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Volume II

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Optimization Study Report on Base Level Analyzer

Contract No. AF 33(657) 13866

Prepared for

Air Force Systems Command U.S.Air Force Wright Patterson Air Force Base, Ohio

o January 1965



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Volume II

**Optimization Study Report** 

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Prepared for

Air Force Systems Command U.S.Air Force Wright Patterson Air Force Base, Ohio

6 January 1965

Prepared by: / Ela Yerd-

Peter N. Dudency Program Manager

Approved by

William G. Langton

Vice President, Science & Engineering

BAIRD-ATOMIC, INC. . 33 University Road Cambridge, Massachusetts 02138



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### 7. SYSTEM ERROR ANALYSIS

# 7.1 Introduction

In the following paragraphs, the operation of the Base Level Analyzer will be examined and all sources of error will be identified and, if possible, described quantitatively. A final conclusion regarding the errors to be expected will be made, summarizing all calculations and error estimates in a conservative manner.

The error sources in the Base Level Analyzer operation fall into four main categories and are:

- a. Oil Sample and/or Standard Errors
- b. Source Variations
- c. Optics Errors
- d. Electronic Signal Processing or Readout Errors

Of these, the last two categories will be emphasized since they represent instrumental errors which are within the scope of the existing contract and since the greatest amount of quantitative data is available in these areas. Source variations would be covered in more detail except for a lack of firm quantitative data. The first category, that of oil sample and standard errors, is not strictly within the scope of the contract and is an extensive problem area in itself. It will be discussed briefly, but no quantitative error estimate will be made.

The Base Level Analyzer is an instrument which does not contain an internal reference standard but compares oil samples of unknown impurity level with supposedly known standards. As a result it is important to consider in this error analysis only those sources of error which affect the calibration repeatability or reproducibility of the instrument over a relatively short time span, say one day. Systematic errors which are either constant or very small are not of major concern. However, such errors will be identified where they exist and described as well as is possible.

7-2

Since many of the errors to be discussed in the optical category are wavelength dependent, being greatest for the spectral extremes, frequent mention in examples will be made of iron. Iron has been selected first because it is of prime interest to the sponsor; second, because the iron line being used falls near the low extreme of the spectrum considered; and third, because the working curve for iron is nearly linear.

### 7.2 Oil Sample and Standard Errors

Quantitative analysis of impurities in lubricating oil by spectrographic methods is based upon a comparison of the oil sample containing the unknown impurities with a set of known standards containing the impurities of interest at various levels of concentration. In order to obtain an accurate and meaningful result from the comparison, the sample must be representative of the whole body of oil from which it was extracted and the set of standards must be reliable.

Obtaining a representative sample is a matter of common sense and good chemical technique. Samples must be obtained in a uniform and careful manner considering oil temperature, time since last agitation, sampling level within the reservoir in the vehicle, and cleanliness of the glassware used. Also, the handling of the samples subsequent to extracting the oil from the vehicle must be such that no chemical reaction or sedimentation occur which might remove impurities from the sampled oil. Finally care must be taken to avoid contamination of the sample or source electrodes when setting up the source prior to burning the sample.

Sets of oil standards are prepared by adding known amounts of organic salts containing the metallic constituents plus solubilizing agents to a base oil assumed to be free of impurities. Generally the highest impurity levels are obtained directly in this way and the initial solution is diluted with



additional oil to obtain standards at the lower impurity levels. The final standards can be in error as a result of inaccurate weighing and mixing of the organic salts, errors in diluting the initial high impurity level standard, failure of all the organic salts to dissolve and remain in solution, chemical reactions causing one or more metallic constituents to leave the solution, dirty glassware, and impurities contained initially in the base oil.

Of these sources of error, the most important and generally the hardest to detect are the failure of the organic salts to dissolve and remain in solution and chemical reactions causing impurities to leave solution. Weighing and diluting measurement errors can be kept small, of the order of two percent. Dirty glassware becomes more important at the lowest impurity levels but should not present a problem if good technique is employed.

Impurities in the base oil cannot be completely eliminated. Generally the assumption is made that an oil is impurity free as it comes from the refinery container despite the fact that it may contain minor amounts of impurities (of the order of 1 or 2 parts per million (ppm)). Larger amounts of impurity additives can be detected in the base oil by comparing supposedly pure oils of different types. Small amounts of impurities (1 or 2 ppm) cause an effective offset in the absolute calibration of the Base Level Analyzer but do not impede its successful use since it is always used as a comparative rather than an absolute measuring tool.

Since the subject of oil sample and oil standard errors is a major problem area outside the scope of the existing contract, no quantitative estimate of overall error has been attempted here.

7.3 Source Variations

Changes in source conditions may excite some unknown lines differentially with respect to reference background. No exact data is available as to the

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7-4

importance of this effect or the ultimate source stability requirements. However, data have been obtained on the Baird-Atomic Research Direct-Reader Spectrograph (RDRS) which indicate that the integrated intensity for a burn of fixed time duration does not vary by more than  $\pm 10$  percent and that the total reproducibility error from all causes, optical and electronic, is of the order of  $\pm 10$  percent. A source reproducibility error much less than  $\pm 10$  percent is indicated by this data despite the lack of a voltage regulated supply for the RDRS source. Hence it will be assumed arbitrarily that reproducibility errors resulting from variations in source conditions are small and are of the order of  $\pm 1$  percent or less for the Base Level Analyzer, which does use a regulated source power supply.



### 7.4 Optics Errors

The optical errors which are to be discussed in this section arise from changes in ambient atmospheric pressure, temperature and humidity which affect the index of refraction of air, the diffraction grating line spacing and the dimensions of the mechanical supporting structure in the optical head. These environmental parameters produce shifts of the spectral lines laterally with respect to the exit slits (dispersion errors) and changes in optical focus (line broadening). Errors are produced when the amounts of integrated power reaching the reference and unknown photomultipliers respectively vary by different scale factors.

In the Base Level Analyzer design discussed in this report the reference photomultiplier uses spectral background as a source of energy and hence its output will not be affected by minor discrepancies in optical alignment or dispersion or by minor focus errors. However, the output of a photomultiplier collecting light from a spectral line of unknown intensity will be affected. Percentage changes in the excitation of such a photomultiplier will produce equal error percentages in the output reading of element concentration in parts per million provided the working curve for the element in question is linear.

In order to have a specific example for use in the discussion which follows, the element iron has been selected. Iron is the element of greatest interest in this program and it has a linear working curve. Also, since the spectral line used for iron (2599 A) is at one extreme of the focal curve, iron will experience nearly all the peak errors discussed below which are a direct function of wavelength separation from the optical alignment servo wavelength.

### 7.4.1 Dispersion Errors

In an earlier section the dispersion errors in terms of lateral shifts for the extreme wavelengths were computed assuming extreme changes in temperature, pressure, and humidity plus an optical alignment servo for maintaining registration between a mean wavelength and the corresponding slit. These shifts for

temperature, pressure, and humidity are respectively 1.9 microns, 14 microns, and 1 micron at 2208 A and 2.2, 16 and 0.8 microns at 4254 A assuming the alignment servo operates at 3126 A.

An additional error of as much as  $\pm 5$  microns can be caused by dispersion in the calcium fluoride plate used in the optical alignment servo. A 5-micron shift is produced at the short wavelength end of the focal curve (2200 A) when the full 100-micron deflection of the calcium fluoride plate is being employed. Since the shift is a direct function of deflection, it normally will be much less than 5 microns, even at 2200 A. Also the maximum shift at 4500 A is only 1.2 microns at full deflection while no shift is experienced at 3147 A.

Dispersion errors can result in important output errors. The amount of each in terms of percentage error at the photomultiplier output can be determined by considering the widths of exit and entrance slits and the spectral line shape:

The Base Level Analyzer entrance slit is 50 microns in width and is imaged by a 1-meter grating onto 25-micron wide exit slits. The intensity profile of the spectrum line has an approximately Gaussian shape as shown in figure 7-1. Translation of the line relative to the slit causes a decrease in the integrated intensity passing through the slit. The power on the photomultiplier is P where

$$P = \int_{X_1}^{X_2} \phi(x) dx$$

 $\frac{1}{\sqrt{2}} \exp \left[-\frac{1}{2} \frac{x^2}{\sigma^2}\right]$ 

In this equation  $X_1$  and  $X_2$  represent the boundaries of the exit slit and







Using the cumulative distribution function

$$\mathbf{P} = \mathbf{\Phi}(\mathbf{X}_2) - \mathbf{\Phi}(\mathbf{X}_1) \quad .$$

where

$$\overline{\Phi}(\mathbf{x}) = \int_{-\infty}^{\mathbf{X}} \Phi(\mathbf{x}) d\mathbf{x}$$

The values  $X_2$  and  $X_1$  always differ by the width of the exit slit  $\sigma$ , or 25 microns. The graph in figure 7-2 shows the effect on the photomultiplier tube output for shifts of various amounts.

It can be seen from figure 7-2 that the maximum dispersion shift of 2 microns caused by temperature, changes in index of refraction of air and grating line spacing produces less than a 0.5 percent change in photomultiplier output. Humidity variations produce a still smaller percentage change, about 0.2 percent. Since the reference channel will not be affected by these environmental effects, a maximum change in calibration in the iron channel of 0.5 or 0.2 percent will be observed for the complete temperature or humidity excursion.

Pressure changes cause a much greater shift, about 15 microns, in the position of the spectral line relative to the exit slit. Figure 7-2 shows that this shift is equivalent to a 15 percent change in photomultiplier output (or error) in the case of iron. An error of this size is intolerable. Consequently, a mechanism has been provided whereby the grating is moved manually toward the center of the focal curve as pressure is reduced so that compensation for the line shift is achieved. A total movement of 0.003 inch is needed to compensate from 0 to 10,000 feet altitude. Provided the grating is exactly adjusted for ampient pressure no error will exist. In practice, however, only an average pressure setting for the height above sea level of the instrument will be used. Weather changes can cuase rms atmospheric pressure variations



about the average pressure setting of the order of 15 millimeters of mercury and corresponding line shifts of 0.8 micron rms. Errors from this source in a reading of iron concentration can therefore be as much as 0.2 percent rms.

The calcium fluoride deflection plate used in the optical alignment servo can produce as much as a  $\pm$  5-micron alignment error, equivalent to a  $\pm$  1.6 percent output error. The amount of this error is proportional to the difference between the wavelength in question and 3126 A as well as to the amount of correction supplied by the servo. Pressure changes from 0 to 10,000-foot altitude produce a  $\pm$  50-micron correction requirement. Temperature changes of  $\pm$  30°C require  $\pm$  6.6-micron corrections while the total range of humidity change from minimum or normal conditions to high temperature wet conditions (23°C and 72 percent relative humidity to 55°C and 100 percent relative humidity) gives only a  $\pm$  3-micron correction. The  $\pm$  1.6-percent error stated above corresponds to the maximum correction that can be made by the servo which is  $\pm$  100 microns.

## 7.4.2 Line Broadening Errors

In the preceding section errors caused by shifts of a spectral line of constant width relative to an exit slit were discussed. This section is concerned with errors produced when the optics become defocused and the width of the spectral line is increased without any lateral shift. If the line is broadened relative to the exit slit wide, , a smaller fraction of the total light will pass through to the photomultiplier. The decrease in photomultiplier output can once again be estimated assuming a Gaussian line shape. If the exit slit is 25 microns wide and the line is 50 +  $\delta$  microns wide, then the slit width in terms of standard deviations (0) is

 $w = \frac{50}{50+5} \sigma$ 

7-10



The power on the photomultiplier is

$$P = \int_{-\frac{X}{0}}^{+\frac{X}{0}} \phi(x) dx$$
$$= 2 \int_{0}^{-\frac{X}{0}} \phi(x) dx$$

where

$$X = \frac{w}{2} = \frac{25}{50 + \delta} \sigma$$

Figure 7-3 shows the relative photomultiplier output plotted versus the broadening  $\delta$ . The relative photomultiplier output is

$$\frac{P_{(50 \pm \delta)\mu}}{P_{50\mu}} = \frac{0}{0} \int_{0}^{X} \frac{\phi(x) dx}{\phi(x) dx}$$

The focus errors which produce the broadening  $\delta$  result basically from changes in ambient temperature. These errors occur in addition to the dispersion errors produced by temperature effects on the air and grating which have already been analyzed in paragraph 4.3.5. No additional registration errors result from temperature changes provided the complete optical structure including focal plane is fabricated of the same material so that it has a uniform temperature coefficient of expansion and is maintained at a uniform temperature throughout.

In order to correct the focusing errors which can result from temperature changes, a servo has been provided to adjust the optical path between entrance slit and diffraction grating. This servo operates by inserting a variable thickness plate of calcium fluoride so that correct focus is maintained despite  $\pm 30$ °C temperature changes and resulting changes in the size of the mechanical structure.



These changes result from the fact that the optical head assembly will be a casting of A 356 aluminum. This material has a linear coefficient of expansion of  $11 \times 10^{-6}$  inches per inch/°C or  $\pm$  0.011 inch over 40 inches with a  $\pm$  25°C temperature change, results in an increase shift of  $\pm$  0.044 inch over a  $\pm$  25°C temperature range. This is due to the fact that the entrance slit moves 0.022 inch relative to the grating and the grating moves 0.022 inch relative.

No appreciable error would result from line broadening due to poor focusing if the servo were designed to operate in a proportional manner and if equal focus correction were provided for all wavelengths. However, neither of these conditions are met. The servo operates stepwise covering the total range of correction in 10 equal steps and the calcium fluoride has noticeable dispersion over the spectral band of interest.

First, consider the error caused by the stepwise servo adjustment. The total servo focus correction is  $\pm 0.044$  inch. Each servo step is therefore 0.0088 inch. Since it is equally probable that the error lies anywhere within = this range, the rms focus error is  $0.0088/2\sqrt{3}$  or 0.0025 inch (63.5 microns). In an f/20 optical system the line broadening to be expected is about 1/20 of this value or 3.2 microns. Reference to figure 7-3 shows that 3.2 micron broadening causes a 5.5 percent error in the photomultiplier output or in the concentration reading of an element such as iron.

These calculations are considerably pessimistic since actual measurement shows that an f/20 one meter diffraction grating system with the entrance slit displaced by 0.050 inch has but 13 microns of line broadening. Linear extrapolation from this value leads to an rms broadening of only 0.75 micron and a corresponding error of 1.1 percent.



