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**A USER'S MANUAL FOR THE
SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS
COMPUTER PROGRAM**

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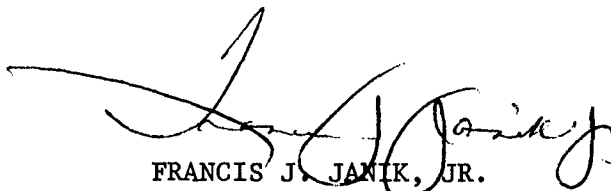
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This Technical Report has been reviewed and is approved.



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FOREWORD

This program was prepared by J. M. Potter of the Solid Mechanics Branch and R. A. Noble of the Experimental Branch, Structures Division, Air Force Flight Dynamics Laboratory. The work was conducted in-house under Project 1347 "Structural Testing of Flight Vehicles", Task 134704 "Structural Testing Criteria". This report covers work accomplished over a time period of 1 October 1972 to 1 May 1973. The essence of the analysis was presented in AFFDL-TM-73-131-FBR in October 1973.

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ABSTRACT

This report presents a detailed description of a computer program to calculate cumulative damage of notched structural members subjected to arbitrary spectra. The Sequence Accountable Fatigue Analysis computer program develops its sequence sensitivity by tracking residual stresses local to a notch throughout the spectrum of loads. Residual stress relaxation analysis is included to increase the generality of the results. An example spectrum and resulting cumulative damage analysis are illustrated.

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SYMBOLS

σ_{res}	Residual stress
σ_{max}	Maximum local stress level
σ_{min}	Minimum local stress level
σ_{ys}	Yield stress
$\sigma_{res_{EQ}}$	Equilibrium component of the residual stress
ϵ_t	Local strain, total
ϵ_e	Elastic component of total strain
ϵ_p	Plastic component of total strain
ϵ_f'	Strain intercept at one reversal on a $\log \epsilon_p$ -log life curve
c	Slope of the $\log \epsilon_p$ -log life curve
S_{max}	Maximum applied stress level
S_{min}	Minimum applied stress level
S_{mean}	Mean applied stress level, $(S_{max} + S_{min})/2$
K_t	Elastic stress concentration Factor
D	Damage
I	Integer describing the level number
N_{EP}	Equilibrium period, number of cycles for the local stresses to return approximately to the equilibrium conditions following an overload
$C1$	Residual stress relaxation constant
$E1, E2$	Relaxation function exponents
N	Number of cycles
E	Modulus of elasticity
N_f	Number of cycles of life at a given stress or strain cycling level

SECTION I

INTRODUCTION

Cumulative damage analyses based upon the local stress-strain behavior at a notch appear to be reasonably successful in anticipating trends in fatigue life behavior of notched specimens subjected to spectrum loading (1-6). The type of behavior that usually occurs is that peak tensile loads tend to increase the fatigue life and peak compressive loads tend to decrease the life of notched structures compared to structures experiencing load spectra not having those peaks (5,6). Local behavior analyses, such as those developed by Smith (7) and Neuber (8), help to explain this phenomenon as being a result of the tensile peak load creating a compressive residual stress at the notch and, conversely, the compressive peak creating a tensile residual stress. The change in life occurs because the residual stress state modifies the subsequent damage accumulation rates.

The Sequence Accountable Fatigue Analysis computer program was developed to incorporate the local stress-strain approach with a recent residual stress relaxation analysis (6) in order to improve the sequence sensitivity of cumulative damage analysis. This technical memorandum presents the details of the resultant computer program and an example of its use. The correlation of predictions made with this analysis to actual results of tests experiencing spectrum loading is presented by Potter (9), and Potter, Gallagher, and Stalnaker (10).

SECTION II
PROGRAM OUTLINE

The Sequence Accountable Fatigue Analysis traces the stress-strain behavior local to a notch throughout an applied load spectrum and calculates the damage based on the local experience. The computer program is divided generally into the four parts or modules outlined in Fig. 1.

The basic input data for the material, specimen geometry, fatigue behavior qualities and spectrum, are developed in Module I. The information required in Module I is discussed further in Section III. Module II takes the input information and determines the local stress-strain behavior. Module III references the Range Pair Counting Method Subroutine to cycle count the local stress spectrum. Module IV determines the damage in the local stress-strain spectrum.

The basic analyses used in Modules II, III and IV are presented below.

Module II - Local Stress-Strain Behavior - The analysis used during the determination of the local stress behavior during the spectrum of loading is a combination of analyses developed by Smith ⁽⁷⁾, Neuber ⁽⁸⁾ and Potter ⁽⁶⁾. Smith's simple analysis indicated that the residual stress could be approximated by assuming that the initial stress-strain behavior was elastic upon unloading following plastic flow. Thus, the residual stress could be calculated knowing the

maximum local stress and the maximum applied stress as in Eq. 1 and in Fig. 2.

$$\sigma_{res_i} = \sigma_{max_i} - K_t S_{max_i} \quad (1)$$

The Sequence Accountable Fatigue Analysis computer program currently incorporates elastic-perfectly plastic stress-strain behavior. Therefore, σ_{max_i} is equal to the yield stress. For the cycles immediately following the peak stress, the residual stress determined in Eq. 1 modifies the elastic solution as shown in Eqs. 2 and 3 (provided that the following maximum applied stress is less than S_{max} and that there is no change in the residual stress due to a minimum applied stress causing reversed yielding).

$$\sigma_{max_i} = \sigma_{res_{i-1}} + K_t S_{max_i} \quad (2)$$

$$\sigma_{min_i} = \sigma_{res_{i-1}} + K_t S_{min_i} \quad (3)$$

The analysis developed by Neuber ⁽⁸⁾ has been extended to cyclic loading by Wetzel ⁽²⁾ and Wetzel, Morrow and Topper ⁽³⁾ and used by many others ^(1,4-6) primarily to determine local stress-strain behavior. It is used in this program only to calculate plastic strains occurring when the residual stress undergoes a step change. The plastic strain calculation routine is accessed only when the σ_{max_i} or σ_{min_i} terms in Eqs. 2 and 3 exceed tensile or compressive yield stress levels, respectively. Figure 3 illustrates the calculation of the plastic strain.

The local stress-strain behavior, according to Wetzel ⁽²⁾ is related to the applied load by Eq. 4

$$\sigma \cdot \epsilon = (K_t S_{max})^2 / E \quad (4)$$

The plastic strain can be found by subtracting the elastic component from the total strain.

$$\epsilon_p = \epsilon_t - \epsilon_e = (K_t S_{\max})^2 / E \cdot \sigma_{\max} - \sigma_{\max} / E$$

Therefore, the plastic strain associated with S_{\max_i} is given in Eq. 5.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - \sigma_{ys} / E \quad (5)$$

If a residual stress existed prior to this plastic strain excursion, the plastic strain associated with that prior excursion is subtracted from Eq. 5 as shown in Eq. 6.

$$\epsilon_{p_i} = (K_t S_{\max_i})^2 / E \sigma_{ys} - (\sigma_{ys} - \sigma_{res_{i-1}})^2 / E \sigma_{ys} \quad (6)$$

A similar calculation is made for plastic strains occurring during the minimum stress peak.

In the analysis developed by Potter ⁽⁶⁾ the residual stress cyclically relaxes toward zero or an equilibrium residual stress as shown in Fig. 4 according to Eq. 7.

$$\sigma_{res_{N=1,2,\dots}} = (\sigma_{res_{N=1}} - \sigma_{res_{EQ}}) \exp(N / N_{EP_i} \ln(0.1)) \quad (7)$$

The N_{EP} term, the Equilibrium Period, is dependent upon the applied stress and the Residual Stress Relaxation Constant.

$$N_{EP_i} = (C1 / \left| K_t S_{\max_i} \right|^{E1} \cdot \left| K_t S_{\text{mean}_i} \right|^{E2}) \quad (8)$$

The Residual Stress Relaxation Constant, $C1$, has not yet been experimentally defined but should be a constant for a material.

Module III - Cycle Counting Method

After the local stress and plastic strain behavior is calculated, the local stress spectrum is Range Pair Counted using a computer program developed by Tischler. ⁽¹¹⁾

Module IV - Damage Calculation

Damage is calculated separately for the plastic strain excursions and the elastic stress spectrum. The damage is determined from the conventional $D = \sum \frac{n}{N}$ calculation. Damage from each of the plastic strain cycles is determined from the Coffin-Manson ⁽¹²⁾ form

$$D_i = 1./N_{f_i} = 1./(\epsilon_{p_i}/\epsilon_{f'})^{1/c}$$

Damage from the elastic stress cycles is determined in a similar manner. The maximum and minimum local stress levels are sequentially compared to unnotched S-N data in a Modified Goodman Diagram format. Damage is summed, and failure of the coupon is defined as the event occurring when the summed damage equals unity.

SECTION III
INPUT DATA REQUIREMENTS

In general, each spectrum analyzed will require slightly different programming in order to get the load history into a usable format for the core program. The basic program requires a certain family of information before any analytical predictions can be made. Appendix I contains a program listing for the Sequence Accountable Fatigue Analysis. The subroutine CORE which accesses the subroutines having to do with RPCM, the Range Pair Counting Method, contains the basic analysis. Subroutine SAL reads the data input and then references subroutine CORE. The subroutine SAL shown is one in which a block of cycles is repeated with optional cycles. A list of the input data cards and the resulting analysis is given in Appendix II.

The specific data requirements are given below.

1. Stress-Strain Behavior - The stress-strain behavior is presumed to be elastic-perfectly plastic with the tensile yield stress being equal to the compressive yield stress. The yield stress value used is an average of the monotonic behavior generally being above the 0.2% yield value and below the engineering ultimate strength.

2. Residual Stress Relaxation - The residual stress relaxation behavior of Eq. 7 and 8 is characterized by C_1 , the Residual Stress Relaxation Constant and E_1 and E_2 , the relaxation equation exponents. The Residual Stress Relaxation Constant, C_1 , has not yet been adequately determined. It should be a material property if the relaxation function

is correct and must be assumed. A reasonably accurate estimate of the Residual Stress Relaxation Constant for aluminum material falls in the range of $5-20 \times 10^6$ (cycles) (Ksi)². Further experimentation on the part of the analyst should develop a CI usable for his set of conditions until actual measurement of residual stress relaxation behavior defines the relaxation function and constants. At present E1 and E2 are considered to be equal to 1.0.

3. Specimen Geometry - The elastic K_t value (if available) is entered into the analysis. If that value is not available then an estimate from some other method may be used. In certain cases, a value may be determined from a constant amplitude fatigue test of a similar structure by fitting several values of K_t to the analysis and determining the best correlation as is done with the K_f solution. Once a stress concentration factor, K_t , is determined for a specimen, that value is not changed from test-to-test of the same coupon configuration.

4. Load Multiplier - Different spectra are presented for analysis in different manners. Some data are presented in percent of maximum stress, others in terms of nominal stress, and others in terms of bending moment. The value of the load multiplier defines the nominal stress history.

5. Cumulative Damage Analysis - The damage from the range-paired elastic stress spectrum is determined by calculating a simple $\frac{n}{N}$ value for each level and accumulating the total. The N_{f_i} value is determined from unnotched coupon S-N data in the Modified Goodman Diagram format.

The program requires the input of four second order equations describing the maximum and minimum stress levels at lives of 10^4 , 10^5 , 10^6 and 10^7 cycles. The coefficients of the equations are derived by least square fitting the S-N data presented in the form of Eq. 9.

$$S_{\max} = A(I)S_{\min}^2 + B(I)S_{\min} + C(I) \quad (9)$$

The A, B, and C coefficients for several typical materials are presented in Appendix IV. The S-N data shown was derived from various sources but usually from the MIL-HDBK-5A (13). The C coefficients correspond to the maximum stress level at zero to maximum applied stress conditions on the unnotched coupons.

The damage from the plastic strain cycles is determined using the Coffin-Manson relation to calculate the N_{f_i} value. The conventional plastic strain intercept at one reversal and the ϵ_p - life slope values are used in the analysis. Specific measured values from the literature are used when available and typical values when they are not available.

6. Analysis or Test Spectrum - The last information needed is the order and magnitude of application of the spectrum used in the test.

SECTION IV
OUTPUT OPTIONS

The Computer Program prints the following output in the process of the analysis.

1. Maximum and minimum applied stress and local stress response through the spectrum. Also printed out is the residual stress, equilibrium stress, applied cycles, and the equilibrium period.

2. The elastic local stress history as input into the Range Pair Subroutine and the resulting Range Paired spectrum.

3. The plastic strain occurrence during the spectrum and the damage associated with each strain reversal.

4. The accumulated damage associated with the plastic strains.

5. The Range Paired elastic stress spectrum and the damage associated with each level.

6. The accumulated damage associated with the current block of loading including the plastic strain damage and the total damage since the initiation of cycling.

