AFAPL-TR-70-44 Part III

AD 749129

A STATE AND A STATE OF A

4

FAILURE MECHANISMS AND ACCELERATED LIFE TESTS OF NICKEL-CADMIUM BATTERIES

J. McCallum

B/ TELLE Columnus Laboratories

and

3. H. Miller Air Force Aero Propulsion Laboratory

TECHNICAL REPORT AFAPL-TR-70-44, Part III

August 1972

Reproduced by NATIONAL TECHNICAL INFORMATION SERVICE U S Department of Commerce Springfield VA 22131

Approved for public release; distribution unlimited,

Air Force Aero Propulsion Laboratory Air Force Systems Command Wright-Patterson Air Force Base, Ohio

Sector Sector



オントロスロスローズがあるまたとう

NOTICE

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.



and a state of the second state of the

Copies of this report should not be returned unless return is required by security considerations, contractual obligations, or notice on a specific document.

			UNIA) SAID			
linglagatetet			ALCONO.			
Security Classification			~~~~~			
DUCL	MENT CONTROL DATA · RAD		9,000			
(Seculty cleastlication of title four of shales	ct and industry annitation must be arraised what	n the everall separt to class(flait)	e. S			
Defigitiating activity (Lugurane evider)	20. MLP 11	UNT SECURITY CLASSIFICATION	. e.W.			
505 King Avenue	25 24. CAG	NP				
Columbus, Ohio 43201			ين لك ي ^{ار}			
Failure Mechanisms and Accelera	ted Life Tests of Nickel-Cadm	ium Batteries	a and a sector			
. DESCRIPTIVE HOTES (Type of report and inclusi	va dalaa)		()*** A			
Final						
MCCallum, J. McCallum, J. Miller, G. H.	*****************		5.11 (17 - 11 - 11 - 11 - 11 - 11 - 11 - 11 -			
. REPORT DATE	74. TOTAL HO. OF PAGES	70. HO. OF REFS	11.11.1			
August 1972	68	26	4. 4. 4 th			
A. CONTRACT OR GRANT NO.	\$9. ORIGINATOR'S REPORT NU	IMBER(S)	11411			
F33615-69-C-1537 A PROJECT NO.						
6. SS. OTHER REPORT NO(5) (Any other numbers that may be seelened the nervi						
	AFAPL-TR-70-44	Part III	NAME OF THE OWNER			
. AVAIL ADILITY/LIMITATION NOTICES			, 1364, ISBN,			
Approved for public release; distribution unlimited, . SUPPLEMENTARY NOTES 12. SPONSORING MILITARY ACTIVITY						
	Air Force Aero Pro	pulsion Laboratory	7,41,81			
	Wright-Patterson /	Air Force Base, Shio	· 新聞 「 王 八 二 王 八 二 三 ノ			
J. ADSTRACT	1					
Preliminary experiments indicat dients and/or with tates of dis testing of sealed batteries.)) small effect on the life of cad without, significant loss of pe tests were recommended for them Supporting studies emphasized (battery testing laboratory, (b) statistical, and physical araly development of a physical basis pots, and their analogs as they calculation of thermal resistan cells, and combinations of cell information Index	ed valid tests were developed charge as the primary variable epth of discharge was indicate minum electrodes. New scaled of formance, were designed and of a) the development of maintena the recommendation of approace sis of large amounts of batter for accelerated life tests bar relate to aging processes, (do ces and thermal capacitances of s in batteries, and (e) the am	with temperature gra- es for accelerated life ed to have only a very ceils having longer life constructed, Verification ance procedures for a ches for empirical, by life data, (c) the used on springs, dash- l) the measurement and of cells, components of plification of a Battery	ባ እና ስጥ የእ -			
1) . form 1/173			the second second			
	<u> </u>	ssified Cutity Classification	and the second se			
	Call Contractor and the Contractor					

 A second design of the second s 17215 WALLAND AND BALL

Unclassified

الكوالعيلا يتعوال كروم معامد

an dat Kettend

(1994) A Market and American

Mathic Ania Series

17.55 Leanatt

24

	Security Classification			1.0-			
14.	NET WORDS	8015	wr	ROLE	w T	ROLF	K C
	Accelerated Life Tests						
Į	Failure Analysis						
	Nickel-Cadmium Cell Fabrication						
	Simulated Orbital Life Testing						
	Thermal Properties of Cells						
	Stress-Strain Models						
1							
i							
ĺ							
			İ				
					ļ		
			}	l			
Į				İ			
					1		
		ļ .					
[Į			
				1			
			1			Į	
						Į	
			ļ			Ì	
		1					
						1	i
Į				1	1	Í	
			l			1	
		Į					
I		1	1	I	[1	Í
ł		1			1	1	
1		1	1	1	1	1	
				ł			
		1		Į	ļ	1	1
ł				1	[1	1
					1	I	

Unclassified Security Classification and a second second second second second second second second second second second second second second second

ALING A STATE AND A STATE

an and the second of the second second second second second second second second second second second second se

FAILURE MECHANISMS AND ACCELERATED LIFE TESTS OF NICKEL-CADMIUM BATTERIES

的学言

しょうざい ちょうじゅう しんしゅう ちょうちんしん ちゅうたいちょう イン

のため、またまたが、またがはなるないないないです。 またまたいかんだい いったいない かんたいかい たいかい いたい いってい

J. McCallum

BATTELLE Columbus Laboratories

and

G. H. Miller Air Force Aero Propulsion Laboratory

Approved for public release; distribution unlimited.

FOREWORD

This report describes work performed during a three-year contract at the Columbus Laboratories of Battelle Memorial Institute, 505 King Avenue, Columbus, Ohio 43201, and at the Aero Propulsion Laboratory, Wright-Patterson Air Force Base, Ohio 45433. The work was done under Contract No. F 33615-69-C-1539, Task 314522-022. The report was submitted by the auchors on June 30, 1972, and covers work done between May 1, 1969 and June 30, 1972.

This project was under the general direction and supervision of Dr. J. McCallum. Battelle personnel who have contributed to the various aspects of the work are:

> Dr. E. W. Brooman - Accelerated Testing, Literature Review, and Failure Analysis
> Mr. P. W. Cover - Increased Battery Life and Cell Fabrications
> Mr. R. L. Gibbs - Facility Maintenance
> Mrs. D. M. Johnson - Literature Review
> Mr. O. L. Linebrink - Facility Maintenance
> Dr. A. H. Reed - Accelerated Testing, Increased Battery Life, Cell Fabrication, Failure Analysis, and Literature Review
> Mr. E. A. Roeger, Jr. - Accelerated Testing, Data Analysis, Facility Maintenance, Failure Analysis, and Literature Review

Dr. R. E. Thomas - Stress-Strain Models and Accelerated Testing

Dr. A. Fleischer, Consultant, 466 South Center Street, Orange, New Jersey 07050, has provided guidance in aspects of the program dealing with cell fabrication and failure analysis.

Mr. Gerald H. Miller, AFAPL/POE-1, is the cognizant Air Force Scientist. His efforts helped direct the work in two laboratories toward the needs of the U.S. Air Force.

Publication of this report does not constitute Air Force Approval of the report's findings or conclusions. It is published only for the exchange and stimulation of ideas.

James D. Reams Chief Energy Conversion Branch Aerospace Power Division Air Force Aero Propulsion Laboratory

ABSTRACT

This report summarizes accomplishments during a three year work period on several tasks and subtasks as follows:

Task I. Accelerated Life Tests. Accelerated life tests for sealed cells were planned and given preliminary examination in the laboratory with 5 groups of 5 cells in each group. Specially prepared demountable cells having internal heaters were constructed with thin, commercially prepared nickel and cadmium electrodes. In one experiment, uniform temperatures of -20, -5, +10, +25, and +40 C were imposed and the effects of temperature on cell life was negligible. In a companion experiment having internal heaters at +40 C, the rates of degradation of cell qualities was accelerated about 10 times for each 30 degree temperature difference between inside and outside. In another experiment different discharge rates of C/2, C, 2C, 4C, and 8C were imposed with uniform cell temperatures of 25 C. A 10-times increase in discharge rate appeared to yield about a 30-times increase in the rate as which cell qualities degraded. In a companion experiment having internal heaters at +40 C, a 10-times increase in discharge rate appeared to yield about a 100-times increase in the quality-degradation rate. In another experiment with limiting cadmium electrodes, the effects of depth of charge were found to be much smaller than anticipated from published information. A factor Of 10-times in the product of (depth of discharge x cycles) was required to yield an apparent degradation of capacities by a factor of only 2-times. A cheoretical basis for accelerated life tests was refined and expanded by analogy with springs and dashpots. Ideal springs always store energy. Ideal dashpots always dissipate energy. This report correlates dashpot analogs though Onsager's reciprocal relations and correlates for the first time, spring analogs with the Tribus relations for thermostatics.

「日本のない」ので、「日本のない」ので、「日本のない」ので、

いたか。 中国の日本の日本の日本の日本の日本の日本の日本の日本の日本の日本の日本の日本の一本の日 「日本の一本の日本の日本の日本の日本の日本の日本の日本の日本の日本の日本の一本の日本

Task II. Improved Cells. Prototype nickel-cadmium cells were built with the following improvements: (1) thicker plaques, (2) thicker substrates, (3) moderate pore fiiling, (4) prolonged final washing of electrodes, (5) electrode matching, (6) polypropylene separator, (7) solid nickel terminals, combs, and tabs, and (8) "Mercurycell"-grade potassium hydroxide. Comparative testing shows longer cell life with the incorporation of these improvements.

Task III. Supporting Studies. (a) Routines for maintaining a battery testing laboratory were worked out. (b) Procedures for analyzing large quantities of test data were recommended. (c) A physical and mathematical basis for combining

iii

springs, dashpots, inertial masses, and their analogs was developed. (d) Thermophysical properties for the components of cells, and __r batteries of cells were determined for resistance/capacitance networks and calculations were experimentally verified. (e) A battery information index having about 2200 documents was produced as a result of active engagements by the project personnel with the published technical literature throughout the course of the contractual program.

TABLE OF CONTENTS

territer and a state of the state of the state of the state of the state of the state of the state of the state

of Strather

State State States

Ń

Ś

.

1.1

- KEE

A CONTRACTOR OF THE OWNER OF THE

Ď299.

Real for

	Page
BACKGROUND AND OBJECTIVES	1
TASK I. ACCELERATED LIFE TESTS	3
A Physian Basis for Accelerated Life Tests	3
T_uperature and Temperature Gradients.	11
Rate of Discharge.	12
Depth of Discharge	1/
Verification Tests	14
Taet Dugion	10
Call Description	10
Coll Characterization and Matching	18
Toot A maratterization and matching.	19
	20
	20
	25
'e 'ire Anai'sia	27
TASK 1 NCREASED BATTERY LIFE	29
Design of Longer Life Cells	29
Cells in Demountable Cases	29
Cells in Hermetically Sealed Cases	31
Evaluation of Longer Life Cells.	34
Cells in Demountable Cases.	34
Conclusions and Recommendations.	42
TASK III, SUPPORTING STUDIES	44
Task III (a). Facility Maintenance.	44
Task III (b). Data Analysis	45
Tack III (c). Stress-Strain Models.	46
Tack III (d) Thermonbusical Properties	40
Task III (a) literature Device	52
lask III (e). Siterature Review	54
Subject Indexing and Furging Procedures	
Retrieval Techniques. ,	57
REFERENCES.	50
APPENDIX	
RECOMMENDED CELL DESIGN FOR MAXIMUM PERFORMANCE AND LIFE RECOMMENDED	
ACCELERATED TEST PROCEDURE	63
LIST OF ILLUSTRATIONS	
Figure	
	~~
1. Circuit Diagram - Test Apparatus	25
2 Harmatically Sealed Coll Design	33
T. HETMETTRATTA SEATER AETT DESTRU	

LIST OF TABLES

<u>Table</u>		Page
I.	Generalized Stress-Strain and Stress-Rate-of-Strain Relationships	6
11.	Verification Test Plan	17
III.	Quantities of KNO3 stu K2CO3 in Commercial Cells	19
IV.	Characterization, Capacaty, and Failure Determination Data For Charge Strain-Rate Post	21
v.	Characterization, Capacity, and Failure Determination Date For Discharge Strain-Rate Test	22
VI.	Summary of Veril Action Test Groups As or June 7, 1972	26
VII.	Preliminary Estimate 68 Acceleration Factors For Verification Test Groups	28
VIII.	Failure Ar lysis Procedure For Cells in Verification Tests	29
IX.	Failure Analysis Data For Demountable Improved Cells Evaluated At 40 C	38
X.	Failure Analysis Fata From Demountable Improved Cells Evaluated at 25 C	39
XI.	Failure Analysis Data For Demountable Improved Cells Evaluated at 10 C	40
XII.	Failure Analysis Data For Demountable Improved Cells Evaluated at -5 C	41
XIII.	Third Version of Generalizec Stress-Strain and Stress-Strain Rate Relatic	48
XIV.	Source Classification of Battery Information Index (Includes Battery Documents B-1 through 52189)	53
xv.	Growth of BII Indexes	55
XVI.	Average Number of Indexing Terms Fer Document in BII System	56
XVII.	Initial Results of Index Purging Procedures	58
(V 111.	Recommended Accelerated Test Procedure Based on 6-Ampere-Hour Sealed Nickel-Cadmium Cells	65

Processing of the Science of the Sci

΄.

2 2 8

-

FAILURE MECHANISMS AND ACCELERATED LIFE TESTS OF NICKEL-CAPMIUM BATTERIES

J. McCallum and G. H. Miller

BACKGROUND AND OBJECTIVES

This final report summarizes a three-year program to develop concepts concerning accelerated testing of sealed spacecraft cells and batteries and to verify the concepts. The theoretical framework previously proposed $(1,2,3)^*$ for both the United States Air Force and NASA shows that degradation of cell and battery performance should be controlled by applied stresses. Each stress is assumed linearly proportional to both the associated strains and to strain rates applied to the system. Furthermore, stresses were identified as intensity gradients for the degradation of various energy forms. The intensity gradients most pertinent to battery life and performance were identified as voltage gradients, temperature gradients, pressure gradients, and concentration gradients. This identification of stresses with gradients led to what was called the "gradient hypothesis", which says: All aging is caused by intensity gradients with their strains and strain rates. Therefore, the main objective of new work was to discover one or more accelerated test procedures. Also, efforts were planned to develop nickel-cadmium cells with longer life by designing the components so they would operate at stress levels lower than those found in present-day cells.

The work on this program was divided into the following three tasks:

「ちょうで、ここである」をいたい、ここのないたいであるというできょうこうないできょうできます。

いたちょうろうちょうとうちょうで、

- Tesk I. Accelerated Testing. The purpose of this task was to provide experimental evidence for the validity of selected tests and their theoretical basis.
- Task II. Increased Battery Life. The work of this task involved designing and developing nickel-cadmium cell components so the cells would operate at lower stress levels. This task complements Task I by using the same stress-strain and stress-strain rate phenomena for opposing purposes.

References are listed on page 58.

Task III. Supporting Studies. This task was divided into five subtasks:

and the second of the second o

- (a) Facility Maintenance. This work involved maintaining the equipment in the Battery Laboratory of the Aero Propulsion Laboratory at Wright-Patterson Air Force Base in & state of readiness to perform the experimental portion of this program.
- (b) Data Analysis. Simulated orbital life tests begun under Contract No. AF 33(165)-3701 were continued under this contract. Cells that failed during these tests were analyzed for failure determinants and failure mechanisms.
- (c) Stress-Strain Models. A mathematical model based on springs and dashpots was developed to describe the aging of an energy storing device. An objective was to improve the theoretical basis for new accelerated test procedures and improved cells in Tasks I and "I.
- (d) Thermophysical Properties. Thermophysical properties were determined for individual cell components so that improved models of heat flow in cells and batteries could be used to predict improved designs for cells and batteries.
- (e) Literature Review. A continuing review of literature in the field of nickel-cadmium battery technology and accelerated testing was performed to help keep project personnel up to date in the field.

Each of these tasks and subtasks is discussed in detail in subsequent sections of this report.

TASK I. ACCELERATED LIFE TESTS

いたいないないないないないないないないないないないないないないできょうかんとうなくてい

20日本には1月1日に、2月1日には1月1日には1月1日に

A Physical Basis for Accelerated Life Tests

In previous work for the U. S. Air Force (1,2) a theoretical basis for accelerated life tests was proposed. Development of fundamental concepts was continued for NASA. (3) During this new contract period, fundamental concepts were refined and enlarged upon, as explained in sections to follow.

