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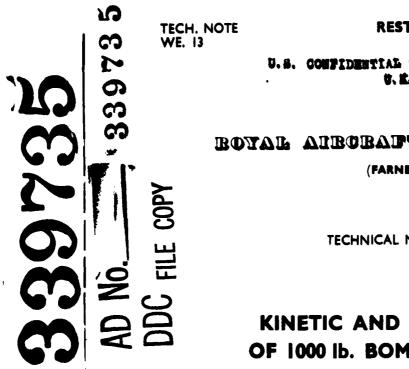
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TECH. NOTE WE. 13

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(FARNBOROUGH)

TECHNICAL NOTE No. WE. 13

# KINETIC AND SOLAR HEATING OF 1000 Ib. BOMBS EXAMINED AT R.A.F. IDRIS, AUGUST - SEPTEMBER, 1962

Ьу

W. F. Fielding, B.Sc., A.Inst.P.

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Technical Note No. WE 13

February, 1963

#### ROYAL AIRCRAFT ESTABLISHMENT

(FARNBOROUGH)

KINETIC AND SOLAR HEATING OF 1000 LB BONBS EXAMINED AT R.A.F. IDRIS, AUGUST-SEPTEMBER, 1962

Ъy

W. F. Fielding, B.Sc., A.Inst.P.

RAE Ref: LSW/247/04

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#### SUMMARY

This Note gives the results of experiments carried out with 1000 lb Mk.10 bombs at R.A.F. Idris, August-September, 1962.

Values are given for the temperatures recorded on and within bombs stored in full sunlight. It is inferred that there is little danger of the filling melting even under the worst tropical conditions.

Results of flight trials confirm the tables of speed restrictions previously issued (R.A.E. Tech. Note No. Arm 704) for external carriage of V.T. fuze No.906 and 1000 lb bombs.

Some information is given regarding the met. conditions at Idris during the period of the trials.

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Technical Note No. WE 13

#### LIST OF CONTENTS

		Page
1	INTRODUCTION	4
2	WEATHER CONDITIONS	4
3	INSTRUMENTATION	5
	3.1 Bomb 3.2 Recording	5 5
4	GROUND TRIALS	6
	4.1 Bomb on ground trolley 4.2 Bomb on pylon 4.3 Results obtained	6 6 7
5	FLIGHT TRIALS	8
	5.1 Bomb in bomb bay 5.2 Bomb on pylon 5.3 Results obtained	8 8 9
6	COMPARISON WITH THEORETICAL FIGURES	10
7	CONCLUSIONS	11
8	ACKNOWLEDGEMENTS	11
LIS	T OF REFERENCES	11
ADV.	ANCE DISTRIBUTION LIST	12
APPI	ENDICES 1 AND 2	13-16
TAB	LES 1 - 8	1 7-24
ILLI	USTRATIONS - Figs.1-12	-
DET.	ACHABLE ABSTRACT CAPDS	-

#### LIST OF APPENDICES

#### Appendix

14

.

1	Met. conditions for the trial period at R.A.F. Idris	13-15
2	The flight recorder	16

#### LIGT OF TABLES

Table		
1	Bomb temperatures for ground trial No.1. 1000 lb Mk.10 bomb, nose pointing to south	17
2	Bomb temperatures for ground trial No.2. 1000 lb Mk.10 bomb, nose pointing to east	18
3	Bomb temperatures for ground trial No.3. 1000 lb Mk.10 bomb, nose pointing to east	19
4	Bomb temperatures for ground trial No.4. 1000 lb Mk.10 bomb on Buccaneer pylon, nose pointing west	20