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Second, consider the errors in focus produced by the dispersion of calcium fluoride focus corrector plate. A plate ranging in thickness from about 3.9 to 11 millimeters will be required. Such a plate has the ability to shift the focal plane by a distance x where

$$\mathbf{x} = \mathbf{d} \left( \mathbf{1} - \frac{\mathbf{I}}{\mathbf{n}} \right)$$

and

d = the plate thickness

n = the index of refraction

Assuming an 11 millimeter thickness and taking the indices of calcium fluoride at 2200 and 4300 A of 1.48112 and 1.43950 respectively, the focal plane shifts at these wavelength are 3363 and 3574 microns. The difference of 211 microns can be split so that the maximum error in focus at the extreme wavelengths is  $\pm 106$  microns. A linear extrapolation of the line broadening measurement previously mentioned leads to a maximum value of line broadening of 1.1 microns and a maximum error (figure 7-3) of 2.0 percent. Typically, of course, the error at the extreme wavelengths is less since the focus correction actually used in a normal temperature environment is about half the maximum.



### 7.5 Electronic Signal Processing Errors

The light emitted by the source during the burn period is dispersed by the diffraction grating and focused on the reference and unknown exit slits. The light passing through each of these slits is detected by a photomultiplier tube whose output current charges an integrating capacitor. Provision is made in the charging circuit for photomultiplier dark current cancellation. At the conclusion of the burn, namely when the reference capacitor voltage reaches a specified level, the capacitors are disconnected from the photomultipliers and their voltages are measured sequentially using a non-linear measurement circuit utilizing the working curves for the elements. The digitized outputs of the non-linear circuits are converted to 9 bit digital numbers and stored in registers whose contents can be displayed at will on a 3-decimal digit set of Nixie tubes.

The whole apparatus is illustrated in block diagram form in figure 7-4 in order to facilitate the following error analysis. In the analysis, each error producing block will be discussed in turn. Those subjects to be covered are:

a. Photomultipliers

b. Integrating Capacitors

c. Reference Capacitor Cutoff Voltage

d. Dark Current Cancellation

e. - Analog Circuits

1. Unity Gain Amplifier

2. Nonlinear Amplifiers

3. Analog-to-Digital Converters

f. Digital Circuits





# 7.5.1 Photomultiplier Errors

The photomultipliers used in the Base Level Analyzer are 1P28 9-stage units operated from a common 1000-volt regulated supply (an Arnoid Magnetic Corp. Model SMU). Three types of errors can be considered to arise from the combination of photomultiplier and supply. These result from

- a. Changes in Supply Voltage
- b. Changes in Individual Photomultiplier Tubes
- c. Photomultiplier Dark Current

The discussion of dark current problems will be deferred to paragraph 7.5.3

Small changes in the photomultiplier supply voltage tend to make large changes in photomultiplier current gain and anode sensitivity. The current gain G is

$$G = \delta^9$$

where

 $\delta$  = the secondary emission ratio of the dynodes (approximately 3). In the region of interest,  $\delta$  is approximately proportional to the voltage per stage (V).

 $G = KV^9$ 

 $K = (K')^9$ 

so that

if

If the voltage changes by a factor X then,

 $\frac{G_x}{G} = \frac{K x^9 v^9}{K v^9} = X^9$ 

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 $A \neq 1$  percent change in voltage, which can be expected from the photomultiplier voltage supply over the design temperature range, can therefore give about a  $\pm 10$  percent change in gain.

However, because of the use of a common supply voltage for all tubes, each should experience the same gain change, at least to a first approximation, so that the ratio of unknown capacitor voltage to reference capacitor voltage should remain constant. It is possible that not all dynodes in the various tubes will have exactly the same gain versus voltage characteristic and that a second order error will result. However, since no data is available upon which to base a quantitative error estimate, it will be assumed arbitrarily that such errors are negligibly small.

Differential aging of the photomultiplier tubes, if it occurred following calibration of the instrument could have a major effect on the useability of the Base Level Analyzer. Differential aging is important for some types of photomultipliers, especially during the first 100 hours of operation. However, it is believed that this effect has not been observed with 1P28 tubes. In any case a burn-in program can be conducted to achieve good photomultiplier stability. Consequently, no assignment of error will be made to account for aging phenomena.

# 7.5.2 Integrating Capacitor Errors

7.5.2.1 Charging Errors -- The voltage across a capacitor C being charged from a current source I is given by

$$V = R_{L} \frac{I}{L} \begin{bmatrix} 1 - e^{-\frac{t}{R_{L}}C} \end{bmatrix} \simeq \frac{It}{C} \begin{bmatrix} 1 - \frac{t}{2R_{L}C} \end{bmatrix} \text{ for } R_{L}C > t$$

R: = the leakage resistance

t = the time



(7-2

The capacitors being used have an intrinsic time constant of 25,000 seconds, except for the reference capacitor which is always shunted by a unity gain amplifier having less than 1.0-nanoampere leakage current, and the charge time is of the order of 10 seconds so that the approximation is valid.

The voltage on the ith capacitor after burning the oil sample is

$$V_{i} \simeq \frac{l_{i} t_{f}}{C_{i}} \left[ 1 - \frac{t_{f}}{2 R_{L_{i}} C_{i}} \right].$$
(7-1)

and the time at the end of burn,  $t_f$ , is found from the equation

 $\mathbf{V}_{\mathbf{R}} \simeq \frac{\left(\mathbf{I}_{\mathbf{R}} - \mathbf{I}_{\mathbf{L}}\right)\mathbf{t}_{\mathbf{f}}}{C_{\mathbf{R}}} \left[1 - \frac{\mathbf{t}_{\mathbf{f}}}{2\mathbf{R}_{\mathbf{L}}\mathbf{C}_{\mathbf{R}}}\right]$ 

where  $V_R$  is the reference threshold voltage and  $I_L$  is the leakage current of the unity gain amplifier. It should be pointed out that leakage phenomena included in these two equations do not of themselves cause output errors. If all leakage phenomena were constant, no output errors would result. It is only when either  $I_L$  or  $R_L$  varies that errors occur. Naturally a large variation in a quantity having only a vanishingly small effect on the cutpute in the first place cannot cause a serious error. Therefore it is desirable to keep leakage currents small as well as constant.

The rms output error, again using iron as our example, is identical to

 $\frac{1}{\Delta V_{i}^{2}}$  1/2

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where

$$\left[ \overline{\Delta V_{i}^{2}} \right]^{1/2} = \left\{ \left( \frac{\partial V_{i}}{\partial I_{R}} \right)^{2} \overline{\Delta r_{R}^{2}} + \left( \frac{\partial V_{i}}{\partial V_{R}} \right)^{2} \overline{\Delta V_{R}^{2}} + \left( \frac{\partial V_{i}}{\partial I_{L}} \right)^{2} \overline{\Delta I_{L}^{2}} \right. + \left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta I_{L}^{2}} \right\}^{1/2} + \left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta R_{L}^{2}} \right\}^{1/2} \right\}^{1/2}$$

$$\left. \left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta R_{L}^{2}} \right\}^{1/2} \right\}^{1/2}$$

$$\left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta R_{L}^{2}} \right\}^{1/2}$$

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$$\left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta R_{L}^{2}} \right\}^{1/2}$$

$$\left. \left( \frac{\partial V_{i}}{\partial R_{L}} \right)^{2} \overline{\Delta R_{L}^{2}} \right\}^{1/2}$$

The value of  $V_i$  in this equation is obtained by eliminating  $t_f$  from equation (7-1) with equation (7-2).

$$V_{i} = \frac{k_{1} \frac{I_{R}}{R} V_{R} k_{2}}{(I_{R} - I_{L})} \left[ 1 - \frac{V_{R} k_{2}}{2 (I_{R} - I_{L}) R_{L}} \right]$$
(7-4)  
$$k_{1} = \frac{I_{i}}{I_{R}}$$
  
$$k_{2} = \frac{C_{R}}{C_{i}}$$

The ratio of the currents charging the reference and unknown capacitors has been replaced by a constant  $k_1$ . This has been done since, for a given unknown element, source variations and photomultiplier power supply variations affect both currents and their ratic remains constant. In other words the current errors are functionally related and correlated and are not statistically independent. The same reasoning applies to the values of the reference and unknown capacitors which are primarily affected by temperature



and hence have a constant ratio. All remaining parameters in equation (7-4) have statistically independent random errors so that the averaging operation removes all but the squared terms from equation (7-3). Evaluating equation (7-3) we find that

$$\frac{1}{\Delta V_{1}^{2}} = \begin{cases} \frac{I_{L}^{2} k_{1}^{2} V_{R}^{2} k_{2}^{2}}{(I_{R} - I_{L})^{4}} & \Delta I_{R}^{2} + \frac{k_{1}^{2} I_{R}^{2} k_{2}^{2}}{(I_{R} - I_{L})^{2}} & \Delta V_{R}^{2} \end{cases}$$

$$+ \frac{k_{1}^{2} I_{R}^{2} V_{R}^{2} k_{2}^{2}}{(I_{R} - I_{L})^{4}} \frac{\Delta I_{L}^{2}}{\Delta I_{L}} - \frac{k_{1}^{2} I_{R}^{2} V_{R}^{4} k_{2}^{4}}{4 (I_{R} + I_{L})^{4} R_{L}} \frac{\Delta R}{\Delta R}$$

Assuming as values for the preceding equation

$$I_{L} = 10^{-9} \text{ ampere}$$

$$I_{R} = 10^{-6} \text{ ampere}$$

$$k_{1} = 1$$

$$C_{R} = 10^{-6} \text{ farad}$$

$$C_{I} = 10^{-7} \text{ farad}$$

$$\frac{\Delta R_{L_{1}}}{R_{L_{1}}} \int_{-\frac{1}{2}}^{1/2} - \frac{1}{2} 20 \text{ percent}$$

$$\frac{\Delta I_{R}}{R_{L_{1}}} = 25,000 \text{ seconds}$$

$$V_{R} = 10 \text{ volts}$$

$$\frac{V_{R}}{V_{R}} = 10 \text{ volts}$$

$$\frac{V_{R}}{V_{R}} = 10 \text{ volts}$$

$$\frac{V_{R}}{V_{R}} = 10 \text{ volts}$$

we find that

$$\left[\frac{1}{\Delta V_1^2}\right] = 7.4 \times 10^{-2}$$

If a typical value of  $V_1$  corresponding to a current of  $10^{-7}$  uniperes is 10 volts, the rms output error is

$$\frac{\left[\frac{1}{\Delta V_{1}^{2}}\right]}{V_{1}} = \pm 0.7^{\pm} \text{ percent}$$

Of the four terms in the expression for

$$\left[\frac{1}{\Delta V_1^2}\right]^{1/2}$$

the second contributes the major portion of the error. The second term represents the effect of uncertainty in the reference capacitor cut-off voltage

7 5.2.2 LeaRage Errors -- In addition to the errors which occur during the charging of the integrating capacitors. which have just been discussed, there can be a variable amount of discharge between the time the capacitor charging is complete and the time that the charge is read out — Ten unknown capacitors are read at intervals of 100 milliseconds. Each of these 0.1 microfarad capacitors has a 25,000 second leakage time constant. Since the voltage during the time the capacitor is allowed to leak is

$$v = V_i \exp\left[\frac{-t}{R_{L_i^{-1}}}\right] \simeq V_i^{+} \left(1 - \frac{t}{R_{L_i^{-1}}}\right) \text{ for } t \ll T_L$$

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the rms error in v for a particular  $V_1$  and t is

$$\begin{bmatrix} \overline{\Delta v}^2 \end{bmatrix}^{1/2} = \left\langle \left( \frac{\partial v}{\partial R_{L_1}} \right)^2 - \overline{\Delta R_{L_1}^2} + \left( \frac{\partial v}{\partial C_i} \right)^2 - \overline{\Delta C_i^2} \right\rangle^{1/2}$$
$$= \frac{V_1}{R_{L_1}} \left\langle \frac{\overline{R_{L_1}}}{R_{L_1}} + \frac{\overline{\Delta C_i^2}}{C_i^2} \right\rangle^{1/2}$$

This equation can be evaluated using the most pessimistic parameter values, namely.

 $V_1 = 10$  volts  $t = 10 \times 100$  milliseconds = 1 second

 $R_{L_i} = 25,000$  seconds



The result is that

 $\frac{\left[\frac{1}{\Delta v^2}\right]}{V} \approx 0.01 \text{ percent, a very small rms variable-capacitor-leakage}$ 



Ar. additional leakage error can occur as each unknown capacitor is read A single high input impedance readout system is used to read all 10 capacitors .After connection is made to a particular capacitor, 30 milliseconds are allowed to pase to assure that all relay contact and amplifier transients have ended and then the capacitor voltage is sampled for 30 milliseconds. The remaining 40 milliseconds of the 100-millisecond cycle is used to disconnect the amplifier from the capacitor. During the sampling operation the voltage leakage increment is

$$\Delta v_{L} = -\frac{V_{i}}{T_{a}} t$$

where

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$$T_a = R_a C_i$$

 $R_{n} \approx$  the amplifier input impedance

The variation in  $\Delta v_L$  as a result of changes in amplifier input impedance and sampling time is the error in this case.

$$\frac{\left[\frac{\Delta v_{L}^{2}}{\nabla_{i}}\right]^{1/2} \cdot \left[\left(\frac{\partial \Delta v_{L}}{\partial R_{a}}\right)^{2} \frac{\Delta R_{a}^{2}}{\Delta R_{a}} + \left(\frac{\partial \Delta v_{L}}{\partial C_{i}}\right)^{2} \frac{\Delta C_{i}^{2}}{\Delta C_{i}} + \left(\frac{\partial \Delta v_{L}}{\partial t}\right)^{2} \frac{\Delta t^{2}}{\Delta t^{2}}\right]^{1/2}}{V_{i}}$$

$$=\frac{t}{\tau_{a}}\left\{\frac{\frac{\Delta R^{2}}{a}}{R^{2}_{a}}+\frac{\frac{\Delta C^{2}_{i}}{\Delta C^{2}_{i}}}{C^{2}_{i}}+\frac{\frac{\Delta C^{2}_{i}}{\Delta t}}{C^{2}_{i}}\right\}^{1/2}$$

The rms error during capacitor readout is 0.01 percent for the parameters





# 7.5.3 Reference Capacitor Final Voltage Errors

The voltage on the reference capacitor at the end of the burn period is. ideally a constant value of the order of 10 volts. However, because of errors occurring in the Adage HA-150 unity gain amplifier, analog-to-digital converter and digital threshold used to monitor the reference capacitor voltage, there will be some small inconsistencies in the final value. These inconsistencies, considered on an rms error basis, represent an output error for an element having a linear working curve. The effect of this type of error was included in the previous section in the form of the term

$$\left(\frac{\partial V_i}{\partial V_R}\right)^2 \frac{1}{\Delta V_R^2}$$
 in equation (7-3).

