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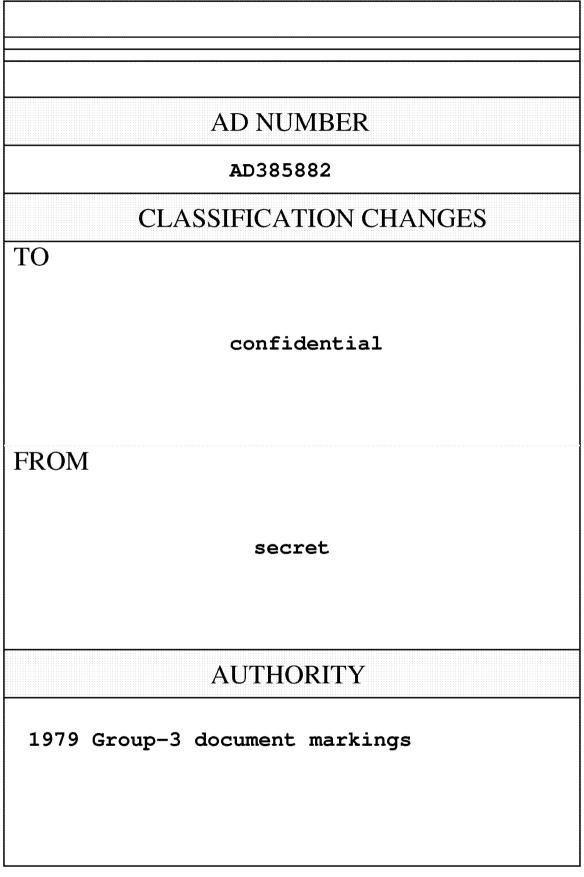
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An Analysis of F-105 Combat Losses in (Out-Country) (Unclassified Title)

> Jerry D. O'Brien Jay M. Meiselman, 2d Lt, USAF

**TECHNICAL REPORT AFFDL-TR-67-118** September 1967

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> Air Force Flight Dynamics Laboratory **Research and Technology Division** Air Force Systems Command Wright-Patterson Air Force Base, Ohio

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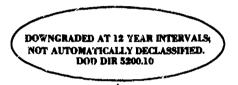
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# An Analysis of F-105 Combat Losses in SEA (Out-Country) (Unclassified Title)

Jerry D. O'Brien Jay M. Meiselman, 2d Lt, USAF



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#### FOREWORD

This report was prepared by the Structures Division of the Air Force Flight Dynamics Laboratory (AFFDL), Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio. This work was performed under Project 1368, "Structural Design Concepts," Task 136814, "Aircraft Vulnerability." Lt Jay M. Meiselman (FDTS) was the project engineer.

This report contains data which has been extracted from the Weapons Systems Evaluation Group (WSEG) Compendium of Aircraft Combat Losses and Damages in Southeast Asia from 1 February 1965 to 31 January 1966 (U), Volume I: Losses, dated July 1966. The subject report is classified SECRET, Group 3, No Foreign Dissemination.

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This report was submitted by the authors July 1967.

This rechnical report has been reviewed and is approved.

FREDERICK C. KRUG, Colonel

Chief, Structures Division

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#### AFFDL-TR-67-118

#### UNCLASSIFIED ABSTRACT

This report contains an analysis of F-105 aircraft losses based on the Weapons Systems Evaluation Group (WSEG) Compendium of Aircraft Losses for the time period of 1 February 1965 to 31 January 1966. The analysis is performed for the purpose of providing an insight into areas such as threat, cause of aircraft loss, and time from initial damage to loss. (In addition to security requirements which must be met, this abstract is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Air Force Flight Dynamics Laboratory (FDTS), Wright-Patterson Air Force Base, Ohio 4543%.)

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#### SECTION I

#### INTRODUCTION

(U) A two-volume compendium containing the damage and losses data was published by the Weapons Systems Evaluation Group (WSEG) to provide a source of information identifying combat data which would be useful for R&D purposes. The information was compiled for the period of 1 February 1965 to 31 January 1966 and was published without being analyzed. The information for the Air Force aircraft losses is limited to North Vietnam and Laos. A case history for each aircraft loss incident is contained in the compendium in the form of an Aircraft Combat Loss Report.

(U) The authors attempt to analyze the WSEG data in order to provide an insight into areas such as threat, cause of loss, and time from damage to loss. Due to many unknowns, an effort was made to estimate a most reasonable answer to many of the parameters presented. This estimate was arrived at by looking back to known information such as enemy defenses and reported location of fire. The results hopefully will lead to less vulnerable aircraft in the future and point out areas where possible fixes are needed on present aircraft by providing an insight into the most vulnerable areas and subsystems, thus allowing a judicious choice for the ABCS of survivability ("A" denoting add protection or armor vulnerable components; "B" being bury vulnerable components behind less vulnerable components; "C" representing concentration of vulnerable components to provide a smaller presented area; and "S" being separate redundant critical vulnerable items so that damage to one will not result in highly probable damage to the other).

(U) Although this report deals solely with the F-105, the analysis methods should be applicable to all aircraft.

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#### SECTION II

#### COMPILATION

(U) This report attempts to make a correlation of the WSEG compendium data for the F-105. The tables are presented as compilations of the information contained in the WSEG reports. There are 62 aircraft listed in the tables, but four of these aircraft were lost to causes other than enemy defenses and are not included in the analysis. The last two tables summarize the data on aircraft lost due to flight control and fire damage, respectively. It is obvious from examination of these tables that much data is lacking. In various figures, this lacking data is presented as "UNK" or unknowns. The handling of the unknowns will be dealt with in Section III of this report.

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#### SECTION III

#### (U) THE THREAT

(S) From Table I it is seen that the F-105 is used on a variety of missions. Among these are flak suppression, armed reconnaissance (recce), strike, and close air support. As such, the F-105 is exposed to a hostile environment for which it was not designed. The environment exposes the F-105 to defenses which include small arms, automatic weapons such as 12.7 mm and 14.5 mm; medium antiaircraft such as 23 mm, 37 mm, and 57 mm; heavy antiaircraft artillery such as 85 mm and 100 mm; and SAM's. (See Section IV for an explanation of the letters used in the Type Kill column of Table I.)

(C) Figure 1 shows the number of aircraft lost as a function of projectile type. The figure contains a large number of the previously mentioned unknowns. Since the lost aircraft are not recovered, it is difficult to determine the exact killing projectile. In Figure 2, an attempt is made to decrease the number of reported kills due to unknown projectiles by making an estimate of projectile type. This estimate is arrived at from an analysis of enemy defenses in the target area. The analysis involves looking at the enemy defenses that the lost aircraft encountered. For example, if the killing projectile was unknown but the defenses were listed as 37 mm and 57 mm, then the kill is placed in the 37-57 mm category in Figure 2.

(S) Finally we have the number of aircraft lost as a function of altitude and projectile size in Figure 3. The most noticeable trends are the appearance of SAM hits at altitudes over 3000 feet and the disappearance of projectiles of less than 37 mm at 3000 feet. The properties of both weapons predict this occurrence.

(S) One should keep in mind the fact that North Vietnam antiaircraft defenses are such that heavy and medium antiaircraft 'batteries are reinforced by many small-arms-firing infantrymen and machine gunners. Thus, many of the unknown kills were relegated to the 37-57 mm category, but the aircraft loss could possibly have been caused by weapons in the category of 14.5 mm or less.

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#### SECTION IV

#### (U) CAUSE OF LOSS

(C) In the field of aircraft vulnerability, aircraft losses due to battle damage are classed in the categories of KK, K, A, and B kills. These are defined as:

#### KK -- Instantaneous

K -- Loss within 5 seconds

A -- Loss within 5 minutes

B -- Loss within 30 minutes

Figure 4 shows the number of aircraft lost in each category.

(S) An examination of the location of hits on the F-105 could possible give a clue as to the most vulnerable areas. Figure 5 shows the number of losses occurring as a function of hit location on the aircraft. It is known (Reference 2) that almost all aircraft are hit on low-level attack missions and that the hits occur on the underside and sides of the aircraft. Examination of Figure 5 indicates that most of the hits causing losses are located between Stations 300 and 777. Station 300 is just aft of the crew station. Station 494 is the start of the engine section, and 777 is the aft end of the aircraft. These stations are for the F-105F but are representative of all F-105 series aircraft. An estimate was used to lower the number of unknowns shown in the Location of Hits column in Table I. The location was estimated in many cases by the observation of damage, i.e., when a fire was reported coming from the tail section it was assumed that the aircraft was hit in the tail section even though the exact location was not reported. It was not possible to account for 21 of the lost aircraft even with this method. Figure 5 has these estimates taken into account. The damage locations have been presented on a view of the F-105 in Figure 6.

(U) An examination of Figures 7a and 7b shows that many critical components of the aircraft are located in the area between Stations 300 and 777. These components are:

a. Engine and compressors

b. Fuel tanks

c. Fuel lines and pumps

d. Stabilizer control actuator

e. Hydraulic lines.

It can also be seen that most of these components are on the underside of the aircraft and, thus, are not masked by less vulnerable or less critical components. Figure 8 gives an indication of the density of hydraulic, fuel, and air lines through the bomb bay.

(C) A possible correlation between cause of loss and hits in a certain location would be to examine the failure mode of the lost aircraft. Figure 9 shows the

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percentage of aircraft lost experiencing various failure modes. From Figure 9, it is noted that flight control failure and fire occur in most of lost aircraft. Figure 7a shows that all the fuel tanks and fuel lines are between Stations 300 and 777 along with a heat or ignition source (engine). The stabilizer actuator is also in this area along with hydraulic lines and the engine-driven hydraulic pumps. A hit in any of these items could lead to a loss with a reported fire. Both the aircraft fuel and hydraulic fluid are highly flammable and even if the projectile does not perform the ignition function the engine can.

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(S) It is interesting to note on Table I that 50% of the losses on which information was available had fires, 32.8% had no fires, and 17.2% were unknown losses.

(S) Figure 10 is a representation of the time that the aircraft remains flyable after the aircraft is hit and fire is observed. Since most of the kills are Category A, the pilot does have some time to try to control the fire or to eject.

(S) The data in Figure 9 also shows that 34.5% of the lost aircraft experienced flight control failure. Since an aircraft can have both fire and flight control failure, the flight control failure due to hydraulic line penetration can lead to a fire or a fire can cause failure of the flight control system through degradation of the hydraulic fluid or seals. Thus, it is often difficult to determine the prime cause of loss. Furthermore, loss of the engine and the associated hydraulic pump power supply driven by the engine can cause loss of flight control before the ram-air turbine, an emergency pump power supply, can be activated. Figure 11 shows the kill category for flight control failure.

(C) Another consideration from an intelligence standpoint is the attacking sequence of the aircraft when damage was received. Figure 12 shows the formation position, if known, of the aircraft when hit. A possible reason for the decrease in losses to following aircraft, as indicated in Figure 12, could be that the defenders are protecting themselves from explosions from the ordnance delivered by the first aircraft. This is particularly true in a flak suppression mission (Reference 3).