# U.S. CONSIDENTIAL MODIFIED HANDLING AUTHORIZED

Technical Note No. WE 13

4

#### LIST OF TABLES (CONTD)

Table		Page
5	1000 lb Mk.10 bomb carried in Buccaneer bomb bay. Temperatures during flight at Idris, 20th September, 1962	21
6	1000 lb Mk.10 bomb carried on Buccaneer port inboard pylon. Temperatures obtained during flight at Idris, 20th September 1962	22
7	Calculated flight limitations for V.T. fuzes Nos.906 and 907 when their temperatures must not exceed 70°C	23
8	Calculated flight limitations for 1000 lb bombs when the bomb filling temperature must not exceed 80°C at a depth of 3 cms after 40 minutes flight	24
	LIST OF ILLUSTRATIONS	
		<u>Fig.</u>
	nce thermometer elements in bomb	1
	Mk.10 bomb on ground. Trial No.1. Temperatures recorded on bomb	•
	Bomb nose to south	2
	Mk.10 bomb on ground. Trial No.1. Temperatures recorded at ce of bomb filling. Bomb nose to south	3
	Mk.10 bomb on ground. Trial No.1. Temperatures recorded at a	)
	of 3 cms into the filling. Bomb nose to south	4
	Mk.10 bomb on ground. Trial No.2. Temperatures recorded on bomb	-
	Bomb nose to east	5
-	Mk.10 bomb on ground, Trial No.2. Temperatures recorded at	-
	ce of filling. Bomb nose to east	6
1000 1b	Mk.10 bomb on ground. Trial No.2. Temperatures recorded at a	
	of 3 cms into the filling. Bomb nose to east	7
	Mk.10 bomb on ground. Trial No.3. Temperatures recorded on bomb	
	Bomb nose to east	8
	Mk.10 bomb on ground. Trial No.3. Temperatures recorded at	•
	ce of filling. Bomb nose to east	9
	Mk.10 bomb on ground. Trial No.3. Temperatures recorded at a	40
	of 3 cms into the filling. Bomb nose to east.	10
	Mk.10 bomb in Buccaneer bomb bay. Temperatures recorded in bomb V.T. fuze during flight. Trial No.5	11
	Mk.10 bomb on Buccaneer port inboard pylon. Temperatures	
	ded in bomb and V.T. fuze during flight. Trial No. 6	12

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#### 1 INTRODUCTION

During the recent Buccaneer tropical trials at R.A.F. Idris, Libya, August-September, 1962, an R.A.E. team attended to study the behaviour of various weapons under tropical or near-tropical conditions.

This Note is concerned solely with tests on the 1000 lb Mk.10 bomb (although the results are applicable equally to the 1000 lb N.1 bomb) and covers the following conditions:-

(a) heating of the bomb when stored on the ground in the sun,

(b) heating of the bomb on the aircraft pylon, with the aircraft standing in the sun on the ground,

(c) heating of the bomb in the aircraft bomb bay during flight,

(d) kinetic heating of the bomb in flight when carried externally on the aircraft pylon.

Case (a) was covered adequately in the period before the aircraft arrived at Idris, but owing to the short time during which a serviceable aeroplane was available (5 days) for all the tropical trials, only one test could be made under each of cases (b) to (d).

In addition to the heat measurements made on the bombs, solar radiation intensity was measured near by and data on shade temperature, wind speed and direction, humidity and low cloud coverage was obtained from the Idris Met. Office.

Most of the trials undertaken had as aim the verification of previouslypropounded theories on bomb heating under near-tropical conditions. This was done, as far as possible, but flight limitations imposed on the aircraft (550 knots maximum speed, with air temperature not to exceed 35°C) made it impossible to obtain as high bomb temperature increases over U.K. summer flight conditions (lower air temperatures but higher aircraft speeds) as had been hoped for.

#### 2 WEATHER CONDITIONS

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Owing to delay in starting the trials (August 21st instead of mid-July) the air temperatures experienced at Idris were not as high as had been expected, being generally in the low 30s centigrade, with the occasional excursion to  $35^{\circ}$ C.

There were many cloudless days, but cloud cover tended to increase towards the end of the trials (mid-September) and for several days at the end of August sand-storms were blowing.



Technical Note No. WE 13

Appendix 1 gives details of :-

shade temperature, humidity, low-cloud cover, wind speed and direction, solar radiation,

for the days on which the trials reported in this Note were carried out.

The Appendix also gives maximum and minimum shade temperatures for the whole period of the trials.

The solar radiation was measured with a Kipp Solarimeter which had been calibrated at Kew Observatory.

