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PERFORMANCE CHARACTERISTICS OF THREE PROPELLERS WITH VARYING PITCH DISTRIBUTIONS ON AN INCLINED SHAFT

James G. Peck

Naval Ship Research and Development Center

Prepared for:

Naval Sea Systems Command

August 1974

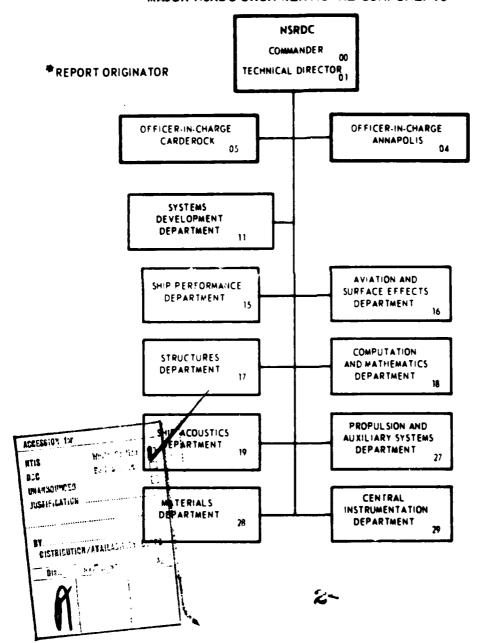
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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION	PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM				
1. REPORT NUMBER SPD-497-02	2. GOVT ACCESSION NO.					
4. TITLE (and Subtitle)		5. TYPE OF REPORT & PERIOD COVERED				
PERFORMANCE CHARACTERISTICS OF TH		FINAL				
WITH VARYING PITCH DISTRIBUTIONS SHAFT	ON AN INCLINED	6. PERFORMING ORG. REPORT NUMBER				
7. AUTHOR(a)		8. CONTRACT OR GRANT NUMBER(4)				
James G. Peck						
9. PERFORMING ORGANIZATION NAME AND ADDRESS	····	10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS				
Naval Ship Research and Developme Bethesda, Maryland 20034	PMS 300 Funding, Code 49501 W.U. 1532-242					
11. CONTROLLING OFFICE NAME AND ADDRESS		12. REPORT DATE				
Naval Sea Systems Command Washington, D. C. 20360		August 1974 13. NUMBER OF PAGES 33				
14. MONITORING AGENCY NAME & ADDRESS(II dilloren	t from Controlling Office)	15. SECURITY CLASS. (of this report)				
		UNCLASSIFIED				
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE				
16. DISTRIBUTION STATEMENT (of this Report)						
APPROVED FOR PUBLIC RELEASE: DI						
17. DISTRIBUTION STATEMENT (of the abetract entered	in Block 20, II different fro	n Report)				
19. SUPPLEMENTARY NOTES 19. KEY WORDS (Continue on reverse side if necessary on	d identify by Nach number)					
Cavitation Erosion N Propellers	oduced by IATIONAL TECHNICAL IFORMATION SERVIC S Department of Commerci	E				
	Springfield VA 22151					
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affecting the performance of the pr	opeller.					

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UNCLASSIFIED

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NOTATION

BF _{ang}	Bearing force angle, measured from the vertical BF_{ang} = arctan F_H/F_V [deg]
^C 0.7	Blade-section length at 0.7 radius [ft]
ח	Propeller diameter [ft]
F _H	Horizontal force, measured perpendicular to the shaft [lb]
$F_{\mathbf{v}}$	"Vertical" force, measured perpendicular to the shaft [lb]
F _{BF}	Bearing force $F_{BF} = \sqrt{F_H^2 + F_V^2}$ [1b]
J ·	Advance coefficient $J = V/nD$
K_{BF}	Bearing-force coefficient $K_{BF} = F_{BF} / \rho n^2 D^4$
K_L	Lift coefficient $K_L = L/\rho n^2 D^4$
K_{Q}	Torque coefficient $K_Q = Q/\rho n^2 D^5$
K _T	Thrust coefficient $K_T = T/\rho n^2 D^4$
L	Lift force $L = T' \sin \alpha + F_V \cos \alpha$ [lb]
n	Revolutions per second of unit time
p _{••}	Ambient static pressure [lb/ft ²]
p_v	Ambient vapor pressure [lb/ft²]
Q	Torque [ft-lb]
T .	Thrust in the horizontal plane $T = T' \cos \alpha - F_V \sin \alpha$ [lb]
T'	Thrust measured on the shaft [lb]
.v	Speed of advance [ft/sec]
α	Angle of shaft inclination [deg]
η	Efficiency $\eta = TV/2\pi Qn$
ρ	Mass density of water [lb - sec ² /ft ⁴]
σ	Cavitation number $\sigma = \frac{P_m - P_v}{1/2 \rho V^2}$

ABSTRACT

A series of three commercial propellers with sysyematic pitch reductions at the hub were characterized at zero and 15 degrees shaft angle, over a range of cavitation numbers and advance coefficients. In addition to the usual thrust and torque forces, horizontal and vertical side forces were measured. These experiments showed that varying the pitch distribution of these propellers changed the type of cavitation on the propeller without significantly affecting the performance of that propeller.

ADMINISTRATIVE INFORMATION

This work was funded by the Naval Sea Systems Command, sponsored by PMS 300 under Funding Code 49501, Work Unit 1532-242.

INTRODUCTION

One of the most persistent roblems which high-performance, small craft have encountered is propeller cavitation erosion in the blade root area. This problem was also observed on a series of constant pitch commercial propellers which were characterized on inclined shafts at the Naval Ship Research and Development Center (NSRDC). The Naval Sea Systems Command (NAVSEA) requested that NSRDC investigate the effect of reducing propeller pitch at the hub on propeller blade root cavitation.

Two propellers were purchased with specific pitch reductions at 0.2R of 80 percent and 90 percent of the pitch at 0.7R. The design

Peck, J. G. and D. II. Moore, "Inclined-Shaft Propeller Performance Characteristics," NSRDC Report 4127 (Apr 1974)

pitch distribution of these propellers and of a constant pitch propeller from a series characterized earlier are shown in Figure 1. The pitch of each blade of the three propellers was measured at several radial stations. An averaged value for the four blades of each propeller was calculated. These values are given in Table 1, and are presented as percent of pitch at 0.7 radius in Figure 1.

EXPERIMENTAL PROCEDURE AND FACILITIES

Propeller open-water characteristics were obtained in the deepwater towing basin using a propeller boat with zero degrees shaft inclination. All three propellers were characterized in open water over a range of advance coefficients (J) from zero velocity to zero thrust loading. Reynolds number for the open-water tests ranged from 6.8×10^5 to 7.4×10^5 .

Cavitation characteristics of the propellers were obtained in the 36-inch variable pressure water tunnel using the right-angle shaft dynamometer. Cavitation experiments were conducted with a strut and shaft system upstream of the propellers simulating a typical full-scale environment. Tunnel water velocities for each propeller were established by setting thrust values at zero degrees shaft inclination equal to the thrust values obtained from the open-water characteristics tests at the same propeller advance coefficient. A water speed of 20 fps was used for all of the experiments. Reynolds number for the cavitation experiments ranged from 1.5×10^6 to 2.2×10^6 .

All three propellers were characterized at cavitation number values from 0.5 to 14.7 over a range of advance coefficients from zero thrust loading to maximum torque of the dynamometer. Characterizations were made at zero and 15 degrees shaft inclination. In addition to thrust and torque, forces perpendicular to the shaft in the vertical plane and in the horizontal plane were measured. Cavitation observations and sketches were made throughout the entire program.

PRESENTATION OF DATA AND DISCUSSION

The open-water characteristics data of the propellers were reduced to the usual nondimensional coefficients of thrust and torque. The characteristic curves of these propellers are presented in Figure 2. These curves show that, except when lightly loaded, the propellers with radially varying pitch distribution increasing toward the tip produce more thrust at the same advance coefficient but are less efficient than the constant pitch propellers.

During the cavitation characteristic experiments, forces perpendicular to the shaft and thrust and torque forces in the shaft were measured. In order to present the results in the usual coordinate system, the thrust and side forces were resolved into horizontal and vertical components as shown in Figure 3. The resultant thrust and torque data from the cavitation experiments were reduced to the usual nondimensional coefficients, K_T and K_Q , for each propeller. The lift coefficient K_L , bearing force coefficient $K_{\rm BF}$, and bearing force angle ${\rm BF}_{\rm ang}$, were also computed from measured data. Efficiencies, K_T/J^2 and K_Q/J^3 were computed from the faired values of K_T and K_Q . All force coefficients are given in Tables 2 through 7.

