CARDE T.N. 1796/68

67918

PROJECT 046-95-51-10

EXCHANGE DOCUMENT 26 9CXR

VELOCITY DECAY IN THE INTERMEDIATE WAKE REGION BEHIND HYPERSONIC SPHERES

200 335





CANADIAN ARMAMENT RESEARCH AND DEVELOPMENT ESTABLISHMENT CENTRE CANADIEN DE RECHERCHES ET PERFECTIONNEMENT DES ARMES

DEFENCE RESEARCH BOARD

22A

NCLOSURE REFERS TO

CONSEIL DE RECHERCHES POUR LA DEFENSE



	(All and a state of the		7883. GA	1	
CRSTI	WRITE SECTI	on of	AL	< 11	NIA
CHG Prachometer	BUFF SEGTI		1 de	$\lambda \mu$	
STELCATIO		A	Pill	1.1	
		(vin	apr	29 A	
	and state of the second se	XUU	1		
OISTS BUTTON	AVAILAND	VF			
DIST					
	TAIL STAT BY	ELIAL		MERT .	13.5
1		39 B.C.M	4. Santa - A		
Contraction of the second s	THE CONTRACT OF A DATE OF	PLACE DALL DOLL	ALCOLAND Z.		the maked on the second

NOTE

5

CARDE Technical Notes are prepared for timely reporting of various types of information which may be tentative or in a form which does not justify a Technical Report.

meaningless since ARPA has cleared paper open puble with CARDE concurrence. SECURITY CAUTION nf ith he express understanding that mai 0 18rights will be rietar paten protected. nation wit o anothe specif proval A Nation etence.

VELOCITY DECAY IN THE INTERMEDIATE WAKE REGION

BEHIND HYPERSONIC SPHERES

by

C. Lahaye*and D. Heckman

*Research Scientist, Computing Devices of Canada Ltd., attached to Aerophysics Division.

Distribution of this document is unlimited Distribution of this d

This research was sponsored jointly by

The Canadian Armament Research and Development Establishment P.O. Box 1427, Québec, Que. Carada Under Projects: D46-95-51-10 D46-99-10-35

The Advanced Research Projects Agency as part of Project DEFENDER ARPA Order 133 Monitored by the US Army Missile C. mmand, Redstone Arsenal, Alabama 35809 Contract DA-01-021-AMC-14468(Z)

CANADIAN ARMAMENT RESEARCH AND DEVELOPMENT ESTABLISHMENT

Québec, Que.

May, 1968.

ABSTRACT

Axial velocity decay data for hypersonic sphere wakes obtained with the sequential spark and probe array techniques have been fitted with an equation of the form $V_W/V_{\infty} = C_1 (X/D)^{C_2}$. This power law is found to fit the data very well for axial distances between 300 and 1000 body diameters behind the body. Values of C_1 and C_2 obtained from sequential spark measurements are given as a function of radial distance from the wake axis. Near the axis, the absolute value of C_2 is slightly greater than unity, decreasing to about one-third at radial distances of about two body diameters.

TABLE OF CONTENTS

																										Page No.
ABSTRACT	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	٠	•	•	•	•	•	•	•	•	•	i
DISCUSSIONS	•	•	•	•	•	•	•	•	•	٠	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	1
REFERENCES .	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	2
FIGURES 1 to	5	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	4 - 8

It has been shown by Townsend (1) that in the case of the incompressible "self-preserving" axisymmetric turbulent wake, the growth of the viscous core varies as the 1/3 power of the distance behind the body, while the decay of the wake velocity varies as the minus 2/3 power of the distance. For the hypersonic turbulent wake, there is a large amount of wake growth data (2,3) existent to show that the 1/3 power law is obeyed over a range of axial distance from 100 to 10,000 body diameters behind a body. Until recently, velocity data has not been available on which to test the minus 2/3 power law in the hypersonic case, where the effect of large density variations might be expected to cause departures from such a law.

Experiments using CARDE hypersonic range facilities are now in progress to measure both velocity and mean density profiles across the turbulent wake (4,5). Consideration here will be restricted to velocity data. Most of this data has been obtained using the sequential spark experiment which measures the velocity profile across the wake at given axial distances behind the projectile.