At the option of the analyst, he can print out all the above items or only two. The IPRINT value controls what data is printed.

If IPRINT = 1, all six items are printed for each flight or block.

If IPRINT = 2, all items except 2. above are printed.

If IPRINT = 3, only items 4. and 6. above are printed.

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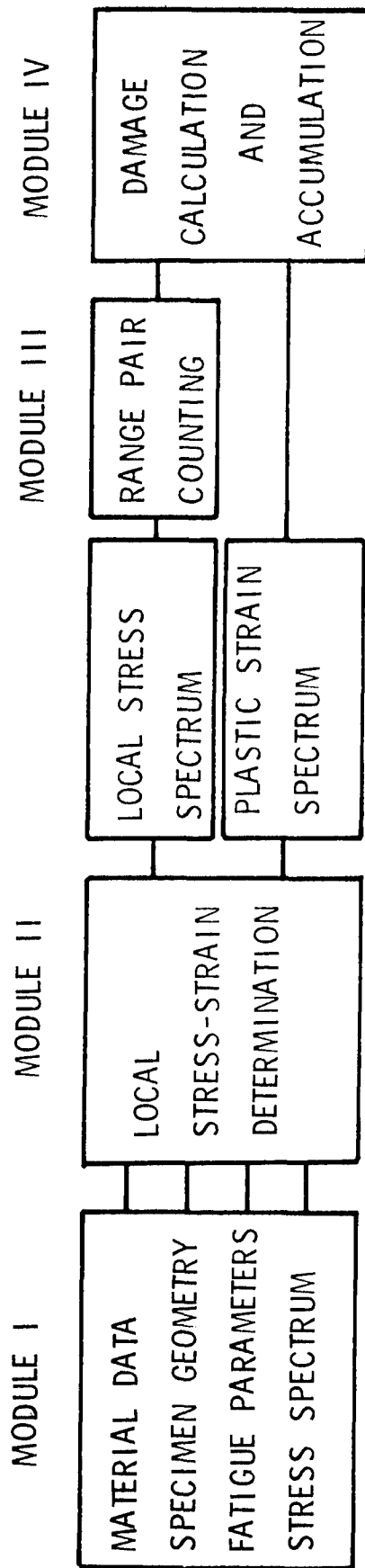


FIGURE 1. PROCEDURE USED IN THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

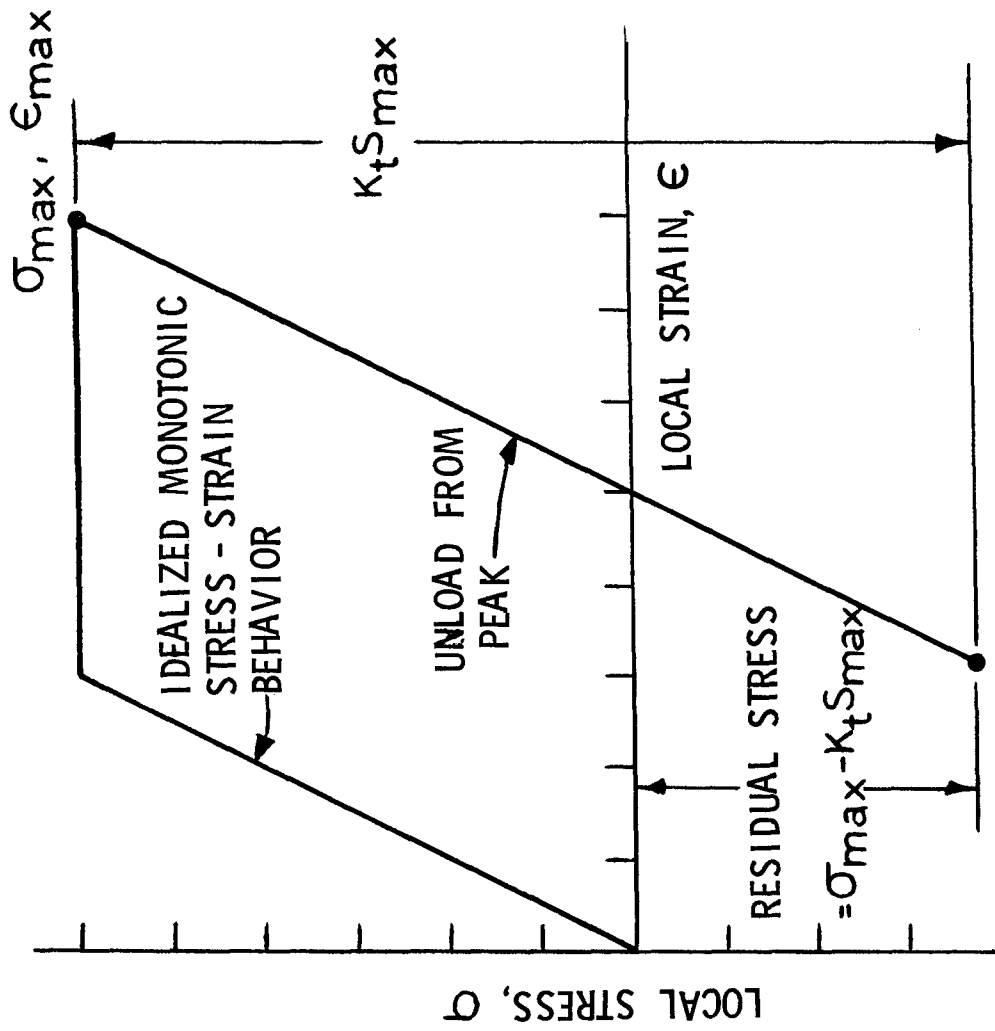


FIGURE 2. METHOD OF DETERMINING THE RESIDUAL STRESS FOLLOWING A PEAK LOAD

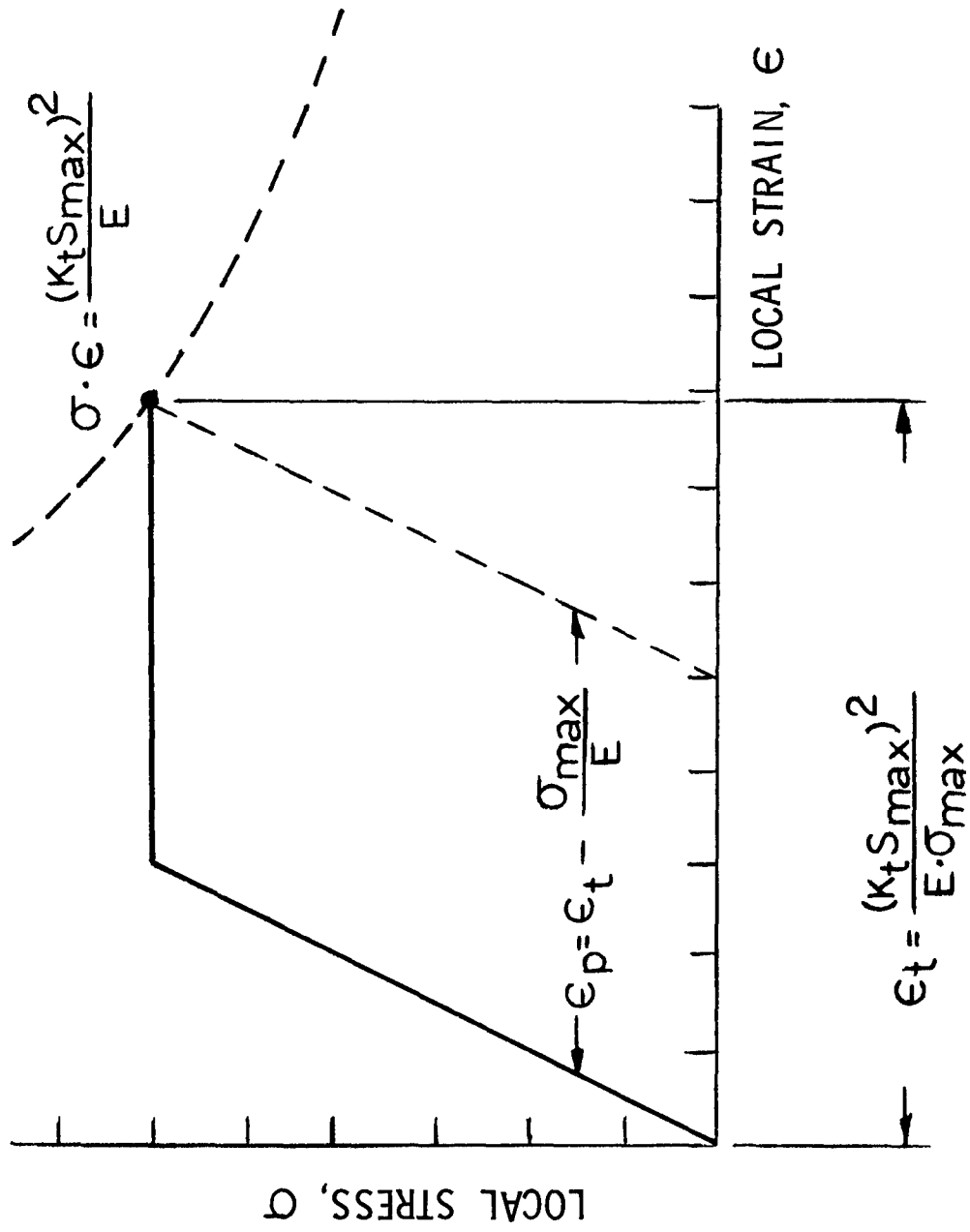
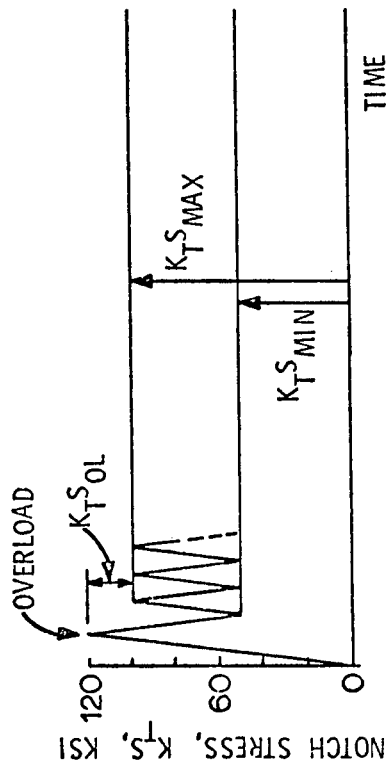