The cavitation characteristics of the three propellers are shown in Figures 4 through 15. Figures 4 through 8 show the performance characteristics of each propeller for various cavitation numbers at zero shaft inclination and 15 degrees shaft inclination. Cavitation inception curves for the three propellers at 15 degrees shaft inclination are given in Figures 10 through 12. The cavitation inception curves represent the limiting values of sigma for cavitation at the section r/R under consideration; i.e., the area above the curves indicates there is no cavitation present at the given radius r/R. At a given

advance coefficient and cavitation number, the type and location of propeller cavitation will be determined by the curves above this point on the chart. These curves clearly demonstrate the effect of pitch reduction at the hub on face cavitation. The pitch reduction causes face cavitation from the hub to 0.3 radius, at much higher cavitation numbers than on the constant pitch propeller.

Figures 13 through 15 show propeller efficiency versus propeller loading for each propeller at various cavitation numbers. These curves indicate the propeller efficiency which might be expected for a given propeller loading at a specific cavitation number.

Photographs of the constant pitch propeller and one of the reduced pitch propeller are shown in Figure 16. These photographs show that, for the same operating conditions, the constant pitch propeller has back cavitation over a larger extent of the blades, whereas the reduced pitch propeller has heavier face cavitation from 0.45 radius to the hub. Figure 17 presents sketches of the cavitation present on the three propellers at 15 degrees shart inclination for two propeller loadings. These sketches illustrate the decrease in back cavitation and increase in face cavitation on a propeller with pitch reduction at the hub compared to a constant pitch propeller with the same pitch at 0.7 radius.

CONCLUSIONS AND RECOMMENDATIONS

These experiments show the changes in propeller cavitation which may be made by varying the radial pitch distribution of propellers having the same pitch at 0.7 radius. Reducing the pitch at the hub decreases back cavitation in that area but increases face cavitation.

The erosion damage caused by the different types of cavitation cannot be determined by the experiments. The performance of these propellers was not greatly affected by the pitch distribution. It is therefore recommended that constant pitch propellers continue to be used on high-performance small craft.

REFERENCES

1. Peck, J. G. and D. H. Moore, "Inclined-Shaft Propeller Performance Characteristics," NSRDC Report 4127 (Apr 1974)

Table 1
Pitch Distribution of Propeller Series

Propeller	4529	4615	4614
r/R	Averaged Measured Pitch Ins	Averaged Measured Pitch Ins	Averaged Measured Pitch Ins
0.2	- .	7.637	7.360
0.3	10.117	8.300	9.120
0.5	10.104	9.588	9.565
0.7	10.175	10.217	10.144
0.9	10.289	10.776	10.505

Table 2 - Performance Characteristics of Propsller 4529 at Zero Shaft Inclination

	INCLINATION	ANGLE =	U.000	PITCH RATIO =	1.000	SIGMA	= 14.700	
J	KTOUT	1000001	FF F L	KT/JS	KQ/J3	μĿ	KRF	Hr ANG
.6500	• 2226	.3756	.6132	•5264	.1368	.0012	.0074	50.5474
.7000	.199A	.3474	.6401	.4078	.1014	.6013	.0027	52.3387
.7500	.1774	.3192	.6634	.3154	.0757	.0013		55.0422
.8000	.155]	.2918	.6769	.2424	-0570	.0013		57.2665
.9500 .9000	•1326 •1093	.2342	.6785 .6684	•1835 •1349	.0431 .0321	.0015	•••	5.7 .4799
.9500	.0848	.1983	.461	.0939	-0231	.0017		58.8037 58.8050
1.0000	.0591	.1554	.6035	.0591	.0156	.001A		59.2896
1.0500	.0330	.1090	.5055	.0299	.0094	.0019		71.0952
1.1000	.0079	.0649	.2125	.0065	.0049	.0019		74.8843
	INCLINATION	ANGLE =	0.000	PITCH RATIO =	1.000	SIGMA	= 3.000	
J	KTOUT	10×0001	EFFIC	KT/JS	KQ/J3	KĽ	KRF	HF ANG
•6500	.2200	.3716	.6124	•5207	•1353	.0010	.0029	69.6555
-7000	•5050	.3457	•6511	•4123	-1008	.0007	.0030	75.7671
•7500 •8000	·1A1,	.3176	.6810	•3222	. 0753	.0005	.0032	AQ.7575
•R500	•1581 •1333	-2884	•6967	•247¢	• 0564	.0006	.0035	79.8802
.9000	•1333 •1076	.2584 .2241	.6979	-1845	•0421	.0011	.0039	74.4612
.9500	•0816	• 2271 • 1845	•68/8 •66()	-1329	-0307	.0016	.0044	69.9103
1.0000	•0557	1394	•6641 •6355	•0905 •0557	•0215	.0018	•005u	67.7328
1.0500	.0296	.0921	.5372	•0268	.0139 .0080	.0016 .0014	.0057	72.5410
1.1000	.0027	.0498	. 0935	•0055	•0037	.0019	.0065 .0071	79.0656 74.1673
	INCLINATION	ANGLE =	U.000	PITCH RATIO :		SIGMA		1441013
	KTOUT	1 GKQUUI	EFF1C	KT/J2	KQ/J3	*1	KBF	HF ANG
J •6500	•1562	.2864	.5640		.1043	۲۱. 0005ء	.0034	8].0644
•PD00	.365A	.2992	.6173		.0872	.0006	.0040	81.7123
.7500	.1635	2975	.6560	.2906	.0705	.0006	.0041	80.8255
.4000	.151A	.2814	.6857	.2372	.0551	.000A	.0041	77.4572
.8500	.1331	.2546	.7072		.0415	.0012	.0042	74.0192
.9000	.1095	.2184	.7169		.0300	.0015	.0046	70.5337
.9500	.0828	.177b	.7044		.0207	.0019	.0054	68.6866
1.0000	•0551	.1355	.6470		.0136	.0022	.0064	69.5798
1.0500	-02A2	.0954	.4949	•0256 •0037	.0082 .0045	.0024	.0074	71.4841
1.1000	•0045	.0603	.1310			.0027	.00A?	71.0915
	INCLINATION	ANGLE =	0.000	PITCH RATIO	1.000	SIGMA		
J.	KTOUT	100000	EFF1C		K0/J3	K.F	KRF	RFANG
•6500	.0754	-167n	.4672		.0608	0002	.0032 .0034	
.7000	.0841	.182A	.5127		.0533	0002	.0040	37.3538
.7500 .8000	.0917 .0963	.1959 .2071	.5586 .5917		.0464 .0405	.0002	.0046	87.7723
.4500	.0955	.2123	.6085		.0346	.0016	.0051	92.7012
.9000	.0970	.2048	.6082		.0281	.0021	.0058	71.1846
.9500	.0692	.1793	.5838	.0767	.0209	.0025	.0065	
1.0000	.0426	.1350	.5027	.0426	.0135	.0030	.0075	
1.0500	.0101	.07Ad	.2135		.0068	.003B	.0095	
1.1000	0221	.0290	-1.3333	0183	•0055	.0052	.0094	23.0141
	INCLINATION	ANGLE =	0.000	PITCH RATIO	= 1.000	SIGMA		
J	KTOUT	IOKONUT	EFF1C	KT/J2	K0/J3	kF	KBF	BFANG
.6500	.0392	.1008	.4021	.0927	.0367	.0009	.0031	A3.7098
.7000	.0491	.1182	.4630	.1003	.0345	0001	.0040	37.2028
.7500	.0554	.1300	.5089	.0985	.0308	0007		-33.8A17
.8000	.0604	.1450	.5298	•0943	.0283	0001	.0050	-46.7173
.8500	.0617	.1571	.5311	.0R54	.0256	.0007	.0058	1.7039
.9000	.0552	.1534	.5155	.0685	.0210	.0019	.0070	59.2888
.9500	.0375	.1233	.4597	.0416	.0144	.0035	.0085	103.5606
1.0000	.0086	.0675	.2040	.0086	.0057	.0050	.0097 .0099	76.3755 18.6409
1.0500	-,0251 0485	.0062 0113	-6.7184 7.5260	0228 0401	.0005 0008	.0059 .0052	.0079	55.0316
1.1000	-• ७ =त7	4113	7.2200	12<	-1440	*****		350.000
				TW -				

Table 3 - Performance Characteristics of Propeller 4615 at Zero Shaft Inclination

