UNCLASSIFIED AD 408994

DEFENSE DOCUMENTATION CENTER

FOR

SCIENTIFIC AND TECHNICAL INFORMATION

CAMERON STATION, ALEXANDRIA, VIRGINIA



. . .

UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.



Contract No. Nonr 140604 Project No. NR 064-429

THE DYNAMIC VISCOPLASTIC EXPANSION OF A CYLINDRICAL TUBE¹

by

Edmund J. Appleby²

Illinois Institute of Technology

Department of Mechanics

May 1963

DOMIIT Report No. 1-24

- 1. Reproduction in whole or in part is permitted for any purpose of the United States Government.
- 2. Visiting Assistant Professor of Mechanics, Illinois Institute of Technology.

THE DYNAMIC VISCOPLASTIC EXPANSION OF A CYLINDRICAL TUBE

by

Edmund J. Appleby

ABSTRACT

The viscoplastic flow of a long thick-walled tube is investigated. The tube is subjected to internal pressure and has its ends restrained from motion in the axial direction. The material of the tube is rigid-viscoplastic and incompressible. The pressure required to produce a specified expansion of the tube is calculated for two examples. In the former the effect of different viscosity coefficients is observed. In the latter example a comparison is made of the effects of perfect plasticity, viscosity and inertia.

I. INTRODUCTION

A study is made of the mechanical behavior of a long hollow circular cylinder, with its end restrained from motion in the axial direction, when it is subjected to internal pressure. The basic problem of plane strain with rotational symmetry, being one of those which most readily yield to treatment in plasticity, is a standard problem in this field, and has been studied by investigators in a variety of different ways for the ideally plastic and the elastic plastic material. Reviews of such solutions have been given by Hill [1], and Prager and Hodge [2]. The present paper considers an ideal material, a viscoplastic Bingham solid [3], which is undeformable until the stresses reach their yielding values, and then under stresses which are in excess of their yielding values, has strain velocities dependent on this stress excess or overstress. The material does not exhibit work hardening, and as it is known that in the fully plastic state volume changes are negligible, it is assumed to be incompressible.

The solution is obtained by using the viscoplastic constitutive equations due to Hohenemser and Prager [4]. The analysis is valid for a general plastic yield condition. An approach by means of a linearized theory of viscoplasticity, in which the flow is specified by Prager's constitutive equation [5], is equivalent.

A general solution is formulated, but in order to simplify the numerical calculations a specific expansion is imposed on the tube in which the interior boundary has a uniform radial acceleration. The pressure-time variations required to maintain this flow for different viscosity coefficients are compared. A different expansion is then imposed, in which the inner

1

radius of the tube expands to one and one half its initial size, beginning and ending with zero velocity. Comparison is made of the effects of perfect plasticity, viscosity and inertia. If the expansion takes place slowly, the effect of inertia on the required pressure is negligible, and the effects of perfect plasticity and viscosity are comparable. On the other hand if the expansion occurs very quickly the effect of inertia becomes comparable with that of viscosity, and the perfectly plastic contribution to the pressure is negligible.

2. BASIC EQUATIONS

Let the space variables be a system of cylindrical co-ordinates r, θ , z in which the z-axis coincides with the axis of symmetry, then the tube is bounded by the cylinders r = a and r = b, where a < b. As the pressure inside the tube increases, the stresses in the material nearest the interior boundary will be the first to reach the yield limit. With further increase of pressure the plastic region will extend until its outer boundary coincides with the tube's outer surface. Until this state is reached, the flow of the plastic innermost region of the tube is restricted by a surrounding rigid region and by the conditions of axially symmetric plane strain on the incompressible material. The whole tube therefore remains rigid, and hence the stresses in the innermost plastic region reach but do not exceed the yield limit. When the whole of the tube becomes plastic, the material is about to flow in an unrestricted manner, since any further increase in the internal pressure will then produce overstress in the material of the tube. The time t is measured from this instant, and the values of a and b for $t \leq 0$ are denoted by a_0 and b_0 .

It is assumed that the tube is sufficiently long to make the stresses and strains independent of the axial co-ordinate, and that at any instant of the flow process each particle of the tube is moving radially outward with a velocity u which depends only on the radial co-ordinate r. The velocity components at a point distance r from the axis at time t are then

$$u_r = u(r,t), u_{\theta} = 0, u_r = 0.$$
 (1)

3

The strain rates are

$$\begin{aligned} \mathbf{\dot{\epsilon}}_{\mathbf{r}} &= \frac{\partial \mathbf{u}}{\partial \mathbf{r}} , \quad \mathbf{\dot{\epsilon}}_{\theta} &= \frac{\mathbf{u}}{\mathbf{r}} , \quad \mathbf{\dot{\epsilon}}_{\mathbf{z}} &= 0, \\ \dot{\dot{\gamma}}_{\theta \mathbf{z}} &= 0 , \quad \dot{\dot{\gamma}}_{\mathbf{zr}} &= 0 , \quad \dot{\dot{\gamma}}_{\mathbf{r}\theta} &= 0 \end{aligned}$$

$$(2)$$

where the dot denotes differentiation with respect to the time.