FIGURE 3. METHOD OF DETERMINING PLASTIC STRAIN LEVELS



15

APPLIED STRESS HISTORY

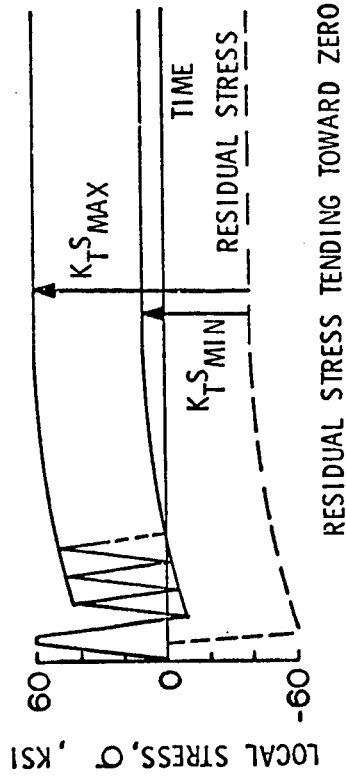


FIGURE 4. LOCAL STRESS RESPONSE FOR APPLIED CONSTANT AMPLITUDE LOADING WITH RESIDUAL STRESS RELAXATION


```

115      FORMAT(16HISPECTRUM FROM ,8A8)
      C
      C      INPUT OF DATA PECULIAR TO A MATERIAL
      C
120      READ(5,10)TM1, TM2, TYS, EPSD, COFMAN, ELMOD
      FORMAT(2A8, 3F18.5, F10.2)
      WRITE(6,12) TM1, TM2, TYS, EPSD, COFMAN, ELMOD
125      *-- F18.5 // 23H LCF STRAIN INTERCEPT =, F18.5 // 31H INVERSE OF COFFI
      *N-MANSON SLOPE, F18.5 // 18H ELASTIC MODULUS =, F18.5)
      READ(5,14) (A(N), B(N), C(N), TITLE1, TITLE2, N=4, 7)
      FORMAT(3F18.5, 2A8)
      WRITE(6,16)
130      *TA)
      FORMAT(58HDCOEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DA
      WRITE(6,18)
      C      SMAX = A(I)*SHIN**2 + B(I)*SHIN + C(I))
      WRITE(6,20)
135      FORMAT(5X, 5H LIFE, 10X, 5H A(I), 14X, 5H B(I), 14X, 5H C(I))
      WRITE(6,22) (N, A(N), B(N), C(N), N=4, 7)
      FORMAT(8H 10**, I2, 3F18.5)
      WRITE(6,56) TITLE1, TITLE2
140      *SUPPLIED FROM ,2A8)
      FORMAT(39HUNNOTCHED COUPON S-N DATA DERIVED FROM/28H INFORMATION
      READ(5,14) C1, E1, E2
      WRITE(6,24)
145      FURMAT(36HRESIDUAL STRESS RELAXATION FUNCTION)
      WRITE(6,26)
      FORMAT(/42H ENEP = C1/(KTSMAX**E1 + KTSHEAN**E2)/)
      WRITE(6,28) C1, E1, E2
150      *3)
      FORMAT(13H*WHERE C1 =, E15.8, 9H , E1 =, F10.3, 10H AND E2 =, F10.
      C
      C
      C      INPUT OF DATA PECULIAR TO A SEQUENCE
      C
      C
155      READ(5,65)AKT
      FORMAT(F18.5)
      READ(5,32) NBLOCK, JLEVEL, NTYPE
      FORMAT(3I10)
      WRITE(6,34) NBLOCK, JLEVEL
160      FORMAT(/ / I10, 23H TIMES THROUGH BLOCK OF, I10, 6H LOADS)
      READ(5,35) TLL
      FORMAT(F18.5)
      WRITE(6,33) TLL
      FORMAT(/ 5X, 13H LOAD LIMIT =, F18.5)
      READ(5,36) (IDUMHY , ITYPE(K), RTMIN(K), RTMAX(K), RNN(K), K=1, JLEVEL)
      FORMAT(I4, 2X, I4, 2X, F18.5, 1X, F18.5, 1X, F18.5, 1X)
      WRITE(6,38)
165      *K , ITYPE(K), RTMIN(K), RTMAX(K), RNN(K), K=1, JLEVEL)
      WRITE(6,39)
      FORMAT(/ / 47H BLOCK TYPE TYPE TYPE TYPE TYPE TYPE TYPE TYPE /)

```

```

170      DO 42 JJ=1,NBLOCK
175      READ(5,40) IDUMHY
180      WRITE(6,40) JJ
185      FORMAT(9I5)
190      CONTINUE
195      SUMENN=0.
200      RES(1)=0.
205      WRITE(6,8) T1,T2,T3,T4,T5,T6,T7,T8
210      WRITE(6,51) (AKT)
215      FORMAT(7H AKT = , (F6.2))
220      WRITE(6,55) C1
225      FORMAT(/24H RELAXATION CONSTANT C1=,F15.2)
230      IF(IRPCM.GE.2)GO TO 59
235      WRITE(6,11)
240      FORMAT(56HOSPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE)
245      *)
250      CONTINUE
255      DO 1002 KFL=1,NBLOCK
260      JJJ=1
265      DO 60 J=1,JLEVEL
270      DO 70 KK=1,NTYPE
275      IF(NN(KFL,KK).EQ.0) GO TO 60
280      IF(IITYPE(J).EQ.NN(KFL,KK))GO TO 150
285      CONTINUE
290      STMIN(JJJ)=RTMIN(J)*TLL
295      STMAX(JJJ)=RTMAX(J)*TLL
300      ENN(JJJ)=RNN(J)
305      JJJ=JJJ+1
310      CONTINUE
315      NLEVEL=JJJ-1
320      CALL CORE(KFL)
325      IPRINT=2
330      CONTINUE
335      597 CONTINUE
340      595 CONTINUE
345      596 CONTINUE
350      580 STOP
355      END

```

```

SUBROUTINE CORE(KFL)
*****
CORE PROGRAM OF
THE SEQUENCE ACCOUNTABLE FATIGUE ANALYSIS

MODULE II

LOCAL STRESS AND STRAIN DETERMINATION
*****
COMMON/MDEC1/SIGMAX(200),SIGMIN(200)
COMMON/MSAL/RNCYC(200),KPHAX,IPRINT
DIMENSION PLSTRA(200),EN(200),EX(200)
COMMON/MCOR1/NLEVEL,IRPCM,IPLST,ELMOD,TYS,EPS0,COPHAN,C1,E2,E1,
*RES(200),AKT,SUMENN,SUMNC
COMMON/MCOR2/STMIN(200),STMAX(200),ENN(200),A(10),B(10),C(10),
*R(10),NN(20,10)
JJ=KFL
IF(JJ.GT.3) IPRINT=3
IRAIN=1
WRITE(6,54) JJ
FORMAT(20H FLIGHT OR BLOCK NO.,I5)
IF(IPRINT.GE.3) GO TO 61
WRITE(6,62)
FORMAT(66H STHAX STMIN SIGMAX SIGMIN RES EQRES ENN
NEP)
*
CONTINUE
DO 570 J=1,NLEVEL
I=J+1
PLSTRA(J)=0.
*****
DETERMINE SEQUENCE ACCOUNTABLE RESIDUAL STRESS AND CORRESPONDING
PLASTIC STRAIN
ASMAX=AKT*STMAX(J)
ASMIN=AKT*STMIN(J)
ASMEAN=(ASMAX+ASMIN)/2.
JA=0
IF(RES(I-1)+ASMIN+TYS) 190,190,200
JA=1
AAA=RES(I-1)+ASMIN
PLSTRA(J)= -1.*(AAA/ELMOD)*(1.+AAA/TYS)
JB=0
IF(RES(I-1)+ASMAX-TYS) 220,210,210
JB=-1
AAA=ASMAX
888=TYS-RES(I-1)
IF(888.GE.TYS) GO TO 214
888=0.
PLSTRA(J)= AAA*AAA/(ELMOD+TYS)-888*888/(ELMOD+TYS)

```



```

115      GO TO 450
120      CYCINT=DUMMY/10.
130      DO 500 K=1,10
140      DECK=FLOAT(K)
150      EN(K)=CYCINT*DECK
160      IF(K.EQ.1) GO TO 490
170      EX(K)=EXP(-2.303*EN(K-1)/ENEP)+EXP(-2.303*EN(K)/ENEP)
180      GO TO 500
190      EX(K)=1.+EXP(-2.303*EN(K)/ENEP)
200      CONTINUE
210      IF(NFLAG.EQ.0) GO TO 530
220      NFLAG2=NFLAG+10
230      DO 520 K=11,NFLAG2
240      EN(K)=2.*DUMMY
250      EX(K)=EXP(-2.303*EN(K-1)/ENEP)+EXP(-2.303*EN(K)/ENEP)
260      DUMMY=2.*DUMMY
270      CONTINUE
280      DO 559 K=1,NFLAG2
290      SIGMAX(IRAIN)=ASMAX+EQRES+DIF*EX(K)/2.
300      SIGMIN(IRAIN)=SIGMAX(IRAIN)-ASMAX+ASMIN
310      RNCYC(IRAIN)=EN(K)
320      IF(K.EQ.1) GO TO 540
330      RNCYC(IRAIN)=RNCYC(IRAIN)-EN(K-1)
340      IF(IPRINT.GE.3) GO TO 551
350      WRITE(6,550)SIGMAX(IRAIN),SIGMIN(IRAIN),RNCYC(IRAIN),IRAIN
360      FORMAT(16H RELAXATION
370      ,2(F7.2,1X),16X,F6.2,31X,I6)
380      CONTINUE
390      IRAIN=IRAIN+1
400      CONTINUE
410      RES(I)=EQRES+DIF*EXP(-2.303*ENN(J)/ENEP)
420      CONTINUE
430      RES(1)=RES(J)
440      IN=IRAIN-1
450      C
460      C
470      C
480      C
490      C
500      C
510      C
520      C
530      C
540      C
550      C
560      C
570      C
580      C
590      C
600      C
610      C
620      C
630      C
640      C
650      C
660      C
670      C
680      C
690      C
700      C
710      C
720      C
730      C
740      C
750      C
760      C
770      C
780      C
790      C
800      C
810      C
820      C
830      C
840      C
850      C
860      C
870      C
880      C
890      C
900      C
910      C
920      C
930      C
940      C
950      C
960      C
970      C
980      C
990      C
1000     C

```

MODULE III
CYCLE COUNTING TECHNIQUE
CALL SUBROUTINE TO RANGE PAIR COUNT SPECTRUM
IF(IRPCM.GT.1)GO TO 591
CALL RPCM(IN)
GO TO 592
CONTINUE
KPMAX=IN
CONTINUE

MODULE IV
DAMAGE ACCUMULATION CALCULATION


```

311 CONTINUE
NPKSN = NPKSN - JMAS
C
C RANGE PAIR COUNT THE ADJUSTED LOAD SPECTRUM
115 6000 I = 1
      KB = 0
      L = JMAX
      KMIN = 0
      KMAX = 0
      LR = 0
      K31 = 0
      1 IF (RNCYC(I) .GT. 1.0) GO TO 400
      IF (KB .NE. 0) GO TO 5
      X1 = SIGMAX(I)
      X2 = SIGMIN(I)
      IND1 = NSTEP(I)
      IND2 = IND1
      I = I + 1
      KB = 1
      GO TO 1
      5 X3 = SIGMAX(I)
      X4 = SIGMIN(I)
      IND3 = NSTEP(I)
      IND4 = IND3
      KMIN = 1
      KMAX = 0
      K31 = 0
      IF (RNCYC(I) .EQ. 1.0) GO TO 6
      KEY = 1
      KIND = 1
      GO TO 415
      6 KEY = 0
      CYCNO = AINT(RNCYC(I)+0.5)
      CALL DECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)
      1000 GO TO (10,10,30),KCYGEN
      10 KB = 1
      I = I + 1
      IF (KMIN .NE. 1) GO TO 36
      IF (I .LE. NPKSN) GO TO 5
      RES(LR+1) = X1
      RES(LR+2) = X2
      INDEX(LR+1) = IND1
      INDEX(LR+2) = IND2
      LRMAX = LR + 2
      GO TO 2000
      30 IF (KMIN .NE. 1) GO TO 35
      12 I = I + 1
      IF (I .LE. NPKSN) GO TO 31
      RES(LR+1) = X1
      RES(LR+2) = X2
      RES(LR+3) = X3
      INDEX(LR+1) = IND1
      INDEX(LR+2) = IND2
      INDEX(LR+3) = IND3

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CORE 366
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 CORE 419
 CORE 420

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170      LRMAX = LR + 3
        GO TO 2000
        X4 = SIGMAX(I)
        IND4 = NSTEP(I)
        KMAX = 1
        KMIN = 0
        K31 = 1
32      IF (RNCYC(I) .GT. 1.0) GO TO 40
175      GO TO 6
        KEY = 1
        KIND = 0
        GO TO 415
35      X4 = SIGMIN(I)
        IND4 = NSTEP(I)
        KMIN = 1
        KMAX = 0
        K31 = 0
        GO TO 32
36      X3 = SIGMIN(I)
        IND3 = NSTEP(I)
        KMIN = 1
        KMAX = 0
        GO TO 12
400     KEY = 1
        IF (KB .NE. 0) GO TO 410
        X1 = SIGMAX(I)
        X2 = SIGMIN(I)
        X3 = SIGMAX(I)
        X4 = SIGMIN(I)
        IND1 = NSTEP(I)
        IND2 = IND1
        IND3 = IND1
        IND4 = IND1
        KMIN = 1
        KMAX = 0
        K31 = 0
        IF (RNCYC(I) .LE. 2.0) GO TO 401
        RNCYC(I) = RNCYC(I) - 1.0
        GO TO 402
401     RNCYC(I) = RNCYC(I) - 2.0
402     KIND = 0
        GO TO 415
410     X3 = SIGMAX(I)
        X4 = SIGMIN(I)
        IND3 = NSTEP(I)
        IND4 = IND3
        KMIN = 1
        KMAX = 0
        K31 = 0
        KIND = 1
        RNCYC(I) = RNCYC(I) - 1.0
        KB = 0
        CYCNO = AINT(RNCYC(I)+0.5)
        CALL DECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)
415     GO TO 1000

```

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CORE 421
CORE 422
CORE 423
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CORE 473
CORE 474
CORE 475