#### 3 INSTRUMENTATION

3.1 <u>Bomb</u>

The 1000 lb Mk.10 bomb was fitted with eighteen platinum-law resistance thermometer elements arranged in two groups of nine as shown in Fig.1.

Of each group of three elements, one was just below the surface of the bomb skin, one was at the surface of the bomb filling and one was 3 cms deep in the filling.

Full details, with diagrams, are given in Ref.1, the only difference being that for the trials described in this Note the cables from the thermometer elements were brought out from the rear fuze well and not from a hole cut in the top of the bomb.

The instrumented bomb was filled with an inert substance having heat properties as close as possible to those of Torpex 2A (see Ref.1, Appendix 1) and connexions were provided into the nose fuze well so that a V.T. fuze, instrumented with two resistance thermometer elements (one on the power pack, one on the amplifier) could be connected if desired.

#### 3.2 <u>Recording</u>

Two methods were used for recording temperatures. For the ground trials, the leads from the thermometer elements were connected to a selector box in which each element could, in turn, be connected to a ratiometer which gave a direct temperature reading.

For flight trials, the same method could not be used and a photographic recorder was constructed to fit inside the bomb tail, mounted on anti-vibration mountings. (See Appendix 2 for details.) Once every minute twelve of the thermometer elements in the bomb (the desired twelve being selected before the trial) were sampled automatically, connected to a ratiometer, and the ratiometer dial together with an adjacent clock photographed after allowing five secs for the ratiometer needle to take up its correct position on the dial.

- 5 -

#### Technical Note No. WE 13

This cycle of twelve samples per minute could be repeated for as long as the recorder was switched on. Power from the recorder was obtained from the pylon or bomb bay (for external or internal carriage respectively) and fed to it either through a two-pin plug in the base of the tail (internal carriage only) or via leads which entered the bomb through the E.L. fuzing position and emerged from the tail fuze well to be connected to the recorder.

In practice, it was found better to sample eleven resistance elements in turn and to leave the twelfth position unconnected to form an identification point once in each cycle.

The accuracy of the system is such as to give mean temperature rises correct to  $\pm 2^{\circ}$ C. Before test, care was taken to ensure that the bomb was stored in such a way that its temperature was reasonably constant throughout. The mean value obtained from a pre-test record was taken as the initial temperature of the bomb. Bomb temperatures throughout the trial were then obtained by adding this result to the values of temperature rise obtained from the film.

#### 4 GROUND TRIALS

#### 4.1 Bomb on ground trolley

Three trials were carried out with the bomb on a bomb trolley inside a compound such that it was effectively shielded from wind but was in the full sun for most of the daylight hours (06.30-20.00 hrs local time, G.K.T. + 2 hrs). The shielding from wind was not intentional.

The trolley was placed in position after dark the previous evening (this was dictated by other trials carried out in parallel) and left in position all night, recording starting at 04.00 hrs the next morning and continuing until 04.00 hrs of the next day.

The conditions of test were as follows:-

- Trial 1, bomb in north-south orientation with nose to south. No tail fitted. V.T. fuze No.906 fitted.
- Trial 2, bomb in east-west orientation with nose to east. No tail fitted. No V.T. fuze fitted.
- Trial 3, bomb in east-west orientation with nose to east. Tail No.114 fitted. No V.T. fuze fitted.

Tables 1-3 show the temperatures attained during the three trials together with the relevant met. data; Figs.2-10 show the temperature data in graphical form.

#### 4.2 Bomb on pylon

One trial (No.4) was conducted with the bomb mounted on the inboard port pylon of the sircraft parked on the airfield with its nose pointing west.

Again, the bomb was put into position the previous evening, but owing to aircraft servicing requirements recording was only possible between 09.30 hrs and 14.00 hrs, during which time the bomb was almost completely shaded from direct sunlight by the aircraft wing. It was not shielded from such wind as was blowing.

Table 4 shows the results obtained.

#### 4.3 Results obtained

4.3.1 Comparison between Figs.2, 5 and 8 shows clearly the different effect arising out of the direction in which the bomb was pointing. In Fig.2, with bomb pointing south, the sun was on the port side during the morning, then changed to the starboard side for the hottest part of the day (14.00-16.00 local time).