Table I, shows earlier relationships proposed for generalized stressstrain and stress-rate-of-strain involving eight forms of energy. For accelerated life testing of batteries, only four forms of energy were deemed to be germane, namely, (1) three dimensional mecha..ical energy in Row 3, (2) electrical energy in Row 4, (3) thermal energy in Row 5, and (4) chemical energy in Pow 6. For each energy form, the primary cause of "aging" and shortened life was attributed to <u>stresses</u>. Stresses were associated with gradients of the intensity factors from Column 3, as given in Column 5 for the "Flux Generality" and in Column 7 for the "Stress-Strain Generality", namely gradients of pressure, voltage, temperature, and concentration. This prior work thereby led to aforementioned "gradient hypothesis".

One of the unusual aspects of "his approach to accelerated life tests was an implication that temperature, itself, was not a stress whereas temperature gradient was a stress. This implication is unusual because of weil established and well known associations of chemical reaction rates with temperature and because of allied associations of chemical reaction rates with "aging". To remain consistent with the gradient hypothesis, therefore it was supposed that an increase of temperature could create or change gradients of concentrations, voltages, and/or pressures. Thereby, the various kinds of intensity gradients should be associated with an absolute value of uniform temperatures. This sort of reasoning was in mind during the planning of the experimental program. Thus, the next section of this report describes two parts of an experiment in which variables were supposedly controlled identically except that in one part of the experiment cell temperatures were held uniform, in the other part, internal temperature gradients were imposed. Similarly, another experimenc, described later under <u>Rate of Discharge</u> has two parts: one with internal temperature gradients and one without temperature gradients.

A third experiment described later under <u>Depth of Discharge</u> extended the testing of the gradient hypothesis, as follows. In the context of Table I, depth of discharge appears as an amount factor per unit in Column 7 of Row 4. Voltage

gradients were thereby implied to be the stress d. sectly associated with the various levels of depth of discharge. At the same time, the earlier theoretical framework implied existance of gradients of pressure, temperature, or concentration somehow coupled with the voltage gradients. Voltage gradient, of course, are directly coupled to electrical charge per unit area, that is, the depth of discharge. Therefore, the Depth of Discharge Experiment was planned to provide physical evidence for how and to what extent these various stresses were related to each other and to the electrical strain called depth of discharge. A fourth e: periment was planned with varying concentrations of carbonate in the electrolyte. This experiment was devised because large carbonate concentrations had frequently been found to be associated with failure mechanisms. (2,4) Moreover, concentration gradients show in Row 6 of Table I as one of the important stresses and an experiment designed to study this stress under controlled conditions seemed to be a very desirable experiment to perform. In the actual course of events, however, the Carbonate Experiment was abandoned when Barney and Catotti published their comprehensive findings on the "Effects of Carbonates on Charge and Discharge Characteristics of Sealed Nickel-Cadmium Cells".⁽⁵⁾ Therefore, the effects of carbonate received little direct study during this contractual period. Nevertheless, the above mentioned knowledge from Barney and Catotti⁽⁵⁾ as well as results from this project (2,4) shows the effects of carbonate concentration to be consistent with the physical basis for structuring an accelerated life test. In the meantime, the physical concepts and relationships continued under development with the three tasks of this program.

In Subtask III(e) on "Styras-Strain Models", the fundamental concepts inherent in Table I continued to be refined and improved. First of all, the stress-strain model was related to mechanical springs and mechanical dashpots. An ideal spring is a mechanical device in which mechanical stress is proportional to mechanical strain. An ideal dashpot is a mechanical device in which stress is proportional to strain rate. Real examples of springs and dashpots are given by the springs and the shock absorbers on automobiles, respectively. The first six rows of energy forms in Table I were then related by analogy with ideal springs and ideal dashpots. The final results of these deliberations are given in Table XIII.

The spring and dashpot analogies were utilized for the purpose of emphasizing these most important points:

> Point (a). Ideal spring (and their analogs) always store energy and never dissipate energy, whereas Foint (b). Ideal dashpots (and their analogs) always

dissipate energy and never store energy. (6)

			TABLE I. GE	NERALIZED STR	ESS-S .IN AND ST	LESS-RATE-OF-S	TRAIN RELATIONSE	IIPS	
=	(1) Energy Form	(2) Ene rgy Source	(3) Intensity Factor, I	(4) Amount Factor, A	(5) Flux Generality $\frac{1}{a} \frac{\partial A}{t} = k_8 \cdot \frac{\partial I}{\partial x}$	(6) Conservation Generality $\frac{\partial I}{\partial t} = k_{17} \cdot \frac{\partial^2 I}{\partial x^2}$	(7) Stress/Strain Generality $\frac{\partial 1}{\partial x} = k_{27} \cdot \frac{A^{(b)}}{a}$	(8) Impulse/Amount Generality) $\Delta 1 = k_{33} \cdot \frac{\partial A}{\partial L}$	(9) Impulse/Rate Generality $\Delta 1 = k_{44} \cdot \frac{\partial^2 A}{\partial t^2}$
1.	Mechanical (one de- gree of freedom)	Morion	Force, 7	Distance, x	Not appropriate	Kot eppropríete	Not appropriate	$\Delta F = k_{34} \cdot \frac{\partial x}{\partial t}$ (Newton's 1st Law, Stoke's Law)	$\Delta F = k_{46} \cdot \frac{\partial^2 x}{\partial t^2}$ (Newton's 2nd Lev)
2.	Mechanical (two du- grees of freedom)	Liquids	Surface tension, Y	Агеа, а	Not appropriate	Not appropriate	Not appropriate	Δγ = k36. 34	$\Delta y = k_{44} \cdot \frac{\partial^2 a}{\partial t^2}$
з.	Hechanical (three da- grees of freedom)	Liquids or gases	Pressure, P	Volume, V	$\frac{I}{a} \cdot \frac{\partial V}{\partial t} = k_{11} \cdot \frac{\partial P}{\partial x}$ (Poiscuille's Lcw)	$\frac{\partial P}{\partial t} = k_{18} \cdot \frac{\partial^2 P}{\partial x^2}$	$\frac{\partial P}{\partial x} = k_{29} \cdot \frac{V}{a}$ (Sound energy)	$\Delta P = k_{37} \cdot \frac{\partial V}{\partial z}$ (Poisruille's Lav)	$\Delta P = k_{44} \cdot \frac{\partial^2 v}{\partial t^2}$
4.	Electrical	Elec- tricity	Voltage, E	Charge, Q	$\frac{I}{a} \cdot \frac{\partial Q}{\partial t} = k_{14} \cdot \frac{\partial E}{\partial x}$ (Ohn's Law)	$\frac{\partial E}{\partial t} = k_{19} \cdot \frac{\partial^2 E}{\partial x^2}$ (Ohm's Law)	$\frac{\partial E}{\partial x} \sim \frac{k_{28}}{k_{28}} \frac{Q}{a}$ (Capacitance relationship)	ΔE * k ₃₈ · 30 (Ohm's Law)	$\Delta E = k_{47} \cdot \frac{\partial^2 Q}{\partial z^2}$ (Inductance relationship)
5.	Thermal	Heat	Tempera- ture, T	Heat, q (calories)	$\frac{1}{a} \frac{\partial q}{\partial t} = k_{15} \cdot \frac{\partial T}{\partial x}$ (Fourier's Law)	$\frac{\partial \mathbf{T}}{\partial t} = k_{20} \cdot \frac{\partial^2 \mathbf{T}}{\partial x^2}$ (Pourier's Law)	$\frac{\partial T}{\partial x} = k_{27} \cdot \frac{q}{a}$	ΔT = k ₃₉ . Jq (Pourier's Lav)	$\Delta T = k_{44} \cdot \frac{\partial^2 q}{\partial t^2}$
6.	Chemical	Chemical specien in solution	Concentra- tion, C	Holes, H	$\frac{1}{a} \frac{\partial M}{\partial t} = \frac{k_{16}}{k_{16}} \frac{\partial C}{\partial x}$ (Fick's lst Law)	$\frac{\partial C}{\partial t} = k_{21} \cdot \frac{\partial^2 C}{\partial x^2}$ (Fick's 2nd Law)	$\frac{\partial C}{\partial x} = k_{30} \cdot \frac{M}{a}$	$\Delta C = k_{40} \cdot \frac{\partial M}{\partial t}$ (Hass Action Law)	$\Delta C = k_{44} \cdot \frac{\partial^2 H}{\partial t^2}$
7.	Kinetic	Notion	Velocity, v	Nomentum, av	$\frac{I}{a} \cdot \frac{\partial(\mathbf{m}\mathbf{v})}{\partial t} = k_{22} \cdot \frac{\partial \mathbf{v}}{\partial \mathbf{x}}$ (Bernoulli's Theorem)	$\frac{\partial v}{\partial t} = k_{24} \cdot \frac{\partial^2 v}{\partial x^2}$	du = k ₂₇ . <u>ev</u>	$\Delta v = k_{41} \cdot \frac{\partial (mv)}{\partial x}$ (Newton's lat Law, Stoke's Law)	$\Delta v = k_{44} \cdot \frac{\partial^2 (\mathbf{m} v)}{\partial t^2}$ (Hooke's Law)
8.	Light	Rediation	Prequency, P	Illumina- tion, L (no. of pnctons)	$\frac{1}{a} \frac{\partial L}{\partial x} = k_{25} \frac{\partial f}{\partial x}$ (The Red)	$\frac{\partial L}{\partial x} = k_{17} \cdot \frac{\partial^2 L}{\partial x^2}$	$\frac{\partial f}{\partial x} = k_{27} \cdot \frac{L}{a}$	$\Delta f = k_{42} \cdot \frac{\partial L}{\partial x}$ (Planck's Law;	Yot appropriate

and the second second second second second second second second second second second second second second second

C(1) (2)

÷ ? ÷ 1.6

APP March

s s .

(a) (b) (c) Strain rate = k*stress. Streag = k*strain Force = k*scceleration.

-

Real springs and dashpots are never ideal and behave as though there were mixtures of springs and dashpots in real devices. Fortunately, the rules are well worked out for combining springs in series or in parallel with other springs or with dashpots. Fortunately too, mathematical descriptions of thermal and electrical analogs of mechanical springs, dashpots, and inertial masses are also well worked out. Unfortunately, however, some of the basic definitions for electrical and thermal components are inverted from analogous definitions for mechanical components. The inversions are unfortunate because they sometimes cause similar physical situations to be described differently. For example, electrical voltages in series are additive and voltages in parallel are the same on each device whereas mechanical forces in series are the same on each device and forces in parallel are additive. Even so, forces are analogous with voltages in several ways and a physical basis for accelerated life tests depends on the proper understanding and use of "forces", "displacements", "stresses", "strains", "strain-rates", and their analogs.

During the new contract period, considerable thought was given to refinements and improvements in the general concepts inherent in Table I. Noteworthy accomplishments are as follows with reference to Table I: 114 B G

- Generalized forces are intensity factors in Column 3.
- Generalized displacements are amount factors in Column 4.
- The mathematical product of force times displacement should give appropriate units of <u>work</u>. Heat energy in Row 5 is an exception to this genera^{1,2}zation. Therefore, suggestions have been made⁽⁷⁾ that intensity and amount factors for heat energy should be sought and developed such that (a) their mathematical product gives units of thermal work and (b) the new intensity factor has all the propervies of temperature. A dimension of velocity is suggested as the appropriate new unit to be developed as the appropriate intensity factor ior thermal energy. Then, heat would be treated as kinetic energy with an amount factor of momentum in accordance with Row 7.
- Generalized stresses and strains differ from generalized forces and displacements by the geometry of the components under consideration. Thus, the mathematical product of stress times strain should give appropriate units of work/volume, whereas force times displacement gives work.

Preceding page blank 7

There are two kinds of stresses. They have different units. They can interact with one another through common units yet they are conceptually quite different. The first kind of stress appears as analogs of springs for the storage of energy, wherein stress is proportional to strain. The second kind of stress appears as analogs of dashpots for the dissipation of energy, wherein stress is proportional to strain rate. The spring analog of thermal stress is <u>temperature</u>; the dashpot analog of thermal stress is temperature gradient. Similarly, the spring analog of chemical stress is concentration and the dashpot analog of a chemical stress is a concentration gradient. At the same time these distinctions are noted about stresses. the associated distinctions about strains and strain-rates must be noted also. The spring analog of strain will have units of amounts/volume. The dashpot analogs of strain rates will describe fluxes having units of amounts/(area.time). Then, the spring analogs of stress times strain always give units of work/volume. The dashpot analogs of stress times strainrate always give units of power/volume = work/(volume time).

w other hands of

• The various dashpot analogs in Column 5 are related to each other through Onsager's reciprocal relations.⁽⁸⁾ Onsager's relations state that any given flux on the left-hand side of the equations in Column 5 is equal to the sum of the contributions to that flux caused by all the intensity gradients on the right-hand side of the equation in Column 5. Readers may note, of course, that forms of energy exist which are not in Table I and, therefore Onsager's relations are not limited to information in Column 5. However, since the important forms of energies in batteries have been judged to be limited to mechanical, electrical, thermal, and chemical, let us use the letters Me, E, T, and C as subscripts to denote these respective energy forms from Rows 3 through 6. Then, using the intensity factors and amount factors from Columns 3 and 4 with their respective symbols, the Onsager relations state as follows:

$$(1/a)(\partial V/\partial t) = k_{MeMe} \cdot \partial P/\partial x + k_{MeE} \cdot \partial E/\partial x + k_{MeT} \cdot \partial T/\partial x + k_{MeC} \cdot \partial C/\partial x$$
(1)

$$(1/a)(\partial Q/\partial t) = k_{EMe} \cdot \partial P/\partial x + k_{EE} \cdot \partial E/\partial x + k_{ET} \cdot \partial T/\partial x + k_{EC} \cdot \partial C/\partial x$$
(2)

$$(1/a)(\partial Q/\partial t) = k_{TMe} \cdot \partial P/\partial x + k_{TE} \cdot \partial E/\partial x + k_{TT} \cdot \partial T/\partial x + k_{TC} \cdot \partial C/\partial x$$
(3)

$$(1/a)(\partial M/\partial t) = k_{CMe} \cdot \partial P/\partial x + k_{CE} \cdot \partial E/\partial x + k_{CT} \cdot \partial T/\partial x + k_{CC} \cdot \partial C/\partial x$$
(4)

To read these equations consider the second, the electrical flux relation, which might be read as follows:

"The amount of electrical charge, ∂Q , per unit area, a, per unit of time, ∂t , is called 'an electrical flux' and the total flux is equal to the sum of the electrical flux caused by a pressure gradient plus the electrical flux caused by a voltage gradient plus the electrical flux caused by a temperature gradient plus the electrical flux caused by a concentration gradient."

The other equations above can be read similarly. Obviously, the major portion of each flow will be associated with its directly related gradient in Rows 3 through 6. But any portion of flux occurring as a result of coupling with the other gradients also contributes to the dissipattion of energy in the device. The above Onsager relations in appropriate combinations are a vital part of a physical basis for accelerated life tests, 8)

• Another vital part of a physical basis for accelerated life tests is indicated by the Tribus relationships. (17) utilize analogs of spring wherein the stress units are intensity factors in Column 3 and strain units are amount/volume. Using, then, the amount factors in Column 4 with brackets to denote the respective amounts/volume, the Tribus relations state as follows:

$$\partial [V]/\partial x = k_{MeMe} \cdot \partial P/\partial x + k_{MeE} \cdot \partial E/\partial x + k_{MeT} \cdot \partial T/\partial x + k_{MeC} \cdot \partial C/\partial x$$
 (5)

$$\partial[Q]/\partial x = k'_{EMe} \cdot \partial P/\partial x + k'_{EE} \cdot \partial E/\partial x + k'_{ET} \cdot \partial T/\partial x + k'_{EC} \cdot \partial C/\partial x$$
 (6)

$$\partial[q]/\partial x = k_{TMe}' \quad \partial P/\partial x + k_{TE}' \quad \partial E/\partial x + k_{TT}' \quad \partial T/\partial x + k_{TC}' \quad \partial C/\partial x$$
(7)

$$9[W]/9x = k^{CMe}$$
, $9B/9x + k^{CE}$, $9E/9x + k^{CL}$, $3L/9x + k^{CC}$, $9C/9x$ (

In the above equations the ∂x values cancel out of all LIMS. When the ∂x term is deleted, the result is a direct stress-strain relationship such that strains are amounts/volume and stresses are intensities and stress times strain has units of work/volume.

