

### **DEFENSE DOCUMENTATION CENTER**

FOR SCIENTIFIC AND TECHNICAL INFORMATION

CAMERON STATION. ALEXANDRIA. VIRGINIA



## UNCLASSIFIED

NOTICE: When a remnant or other drawings, specifications or other data are used for any 1 mpose other than in connection with a definitely related government procurument operation, the U.S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other fate is not to be regarded by implication or othervise as in any memor licensing the holder or any other person or corporation, or conveying any rights or permission to manifecture, use or sell any patented invention that why in any way be related thereto.

- 27

\$

ikt of the second second

19

.0

1



#### A VARIATIONAL SOLUTION OF SOLID AND FREE-FLOODING CYLINDRICAL SOURD RADIATORS OF FINITE LENGTH

. لە

:

•

Miguel C. Junger

1 March 1964

Technical Report U-177-48 Prepared for Office of Naval Research Acoustics Programs - Code 468 Contract Hoar-2739(00) Task NR 185-301

Reproduction in Whole or in Part is Permitted for any Purpose of the United States Government

CANERIDGE ACCUSTICAL ASSOCIATES, INC. 129 Mount Auburn Street Cambridge, Malsachusetts 02138

N 6

#### ACKROWLEDGHESP

Ţ

An analytical study such as the one here presented is of practical value only if it can be readily used to obtain numerical results. This analysis escapes the fate of being a mere mathematical exercise because Dr. Joshua B. Greenspon, of J G Engineering Research Associates, Ealtimore, Maryland, has applied to this analysis his wast experience in evaluating the complicated integrals characteristic of radiation problems in cylinarical coordinates. These quantitative results will be presented by Dr. Greenspon in a compenion report: "Axially Symmetric Green's Functions for Cylinders." Drs. Alexander Silb'ger and Ewald G. Eichler of this firm contributed useful comments.

#### ABBITRACT

The radiation impedance of a cylimitrical sound source of finite length can be expressed as the sum of two components:

 $Z = Z_r + Z_{\alpha}$ 

where  $Z_{T}$  is the impedance evaluated by means of the generaty used technique originated by Robey, which assumes that the radiating surface is bracketed between two rigid semi-infinite cylindrical baffles.  $Z_{T}$  is associated with the radial velocity distribution  $\alpha(z)$  over these two semi-infinite cylindrical surfaces and is therefore in the Lature of a correction factor to Robey's impedance.  $\alpha(z)$  is an mknown function which satisfies a non-homogeneous Fredholm integral equation. A functional  $J[\alpha]$ is contructed which is staticnery and proportional to  $Z_{tr}$  for the correct solution  $\alpha(z)$ :

$$\delta J[\alpha]/\delta \alpha = 0, \ Z_{\alpha} \propto J[\alpha]$$

From this variational principle a value of  $Z_{\rm C}$  is calculated by means of a Rayleigh-Ritz-type procedure. Finally, the far field is evaluated. The variational principle used here parallels the Levine-Schwinger principle widely used to obtain scattering cross sections. Variational solutions are presented for solid and free-flocding cylinders for axisymmetric and for arbitrary velocity distributions. A variational solution ic given for "squirters" of finite wall thickness, but it is ret ricted to thin walled transducers. In an Appendix, non-variational solutions of the integral equation for  $\alpha(z)$  are presented for "squirterr" of greater wall thickness.

In a companion report, the procedures developed here are applied to the evaluation of the self- and mutual-radiation impedances of elements in an array of coexial, free-flooding axially spaced ring transducers. Dr. J. E. Greenspon, of J G Engineering Research Associates, evaluated the inverse Fourier transforms required for these solutions and obtained quantitative results which he will present in a separate report.

11

and the second

#### TABLE OF CONTENTS

Page

ł

Acknowled	dgment	1
Abstract		ii
List of 8	Symbols	۷
List of !	Tables	ii
List of 1	Figures	<b>i</b> 1
I S	cope of Study	1
II A C	Review of Published Analytical Approaches to the Finite	1
III D	escription of the Present Amproach	4
IV I	integral Soundion Formulation of the Solid Cylinder Problem	6
ה ע	erivation of the Variational Principle for the Radiation	•
Ŀ		9
VI Br	valuation of the Radiation Impedance from the Variational rinciple	13
VII T	he Far Field Potentials	18
VIII 2	he Open-Ended Free-Flooding ., lindrical Radiator of Vanishing	
We	all Thickness	21
IX T	he Free-Flooding Cylindricel Transducer or "Squirter"	53
X Cy	ylinders Vibrating in Longitudinal and in Non-Axisymmetric Medes	29
Apperdix	A: Derivation and Evaluation of the Green's Function G, for	
	the Cylindrical Region r < a	33
	1. Construction of the Green's Austion	33
	2. Evaluation of the Laverse Fourier Transform	34
Appendix	B: Hon-Variational Techniques for Solving the "Squirter" Integral Equation, Eq. IX.11	39
Reference	68	.7

iv

a. . .

	Symbols"
	(Alternative subscripts, viz $a_{0}$ , are used to condense two equations
	into one, the upper subscript on the left side of the equation being associated with the upper signs and subscripts on the right side of the equation, and vice versa.)
8	radius of cylindrical scurce (Fig. 2); mean radius of "squirter" (Fig. 3)
a, ,a	inner and outer radius of "squirter," respectively (Fig. 3)
1'0 c	sound velocity in fluid medium
€ <sub>i</sub> ,G <sub>o</sub> ,	$g_{1}, g_{0}, G_{1+}, G_{1-}, (z-z')$ , and $H(z)$ Green's functions and related functions defined in table 1, p. 1
H	Hankel function of the first kind, of order a (with this notation, a massive reactince is negative)
h	half thickness of "squirter" (Fig. 3)
J	Bessel function of order m
k	wave number, equal to w/c
×,	radial wave number, equal to $(k^2 k_z^2)^{\frac{1}{2}}$
ĸ,	axial wave number
L	half length of cylindrical radiator (Figs. 2 and 3)
P	sound pressure
r,o,z	cylindrical coordinates
R,0	spherical coordinates
U	radial velocity amplitude of cylindricel source (Fig. 3)
u(z)	radial velocity distribution on cylindrical surface $(r=a)$ , positive outward (Fig. 3)
Z	radiation impedance of cylindrical source in units of force/velocity equal to $(Z_r + Z_\alpha)$ (Fig. 2)
7.	reliation impedance obtained from Robey's model (Fig. 1b) for which $\alpha(z) = 0$
²α	correction factor associate with verticity distribution $\alpha(z)$ and to be udded to $Z_r$ (Fig. 2)
a(z)	radial velocity in the regions $ z  > L$ (Fig. 2) normalized to velocity amplitude U of cylindrical surface
υ	Poisson's ratio of transducer material
٥	density of fluid medium

- $\label{eq:velocity potential (outward velocity is -d //dr); subscripts "i" and "o" refer respectively to regions <math>r \leq s$  and  $r \geq s$
- a circular frequency [harmonic time dependence factor exp(-imt) which multiplies the velocities and the potentials, has been suppressed throughout this report]

2.8

"Other symbols are defined in the text.

î

1.5

è

Table 1

5

C

TI.

~

1

#### GREEN'S FUNCTIONS AND FRIATED FUNCTIONS

Symbol	Equation	Region where Appli- cable	Sound Source Configuration to Laich Applicable	Description
G <sub>0</sub> (r,a,z-z')	IV.1	r > _	All configurations with axisymmetric velocity distribution	General form of
$\overline{G_i(r,a,z-z')}$	IA.5	r < a	Free-flooding cylinder with axisymmetric velocity distribution	axisymmetric Green's function
g <sub>0</sub> (r,s,z-z')	IV.6	r > a	All configurations with axisymmetric velocity dis- tribution symmetrical about z = 0	Even component of G <sub>O</sub>
g <sub>1</sub> (r,a,z-z')	VIII.2	r < a	Free-flooding cylinder with axisymmetric velocity dis- tributival symmetrical about 2 = C	Even component of G
$G_{i+}(r,a;z-z')$	IV.3 and 4	r < a z > L	Solid cylinder with axisymmetric velocity distribution	Green's function whose normal de- rivative $\partial/\partial z$ vanishes cu end cap $z = L$
$\overline{G_{i-\binom{r}{2-z}}}$	IV.3 and 4	r < e z < L		Green's function whose normal de- rivative $\partial/\partial z$ van- ishes on end cap z = -L
$\frac{\overline{G_{om}(r, i)}}{z-z'}$	X.4	r>a	Cylinder with arbitrary velocity distribution ex-	General form of non-axisymmetric
$G_{in}(r,s; \phi-\phi; z-z')$	<b>~.</b> 6	r < a	in $\varphi$ (Eq. X.?)	
Γ(z-z')	IV.11	r=a	Solid cylinder	
	VIII.4a and VIII.5	r = 8	Free-flooding cylinder of vanishing wall thickness	Linear combination of Green's functions
	IX.14a	r=8	"Squirter"	
[](z)	IV.8	L	Solid cylinder	Į
	VIII.4b		Free-flooding cylinder of vanishing wall thickness	Integral of Green's function over
	IX.14b		"Squirter"	ractaving surface

vi

#### LIST OF TABLES

٤.

1.	GREEN'S FUNCTIONS AND RELATED FUNCTIONS	
2.	COMPARISON OF VARIOUS ANALYTICAL APPROACHES TO THE CYLINDRICAL SOURCE OF FINITE LENGTH	

Page

Page

#### LIST OF FIGURES

# 1. REVIEW OF PUBLISHED ANALYSES OF CYLINDRICA, RADIATORS. 43 2. SUMMARY OF PRESENT APPROACH. 44 3. GEOMETRY OF "SQUINTER" 45 4. CONTOUR INTEGRATION OF THE INVERSE TRANSFORM OF THE GREET'S FUNCTION G1. 46

vii

#### I. Scope of Study

In this report a variational technique is used to derive expressions for the radiation loading of radially pulsating cylinders of <u>finite length</u>. End effects are accounted for, no restrictions being placed on the circulation of the acoustic fluid around the edges of the cylinder. The far field potentials, on the axis of the cylindrical radiator and for other bearings are also given. The analysis is performed for (1) a solid cylinder, (2) an open-ended, free-rlooding cylinder of vanishing wall thickness, and (3) a "squirter" of small, but non-vanishing thickness-to-radius ratio. The final section presents an extension of the  $\epsilon$  al-ysis to cylindrical radiators embodying an arbitrary, non-axisymmetric velocity distribution. A non-variational technique presented in Appendix B extends this study to "squirters" or larger wall thickness.

With the present approach, the need for machine calculations has been confined to the evaluation of Green's functions in the form of Robey's integrals. As mentioned in the acknowledgment, these integrals are being evaluated by Dr. Greenspon, J G Engineering Research Associates. The technique developed in this study has been extended to the evaluation of the mutual radiation impedances between elements of an array of free-flooding ring transducers.<sup>4</sup> Numerical results for this configuration are also being obtained by Dr. Greenspon and vill be included in his report: "Axially Symmetric Green's Functions for Cylinders."

#### II. <u>A Review of Published Analytical Approaches to the Finite Cylindrical</u> <u>Radiator</u>

The fluid potential  $\Phi$  generated by a sound radiator is given by the familiar Helmholtz integral equation  $^{1}$ 

$$\vartheta(\overline{R}) = \int_{S'} [G(\overline{R}, \overline{R}') \frac{\vartheta \vartheta(\overline{R}')}{\partial n'} - \frac{\partial C(\overline{R}, \overline{R}')}{\partial n'} \vartheta(\overline{R}')] dS'$$
(II.1)

"This analysis will be presented in a report to be published in March 1964, "Mutual Radiation Impedance for Spaced, Coa.ial, Free-Flooding Ring Transducers," CAA Report U-178-48, Contract Nonr-2739.



where S' is the radiating surface, and n' is the outward normal to the surface of integration. The first term in the integrand can be readily evaluated, if the normal velocity  $u(\overline{S}')$ , of the radiating surface, which equals  $-\partial \Phi/\partial n'$ , is known. The second term in the integrand involves the unknown potential on the radiating surface,  $\Phi(\overline{R}')$ . We must therefore, in general, solve an integral equation to obtain this potential. In the last two years, the general availability of large, digital computers has made it practical to use a finite-difference method to obtain numerical solutions of the Helmholtz integral equation for the finite cylindrical radiator<sup>2,3</sup> (Fig. 1a and the Table on p.3). The drawback of this approach is that the large computational effort involved must be repeated for every combination of length-to-radius ratic, of ka, and of surface velocity distribution.

Another successful approach to the finite cylinder problem, which circumvents the Helmholtz integral equation, uses an expansion of the potential in sphericel wave harmonics.<sup>4</sup> For this approach the volume of calculations is less than for the finite-difference approach described above. Thus, approximate results were obtained in ref. 4 without the help of electronic computers, by confining the series expansion to only a few terms. For practical applications, this method also requires computer facilities.

An approximate method which, historically, precedes the approaches described above, consists in constructing a Green's function whose derivative  $\partial G/\partial a'$  vanishes on the infinite cylindrical surface r=s. If we now prolong the cylindrical radiator by two semi-infinite rigid cylindrical baffles of the same diameter, the surface integral in Eq. II.1 is confined to the cylindrical surface (Fig. 1b). Over this surface, the second term in the integrand, which involves the unknown potent'al, has been eliminated by our choice of the Green's function. We can therefore obtain an approximate expression for the potential without having to solve an integral equatica:

$$\widehat{\mathbf{R}} = - \int_{\mathbf{S}'} u(\overline{\mathbf{R}'}) \mathbf{G}(\overline{\mathbf{R}}, \overline{\mathbf{R}'}) \, d\mathbf{S}'$$
(II.2)

With this approach Laird and Cohen<sup>5</sup> derived an analytical expression for the far field potential, the integrals being evaluated by the method of stationary phase. These integrals, which for the axisymmetric velocity distribution are known as Robey's integrals, must unfortunately be evaluated numerically if the potential on or near the radiating surface is required.<sup>6</sup> Greenspon has simplified the technique for performing this integration.<sup>7</sup> He and Shorzen also evaluated these

CONDARTEON OF VARIOUS ANALYTICAL APPROACHES TO THE CYLINDRICAL BOURCE OF FINITYE LEMOTR

ى

.

. . .

Tanana

1

7

1.

j.

۰.

-

1 . **X** .

