§	T/2	L	65 000	44 500	20.5	44 800	—	103 800	123 100
	0.083	0.27	318086	T/2	LT	68 100	43 500	18.0	47 500	—	98 400	121 200
	0.430	2.1	340241§	T/2	L	67 100	42 400	21.0	44 900	—	100 800	125 000
	0.500	0.64	340243	T/2	LT	72 400	48 200	18.5	55 200	38 800	107 500	128 800
	2.562	6.4	340245	T/4,W/4	L	79 800	54 800	17.5	55 200	39 400	90 200	—
					LT	71 600	47 800	18.0	58 700	40 900	106 800	134 500
					L	82 100	59 500	15.5	50 000	37 100	91 800	104 300
					LT	69 700	47 400	14.5	57 100	39 100	104 300	130 300
					L	80 900	56 500	15.5	52 100	38 500	92 800	126 300
					LT	61 300	45 200	6.5	47 300	—	98 300	126 300
					ST	64 000	44 000	8.5	47 300	41 700	—	—
2024-T62	0.064	0.27	318059§	T/2	L	65 700	54 900	7.5	54 100	—	107 500	132 200
	0.083	0.27	318087	T/2	LT	68 500	55 200	11.0	57 900	—	106 900	102 500
	0.430	2.1	340242§	T/2	L	68 800	54 200	7.0	56 900	—	109 300	135 400
	0.500	0.64	340244	T/2	LT	68 300	57 700	9.0	57 500	—	94 100	111 700
	2.562	6.4	340246	T/4,W/4	L	67 800	57 700	12.0	61 400	38 700	81 700	99 300
					LT	71 600	59 200	11.0	40 000	102 800	82 300	100 000
					L	68 100	58 000	9.0	—	—	—	—
					LT	71 200	58 100	10.5	59 500	40 000	101 400	131 900
					L	66 100	55 300	7.0	58 400	—	96 900	127 400
					LT	67 800	53 800	10.5	56 200	38 200	99 400	120 200
					L	62 000	51 500	6.5	53 700	36 500	96 900	120 200
					LT	63 000	52 700	6.0	55 600	38 400	—	—

\* T - Thickness; W - Width

† L - Longitudinal; LT - Long-Transverse; ST - Short-Transverse

‡ Offset equals 0.2 per cent

§ Producer B; all others from Producer A

\*\*Specimens and fixtures cleaned ultrasonically in Toson 3 solvent

††Offset equals 2 per cent of pin diameter

TABLE X (Concluded)  
MECHANICAL PROPERTIES OF EXTRUSIONS IN THE "HEAT-TREATED-BY-USER" TEMPER  
AF3(615)-3530

Alloy and Temper	Section Cross- Sectional Area, in. <sup>2</sup>	Sample Number	Loca- tion*	Tensile Stress, psi	Tensile Yield Stress, psi	Elongation in 2 in. or 4D,	Comp. Yield Str., psi	Shear Ultimate Stress, psi	Platwise			Beadwise					
									Ultimate Stress, psi		Yield Stress, psi	Ultimate Stress, psi		Yield Stress, psi			
									370-1.5	370-2.0	370-2.0	370-1.5	370-2.0	370-2.0			
6061-T62	0.246	4.6	318090 T/2	L	48 000	42 800	15.5	45 100	80 200	104 000	69 400	80 500	--	--			
	1.625	3.9	418091 T/4, W/4	L	46 100	41 500	18.5	42 500	82 200	105 200	73 300	85 400	--	--			
	0.300	1.7	318092 T/2, W/2	L	45 000	39 400	14.0	42 200	75 900	93 000	67 500	77 700	--	--			
	1.225	21.2	318096 T/2, W/4	L	45 600	41 500	19.0	43 200	76 300	96 300	69 200	76 700	.74	.600			
	2.250	4.1	318100 T/4	L	80 700	70 500	11.0	74 900	--	118 000	152 200	96 900	118 600	--	--		
	7.178-T62	0.050	318095 T/2	L	79 200	70 000	13.0	76 700	--	--	--	--	--	--			
	0.126	0.17	318093 T/2	L	91 000	83 200	10.0	88 400	48 700	124 000	158 500	106 600	125 700	--	--		
	0.300	1.7	318097 T/2	L	84 800	76 000	9.4	85 900	47 600	124 900	153 200	110 200	124 200	--	--		
	1.225	21.2	318098 T/2, W/4	L	87 100	78 000	10.0	81 800	48 200	124 700	150 300	110 200	124 200	--	--		
	2.250	4.1	318101 T/4	L	84 600	72 400	7.5	79 300	46 500	124 000	156 800	107 900	125 000	121 800	144 900		
	7.178-T62	0.050	318095 T/2	L	80 000	71 400	6.5	78 500	47 200	123 200	151 100	107 300	123 200	127 600	127 200		
	0.126	0.17	318093 T/2	L	92 700	85 100	10.5	88 000	48 600	122 200	152 800	106 500	125 000	127 600	127 200		
	0.300	1.7	318097 T/2	L	89 300	81 900	9.5	83 600	47 100	120 300	149 100	103 800	122 200	125 000	127 200		
	1.225	21.2	318099 T/2, W/4	L	76 700	66 100	7.8	72 400	46 200	--	--	--	--	--	--		
	2.250	4.1	318101 T/4	L	71 300	61 300	10.5	67 100	--	105 800	139 100	27 000	106 800	--	--		
	7.178-T62	0.050	318095 T/2	L	70 500	58 200	11.0	67 800	--	--	--	--	--	--	--		
	0.126	0.17	318093 T/2	L	76 800	67 000	12.0	71 800	122 700	145 500	93 200	113 100	--	--	--	--	
	0.300	1.7	318097 T/2	L	74 300	64 300	11.7	68 300	42 200	113 100	148 000	98 900	119 300	97 500	138 800	115 000	
	1.225	21.2	318099 T/2, W/4	L	78 400	69 600	10.5	73 700	43 700	114 200	144 900	98 000	117 200	100 600	131 600	97 500	115 000
	2.250	4.1	318101 T/4	L	76 200	64 700	10.5	70 200	43 600	114 100	144 900	95 200	115 200	101 900	122 200	97 500	114 400
	7.178-T62	0.050	318095 T/2	L	79 800	72 700	11.5	73 800	45 700	118 800	150 600	96 000	118 000	115 200	145 200	96 500	116 500
	0.126	0.17	318093 T/2	L	75 100	67 300	11.0	72 200	42 100	119 900	142 400	93 800	110 900	--	--	--	--
	0.300	1.7	318097 T/2	L	76 800	59 900	8.6	63 400	42 900	--	--	--	--	--	--	--	
	1.225	21.2	318099 T/2, W/4	L	70 700	68 600	11.5	77 700	--	135 400	166 100	117 300	133 900	--	--	--	--
	2.250	4.1	318101 T/4	L	96 700	88 600	11.5	97 700	51 900	136 400	167 900	125 900	147 000	134 100	167 900	126 400	149 900
	7.178-T62	0.050	318095 T/2	L	96 400	85 700	7.5	98 200	50 300	141 500	164 800	120 800	149 900	--	--	--	--

\* T - Thickness; W - Width.

† L - Longitudinal; LT - Long-Transverse; ST - Short-Transverse.

‡ Offset equals 0.2 per cent.

§ Producer A; all others from Producer B.

\*\* Specimens and fixtures cleaned ultrasonically in Tosten 3 solvent.

†† Offset equals 2 per cent of pin diameter.

(Concluded)

TABLE X (Concluded)

TABLE XI

 SPECIFIED MINIMUM VALUES\* FOR ALUMINUM ALLOY EXTRUSIONS  
 [AF33(615)-3580]

Alloy and Temper	Thickness, in.	Area, sq. in.	Tensile			Federal Specification
			Ultimate Stress, psi	Yield Stress, † psi	Elongation 2 in. or 4D, %	
2014-T62	≤0.749	All	60 000	53 000	7	
-T6510	≤0.499	All	60 000	53 000	7	
	0.500-0.749	All	64 000	58 000	7	
	≥0.750	≤25	68 000	60 000	7	
2024-T3510, -T3511	≤0.249	All	57 000	42 000	12	
	0.250-0.749	All	60 000	44 000	12	
	0.750-1.499	All	65 000	46 000	10	
	≥1.500	≤25	70 000	52 000	10	
	≥1.500	≥25, ≤32	68 000	48 000	8	
-T42	≤0.749	All	57 000	38 000	12	
	≥1.500	≤25	57 000	38 000	10	
-T8510, -T8511	0.050-0.249	All	64 000	56 000	4	
	0.250-1.499	All	66 000	58 000	5	
	≥1.500	≤32	66 000	58 000	5	
-T62	≤0.749	-	--	--	-	
	≥1.500	-	--	--	-	
6061-T62*, -T6510	≤0.249	All	38 000	35 000	8**	
	≥0.250	All	38 000	35 000	10	
7075-T62‡, -T6510	≤0.249	All	78 000	70 000	7	
	0.250-0.499	All	81 000	73 000	7	
	0.500-2.999	All	81 000	72 000	7	
	3.000-4.499	≤20	81 000	71 000	7	
-T73X, †† -T73510	≤0.249	-	--	--	-	
	0.250-0.499	-	--	--	-	
	0.500-1.499	-	--	--	-	
	1.500-2.999	-	--	--	-	
7079-T62, * -T6510	≤0.249	≤20	75 000	67 000	7	
	0.250-0.499	≤20	77 000	68 000	7	
7178-T62‡	≤0.061	≤20	79 000**	73 000**	5	
	0.250-1.499	≤20	82 000**	74 000**	5	
-T6510	0.062-0.249	≤20	84 000**	76 000**	5	
	0.250-1.499	≤25	87 000**	78 000**	5	

\* All values are as shown in the Aluminum Association Booklet, "Standards for Aluminum Mill Products," 1967.

† Offset equals 0.2 per cent.

‡ In QQ-A-200/8b, 11b, 12b and 13, values for T6 temper apply also for extrusions heat treated and aged by user (T62 temper).

\*\* Lower than in Federal specifications.

†† "T73X" signifies T73-type temper for 7075 when heat treated and aged by user. Standard designation not yet assigned.

