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1. The type A0009 20101 altimeter, enclosure (1), figure 1, was installed in P-3A airplane BuNo 148889, TF-9J airplane BuNo 142441, and F-8D airplane BuNo 147044. The type B0001 22107 altimeter, enclosure (1), figure 2, was installed in F-4A airplane BuNo 146819. The altimeters were similar in operation and presentation. Scales, numeral size and pointer size were identical. The B0001 altimeter was integrally lighted and incorporated a counter-type barometric setting scale; the A0009 altimeter was post lighted and contained the window and dial-type barometric setting scale.

2. The A0009 and B0001 altimeters were evaluated as follows:

Altimeter	<u>Acft</u>	No. of Pilots	Flights	Flight <u>Hours</u>	Instrument Hours
80009	P-3A	16	20	183.1	57,5
	TF-9J	3	5	8.3	5.2
	F-8D	4	8	16.0	2.2
B0001	F-4 A	6	12	20.8	5 JUN 1 1 1968
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ST32-23R-63

3. Test flights included climbs, dives, penetrations, pullups, instrument approaches, radar intercepts, and aerobatic maneuvers in smooth and turbulent air in an altitude range of 0 to 50,000 ft, an airspeed range of 0 to 690 KIAS, and a Mach range of 0 to 2.0 IMN.

4. Evaluation of the A0009 and B0001 altimeters indicated that:

a. Readability (size and clarity of pointer and numerals) was satisfactory; however, the pointer hand partially obscured the one-thousand- and ten-thousand-foct digits when positioned within the 650- to 800-ft range as shown in enclosure (1), figure 3.

b. Interpretability of the altimeters was unsatisfactory; reading errors of one thousand feet and ten thousand feet were frequent. Portions of as many as four digits were visible on the numeral drum at some altitudes (enclosure (1), figure 4) making interpretation difficult and significantly increasing comprehension time. Movement of the thousand-foot drum, as a direct function of pointer displacement, facilitated the monitoring of altitude change and interpretation of indicated altitude. The drum feature was desirable, but the presentation of independently rotating digits for the multiple digit readout was unsatisfactory.

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c. Barometric pressure-setting provisions were satisfactory on both altimeters; however, the direct digital readout type provided in the B0001 altimeter was distinctly superior.

d. Both integral lighting of the BOOOl altimates and post lighting of the AOOO9 altimates were satisfactory.

5. The BOOOI altimeter, installed in F-4A BuNo 146819, was subjected to four field carrier arrestments to airplane limit load factor and three field catapult launches to within 0.7 g of airplane limit load factor (6.1 g). Results of these tests were satisfactory.

6. No malfunctions developed on either altimeter during the course of the evaluation.

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7. Installation time and procedures for the A0009 altimeter in the P-3A, TF-9J and F-8D airplanes were comparable to currently used altimeters. Installation of the B0001 altimeter in the F-4A airplane was identical in procedure except a pigtail on the airplane wiring was required to adapt the lighting plugs to the indicator. The pigtail was constructed of three ten inch leads, one Amphenol plug #AN3106A-105L-35-(c) and one Amphenol plug #AN3100A-105L-3P. Installation time required for the B0001 altimeter was 0.5 man-days.

8. Counter-pointer altimeters presently in use characteristically exhibit pointer sticking or hang-up tendencies during one-thousand-foot altitude transitions. The A0009 and B0001 altimeters showed no sticking tendencies; pointer transit was smooth.

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ST32-23R-63 RAV09P003 27

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FIGURE 1

ASTEK LARGE NUMERAL DRUM-POINTER ALTIMETER TYPE A0009 20101

> PAGE 1 OF 4 ENCLOSURE (1)

ST32-23R-63 RAV09P003 27

FIGURE 2

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ASTEK LARGE NUMERAL DRUM-POINTER ALTIMETER TYPE B0001 22107

> PAGE 2 OF 4 ENCLOSURE (1)

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FIGURE 3

ASTEK TYPE A0009 20101 DRUM-POINTER ALTIMETER SHOWING PARTIAL OBSCURATION OF NUMERAL DRUM DIGITS

> PAGE 3 OF 4 ENCLOSURE (1)

ST32-23R-63 RAV09P003 27

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FIGURE 4

ASTEK TYPE A0009 20101 DRUM-POINTER ALTIMETER SHOWING MULTIPLE PORTIONS OF NUMERAL DRUM DIGITS

PAGE 4 OF 4 ENCLOSURE (1)