In Figs.5 and 8, the bomb nose pointed to the east and the starboard side was in the sun all day long, with the port side partly shaded from direct sunlight by the top of the bomb.

For all three cases, the bottom of the bomb never received full, direct sunlight and the temperature rises there are only slightly greater than those of the ambient shade temperature, augmented, perhaps, by some slight reflexion from the ground.

Figs.3, 6 and 9 show that in only one case does the temperature at the surface of the bomb filling exceed  $50^{\circ}$ C, and that only marginally.

During the hours of darkness the bomb cools steadily and by dawn next morning is completely cooled to air ambient temperature which, for the duration of the trials, was generally slightly below 20°C.

Considering the state of affairs at a depth of 3 cms into the filling (Figs.4, 7 and 10) we see that the maximum mean temperatures for the three trials are 42, 34 and 38°C respectively with an over-all mean of 38°C. Individual points exceed these values for a time, but nowhere do the temperatures approach near to the melting point of Torpex (70-80°C). The amount of filling to a depth of 3 cms is about  $\frac{1}{3}$  the capacity of the bomb and consequently the major part of the filling will be at a temperature of less than the 38°C quoted above, which itself represents a temperature rise of about 18°C.

Even under maximum temperature atmosphere conditions the initial morning temperature would probably be not very much in excess of 20°C. If the bomb filling temperature rose by twice the amount found in these trials (and this is most unlikely) it would still be considerably below melting point.

It appears, then, that there is little danger of the bomb filling melting (except perhaps for very local melting at the surface of the filling) during ground storage and that a maximum mean temperature of 35°C for the filling, as has been assumed previously (Ref.1), is reasonable.

Some protection is usually provided for stored bombs, and even a tarpaulin sheet would reduce local temperature rises considerably (see para.4.3.3).

- 7 -

#### Technical Note No. WE 13

4.3.2 The V.T. fuze, (Trial No.1, Table 1) did not appear to heat up quite so much as might have been expected. It is, however, a light structure, very susceptibel to cooling from changes of ambient temperature and by conduction into the bomb.

4.3.3 On the pylon, very little temperature change was noticed during the period of the test, since the bomb was almost completely in the shade from the aircraft wing throughout the whole period and was also exposed to the wind (up to 7 knots from a south-easterly direction). Table 4 shows the temperatures reached and the met. conditions for the day.

This trial shows how effective some shielding from direct sunlight can be in keeping down bomb temperatures; trials carried out on other stores at the same period showed that a simple awning of dark cloth on a light frame made a considerable reduction in store temperature as against the case of no shielding.

It is probable that on the Buccaneer less shading would have occurred on the outboard pylons, but as these are not yet available, it could not be tried.

Only one trial could be carried out on the pylon since this was the only occasion on which the aircraft stood out in the sun for a long period.

#### 5 FLIGHT\_TRIALS

#### 5.1 Bomb in bomb bay

Since the available flying time of the aeroplane was so short, only one flight could be carried out with the instrumented bomb and three uninstrumented ones in the bomb bay, with the instrumented one in the forward port position when the bomb doors were closed.

The 906 V.T. fuze was fitted to the bomb and in addition "Temp-plate" temperature indicators were attached to the ejector release units as close to the cartridges as possible.

The flight was made with bomb doors closed under the following conditions (Trial No.5):-

Speed 550 kts (M = 0.81). Height 500 ft. Duration 40 minutes. Air temperature 30°C.

Table 5 gives the temperatures recorded in bomb and fuze. Fig.11 is a plot of the temperatures.

#### 5.2 Bomb on pylon

Again, only one flight was possible (Trial No.6) with the instrumented bomb on the port inner pylon and an uninstrumented one on the corresponding starboard pylon.

- 8 -

Technical Note No. WE 13

Before arriving at Idris the pylons had not been fitted to this aircraft, even for trial, and consequently no power supply to run the recorder was available in the pylon. One was eventually improvised, sufficient for this one flight.