TABLE XII

RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES  
OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580]

Alloy and Temper	Sample Number	Cross-Sectional Area, in. <sup>2</sup>		Location*	Tensile Strength, T <sub>S</sub> (L)		Tensile Strength, T <sub>S</sub> (ST)		Tensile Strength, T <sub>S</sub> (LT)		Tensile Strength, T <sub>S</sub> (ST)		Tensile Strength, T <sub>S</sub> (LT)		Shear Strength, S <sub>T</sub> (L)		Shear Strength, S <sub>T</sub> (ST)		Shear Strength, S <sub>T</sub> (LT)	
		in.	in.		T <sub>S</sub> (L)	T <sub>S</sub> (ST)	T <sub>S</sub> (LT)	T <sub>S</sub> (ST)	T <sub>S</sub> (LT)	S <sub>T</sub> (L)	S <sub>T</sub> (ST)	S <sub>T</sub> (LT)	S <sub>T</sub> (ST)							
2014-T6510	0.061	0.30	317950	T/2	1.11	--	1.10	--	1.04	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.070	0.24	318017†	T/2	--	--	--	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.072	0.16	317951	T/2	--	--	1.06	1.03	1.01	--	1.03	--	1.03	--	1.03	--	1.03	--	1.03	
	0.246	0.45	318130†	T/2	--	--	1.06	1.01	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.250	3.7	340154	T/2	--	--	1.06	1.01	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.671	0.40	317994	T/2	--	--	1.06	1.01	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.625	0.50	317952	T/2	--	--	1.06	1.01	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.750	0.55	340291	T/2	--	--	0.92	0.89	0.89	--	0.89	--	0.89	--	0.89	--	0.89	--	0.89	
	1.657	2.2	318046	T/2	--	--	0.92	0.91	0.91	--	0.91	--	0.91	--	0.91	--	0.91	--	0.91	
	0.075	0.70	318132†	T/2	1.07	--	0.91	--	0.85	1.09	1.09	--	1.09	--	1.09	--	1.09	--	1.09	
2024-T3510	0.094	0.30	318017†	T/2	--	--	--	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.101	0.33	317885	T/2	--	--	--	--	0.92	--	0.92	--	0.92	--	0.92	--	0.92	--	0.92	
	0.106	0.31	317904†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.120	0.27	318018†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.151	0.82	317886	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.225	2.8	317942	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.258	4.2	318047	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.275	6.2	317943	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.510	10.1	317926	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.525	1.9	318020†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
2024-T8510	0.550	1.9	317856	T/2	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	1.01	
	0.612	5.8	317945	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	0.920	4.6	317944	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	1.150	5.6	318077†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	1.200	3.9	317946	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	1.450	7.3	318021†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	1.705	4.8	340223	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	2.520	8.8	318133†	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	4.000	24.0	340221	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
	2.760	29.6	318048	T/2	--	--	0.96	0.96	0.96	--	0.96	--	0.96	--	0.96	--	0.96	--	0.96	
2024-T8510	0.075	0.70	318022†	T/2	1.07	--	1.02	--	1.02	--	1.02	--	1.02	--	1.02	--	1.02	--	1.02	
	0.095	0.30	318134†	T/2	--	--	--	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	--	1.01	
	0.101	0.33	317887	T/2	--	--	--	--	1.05	--	1.05	--	1.05	--	1.05	--	1.05	--	1.05	
	0.106	0.31	317888	T/2	--	--	--	--	0.99	--	1.02	--	1.02	--	1.02	--	1.02	--	1.02	
	0.120	0.27	318023†	T/2	--	--	1.05	1.05	1.05	--	1.05	--	1.05	--	1.05	--	1.05	--	1.05	
	0.151	0.82	317889	T/2	--	--	0.99	0.99	0.99	--	0.99	--	0.99	--	0.99	--	0.99	--	0.99	
	0.255	2.8	318082	T/2	--	--	0.97	0.97	0.97	--	0.97	--	0.97	--	0.97	--	0.97	--	0.97	
	0.258	4.2	317891	T/2	--	--	0.97	0.97	0.97	--	0.97	--	0.97	--	0.97	--	0.97	--	0.97	
	0.375	6.2	317892	T/2	--	--	0.98	0.98	0.98	--	0.98	--	0.98	--	0.98	--	0.98	--	0.98	
	0.510	10.1	318024†	T/2	1.9	--	0.99	--	0.99	--	0.99	--	0.99	--	0.99	--	0.99	--	0.99	

\* T = Thickness; W = Width

† Producer B; all others from Producer A

‡ Samples were in the T3511 temper

§ Sample was in the T8511 temper

TABLE XII

TABLE XII  
(Continued)  
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES  
OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-5580]

Allloy and Temper	Sample			Location*	Tensile Properties						Compressive Properties						Shear Properties								
	Cross- Sectional Thickness, in.	Section Area, in. <sup>2</sup>	Number		TBS (LT)	TBS (ST)	TBS (UT)	TBS (SL)	TBS (SU)	TBS (ST)	TBS (LT)	TBS (UT)	TBS (SL)	TBS (SU)	TCS (LT)	TCS (ST)	TCS (UT)	TCS (SL)	TCS (SU)	TSS (LT)	TSS (ST)	TSS (UT)	TSS (SL)	TSS (SU)	
2024-T8510	0.550	1.9	317922	T/2, V/4	0.97	-	0.98	-	1.04	1.02	1.02	1.02	1.02	1.02	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.642	5.8	317844	T/2, V/4	0.95	-	0.96	-	1.08	1.04	1.04	1.04	1.04	1.04	0.57	0.57	-	-	-	0.56	0.56	-	-	-	
	0.950	4.6	317623	T/2, V/4	0.97	-	0.98	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	1.150	5.6	318078	T/2, V/4	0.97	-	0.98	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	1.200	3.9	317895	T/2, V/4	0.97	-	0.97	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	1.450	7.3	318025†	T/2, V/4	0.97	-	0.97	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	1.705	4.8	340169	T/4, V/4	0.97	-	0.98	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	2.760	29.6	318079	T/4, V/4	0.91	-	0.94	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	4.000	24.0	340225	T/4, V/4	0.97	0.95	1.00	0.96	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	6061-T6510	0.050	318136†	T/2	0.98	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
6061-T6	0.075	0.42	317857	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.090	0.59	318027†	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.125	0.61	317846	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.126	0.70	317847	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.250	0.36	317848	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.210	6.3	317905	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.315	5.8	317953	T/2	0.98	0.94	0.95	0.94	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.375	8.6	317927	T/2	0.97	-	0.95	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.918	7.7	318082	T/2	0.99	0.98	0.98	0.98	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	1.240	2.7	317906	T/2	0.99	0.98	0.98	0.98	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
1.960	4.4	317896	T/4, V/4	0.89	-	0.86	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-		
	3.000	15.0	340226	T/4, V/4	0.87	-	0.86	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	6.500	33.2	317897	T/4, V/4	0.91	-	0.86	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	7075-T6510	0.065	0.18	317889	T/2	0.97	-	0.96	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-
	0.065	0.27	318021†	T/2	0.97	-	0.96	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
	0.080	0.18	317856	T/2	0.98	-	0.96	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	
0.133	0.97	318029†	T/2	0.98	-	0.96	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-		
	0.133	0.97	318029†	T/2	0.98	-	0.96	-	1.05	1.05	1.05	1.05	1.05	1.05	0.56	0.56	-	-	-	0.55	0.55	-	-	-	

\* T - Thickness; V - Width  
† Producer B; all others from Producer A  
‡ Samples were in the T3511 temper  
§ Sample was in the T6511 temper

TABLE XII (Continued)

TABLE XII  
(Concluded)  
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES  
OF STRESS-RELIEVED STRETCHED ALUMINUM ALLOY EXTRUSIONS  
[AF33(65)-3580]

Alloy and Temper	Sample			Tensile Strength, in.²	Tensile Strength, lb/in.²	Tensile Strength, kg/cm²	Compressive Strength, lb/in.²	Compressive Strength, kg/cm²	Shear Strength, lb/in.²	Shear Strength, kg/cm²
	Cross- Sectional Thickness, in.	Number	Location*							
7075-T6510	0.160	0.26	318020†	T/2	0.98	—	—	—	—	—
	0.260	1.2	318028†	T/2	—	—	—	—	—	—
	0.313	0.51	317908	T/2	—	—	—	—	—	—
	0.375	2.2	317924	T/2	0.93	0.92	—	—	—	—
	0.438	7.2	317889†	T/2	0.94	0.93	—	—	—	—
	0.463	1.9	318022†	T/2	0.95	0.95	—	—	—	—
	0.935	7.2	340155†	T/2, W/2	0.95	0.95	—	—	—	—
	1.025	1.8	318033†	T/2, W/4	—	—	—	—	—	—
	1.188	27.1	317860	T/2, W/2	0.97	0.96	—	—	—	—
	1.500	1.8	317955	T/2, W/2	0.85	0.78	—	—	—	—
2.190	2.000	3.1	317861	T/2, W/2	—	—	—	—	—	—
	3.040	17.0	318137†	T/2, W/4	0.87	0.82	—	—	—	—
	3.040	13.8	318138†	T/2, W/4	0.86	0.91	—	—	—	—
	7075-T73510	0.080	0.18	317862	T/2	—	—	—	—	—
	0.313	0.51	317909	T/2	0.97	0.97	—	—	—	—
	0.375	2.2	317900	T/2	0.99	1.00	—	—	—	—
	0.438	7.2	317910	T/2	0.96	0.96	—	—	—	—
	0.935	7.2	340282	T/2, W/4	—	—	—	—	—	—
	1.500	1.8	317956	T/2, W/4	0.90	0.85	—	—	—	—
	2.000	3.1	317948	T/2, W/4	—	—	—	—	—	—
7079-T6510	0.161	0.72	340252	T/2	0.97	—	—	—	—	—
	0.251	0.82	340253	T/2	0.95	—	—	—	—	—
	0.653	0.37	317902	T/2	0.99	—	—	—	—	—
	0.142	1.0	318016	T/2	0.99	—	—	—	—	—
	0.154	0.42	318025†	T/2	—	—	—	—	—	—
7178-T6510	0.162	0.49	317903	T/2	—	—	—	—	—	—
	0.265	0.88	317996	T/2, W/4	0.97	0.97	—	—	—	—
	0.625	6.9	317997	T/2, W/2	0.97	0.97	—	—	—	—
	0.780	3.7	340254†	T/2, W/4	0.91	0.91	—	—	—	—
	1.200	6.4	318139†	T/2, W/2	0.94	0.94	—	—	—	—
2.180	1.432	6.4	317937	T/4, W/4	0.91	0.91	—	—	—	—
	15.5	318140†	T/2, W/2	0.93	0.86	0.90	0.84	1.00	0.99	0.51

\* T - Thickness; W - Width  
 † Producer B; all others from Producer A  
 ‡ Samples were in the T8511 temper  
 § Sample was in the T8511 temper

TABLE XII (Concluded)

TABLE XIII  
RATIOS AMONG THE TENSILE, COMPRESSIVE AND SHEAR PROPERTIES  
OF ALUMINUM ALLOY EXTRUSIONS IN THE "HEAT-TREATED-BY-USER" TEMPS  
[AF23(615)-3580]