The unity gain amplifier has errors due to gain stability and drift or offset variations. The output voltage,  $V_{o}$ , is given by

 $V_0 = KV_i + \Delta$ 



where

- K =the gain
  - $V_i =$ the input voltage
  - $\Delta$  = the offset voltage.

Using this equation the rms error in output is

$$\begin{bmatrix} \overline{\Delta V_{o}^{2}} \end{bmatrix}^{1/2} = \left\{ \begin{pmatrix} \frac{\partial V_{o}}{\partial K} \end{pmatrix}^{2} \quad \overline{\Delta K_{2}} + \begin{pmatrix} \frac{\partial V_{o}}{\partial \Delta} \end{pmatrix}^{2} \quad \overline{\Delta \Delta^{2}} \right\}^{1/2}$$
$$= \left\{ V_{i}^{2} \quad \overline{\Delta K^{2}} + \overline{\Delta \Delta^{2}} \right\}^{1/2}$$

Using the values

$$V_1 = 10$$
 volts

 $\begin{bmatrix} 1/2 \\ \Delta \Delta^2 \end{bmatrix}_{1}^{1/2} = \pm \frac{100 \text{ microvolt}}{\sqrt{3}} = \pm 58 \text{ microvolts rms}$ 

$$\frac{\left[\frac{1}{\Delta K^2}\right]^{1/2}}{K} = 10^{-6}$$

V = 10 volts

the equation gives for the unity gain amplifier error

$$\frac{\left[\frac{\Delta V_o^2}{\Delta V_o}\right]^{1/2}}{V_o} = 6 \times 10^{-4} \text{ percent rms}$$



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The analog-to-digital converter has errors due to its finite sampling interval, drift in its voltage-reference, and the resolution of the digital output.

Since the sampling interval is 10 milliseconds and since the ramp input to the converter has a slope of 10 volts in 10 seconds or 1 volt per second, the voltage change between samples is  $10^{-2}$  volt or 0.1 percent of full scale. The rms error is 0.1 percent/ $2\sqrt{3}$  or 0.03 percent as a result of the finite sampling interval. The digital resolution is one part in 512 for a 9-bit system, an rms error of 0.06 percent and the drift in the voltage reference is 0.01 percent rms.

Combining all errors for both analog-to-digital converter and unity gain amplifier, the total uncertainty in the reference capacitor cutoff voltage is approximately 0.07 percent rms<sup>\*</sup>.

# 7.5.4 Dark Current Cancellation Error

Reversing relays and a dark current cancellation shutter for the source are provided in the Base Level Analyzer for the elimination of dark current contributions to the charge stor ' on the integrating capacitors. Charge is stored over a number of regular cycles, light source produced photocurrent plus dark current being added to the capacitor in one half cycle and dark current alone being subtracted in the next half cycle. Figure 7-5 shows the approximate effect on dark current alone. Starting with one-half of a positive dark current pulse assures that the mean dark current charge increment will be zero, since it is equally likely that the threshold will be reached and shutoff will occur at anytime during the positive dark current pulse, a result of the decision to control the final reference capacitor voltage rather than the source burn time.

The effect of this error is included in paragraph 7.5.2.1 and need not be considered again separately.



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Although the mean dark current voltage increment shown in figure 7-5(b) is zero, the rms increment is non-zero

$$\begin{bmatrix} \frac{1}{2} \end{bmatrix}_{D}^{1/2} \begin{bmatrix} \frac{1}{2} \\ \left( \begin{array}{c} P \\ s \end{array} \right) V_{D}^{2} \\ \frac{1}{2} \\$$

......

$$= \left( \begin{array}{c} T & 1 & 2 \\ T & T & 1 & 2 \\ T & 0 & C^{2} \end{array} \left[ -\frac{T}{2} + t_{s}^{2} \right]^{2} dt_{s} \right) = \frac{I_{D}T}{4\sqrt{3}C}$$
(7-6)

The full rms voltage increment given in equation (7-6) does not represent an output error since comparable errors of different magnitude appear in both reference capacitor voltage and unknown capacitor voltage and it is the ratio of these voltages which is of interest. If the subscripts, i and R represent unknown and reference respectively then the fractional error is the voltage ratio is

$$\epsilon = \frac{\frac{V_{1} + V_{D_{1}}}{V_{R} + V_{D_{R}}} - \frac{V_{1}}{V_{R}} - \frac{V_{D_{1}}}{V_{1}} - \frac{V_{D_{R}}}{V_{R}}}{V_{R} - \frac{V_{D_{1}}}{V_{1}} - \frac{V_{D_{1}}}{V_{R}}} - \frac{V_{D_{1}}}{V_{1}} - \frac{V_{D_{R}}}{V_{R}}}{V_{1} - \frac{V_{R}}{V_{R}}}$$
(7-7)



Using rms values of dark current voltage increment



and since

$$V_{i} = \frac{NT I}{C_{i}}$$

where N is the number of dark current reversals

$$\epsilon \simeq \frac{I_{\rm D}}{4\sqrt{3}'\,\rm N} \left[\frac{1}{I_{\rm i}} - \frac{1}{I_{\rm R}}\right] \tag{7-8}$$

Equation (7-8) can be evaluated using as pessimistic values of the parameters. typical values for low unknown impurity concentrations and maximum dark currents

$$I_{D} = 10^{-8} \text{ amperes}$$

$$N = 10$$

$$I_{i} = 10^{-9} \text{ amperes (5 ppm impurity level)}$$

$$I_{R} = 10^{-6} \text{ amperes}$$

These values lead to an rms error,  $\epsilon$ , of  $\pm 14.4$  percent at maximum levels of dark current and low levels of signal current for the unknown. If lower dark current values can be achieved by selection of photomultipliers, say  $10^{-9}$  amperes, the error is decreased to 1.4 percent. Additional improvement



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can be achieved by increasing the number of reversing cycles, N. Since this decreases relay life, only a modest increase, say a factor of two, is possible.

A problem of accuracy exists at the high operating temperature levels which may be experienced (130°F). At these temperatures dark current in photomultipliers may increase as much as a factor of six<sup>\*</sup> over room temperature values which are typically around  $10^{-9}$  amperes in selected tubes. At the 10-ppth impurity level, the error would be increased from 0.7 percent, the room temperature value, to 4.2 percent at 130°F.

# 7.5 5 Readout Circuit Errors

In order to measure the voltage stored on an unknown integrating capacitor after the source burn and to convert this voltage nonlinearly to a new voltage proportional to unknown concentration, a special readout circuit is employed. This circuit consists of a high input impedance unity gain amplifier, an operational amplifier with non-linear feedback networks and an analog-to-digital converter. In addition digital storage registers and readout are used.

The error contribution of the Adage HA-150 unity gain amplifier results from variations or drifts in offset voltage, noise and gain instability. The offset voltage can vary between  $\pm$  100 microvolts so that the rms error is 58 volts. The amplifier noise is 70 microvolt peak-to-peak or 10 microvolts rms. In addition the gain variations are 1 ppm. Assuming that all these errors are statistically independent and Gaussian they may be combined by taking the square root of the sum of the squares. If a 500-ppm impurity level is equivalent to a 10-vol charge on the unknown capacitor, the combined errors of the unity gain amplifier at a 500-ppm level are 0.006 percent. At 10 ppm, the unity gain amplifier error becomes 0.03 percent and at1 ppm, 0.3 percent.

RCA Technical Manual PT-60, "Phototubes and Photocells", p. 96, Radio Corporation of America, Lancaster, Pa., 1963.

The operational amplifier error contribution arises from gain instability and offset voltage drifts with changing temperature and supply voltage. The gain stability is determined by the stability of the resistances in the feedback loop. If these have 25-ppm stability, then the gain stability is 0.004 percent. The temperature produced offset is 3 microvolts per °C or  $\pm$  90 microvolts over the temperature range. The supply voltage produced offset is  $\pm$  200 microvolts per volt. Since the 28-volt supply used has a 0.05 percent per °C temperature coefficient, an additional 72 microvolts can be added for a total offset voltage variation with temperature of  $\pm$  162 microvolts. At the 500-ppm impurity level the total operational amplifier error is 0.004 percent rms and results from gain instability. At the 10-ppm impurity level, the operational amplifier drift gives a  $\pm$  0.1 percent error over the temperature range while at the 1-ppm level there is a  $\pm$  1 percent error over the temperature range.

The analog-to-digital converter errors result from the instability of its voltage reference, which is known to be  $\pm 0.01$  percent, and from the digital resolution of a 9 bit binary system. The latter is a one part in 512 peak-to-peak error or a 0.06 percent of full scale rms error so that it predominates. The total analog-to-digital converter error is 0.06 percent rms or 0.3 ppm at 500 ppm, 3 percent rms or 0.3 ppm at 10 ppm, and 30 percent rms or 0.3 ppm at 1 ppm.

The digital portion of the readout system, including registers and Nixie tubes is assumed to be error free

## 7.6 Summary of Base Level Analyzer Errors

A summary of the errors calculated in the previous paragraphs is given in table 7-1. No attempt has been made to combine all errors since many reach meaningful levels only under particular adverse environmental conditions However, the most important error sources are imperfect dark current Table 7-1

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# Summary of Base Level Analyzer Errors

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cance laws and electronics readout errors (primarily digital resolution error). Poth of these sources of error are most important at impurity concentrations below 10 ppm and together can produce a 5 percent rms error at high temperatures at the 10-ppm impurity level.

One of the worst possible combinations of conditions is found in a high altitude, hot, humid location. If the maximum limit of the temperature, pressure and humidity specifications all occur together, iron ( $\lambda = 2599$  A) experiences a 5.5 percent rms or 0.55 ppm error at the 10 ppm level plus offset errors totaling 2 percent. The offset error remains constant at lower impurity levels while the rms error is inversely proportional to concentration. Thus, at the 1-ppm level of iron the error will be about 55 percent rms or or 0.55 ppm. The error for low concentrations of lead may well exceed that of iron since the lead line used is also at short wavelengths (2203 A) and the sensitivity shown by the lead working curve is less than that for iron.

# 8. RELIABILITY

# 8.1 Reliability Program Plan

# 8.1.1 Basic Philosophy

The basic philosophy of the Base Level Analyzer Reliability Program is to emphasive adequate reliability controls during the design phase of the project. These controls reflect the requirements of the procuring agency's work statement and are the cornerstone of the reliability structure. Baird-Atomic takes the position that reliability, by definition, must be inherent in the basic design, is no amount or reliability controls in areas other than the design phase can ever correct or make up for basic design defects.

As a support or a back up function to the design control phase, the reliability effort will also be directed toward ensuring that the procurement, manufacture ing, safety human factors, and maintainability elements of the total program are integrated to that point where each element contributes a minimum degradation to the reliability of the analyzer.

# 8 1.2 Peliability Orgalization

The organization structure of reliability, and its immediate access to top management is is shown in figure 8-1

# 8 1 5 General Reliability Requirements

The Baird-Atomic Reliability Program requirements are as specified in the following documents:

# a MIL-STD-441

b MIL-R-26484A except that MIL-T-21200D shall be substituted for MIL-E-5400, in accordance with RFQ 33-657-65-5001, dated 31 July 1964

In general the Reliability Program will consist of the following basic \* elements:

a. Design Review Program



- b. Parts Program
- c. Milestone Identification and Schedule
- d. Monitoring Program
- e. Failure Analysis and Reporting
- f. Reporting

# 8.1.4 Design Review Program

During the design phase of the project, the reliability of each subsystem proposed is carefully considered before final design decisions are made. The reliability design review shall take into consideration the following factors:

- a. Concept and Design Approach
- b. System and Component Design and Performance Requirements
- c. Environmental Requirements
- d. Maintainability Requirements
- e. Safety Requirements

In the area of mechanical design, stress analyses will be performed to assure that the critical mechanical structures will not fail in the environment specified or will not fail under the shock and vibration tests specified. In the area of optical design, tolerance studies will be made to assure that the manufactured optical system will have the same performance as the computed requirements. Also, in the electronic design area, preferred parts and circuits will be selected for their reliability aspects and will be employed with proper derating. Circuits will be simple, have a minimum number of parts, and incorporate redundancy where it appears de irable.

The design review program will consist of a critical failure mode analysis of each major subassembly. A system diagram will be used for the initial prediction. Each subassembly will be assigned a reliability apportionment,



and a detailed analysis of all parts comprising each subassembly will be made relative to the apportionment. Updating and reliability trade-offs will be made as required by the program position.

The analyzer reliability shall be assumed to be in accordance with an exponential distribution; that is,

 $R = e^{-\lambda_s t}$ 

where

 $R = system reliability \ge 0.9989$ 

= system failure rate =  $1/MTBF \le 1/150^{failures/HR}$ 

t = required operational time; 10 minutes

. . . .

As a basis for total life requirements, maintainability, etc., it will be assumed that the longevity (total operating life) requirement, in accordance with MHJ-R-2648A, shall be 3000 hours, minimum. It is implicit that at any time the equipment is operated during this 3000 hour operational period, the probability of it completing a 10-minute run will be 0.9989.

RADC Reliability Notebook, AD-148868, and MIL Handbook 217 will be used in the assignment of failure rates when documented evidence from the supplier of a specific part or subassembly is not available upon demand. It is understood that both the RADC Notebook and MIL Handbook 217 offer only engineering estimates of failure rates and as such do not constitute proof of a failure rate specified. Moreover, it is also understood that a failure rate specified by a vendor of a component or subassembly for the analyzer may be an average failure rate attained over a period of time when the vendor has controlled manufacturing and testing techniques. The failure rate specified by the vendor does not have to necessarily reflect the lot from which the component or subassembly supplied to Baird-Aton... was selected.

# 8.15 Parts Program

Baird-Atomic reliability shall sign off all requisitions relative to end item procurement of all major parts and/or subassemblies of the analyzer. In this way, complete control is maintained by the reliability organization.

# 8.1.6 Milestone Identification and Schedule

Milestone identification and schedule will be in accordance with figure 8-2.

# 8.1.7 Monitoring Program

Reliability monitoring will be involved with the following

a. Procurèment

b. Development Testing

c. Manufacturing

d. Acceptance Testing

e. Maintainability Evaluation

8, 1.7.1 <u>Procurement</u> -- Where possible, each part and/or subassembly of the analyzer will be a "preferred" item and will be purchased in accordance with the pertinent specification Qualified Parts List (QPL).