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```

2000 LMAX = L
    IF (LRMAX .LT. 4) GO TO 5000
    IF (NCYNO .EQ. 0) GO TO 5000
C
C RANGE PAIR COUNT OF RESIDUE SPECTRUMS
C
    NRES = NRES + 1
    CALL DECRES(LRMAX,NCYNO)
    GO TO 2000
230 5000 IF (LRMAX .LE. 1) GO TO 3000
C
C COUNT THE LAST RESIDUE SPECTRUM - RANGE PAIR COUNTING WILL YIELD N
C ADDITIONAL CYCLES
C
    KK = 0
    RESMAX = RES(1)
    RESMIN = RES(1)
    IMAX = 1
    IMIN = 1
240 DO 500 I = 2,LRMAX
    IF (RES(I) .LT. RESMAX) GO TO 490
    RESMAX = RES(I)
    IMAX = I
    GO TO 500
245 490 IF (RES(I) .GT. RESMIN) GO TO 500
    RESMIN = RES(I)
    IMIN = I
500 CONTINUE
    CALL CYCRES(RESMAX,RESMIN,1.0,INDEX(IMAX))
    KK = KK + 1
250 510 J = IMAX - 2
    IF (J .LE. 0) GO TO 550
    CALL CYCRES(RES(J),RES(J+1),1.0,INDEX(J))
    KK = KK + 1
    IMAX = J
    GO TO 510
255 550 J = IMIN + 2
    IF (J .GT. LRMAX) GO TO 575
    CALL CYCRES(RES(J-1),RES(J),1.0,INDEX(J-1))
    KK = KK + 1
    IMIN = J
    GO TO 550
260 575 KHAX = KK
    LMAX = L
C
C SORT THE ANALYSIS SPECTRUM TO PRODUCE THE RANGE PAIR COUNTED SPECT
C
3000 KP = 0
    DO 605 JJ = 1,NPKS
    KC = 0
    DO 600 I = 1,LMAX
    IF (NNSTEP(I) .NE. JJ) GO TO 600
    KP = KP + 1
    KC = KC + 1
    NSTEP(KP) = KP
    END

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CORE 476
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280   SIGMAX(KP) = CYCLE(I,1)
      SIGMIN(KP) = CYCLE(I,2)
      RNCYC(KP) = RNECYC(I)
      IF (KC .LT. 2) GO TO 600
      IF (SIGMAX(KP) .NE. SIGMAX(KP-1)) GO TO 600
      IF (SIGMIN(KP) .NE. SIGMIN(KP-1)) GO TO 600
      KP = KP - 1
      RNCYC(KP) = RNCYC(KP) + 1.0
      600 CONTINUE
      605 CONTINUE
      KPMAX = KP
      IF (IPRINT.GE.2) GO TO 104
      WRITE(6,2010)
      2010 FORMAT(1H1,48X,33HRANGE PAIR CYCLE COUNTED SPECTRUM//)
      WRITE(6,22)
      102 FORMAT(5X,3F10.2)
      104 CONTINUE
      END

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531 CORE
 532 CORE
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SUBROUTINE CYCGEN(Y1,Y2, CYCPF,NSTPEP)
COMMON/MCYG/CYCLE(200,2),RNECYC(200),NNSTEP(200)
COMMON/MCGOE/L,LIND
C
C THIS SUBROUTINE GENERATES CYCLES FOR THE ANALYSIS SPECTRUM FROM OA
C SUPPLIED BY SUBROUTINE DECIDE
C
LIND = 0
L = L + 1
CYCLE(L,1) = Y1
CYCLE(L,2) = Y2
RNECYC(L) = CYCPF
NNSTEP(L) = NSTPEP
IF (L.EQ.1) GO TO 100
IF (CYCLE(L-1,1) .NE. CYCLE(L,1)) GO TO 100
IF (CYCLE(L-1,2) .NE. CYCLE(L,2)) GO TO 100
10 L = L - 1
RNECYC(L) = RNECYC(L) + 1.0
LIND = 1
100 RETURN
END

```

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CORE 550
CORE 551
CORE 552
CORE 553
CORE 554
CORE 555
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CORE 558
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CORE 562
CORE 563
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CORE 566
CORE 567
CORE 568
CORE 569
CORE 570

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SUBROUTINE DECIDE(X1,X2,X3,X4,KEY,I,CYCNO,KCYGEN)
COMMON/MDEC1/SIGMAX(200),SIGNIN(200)
COMMON/MDEC2/NSTEP(200),LR,KMAX,KMIN,K31
COMMON/MDECR/RES(450),INDEX(450),IND1,IND2,IND3,IND4,KIND
COMMON/MCYG/CYCLE(200,2),RNECYC(200),NNSTEP(200)
COMMON/MCGDE/L,LIND
C THIS SUBROUTINE DECIDES WHETHER OR NOT THE VALUES X1,X2,X3, AND X4
C FROM THE ADJUSTED LOAD SPECTRUM SATISFY THE RANGE PAIR COUNTING CO
C
KFIRST = 0
IF (K31 .NE. 0) GO TO 11
IF (X3 .LE. X2) GO TO 200
11 IF (X2 .GT. X1) GO TO 210
IF (X2 .LT. X4 .OR. X3 .GT. X1) GO TO 500
150 IF (X2 .GT. X3) GO TO 151
CALL CYCGEN(X3,X2,1.0,NNSTEP(I))
GO TO 152
151 CALL CYCGEN(X2,X3, 1.0,NNSTEP(I))
152 X1 = X1
X2 = X4
IF (IND3 .NE. IND2) LIND = 1
IND2 = IND4
KCYGEN = 1
IF (KEY .NE. 0) GO TO 110
RETURN
210 IF (X2 .GT. X4 .OR. X3 .LT. X1) GO TO 500
GO TO 150
200 X1 = X1
X2 = X4
IND2 = IND4
KCYGEN = 2
IF (KEY .EQ. 0) RETURN
CYCNO = CYCNO - 1.0
GO TO 110
C ADD X1 TO THE RESIDUE SPECTRUM
C
500 LR = LR + 1
RES(LR) = X1
INDEX(LR) = IND1
X1 = X2
X2 = X3
X3 = X4
IND1 = IND2
IND2 = IND3
IND3 = IND4
KCYGEN = 3
IF (KEY .NE. 0) GO TO 110
RETURN
110 GO TO (1150,1200,1500),KCYGEN
1150 IF (CYCNO .GT. 1.0) GO TO 1151
IF (CYCNO .LE. 0.0) RETURN
1153 CYCNO = CYCNO - 1.0
GO TO 1152

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CORE 571
CORE 572
CORE 573
CORE 574
CORE 575
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CORE 623
CORE 624
CORE 625

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1151 IF (LIND .EQ. 1) GO TO 1153
IF (IND3 .NE. IND4) GO TO 1153
RNECYC(L) = RNECYC(L) + CYCNO - 2.0
CYCNO = 1.0
1152 IF (KMAX .NE. 1) GO TO 111
X3 = SIGMIN(I)
IND3 = NSTEP(I)
IF (CYCNO .GT. 0.0) GO TO 112
KMIN = 1
KMAX = 0
KCYGEN = 3
RETURN
1200 IF(CYCNO .LE. 0.0) RETURN
CYCNO = CYCNO - 1.0
X3 = SIGMAX(I)
X4 = SIGMIN(I)
X4 = SIGMIN(I)
KFIRST = 1
GO TO 113
111 X3 = SIGMAX(I)
X4 = SIGMIN(I)
IF (KFIRST .NE. 0) GO TO 113
CYCNO = CYCNO - 1.0
KFIRST = 1
113 IND3 = NSTEP(I)
IND4 = IND3
KMIN = 1
KMAX = 0
GO TO 10
1500 IF (KMAX .NE. 0) GO TO 1510
IF(CYCNO .LE. 0.0) RETURN
CYCNO = CYCNO - 1.0
112 X4 = SIGMAX (I)
IND4 = NSTEP(I)
KMAX = 1
KMIN = 0
GO TO 11
1510 X4 = SIGMIN(I)
IND4 = NSTEP(I)
KMAX = 0
KMIN = 1
GO TO 10
END

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CORE

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CORE


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SUBROUTINE DEGRES(LRMAX,NCYNO)
COMMON/MGDE/L,LIND
COMMON/MDECR/RES(450),INDEX(450),INC1,IND2,IND3,IND4,KIND
C
C THIS SUBROUTINE DECIDES WHETHER OR NOT THE ELEMENTS OF THE RESIDUE
C SPECTRUM SATISFY THE RANGE PAIR COUNTING CONDITIONS
C
K = 0
NCYNO = 0
X1 = RES(1)
X2 = RES(2)
X3 = RES(3)
X4 = RES(4)
IND1 = INDEX(1)
IND2 = INDEX(2)
IND3 = INDEX(3)
IND4 = INDEX(4)
J = 4
10 IF (X2 .GT. X1) GO TO 100
IF (X2 .LT. X4 .OR. X3 .GT. X1) GO TO 500
150 IF (X2 .GT. X3) GO TO 151
CALL CYCRES(X3,X2,1.0,IND3)
GO TO 152
151 CALL CYCRES(X2,X3,1.0,IND2)
152 NCYNO = NCYNO + 1
X1 = X1
X2 = X4
IND2 = IND4
IF (J .EQ. LRMAX) GO TO 300
IF ((J + 1) .EQ. LRMAX) GO TO 315
X3 = RES(J+1)
X4 = RES(J+2)
IND3 = INDEX(J+1)
IND4 = INDEX(J+2)
J = J+2
GO TO 10
100 IF (X2 .GT. X4 .OR. X3 .LT. X1) GO TO 500
500 K = K + 1
RES(K) = X1
INDEX(K) = IND1
J = J + 1
IF (J .GT. LRMAX) GO TO 330
X1 = X2
X2 = X3
X3 = X4
X4 = RES(J)
IND1 = IND2
IND2 = IND3
IND3 = IND4
IND4 = INDEX(J)
GO TO 10
300 K = K + 1
RES(K) = X1
RES(K+1) = X2

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CORE 668
CORE 669
CORE 670
CORE 671
CORE 672
CORE 673
CORE 674
CORE 675
CORE 676
CORE 677
CORE 678
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CORE 680
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CORE 712
CORE 713
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CORE 715
CORE 716
CORE 717
CORE 718
CORE 719
CORE 720
CORE 721
CORE 722

```
60 INDEX(K) = IND1
   INDEX(K+1) = IND2
   LRMAX = K + 1
   RETURN
315 K = K + 1
   RES(K) = X1
   RES(K+1) = X2
   RES(K+2) = RES(J+1)
   INDEX(K) = IND1
65 INDEX(K+1) = IND2
   INDEX(K+2) = INDEX(J+1)
   LRMAX = K + 2
   RETURN
330 K = K + 1
   RES(K) = X2
   RES(K+1) = X3
   RES(K+2) = X4
75 INDEX(K) = IND2
   INDEX(K+1) = IND3
   INDEX(K+2) = IND4
   LRMAX = K + 2
   RETURN
   END
```

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CORE 723
CORE 724
CORE 725
CORE 726
CORE 727
CORE 728
CORE 729
CORE 730
CORE 731
CORE 732
CORE 733
CORE 734
CORE 735
CORE 736
CORE 737
CORE 738
CORE 739
CORE 740
CORE 741
CORE 742
CORE 743
CORE 744
CORE 745
```

SUBROUTINE CYCRES

SUBROUTINE CYCRES(Y1,Y2, CYCPF,NSTEPP)
 COMMON/MCYG/CYCLE(200,2),RNECYC(200),NNSTEP(200)
 COMMON/MCDE/L,LIND
 CORE 746
 CORE 747
 CORE 748
 CORE 749
 CORE 750
 CORE 751
 CORE 752
 CORE 753
 CORE 754
 CORE 755
 CORE 756
 CORE 757
 CORE 758
 CORE 759

5 C THIS SUBROUTINE GENERATES CYCLES FOR THE ANALYSIS SPECTRUM FROM DA
 C SUPPLIED BY SUBROUTINE DECS
 C

L = L + 1
 CYCLE(L,1) = Y1
 CYCLE(L,2) = Y2
 RNECYC(L) = CYCPF
 NNSTEP(L) = NSTEPP
 RETURN
 END

10

APPENDIX II

SAMPLE PROBLEM WITH INPUT DATA LISTING

1 DATA DECKS ARE TO BE PROCESSED.
SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE
SEQUENCE ACCOUNTABLE FATIGUE EVALUATION

SPECTRUM FROM B-1 SPECTRUM -- TRUNCATION LEVEL 270,000 CYCLES
 MATERIAL TYPE -- 2219-T851 AL

TENSILE YIELD STRESS (KSI) -- 55.00000
 LCF STRAIN INTERCEPT = .40000
 INVERSE OF COFFIN-MANSON SLOPE -1.83600
 ELASTIC MODULUS = 10000.00000