- The coefficients in both the Onsager and in the Tribus relationships are called "reciprocal" relations because cross coefficients with the same subscript symbols are numerically equal. For example, k_{MeE} in the first Onsager relation Equation (1) above is equal to k_{EMe} in Equation (2). Likewise, k_{MeE} in the first Tribus relation Equation (5) is equal to k^{*}_{EMe} in Equation (6).
- Insofar as the degradation of qualities is brought about by the physical movement or rearrangement of physical matter, only interactions of stresses and fluxes involving matter cause aging. This means k_{EE} , k_{ET} , k_{TE} , k_{TT} , (1/a)($\partial Q/\partial t$), and (1/a)($\partial q/\partial t$) are not directly associated with failure mechanisms. All other terms and couplings in the Onsager relations are.
- <u>All</u> the Onsager and Tribus relations behave as a simple Maxwell element of springs and dashpots in a series. Therefore, the time to failure is given by an Equation of the type:

$$t_{f} = (b/k) \ln[x(o)/(L-L_{g})]$$
 (9)

Here, L_p is the predefined failure extension of the Maxwell element.

- L is the experimental constant extension at which the device is held from time to to time t.
- x(o) is the difference between the equilibrium extension of the device at time t and extension L.
- b is the combined dashpot constant.
- k is the combined spring constant .

Because x(o), L, and L_F will be fixed for any given test condition, it is seen that the time to failure depends primarily on the ratio of the dashpot constant to the spring constant, b/k.

For a valid accelerated life test the above vatio of constants,
 b/k, must be made to decrease with increasing stress levels. If
 values of b/k can be determined to be a function of some experimentally

controlled variable z such that b/k = f(z), this relation may then be used to obtain an extrapolated time to failure at a "normal" condition of stress levels.

The above conclusions and accomplishments with regard to a physical basis for accelerated life tests require further quantitative development. However, the general Onsager and Tribus relationships in combination appear to provide a sufficient basis for describing accelerated life tests. Further work is needed to autheuicate the general validity of these relationships.

Temperature and Temperature Gradients

Details of the two experiments summarized below are given in a special report titled, "Temperature and Temperature Gradients for Accelerated Life Tests (Nickel-Cadmium Cells".⁽⁹⁾ Briefly summarized, cycling tests were performed on groups of five cells each at ambient temperatures of -20, -5, 10, 25, and 40 C and both with and without temperature gradients. Special demountable cells with internal heaters were made and used to provide an internal cell temperature of 40 C for those cells with temperature gradients.

In all these cells, the failure determinant was a loss of capacity at the positive electrode. Actually, both positive and negative electrodes lost substantial ampere-howr capacity during the tests and the negative electrodes lost the most. However, failures to discharge within test specification times were caused by inadequate capacities at the positive electrodes and so failure determinants were associated with the positives.

All cells in these experiments were made with thin, commercial electrodes from special aircraft batteries. As part of the failure analysis, these electrodes were found to contain substantial quantities of soluble nitrate impurities. A strong odor of ammonia emmanated from all demountable cells after the tests. As a result of these and other findings during failure analysis, the failure mechanism was associated with a "nitrate-nitrite shuttle" which acted like a partial internal electrical short. The same mechanism appeared to control all failures and degradations of qualities. Because the failure mechanism appeared to be the same for all cells, the accelerated life test was deemed to be a valid life test. بالمجافية المراجع المسابق المحافية والمحاوية المحافظ والمحافظ والمحافظة المحافظة المحافظ

In these experiments. the effect of a constant temperature level on degradation of cell qualities appeared to be negligible. On the other hand, cells with maximum temperature gradients gave an acceleration factor of about 35 compared

with cells in the same tests but with no temperature gradient. Thus, in the five tests at +40, +25, +10, -5, and -20 C both inside and outside the cells, the rate of change of voltage qualities per cycle was practically the same for all cells. On the other hand, with the five tests having the same five external temperatures but with all cells having the same internal temperature of +40 C for all tests, the acceleration factors appeared to increase about 10 times for each 30-degree differential between internal and external temperatures. Both tests had large and mostly unexplained differences between cells but the statistics suggest that the above conclusions are correct with at least a 70 percent probability.

Cell temperatures varying between +40 and -20 C were expected to show significant differences in the rates at which cells failed. The fact that no such effect was observed is attributed to the particular failure mechanism, namely the "nitrate shuttle". Thus, it is supposed that the concentration of the very soluble nitrates was unaffected by the absolute level of temperature. The rate of the nitrate-nitrite reactions with active electrode materials also appeared to be unaffected by the absolute temperature level. The net result was an absence of any observable effect of temperature level on cell life. On the other hand, the effect of temperature gradients was to bring about nonuniform gradients of all kinds within cells as explained in the next section of this report.

From the viewpoints of a physical basis for accelerated life tests described in the preceding section, temperature and temperature gradients are possibly important any time temperature appears in the Onsager or in the Tribus relationships. Thus, from the previously written Onsager relations, temperature gradients can cause:

- (a) Volumes to move by convection or by osmosis
- (b) Charge to move by the Soret effect or by concentration gradients
- (c) Chemicals to move by convection or by concentration gradients.

The absolute temperature itself can increase or decrease:

- (a) Volume by expansions or by contractions
- (b) Electrical charge through ionizations or by changing solubilities
- (c) Chemical concentration by changing solubilities or by changing chemical reaction rates as expected from the Arrhenius equation.⁽⁷⁾

A BARANCE AND A STATE

concentration gradients may have been responsible for the accelerations found with temperature gradients but not with uniform temperatures.

A State of Long and the second second second second second second second second second second second second se

In future experiments involving temperature gradients it is recommended that heaters be placed outside rather than inside, as was done with these experiments. Thereby, the variations between cells in the same tests would be expected to be much more closely controlled.

Rate of Discharge

Details of the two experiments summarized below are given in a special report titled, "Rate of Discharge for Accelerated Life Tests of Nickel-Cadmium Cells". (10) Briefly summarized, cycling tests were performed on groups of five cells each discharged to 50 percent depth and at C/2, C, 2C, 4C, and 8C rates. Environmental temperature was 25 C for both experiments but one experiment of five groups had specially constructed internal heaters maintained at +40 C.

The failure determinant and failure mechanism for these cells were found to be the same as those described in the previous section, namely, an excessive loss of capacity at positive electrodes caused by the "nitrate-nitrite shuttle".

The effect of temperature gradients with these two experiments confirmed the effects mentioned in the previous section, namely, a temperature difference of about 30 degrees between heaters inside and environmental temperature outside appeared to give about a 10 times increase in acceleration factors compared with similar cells having uniform temperature. In these rate of discharge experiments, however, the effect of temperature gradient on an acceleration factor appears to be favored with about a 95 percent probability compared to a 70 percent probability in the previous two temperature experiments.

The effect of rate of discharge was relatively large. Thus, acceleration factors increased about 30 times for each 10-times increase in discharge rates. Unfortunately, the observed differences were also large between cells in the same group and, because of these large variations, confidence in the above conclusion is not as high as it should be for a widely accepted accelerated life test. The probability is at least 70 percent in favor of rate of discharge as a useful strain-rate for accelerating the life tests of batteries.

The acceleration factors for temperature gradients combined with acceleration factors from rates of discharge give about a 100-times increase and the state of the second of the second second second second second second second second second second second

in acceleration factors for each 10-times increase in rate of discharge. These important findings imply that future life tests with nickel-cadmium batteries might be accelerated two ways. First, by discharging 10-times faster and with an imposed temperature gradient, a cell could be made to degrade in quality perceptably with, perhaps 1 percent of the normal number of cycles to the same level of degradation. Second, if discharge rates are increased 10times, it follows that the duration of each cycle could be shortened. The combination of discharge rates with temperature gradients, therefore might shorten the duration of an accelerated life test as much as 1000 times.

SALVER AND STREET

From the viewpoints of a physical basis for accelerated life tests, rate of discharge is possibly important whenever rate of discharge can be associated with one of the parallel stresses in the Onsager and Tribus relations. The main effect to be associated with rates of discharge would be expected to be voltage gradients. This expectation follows because variation in rate of discharge for any given cell design becomes a current density variable. However, concentration gradients associated with current densities may be more important, insofar as the physical movement and rearrangement of matter is concerned. Thereby, concentration gradients become directly coupled with current densities. In addition, the relationship between current densities and temperature becomes more pronounced with increasing current densities because Joule heating is proportionate to the square of the current. With Joule heating, nonuniformities of temperature are to be expected. With nonuniformities, aging effects can be expected to accumulate faster as smaller areas are "attacked" separately. In this particular experiment, the main effect of increasing current densities appears to have been the associated increased concentration gradients which, in turn, enhanced the adverse effect of the "nitrate-nitrits shuttle".

Depth of Discharge

Details of this experiment are given in a special report titled, "Depths of Discharge for Accelerated Life Tests of Cadmium Electrodes. ⁽¹¹⁾ Briefly summarized, cycling tests were performed with thin cadmium electrodes taken from the same commercial cells as used to build demountable cells for the previously mentioned experiments. Portions of the commercial cadmium electrodes were cut to 1-1/2 by 3 inches and placed between two pieces of nickel-oxide electrodes, 2-3/4 by 5-1/2 inches. The electrodes were wrapped with perforated polypropylene separators, placed in special Lucite cells, and flooded with KOH electrolyte. After careful matching of electrodes for similar electrical performances

「「「「「「」」」」」」

ふうちょうちょう いろちょう しんちょうろう う

alura bidat and marken all all

2.00

and for state of charge, five cells were placed in each group and the five test groups were cycled as follows:

Additional Depth of Discharge Beyond an Initial 10 percent	Discharge Time, minutes	Charge Time, minutes
10	6	12
20	12	.24
40	24	48
60	36	72
80	48	96

One electrode was removed, from each test group for characterization and analysis after 100, 200, 400, 800, and 1600 cycles. Otherwise, all cells and groups were supposedly identical in content and in experimental conditions.

The net result of this fairly sophisticated experiment was that depth of discharge has a relatively small effect on the quality of cadmium electrodes. There difinitely was a decrease in utilizable cadmium in those electrodes having the combinations of large depth of discharge with large number of cycles. A factor of about 10 times in this product of (depth of discharge) (number of cycles) was required to observe a 2 times decrease in available capacity. Very little cadmium was observed to migrate out of the electrodes. The relatively small losses of capacity were attributed to a change in structure of the charged cadmium. The charged cadmium may also have become buried or hidden in pores with the extensive cycling so that it became electrically isolated and could not be oxidized. Evidence for this latter mechanism is indicated by a series of photomicrographs and chemical analyses for portions of electrodes going through these depth of discharge tests.⁽¹¹⁾

From a practical point of view, the use of depth of discharge to arrive at a valid accelerated life test was anticipated because of widespread information in the literature which makes, or implies, such claims. Francis reviews some of the relevant information. ⁽¹²⁾ If depth of discharge does, in fact, accelerate the failure rate of sealed nickel-cadmium cells, such an effect must be associated with the positive electrodes or with other variables not studied in this experiment.

From the viewpoint of the previously described physical basis for an accelerated test, there is no reason to expect that depth of discharge, itself, would have anything directly to do with aging mechanism. This reasoning follows because depth of discharge is analogous with releasing some of the displacement

on a spring. So long as the spring, or its analog is an ideal spring, energy would either be stored or be given back from storage and only the associated dashpot components would contribute to irreversibility of the processes. Future work will be required to identify the dashpot analogs that should be associated with depth of discharge. Even more work will be required to identify experimente' conditions where the dashpot components can be measured and where the dashpot coefficient, "b", will have a significant effect on the time-to-failure, t_f in the previously itemized failure Equation (9).

Verification Tests

Contractually, the verification tests wer 2 designed to "include normal (real-time) and accelerated tests in order to verify that the failure rates and mechanisms of the accelerated tests correspond to those experienced in the realtime testing". The verification tests were set up with commercial cells to confirm evidence described previously in preliminary tests using specially demountable cells. Two new tests were proposed, (1) a charge strain-rate test using 20 amperehour sealed cells and (2) a discharge strain-rate test using 6 ampere-hour sealed cells.

<u>Test Design</u>

<u>Charge Strain-Rate Test</u>. This test consists of five groups of six cells each. The test parameters are listed in Table II. Each group is charged at different Step 1 rates of charge to a voltage limit, at which time the charge is switched to the Step 2 rate of charge. Although the charge rate switchover is caused by a voltage controller on the group voltage, the switchover is set up on the first cycle so that it occurs exactly when the cells become 100 percent recharged. The voltage of the group at 100 percent recharge on the first cycle is then used for the switchover voltage during charge for the remainder of the life tests. The Step 1 and Step 2 charge times add to a total 3-hour charge period. This procedure was used because of the impurities found in these cells (described later in the report).

The discharge portion of each cycle is 7.5 amperes for 1 hour so . t the lowest charge rate is able to replace the discharged capacity with aimum overcharge of 13 percent.

Group Nc.	Charge (Step 1, amp	Current Step 2, ainp	High Charge Time, min	Discharge Current amp	Discharge Time, min
	Cha	rge Strain-Ra	<u>ate Test (20 amp</u>	-hr Cells)	
C-1	48.0	2.0	9.4	7.5	60.0
C-2	24.0	2.0	18.8	7.5	60.0
C-3	12.0	2.0	37.5	7.5	60.0
C-4	6.0	2.0	75.0	7.5	60.0
C-5	3.0	2.0	1.50.0	7.5	60.0
	<u>Disch</u>	arge Strain-	Rate Test (6 amp	-hr Cells)	
D-1	3.0	0.6	60.0	3.0	60.0
D-2	3.0	0.6	60.0	6.0	30.0
D-3	3.0	0.6	60.0	12.0	15.0
D-4	3.0	0.6	60.0	24.0	7.5
D-5	3.0	0.6	60.0	48.0	3.8

液気治気 生きる ちょうき

العليلية بالبالية المراقبة والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع والمراجع وال

and the state of the state of the second second second second second second second second second second second

TABLE II. VERIFICATION TEST PLAN

: Conte

Selected cells are to be removed from each group for parametric failure during the life of the group. One cell will be removed when its quality has degraded one-third of the difference between its initial and failure qualities. A second cell will be removed when its quality has degraded by two-thirds. In this manner, analysis can be performed on two parametric failures and still leave four cells to complete the test. Any cell's quality which fails below its failure quality will be removed immediately from test for failure analysis.

Discharge Strain-Rate Test. This test consists of five groups of six cells each. The test parameters are also listed in Table II. Each group is charged at the Step 1 rate of charge (3.0 amperes) for 60 minutes and then switched to the Step 2 rate of charge (0.6 ampere) for the remainder of the 3-hour charge period. Although the charge rate switchover is caused by a voltage controller on the group voltage, the switchover is set up on the first cycle so that it occurs exactly after the cells have been charged at the Step 1 rate for 1 hour. The cells are then subjected to 2 hours at the Step 2 rate of charge. This procedure was used because of the impurities found in the cells, as explained in the section on <u>Cell Descriptions</u>. Each group is discharged to 50 percent of its rated capacity (6 amperehours). The discharge rates for the various groups are given in Table II. Those groups which require less than 1 hour for their discharge remain on open circuit for the remainder of the 1-hour discharge period.

Two cells in each of these groups are to be removed for parametric failures. These cells are to be removed in the same manner as explained for the cells removed in the Charge Strain-Race Test.

Cell Descriptions

As previously explained, the Charge Strain-Rate Test uses commercial 20 ampere-hour cells while the Discharge Strain-Rate Test uses commercial 6 ampere-hour cells. These cells had been purchased under a previous contract (Contract AF 33(615)3701) and had been stored shorted for over 3 years before being used.

Because of the difficulties associated with the amount of carbonate and nitrate found in cells from previous experiments (9,10), one 6 ampere-hour and one 20 ampere-hour cell were subjected to a chemical analysis for carbonate and nitrate. Samples of positive electrodes, negative electrodes, and separators from

each cell were analyzed. The results of these analyses were used to calculate the total weight of nitrate and carbonate in the cells as KNO_3 and K_2CO_3 . The quantities of carbonate and nitrate found in the commercial cells are listed in Table III.