With the pylons supplied with this aircraft there were no fairings to fill the gaps between bomb and sole plate at bomb nose and tail, and in consequence there was a free path for air to be forced into the pylon and over the E.R.U. with a possibility of seriously overheating it<sup>2</sup>.

Consequently an initial flight was carried out at a speed of 450 kts (with the instrumented bomb replaced by an uninstrumented one) with the E.R.Us fitted with "Temp-plate" indicators as for the internal carriage. After the flight, the E.R.U. temperatures had not exceeded 65°C and it was decided to be safe to carry the instrumented bomb at 550 kts (aircraft maximum for those temperature conditions).

Table 6 gives the temperatures measured; Fig.12 gives the temperature data in graphical form.

#### 5.3 <u>Results obtained</u>

5.3.1 In the bomb bay, most of the heat is derived from internal sources (hot air exhausted into the bomb bay) rather than from kinetic heating of the circraft skin. This internal heating takes place whenever the engines are run on the ground or in the air, and pre-flight running of them accounts for the high  $(32^{\circ}C)$  initial bomb temperature (Fig.11).

Only at the end of the flight was the temperature beginning to level off on the bomb skin, whilst it was still rising inside. In an empty bomb bay, air temperatures of about  $80^{\circ}$ C have been recorded. The large thermal inertia of 4 × 1000 lb bombs is such that they are here settling down to a steady temperature well below  $80^{\circ}$ C. This accords with the results of trials undertaken on a mock-up bomb bay<sup>2</sup> to investigate the kinetic heating of bombs placed in it. Those bombs took longer to heat, since the heating was purely by convection from the heated bomb bay skin, there being no flow of hot air as in this case, but in no case, even after prolonged heating, did they attain a temperature anything like the air temperature reached in an empty bomb bay under identical heating conditions.

For the flight conditions used, the aircraft skin temperature would have risen to about  $63^{\circ}$ C, slightly lower than recovery temperature ( $66^{\circ}$ C), so there would have been some tendency for heat from the hot air flow to flow outwards from the bomb bay as well as into the bombs.

The V.T. fuze, being a light structure, heats up more than does the bomb and its final temperature ( $65^{\circ}$ C mean) is probably a reasonable assessment of effective bomb bay temperature. In this flight its temperature was such that its sensitivity would be almost halved; when higher speeds become possible with this aircraft it will over-heat even more unless the bomb bay hot air flow can be reduced.

> - 9 -RESTRICTED

#### Technical Note No. WE 13

The E.R.U. cartridges did not overheat in the bomb bay. Their maximum temperatures being less than 65°C.

5.3.2 On the pylon, the bomb skin quickly rises to within a few degrees of recovery temperature, as would be expected, but there is no serious overheating of the bomb filling during the flight period. The limitation is that the filling temperature must not exceed 80°C at a depth of 3 cms. Under the conditions of this flight (speed M = 0.81, ambient temperature  $30^{\circ}$ C) Ref.1 shows that flight should be possible at M = 1.65 for 40 minutes before overheating of the filling occurred.

The same paper also shows that the corresponding flight limitation on the V.T. fuze (before its temperature reached  $70^{\circ}$ C) would be M = 0.9. This agrees well with the 65°C found in this flight at M = 0.81.

After the flight the E.R.U. cartridges were found to have reached a temperature of  $75^{\circ}$ C which exceeds the temperature (70°C) to which they are tested.

Until fairings are provided on the pylon, it seems likely that E.R.U. and V.T. fuse temperatures will provide flight limitations rather than bomb temperatures.

After the flight, the two upper tail attachment sorews (No.114 tail) were loose by about one turn, although they were tightened as hard as possible before flight.

#### 6 COMPARISON WITH THEORETICAL FIGURES

A previous paper<sup>4</sup> showed that for a homogeneous body such as a conventional bomb, a good approximation to the temperatures occuring at any point within the filling after a given time from start of heating, could be obtained by treating the bomb as a cylinder of explosive and using standard conduction theory for such a body, with a known skin temperature.