Alloy and Temper	Sample				Tensile Properties						Compressive Properties						Shear Properties					
	Section Thickness, in.	Cross-sectional Area, in. <sup>2</sup>	Number	Location*	Tensile Strength, $\frac{T_{UT}}{T_{TS(L)}}$	Tensile Strength, $\frac{T_{TS(L)}}{T_{TS(T)}}$	Tensile Strength, $\frac{T_{TS(T)}}{T_{TS(S)}}$	Tensile Strength, $\frac{T_{TS(S)}}{T_{TS(L)}}$	Compressive Strength, $\frac{C_{TS(L)}}{T_{TS(L)}}$	Compressive Strength, $\frac{C_{TS(T)}}{T_{TS(L)}}$	Compressive Strength, $\frac{C_{TS(S)}}{T_{TS(L)}}$	Shear Strength, $\frac{S_{UT}}{T_{TS(L)}}$	Shear Strength, $\frac{S_{TS(L)}}{T_{TS(L)}}$	Shear Strength, $\frac{S_{TS(T)}}{T_{TS(L)}}$	Shear Strength, $\frac{S_{TS(S)}}{T_{TS(L)}}$							
2014-T62	0.300 0.499	6.3 1.4	318084 318085	T/2	1.01 0.98	--	0.98	--	1.04 1.03	1.08 1.03	--	0.58 0.65	0.56 --	--	--	--	--	--				
2021-T42	0.064 0.083 0.470 0.500 2.562	0.27 0.27 2.1 0.64 6.4	318088† 318086† 340242† 340243 340245	T/2 T/2 T/4, W/4 T/4, W/4 T/2, W/2	1.06 1.02 1.00 0.99 0.85	--	0.98 1.02 0.99 0.81 0.80	--	1.01 1.06 1.01 1.00 1.01	1.07 1.06 1.05 0.85 0.84	--	--	--	--	--	--	--					
2021-T62	0.064 0.083 0.430 0.500 2.562	0.27 0.27 2.1 0.64 6.4	318089† 318087† 340242† 340244 340246	T/2 T/2 T/4, W/4 T/4, W/4 T/2, W/2	1.06 1.03 0.99 0.95 0.93	--	1.01 0.98 0.98 0.96 0.96	--	0.99 1.03 1.04 1.02 1.04	1.05 1.03 1.04 1.01 1.03	--	--	--	--	--	--	--					
6061-T62	0.246 1.625	4.6 3.9	318090 313091	T/4, W/4 T/2, W/2	0.96 0.98	--	0.95 0.97	--	1.00 1.03	1.00 1.04	--	0.74 0.70	0.73 0.68	--	--	--	--	--				
7075-T62	0.062 0.126 0.300 1.225	0.74 0.17 1.7 21.2	318094† 318092† 318096† 318098†	T/2 T/2 T/2, W/4 T/2, W/4	0.98 0.95 0.95 0.95	--	0.99 0.93 0.92 0.95	--	1.06 1.05 1.05 1.05	1.09 1.03 1.02 1.03	--	--	--	--	--	--	--					
7075-T73X	0.062 0.126 0.300 1.225	0.74 0.17 1.7 21.2	318100 318100	T/2 T/2	0.86 0.86	--	0.81	--	1.05 1.05 1.05 1.05	1.06 1.03 1.02 1.03	--	--	--	--	--	--	--					
7178-T62	0.050 0.051 0.403	0.15 0.20 3.0	340247† 318102† 340249	T/2 T/2 T/2	--	--	--	--	1.06 1.07 1.06 1.05	1.06 1.02 1.03 1.02	--	--	--	--	--	--	--					
									1.09													

\* T - Thickness; W - Width

# Producer B; all others Producer A

TABLE XIII

**TABLE XIV**  
**RATIOS OF BRAKING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED  
 ALUMINUM ALLOY EXTRUSIONS**  
**[AF33(615)-350]**

Alloy and Temper	Sample Number	Gross- Section- Area, in. <sup>2</sup>	Loos- tion*	Flatwise				Endwise			
				TBS(L) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(T) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(W) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(H) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(L) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(T) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(W) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0	TBS(H) e/D=1.5 e/D=2.0 e/D=2.5 e/D=3.0
2014-T6210	0.061	0.20	317950	T/2	1.57	2.07	1.42	1.67	--	--	--
	0.070	0.24	318017†	T/2	1.61	2.14	1.43	1.68	--	--	--
	0.216	0.45	318130	T/2	1.72	2.18	1.58	1.78	--	--	--
	0.250	0.70	3180154	T/2	1.58	2.07	1.46	1.65	--	--	--
	0.271	0.70	317994	T/2	1.43	1.86	1.32	1.49	--	--	--
	0.270	0.70	317924	T/2	1.46	1.87	1.33	1.56	--	--	--
	1.057	2.2	318046	T/2	1.45	1.81	1.29	1.50	--	--	--
2024-T5510	0.075	0.70	318132†	T/2	1.47	1.81	1.48	1.78	--	--	--
	0.094	0.70	318019†	T/2	1.52	1.86	1.48	1.79	--	--	--
	0.101	0.53	317985	T/2	1.51	1.95	1.47	1.88	--	--	--
	0.101	0.52	317904†	T/2	1.54	1.96	1.44	1.89	--	--	--
	0.120	0.51	318018†	T/2	1.46	1.77	1.21	1.78	--	--	--
	0.151	0.82	317886†	T/2	1.46	1.90	1.23	1.90	--	--	--
	0.255	2.8	317942	T/2	1.25	1.56	1.23	1.59	--	--	--
	0.258	2.8	318047	T/2	1.27	1.60	1.26	1.60	--	--	--
	0.375	4.2	317943	T/2	1.67	1.67	1.28	1.74	--	--	--
	0.510	10.1	317926	T/2	1.23	1.26	1.23	1.26	--	--	--
	0.520	10.1	318020†	T/2	1.63	1.63	1.23	1.68	--	--	--
	0.550	11.9	317856	T/2	1.23	1.26	1.23	1.26	--	--	--
	0.682	5.8	317945	T/2	1.31	1.27	1.27	1.31	--	--	--
	0.910	4.6	317944	T/2	1.27	1.27	1.27	1.27	--	--	--
	1.150	5.6	318077	T/2	1.32	1.32	1.32	1.32	--	--	--
*											
1.200	3.9	317946	T/2	1.21	1.21	1.19	1.17	1.21	1.23	1.45	--
1.250	7.3	318021†	T/2	1.31	1.31	1.31	1.31	1.27	1.63	1.21	--
1.705	4.6	340213	T/2	1.34	1.34	1.34	1.34	1.26	1.60	1.19	--
2.520	8.8	318133†	T/2	1.32	1.32	1.32	1.32	1.24	1.55	1.27	--
4.000	24.0	340214	T/2	1.28	1.28	1.28	1.28	1.22	1.54	1.18	--
2.760	29.6	318048	T/2	1.22	1.22	1.22	1.22	1.20	1.51	1.25	--
*											

\* T - Thickness; W - Width

† Producer B; all others from Producer A

# Samples were in the T2511 temper

§ Sample was in the T6511 temper

\*\* Bearing specimen failed before reaching yield stress (2 per cent offset).

NOTE: L - Longitudinal; T - Transverse

TABLE XIV

TABLE XIV (Continued)  
RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED  
ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580]

Sample	Cross- Sec- tional Area, in. <sup>2</sup>	Section Thickness, in.	Loca- tion*	Flatwise						Edgewise						
				BUS(L) TUS(L)												
2024-T8510	0.075	0.70	318022+	T/2	1.52	1.95	1.45	1.68	1.71	1.75	1.71	1.75	1.71	1.75	1.71	1.75
	0.075	0.70	318134+	T/2	1.46	1.92	1.44	1.71	1.75	1.73	1.71	1.73	1.71	1.73	1.71	1.73
	0.101	0.23	317887	T/2	1.52	2.04	1.44	1.75	1.72	1.76	1.74	1.72	1.74	1.72	1.74	1.72
	0.106	0.31	317888	T/2	1.52	2.00	1.40	1.76	1.71	1.75	1.70	1.71	1.70	1.71	1.70	1.71
	0.120	0.27	318022+	T/2	1.52	2.05	1.48	1.78	1.72	1.77	1.71	1.72	1.70	1.71	1.70	1.71
	0.151	0.82	317889	T/2	1.52	2.09	1.41	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.255	2.8	317890	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.258	2.8	318062	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.375	0.62	317891	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.510	10.1	317892	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.525	0.25	318024+	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.550	1.9	317922	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.642	5.8	317894	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	0.950	5.6	317893	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	1.150	5.6	318078	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	1.200	3.9	317895	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	1.450	7.3	318025+	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	1.705	4.8	340169	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	2.760	29.6	318079	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
	4.000	24.0	340225	T/2	1.52	2.09	1.40	1.82	1.75	1.88	1.78	1.82	1.74	1.82	1.74	1.82
6061-T6510	0.050	0.42	318136+	T/2	1.76	1.98	1.72	1.72	1.72	1.72	1.72	1.72	1.72	1.72	1.72	1.72
	0.075	0.59	317857	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	0.090	0.61	317827+	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	0.125	0.63	317905	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	0.210	3.7	317923	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	0.375	8.6	317927	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	0.918	7.7	318082	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	1.240	2.7	317906	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	1.960	4.4	317907	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	3.000	15.0	340226	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77
	6.500	33.2	317897	T/2	1.76	2.02	1.67	1.69	1.72	1.77	1.76	1.77	1.76	1.77	1.76	1.77

TABLE XIV. (Continued)

\*  $T = \text{Tau} \times \text{slope} \times W = \text{Width}$

† Producer B; all others from Producer A  
Samples were in the 3511 tempers

Samples were in the T3511 tempers  
Sample was in the T8511 temper

Sample was in the T8511 temper

\*\* Bearing specimen failed before reaching yield stress (2 per cent offset).