COEFFICIENTS OF SECOND ORDER LEAST SQUARE FIT OF S-N DATA

LIFE	A(I)	B(I)	C(I)
10** 4	-.00217	.22041	55.82462
10** 5	-.00178	.33210	48.26657
10** 6	-.00149	.46283	39.65455
10** 7	-.00243	.64199	31.70955

UNNOTCHED COUPON S-N DATA DERIVED FROM
 INFORMATION SUPPLIED FROM NORTH AMERICAN

RESIDUAL STRESS RELAXATION FUNCTION

ENEP = C1/(KTSMAX**E1 * KTSMEAN**E2)
 WHLR = C1 = .25000000E+07, E1 = 1.000 AND E2 = 1.000

7 TIMES THROUGH BLOCK OF 44 LOADS

LOAD LIMIT = 26,90000

STEP	TYPE	STMIN	STMAX	ENN
1	1	-.14800	-.02900	1.00000
2	3	.58400	1.00000	1.00000
3	2	.58400	.89500	1.00000
4	1	.58400	.68400	1.00000
5	1	.45500	.58400	1.00000
6	1	.64800	.69300	1.00000
7	1	.39500	.69900	1.00000
8	1	.48800	.56300	3.00000
9	1	.23500	.50900	1.00000
10	1	.50900	.62700	2.00000
11	1	.43400	.50900	1.00000
12	1	.36400	.69000	1.00000
13	1	.13300	.36400	1.00000
14	1	.36400	.48200	6.00000
15	1	.30400	.36400	1.00000
16	1	.30700	.75600	1.00000
17	1	.31600	.52700	1.00000
18	2	.54500	.70800	1.00000
19	1	.33400	.61700	1.00000
20	1	.40100	.51200	1.00000
21	1	.12700	.54800	7.00000
22	1	.22900	.37700	1.00000
23	1	-.00600	.34000	48.00000
24	1	.13000	.24400	1.00000
25	1	.07800	.72300	35.00000
26	1	.11100	.55700	1.00000
27	1	.18700	.37700	9.00000
28	3	.62700	.90000	10.00000
29	1	.62700	.90000	1.00000

29	2	-.12000	.88300	1.00000
30	1	.52700	.74700	1.00000
31	1	-.14900	-.02900	1.00000
32	1	.55700	.84600	1.00000
33	1	.50900	.88000	1.00000
34	1	.33100	.52700	1.00000
35	1	.52700	.61700	19.00000
36	1	.46400	.52700	2.00000
37	1	.48200	.81000	1.00000
38	1	.58100	.68700	1.00000
39	1	.32200	.70200	1.00000
40	1	.39500	.4900	9.00000
41	1	.43100	.59000	5.00000
42	1	.46100	.54500	29.00000
43	1	-.14800	-.02900	1.00000
44	1	-.13600	-.03900	1.00000

BLOCK TYPE TYPE TYPE TYPE TYPE TYPE

1	1	-0	-0	-0
2	1	-0	-0	-0
3	2	1	-0	-0
4	1	-0	-0	-0
5	3	1	-0	-0
6	1	-0	-0	-0
7	1	-0	-0	-0

SPECTRUM FROM B-1 SPECTRUM -- TRUNCATION LEVEL 270,000 CYCLES

AKT = 4.50

RELAXATION CONSTANT C1= 2500000.00

SPECTRUM SUBJECTED TO THE RANGE-PAIR COUNTING TECHNIQUE

FLIGHT OR BLOCK NO. 1

STIMX	STMIN	SIGMAX	SIGMIN	RES	EQRES	ENN	NEP
-1.78	3.98	-3.51	-17.92	0.00	0.00	1.00	.13025831E+05
18.40	15.71	55.00	42.89	0.00	-27.80	1.00	.39342782E+03
15.71	12.24	42.89	27.28	-27.80	-15.69	1.00	.56235689E+03
18.40	17.43	55.00	49.55	-27.75	-28.89	1.00	.36717948E+03
15.80	10.63	55.00	18.20	-28.89	-29.61	1.00	.44621680E+03
15.14	13.13	38.54	29.46	-29.61	-13.15	3.00	.57857259E+03
13.69	6.32	32.20	-9.97	-29.42	-0.61	1.00	.90105168E+03
18.87	13.69	46.54	32.25	-29.36	-20.90	2.00	.47906511E+03
13.69	11.67	32.34	23.26	-29.28	-0.61	1.00	.71090397E+03
13.56	9.79	54.32	14.86	-29.21	-28.52	1.00	.46919213E+03
9.79	3.58	14.86	-13.10	-29.20	0.00	1.00	.18861779E+04
12.97	9.79	29.18	14.90	-29.17	-3.35	6.00	.83680259E+03
9.79	8.18	15.32	8.06	-28.74	0.00	1.00	.14033389E+04
20.34	8.26	55.00	.65	-28.70	-36.51	1.00	.42460525E+03
14.18	8.50	27.28	1.74	-36.51	-8.79	1.00	.76807252E+03
15.60	8.98	38.26	4.00	-36.43	-19.69	1.00	.58153350E+03
13.77	10.79	25.61	12.18	-36.36	-0.98	7.00	.72996103E+03
14.74	3.42	30.61	-20.35	-35.72	-11.34	1.00	.92247814E+03
10.14	6.16	9.97	-7.94	-35.66	0.00	4.80	.14935730E+04
RELAXATION		10.11	-7.81			4.80	
RELAXATION		10.37	-7.55			4.80	
RELAXATION		10.63	-7.29			4.80	
RELAXATION		10.89	-7.03			4.80	
RELAXATION		11.14	-6.77			4.80	
RELAXATION		11.40	-6.52			4.80	
RELAXATION		11.65	-6.27			4.80	
RELAXATION		11.90	-6.02			4.80	
RELAXATION		12.15	-5.77			4.80	
RELAXATION		12.40	-5.52			4.80	
9.15	-.16	8.04	-33.84	-33.12	0.00	1.00	.30047963E+04
6.56	3.50	-3.56	-17.36	-33.09	0.00	35.00	.37392025E+04
RELAXATION		-3.52	-17.32			3.50	
RELAXATION		-3.45	-17.25			3.50	
RELAXATION		-3.38	-17.18			3.50	
RELAXATION		-3.31	-17.11			3.50	
RELAXATION		-3.24	-17.04			3.50	
RELAXATION		-3.17	-16.97			3.50	
RELAXATION		-3.10	-16.90			3.50	
RELAXATION		-3.03	-16.83			3.50	
RELAXATION		-2.96	-16.76			3.50	
RELAXATION		-2.89	-16.69			3.50	
19.45	2.10	55.00	-23.08	-32.39	-32.52	1.00	.58920931E+03
14.98	2.99	34.91	-19.08	-32.52	-12.42	9.00	.91708325E+03
10.14	5.03	13.57	-9.43	-32.07	0.00	10.00	.16047965E+04
20.09	14.18	55.00	26.37	-31.61	-35.42	1.00	.35855054E+03
-1.78	-3.98	-38.93	-53.34	-35.42	0.00	1.00	.13025831E+05
22.76	14.98	55.00	20.02	-35.42	-47.41	1.00	.28748314E+03
23.67	13.69	55.00	10.09	-47.41	-51.52	1.00	.27916148E+03
14.18	8.90	12.27	-11.46	-51.52	-8.79	1.00	.75464468E+03
16.60	14.18	23.29	12.40	-51.39	-19.69	19.00	.48342514E+03
RELAXATION		23.44	12.54			1.90	
RELAXATION		23.72	12.83			1.90	

RELAXATION	24.00	13.11	1.90	50
RELAXATION	24.28	13.39	1.90	51
RELAXATION	24.56	13.66	1.90	52
RELAXATION	24.83	13.94	1.90	53
RELAXATION	25.11	14.21	1.90	54
RELAXATION	25.37	14.48	1.90	55
RELAXATION	25.64	14.75	1.90	56
RELAXATION	25.91	15.01	1.90	57
14.13	12.48	15.14	2.00	58
21.79	12.97	49.68	1.00	59
18.48	15.63	34.83	22.00	60
18.98	8.66	36.76	-9.24	61
12.08	10.63	6.23	-0.31	62
15.87	11.59	24.39	5.14	63
14.66	12.40	19.56	9.39	64
RELAXATION	19.75	9.58	2.90	64
RELAXATION	20.12	9.95	2.90	65
RELAXATION	20.50	10.33	2.90	66
RELAXATION	20.86	10.70	2.90	67
RELAXATION	21.23	11.06	2.90	68
RELAXATION	21.59	11.42	2.90	69
RELAXATION	21.94	11.78	2.90	70
RELAXATION	22.30	12.13	2.90	71
RELAXATION	22.65	12.48	2.90	72
RELAXATION	22.99	12.82	2.90	73
-0.78	-3.98	-40.60	-55.00	74
-1.05	-3.66	-41.80	-53.54	75

STEP	MAXIMUM	SIGMA	MINIMUM	COUNTER K
1	-.351045E+01		-.179154E+02	1.00000
2	.550000E+02		.428950E+02	1.00000
3	.428950E+02		.272795E+02	1.00000
4	.550000E+02		.495527E+02	1.00000
5	.550000E+02		.182008E+02	1.00000
6	.385372E+02		.29585E+02	3.00000
7	.321966E+02		-.971139E+00	1.00000
8	.465387E+02		.322548E+02	2.00000
9	.323357E+02		.232570E+02	1.00000
10	.543191E+02		.148568E+02	1.00000
11	.148601E+02		-.131024E+02	1.00000
12	.291797E+02		.148958E+02	6.00000
13	.153186E+02		.805562E+01	1.00000
14	.550000E+02		.648550E+00	1.00000
15	.272795E+02		.173800E+01	1.00000
16	.582570E+02		.399989E+01	1.00000
17	.256130E+02		.121764E+02	7.00000
18	.306127E+02		-.203494E+02	1.00000
19	.101054E+02		-.781000E+01	4.80000
20	.103674E+02		-.754800E+01	4.80000
21	.106275E+02		-.728793E+01	4.80000
22	.108856E+02		-.702978E+01	4.80000
23	.111419E+02		-.677353E+01	4.80000
24	.113962E+02		-.651917E+01	4.80000
25	.116487E+02		-.626569E+01	4.80000
26	.118993E+02		-.601607E+01	4.80000
27	.121481E+02		-.576729E+01	4.80000
28	.123950E+02		-.552036E+01	4.80000
29	.803921E+01		-.384441E+02	1.00000
30	-.35209E+01		-.173203E+02	3.50000
31	-.344941E+01		-.17491E+02	3.50000
32	-.337838E+01		-.171781E+02	3.50000

THE NUMBER OF PEAKS OR VALLEYS IN THE INPUT LOAD SPECTRUM = 75

33	-.330750E+01	-.171072E+02	3.50000
34	-.323678E+01	-.170365E+02	3.50000
35	-.316621E+01	-.169659E+02	3.50000
36	-.309579E+01	-.168952E+02	3.50000
37	-.302522E+01	-.168246E+02	3.50000
38	-.295440E+01	-.167541E+02	3.50000
39	-.288544E+01	-.166835E+02	3.50000
40	-.550000E+02	-.230772E+02	1.00000
41	-.349057E+02	-.190826E+02	9.00000
42	-.135638E+02	-.343374E+01	10.00000
43	-.550000E+02	-.283690E+02	1.00000
44	-.389348E+02	-.533397E+02	1.00000
45	-.550000E+02	-.200165E+02	1.00000
46	-.550000E+02	-.100902E+02	1.00000
47	-.122694E+02	-.114564E+02	1.00000
48	-.234369E+02	-.125424E+02	1.00000
49	-.237213E+02	-.128268E+02	1.00000
50	-.240031E+02	-.131036E+02	1.00000
51	-.242824E+02	-.133979E+02	1.00000
52	-.245592E+02	-.136676E+02	1.00000
53	-.248355E+02	-.139302E+02	1.00000
54	-.251032E+02	-.142108E+02	1.00000
55	-.253747E+02	-.144832E+02	1.00000
56	-.256416E+02	-.147471E+02	1.00000
57	-.259062E+02	-.150117E+02	1.00000
58	-.151434E+02	-.751720E+01	2.00000
59	-.496805E+02	-.997609E+01	1.00000
60	-.348288E+02	-.219975E+02	1.00000
61	-.367628E+02	-.923622E+01	1.00000
62	-.622539E+01	-.311306E+00	9.00000
63	-.243886E+02	-.514165E+01	5.00000
64	-.197465E+02	-.957830E+01	2.00000
65	-.201228E+02	-.395453E+01	2.00000
66	-.204951E+02	-.103269E+02	2.00000
67	-.212277E+02	-.106952E+02	2.00000
68	-.215882E+02	-.110535E+02	2.00000
69	-.219448E+02	-.114200E+02	2.00000
70	-.222977E+02	-.117766E+02	2.00000
71	-.226467E+02	-.121235E+02	2.00000
72	-.229920E+02	-.124785E+02	2.00000
73	-.405950E+02	-.128238E+02	2.00000
74	-.417990E+02	-.155000E+02	1.00000
75	-.417990E+02	-.153540E+02	1.00000