Table 6 and Fig.12 show the temperatures calculated on this basis for a point 3 cms into the bomb filling, using the measured values of bomb skin temperature. These calculated values agree with the measured ones at the same depth to within  $2-3^{\circ}$ C, the calculated ones being slightly higher than the measured ones.

This is to be expected, since the theory neglects, for simplification, the bitumen layer between bomb case and filling, but the agreement is close enough to permit use of theoretical values in future, at least initially, with, possibly, confirmatory flight tests later on.

The values of flight limitations given in Ref.1 should now be considered as accurate for external carriage of 1000 lb bombs and 906 V.T. fuzes. For convenience, these flight limitations are reproduced here as Tables 7 and 8.

- 10 -

Technical Note No. WE 13

#### 7 <u>CONCLUSIONS</u>

(1) The use of theoretical temperatures for a regularly-shaped and homogeneous store, as advocated in R.A.E. Tech. Note No. Arm 702, is justified, and gives results slightly higher than would be obtained in practice.

(2) Bombs stored out in the sum under tropical or semi-tropical conditions will have a maximum mean filling temperature of about 35°C. At no point are they likely to become hot enough to melt the filling.

A simple awning on a light frame would reduce the heating considerably.

(3) When the Buccaneer is standing on the ground, bombs on the inboard pylons do not suffer much temperature increase, due to shielding from the sun by the aircraft wing.

(4) No.906 V.T. fuses overheat when carried in the Buccaneer bomb bay due to the high temperatures there, arising from the injection of hot air from elsewhere in the aircraft.

(5) The tables of flight limitations for externally-carried 1000 lb bombs and No.906 V.T. fuzes given in Tech. Note No. Arm 704 are reasonably accurate.

(6) Fairings are required between bomb and pylon sole plate to prevent hot air entering the pylon and overheating the ejector release units.

#### 8 ACKNOTLEDGERENTS

Acknowledgement is made to Mr. R.S. Howell, Jeapons Dept., R.A.E., who built and tested the photographic recorder and to the O.C, and personnel of The Overseas Experimental Unit, at R.A.F. Idris for the invaluable assistance rendered in many ways throughout the period of the trial.

Ref. No.	Author	<u>Title, etc.</u>
1	Fielding, <sup>W</sup> .F.	The kinetic heating of externally-carried conven- tional aircraft bombs. R.A.E. Tech. Note No. Arm 704. October, 1961.
2	Russell, C.P. Jnr	Investigation of aerodynamic heating of external stores (F-100, F-101, F-105 aircraft). APGC-TR-60-9. April, 1960.
3	Chaloner, J.H. Fielding, W.F.	Tests to simulate the effects of kinetic heating of bombs and fuses carried internally in the TSR2. R.A.E. Tech. Note No. Arm 677. December, 1960.
4	Fielding, W.F.	A theoretical examination of the problem of the kinetic heating of externally-carried aircraft bombs. R.A.E. Tech. Note No. Arm 702, October, 1961.
		- 11 -
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LIST OF REFERENCES

#### Technical Note No. WE 13

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#### ATTACHED:

Appendix 1 (including Figs. App.1/Figs.1 & 2, Drg. Nos. WE.R.3192, 3193) Appendix 2 (including Figs. App.2/Figs.1-3, (Drg. Nos. WE.R.3194, 3195 and Tables 1-8 (Neg. No. 160,661.) Illustrations, Figs.1-12 (Drg. Nos. WE.R.3196-3207) Detachable Abstract Cards

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#### APPENDIX 1

#### MET. CONDITIONS FOR THE TRIAL PERIOD AT R.A.F. IDRIS

During the period of the trials solar radiation intensity was measured on a good many days at a site adjacent to the trial.

Forevery day of the trials (21st August to 20th September) the following data was obtained from the Idris Met. Office for hourly periods throughout the whole day:-

Shade temperature (<sup>o</sup>C). Wind speed and direction. Humidity. Sea-level pressure. Low cloud cover.

Table 1 of this Appendix gives this data for the days on which Trials 1-6 were carried out.