(S) A brief analysis is performed on pilot survivability according to type of aircraft kill. Survivability is defined as getting out of the aircraft alive regardless of whether the pilot was cartured or returned to his unit. This analysis is based on the data from Table I (Type Kill column) and Table II (Pilot Status column). The results of this analysis are shown in Figure 13. Note that 65% of A kill and 83.3% of B kill pilots survive, indicating the natural conclusion that the longer the aircraft remains under control, the greater the chances of crew survival.

(S) The following table gives the pilot status for aircraft lost as a function of altitude.

Altitude	Number of Aircraft Lost	Pilot Survives	Pilot Recovered	KIA or <u>MIA</u>
0 to 1000 ft	10	5	3	5
1000 to 3000 ft	7	6	5	1
3000 to 6000 ft	12	10	8	2
greater than 6000 ft	5	2	2	3
unknown	24	9	7	15

(U) The pilot status is shown as a percentage of total losses at altitude bands in Figure 14.

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#### SECTION V

#### (U) RESULTS

(U) The objective of this analysis was to determine those factors which contributed to the losses of the F-105 aircraft when subjected to the combat environment encountered in Southeast Asia. Although detailed information was lacking in many of the combat losses reports contained in the WSEG Compendium, it was possible to obtain a general indication of those items which eventually lead to the loss of the aircraft.

(S) Based on the known information and various estimates, the analysis shows that most aircraft losses were due to subsystem failure or fire rather than primary structural failure. Furthermore, other studies under Air Force Project 5105 have indicated that this is the case for all SEA combat fighter/bomber aircraft.

(C) It should be noted that there are many items of information missing in the reported data. It is realized that it is difficult to obtain information from an unrecovered aircraft. However, there are many gaps in the data due to unsatisfactory reporting procedures. Efforts are currently under way to attempt to improve the reporting procedures.

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#### SECTION VI

#### (U) CONCLUSIONS

(C) As a result of the this analysis, it can be concluded that more detailed information is needed. In many cases, information such as cause of loss or damaging projectile cannot be recovered. However, much of the data needed to achieve a higher degree of confidence in portions of the analysis should be available but is not reported.

(S) The primary cause of F-105 loss in SEA (Out-Country) during the period studied was due to damage from medium antiaircraft (37-57 mm) weapons with the failure mechanism being fire and subsystem loss, primarily the flight control system. Fires were reported in 50% of the losses. Flight control failure was reported in 34.5% of the losses. Fifty percent of the losses were attributed to ground fire in the 37-57 mm range.

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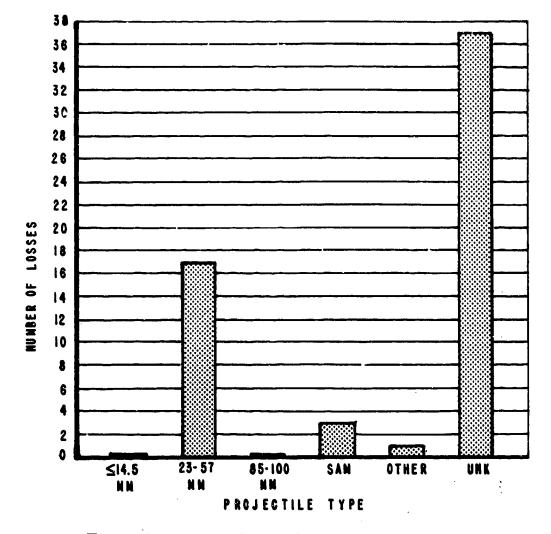
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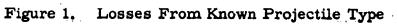
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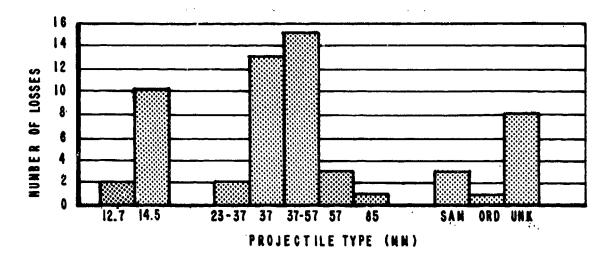


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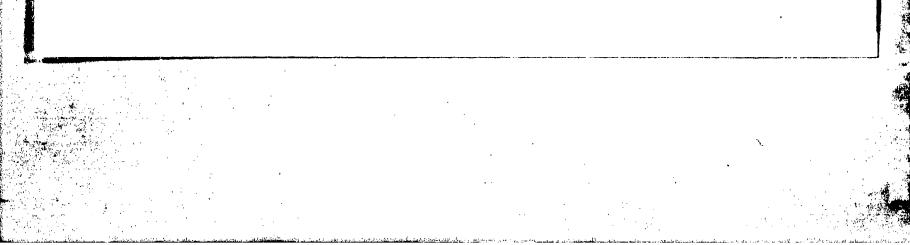
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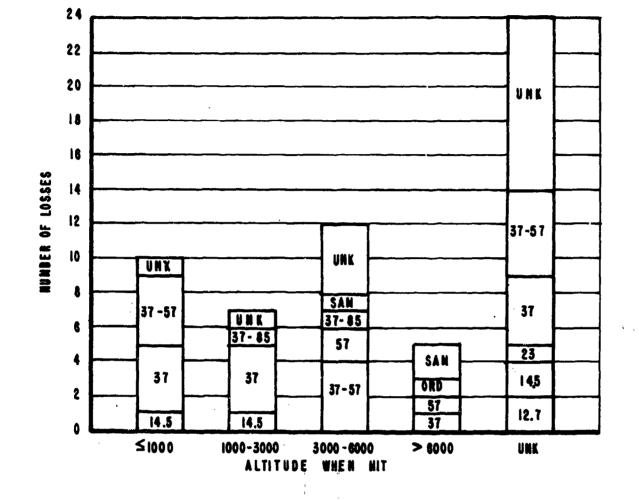


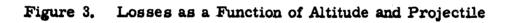








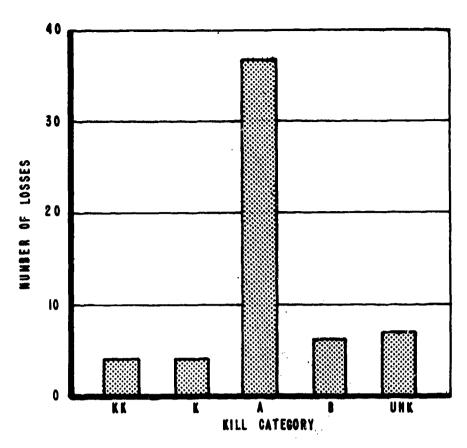




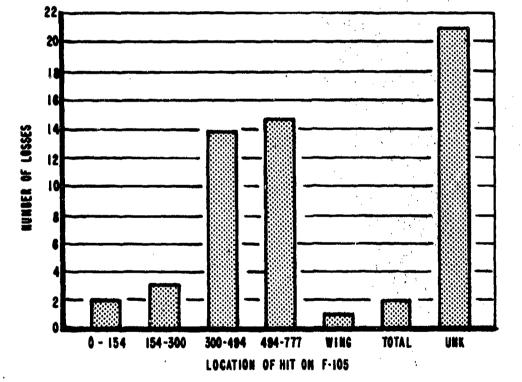
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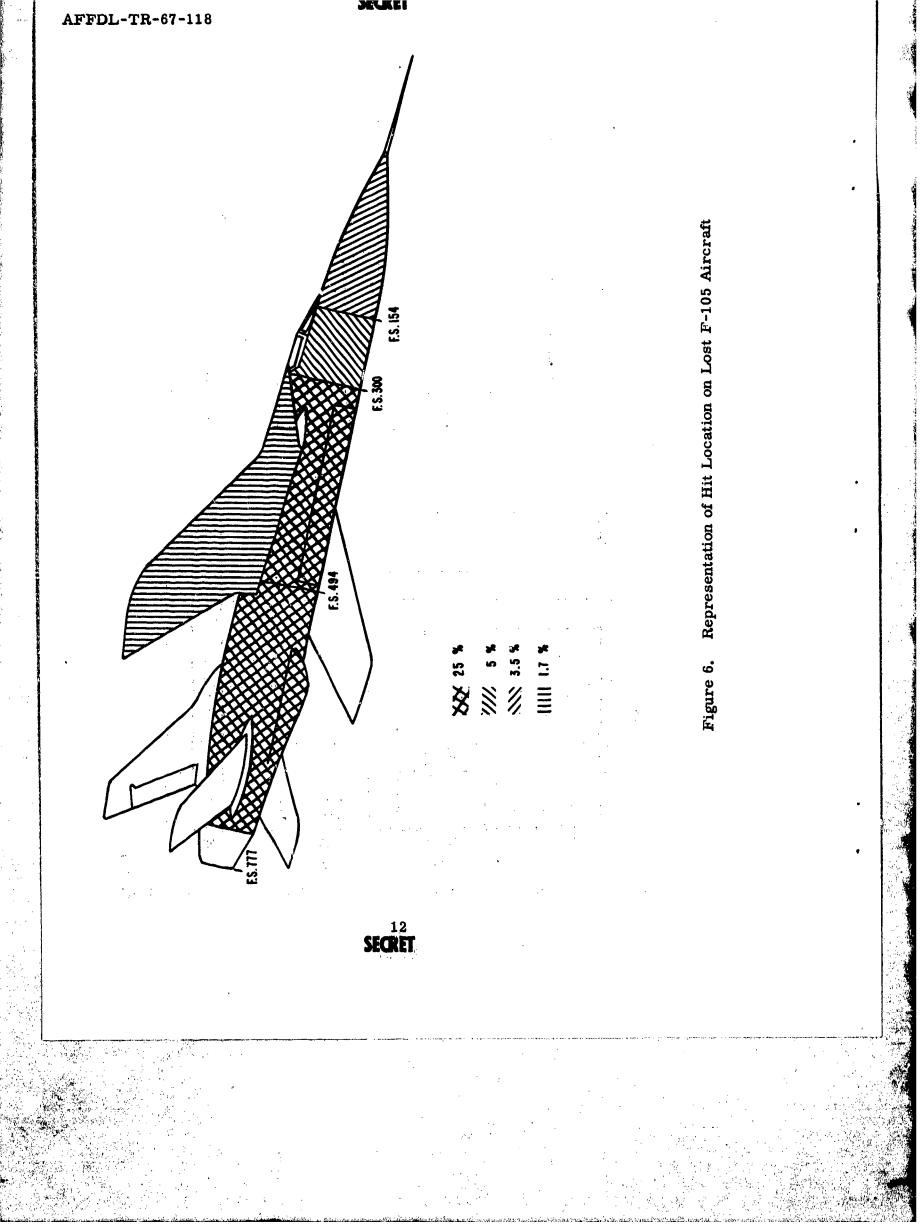
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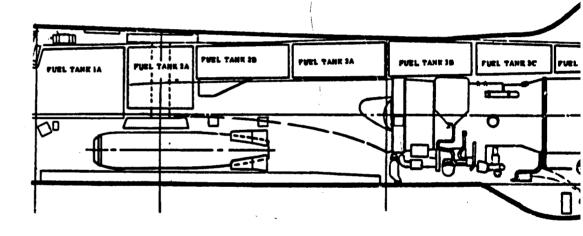
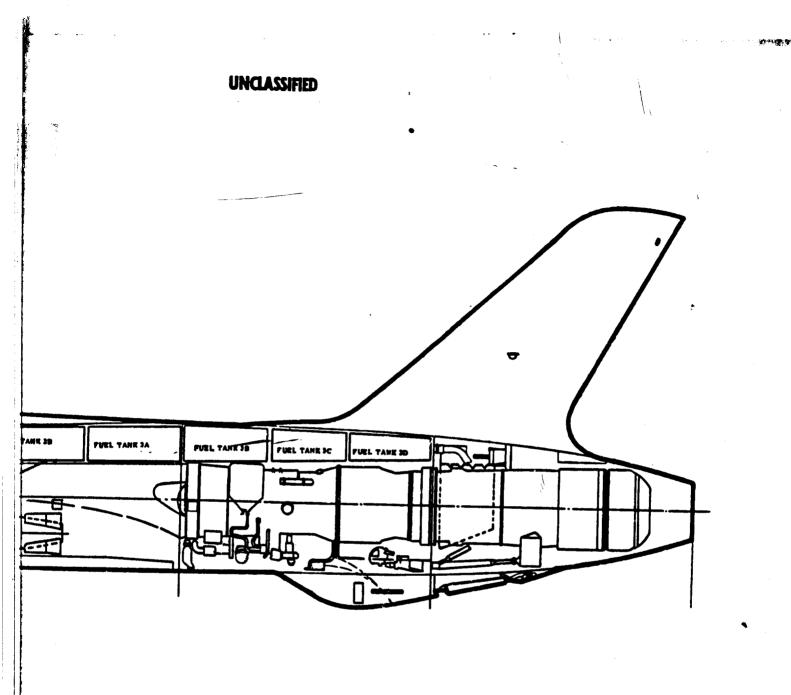
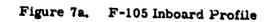