	Salient Feature of Approach	Authors	Formulation of Problem	Electronic Computer Reguirements
ч	Finite-Difference Calculations (Fig. 10)	Baron, Mattheve end Bleich; <sup>3</sup> Chon and Schweikert <sup>3</sup>	Helmholtz Integral Equation over radiating aurface using free-space Oreen's lunction, approximated by set of similtareous algebraic finite-difference equations	Extensive
N	Series of Spherical Vave Rarmonicu	Parke and Williamo <sup>4</sup>	Integral Equation Circurvented by expansion of "otential in a series of spherica" www harmonics	Modera te
m	Semi-Infinite, Rigid Cylindrical Baffles Extending Radiator (Fig. 1b)	Laird and Coken <sup>4</sup> for far ficidi Rubey, <sup>6</sup> and Greenspon and Sherman <sup>6</sup> for near field	Helmholtz Integral Equation circum- vented by (1) constructing Green's function whose radial derivative vaniahes on cylindri- cal surface, (2) extending radiator to infinity by rigid baffles	None for far fleld solution: moderate for radistion loading calculation ("Robey's integral")
4	Variational Frinciplo for rudiation loading (Fig. 2)	Present report (Related to Schwinger's variational principle for scattering cross section)	Bimilar to approach (3), but unknown velocity distribution $\alpha(z)$ takes the place of scmi- infinite bafflos; $\alpha(z)$ satisfies integral equation; radiation impedance, stationary with respect to correct $\alpha(z)$ , is solved by Rayleign-Ritz-style calcula- tion	Modorato (Same as "Robey's integral" evaluation in approach 3)

3

Tr ble 2

: 7

•

-

integrals for non-axisymmetric velocity distributions.<sup>8</sup> The drawback of Robey's mathematical model is that it does not permit circulation of the fluid around the edges of the transducer, because of the assumption of two semi-infinite cylindrical baffles. Neither does this model lend itself to the evaluation of axially vibrating solid cylinders, or of the free-flooding open-ended cylinders known in acoustical vernecular as "squirters." Hobey originally approximated the radiation loading of such free-flooding transducers by assuming that the fluid column inside the "squirter" is terminated by pressure-release pistons.<sup>9</sup> He then refined his analysis by assuming the terminal impedance to be that of a piston in an infinite plane baffle<sup>10</sup> (Fig. 1c).

In summary, existing analyses use either a large computational effort which must be repeated for every particular combination of sound source parameters, or an elegant approximate technique which, however, does not account for the circulation of fluid around the two extremitions of the cylinder.

#### III. Description of the Present Approach

The present approach makes use of a Green's function similar to Robey's, tius eliminating from the Ashcholtz equation the term containing the potential on the cylindrical surface r=a. However, instead of assuming the source to be bracketed by rigid beffles, the potential is expressed in terms of an unknown radial velocity dustribution, a(z), over the two semi-infinite cylindrical boundaries prolonging the sound source (Fig. 2). The potential in the cylindrical column in the rigion r < a is then formulated with the help of a suitable Green's function whose normal derivative vanishes on the boundary rea. The potential in this cylindricsi region is also expressed in terms of the unknown velocity distribution  $\alpha(z)$ . By requiring continuity of these two potentials across the cylindrical boundary res, |z| > L, as integral equation for  $\alpha(z)$  is obtained. If we conpore this formulation to the free-space Green's function formulation in ref. 2 and 3, we see that the unknown function C(z) and the surface r=a, |z| > L take, respectively, the place of  $f(\overline{R}')$  and of the radiator surface. One advantage of the present approach is that a given error in the expression for  $\alpha(z)$  can be expected to result in a smaller error in the reliation impedance that would result from a similar error in  $\dot{v}(\mathbf{\bar{R}}')$  in ref. 2 and 3. The principal advantage, however, is that this approach leads itself to an approximate variation 1 solution both of

the solid and open-ended finite cylinder.\*

z

The radiation impedance Z can be written as the sum of the impedance  $Z_r$  obtained by setting  $\alpha(z) = 0$ , and of an impedance  $Z_{\alpha}$  associated with  $\alpha(z)$ , the unknown velocity distribution in the region |z| > L:

$$= Z_r + Z_\alpha$$
 (III.1)

 $Z_{\alpha}$  is thus in the nature of a correction factor to impedances computed by Robey,<sup>6</sup> dreenspon,<sup>7,8</sup> and Sherman<sup>8</sup> from Robey's mathematical model. By virtue of the variational principle to be derived in Section V, for the correct solution of the integral equation  $\alpha(z)$ ,  $Z_{\alpha}$  is proportional to a functional  $\hat{J}[\alpha]$  which is stationary with respect to first order variations of  $\alpha(z)$ :

$$Z_{\alpha} \propto J[\alpha]$$

$$\frac{\delta J[\alpha]}{\delta \alpha} = 0$$
(III.2)

Furthermore,  $J[\alpha]$  depends on the functional form of  $\alpha(z)$  but not on its amplitude. The technique for computing  $Z_{\alpha}$  is similar to the Rayleigh-Ritz technique for evaluating the paturel frequency.

This approach parellels the use of the Levine-Schwinger variational principle for acattering cross sections, which has been applied to a large number of diffraction problems.<sup>12</sup> The equivalent principle for radiation impedances is proved in its general form, using free-space Green's functions, by Morse and Feshbach.<sup>13</sup> These authors do not, however, use it to solve any particular problem. Apparently, only Storer<sup>14</sup> applied this principle to a specific problem, viz. the effect of a finite circular baffle on the radiation loading of a commission. In 1954, Professor Storer, of Harvard University, suggested to the author of this report that the axisymmetrically vibrating cylinder of finite length could also be analyzed in this fashion. Consequently a rather sketchy variational solution

<sup>&</sup>quot;A rigorous Wiener-Hopf type solution of the integral equation is possible for semi-infinite cylindrical radiator problems formulated in this fashion, according to Levine and Schwinger.<sup>11</sup> "Encre authors mention this formulation as an alternative to the one they actually used in their analysis of sound radiation from a semi-infinite pipe. Levine,<sup>12</sup>S extended this study to pipes of arbitrary cross section. One of his approximations for the reflection coefficient is obtained from a variational solution of an integral equation (his "variational principle A") appliable over the semi-infinite cylindrical surface extending the pipe, and is therefore of the form of the integral equation used in this report.

of the solid cylindrical source was presented in an internal memorandum of the Harvard Acoustics Research Laboratory.<sup>15</sup> A detailed analysis was not carried out because numerical results depended on the evaluation of Robey's integral, which at that time was not available. Since, as mentioned earlier, such integrals can now be readily evaluated, and since Dr. Greenspor <u>kindly egreed</u> to apply his experiience in this type of calculation to the problem at hand, it is now worthwhile to use the variational formulation to obtain a solution to the finite cylinder problem.

#### IV. Integral Equation Formulation of the Solid Cylinder Problem

The infinite region surrouncing the cylindrical radiator is subdivided into three regions (Fig. 2): an outer region, r > a, identified by subscript o; and two semi-infinite inner regions, r < a, one corresponding z > L and identified by the subscript i+, and a second inner region corresponding to values of z < L, identified by the subscript i-. The Green's function for the outer region catisfying the condition  $\partial G/\partial r'=0$  for r'=z, was constructed by Robey:

$$G_{o}(r,a,z-z') = \frac{1}{\frac{1}{4\pi^{2}a}} \int_{-\infty}^{\infty} \frac{H_{o}(k_{r}r)}{k_{r}a_{1}(k_{r}a)} \exp[ik_{z}(z-z')] dk_{z}$$
(IV.1)

The evaluation of G<sub>0</sub> is the subject of references 6 and 7. Since  $k_r = (k^2 - k_z^2)^{\frac{1}{2}}$ , the functions or  $x_r$  in the integrate are even in  $k_z$ . Hence, only the real component of the exponential function,  $\cos[k_z(z-z^*)]$  contributes to the integral. The same comment applies to the Green's function for the infinite cylindrical region r < a:

$$G_{1}(r,s,z-z') = -\frac{1}{4\pi^{2}s} \int_{\infty}^{\infty} \frac{J_{0}(k_{r}r)}{k_{r}J_{1}(k_{r}s)} \exp[ik_{z}(z-z')] dk_{z}$$
(IV.2)

This function is derived and evaluated in Appendix A.

In the case of the solid cylindrical radiator it is convenient to combine two Green's functions of the form of Eq. IV.2 so that the normal derivative of the resultant Green's function vanishes over the end caps of the cylinder, i.e., in the two circular regions r < a, z = L. This will eliminate the potential term from the Helmholtz integral over the end caps, as well as over the cylindrical surface.

Such Green's functions are readily constructed by introducing image sources:

$$G_{i+}(r,a,z-z') = G_{i}(r,a,z-z') + G_{i}(r,a,z+z'-2L)$$

$$G_{i-}(r,a,z-z') = G_{i}(r,a,z-z') + G_{i}(r,a,z+z'+2L)$$
(IV.3)

These Green's functions can be written more concisely as\*

$$G_{it}(r,a,z-z') = -\frac{1}{2\pi^2 a} \int_{-\infty}^{\infty} \frac{J_{o}(k_{r}r)}{k_{r}J_{1}(k_{r}a)} \cos[k_{z}(z\neq L)]\cos[k_{z}(z'\neq L)]dk_{z}$$
(IV.4)

We can now write the potentials in these three regions by making use of the modified Helmholtz integral in Eq. II.2. If we assume that the end caps are rigid, the surface integral reduces to the cylindrical surface:\*\*

$$\Phi_{0}(\mathbf{r},z) = 2\pi a \int_{-\infty}^{z} u(z') G_{0}(\mathbf{r},a,z-z') dz'$$
 (IV.5a)

$$\phi_{i\pm}(\mathbf{r},z) = \overline{4}2\pi a \int_{\pm L}^{\pm \infty} u(z')[G_{i}(\mathbf{r},a,z-z') + G_{i}(\mathbf{r},a,z+z' + 2L)] dz' \qquad (IV.5b)$$

The time dependence of u(z') and of the potentials is harmonic. The time-dependent function  $\exp(-imt)$  has been united, for the sake of brevity throughout this report. These integrals are of opposite sign, because  $\partial \Phi/\partial n'$  in the Helmholtz equation equals  $-\partial \Phi_0/\partial r=u$  in the outer region, and  $+\partial \Phi/\partial r=-u$  in the inner region. With this sign convention, the sound pressure equals  $\rho \Phi$ . If the velocity distribution of the radiator is symmetrical about the plane z=0, the two inner regions will have identical potentials and we need concern ourselves with only one inner region which we will designate by the subsomipt i. For the case of a symmetrical velocity distribution, only the part of the Green's function which is symmetrical

7

<sup>\*</sup>Here, and elsewhere in this report, alternative subscripts and signs have been used, for the sake of brevity, to condense two equations into one, the upper subscript on the left side of the equation being associated with the upper sign on the right side of the equation, and vice versa.

<sup>&</sup>lt;sup>64</sup>In section X expressions are given for an arbitrary velocity distribution over the radiating surface. The potential contributed by vibrating end caps is given in Eq. X.1 and 8.

about z'=0 condributes to the potential. We will denote this even component of  $C_0$  by  $F_0$ :

$$g_{o}(r,a,z-z') = \frac{1}{2\pi^{2}a} \int_{0}^{\infty} \frac{B_{o}(k_{r}r)}{k_{r}H_{1}(k_{r}a)} \cos k_{z}^{2} \cos k_{z}^{2} dk_{z}$$
 (IV.6)

We further specialize the problem by assuming that the radial velocity of the sound source is constant and equal to U. u(z') is therefore a known function for  $|z| \leq L$ . The velocity distribution along the cylindrical boundaries z > L, rea, is an unknown function, say Ux(z). The integrals for the potentials now because

$$\delta_{1}(r,z) = -2\pi a \ \overline{U} \int_{L}^{\infty} \alpha(z') [G_{1}(r,a,z-z') + G_{1}(r,a,z+z'-2L)] dz'$$
 (IV.7a)

$$\Phi_{O}(\mathbf{r},z) = 4\pi a \, \Psi[\int_{C}^{L} g_{O}(\mathbf{r},a,z-z') \, dz' + \int_{L}^{\infty} \alpha(z')g_{O}(\mathbf{r},a,z-z') \, dz'] \qquad (IV.7b)$$

For the sake of brevity we will from new on express the known component of the potential  $\frac{1}{2}_{0}$ , evaluated on the cylindrical surface rea, as a function, say H(z), rather than as an integral

$$H('z) = \int_{0}^{L} g_{0}(a, a, z-z') dz'$$
 (IV.8)

We can now construct the integral equation which the unknown function  $\alpha(z)$  must satisfy. This integral equation is derived from the requirement that the potentials be continuous across the cylinder boundary r=a, z > L:

$$\phi_{0}(a,z) - \phi_{1}(a,z) = 0, \text{ for } z > L$$
(IV.9)

When we substitute Eqs. IV.5, this continuity condition takes the form of a nonhomogeneous Fredholm integral equation of the first kind

$$\int_{L}^{\infty} \Gamma(z-z') \alpha(z') dz' = -S(z), \text{ for } z > L$$
 (IV.10)

where the kernel of this equation is given by

$$I(z-z') = g_0(a,a,z-z') + \frac{1}{2}G_1(a,a,z-z') + \frac{1}{2}G_1(a,a,z+z'-2L)$$
 (IV.11)

We will now show that the function  $\alpha(z')$  which satisfies this integral equation gives a stationary value for the radiation impedance, with respect to variations  $\delta \alpha$ .

#### V. Derivation of the Variational Principle for the Radiation Impedance

We note for future use that the regultant radiation impedance on the cylinder is obtained by integrating the pressure

$$p(a,z) = \rho \delta_0(a,z)$$
  
=  $-i\omega\rho \delta_0(a,z)$  (V.1)

over the surface of the cylinder:

$$Z = -\frac{4i\pi a \omega_0}{U} \int_{0}^{L} \Phi_0(a,z) dz$$
  
=  $-i(4\pi a)^2 \omega_0 \int_{0}^{L} [H(z) + \int_{L}^{\infty} \alpha(z')g_0(a,a,z-z') dz'] dz$  (V.2)

The radiation impedance is thus clearly the sum of two component impedances, as indicated in Eq. III.1. The impedance computed from Robey's model is

$$Z_{r} = -(L_{RA})^{2} i \exp \int_{C}^{L} H(z) dz$$
  
= -(L\_{RA})^{L} i \exp \int\_{0}^{L} \int\_{0}^{L} g\_{0}(a,a,z-z^{\*}) dz^{\*} dz (V.3)

Like H(z),  $Z_r$  is a known quantity since it does not involve the unknown function  $\alpha(z)$ . The correction term in Eq. III.1, which embodies the contribution of the

flow across the cylindrical boundaries prolonging the source is

$$Z_{\alpha} = -(4xa)^{2} \lim_{\Omega \to 0} \int_{0}^{L} \left[ \int_{0}^{\infty} \alpha(z') g_{0}(z, a, z-z') dz' \right] dz \qquad (V.4)$$

Since the Green's function is symmetrical in z and z' the order of integration in Eq. 9.4 can be inverted:

$$Z_{\alpha} = -(4\pi a)^{2} i \omega \rho \int_{I}^{\infty} \left[ \int_{0}^{L} g_{0}(a,a,z-z') dz' \right] \alpha(z) dz$$
$$= -(4\pi a)^{2} i \omega \rho \int_{U}^{\infty} H(z) \alpha(z) dz \qquad (V.5)$$

A functional  $J[\alpha]$  will be defined below, Eqs. V.9 and 10. For future reference we note that for the correct function  $\alpha(z)$ , i.e., for the function which satisfies the integral equation, Eq. IV.10, this functional con also be written as

$$J[\alpha] = -\int_{L}^{\infty} H(z) \alpha(z) dz, \text{ for } \alpha(z) \text{ solution of Eq. IV.10.} \qquad (V.6)$$

Comparing this with Eq. V.5, we can relate this functional and the impedance  $Z_{\alpha}$ :

$$Z_{\alpha} = imp(4\pi a)^2 J[\alpha], \text{ for } \alpha(z) \text{ solution of Eq. IV.10.}$$
 (V.7)

We will now prove that  $J[\alpha]$  is stationary with respect to first order variations for about the correct function  $\alpha(z)$ . For this purpose, we relate  $J[\alpha]$  to the integral equation, Eq. IV.10. We multiply both sides of this equation by  $\alpha(z)$  and integrate with respect to z over the region z > L. We then divide both sides of the equation thus obtained by

$$\left[\int_{L}^{\infty} H(z) \alpha(z) dz\right]^{2}$$

Our original integral equation now takes the form

10

*.* 

. .