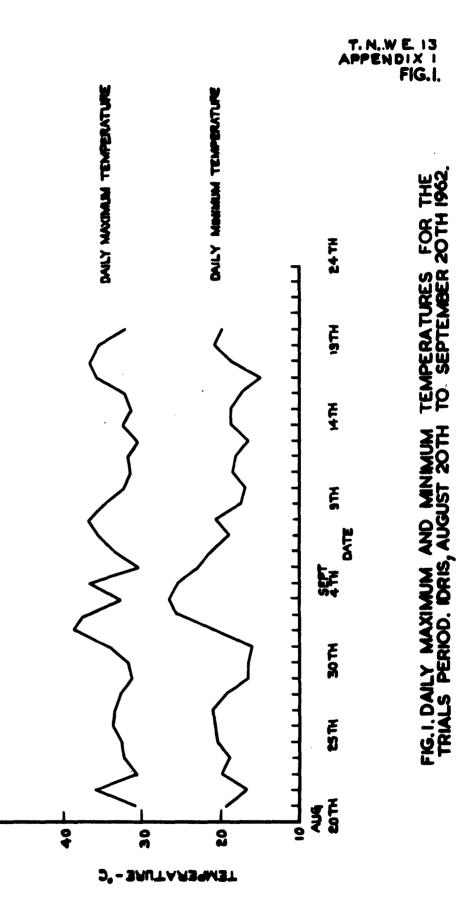
Figs.1 and 2 of this Appendix shows the daily maximum and minimum temperatures throughout the whole period together with the number of hours per day on which the temperature:

- (a) exceeded 30°C
- (b) exceeded 35°C.

ATTACHED:

Drg. Nos. WE.R. 3192-3193

- 13 -



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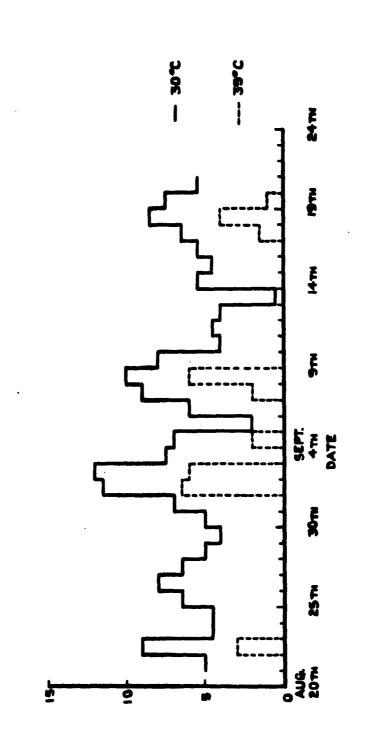
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T.N. WE.IL APPENDIX 1. FIG.2.

FIG.2. NUMBER OF HOURS PER DAY FOR WHICH THE TEMPERATURE EXCEEDED 30°C AND 35°C AT IDRIS, AUGUST 20TH TO SEPTEMBER 20TH 1962.