Figure 7a. F-105 Inboard

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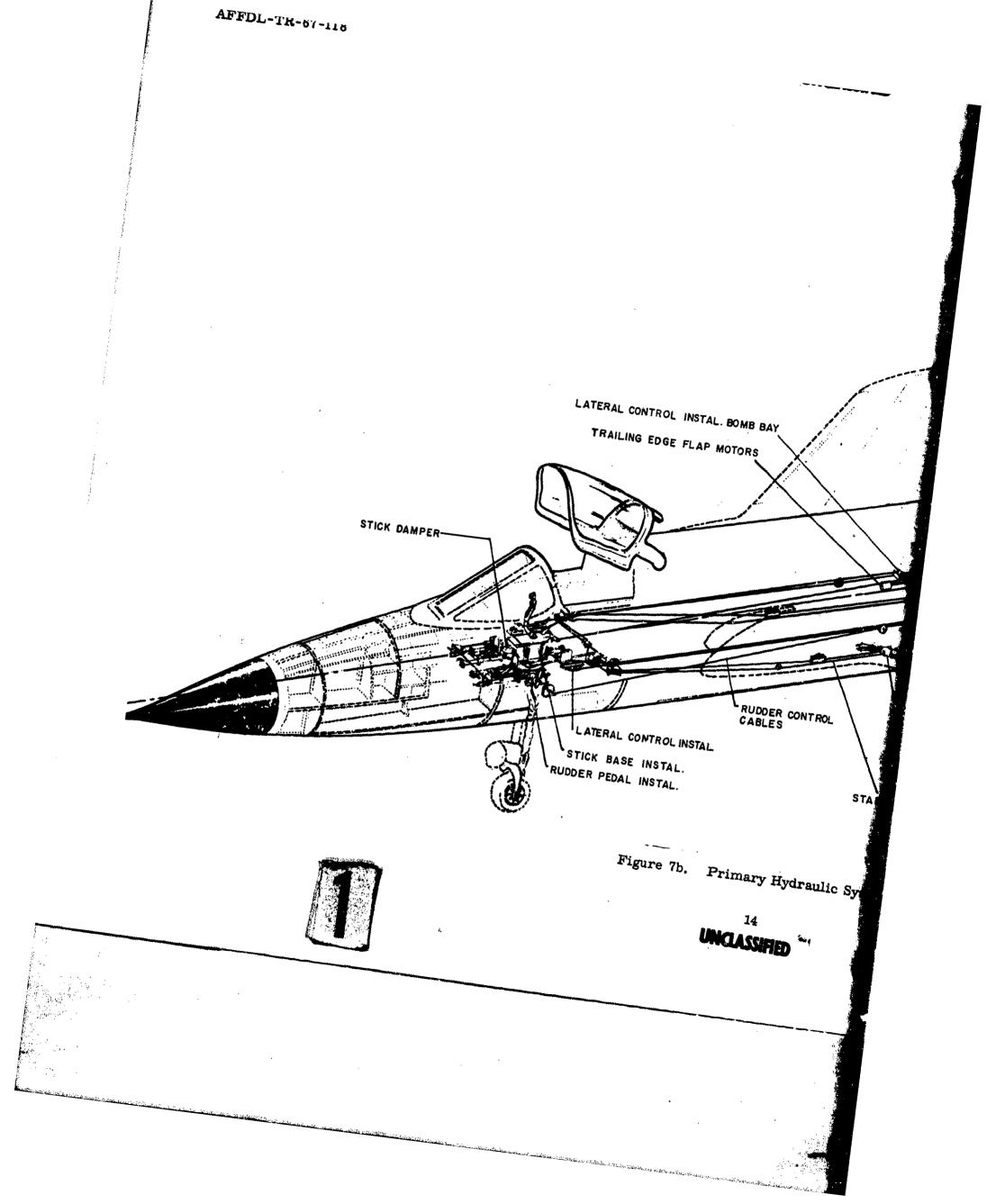


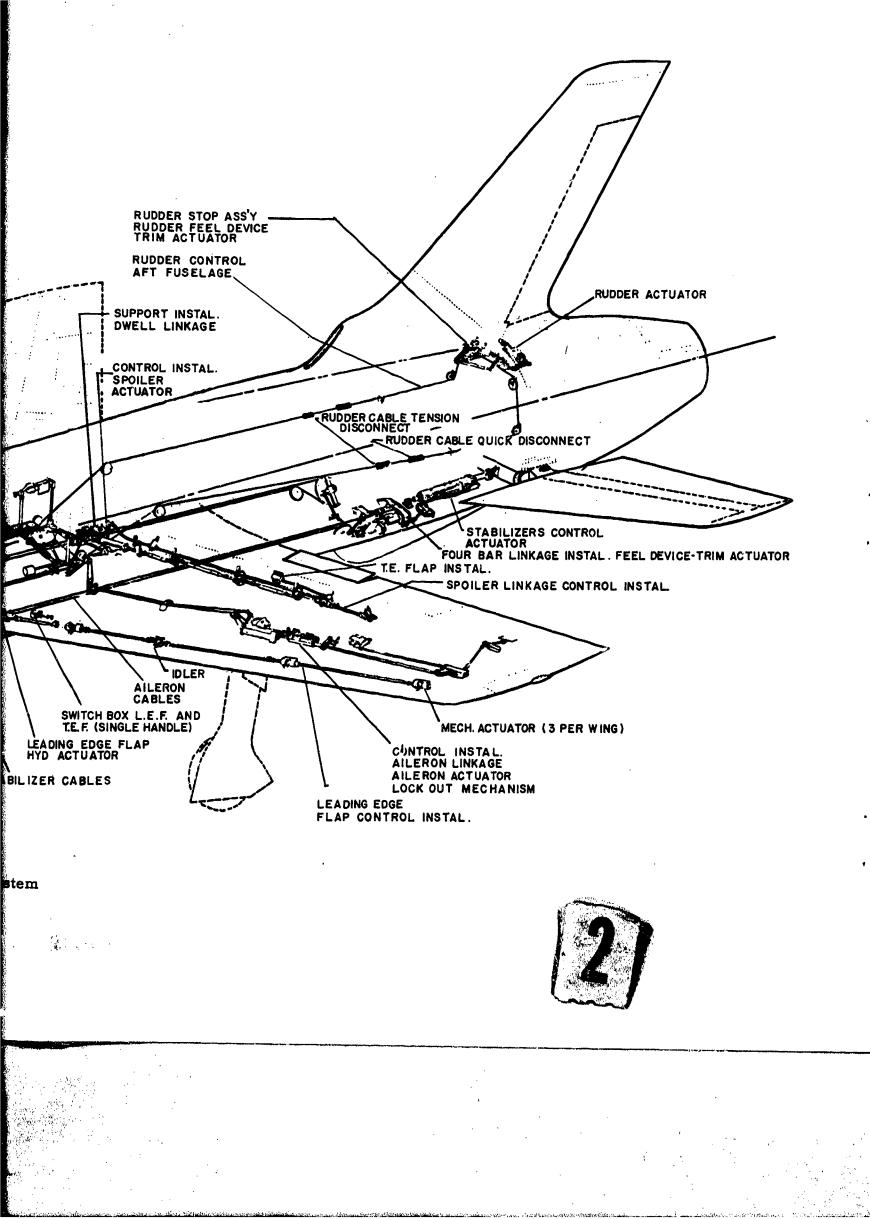




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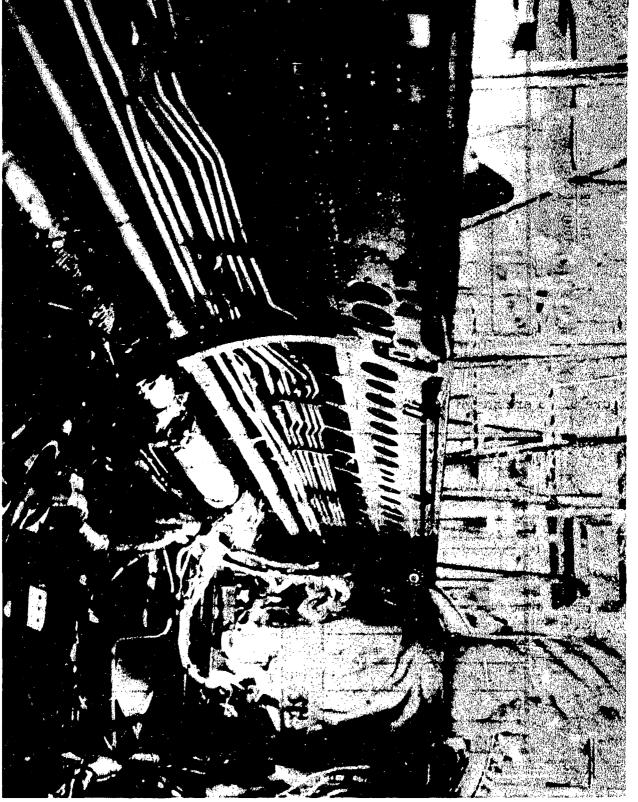
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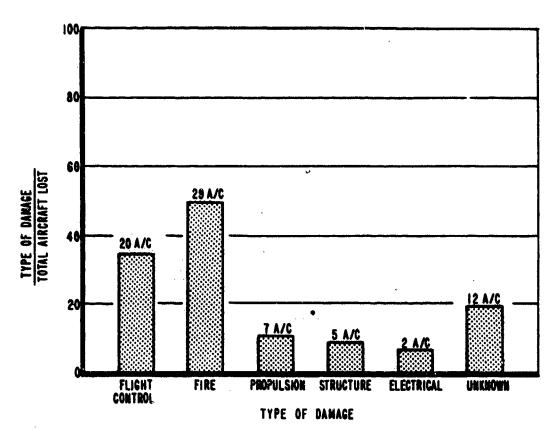
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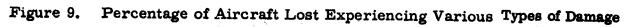