· · · · ·

$$\frac{\int_{L}^{\infty} \int_{L}^{\alpha(z)\Gamma(z-z')} \alpha(z') dz' dz}{\left[\int_{L}^{0} H(z)\alpha(z) dz\right]^{2}} = \frac{-1}{\int_{L}^{\infty} H(z)\alpha(z) dz}$$
(V.8)

The functional  $J[\alpha]$  is defined as the reciprocal of the left side of this equation:

$$J[\alpha] = \frac{A^2[\alpha]}{B[\alpha]}$$
(V.9)

where

1

.

$$A[\alpha] = \int_{L}^{\infty} H(z) \alpha(z) dz \qquad (V.10a)$$

$$B[\alpha] = \int_{L}^{\infty} \int_{L}^{\infty} \alpha(z) \Gamma(z-z') \alpha(z') dz' dz \qquad (V.10b)$$

For a function  $\alpha(z)$  which satisfies the integral equation, and therefore the equality in Eq. V.8, the reciprocal of the right side of Eq. V.8 is also equal to the  $J[\alpha]$ , as already indicated in Eq. V.6. If the functional derined in Eq. V.9 is indeed stationary with respect to small variations of the function  $\alpha(z)$ , then, by definition, the increment  $\delta J[\alpha]$  associated with an increment  $\delta \alpha$  is zero:

$$\delta J[\alpha] = \frac{2E[\alpha]}{B[\alpha]} \cdot \int_{L}^{\infty} H(z) \ \delta \alpha(z) \ dz - \frac{A^{2}[\alpha]}{B^{2}[\alpha]} \cdot \int_{L}^{\infty} \Gamma(z-z')[\alpha(z)\delta\alpha(z')+\alpha(z')\delta\alpha(z)] \ dz' \ dz = 0 \quad (V.11)$$

Like the Green's functions in Eq. IV.11,  $\Gamma(z-z')$  is symmetrical with respect to z and z'. We can therefore invert the order of integration in the former of the two terms of the integrand of the double integral in Eq. V.11. The double integral can thus be condensed to

$$\int = 2 \int_{L}^{\infty} \Gamma(z-z') \alpha(z) \delta \alpha(z) dz' dz \qquad (V.12)$$

From the integral equation, Eq. IV.10, we see that the integral over z' equals -H(z) for the correct function  $\alpha(z)$ . The double integral can thus finally be written as

$$\iint_{L} = -2 \int_{L}^{\infty} H(z) \, \delta \alpha(z) \, dz \qquad (V.13)$$

When we substitute Eq. V.13 in place of the double integral in Eq. V.11 and multiply the terms of this equation by the ratio  $B^2[\alpha]/2A[\alpha]$ , we obtain

١

$$\{H[\alpha] + A[\alpha]\} \cdot \int_{L}^{\infty} H(z) \, \delta\alpha(z) \, dz = 0 \qquad (V.1h)$$

Since the integral is not identically zero the sum in brackets, which multiplies this integral, must vanish, i.e.,

$$B[\alpha] = -A[\alpha]$$
 (V.15)

When we substitute the definitions of these two functionals, Eqs. V.10, this because:

$$\int_{L}^{\infty} \int_{L}^{\infty} \alpha(z) \Gamma(z-z') \alpha(z') dz' dz = - \int_{L}^{\infty} H(z) \alpha(z) dz \qquad (\forall .16)$$

This equation is obviously satisfied if  $\alpha(z')$  satisfies our original integral equation, Eq. IV.10. We have thus shown that the functional J[ $\alpha$ ], as defined in Eq. V.9, does indeed take on a stationary value for the correct value of the function  $\alpha(z)$ . Since the functional J[ $\alpha$ ] is stationary with respect to the correct function, the error in J[ $\alpha$ ] is of a higher order than the error in  $\alpha(z)$ .

We shall now illustrate the evaluation of the radiation impedance by means of the variational principle just derived.

#### VI. Evaluation of the Radiation Impedance from the Variational Principle

We first proceed to select the simplest trial function  $\alpha(z)$  which yields a far field potential  $\hat{\Phi}_i$  in the desired form of a spherically spreading wave,

$$\Phi_1(a,z) = \frac{A}{|z|} \exp(ikz)$$
, for large  $|z|$  (VI.1)

If the cylindrical boundary r=a, |z| > L were in the form of a rigid pipe, the far field potential would only decay as a result of viscous losses, as embodied in an imaginary component of the wave number. The far field potential in a rigid pipe can therefore only decay exponentially. The desired spherical spreading loss, Eq. VI.1, must therefore be the result of energy flow across the cylindrical boundary associated with the velocity  $\alpha(z)$ . The rate of energy outflow, per unit axial diskspee, along the "pipe" is

$$\frac{\partial \tilde{z}(z)}{\partial z} = \pi kapc U \left| \frac{k_1}{4}(a, z) \right| x(z) \right|, \qquad (VI.2)$$

This must balance the decrease, per unit axial distance, of the acoustic energy propagating down the pipe:

$$\frac{\partial \vec{x}(z)}{\partial z} = \frac{\partial}{\partial z} \left[ \pi \operatorname{ock}^2 \int_0^u \left[ \mathbf{a}_1(r, z) \right]^2 r dr \right]$$
 (VI.3)

In the far field, and for the values of kL characteristic of transducers,  $s_1(r,z)$  can be set equal to  $s_1(a,z)$ .<sup>#</sup> We can thus replace the integral in Eq. VI.3 by  $a^{\frac{3}{2}}_{1}(a,z)^{\frac{3}{2}/2}$ . When we substitute Eq. VI.1 for  $s_1$  in Eqs. VI.2 and 3, we can solve for  $|\alpha(z)|$ :

$$|\alpha(z)| = \frac{kaA}{U} \frac{1}{|z|^2}$$
, for large  $|z|$  (VI.4)

We thus conclude, that in the far field,  $\alpha(z)$  must be of the form

$$\alpha(z) \propto \frac{\exp(ikz)}{z^2}$$
, for large  $|z|$  (VI.5)

We will now illustrate the use of the variational principle by celecting the

"Even for large ka, the value of  $\phi_i$  averaged over the cylindrical cross section is equal to a constant times  $\phi_i(a,z)$ . The functional relation derived in Eq. VI.4 therefore still holds, but the constant in this equation will not equal A.

simplest trial function which satisfies Eq. VI.5 and which can also account for a more rapidly decaying near field. Such a function requires the use of at least two unknown coefficients,  $x_1$  and  $x_2$ :

$$\alpha(z) = \left(\frac{z_1}{z^2} + \frac{x_2}{z^3}\right) \exp(ikz)$$
 (VI.6)

Since  $x_1$  and  $x_2$  are generally complex quantities, this expression allows for a phase shift between the two components of the potentials. By virtue of the variational principle, the best values of the unknown coefficients are those which give a stationary value to the functional  $J[\alpha]$ :

$$\frac{\partial J[\alpha]}{\partial x_2} = 0 \qquad (VI.7)$$

When we substitute Eq. V.9 for the functional  $J[\alpha]$  we can write these equations, after some manipulation, as

$$2 \frac{\partial A[\alpha]}{\partial x_1} A[\alpha] - \frac{\partial B[\alpha]}{\partial x_1} J[\alpha] = 0$$
 (VI.8)

etc.

When we combine the assumed trial function, Eq. VI.6, with the definitions of the functionals  $A[\alpha]$  and  $B[\alpha]$ , Eqs. V.10, these two functionals are found to be, respectively, linear and quadratic in  $x_1$  and  $x_2$ ,

$$A[\alpha] = a_{1}x_{1} + a_{2}x_{2}, \quad \partial A/\partial x_{1} = a_{1}, \quad \partial A/\partial x_{2} = a_{2}$$
  

$$B[\alpha] = b_{1}x_{1}^{2} + b_{2}x_{2}^{2} + 2b_{12} x_{1}x_{2}$$
  

$$\partial B/\partial x_{1} = 2b_{1}x_{1} + 2b_{12}x_{2}, \quad \partial B/\partial x_{2} = 2b_{2}x_{2} + 2b_{12} x_{1} \qquad (VI.9)$$

where coefficients  $a_1$ ,  $a_2$ ,  $b_1$ ,  $b_2$  and  $b_{12}$  are known, complex quantities. When expressions VI.9 are substituted in Eqs. VI.8 a set of two simultaneous linear equations is obtained:

$$2(a_{1}^{2} - b_{1} J[\alpha])x_{1} + 2(a_{1}^{2} - b_{12} J[\alpha])x_{2} = 0$$

$$2(a_{1}^{2} - b_{12} J[\alpha])x_{1} + 2(a_{2}^{2} - b_{2} J[\alpha])x_{2} = 0$$
(VI.10)

Since this set of equations is homogeneous, its coefficient matrix must vanish:

$$S(a^{1}a^{5}-p^{15}2[\alpha]) = S(a^{5}-p^{5}2[\alpha]) = 0$$

$$S(a^{1}a^{5}-p^{15}2[\alpha]) = 0$$
(AI'17)

Expanding this matrix we only obtain terms proportional to  $J^2[\alpha]$  and  $J[\alpha]$ . The equations can therefore be solved for  $J[\alpha]$ :

$$J[\alpha] = \frac{a_1^2 b_2 + a_2^2 b_1 - 2a_1 a_2 b_{12}}{b_1 b_2 - b_{12}^2}$$
(VI.12)

The remarkable feature of this result is that the functional is independent of  $x_1$ and  $x_2$ . Like the coefficients  $a_1$ ,  $a_2$ ,  $b_1$ ,  $b_2$  and  $b_{12}$  the functional in Eq. VI.12 is complex. We see, by referring to Eq. V.7, that the imaginary component of  $J[\alpha]$  embodies the radiation resistance, and its real component, the reactance.

13. **24** 24

If we are merely interested in the radiation loading we need not evaluate the unknown coefficients  $x_1$  and  $x_2$ . If, however, we wish to compute the far field potentials, we must substitute the expression for  $\alpha(z)$  in Eqs. IV.8, and therefore require the values of  $x_1$  and  $x_2$ . The ratio of these two coefficients is obtained from either of the two homogeneous equations, Eq. VI.10

$$\frac{x_2}{x_1} = \frac{(-a_1^2 + b_1 J[\alpha])}{a_1 a_2 - b_{12} J[\alpha]}$$
(VI.13)

where the value of  $J[\alpha]$  is known from Eq. VI.12. The coefficient  $x_1$  is obtained by substituting the ratios  $x_2/x_1$ , Eq. VI.13 in Eq. VI.6, which is then substituted for  $\alpha(z')$  in the integral equation, Eq. IV.10. Unless the functional dependence of the trial function, Eq. VI.6, on z is the correct one, the coefficient  $x_1$  can not be selected so as to satisfy the integral equation in the whole range |z| > L. It is advantageous to select a coefficient  $x_1$  which satisfies the integral equation for a value of z associated with a relatively large value of  $\alpha(z)$  and hence with a large contribution to the far field potential, viz. for  $z \ll L$ , say  $L + \epsilon$ :

x

$$\frac{-H(I+s)}{\int_{L}^{\infty} \Gamma(I+s-z') \cdot (\frac{1}{z'^{2}} + \frac{x_{2}}{x_{1}} \frac{1}{z'^{3}}) \exp(ikz') dz'}$$
(VI.14)

An alternative procedure, which tives more nearly equal weight to the whole region of z where the integral equation applies, is based on the fact that for the correct function  $\alpha(z)$ ,  $J[\alpha]$  equals  $-A[\alpha]$ , from Eqs. V.6 and V.10a. Hence, substituting the ratio  $x_2/x_1$ , from Eq. VI.13, and the value of  $J[\alpha]$  from Eq. VI.12, in the expression for  $A[\alpha]$ , Eq. VI.9, we can solve for  $x_1$ 

$$x_{1} = \frac{-J[\alpha]}{a_{1} + a_{2}(x_{2}/x_{1})}$$
(VI.15)

ĩ

2

1

Experience with numerical calculations will indicate which procedure is preferable.

To refine the selection of the trial function further, we can introduce additional unknown coefficients associated, for example, with non-propagating incompressible near field components of the potentials. The trial function might thus, for example, be expressed in terms of three coefficients:  $x_1$  and  $x_2$ . associated with propagating components of the potentials, and  $x_3$  with an incompressible, nearfield component decaying rapidly with distance:

$$\alpha(z) = \left(\frac{x_1}{z^2} + \frac{x_2}{z^3}\right) \exp(ikz) + \frac{x_3}{z^3}$$
(VI.6a)

This yields three simultaneous equations of the rorm of Eq. VI.8. Once again, we will find that this equations are linear in the three unknown coefficients and, of course, homogeneous. We can therefore construct a third order determinant similar to Eq. VI.11. The constant term and the linear term in  $J[\alpha]$  are found to cancel, leaving only a cubic and a quadratic term in  $J[\alpha]$ . The determinant thus yields a single root  $J[\alpha]$ :

$$J[\alpha] \approx [a_1^2(b_2b_3 - b_{23}^2) + a_2^2(b_1b_3 - b_{13}^2) + a_3^2(b_1b_2 - b_{12}^2) + + 2a_1a_2(b_{13}b_{23} - b_3b_{12}) + 2a_1a_3(b_{12}b_{23} - b_{22}b_{13}) + 2a_2a_3(b_{12}b_{13} - b_1b_{23})] \cdot [b_1b_2b_3 + 2b_{12}b_{13}b_{23} - (b_1b_{23}^2 + b_2b_{13}^2 + b_3b_{12}^2)]^{-1}$$
(YI.12a)

We then solve three of the set of three homogeneous equations, Eqs. VI.8, for two ratios of undertermined coefficients. Finally, we solve for the amplitude of the one remaining coefficient by satisfying the integral equation z = 1 + z, or in the manner indicated in Eq. VI.15.

If  $\alpha(z)$  is expressed in terms of N unknown coefficients, the functionals  $A[\alpha]$  and  $B[\alpha]$  and their derivatives take on the following form:

$$A[\alpha] = \sum_{n=1}^{N} a_n x_n , \quad \frac{\partial A}{\partial x_n} = a_n$$
  

$$B[\alpha] = \sum_{n=1}^{N} (b_n x_n^2 + 2 \sum_{m \neq n}^{N} b_{nm} x_n x_m)$$
  

$$\frac{\partial B}{\partial x_n} = 2b_n x_n + 2 \sum_{m \neq n}^{N} b_{nm} x_m$$
(VI.9e)

The set of N homogeneous linear equations for the \_known coefficients corresponding to Eqs. VI.10 is of the general form

$$2(a_{n}^{2}-b_{n}J[\alpha])x_{n} + 2\sum_{m\neq n}^{N} (a_{n}a_{m}-b_{nm}J[\alpha])x_{n} = 0$$
 (VI.10a)

When the coefficient matrix of this set of equations is set equal to zero it will be found that only the terms containing the two highest powers of  $J[\alpha]$ , N and N-1, do not cancel. When both terms are divided by  $(J[\alpha])^{N-1}$ , a linear equation in  $J[\alpha]$  is obtained. The Nth order determinant for  $J[\alpha]$  has a single non-vanishing root. This is consistent with the requirement that the integral equation, Eq. IV.10, have only one solution.