NOTE: L - Longitudinal; LT - Long-Transverse

TABLE XIV (Continued)

TABLE XIV (Concluded)  
RATIOS OF BEARING TO TENSILE PROPERTIES OF STRESS-RELIEVED STRETCHED  
ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580]

Alloy and Temper	Sample Number	Section Thickness, in.	Cross-sectional Area, in. <sup>2</sup>	Location*	Flatwise								Edgewise								
					B/S (L)				B/S (T)				B/S (L)				B/S (T)				
					T/S (L)	T/S (T)	e/D=1.5 e/D=2.0	e/D=1.5 e/D=2.0	T/S (L)	T/S (T)	e/D=1.5 e/D=2.0	e/D=1.5 e/D=2.0	T/S (L)	T/S (T)	e/D=1.5 e/D=2.0	e/D=1.5 e/D=2.0	T/S (L)	T/S (T)	e/D=1.5 e/D=2.0	e/D=1.5 e/D=2.0	
7075-T6510	0.065	0.18	317899	T/2	1.46	1.82	1.35	1.35	1.55	1.62	1.34	1.34	1.55	1.62	1.34	1.34	1.55	1.62	1.34	1.34	
	0.065	0.27	317851†	E/2	1.42	1.86	1.42	1.42	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.080	0.18	317858	E/2	1.42	1.81	1.34	1.34	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.123	0.26	317897	E/2	1.42	1.83	1.42	1.42	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.160	0.26	317828†	E/2	1.47	1.85	1.45	1.45	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.260	1.25	317954	E/2	1.42	1.82	1.42	1.42	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.375	2.2	317859	E/2	1.47	1.82	1.47	1.47	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.438	7.2	317802†	E/2	1.45	1.85	1.45	1.45	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.463	7.2	340155	T/2, W/4	1.45	1.83	1.45	1.45	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	0.935	7.2	317803	T/2, W/4	1.46	1.84	1.46	1.46	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	1.023	1.8	317880	T/2, W/4	1.37	1.77	1.37	1.37	1.49	1.57	1.38	1.38	1.49	1.57	1.38	1.38	1.49	1.57	1.38	1.38	
	1.188	27.1	317885	T/2, W/4	1.47	1.83	1.47	1.47	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	1.52	1.60	1.41	1.41	
	1.500	1.8	317881	T/2, W/4	1.36	1.67	1.36	1.36	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	
	2.000	3.1	317895	T/2, W/4	1.36	1.76	1.36	1.36	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	
	2.190	17.0	317837†	T/4, W/4	1.32	1.71	1.32	1.32	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	
	3.040	13.8	318138†	T/4, W/4	1.32	1.75	1.32	1.32	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	1.46	1.54	1.38	1.38	
	7075-T73510	0.080	0.18	317862	E/2	1.52	1.94	1.45	1.45	1.69	1.74	1.54	1.54	1.69	1.74	1.54	1.54	1.69	1.74	1.54	1.54
	0.275	2.2	317900	E/2	1.53	1.88	1.47	1.47	1.60	1.65	1.50	1.50	1.60	1.65	1.50	1.50	1.60	1.65	1.50	1.50	
	0.438	7.2	317910	E/2	1.53	1.83	1.50	1.50	1.60	1.65	1.51	1.51	1.60	1.65	1.51	1.51	1.60	1.65	1.51	1.51	
	1.500	1.8	317956	E/2	1.39	1.85	1.41	1.41	1.51	1.56	1.31	1.31	1.51	1.56	1.31	1.31	1.51	1.56	1.31	1.31	
	2.000	3.1	317948	E/4	1.44	1.85	1.44	1.44	1.51	1.56	1.31	1.31	1.51	1.56	1.31	1.31	1.51	1.56	1.31	1.31	
	7079-T6510	0.161	0.72	340252	T/2	1.44	1.81	1.33	1.33	1.55	1.50	--	--	--	--	--	--	--	--	--	
	0.251	0.82	340253	T/2	1.46	1.82	1.33	1.33	1.55	1.50	--	--	--	--	--	--	--	--	--	--	
	7178-T6510	0.063	0.37	317902	T/2	1.40	1.77	1.31	1.31	1.55	1.60	1.31	1.31	1.55	1.60	1.31	1.31	1.55	1.60	1.31	1.31
	0.154	0.42	318016	T/2	1.41	1.79	1.32	1.32	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	
	0.162	0.42	318025†	T/2	1.44	1.87	1.34	1.34	1.56	1.61	1.34	1.34	1.56	1.61	1.34	1.34	1.56	1.61	1.34	1.34	
	0.265	0.88	317946	T/2	1.44	1.79	1.32	1.32	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	
	0.788	1.7	340254	T/2, W/4	1.39	1.75	1.39	1.39	1.56	1.61	1.35	1.35	1.56	1.61	1.35	1.35	1.56	1.61	1.35	1.35	
	0.625	6.9	317997	T/2, W/4	1.42	1.46	1.42	1.42	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	1.56	1.61	1.32	1.32	
	1.200	3.9	318139†	T/2, W/4	1.39	1.76	1.42	1.42	1.56	1.61	1.34	1.34	1.56	1.61	1.34	1.34	1.56	1.61	1.34	1.34	
	2.158	6.4	317997	T/2, W/4	1.31	1.77	1.31	1.31	1.56	1.61	1.31	1.31	1.56	1.61	1.31	1.31	1.56	1.61	1.31	1.31	
	2.180	15.5	318140†	T/2, W/2	1.36	1.77	1.31	1.31	1.56	1.61	1.31	1.31	1.56	1.61	1.31	1.31	1.56	1.61	1.31	1.31	

TABLE XIV (Concluded)

\* T - Thickness; W - Width  
† Producer B; all others from Producer A

# Samples were in the T8511 temper

\$ Sample was in the T8511 temper

\*\* Bearing specimen failed before reaching yield stress (2 per cent offset).

NOTE: L - Longitudinal; LT - Transverse

**TABLE XIV**  
 RATIOS OF HEATINGS TO TENSILE PROPERTIES OF ALUMINUM ALLOY EXTRUSIONS  
 IN THE "HEAT-TREATED-BY-USER" TEMPER  
 [ASTM (615)-3580]

Alloy and Temper	Section Thickness, in.	Sample Number	Tensile						Elongation					
			Gross- Sec- tional Area, in. <sup>2</sup>	Thickness, in.	Elong., % D-1.5 e/D-2.0 e/D-3.0									
2014-T62	0.185 0.200 0.499	1.0 6.3 1.4	310240† 310804 310805	T/2 T/2 T/2	1.59 1.51 1.54	2.06 1.95 1.98	1.50 1.45 1.43	1.78 1.71 1.69	1.52 1.44 1.41	1.93 1.74 1.69	1.72 1.72 1.72	— — —	— — —	
2024-T42	0.064 0.085 0.490 0.500 2.562	0.27 0.27 2.1 0.64 0.64	3108088† 3108086 340241† 340223 340245	T/2 T/2 T/2 T/2 T/2,W/2	1.60 1.49 1.29 1.25 1.29	1.89 1.85 1.73 1.64 1.61	1.68 1.65 1.60 1.59 1.54	1.95 2.11 1.80 1.75 1.70	— — — — —	— — — — —	— — — — —	— — — — —		
2024-T62	0.064 0.085 0.490 0.500 2.562	0.27 0.27 2.1 0.64 0.64	3118089† 3118087 340242† 340244 340246	T/2 T/2 T/2 T/2 T/2,V/2	1.61 1.62 1.43 1.43 1.41	2.01 2.02 1.87 1.85 1.84	1.58 1.62 1.72 1.69 1.65	1.87 2.00 1.72 1.69 1.65	— — — — —	— — — — —	— — — — —	— — — — —		
6061-T62	0.246 1.625	4.6 3.9	3118090 3118091	T/2 T/2,V/2	1.67 1.65	2.17 2.13	1.58 1.64	1.84 1.89	1.72 1.67	2.19 1.85	1.68 1.63	— —	— —	
7075-T62	0.063 0.300 1.225	0.24 1.7 21.2	3118094† 3118096 3118098†	T/2 T/2,V/2 T/2,V/2	1.46 1.36 1.43	1.89 1.74 1.76	1.37 1.31 1.41	1.68 1.54 1.58	— — —	— — —	— — —	— — —		
7075-T73K	0.063 0.300 1.225	0.24 1.7 21.2	3118095† 3118097 3118099†	T/2 T/2,V/2 T/2,V/2	1.50 1.47 1.44	1.65 1.69 1.69	1.32 1.39 1.47	1.74 1.69 1.67	— — —	— — —	— — —	— — —		
7178-T62	0.050 0.052 0.403	0.15 0.20 3.0	-340247 318102† 340249	T/2 T/2 T/2	1.40 1.44 1.37	1.72 1.74 1.71	1.51 1.64 1.52	1.32 1.40 1.30	— — —	— — —	— — —	— — —		

\* T - Thickness; W - Width

† Producer B; all others from Producer A

NOTE: L - Longitudinal; LT - Transverse

**TABLE XVI**  
**RATIOS AMONG THE MECHANICAL PROPERTIES AT DIFFERENT LOCATIONS**  
**[AF53(615)-3586]**

Alloy and Temper	Sample			Location*	Stress-Relieved $\frac{W}{2}, \frac{W}{4}$	Tensile Ultimate Stress	Tensile Yield Stress	Compressive Yield Stress	Shear Ultimate Stress	Bearing			
	Section Thickness, in.	Cross- Sectional Area, in. <sup>2</sup>	Number							1.00	0.98	0.97	0.97
2014-T6510	1.657	2.2	318146	L	W/2, W/4	0.99	1.00	1.03	0.98	--	--	--	--
2024-T5510	0.525	1.9	318020**	L <sup>t</sup>	W/2, W/4	1.03	1.02	1.07	1.01	0.99	1.01	0.97	0.98
1.150	5.6	318077	L <sup>t</sup>	W/2, W/4	0.97	0.96	0.94	0.99	0.99	0.98	0.99	0.98	
1.450	7.3	318021**	L <sup>t</sup>	W/2, W/4	1.01	0.97	0.97	0.97	0.99	0.98	0.97*	0.99*	
1.705	4.8	340213	L <sup>t</sup>	TW/2, TW/4	1.01	1.00	1.01	1.02	0.96	0.98	0.98	0.97	
2.520	8.8	318133**	L <sup>t</sup>	TW/2, TW/4	1.00	1.02	1.02	1.02	--	--	--	--	
4.000	24.0	340214	L <sup>t</sup>	TW/2, TW/4	1.00	1.02	1.02	1.02	--	--	--	--	
2.760	29.6	318048	L <sup>t</sup>	TW/2, TW/4	1.00	1.00	1.00	1.00	0.99	0.99	0.97*	0.99*	
					--	--	--	--	0.96*	0.97*	1.02*	0.99*	
					1.03	1.08	1.04	1.04	1.01	0.99	0.99	1.00	
2024-T8510	0.525	1.9	318024**	L	W/2, W/4	0.99	0.98	0.99	0.99	--	--	--	--
1.150	5.6	318078	L <sup>t</sup>	W/2, W/4	1.00	1.00	1.00	1.00	0.99	1.00	0.99	1.00	
1.450	7.3	318025	L <sup>t</sup>	W/2, W/4	1.00	1.00	1.00	1.00	0.99	1.02	0.98	1.01*	
1.705	4.8	340169	L <sup>t</sup>	TW/2, TW/4	1.01	1.02	1.00	1.00	--	0.99*	1.01*	1.01*	
2.760	29.6	318079	L <sup>t</sup>	TW/2, TW/4	1.00	1.03	1.00	1.00	0.98	1.01	1.02	1.03	
4.000	24.0	340225	L <sup>t</sup>	TW/2, TW/4	1.01	1.02	1.02	1.02	1.05	1.02	1.02	1.02	
					0.98	0.98	0.98	0.98	0.98	0.98	0.98	0.98	
6061-T6510	1.240 1.960 3.000	2.7 4.4 15.0	317907 317896 340226	L	W/2, W/4 W/2, W/4 TW/2, TW/4	1.01 1.01 0.98	1.01 1.01 0.98	0.98 1.02 0.97	0.98 1.00 0.98	0.99 0.99 0.99	1.00 0.99 0.98	0.97 0.99 0.97*	0.97 0.99 0.97*