	INCLINATIO	N ANGIF =	0.000	PITCH MATIC	J = 1.000	SIGMA	= 14.700	
		I OKGOUT	EFF1C	KT/J2	KQ/J3	ıc.	KĄF	BFANG
J	KIOUT	•	E++ 10	_			-	
.6500	.2270	.3681	.6381	.5374		-0012		9.3566 3.0760
.7000	.2027 .1794	.33H0 .3086	.6681 .6939	.4136 .3189		.0010 .0011		3.2014
.7500 .9000	.1565	.2769	.7197	.2445		.0014		1.1506
.4500	.1330	.2423	.7426	-1841		.0018		9.1542
.9000	.1079	.2055	.7524	.1332	.0282	.0020	.0057	9.2358
.9500	. 1809	.1669	.7330	.0896		.0920		2.1917
1.0000	.0523	.1252	.6648	.0523		.0017		6.5710
1.0500	.023A	.0763	.5207	.0216	.0966	.001A .0071		7.6556 6.4400
1.1000	0016	.0111	2498	0013	.0009	•110 11	•0000	
	INCLINATION	N ANGLE =	0.000	PITCH RATIO	= 1.000	SIGMA	s 3.000	
J	KTOUT 1	OKQOUT	EFFIC	SLVTX	KU/J3	KI.	KHF	BF ANG
.6500	.2279	.3746	.6103	.5300	.1382	.0007	.0028	75.4496
.7000	.2039	.3514	.6456	.4162	.1026	.0000	.0079	74.1267
.7500	.1797	.3146	.6731	•3194	.0755	.0011	.0031	69.4587
.8090	.1546	.2A37	.6936	.2415	.0554	.0115	-0035	65.4841
.8500	•129 <u>5</u>	.7484	.7050	.1792	.0405	.001A	.0043	64.2182 65.9371
.9000	.1037	.21.H	.7010	.1280	.0291	.0019 .0019	.0047 .0057	69.6126
.9500	.0762	.1726 .129#	.6674 .5736	.0844 .0468	.0130	.0019	.0066	73.1461
1.0000	.0468 .0173	.0846	.3420	.0157	.0073	0020	.0073	73.7539
1.0500 1.1000	0073	.0410	3103	0060	.0031	.0025	.0064	68.2906
141000	INCLINATION	•	0.100	PITCH RATTO	= 1.000	SIGMA :	1.50 0	
J	KTOUT	1049001	EFFIC	K1/75	KQ/J3	KL.	KRF	AFANG
4500	.1676	.30AH	.5615	. 3968	.1125	.1005	.0033	A1.4845
.6500 .7000	.1719	.3181	.6022	3509	.0927	.0005	.0034	81.6400
.7500	.1661	.3170	.6375	.2953	.0737	.0007	.0033	77.6019
.8000	.1510	.2914	.6587	.2759	.0570	-0011	.0035	71.3416
. 4500	.12AA	.2638	.6604	.1782	.0429	.0017	.004U .004H	55.1661 52.1385
.9000	.1020	ARSS.	.6387	.1259	.0314	.0022 .0024	.0056	63.4995
.9500	.0730	.1880	.5868 .4790	.0A09 .0426	.0142	.0025	.0054	67.2883
1.0000	.0476	.1415 .0881	.1925	.0087	.0076	.0030	.0073	66.7631
1.0500 1.1000	.0096 0302	.0255	-2.0761	0250	.0019	.0851	.0086	48.M228
1.1000			0.000	PITCH PATTO	= 1.000	SIGMA :	.750	
	INCLINATION			-	KQ/J3		KSF	HFANG
J	KINUT	1042001	EFF1C	K1/JS		0005	.0014	-86.1424
.4500	.0999	.194H	.5301	.2363 .2071	.0710 .0591	0002	.0025	-12.0605
.7000	.1025	.2026	.5634 .5823	.1#01	.0492	.0005	.0032	40.4590
.7500	.1013	.2076 HP05.	.5959	.1534	.0410	.0014	.0041	49.6356
. 2000	.0982 .0915	.205u	.6041	.1267	.0334	.0023	.0050	77.5663
.9588 .9080	.0791	.1874	.5972	.0964	.0257	-0012	.0057	49.4148
.9500	.0551	.1518	.5443	.0611	.0177	.0043	.0067	34.996
1.0000	.0220	.0954	. 3654	.0220	.0096	•0054 •0066	.0046	44.3829 51.4588
1.0500	0178	.0271	-1.3404	0161 0450	.0019 0045	-0072	.0067	
1.1000	0545	0593	1.6097				****	
	INCL INATI	ON ANGLE		PITCH RAT			KBF	HF ANG
J	KTOUT	19KOOUT	EFFIC	KT/·IS	K6/J3	21.	.0026	-24.5531
.6500	.0598	.1420	.435P	-1416	.05)7	0000 0004	0025	-64.25R9
.7000	.0676	.1550	••	.1379	.0452 .0365	-,0002	.0029	-15.1510
.7500	.0667	.1539	. >132	.1177 .0905	.0291	.0007	.0034	47.5079
.A000	.0580	.14AH	.4960 .4310	.0410	.0225	.0018	.0055	87.4570
.8500	.0441	.1384 .1173	.3112	.0315	.0161	.0024	.0071	97.7087 84.8195
.9000	.0255 .0032	•0455	.0586	.0035	.0096	.0020	.00HZ	
.9590 1.0000	8209	.0391	4490	0209	.0039	.0004	.0078 .0051	
1.0580	0432	.0094	-7.2693	0392	.0009	001A 0072	0014	
1.1000	0594	.0394	-2.5935	04631	.3<**3*	-,011 12	••••	

Table 4 - Performance Characteristics of Propeller 4616 at Zero Shaft Inclination

	INCLINATI	ON ANGLE	= 0.000	PITCH RAT	TIO = 1.00	00 S1G*	4A = 14.70	0
J	KTOUT	10KGOUT	EFFIC	KT/J2	KQ/J3	KL	KBF	B+ ANG
.4500	.2239	.3668	•6316	•5300	1224	.0017	.002H	53.4108
.7000	.2006	3405	•6560	•4093	•1336 •0993	.0017	.0021	57 .5 566
.7500	-1781	.3132	.6790	.3167	.0742	.0017	.0035	60.7193
.9000	.1549	.2799	.7048	.2421	.0547	.001A	.0039	62.7819
.8500	.1305	.2423	.7286	.1806	.0394	.0019	.0043	63.9353
.9000	.1048	.2040	.7361	.1294	.0280	.0020	.0046	64.5143
•9500	.0793	.1640	.7051	.0A6A	.0196	.0022	.0052	64.8348
1.0000	.0511	.1324	.6117	.0511	.0133	.0025	.0060	65.0301
1.0500	.0274	.0897	.4179	.0203	.0077	.0129	.0059	64.8883
1.1000	0092	.0167	8661	0076	.0014	.0035	.0078	63.6885
	INCLINATIO	N ANGLE =	0.000	PITCH RATI	0 = 1.000	SIGMA	= 3.000	
J	KTOUT	10KQ0UT	EFFIC	K1\75	KO/J3	ĸĽ	KBF	BF ANG
.6500	.2741	.3792	.6113	.5304	.1361	.0008	.0025	70.3770
.7000	.2040	.3532	.6433	.4163	.1030	.000A	.0030	74.3377
.7500	.1803	. 3224	.6074	.3205	.0764	.0007	.0034	78.6387
.8000	.1555	.2897	.6832	.2429	. 0566	.0007	.003H	79.6128
.4500	.1301	. 2565	.6861	.1R00	.0418	.0009	.0040	77.5858
.9000	.1036	.2226	.6668	.1279	.0305	.0011	.0043	74.9620
.9500	.0755	.1870	.6103	.OR36	.0218	-0012	.0044	74.3102
1.0000	.0455	.1475	.4912	.0455	.0148	.0013	.0055	76.4501
1.0500	.0151	.1014	.2485	.0137	.00AB	.0014	.0063	78.5378
1.1000	0]24	.0453	4779	0102	.0034	.0017	.0070	72.1520
1	NCLINATION	ANGLE = 0	.000 P	ITCH RATIO	= 1.000	SIGMA =	•750	
J	KTOUT	IOKOOIIT	EFFIC	KT/J2	KO\J3	ĸĽ	KAF	BFANG
-6500	.0937	.1879	.5139	•5509	.0684	0005	.0031	*13.7666
.7000	•0965	.1969	.5444	-1964	•0574	0003	.0038	-80.0924
•7500	•0955	.2071	.5609	-1697	•0482	0001	.0045	-4.0161
.P000	.093A	.2067	.5777	•1465	.0404	•0002	.0051	59.4923
·8500	.0890	.2022	•5956	•1535	•0329	.0005	.0055	90.8262
.9000	•0770	.1829	.602A	.0950	.0251	-0010	.0060	91.1750
.9500	.0542	.1445	.5666	.0600	.0169	.0019	.0065	75.4572
1.0000	•0205	.0890	.3667	.0205	-00A9	• 0059	.0072	62.2532
1.1000	0177 0453	.02A8 0046	-1.0271 8.2429	0161	•0025	.003A	.0076	58.6377
111000		0070	004467	0374	0007	.0040	.0073	54.6130
IN	CLINATION AN			TCH RATIO =	1.000	SIGMA =	1.500	25.442
J	KTOUT	1 UK GOU I	EFFIC	KT/JS	KO/J3	KL	KRF	BFANG
.4500	.1670	.2973	.5A10	.3952	.10A3	0007	.0035	-69.9498
.7000	.1734	.3107	.6217	.3539	.0906	0008	.0044	-80.6324
.7500	.1676	.3069	.6518	.2979	.0727	0007	•005v	-95.881#
.8000	.1515	.2A65	.6731	-2367	.0560	0004	.0051	-79.697 0 -19.2312
.8500	.1279	.2541	.6809	.1770	.0414	.0002 .0013	.0055	64.2581
.9000	.0998	.2152	.6643	.1732	.0295		•0004	104.4582
.9500	.0697	.1744	.6045	.0772	.0203 .0133	.0028 .0042	.0093	-20.9486
1.0000	.0391	.1328	.4682	•0391 •0069	.0133	.0044	.0024	*01.2360
1.0500	.0076 0273	.0860 .0214	.1476 -2.2317	0226	.0016	.0017	0193	*03.5A37
.0100	INCLINATION			PITCH RATIO	= 1.000	SIGMA	500	
J	KTOUT	10KQ0UT	EFFIC	KT/JZ	KQ/J3	KL	KHF	BFANG
.4500	.0601	.1411	.4403	.1422	.0514	.0005	.0030	81.0092
.7000	.0654	.1523	.4784	.1335	.0444	.0004	.0032	A2.20A9
.7500	•0667	.1596	.4992	.1187	.0378	.0005	.0032	A1.3843
.0000	.063H	.1617	.5023	.0997	.0316	.0005	.0032	HO.2353
.4500	.0543	.1557	.4887	.0779	.0254	-0006	.0034	74.2654
.9000	.0427	.1373	.4456	.0527	.0188	.0008	•003H	78.2373
.9500	.0195	.1014	.2910	.0216	.0118	.0011	.0045	76.6286
1.0000	0205	.0422	7744	0205	.0042	.0015	.0053	74.0876
1.0500	0491	0460	3.2352	0808	0040	.0020	•0065	70.8892
1.1000	2038	16H7	2.1153	1684.1	A 20127	.0025	.0064	68.3903
	= · · · · ·	-	_	- No 1	42			