Since the material is assumed incompressible

$$\frac{\partial u}{\partial r} + \frac{u}{r} = 0.$$
 (3)

This differential equation has the solution

$$u = \frac{\dot{\Psi}(t)}{r} , \qquad (4)$$

in which $\dot{\Psi}(t)$ is an arbitrary function of the time. Expressing the velocity u of a particle, distance r from the axis at time t as dr/dt, and integrating (4), we find that a particle initially at distance r_0 from the axis is at distance $\sqrt{r_0^2 + 2\Psi(t)}$ after a time t. The radial and circumferential strain rates can now be written

$$\dot{\epsilon}_{r} = -\frac{\dot{\Psi}}{r^{2}}, \quad \dot{\epsilon}_{\theta} = \frac{\dot{\Psi}}{r^{2}}. \quad (5)$$

By the rotational symmetry of the flow field, the shearing stresses with respect to the cylindrical co-ordinates are zero, so the only equation of motion which is not identically satisfied is

$$\frac{\partial \sigma_{\mathbf{r}}}{\partial \mathbf{r}} + \frac{\sigma_{\mathbf{r}} - \sigma_{\theta}}{\mathbf{r}} = D \frac{\mathrm{d}u}{\mathrm{d}t} , \qquad (6)$$

where D is the density of the material. When u is replaced by the expression in (4), the above equation can be written

$$\frac{\partial \sigma_{\mathbf{r}}}{\partial \mathbf{r}} + \frac{\sigma_{\mathbf{r}} - \sigma_{\theta}}{\mathbf{r}} = \frac{D}{\mathbf{r}} \begin{bmatrix} \mathbf{\dot{\Psi}} - \frac{\mathbf{\dot{\Psi}}^2}{\mathbf{r}^2} \end{bmatrix}$$
(7)

The boundary conditions throughout the yielding process are

$$\sigma_{r} = -p \text{ at } r = a,$$

$$\sigma_{r} = o \text{ at } r = b;$$
(8)

it being assumed that there is no pressure on the external boundary.

The analysis so far is independent of the constitutive equations, and is the same for all materials.

For any incompressible isotropic material in plane strain Geiringer [6] has shown that a general yield condition can be expressed as a function of the single variable $\sigma_1 - \sigma_2$; σ_1 and σ_2 being the principle stresses in the plane z = const. For the rotationally symmetric problem the principle stresses are σ_r and σ_{θ} , so that the yield function F can always be written in the form

$$\mathbf{F} = |\boldsymbol{\sigma}_{\boldsymbol{A}} - \boldsymbol{\sigma}_{\boldsymbol{a}}| - 2\mathbf{k}, \qquad (9)$$

where k is the yield stress in shear. In particular if $\sigma_{\theta} \ge \sigma_r$, an assumption which may be verified <u>a posterior</u>, (9) can be replaced by

$$\mathbf{F} = \sigma_{\theta} - \sigma_{r} - 2\mathbf{k} \tag{10}$$

The viscoplastic flow rule of Hohenemser and Prager can be written in the general form

$$\lambda \dot{\epsilon}_{ij} = \langle F \rangle \frac{\partial F}{\partial \sigma_{ij}} , \qquad (11)$$

where the notation is defined by:

 $\langle F \rangle = F \text{ if } F \ge 0,$ $\langle F \rangle = 0 \text{ if } F < 0,$

 λ is the viscosity coefficient of the material, and F is the yield function. For the yield function (10), therefore, the only nonvanishing strain rate components for t > 0 are given by

 $\lambda \dot{\epsilon}_{r} = -(\sigma_{\theta} - \sigma_{r} - 2k), \qquad \lambda \dot{\epsilon}_{\theta} = \sigma_{\theta} - \sigma_{r} - 2k.$ (12)

It is interesting to note that since (10) is a linear function, (12) may also be obtained as a special case of the piecewise-linear visco-plastic flow law proposed by Prager [5].

3. SOLUTION

Equations (12) can alternatively be written

$$\frac{\lambda \dot{\Psi}}{r^2} = \sigma_{\theta} - \sigma_r - 2k.$$
(13)

The substitution of equation (13) into the equation of motion (7) gives

$$\frac{\partial \sigma_{\mathbf{r}}}{\partial \mathbf{r}} = \frac{2\mathbf{k}}{\mathbf{r}} + \frac{\lambda \dot{\Psi}}{\mathbf{r}^3} + \frac{D}{\mathbf{r}} \left[\ddot{\Psi} - \frac{\dot{\Psi}^2}{\mathbf{r}^2} \right]. \tag{14}$$

Integrating this equation with respect to r, and employing the boundary condition, $\sigma_r = 0$ when r = b, to evaluate the arbitrary function of time furnishes the stress components:

$$\sigma_{r} = 2k \log \frac{r}{b} + D \ddot{\Psi} \log \frac{r}{b} + \frac{\lambda \dot{\Psi}}{2} \left[\frac{1}{b^{2}} - \frac{1}{r^{2}} \right] - \frac{D \dot{\Psi}^{2}}{2} \left[\frac{1}{b^{2}} - \frac{1}{r^{2}} \right],$$

$$\sigma_{\theta} = 2k \left[1 + \log \frac{r}{b} \right] + D \ddot{\Psi} \log \frac{r}{b} + \frac{\lambda \dot{\Psi}}{2} \left[\frac{1}{b^{2}} + \frac{1}{r^{2}} \right] - \frac{D \dot{\Psi}^{2}}{2} \left[\frac{1}{b^{2}} - \frac{1}{r^{2}} \right].$$