Experience with numerical calculations will show whether the radiation impedance is sensitive to the selection of the trial function  $\alpha(z)$ . If this should be the case, the functional dependence of the near field potentials on z, particularly of the non-propagating incompressible components, can be studied more closely so as to construct a more sophisticated trial function than Eqs. VI.6 or 6a. Theoretical insight into this functional relation can be gained from the fluid mechanics literature dealing with accessions to inertia of vibrating solids. Comparison with the results of the non-variational solutions presented in Appendix B can also be used to evolve more refined expressions of  $\alpha(z)$ .

The fact that  $J[\alpha]$  (or, referring to Eq. V.7,  $Z_{\alpha}$ ) can be evaluated from a variational principle, without previously determining the amplitude of the

unknown coefficients of the trial function G(z), has already been related to the Levine-Schwinger variational principle for scattering cross sertions (Section III). Another parallel which may be more familiar to sume readers is found in the Raylsigh-Ritz method for optimizing the natural frequencies obtained from Rayleigh's principle.<sup>16</sup> In this method a trial function is assured for the dynamic configuration of the vibrating system. The best choice of the coefficients in this trial function is determined by giving a stationary value to the natural frequency obtained from Ray)sigh's principle. If B unknown coefficients are used in expressing the trial function, a set of N linear homogeneous equations is obtained with the coefficients as unknown quantities. By setting the coefficient matrix of this set of equations equal to zero, values of matural frequencies are obtained, withour ever baving to compute the unknown coefficier's themselves. The fundamental natural frequency thus obtained is equivalent to the subctional  $J[\alpha]$ . To compute the ratio of the undetermined coefficients at that frequency one substitutes this value of the fundamental frequency back in the set of N hozogoneous equations and solves for E-1 ratios. The amplitude of the Mth unknown coefficient is finally obtained from an inhomogeneous equation of motion.

To conclude our study of the solid cylindrical radiator, we now turn to the evaluation of the far field potentials.

#### VII. The Far Field Potentials

To evaluate the potential in the region r < a, z > L we substitute the Green's function, G<sub>4</sub>, Eq. IV.2, in Eq. IV.7a:

$$\hat{e}_{1}(r,z) = \frac{U}{2\pi} \int_{L}^{\infty} \alpha(z') dz' \int_{-\infty}^{\infty} \frac{J_{0}(k_{r}r)}{k_{r}J_{1}(k_{r}e)} \{ \exp[ik_{z}(z-z')] + \exp[ik_{z}(z+z'-2L)] \} dk_{z} \quad (VII.1)$$

For  $\alpha(z')$ , we substitute a trial function of the form of Eq. VI.6, with the unknown coefficients expressed in terms of  $J[\alpha]$ , as described in 3ection VI. The  $k_z$ -integral associated with the first exponential term in braces in Eq. VII.1 is

"In contrast to the variational principle used here, which yields a single solution  $J(\alpha)$ , the Rayleigh-Ritz techniques yields a number of natural frequencies event to the number of unknown coefficients in the assumed trial function.

is given in Appendix A, Eq. A.14. Substituting (-z'+2L) in place of z', we obtain the  $k_z$ -integral associated with the second exponential torm. When we add the two integrals, we obtain

$$\int_{-\infty}^{\infty} dk_{z} = 2\pi i \left[ \frac{\exp(ik|z-z'|)}{k^{\alpha}} + \frac{\exp[ik(z+z'-2L)]}{k^{\alpha}} \right] + 4\pi i \sum_{n=1}^{\infty} \frac{J_{o}(k_{n}r)}{(k^{2}-k_{n}^{2})^{\frac{1}{2}} a[J_{o}(k_{n}a) - J_{c}(k_{n}a)]} \cdot \left[ \exp[i(k^{2}-k_{n}^{2})^{\frac{1}{2}} |z-z'|] + \exp[i(k^{2}-k_{n}^{2})^{\frac{1}{2}} (z+z'-2L)] \right] \quad (VII.2)$$

where  $(k_n a)$  is the nth root of the Bessel function  $J_1$ . The integral over z' in Eq. VII.1, must be split into two regions of integration: (1) from L to z, where |z-z'| is taken equal to (z-z'), and (2) from z to  $\infty$ , where |z-z'| equals -(z-z'). Since even the lowest root, k\_a=3.83, is generally larger than the ka-value of resonant piezoelectric or magnetostrictive transducers, the terms under the summation sign decay exponentially with increasing |z-z'|. Because the source distribution  $\alpha(z)$  extends to infinity, these "near field" terms contribute to the far field. By using energy flow considerations it was shown in Section VI that the desired for field behavior of  $\phi_i$ , Eq. VI.1, requires that the function  $\alpha(z)$  embody terms of order  $|z|^{-2}$  and higher. The dominant, plane-wave components of the inverse transforz of the Green's function, Eq. VII.2, do not decay with increasing z'. In combination with the far field term of  $\alpha(z')$ , these plane wave components of the Green's function therefore give rise to a far field potential whose absolute value varies as  $\int |z|^{-2} dz = -|z|^{-1}$ . This result is consistent with the potential, Eq. VI.1, used in deriving the functional form of  $\alpha(z)$  in the far field. The evaluation of the inverse transform of the Green's function, Eq. A.14, is thus consistent with the energy flow analysis in Eqs. VI.2 to 4.

The for field in the region r > a is obtained by substituting the appropriate Green's function, Eq. IV.1, in Eq. IV.5a:

$$\phi_{0}(r,z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} \int_{-\infty}^{\infty} \frac{H_{0}(k_{r}r)}{k_{r}H_{1}(k_{r}a)} \exp[ik_{z}(z-z')] u(z') dk_{z} dz' \qquad (VII.3)$$

The integration over z' can be carried out immediately by making use of the definition of the Fourier transform of the velocity distribution u(z'):

$$u(k_z) = \int_{-\infty}^{\infty} u(z') \exp(-ik_z z') dz' \qquad (VII.4)$$

The expression for the potential now becomes

$$\Psi_{0}(\mathbf{r},z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} \frac{u(k_{z})H_{0}(k_{r}r)}{k_{r}H_{1}(k_{r}a)} \exp(ik_{z}z) dk_{z}$$
(VII.5)

When the asymptotic, large argument expression for the Hankel function,

$$H_{o}(k_{r}r) = \left(\frac{2}{4k_{r}r}\right)^{\frac{1}{2}} \exp[i(k_{r} - \frac{\pi}{4})]$$
(VII.6)

is substituted in the integrand in Eq. VII.5, and using spherical coordinates R and  $\theta$  in lieu of the cylinder coordinates,

$$z = R \cos\theta$$
 and  $r = R \sin\theta$ ,

the far field potential becomes

$$s_{o}(\mathbf{R}, \boldsymbol{\theta}) = \frac{\exp(-i\pi/4)}{(2\pi^{3}\mathbf{R} \sin\theta)^{\frac{1}{2}}} \int_{-\infty}^{\infty} \frac{\exp[i\mathbf{R}(\mathbf{k}_{r}\sin\theta + \mathbf{k}_{z}\cos\theta)]}{\mathbf{k}_{r}^{3}\mathbf{H}_{1}(\mathbf{k}_{r}a)} d\mathbf{k}_{z} \qquad (\text{VII.7})$$

This integral was evaluated by Laird and Cohen<sup>5</sup> using the method of stationary phase. The result thus obtained is

$$\phi_{0}(R,\theta) = \frac{-iu(k \cos\theta) \exp(ikR)}{\pi kR \sin\theta H_{1}(ka \sin\theta)}$$
(VII.8)

The velocity transform  $u(k \cos\theta)$  can be written more explicitly in terms of the trial function  $\alpha(z')$  as

$$u(k \cos\theta) = 2U[\frac{\sin(kL \cos\theta)}{k \cos\theta} + \int_{L}^{\infty} \alpha(z') \cos(kz' \cos\theta) ds'] \qquad (VII.9)$$

We now have concluded the analysis of the solid cylindrical radiator. and proceed with the open-ended cylinder.

#### VIII. The Open-Ended Free-Flooding Cylindrical Radiator of Vanishing Wall Thickness

We consider a cylindrical radiator whose wall thickness is negligible compared to both its radius and the acoustic wavelength. For such a source configuration we can construct a single potential  $\phi_i$  for the region r < a extending now from  $-\infty < z < \infty$ . This is in contrast to the solid radiator where we had to define two potentials  $\phi_i$  and  $\phi_i$  each valid in a semi-infinite region. The outer potential  $\phi_0$  is similar to the outer potential derived for the solid cylinder, Eqs. IV.5a and 7a. The inner potential is of the form

$$\delta_{i}(r,z) = -2\pi a \int_{-\infty}^{\infty} u(z') G_{i}(r,a,z-z') dz'$$
 (VIII.1)

where the Green's function  $G_i$  is given in  $\Xi_i$ . IV.2. Organing this with Eq. IV.5b we see that the potentials  $\Phi_i$  defined, respectively, for the free-flooding and solid case differ as to the range of z'over which the integration is performed as well as to their Green's functions. As in the case of  $G_0$ , it is convenient to define separately the component of  $G_i$ , which is symmetrical about z'=0, and which alone contributes to the potential when the velocity distribution is similarly symmetrical:

$$g_{1}(r,a,z-z') = -\frac{1}{2\pi^{2}n} \int_{0}^{\infty} \frac{J_{0}(k_{r}r)}{k_{r}J_{1}(k_{r}a)} \cos k_{z}z \cos k_{z}z' dk_{z}$$
(VIII.2)

Assuming a constant velocity U over the radiating surface, this potential is again expressed in terms of an unknown velocity distribution U  $\alpha(z')$ :

$$\dot{g}_{1}(r,z) = -4\pi a \, \Im \left[ \int_{0}^{L} g_{1}(r,a,z-z') \, dz' + \int_{L}^{\infty} \alpha(z') g_{1}(r,a,z-z') \, dz' \right] \quad (VIII.3)$$

The continuity condition at the boundary rea, z > L, is again in the form of Eq. IV.9. The corresponding integral equation is therefore also, formally at least, similar to the integral equation derived for the solid cylinder, Eq. IV.10. Now,

however, the kernel  $\Gamma(z-z')$  and the non-homogeneous term E(z) are

$$\Gamma(z-z') = g_{i}(s, a, z-z') + g_{o}(a, a, z-z')$$
(VIII.4a)

$$H(z) = \int_{0}^{L} [g_{1}(a, a, z-z') + g_{0}(a, a, z-z')] dz' \qquad (VIII.4b)$$

The expression for  $\Gamma(z-z')$  can be simplified by using the Wronskian relation for  $H_{c}$  and  $J_{c}$ :

$$\Gamma(z-z') = \frac{1}{2\pi^2 a} \int_{0}^{\infty} \frac{H_{0}(k_{r}a)}{\sum_{r=1}^{m} H_{1}(k_{r}a)} - \frac{J_{0}(k_{r}a)}{k_{r}J_{1}(k_{r}a)} \cos k_{z}z \cos k_{z}z' dk_{z}$$
$$= \frac{1}{\pi^3 a^2} \int_{0}^{\infty} \frac{\cos k_{z}z \cos k_{z}z'}{k_{r}^2 J_{1}(k_{r}a)H_{1}(k_{r}a)} dk_{z} \qquad (VIII.5)$$

The radiation impedance of the free-flooding shall differs from that of the solid cylinder in that the pressure on both the outer and inner surface contribute to it

$$Z = -\frac{4\pi a \operatorname{imp}}{U} \int_{0}^{L} \left[ \frac{\phi}{a,z} - \frac{\phi}{i}(a,z) \right] dz \qquad (VIII.6)$$

The two components of this impedance stated in Eq. III.1 can again be separated. The impedance associated with Robey's mathematical model is:

$$Z_{r} = -(4\pi a)^{2} im \rho \int_{0}^{L} \int_{0}^{L} (z-z') dz' dz$$
  
= -(4\pi a)^{2} im \rho \int\_{0}^{L} H(z) dz (VIII.7)

The correction term resulting from fluid flow across the cylindrical boundary rma, |z| > L is

$$Z_{\alpha} = -(4\pi a)^{2} i m \rho \int_{0}^{L} \left[ \int_{L}^{\infty} \alpha(z') \Gamma(z-z') dz' \right] dz \qquad (VIII.8)$$

Since  $\Gamma(z-z')$  is symmetrical in z and z', we can invert the order of integration and, using the definition of H(z) in Eq. VIII.4b, write the unknown component of the radiation impedance as follows:

$$Z_{\alpha} = -(4\pi a)^{2} i \omega_{\beta} \int_{L}^{\infty} \alpha(z) H(z) dz \qquad (\forall III.9)$$

The construction of the functional and the proof that it is stationary for the correct form of the unknown function  $\alpha(z)$  parallels formally the proof given in Eqs. V.8 to 16 for the solid cylinder. The relation between the unknown impedance  $Z_{\alpha}$  and the functional  $J[\alpha]$ , Eq. V.7, is also applicable. The variational solution of the free-flooding thin-walled shell is therefore formally identical with that of the solid cylinder provided the definitions of  $\Gamma(z-z')$  and H(z) given in Eqs. VIII.4a and b are used, instead of the corresponding definitions, Eq. IV.11 and 8, respectively, which apply to the solid cylinder case. We will see in the next section that this parallel does not hold when we assume a realistic free-flooding transducer or "squirter" whose wall thickness is not negligible.

The expression for the for field potential  $\hat{*}_0$  is still given by Eqs. VII.8 and 9. The expression for the potential  $\hat{*}_1$  is somewhat different, because of the contribution of the region |z'| < L which is absent in the case of the solid cylinder:

$$\Phi_{1}(r,z) = \frac{1}{2\pi} \int_{-\infty}^{\infty} u(z') \int_{-\infty}^{\infty} \frac{J_{0}(k_{r}r)}{k_{r}J_{1}(k_{r}a)} \exp[ik_{z}(z-z')] dk_{z} dz' \qquad (VIII.10)$$

The integral over  $dk_z$  is given in Appendix A, Eq. A.14. The comments used in connection with Eqs. VII.1 and 2 apply.

#### IX. The Free-1\_coding Cylindrical Transducer or "Squirter"

.`

When we drop the assumption of a vanishing wall thickness, we must formulate the analysis in terms of three coaxial cylindrical boundaries and their respective radial velocities (Fig. 3).

Each of these cylindrical boundaries is of infinite extent in the z-direction.