\* L - Longitudinal; LT - Long-Transverse

† T - Thickness; W - Width

\* Edgewise bearing specimens; others - flatwise specimens

\*\* Producer B; all others from Producer A

†† Sample was in the T3511 temper

\*\* Sample was in the T8511 temper

TABLE XVI

TABLE XVI (Concluded)  
RATIOS AMONG THE MECHANICAL PROPERTIES AT DIFFERENT LOCATIONS  
[AF23(615)-2580]

Alloy and Temper	Sample				Tensile Ultimate Stress	Tensile Yield Stress	Compressive Yield Stress	Shear Ultimate Stress	Ultimate Stress $\sigma_{UTS}/\sigma_{Yield}$	Yield Stress $\sigma_{Yield}/\sigma_{UTS}$	Bearing
	Section Thickness, in.	Cross-sectional Area, in. <sup>2</sup>	Number	Direction* Location							
6061-T6510	6.500	35.2	317897	L IR	T/2,W/4 0.99 0.99	0.99 0.97	0.99 0.97	0.99 0.99	0.98 0.99	0.99 0.99	0.95
7075-T6510	0.935	7.2	340155	L IR	W/2,W/4 0.98 0.98	0.97 0.98	0.98 0.97	0.99 0.99	0.98 0.99	0.97 0.98	0.98
	1.188	27.1	317860	L IR	W/2,W/4 0.98 0.98	0.98 0.98	0.98 0.97	0.99 0.99	0.98 0.99	0.98 0.99	0.93
	2.000	2.1	317861	L IR	T/2,T/4 0.97	0.98 0.97	0.98 0.95	0.97 0.96	1.03 1.03	1.04 1.04	1.02 <sup>#</sup>
	2.190	17.0	318137**	L IR	T/2,T/4 0.97	0.98 0.97	0.98 0.95	0.97 0.96	0.95 0.96	0.95 0.96	0.93 <sup>#</sup>
	3.040	13.8	318138**	L IR	TW/2,TW/4 0.97	0.97 0.97	0.96 0.95	0.95 0.94	1.00 1.00	1.03 <sup>#</sup> 0.97 <sup>#</sup>	0.98 <sup>#</sup>
					0.99	1.00	0.99	--	0.99 <sup>#</sup>	0.98 <sup>#</sup>	0.97 <sup>#</sup>
7075-T73510	0.935	7.2	340292	L IR	W/2,W/4 0.98	0.98 0.98	0.98 0.98	0.97 0.98	0.98 0.99	0.98 0.99	--
	2.000	3.1	317948	L IR	T/2,T/4 0.99	0.98 0.98	0.98 0.98	0.99 0.98	1.01 1.01	0.98 1.01	1.02
7178-T6510	0.625	6.9	317997	L IR	W/2,W/4 0.98	0.99 0.98	0.99 0.98	0.98 0.98	1.00 1.01	1.01 1.01	0.96
	1.200	3.9	318139**	L IR	W/2,W/4 0.98	0.98 0.98	0.98 0.98	0.98 0.98	1.00 1.01	1.00 1.01	0.98
	2.180	15.5	318140**	L IR	TW/2,TW/4 0.97	0.95 0.95	0.95 0.95	0.95 0.95	0.99 0.99	0.99 0.99	0.99
					1.00	0.98	0.96	0.96	0.96 <sup>#</sup>	1.00 <sup>#</sup>	0.95 <sup>#</sup>
2024-T42	1.562	6.4	340245	L IR	External Size of the Treated bar - IR WT/2,WT/4 0.99	0.97 0.97	0.97 0.97	0.96 0.96	0.96 0.96	0.96 <sup>#</sup> 0.95 <sup>#</sup>	0.95 <sup>#</sup>
					0.88	0.96	0.95	--	0.94 <sup>#</sup> 0.94 <sup>#</sup>	0.97 <sup>#</sup> 0.97 <sup>#</sup>	0.95 <sup>#</sup>
2024-T62	1.562	6.4	340246	L IR	WT/2,WT/4 0.95	0.93 0.94	0.94 0.93	0.96 0.92	0.94 --	0.95 1.00 <sup>#</sup>	0.97 1.01 <sup>#</sup>
6061-T62	1.625	3.0	318091	L IR	WT/2,WT/4 1.00	1.01 1.01	1.01 1.02	1.02 1.02	1.02 1.02	1.02 1.02	0.95
7075-T62	1.225	21.2	318098	L IR	W/2,W/4 0.97	0.96 0.96	0.96 0.96	0.99 0.97	1.02 <sup>#</sup> 0.99	1.02 <sup>#</sup> 1.00 <sup>#</sup>	1.01 <sup>#</sup>
	2.250	4.1	318140	L IR	T/2,T/4 0.97	0.96 0.96	0.96 0.96	0.97 0.95	0.96 0.96	0.96 0.96	0.97
	2.225	21.2	318145	L IR	W/2,W/4 0.97	0.96 0.96	0.96 0.95	0.98 0.97	1.03 <sup>#</sup> 1.03 <sup>#</sup>	0.98 <sup>#</sup> 0.98 <sup>#</sup>	0.95 <sup>#</sup>
	2.250	4.1	318101	L IR	T/2,T/4 0.97	0.96 0.96	0.96 0.95	0.97 0.94	1.01 1.01	0.95 0.95	0.94

\* L - Longitudinal; IR - Long-Transverse  
 † T - Transverse; W - Width  
 # Edgewise bearing specimens; others - flatwise specimens

\*\* Producer B; all others from Producer A  
 §§ Sample was in the T3511 temper  
 # Sample was in the T8511 temper

TABLE XVI (Concluded)

TABLE XVI  
RATIOS OF BEARING PROPERTIES IN THE EDGewise DIRECTION TO THOSE IN THE FLATwise DIRECTION FOR ALUMINUM ALLOY EXTRUSIONS  
[AF23(615)-2580]

\*\* Producer B; all others from Producer A  
Because specimen failed test machine

<sup>††</sup> Bearing Specimen failed before reaching yield stress.

L - Thickness; W - Width; LT - Long-Transverse  
L - Longitudinal; WT - Wide Transverse  
Producer A: All others from  
Producer B: All others from  
Producer C: All others from  
Producer D: All others from  
Producer E: All others from  
Producer F: All others from  
Producer G: All others from  
Producer H: All others from  
Producer I: All others from  
Producer J: All others from  
Producer K: All others from  
Producer L: All others from  
Producer M: All others from  
Producer N: All others from  
Producer O: All others from  
Producer P: All others from  
Producer Q: All others from  
Producer R: All others from  
Producer S: All others from  
Producer T: All others from  
Producer U: All others from  
Producer V: All others from  
Producer W: All others from  
Producer X: All others from  
Producer Y: All others from  
Producer Z: All others from

Sample was in the T3511 temper bearing specimen failed before reaching yield stress (2 per cent offset).

TABLE XVII  
RESULTS OF FRACTURE TOUGHNESS TESTS OF SINGLE-EDGE-NOTCHED SPECIMENS  
OF ALUMINUM ALLOY EXTRUSIONS  
[AF33(615)-3580].

Alloy and Temper	Section Thickness, in.	Cross-Sectional Area, in. <sup>2</sup>	Sample Number	Specimen		Nature of pop-in, in.	Load, (P) Lb.	Original Crack Length (a), in.	Gross Stress ( $\sigma$ ), psi	# Net Stress ( $\sigma_n$ ), psi	K <sub>IC</sub> , in.-lb/in. <sup>2</sup>
				Direction* and Specimen No.	Type (See Fig. 1)						
2014-T6510	0.271	0.40	317894	T1	2	1.502	0.271	I	4,600	0.32	766
2024-T8510	0.255	2.8	317890	T1	2	1.502	0.271	S, S	5,100	0.52	700
				T2	2	1.502	0.219	I	4,950	0.51	777
				T1	1	0.937	0.196	S, S	5,100	0.51	62
				T2	1	0.939	0.200	P	2,060	0.32	64
0.510	10.1	317892	T1	3	2.250	0.500	P	10,375	0.78	454	
0.642	5.8	317894	T1	4	3.000	0.627	P	10,100	0.78	444	
1.450	7.3	318025**	T1	4	3.000	0.613	P	9,750	0.78	444	
2.760	29.6	318079	T1	1	0.937	0.374	S, M	25,400	1.02	717	
4.000	24.0	340225	T1	2	1.501	0.500	S, M	25,700	1.01	654	
			T2	2	1.501	0.495	S, M	27,300	1.01	717	
			T1	1	0.937	0.373	S, M	310	0.33	717	
			T2	1	0.935	0.377	S, M	3570	0.31	717	
			T1	1	0.217	P	S, M	1,620	0.34	717	
			T2	1	0.938	0.217	Specimen failed in fatigue	1,680	0.34	717	
			T1	4	3.000	1.010	I	8,300	0.47	717	
			T2	4	3.000	1.010	S, M	8,100	0.51	717	
			T1	1	0.937	0.373	S, M	8,500	0.47	717	
			T2	1	0.935	0.377	S, M	8,000	0.50	717	
			T1	1	0.937	0.374	Specimen failed in fatigue	3,125	0.37	717	
			T2	1	0.937	0.374	I	4,800	0.58	717	
			T1	2	1.501	0.500	I	4,875	0.58	717	
			T2	2	1.501	0.495	S, M	5,600	0.52	717	
			T1	1	1.501	0.495	S, M	5,625	0.52	717	

TABLE XVIII

TABLE XVIII (Concluded)  
RESULTS OF FRACTURE TOUGHNESS TESTS OF SINGLE-EDGE-HOTCHED SPECIMENS  
OF ALUMINUM ALLOY EXTRUSIONS

[AF33(615)-350]