(15)

Since a and b can be expressed in terms of their initial values a_0 and b_0 and the function $\Psi(t)$, equations(15) furnish the stress distribution at any instant in terms of Ψ and its first and second derivatives. The functions Ψ and p must satisfy the boundary condition $\sigma_r = -p$ on r = a. Hence, by using the expression $r = \sqrt{r_0^2 + 2\Psi}$,

$$\begin{split} \ddot{\Psi}D\log\left[\frac{a^{2}_{0}+2\Psi}{b^{2}_{0}+2\Psi}\right] + \frac{\lambda\dot{\Psi}^{(a^{2}_{0}-b^{2}_{0})}}{(a^{2}_{0}+2\Psi)(b^{2}_{0}+2\Psi)} + \frac{\dot{\Psi}^{2}_{D}(b^{2}_{0}-a^{2}_{0})}{(a^{2}_{0}+2\Psi)(b^{2}_{0}+2\Psi)} \\ &= -2p - 2k\log\left[\frac{a^{2}_{0}+2\Psi}{b^{2}_{0}+2\Psi}\right]; \end{split}$$
(16)

the conditions at t = 0 being $\Psi = 0$ and $\dot{\Psi} = 0$, that is $\sqrt{r_0^2 + 2\Psi}$ is initially r_0 and the initial radial velocity is zero.

If the effects of inertia and viscosity had been neglected, the left hand side of equation (16) would be zero, and hence $p = k \log [1 + (b_0^2 - a_0^2)/a^2]$, which is the internal pressure required in the case of an ideally plastic tube to maintain it in a state of unrestricted flow (cf. [2], page 118). If, on the other hand, the inertia term alone is neglected and p is regarded as a constant internal pressure, equation (16) gives a formula for $\hat{\Psi}$ in terms of Ψ which does not contain t explicitly, namely

$$\dot{\Psi} = \frac{2(a_0^2 + 2\Psi)(b_0^2 + 2\Psi)}{\lambda(b_0^2 - a_0^2)} \left\{ p + \log\left[\frac{a_0^2 + 2\Psi}{b_0^2 + 2\Psi}\right] \right\}.$$
 (17)

Integration of equation (17) by means of several substitutions yields

$$\Psi = \frac{b_0^2 - a_0^2 \exp(x)}{2[\exp(x) - 1]} ,$$

where $x = \frac{p}{k} [1 - \exp(kt/\lambda)] + \exp(kt/\lambda) \log \frac{b_0}{a_0^2} .$ (18)

The fields of radial velocity and radial and circumferential strain rate can now be written down from equations (4) and (5). The stress field is obtained by substituting for Ψ and $\dot{\Psi}$ as functions of the time in the following equations:

$$\sigma_{\mathbf{r}} = \mathbf{k} \log \frac{\mathbf{r}^{2}}{(\mathbf{b}^{2}_{0} + 2\Psi)} + \frac{\lambda \dot{\Psi}}{2} \left[\frac{1}{(\mathbf{b}^{2}_{0} + 2\Psi)} - \frac{1}{\mathbf{r}^{2}} \right],$$

$$\sigma_{\theta} = 2\mathbf{k} \left[1 + \frac{1}{2} \log \frac{\mathbf{r}^{2}}{(\mathbf{b}^{2}_{0} + 2\Psi)} \right] + \frac{\lambda \dot{\Psi}}{2} \left[\frac{1}{(\mathbf{b}^{2}_{0} + 2\Psi)} + \frac{1}{\mathbf{r}^{2}} \right].$$

(19)

· ·

· · · ·

9·

4. COMPARISON OF THE VISCOUS EFFECT FOR DIFFERENT VISCOSITY COEFFICIENTS

0

It will now be assumed that p varying with the time, is the internal pressure required to keep the expanding tube flowing unrestrictedly in a certain manner. In order to simplify the problem the condition that the interior boundary of the tube expands with a uniformly increasing speed is imposed. The pressure p(t) which is required to produce this effect is found. The expansion of a is of the form

$$a = a_0 + ct^2, \qquad (20)$$

where c is a positive constant. The function $\Psi(t)$ is found to satisfy

$$2\Psi = c^{2}t^{4} + 2a_{0} ct^{2},$$

giving $\hat{\Psi} = 2ct (a_{0} + ct^{2}),$
and $\ddot{\Psi} = 2c (a_{0} + 3ct^{2}).$ (21)