Table 5 - Performance Characteristics of Propeller 4529 at 15 Degrees Shaft Inclination

	THE THAT	ION ANGLE =	15.000	PITCH RAT	10 = 1.000	SIGNA	= 14,78	3
							KBF	BFANG
J	KTOUT	16K90UT	EFFIC	SLVTX	K9/J3	KL	אטר	9, 7,40
.6500	.2191	. 3855	.5879	-5185	-1404	.0809	.0227	-13.6048
.7000	-1977	.3595	.6129	·4036	.1048	.0770	.0241	-12.7635
.7500	.1775	.3349 .3103	.6327	•3156 •2466	.0794 .0606	•0737	.0251	-11.5613 -10.3901
.8000	•1578 •1379	.2841	.6477 .6565	.1908	.0463	.0708 .0680	.0275 .0310	-9.4135
.9500 .9000	.1168	.2553	.6555	.1442	.0350	.0651	.0339	-8.6528
.9500	.0940	.2230	.6376	-1042	.0260	.0619	.0356	-8.0720
1.0000	•0693	.1872	.5891	•0693	.0187	.0584	.0371	-7.6640
1.0500	.0429	.1484	.4828	.0389	.0128	.0546	.0436	-7.5361
1.1000	-0160	.1084	.2581	.0132	.0081	.0507	.0654	-7.9951
	INCLINATI	ON ANGLE =	15.000	PITCH RATI	0 = 1.000	SIGMA	= 3.000	
J	KTOUT	10K90UT	EFFIC	KT/J2	K0/J3	KL	KBF	BFANG
•6500	•2107	. 3884	.5612	.4986	.1414	.0795	.0225	-7.6808
.7000	.1970	. 3666	15987	.4020	.1969	.0774	.0241	-9.8460
.7500	.1786	.3408	.6254	•3175	.0808	.0740	.0257	-10.7821
.A000	•15A3	.3139	.6421	.2474	.0613	.0704	.0275	-10.7347
.A500	•1371	.2861	.6480	-1897	.0466	.0668	.029.	-10.0354
.9000	.1146	.2565	.6401	.1415	.0352	.0634	.0319	-9.0544
.9500	•0905	.2238	.6110	.1002	.0261	.0599	.0348	-8.1532
1.0000	.0644	.1876 .1492	.5465	.0644	.0188	.0561	.0379	-7.6373
1.1000	.0377 .0137	.1128	.4226 .2123	.0342 .0113	.0129 .0085	.0522	.0410 .0436	-7.7088 -8.4192
1.1000	•0137	•1160	*2123	••113	• • • • • • • • • • • • • • • • • • • •	• • • • • • • • • • • • • • • • • • • •	. 4430	-0.4172
	INCL INAT	ION ANGLE =	15.000	PITCH RAT	10 = 1.000	SIGMA	= 1.500)
J	KTOUT	10K90UT	EFFIC	KT/J2	KQ/J3	KL	KBF	BFANG
•6500	.1450	.2785	-5384	-3431	.1014	•053A	.0147	10.7725
.7000	•1525	.2886	.5889	•3113	.0841	.0608	.0193	5.2480
.7500	.1554	.2943	.6301	.2762	.0698	.0659	.0234	9895
.A000 .A500	.1507 .1381	•2872 •2666	.6681 .7009	•2355 •1912	.0561	.06A1 .0676	.0269	-5.7514
.9000	-1189	.2367	.7193	.1467	.0434 .0325	.0650	.0298 .0324	-8.3353 -9.0767
.9500	.0950	.2029	.7080	.1053	.0237	.0613	.0351	-8.9008
1.0000	.0690	.1687	.6510	.0690	.0169	.0574	.0381	-8.8743
1.0500	.0429	.1325	•5406	.0389	.0114	9537	.0414	-9.7574
1.1000	.0173	.0836	.3631	-0143	.0063	1499	.0449	-11.5552
	INCL INAT	ION ANGLE =		PITCH RAT				-
J	KTOUT	10KQOUT	EFFIC	KT/J2	KG/J3	KL	KBF	RFANG
-4500	.0759	.1888	.4158	•1796	.0637	.0256	.0054	17.6763
.7000	.0814	.2026	.4474	.1660	.0591	.0300	.0083	16.4611
.7500	.0866	.2144	.4820	.1539	.0508 .0436	.0348	.0116	15.3385
.8000	.0900 .0894	.2231 .2258	.5138 .5354	•1407 •1237	.0368	.0400 .0449	.0156 .0204	11.8499 5.9127
.8500 .90 0 0	•0823	.2193	.5376	.1016	.0301	.0486	.0257	-1.1174
.9500	.0675	.2013	.5069	.0748	.0235	.0501	.0312	-7.3837
1.0000	.0452	.1717	.4192	.0452	.0172	.0490	.0363	-11.4660
1.0500	.0185	.1339	.2304	.0168	.0116	.0457	.0406	-13.3180
1.1000	0064	.0962	1166	0053	.0072	.0420	.0437	-15.2050
	INCLINA	TION ANGLE	= 15.000	PITCH RA	TIO = 1.00	0 \$1GM		
J	KTOUT	1 0K 00UT	EFFIC	KT/J2	K9/J3	KL	KBF	BFANG
.6500	.0397	.1149	.3576	.0940	.0418	.0160	.0054	19.1620
.7000	.0443	.1255	.3928	.0903	.0366	.0172	.0053	14.8735
.7500	.0516	.1438	.4280	.0916	.0341 .0307	.0204	.0066	13.0623
.2000	.0562	.1572	.4549 .4655	.0878 .0771	.0264	.0240	.0088	11.7246 9.4117
.A500	.0557 .0499	.1619 .1589	.4498	.0616	.0218	.0274 .0307	.0168	5.3980
.9000 .9500	.0397	.1511	.3970	.0440	.0176	.0339	.0226	1502
1.0000	.0262	.1394	.2989	.0262	.0139	.0366	.0288	-6.0073
1.0500	.0098	.1195	.1377	.0089	.0103	.0369	.0335	-9.7192
1.1000	0105	.0784	2351	0087	.0059	.0317	.0332	-7.4358
				1.5	5 <			

15<

Table 6 - Performance Characteristics of Propeller 4615 at 15 Degrees Shaft Inclination