Alloy and Temper	Sample		Specimen				Nature of pop-in Indication <sup>a</sup>	Original Crack Length( $a_0$ ), in.	Gross Stress( $\sigma_g$ ), psi	Net Stress ( $\sigma_n$ , psi) <sup>b</sup>	$\sigma_n/a_0$	K <sub>IC</sub> off, psi/in. <sup>c</sup>	$\sigma_{yc}/\sigma_n$	
	Section Thickness, in.	Cross-Sectional Area, in. <sup>2</sup>	Specimen No.	Direction and Specimen Type (See Fig. 1)	Width(W) in.	Thickness,(B) in.								
7075-T6510	0.438	7.2	317059	L1	4	3.000	0.390	P S	9 250 9 550	30 900 31 300	0.31 0.39	26 200 26 500	66 68	
				L2	4	3.000	0.401	I S	9 125 10 150	31 000 34 400	0.39 0.43	26 100 29 100	66 61	
				T1	3	2.250	0.390	P S	7 300 7 500	38 500 38 500	0.50 0.50	27 800 27 800	74 74	
				T2	3	2.250	0.390	P S	7 500 7 550	38 500 38 500	0.49 0.49	27 800 27 800	74 74	
				L1	4	3.000	1.010	I S	25 300 26 200	31 300 32 400	0.41 0.43	26 700 26 800	68 69	
	1.188	27.1	318137	L2	4	3.000	1.010	I S	25 400 27 200	31 400 33 700	0.41 0.44	26 800 26 700	74 79	
				T1	2	1.501	0.501	P S	7 450 7 500	38 200 38 400	0.55 0.56	25 100 25 300	51 52	
				T2	2	1.500	0.500	S, M	6 900	38 600	0.51 0.56	25 200 25 300	51 52	
				T1	1	0.939	0.375	I S, M	3 750 4 250	40 900 42 300	0.61 0.63	19 000 20 400	37 40	
				T2	1	0.939	0.376	I S	3 800 3 830	40 600 40 800	0.63 0.69	21 000 22 000	46 46	
7075-T73510	0.438	7.2	317910	L1	4	3.000	0.395	I S	12 750 12 750	42 100 42 100	0.62 0.62	36 100 36 100	125 125	
				L2	4	3.000	0.400	I S	12 650 12 250	41 800 40 500	0.62 0.60	35 800 34 600	123 115	
				T1	3	2.250	0.400	P S	9 600 9 600	42 300 42 300	0.63 0.63	31 400 31 400	95 95	
				T2	3	2.250	0.400	P S	9 750 9 750	42 800 43 800	0.65 0.65	32 500 32 500	101 101	
				L1	3	2.250	0.620	P S	9 300 9 300	6 700 6 700	0.29 0.29	18 100 18 100	31 31	
7178-T6510	0.625	6.9	317997	Specimen failed in fatigue				Specimen failed in fatigue				Specimen failed in fatigue		
				T1	2	1.500	0.374	P S	2 750 2 750	6 700 6 700	0.38 0.38	18 000 18 000	31 31	
				T2	2	1.500	0.374	P S	2 750 2 750	30 600 30 600	0.38 0.38	18 000 18 000	31 31	
				L1	4	3.000	1.010	I S	16 000 18 600	20 100 23 300	0.24 0.26	17 000 19 800	28 33	
				L2	4	3.000	1.010	I S	16 200 21 000	20 300 19 500	0.26 0.23	16 600 16 600	26 26	
	2.180	15.5	318140**	T1	2	1.500	0.498	S, M	5 800	7 800	0.41 0.41	18 500 18 500	33 33	
				T2	2	1.500	0.500	P S	6 000 6 300	8 000 8 400	0.41 0.43	18 500 19 500	33 36	
				L1	4	3.000	1.010	I S	16 000 18 600	20 100 23 300	0.24 0.26	17 000 19 800	28 33	
				L2	4	3.000	1.010	I S	16 200 21 000	20 300 19 500	0.26 0.23	16 600 16 600	26 26	
				T1	2	1.500	0.498	S, M	5 800	7 800	0.41 0.41	18 500 18 500	33 33	
2014-T62	0.300	6.3	318084	Extrusions in the "Heat-Treated-By-User" Temperature				Extrusions in the "Heat-Treated-By-User" Temperature				Extrusions in the "Heat-Treated-By-User" Temperature		
				L1	1	0.935	0.296	I S	1 750	13 700	55 600	0.83 0.83	26 900 26 900	68 68
				L2	1	0.938	0.296	P S	1 900 4 080	14 800 14 100	54 400	0.80 0.81	26 400 26 700	68 72
				T1	1	0.933	0.269	I S	3 750	13 300	52 100	0.88 0.88	25 300 25 300	60 60
				T2	1	0.933	0.268	P S	3 600 3 870	14 300 15 500	56 000 57 800	0.89 0.91	26 400 26 500	65 75
7075-T62	1.225	21.2	318098**	T1	2	1.499	0.500	I S	7 000	9 300	37 100	0.52 0.52	22 300 24 000	48 45
				T2	2	1.499	0.500	I S	7 100	10 000	39 700	0.50 0.53	22 600 23 500	49 53
(7075-T73X)	1.225	21.2	318099**	T1	2	1.499	0.501	I S	6 400	8 500	39 100	0.50 0.50	23 200 24 800	52 59
				T2	2	1.500	0.500	P S	6 700 8 125	8 900 10 100	40 600	0.63 0.63	24 400 24 800	57 59

L - Longitudinal; T - Transverse

Nature of Pop-in Indication: W - Maximum Load

P - Clear Pop-in, below maximum load without significant prior deviation from linearity

I - Initial significant deviation from linearity

S - 5 per cent secant offset

At start of fracture toughness test, i.e., after fatigue cracking

# P/MB

\*\* Combined (P/A + Mo/I) stress at tip of crack

$$\text{#} \quad K_{Ic} = K_{Ic}^2 (1 - \mu^2) = \frac{\sigma^2}{E} \left[ 7.59 \frac{1}{\mu} - 32 \left( \frac{1}{\mu} \right)^2 + 117 \left( \frac{1}{\mu} \right)^3 \right]; \text{ ref: ASTM STP 411.}$$

E = modulus of elasticity, psi

$K_{Ic}$  = plane-strain strain-energy release rate, in.-lb/in.<sup>2</sup>

$K_{Ic}$  = plane-strain stress-intensity factor, psi<sup>0.5</sup>/in.

$\mu$  = load, lb, at unstable crack growth

B = thickness, in.

W = gross width, in.

\*\* Producer B, all others are Producer A

$\mu$  = Poisson's ratio = 0.33 for aluminum alloys

$$\alpha = a_0 + \frac{K_{Ic}}{\sigma_{yc}} a_0^{0.5}$$

$a_0$  = actual crack length, in., plus plastic-zone correction factor, in.

$\sigma_{yc}$  = tensile yield stress, psi

TABLE XIX  
RESISTANCE TO STRESS-CORROSION CRACKING OF STRESS RELIEVED  
STRETCHED ALUMINUM ALLOY EXTENSIONS

Alloy	Section Thickness, in.	Number	Stress - 75% Yield Strength*						Short Transverse P/N Days++	
			Longitudinal Per Cent Long in Test			Transverse Tensile Strength***				
			P/N	Days++	Tensile Strength	P/N	Days++	Tensile Strength		
2014-T6510	0.250	340154	0/2 Not in Test	84	29	0/2 Not in Test	84	22	--	
2021-T3510	0.255	317942	0/2	84	18	0/2 F81, (OK 84)	84	25	2/2	
	0.510	317926	0/2	84	14	2/2 F-12, 26*	77	42	F-6, 6	
	0.950	317944**	0/2	84	—	2/2 F-12, 26*	—	—	F-6, 6	
	1.200	317946*	0/2	84	—	2/2 F-12, 26*	—	—	F-40 (1 OK 84)	
	2.760	318048	0/2	57	—	Not in Test	—	—	Not in Test	
2024-T8510	0.255	317890	0/2	84	5	0/2 Not in Test	84	8	--	
	0.510	317892	0/2	84	6	0/2 Not in Test	84	7	--	
	0.950	317894**	0/2	84	6	0/2 Not in Test	84	9	0/2	
	1.200	317895**	0/2	84	—	0/2 Not in Test	84	10	0/2	
	2.760	318025	0/2	84	—	0/2 Not in Test	84	—	0/2	
6061-T6510	0.375	317923	0/2	84	4	0/2 Not in Test	84	0	Not in Test	
	1.240	317927	0/2	84	0	0/2 Not in Test	84	100	--	
	1.960	317907	0/2	84	0	0/2 Not in Test	84	—	--	
	3.40226	317890	0/2	84	—	0/2 Not in Test	84	—	--	
7075-T6510	0.275	317954	0/2	84	3	1/2 Not in Test	84	7	Not in Test	
	0.438	317859	0/2	84	3	1/2 Not in Test	84	—	--	
	0.925	340155*	0/2	84	—	3/2 Not in Test	84	6	F-6, 6	
	1.188	317860**	0/2	84	—	1/2 Not in Test	84	—	F-4, 4	
	2.190	318137	0/2	57	—	2/2 F-8, (OK 57)	—	—	F-9, 9	
	3.040	318138	0/2	57	—	2/2 F-4, 4	—	—	—	
7075-T73510	0.375	317900	0/2	84	2	0/2 Not in Test	84	0	Not in Test	
	0.438	317910	0/2	84	—	0/2 Not in Test	84	3	Not in Test	
	0.935	340292	0/2	84	—	Not in Test	—	—	—	
7079-T6510	3.40253	317900	0/2	84	—	1/2 F81, (OK 84)	—	14	2/2	
7178-T6510	0.625	317997	0/2	84	9	2/2 F-7, 7	—	—	2/2	
	1.200	318139	0/2	57	—	2/2 F-4, 4	—	—	2/2	
	2.180	318140	0/2	57	—	—	—	—	—	

Notes: \* Specimens and test environment are described in the first Quarterly Report.

+ P/N denotes number of specimens failed over number exposed.

++ Tests in progress for periods shown, with maximum duration of 84 days.

+++ Results are average values for tension tests of specimens which did not fail by stress-curtion cracking.

\*\* Short transverse yield strengths determined by tests of duplicate 0.050" diameter tension specimens.

\$ The directionality of this section is being examined microscopically.