The second second second second second second second second second second second second second second second s

Cell	Total Weight KNO ₃ , mg	Total Weight K ₂ CO ₃ , g
20 amp-hr	93	24.1
6 amp-hr	24	7.4

TABLE III. QUANTITIES OF KNO3 AND K2CO3 IN COMMERCIAL CELLS

The quantities of impurities in the two cells are in about the same ratio as their rated capacities and appear to be high enough that they could be detrimental to the cells operation. Since no other cells were available, these commercial cells were still used for the verification tests. Because of relatively large concentrations of carbonate and nitrate, recommendations were made that the effect of carbonate on charging voltages be taken into account by emphasizing matching of cells at end of charge and by appropriate consideration of "charged qualities" in terms of initially measured values.

Cell Characterization and Matching

In order to characterize and obtain the data required to match the cells, both the 6 ampere-hour and 20 ampere-hour cells were subjected to the following steps:

- (1) Weigh cells and visually inspect for leaks.
- (2) Charge at C/20 for 48 hours with 1.50-volt limit.

(C = manufacturer's rated capacity)

- (3) Discharge at C/3 to 1.0 volt.
- (4) Short with 1-ohm resistors for 16 hours then dead short for 1 hour.
- (5) Charge at C/10 for 24 hours record voltage.
- (6) Discharge at C/2 to 1.0 volts record voltage and time.

- (7) Short with 1-ohm resistors for 16 hours then dead short for 1 hour.
- (8) Charge at C/10 for 24 hours record voltage and measure internal resistance.
- (9) Discharge at C/2 to 1.0 volts measure internal resistance and calculate capacities to 1.1 volts and 1.0 volt.

The results of these characterization and capacity determination steps are given in Tables IV and V.

The cells were matched into groups using their capacities determined in Step (9) and their internal resistances and charge voltages obtained in Step 8.

Test Apparatus

Once matched into six-cell groups, the cells were mounted in environmental chambers on Plexiglas base plates. A thermocouple was placed on one side of the cell in the approximate electrode center between the cell case and the retaining plate. The cells were then wired in series to a charge-discharge cycling rack to provide automatic testing. Figure 1 shows a circuit diagram of this test apparatus. The apparatus requires testing the cells as a group with no facilities for automatically disconnecting a cell if its voltage fall, below acceptable levels. A group safety feature is provided such that the group will be open circuited if the group voltage falls below a preset limit.

The individual cell discharge voltages, individual cell temperatures, group voltages and group currents are recorded on 24-point recorders. The voltage recorders have been modified so that data are recorded only during the last 40 to 45 seconds of each discharge. During this recording period, the speed of the recorders is such that one of the 24 recorder points is measured and recorded every 1.67 seconds. Thus, it takes 40 seconds to read all 24 recorder points.

and the first of the state of the second of the state of the state of the state of the state of the state of the

SZSZ AR SARAN SZALZA, IN A 11 - 1943 - 2001 A 2001 A 2001 A 2001

Data Analysis

Because of the data problems previously discussed, ⁽⁹⁾ a simplified method of data handling was devised for the verification tests. As explained in the previous section, individual cell data are recorded only during the last 40 beconds of each discharge cycle. This data, along with the discharge current

CHARACTERIZATION, CAPACITY, AND FAILURE DETERMINATIC:: DATA FUR CHARGE STRAIN-RATE TEST TABLE IV.

No. of Concession, Name

(Statements)

		Cell	End of	Internal	Capac	ity,					Fa	ilure
Group No.	Cell	Weight, erams	Charge, volve	Resistance, 0 v 10-3	To 1 1 .	-hours	n,	8,	Th 10-2	ļ	Volt,	Quality,
		0			A 1.1 DI	A 0.1 01	liours	g/ amp-nr	- 01 X XT	ugtk	volt	amp/gram
C-1	750	886.4	1.47	2.99	26.7	27.2	2.667	44.32	2.24	2.65	0.978	0.378
	752	890.3	1.48	2.96	26.7	27.3	=	44.52	2.22	2.64	0.978	0.379
	759	888.1	1.47	2.90	26.5	26.8	=	44.41	2.18	2.58	0.978	0.388
	761	892.6	1.46	2.85	26.5	26.8	:	44.63	2.14	2.55	0.979	0.392
	764	888.8	1.48	2.87	26.5	27.2	=	44.44	2.15	2.55	0.978	0.392
	772	885.2	1.47	2.65	26.5	26.8	:	44.26	1.99	2.35	0.980	0.425
C-1	753	RR6.9	1 47	2 06	5 70	0 76						
T F	755	5 000 5 004	1 47	2 03	1 0 7 C	0.01	100.42		77.7	Z.03	0.978	0.380
	756	2 200		, t t J J	- 07	0.12		44./3	2.20	2.62	0.978	0.382
	076	0.000	- t - r	40°7	707	21.0	2 3	44.28	2.17	2.56	0.978	0.391
		88/ •U	1•4/	2 ,88	26.8	27.2	=	44.35	2.16	2.55	0.978	0.391
	/63	891.3	1.47	2.84	26.7	27.2	2	44.57	2.13	2.53	0.979	0.395
	1//	ð86 •5	1.47	2.75	26.7	27.2	2	44.33	2.06	2.44	0.979	0.410
с-3 С-3	743	887.0	1.47	2.98	26.3	0 20	7 667	11. 25		22 6		
	749	201.6	1 47						+7 • 7	C0•7	0.410	0.5/8
	147		1 - 4 /	2.20	20.3	. / .	: :	44,23	2.24	2.64	0.978	0.379
	104	88/.9	1.47	2.90	26.5	27.0	=	44.40	2.18	2.58	0.978	0.388
	(9)	887.3	1.47	2.88	26.3	26.7	=	44.37	2.16	2.56	0.978	0.391
	769	882.8	1.48	2.70	26.2	26.7	=	44.14	2.03	2.39	0.980	0.419
	773	187.3	1.47	2.85	26.2	26.8	=	44.37	2.14	2.53	0.979	0.395
C-4	751	891.1	1.47	2.89	27.0	27.2	2.667	44.56	2.17	2.58	0.978	0.388
	762	389.7	1.47	2.93	27.2	27.5	=	44.49	2.24	2.66	0.978	0.376
	768	885.0	1.48	2.64	27.0	27.7	=	44.25	1.98	2.34	0.980	0.427
	770	890.1	1.47	2.82	26.8	27.3	=	44.51	2.12	2.52	0.979	0.397
	774	388. 2	1.49	2.78	27.2	27.7	=	14.41	2.09	2.48	0.979	0.403
	775	885.8	1.48	2.75	27.0	27.3	=	44.29	2.06	2.43	625.0	0.411
C-5	744	887.5	1.48	3.03	26.2	27.0	2.667	44.38	7.27	2.69	0.977	0 372
	745	887.7	1.47	3.01	25.7	26.3	=	44.39	5.26	29.0	0 077	0.372
	747	888.8	1.49	2.98	25.5	26.0	=	44.44	2.24	2 45	0.078	0,010
	757	884.4	1.47	2,88	25.7	26.2	=	44.22	2 16 2 16	0 1 1 1 1 1 1 1 1 1	0,070	
	758	883.9	1.47	2.89	26.2	26.5	=	44.20	2.17	4 C	0 0 0 0	2010
	767	893.1	1.48	2.84	25.5	26.5	=	44.66	2,13	2.54	0.070	76C.0
											~ ~ ~ ~ ~ ~	

antiste and the

3

:

č,

¢

÷

9

. આ આ ગામ આ તેમ હામ છે. આ આ ગામ આ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ ગામ ગામ આ ગામ ગામ ગામ ગામ

TABLE V . CHARACTERIZATION, CAPACITY, AND FAILURE DETERMINATION DATA FOR DISCHARGE STRAIN-RATE TEST

9<u>17</u>

Lughter and

the barrand and the state of the second state of the second states of th

and the

		Cell	End of	Internal	Capac	itv.		الم التي التي التي التي التي التي التي التي			Fa	i lure
Group No.	Cell No.	Weight, grams	Charge, volts	Resistance, C x 10 ⁻³	ampere To 1.1. v	To 1.0 v	n, hours	g, g/amp-hr	IR × 10 ⁻²	ngIR	Volt, volt	Quality, amp/gram
	362	271.1	1.48	3.1	7.30	7.35	2.0	45.18	0.93	0.34	1.9.0	1.19
	364	271.1	1.48	3.1	7.30	7.40	=	45.18	0.93	0.84	166.0	1.19
	378	271.5	1.48	2.9	7.30	7.35	=	45.25	0.87	0.79	0.991	1.27
	380	269.7	1.48	2.8	7.30	7.35	=	44.95	0.84	0.76	0.992	1.32
	381	272.2	1.48	2.9	7.30	7.40	=	45.37	0.87	0.79	166.0	1.27
	382	268.8	1.48	2.8	7.25	7.40	:	44.80	0.84	0.75	0.992	1.33
D-2	356	272.7	1.48	3.2	7.15	7.25	1.0	45.45	1.92	0.87	0.981	1.15
1	359	272.3	1.48	3.2	7.20	7.25	=	45.38	1.92	0.87	0.981	1.15
	361	270.8	1.48	3.1	7.20	7.35	=	45.13	1.86	0.84	0.981	1.19
	365	269.5	1.47	3.0	7.20	7.30	=	44.92	1.80	0.81	0.982	1.23
	366	271.0	1.47	3.1	7.25	7.30	=	45.17	1.86	0.84	0.981	1.19
	376	272.8	1 .49	3.0	7.25	7.40	=	45.17	1.80	0.82	0.982	1.22
0-3	362	270.5	1.48	3.1	7.40	7.50	0.5	45 . 08	3.72	0.84	0.963	1.19
	367	270.5	1.48	3.0	7.40	7.50	=	45.08	3.60	0.81	0.964	1.23
	369	275.1	1.48	3.2	7.45	7.60	=	45.85	3.84	0.88	0.962	1.14
	371	269.1	1.48	3.2	7.40	7.50	=	44.85	3.84	0.86	0.962	1.16
	372	273.0	1.48	3.2	7.40	7,50	=	45.50	3.84	0.87	0.962	1.15
	387	272.8	1.48	3.0	7.40	7.45	=	45.47	3.60	0.82	0.964	1.22
D-4	360	274.0	1.48	3.1	7.55	7.60	0.25	45.67	7.44	0.85	0.926	1.18
	368	271.3	1.48	3.3	7.60	7.65	=	45.22	7.92	06*0	0.921	1.11
	373	271.7	1.48	3.2	7.50	7.55	=	45.28	7.68	0.87	0.923	1.15
	375	273.9	1.48	2.9	7.60	7.70	=	45.65	6•96	0.79	0.930	1.27
	385	268.6	1.48	3.0	7.45	7.50	=	44.77	7.20	0.81	0.328	1.23
	386	271.7	1.49	3.1	7.65	7.80	=	45.28	7.44	0.84	0.926	1.19
D-5	370	271.6	1,47	3.2	7.35	7.40	0.125	45.27	15.36	0.87	0.846	1.15
	377	270.2	1.48	2.8	7.35	7.40	=	45.03	13.44	0.76	0.866	1.32
	379	272.0	1.48	2.9	7.35	7.40	2	45.33	13.92	0.79	0.861	1.27
	383	272.3	1.48	3.0	7.35	7.55	=	45.38	14.40	0.82	0.856	1.22
	384	273.7	1.48	2.9	7.35	7.45	=	45.53	13.92	0.79	0.861	1.27
	388	272.7	1.49	2.8	7.35	7.45	=	45.45	13.44	0.76	0.866	1.32

ì.

۶ ۵

· · · ·

, **.**.,

•



and charge time to the Step 1 to Step 2 switchovers, are copied onto punch card transcript forms daily. Once a week these punch card transcripts forms are transferred to the Computer Science Center at WPAFB for card punching and then computer analysis. The computer analysis consists of calculating the individual cell qualities, an average group quality, and a group charging voltage quality. The computer program then prints out the results of these calculations along with the cycle number, discharge current, and recording date for each group by ascending cycle number. Once 200 cycles of data have been accumulated for a group, average-cell-quality-values-versus-cycle, and group-charging-voltagequality-versus-cycle plots are made.

A new definition of electrical discharge quality was leveloped for these tests. This new definition is:

$$Q_{E} = \frac{E_{eod} + IR}{ngIR}$$
(10)

where

 E_{eod} = the end-of-discharge voltage,

- I = the discharge current in amps,
- R = the initial internal resistance in ohms,
- n = the number of hours required to fully diacharge the cell at Current I, and

g = the specific capacity of the cell in grams per amp-hr.

This definition of quality has units of amps/gram. The definition was developed such that it tends to be independent of charge or discharge rate and cell size. The only experimental parameters left to assign for real quality measurements are (1) the temperature and (2) the depth of discharge. The manufacturer's rated capacity was used to determine the values of I, g, and the depth of discharge. The actual capacity could have been used since the n and I exactly compensate one another in the quality calculation. The values required for individual cell quality measurements are given in Table IV and V. The individual cell failure qualities are also calculated by Equation (10), except that 1.00 volt. compensated for IR, was used instead of the end-of-discharge voltage (E_{eod}).

Although the quality in Equation (10) is directly proportional to a corrected end-of-discharge voltage, there are several important reasons for retaining the term "quality". The first reason deals with the conceptual basis for quality and a need to measure both initial values and their rates of degradation. A history of requirement: for the development of a quality equation are presented in a special

report "Rate of Discharge for Accelerated Life Tests of Nickel-Cadmium Cells".⁽¹⁰⁾ The second reason for using quality deals with the necessity of comparing the results of one test with the results of another. The meed for having to specify such things as cell sizes, charging rates, and discharging mates have been taken into account with the use of quality as defined in Equation (10).

The previously mentioned "group charging voltage quality" was calculated by:

$$Q_{\rm cv} = \frac{{\rm amps \ x \ time}}{{\rm AH}}$$
(11)

where amps = Step 1 constant charging current in amps,

time = time in hours required to reach the Step 1 to Step 2
switchover point, and

AH = ampere-hours removed during cycling tests.

This group quality was used in order to help relate the experimental results to a theoretical framework for accelerated life tests.

Test Results

Table VI lists the status of the verification test groups as of June 7, 1972. All cell groups listed as cycling are actually on open circuit awaiting safety features to be added to the cycling apparatus. Once the safety features have been added, the cycling of these groups will be resumed.

Group C-3 failed because of mechanical problems in the cycling equipment. The discharge relay for this cycling rack stuck closed during the discharge of Cycle 244. As a result, the cells were subjected to an additional discharge of about 335 ampere-hours before the group was manually put on open circuit. All cells had burst their retaining plates and Cell 765 had ruptured. A small hole was drilled into each cell to relieve the pressure buildup. All cell, are stored in plastic bags awa**iting** further examination.

Two cells (752 and 772) of Group C-1 also Jurst their retaining plates. Preliminary investigations indicated the group was subjected to the high rate (48.0 amperes) of charge for the entire 3-hour charge period. This overcharge appears to have happened during Cycles 94, 95, and 96 before the group was open circuited. There appeared to have been no other equipment malfunctions.