When a required item does not conform to either the RADC Notebook or MIL Handbook 217, the project manager will be so informed by Reliability. The supplier of this item will be requested to inform Baird-Atomic Reliability as to a failure rate relative to our specific usage. A reliability report will be compiled on each of these items outlining all judgments as to predicted reliability.

8.1.7.2 Development and Acceptance Testing -- Development and acceptance testing will supply all data relative to justification of the design reliability prediction. All data gathered during these tests will be used to verify the reliability prediction by means of a sequential test analysis technique. 

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8.1.7.3 <u>Manufacturing</u> -- Techniques of manufacturing and assembly of each analyzer will be monitored by reliability in close cooperation with Quality Assurance personnel in order to insure a minimum degradation of the analyzer design reliability.

8.1.7.4 <u>Maintainability Evaluation</u> -- During the prototype testing phase, the maintainability shall be verified by simulations in accordance with paragraph 40.1.1 of MIL-M-26512B.

8, 1, 8 Failure Analysis and Reporting

A test summary tabulated against consecutive failure numbers shall be madeincluding both relevant and nonrelevant failures. The test summary shall contain the following items:

- a. Failure Number
- b. Failure Analysis Number
- c. Date
- d. Part Name.
- e. Part Number and Symbol
- f. Subassembly
- g. Major Assembly
- h. Serial Number of Equipment
- i. Hours of Operation (both equipment and total accumulated)
- j. Mode of Operation
- k. Manufacturer of Failed Part
- 1. Symptom of Failure

m. Cause of Failure

- n. Corrective Action for Test Continuation
- o. Diagnosis Time for Each Skill Level
- p. Repair and Retest Time for Each Skill Level
- q. Categorization of Failure
- r. Result of Failure (catastrophic abort, degradation, etc.)

A full tabulation of failures will be submitted with every monthly report.



# 8,1.9 Renability Reporting

Baird-Atomic Reliability will submit a progress report each month as part of the Contract Progress Report,

Each monthly report will contain a complete treatment of any trouble areas and will emphasize corrective action required and/or completed. Updated system reliability prediction will be submitted in accordance with the timing specified in figure 8-2. Each updated prediction will include an updated electronic parts list in accordance with figure 8-3.

The final report, at the end of each phase of the program, will consist of a complete analysis and prediction of the probability of success of the analyzer.

# 8.2 Reliability Analysis and Prediction

# 8.2.1 Introduction

In accordance with the parameters as specified in RFQ 33-657-65-5001 and the basic reliability requirements as outlined in MIL-R-26484A (USAF), this report includes those aspects relative to the predicted reliability of the Base Level Analyzer design.

The operational life of the instrument is defined as 3000 hours minimum in accordance with MIL-R-26484A (USAF). It is assumed that the 3000-hour operational life of the instrument is defined as system ON time, not specimen analysis time.

Throughout this analysis, failure rate assignments are made in accordance with the following:

a. The failure rate quoted by the proposed supplier, (where documentary evidence to justify the supplier's claim is available to Baird-Atomic upon demand).

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b. MIL-Handbook 217, where component will be purchased to the appropriate MIL specification.

Best engineering estimate of the proposed supplier.

8.2.2 Summary

use period of 10 minutes. These criteria specify an MTBF of 150 hours.

In accordance with the positions outlined in the Reliability Program Plan, the Base Level Analyzer design has a predicted reliability and associated Mean Time Between Failures (MTBF) as shown in figure 8-4.

Figure 8-4 shows that the reliability and MTBF of the analyzer design ranges from a 0.'999799 and 820 hours respectively at 60 runs per hour to 0. 993928 and 2270 hours respectively at 8.4 runs per hour, this being the maximumminimum number of estimated runs per hour on the basis of a one man operation.

On a design basis, therefore, the Base Level Analyzer exceeds the required MTEF of 150 hours by a minimum factor of 5.5 at 60 runs per hour and a maximum factor of 15.0 at 8.4 runs per hour.

For completeness, Baird-Atomic Reliability has investigated the concept of treating the 10 integration X channels as a parallel system and has determined the increased reliability of the system assuming some X-channels may fail without system degradation — Paragraph 4-3 gives a complete treatment of the analysis which shows an increased system MTBF of from 114.5 to 131-3 percent.

It is the conclusion of Baird-Atomic Reliability that the analyzer design as reflected in deliverable hardware will more than meet requirements.

8-10



Figure 8-4. Reliability and MTBF versus Number of Sample Runs per Hour



# 8.2.3 Reliability Prediction

8.2.3.1 <u>General Introduction</u> -- In accordance with MIL-R-26484A (USAF) the exponential distribution is assumed for the proposed design. <sup>m</sup> erefore

$$-\lambda t$$
  
R = e

where

**R** = system reliability = 0.9989

$$\lambda_{s}$$
 = system failure rate =  $\frac{1}{MTBF} = \frac{1}{150}$ 

t = operational time period = 10 minutes

Since the required MTBF = 150 hours, the failure rate design goal of the Base Level Analyzer system is as follows:

 $\lambda_{s} = \frac{1}{150} = 6,667,000 \times 10^{-9}$  failures per hour

8.2.3.2 Functional Operation of the Base Level Analyzer -- The first step in the reliability analysis was to generate a reliability functional operation diagram of the proposed system (figure 8-5).

8.2.3.3 <u>Reliability Block Diagram</u> -- The second step in the reliability analysis was to generate a reliability block diagram of the proposed system (figure 8-6). It will be noted that there are 10 major subdivisions of the analyzed system as follows:

 $\mathbf{R}_1 = structures$ 

R,

- R<sub>2</sub> = miscellaneous electrical components
  - = excitation source system
  - # source shutter mechanism
  - = optical system







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Figure 8-6. Reliability Block Diagram



 $R_6 = monitor servo system$ 

R<sub>7</sub> = tilt-normal system

 $R_8 = signal receiving system$ 

R<sub>o</sub> = signal processing systems

 $R_{10}$  = signal readout system

The following analysis takes each of the above major subdivisions specified together with their associated subassemblies and fully outlines the position taken relative to each failure rate assignment.

8.2.3.4 <u>Subsystem Analysis</u> -- In paragraphs 8.2.3.4.1 through 8.2.3.4.10 and tables 8-1 through 8-6, all subassemblies are noted for the sake of continuity. However, when a particular subassembly and its operation is a function of the number of analyses made per hour, the failure rate is not stated. Paragraph 8.2.3.4.11 and table 8-7 contain all failure rate assignments as a function of analyses made per hour.

8.2.3.4.1 <u>Structures</u> -- Structures used in the Base Level Analyzer design include the following:

R<sub>11</sub> = case and support structure R<sub>12</sub> = exit/entrance slits, lens and mirror mounts R<sub>13</sub> = optical encasement

It is assumed that all structures reflect a mature design with the incorporation of proper safety factors as determined by structural analyses. Therefore, all structures are defined as being represented by a rectangular failure density function, and the apportioned reliability value is assumed to be 1.00, or

 $\lambda_{11} = \lambda_{12} = \lambda_{13} = \lambda_{1} = 0$ 

8.2.3.4.2 <u>Miscellaneous Electrical Components</u> -- There are eight subgroups under miscellaneous electrical components, R<sub>2</sub> as follows:

 $\mathbf{R}_{21}$  = source setup switch

 $R_{22}$  = photomultiplier gain adjust

 $R_{23} = start switch$ 

 $R_{24} = programmer$ 

 $R_{25}$  = system air intake

 $R_{26}$  = interlocks

 $R_{27}$  = air port actuators

 $R_{28}$  = power supplies

In accordance with table 8-1 for that part of the total failure rate,  $\lambda_{2}$ , which is not effected by the number of analyses per hour, is as follows:

 $\lambda_{2} = 110,380 \times 10^{-9}$ 

8.2.3.4.3 Excitation Source System -- There are six subgroups under excitation source system, R<sub>3</sub> as follows:

R<sub>31</sub> = electrodes R<sub>32</sub> = gap adjustment mechanism R<sub>33</sub> = RF meter R<sub>34</sub> = control gap air pump R<sub>35</sub> = mercury lamp R<sub>36</sub> = rotary electrode drive

Since the analytical electrodes are replaced every run and the control gap adjustment mechanism is manually operated, it is assumed that  $R_{31}$ and  $R_{32}$  can be defined as a rectangular failure density function, and the apportioned reliability value is assumed to be 1.00 or  $\lambda_{31} = \lambda_{32} = 0$ . Note: See paragraph 8.2.3.4.6 for a full treatment of the mercury lamp.

-	Relia	ability Data S	shee	st. Miscella	neous Elec	ctrical Comp	onents,	R2	<b>b</b> erne an a some ton set at the set of	<b></b>
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	(c) Relay Purer	28-	-	Elec Research. Inc.	MS 284	M11E×4400		<b>9</b> to 30°C	Electric Corporation of America	25, 000
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In accordance with table 8-2 the total failure rate for that part of  $\lambda_3^{(*)}$ , which is not effected by the number of analyses per hour, is as follows:

$$\lambda_3 = 0 \times 10^{-9}$$

8.2.3.4.4 Shutter Mechanism -- There are two subgroups under the shutter mechanism,  $R_A$ , as follows:

 $R_{41} =$ shutter relay  $R_{42} =$ shutter switch

Since both  $R_{41}$  and  $R_{42}$  are a function of the number of analyses made per hour, the failure rate,  $\lambda_4$ , which is not effected by the number of analyses made per hour, is as follows:

$$\lambda_4 = 0 \times 10^{-9}$$

8.2.3.4.5 Optical System -- There are three subgroups under the optical system, R<sub>5</sub>, as follows:

 $R_{51}$  = grating  $R_{52}$  = lens, mirror, and filter systems  $R_{53}$  = focal plane

 $\lambda = \lambda = \lambda = \lambda = \lambda$ 

Since the optical elements will be structurally sound and the transmission and reflectance characteristics are not expected to degrade over the 3000-hour operational period, a rectangular failure density function is assumed and the apportioned reliability values are assumed to be 1.00 for each of the subgroups. Therefore,

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# Table 8-3 Reliability Data Sheet, Monitor Servo System R<sub>6</sub>

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8.2.3.4.6 <u>Monitor Servo System</u> -- There are eight subgroups under the monitor servo system, R<sub>b</sub>, as follows:

R <sub>61</sub>	8	servo amplifiers
R <sub>62</sub>	=	servo motor
R <sub>63</sub>	Ħ	corrector plate
R <sub>64</sub>	=	photomultiplier tubes and recepticles
R <sub>65</sub>	=	mercury lamp
R <sub>66</sub>	H	bar adjust (manual)
R <sub>67</sub>	=	manual ON-OFF switch
R <sub>68</sub>	=	servo shutter mechanism

It is assumed that all structure mechanisms reflect a mature design with the incorporation of proper safety factors as determined by a structural analysis. Therefore, all structures are defined as being represented by a rectangular failure density function, and the apportioned reliability value for  $R_{63}$  and  $R_{66} = 1$ , or  $\lambda_{63} = \lambda_{66} = 0$ .

In accordance with table 8-3 (page 8-19), the total failure rate for that part of the monitor servo system,  $\lambda_6$ , which is not effected by the number of analyses per hour, is as follows:

 $\lambda_6 = 59,100 \times 10^{-9}$ 

The mercury lamps (one in the high voltage supply near the auxiliary gap and the other the reference mercury line source in the monitor servo system) are purchased from the commercial division of Westinghouse as their type 794H.

Westinghouse has supplied Baird-Atomic with their best engineering estimate of lamp life characteristics as follows:

Random Failure Rate 80 percent probability that any lamp will attain a life of 6000 hours



# Mean Life

6000 hours with a standard deviation of approximately 1400 hours

The random failure rate is computed as follows:

 $\begin{array}{c} -6000 \ \lambda \\ random \end{array}$ 

where

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 $\lambda$  = 37,200 x 10<sup>-9</sup> random

Assuming four standard deviations from the mean, the wearout probability at (6000 - 4 x 1400) = 400 hours is 0.000032 or the probability of no wearout is 0.399968. This probability of no wearout when considered at a time of 400 hours converts to a wearout,  $\lambda_{i}$ , of zero.

Therefore, the failure rate for the mercury lamp is  $37,200 \times 10^{-7}$  when the lamp is replaced every 400 hours. This will be considered a basic maintainability requirement.

8.2.3.4.7 <u>Tilt-Normal System</u> -- There are four subgroups under the ultnormal system, R<sub>7</sub>. as follows:

 $R_{71} = tilt-normal switch$   $R_{72} = tilt-normal meter$   $R_{73} = photomeltiplier tube and recepticle$  $R_{74} = meter shunt$ 

The tilt-normal switch failure rate is identical with item R<sub>21</sub> failure rate, Therefore, from table 8-1,

 $71 = 860 \times 10^{-9}$ 



8-22

The tilt-normal meter failure rate is identical with item  $R_{33}$  failure rate but with a duty cycle of 1/120. Therefore, from table 8-2,

$$\lambda_{72} = \frac{1}{120} \times 5100 \times 10^{-9}$$

or

$$\lambda_{72} = 40 \times 10^{-9}$$

The photomultiplier tube and recepticle failure rate is equal to one-half of item R failure rate. Therefore, from table 8-3, 64

$$\lambda_{73} = 5950 \times 10^{-9}$$

The meter shunt is a standard potentiometer and MIL-HDBK-217 defines such as follows:

$$\lambda_{74} = \frac{1}{120} \times 300 \times 10^{-9}$$

or

$$\lambda_{74} \cong 0$$

The failure rate for that part of the tilt-normal system  $\lambda_7$  which is not effected by the number of analyses made per hour is as follows:

 $\lambda_7 = \lambda_{71} + \lambda_{72} + \lambda_{73} + \lambda_{74}$ 

or

 $\lambda_{m} = 6850 \times 10^{-9}$ 

8.2.3.4.8 Integrated relay network -- There are seven subgroups under integrated relay network, R<sub>g</sub>, as follows:

 $R_{81} = photomultiplier tubes and recepticles$   $R_{82} = integration capacitors$   $R_{83} = "S" relays$   $R_{84} = "R" relays$   $R_{85} = "M" relays$   $R_{86} = "C" relays$   $R_{87} = clear register network$ 

In accordance with table .8-4, the total failure rate for that part of  $\lambda_8$ , which is not effected by the number of analyses made per hour. is as follows:

$$\lambda_8 \approx 65,450 \times 10^{-9}$$

8.2.3.4.9 Signal Processing System -- There are five subgroups under signal processing system, R<sub>q</sub>, as follows:

 $R_{91}$  = unity gain amplifiers  $R_{92}$  = relay network and nonlinear shapers  $R_{93}$  = operational amplifier  $R_{94}$  = analog-to-digital converter  $R_{95}$  = register assembly

In accordance with table 8-5 the total failure rate of that part of  $\lambda_9$ , which is not effected by the number of analyses per hour, is as follows:

 $\lambda_0 = 71,430 \times 10^{-9}$ 

8.2.3.4.10 Signal Readout System -- There are four subgroups under signal readout system, R<sub>10</sub>, as follows:

 $R_{101} = 10$ -position manual readout switch  $R_{102} =$  accumulator count-down circuit



8-24

# Table 8-4 Reliability Data Sheet Integrated and Relay Network

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## Reliability Data Sheet. Signal Processing System, R9

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# Reliability Data Sheet, Signal Readout System, R

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R<sub>103</sub> = pulse counter circuit R<sub>104</sub> = nixie display assembly

In accordance with table 8-6 (page 8-25), and since all four items are effected by the number of analyses made per hour, the failure rate,  $\lambda_{10}$ , which is uneffected by the number of analyses made per hour is as follows:

$$\lambda_{10} = 0 \times 10^{-9}$$

8.2.3.4.11 Failure Rates Effected by Number of Analyses per Hour -- Table 8-7 presents a complete analysis of those items effected by the number of analyses made per hour on the Analyzer.