	INCLINA	TION ANGLE	= 15.000	PITCH F	RATIO = 1.0	00 SIG	MA = 14.7	00
J	KTOUT	LOKOOUT	EFFIC	KT/J2	KQ/J3	KL.		
4500	2222			-		~	KBF	BFANG
.6500 .7000	.2223	.3977	.5783	.5262	.1448	.0819	.0225	-1E 7000
	.199A	.3717	.5988	.4078	.1084	.0777	.0241	
•7500	-1790	. 3482	.6136	.3182	.0825	.0743	.0591	-14.5626
.4000	•15A3	.3251	.6257	.2473	. 9629	.0709	.0281	-13.0000
.8500	.1363	.2929	.6295	.1886	.0477	.0674	.0304	-11.5607
.9000	-1120	.2622	.6120	.1383	-0360	.0636	.0329	-10.4115
.9500	.GA53	.2313	.5575	.0945	.0270	.0593		-9.5514
1.0000	.0567	.1982	•4553	.0567	.0198	.0548	.0356	-8.9350
1.0500	1850°	.1556	.3016	.0255	.0134	.0504	.0387	-8.5965
1.1000	.0027	.0880	.0529	.0022	.0066	.0464	.0419	-8.7734
				•			.0448	-10.0298
	INCLINAT	TION ANGLE	= 15.000	PITCH R	ATIO = 1.0	00 SIGN	IA = 3.00	0.0
J	KTOUT	100000.7						•
3	K 1001	10KDOUT	EFFIC	KT/J2	KQ/J3	KL.	KBF	BFANG
.6500	2142				**	~~	NO.	BF ANG
•7000	.2163	.3900	•5737	.5119	.1420	.0A0a	.0222	-4 0/50
.7500	-2014	.3717	.6039	.4111	-1084	.0791	.0245	-4.8659
.A000	-1824	.3451	.6308	.3242	.0818	.0758	.0265	-8.7217
.A500	-1610	.3156	.6494	·2515	.0616	.0719	\$850.	-10.5810
•9000	-1380	.2851	.6548	-1910	.0464	•0676	.0301	-10.7729
	•1136	.2535	.6421	.1403	.0348	.0634		-9.9346
•9500	.0878	.2193	•605 <u>2</u>	.0972	•0256	•0590	.0325	-8.8145
1.0000	-0606	.1807	•5339	.0606	-0181	.0547	.0347	-6.0742
1.0500	•0333	.1370	-4062	.0302	.0116		.0375	-8.0912
1.1000	.00Al	.0889	.1591	.0067	•0067	•0505 •0469	.0407	-8.7612
	.				*****	• 0 • 10 •	.0437	-9.3008
	INCLINAT	TON ANGLE	= 15.000	PITCH R	ATIO = 1.00			_
J	KTOUT	10KQOUT			= '	00 \$1GH	A = 1.50	10
•		100001	EFFIC	KT/J2	KØ/J3	KL.	KBF	05.444
•6500	.1607	2002					N DF	RFANG
.7000	1657	.2983	.5574	.3804	-1086	-0604	.0172	12 4455
.7500	•1643	.3082	.5989	.3381	.0898	•0655		13.4653
.6000	•16 • 3	•3075	.6379	.2921	.0729	•0690	.0206	8.0264
.8500	•1399	-2945	.6726	.2431	.0575	.0701	.0241	1.6228
.900g		.2707	•6993	-1937	.0441	.0688	•0275	-3.8633
.9500	-1187	.2392	.7111	.1466	.0328	•0656	.0305	-7.6473
1.0000	•0937	.2028	•6986	-1038	.0237	.0614	.0331	~9.7678
1.0500	•0663	.1627	-6482	.0663	.0163	•0569	.0358	-10.8117
1.1000	•0369	.1163	•5302	•0335	.0100	•0522	.0386	-11.6402
147000	.0045	.0557	.1428	.0038	.0042	.0470	.0419	-13.1143
	74404 7440					.0470	.0462	-15.8200
	IMCETM	ITION ANGLE	= 15.000	PITCH F	RATIO = 1.0	000 SIG	MA = .7	50
J	KTOUT	10KQQUT	EEE10	×= 4 10			= •,	30
•		2 UN 400 !	EFFIC	KT/J2	KQ/J3	KL.	KRF	BF ANG
-6500	.0805	.1708	4000			_		01 4140
.7000	.0897	.1882	•4893	.1915	• 9622	.0278	.0063	19.4317
.7500	.0940	.1985	.5312	.1831	. 0549	.0341	.0100	12.7222
.8000	•0933	.2004	.5649	-1670	•0471	.0389	.0134	6.9270
.8500	.0873	.1928	.5929	.1458	•0391	.0422	.0166	2.1074
.9000	.0753	.1755	.6121	-1208	.0314	.0440	.0199	-2.1944
.9500	.0575	.1488	.6144	.0929	-0241	.0446	.0237	
1.0000	.0349		.5839	.0637	.0174	.0442	.0284	-6.5619
1.0500	.0102	.0734	.4874	.0349	.0114	.0427	.0335	-11.3102
1.1000	0122		.2320	•0092	•0063	.0398		-16.0917
		.0305	7089	0101	•0023	.0344		-19.5026 -18.6884
	THE THAT	ION ANGLE :		_			.0370	-10.0084
	INCC INA I	TOM MHOLE :	12.000	PITCH RA	TIO = 1.00	O SIGM	.50	D
J	KTOUT	10KOOUT	EFFIC	KT/J2	K9/J3	-		-
			21110	K1702	WAY 33	KL	KBF	BFANG
.6500	.0514	.1345	.3953	.1217	. 6466	****		
.7000	.0584	.1552	.4193	.1192	-0490	-0188	.0058	18.3419
.7500	•0633	.1725	.4378	.1125	.0452	.0227		10.8549
.8000	•0658	.1821	.4600	•1125	.0409	.0277	.009A	4.7211
.8500	.0622	.1617	.4630		•0356	.0314	.0132	.9603
.9000	.0487	.1714	.4968	•0601	•0296	.0328	.0163	8112
.9500	.0246	.1527	.2431	.0272	•0235	.0320	.0190	-1.6174
1.0000	0044	.1287	0541	0044	-0178	.0295		-2.3599
-0500	0226	.1031	3659	0205	.0129	•0258		-3.0555
1.1000	0014	.0801	0312			•0207		-2.0725
			.4316	0012		.0129	.0189	4.6314
				· • 4 c	• •			

Table 7 - Performance Characteristics of Propeller 4616 at 15 Degrees Shaft Inclination

	INCLINAT	ION ANGLE :	= 15.000	PITCH RATIO	= 1.000	SIGMA	= 14.70)
J	KTOUT	10KQ0UT	EFF1C	KT/J2	K 0/ J3	XI.	KBF	BFANG
-6500	.2189	.3884	.5830	•5181	.1414	.0810	.02 2	-13.5094
.7000	.1971	. 3639	.6035	•4023	-1061	.0772	.0241	-12.3259
•7500	.1774	.3429	.6176	•3154	.0813	.0740	.0260	-10.7703
.8000	•1572	.3191	.6272	•2456	•0623	.0708	.0280	-9.3467
-8500	•1353	.2907	.6294	•1872	.0473	.0672	.0302	-8.3020 -7.4055
.9000	•1113	.2583 .2234	•6173 •5797	•1374 •0949	•0354 •0261	.0633 .0592	.0326 .0353	-7.6955 -7.4679
.9500 1.0000	.0856 .0588	.1867	•5009	•0588	.0167	.0550	.0383	-7.5112
1.0500	.0309	.1465	.3531	.0281	.0127	.0509	.0416	-7.7373
1.1000	.0019	.0965	.0341	•0016	.0073	.0469	.0453	-8.1480
	INCLINAT	ION ANGLE :	= 15.000	PITCH RATIO	= 1.000	SIGMA	= 3.000)
J	KTOUT	10KQOUT	EFFIC	KT/J2	KQ/J3	1 01.	KBF	BFANG
•6500	•2159	.3776	•5916	•5111	1275	.0810	.0225	-4 484E
.7000	.1995	•3542	.6275	•4071	•1375 •1033	.0788	.0248	-6.6045 -9.6279
•7500	.1803	.3264	.6594	•3206	•0774	.0755		-10.5978
.A000	.1596	.2977	.6827	-2494	.0581	.0718		-10.2386
.8500	.1375	.2690	.6918	-1904	.0438	.0679	.0304	-9.2993
.9000	.1138	.2394	.6807	.1404	.0328	.0639	.0327	-8.4377
.9500	.0879	.2066	.6430	.0974	.0241	.059A	.0354	-8.1061
1.0000	.0598	.1670	•5698	• 0598	-0167	•0555	.0385	-8.4360
1.0500	.0302	.1160	•4349	.0274	.0100	·0508	.0418	-9.1231
1.1000	.0009	.0483	.0311	.0007	.0036	.0458	.0446	-9.3124
	INCLINATIO	ON ANGLE =		PITCH RATIO	= 1.000	SIGMA =	1.500	
J	KTOUT	10K90UT	EFFIC	KT/J2	KQ/J3	KL.	KBF	BFANG
.6500	.1596	.3110	.5308	•3777	.1132	.0597	.0167	13.4178
.7000	.1657	.3230	.5716	•3381	.0942	.0660	.0210	5.3333
.7500	.1654	.3253	.6070	•2941	.0771	.0702	.0251	-2.1391
.8000	•1562	.3153	.6308	.2440	.0616	.0710	.0284	-7.0964
.8500	.1385	.2930	•6396	.1918	.0477	.0689	.0310	-9.4678
.9000	-1149	.2610	.6307	•1419 •070	.0358 .0260	.0649	.0335	-10.2548 -10.7713
.9500	.0883	.2227	.5995 .5312	.0979 .0606	.0182	.0607 .0570	.0365 .0402	-11.8831
1.0000	.0606 .0316	.1817 .1405	.3764	.0287	.0121	.0530	.0441	-13.2488
1.1000	0027	.0997	0472	0022	.0075	.0458	.0464	-12.5591
111000			-	PITCH RATIO =		SIGMA =	.750	
	INCLINATION		EFFIC	KT/J2	KQ/J3	KL.	KBF	BFANG
J	KTOUT	10K00UT					.0074	16.3909
.6500	.0879	.2015	•4513	-2081	.0734	.0309	.0109	13.6581
.7000	•0939	.2174	.4814	.1917	.0634	.0361 .0411	.0144	9.7134
.7500	-0985	•2296	•5122 •5242	•1751 1560	• 0544 • 0465	.0454	.0181	4.1701
.8000 .8500	.0999 .0958	.2380 .2401	.5342 .5398	•1560 •1326	• 0 1 0 3 9 1	.0486	.0222	-2.1941
.9000	•0846	.2326	.5210	.1045	.0319	.0502	.0269	-8.1039
.9500	.0656	.2127	.4660	.0726	.0248	.0499	.0319	-12.4520
1.0000	.0396	,1796	.3510	.0396	.0180	.0475	.0369	-14.9654 -16.8716
1.0500	.0100	.1353	.1231	•0090	.0117	.0434	.0412	-21.5646
1.1000	0172	.0866	3469	0142	.0065	.0387	.0439	
	INCLINAT	ION ANGLE :	15.000	PITCH RATIO		SIGMA		
J	KTOUT	1 OK GOUT	EFFIC	KT/J2	K0/J3	Æ.	KBF	BFANG
•6500	.0638	.1750	.3773	-1511	•0637	.0221	.0051	18.2236 14.8623
.7000	.0742	.1979	.4177	•1514	-0577	.0284	.0085 .0119	11.0062
.7500	.0803	.2155	•4450	.1428	-0511	.0337	.0119	5.8148
.8000	.0811	.2256	•4578	•1267	.0441 .0369	.0373	.0185	2497
.8500	.0757	• 2269	.4512	•1047	.0309	.0403	.0226	-6.0688
.9000	.0637	.2189	.4170 .3410	•0787 •0504	.0235	.0404	.0277	-10.5140
.9500	.0455	.2016 .1753	.1985	.0219	.0175	.0394	.0332	-13.0932
1.0000	.0219 0055	.1401	0651	0050	.0121	.0360	.0375	-14.5972
1.1000	0340	.0955	6229	0281	.0072	.0280	.0374	-17.7461
				17	٠			