ALTER AND SHOW

Group	Charge C	urrent,	Dicharge Current,	Status,
No.	High	Low	amp	cycles
		<u>Charge Stra</u>	<u>in-Rate_Test</u>	
C-1	48.0	2.0	7.5	Cycling (97)
C-2	24.0	2.0	7.5	To be started
C-3	12.0	2.0	7.5	Failed (244)
C-4	6.0	2.0	7.5	Cycling (791)
C-5	3.0	2.0	7.5	Cycling (1508)
	D	ischarge St	rain-Rate Test	
D-1	3.0	0.6	3.0	Cycling (792)
D-2	3.0	0.6	6.0	Cycling (618)
D-3	3.0	0.6	12.0	Cycling (560)
D-4	3.0	0.6	24.0	Cycling (301)
D-5	3.0	0.6	48.0	Cycling (171)

TABLE VI. SUMMARY OF VERIFICATION TEST GROUPS AS OF JUNE 7, 1972

A suggested explanation is that a thermal runaway condition exist in which the temperature of the ceils exceeded 70 degrees centigrade durit and the switchover for the Step 1 to the Step 2 rate of charge was never x If this explanation is correct, the failure mechanism may be different frc expected at normal charging rates, and a limit on stress levels may have b identified for the first time on this resear 'h program. If poor controls responsible for the thermal runaway, then the test conditions were changed than a limiting failure mechanism. Investigation of the occurrence surrou this problem are continuing. The results of the verification tests to date were used for pre-nary estimates of the acceleration factors for the various groups. These results are listed in Table VII. Since the degradation of cell quality do appear to be linear, the acceleration factors listed in Table VII are exp change with coling. As a result, an in-depth analysis of the data was no performed for this report. Several points concerning the data in Table VII need to be discu follows: (1) The data indicate that both the Charge Strain-Rate Test and Strain-Rate Test should lead to valid accelerated tests. (2) The results Group D-3 appear inconsistent with other groups the test. An examination dat: showed experimental test problems which will word to the A suggested explanation is that a thermal runaway condition existed in which the temperature of the ceils exceeded 70 degrees centigrade during charge, and the switchover for the Step 1 to the Step 2 rate of charge was never reached. If this explanation is correct, the failure mechanism may be different from that expected at normal charging rates, and a limit on stress levels may have been identified for the first time on this resear h program. If poor controls were responsible for the thermal runaway, then the test conditions were changed rather than a limiting failure mechanism. Investigation of the occurrence surrounding

Married Married

「「ないのちまえんない」ときます

こうちょうちょうちょうちょう いちのちょうちょうちょうちょうちょう

a table in the sur

とくそう チャッチラ

× P

and the state of the second state of the state of the second state of t

The results of the verification tests to date were used for preliminary estimates of the acceleration factors for the various groups. These preliminary results are listed in Table VII. Since the degradation of cell quality does not appear to be linear, the acceleration factors listed in Table VII are expected to change with cucling. As a result, an in-depth analysis of the data was not

Several points concerning the data in Table VII need to be discussed as follows: (1) The data indicate that both the Charge Strain-Rate Test and Discharge Strain-Rate Test should lead to valid accelerated tests. (2) The results of Group D-3 appear inconsistent with other groups the test. An examination of the data showed experimental test problems which will need to be analyzed with regard to the data. However, a separate entry was made in Table VII which shows that Group D-3 had the same change in quality but that it occurred in only half as many cycles. If this correction is used, the overall trend becomes self-consistent. (3) The acceleration factors are based on only two points per group and therefore will require further data and physical analysis. (4) The results listed in Table VII were obtained without the data manipulations required of previous tests. (9,10) These encouraging results imply significant improvements have been made in the operations involved in both testing and in data collection.

Failure Analysis

A failure analysis will be performed on all failed cells in order to demonstrate that the failure mechanism at a higher stress level is the same as the failure mechanism at the lower, or normal, stress level. Table VIII lists a failure analysis procedure which is recommended for use on the cells from the Charge Strain-Rate and Discharge Strain-Rate tests. This procedure should provide the data required to determine failure determinates. After the failure determinates, the failure mechanisms should be determined. The results of the failure analysis procedure should also assist in this determination of mechanisms.
TABLE VII. PRELIMINARY ESTIMATE OF ACCELERATION FACTORS FOR VERIFICATION TEST GROUPS

の記録のとい

对中国的建立中国的

	mating conditions.	present "normal" ope	at 1.0 to rep	ictor indexed	Acceleration fa	(8)
1.9	0.000188	277	1.388	1.440	12.0	D-3
9.1	0.000893	168	1.285	1.435	48.0	D-5
5.2	0.000510	286	1.285	1.431	24.0	D-4
1.0	0.000095	546	1.388	1.440	12.0	D-3
1.5	0.000152	598	1.4,05	1.486	6.0	D-2
1.0 ^(a)	0.00098	780	1.504	1.580	3.0	D-1
	<u>ells)</u>	ite Test (6 amp-hr C	arge Strain-Ra	Disch		
Acceleration Factor	Change in Q Units Per Change in Cycles	Cycles Between Initial and Last Printout	Average Q at Lest Printout	Average Initial Q	Discharge Current, Sr?	Group No.
1,0 ^(a)	0.000016	1,495	0.463	0.487	3.0	C-5
1.4	0.00023	780	0.488	0.506	6.0	C-4
4.3	0.000068	234	0.477	0.493	12.0	с-3
					24.0	C-2
13.9	0.000223	94	0.474	0.495	48.0	C-1
	<u>11s</u>	e Test (20 amp-hr Ge	ge Strain-Rate	Char		
Acceleration Factor	change in Y units Per Change in Cycles	Cycles between initial and Last Printout	Average V at Last Printout	Average Initial Q	Step I Charge, amp	Group No.
	others day of matter	Ē				

いちん ほうしんかがくないない いい

the second strategy and the state of a second strategy and

こうごうかいたちからしることなどがないないないないないないないないないできょうかいできょう

1

i

TABLE VIII. FAILURE ANALYSIS PROCEDURE FOR CELLS IN VERIFICATION TESTS

and the state of the later

いいないないないないないないである うちない イストット・コスション・コン いたいちょう しょうしょう ちょうちょうちょうちょう ちょうちょう ちょうちょう ちょうちょう ちょうちょう ちょうちょう ちょうちょう しょうしょう しょうしょう しょうしょう しょうしょう しょうしょう

ited as so interesting the states of the states of the states of the states of the second of the second of the

1. Physical Examination

- 1.1 Remove retainer plates, weigh cell, compare with original weight
- 1.2 Check leak, use 0.25 percent phenolphthalein solution
- 2. Electrical Performance
 - 2.1 Replace retainer plates, measure open-circuit voltage
 - 2.2 Measure internal resistance
 - 2.3 Discharge at C/2 rate to 1.0 volu using the manufacturer's rated capacity
 - 2.4 Charge at C/10 for 24 hours with 1.50-volt limit, record time to 1.50 volts or voltage after 24 hours
 - 2.5 Repeat 2.3
 - 2.6 Repeat 2.4
 - 2.7 Repeat 2.5
 - 2.8 Check for internal shorts by:
 - (a) Discharge through 1-ohm resistor until cool
 - (b) Short terminals for 16 hours
 - (c) Charge 5.0 minutes at C/10 rate
 - (d) Read open-circuit voltage immediately and after 24-hour stand. A decay of more than 0.10 volt indicates possible short.
- 3. Disassembly
 - 3.1 Remove cell can avoiding damage to internal components; note any irregularities
 - 3.2 Place electrode pack in Soxhlet extractor and extract the electrolyte for 48 hours under nitrogen atmosphere. Analyze the extracted electrolyte for KOH and K₂CO₃ by titrating with standard 1.0 N acid.
 - 3.3 After removing the electrode pack from the extractor, dry the pack at 60 C for 24 hours. Weigh the dry pack.
 - 3.4 Place electrode pack between Lucite retainer plates clamped to hold the cell tightly compressed. Place pack in beaker with sufficient 30 weight percent KOH to cover electrodes and separators. Allow pack to soak for 24 hours. Charge at C/3 rate for a time equal to 1.5 times the discharge time of 2.7.
 - 3.5 Discharge at C/2 rate in flooded state. Record potentials of positives and negatives against Hg/HgO reference electrode as function of time during discharge. Continue discharge until both sets of electrodes have been fully discharged.
 - 3.6 Disassemble electrode pack and note any irregularities such as burned cr disc lored separator and damaged electrodes.

Martin and the second states of the second states o

During charge or discharge of sealed cells, the major portion of gradients of pressure, voltage, temperature, or concentration tends to exist between electrodes. Therefore, it is reasoned that thicker electrodes will allow more efficient use of active material over a longer period of time because the number of gradients will be decreased and the severity of gradients will be essentially unchanged. For the same reason, the physical strains of depth of discharge, heat content, volumes of gas, and charge densities can be increased without any commensurate increase in degradation rates. The thicker substrates and moderate pore filling both have to do with mechanical stresses. It is well known, for example, that cadmium expands 230 percent in going to cadmium hydroxide. Room must be allowed for this expansion to take place. A moderate pore filling will allow a more effective use of material that is present without commensurate stre: ~ and strain on the mechanical structure of the porous electrodes. The prolonged final washing of electrodes to eliminate carbonate should also enable the thorough removal of nitrate. Preliminary tests on this program, using commercial electrodes, showed a great deal of difficulty with the "nitrate shuttle". For longer life cells, the extra precautions of removing all carbonates and nitrates cannot be over emphasized. With regard to electrodes matching on the basis of impregnation weight gains, it was reasoned that not all of the active materials impregnated into electrodes are ever available as an electrical capacity. Consequently, two electrodes which showed the same electrical capacity but differed in weight gain would have a substantial difference in pore filling or a substantial difference in some other property which limits the access to this active material. It seemed logical to expect that if cells were matched on both electrical and weight capacities, that the physical performance of electrodes in a cell would tend to be more identical and therefore a combination of electrodes would tend to be longer lived. Polypropylene separator was suggested because nylon separators have been found to oxidize particularly under over-charge conditions. Separator oxidation is the main source of carbonate pickup during use of sealed cells and polypropylene is relatively stable to oxidation. The solid-nickel components and mercury-grade potassium hydroxide are recommended because metallic and ionic impurities are to be avoided in any cells having long life.

「おうちをないろうないないのでないないとういろ、 いまい、

「 」というかっているというでないないないで、ころものないないであるともないないで、 しゃ こうちょうている し

いたかいいでもうないのでいたいとうな

The fabrication of these longer life cells was described in detail in the Second Annual Report. ⁽¹⁸⁾ Plaques were made 0.069 inch thick with Type 287 carbonyl nickel powder. The plaques were impregnated with nickel or cadmium hydroxide using a standard vacuum impregnation technique followed by cathodic polarization. Five impregnation cycles were used for each electrode. The impregnated electrodes were formed in excess 30 weight percent KOH containing 0.1N LiOH. After the formation cycles were completed, the electrodes were washed with 40 changes of deionized water to insure complete removal of the KOH electroyte so as to prevent any subsequent formation of $K_2^{CO}_3$. The electrodes were stacked between layers of filter paper and vacuum dried for 3 days at 6J C.

ななどのないではないとないであるというないであるというであるという

المراجع المراجع والمحالية المحافظ والمحافظ The 3.0 x 6.0 x 0.060-inch plaques had an average weight of 35.8 grams. Their everage porosity was 78.0 percent with a standard deviation of 0.9 percent. The average weight gain after five impregnation cycles for the cadmium electrodes was 26.9 grams with a standard deviation of 1.1 grams. The average weight gain after five impregnation cycles for the nickel electrodes was 15.8 grams with a standard deviation of 0.9 gram. These figures correspond to a loading of 2 12 grams of Cd(OH)₂ per cm³ of pore volume for the negative electrodes and 1.25 grams of Ni(OH) $\frac{1}{2}$ per cm³ of pore volume for the positive electrodes. These loadings are slightly higher than the loadings of 1.75 and 1.0 gram per cm³ of pore volume used by Fleischer for the negative and positive electrodes respectively.⁽¹⁹⁾ The loadings in the present work correspond to 44.4 percent of the porous volume filled with Cd(OH), of density 4.79 g/cm³ and 31.8 percent filled with Ni(OH), of density 3.94 g/cm³.⁽²⁰⁾ By comparison, the loadings of commercial cells calculated from tear-down analysis data⁽⁴⁾ were from 50 to 58 percent of the pore volume filled with Cd(OH), and from 42 to 55 percent of the pore volume filled with Ni(OH). The lighter loading of the electrodes for use in the cells designed to have increased life fulfills the third irem above "moderate pore filling to lower mechanical stresses".

Cells in Hermetically Sealed Cases

A small number of cells were fabricated in hermetically sealed cases to demonstrate the capability for complete cell fabrication at the Aero Propulsion Laboratory. These cells were assembled in stainless steel cans with nickel seal covers. The cans and cover assemblies (part numbers 201A3019-22 and 152B504AD-1, respectively) were purchased from the Battery Business Section of the General Electric Company. Figure 2 shows a drawing of the cell and its assembly details.

The sintered nickel plaques used in these electrodes had dimensions of 2.5 x 6 x 0.060 inches. Their average porosity was also 78 percent. The average weight gain after five impregnation cycles was 23.1 grams for the negative electrodes and 11.3 grams for the positive electrodes. These figures correspond to a loading of 2.56 grams of $Cd(OH)_2$ per cm³ of pore volume of the negative electrodes. Expressed as percent of the pore volume filled, these figures correspond to 53.7 percent for the negative electrodes and 32.0 percent for the positive electrodes. Each of these impregnated master plaques was cut into three electrodes after they were formed in 30 weight percent KOH.

Electrolyte Volume in Sealed Cells. The decision of how much electrolyte to use in a sealed nickel-cadmium cell which operates in a scarved electrolyte condition requires a number of considerations. If too litule electrolyte is used, the cell will have a high internal resistance and a low ampere-hour capacity. much electrolyte is used, high internal pressures may occur because the If oxygen recombination is decreased. Lunn and Parker⁽²¹⁾ have studied this rate problem of how much electrolyte to add to sealed nickel-cadmium cells. They found that the amount was critical to maintain both low internal resistance and low internal pressure during charging. The optimum amount of electrolyte in their study was equal to the volume that filled the pores of the plates plus about 25 percent of the total quantity retained by the separator after immersion in the electrolyte and draining off the surplus. Care must be taken in the use of such a rule of thumb because the state of charge was not given by Lunn and Parker. Fleischer⁽²²⁾ has stated that the rule of thumb for patteries for space applications which contain polypropylene separators is to use a weight of electrolyte equal to 17 percent of the element weight. All such rules of thumb imply that certain other parameters of cell design such as porosity, loading, and amount of separator are within certain limits for the rules to be valid.

The average total weight of the elements in the demountable improved cells (three negative electrodes, two positive electrodes, plus separator) was 175 grams. The volume of 30 percent KOH of density 1.285 g/cm³ which correspond to various percents of the element weight are



33

2 és s

OF STATISTICS

Percent of Element Weight	Volume of 30% KOH, cm
15	20.2
16	21.8
17	23.2
18	24.4
19	25.8

The demountable improved cells were filled with 20.0 cm³ of 30 weight percent KOH. According to the rule described above for calculating the quantity of electrolyte (17 percent of the element weight), this was too little electrolyte. However, in view of the fact that the electrodes in these cells were more lightly loaded with active material than most commercial cells, it was felt that slightly less electrolyte would be required.

The elements of these cells were almost dry to the touch when the cells were disassembled during subsequent failure analysis. Furthermore, as is discussed in connection with failure analysis, one cell which was inadvertently filled with about 25 cm³ of electrolyte instead of 20 cm³ had a slightly lower initial internal resistance and did not fail during its evaluation. For these reasons the hermetically sealed improved cells should be filled with 25.0 cm³ of 30 weight percent KOH. A volume of 25.0 cm³ of 30 percent KOH would constitute 18.4 percent of the element weight in the hermetically sealed improved cells.

Evaluation of Longer Life Cells

Cells in Demountable Cases

Four groups of five cells were selected and matched for evaluation. One of the groups was evaluated at each of the following temperatures: 40 C, 25 C, 10 C, and -5 C. The cycling regime for all four groups was the same. Cells were charged at the C/2 rate until the group voltage averaged 1.45 volts per cell at a temperature of 25 C. After reaching this voltage, the cells were switched to the C/10 charge rate for the remainder of a three hour charge period. The switchover voltage was compensated for temperatures other than 25 C using a temperature coefficient of -0.45 mV/deg C. Discharge was at the C rate for one-half hour, followed by one-half hour of open circuit stand. Thus, the total cycle time was four hours, three hours of charge, one-half hour of discharge, and one-half hour of open circuit stand. Cell failure was defined as the cell's voltage falling

below 1.1 volts, compensated for temperature and internal resistance, before the end of the specified discharge time.

Cells were removed from test at the end of discharge after they had failed. It was planned that cycling of a group of cells would be terminated when four cells had failed. However, since the cycling equipment was required for the verification tests before all four groups had failed, the evaluation of the demountable improved cells was terminated before all groups had failed. Nevertheless, it was shown that the improved cells had longer cycle lives than did cells used for the temperature experiment under the same conditions. Following is a summary of the results of the evaluation of the demountable improved cells and the temperature-experiment cells made with commercial electrodes. 「「「「「「「「「」」」」

Operating Temp., C	Cycles to Failure Temp-Experiment Cells(a)	Cycles to Failure Improved Cells
40	12	25 ^(a)
25	257	1169 ^(a)
10	163	575 ^(b)
-5	81	772 ^(b)

(a) Test completed - four failed cells.