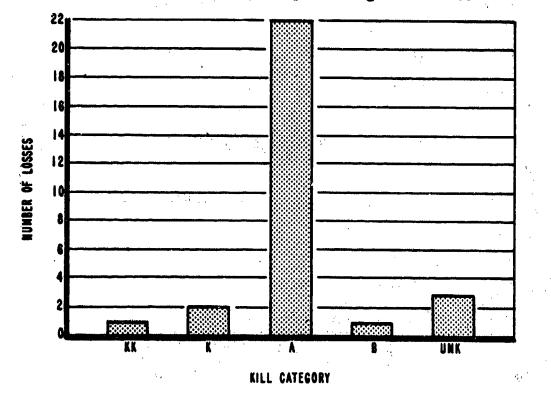
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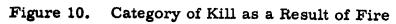
Figure 8. F-105 Bomb Bay











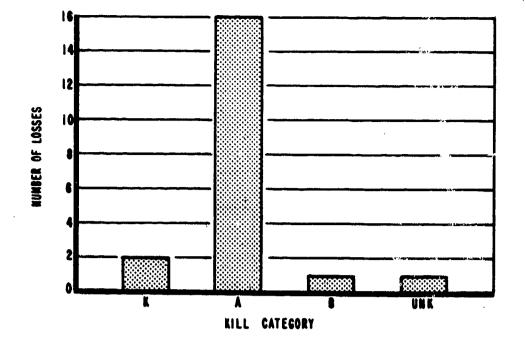
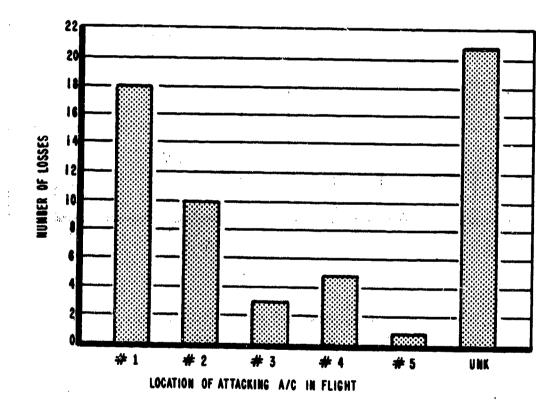
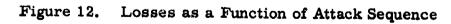
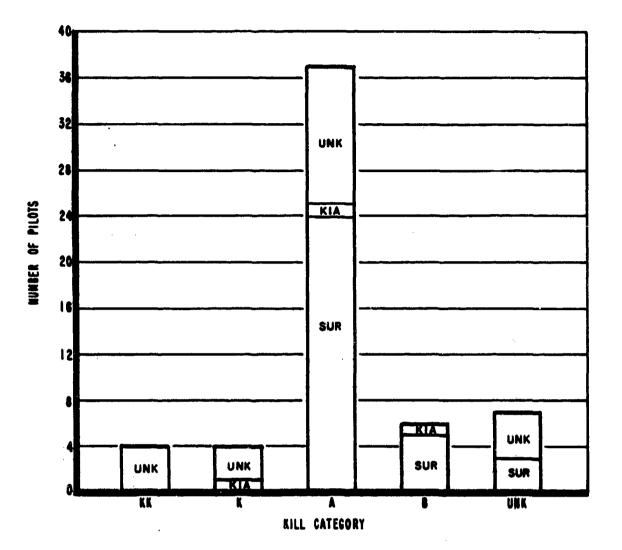


Figure 11. Category of Kill as a Result of Flight Control Failure

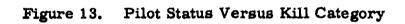








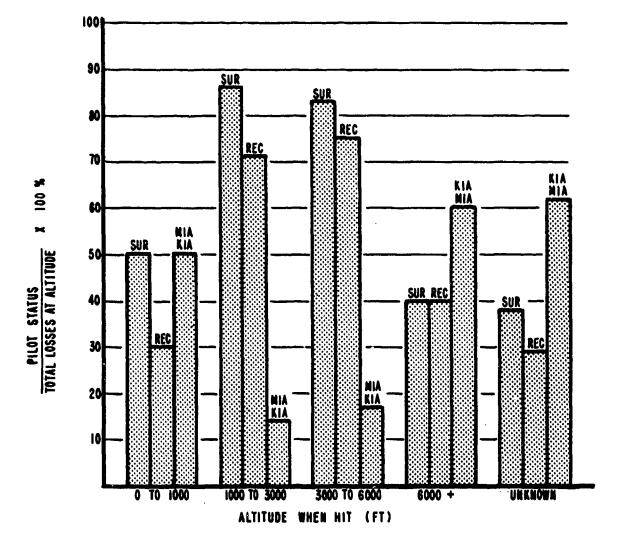
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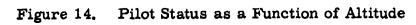


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INCIDENT	SORTIE TYPE	SORTIE LOCATION	TARGET TYPE	WEATHER	ENENY DEFENSE	WEAPONS OF AIRCRAFT	NO. OF A IRCRAFT IN MISSION	STAGE WHEN HIT
1	FLAK SUPPRESSION	ROLLING THUNDER 5	NUNITIONS Storage	20% CLOUD, IONI.VISIBILITY	37 OR 57 NM	8 - 750	24 (LEAD OF 4)	ATTACK
2	STRIKE	-	MUNITIONS STORAGE	20% CLOUD, 10 MI. VISIBILITY	GROUND FIRE (HEAVY)	8 - 750	24	ATTACK
3	FLAK SUPPRESSION		AMMUNITION STORAGE	CLEAR WITH HAZE	AUTO WEAPONS, 37 NM	2 - C B U	4	ATTACK
4	FLAK SUPPRESSION	-	AA BOATS OFF Shore	CLEAR	14.5 MN (MODERATE)	BONBS, Rockets	4 (LEAD NAN)	ATTACK
5	STRIKE	THANH HOA	RAILROAD WITH HIWAY BRIDGE	CLEAR 5 MI. VIS., HAZE 0-1200 FT	GROUND FIRE	750 LB Bombs	48	ATTACK
6	STRIKE	THANH HOA	RAILROAD WITH HIWAY BRIDGE	_	MIG CANNON		48 (FLIGHT LEAD)	ORBIT PRIO To Attack
7	STRIKE	THANH HOA	RAILROAD WITH HIWAY BRIDGE	SCATTERED HAZE	MIGS		48	ORBIT PRIOR To Attack
8	ARNED RECCE.		TRUCKS		37 NN	_	4 (THIS NAN #4)	PULLOUT FROM Strafing Pas
*9	ARNED Recce.					_	4	
10	STRIKE	THANH HOA	BRIDGE	40% CLOUD, 25 NI VISIBILITY	AUTO WEAPONS, 37 & 57 MM	8 - 750	42	ATTACK
11	AR NED RECCE	STEEL TIGER	FLAK POSITION	SCATT. TO BROKEN TO OVERCAST	(HEAVY)	ROCKET POD, M 61 CANNON	2 (THIS MAN <sup>\$</sup> 2)	
12	RECCE	ROUTE 7 FROM VINH TO LAOS	TARGETS OF Opportunity	SCATTERED TO BROKEN CLOUDS	14.5, 37 MM HEAVY AUTO, SM. ARMS	ROCKET PODS, 20 NN	4 (THIS MAN #4)	ROAD RECCE
13	STRIKE BARREL ROLL		ROAD	SCATTERED TO BROKEN OVERCST. 10 NI.VISIBILITY	GROUND FIRE, LIGHT AA, AUTO. WEAPONS	8-750, Ngi cannon	4 (LEAD NAH)	ATTACK
14	ROAD R ECCE	ROUT'ES A, B, & C	BRIDGE, ROADS	50% CLOUD, ISHI. VISIBITITY	AUTO. WEAPONS (LIGHT)	ROCKETS	12	—
15	STRIKE	THAN HOA	BRIDGE	15-16,000 FT CLOUD LAYER	GROUNDAFIRE	8 - 750	4	ATTACK
16	RESCAP For A4C	LAOS			GROUND FIRE (HEAVY)			ORDINANCE Delivery pas
17	ARNED Recce	· · · · · · · · · · · · · · · · · · ·	WOODEN Bridge	70% CLOUD, 10 MI. VISIBILITY	GROUND FIRE (HEAVY)		(LEAD NAN)	ATTACK
18	STRIKE	BON XON LON	ARNY BARRAKS	li,000 FT, Broken	37, 57, <b>8</b> 85 MH AUTO. WEAPONS	, 8 - 7 50	4 (LEAD NAN)	ATTACK
19	ARNED RECCE	ROLLING THUNDER 19 C	CONCRETE BRIDGE	4500 FT,	AUTO WEAPONS (HEAVY)	8 - 750	4	ATTACK