Omitting details of the calculation, we find that equation (16) now furnishes

$$P(h) = \frac{1}{2} \log \left[1 + \frac{\binom{2}{(\alpha_{0}^{2} - 1)}}{(1 + h^{2})^{2}} \right] + \frac{R}{B} \left\{ \frac{(3h^{2} + 1)}{2} \log \left[1 + \frac{(\alpha_{0}^{2} - 1)}{(1 + h^{2})^{2}} \right] + \frac{(1 - \alpha_{0}^{2})h^{2}}{(\alpha_{0}^{2} + h^{4} + 2h^{2})} \right\} - \frac{1}{2B} \frac{(1 - \alpha_{0}^{2})h}{(\alpha_{0}^{2} + h^{4} + 2h^{2})(1 + h^{2})},$$
(22)

here
$$\alpha_0 = b_0/a_0$$
,
h = t $(c/a_0)^{\frac{1}{2}}$, a dimensionless time,

and P(h) = p(t)/2k, the dimensionless pressure. Also R and B are two dimensionless parameters analogous to the Reynolds number and the Bingham number and defined by

$$R = \frac{D a_0^{c}}{\lambda} \left(\frac{a_0}{c}\right)^{\frac{1}{2}}, \quad B = \frac{k}{\lambda} \left(\frac{a_0}{c}\right)^{\frac{1}{2}}$$

For the present purpose it will be more convenient to utilize less significant parameters defined by

$$R_{p} = \frac{D a_{0} c}{k} , \qquad B_{p} = \frac{\lambda}{2k} \left(\frac{c}{a_{0}}\right)^{\frac{1}{2}}$$
(23)

The pressure can then be written

$$P(h) = \frac{1}{2} \log \left[1 + \frac{\binom{2}{\alpha_{0}^{2} - 1}}{(1 + h^{2})^{2}} \right] + R_{p} \left[\frac{(3h^{2} + 1)}{2} \log \left[1 + \frac{\binom{2}{\alpha_{0}^{2} - 1}}{(1 + h^{2})^{2}} \right] + \frac{(1 - \alpha_{0}^{2})h^{2}}{\binom{2}{\alpha_{0}^{2} + h^{4} + 2h^{2}}} \right] - B_{p} \frac{(1 - \alpha_{0}^{2})h}{\binom{2}{\alpha_{0}^{2} + h^{4} + 2h^{2}}(1 + h^{2})}$$

$$(24)$$

Equation (24) gives the dimensionless pressure required to produce an unrestricted flow of the tube in which the interior boundary has a uniform acceleration. The stress field is then given by:

$$\frac{\sigma_{r}}{2k} = \frac{1}{2} \log \left[1 + \frac{(\rho_{0}^{2} - \alpha_{0}^{2})}{(\alpha_{0}^{2} + h^{4} + 2h^{2})} \right] + R_{p} \left[\frac{(3h^{2} + 1)}{2} \log \left[1 + \frac{(\rho_{0}^{2} - \alpha_{0}^{2})}{(\alpha_{0}^{2} + h^{4} + 2h^{2})} \right] - \frac{h^{2} (h^{2} + 1)^{2} (\rho_{0}^{2} - \alpha_{0}^{2})}{(\alpha_{0}^{2} + h^{4} + 2h^{2}) (\rho_{0}^{2} + h^{4} + 2h^{2})} \right]$$

$$+ \frac{B_{p}h}{(\alpha_{0}^{2} + h^{4} + 2h^{2})} (\rho_{0}^{2} - \alpha_{0}^{2})}{(\alpha_{0}^{2} + h^{4} + 2h^{2}) (\rho_{0}^{2} - \alpha_{0}^{2})}$$
(25)

$$\frac{\sigma_{\theta}}{2k} = 1 + \frac{1}{2} \log \left[1 + \frac{(\rho_0^2 - \alpha_0^2)}{(\alpha_0^2 + h^4 + 2h^2)} \right] + R_p \left[\frac{(3h^2 + 1)}{2} \log \left[1 + \frac{(\rho_0^2 - \alpha_0^2)}{(\alpha_0^2 + h^4 + 2h^2)} \right] - \frac{h^2 (h^2 + 1)^2 (\rho_0^2 - \alpha_0^2)}{(\alpha_0^2 + h^4 + 2h^2) (\rho_0^2 + h^4 + 2h^2)} \right] + \frac{B_{h}h (1 + h^2) (\rho_0^2 + \alpha_0^2 + 2h^4 + 4h^2)}{(\alpha_0^2 + h^4 + 2h^2) (\rho_0^2 + h^4 + 2h^2)}, \quad (25)$$

and the velocity field u is found to satisfy

$$u^{2} = \frac{4(1+h^{2})^{2}}{(\rho_{0}^{2}+h^{4}+2h^{2})} \quad a_{0}ch^{2},$$
(26)

where $\rho_0 = r_0/a_0$.

As estimate of the effect of viscosity on plastic flow of the type specified above was obtained from equation (24). The numerical values used in the investigation were as follows in c. g. s. units: $a_0 = 5, \alpha_0 = 2$, c = 0.1; D = 8.5; and $k = 23.5 \times 981 \times 10^5$, appropriate for a thick walled brass (Zn 30 percent, Cu 70 percent) tube. The viscosity coefficient of the metal in these units would be of order 10^{10} . Two values $\lambda = 5 \times 10^{10}$ and $\lambda = 10^{11}$ were chosen, and the results compared with that for $\lambda = 0$ corresponding to zero viscosity. With the above parameters the second term in (24), the inertia term, becomes comparable with the other terms only in the final stage of yielding, and for all practical purposes is negligible. The variations of the dimension-less pressure with dimensionless time, for the three assigned values of the viscosity, are shown in Fig. 1. It is seen that the viscosity has a considerable effect during the initial stages of the yield process, and that in order to maintain the same yield with greater viscosity, the initial rate of loading must be correspondingly increased. The required pressure attains its maximum and may be allowed to decrease at a rate more nearly comparable with the rate of unloading in the nonviscous case. In each case the required pressure approaches zero asymptotically with increasing time, as might be expected in the absence of fracture, since the thickness of the tube is steadily decreasing.