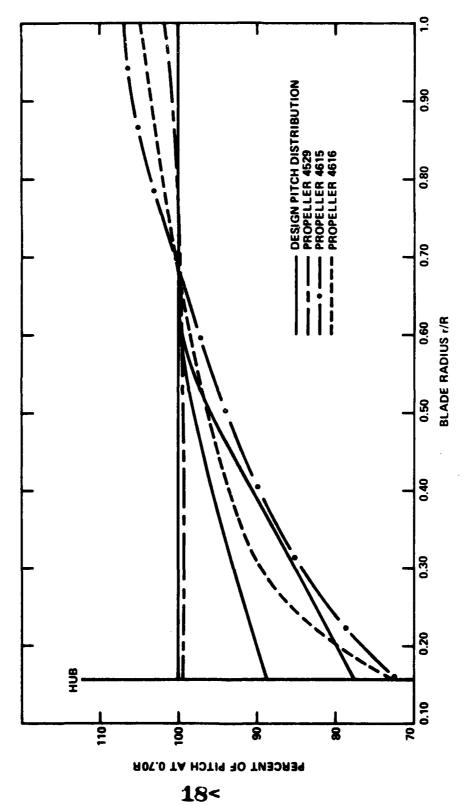


Figure 1 - Pitch Distribution on Propellers 4529, 4615, and 4616

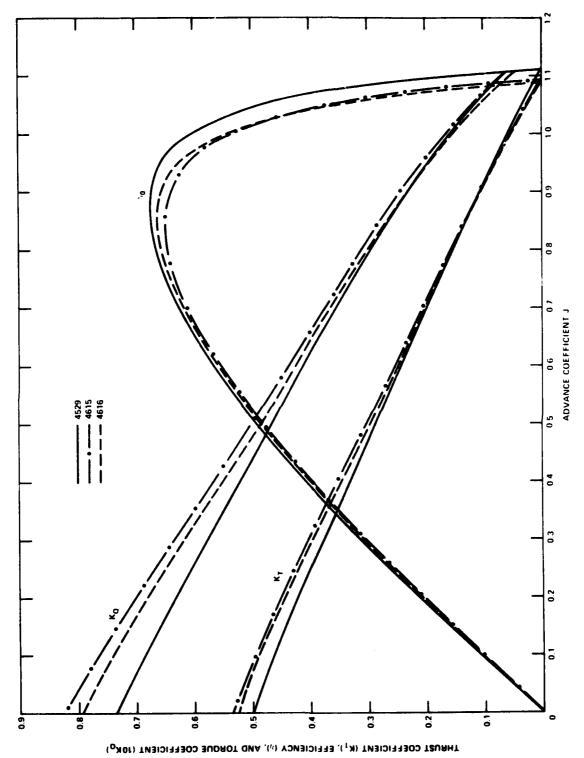


Figure 2 - Open Water Characteristics of Propellers 4529, 4615, and 4616

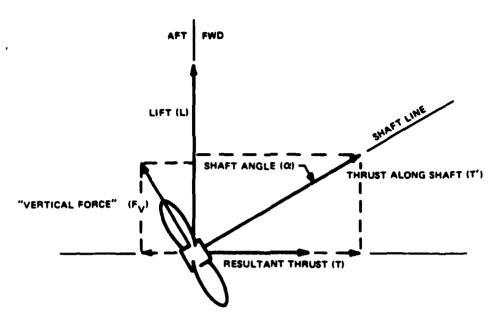


Figure 1a - Side View

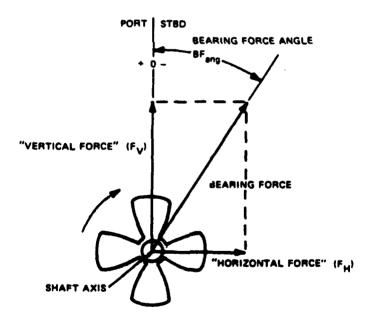


Figure 1b - End View

Figure 3 - Force Diagram for Side Force Measurements

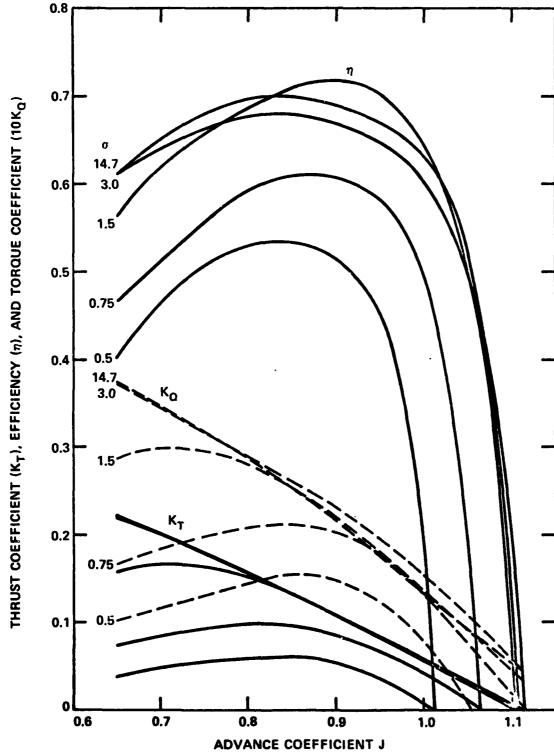


Figure 4 - Cavitation Characteristics of Propeller 4529 at Zero Shaft Inclination 19<

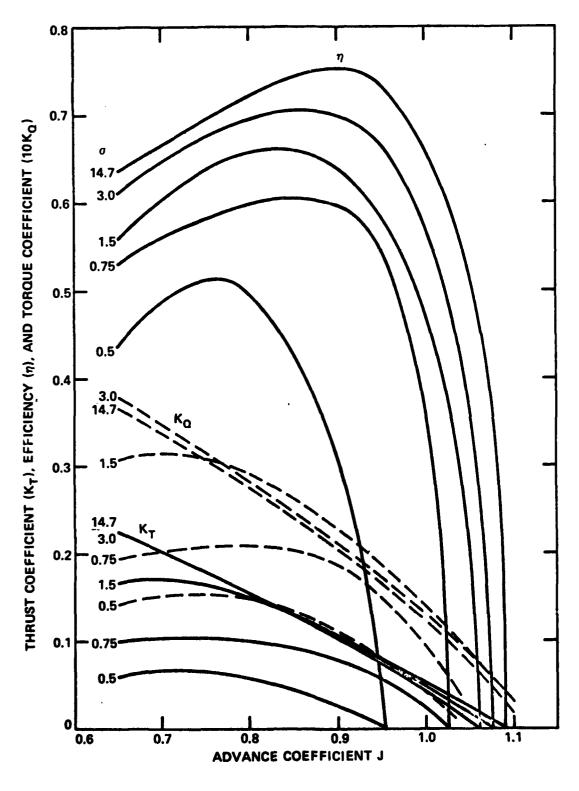


Figure 5 - Cavitation Characteristics of Propeller 4615 at Zero Shaft Inclination 20