The sequential spark technique, which is based on stereo measurement of the displacement history of an illuminated ionized path across the wake, has been described in recent publications (4,6). Considerable data has been generated for spheres over a range of pressures at hypersonic velocities. In addition measurements of wake velocity have been obtained at CARDE using arrays of electrostatic probes (7) and this data is in good agreement with the sequential spark data (8). The purpose of this note is to present preliminary results regarding the decay of the hypersonic turbulent wake velocity behind spheres as a function of axial distance downstream and of radial distance from the wake axis.

The velocity data obtained from measurements in a large number of firings at various axial distances (X/D in body diameters) has been separated according to various values of radial distance (R/D in body diameters) from the wake axis. For each value of R/D, the pertinent data has been fitted by the least mean squares method on an equation of the form

 $v_{v}/v_{\infty} - c_{1} (x/D)^{c_{2}}$

where Vw/V_{∞} is the normalized wake velocity and C_1 and C_2 are constants for a given R/D.

Figure 1 gives an example of such a power law fit of sequential spark data at a radial distance of 0.7 B.D. (body diameter). The spark data (open points) pertains to 1.0 inch diameter sphere firings at 40 torr (P_{co} D = 100 torr cm) at velocities of 12,000 to 15,000 ft/sec and is concentrated at axial distances of 300, 600 and 1,000 B.D. behind the projectile. Also shown for comparison are velocity results obtained with arrays of probes (solid points) using 2.7 inch diameter sphere firings at 20 torr (P_{co} D = 135 torr cm) over a range of velocity from 14,000 to 15,000 ft/sec. The agreement between the two sets of results is very

good. Figures 2 and 3 show similar comparisons at values of R/D of 0.9 and 1.2 B.D. respectively.

In Figures 4 and 5, the values of C1 and C2 obtained from least mean square fits of sequential spark data are plotted as functions of R/D. Because of the large "catter in the data, which masks a weak dependence on pressure and a weaker dependence on Mach number, it was desirable to lump as much of the available data as possible to increase statistical accuracy. The crosses on the figures indicate data obtained from rounds at 20, 27, 40 and 76 torr, with velocities from 12,000 to 15,000 ft/sec. (Assuming a linear dependence of wake velocity on ambient pressure, the distributions of points are reasonably balanced, giving an average pressure of 40 torr.) These results may be compared with the open circle points, in which only data obtained at 40 torr is used. In the case of the "lumped" data, the number of data points is roughly 20 at 300 B.D., 10 at 600 B.D. and 40 at 1000 B.D. In the 40 torr case there are about 14 points at 300 B.D., 8 at 600 B.D. and only 4 at 1000 B.D. It is possible that these differences in the distribution of data points between the two cases may affect the slopes obtained from the power law fits, since the sequential spark data indicates a curvature departing somewhat from the power law fit at small R/D (Figures 1 and 2). However, despite the differences, both cases show the same trend.

Considering the exponent C_2 , it may be seen that the velocity at small R/D decays approximately as the power of minus unity, while weaker values of the exponent occur at larger radial distances. Plotting the velocity decay at constant values of radial distance normalized to the wake width would result in higher absolute values of C_2 . The values of C_1 in the two cases show comparable trends, except for the apparent deviation at small values of radial distance. The solid black circles on Figures 4 and 5 are values of C_1 and C_2 obtained from the electrostatic probe array velocity data shown on Figures 1 - 3.

The most important result of this work is that the wake velocity at intermediate values of axial distance (300 - 1000 B.D.) behind spherical projectiles appears to decay with a power law dependence on axial distance. The value of the exponent varies smoothly between values of the order of minus unity near the wake axis and about minus 1/3 at radial distances of about 2.4 body diameters.

- 1. Townsend, A.A. "The Structure of Turbulent Shear Flow," (Cambridge University Press, Cambridge, England, 1956), Chapter VII, pp 169-171.
- Lyons, W.C., Brady, J.J. and Levensteins, Z.J. "Hypersonic Drag, Stability, and Wake Data for Cones and Spheres," AIAA Journal, Vol 2, No.11, pp 1948-1956, November 1964.
- 3. Clay, W.G., Labitt, M. and Slattery, R.E. "Measured Transition from Laminar to Turbulent Flow and Subsequent Growth of Turbulent Wakes," AIAA Journal, Vol 3, No.5, pp 837-841, May 1965.