5. COMPARISON OF PLASTIC, VISCOUS AND INERTIA EFFECTS

It is also of interest to compare the effects of perfect plasticity, viscosity, and inertia on the expanding tube. For this purpose an expansion

$$a = (a_0^{\prime}/4) [5 + \sin(\omega t - \pi/2]$$
 (27)

of the inner radius of the tube is imposed during the time interval t = 0 to $t = \pi/\omega$ When t = 0, $a = a_0$ and da/dt = 0; when $t = (\pi/\omega)$, $a = 3a_0/2$ and da/dt = 0. See Fig. 2. Hence (27) corresponds to an expansion of the inner radius to one and a half times its initial value in a time $T = \pi/\omega$, starting and finishing with zero velocity. It can then easily be shown that

$$\Psi = \frac{a^{2}}{32} \left[5 + \sin\left(\omega t - \frac{\pi}{2}\right) \right]^{2} \frac{a^{2}}{2} ,$$

$$\dot{\Psi} = \frac{a^{2}}{32} \left[5 + \sin\left(\omega t - \frac{\pi}{2}\right) \right]^{2} \frac{a^{2}}{2} ,$$

$$\dot{\Psi} = \frac{a^{2}}{32} \frac{\omega}{32} \sin \left(\omega t - \frac{\pi}{2}\right) + \frac{10a^{2}}{32} \frac{\omega}{32} \cos\left(\omega t - \frac{\pi}{2}\right) ,$$

$$\dot{\Psi} = \frac{a^{2}}{16} \frac{\omega^{2}}{16} \cos \left(\omega t - \frac{\pi}{2}\right) - \frac{10a^{2}}{32} \frac{\omega^{2}}{32} \sin\left(\omega t - \frac{\pi}{2}\right) .$$
(28)

The pressure variation needed to produce expansion (27) can be written as

$$p_{\underline{t}}(t) = P_{\underline{P}}(t) + P_{\underline{I}}(t) + P_{V}(t)$$
, (29)
2k

where $P_{P}(t)$, $P_{I}(t)$ and $P_{V}(t)$ are the contributions due to perfect plasticity, inertia, and viscosity respectively, and given by

$$P_{\rm p}(t) = \frac{1}{2} \log \left[\frac{b_0^2 + 2\Psi}{a_0^2 + 2\Psi} \right] ,$$

$$P_{\rm I}(t) = \frac{D}{4k} \left\{ \frac{\cdots}{\Psi} \log \left[\frac{b_0^2 + 2\Psi}{a_0^2 + 2\Psi} \right] - \frac{\left(\frac{b_0^2 - a_0^2}{a_0^2 + 2\Psi} \right)^2}{\left(\frac{a_0^2 + 2\Psi}{a_0^2 + 2\Psi} \right)^2} \right\} ,$$

$$P_{\rm V}(t) = \frac{\lambda}{2} \frac{\left(\frac{b_0^2 - a_0^2}{a_0^2 + 2\Psi} \right)^2}{\left(\frac{a_0^2 - 2\Psi}{a_0^2 + 2\Psi} \right)^2} .$$
(30)

The numerical values of a_0 , b_0 , D and k are chosen the same as previously. The value of λ is now fixed at 5×10^{10} . By varying ω the expansion can be made to take place in any desired time. Three values of ω were chosen. These correspond to an expansion of the tube in 1 sec., 10^{-6} sec., and 10^{-7} sec. respectively. The pressure variation with time required to produce the expansion was calculated in each case from equations (28) through (30). The dimensionless pressure contributions P_p , P_I , and P_V were plotted against t in each case. When T = 1, corresponding to a slow expansion, inertia has a negligible effect on the pressure required, but the perfectly plastic and viscous contributions to this pressure are of the same order. See Fig. 3. When T = 10^{-6} the required pressure is considerably increased and the perfectly plastic contribution is a negligible part, but the inertia effect begins to be apparent. See Fig. 4. For T = 10^{-7} the inertia term in p(t)/2k has a considerably greater effect (see Fig. 5).

It is noted that Figs. 4, 5 show that a negative pressure is required near the end of the period to produce the required expansion. By referring to equations (14), (15) and (16) it is seen that p can be replaced by a pressure ($p_i - p_e$), where p_i and p_e are an internal pressure and an external pressure respectively. Thus, imposing a negative value of the pressure p is exactly equivalent to an application of external pressure and hence is physically possible.

The order of magnitudes involved in P_V and P_I is apparent from equations (28) and (30). When the time of expansion T is altered by some factor K , the viscous pressure contribution is altered by a factor K^{-1} , and the inertia contribution by a factor K^{-2} ; the perfectly plastic pressure contribution is, of course, independent of T. For the type of problem considered, this shows an interesting comparison over the full range of expansion times of the three effects mentioned. In a slow expansion the flow approximates that of the plastic quasi-static theory in which the strain rates are very small and the inertia effects are neglected. As the speed of expansion increases the viscous effect becomes important and then dominant, whereas the inertia effect is still negligible. For even faster expansions the effect of inertia becomes important and finally predominates.