\* NOT USED IN ANALYSIS



### TABLE I

### WSEG DATA SUMMARY

NS OF NAFT	NO. OF A IRCRAFT IN MISSION	STAGE WHEN HIT	BÀSIC NANEUVER	STARTING Altitude	ALTITUDE When hit	VELOCITY When hit	PROJECTILE Type	DIRECT OR FRAGMENT	LOCATION OF HIT	RESU DA N
0	24 (LEAD OF 4)	ATTACK	PULLOUT AFTER BONBING	1	_	_	37 OR 57 MM	-	TAIL	HYDRAU Flight C
0	24	AT TA CK	-			_		-		AC CENE
U	4	ATTACK	LEVEL FLIGHT	10,000	20 FT	500 KIAS	37 NM	DIRECT	NOSE, LT. WING ENG COMP ELEC. SYS.	ELECT
s	4 (LEAD NAR)	ATTACK		-		_	-	-	AFT SECTION	FIRE IN F
	48	ATTACK	BOMB RUN	-			-			FIRE IN SECTIO
	48 (FLIGHT LEAD)	ORBIT PRIOR TO ATTACK	ORBITING	_		-	CANNON FIRE NIG		-	DA NAGEL CANNON
	48	ORBIT PRIOR To Attack	ORBITING	-			CANNON FIRE MIG 23 & 37 M M	DIRECT	LT. SIDE FROM SP. ' BK. TO CANOPY	LOST A Contri
	4 (THIS MAN #4)	PULLOUT FROM STRAFING PASS	STRAFING PASS			420 KTS	-			FIRE AN Contro
	4	-	ROCKET Attack	6000		305 KTS			(HIT HILL IN DIVE)	
j0	42	ATTACK	DIVE BOMB Attack	13-14,000	4500 FT	<b></b>	—		RICHT WING Root	STABILI AUGNEN
OD, Non	2 (THIS NAN <sup>*</sup> 2 )	-				-	_		?	
ODS,	4 (THIS MAN #4)	ROAD RECCE	ROAD RECCE	3000	3000	450	37 MN	DIRECT	FUEL CONT. OR MAIN FUEL LINE	ENGINE
NON	4 (LEAD NAH)	ATTACK	DIVE BONB Delivery		3000 AGL	-	-		?	FIRE
	12				1000 AGL			<b>—</b> ,	ELECTRONICS IN FUSELAGE BOTN.	FIRE
	4	ATTACK	DIVE BONB Attack	18,000		-				FIRE
		ORDINANCE DELIVERY PASS	ORD IN ANCE DELIVERY PASS		4000	_				LOSS OF TANK &
	(LEAD NAN)	ATTACK	EVASIVE CLIMB AFTER PASS		4000		GROUND FIRE	-	AFT SECTION	FIRE
)	4 (LEAD NAN)	ATTACK	DIVE BONB PASS	7000 AGL	4000 AGL	500				
	4	ATTACK	DIVE BOMB RUN, BOMBS DID NOT RELEASE	4500 MSL I500 PULLOUT		300 KTS	AUTO WEAPON Fire			FIRE OB Coming Tail
	·····					····			· · · · · · · · · · · · · · · · · · ·	A contract of the second

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STARTING Altitude	ALTITUDE When hit	VELOCITY When hit	PROJECTILE Type	DIRECT OR FRAGMENT	LOCATION OF HIT	RESULTANT DANAGE	TYPE Kill
_		_	37 OR 57 NM		TAIL	HYDRAULIC PRESS. FLIGHT CONTROL	B
	-	-				AC GENERATOR, LOST NAV. AIDS	B
10,000	20 FT	500 KIAS	37 NN	DIRECT	NOSE, LT. WING ENG COMP ELEC. SYS.		A
-	J	1	-	1	AFT SECTION	INTAKES, COMP. STALL	-
	_		_	-	_	FIRE IN AFT Section	_
-	-	-	CANNON FIRE Wig			DAMAGED BY Cannon Fire	B
-			CANNON FIRE MIG 23 & 37 M M	DIRECT	LT. SIDE FROM SP." BK. TO CANOPY	LOST ALL Control	
_	-	420 KTS				FIRE AND CONTROL LOSS	Å
6000	-	305 KTS			(HIY HILL IN DIVE)		-
13-14,000	4500 FT	ļ			RIGHT WING Root	STABILITY AUGNENTATION	A
					?	?	KK
3000	3000	450	37 NN	DIRECT	FUEL CONT. OR MAIN FUEL LINE	ENGINE FLANEOUT	A
<b>—</b> .	3000 AGL		—		?	FIRE	٨
	1000 AGL				ELECTRONICS IN FUSELAGE BOTN.	FIRE	B
18,000	-				_	FIRE IN TAIL	A
	4000					LOSS OF RT. WING TANK & PYLON	B
	4000		GROUND FIRE	-	AFT SECTION	FIRE	Å
7000 AGL	4000 AGL	500					A:-
4500 MSL 1500 PULLOUT		300 KTS	AUTO WEAPON . FIRE			FIRE OBSERVED Coming From Tail	A
	ALTITUDE 	ALTITUDE       WHEN HIT                 10,000       20 FT   6000          13-14,000       4500 FT             3000       3000          3000 AGL         1000 AGL       4000         18,000           4000         7000 AGL       4000 AGL	ALTITUDE       WHEN HIT       WHEN HIT                   10,000       20 FT       500 KIAS  420 KTS         6000        305 KTS         13-14,000       4500 FT               3000       3000       450              3000       3000       450          1000 AGL           4000           4000           4000          7000 AGL       4000 AGL       500	ALTITUDE       WHEN HIT       WHEN HIT       TYPE         -       -       37 OR 57 MM         -       -       -         10,000       20 FT       500 KIAS       37 MM         -       -       -       -         10,000       20 FT       500 KIAS       37 MM         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         -       -       -       -         6000       -       305 KTS       -         13-14,000       4500 FT       -       -         -       -       -       -         -       3000       3000       450       37 MM         -       -       -       -       -	ALTITUDE         WHEN HIT         WHEN HIT         TYPE         FRAGMENT             37 OR 57 MM                    10,000         20 FT         500 KIAS         37 MM         DIRECT   CANNON FIRE               CANNON FIRE               CANNON FIRE         DIRECT                   6000          305 KTS              13-14,000         4500 FT               3000         3000         450         37 MM <td>ALTITUDE         WHEN HIT         WHEN HIT         TYPE         FRAGMENT         OF         HIT             37 OR 57 MM          TAIL               TAIL               TAIL               TAIL               TAIL               TAIL               TAIL                      AFT SECTION              CANNON FIRE               CANNON FIRE                CANNON FIRE               420 KTS               6000          305 KTS           Riget Aunt Winge Root</td> <td>ALTITUDE         WHEN HIT         WHEN HIT         TYPE         FRASMENT         OF         HIT         DAMAGE          </td>	ALTITUDE         WHEN HIT         WHEN HIT         TYPE         FRAGMENT         OF         HIT             37 OR 57 MM          TAIL               TAIL               TAIL               TAIL               TAIL               TAIL               TAIL                      AFT SECTION              CANNON FIRE               CANNON FIRE                CANNON FIRE               420 KTS               6000          305 KTS           Riget Aunt Winge Root	ALTITUDE         WHEN HIT         WHEN HIT         TYPE         FRASMENT         OF         HIT         DAMAGE



INCIDENT	SORTIE TYPE	SORTIE LOCATION	TARGET TYPE	WEATHER	E NENY DEFENSE	WEAPONS OF AIRCRAFT	NO. OF AIRCRAFT IN MISSION	STAGE WHEN HIT
20	A R MED RECCE	ROLLING THUNDER 210	BRIDGE	20,000 FT, THIN OVERCAST	57 MN (HEAVY)	6-750, 2 LAU3A PODS	4 (THIS NAN #2)	ATTACK
21			RESCAP FOR DOWNED NAVY GREW	3000 FT, OVERCAST	37 NN (light)	20 N.N.	4 (LEAD WAN)	RESCAP
22	AR NED RECCE	-	BARRACKS AREA		37, 57 NN AUTO WEAPONS (INTENSE)	_	Û	AP PROACH TO TARGE T
23	STRIKE	-	CAN DOI BKS.	CLEAR, IO NI VISIBILITY	37-57 NN?AUTO WEAP (HEAVY)	-	15	ATTACK
24		HAN01	SAM SITE # 7	CLEAR, UNLIN- ITED VISIBILITY	37 NN & AUTO WEAP. (INTENSE)		12 (THIS NAN # 2)	AFTER WEAPON RELEASE
25		?	SAM SITE #7	CLEAR, UNLINIT- ED VISIBILITY	GROUND FIRE (HEAVY)	-	12	AFTER NAPALN Release
*26		?	SAN SITE #6		GROUND FIRE (VERY HEAVY)		11	OVER TARGET
*27		?	SAN SITE #6		GROUND FIRE (VERY HEAVY)		11	
28	AR NED RECCE	ROLLING Thunder 25-C4	BRIDGE	10,000 OVERCST. 5 ML VISIBILITY	37, 57, 85,1 100 NN AUTO WEAP	2-3000	5	PRIOR TO BOMB RELEASE
29	ARMED Recce	ROLLING Thunder 25-C5	BARGES	80% CLOUD COVER AT 20,000 FT	12.7 NN AUTO WEAPONS (HEAVY)	38 - 2.75" Rockets	? (THIS NAN #1)	ATTACK
• 30	STRIKE	ROLLING THUNDER 25-A-3	BRIDGES	90% CLOUD Cover At 25,000 Ft	37NN (light)	2 - 3000	4 (THIS NAN #4)	ATTACK
31	AR MED R E C CE	ROLLING THUNDER 26 - C - 6	RADAR SITE	CLEAR, 15 MI VISIBILITY	37NN . (light )		4	AT TAC K
32 .	A R MED RE CCE	ROLLING THUNDER 28-C-5						ATTACK
33	AR NED RECCE	ROLLING Thunder 29-C-2	BARRACKS AREA	SCATTERED CLOUDS 7000'		20 MM SNAKE Bombs	4	ATTAČK
34	ARNED RECCE	ROLLING THUNDER 29-C-4	YEN BAY Arsenal	2000' BROKEN TO OVERCAST, IO MI. VISIBILITY	20 MN, 37 MN (HEAVY)	6 - 750	4 ( THIS MAN # 4 )	ATTACK
35	ARMED RECCE	LEFT HOOK Alert	BRIDGE	CLEAR, 10 NI. VISIBILITY	37 N N, 57 NN (NODERATE)	NAPALN	2.	ATTACK
36	ARNED Recce	ROLLNG THUNDER 29-C-7	BRIDGE	SCATTERED TO BROKEN, GOOD VISIBILITY	37 NM OR 57 MM (LIGHT)	6 - 7 50 2 LAU	4	ATTACK

\*NOT USED IN ANALYSIS

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## CABLE I (Cont'd)