TABLE XIX (Continued)

**RESISTANCE TO STRESS-CORROSION CRACKING OF ALUMINUM ALLOY  
EXTRUSIONS - HEAT TREATED AND AGED BY USER\***  
(TESTS STILL IN PROGRESS)++

Alloy	Section Thickness, In.	Sample Number	Stress - 75% Yield Stress*					
			Longitudinal		Transverse		Short Transverse	
			F/N+	Days	F/N	Days	F/N	Days
2014-T62	0.300"	318084	1/2	F-24***, (OK57)	1/2	F-24 (OK 57)	---	---
2024-T42	0.430"	340241	Not in test		Not in test		---	---
		340245	Not in test		Not in test		Not in test	---
2024-T62	0.430"	340242	Not in test		Not in test		---	---
		340246	Not in test		Not in test		Not in test	---
6061-T62	0.246" 1.625"	318090 318091	0/2 0/2	57 57	0/2 0/2	57 57	0/2 0/2	55
7075-T6	0.350" 1.225"	318096 318091	0/2 0/2	57 57	0/2 0/2	57 57	2/2	---
7075-T73X	0.350" 1.225"	318097 318099	0/2 0/2	57 57	0/2 0/2	57 57	0/2	62
7178-T6		340249	Not in test		Not in test		---	---

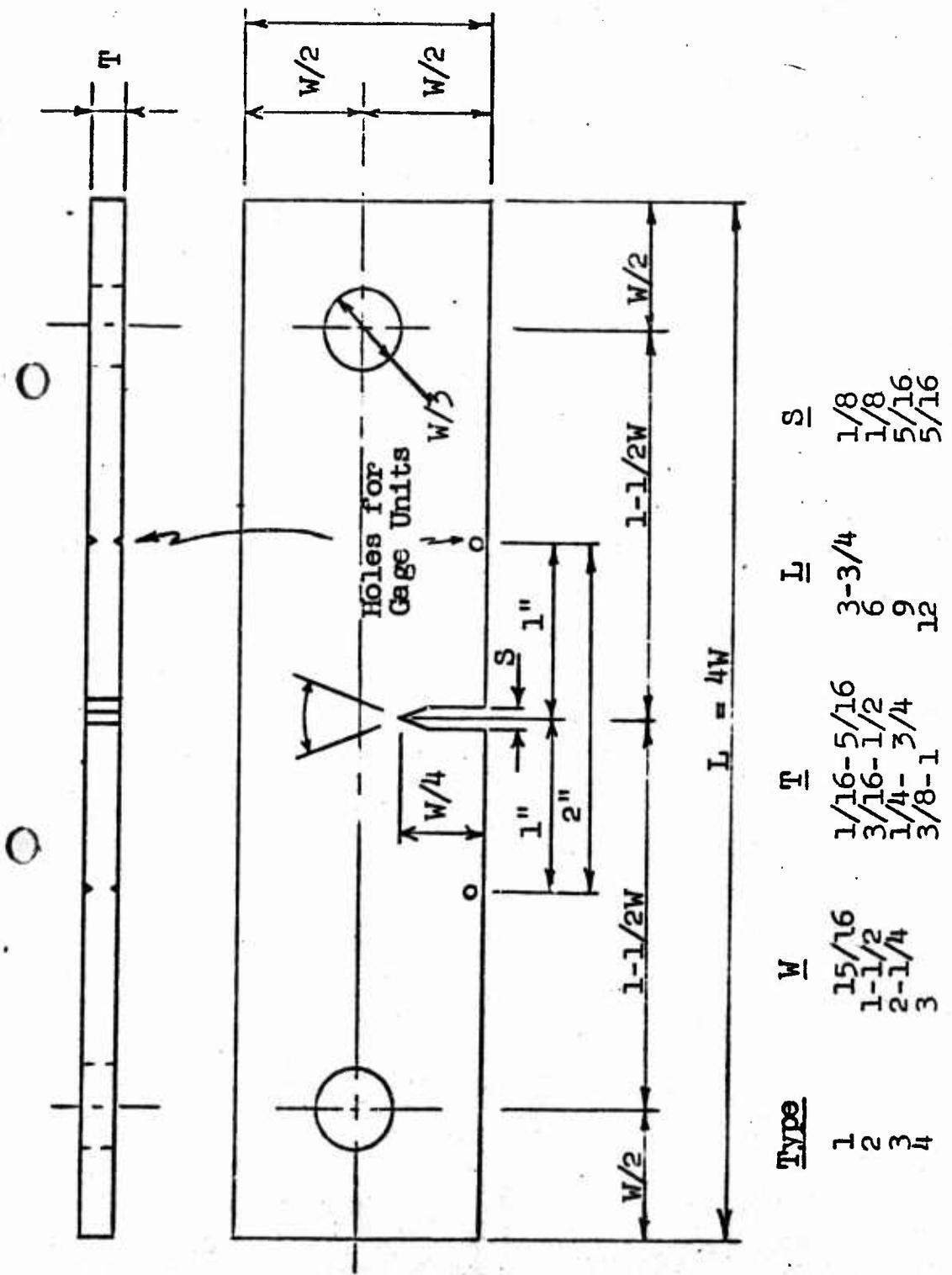
\* Sections submitted in the -O temper. Solution heat treated and aged at these Laboratories.  
\*\* Specimens and test environment are described in the first Quarterly Report.

+ F/N denotes number of specimens failed over number exposed.

++ Tests in progress for periods shown, with maximum duration of 84 days.

+++ Failed in shoulder, outside the gauge length.

TABLE XIX (Continued)