REFERENCES

- 1. R. Hill, "The Mathematical Theory of Plasticity", Oxford University Press, 1950.
- 2. W. Prager and P. G. Hodge, "Theory of Perfectly Plastic Solids", John Wiley and Sons, Inc., New York 1951.
- 3. "Studies in Mathematics and Mechanics presented to Richard von Mises", Academic Press, Inc., New York, 1954, p. 208.
- 4. K. Hohenemser and W. Prager, "Über die Ansätze der Mechanik der Kontinua", Zeitschrift f. angew. Math. u. Mech., vol. 12, 1932, p. 216.
- 5. W. Prager, "Linearization in Viscoplasticity", Oestereichisches Ingenieur - Archiv, vol. 15, 1961, p. 152.
- 6. H. Geiringer, Encyclopedia of Physics, edited by S. Flügge vol. 6, Springer, 1958, sec. 63.

ACKNOWLEDGMENT

The author is grateful to Professor W. Prager of Brown University and Professor P. G. Hodge, Jr., of Illinois Institute of Technology for helpful discussions and suggestions.

LIST OF FIGURES

Fig. 1. Pressure-Time Curves.

.

.

÷

- Fig. 2. Expansion of Inner Boundary.
- Fig. 3. Pressure Contributions when T = 1 sec.
- Fig. 4. Pressure Contributions when $T = 10^{-6}$ sec.
- Fig. 5. Pressure Contributions when $T = 10^{-7}$ sec.





-

.

INNER BOUNDARY





FIG. 4. PRESSURE CONTRIBUTIONS WHEN T=10⁻⁶ SEC.



DISTNIBUTION LIST FOR UNCLASSIFIED TECHNICAL REPORTS ISSUED UNDER CONTRACT TARK HE 064-429

Chief of Naval Research Chief of Mayal Messard Department of the Navy Washington 25, D. C. Attm: Code 438 Code 463 (2) (1) mading Officer Office of Neval Research Branch Office 495 Summer Street Boston 10, Massachusetts (1) Commanding Officer Office of Naval Research Branch Office John Crerar Library Building anding Officer 86 S. Randolph Stree Chicago 11, Illinois **(1)** manding Officer Office of Naval Research Branch Office 346 Broadway New York 13, N. Y. (1) Commanding Officer Office of Naval Research Branch Office 1030 S. Green Street Pasadena, California (1)Commanding Officer Officer of Naval Research Branch Office 1000 Geary Street Jan Francisco, California (1) Commanding Officer Office of Neval Research Navy #100, Fleet Post Office . (25) New York, N. Y. Director Naval Research Laboratory Nuvel Research Laborator; Jashington 25, D. C. Attn: Tech Info Officer Code 6200 Code 6205 Code 6250 (6) (1) (i) (ii) Code 6260 (1)Arned Services Technical Information Agency Document Service Center Arlington Hall Station Arlington 12, Virginia (10) Office of Technical Jervices Department of Commerce Jashington 25, D. C. (1) Office of the Secretary of Defense Research and Development Division Research and Development The Pentagon Jashington 25, D. C. Attn: Technical Library **(1)** Chief Armed Forces Special Weapons Project The ientagon dashington 25, D. C. Attn: Technical Information Division (2) weapons iffects Division (1) Special Field Projects (1) Blast and Shock Branch (1) Office of the Secretary of the Army The Pentagon Washington 25, D. C. Attn: Army Library (1) Chief of Staff Chief of Starf Department of the Army Washington 25, D. C. Attn: Development Branch (B&D Div) (1) Research Branch (B&D Div) (1) Special Weapons Br.(B&D Div) (1) Office of the Chief of Engineers Department of the Army Washington 25, D. C. Attn: 200-HL Lib. Br., Adm. Ser. Div. 203-WE ing. Div. Civil Works (1) ົຒ

Gommanding Officer Engineer Research Development Laboratory Fort Belvoir, Virginia (1) Office of the Chief of Ordnance Department of the Army Weahington 25, D. C. Attn: Research & Materials Branch (Ord BhD Div.) (1) Office of the Chief Signal Officer Department of the Army Washington 25, D. C. Attn: Engineering & Technical Division (i) Commanding Officer Gommanuing Cilics Watertown Arsenal Watertown, Massachusetts Attn: Laboratory Division (1) Commanding Officer Frankford Arsenal Bridesburg Station Philadelphia 37, Pennsylvania Attn: Laboratory Division (1) Office of Ordnance Research 2127 Myrtly Drive Duke Station Duke Station Durham, North Carolina Attn: Division of Engineering Colonces (1) Commanding Officer Squier Signal Laboratory Fort Monmouth, New Jersey Attn: Components & Materials Branch (1 $\overline{(1)}$ Chief of Naval Operations Department of the Navy Washington 25, D. C. Attn: 0p 37 **(1)** Commandant, Marine Corps He dquarters, U.C. Marine Corps sashington 25, D.C. (1) (i) Chief, Bureau of Ships Department of the Navy Authington 25, D.C. Attn: Gode 312 Gode 376 Gode 377 Code 420 (2) (1) (1) ii) Code 425 Code 442 (2) (2) Chief, Bureau of Mer nautics Department of the Navy Washington 25, D.C. Attn: AV-54 AD-2 (1)RS-7 RS-8 (1) άĵ TD-+2 Chief, Bureau of Ordnance Department of the Navy Washington 25, D.C. Attn: Ad3 (1) (1)(1)(1)Re Reu ReS5 HeS1 (1) ā Ren Chief, Bureau of Yurds and Dock Department of the Mavy Washington 25, D.C. Attm: Codes D-202 (1) (1)(1)(1)(1)(1)D-202.3 D-220 D-222 D-4100 D-440 (1)D-500 **(1)** Commanding Officer and Director David Taylor Model Basin Washington 7, D.C. Atta: Code 140 0000000 600 700 720 725 731 740 (2)