STAGE HEN HIT	BASIC MANEUVER	STARTING ALTITUDE	ALTITUDE WHEN HIT	VELOCITY WHEN HIT	PROJECTIL E TYPE	DIRECT OR FRAGMENT	LOCATION OF HIT	RESULTANT DAMAGE	TYPE KILL
TTACK	DIVE BONB	17,000	5000	450	AUTO WEAPON FIRE	DIRECT	AFT OF Cockpit	FIRE FROM WING ROOT	A
ESCAP	TURNING,GOING INTO ORBIT	3000 AGL	3000 AGL	350-400 KTS	37 NN	DIRECT	ACCESSORY SECTION	FIRE, LOST STABILIZER CONTROL	٨
P PROACH TO NRGE T	LEFT BREAK To avoid Flak		800	500	AUTO WEAPON Fire			ENGINE POWER LOSS -FLIGHT CONTROL FAILURE	٨
TTAC K	NAPALM RUN	100 AGL	100 AGL	500 KTS		<u> </u>		FIRE LT. WING RT- PART OF LEFT WING MISSING	À
FTER WEAPON Elease	EVADING AA FIRE	_					·	POSSIBLE FLIGHT CONTROL FAILURE	٨
FTER NAPALN ELEASE	LOW-LEVEL NAPALN DELIV.		50 AGL					BURNING FROM FWD. OF INTAKES	Å
VER TARGET		-					FLAK HIT NOSE Section		A
							NID AIR Collision		1
RIOR TO BOMB ELEASE	DIVE RECOVERY AT REL. POINT	10,000	5000	450	37 OR 57 MM	FRAGMENTS	BETWEEN RIGHT WING & FUSELAGE	FIRE AT HIT LOC. , FLT. CONT. LOSS	A
TTACK	ROCKET PASS				·			FIRE OBSERVED AT AFT. SECT., FLIGHT CONTS & HYD.	٨
TTACK	DURING DIVE RUN OR AFTER PULL-UP	16,000		450	37 NN		ENGINE INTAKE		B
TTACK	START OF PULL- UP FROM STRAFING PASS						-	FUEL INLET PRESS, FUEL PUNP, LIGHTS ON STAB. AUGNENT.	A
TACK.	PULL-UP FROM Attack		6-7000	500				FIRE	٨
TTACK	WEAPONS DELIVERY PASS							FIRE IN TAIL PIPE, HIS GUN EXPLODED	A
TTACK	LOW-LEVEL HIGH SPEED ATTACK	50 - 100	50-100	550 KTS	20 OR 37 MM			FIRE IN BATTERY COMP., ELECT. & FLIGHT CONTROLS	A
TTACK	DIVE HIT AT NAPALN DROP PT.	8000	3000	450	37 NN	DIRECT	FUSELAGE- BOMB BAY TO AFT SECT	FLIGHT CONTROL	٨
TTACK	DIVE BONB Run	10,000		500 - 530	37 OR 57 MM	DIRECT (ASSUMED)		NO FIRE, POSSIBLY FLIGHT CONTROL	K

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INCIDENT	SORTIE TYPE	SORTIE LOCATION	TARGET TYPE	WEATHER	ENENY Defense	WEAPONS OF Aircraft	NO. OF AIRCRAFT In Mission	STAGE WHEN HIT
37	ARNED RECCE	ROLLING THUNDER 30-C-4	RADIO COMMUN- ICATIONS SITE	70% CLOUD COVER AT 13,000'	37,57 NN AUTO WEAPONS (HEAVY)	_	(THIS MAN #2)	ATTACK
38	STRIKE	IRON HAND	SAN SITE	CLEAR, 20 NI VISIBILITY	37 NM, 57 NM ( NE DIUM)		(LEAD MAN)	
39	STRIKE	IRON HAND	SAN SITE	CLEAR, 15 NI VISIBILITY	SMALL ARMS Or 37 Mm	NAPALN AND 20 MM	2 ( LEAD HAH)	RUN-IN TO, TARGET
40	AR NED Recce	ROLLING THUNDER 32-C-1	BARRACKS Building	CLEAR,UNLIN- ITED VISIBILITY	37NM (NODERATE)	2 PODS OF 2.75" Rockets	4 (THIS MAN #2)	ATTACK
41	ARNED RECCE	ROLLING Thunder 32-C-4	BRIDGE	-			3 (THIS MAN #2)	ATTACK
42	ARNED RECCE	DONG BAI NVN	BRIDGE	CLOUD BASE 6000', 7ML VISIBILITY, 80% CLOUD COVER	GROUND FIRE (HEAVY)		4	ATTACK
43	ARNED Recce	STEEL TIGER 311	MILITARY BUILDINGS	CEILING UNLIM - ITED, 15 NI. VIS.	AUTO WEAPONS 57 NM (NEDIUN)		2 (LEAD NAN)	ATTACK
44	AR NED RÉCCE	ROLLING Thunder 33-C-7	NINH BINH HIWAY BRIDGE	26,000'CEILING 7 NI. VISIBILITY	SAN'S	NONE		ORBITING DIRECTING STI
45	ROLLING Thunder		LANG MET HIWAY BRIDGE	-	GROUND FIRE (HEAVY)		4	
46	STRIKE	-	LANG MET HIWAY BRIDGE	BROKEN LAYER AT 3000', RAIN, THUNDERSTORMS, SHOWERS	37 N N, 57NN , 85 N N	6 - 750	3 (LEAD NAN)	ATTACK
47	-	· •	BRIDGE			-	4 (LEAD)	BDA RUN AFT Attack
48	ARMED RECCE	ROLLING TeathDER 36-C-D	BARRACKS & Storage Area	7000'OVERCAST, IO XI. VISIBILITY	20 GUNS OF 37-57 VARIETY	8 <sup>°</sup> -750 GP Bonbs	4 (LEAD)	ATTACK
49	ARMED RECCE	ROLLING THUNDER 36-C-I	BARRACKS & Storage Area	7000' OVERCAST IO MI. VISIBILITY	20 GUNS OF 37-57 VARIETY	8-750 GP Bonbs	4 (THIS NAN ₩2)	RESCAP
50	ARNED Recce	ROLLING THUNDER 37-C-I	SAN SITE	2000'SCATTERED 5 NI. VISIBILITY IN HAZE	37,57 NN AUTO WEAPONS (HEAVY)	CBU'S AND Shakeye Bonbs	<b>4</b> (lead)	ATTACK
*51	ARNED RECCE	—	BRIDGE	12,000'OVERCAST 7 NI. VISIBLITY		6 BOMBS	4 (THIS MAN #4)	ATTACK
52	(AIRBORNE IRC	N HAND)	TARGETS OF OPPORTUNITY	1500-2500 FT OVERCAST 2 NI.VISIBILITY	SAN'S (3)	750 LB BOMBS	4 (LEAD)	A PPROACH TARGET ARE
53	ARNED RECCE	ROLLING THUNDER 40	R0/	CLOUDS BROKEN AT 3000 FT, 7 NI VISIBILITY IN RAIN	SAN'S (2)	2.75"ROCKETS	(THIS NAN #4)	OUT BOAR FROM TAR

\*NOT USED IN ANALYSIS



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TABLE 1 (Cont'd)

INS OF CRAFT	NO OF AIRCRAFT IN MISSION	STAGE WHEN HIT	BASIC NANEUVER	STARTING ALTITUD E	ALTITUDE WHEN HIT	VELOCITY WNEN HIT	PROJECTILE TYP E	DIRECT OR FRAGMENT	LOCATION OF HIT	
	(THIS NAN #2)	ATTACK	PULL-UP AFTER RELEASE		6000	500 KTS	-		TAIL SECTION	ELI
1177	(LEAD MAN)		Joseph T		MININUM ALTITUDE					EN
LM AND	2 ( LEAD NAN )	RUN-IN TO TARGET	LOW -LEVEL RUN	150 FT	150 FT	510 KTS	SMALL ARNS OR 37 NN	DIRECT	BONB BAY	FIR
s of Rockets	(THIS MAN #2)	ATTACK	PULLOUT ON ROCKET PASS	8000	500 AGL		-		·	EXI AF
	(THIS NAN #2)	ATTACK	ROCKET PASS	7 - 80 00	Trinc.	350 KTS		~~~~		NA WH
	4	ATTACK	DIVE BONB DELIVERY	7000	_		-			A F ON
	(LEAD NAN)	ATTAC K	ROCKET PASS CLIMB OUT	anna an anna an anna an anna an anna an an	45 00	500 KTS	57 MM			FI
		OR BITING DIRECTING STRIKE	ORBITING AS MISSION COORD.		18,000		SA M	?		<b>†</b>
	4			an a						
750	3 (LEAD NAN)	ATTAC K	DEPARTING AREA AFTER BOMB DROP	15.00	1500	480				F IR OF FLI
	4 (LEAD)	<b>ODA</b> RUN AFTER Attack	TURN TO RIGHT ON BDA RUN	A said and any film of a statement of the	2000 AGL				REAR PART OF AIRCRAFT	HYD
0 GP 8	4 (LEAD)	ATTACK	PULL UP AFTER PASS ON TARGET	in a filling a f			37-57 NM			1
0 GP 6	(THIS MAN #2)	REBCAP	Million P	taica yer	2				UNKNOWN	1
AND IV E	(LEAD)	ATTAOK	LOW LEVEL HIGH SPEED OBU DELIVERY	400 AGL	400 AGL	550	A UTO WEAPONS	DIRECT		FIR FUS BO
186	(THIS NAN #4)	ATTACK	DIVE BOMB RUN	11,000 NSL	increase in the second s				HIT MOUNTAIN	HIT
BOMBS	4 (LEAD)	APPROACH TO TARGET AREA	APPROACH	<del>6</del> 500	6500 -,	450 K (EAS)	SAM (SA - 2)	FRAGMENTS	SAN DETONATED 20 FT UNDER AIRCRAFT	D BUI A/ TO
ROCKETS	(THIS MAN #4)	OUT BOARD FROM TARGET			4000	400	(PILOT SAID IT WAS A SAN)			FI

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STARTING ALTITUDE	ALTITUDE WHEN HIT	VELOCITY When hit	PROJECTILE Type	DIRECT OR FRAGMENT	LOCATION OF HIT	RESULTA NT DANAGE	TY PE KILL
	6000	500 KTS	-			ELECTRICAL ENG. FLIGHT CONTROL	A
	MININUN Altitude					ENGINE AFFECTED	A
150 FT	150 FT	510 KTS	SMALL ARMS OR 37 NM	DIRECT	BOMB BAY	FIRE IN BOMB BAY ENGINE	٨
8000	500 AGL			—	· -	EXPLOSION IN AFT SECTION	A
7 - 8000		350 KTS				NAY HAVE HIT TREES WHEN TURNING	KK
7000			_		-	AFT SECTION On Fire	٨
-	45 00	500 KTS	57 NN	·	·	FIRE	٨
	18,000		SAM	?			KK
	_						
1500	1500	480				FIRE LEFT SIDE OF AIRCRAFT FLIGHT CONTROLS	K
	2000 AGL				REAR PART OF AIRCRAFT	HYDRAULIC SYS. & Flight control	A
			37-57 NN	-			٨
				_	UN K NOW N	UNKNOWN	
400 AGL	400 AGL	550	AUTO WEAPONS	DIRECT		FIRE ON FUSELAGE BOTTON	٨
11,000 MSL				-	HIT MOUNTAIN	HIT NOUNTAIN	
6500	6500 -	. 450 K (EAS)	SAN (SA-2)	FRAGMENTS	SAN DETONATED 20 FT UNDER AIRCRAFT	BURST IN FLANES A/C SHREDDED TO PIECES	KK
	4000	400	(PILOT SAID IT WAS A SAN)			FIRE, AIRCRAFT DISINTEGRATED	K