The required modifications in the expressions for the potentials, Eqs. IV.5a and VIII.1, are self-evident and involve merely labeling a and u with the appropriate subscripts i or o. There is an equally obvious change in Eq. VIII.6 for the radiation impedance, where the potentials  $\frac{1}{9}_{0}$  and  $\frac{1}{1}_{1}$  must now be multiplied, respectively, by the ratios  $a_{0}/a$  and  $a_{1}/a$ . A non-trivicl change must be introduced in the statement of the continuity condition, which now no longer takes the form  $\frac{1}{2}(a,z)=\frac{1}{9}(a,z)$ . Rather, we assume an incompressible potential in the annular n gion  $a_{1} < r < a_{0}$ . With this assumption the difference between the potentials  $\frac{1}{9}_{1}$  and  $\frac{1}{9}_{2}$  must be matched to the inertia force exerted by the fluid located in this annular region.

$$(1 + \frac{h}{a}) \stackrel{e}{}_{o}(a_{o}, z) - (1 - \frac{h}{a}) \stackrel{e}{}_{i}(a_{i}, z) = -2h\alpha(z)U, \text{ for } |z| > L$$
 (IX.2)

The assumption of an incompressible potential implies that the ratio  $2h/\lambda$  is small. This condition is generally satisfied.<sup>(\*)</sup>

We will now derive relations between the radial velocities on the three cylindrical surfaces defined above. As a result of the assumption of an incompressible potential in the annular region  $a_i < r < a_o$  the velocities in the two semi-infinite regions prolonging the transducer can be derived from the requirement that inflow must balance outflow across the cylindrical boundaries rea,  $a_i$ , and  $a_i$ 

$$\begin{array}{c} u_{1} = u_{3}/a_{1} \\ = u/(1-h/a) \\ u_{0} = u^{3}/a_{0} \\ = u/(1+h/a) \end{array}$$
 for  $|z| > 1$  (IX.3)

For the MRL magnetostictive transducer ring  $2h/\lambda$  is approximately 0.07. For transducers where this assumption is not valid, a compressible potential,  $\hat{\Psi}_a$ , must be constructed for the region  $a_i < z < a_0$ . The three potentials must satisfy two continuity conditions, viz:  $\hat{\Psi}_i(a_i) = \hat{\Psi}_a(a_i)$  and  $\hat{\Psi}_a(a_0) = \hat{\Psi}_0(a_0)$ . Instead of one, two unknown radial velocity distribution over the boundaries, reai and read must be determined from the two simultaneous integral equations arising from the two potential continuity requirements. These equations have been constructed, but it has not yet been verified whether a variational principle can be applied to their solution.

For |z| < L, i.e., in the region of the transducer, different relations must be used. The form of the equations relating these three velocities depends upon whether we are dealing with piezoelectric or magnetostrictive elements. In the case of <u>piezoelectric transducers</u>, the voltage applied to the electrodes located on the inner and outer surfaces of the ceramic ring produces a radial strain  $e_{p}$ . The corresponding circu ferential significance of the mean surface of the element is the result of Poisson coupling:

$$e_{\omega} = -v e_{r}$$
 (IX.4)

where v is Poisson's ratio. This circumferential strain is related to the radial velocity u of the mean surface as follows:

$$e_{g} = u/(-i\omega s)$$
 (D.5)

Noting that  $e_r$  is of opposite sign than  $e_{\varphi}$  and hence than the displacement  $u(-i\omega)^{-1}$ , we find that the velocity of the outer and inner surface are respectively reduced and increased by the radial strain

$$u_{i} = u - h\dot{e}_{r}$$

$$u_{o} = u + h\dot{e}_{r}$$
(IX.6)

where a contraction corresponds to negative  $\varepsilon_{\rm p}$  . Combining these equations we finally have

$$u_{i} = u(1 + \frac{h}{va})$$

$$u_{o} = u(1 - \frac{h}{va}) \qquad (IX.7)$$

In the case of a ring-shaped magnetostrictive transducer the current in the solenoid produces a circumferential strain  $\epsilon_{\varphi}$ . In this case it is the radial strain that results from Poisson coupling:

$$e_r = -v e_{gr}$$
  
= -v u(-102a)<sup>-1</sup> (IX.8)

Combining these equations we now have

$$u_{2} = u(1 + \frac{vh}{a})$$

$$u_{0} = u(1 - \frac{vh}{a})$$
(IX.9)

For the sake of brevity, we define the following coefficients, which tend to unity for small values of h/a:

$$\beta_{i} = (1 \neq \frac{h}{a}) (1 \neq \frac{h}{va}), \text{ for piezoelectric transducers}$$

$$\beta_{i} = (1 \neq \frac{h}{a}) (1 \neq \frac{vh}{a}), \text{ for magnetostrictive transducers} (IX.10)$$

If we now substitute these velocities in the expressions for the potentials used in the boundary condition, Eq. IX.2, we obtain a more complicated integral equation than for the two earlier configurations:

$$\frac{h}{2\pi a} \alpha(z) + \int_{L}^{L} \alpha(z')((1+\frac{h}{a})g_{0}(a_{0},a_{0},z-z') + (1-\frac{h}{a})g_{1}(a_{1},a_{1},z-z')] dz'$$

$$= -\int_{0}^{L} [\beta_{0}(1+\frac{h}{a})g_{0}(a_{0},a_{0},z-z') + \beta_{1}(1-\frac{h}{a})g_{1}(a_{1},a_{1},z-z')] dz' \quad \text{for} |z| > L$$
(IX.11)

As will be seen shortly, the presence of the linear  $t - in \alpha(z)$  outside the integral sign, which makes this into a Fredholm int ... i equation of the second kind, does not interfere with the application of t: variational principle. The presence of the coefficients  $\beta_i$  and  $\beta_o$  in the non-homogeneous term of the integral equation does unfortunately make the application of the variational principle impractical because the stationary potential  $J[c_i]$  which can be constructed, is no longer proportional to  $Z_{\alpha}$ , as stated in Eq. V.7. To make this simple relation applicable, we must assume

$$\theta_1 \approx \theta_0 \approx 1$$
 (DX.12)

The error in this procedure is seen from Eq. IX.10 to be

$$e = \frac{h}{a} (1-v) - \left(\frac{h}{a}\right)^2 v \qquad (IX.13a)$$

for magnetostrictive transducers. For piezoelectric transducers the error is larger

$$e = \frac{h}{a} \left(1 - \frac{1}{v}\right) - \left(\frac{h}{a}\right)^2 \frac{1}{v}$$
 (IX.13b)

The simple variational technique is therefore better suited to magnetostrictive than to piezoelectric transducers. For the magnetostrictive NRL transducer

wing  $(2a=5-7/8 \text{ in.}, 2h=\frac{1}{2} \text{ in.})$ , the error computed from Eq. IX.13a is approximately 6 percent. It may seen inconsistent to introduce assumption IX.12 and not to drop the linear term in  $\alpha(z)$  from Eq. IX.11, and the ratio h/s from terms of the form  $(1 \pm h/a)$ . Further work is necessary to determine whether retention of these terms increases accuracy. Until this is done, we shall retain these terms, because they do not complicate the variational technique. In addition to introducing the assumption stated in Eq. IX.11, we give a new definition of the function  $\Gamma(z-z')$  and of H(z)

$$\Gamma(z-z') = (1 + \frac{h}{a})g_0(a_0, a_0, z-z') + (1 - \frac{h}{a})g_1(a_1, a_1, z-z')$$
 (IX.14a)

$$H(z) = \int_{0}^{L} [(1 + \frac{h}{a})g_{3}(a_{a}, a_{0}, z-z') + (1 - \frac{h}{a})g_{1}(a_{1}, a_{1}, z-z')] dz' \qquad (IX.14b)$$

Furthermore, we make the integral equation, Eq. IX.11, formally into a Fredholm equation of the first kind by the artifice of adding a Dirac delta function  $\delta(z-z')$  to the kernel:

$$\int_{L}^{\infty} \left[ \Gamma(z-z') + \frac{h}{2a\pi} \delta(z-z') \right] \alpha(z') dz' = -H(z), \text{ for } |z| > L \qquad (IX.15)$$

The functional  $J[\alpha]$  is still of the same form as in the two earlier analyzes, Eq. V.9, and the definition of the functional  $A[\alpha]$ , also remains formally the same, Eq. V.10a. The functional  $B[\alpha]$  is, however, different.

$$B[\alpha] = \int_{L}^{\infty} \int_{L}^{\infty} \alpha(z) [\Gamma(z-z') + \frac{h}{2a\pi} \delta(z-z')] \alpha(z') dz dz' \qquad (IX.16)$$

We will now show that the stationary character of  $J[\alpha]$  can be established as before. Setting the increment of the functional equal to 0, we have

$$\delta J[\alpha] = 0 = \frac{2A[\alpha]}{B[\alpha]} \int_{L}^{\infty} f(z) \ \delta \alpha(z) \ dz = - \frac{A^{2}[\alpha]}{B^{2}[\alpha]} \int_{L}^{\infty} \int_{L}^{\infty} \int_{L}^{\infty} [\Gamma(z-z') + \frac{h}{2a\pi} \delta(z-z')][\alpha(z)\delta\alpha(z') + \alpha(z')\delta\alpha(z)]dz \ dz' \qquad (IX.17)$$

The order of integration of the  $\alpha(z)\delta\alpha(z')$  term in the double integral can be inverted.

$$\int_{L}^{\infty} \int_{L}^{\infty} = 2 \int_{L}^{\infty} \int_{L}^{\infty} [\Gamma(z-z') + \frac{h}{2ax} \delta(z-z')] \alpha(z') \delta\alpha(z) dz dz'$$
(IX.18)

From here on, the proof parallels exactly the steps from Eqs. V.13 to 16 and will therefore not be repeated. The radiation impedance correction factor  $Z_{\alpha}$ , is once again formally given by Eq. VIII.8, with  $\Gamma(z-z^{\prime})$  defined in Eq. IX.14s. Thus by setting the coefficients  $\beta$  equal to unity,  $Z_{\alpha}$  can still be expressed in terms of the functional J[ $\alpha$ ] as in Eq. V.7. The component of the radiation impedance associated with Robey's mathematical model is

$$Z_{r} = -i \alpha o (4\pi a)^{2} \int_{0}^{L} \int_{0}^{L} \left[ \beta_{o} (1 + \frac{h}{a}) g_{o}(a_{o}, a_{o}, z - z') + \beta_{i} (1 - \frac{h}{a}) g_{i}(a_{i}, a_{i}, z - z') \right] dz' dz \qquad (IX.19)$$

Even if the coefficients  $\beta$  had not been set equal to unity, a stationary potential  $J[\alpha]$  could mave been constructed with

$$A[\alpha] = \int_{L}^{a} \alpha(z) \{ \int_{0}^{L} [\beta_{0}(1+\frac{h}{a})g_{0}(a_{0},a_{0},z-z')+\beta_{1}(1-\frac{h}{a})g_{1}(a_{1},a_{1},z-z')] dz' \} dz \quad (I.20)$$

instead of the expression in Eq. V.10a. The usefulness of the variational method is however impaired, because the radiation impedance component  $Z_{\alpha}$  does not change in the same manner as A[ $\alpha$ ]:  $Z_{\alpha}$  is given, is before, by Eq. VIII.8 with  $\Gamma(z-z')$  as defined in Eq. IX.14a. It therefore does not involve the coefficients  $\beta$ , whether we set the coefficients equal to unity or not.  $Z_{\alpha}$  is therefore not proportional to A[ $\alpha$ ], Eq. IX.20, and Eq. V.7 relating  $Z_{\alpha}$  and J[ $\alpha$ ] does not apply. Thus even though the variation: method can be used for "squirters" whose walls are too thick to permit setting the coefficients  $\beta$  equal to unity, the unknown coefficients in the trial function  $\alpha(z)$  must be solved for before computing  $Z_{\alpha}$ . Whether such a procedure is competitive with the non-variational solutions presented in Appendix B for thick-walled "squirters" can best be verified empirically after numerical calculations have been performed.

We shall now extend the variational technique to arbitrary non-axisymmetric velocity distributions of the radiating surface.

#### X. Cylinders Vibrating in Longitudinal and in Mon-Axisymmetric Modes

The analysis of the "squirter" and of the solid cylinder can be directly adapted to the case of nonuniform axisymmetric velocity distributions over the radiating surface, by replacing the constant velocity U in the region |z| < L with a zdependent velocity u(z). If u(z) is not symmetrical about z=0, the most convenient approach is to consider the velocity distribution as the sum of a symmetrical distribution ug and of 24 antisymmetric distribution ug. As we are dealing with a linear problem, we can add the corresponding potentials. We first compute the potential associated with  $u_s$  as in the preceding analysis, setting  $\alpha_s(z) = \alpha_s(-z)$ , and using the corresponding Green's functions g, Eq. IV.6, and g, Eq. VIII.2 (the latter in the case of the open-ended cylinder). To this we add the potential resulting from the velocity distribution  $u_{a}$  for which  $G_{a}(z) = -G_{a}(-z)$ . The suitable partial Green's functions are obtained by modify'ng the expressions for g and g, given, respectively, in Eqs. IV.6 and VIII.2, sin kgs sin kgz' being substituted for the product of cosines. We must thus solve two uncoupled integral equations for the two unknown velocity distributions,  $\alpha_{\rm g}$  and  $\alpha_{\rm g}$ .\* Unless we proceed in this fashion, the solid cylinder with arbitrary velocity distribution u(z) gives rise to three different potentials  $\phi_0$ ,  $\phi_{i+}$ ,  $\phi_{i-}$  which in turn result in two distinct boundary conditions corresponding, respectively, to the regions z < L and z > L. The two resulting integral equations will thus be coupled, each involving both unknown velocity distributions,  $\alpha(z < 0)$  and  $\alpha(z > 0)$ .

In the case of a piston or ring vibrating in phase on a finite cylindrical baffle or array, the velocity distribution of the active element is of course constant and hence symmetrical over the midplane (z=0) of the element, but unless this element is centrally located with respect to the baffle or array, the velocity distribution  $\alpha(z)$  will not be symmetrical. In this respect, the present mathematical model differs from Robey's model, in which the potential and the velocity distribution are always symmetrical about the plane of symmetry of the active element.

A configuration of practical interest is that of a solid cylinder whose end caps reciprocate in the axial direction. This situation arises as a result of Poisson coupling with predominantly radial, axisymmetric modes. End cap motion can contribute the major portion of the sound field in the case of the so-called

<sup>&</sup>quot;This procedure will be illustrated in the report dealing with an array of ring transducers (see footnote on p. 1).



accordion modes, which are predominantly longitudinal. To account for an axisymmetric velocity distribution v(r') over the end caps, the following integral is added to the surface integrals in Eq. IV.55:

$$\Delta \Phi_{i\pm}(r,z) = \mp 4\pi \int_{0}^{a} \nabla(r') G_{i}(r,r',z \neq L) r' dr' \qquad (X.1)$$

where v has been taken positive in the positive z-direction. If the cylindrical surface of the radiator is motionless, the potential  $\hat{*}_0$ , in the region r > a is associated entirely with the velocity distribution  $\alpha(z)$  across the two surfaces (r=a.  $z \ge b$ ).