RANGE PAIR CYCLE COUNTED SPECTRUM

STEP	MAXIMUM	SIGMA	MINIMUM	COUNTER K
1	.351045E+01		.179154E+02	1.00000
2	.550000E+02		.272795E+02	1.00000
3	.550000E+02		.495527E+02	1.00000
4	.550000E+02		.550000E+02	1.00000
5	.321966E+02		.294585E+02	1.00000
6	.465387E+02		.322548E+02	1.00000
7	.465387E+02		.232570E+02	1.00000
8	.323357E+02		.322548E+02	1.00000
9	.543191E+02		.971139E+00	1.00000
10	.148601E+02		.148568E+02	1.00000
11	.291797E+02		.148958E+02	5.00000
12	.291797E+02		.805562E+01	1.00000
13	.153186E+02		.148958E+02	1.00000
14	.550000E+02		.131024E+02	1.00000
15	.272796E+02		.173300E+01	1.00000
16	.382570E+02		.648550E+00	1.00000
17	.306127E+02		.121764E+02	1.00000
18	.101054E+02		.781300E+01	5.00000
19	.103674E+02		.754800E+01	5.00000
20	.106275E+02		.728735E+01	5.00000
21	.108856E+02		.702978E+01	5.00000
22	.111419E+02		.677353E+01	5.00000
23	.113962E+02		.654917E+01	5.00000
24	.116487E+02		.626669E+01	5.00000
25	.118932E+02		.601607E+01	5.00000
26	.121481E+02		.576729E+01	5.00000
27	.123950E+02		.552036E+01	4.00000
28	.123950E+02		.203494E+02	1.00000
29	.803921E+01		.552036E+01	1.00000
30	.344941E+01		.173303E+02	4.00000
31	.337838E+01		.172491E+02	4.00000
32	.330750E+01		.171781E+02	4.00000
33	.323678E+01		.171072E+02	4.00000
34	.316621E+01		.170365E+02	4.00000
35	.309579E+01		.169599E+02	4.00000
36	.302552E+01		.168955E+02	4.00000
37	.295540E+01		.168232E+02	4.00000
38	.288544E+01		.167515E+02	4.00000
39	.550000E+02		.166851E+02	4.00000
40	.135588E+02		.190826E+02	4.00000
41	.550000E+02		.943374E+01	1.00000
42	.550000E+02		.338441E+02	10.00000
43	.550000E+02		.201655E+02	1.00000
44	.550000E+02		.533397E+02	1.00000
45	.237213E+02		.125424E+02	1.00000
46	.240031E+02		.128268E+02	2.00000
47	.242824E+02		.131096E+02	2.00000
48	.245592E+02		.133879E+02	2.00000
49	.248335E+02		.136647E+02	2.00000
50	.251053E+02		.139390E+02	2.00000
51	.253747E+02		.142108E+02	2.00000
52	.256416E+02		.144802E+02	2.00000
53	.259062E+02		.147471E+02	2.00000
54	.259062E+02		.150117E+02	1.00000
55	.259062E+02		.751720E+01	1.00000
56	.151434E+02		.150117E+02	1.00000

57	.151434E+02	.751720E+01	1.00000
58	.496805E+02	-.114564E+02	1.00000
59	.348288E+02	.219975E+02	1.00000
60	.367628E+02	.397609E+01	1.00000
61	.243886E+02	.514165E+01	4.00000
62	.243886E+02	-.311306E+00	1.00000
63	.197465E+02	.357830E+01	3.00000
64	.201228E+02	.395499E+01	3.00000
65	.204951E+02	.103289E+02	3.00000
66	.208634E+02	.106952E+02	3.00000
67	.212277E+02	.110595E+02	3.00000
68	.215882E+02	.114200E+02	3.00000
69	.219448E+02	.117766E+02	3.00000
70	.222977E+02	.121295E+02	3.00000
71	.226467E+02	.124785E+02	3.00000
72	.229920E+02	.128286E+02	2.00000
73	.229920E+02	.514155E+01	1.00000
74	-.4417990E+02	-.535408E+02	1.00000

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

STEP	PLASTIC STRAIN	MAX OR MIN	DAMAGE
2	.00636	MAX	.589085E-03
4	.00035	MAX	.236733E-05
5	.00022	MAX	.105771E-05
14	.00249	MAX	.891548E-04
22	.00004	MAX	.498022E-07
25	.00123	MAX	.242930E-04
27	.00420	MAX	.233133E-03
28	.00156	MAX	.379281E-04
38	.00063	MIN	.718916E-05

DAMAGE FROM PLASTIC STRAINS= .98426328E-03

SIGMAX	SIGMIN	RNCYC	CYCLES	ENNYCYC
-3.51	-17.92	1.	.1000000E+10	.1000000E-08
55.00	27.28	1.	.14807856E+06	.87531722E-05
55.00	49.55	1.	.69900694E+09	.14306010E-08
55.00	-55.00	1.	.1000000E+05	.1000000E-03
32.20	29.46	1.	.46901918E+12	.21321090E-11
46.54	32.25	1.	.11536870E+09	.86678619E-08
46.54	23.26	1.	.51785551E+07	.19310405E-06
32.34	32.25	1.	.36726287E+13	.27228454E-12
54.32	-.97	1.	.14931857E+05	.66970305E-04
14.86	14.86	1.	.50066416E+12	.13973463E-11
29.18	14.90	5.	.1265797E+10	.39476395E-08
29.18	8.06	1.	.14052164E+09	.71163418E-08
15.32	14.90	1.	.42187482E+12	.23703714E-11
55.00	-13.10	1.	.1000000E+05	.1000000E-03
27.28	1.74	1.	.53004668E+08	.18836263E-07
38.26	.65	1.	.27966528E+09	.57293423E-06
30.61	12.18	1.	.1000000E+10	.35757031E-08
10.11	-7.81	5.	.1000000E+10	.5000000E-08
10.37	-7.55	5.	.1000000E+10	.5000000E-08
10.63	-7.29	5.	.1000000E+10	.5000000E-08
10.89	-7.03	5.	.1000000E+10	.5000000E-08
11.14	-6.77	5.	.1000000E+10	.5000000E-08
11.40	-6.52	5.	.56054816E+09	.89198403E-08
11.65	-6.27	5.	.56038584E+09	.83144700E-08
11.90	-6.02	5.	.56115089E+09	.83102593E-08
12.15	-5.77	5.	.56146456E+09	.83052816E-08
12.40	-5.52	4.	.56151397E+09	.71235864E-08
12.40	-20.35	1.	.27376289E+08	.36527961E-07
8.04	-5.52	1.	.1000000E+10	.1000000E-08
-3.52	-17.32	4.	.1000000E+10	.40000000E-08
-3.45	-17.25	4.	.1000000E+10	.40000000E-08

-3.38	-17.18	4.	.10000000E+10	.40000000E-08
-3.31	-17.11	4.	.10000000E+10	.40000000E-08
-3.24	-17.04	4.	.10000000E+10	.40000000E-08
-3.17	-16.97	4.	.10000000E+10	.40000000E-08
-3.10	-16.90	4.	.10000000E+10	.40000000E-08
-3.03	-16.83	4.	.10000000E+10	.40000000E-08
-2.96	-16.76	4.	.10000000E+10	.40000000E-08
-2.89	-16.69	4.	.10000000E+10	.40000000E-08
55.00	-19.08	10.	.16603817E+09	.60227117E-07
55.00	-33.84	1.	.10000000E+05	.10000000E-03
55.00	28.02	1.	.635+6690E+05	.15736461E-04
55.00	-53.34	1.	.10000000E+05	.10000000E-03
23.44	12.54	2.	.52786328E+10	.3788599E-09
23.72	12.83	2.	.52807566E+10	.37875000E-09
24.00	13.11	2.	.52715248E+10	.37905745E-09
24.28	13.39	2.	.52733259E+10	.37939687E-09
24.56	13.66	2.	.52762450E+10	.37985421E-09
24.83	13.94	2.	.52715248E+10	.38042931E-09
25.11	14.21	2.	.52572184E+10	.38112208E-09
25.37	14.48	2.	.52476624E+10	.38112208E-09
25.64	14.75	2.	.52365271E+10	.38112208E-09
25.91	15.01	1.	.37539154E+09	.266603419E-08
25.91	7.52	1.	.48307951E+12	.20700526E-11
15.14	15.01	1.	.15515977E+11	.64449697E-10
15.14	7.52	1.	.24152055E+05	.41404344E-04
49.68	-11.46	1.	.15814258E+10	.63234078E-09
34.83	22.00	1.	.15038306E+08	.66496353E-07
36.76	9.98	1.	.31310615E+09	.12775220E-07
24.39	5.14	4.	.67337984E+10	.12915152E-07
19.75	9.58	3.	.67337984E+10	.44551379E-09
20.12	9.95	3.	.67738754E+10	.44248601E-09
20.50	10.33	3.	.58230691E+10	.43968484E-09
20.86	10.70	3.	.68633296E+10	.43710563E-09
21.23	11.06	3.	.69006119E+10	.43474405E-09
21.59	11.42	3.	.69348759E+10	.43259609E-09
21.94	11.78	3.	.69660840E+10	.43065799E-09
22.30	12.13	3.	.69942085E+10	.42892530E-09
22.65	12.48	3.	.70192214E+10	.42739783E-09
22.99	12.82	2.	.70411021E+10	.42404644E-09
22.99	5.14	1.	.49378912E+09	.20251560E-08
-41.80	-53.54	1.	.10000000E+10	.10000000E-08

DAMAGE PER THIS SET= .16162546E-02

TOTAL ENN/CYC =, .16162546E-02

FLIGHT OR BLOCK NO.	STHIN	SIGMAX	SIGMIN	RES	EQRES	ENN	NEP	TOTAL ENN/CYC =,	DAMAGE PER THIS SET=
18.78	-3.98	-40.58	-54.99	-37.07	0.00	1.00	.13025831E+05	1	.16162546E-02
18.40	15.71	45.73	33.63	-37.07	-27.80	1.00	.39342782E+03	2	.16162546E-02
15.71	12.24	33.68	18.07	-37.01	-15.69	1.00	.56235689E+03	3	.16162546E-02
18.64	17.43	46.96	41.52	-36.92	-28.89	1.00	.36717948E+03	4	.16162546E-02
18.80	10.63	47.74	10.94	-36.87	-29.61	1.00	.44621680E+03	5	.16162546E-02
15.14	13.13	31.31	22.24	-36.84	-13.15	3.00	.57667259E+03	6	.16162546E-02
13.69	6.32	25.06	-8.11	-36.55	-6.61	1.00	.90105168E+03	7	.16162546E-02
16.87	13.69	39.42	25.14	-36.48	-20.90	2.00	.47906511E+03	8	.16162546E-02
13.69	11.67	25.29	16.21	-36.33	-6.61	1.00	.71090397E+03	9	.16162546E-02
18.56	9.79	47.29	7.83	-36.23	-28.52	1.00	.46919213E+03	10	.16162546E-02
9.79	3.58	7.87	-20.10	-36.20	0.00	1.00	.18851779E+04	11	.16162546E-02
12.97	9.79	22.20	7.91	-36.15	-3.35	6.00	.83680259E+03	12	.16162546E-02
9.79	8.18	8.45	1.19	-35.61	0.00	1.00	.14033889E+04	13	.16162546E-02
20.34	8.26	55.00	.65	-35.56	-36.51	1.00	.42460525E+03	14	.16162546E-02
14.18	8.50	27.28	1.74	-36.51	-8.79	1.00	.76807252E+03	15	.16162546E-02
16.60	8.98	38.26	4.00	-36.43	-19.69	1.00	.58153350E+03	16	.16162546E-02
13.77	10.79	25.61	12.18	-36.36	-6.98	7.00	.72996103E+03	17	.16162546E-02