8.2.4 Conclusions

8.2.4.1 Predicted MTBF -- A summary of the predicted failure rates is in accordance with table 8-8. The resulting system failure rate is as follows:

System fàilure rate.  $\lambda_{g} = [313, 210 + 15, 084 \text{ X}] \times 10^{-9}$  failure per hour

where

 $\mathbf{X}_{i}$  = number of analyses made per hour

Predicted MTBF =  $\frac{1}{\lambda_s}$  =  $\left[\frac{1}{313,210+15,084 \text{ X}}\right] \times 10^9$ 

24.2 Predicted Reliability -- Using the exponential,

Table 8-7Reliability Data Sheet, All Assemblies Effected

		• •	Operating Data am	ð	-	Effer ted E al	lute Rate			
	ź	Discription	Figure Number	11	r Fatlur	FALET, X 10.4	F'allures/Obseration in 10"4	Time Par	Operations Per	Failures For
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Table 8-8 Failure Rate Summation

	Number of Analyses/		. Failure Rates Affects Number of Analyses/l	d by Hr
R. No. 8 Subgroup Identification	Reference		λ	A _ Reference
	Para 8.2.3.4.1	Ö,	0	Para 8. 2. 3. 4. 1
R2 Miscellaneous electronic components	Table 8-1	1140, 380 a	253.4X	Table;8-7
R35 1 Xcitation source system.	Table 8+2	, o ,	513, 8X	Table 8-7
Real Squates mechanism	Para 8.2.3.4.4	· · · ·	8ć9. 4X	Table 8-7
ASS Optical system	Para 8. 2. 3. 4. 5	O ↔	0	Para 8, 2, 3, 4, 5
Kolding Monitors servo system	Table 8-3	59.100	1,490.0X	Table 8-
Kows Tilt normal oystem	Para 8.2.3.4.7	6,850	0	Table 5-7
Re Signal receiving system	Table 8-4	65,450	:9.465.0X	Table 8-7
《大学》》: 《 Sugral processing system	Table 8-5	71,430	1,071.5X	Table 8-7
[[R]]0%]] [] Signal Feadout system	Table 8-6	0	I,420.8X	Table 8-7
	Totals	24, = 313, 210 × 10-9	EA_X =15.084X × 10-9	

failure rate of those systems UNAFFECTED by the number of analyses per hour, failures per hour

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a number of analyses made per hour

# failures per analysis

X = failure rate of those systems AFFECTED by the number of analyses per hour, failures per hour

System fainte rate  $\lambda_s = \Sigma \lambda_1 + \Sigma \lambda_a X$  or  $\lambda_s = \{313, 210 + 15, 084X\} \times 10^{-9}$  failures per hour





8=29

where

R = system reliability

 $\lambda_{c}$  = system failure rate = [313,210 + 15,084 X] x 10<sup>-9</sup> failures per hour.

t = operational use period = 10 minutes

then

$$R_s = \exp \left[ -\frac{1}{6} (313, 210 + 15, 084 \text{ X}) \times 10^{-9} \right]$$

Figure 8-4 is a plot of both MTBF and  $R_s$  as a function of X, where X is the number of analyses made per hour.

If for a one-man operation it is assumed that each analysis takes five minutes and the operator exhibits a 70 percent efficiency, then the number of analyses made per hour will be as follows:

 $\frac{1 \text{ analysis}}{5 \text{ minutes}} \times \frac{60 \text{ minutes}}{\text{hour}} \times 0.70 = 8.4 \text{ analyses per hour}$ 

On this basis, that is, X = 3.4 analyses per hour,

 $\mathbf{R}_{1} = 0'.999928$ 

 $\mathbf{MTBF} = 2273 \text{ hours}$ 

at 8.4 analyses per hour

If the Analyzer is operated at 60 analyses per hour,

R = 0.999799

MTBF = 820.9 hours

at 60 analyses per hour



Minimum Excess =  $820.9/150 \approx 5.5$ 

Maximum Excess =  $2273/150 \approx 15.0$ 

8.2.4.3 Parallel Channel Considerations -- The purpose of this section is to determine the magnitude of increased system reliability if some failures are permissable in the ten X-channels.

There will be three cases considered:

a. Case I -- The operation of three specific X-channels will be assumed to be mandatory. This investigation will determine what increase in reliability will be attained assuming one up to seven failures are permitted in the balance of seven X-channels.

b. Case II -- The operation of four specific X-channels will be assumed to be mandatory. This investigation will determine what increase in reliability will be attained assuming one up to six failures are permitted in the balance of six X-channels.

c. Case III -- The operation of five specific X-channels will be assumed to be mandatory. This investigation will determine what increase in reliability will be attained assuming one up to five failures are permitted in the balance of five X-channels.

Those portions of the system that are included in a parallel channel are in accordance with table 8-9.

From paragraph 8.2.4.1. the system failure rate (with no permissable failures) is defined as follows:

 $\lambda = [313,210 + 15,084 \text{ X}] \times 10^{-9}$ 



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## Table 8-9

X-Channel Breakdown

ltem	Channel Component and Breakdown	$\lambda_{u} \times 10^{-9}$	$\lambda_a X \times 10^{-9}$	Table Reference
l Photomultiplier Tube l Photomultiplier Receptacle	Ĵ.	5950		8-4
Relay Network and Integrator	l "R" Relay l "S" Relay l "M" Relay l "C" Relay		66 4X 66 4X 66 4X 66 4X	8-7
Relay Network and Nonlinear Shapers	1.4 "C" Relays 2 Diodes 5 Resistors 6 Potent ométe	<b>1</b> 8	93 0X 0.006X 0.010X 0.050X	8-7
Register	2 Modules	4700		8-5
Photomultiplier Gain Adjust		809		87-1
	Totals	$\sum_{a} = 11,450 \times 10^{-9}$	$\sum_{a} \lambda_{a} X = 358.7$	X x 10 <sup>-9</sup>

The failure rate for one of the ten channels under parallel circultry considera-

tion is as follows:

-9 11, 450 + 358. 7X x 10  $\lambda_{1} \equiv =$ 

where X = number of analyses made per hour-



From table 8-9, the failure rate for one channel under the parallel circuitry consideration is

$$= [11, 450 + 358.7 \text{ X}] \times 10^{-9}$$

and for the ten channels, the total failure rate is

 $10 \lambda_1 = [114, 500 + 3587 X] \times 10^{-9}$ 

 $= e^{-\lambda}$  series t

If the system that is not included in the parallel circuitry is defined as the series portion,  $R_{series}$ , then  $R_{series}$  is defined as follows:

= 10 minutes

R scries

where

 $\lambda_{10} = \lambda_{10} \lambda_{1} = [313, 210 + 15, 084 \text{ X}] \times 10^{-9} - [214, 500 + 3587 \text{ X}] \times 10^{-9}$ 

 $\hat{\lambda}_{series} = [198, 710 + 11, 497 X] \times 10^{-9}$ 

For system reliability, when considered from the parallel circuitry point of view

R system parallel x R series

R series = [198,710 + 11,497 %] 1 x 10<sup>-9</sup>

R and a determined relative to the particular case involved.



a. Case I -- Case I requires that three specific X-channels of the ten available operate. Therefore, R parallel is that probability of success of the seven remaining channels operating when one or more these seven channels may fail

$$R_{parallel} = \sum_{k=0}^{r} {\binom{n}{k}} {\binom{n}{k}} {\binom{n-k}{l}} {\binom{1-R_{p}}{k}}^{k}$$

where

- n = seven channels
- r = maximum number of channel failures permitted

÷.

 $k = failures considered, ranging from 0 \rightarrow r$ 

 $R_n =$  the probability of successful operation of one channel = e\_

or

 $-\frac{1}{6} [11, 450 + 358.7 \text{ X}] \times 10^{-9}$ R

If we consider 60 analyses per hour as the worst case,

$$R_{p} = e^{-\frac{1}{6} [11, 450 + 358.7 \times 60] \times 10^{-9}} = e^{-5495 \times 10^{-7}}$$

or

R = 0.99999455

Since  $R_p = 0.99999455$  is so close to 1.0, even one permitted failure will, make  $R_p$  be equivalent to 1.00 by observation.

Therefore, for Case I

 $R_{system} = R_{series} \times R_{3 channels}$ 

$$-[198,710 + 11,497 \text{ X}] \frac{1}{6} \times 10^{-9} -3 [11,450 + 358.7 \text{ X}] \frac{1}{6} \times 10^{-9}$$
  
= e x e

or  

$$R_{system} = e^{-[38, 843 + 2, 096 X] \times 10^{-9}}$$

and

Sandy and the start of the

$$MTBF_{system} = \frac{1}{[233,060 + 12.573 \text{ X}] \times 10^{-9}}$$

b. Case II -- Case II requires that four specific X-channels of the ten available operate. Using the same arguments as in Case I,  $R_p = 0.99999455$ and therefore  $R_{parallel} = 1.0$  when at least one failure is permitted.

$$R_{system} = R_{series} \times R_{4 \text{ channels}}$$

$$-[198,710 + 11,497 \text{ X}] \frac{1}{6} \times 10^{-9} -4 [11,450 + 358.7 \text{ X}] \frac{1}{6} \times 10^{-9}$$

$$= e \times e$$

or

$$\mathbf{R}_{system} = e^{-[40,752 + 2,155 X] \times 10^{-9}}$$

and

$$MTBF_{system} = \frac{1}{[244, 510 + 12, 932 X] \times 10^{-9}}$$

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a. Case III -- Case III requires that five specific X-channels of the ten available operate. Again using the same arguments of Case I.

 $R_{system} = R_{series} \times R_{5 channels}$ 

$$-[198,710+11,497 X] \frac{1}{6} \times 10^{-9} -5[11,450+358.7 X] \frac{1}{6} \times 10^{-9}$$

or

 $R_{system} = e^{-[42,660 + 2.215 X] \times 10^{-9}}$ 

and

$$MTBF_{system} = \frac{1}{[255,960 + 13,291 X] \times 10^{-9}}$$

Looking at the effects of Cases I, II, and III, with respect to the extremes of X, (the number of analyses per hour), the system Reliability and system MTBF are in accordance with table 8-10.

The minimum excess factor = 6.3 at 60 operations per hour

The maximum excess factor = 19.7 at 8.4 operations per hour

Therefore, the parallel circuitry approach will increase the MTBF of the analyzer as follows:

 $\frac{6.3 - 5.5}{5.5} = 14.5$  percent minimum

 $\frac{19.7 - 15}{15} = 31.3 \text{ percent maximum}$ 

The minimum and maximum increase in MTBF is therefore, 114.5 and 131.3 percent respectively.

	Derivation of	System Failure Rate	
Case -		II.	III
Number of X- Charnels that must operate	ŝ	4	ŝ
Basic Reliability	$e^{-[38, 843 + 2, 096X] \times 10^{-9}}$	$e^{-[40,752+2,155X] \times 10^{-9}}$	$e^{-\left[\frac{4}{2}, 660 + 2, 215X\right] \times 10^{-9}$
System X = 8.4	0. 999944	0.999942	0: 999939
$\mathbf{Rehability} \mathbf{X} = 60$	0, 999837	0. 999832	0. 999826
Badic MIBF		1	
Equation	$[233,069+12,573X] \times 10^{-7}$	$[244, 510+12, 932X] \times 10^{-7}$	$[255,960+13,291X] \times 10^{-7}$
System X = 8, 4	2953	2832	2720
MTBF (hours) X = 60	1013	980 ·	949
Factor of $X = 8.4$	19.7	18.9	18.1
Improve-   X = 60 ment over    X = 60	6.7	6.5	6.3
150 MTBF requirement			
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### 9. MAINTAINABILITY PLAN

### 9.1 Basic Philosophy

The basic philosophy of the Baird-Atomic maintainability program is to introduce within the design frame work concepts and techniques that will ensure ease of maintenance and operation within the limitations specified by the procuring activity.

### 9.2 Maintainability Organization

The organization structure of maintainability, and its immediate access to top management, is in accordance with figure 9-1.

It will be noted that reliability, maintainability, and safety engineering come under one head. Baird-Atomic takes the position that because of the interrelationship of these three groups, combining them will unify their effort and best serve the customer.

### 9.3 General Maintainability Requirements

The Baird-Atomic Maintainability Program requirements are as specified in the following documents:

a. AFSC Manual 80-5 (Ground Equipment)

b. AFSC Manual 80-6 (Ground Support Equipment)

c. MIL-M-26512C, except that maintainability demonstration shall be in accordance with paragraph 40.1 1 of MIL-M-26512B

d. MIL-STD-803 (Human Engineering)

In general the Maintainability Program will consist of the following basic elements:

a. Operational and Support Concepts and Requirements

b. Design Assistance, Review, and Analysis Frogram

c. Milestone Identification and Schedule

d. Maintainability Controls

e. Maintainability Demonstration

f. Records

r. Reporting





### 9.4 Operational and Support Concepts and Requirements

Baird-Atomic maintainability shall be responsible for the review and delineation of the import of the referenced manuals and specifications relative to the maintainability requirements of the Base Level Analyzer design. Specific design guides shall be made known to design engineering and monitored as to effective usage.

It is understood that the maintenance manpower goal is 4 percent of the operating hours of the equipment with the minimum operating hour requirement of the equipment defined as 3000 hours.