NCIDENT	SORTIE TYPE	SORT IE LOCATION	TARGET TYPE	WEATHER	ENENY DEFENSE	WEAPONS OF Aircraft	NO. OF AIRCRAFT In Mission	STAGE When hit	BA MÁN
54				_	—		(LEAD)	A T TACK	CLIN Posi Rock
55		ROLL ING THUNDER	BARRACKS	2000'OVERCAST, 3 NI. VISIBILITY IN HAZE	—		(THIS NAN #2)	ATTACK	HIGH STRA
56		ROLLING THUNDER	RR BRIDGE	3000-5000FT SCATTERED, 10 NI. VISIBILITY	57 TO 85 NN (HEAVY & Accurate)	3000 LB B0 NB	4 (THIS NAN #3)	ATTACK	POP- Ordin Deli
57		ROLLING TH UNDER	POWER PLANT	OVERCAST AT 2500 FT, 7 NI VISIBILITY IN HAZE	37 NM AUTO WEAPONS (light)		? (lead)	TARGET Arproach	
58		VU CHUA, NVN	RR BRIDGE	1000 FT Overcast	37 MM OR 57 MM	6 - 750	4 (THIS NAN #3)	LETDOWN TO TARGET	LEFT BANK DOWN
59		(ROLLING THUNDER)	BRIDGE		37& 57 MN (NODERATE)	6 - 750	4 (THIS MAN #2)	ATTACK	P ULL BO NB
60	BARREL ROLL	(LAOS)	BRI D G E	CLEAR WITH LIGHT HAZE, 7 MI VISIBILITY	AUTO WEAPONS (light)	8 - 750	4 (THIS NAN #3)	ATTACK	PULI FROI Bone
61	CLOSE AIR SUPPORT		AUTO WEAPONS POSITIONS & TRENCHES	HEAVY CUMULUS SCATTERED BASE, 8-9000 FT	GROUND FIRE	20 M M	5 (THIS NAN #5)	ATTACK	STRA
62	A RMED RECCE	ROLLING Thunder 48-0-4	ROUTES IA & WATERWAY BARGES	LOW SCATTERED AT 1200-1600 FT Solid At 2000 FT	37 OR 14.5 N N	ROCKETS 20 NN	4 (THIS NAN # 2)	RECCE	LOW



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# ABLE I (Cont'd)

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AGE N HIT	BASIC NANEUVER	STARTING ALTITUDE	ALTITUDE WHEN HIT	VELOCITY When hit	PROJECTILE Type	DIRECT OR FRAGMENT	LOCATION OF HIT	RESULTANT DANAGE	TYPE Kill
NCK	CLINB TO POSITION FOR ROCKET PASS		5000			_		FUEL COMING OUT OF A/C, FIRE FLIGHT CONTROLS	٨
IC K	HIGH ANGLE STRAFE PASS	7000 <b>A</b> gl	_	<u> </u>	-			FLIGHT CONTROLS	A
CK	POP-UP FOR ORDINANCE DELIVERY		8 - 9000 FT	515	37 N N	DIRECT	-	· -	K
BET Roach			2500	450	37 MM -	-	-	FIRE (TRAILING AIRCRAFT)	-
OWN TO Et	LEFT TURN & BANK TO LET- DOWN ON TARGET		4000	350 KTS		- (PILOT FELT THUMP)	_	FIRE, LOST Flight controls	٨
C K	PULLOUTAFTER BOMB RELEASE		7-8000 FT	450	57 NN	DIRECT	TAIL HOOK AREA	FIRE, UTILITY & HYDRAULIC SYS, FLIGHT CONTROLS	٨
AC K	PULLOUT FRON OWN Bonb Run		4 500 -	500	A/C OWN BONB (PREMATURE)			FIRE	Å
TACK	STRAFE PASS						_	_	-
CCE	LOW-LEVEL Arned recce		_	_			_	_	PROB A
			1	l	[	<u> </u>	<u> </u>	<u></u>	L



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TABLE II DATA SOURCES

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TABLE II (Cont'd).

									7	_	_		[ <b>-</b> ]	1			-				1				-		-					l
SEARCH AND Rescue Report	×	×	×	X	X	×			×		*	X				×	×	-			×		×	~		*	X	*	*	X	×	
FORM 484		×	X		X		X	X	X	X	X		X	X	×		X	-	×	X	×			×	×					X	×	
PILOT STATEMENT	*			X		X						~				*											×	X	×	×		
PILOT				X																								×				
WINGMAN STATEMENT		X	X		X		X	X	X	X	X		×	X	×		X	X		X	X			X	X						×	
WINGMAN Interviewed	•				X			X							X				X													
ESCAPE AND EVASION REPORT			-													X													X			
MISHAP REPORT	X			×		×	×	X	X	X	X	×	*	X	X	X	X	X	X	X	X	X	X		I	X	X	X	X	X	X	
COACT		X	X	×	×	×	X	×	×	×		×							X		X	X			X	X			X		X	
USAF COMMAND POST RECORDS	X	х	X	X	×	×	۲	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	X	×	X	X	X	X	X	X	
PILOT STATUS	RECOVERED	N I N	DETAIKED	RECOVERED	KIA	RECOVERED	MIA	NIA	KIA.	MIA	<b>KIA</b>	RECOVERED	MLA	MIA	NIA	RECOVERED	MIA	MIA	NIA	MIÅ	NIA NIA	KIÅ	RECOVERED	MIA	MIA	RECOVERED	RECOVERED	RECOVERED	RECOVERED	N I N	HIA HIA	
ORCANIZATION	6441 TFW	IS TFW	IS TEW	18 TFW	6441 TFW	18 TFW	18 TFW	18 TFW	18 TFW	18 TFW	334 TFS	562 TFS	334 TFS	6441 TFW	6441 TFW	562 TFS	6441 TFW	6441 TFW	6441 TFW	23 TFW	355 TFW	6234 TFW	6234TFW	355 TF#	355 TFW	355 TF W	355TFW	6234 TFW	355 TFW	355TFW	6234 TFW	
TAIL Number	62-4235	63 - 8282	61-0193	61-0185	62-4389	62-4337	61-0189	61-0217	62-4247	62-4328	60-0382	61-0200	£110-05	62-4285	62 - 43 76	0910-19	62-4333	62-4305	62-4350	61-0163	62-4342	62-4332	61-062	62-4285	51-0182	62-4363	61-0090	59-1823	59-1736	59-1719	61-0210	ANALYSIS
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		F I KE		0 <b>X</b>	YES	2	0 N	YES	0¥	Ŷ	YES	0¥
	1	FIRAL MANEUVER	A/C ROLLED TO RIGHT. UNABLE To STOP WITH RUDDER	PILOT LOST ALL CONTROL Except rudder	LOST CONTROL AFTER WARNING LIGHTS CAME ON, A/C NOSED OVER	WAS TURNING TO LEFT WHEN P <sub>2</sub> Went to zero	DURING LEVEL-OFF AFTER A CLIMBING TURN, PILOT LOST CONTROL OF STABILIZER	NUFFLED EXPLOSION FOLLOWED BY LOSS OF CONTROL, THEN FLAMEOUT	PO SSIBLE FLIGHT CONTROL FAILURE EVIDENCED BY PORPOISING	BALLISTIC TYPE TRAJECTORY		
ILURES	E LOST ON		5 MIN AFTER RAM AIR EXTENDED	!	l		l	!	1	1	LOST	1
TABLE III FLIGHT CONTROL FAILURES	HYDRAULIC PRESSURE LOST	P2	FLUCTUATED AT FIRST, WENT OUT			WENT OUT Few NIN- Utes After P <sub>1</sub>	I	1	l	1		1
TABLE III HT CONTROL	HYDRAULI	ď	AFTER CLIMB-OUT	<u>l</u> .		GRADUALLY DECAYED TO ZERO		}	1	l	LOST	1
FLIG	DESCRIPTION OF HIT OR DANAG	RESULTED IN LOSS OF FLIGHT CO	A/C HIT IN TAIL SECTION AND VERTICAL STABILIZER MEAR FUSELAGE.	HIT BY CANNON FIRE ON LEFT SIDE OF A/C FROM Speed Brake to Approximately 5 FT rear of Canopy.	PILOT FELT THREE THUMPS AT PULLOUT. OVERNEAT AND FIRE WARNING.	RIGHT WING ROOT. PILOT THOUGHT STABILITY Augmentation "Kicked - Off"	A/C RECEIVED DIRECT HIT IN THE Accessory section.	PILOT DID NOT FEEL HIT FIRE WARNING LIGHT CAME DN, FIRE CAME LATER	A/C STREAMED WHITE SMOKE FROM AFT Section	A/C HIT ON RIGHT SIDE WHERE FLIGHT CONTROLS ARE.	FLANES WERE SEEN COMING FROM A/C REAR	PILOT FELT HITS. FUEL PUMP AND STABILITY AUGMENTATION LIGHTS CAME ON
	INCIDENT	וווערועכעו	_	F==-	80	0	51	52	24	<b>68</b> 7	29	31