14.74	3.42	30.61	-20.35	-35.72	-11.34	1.00	.92247814E+03	18	13
10.14	6.16	9.97	-7.94	-35.66	0.00	4.80	.14935730E+04	19	19
RELAXATION		10.11	-7.81			4.80			19
RELAXATION		10.37	-7.55			4.80			20
RELAXATION		10.63	-7.29			4.80			21
RELAXATION		10.89	-7.03			4.80			22
RELAXATION		11.14	-6.77			4.80			23
RELAXATION		11.40	-6.52			4.80			24
RELAXATION		11.65	-6.27			4.80			25
RELAXATION		11.90	-6.02			4.80			25
RELAXATION		12.15	-5.77			4.80			27
RELAXATION		12.40	-5.52			4.80			28
9.15	-1.16	8.04	-33.84	-33.12	0.00	1.00	.30047963E+04	20	23
0.56	3.50	-3.56	-17.36	-33.09	0.00	35.00	.37392025E+04	21	30
RELAXATION		-3.52	-17.32			3.50			30
RELAXATION		-3.45	-17.25			3.50			31
RELAXATION		-3.38	-17.18			3.50			32
RELAXATION		-3.31	-17.11			3.50			33
RELAXATION		-3.24	-17.04			3.50			34
RELAXATION		-3.17	-16.97			3.50			35
RELAXATION		-3.10	-16.90			3.50			36
RELAXATION		-2.96	-16.76			3.50			37
RELAXATION		-2.89	-16.69			3.50			38
13.45	2.10	55.00	-23.08	-32.39	-32.52	1.00	.58209331E+03	22	39
14.98	2.99	34.91	-19.08	-32.52	-12.42	9.00	.9170325E+03	23	40
10.14	5.03	13.57	-9.43	-32.07	0.00	10.00	.16047965E+04	24	41
20.09	14.18	55.00	28.37	-31.61	-35.42	1.00	.35855054E+03	25	42
-7.8	-3.98	-38.93	-53.34	-35.42	0.00	1.00	.13025831E+05	26	44
22.76	14.98	55.00	20.02	-35.42	-47.41	1.00	.28748314E+03	27	45
23.67	13.69	55.00	10.09	-47.41	-51.52	0.00	.27916148E+03	28	46
14.18	8.90	12.27	-11.46	-51.52	-8.79	1.00	.75464468E+03	29	47
15.60	14.18	23.29	12.40	-51.39	-19.69	19.00	.48342514E+03	30	48
RELAXATION		23.44	12.54			1.90			48
RELAXATION		23.72	12.83			1.90			49
RELAXATION		24.00	13.11			1.90			50
RELAXATION		24.28	13.39			1.90			51
RELAXATION		24.56	13.66			1.90			52
RELAXATION		24.83	13.94			1.90			53
RELAXATION		25.11	14.21			1.90			54
RELAXATION		25.37	14.48			1.90			55
RELAXATION		25.64	14.75			1.90			56
RELAXATION		25.91	15.01			1.90			57
14.18	12.48	15.14	7.52	-48.65	-8.79	2.00	.65336542E+03	31	53
21.79	12.97	49.68	9.98	-48.37	-43.05	1.00	.32605652E+03	32	59
18.48	15.63	34.83	22.00	-48.33	-28.16	1.00	.39170979E+03	33	60
10.38	8.66	36.76	-9.24	-48.21	-29.98	1.00	.47468264E+03	34	61
12.03	10.63	6.23	-31	-48.13	0.00	9.00	.90043347E+03	35	62
15.87	11.59	24.39	5.14	-47.03	-16.42	5.00	.56645141E+03	36	63
14.66	12.40	19.56	9.339	-46.41	-10.97	29.00	.62236611E+03	37	64
RELAXATION		19.75	9.58			2.90			64
RELAXATION		20.12	9.95			2.90			65
RELAXATION		20.50	10.33			2.90			66
RELAXATION		20.86	10.70			2.90			67
RELAXATION		21.23	11.06			2.90			68
RELAXATION		21.59	11.42			2.90			69
RELAXATION		21.94	11.78			2.90			70
RELAXATION		22.30	12.13			2.90			71
RELAXATION		22.65	12.48			2.90			72
RELAXATION		22.99	12.82			2.90			73
-7.8	-3.98	-40.60	-55.00	-42.81	0.00	1.00	.13025831E+05	38	74
-1.05	-3.66	-41.80	-53.54	-37.08	0.00	1.00	.14337171E+05	39	75

LOCAL STRESSES AND PLASTIC STRAINS W/RESULTING FATIGUE LIFE

STEP 14 22 25 27 28 38

PLASTIC STRAIN
 .00032
 .00004
 .00123
 .00420
 .00156
 .00063

DAMAGE FROM PLASTIC STRAINS= .30462764E-03

DAMAGE
 .202856E-05
 .498022E-07
 .242990E-04
 .233135E-03
 .379281E-04
 .718916E-05

STEP	SIGMAX	SIGMIN	RNCYC	CYCLES	ENNV/CYC
14	-40.58	-54.99	1.	.1000000E+10	.1000000E-08
22	45.73	18.07	1.	.2403377E+07	.4150493E-06
25	33.68	33.63	1.	.4320607E+13	.23144895E-12
27	46.96	41.52	1.	.83661145E+11	.42263381E-10
28	47.74	-20.10	1.	.19730666E+05	.50554416E-04
38	25.06	22.24	1.	.27937764E+12	.35793846E-11
	39.42	25.14	1.	.50516846E+09	.13795377E-08
	39.42	16.21	1.	.24432813E+08	.40928567E-07
	25.29	25.14	1.	.15753982E+13	.63476016E-12
	47.23	-8.11	1.	.58801786E+05	.17006285E-04
	7.87	7.83	1.	.1000000E+10	.1000000E-08
	22.20	7.91	5.	.15461310E+10	.32338787E-08
	22.20	1.19	1.	.21206560E+09	.47155219E-08
	8.45	7.91	1.	.1000000E+10	.1000000E-08
	55.00	-53.34	1.	.1000000E+05	.1886263E-07
	27.28	1.74	1.	.53004668E+08	.1886263E-07
	38.23	.65	1.	.17454308E+07	.37293429E-06
	30.61	12.18	1.	.27956528E+09	.35757031E-08
	10.11	-7.81	5.	.1000000E+10	.5000000E-08
	10.37	-7.55	5.	.1000000E+10	.5000000E-08
	10.63	-7.29	5.	.1000000E+10	.5000000E-08
	10.89	-7.03	5.	.1000000E+10	.5000000E-08
	11.14	-6.77	5.	.56054816E+09	.39198403E-08
	11.40	-6.52	5.	.56088584E+09	.89144700E-08
	11.65	-6.27	5.	.36115089E+09	.89102593E-08
	11.90	-6.02	5.	.56134367E+09	.89071993E-08
	12.15	-5.77	5.	.36116456E+09	.89052816E-08
	12.40	-5.52	4.	.56151397E+09	.71235984E-08
	12.40	-20.35	1.	.27376289E+08	.36527361E-07
	8.04	-5.52	1.	.1000000E+10	.1000000E-08
	-3.52	-17.32	4.	.1000000E+10	.4000000E-08
	-3.45	-17.25	4.	.1000000E+10	.4000000E-08
	-3.38	-17.18	4.	.1000000E+10	.4000000E-08
	-3.31	-17.11	4.	.1000000E+10	.4000000E-08
	-3.24	-17.04	4.	.1000000E+10	.4000000E-08
	-3.17	-16.97	4.	.1000000E+10	.4000000E-08
	-3.10	-16.90	4.	.1000000E+10	.4000000E-08
	-3.03	-16.83	4.	.1000000E+10	.4000000E-08
	-2.96	-16.76	4.	.1000000E+10	.4000000E-08
	-2.89	-16.69	4.	.1000000E+10	.4000000E-08
	55.00	-19.08	10.	.16603817E+09	.60227117E-07
	55.00	-33.84	1.	.1000000E+05	.1000000E-03
	55.00	20.02	1.	.63546690E+05	.15736461E-04
	55.00	-55.00	1.	.1000000E+05	.1000000E-03
	23.44	12.54	2.	.52786328E+10	.3788539E-09
	23.72	12.83	2.	.52835280E+10	.37875000E-09
	24.00	13.11	2.	.52807566E+10	.37873361E-09
	24.28	13.39	2.	.52793259E+10	.3783625E-09
	24.56	13.66	2.	.52762450E+10	.37905745E-09
	24.83	13.94	2.	.52651779E+10	.37939687E-09
	25.11	14.21	2.	.52572184E+10	.37985421E-09
	25.37	14.48	2.	.52476624E+10	.38042931E-09
	25.64	14.75	2.	.52365271E+10	.38112208E-09
	25.91	15.01	1.	.37589154E+09	.19096626E-09
	25.91	7.52	1.	.37589154E+09	.266003419E-08

FLIGHT OR BLOCK NO.	SIGMAX	SIGMIN	SIGMIN	RES	EQRES	ENN	NEP	DAMAGE PER THIS SET=	ENM/CYC =;
15.14	15.01	1.0	1.0	1.0	1.0	1.0	1.0	.83070929E-03	.20700266E-11
15.14	7.52	1.0	1.0	1.0	1.0	1.0	1.0	.24469639E-02	.54449697E-10
49.68	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		.41040344E-04
34.83	22.00	1.0	1.0	1.0	1.0	1.0	1.0		.63234078E-09
36.76	9.98	1.0	1.0	1.0	1.0	1.0	1.0		.66496893E-07
24.39	9.14	1.0	1.0	1.0	1.0	1.0	1.0		.12915162E-07
24.39	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		.44551379E-09
19.75	9.59	1.0	1.0	1.0	1.0	1.0	1.0		.44248601E-09
20.12	9.95	1.0	1.0	1.0	1.0	1.0	1.0		.43710563E-09
20.50	10.33	1.0	1.0	1.0	1.0	1.0	1.0		.43259609E-09
20.86	10.70	1.0	1.0	1.0	1.0	1.0	1.0		.43065799E-09
21.23	11.06	1.0	1.0	1.0	1.0	1.0	1.0		.42892630E-09
21.59	11.42	1.0	1.0	1.0	1.0	1.0	1.0		.42739783E-09
21.94	11.78	1.0	1.0	1.0	1.0	1.0	1.0		.28404544E-09
22.30	12.13	1.0	1.0	1.0	1.0	1.0	1.0		.20251560E-08
22.65	12.48	1.0	1.0	1.0	1.0	1.0	1.0		.10000000E-08
22.99	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
22.99	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
-41.80	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
3									
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.78	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.13	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.48	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.78	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.13	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.48	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.78	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.13	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.48	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.78	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.13	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.48	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.78	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.13	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	12.48	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	12.82	1.0	1.0	1.0	1.0	1.0	1.0		
2.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	5.14	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-53.54	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	15.01	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	7.52	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	-11.46	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	22.00	1.0	1.0	1.0	1.0	1.0	1.0		
1.0	9.98	1.0	1.0	1.0	1.0	1.0	1.0		
4.0	9.14	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	-3.1	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.59	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	9.95	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.33	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	10.70	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.06	1.0	1.0	1.0	1.0	1.0	1.0		
3.0	11.42	1.0	1.0	1.0					