### 9.5 Design Assistance, Review, and Analysis Program

### 9.5.1 Design Assistance

Maintainability shall assist design engineering by specifying design guides, and monitoring their usage. The minimum design guide listing is as follows:

a. Every component or subassembly used in the proposed design shall, where possible, be MIL approved.

b. Every component or subassembly used in the proposed design shall, where possible, have a document of failure rate available to Baird-Atomic upon demand.

c. Every phase of the design shall be, where possible, in accordance with design and construction techniques as specified in MIL-T-21200D.

d. Each major portion of the equipment shall contain, where applicable, a suitable elapsed time indicator to provide appropriate data for maintenance records during the testing program.

e. Every failure in any type of development or final test shall be recorded including an analysis of cause, description of the defective part, and corrective action recommended.

f. Every major portion of the design shall be provided with test points in such a manner as to provide the means to locate a defective subassembly in the least possible time.

g. All parts and subassemblies shall be so arranged that they are easily accessible for testing and replacement.

h. Every consideration shall be given to keeping the cost of modules and subassemblies at a minimum to allow for the possibility of disposal at failure.



### 9.5.2 Design Review

The maintainability design review phase of the project will be in accordance with that specified in the Reliability Program Plan.

### 9.5.3 Maintainability Analysis

The maintainability analysis phase of the project will include the determination of the following:

a. Inherent mean and maximum downtime.

b. Inherent availability.

c. A schedule of all preventive maintenance tasks with an estimate of the time to accomplish such tasks computed on the basis of one man per task.

The basic work sheet used in accomplishing the maintainability analysis shall be in accordance with figure 9-2.

### 9.6 Milestone Identification and Schedule

The milestone identification and schedule shall be in accordance with that specified in figure 9-3.

### Maintainability Controls

9:7

In order to effect control of maintainability principles in the design of the analyzer, the maintainability organization will participate in all of the design reviews as organized by design engineering, and will be on the distribution list for all design changes.

### 9.8 Maintainability Demonstration

Maintainability demonstration shall be in accordance with paragraph 40.1.1 of MIL-M-26512B.

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Figure 9-2. Milestone Identification and Schedulc



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### 9.9 Records

The maintainability records to be established and maintained are as follows:

- a. Description of Changes Caused by Maintainability Recommendations
- b. Maintainability Design Recommendations Rejected or Trade-Offs
- c. Design Review Results and Current Status
- d. Progress and Deviation Reports
- e. Supporting Documents
- f. Documented Sign-Off of Design Specifications and Drawings

### 9.10 Reporting

Baird-Atomic Maintainability will submit a progress report each month that will be a part of the Contract Progress Report.

Each monthly report will contain a complete treatment of any trouble areas and emphasize corrective action required and/or completed.



### ). SAFETY ENGINEERING PLAN

### 10.1 Basic Philosophy

The basic philosophy of the Base Level Analyzer Safety Engineering Program is to specify the requirements and establish the responsibility for achieving a comprehensive system safety effort. Safety engineering is considered a prime effort.

The system safety effort shall be integrated with system design, development, manufac ure, test, checkout, installation and operation by the procuring agency.

Every effort shall be made to obtain the highest degree of inherent safety through the selection of appropriate design features, proven qualified components, and operating principles.

### 10.2 Salety Engineering Organization

The organization structure of safety engineering, and its immediate access to top management, is in accordance with figure 10-1.

It will be noted that safety, maintainability, and reliability engineering come under one head. Baird-Atomic takes the position that because of the interrelationship of these three groups, combining them will unify their effort and best serve the customer.

### 10:3 Safety Engineering Requirements

The Baird-Atomic Safety Engineering Program requirements are as specified in MIL-S-38130 (USAF), dated 30 September 1963.

In general, the Safety Engineering Program will consist of the following elements:

### a. Subsystem Safety Analysis.

b. System Safety Analysis

c. Salety Recommendations

d. Safety Procedures

e. Reporting





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V. Land, M. Ballel, and M. Mallah



### 10.4 Subsystem Safety Analysis

Each major subsystem of the analyzer shall be reviewed in order to identify all potential hazards and determine their criticalness. Hazardous failure modes shall be identified and probability of occurrence predicted on the basis of the reliability analysis of parts and subassemblies. As a result of this subsystem, safety analysis the system safety analysis will be performed.

### 10.5 System Safety Analysis

The system safety analysis utilizes the results from the subsystem safety analysis and develops these results as to their effect on the interface safety aspects of the total system. The combined system design is then reviewed in detail in order to define likely modes of failure including personnel error and the effects on system safety.

### 10.6 Safety Recommendations

Upon completion of the system safety analysis a review of the design precepts will be made in order to eliminate all classes of hazard exceeding class II.

Suitable safety and warning devices will be employed, where required, in order to effect class II as the maximum hazard condition.

Recommendation: shall include all aspects relative to personnel, procedures, and equipment for conducting maintenance, support, and test operations.

### 10.7 Safety Procedures

Safety Engineering will be responsible for initiating or review of all safety procedures in the operation, maintenance, or test of the Analyzer.

### 10.8 Reporting

At the end of the program a safety engineering report will be submitted relative to safety engineering aspects of the Analyzer.



### A.1 RESEARCH DIRECT READER SPECTROGRAPH (RDRS) DATA REDUCTION PROGRAM

### A.1.1 Object

Take RDRS data from analysis of standard samples and convert from concentration versus clock time to concentration versus capacitor voltage formatwith voltages normalized to a 10 volt charge on the reference capacitor.

A.1.2 Equations

$$V_{i} = 1.50 e^{\left[\frac{t_{cv} - t_{i}}{20}\right]}$$

where

 $V_i$  = voltage on capacitor for i<sup>th</sup> element

t = time in clock units require to discharge reference capacitor to 1.5 volts

t. = time in clock units required to discharge reference capacitor to i<sup>th</sup> element capacitor voltage

$$V_0 = 1.50 \text{ e} \left[\frac{t_{\rm cv}}{20}\right]$$

where

V = initial voltage on reference capacitor

The values obtained from equations (A-1) and (A-2) are scaled so that all values are normalized to  $V_0 = 10$  volts. All averaging operations and all computer data printouts are done subsequent to conversion from clock units to voltage and normalization.



## A.1.3 Computer Printout

Three numbers are printed by the computer for each concentration of each impurity type. These are the mean  $\overline{V}$ , the maximum difference  $V_{max} - V_{min}$  and the rms, v. These are obtained as follows

$$\overline{\mathbf{V}} = \frac{\mathbf{j} = 1}{\frac{\mathbf{N}}{\mathbf{N}}}$$
(A-3)

$$\sum_{v=\frac{j=1}{N}}^{N} (v_{j} - \overline{v})^{2}$$
(A-4)

On the output data sheets, (which follow) line 1 gives the average, line 2 the maximum minus the minimum, and line 3 the standard deviation.



7808/GB/BA Standard A, Normalized Output Data (Sheet 1 of 2)

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7808/GB/BA Standard A, Normalized Output Data (Sheet 2 of 2)

7808/GB/BA Standard A

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7808/GB/BA Standard A, Input Data, Concentration versus Clock Readings

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7808/GB/BA Standard A, Input Data, Concentration versus Clock Readings (Sheet 2 of 3)

Input Data 7808/GB/BA Standard A

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## Table A-2

7808 GB BA Standard A. Input Data, Concentration versus Clock Readings

(Sheet 3 of 3)

Input Data 7808/GB/BA Standard A

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		73.4	62.2	62.7	67-1			4781	72.5	67.2	58.8	30.0	
۰	-	-72.6	- 61.6		~ ****	*4:7			16.87	66.2	160.4	32.0	
		75.9	60.7	61.6	26.2	64 . 1	58.6	•		0002	50.A	31.7	
1	1.4	0.0	68.1	64.8	62.3	63.8	69.2	53.2			<i>,,,,</i> ,	34.4	· · · · ·
, <b>-</b>	••		60.7	67.9		66.7		52.2	0.0			37.6	
				68.8	65.4	64.1	69.3	52.3	78.8	82.9	-	32.8	1991 - Alexandre - Alexandr Alexandre - Alexandre -
					66.5	62.8	68.8	52.5	82.0	80.5	-61.6	32.3.	
		00.0				65.9	68.4	51.7	79.3	80.3	62.6	33.4	······································
		00.0	66.1				68.3	52.3	79.4	81.3	62.9	33.1	
		00.0	66.1	68.2				52.4	80.7	80.8	62.7	32.8	
`ı		00.0	65.6	69.0	69.0		alier and relation in graves, and		79.8	80.5	63.2	33.6	
1		00.00	67.0	68.5	67.6	65.0				79+7	62.5	33'14	
		00.0	66.4	69.6	66.7	65.8	67+1			<u>*</u> ^	61.1	33.2	
1	0A	00.0	70.1	66.0	67.7	67.7	73.9	55.3				35.0	
			68.8	72.6	67.6	66.9	73+4	54.3	00.0			-33+6	
<b></b>				70.6	63+6	64+2	73.2	53.9	81.4	87.8		33.8	
'		<u>.</u>		•	67+Z		7441	23.4	87.2	0.66	07.04	3361.	
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# 7808/GB/BA Standard B. Normanized Output Data (Sheet 1 of 2)

		7	808/G <b>8</b> /	'BA Stan	dard B			-						
<b>f - 1 - 1</b>	OIL TYPE	····				~				E	<b></b>		·····	
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'	5008	6.697	6.093	8.175	3.052	5.496	8.093	7.429	5.791	6.644	7.002	10.00	1.293	ar er - a maa sõõrsamedide
	SPREAD '	•951	1+/12 +625	3.976	•248	• <b>996</b> • <b>3</b> 70	1+841 •755	14,428 +388	1.322 •396	•506	• 764		-	
•, -	4008	5,813	5.485	7.105	2.460	5.258	7:439	6.393	4.882	6.205	5.804	10.00	1.373	······································
\$ 1	SPREAD	• 796	.380	3.487	•419	• 96 9 • 286	1.041 e473	1.391 .457	.282	•466	•471			
	3008	4.916	4.571	5.820	1.922	4.004	6.442	t. 9	3.665	5,375	4.522	1,0.00	1.339	
	- SPREAD-	•463 • <b>•180</b>	• 193 • <del>25</del> 2	2.4499	•429	•896 <b>292</b>	1.143	.::7 	1.081	767	.475			
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	2008	10.0 5+011	34409	-4.340	1+491		1.384		1.051	4.021		10.00	14643	· •• · · · · · ·
	SPREAD	•187	.189	•595	+116	.130	•462	+158	•330	•136	,140			
,; <i>"</i>	· 1008 -	-26481	2 - 258	-2-641		1.572	3.567	2.+32	1.+78	2.588	1.793	10.00	1.734"	
: : :	SPREAD	•362 •119	•324	1•536 •457	e 127 • 041	•239 •077	•883 •329	•400 •153	•216 •073	•754	•246 •081		•	
•	50B	1.603	1.414	1.584	•6,15	.938	2.133	1.648	.957	1,719	1.067	10.00	1.261	
		•239	•310	1.123	•198	.065	•645	•190	.108	•453	•139			• •



7808/GB/BA Standard B, Normalized Output Data (Sheet 2 of 2)

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7808/GB/BA Standard B SARE AD .105 .090 .312 .065 .039 .026 .033 .643 .207 .147 206 .915 .758 .796 .67? .546 .976 .456 1.265 1.207 .732 10.00 1.299 .145 .914 .288 .047 .409 .951 .577 .423 .892 .332 SPREAD .288 .295 .044 .085 . Ć15 .126 .300 .185 .133 .112 10 . 498 .535 ,361 .579 .406 .929 .344 .572 TJ.00 1.181 57 A 4 .736 .070 .301 .046 .230 .020 .135 .082 .046 .093 .025 .044 SPREAD .006 .024 .016 .023 •066 .090 .013 .029 .008 58 .251 .449 .433 .363 .492 .529 .829 .258 . 342 .523 10.00 1.242 .131 .081 .079 .052 .448 .063 • 053 .011 .074 .054 SPREAD .149 .026 .039 .024 .015 .023 .018 .004 .021 .019 18 .325 . 362 .384 .311 .710 .178 .000 .328 .173 .444 10.00 1.236 .047 .122 .071 .115 .000 .056 .023 .022 .038 .050 .900 .016 SPREAD .035 .022 .046 .008 .017 .008 .015 .014 - 0 .000 .318 .353 .247 .692 .150 .116 .337 .301 .419 10.00 1.299 000 .042 .075 .026 .051 .044 .096 .104 .013 .024 SPREAD .013 .028 •031 .023 .008 •016 .004 .008 .013. -. . ... ŗ • • • L A-1>



7-08 GB BA Standard B Input Data, Concentration versus Clock Readings (Sheet 1 of 3)