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TABLE	

	ESSURE LOST ON FINAL MANEUVER FIRE TYPE	FIRE MAY HAVE CAUSED FLIGHT YES A CONTROL LOSS	BEST CONTROL WAS PITCH. NO A CONTROLS DETERIORATED.		CONTROL STICK WOULD NOT MOVE NO A Left of Center in Roll Axis, Engine Surged	FIRE MAY HAVE CAUSED FLIGHT YES K Control Loss.	NOSE OF A/C DROPPED. BACK NO A PRESSURE RAISED IT, BUT MOT THE SECOND TIME IT HAPPENED	LOST FLIGHT CONTROLS YES A	PILOT USED RAM AIR AND STATED MO A That it helped, then transmitted that he had to eject	V č	STICK FROZE FROM LOSS OF YES A HYDRAULIC FLUID.
	URE LOST ON URE LOST		BEST CONTROL WAS PITCH CONTROLS DETERIORATE		CONTROL STICK WOULD NOT Left of Center in Roll J Engine Surged	FIRE MAY HAVE CAUSED FLIG Control Loss.	NOSE OF A/C OROPPED. BACK PRESSURE RAISED IT, BUT THE SECOND TIME IT HAPP		PILOT USED RAM AIR AND ST THAT IT HELPED, THEN TRAN THAT HE HAD TO EJECT		STICK FROZE FROM LOSS HYDRAULIC FLUID.
DESCRIPTION OF HIT OR DAMAGE WHICH RESULTED IN LOSS OF FLIGHT CONTROL SYS. A/C OBSERVED TO BE ON FIRE. AS FIRE GOT BATTERY COMPARTMENT AREA. AS FIRE GOT BATTER HIT, ALL LIGHTS ON EXCEPT FIREWARNING. RUDDER CONTROLS SEVERED OBSERVER NOTICED FLAK BEHIND A/C FOLLOWED BY STREAMING. WHITE VAPOR. NO FLIGHT CONTROL DATA. MASTER CAUTION AND RUDDER TRAVEL LIGHTS CONTROL DATA. MASTER CAUTION AND RUDDER TRAVEL LIGHT CONTROL DATA. MASTER CAUTION AND RUDDER TRAVEL LIGHTS CONTROL DATA. MASTER CAUTION AND RUDDER TRAVEL LIGHTS CONTROL DATA. FIRE ON LEFT SIDE OF A/C. A/C PITCHED DOWN. FIRE AND FIRE COMING OUT OF A/C LOWER LEFT FULOT FELT HIT PILOT FELT HIT A/C HIT; COCKPIT FILLED WITH SMOKE, FLIGHT CONTROLS THER LOST. PILOT FELT HIT PILOT FELT WIT SYSTEM WERE OUT. FXTERMALL FIRE FED BY STREAMIG HYDRAULIC FLUID. FIRE FED BY STREAMIGH HYDRAULIC FLUID.	HYDRAULIC Pi				LOST	<u> </u>			FAILED		LOST About I <mark>-</mark> min
		A/C OBSERVED TO BE ON FIRE, FIRE COMING FROM Battery compartment area. As fire Got Worse, a/C started pitching and yawing.	AFTER HIT, ALL LIGHTS ON EXCEPT FIREWARNING. RUDDER CONTROLS SEVERED	<b>N</b> H	MASTER CAUTION AND RUDDER TRAVEL LIGHTS Came on. After Hits Lost Both Boost Pumps And Ac generator.	FIRE ON LEFT SIDE OF A/C. A/C PITCHED DOWN.	A/C HIT IN REAR. FUEL FLOW INDICATOR Spinning Rapidly and hydraulic system Failing.		PILOT FELT HIT	A/C HIT; COCKPIT FILLED WITH SWOKE, FLIGHT CONTROLS THEN LOST.	PILOT FELT EXPLOSION AND NOTED THAT P <sub>2</sub> and utility systems were out. External fire fed by streaming hydraulic fluid.

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# TABLE IV

# FIRE

INCIDENT	RESULTANT DANAGE	LOCATION OF HIT	PROJECTILE Type	TYPE KILL
4	FIRE IN FUSELAGE AND ENGINE INTAKES, CONPRESSOR STALL.	AFT SECTION	~	-
5	FIRE IN AFT SECTION	—	_	-
8	FIRE AND CONTROL LOSS	_	-	A
13	FIRE	?	-	A
14	FIRE	ELECTRONICS 1N Fus. Bottom		В
15	FIRE IN TAIL		_	A
17	FIRE	AFT SECTION	GROUND FIRE	•
19	FIRE OBSERVED CONING FROM TAIL		AUTO WEAPON FIRE	A
20	FIRE FROM WING ROOT	AFT OF COCKPIT	AUTO WEAPON FIRE	
21	FIRE, LOST STABILIZER CONTROL	ACCESSORY SECTION	. 37 NM	A
23	FIRE LEFT WING ROOT; PIECES FROM A/C, PART OF LEFT WING MISSING	_	-	<b>A</b>
25	BURNING FROM FORWARD OF INTAKES		_	•
28	FIRE AT HIT LOCATION , FLIGHT CONTROL LOSS	BETWEEN RIGHT WING AND FUSELAGE	37 OR 57 MN	A
29	FIRE OBSERVED AT AFT SECTION, FLIGHT CONTROLS AND HYDRAULICS.			•
32	FIRE	-	-	<b>A</b> .
33	FIRE IN TAIL PIPE, HIS GUN EXPLOADED		-	٨
34	FIRE FWD OF COCKPIT IN-BATTERY COMPARTMENT, ELECTRIC AND FLIGHT CONTROLS	_	20 OR 37 NM	A
39	FIRE IN BONBBAY ENGINE	BOMB BAY	SNALL ARNS OR 37 NM	•
42	AFT SECTION OM FIRE	-	-	A
43	FIRE	_	57 N.N	•

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INCIDENT	RESULTANT DAMAGE	LOCATION OF HIT	PROJECTILE Type	TYPE Kill
46	FIRE, LEFT SIDE OF AIRCRAFT FLIGHT CONTROLS		_	K
50	FIRE ON BOTTON OF FUSELAGE	-	AUTO WEAPON	•
52	BURST INTO FLAMES, AIRCRAFT SHREDDED	DETONATED 20 FT UNDER AIRCRAFT BOTTON	SAM	KK
53	FIRE, AIRCRAFT DISINTEGRATED	-	(SAN )	ĸ
54	FIRE, LOST FLIGHT CONTROL		_	A
57	FIRE (TRAILING)	-	37 N N	-
58	FIRE, LOST FLIGHT CONTROL	-	-	A
59 -	FIRE, LOST UTILITY HYDRAULIC SYSTEN.LOST Flight control	TAIL HOOK AREA	57 MM	A
60	FIRE		OWN BONB	A

# TABLE IV (Cont'd)

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DESCRIPTIVE NOTES (Type of report and inclusive dates)	······		
AUTHOR(S) (Lost name, first name, initial)			
O'Brien, J. D.			
Meiselman, Jay M., 2d Lt, USAF			
REPORT DATE September 1967	78. TOTAL NO. 01	F PAGES	76. NO. OF REFS
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. Task 136814			
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AVAILABILITY/LIMITATION NOTICES In addition	n to security :	requirem	ents which apply to
this document and must be met, each Government must have prior approva	n transmittal	outside t	he agencies of the U.
tory (FDTS), Wright-Patterson Air			
SUPPLEMENTARY NOTES	12. SPONSORING N	ILITARY ACT	IVITY
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ABSTRACT (UNCLASSIFIED)			
This report contains an analysis o ons Systems Evaluation Group (WSE) time period of 1 February 1965 to 31 for the purpose of providing an insig	G) Compendiu . January 1960 ht into areas	m of Air 5. The a such as f	craft Losses for the nalysis is performed threat, causes of air-
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DEPARTMENT OF THE AIR FORCE HEADQUARTERS 88TH AIR BASE WING (AFMC) WRIGHT-PATTERSON AIR FORCE BASE OHIO

### MEMORANDUM FOR DTIC-RS

1 4 JUN 2002

ATTN: Kelly Akers Defense Technology Information Center 8725 John J. Kingman Rd, Suite 0944 Ft Belvoir VA 22060-6218

FROM: 88 CG/SCCMF 4375 5<sup>th</sup> Street Rm 150 WPAFB OH 45433-7802

SUBJECT: Change of Classification and Distribution Statement for Document Number's AD-C016-682 and AD-385-882

1. The attached 16 April 2001 letter from W. Howard Plunkett requests classification review of subject technical reports and change of distribution requirements from "Limited Distribution" to "Approved for Public Release; Distribution Unlimited."

2. The requestor handcarried this request to the FOIA office, therefore it was treated as a FOIA request. Subsequently, it was reviewed by the Subject Matter Expert, Don Voyls, 46 OGM/OL-AC. His analysis states that the documents appear to be fully releasable. Capt Stephanie Masoni, his Security Manager, attached a memo indicating that she concurs to full release of the reports.

3. Please take the appropriate action to make subject technical reports available for public dissemination. The requester has been notified of this action. Point of contact at 88 CG/SCCMF is Lynn Kane at DSN 674-8189.

Sincerely,

how M. Am

SHEREE M. COON Freedom of Information Act Manager Management Services Branch Information Management Division

Attachments:

- 1. AFMC Form 559, 6 June 2002
- 2. 46 OG/OGM/OL-AC Memo, 6 Jun 2002
- 3. Don Voyls Memo, 5 Jun 2002
- 4. Initial Request Letter, 16 Apr 2001
- 5. AD 385-882
- 6. AD C016 682
- 7. 88CG/SCCMF Ltr to Requestor, 14 Jun 02



### DEPARTMENT OF THE AIR FORCE HEADQUARTERS 46TH TEST WING (AFMC) EGLIN AIR FORCE BASE, FLORIDA

6 June 2002

## MEMORANDUM FOR 46 OG/OGM/OL-AC (Mr. Richard E. Colclough)

### FROM: CAPT STEPHANIE MASONI (Unit Security Manager)

SUBJECT: Classification and Limited Distribution Requirement Review for Freedom of Information Act (FOIA) Case #010421LK, W. Howard Plunkett.

I have reviewed the two documents in support of the attached FOIA request, and concur with Mr. Donald Voyls(memo attached); both documents are fully releasable to the public.

Stephonie C Mason

Stephanie C. Masoni, Capt, USAF 46 OG/OGM/OL-AC Security Manager

Attachment Memo dated 5 June 02 (Mr. Voyls)

## MEMO TO: 46<sup>TH</sup> OG/OGM/OL-AC

### SUBJECT: Freedom of Information Act (FOIA) Case #010421LK, W. Howard Plunkett

I have reviewed the two documents in support of the attached FOIA request iaw the guidelines provided. Based on the instructions and directions provided (DoD Regulation 5400.7, Chapter 3 and DoD Directive 5230.25) I could not find any exemptions for withholding these documents from public disclosure. Therefore the documents appear to be Fully Releasable.

Review of AFFDL-TR-67-118, An Analysis of F-105 Combat Losses in SEA resulted in the following findings. The report was declassified 22 August 1990. The F-105 has been out of the USAF inventory for some time and I do not know of any other country operating that aircraft. The report draws general trends on F-105 combat losses. Some limited identification on the cause of loss based on hit location is provided on page 4 but in my opinion is common knowledge. In conclusion the author states that more detailed data was needed to increase the confidence of the analysis.

Review of AFFDL-TR-77-115, A Comparative Analysis of USAF Fixed Wing Aircraft Losses in Southeast Asia Combat, AD-C016682 resulted in the following findings. The report was declassified 31 December 1988. The aircraft covered in this report with the exception of the B-52 and C/AC-130 are no longer in the USAF inventory. To my knowledge the German Air Force still flies the F-4. The report provides statistics on aircraft and crew losses relative to a variety of situations such as altitude, location, threat category, date, etc for all the aircraft except the B-52 and AC-130. Statistics on losses due to general causes (fuel fire, engine failure, flight controls) were provided. Data on the B-52, page 51, and AC-130, page 56, were limited to reasons for crash with no threat stated. Comparative analyses on the F-4 vs. F-105, one engine vs. two, and effectiveness of vulnerability reduction features were included. Though I personally have some small reservation on releasing the F-4 portion of this report to the public, I could not find any justification in the directives to withhold it.

According to the FOIA office, a memo from the classification officer stating that the reports are properly classified and marked is needed. Also, to aid in filling out form DD Form 2086 block 2, I spent 2 hours on Coordination/Approval/Denial and 11 hours on the review.

Let me know if you have any questions.

Don Voyls, 5 June 2002