The variational analysic in be extended further, to include an arbitrary nonaxisymmetric velocity distribution

$$u(\varphi,z) = \frac{1}{2\pi} \sum_{m=-\infty}^{\infty} U_m f_m(z) \exp(im\varphi), \text{ for } |z| < L \qquad (X.2)$$

 $U_m$  is a modal velocity amplitude, the maximum value of the function  $f_m(z)$  being unity. To each Fourier component  $U_m$  of the velocity, corresponds a partial potential  $\phi_m(r,z)$ , the total potential being of the form

$$\hat{\mathbf{e}}(\mathbf{r},\mathbf{z}) = \sum_{\underline{f} \neq \underline{a}}^{\infty} \hat{\mathbf{e}}_{\underline{n}}(\mathbf{r},\mathbf{z}) \exp(\mathbf{i}\mathbf{n}\mathbf{x}) \qquad (X.3)$$

The partial potentials are obtained from Eqs. IV.5 and Eq. VIII.1, by substituting in place of the exisymmetric Green's functions given in Eqs. IV.1 and 2 the following.

$$G_{oct}(r,z,\phi-\phi',z-z') = -\frac{1}{\frac{1}{4\pi^2 a}} \exp[i\pi(\psi-\phi')] \int_{-\infty}^{\infty} \frac{b_{m}(k_{r}r)}{k_{r}H_{z}'(k_{r}a)} \exp[ik_{z}(z-z')]dk_{z}, \text{ for } r>a \quad (X.4)$$

٤

The near field value of this integral has been evaluated by Greenspon and Shervan.<sup>6</sup> Its asymptotic far field valus is given by Laird and Cohen:<sup>5</sup>

$$G_{om}(R,\theta,\phi-\phi') = \frac{1 \exp(ikR)}{2\pi^2 ak R \sin\theta} \frac{\exp[im(r_{-}\phi'+\pi/2)]}{H'_{a}(ka \sin\theta)}, \text{ for large } R \quad (X.5)$$

_	4	
-	ſ	۰.
-	۰.	٠
-		

The corresponding Green's function in the cylindrical region is derived in Appendix A, Eq. A.6:

$$G_{in}(r,u,z-z',\varphi-\varphi') = \frac{1}{k_x^2} \exp[in(\varphi-\varphi')] \int_{-\infty}^{\infty} \frac{J_{E}(k_r r)}{k_r J_{R}(k_r a, -\infty)} \exp[ik_z(z-z')] dk_g, \text{ for } r$$

The integral is evaluated in Eq. A.13. Each potential  $\frac{4}{2n}$  is the sum of two compoments: a potential  $\frac{4}{2n}$  casociated with the known modal velocity distribution  $U_{\text{max}}(z)$  over the radiating surface, and a component  $\frac{4}{2n}$  associated with the unknown modal velocity distribution  $U_{\text{max}}(z)$  in the two regions |z| > L. The former component  $\frac{4}{2n}$  is of course the component computed from Robey's mathematical model of the cylinder prolonged by two semi-infinite cylindrical baffles. The modal impedance associated with the math mode of the radiator can be expressed, as in Eq.111.1, as the sum of Knowy's impedance, associated with  $\frac{4}{2n}$ ;

$$Z_{\rm MF} = -\frac{10000\pi}{U_{\rm m}} \int_{-L}^{L} \phi_{\rm MF}(a,z) f_{\rm m}(z) dz$$

$$Z_{\rm MF} = -\frac{10000\pi}{U_{\rm m}} \int_{-L}^{L} \phi_{\rm MF}(a,z) f_{\rm m}(z) dz \qquad (X.7)$$

This impedance can be used to compute the generalized force associated with radiation loading of the ath elastic mode of the cylinder, and hence the model impedances and natural frequencies of the subme yed cylinder. Model radiation impedances can also be combined to compute the self-radiation impedance of rigid pistons in <u>finite</u> cylindrical befoles.\* Because of the similarity in the form of the Green's functions of the axisymmetric case analyzed in detail in this report and of the non-axisymmetric radiator configurations, it is obvious that the integral equations which  $\alpha_{n}(z)$  must satisfy are of the same form as the integral equations thich define  $\alpha(z)$  in the axisymmetric case. Functionals  $J_{m}(\alpha_{m})$  stationary with respect to the correct function  $\alpha_{m}(z)$  can be constructed and are found to be of the same form as ti functional  $J[\alpha]$  constructed carlier for the axisymmetric radiator. The proof that the impedance  $Z_{m}$  is proportional to  $J_{m}(\alpha_{m})$  for the

<sup>&</sup>quot;The component  $Z_{min}$  of this impedance is computed by Greenspon and Sherman.<sup>8</sup>

<sup>31</sup> 

correct form of  $\alpha_{\rm m}(z)$  parallels the proof in Section 7. The variational technique illustrated in Section VI can therefore be used to evaluate  $2_{\rm MC}$ , and need not te repeated here.

In the case of the solid cylinder, nonrigid vibrating end caps can be accounted for by adding to the expression for the potentials  $\delta_{1\pm} + \alpha$  surface integral over the end caps:

$$\Delta \bar{\Psi}_{i\pm} = \bar{\mp} 2 \sum_{m=-\infty}^{\infty} \int_{0}^{2\pi} (v(r', \psi') G_{in}(r, r', v-\psi', z \bar{\mp} L) r' dr' d\psi' \qquad (X.8)$$

#### Appendix A#

DERIVATION AND EVALUATION OF THE GREEN'S FUNCTION G FOR THE CYLINDRICAL REGION  $\mathbf{r} < \mathbf{a}$ 

#### 1. Construction of the Green's Function

This derivation parallels the construction of the Green's function  $G_0$  for the region r > a given by Robey<sup>6</sup> for Keumann boundary conditions ( $\partial \phi/\partial n$  known,  $\partial G/\partial n$  maje to vanish on boundary) and by Papas<sup>18</sup> for Dirichlet boundary conditions ( $\phi$  known, G made to vanish on boundary). The Green's function associated with the mth Fourier component of the velocity distribution in  $\phi$  can be expressed in terms of an inverse Fourier transform in (z-z'):

$$G_{ig}(\mathbf{r},\mathbf{r}',\mathbf{\phi}-\mathbf{\phi}',z-\mathbf{r}') = \frac{\exp[i\pi(\mathbf{\phi}-\mathbf{\phi}')]}{4\pi^2} \int_{-\infty}^{\infty} G_{i\pi}(\mathbf{r},\mathbf{r}',\mathbf{k}_z) \exp[i\mathbf{k}_z(z-z')] d\mathbf{k}_z \qquad (A.1)$$

The Green's function satisfies, by definition, the non-homogeneous Helmholtz equation expressed in cylindrical coordinates. Consequently, the transform of the Green's function satisfies the following equation:

$$\left[\frac{1}{r}\frac{\partial}{\partial r}\left(r\frac{\partial}{\partial r}\right) + k^{2} - k_{2}^{2} - \frac{m^{2}}{r^{2}}\right]G_{m}(r,r',k_{z}) = -\frac{\delta(r-r')}{r} \qquad (A.2)$$

The steps leading from the non-homogeneous Helmholtz equation to Eq. A.2 are presented in detail in reference 11. Except when r=r', Eq. A.2 is of the form of Bessel's equation. A suitable solution to this equation must be regular when r < vanishes (r < and r > are, respectively, the smaller and the larger of thequantities r and r'). The solution of Eq. A.2 must therefore contain only Besselfunctions of argument proportional to <math>r <; Neumann or Hankel functions can only have arguments proportional to r > or s. A combination of cylinder functions which satisfies these conditions, and whose radial derivative  $\partial \sigma / \partial r'$  vanishes on

<sup>&</sup>lt;sup>\*</sup>This material is included here because it does not appear to be available in the literature. Appendix A is condensed from a Harvard Acousticr Research Laboratory Memorendum.<sup>17</sup> Amalytical details and proofs which had to be omitted to keep the length of this report within resson, can be found in reference 1?.

the cylindrical boundary r > = r', is of the form:

$$G_{\underline{i}\underline{m}}(\mathbf{r},\mathbf{r}',\mathbf{k}_{z}) = A_{\underline{n}}J_{\underline{n}}(\mathbf{k}_{r}\mathbf{r} <)[H_{\underline{n}}'(\mathbf{k}_{r}a)J_{\underline{n}}'(\mathbf{k}_{r}\mathbf{r} >) - H_{\underline{n}}(\mathbf{k}_{r}\mathbf{r} >)J_{\underline{n}}'(\mathbf{k}_{r}a)]$$
(A.3)

The coefficients  $A_{\underline{N}}$  are determined from the equation defining the discontinuity of the first derivative of the Green's function:

$$r \frac{\partial}{\partial r} G_{im}(r, r', k_z) \begin{cases} r=r'+\\ = -1 \\ r=r'- \end{cases}$$
(A.4)

After some transformations the coefficient  $A_m$  is found to be

$$A_{m} = \frac{-\pi i}{2J_{m}'(k_{r}^{a})}$$
(A.5)

When we substitute this coefficient in Eq. A.3, set r' = r > = a and r = r <, and use the Wronskisn relation between  $H_m(k_r a)$  and  $J_m(k_r a)$ , we finally obtain the following expression for the Green's function

$$G_{im}(r,a.\phi-\phi',z-z') = \frac{\exp[im(\phi-\phi')]}{4\pi^2 a} \int_{-\infty}^{\infty} \frac{J_m(k_r r)}{k_r J_m'(k_r a)} \exp[ik_z(z-z')] dk_z \qquad (A.6)$$

For the axisymmetric case, we bave  $J_m'=J_0'=-J_1$ , which yields the Green's function given in Eq. IV.2. We will follow the notation used in the body of the report, whereby the subscript m is omitted when m=0, i.e., in what follows,  $G_i$ , I,  $R_n$  and  $k_n$  indicate, respectively,  $G_{i0}$ ,  $I_0$ ,  $R_{0n}$  and  $k_{0n}$ .

#### 2. Evaluation of the Inverse Fourier Transform

•

and for a

We will not evaluate the infinite integral in Eq. A.6. For the purpose of analysis the wave number is assumed to have a small imaginary component

$$\mathbf{k}_{\mathbf{z}} = \mathbf{\xi} + \mathbf{i}\mathbf{\eta} \tag{A.7}$$

The integration in Eq. A.6 will be performed in the complex plane along a closed counterclockwise contour including the real axis and a half circle of infinite radius (Fig. 4). By the residue theorem, <sup>19</sup> the value of the contour integral is  $2\pi i$  times the sum of the n residues at the poles  $k_{min}$  of the integrand.

$$\int_{n}^{\infty} I_{m} dk_{z} = 2\pi i \sum_{n}^{R} R_{mn+}, \text{ for } z-z' > 0, \text{ and } k_{mn} \text{ in upper half-plane}$$
$$= -2\pi i \sum_{n}^{R} R_{mn-}, \text{ for } z-z' < 0, \text{ and } k_{mn} \text{ in lower half-plane} \qquad (A.8)$$

where  $I_m$  stands for the integrand in Eq. A.6. This contour integral equals the integral along the real axis if the contribution of the half-circle vanishes. To achieve this condition the integrand must vanish as  $k_z$  tends to infinity, i.e.,  $\exp[ik_z(z-z')]$  in the integrand must decrease exponentially with increasing  $k_z$ . Hence

- $\eta > 0$ , for z-z' > 0 (Integration in upper half-plane)
- $\eta < 0$ , for z-z' < 0 (Integration in lower half-plane) (A.9)

Like  $k_z$ , k can be assumed to be complex. Its infinitesimal imaginary component can be associated with viscous losses in the acoustic medium, if a physical interpretation is desired. The complex quantity +k lies just above the real axis, and -k just below it. The contours of integration are then as shown in Fig. 4.

We will first evaluate the axisymmetric Green's function. The integrand I has poles at  $k_r=0$ , i.e., at  $k_z=\pm k$ . Taking the asymptotic expression of the Bessel functions  $J_0$  and  $J_0'$  as their argument tends to zero, we find that the integrand tends to

$$I(k_{z}) \rightarrow \frac{2 \exp[ik_{z}(k_{z}-z')]}{a^{2}(k_{z}+k)(k_{z}-k)}, a_{z} k_{z} \rightarrow \frac{1}{k}$$
(A.10)

The two simple poles at  $k_z = \frac{1}{2} k$  give rise to the following residues

$$R_{0+} = \frac{\exp[ik(z-z')]}{ka^2}, \text{ for } z-z' > 0$$

$$R_{0-} = \frac{-\exp[ik(z-z')]}{ka^2}, \text{ for } z-z' < 0 \quad (A.11)$$

Other poles, all of them simple, occur at the roots  $(k_n^a)$  of  $J'_O(k_r^a)$ . The corresponding residues are

$$R_{n_{*}^{\pm}} = \frac{t2 \exp[\pm i(k^{2}-k_{n}^{2})^{\frac{3}{2}}(z-z')]}{a^{2}(k^{2}-k_{n}^{2})^{\frac{1}{2}}} \frac{J_{0}(k_{n}r)}{J_{0}(k_{n}a)-J_{2}(k_{n}a)}$$
(A.12)

2	
٩	7
~	-

Because

3

$$R_{n+}(z-z') = -R_{n-}(z'-z)$$
, (A.13)

the integral can be stated as follows, without regard for the relative magnitude of z and z':

$$\int_{-\infty}^{\infty} I dk_{z} = \frac{2\pi i}{a} \left[ \frac{\exp(ik|z-z'|)}{ka} + 2 \sum_{n=1}^{\infty} \frac{\exp[i(k^{2}-k_{n}^{2})|z-z'|]}{(k^{2}-k_{n}^{2})^{\frac{1}{2}}a} \frac{J_{0}(k_{n}r)}{J_{0}(k_{n}a)-J_{2}(k_{n}a)} \right] (A.14)$$

For all roots  $k_n^a$  of  $J'_0$  valch exceed ka, the terms under the summation sign decay exponentially with increasing |z-z'|. For higher order roots, the terms under the summation sign are proportional to

$$\frac{\exp(-n\pi |z-z'|/s)}{n(ra)^2}, \text{ for n large}$$
 (A.15)

The series expression in Eq. A.14 is thus seen to be convergent except for z=z', which fulfills the requirement of a Green's function.

The convergence of the Green's function as |z| tends to infinity is not spherical, but relies on the small imaginary component of the wave number k. The reason is that we have constructed a Green's function suitable for cylindrical region, viz. a circular pipe, where only viscous losses, but no spreading losses occur. The potential  $\phi_i$  does, however, vary as  $|z|^{-1} \exp(ikz)$  for large |z|, because the radial velocity  $\alpha(z)$  gives rise to a net outflow of acoustic energy from the region r < a (see Eqs. VI.1 to 6, and comments following Eq. VII.2).

We now turn to the evaluation of the non-axisymmetric Green's function. At  $k_{z} = \pm k$ , for m > 0 the integrand tends to

$$I_{\underline{m}} \rightarrow \frac{(r/a)^{\underline{m}}}{\underline{m}} \exp[ik_{z}(z-z')] \quad as k_{z} \rightarrow \pm k \qquad (A.16)$$

There are therefore no poles at  $k_2 = \frac{1}{2} k$ , only for m=0. For m≠0, all the residues are associated with the roots of  $J'_m$ :

$$R_{mn}t^{*} t \frac{4\exp[ti(k^{2}-k_{nn}^{2})^{\frac{1}{2}}(z-z')]}{a^{2}(k^{2}-k_{mn}^{2})^{\frac{1}{2}}} \frac{J_{m}(k_{mn}r)}{2J_{m}(k_{mn}a)-J_{m-2}(k_{mn}a)-J_{m+2}(k_{mn}a)} \text{ for } m\neq 0 \quad (A.17)$$

~	-
2	•
•	
~	-
-	

Once again the integrand can be expressed without regard for the sign of (z-z'):

$$\int_{-\infty}^{\infty} I_{m} dk_{z} = \frac{\theta_{\pi 1}}{a^{2}} \sum_{n=1}^{\infty} \frac{\exp[i(k^{2}-k_{mn}^{2})^{\frac{1}{2}}|z-z'|]}{(k^{2}-k_{mn}^{2})^{\frac{1}{2}}} \frac{J_{m}(k_{mn}a) - J_{m-2}(k_{mn}a) - J_{m+2}(k_{mn}a)}{2J_{m}(k_{mn}a) - J_{m-2}(k_{mn}a) - J_{m+2}(k_{mn}a)}$$
for m/O (A.18)

The higher order terms are again found to be proportional to the expression in Eq. A.15, and thus to conform to Green's function requirement by converging for  $z \neq z'$ . A proof was given in ref. 17 of the fact that even though the function  $(k^2 - k_z^2)^{\frac{1}{2}}$  has a branch cut in the region  $|k_z| \leq k$  of the real axis, the integrand in Eq. A.6 does not have branch points at any of its poles.