**Fig. 1** Single-Edge-Notched Fracture-Toughness Specimens.

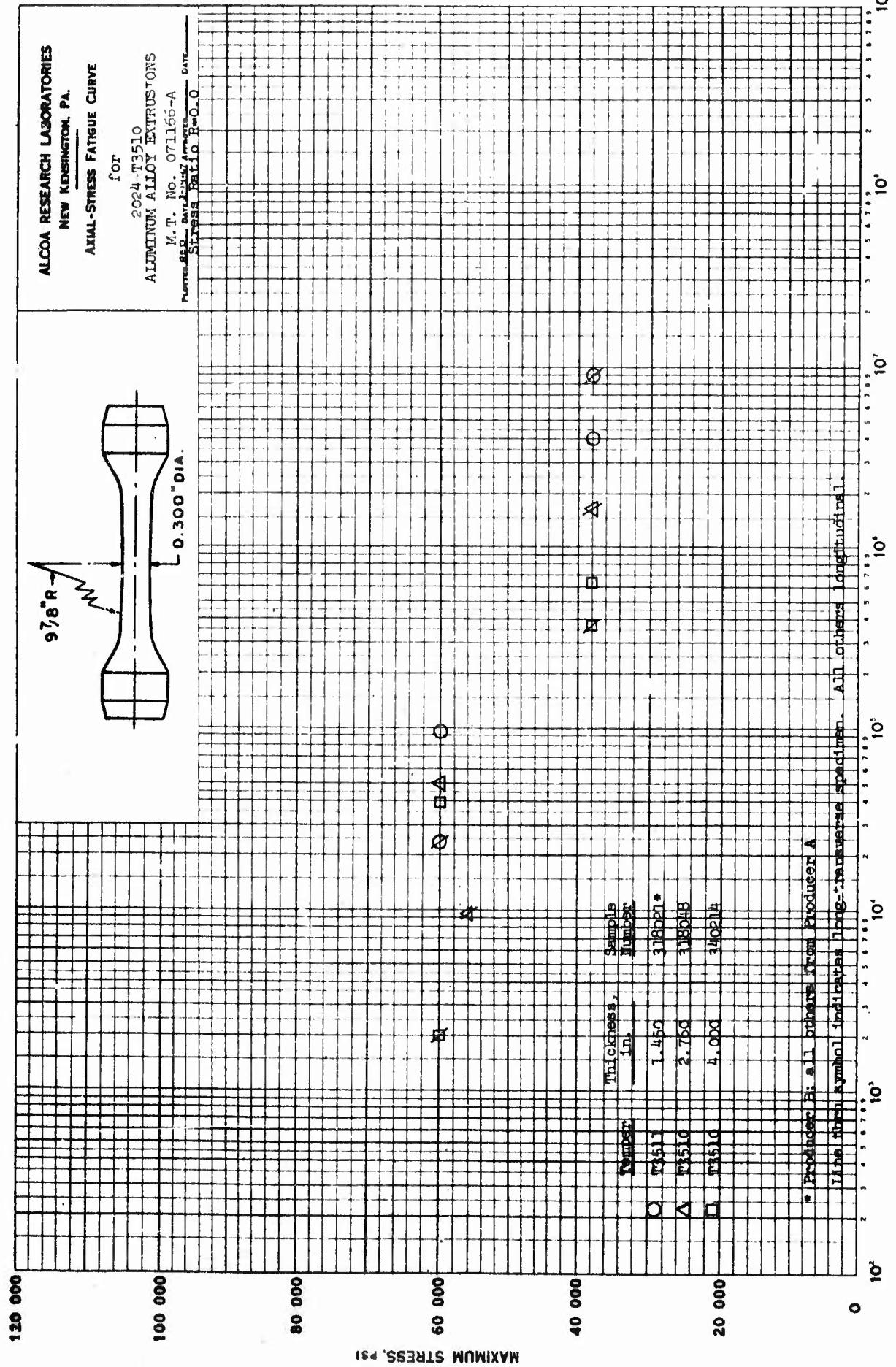


FIG. 2

120 000

ALCOA RESEARCH LABORATORIES

NEW KENNSINGTON, PA.  
ANAL-STRESS FATIGUE CURVE

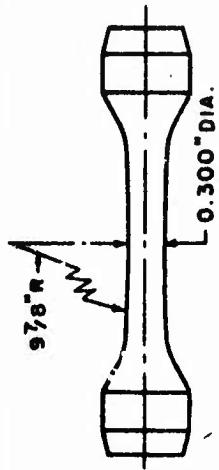
for

2024-T8510

ALUMINUM ALLOY EXTRUSIONS

M.T. No. 071166-A

Normal Stress Ratio R=0.0



MAXIMUM STRESS, PSI

40 000

60 000

100 000

80 000

120 000

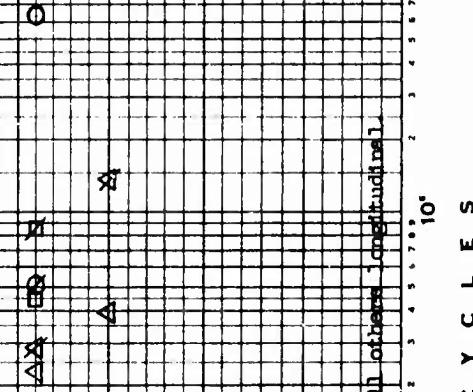
Thickness, in.  
Symbol  
Number

1.450  
118225\*

2.760  
318079

4.000  
342225

4.510  
342225

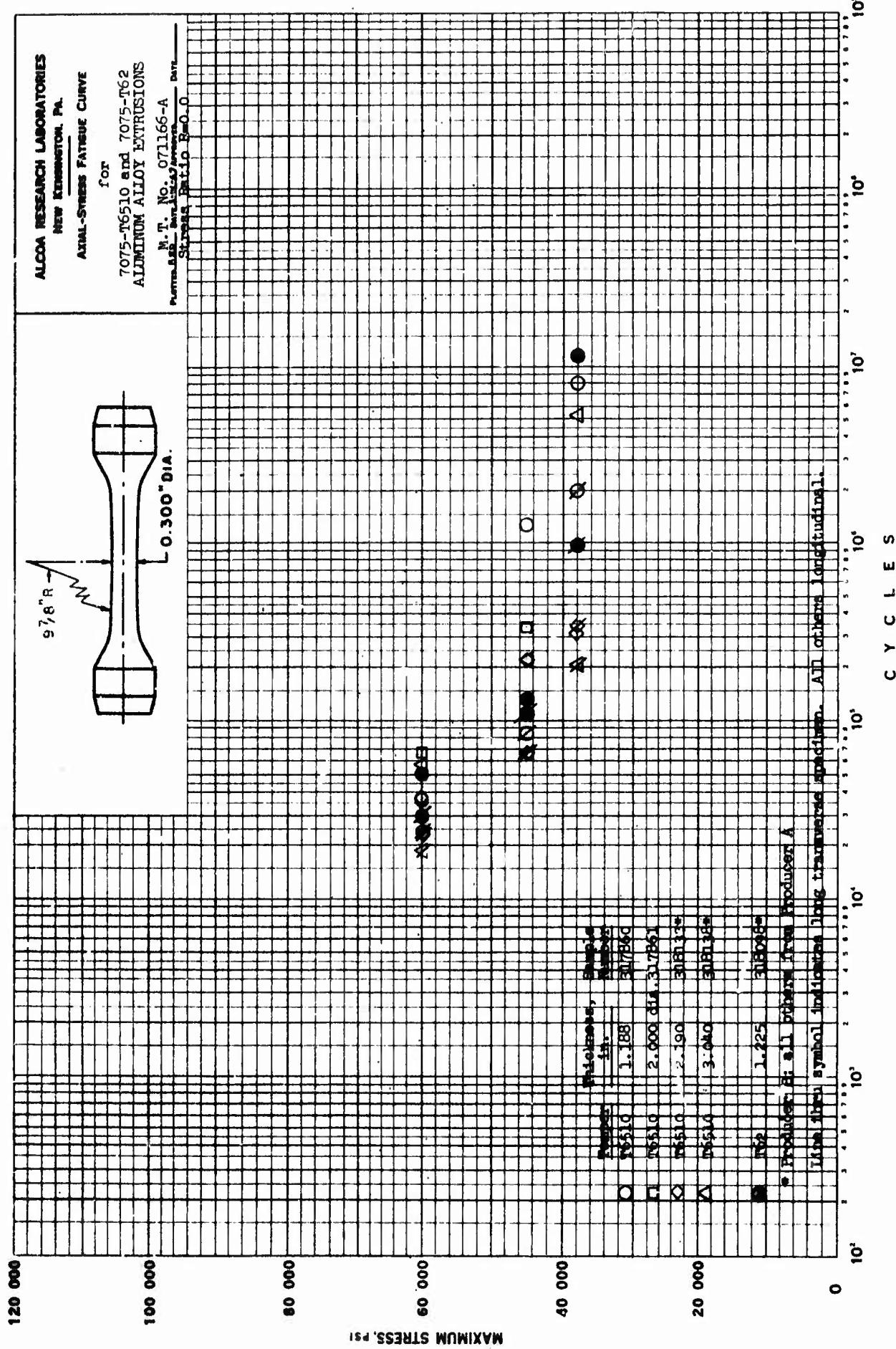


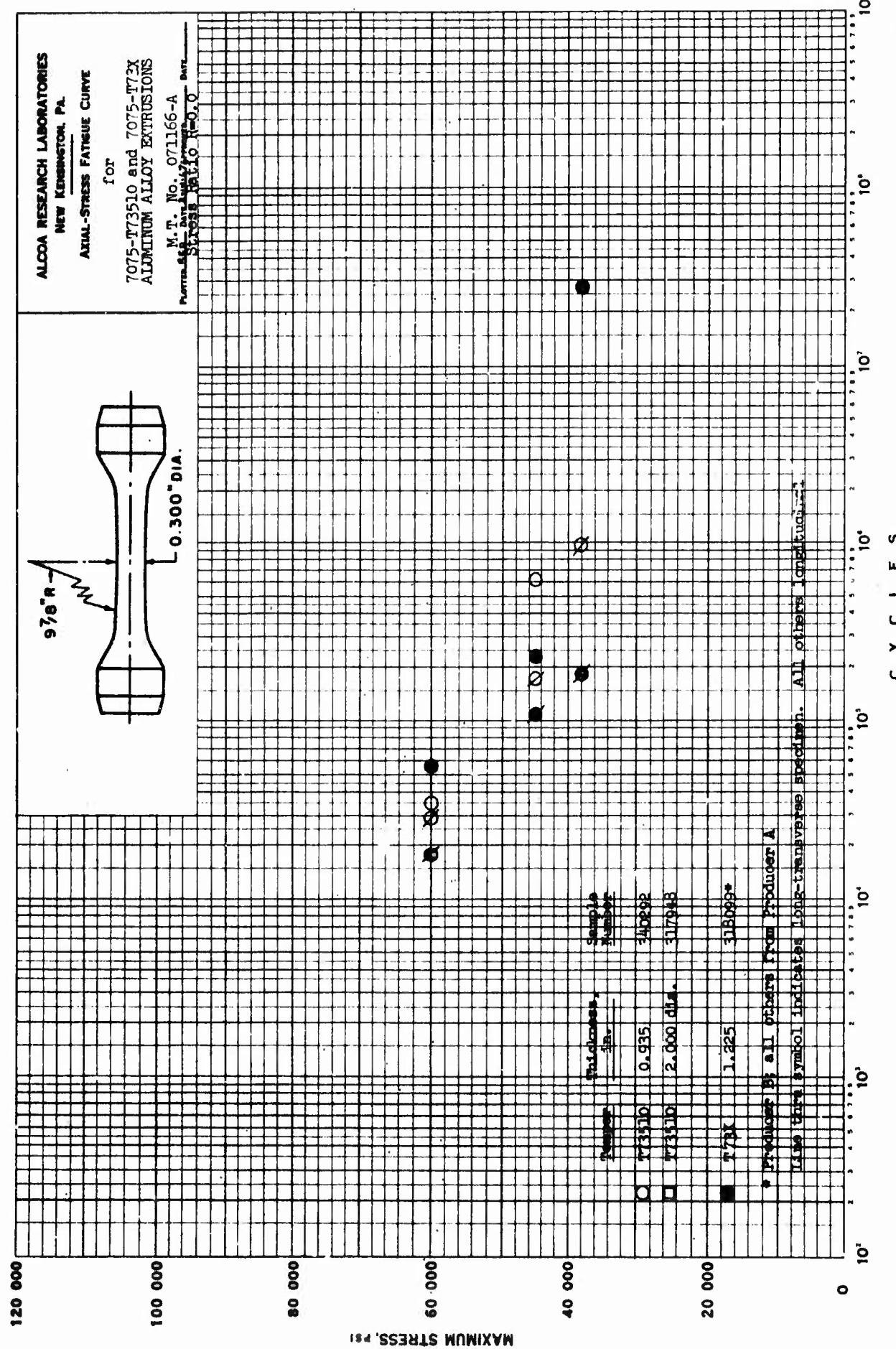
\* Producer B all stresses from Producer A

This graph symbol indicates longitudinal tension. All others longitudinal.

CYCLES

FIG. 3





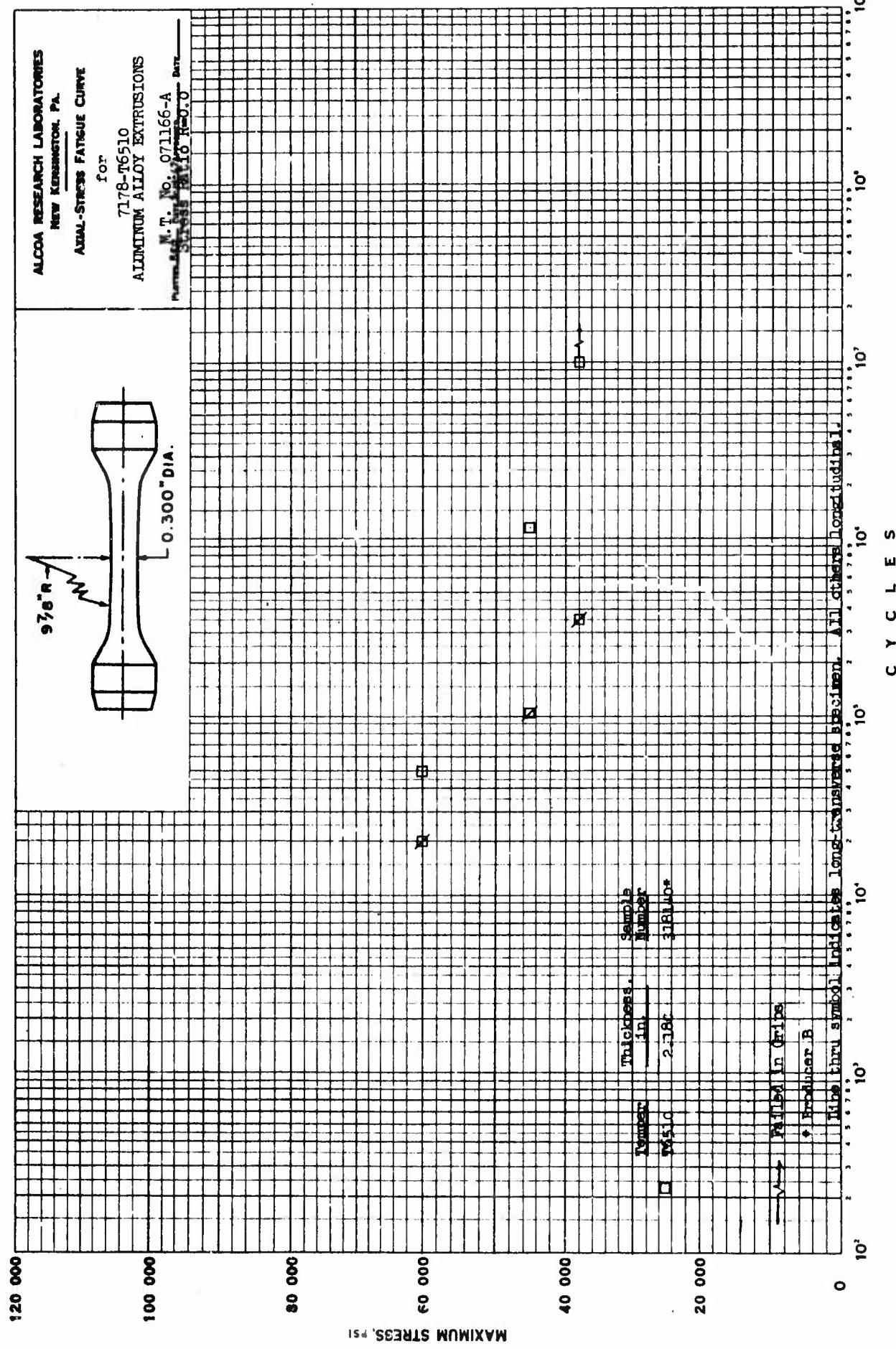


FIG. 6