- 4. Lahaye, C., Léger, E.G. and Lemay, A. "Wake Velocity Measurements Using a Sequence of Sparks," AIAA Journal Vol.5, No.12, pp 2274-76, December 1967.
- 5. Tardif, L. and Dionne, J.G.G. "Density Distribution in Turbulent and Laminar Wakes," Submitted to the AIAA.
- 6. Lahaye, C., Léger, E.G. and Lemay, A. "Radial and Axial Velocity Profiles of Hypersonic and Supersonic Wakes Measured by the Sequential Spark Method," Conference Proceeding No.19 of the AGARD Specialists Meeting on the Fluid Physics of Hypersonic Wakes, Colorado University, May, 1967.
- 7. Kirkpatrick, A., Heckman, D. and Cantin, A. "Wake Plasma Turbulence Study Using an Electrostatic Probe Array," AIAA Journal Vol.5, No.8, pp 1494-95, Aug 1967.
- 8. Heckman, D., Cantin, A., Emond, A. and Kirkpatrick, A. "Convection Velocity Measurements in Hypersonic Sphere Wakes." AIAA Journal, Vol.6, (To be published).





FIGURE 1 - Power law fits of sequential spark and electrostatic probe array wake velocity data at R/D = 0.7. The spark data is from 1 inch diameter sphere firings at 40 torr and velocities from 12000 to 15000 feet/sec while the probe data is from 2.7 inch sphere firings at 20 torr and velocities from 14000 to 15000 feet/sec.



FIGURE 2 - Power law fits of sequential sparks and electrostatic prote array wake velocity data at R/D = 0.9. The spark data is from 7 inch diameter sphere firings at 40 torr and velocities from 12000 to 15000 feet/sec while the probe data is from 2.7 inch sphere firings at 20 torr and velocities from 14000 to 15000 feet/sec.



FIGURE 3 - Power law fits of sequential sparks and electrostatic probe array wake velocity data at R/D = 1.2. The spark dats is from 1 inch diameter sphere firings at 40 torr and velocities from 12000 to 15000 feet/sec while the probe data is from 2.7 inch sphere firings at 20 torr and velocities from 14000 to 15000 feet/sec.



FIGURE 4 - Constant C1 as a Function of R/D



FIGURE 5 - Value of Exponent C_2 as a Function of R/D

UNCLASSIFIED		£.
DOCUMENT CONT	ROL DATA · R & D	
(Security elevelification of title, body at abetravi and indexing 1. ORIGINATING ACTIVITY (Corporate suffer)	ennotation must be entered when the everell report to classified)	
	Unclassified	
Canadian Armament Research and Developmen	nt standard	
A REPORT TITLE	shment N/A	
• Velocity Decay in the Intermediate Wake H	Region Behind Hypersonic Spheres	
4. DESCRIPTIVE NOTES (Type of report and inclusive dates)		
Technical Note s. AutHoRisi (First name, middle Inilial, last name)		
C. Lahaye - D. Heckman		
6. REPORT DATE	74. TOTAL NO. OF PAGES 75. NO. OF REFS	
HAFCH 1700	0 0	
ARPA Urder 133 in conjunction with Contract DA-01-021-AMC A. PROJECT NO. 14468(Z)	CARDE T.N. 1796/68	
040-9 <u>7-51-10</u> •.	OD. STHEN REPORT HOIS (Any other manbers that may be seeing	nod
	and report)	
10. DISTRIBUTION STATEMENT	N/A	
Distribution Unlimited		
Joint Canada/US Research Program sponsored by CARDE and ARPA	Washington, D.C U.S. Army Missile Command, Redstone Arsenal, Alabama, 35809	•
Axial velocity decay data the sequential spark and probe are equation of the form Ww/Voo = C1 data very well for axial distances the body. Values of C1 and C2 obd are given as a function of radial the absolute value of C2 is slight one-third at radial distances of a N Lubw N Luby	for hypersonic sphere wakes obtained wit ray techniques have been fitted with an $(X/D)^{C_2}$. This power law is found to fit s between 300 and 1000 body diameters beh tained from sequential spark measurements distance from the wake axis. Near the a tly greater than unity, decreasing to abo about two body diameters.	h the ind xis, nut
DD 1473	UNCLASSIFIED	