37

#### Appendix B

NON-VARIATIONAL TECHNIQUES FOR SOLVING THE "SQUIRTER" INTEGRAL EQUATION, EQ. IX.11

In Section IX it was shown that the variational technique developed for the solid cylinder is applicable to the free-flooding cylinder only when the coefficients  $\beta_i$  and  $\beta_o$ , Eq. IX.10, can be set equal to unity, i.e., when the ratio of wall thickness to radius 2h/a is small. It was also pointed out that when the ratio of wall thickness to accustic wavelength  $2h/\lambda$  is not small enough to make the compressibility of the fluid annulus in the region  $a_i < r < a_o$  negligible, a complicated analysis involving two coupled simultaneous integral equations must be used. The purpose of this Appendix is to present a technique for dealing with a "squirter" for which the ratio 2h/a is not small enough to permit setting the coefficients 6 in Eq. IX.10 equal to unity, even though the corresponding ratio  $2h/\lambda$  is sufficiently small to allow us to ignore the compressibility of the fluid in the annulær region. The most straightforward approach is to evaluate the unperturbed potentials, which are obtaized by setting  $\alpha(z) = 0$ , and to use these potentials to compute a perturbation solution of  $\alpha(z)$  from Eq. IX.2:

. -

$$\alpha^{(0)}(z) = -\frac{2\pi a}{h} \int_{0}^{L} [\beta_{0}(1+\frac{h}{a})g_{0}(a_{0},a_{0},z-z') + \beta_{1}(1-\frac{h}{a})g_{1}(a_{1},a_{1},z-z')] dz' \qquad (B.1)$$

The perturbation solution of the impedance correction factor  $Z_{\alpha}$  is obtained by substituting this expression for  $\alpha(z)$  in Eq. VIII.8, with  $\Gamma(z-z')$  as defined in Eq. IX.14. The far field potentials can of course also be obtained in a straightforward fashion by substituting  $\alpha^{(0)}(z)$  in Eq. VII.8 and 9, for  $\phi_0$ , and Eq. VIII.3 ior  $\phi_4$ .

The perturbation solution can be improved by iteration as follows: One substitutes  $\alpha^{(0)}(z)$  for  $\alpha(z')$  in the z'-integral in Eq. IX.11 and solves for  $\alpha(z)$ . This amounts to solving Eq. IX.1 for  $\alpha(z)$  using the perturbation solutions of  $\tilde{*}_0$ and  $\tilde{*}_i$ . If this iteration process is repeated p times, one finds that the pth iterste of  $\alpha(z)$  is related to the (p-1)th iterate as follows:

$$u^{(p)}(z) = -\frac{2\pi a}{h} \left\{ \int_{0}^{L} \left[ \beta_{0}(1+\frac{h}{a})g_{0}(a_{0},a_{0},z-z') + \beta_{1}(1-\frac{h}{a})g_{1}(a_{1},a_{1},z-z') \right] dz' + \int_{L}^{\infty} \alpha^{(p-1)}(z') \Gamma(z-z') dz' \right\}$$
(B.2)

where  $\Gamma(z-z')$  is defined in Eq. IX.14a.

We will now present a finite-difference procedure for solving the integral equations. Instead of requiring that the integral equation, Eq. IX.11, be satisfied for all values of z in the region |z| > L, we satisfy it at a finite number of points  $z_n = z_1, z_2, z_3, \ldots, z_N$  separated by intervals  $2d_n$ . These intervals should be selected smaller in regions close to the transducer extremities, which make a more important contribution to the potentials than more distant regions. Furthermore, we assume that the unknown function  $\alpha(z)$  has a constant value  $\alpha_n$  in each interval  $(z_n - d_n) < z < (z_n + d)$  and varies discontinuously from one interval to the next. We thus arrive at a set of N simultaneous equations in  $\overline{n}$  unknown quantities  $G_n$ :

$$\begin{bmatrix} \frac{h}{2\pi a} + 2d_{1}\Gamma(0) & 2d_{2}\Gamma(z_{1}-z_{2}) & \dots & 2d_{N}\Gamma(z_{1}-z_{N}) \\ 2d_{1}\Gamma(z_{2}-z_{1}) & \frac{h}{2\pi a} + 2d_{2}\Gamma(0) & \dots & \dots \\ \dots & \dots & \dots & \dots & \dots & \dots \\ 2d_{1}\Gamma(z_{N}-z_{1}) & \dots & \dots & \frac{h}{2\pi a} + 2d_{N}\Gamma(0) \end{bmatrix} \begin{pmatrix} \alpha_{1} \\ \alpha_{2} \\ \vdots \\ \alpha_{N} \end{pmatrix} = \begin{pmatrix} F(z_{1}) \\ F(z_{2}) \\ \vdots \\ F(z_{N}) \end{pmatrix}$$

where

$$P(z_{n}) = \int_{0}^{L} \left[\beta_{0}(1+\frac{h}{a})g_{0}(a_{0},a_{0},z_{n}-z') + \beta_{1}(1-\frac{h}{a})g_{1}(a_{1},a_{1},z_{n}-z')\right] dz'$$
(B.3)

The two Green's functions which enter into the linear combination  $\Gamma(z-z')$ , Eq. IX.14e. have a pole at z=z'. The diagonal terms in the above matrix do not, however, display a singularity, since they are equivalent to an integral of  $\Gamma(z-z')$  over z', which, like the potential, is well behaved.

Solving this set of equations for the values of  $\alpha_n$  we can compute the radiation impedance component  $Z_n$  from Eq. VIII.8

$$Z_{\alpha} = -2(4\pi s)^{2} i \omega_{p} \sum_{n=1}^{N} d_{n} \alpha_{n} \int_{0}^{L} \Gamma(z-z_{n}) dz$$
 (B.4)

Robey's impedance component,  $Z_{p}$ , is given in Eq. IX.19.

•

When applying the finite-difference method to the non-axisymmetric velocity distributions discussed in Section X, it is not necessary to construct a twodimensional grid of points  $(z_n, \varphi_p)$  over the two cylindrical surfaces r=a, |z| > L. Rather, a one-dimensional set of finite-difference equations in z, of the form of Eq. B.3 applies to each modal velocity distribution  $\alpha_m(z)$  essociated with the non-axisymmetric Green's functions, Eq. X.3 and 5.

In conclusion, it is recalled (end of Section IX) that a variational solution is applicable, even when 2h/a is not small, but that  $Z_{\alpha}$  mannot be obtained directly from the functional J[ $\alpha$ ], without also solving for the unknown coefficients in the trial function  $\alpha(z)$ .

(a) <u>Finite-Difference Calculation</u> (Baron, Matthews and Bleich,<sup>3</sup>

(Baron, Matthews and Bleich,<sup>3</sup> Chen and Schweikert<sup>3</sup>)

Grid of point-sources approximates radiating surface. Source strength determined from finite-difference solution of Helmholtz integral equation



(b) Robey's Mathematical Model of the Cylindrical Source

(Luird and Cohen,<sup>5</sup> Robey,<sup>6</sup> Greenspon,<sup>7,8</sup> and Sherman<sup>3</sup>)

Integral equation circumvented by assuming rigid cylindrical baffles r=a, |z| > L, and by constructing Green's function for which  $(\partial G/\partial r')=0$  for r'=0

- 1) Known velocity distribution over radiating surface
- 2) Semi-infinite rigid beffles



(c) <u>Robey's Mathematical Model of "Squirter"</u> (Robey<sup>D</sup>)

Same as Fig. 1(t) but plane baffles in regions r > a,  $z = \frac{1}{r} L$ 

(1) Infinite rigid baffles

•



Fig. 1. REVIEW OF PUELISHED AMALYSES OF CYLINDRICAL RADIATORS (See Table 2 on p. 3)



1:



1) Known velocity distribution of the radiating surface (-L < z < L, rms)

2) Unknown velocity distribution u(z) satisfies integral equation on surfaces r=s, |z| > 2 [z-dependent phase shift of  $\alpha(z)$  is not indicated]

Radiation impedance =  $Z_r + Z_{\alpha}$ where  $Z_r$  = impedance computed from Robey's mathematical model, Fig. 1b  $Z_{\alpha}$  = correction associated with unknown velocity distribution  $\alpha(z)$ 

Variational principle: for correct  $\alpha(z)$ ,  $Z_{\alpha} \propto J[\alpha]$ 

$$\frac{\delta J[\alpha]}{\delta \sigma} = 0$$









\* 4

د\*

. ф

•

رتج

C.2

#### References

- See, for example, P. M. Morse and H. Feahbach, <u>Methods of Theoretical</u> <u>Physics</u>, Vol. 1, New York: McGraw-Hill Book Company, Inc. 1953, pp. 504-811.
- N. L. Baron, A. T. Matthews, and H. H. Bleich, Forced Vibrations of an Elastic Circular Cylindrical Eody of Finite Length Subserged in an Accustic Fluid, OFR Contract Nonr-3454(00) FHM, Tech. Rept. No. 1, June 1962.
- L. A. Chen and D. G. Schweikert, "Sound Rediation from an Arbitrary Body," J. Acoust. Soc. Am. <u>35</u>, 1626 (1963).
- 4. N. G. Parke, III and W. Williams, Jr., <u>Mathematical Methods in Transducer</u> <u>Field Theory: The Finite Cylinder</u>, Parke Mathematical Laboratories, Inc. Final Report, U.S. Navy Underwater Sc. and Laboratory Contract N140(70024)-72788B, December, 1962.
- 5 D. T. Laird and H. Cohen, "Directionality Patterns for Accustic Radiation from a Source on a Rigid Cylinder," J. Acoust. Soc. Am. 24, 46 (1952).
- D. H. Robey, "On the Radiation Impedance of an Array on Finite Cylinders," J. Acoust. Soc. Am. <u>27</u>, 706 (1955).
- J. E. Greenspon, "An Approximation to the Remainder of Robey's Reactive Impedance Integral," J. Acoust. Soc. Am. 33, 1428 (1961).
- J. E. Greenspon and C. H. Shersan, "Mutual Impedance and Near Pield Pressure for Pistons on a Cylinder," J. Accust. Boc. Am. <u>36</u>, 148 (1964).
- D. H. Robey, "On the Contribution of a Contained Viscous Liquid to the Acoustic Impedance of a Radially Vibrating Tube," J. Acoust. Soc. Am. <u>27</u>, 24 (1955).
- D. H. Robey, "On the Radiation Impedance of a Liquid-Filled Squirting Cylinder," J. Acoust. Soc. Am. <u>27</u>, 711 (1955).
- H. Levine and J. Schwinger, "On the Radiation of Sound from an Unflanged circular Pipe," Phys. Rev. 73, 3<sup>p</sup> (1948).
- 12. This is a representative chronological listing of variational solutions of diffraction problems (acoustic, electromagnetic and optical):
  - (a) H. Levine and J. Schwinger, "On the Theory of Diffraction by an Aperture in an Infinite Plane Screen, I," Phys. Rev. 74, 950 (1948); and II, Phys. Rev. 75, 1423 (1949). (A surgary of this paper is given by B. B. Baker and E. T. Copson, The Homatical Theory of Raygens' Principla, Onford: Charendon Press, 2nd ed., 1953: Section 6.2: "The Variational Principle of Levine and Schwinger."
  - (b) J. W. Miles, "On Diffraction Through & Circular Aperture," J. Acoust. Soc. An. <u>21</u>, 140 (1949).
  - (c) H. Levine, "Variational Principles in Acoustic Diffraction Theory," J. Acoust. Soc. Am. <u>20</u>, 48 (1950).

- (d) H. Levine and J. Schwinger, "On the Theory of ElectromAgnetic Wave Diffraction by an Aperture in an Infinite Plane Conducting Screen," Commun. Pure Appl. Math. 3, 355 (1950).
- (e) J. W. Miles, "On Acoustic Diffraction Through an Aperture in a Plane Screen," Acustica 2, 287 (1952).
- (f) Reference 1, Vol. 2, p. 1516-1517 (acoustic diffraction through an iris diaphragm in a pipe).
- (g) H. Levine, <u>On the Theory of Sound Reflection in an Open-Ended Cylindrical</u> <u>Tube</u>, Acoustics Research Laboratory, Harvard University Tech. Memo No. 32, 1 April 1953. A condensed version can be found in J. Acoust. Soc. Am. <u>26</u>, 200 (1954).
- (h) F. B. Sleator, "Veriational Solution to the Problem of Scalar Scattering by a Prolate Spheroid," M.I.T., Math. and Phys. (January 1960).
- (i) P. M. Morse and K. V. Ingard, "Linear Acoustic Theory," <u>Handbuch der Physik</u>, Vol. XI/1, Berlin: Springer Verlag 1961. Sections 29 and 30, (scattering from discs with arbitrary surface impedance, in free space and on enclosures).
- 13. Ref. 1. Vol. 2, p. 1136-37.
- J. B. Storer, "The Impedance of an Antenna over a Large Circular Screen" J. Appl. Phys. <u>22</u>, 1058 (1951).
- M. C. Junger, Sound Rediation from a Redially Pulsating Cylinder of Finite Length, Acoustic Research Labortory, Harverd University Internal Report, 24 June 1955.
- See, for example, S. Timoshanko. <u>Vibration Problems in Engineering</u>, New York: D. Van Nostrand Company, Inc., 3rd ed., 1955, pp. 383-385.
- M. C. Junger, <u>Sound Radiation within an Infinite Cylinder for Arbitrary</u> <u>Scurce Distribution on Its Surface</u>, Acoustics Research Laboratory, Harvard University Internal Memoranduz, 20 March 1952.
- C. H. Papas, "Redistion from a Transverse Slot in an Infinite Cylinder," J. Math. Phys. <u>28</u>, 227 (1950).
- See, for example, R. V. Churchill, <u>Introduction to Complex Variables and</u> <u>Applications</u>, New York: McGraw-Hill Book Company, Inc., 1945, Chapter VVI.

#### DISTRIBUTION LIST

Chief of Naval Research Department of the Navy Washington 25, D. C. Attn: Code 439 (2 copies) Code 468 (2 copies)

Commanding Officer Office of Naval Research Branch Office 495 Summer Street Boston, Mass. 02110

Commanding Officer Office of Haval Research Branch Office John Crerar Library Building 86 E. Eandolph Street Chicago 11, Illinois

Commanding Officer Office of Naval Research Branch Office 207 West 24th Street New York 11, New York

Commanding Officer Office of Naval Research Branch Office Navy \$100, Fleet Post Office New York, New York (5 copies)

Director Naval Research Laboratory Washington, D. C. 20390 Attn: Code 2009, Tech Info Of (6 copies) Code 6200, Mechz Div. Code 6250, Shock and Vibration Code 6250, Structures Code 5500, Sound Division

Defense Documentation Center (20 copies) Cameron Station, Building #5 5010 Duke St., Alexandris, Va. 22314

Office of Technical Services Department of Commerce Washington 25, D. C.