いたなためがたいにしたのためになったいというない

(b) Test terminated - two failed cells.

<u>Failure Analysis</u>. The following abbreviated failure analysis procedure was used in the demountable improved cells which failed during evaluation.

- (1) Physical Examination
 - 1.1 Weigh cell, compare with weight obtained after assembly.
 - 1.2 Check leak, visually, by observing any crusty deposits.

- (2) Electrical Performance
 - 2.1 Measure open circuit voltage.
 - 2.2 Measure internal resistance.
 - 2.3 Measure capacity of each set of electrodes according to the following procedure. Charge at C/2 to replace exactly the capacity removed during the last discharge on test. (Since cells were removed from test at the end of discharge cycle and the depth of discharge was 50 percent, the charge time required in this step was one hour.) Record the potentials of both electrodes with respect to the cell case as a reference electrode during charge. Discharge at C/2 rate until both electrodes are completely discharged as indicated by their potential-time curves with respect to the case reference electrode. Determine the limiting set of electrodes, their capacity, and the excess capacity of the other set.
 - 2.4 Check for internal shorts by (a) short cell terminals for 16 hours, (b) charge at C/10 for 5.0 minutes, (c) read open circuit voltage immediately and after 24 hour using a meter with an impedance of at least 100,000 ohms/volt. A decay of more than 0.10 volt indicates a possible internal short.

- (3) Disassembly
 - 3.1 Open cell and remove cell element (electrode and separator stack). Examine for burns, broken tabs, or other signs of damage.
 - 3.2 The element from the last cell which fails in each group will have its electrolyte extracted in a Soxhlet extractor and analyzed for KOH and K_2CO_3 . The weight of the dry separator after extraction will be compared with the initial separator weight.
 - 3.3 The separator from cells other than the last to fail in each group will be removed from the element after Step 3.1 and washed in deionized water and dried. Its weight will be compared with its initial weight.

Due to lack of time and funds, the complete failure aralysis described above was not performed on all the cells. In most cases, only the physical examina-

tion and electrical performance portions were performed. The data obtained during failure analysis are given in Tables IX, X, XI, and XII.

CONSIGNATION OF CONSIGNATION OF CONSIGNATION OF CONSIGNATION OF CONSIGNATION OF CONSIGNATION OF CONSIGNATION OF

語言させていいる

The data in these four Tables do not always indicate that a cell was a failure because the capacities of both electrodes are greater than the 90 ampminutes required for discharge during cycling. However, these capacities are measured to the knees of the discharge curves of the individual electrodes and include capacity that was available for discharge at a voltage of less than 1.10 which was specified as the failure limit. For example, Cells 203, 210, and 213 in Table X all have capacities greater than 90 amp-minutes, but their capacities to a cell voltage of 1.10 were

<u>Cell</u>	Capacity to 1.10 volts, amp-min
203	85.5
216	88.5
213	96.0

Since the cells were required during test to provide 90 amp-min at a voltage greater than 1.10, Cells 203 and 210 had definitely failed. Cell 213 would not be considered a failure during the failure analysis procedure since it provided 96 amp-min at a voltage greater than 1.10 volts. However, the cell could possibly have recovered during the time after it was removed from cycling so that it met the requirements during failure analysis even though it did not meet them during cycling.

During discharge, the potentials of both electrodes in each cell decreased with respect to the stainless steel case used as a reference electrode. The decreases of electrode potential from the start of discharge until the cell voltage reached 1.10 were

	Decrease in Elec	trode Potential
<u>Cell</u>	Negative, volts	Positive, volts
203	0.08	0.24
210	0.05	0.24
213	0.05	0.24

TABLE IX.	FAILURE ANALYSIS DATA	FOR DEMOUNTABLE	IMPROVED	CELLS
	EVALUATED AT 40 C			

Cell	202	205	208	221	224
Cycles to Failure	7	8	25	7	NF
Weight Change, g	+0.4	0.0	-0.2	0.0	-2.4
Open Circuit Voltage	1.17	1.21	1.22	1.10	1.18
Internal Resistance, milliohms					
Before Cycling	8.8	6.5	7.7	4.8	5.1
After Cycling	21.3	16.1	16.8	10.9	5.0
Positive Capacity, amp-min	68.3	87.0	114.0	108.0	80.3
Negative Capacity, amp-min	208.5	243.8	303.0	81.0	606.8
Voltage Loss During Internal Shorts Test	0.62	0.18	0.17	0.85	0.14
Electrolyte Analysis					
KOH, g			8.60		
к ₂ со ₃ , g			0.80		

			فاعت وعيوب غالك فل	يعبد الأعربية عيديني والان	
Cell	201	216	203	210	213
Cycles to Failure	317	317	1,169	1,169	1,169
Weight Change, g	-0.2	-0.6	-0.7	-0.6	-0.5
Open Circuit Voltage	1.26	1.24	1.25	1.28	1.27
Internal Resistance, milliohms					
Before Cycling	7.8	9.2	8.7	11.3	9.7
After Cycling	46.1	156.0	17.6	16.2	11.1
Positive Capacity, amp-min	90.0	137.0	159.0	141.0	164.3
Negative Capacity, amp-min	210.0	69.0	288.8	347.3	422.3
Voltage Loss During Internal Shorts Test		0.52	0.21	0.16	0.07

TABLE X. FAILURE ANALYSIS DATA FROM DEMOUNTABLE IMPROVED C.37LS EVALUATED AT 25 C

DNum

!

こうかん あん いっしょう しょうかい ないない ないない ないない ないない ないない かいしょう しょうしょう しょうしょう しょうしょう しょうしょう しょうしょう

A LEASE AND

A STATE AND A STATE AND A STATE AND A STATE AND A STATE AND A STATE AND A STATE AND A STATE AND A STATE AND A S

行其前已的自然

	•				
Cell	206	214	2 J4	212	218
Cycles to Failure	236	236	575 (NF)	575(NF)	575(NF)
Weight Change, g	-0.3	-0.2	-0.5	0.0	-0.8
Open Circuit Voltage	1.23	1.26	1.23	1.23	1.22
Internal Resistance, milliohms					
Before Cycling	5.1	7.1	4.4	8.7	7.3
After Cycling	9.1	8.7	3.1	6.5	7.9
Positive Capacity, amp-min	87.0	163.5	94.5	106.5	106.5
Negative Capacity, amp-min	315.8	385.5	274.5	307.5	420.0
Voltage Loss During Internal Shorts Test	0.77	0.36	0.08	0.64	0.35
Electrolyte Analysis					
KOH, g		8.24			
K ₂ CU ₃ , g		1.11			

and the state of the second state of the secon

TABLE XI. FAILURE ANALYSIS DATA FOR DEMOUNTABLE IMPROVED CELLS EVALUATED AT 10 C

228/04093 Constant

The second of the second of the second of the second of the second of the second of the second of the second of

Call	207	215	211	217	219
JEAT					
Cycles to Failure	341	361	.'72 (NF)	772 (NF)	772(NF)
Weight Change, g	+0.2	+0.1	-0.5	0.0	-0.8
Open Circuit Voltage	1.24	1.24	1.26	1.26	1.26
Internal Resistance, milliohms					
Before Cycling	4.5	7.9	6.6	3.3	4.1
After Cycling	5.6	7.2	4.5	2.7	2.2
Positive Capacity, amp-min	170.3	162.0	177.0	159.0	192.0
Negative Capacity, amp-min	356.3	336.0	381.0	304.5	550.5
Voltage Loss During Internal Shorts Test	0.31	0.83	0.13	0.91	0.06
Electrolyte Analysis					
KOH, g		7.77			12.11
к ₂ со ₃ , g		1.76			1.04

TABLE XII. FAILURE ANALYSIS DATA FOR DEMOUNTABLE IMPROVED CELLS EVALUATED AT -5 C

Ьr.

STATISTICS.

an the second second

- 10

日本語のなどには日本の

સ્ટાર્ગ કે આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગા આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગામ આ ગ

Are - 58

These data lead to the conclusion that the failure determinant for these three cells was loss of potential at the positive electrode. The electrodes still had sufficient capacity as shown in Table X, but it was available only at a potential below that required by the cycling conditions. Similar behavior was observed in other cells used in the evaluter of demountable improved cells.

The electrolyte analyses show a slight increase in K_2CO_3 concentration over the electrolyte that was used to fill the cells. The K_2CO_3 concentrations in the electrolytes extracted from the cells after evaluation are in the range of 40 to 90 g/l whereas the electrolyte with which the cells were filled had a K_2CO_3 concentration of 9.5 g/l. The increase in carbonate evidently came from carbonate formed in the electrodes during prior operations even though precautions were taken to minimize carbonate formation.

Children Statistics and Statistics

NEW ROOM DESCRIPTION OF THE PARTY OF THE PAR

The analysis of Cell 219 evaluated at -5 C showed that it was filled with more than 20 cm³ of electrolyte. Its internal resistance was the second lowest prior to cycling and the lowest after cycling. Also it had not yet failed when its evaluation was terminated. These facts suggest that the controlyte volume used in the demountable improved cells was somewhat less the continuum.

Conclusions and Recommendations

The performance of the improved demountable cells was shown to be superior to that of the cells used in the earlier Electrical Stress-Rate of Strain and temperature experiments. This improvement in performance is thought to be the result of at least 3 improvements: (1) polypropylene separators, (2) techniques to minimize chemical impurities, and (3) thicker electrodes in the cells.

The same improvements were incorporated into the hermetically sealed cells assembled at APL. Thus they are expected to have cycle lives similar to the demountable cell, when operated under the same conditions. Based solely on the problem encountered during assembly, two changes are recommended in the design of the hermetically sealed cells. (1) The width of the nickel combs as shown in Figure 2 should be increased from one-half inch to approximately 0.65 inch to make it stronger and less subject to deformation during assembly of the electrode tabs into the slots in the comb. (2) The height of the 'abs on the outermost electrodes should be increased sufficiently so that the tabs of these electrodes can fit well into the slots in the combs. This means that the tabs for the outermost electrodes must be higher than the tabs on the inner

electrodes before assembly so that after assembly the bottoms of all electrodes will be properly aligned. Detailed measurements and drawings will be required in order to determine the correct height of tab for each electrode.

Contraction of the second

のいのないないないのないで、「いってい」を見ていたかいないとうと

and the line products of the

ひっちっていたちないないないでいたかいとうで、

Weeks M.

Evaluations. The performance of the hermetically sealed improved cells will need to be evaluated and compared with the performance of commercial 6 ampere-hour cells by APL personnel or as part of a separate effort. The chargedischarge regime planned for the evaluation is a typical near earth orbital profile. The test temperature is to be 20 C. The total time per orbit for this regime is 110 minutes of which 35 minutes is discharge and 75 minutes is charge. A 50 percent depth of discharge is desired. Thus, the discharge rate must be 6/7C in order to discharge 50 percent of the cell capacity in 35 minutes. Charging is to be done at a constant current C/2 rate to a voltage limit selected on the first cycle followed by a trickle charge at a C/10 for the remainder of the 75 minute charge time. The voltage limit selected on the first charging cycle will be that group voltage obtained after exactly 1 hour of charging at a C/2 rate.

TASK III, SUPPORTING STUDIES

المالية الماسمة والمعالية المحاطرية فالمعالمة والمحاطرة والمحالية المحالية والمحالية والمحالية والمحالية والمحالية

Any research program which seeks to develop accelerated life tests for objects as complex as sealed spacecraft batteries will require a great deal more than a few simple experiments to establish valid tests. A theoretical basis is necessary to gain acceptance of results by electrochemists and by other physicists and physical chemists. A large amount of reliable data will be desired. Ancillary knowledge such as thermophysical properties and statistical methods must be considered and taken in account. A cognizance of relevant published literature is more or less assumed to be a part of a research program. Therefore, this section of the Final Report is to summarize some of the activities and accomplishments from studies to support the main efforts described as Tasks I and II of this program.

Task III (a). Facility Maintenance

The maintenance program developed for the Battery Section of the Aero Propulsion Laboratory at WPAFB has been described previously.⁽²³⁾ The purposes of the program were to have each piece of equipment in a state of readiness to perform its intended functions with a minimum amount of preuse conditioning; to reduce the number of equipment failures; and to provide for speedy repair when failures did occur.

A program was developed which consisted of the following four interrelated parts: (1) record of service-this consisted of a card-file system with an entry for each piece of equipment. This system provided a means of reviewing the history of a piece of equipment for regular and emergency services and for deciding to continue to repair or to replace the item; (2) schedule of repetitive services--a schedule was developed which provided the periodic maintenance and service required by the various pieces of equipment in the system; (3) service procedures--written service procedures were provided for each group of equipments. These procedures included a list of steps for the servicing routine and emphasized any feature that required special attention; and (4) inventory of spare perts--an inventory of spare parts was maintained to provide a list of components available and as an indicator of when to purchase additional parts. The system was later expanded to include a schedule and record of the calibration service performed on equipment in the Battery Section by another group at APL. This additional section of the maintenance program included a listing of all items which required calibration, a schedule of calibration, and a record of any service or calibration performed.

Task III (b). Data Analysis

This data analysis section deals with data coll_cted from simulated orbital life tests started under Contract AF 33(615)-3701 and continued under this contract. Two groups of cells (each consisting of 10 sealed nickelcadmium cells) were evaluated in each of the simulated synchronous (24 hours) and polar (5.31 hours) orbital routines. Some of the synchronous orbit cells successfully completed 1471 days or orbits about the earth with 414 charge-discharge cycles during nine eclipse seasons. These cells were tested at a maximum 80 percent depth of discharge. Some of the polar orbit cells completed 929 calendar days of operation representing 4180 orbits about the earth with 1152 charge-discharge cycles during six eclipse seasons. The polar orbit cells were discharged at a maximum 40 percent depth of discharge. Two special reports ^(24,25) have been published which contain all the information obtained from these orbital tests. The special report entitled "Simulated Orbital Life Tests for Spacecraft Cells, Part I. Procedures and Manually Acquired Data"⁽²⁴⁾ contains information regarding the preparation and matching of cells for life tests and the subsequent life-testing routines for the scaled nickel-cadmium cells in simulated synchronous and polar orbital applications. In essence, this Part I report contains all the manually acquired data such as capacity determinations, internal resistance measurements, and failure analysis data. This report also described problems associated with testing, their possible affects on the cells, and what corrective actions were taken. The failure analysis studies showed an increase in carbonate level over uncycled cells and even over cells which had been cycled but had not failed. It is believed that the increased carbonate levels were a result of oxidation of the separators by the charged positive electrodes.