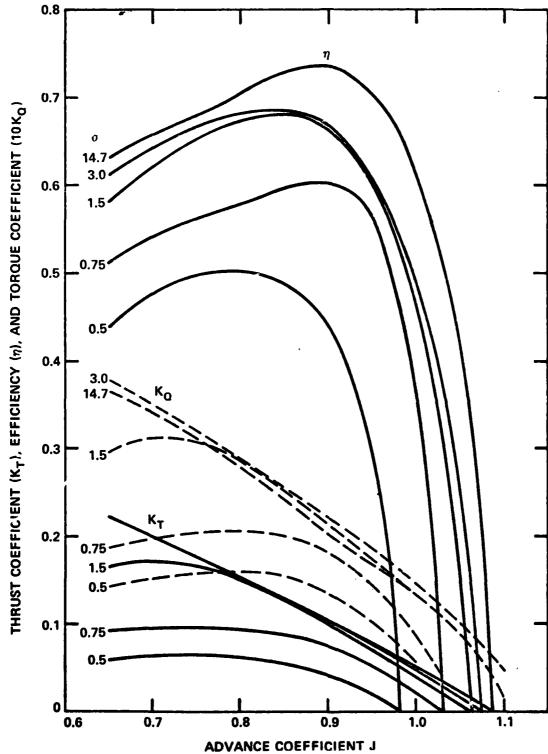


Figure 6 - Cavitation Characteristics of Propeller 4616 at Zero Shaft Inclination

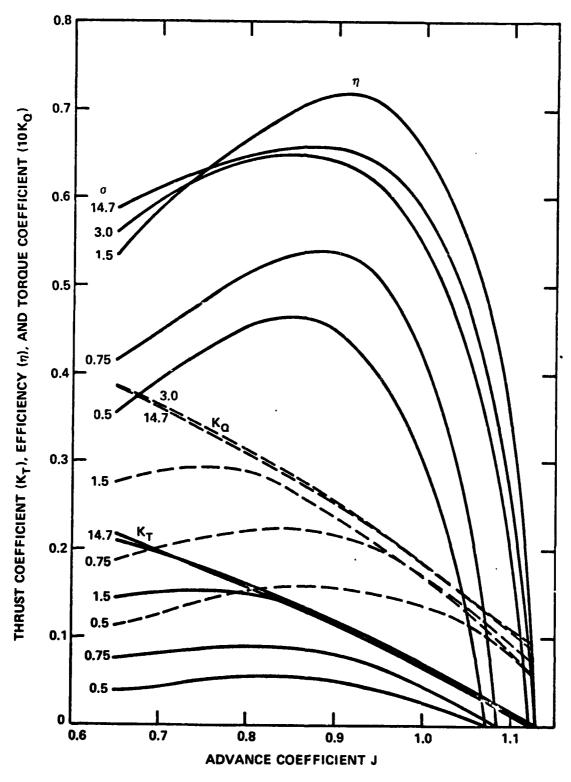


Figure 7 - Cavitation Characteristics of Propeller 4529 at 15 Degrees Shaft Inclination

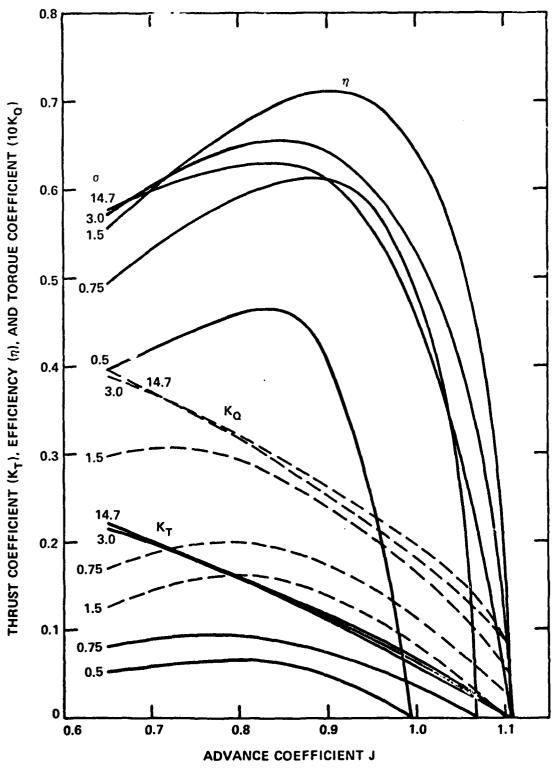


Figure 8 - Cavitation Characteristics of Propeller 4615 at 15 Degrees Shaft Inclination 23<

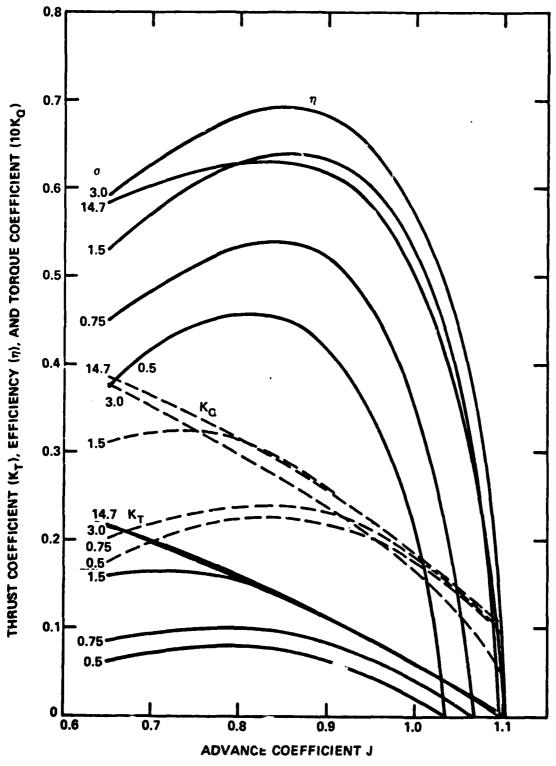


Figure 9 - Cavitation Characteristics of Propeller 4616 at 15 Degrees Shaft Inclination 24

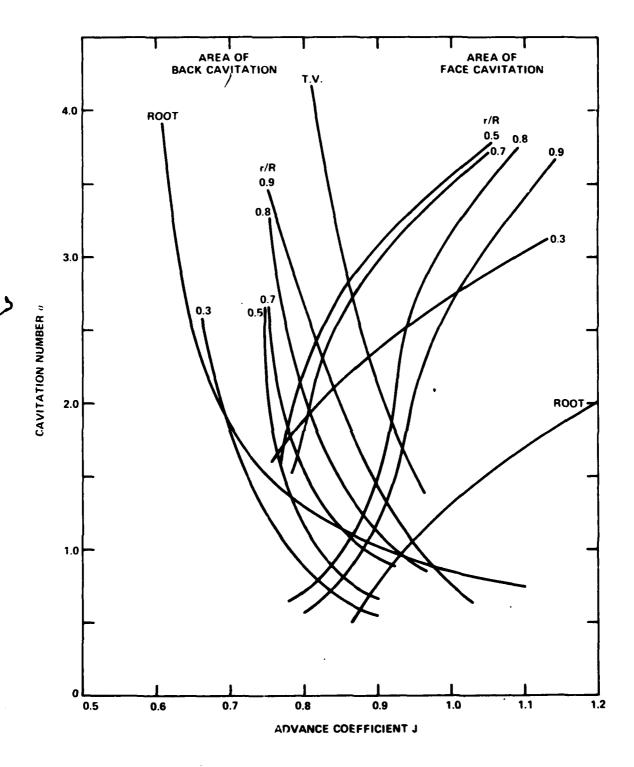


Figure 10 - Cavitation Inception Curves for Propeller 4529 at 15 Degrees Shaft Inclination 25<