Commender U.S. Maval Ordnance Laboratory White Oak, Maryland Atta: Technical Library (2) Technical Avaluation Dept. (1)

Director, Naterials Laboratory New York Naval Shipyard Brocklym 1, New York (1)Communing Officer & Director U. S. Maval Electronic Laboratory Officer-in-Charge U. S. Navel Civil Engineering Research and Evaluation Laboratory Navel Construction Enttalion Center Port Houseme, California Director, Naval Air Experiment Station Naval Air Material Center Naval Base Navel mass Philadelphia 12, Pennsylvania Atta: Materiale Laboratory (1) Structures Laboratory (1) Officer-in-Charge Underwarer Explosion Research Division Norfolk Naval Shippard Portsmouth, Virginia Attn: Dr. A. H. Keil (2) Commander, U. S. Naval Proving Ground Dahlgren, Virginia (1) Superintendent, Naval Gun Factory Washington 25, D. C. (1) Commander, Naval Ordnance Test Station Invokern, China Lake, California Attn: Physics Division (1) Mechanics Branch (1) Commander, Naval Ordnance Test Station Underwater Ordnance Division 3202 E. Foothill Boulevard l'asadena 8, California Attn: Structures Division (1) Commanding Officer and Director Naval Angineering Experiment Station Annapolis, Maryland (1) Superintendent, Naval Fostgraduate School Montorey, California (1) manuant, Marine Corps Schools Juantico, Virginia Attn: Director, Marine Corps Development Center (1) Attn: Commanding General U.S. Air Force Washington 25, D.C. Attn: kesearch and Development Division (1) Commander, WAUD Wright-Fatt reon Air Forde Base Dayton, Ohio Attn: WWRC WWRMDG (1) (1)WIRNOT (ii) Commander, Air Material Command Wright-Patterson Air Force Base Dayton, Ohio Attn: WDCLR (1)Attn: Structures Div. (1) Commander, U.S. Air Force Institute of Technology wright-Fatterson Air Force Base Dayton, Ohio Attn: Chie Chief,Applied Mechanics Group (1) Group Director of Intel-igence Heudquarters, W.S Air Force Washington 25, D.C. Attn: P.V. Branch (Air Turgets Divisios) (1) Commandur, Air Force Office of Scientific Research Washington 25, D.C. Atta: Hechanics Division (1) Commanding Officer USHNOW Kirtland Air Force Base Albuquerque, New Mexico Atta: Code 20 (1) (Dr. J.N.Brensen) U.S. Atomic Energy Commission Washington 25, D.C. Atta. Director of Research(2) Director, National Bureau of Standards Washington 25, D.G. Atts: Division of Mechanics (1) Engineering Mechanics Sect. (1) Aircraft Structures (1)

Commanduat, U.S.Coust Guard 1300 d. Street, N.d. Washington 25, D.C. Attn: Chief, Testing and Development Division (1) H.S. Maritire Administration General Administration Office Blug. Washington 25, D.C. Attn: Chief, Division of Freliminary Design (1)National Advisory Committee for Adronautics 1512 H. Street, N.J. Jashington 25, D.C. Attn: Loads and Structures Div. (2) Director, Langley Aeronautical Lab. Langley Field, Virginia Attn: Structures Division 10 Director, Forest inclucts Lab. 'n Madison, sisconsin Civil Aeronautics Administration Department of Commerce Washington, 25, D.C. Attn: Chief, wireraft dag. Div. (1) Chief, Airflame & Eq. Br.(1) National Academy of Sciences 101 Constitution Ave. Ashington 25, 0.C. Attn: Technical Director, Committy on Ship's Structural Desi n (1) Recutive Secret ry, Committee on Undersea darfare (1) Legislative Reference Service Library of Congress Jushington 25, D.C. Attn: Dr. 5. Menk rofessor Lynn S. Beedle Fritz angineering Laboratory Lohigh University Bethlehem, Lennaylvania (1)frofuseor R.L. Bisplinghoff Department of Aeronautical Engineering Matsuchusetts Institute of Technology Cambridge 39, Massachusetta (1) Frofessor H.H. Eleich Department of Civil Engineering Columbia University New York 27, New York (1)Professor B.A. Boley Department of Civil Engineering Columbia Univ.rsity

New York 27, New York Professor Augene J. Brunell, Jr. Department of Aeronautical Angineering Frinceton University Frinceton, New Jersey (1)

(1)

Dr. John F. Brahts, Messager Construction Sciences Research Stanford Ressurch Institute 820 Mission Street South Pasadena, California (1)

Professor B. Budiansky Department of Mech. Engineering School of Applied Sciences Marvard University Cambridge 33, Massachusetts (1)

Professor G.F. Carrier Pierce Hall, Marvard University Casbridge 35, Massachusetts (1)

Professor J.E. Cormsk Department of Civil Angineering Colorado State University Fort Colline, Colorado (1 **(1)**

Professor Walter T. Daniels School of Engr. and Architecture Remard University Vachington 1, D.G. (1)

Professor Nerbort Derestovids Department of Givil Ingineering Columnia Waiversity 632 W. 1256 Street New York 22.