7808, GB/BA Standard B

1 7 • 7 8 • 0 7 • 6 7 • 7	2 * 13.9 8.5 10.8 8.5 14.2 11.7	3 2.4 1.1 12.0 4.0 2.4 3.6 6.0 14.4	4 25.8. 25.3 24.0 21.8 22.1 29.7 28.0	5 10•4 5•2 8•9 7•3 7•5	5 5.7 1.8 6.3 6.1 3.9 2.1 6.2	7 8+2 5+8 5+6 5+9 5+9 4+3 6+1	8 13.9 12.0 10.4 10.7 10.2 10.4 9.2	9 9+6 9+4 8+6 5+8 9+4	10 7.1 6.8 8.4 7.2	31 • 2 35 • 1 34 • 8 33 • 4 32 • 9 35 • 4 32 • 1	
7 • 7 8 • 4 7 • 6 7 • 7	13.9 8.9 10.8 8.5 14.2 11.7	2.4 1.1 12.0 4.0 2.4 3.6 6.0 14.4	25.6. 25.3 24.0 21.8 22.1 29.7 28.0	10•4 5•2 8•9 7•3 7•5	5 • 7 1 • 8 6 • 3 6 • 1 3 • 9 2 • 1 6 • 2	8 • 2 5 • 8 5 • 6 5 • 9 5 • 9 5 • 9 4 • 3 6 • 1	13.9 12.0 10.4 10.7 10.2 10.4 9.2	9•6 9•4 8•5 5•8 9•4	7•1 6•8 8•4 7•2	31.2 35.1 34.8 33.4 32.9 35.4 32.1	
7 • 7 8 • 9 7 • 6 7 • 6 7 • 6 7 • 6 7 • 6 7 • 7	8.5 20,8 8.5 14.2 11.7	1 • 1 12 • 0 4 • 0 2 • 4 3 • 6 6 • 0 14 • 4	<b>25.3</b> 24.0 21.8 22.1 29.7 28.0	5 • 2 8 • 9 7 • 3 7 • 5	1 •8 6 • 3 6 • 1 3 • 9 2 • 1 6 • 2	5 • 8 5 • 6 5 • 9 - 5 • 9 4 • 3 6 • 1	13.9 12.0 10.4 10.7 10.2 10.4 9.2	9+6 9+4 8+6 5+8 9+4	7•1 6•8 8•4 7•2	35 • 1 34 • 8 33 • 4 32 • 9 35 • 4 32 • 1	·
7 • 7 8 • 9 7 • 6 7 • 6 7 • 6 7 • 6 7 • 6 7 • 7 6 • 7	8.9 20,8 8.5 14.2 11.7	4.0 2.4 3.6 6.0 14.4	24.0 21.8 22.1 29.7 28.0	8•9 7•3 7•5	6 • 3 6 • 1 3 • 9 2 • 1 6 • 2	5•5 5•9 -5•9 4•3 6•1	12.0 10.4 10.7 10.2 10.4 9.2	9+6 9+4 8+6 5+8 9+4	7•1 6•8 8•4 7•2	34.8 33.4 32.9 35.4 32.1	• • • • • • • • • • • • • • • • • • •
7.7 8.9 7.6 7.6 7.6 7.6 7.6 7.6 7.6 7.6 7.6 7.5 9.9 9.9 9.9 0.0	8.9 10.8 8.5 14.2 11.7	4.0 2.4 3.6 6.0 14.4	21.8 22.1 29.7 28.0	7•3 7•5 14•5	6 • 1 3 • 9 2 • 1 6 • 2	5•9 -5•9 4•3 6•1	10.4 10.7 10.2 10.4 9.2	9•4 8•6 5•8 9•4	7+1 6+8 8+4 7+2	33•4 32•9 35•4 32•1	·
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8.5 7.6 7.6 7.7 7.7 7.6 7.7 7.6 7.6 7.6 7.6	8+9 10,8 8+5 14+2 11+7	4.0 2.4 3.6 6.0 14.4	22•1 29•7 28•0	14.5	<b>2•1</b> 6•2	4•3 6•1	10•2 10•4 9•2	5•8 9•4	8•4 7•2	35.4 32.1	
7.6 7.0 (.7 9.9 0.5 0.0	10,8 8,5 14,2 11,7	4 • 0 2 • 4 3 • 6 6 • 0 14 • 4	22•1 29•7 28•0	14.5	6•2	6•1	10•4 9•2	9.4	7.2	32.1	
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9.9 0.5 0.0	1	6.0 14.4	28.0	13.2		9.9				33.2	
9.9 0.5 0.0		14•4		1 J 🛛 C	5.6	7.3	10.6			33.6	•
9.9 1.5 0.0			2100	12.2	8.6	9.0	15.3	12.6		33.6	
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0.0				11.1	6.6	7.4	12.8	10.3	11.4	34+2	
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	11.6	6.7	27.3				13.4	9.8	11.4	32.7	•
1.1	13.1	7.1	-28.2	13.2				9.7	10.6	32.9	1
1.8	12.7	6.9	29.2	13.5	5.3				11.1	31.6	~
5.3	17.6	8.9	34.8	20.7	9.8	13.7				33.7	,
	14.3	9.2	33.6	17.1	7.3	12.8	21.5			34.3	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
		18.3	31.9	16.9	10.7	12.1	18.9	14.2		34.9	
			30.5	16.2	9.5	10.9	16.9	12.7	15.6	34.3	
1.7		<b>.</b> .		18.8	-7.6"	-13.5-	22.7	11.5	16.9	33.7	
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4.0	14.1	8.6				12.5	21.2	11.3	15.0	33.4	
3.4	16.3	11.0	32.8				19.2	• 13.6	15.8	31.5	
3.5	17.9	10.7	32.7	17.4				11.5	14.8	33.5	
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		2441	3703 NO O	2707	15.0	20.7	36.0	19.8	95.5	33.0	
n "			4047	2107	1200	20.0	20.3	10.0	24.3	32.3	
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7808/GB/BA Standard B, Input Data, Concentration versus Clock Readings (Sheet 3 of 3)

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1 18	73.9 73.9 73.5 74.4 00.0 00.0	62.9 61.4 61.0 60.7 67.4 65.9 66.8 67.6	64.4 62.9 62.8 62.5 64.3 69.1 68.6	68 • 1 63 • 7 65 • 9 67 • 4 68 • 8 69 • 4 66 • 7	62.7 63.44 66.8 65.9 61.8 61.7 66.8	57.9 69.7 63.8 69.8 69.8 69.8 69.5 68.4	<b>*9*5</b> 50*2 53*1 52*7 52*4 53*9 5?*3 52*7 53*1	72+9 72+8 73+6 80+8 78+8 80+5 81+4 80+1	53.3 67.4 68.3 66.6 53.3 82.8 79.8 79.7 83.8	59.8 60.2 58.1 58.8 59.6 62.1 63.6 61.8 61.3	32.8 33.3 30.6 31.6 33.5 33.8 33.5 33.2 34.1 34.9 33.0 32.6	
1 08		55.5 65.0 66.2 68.7 69.1	59.8 67.4 69.4 66.0 71.0 70.2	67•7 71•2 67•4 68•7 66•8 67•4	67•5 65•6 63•9 56•7 66•9	69•8 74•3 73•8 72•7	54•2 53•1 53•4	81+4 00+0 82+6	79•3 79•8 87•3	62•4 62•3 62•6	32+2 31+7 33+7 34+4 31+9 31+8	
,		67•4 67•2 67•5 68•2	70.6 70.3 72.1	73•8 68•7 68•2	67•7 68•0	73.4 74.8 74.3	53•4 54•0 52•7 53•1	53.9 84.2 84.4 84.4 84.4	88.7 91.1 91.5 58.9 87.6 88.2	63+4 64+5 62+9 62+4 63+3 64+1	32.9 34.7 31.8 31.8 32.4 34.4	2. 2. 2.
	-00.0.	66.6	. 70.5	. 59 6	6/+4	74+0				63+6	32+1	
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7808/GB/BA Standard C, Normalized Output Data (Sheet 1 of 2)

7808/GB/BA Standard C

OIL TYPE	-							DAT	E				-	, 
REF. CAP	VALU	Ē			MFD	•								
	CAP.	VALUE			MFD	•								
BREAKS P		FCYCL	E	10 2 2 10 10 10 10 10 10 10 10 10 10 10 10 10							**			
			-											-
Kr AMPS.	•- · ·											-		
ENTRANCE	SLIT				MICE.0	NS								
EXIT	SLIT	**** * ** ** **			MICRO	NS		• • • • • • • • • • •					•	······································
TEMPERATI	JRE													- 4-ar <del>man</del> anti-t-
CNCTRATN	MG	CR	Ai	PB	ST	AG	NI	FF	cu	SN	cv	DIVSR		
PPM	1	2	3	- 4	5	6	-7	8	<u> </u>	10				
500C	6.69()	6.429	8.509	3.140	6.508	8.662	7.067	5.903	6.696	7.231	10.00	1.286		
(	.846	1.837	4.081	.649	1.356	2.405	1.053	T.211	1.073	-896				
SPREAD	•314	• 288	1+419	• 236	•452	•841	•319	• 330	• 359	• 308				
4000	5.781	5.438	7.415	2.446	3.247	7.590	6.402	4.676	5,993	5.829	10.00	1.285		~
CODEAN	.306	<b>378</b>	3.576	• 406	• 696	1.396	•712	•877 •301	•906	•527				
SPREAU	• 300	• 1 6 0	1.011	• • • • •	• 2 1 0		1205		• • • • • • • • • • • • • • • • • • • •	• • / / /				
3000	4.823	4.602	5.915	1.870	4.029	6.500	5.198	3.664	5.097	4.525	10.00	1.230		
- CINDE TO	•671	.822	2.853	• 408	• 561	1.499	•669	•806	•799	• 560		<b></b>		
SPREAU	•203	4213	.011	****	120	+0/0		+23¥	0310	0 X (0 G			•	•
2000	3.897	3.534	4.367	1.314	2.744	5.283	3.957	2.586	4.078	3.138	10.00	1.339		
	•541	•500	2:343	• 197	•46T	1.544	•432	•728	.656					
SPREAD	•202	.139	•664	•076	•140	•552	•158	•251	•227	•098				
1000	2.491	2.268	2.618	.795	11505	3.349	2.374	1.361	2.710	1.744	10.00	1.199		·····
	.327	.462	1.257	.088	.090	1.032	. 351	.236	.597	.158				
SPREAD	.119	.150	.413	•033	.030	• 373	.120	•086	• 3 76	•053				
	1.634	1.454	1.561	. 604	1.000	2.237	1.657	.843	1.789	1.104	10.00	1.273		
<u>500</u>				=	10000	ニョニノリ		~ 0 - 7						



7808/GB/BA Standard C, Normalized Output Data (Sheet 2 of 2)

7808/G8/BA Standard C

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PREAD	.109		1199	-031	.062	+217	-099	•028	•117	- 040		
. 20C	•737	.703	•747	.440	.578	1.038	1.039	• 445	.802	•662	10.00	1.380
	.176	.192	.158	.048	.167	.248	.320	.095	.079	.046		
SPREAD	.052	•049	•049	•016	.057	.083	.103	.033	•024	•017		
100	.492	. 559	.574	.407	.519	.723	. 932	• 354	• 523	-578	10,00	1.373
	.054	.067	.295	.048	.199	•292	.110	•059	.107	• 046		
SPREAD	.021	.019	•082	.013	•071	•095	.034	•019	•038	•016		
5C	•251	.450	.449	• 363	.409	.483	.812	•248	• 322	.499	10.00	1.450
	•072	.121	.154	.069	÷022	,303	.109	•148	•230	.084		
SPREAD	.022	.038	.044	.020	.008	<b>•088</b>	.035	•041	•075	025		
10	.000	.369	.340	.345	.383	.307	•64î	.170	•172	.438	10.00	1.293
	.000	.027	.079	.109	.133	.062	.577	.032	.038	.033		ي د مومونين
SPREAD	.000	.009	.025	• 035	.048	•019	•196	•010	•012	.008		
<del></del>	000	. 334	.300	.326	.398	.290	+689	.134	•113	.430	10.00	1.319
	.000	.040	.116	.027	.100	.014	+062	•C12	. 325	+042		
SPREAD	.000	.013	.034	.007	.032	.005	.021	•004	•009	.012		

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7808, GB/BA Standard C, Input Data, Concentration versus Clock Readings (Sheet 1 of 3)

		7838/	GB/ BA S	standard	С							•	
						<del></del>			FE				
		1	2	3	4	5	6	7	8	9	10		
1	500C	9.9	12.1	1.6	25.6	11.4	4.2	7.1				32.9	
			5.3	0.3	24.0	8.2	0.3	4.3	12.6			35.4	
				11.0	23.3	8.2	5.9	4.6	10+6	9 . 1		35.4	
					21.4	8.3	3.8	5.4	10.7	9.6	7.9	33.5	
		7.3		·		7.1	7.8			8.2	6.4	- 32.9	
		8.1	8.6				0.8	4.9	11.0	6.4	6.9	35.2	
		7.7	9.2	2.7			•••	5.7	10.3	7.7	5.4	32.8	
		7.3	8.4	2.2	21.8				8.5	7.3	5.9	32.9	- •
,	100	11.1	13.0	2.8	28.0	12.0	67	9 8			207	23 9	
			11.6	5.1	20.0	12 7	<b>1</b>	0.7	14 0			22.0	
		-	110	761 - 17. T		3054 	••0 •••	786	1010	·		3380	
				1344	27.04	12.04	7 64	0.0	12+0	11+0	• • •	2442	-
		10.0			6107	1202	1.0	<b>7.</b> 0	10+1	1203	11+0	3203	
		10.0				11.02	4.5	01.0	14+3	09.3	00.3	34+0	
		11.1	12+3				3.8	09 • Z	15.7	09+4	11.1	32.8	
		10.3	12.0	05.8				09.1	14.3	09+2	10+8	32.4	
		29.5	.1.7	05.4	26+4				13.1	0965	09.8	32.3	
		12:5	12.5	05.6	29.4	13.8				10.3	10.9	33.4	
		12.4	12.02	5,3	29.7	13.4	5.0				11.1	32.9	
	3000	15.2	16.5	7.6	34+6	19.7	9.9	13.1				33.7.	
	-		14.7	8.5	32.8	17.4	6.3	12.5	- 20.5'			33.8	a human ana
				18.4	32.9	17.7	10.9	13.1	20.8	14.5		34.2	
					31.5	16.9	9.5	12.0	17.5	13.2	15.5	33.9	
-		13.9				18.2	10.7	13.5	20.3	14.7	17.2	32.8	
		14.2	14.9				6.4	12.9	21.3	11.7	13-6	33.8	•
		14.5	16.6	10.5				14.6	27.8	13.9	16.4	30.5	
		13	1818-	9.9	37.7				18.8	11.8	-15.4	37.8	
		16.1	.7.5	11.5	35.0	10.1			1000	14.9	14.1	21 0	
		15.0	13.0	A.A	34.6	10.4	7.2			1400	1001	33.6	
	Jean .	+99									1460		
	7000	10.00	20.4	12.00	43 0	24 4	1201	10.4	20 4			22 - 1	
			2000	10.1	41.07	2044	12.00	14+0	2780			22+0	
				2402	37+1	29.4	10.0	10+3	20.0	20+1	<b>.</b> .	34.4	
		•			- 39•3	23.3	1947	10.5	20.3	10.8	23.8	33.8	
		.1.5			•	24 . 4	12.0	17.8	22.42	17.0	23.4	34.1	
		18.3	19.2				10.3	17.6	27.6	16.8	22+4	33.8	
		19.3	ZZ + O					19.8	30.2	-18+5		-91.3-	
		17.6	20.9	16.6	39.9				24 • 5	16.9	22.6	32.2	
		21	21.1	16.7	42.1	26.6				17+7	2228	32.9	
		20.4	20+8	t4.5	4]•É	27.8	10+2				23+t	32.1	
	1 10	28.9	2.4	24.8	51.3	37.9	22.3	29.8				32.8	
			28.5	24.9	51.4	37.2	19.6	28.1	43.6			34.2	



7808/GB/BA Standard C, Input Data, Concentration versus Clock Readings (Sheet 2 of