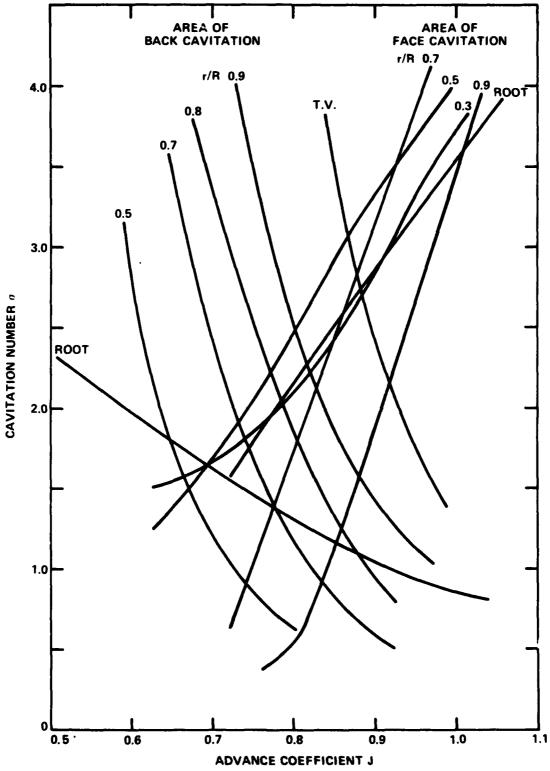


Figure 11 - Cavitation Inception Curves for Propeller 4615 at 15 Degrees Shaft Inclination 26<

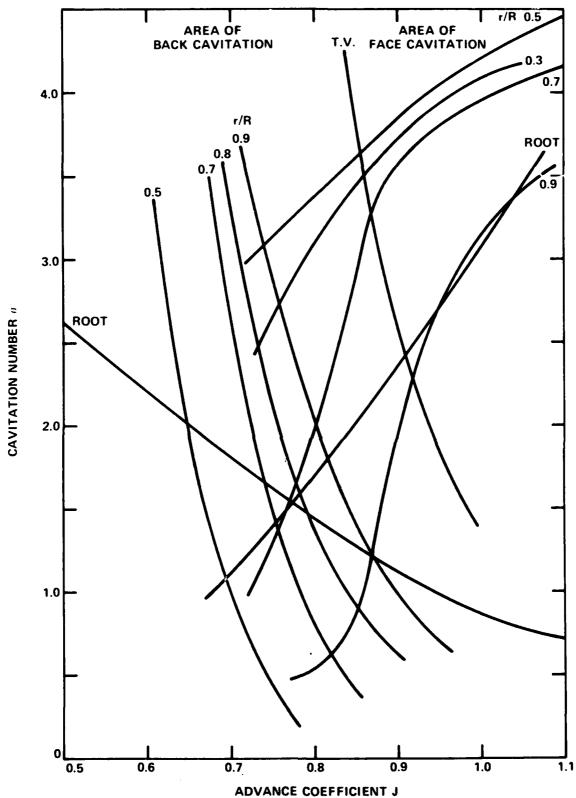


Figure 12 - Cavitation Inception Curves for Propeller 4616 at 15 Degrees
Shaft Inclination 274

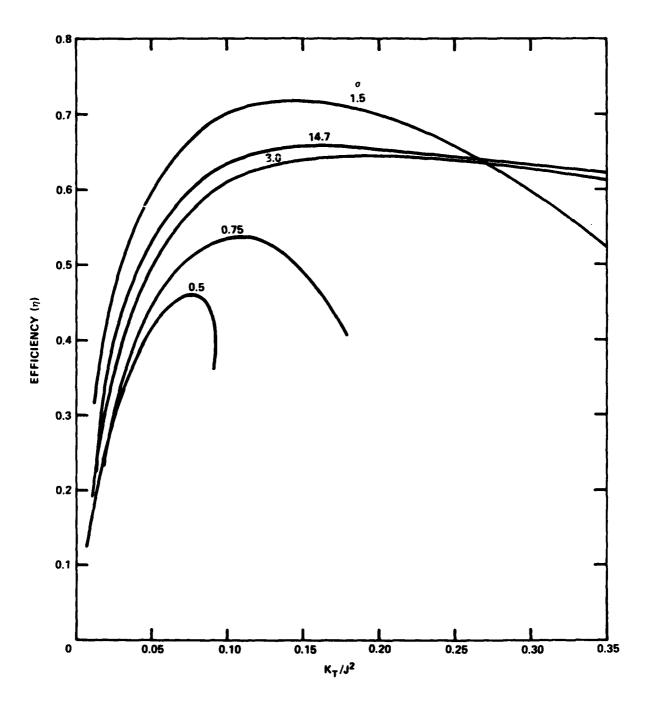


Figure 13 - Propeller 4529 Efficiencies versus K_T/J^2 for Various Cavitation Numbers at 15 Degrees Shaft Inclination

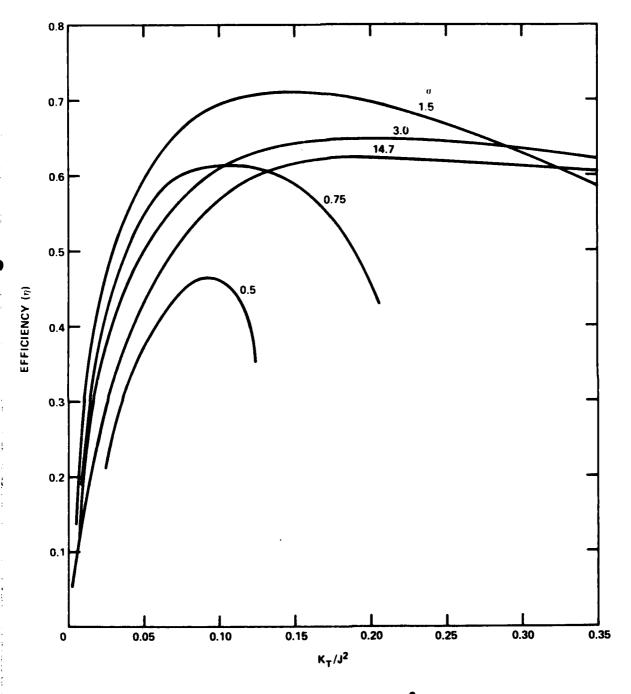


Figure 14 - Propeller 4615 Efficiencies versus K_T/J^2 for Various Cavitation Numbers at 15 Degrees Shaft Inclination

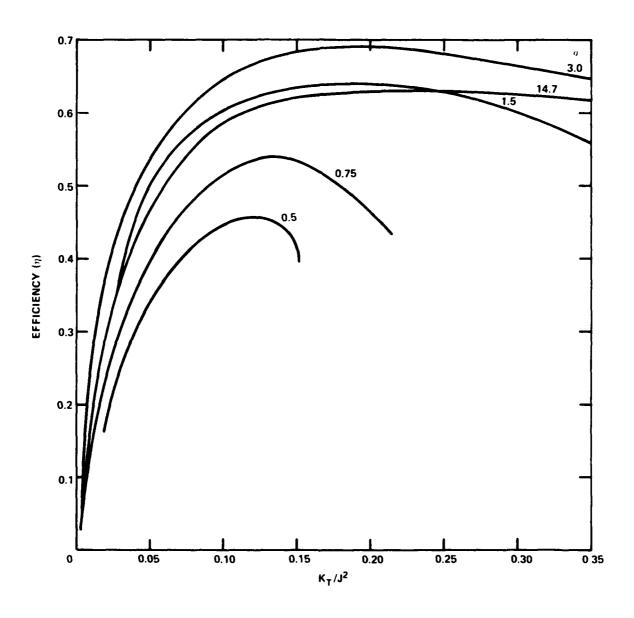


Figure 15 - Propeller 4616 Efficiencies versus K_T/J^2 for various Cavitation Numbers at 15 Degrees Shaft Inclination 30



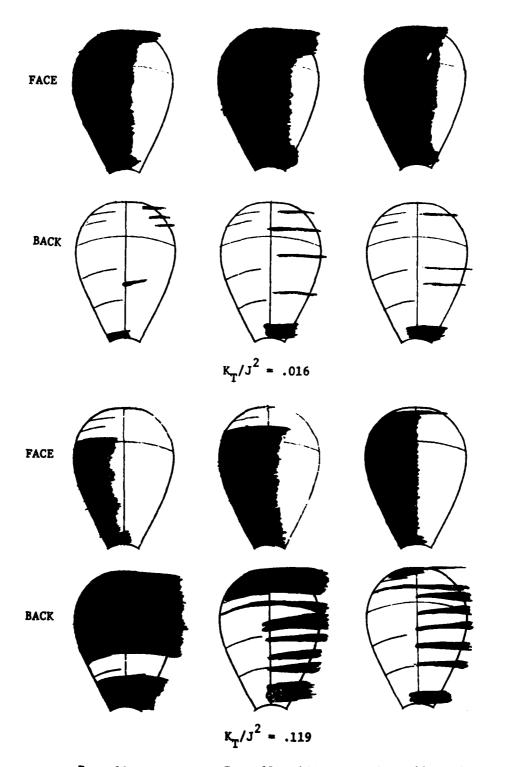


Propeller 4616 Reduced Pitch at the Hub

Propeller 4529 Constant Pitch Distribution

Figure 16 - Comparison of Cavitation on Propellers 4529 and 4616 at J = 0.9, σ = 0.75, and 15 Degrees Shaft Inclination

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Propeller 4529 Propeller 4615 Propeller 4616

Figure 17 - Sketching of Cavitation Present on Propellers at 15 Degress Shaft Inclination, $\sigma = 0.75$, and Two Thrust Loadings