.. . •

second of a second second

Professor D.C. Drucker, Chairman Proteiner D.C. Drucker, Calif-Division of Engineering Brown University Providence 12, Rhode Island(1)

Professor A.C. Gringen Department of Aeronautical dag. Purdue University Lafayette, Indians (1)

Professor 4. Flugge Department of Mechanical Eng. Stanford University (1)Stanford, California

krofessor L. d. Goodman Anginiering Experiment Station University of Minnesota Minneapolis, Minnesota

Mr. Murtin Goland, Vice irss. Southwest Research institute 8500 Calebra Road San Antonio, Texas (1)

Professor J.N. loodiar Dept. of Mechanical Ang. Stanford University Stanford, California

Frofessor #.J. Hall Department of Civil Ang. Univ raity of Illinois Urbana, Illinois

Frofessor R.F. Harrington, Head Department of Aeronautical Angineering University of Cincinnati Cincinnati 21, Ohio (1)

ω

Professor M. Hetenyi The Tichnological Institute Northwe tirn University Evanston, Illinois (1)

Profuser Inilip 3. Hodge, Jr. Department of Mechanics Illinois In titute of Fechnology Chicago 16, Illinois(1)

trofessor N.J. Hoff Division of Acronautical Angineering Stanford University Stanford, California (1)

Frofessor W.H. Hoppmann, II Department of Mechanics Rensselser / olytechnic Institute Troy, New York (1)

Professor J. Kempner Dept. of Adronautical Engineering and Applied Mechanics Polytechnic Institute of Broskiyn 333 Jay Street Brooklyn 1, New York **(1)**

Profissor H. Kolsky Providence 12, Rhode Island(1)

Mr. K.H. Koopman, Secretary delding Research Council of the ingine ring Foundation 29 W. 398 Street New York 18, New York (2)

Professor H.L. Longh Department of Theoretical & Applied Mech. University of Illinois (1)

Professor B.J.Lasan, Director Engineering Experiment Station University of Minnesota Minnespolis 14, Minnesota (1)

Professor E.H.Lee Division of Applied Mathematics Brown University Providence 12, Mode Taland(1)

Professor George H. Lee Prirector of Desearch Hesseelser Polytechnic Institute Troy, New York (1)

Mr. N.H.Lomson Southwest Bosenrch Institute ASCO Galebra Boad -An Antonio 6, Teaus

Nr. S.Levy, Manager General Letris Special Projects Dept. 3195 Chestaut Street Philadelphia 4, Pennsylvania (1)

Professor Paul Lieber Geology Department University of California Berkeley 4, California (1)

Professor Joseph Marin, Head Department of Engineering Mechanics The Pennsylvania State University University Fark, Fennsylvania (1)

Professor R.D. Mindlin reconsect with mindlin Department of Civil Engineering Columbia University 632 V. 1250 Street New York 27, Naw York (1)

Professor Paul N. Naghdi Protessor rati n. Anglai Department of Engineering Nechanics University of California Berkeley 4, California

Professor dilliam A. Nach Department of Engineering Mechanics University of Florida (1)Jainesville, Florida

Professor N.M. Newmark, Head Department of Civil Engineering University of Illinois Urbana, Illinois (1)

Frofessor S. Orowan Department of Machanical Engineering Massachusetts Institute of Technology Cambridge 59, Massachusetts (1)

Professor Nicholas Ferrone Angineering Sciume - Department Pratt Institute Brooklyn 5, Now York (1)

Frofessor Aris Shillips Department of Civil Engineering 15 Prospect Street Yale University Now Maves,Connecticut (1)

Professor V. Prager L. Nerbert Hallou Univ. Professor Brown University Providence 12, Scode Island (1)

irofessor & Reissoer Department of Mathematics Massachusetts Institute of Techn logy Cambridge 39, Nassachusetts (1)

Professor N.A. Sadowsky Protocol H.R. Subvery Department of Mechanics Renueslaer Polytechnic Institute Troy, New York (1)

Profussor B.W. Shaffer Department of Machanical Angineuring New York University Unive sity Heights New York, N.Y. (1)

Frofussor J. Stallmayer Department of Civil Ingineering University of Illinois Urbana, Il.inois (1)

Professor Ali Sternberg Provision of Applied Mathematics Brown University Provd ace 12, Rhode I:: 1 nd (1)

Professor A.S.Veleston Department of Civil Engineering University of Illimois Urbann, Illinois (1)

Professor Dama 1 Tale University New Naves, Comm in Tound at Lout **(1)** Project Staff (10)

.

For your future distribution (10)