7808/GB/BA Standard C

1.			36.7	49.4	37.7	24.8		-20.g	28.3		
			3044	49.5	37.7	19.9	29.2	40.3	26.3	35.7	33.1
. •	26.3	28.6				18.7	26.9	37.8	23.9	33.9	33.8
	28.1	30.2	25.9				29.2	41.2	26.5	34.5	31.0
· .	26.8	29.7	25.8	50.3				38.7	25.7	35.4	32.1
	28.6	30.4	27.1	51.6	38.4				20.2	34.8	32.7
	28.2	28.2	25.1	30.8	38.3	- 19.2	-			35.3	34.3
1 500	38.3	40.9	35.2	56.9	43.8	32.6	37.3				32.8
1		37.8	35.5	57.6	46.4	28.8	35.4	49.7			33.6
	-		44.2	56.0	46.5	33.7	35.7	49.5	37.1		34-4
-				54.7	46.6	29.6	35.9	49.8	35.9	44.9	33.2
	34.7				45.2	28.1	35.4	48.3	34.1	43.9	34.6
· · · · · · · · · · · · · · · · · · ·	36.2	- 37-6-	1			78.4	35.4	50.3	33.8	41.5	34.3
1	35.7	38.6	36.4	·			36.6	50.0	33.1	44.3	33.7
	35.2	39.5	37.8	54.9				48.7	33.6	44.1	31.8
· · · · · · · · · · · · · · · · · · ·	38	37.8	36.1	57.1	48.1				33.6	44.0	34.7
	35.6	37.7	36.1	55.8	45.9	29.1			2240	47.7	33.1
1 200	55.6	56.6	51.0	61.4	59.6	48.3	46.2			~ 2 • 1	31.8
,		57.8	50.8	63.2	58.8	43.8	44.4	64.4			31.4
		2240	55.3	67.4	58.1	46.4	43.7	62.0	50.8		34.9
•				62.8	55.4	46.6	44.7	62.0	51.6	55.1	34.9
				02.00	58.2	44.3	· 44 . 8	64.2	50.1	53.7	34.3
	51.6	52.0				44-8	44.0	60.8	50.3	54.8	33-4
	51.1	52.3	51.7				50.4	60.1	50.1	54.4	32.7
,	50.6	53.6	51.4	67.1					39.6	51.7	30.5
	52.4	52.7	51.5	63.6	55.7			010/	50.8	54.6	32.8
	51.8	52 3	51.8	61.6	53.9	43.4				53.8	31.5
	59.1	57.4	51.8	63.8	50.8	49.4	46.6			,,,,,	31.8
		58.7	57.8	64.3	61.4	51.3	47.5	66.7			32.5
			61.8	63.6	56.4	54.4	47.0	65.9	61.5		33.4
				64.2	-54-4	-54-5-	47.4	86.3			32.0
	59.7				60.5	57.0	47.4	65.4	58.2	57.9	35.1
	59.4	58.2			-	50.9	47.8	68.6	57.7	57.3	33.0
	59.9	57.6	5874	-			46.6	00.0	57.4	46.9	33.2
	61.2	57.6	57.6	65.2				67.7	58.2	57.6	32.8
	61.1	57.9	57.0	64.1	60.2				59.4	56.3	32.2
	-61-3	- 56.3	55 . I	- 67.8-	61.6	50.6				-56-3	31.5
) 5C	77.1	62.1	59.2	66.4	00.0	59.9	50.4				34.4
		61.5.	62.7	66.3	64.3	58.0	49.9	73.4			33.2
			66.3	63.9	63.4	50.3	50.2	69.8	68.7		33.4
				66.5	63.6	59.7	49.5	73.6	65.3	59.7	33.8
	72-5			400/	64-5	62.3	49.7	72.5	65-4	59-6	32.7
:				•							
ı.											



7808/GB/BA Standard C, Input Data, Concentration versus Clock Readings (Sheet 3 of 3)

	7808/0	GB/BA S	tandard	С							,	.#	
e	73.6 74.1	61.7 61.5	50.9 62.8	66.8	*****		52.5 49.5	69.1 73.5 73.7	68.6 69.1	59+8 59+3	33.8 31.6	*	
n un sin des angestes s	~- <del>73</del> •⁴ 71∙2	• <del>60</del> •6 60•7	61.8 61.3	67•6 66•8	64•1 63•7	56.3			65+4	60.3 58.8	33.9		* ****** =
1C	00.0	66.8	64.3	63.0	65.7	72.4	53.1				34.5		7 <u>.</u>
•			68.7	67.4	60.9	69.4	52.7	79.3	82.4		34.6		
· · · · · · · · · · · · · · · · · · ·		-	-	68.3	62.6	68.4	52+1	81.4	75.3	62.8	34-1		
	00.0	66.4			0000	73.4	52.3	83.1	83.8	62.03	33.2		•
	00.0	65.9	68.3				52.7	81.8	82.3	62.3	32.6		
	00.0	65.7	68.5	57.7				82.5	79.8	61.9	32.4	·	
	00.0	65.3	68.0	68.9	67.5	68.7			80+3	63.3	32.8		
1 00	- 00.0	68,5	65.5	68.4	62.6	73.9	54.1	-	- ~	04.03	34.2		
		69.0	70.0	67.6	67.6	73.8	54.0	00.0	•		33.7	-	
			70.0	68.7	67.6	73.5	54.4	84.0	91+6		34.4		
3	00.0			0734	67.1	74.4	53.3	82.6	91.7	94+1 52+8	33.4		
	00.0	68.1				73.4	53.0	82.9	88.1	62.8	33.1		
	··· 00.0	67.2	71.2				52.6	83.9	90.4	6Z.8	31.7	· ·	وريد اوي ، ماسيت من عد . و ع
	00.0	68.8	72.9	69+3	67.9			84•2	90+3 88+0	62+3	31.01 33.4		
arranna Briggenre velarar vanskadar 2003	0.00	65.6	71.3	58.5	56.9	74.2				62.8	32.4	*****	
					•								
			··· •										· ····
անու հեւրերեւտուդան է փինի պարեր։	•						<b>-</b>	(* - · * *					**************************************
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Figure A 21 Oil Type 7808 GB/BA Standard C.

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Figure A-23 Oil Type 7808 GB/BA Standard C



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	7803 GB/B	A Stand	dard B	and C									
-	OTL TYPE	,							DATE				
	REF. CAP.	VALUE		•	10	MFD.							
	X CHANNEL	CAP.	VALUE		01	MF D.			•				
·** .	BREAKS PE	R HALF	CYCLE										
	RE AMPS.				4								
•	ENTRANCE	-		-	50	MICRON	ç						
						WICTON	<b>.</b>						
	CALL	SELL			23	MICKON	3						
	TEMPERATU	RE								-		•	
	CNCTRATN										AL	c٧	DIVSR
	PPM		•										
	4003	•000	.000	.000	.000	.000	•000	•000	•000	•000	6.052	10.00	1.436
	SPREAD	•000 •000	•000 •000	.000 .000	.000 .000	.000 .000	000. 000	000. 000	000. 000	•000 •000	•689 •290		
	3008	.000	-000	-000		.000	-000	.000	•000	.000	5.398	10-00	1.458
	200,0	.000	.000	. 000	.000	.000	.000	.000	.000	•000	•243	10000	
	SPREAD	.000	•000	.000	•000	•000	•000	_•000	•000	•000	•104	···· 4·	************
	200B	.000	.000	.000	.000	.000	.000	.000	.002	• ^00	3.972	10.00	1.345
		.000	.000	.000	.000	.000	.000	.000	•000	.000	•159		
	SPREAD	.000	.000	•000	.000	.000	.000	.000	•000	.000	.067		a,
	1008	000	.000	000	000	000	000	000	000		2 200	10.00	1.450
	1008	. 500	-000	-000	-000 -000	.000	000	•000 •000	-000	•000	177	1000	10420
	SPREAD	.000	.000	.000	.000	.000	.000	.000	•000	• 000	• 380		
		•		-		-							
	50B	.000	.000	\$000	.000	.000	.000	.000	.000	.000	1.385	10.00	1.299
	( D D - · -	.000	.000	•000	•000	.000	.000	.000	•000	•000	•027		
	SPREAD	•000	•000	•000	•000	•000	•000	•000	•000	• 000	•013		*****
					<b>.</b> .								
	20 <b>B</b>	.000	•000	_000	<u>,000</u>	<b>.</b> 000	-000	<u></u>	<b>_</b> 000		.750	10.00	1,199

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7808/GB/BA Standards B and C, Normalized Output Data. (Sheet 2 of 3)

780 3 GB/BA Standard B and C

	SPREAD	.000	.000	.000	• 200	.000	.000	•000	.000	• ^ ^ O	• 721		
	108	.000	.000	.000	.000	.000	.000	.000	.000	.000	•553	10.00	1.242
		.000	.000	.000	.000	• <b>20</b> 0	.000	.010	•00C	.000	.013		
	SPREA)	.000	•000	.000	•000	.000	•000	•000	• <b>1</b> 00	• ^ 0 0 0	• 205		
	5B	.000	.000	.000	.000	.000	.000	.000	100	•000	.408	10.90	1.465
		.000	.000	.000	.000	.000	.000	.000	•U00	.000	•016		
· - <u>-</u> ·	SPREAD	.000	•000	•000	•000	•000	•000	•000	• 000	•000	• 207		
	18	.000	.000	.000	.000	.000	.000	• 000	•^^0		.288	10.00	1.236
_		.000	.000	.000	.000	.000	.000	. 000	.000	. 100	.005		_
	SPREAD	.000	.000	.000	.000	• 200	000	.000	•00c	•000	••••2	•	-
	08	•000	•000	.000	.000	.000	.000	.000	•000	•000	.250	10.00	1.273
		.000	000	,000	.000	.000	.000	.000	.000	.000	.029		
	SPREAD	000	.000	.000	.000	.000	.000	.000	•000	•000	.013		
	4000	.000	.000	.000	.000	.000	.000	.000	•000	.000	6.149	10.00	1.325
•		.000	.000	000ء	.000	.000	.000	.000	•000	•000	•403		
	SPREAD	.000	•000	.000	•000	• 000	•000	•000	•000	•000	.183		
	3000	.000	.000	.000	.000	.000	.000	.000	.000	.000	5.565	10.00	1.293
		.000	.000	.000	.000	.000	.000	.000	.000	• 200	.439		
	SPREAD	.000	•000	.000	,000	.000	.000	.000	•000	•000	.207		
	2000	.000	•000	.000	.000	•000	.000	.000	.000	.000	4.206	10.00	1.458
		•000°	.000	.000	.000	<ul> <li>309</li> </ul>	•000	•000	•000	• 100	• 360		
	SPREAD	•000	•000	.000	•000	• >>>	•000	•000	•000	•000	•157		
	1000	,000	,000	.000	.000	.000	.000	.000	.000	.000	2.489	10.00	1.386
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7808 GB/BA Standards B and C, Normalized Output Data (Sheet 3 of 3)

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7808 GB/BA Standards B and C Input Data, Concentration versus Clock Readings (Sheet 1 of 2)

Input Data 7808 GB/BA Standards B and C

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# 7808 GB/BA Standards B and C Input Data, Concentration versus Clock Readings (Sheet 2 of 2)

Input Data 7808 GB/BA Standards B and C

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# APPENDIX B

### ELECTROMAGNETIC INTERFERENCE

# TEST REPORT

# ON

# BAIRD-ATOMIC OIL ANALYZER

.

## ELECTROMAGNETIC INTERFERENCE

# TEST REPORT

# ON BAIRD ATOMIC OIL ANALYZER

NOVEMBER 24, 1964

### TEST REPORT NO. 248

TEST PERFORMED BY:

TEST PERFORMED FOR: BAIRD ATOMIC, INC. 95 SECOND AVENUE WALTHAM, MASS.

> INTERFERENCE CONTROL SANDERS ASSOCIATES, INC. 95 CANAL STREET NASHUA, N. H.

### PURPOSE

An evaluation to determine the interference level of the pil analyzer and suppress the level below MIL-I-26600 specification limit.

### PROCEDURE

Conducted and radiated measurements were performed in accordance with the procedure outlined in MIL-I-26600.

### TEST RESULTS

Broadband conducted measurements were performed on the unit as received. The levels, shown on Graph 1, varied from 108db uv/mc to 155db uv/mc : which is 5db to 70db above the specification limit. There was a 10db to 20db decrease when the gaps were altered as shown on Graph 2. The unit was still 65db above the specification limit at 4Mc. The analyzer was enclosed integrated container and interference filters were installed on the input power leads. The level dropped below the specification limit except at 0, 15Mc where it was 6db above the specification. Graph 3 shows this level.

Broadband radiated interference measurements were conducted on the oil analyzer as received. The interference level, shown on Graphs 4 and 5, varied from 16fdb uv/mc at .15Mc to 52db uv/mc at 10KMC, which is 30db to 80db above the specification limit. A copper screen was placed over the unit and grounded to the base (ground plane). This screen improved the level at the low end only as shown on Graphs 6 and 7. The coaxial loop was installed inside the container to determine its effect. No change was observed as shown on Graph 8. An aluminum shield was then placed over the coaxial line loop but no decrease in level was noted (Graph 9). A galvanized enclosure was placed over the copper enclosure. No basic improvement was observed except for a slight variation at the low frequency end.

**Probing** indicated that the leakage radiated from the prime power leads and from the base of the shield. Line filters were installed on the prime power leads. This decreased the level across the whole frequency range. The unit was still 0db to 20db above the specification limit above 30Mc as shown on Graphs 11 and 12.

#### SUPPRESSION

The oil analyzer was enclosed by a copper screen furnished by Saird Atomic. A gaivanined container also furnished by Baird Atomic was placed over the copper screen. A better seal at the base of the container was necessary to decrease the interference level. The power leads were filtered with a Sprague JN17-989A filter to stop power line radiation.

### CONCLUSIONS

Broadhand conducted interference measurements were conducted on the oil analyzer as received. The levels varied from 108db uv/mc to 155db uv/mc (50b to 70db above the specification limit. Except for one point, the level dropped below the specification when the analyser was addiesed in a galvanized container and suppression filters were placed on the input power leads. This point occurred at 0.15Mc where it was 5db above the specification limit.

Broadband registed interference measurements were conducted on the oil ensivers as received. The levels varied from 165db w/mc at 15Mc to 52db ur/mc at 10KMC (30db to 80db above the specification limit). A 30db to 50db improvement was observed at the low frequency end when a copper screen and a galvanised screen was placed over the oil analyzer. Probing indicated that the leakage radiated from the prime power leads and from the base of the chield. Line filters were installed and the radiation remeasured. The level was within specification up to 30Ms and was out of specification by 6db to 20db shove 39Mc.

It was felt that tightening the base of the enclosure and the spening for the air will decrease the radiation. Fiftering the power leads with a Spragur JN17-989A filter will further decrease the interference level.

Test Engineer



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