The second special report entitled "Simulated Orbital Life Tests for Spacecraft Cells, Part II. Automatically Acquired Data, Review and Recommendations"⁽²⁵⁾

described.⁽²⁵⁾

Task III (c). Stress-Strain Models

Some of the important starting aspects of this subtask were summarized under Task I and "A Physical Basis for Accelerated Life Tests". The concluding situation is summarized in Table XIII. This supporting effort on modeling continued piecemeal through the entire program because the ideas of stress and strain are seldom generalized for use by electrochemists or in other nonmechanical subjects. Even so, important progress appears to be in the making in the sense that the third version of generalized stresses and strains in Table XIII appears to be more generally applicable than is the starting generalities in Table I. Onsager and Tribus relations appear to have all the necessary and sufficient conditions for understanding and designing suitable accelerated life tests. If something happens to batteries that cannot be handled with those fundamental relations, it appears that catastrophic events must be turned to for explaining failures. By "catastrophic event" it is meant an accidental impact or similar unpredictable event. Conversely, the Onsager and Tribus relations appear sufficient to explain any foresceable aging processes. Even so, a large amount of additional work appears to be needed and justified. Some suggestions for further work are as follows:

and the state of the state of the state of the state of the state of

ž

1971、大学学校の主要である。

ANTINE TO THE STATE

こうちょうがい いたいたいがい ちょうちょう

いろう ひろうち ひたいちかい ちょうかんちょう ちょうちょうちょう ちょうちょう ちょうしょう ちょうしょう

- Quantitative numbers for the Onsager and Tribus coefficients need to be experimentally determined. Special and careful experimentation will be required.
- The use cf three-dimensional stress, strain, strain-rate, and inertial mass concepts for surface tension and other surface phenomena need to be explored.
- A correction for gaseous viscosities needs to be studied further as a more correct explanation for deviations from the ideal gas law than is the present use of "Van der Waal forces". The fact that gas-viscosities do not approach zero values at zero pressure needs to be explained.
- Appropriate amount factors and intensity factors for thermal energy need to be established. The concepts of temperature

	(1) Ene:gy Form(a)	(2) Amount factor(h)	(3) Intensity Factor ^(C)	(4) Spring Analog of Strain(d)	(5) Spring Analog of Stress(d)	(6) Dashpot Analog of Strain Race(e)	(7) Dashpot Analog of Stress(e)	(8) Inertial Mass Analog(f)
Row 1	Mecha lical	Volume, V	Pressure, P	dv v	dP	dV a·dt	dP dx	$\Delta P = k_1 \cdot \frac{d^2 v}{dt^2}$
Row 2	Electrical	Charge, Q	Voltage, E	<u>व0</u> v	.4E	dQ n·dt	dE dx	$\Delta E = k_2 \cdot \frac{d^2 Q}{dt^2}$
Row 3	Thermal	Heat, q	Temperature, T	<u>v</u>	dT	dq a•dt	dT dx	$\Delta T = k_3 \cdot \frac{d^2 q}{dt^2}$
		(Momentum, mv?)	(Velocity, v?)	dmv v?		<u>dav</u> a•dt?	dv dx?	$(\Delta v = \frac{kd^2(mv)}{dt^2?})$
Row 4	Chemical	Moles, M	Concentration, [C]	dm V	d[c]	<u>dm</u> a•dt	d[c]	$\Delta[c] = k \cdot \frac{d^2 M}{dt^2}$

TABLE XIII. TH.RD VERSION OF GENERALIZED STRESS-STRAIN AND STRESS-STRAIN RATE RELATIONSHIP

THE THE PARTY STREET

NAME OF BRIDE

·_•

1.24516 84.4

58.56

A CONTRACTOR

4194381542

35431 T. 123.C.Z.745

2010 A 11 8 16 16

Energy forms associated with performance and aging of batteries. Generalized displacements. Generalized forces. (a)

(b) (c)

WITHIN THE PARTY OF THE PARTY OF THE

İ

(d) Tribus Kelations: Strain₁ = n~Row 1 n=Row 4

(f) Generalized Inertial Forces - Generalized Accelerations.

Preceding page blank

A SHELFT

48

Service Barris

and enclopy are well established, but early controversies have never been settled. The use of "momentum" and "velocity" as proper amount and intensity factors, respectively, has been suggested.⁽⁷⁾ A concerted effort by several persons appears to be desirable for making progress in a technical field of such magnitude as thermal energy.

• Versions of stress-strain relationships should be sought that have more universal applicability then those in Table XIII.

In the mean time the various terms in Row 1 of Table XIII are related to pressure and volume relationships for gases or to pressure-volume relationships for osmotic affects through separators or through pressure-volume relationships during the expansion and contraction of active materials. In Row 2, the spring analog is related with depth of discharge whereas the dashpot analog is related to rates of discharge as measured in current densities. In Row 3, the spring analog refers to heat-capacity relationships where as the dashpot analog is related to heat-conduction phenomena. In Row 4, the chemical analogs have to do with Fick's difusion laws. The ratio of spring and dashpot constants leads to so called time constants which characterize the rate at which steady-state conditions tend to be reached.

Task III (d). Thermophysical Properties

During this contractural period three special reports and two journal publications have described work done pertaining to temperature effects in nickel-cadmium batteries. (9,13,14,15,16)

"An Annotated Bibliography of the Thermal Properties of Primary and Secondary Cells"⁽¹³⁾ gives complete details and condensed tabulations of published knowledge in six sections:

> Section 1. Thermal conductivity, resistance data, and resistancecapacitance networks

Section 2. Specific heat and heat capacity data

Section 3. Enthalpy, entropy, and temperature coefficient of EMF data

Section 4. Heat-generation prediction and measurement

Section 5. Thermal design of cells and batteries

Section 6. Environmental temperature effects on cell or battery operation and life

All the information is indexed by Subject, Author, and Facility. The report, therefore, is a good resource for both designers and users of batteries.

Another special report on "The Thermal Conductivity of Sealed Nickel-Cadmium Cells" gives practical information for calculating thermal properties of cells and batteries.⁽¹⁴⁾ Conclusions of that work are summarized next, where "Types I, II, III, or IV" refer to commercial 20-ampere-hour cells from four different manufacturers.

In the x-direction (perpendicular to the plane of the electrodes) the measured conductivities of cell Types I, II, and III are in the range 0.0026 through 0.0038 cgs unit, and average 0.0030 cgs unit. The measured k_x value for cell Type IV in comparison is 0.0086 cgs unit. The predicted (calculated) average conductivity for all four types of cell is 0.0025 \pm 0.0004 cgs unit. The high k_x value cell Type IV remains unexplained.

In the y-direction (parallel with the plane of the electrodes) the cell conductivities are quite dependent on the presence of gas gaps and the amount of wetted separator extending beyond the edges of the electrodes. Even so, the calculated k values for the Type I, II, and III cells agreed with "easured values to within ± 8 percent. Values measured are in the range 0.0056 through 0.0071 cgs unit, while predicted values are in the range 0.0047 through 0.0067 cgs unit. In calculating and predicting cell thermal conductivities, the networks given in previous reports were combined with the conductivity data for individual constructional materials and with the conductivity for the "active" materials.

The thermal anisotropy of the cells, as given by the ratio k_y/k_x , varies according to cell type, and values in the range 1.76 through 2.63 are obtained. Values in the range 2.34 through 20 have been reported in the literature. It is postulated that the anisotropy of the cells is largely dependent on the presence of gas gaps at the cell edges and bottom. If large gaps exist, a low conductivity value is expected, and vice versa. Thermal-conductivity measurements can therefore identify the presence of gaps, which otherwise is difficult. Neutron radiography might be used to identify gaps, but such an application has not been found in the literature.

The conductivity of a nickel-cadmium cell appears to increase with increasing state of charge, up to full charge, in approximately a linear manner. The measured differences between discharged and 105 percent recharged cells are about 12-14 percent, and the experimental error is estimated to be about ± 5 percent. Originally it was hypothesized that the increase in conductivity of charged cells might be due to the higher proportion of metallic

cadmium (of high relative conductivity) in the charged negative electrodes. However, measurements on individual charged and discharged electrodes do not substantiace this hypothesis. Instead, the changes in conductivity of the individual electrodes are attributed to a redistribution of electrolyte.

Because most heat is usually generated during discharge, and because it is easier to measure discharged cell conductivities, the conductivity values given for discharged cells or electrodes are recommended for thermal analyses.

The thermal resistance of the heat-transfer path in the x-direction was calculated to be the lowest of any of the paths considered. A low value for R_x ensures that most of the heat generated can be dissipated by heat inks on the yz faces (i.e., on the sides parallel with the electrodes). Therefore, these yz faces should be relatively large compared with the xz faces (cell edges). However, the present research program has shown that the measured conductivity in the y-direction is greater than in the x-direction, for the cells studied. Therefore, heat extraction in the y-direction, through the xz faces, could be more efficient than in the x-direction, through the yz faces, depending upon the relative magnitudes of the xz and yz face areas, and the measured k and k conductivity values.

Experiments with six-cell battery modules show that the packing arrangement of the cells in the modules has a noticeable effect on the temperature distribution throughout the modules during cycling because of the thermal anisotropy of the cells. Thermal-resistance networks set up for the battery modules, which include the battery hardware, enable the temperature distributions throughout the modules to be estimated and the approximate overall module temperature changes to be calculated. The most efficient stacking arrangement is found to be primarily a function of the thermal anisotropy of the individual cells. However, the type of external battery or cell surfaces available for heat extraction is another important factor to be considered.

In constructing batteries, the individual cells should be thermally matched as well as electrically matched. The two thermophysical parameters which should be matched are the overall thermal resistance, R_T , and capacitance, C_T . If, in a battery, these are matched as closely as possible, then the thermal time constant for each cell (the product of R_T and C_T) will be similar also. Simple experimental procedures for determining R_T and C_T have been outlined in the special report. However, if it is not practical to perform these tests, the cells could be matched by measuring the temperature rise on discharge under some arbitrarily chosen conditions and cells showing similar temperature changes should be used in the battery if also electrically

matched. The result of thermally matching cells will be a more-balanced battery operation. All cells should be subjected to similar temperature excursions so that all cells in a battery will have a better chance of coming to a state of full charge or discharge at the same time, thus avoiding the development of electrical mismatch.

A third special report relating to thermophysical properties was the earlier mentioned report on "Temperature and Temperature Gradients for Accelerated Life Tests of Nickel-Cadmium Cells". ⁽⁹⁾ Similar use of temperature information was made in the special report "Rate of Discharge for Accelerated Life Tests of Nickel-Cadmium Cells". ⁽¹⁰⁾ These reports are mentioned here because the experiments described therein depended greatly on these supporting studies on thermophysical properties. Thus, the experimental cells in unmountable cases all had specially made heaters in them. Half of the cells had the heaters off and half had the heaters on, at preselected settings so that their internal temperatures were at a constant + 40 C regardless of environmental temperatures from + 40 to -20 C. The successful execution of these plans and the observed agreement between actual and predicted temperature gradients gave evidence and confidence to the "RC-Network" approach to thermal analysis for sealed cells which has been developed in this Subtask III (d).

Task III (e). Literature Review

Published information relating to nickel-cadmium and other aspects of battery technology was collected, organized, and indexed through the Battery Information Index (BII). This information effort was originally initated under Contract AF 33 (615)-3701 and continued under the present Contract F33615-69-C-1537. Information activities during the present program resulted in the reduction of practice of those recommendations suggested in the earlier research contract. ⁽²⁶⁾

Development of the BII is summarized in the following sections: (1) Growth and Status, (2) Subject Endexing and Purging Procedures, and (3) Retrieval Techniques.

Growth and Status

The Battery Information Index contains approximately 2200 documents relating to aerospace battery technology. Input is restricted to professionally selected documents containing technical information deemed useful or relevant to

TABLE XIV .SOURCE CLASSIFICATION OF BATTERY
INFORMATION INDEX (INCLUDES
BATTERY DOCUMENTS B-1 THROUGH
B-2189

U.S. Journal Articles	248
Foreign Journal Articles	140
U.S. Patents	280
Foreign Patents	84
Books	29
Government Reports	1,051
Industrial Literature	15
Symposium Fapers	338
Bibliographies	4
Total	2,189

51 M.

100.00

555278 42771

53

GUNG SSLADO

the contractural work topics. Table XIV shows a breakdown of source classifications that have yielded useful input for the BII. Approximately half of the documents in the system are government reports. Journal articles, bibliographies, symposium papers, and books of the type expected in libraries contribute an additional 35 percent. The other holdings are industrial literature and patents.

Growth of the BII is shown from statistics on index pagination as presented in Table XV. These figures refer to a hard copy version of all indexes containing approximately 50 single-spaced lines per page. The subject, author, racility, and contract numbers indexes are all given in a single-column format. The June, 1972, statistics indicate these four indexes required 163 pages and contained approximately 8'50 index entries. In addition, patent numbers are presented in a double-column format. Five pages with 2 columns per page and 50 lines per page gives about 500 index entries. Both AD- and N-Numbers are presented in a three-column format. Ten pages of 150 items per page gives 1500 index entries. The approximate total of individual entries in the June, 1972, index is 8150 + 500 + 1500 or 10,150 index entries. The June, 1972, index provided coverage for 2189 documents. Therefore, the number of indexing terms per document averages about 4.6. Table XVI shows a comparison of average numbers of terms per document in earlier indexes. The amount of indexing has not changed significantly. 大学校である。「大学校学校では、「大学校では大学校のないなどであった」」で、「「「大学校」」では、「「「大学校」」であった。「「「「大学校」」であった。「「大学校」であった。 大学校会社では、大学校学校では、「大学校では、大学校のないなどであった」では、「「「「大学校」」では、「「「大学校」」であった。「「「「大学校」」では、大学校」では、大学校、大学校では、サイト・マック

シュリード ちょうせいしゅう とうながなた ひがんし かたいない あたいてい せんかい ちょうちゅう

ukatik je je o jej justa i 15 manuar ukatori i 15 mala je je sa sa

いんごう こうかんかん あい さいかいない かかい 読み 読み ひかいかん かんがく シュー・・・・・・

Subject Indexing and Purging Procedures

All of the subject entries presently in the index have been technically approved by professional members actively working on Air Force battery projects. The subject indexing policy adopted for the BII is: (1) Any technical member of the project may enter any or all terms he considers to be useful retrieval tools, (2) Those terms, in essence, belong to him (for as long as he is associated with the project), (3) Subject entries may not be deleted or modified without the approval of the engineer who originally entered them into the system, and (4) The resulting subject index is a list of technically approved terms that tend to be "standardized" by engineer usage rather than by a formal thesaurus.

During 1971, a monitoring procedure was initiated to provide a continuous quality control for terms listed in the subject index. In an effort to retain consistency of term format; informal memoranda were sent to project

TABLE XV . GROWTH OF BII INDEXES

RTH REPUBLIC HISE

a fille and the

		Number	of Pages For Comp	olete Index	
Index Type	July, 1970, B-1 Through B-1580	March, 1971, B-1 Through B-1814	June, 1971, B-1 Through B-1925	September, 1971, B-1 Through B-1993	June, 1972, B-1 Through B-2189
Subjects	85	105	106	103	108
Authors	15	29(a)	29	31	35
Facilities	σ	10	10	11	12
Contract numbers	Ń	8 ^(a)	œ	33	8
Patent numbers	S	ß	4	4	5
AD-numbers	4	Ŋ	S	S	2
N- numbers	4	5	5	5	2
Totals	125	165	167	167	178

A

249 No WN 25

(a) Format change - from double column to single column per page.

11. 49.4 411.2

•

55

and the state of the second

TABLE XVI.	AVERAGE NUMBER	OF INDEXING	TERMS	PER DOCUMENT	
	IN BII SYSTEM				

Index Updates	Number of Pages	Number of Index Entries	Average Number of Terms/Document
July, 1970 (B-1 through B-1580)	125	7,450	4.7
March, 1971 (B-1 through B-1814)	165	9,400	5.1
June, 1971 (B-1 through B-1925)	167	9,550	4.9
September, 1971 (B-1 through B-1993)	167	9,550	4.7
June, 1972 (B-1 through B-2189)	178	10,150	4.5

a Data Projection

members from the information specialist to seek approvals for deletion or for modifications of terms which were criginally entered into the subject index. The distribution of approximately 175 memoranda resulted in over 140 approvals for term revisions and only 6 disapprovals. This method of initiating and approving indexes appears to be a practical solution to information obsolescence and redundancy. The effectiveness of this procedure is reflected in the September, 1971, index update (Table XV). Notice that the subject index contained three pages (approximately 150 terms) less than the previous revision even though index coverage was provided for an additional 179 battery documents. The reduction of index pages was accomplished by the delection of approximately 200 terms (feedback from informal memoranda) and the addition of approximately 200 new index terms from recent inputs. Table XVII shows changes in numbers of index entries resulting from usage of purging procedures.

Consideration was given to future subject indexing procedures that would ensure high quality retrieval from larger reservoirs of information containing numerous technical areas of battery technology. The following procedures were discussed and demonstrated to echnical members on the project: (1) categorical indexing, (2) permuted indexing, and (3) reduced permutation indexing. It was concluded that technical persons consider some of these procedures to be outside their area of interest. Technical persons prefer to have their own indexing formats with guidance and help from an information specialist.

Retrieval Techniques

Retrieval of information from the BII is accomplished in the following manner:

> Step 1. Identification of battery document by B ber from one of seven indexes provided on -_crofiche The seven separate indexes include: (a) Subjects, (b) Authors, (c) Facilities, (d) Government Contract Numbers, (3) Patent Numbers (both United States and foreign, (f) AD-Numbers to Government reports, and (g) N-Numbers to Government reports. All indexes are updated quarterly.