Office of Naval Research Department of the Navy Washingtom 25, D. C. Attn: Dr. F. J. Weyl

Applied Physics Laboratory Johns Hopkins University 8621 Georgia Avenue Silver Spring, Maryland Attu: Dr. W. H. Avery U.S. Eaval Electronics Laboratory San Diego 52, California Attn: Dr. R. J. Christensen

Chief, Bureau of Ships Department of the Havy Washington 25, D. C. Attn: Capt. W. H. Cross, Code 403

Woods Hole Oceanographic Institute Woods Hole, Massachusetts Attn: Dr. P. M. Fye

Chief, Bureau of Maval Weapons Department of the Navy Washington 25, D. C. Attn: Dr. E. S. Lamar, CR-12

U.S. Kaval Ordnance Test Station China Leke, California Attn: Dr. T. Phipps

Office of Naval Research Department of the Navy Washington 25, D. C. Attn: Capt. W. T. Sawyer, Code 406

Chief, Bureau of Ships Department of the Mavy Washingtom 25, D. C. Attn: Dr. G. Sponsler, Code 315

Director, Naval Research Laboratory Department of the Navy Washington 25, D. C. Attn: Mr. P. Waterman, Code 5360

Missile and Syace Division Lockheed Aircraft Corporation Palo Alto, California Attn: Dr. W. F. Whitzore

Special Projects Office (SP-114) Bureau of Maval Weapons Department of the Mavy Washington 25, D. C. Attn: LCDR R. H. Yerbury (Executive Secretary)

Special Projects Office (SP-1142) Bureau of Maval Weepons Department of the Mavy Washington 25, D. C. (3 copies)

Director of Defense Research and Eng. The Pentagym Washington 25, D. C. Attn: Technical Library Chief, Defense Atomic Support Agency The Pentagon Washington 25, D. C. Attn: Tech. Infc. Division Blast and Shock Brancb

Office of the Secretary of the Army The Pentagon Waehington 20, D. C. Attn: Army Library

Chief of Staff Department of the Army Washington 25, D. C. Attn: R and D Division

Office of the Chief of Engineers Department of the Army Washington 25, D. C. Attn: ENG-HL Lib. Br., Adm.Ser. ENG-NB Special Engr.Br., RMD DIV.

Commanding Officer Engineer Research Development Laboratory Fort Belvoir, Virginia

Commanding Officer Satertown Arsenal Watertown, Mass. 02172 Attn: Laboratory Division

Commanding Officer Frankford Arsenal Bridesburg Station Philadelphia 37. Pennsylvania Attr: Laboratory Division

U.S. Army Research Office 2127 Myrtle Drive Duke Station Durham, North Carolina Attn: Div. of Engrg. Sciences

Chicf of Haval Operations Department of the Havy Washington 25, D. C. Attn: Op 07F

Commandant, Marine Corps Headquarters, U. S. Marine Corps Washington 25, D. C.

Commanding Officer USNNOWU Kirtland Air Force Base Albuquerque, New Mexico Attu: Code 20 (Dr. J. N. Brennan) DTMB Underwater Explosion Res. Div. Norfolk Naval Shipyard Portsmouth, Virginia Attn: Mr. D. S. Cohen

Chief, Bureau of Ships Department of the Navy Washington 25, D. C. Attn: Code 335, Tech. Info. Div. Code 345, Hr. F. Vane Code 420 Code 421 Code 423 Code 425 Code 440 Code 442 Code 443 Code 689, Mr. I. Cook Code 1500 Chief, Bureau of Naval Weapons Department of the Navy Washington 25, D. C. Attn: RAAD, Airframe Design DLI-3, Tech. Library R-12, Chief Scientist HMGA, M3 and Airframe Br. RU, ASW Division RRRE, Research Br. Special Projects Office Bureau of Naval Weapons Department of the Havy Washington 25, D. C. Attn: Code SP-001, Chief Scientist Code SP-20, Tech. Director Chief, Bureau of Yards and Docks Department of the Navy Washington 25, D. C. Attn: Code 70, Research

Code E228 Tech. Library Commanding Officer and Director

David Taylor Model Basin Washington, D. C. 20007 Attn: Code 108, Mr. R. T. McGoldrick Code 108B, Dr. M. Strasberg Code 140, Tech. Info. Div. Code 538, Dr. F. Theilheimer Code 563, Mr. A. O. Sykes Code 700, Dr. A. H. Kell Code 720, Mr. E. B. Johnson Code 731, Mr. J. G. Pulos Code 740, Dr. W. J. Sette Code 760, Mr. E. Nooman Code 761, Dr. E. Buchman Code 771, Dr. R. Liebowitz

Commanding Officer and Director U.S. Navy Underwater Sound Laboratory Port Trumbull New London, Connecticut Commander U.S. Maval Ordnance Laboratory White Oak, Haryland Attn: HL, Technical Library D, Technical Director OU, Underwater Weapons RS, Acoustics Division Commanding Officer U.S. Naval Mine Defense Laboratory Panama City, Florida Commander U.S. Naval Air Development Center Johnsville, Pennsylvania Director U.S. Mavy Underwater Sound Ref. Lab. Office of Maval Research P. 0. Box 8337 Orlando, Florida Commanding Officer and Director U.S. Navy Electronics Laboratory San Diego 52, California Attn: Mr. George S. Coleman, Code 2323 Commander Portsmouth Haval Shipyard Portsmouth, New Hampshire Commander Mire Island Maval Shipyard Vellejo, California Director, Materials Laboratory New York Naval Shipyard Brooklyn, New York 11251 Officer-in-Charge Naval Civil Engineering Research

1

and Evaluation Laboratory U.S. Haval Construction Battalion Center Port Hueneme, California

Director Naval Air Experiment Station Naval Air Material Center Naval Base, Philadelphia 12, Penna. Attn: Structures Laboratory

Officer-in-Charge David Taylor Model Basin Underwater Explosion Research Division Norfolk Naval Shipyard Portsmouth, Virginis Attn: Dr. H. M. Schauer Commander U. S. Naval Proving Ground Dahlgren, Virginia

Supervisor of Shipbuilding, USH and Maval Inspector ... Ordiance General Dynamics Corporation Electric Boat Division Groton, Connecticut 06340

Commander Maval Ordnance Test Station, China Take. Attn: Physics Division California Mechanics Division

Commanding Officer and Director U.S. Naval Engineering Experiment Sta. Annapolis, Maryland

Superintendent U.S. Haval Postgraduate School Monterey, California

Commander Air Material Command Wright-Patterson Air Force Base Deyton, Ohio Attn: MCREX-B Structures Division

Commander, WADD Wright-Patterson Air Force Base Ohio Attn: WWRC WWRMDS WWRMDD

Director of Intelligence Headquarters, U.S. Air Force Washington 25, D. C. Attn: P. V. Branch (Air Targets Div)

Commander Air Force Office of Scientific Res. Washington 25, D. C. Attn: Mechanics Division

U.S. Atomic Energy Commission Washington 25, D. C. Attn: Director of Research

.

Director National Bureau of Standards Washington 25, D. C. Attn: Division of Mechanics

National Aeronautics and Space Adm. 1512 H. Street, N. W. Washington 25, D. C. Attn: Chief, Div. of Research Information

Director National Aeronautics and Space Adm Langley Research Center Tangley Field, Virginia Attn: Structures Division

Dr. M. L. Baron Paul Weidlinger, Consulting Engineer 770 Lexington Avenue New York 21, New York

Professor H. H. Bleich Department of Civil Engineering Columbia University 618 Haudd Building New York 27, Kew York

Professor B. A. Boley Department of Civil Engineering Columbia University 618 Mudd Building New York 27, New York

Professor Nicholas J. Hoff Dept. of Aeronautical Engineering Stanford University Stanford, California

Dr. C. W. Horton Defense Research Laboratory University of Texas Austin 12, Texas

Dr. M. A. Brull Engineering Mechanics Division University of Pennsylvania Philadelphia <sup>1</sup>, Pennsylvania

Dr. F. DiMaggio Dept. of Civil Engineering Columbia University 618 Mudd Building Eaw York 27, New York

Professor D. C. Drucker Division of Engineering Brown University Providence 12, Rhode Island Dr. Ira Dyer Bolt Beranek and Newman, Inc. 50 Moulton Street Cambridge, Massachusetts 02138

Professor A. C. Eringen Dept. of Aeronautical Engineering Purdue University Lafayette, Indiana

Dr. Martin Goldberg Special Projects, Applied Mechanics Grussman Aircraft Engineering Corp. Bethpage, Long Islanů, New York

Professor J. N. Goodier Department (, 'Mechanical Engineering Stanford University Stanford, California

Dr. Josh E. Greenspon J G Engineering Research Assoc. 3779 Callaway Avenue Bultimore 15, Maryland

Professor Philip G. Hodge Armour Research Foundation 10 West 35th Street Chicago, Illinois

Professor W. Prager, Chairman Physical Sciences Council Brown University Providence 12, Khode Island

Professor Joseph Kempner Dept. of Aeronautical Engineering and Applied Mechanics Polytechnic Institute of Brooklyn 333 Jay Street Brooklyn 1, New fork

Professor J. M. Klosner Dept. of Aeronautical Engineering and Applied Mechanics Folytechnic Institute of Brooklyn 333 Jay Street Brooklyn 1, New York

Professor E. R. Les Brown University Division of Applied Mathematics Providence 12, Rhode Island

Professor R. D. Mindlin Pept. of Civil Engineering Columbia University 613 Mudd Building New York 27, New York Professor P. M. Kaghdi University of California College of Engineering Berkeley, California 04-0371

Professor N. M. Newmark, Head Department of Civil Engineering University of Illinois Urbana, Illinois

Professor F. Pohle Department of Mathematics Adelphi College Garden City, New York

Woods Hole Oceanographic Institution Woods Hole, Massachusetts

Professor F. V. Romano Dept. of Aeronautical Engineering and Applied Mechanics Polytechnic Institute of Brocklyn 333 Jay Street Brocklyn 1, New York

Professor E. Reiss Institute of Mathazatical Sciences New York University 25 Waverly Place New York 3, New York

Dean V. L. Salerno College of Science and Engineering Fwirleigh Dickinson University Tesneck, New Jersey

Professor A. S. Veletsos Department of Civil Engineering University of Illinois Urbana, Illinois

Brown University Research Analysis Group Providence, Rhode Island

Hydrospace Research Corporation 1749 Rockville Pike Rockville, Maryland

Hudson Laboratories Columbia University 145 Palisades Street Dobbs Ferry, Hew York

Lamont Geological Observatory Columbia University Torre Cliffs Pelicades, New York Orinance Research Laboratory University of Pennsylvania P. O. Box 30, State College, Pa. 16801

Dr. Richard Waterhouse American University Physics Department Washington 16, D. C.

Mr. Stan Lemon Chesapeake Instrument Corporation Shadyside, Maryland

Dr. M. M. Backus Texas Instruments Incorporated 100 Exchange Park North Dallas, Texas

Dr. M. Basin Hughes Aircraft Company P. O. Box 2097 Fullerton, Jalifornia

Mr. John Mahoney OMR Resident Representative Harvard University 473 Broedway Cambridge, Mass. 02138

Naval Ordnance Laboratory White Oak Silver Spring, Maryland Attn: Dr. D. F. Bleil

1. Munice - acoustics .. Sonar transducers -redistion londing 1. Physics - acoustics 2. Bomer treasducers -rediction londing X. C. UMBE N. C. Junger Finite indicts effects an solid cylindrical reservation of a vertained sources and in free-flooding "squirters" are enclosed by means of a vertainon if the lipip for differention problems. The reduction prime set is an of two components: (1) The impedance computed from Rody's submatical acceleration and the sources are accounted with the reduct and set of the impedance set of the second for and set of the impedance set of the reduction problems are accounted at the reduct and set of the impedance set of the reduction problems are accounted at the reduct of (1) the impedance set of the reduction of the reduction of the second for the reduction of the reduction o Tiaite length effects is solid cylindrical to bound sources and in free-floading "equivers" to end the free-floading "equivers" are also been also be a sublement of the free floading to the second sources to the second sources to the second sources to the second sources the second sources the second sources the second sources to the second sources the s Constitue Accountion Associates, Inc. Constitue Accountion Associates, Inc. Constitue, Massachanetts, Peper U-177-45 A VALATIONAL BOLFTICH OF ROLF AND FRE-FLOODED CTLEDDIOL BOLFTICH OF ROLF-45 FLOODED CTLEDDIOL BOLFTICH OF ROLF-45 TIPTTE LEDDIN, by M. C. JAWER, FAST J. March 127-501, Contract Real-5736(00) (Pairs RR 127-501, Contract Real-5736(00) OLUSYDIN Chaptinger Accuational Associates, lac. Chamiltone, Nanoa denserts. Nanoar U-177-45 A VARATIONAL SOUTION OF SOLID AND FAIL-FLOODED CTLINENCOL SOUND AND/ATOND OF TINTE LENDIN, SU SY, A file. File. There like, wid Sy, A file. File. (but AN like - 0.1, Contract Henre 7799(K) cusfigurations. A reboop 0-175-45) estembs this and central free-flooding rink is also applied to low solid cylinder, and to 1. Physics - scoustics C. Somer transducers -redimine longing Scener transducers -redistion longing L. Parstra + anounter N. C. Jager R. C. Jager Fisite langth effects is solid granters soul everyon and in free-flocting "equirers" are restanted by means of a verificiant resultion with perilatis the ferein desidence resultion initiation of a second second second second and the formation of the resultion invisit (3) the impressed as the sea of two compo-ments: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-ingles: (3) the impressed as the sea of two compo-sed as a state of the sea of two impressed as the sea of two implied to induction in the result is also applied to induction in the results. It also applied to induction in the results is a state of the results the results the second the solid of induction induction in the results. It also applied to induction in the results are as it of the results the results the results the results the is attraction of the induction in the results. It also applied to induction in the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the results the results the results the solid of the results the Finite longth effects in solid symmetries to an accurate with street in solid symmetries its ensured by anno of a seriestand localdour with permittels the large schedurger withtich principle for altfractum problem. The relation principle for altfractum problem. The relation represent (1) the laphance computer from heavy's relation of a symmetry and the per-sents: (1) the laphance computer from heavy's relation of the laphance computer from heavy's relation of a strength of the laphance and account for and which has been all account for and effects (2) the laphance and account for and effects (2) the laphance and account for and effects (2) the laphance strength the relation of laber and and the Combridge Accurical Associates, Inc. Comprise, Neuer Beenets. Process P171-19 A VARATIONAL BOUFFORM OF BOULD AND FNEL. FLOODED CILLEDIOL BOUND AND AND FNEL. FLOODED CILLEDIOL BOUND AND AND FNEL. FLOODED CILLEDIOL BOUND AND FNEL. I Brief LEAS, 10 pt. V. CLA. FVEL. The right Accentical Associates, Jac. Therizate, Based charits. March U.T. -A VULTIONL SOUTION OF SOLD AND FAC. FLOODING CRIMENOL ACUTO MOLACOME OF FLOODING CRIMENOL ACUTO FLOODING CRIMENOL ACUTO MOLACOME OF FLOODING CRIMENOL ACUTO FLOOD 5 8 8 <u>ð</u> is also applied to longitudinal me solid cyllader, sed to man-malaym cosfigurations. A subsequent repo V-1 0--2 extends this analysis to Tatis inter-Dooling ringet

ł

I

I