-7.56	1.	10000000E+10	10000000E-08
6.79	5.	10000000E+10	10000000E-08
6.79	1.	10000000E+10	10000000E-08
-6.70	1.	10000000E+10	10000000E-08
-35.52	1.	26004162E+05	38425383E-04
13.19	1.	97971982E+08	10207000E-07
35.22	1.	65291215E+06	15315996E-05
11.74	7.	19249496E+10	36364588E-08
17.04	1.	66014698E+08	15148142E-07
-3.02	5.	10000000E+10	50000000E-09
-2.66	5.	10000000E+10	50000000E-08
-2.31	5.	10000000E+10	50000000E-08
-1.96	5.	10000000E+10	50000000E-08
-1.60	5.	10000000E+10	50000000E-08
-1.26	5.	10000000E+10	50000000E-08
-.91	5.	10000000E+10	50000000E-08
-.57	5.	10000000E+10	50000000E-08
-.23	4.	10000000E+10	50000000E-08
-.23	1.	10000000E+10	40000000E-08
-.23	1.	10000000E+10	10000000E-09
-4.54	4.	10000000E+10	10000000E-08
-16.07	4.	10000000E+10	40000000E-08
-29.87	4.	10000000E+10	40000000E-08
-29.77	4.	10000000E+10	40000000E-08
-29.68	4.	10000000E+10	40000000E-08
-15.88	4.	10000000E+10	40000000E-08
-15.78	4.	10000000E+10	40000000E-08
-15.68	4.	10000000E+10	40000000E-08
-29.48	4.	10000000E+10	40000000E-08
-29.38	4.	10000000E+10	40000000E-08
-15.49	4.	10000000E+10	40000000E-08
-29.29	4.	10000000E+10	40000000E-08
-29.19	4.	10000000E+10	40000000E-08
-29.09	4.	10000000E+10	40000000E-08
-15.29	4.	10000000E+10	40000000E-08
-15.20	4.	10000000E+10	40000000E-08
-42.84	4.	10000000E+10	40000000E-08
1.72	1.	24336594E+05	41006136E-04
60.71	10.	10000000E+10	10000000E-07
55.60	1.	10000000E+10	10000000E-03
55.00	1.	10000000E+10	10000000E-03
-55.00	1.	63546690E+05	15736461E-04
23.44	2.	10000000E+10	10000000E-03
23.72	2.	52796328E+10	3788599E-09
12.83	2.	52805280E+10	3787500E-09
13.11	2.	52807566E+10	37873361E-09
13.39	2.	52733259E+10	37883625E-09
13.66	2.	52752450E+10	37905745E-09
13.94	2.	52715248E+10	37939687E-09
14.21	2.	52651779E+10	37985421E-09
14.48	2.	52572184E+10	38042931E-09
14.75	2.	52476624E+10	38112208E-09
15.01	1.	52365271E+10	19096626E-09
7.52	1.	37539154E+09	26603419E-08
15.01	1.	48307951E+12	20700326E-11
7.52	1.	15515977E+11	64449697E-10
-11.46	1.	24152055E+05	41404344E-04
22.00	1.	15814258E+10	63234078E-09
9.98	1.	15038306E+08	66496853E-07
5.14	4.	31310615E+09	12775220E-07
-31	3.	77428372E+08	12915162E-07
9.58	3.	6737984E+10	44551379E-09
9.95	3.	67798754E+10	44248601E-09
10.33	3.	68230691E+10	43968484E-09
10.70	3.	68633296E+10	43710563E-09
20.86	3.	6906119E+10	43474405E-09
21.23	3.	69348755E+10	43259609E-09
21.59	3.	6960846E+10	43065799E-09
11.78	3.	69942085E+10	42892630E-09
12.13	3.	70192214E+10	42739783E-09
12.48	3.	70411021E+10	42404644E-09
12.82	2.	49378912E+09	20251560E-08
5.14	1.	10000000E+10	10000000E-08
-41.80	1.	10000000E+10	10000000E-08

DAMAGE PER THIS SET= .12645622E-02
TOTAL ENN/CYC =, .37115261E-02
DAMAGE FROM PLASTIC STRAINS= .30462764E-03
DAMAGE PER THIS SET= .83070929E-03
TOTAL ENN/CYC =, .45422354E-02
DAMAGE FROM PLASTIC STRAINS= .22665690E-02
DAMAGE PER THIS SET= .26509046E-02
TOTAL ENN/CYC =, .71931400E-02
DAMAGE FROM PLASTIC STRAINS= .30462764E-03
DAMAGE PER THIS SET= .83070929E-03
TOTAL ENN/CYC =, .80238493E-02
DAMAGE FROM PLASTIC STRAINS= .30462764E-03
DAMAGE PER THIS SET= .83070929E-03
TOTAL ENN/CYC =, .88545586E-02

FLIGHT OR BLOCK NO. 4

FLIGHT OR BLOCK NO. 5

FLIGHT OR BLOCK NO. 6

FLIGHT OR BLOCK NO. 7

3-1 SPECTRUM -- TRUNCATION LEVEL 270,000 CYCLES

219-1851 ML 55
 .00217 22041
 .00178 33210
 .01149 46283
 .01243 64199
 25.00000
 4.5

55.82462
 48.26597
 39.66455
 31.70955
 1.
 1.

-1.836
 NORTH AMERICAN 4
 NORTH AMERICAN 5
 NORTH AMERICAN 6
 NORTH AMERICAN 7
 10000.

10	7	44	3						
1	26.9								
2	3	584	1	148					
3	2	584	1	825					
4	1	584	1	684					
5	1	455	1	534					
6	1	643	1	643					
7	1	395	1	699					
8	1	488	3	563					
9	1	235	1	509					
10	1	519	2	627					
11	1	434	1	509					
12	1	364	1	63					
13	1	133	1	354					
14	1	354	6	482					
15	1	304	1	364					
16	1	307	1	756					
17	1	316	1	527					
18	2	545	1	778					
19	1	334	1	617					
20	1	421	7	512					
21	1	127	1	548					
22	1	229	1	377					
23	1	006	1	340					
24	1	13	1	244					
25	1	178	1	723					
26	1	111	9	557					
27	1	187	1	377					
28	3	527	1	941					
29	2	12	1	883					
30	1	527	1	747					
31	1	148	1	029					
32	1	557	1	846					
33	1	509	1	830					
34	1	331	1	527					
35	1	527	19	617					
36	1	464	2	527					
37	1	432	1	81					
38	1	591	1	687					
39	1	322	1	702					
40	1	335	9	449					
41	1	431	5	59					
42	1	461	29	545					
43	1	148	1	029					
44	1	135	1	039					

1 1
2 1
3 2 1
4 1
5 3 1
6 1
7 1

60

APPENDIX III

LIST OF COMPUTER PROGRAM SYMBOLS AND DEFINITIONS

- A Coefficient of the x^2 term in the equation of a line on a constant life fatigue diagram where minimum stress is x and maximum stress is y . ($R = Ax^2 + Bx + C - y$)
- AA An assigned value of +1. or -1.
- AAA A stress used in the calculation of plastic strain.
- ABDIF The absolute value of DIF.
- ABM The absolute value of ASMAX or of ASMIN, as assigned.
- ABMAX The absolute value of ASMAX.
- ABMEAN The absolute value of ASMEAN.
- ABMIN The absolute value of ASMIN.
- ABR4 The absolute value of R(4).
- ABR7 The absolute value of R(7).
- ABS The name of a routine calling for the absolute value of a quantity.
- AKT Stress concentration factor, K_t
- ASMAX The product (AKT) (STMAX)
- ASMEAN The quantity (ASMAX + ASMIN)/2
- ASMIN The product (AKT) (STMIN)
- AVSGMN Average value of SIGMIN over an interval.
- AVSGMX Average value of SIGMAX over an interval.

B Coefficient of the x term. (See A.)

BBB A stress used in the calculation of plastic strain.

C The constant. (See A.)

COFMAN Inverse of the Coffin-Manson slope.

CYCINT The number of cycles in an interval.

CYCLES The calculated number of cycles expected to be indicated
on a constant life fatigue diagram for the applied combination
of maximum and minimum stress.

C1 The Residual Stress Relaxation Constant (See ENEP.)

DAM Damage.

DECK Decimal or real value of integer K after conversion.

DEL2 A portion of a least-squares-method solution.

DIF The difference between residual stress and equilibrium
residual stress. (RES(I) - EQRES)

DO2 A portion of a least-squares-method solution.

DUMMY A variable used in the calculation of the number of cycles
to be considered as an interval for relaxation determination.

ELMOD The elastic modulus.

EN The number of cycles from the beginning of the relaxation
process to the end of the current interval.

ENEP The number of cycles required for overload residual stress
effect to return to within one-tenth of its original difference
from equilibrium conditions.

$$(N_{ep} = C1 / (ABM)^{E1} (ABMEAN)^{E2})$$

ENN The number of applied cycles at a load level.

ENNCYC The ratio of the number of applied cycles to the number
of cycles to failure. (ENN/CYCLES)

EPSD LCF strain intercept.

EQRES Equilibrium residual stress.

EX An exponential function depicting the relaxation of
residual stress.

EXP The name of a routine calling for the exponential value
of a quantity.

EXPO An exponent. The power of 10 which indicates the number
of cycles to failure.

E1 } Residual stress Relaxation Exponents.
E2 }

FLOAT The name of a routine calling for integer-to-real conversion.

I A variable subscript.

IBLOCK The identifying number of a block the blocks being
numbered consecutively from 1 to NBLOCK.

IFIX The name of a routine calling for real-to-integer conversion.

IN The number of steps input to the range pair counting
subroutine.

IPRINT Value controlling the WRITE statements.

IRAIN A counter.

IRPCM Value controlling entry into the range pair counting
subroutine.

ISTEP The identifying step number, the steps being numbered
 from 1 to NLEVEL.

ITYPE The identifying type number, the types being numbered
 from 1 to NTYPE.

J A variable subscript.

JA Value of +1 or 0, as assigned for branch determination.

JB Value of -1 or 0, as assigned for branch determination.

JJ An index variable.

JJJ An index variable.

JKL An index variable.

K An index variable.

KK An index variable.

KPMAX The number of steps output from the range pair counting
 subroutine.

L An index variable.

LMN An index variable.

M An index variable.

N An index variable with values of N=4-7 indicating the
 power of 10, and thus identifying a particular life cycle
 curve.

NBLOCK The total number of times to execute a block of loads.

NDECK The number of data decks to be run sequentially.

NFLAG An integer used as a counter.

NFLAG2 An integer used as a counter.

NLEVEL The total number of steps, or levels, of loads in a block.

NN A subscripted variable used to indicate which types of loads are experienced in which blocks.

NTYPE The total number of different types.

PLSTRA Plastic strain.

R Residue term in damage calculation.

RES Residual stress.

RNCYC The number of cycles for a level after exiting the range pair counting subroutine.

SIGMAX Maximum stress.

SIGMIN Minimum stress.

STMAX Maximum applied stress.

STMIN Minimum applied stress.

SUMDEL Summation of damage for a flight.

SUMENN Accumulated total of applied cycles. (Summation of ENN).

SUMNC Accumulated cycle ratio. (Summation of ENN/CYCLES).

SUMR Summation of $R(N)$, $N=4,7$.

SUMRN Summation of $NR(N)$, $N=4,7$.

SUMR2 Summation of $R(N)^2$, $N=4,7$.

SUMR2N Summation of $NR(N)^2$, $N=4,7$.

SUMR3 Summation of $R(N)^3$, $N=4,7$.

SUMR4 Summation of $R(N)^4$, $N=4,7$.

TITLE1,TITLE2 Identification of the source of the SN data.

TLL Tensile load limit.

TM1, TM2 Material type.

TTYS One-fifth of tensile yield stress.
TYS Tensile yield stress.
T1,T2,T3,T4,T5,T6,T7,T8 Test identifying information.
X Variable equivalent to SIGMIN.
Y Variable equivalent to SIGMAX.

APPENDIX IV

FATIGUE LIFE INPUT DATA FOR SEVERAL MATERIALS

MATERIAL	YIELD STRESS	STRAIN INTERCEPT	INVERSE OF SLOPE	LIFE, 10^I	S-N LIFE COEFFICIENTS		
					A(I)	B(I)	C(I)
2024-T4	58.	0.4 (2)	-1.836 (2)	4	-.0020	.2091	62.6 (3)
				5	-.0032	.4366	51.4
				6	-.0035	.6207	42.2
				7	-.0042	.7003	36.1
2219-T851	55.	0.4 (2)	-1.836 (2)	4	-.0022	.2204	55.8 (4)
				5	-.0018	.3320	48.3
				6	-.0015	.4628	39.7
				7	-.0024	.6420	31.7
7075-T6	72.	0.4 (1)	-1.836 (1)	4	-.0020	.2801	71.7 (3)
				5	-.0022	.5154	56.3
				6	-.0014	.6141	44.6
				7	-.0013	.6838	38.1
RQC-100	125.	0.54 (5)	-1.493 (5)	4	0.0	.2136	98.3 (5)
				5	0.0	.2927	88.5
				6	0.0	.3669	79.1
				7	0.0	.4376	70.3
Man-Ten	55.	1.11 (5)	-1.667 (5)	4	0.0	.2257	63.5 (5)
				5	0.0	.3520	53.1
				6	0.0	.4669	43.7
				7	0.0	.5678	35.4
4340 Steel	160.	0.4 (2)	-1,836 (2)	4	-.0002	.2567	162.4 (3)
				5	-.0007	.5248	126.9
				6	-.0005	.5557	113.5
				7	-.0005	.5557	108.5
Ti-6-4	158.	0.4 (2)	-1.836 (2)	4	-.0009	.2368	154.2 (3)
				5	-.0006	.4640	110.3
				6	-.0000	.4650	88.9
				7	.0001	.4752	84.2

- (1) Data from Endo, T., and Morrow, J., NAEC-ASL-1105, Naval Air Engineering Center, Philadelphia, PA, June 1966.
- (2) Data not available - Source 1 data considered typical.
- (3) Derived from Metallic Materials and Elements for Aerospace Vehicles Structures, MIL-HDBK-5A, Dept. of Defense, Washington, D.C., February 1966.
- (4) Information supplied from Rockwell International.
- (5) Information supplied by Society of Automotive Engineers Cumulative Damage Division, Courtesy of Mr. H.R. Jaekel.