	June, <u>Complet</u> New	1971, <u>e Index</u>	Septembe <u>Complet</u> New	er, 1971, ce Index	June, <u>Complet</u> New	1972, te Index
Index Type	Entries ^{(a})Deletions	Entries(a)Deletions	Entries (a	Deletions
Authors	100 [.]	O	5	0	123	0
Facilities	19	0	5	0	25	0
Subjects	257	32	200	188	284	142
N-numbers	19	0	16	0	16	0
AD-numbers	10	0	13	0	12	0
Patent numbers	33	0	36	0	33	0
Contract numbers	10	0	4	0	13	0

TABLE XVII. INITIAL RESULTS OF INDEX PURGING PROCEDURES

Source and the second second

いいたいがいたかがないないがいたいがいたいないであったいでいました。 たんかい いちょうてい いちょうのう

ACCESSION OF THE OWNERS

Spirit Marine State

(a) New entries are completely new indexing terms and are not to be considered as additional B-numbers added to proviously existing terms.

- Step 2. Varification of selected battery document from total Accessions List on microfiche or from abstract card files.
- Step 3. Retrieval of document by B-Number from microfiche report file or from hard-copy document file.

Users dev lop a skill in retrieving information from the BII as they gain familiarity and experience with the system. A given battery document may be retrieved through any of the above seven indexes depending upon the starting information. While it is true that any of several indexes will deliver the needed B-Number, it is also true that usage of combined indexes will provide B-Numbers faster depending upon the amount of available starting information. Retrieval of battery information by usage of combined indexes is described in detail on pages 73-76 of Technical Report AFAPL-TR-70-44 Part II.⁽¹⁸⁾

م یا کارکا اللمام کی برز را ا

REFERENCES

- McCallum, J., Thomas, R.E., and Roeger, E.A., Jr., Battelle's Columbus Laboratories, "Failure Mechanisms and Accelerated Life Tests For Batteries", Technical Report AFAPL-TR-68-83 (July 1968), Wright-Patterson Air Force Base Contract AF 33(615)-3701.
- (2) McCallum, J., and Faust, C.L., Battelle, Columbus Laboratories, "Failure Mechanisms in Sealed Batteries", Technical Report AFAPL-TR-67-48, Part VI (June 1969), Wright-Patterson Air Force Base Contract AF 33(615)-3701.
- (3) Thomas, R.E., Brooman, E.W., Waite, J.K., Linebrink, O.L., and McCallum, J., Battelle, Columbus Laboratories, "Study of Space Battery Accelerated Testing Techniques, Phase II Report: Ideal Approaches Tow is Accelerated Tests and Analysis of Data", NASA CR-107114 (August 1969), NASA Coddard Space Flight Center Contract NAS 5-11594.
- Reed, A.H., Cover, P.W., and McCallum, J., Battelle, Columbus Laboratories, "Failure Mechanisms and Analyses of Sealed Batteries", Technical Report AFAPL-TR-69-74 (September 1969). Wright-Patterson Air Force Base Contract AF 33(615)-3701.

 Barney, D.L., Catotti, A.J., and Fensabene, S.F., General Electric Company, "Effects of Carbonate on Charge and Discharge Characteristics of Sealed Nickel-Cadmium Cells", Preprint Paper 8 presented at the Seventh International Power Sources Symposium, Brighton, Sussex, England (September 15-17, 1970).

- (6) Bland, D.R., The Theory of Linear Viscoelasticity, Pergamon Press, New York (1950).
- (7) McCallum, J., Thomas, R.E., and Roeger, E.A., Jr., Battelle's Columbus Laboratories, "Development of Stress-Strain Models for Energy-Storing Systems", Technical Report AFAPL-TR-72-68 (August 1972), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (8) Onsager, L., "Reciprocal Relations in Irreversible Processes", Physical Review, <u>37</u>, 405 (1931).
- McCallum, J., Brooman, E.N., Roeger, E.A., Jr., Battelle's Columbus Laboratories and Miller, G.H., Air Force Aero Propulsion Laboratory, "Temperature and Temperature Gradients For Accelerated Life Tests of Nickel-Cadmium Cells", Technical Report AFA⁻¹, TR-72-66 (August 1972), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (10) McCallum, J., and Roeger, E.A., Jr., Battelle's Columbus Laboratories and Miller, G.H., A.* Force Aero Propulsion Laboratory, "Rate of Discharge For Accelerated Life Tests of Nickel-Cadmium Cells", Technical Report AFAPL-TR-72-67 (August 1972), Wright-Patterson Air Force Base Contract r33615-69-C-1537.

- (11) Reed, A H., and McCallum, J., Battelle's Columbus Laboratories, "Depths of Discharge For Accelerated Life Tests of Cadmium Electrodes", Technical Report AFAPL-TH-72-20 (May 1972), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (12) Francis, H.T., Armour Research Foundation, <u>Space Battery Handbook</u>, NASA Contract NASw-401 (April 15, 1963).
- (13) Brooman, E.W., Battelle, Columbus Laboratories, "An Annotated Bibliography of the Thermal Properties of Primary and Secondary Cells", Technical Report AFAPL-TR-70-34 (July 1970), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (14) Brooman, E.W., and McCallum, J., Battelle's Columbus Laboratories, "The Thermal Conductivity of Sealed Nickel-Cadmium Cells", Technical Report AFAPL-TR-71-75 (February 1972), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- Brooman, E.W., and McCallum, J., Battelle's Columbus Laboratories,
 "Thermal Conductivity Measurements of Nickel-Cadmium Aerospace Cells.
 Part 1: Cell Conductivities", Journal of the Electrochemical Society, <u>118</u> (9), 1518-1523 (September 1971).
- Brooman, E.W., and McCallum, J., Battelle's Columbus Laboratories,
 "Thermal Conductivity Measurements of Nickel-Cadmium Aerospace Cells.
 Part II: Component Conductivities", Journal of the Electrochemical Society (accepted for publication during September 1972).

- (17) Tribus, M., <u>Rational Description</u>, <u>Decisions</u>, and <u>Designs</u>, Pergamon Press, New York, New York (1960).
- (1d) McCallum, J., Battelle's Columbus Laboratories, and Miller, G.H., Air Force Aero Propulsion Laboratory, "Fallure Mechanisms and Accelerated Life Tests of Nickel-Cadmium Batteries", Technical Report AFAPL-TR-70-44, Part II (September 1971), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (19) Fleischer, A., Nickel Cadmium Battery Corporation, "Sincered Plates For Nickel-Cadmium Batteries", Transactions of the Electrochemical Society, <u>ab</u> (6), 289-299 + 400-401 (December 1948).
- (20) Calrns, R.W., and Ott, E., Jour. Am. Chem. Soc., <u>55</u>, 527-533 (1933).
- Batteries 2, Edited by D.H. Collins, Proceedings of the 4th International Symposium, Brighton, Sussex, England (September 1964), Pergamon Press, New York (1965), "Reactions in Sealed Nickel-Cadmium Cells" (Lunn, H.B., and Parker, J.), pp 129-145.

61

With States

- (22) Fleischer, A., private communication, July, 1970.
- (23) McCallum, J., Battelle's Columbus Laboratories and Miller, G.H., Air Force Aerc Propulsion Laboratory, "Failure Machanisms and Accelerated Life Tests of Nickel-Cadmium Batteries", Technical Report AFAPI-TR-70-44 (July 1970), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (24) Roeger, E.A., Jr., and McCallum, J., Battelle's Columbus Laboratories and Miller, G.H., Air Force Aero Propulsion Laboratory, "Simulated Orbital Life Tests For Space: aft Cells. Part 1: Manually Acquired Data", Technical Report AFAPL-TR-71-74, Part I (January 1972), Wright-Patterson Air Force Byse Contract F33615-69-C-1537.
- (25) Roeger, E.A., Jr., and McCallum, J., Battelle's Columbus Laboratories; Waite, J.H., Cryptanalytical Computer Sciences, Inc.; and Miller, G.H., Air Force Aero Propulsion Laboratory, "Simulated Orbital Life Tests For Spacecraft Cells. Part II: Automatically Acquired Data and Recommendations", Technical Report AFAPL-TR-71-74, Part II (August 1972), Wright-Patterson Air Force Base Contract F33615-69-C-1537.
- (26) Reference (17), p 20.

(27) Thomas, R.E., Brooman, E.W., Linebrink, O.L., Waite, J.H., and McCallum, J., "Study of Space Battery Accelerated Testing Techniques, Phase III Report: Recommended Specifications for Experimental Design and Test Facilities". not published, (1970) NASA Goddard Space Flight Center, Contract No. N.S 5-11594.

APPENDIX A

Prozentise man

RECOMMENDED CELL DESIGN FOR MAXIMUM PERFORMANCE AND LIFE RECOMMENDED ACCELERATED TEST PROCEDURE

This appendix is written in fulfillment of that part of paragraph 2.6 of the statement of work which states "submit as appendixes to the Final Report, a 'Ceil Design for Maximum Performance and Life' and an 'Accelerated Test Procedure'". Even though the experimental verification tests have not been completed, the recommended cell design and test procedures are based on information received to date from this program.

The recommended design of cel's is described in this report under the section entitled "Task II. Increased Battery Life". As explained in that Task II section, the fabrication of the cells is described in detail in the Second Annual Report ⁽¹⁸⁾ pages 32 through 35. Although the specifications given are for 6-ampere-hour hermetically sealed cells, the actual dimensions will depend upon the size of cell required to fulfill the power requirements given for the typical near-earth orbital profile previously described under Evaluations in Task II.

Two changes are recommended in the previously mentioned design of hermetically sealed cells. (1) The width of the nickel combs shown in Figure 2 of this report should be increased from 1/2 inch to approximately 0.65 inch to make them stronger and less subject to deformation during assembly of the electrode tabs into the slots of the comb. (2) The height of the tabs on the outermost electrodes should be increased sufficiently so that the tabs of these electrodes can fit well into the slots in the combs. This height adjustment means that the tabs for the outermost electrodes must be longer than the tabs on the inner electrodes before assembly so that after assembly the bottoms of all electrodes will be properly alined.

After the cells are assemblied, they should be filled with a volume of 30 percent KOH (density of 1.285 g/cm^3) equivalent to 18.5 percent of their element weight. The element weight is the total dry weight of the separator and the positive and negative electrodes. The cells should stand for at least 24 hours after adding the electrolyte before proceeding with the electrochemical conditioning.

After the 24-hour stand, the cells should be subjected to the following steps to condition and set the state of charge of the cells:

一一 当时 新校市 小子子

- Step 1. Charge for 6 hours at C/3 rate or until a voltage limit of 1.5 is reached.
- Step 2. Discharge at C/3 rate to zero volts.

- Step 3. Charge at C/3 rate for exactly 5.0 hours and seal. Measure cell voltage within 2 minutes of end-ofcharge.
- Step 4. Determine capacity by the discharge time to 1.0 volt at the C/3 rate.
- Step 5. Charge at C/3 rate for 3 hours and measure internal resistance.

The cells should then be matched into groups primarily according to the capacity determined in Step 4 above. Secondary consideration should be given to internal resistance (Step 5) and the end-of-charge voltage (Step 3).

ACCELERATED TEST PROCEDURE

The normal stress level or conditions for an accelerated life test are represented by Group 1 in Table XVIII. The remaining groups in Table XVIII represent the recommended accelerated test for a typical near-earth orbital profile.

Each group should consist of eight cells, three of which should be removed during the course of the tests. One of the three cells should be removed after its quality had degraded by one-fourth of its original-minus failure-quality. A second cell should be removed after its quality degraded by one-half and the third cell removed after its quality had degraded by threefourths. The cells to be removed from each group should be selected randomly prior to starting the tests.

Each group should be tested in a 110-minute orbit consisting of 75 minutes of charge and 35 minutes allotted for discharge. Those groups not requiring 35 minutes of discharge should be placed on open circuit to complete the allotted 35 minutes of discharge time. All groups should be tested at a 48.7 percent depth of discharge. The charge should consist of a C/2 charge
TABLE XVIII. RECOMMENDED ACCELERATED TEST PROCEDURE BASED ON 6-AMPERE-HOUR SEALED NICKEL-CADMIUM CELLS

1997 - 11 - 1

Group Number	Chamber Temperature, C	Induced Temperature Gradient, C	<u>Charg</u> Step 1	e Rate Step 2	Discharge Current, amps	Discharge Time, min
1	20	none	C/2	C/10	5	35.0
2	20	Ditto	Ditto	Ditto	10	17.5
3	20	**	11	tr	20	8.5
4	20	11	**	"	40	4.4
5	20	**	81	11	80	2.2
6	20	40	C/2	c/10	5	35.0
7	20	40	Ditto	Ditto	10	17.5
8	20	40	11	11	20	8.8
9	20	40	"	11	40	4.4
10	20	40	11	t1	80	2.2

(a) Depth of discharge should be 48.7 percent for all groups.

65

a margane average

rate to a selected group voltage limit and then a C/10 charge rate for the remainder of the 75 minute charge period. The group voltage limit should be selected on the first cycle by charging the cells at the C/2 rate for 1 hour. The group voltage after this 1-hour charge should then be used during subsequent cycles as the switchover point from the C/2 to the C/10 rate of charge.

All groups should be tested in an environmental chamber with its internal temperature set at 20 degrees centigrade. Each cell in Groups 6 through 10 should then have an induced temperature gradient supplied by an external heater. This heater should be set to provide a temperature of 40 degrees centigrade on one of the cell case slides parallel to the face of the cell's electrode pack. A heater of the type previously described (23) would seem to be suitable for this application.

To provide data required for analysis of this experiment, the current, group voltage, individual cell temperatures, and individual cell voltages should be recorded. Also, the charge time during C/2 rate should be measured and recorded. The cell temperatures should be measured by a thermocouple placed on the cell case side opposite the external heater. The cell voltages should be recorded within the last 20 seconds of the end of discharge. In fact, this time should be as close to the end of discharge as possible.

The C/2 charge time should be used to calculate a "group charging voltage quality" by the equation:

$$Q_{cv} = \frac{amps \times time}{AH}$$
, (A-1)

where

amps = C/2 constant charging current in amps, time = time in hours required to reach the C/2 to C/10 switchover point,and

AH = ampere-hours removed during cycling tests.

This calculation can be used as an indication of changes in the charging quality of the group of cells and of charging efficiencies.

Primary life predictions from the tests should be based on the "electrical discharge quality" of the cells. This "quality" should be calculated from the equation:

$$Q_{\rm E} = \frac{E_{\rm eod} + IR}{ngIR} , \qquad (A-2)$$

こうちょう うちょうしょう 日本部長を実施したいないない いっしょうしょうしょう ひょうちょう かんしょう ちょうしょう

66

where

いたいになったいため、ためになるというとき

 E_{eod} = the end-of-discharge voltage,

I = the disc: rge current in amps,

- R = the initial internal resistance in ohms,
- n = the number of hours required to fully discharge
 the cell at current 1, and

g = the specific capacity of the cell in grams
 per amp-hr.

This definition of quality has units of amps/gram. The manufacturer's rated capacity of the cells under test should be used to determine the values of I, g, and the depth of discharge.

The individual cell <u>failure</u> qualities should also be calculated by Equation (A-2), by using 1.00 volt, compensated for IR, in place of the end-of-discharge voltage (E_{eod}).

During the testing of the cell groups, predictions should be made concerning their expected life. One method for making such predictions concerns parametric failures and is described elsewhere. (27) These predictions should be made by using the cells' initial quality and failure quality calculated with Equation (A-2).

All parametric failures, actual cell failures and the cells remaining at the conclusion of the test should be subjected to the failure analysis procedure listed in Table VIII of this report. The purpose of the failure analysis procedure is to determine whether the failure mechanisms are the same at all stress levels. One requirement for a valid accelerated life test is that the failure mechanism is the same at all stress levels.

Also after the tests have been completed, acceleration factors should be determined for each group based on the normal stress of Group 1, Table XVIII. The presently recommended method for determining the acceleration factors is described in this report in the section entitled "Verification Tests, Test Results". The student T test is recommended to calculate confidence levels for the accelerated test procedure recommended in Table XVIII.

The experiment just described is the recommended test procedure for a typical near-earth orbital profile. If for some reason the cells, equipment, or

monies were not available to run this experiment, a minimal experiment would be to test Groups 1, 6, 7, 8, 9, and 10 with five cells instead of eight. This minimal experiment would provide only the minimum amount of statistical information involving two extrapolations and would <u>not</u> provide the electrochemist with the type of information he would like from such an experiment as was recommended. This minimal experiment would provide testing at the most stressful conditions (Group 10) and a means to relate this information back to the normal or operational test conditions (Group 1).

- --- Carbon Strict

With the above information and the cited references, sufficient detail is believed to be available to permit a skilled technician to carry out the recommendations.

1000