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NUMBER OF HOURS

TABLE 1 Met, conditions for the days of Trials

										دو منصوف د ده مد به			
Quantity peasured	Date	01 <b>.00</b>	02.00	03,00	04.00	05,00	06.00	07,00	08,00	09.00	10.00	11.00	12,00
Shade temperature (°C) Wind speed (knots) Wind direction (degrees) Humidity (%) See-level pressure (128) Low cloud cover Solar intensity (milliwatts/cm <sup>2</sup> )	22nd Aug. 1962	19.8 Calm Calm 85 1014.1 N11	19.1 Calm Calm 85 1013.9 Nil	17.6 Calm Calm 84 1013.7 N11		17.0 Calm Calm 81 1013.7 N11 -	16,6 Calm Calm 77 1013,3 Nil	17,8 Calm Calm 61 1013,5 N11	21.4 Calm Calm 53 (013.5 N11 5.6	22,4 Calm Calm 60 1013,9 111 36,3	30.0 02 320 37 1014,2 N11 36.9	31.4 06 350 30 1014.5 N11 71.5	33.2 06 340 26 1014.7 N11 81.6
Shade temperature (oC) Wind speed (knots) Wind direction (degrees) Humidity (%) See-level pressure (MES) Low aloud cover Solar intensity (milliwatts/cm <sup>2</sup> )	24th Aug. 1962	21,2 02 060 86 1018,8 1/8	20,2 02 270 89 1018,4 1/8	19.4 02 240 88 1018.1 1/8	19.0 03 240 91 1018.0 111 -	18.8 02 220 92 1018.2 2/8 -	19,2 02 210 92 1013,5 2/8 -	19,2 01 :30 92 1019,0 2/8	21,2 05 230 88 1019,6 3/8 10	07 260 75 1019.3 1/8	26,6 03 280 71 1019,7 2/8 60,5/35	28,5 08 290 64 1019,9 7/8 83,4/32	29.{ 03 54 1019.1 6/8 £7/33.{
Shade temperature (°C) Wind speed (knots) Wind direction (degrees) Humidity (%) Sem-level pressure (iES) Low cloud cover Solar intensity (milliwatts/cm <sup>2</sup> )	25th Aug. 1962	23.8 05 090 83 1019.7 N11	22,8 06 060 86 1019,3 N11 -	22.4 07 070 88 1019.1 N11 -	22.0 04 090 92 1013.9 N11 -	21.0 04 060 91 1019.0 N11	20,8 03 090 93 1019,0 1/8	20,2 03 110 94 1019,5 1/8 -	23,2 06 110 85 1019,8 1/8 -	25.8 12 140 73 1019.9 1/8 -	27.8 13 120 62 1020.3 1/8	30.0 12 110 50 1020.0 2/8	31,1 08 140 43 1019,1 2/3
Shade temperature (°C) Wind speed (knots) Wind direction (degrees) Humidity (%) Sea-level pressure (NDS) Low sloud sever Solar intensity (milliwatts/cm <sup>2</sup> )	27th Aug. 1962	23.4 06 090 75 1017.9 111	23,2 06 100 80 1017.4 N11	22.9 04 090 81 1017.0 N11 -	22.0 06 100 83 1016.7 N11	21.6 06 100 82 1016.3 N\$1 -	21.0 06 110 84 1016.5 N11 -	20,8 05 090 36 1016.9 Nil	23.2 06 73 1017.4 N11 6.5	25.1 05 090 65 1018.1 N11 39	27.0 07 080 55 1018.6 N11 64	29,2 06 010 41 1013,6 311 73,2	
Shade temperature (°C) Wind speed (knots) Wind direction (degrees) Hulidity (%) Sea-level pressure (128) Low sleud cover Solar intensity (milliwatts/cm <sup>2</sup> )	28th Aug. 1962	22,4 04 100 83 1017,8 111 -	22,0 02 130 79 1017,4 N11 -	21,2 02 040 82 1017,1 N11	21,2 02 030 82 1017,0 N11	19.6 02 040 85 1016.7 N11	19,1 02 030 85 1016,9 Nil	19.7 02 030 84 1017.4 311	21.4 03 070 80 1018.0 N11 7.7	25.0 05 090 64 1018.0 N11 42.6	27.4 08 060 51 1013.7 N11 55.5	29,2 06 050 39 1018,8 141 58	30.1 07 030 32 1018.6 1/8 79
Shade temperature (°C) Wind speed (knots) Wind direction (degrees) Humidity (5) Sem-level pressure (:ES) Low eloud cover Solar intensity (milliwatts/om <sup>2</sup> )	15th 8apt. 1962	19,6 Calm Calm 85 1019,6 Nil	18.9 0.2 110 87 1018.8 N11 -	18,2 01 140 90 1016,5 N11	17.4 Calm Calm 90 1018.1 Nil	16,6 Calm Galm 92 1013,1 Nil	16,4 Calm Calm 92 1010.3 Nil	17.0 Calm Calm 91 1016.4 H11	19.0 02 160 83 1016.0 1/8	23,2 06 180 68 1019,1 111	25,5 05 130 22 1019,1 N11 55	27.6 05 120 54 1013.3 1/8 68.8	27,( 07 060 53 1018,( 5/8 84,(

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NOTE: Shado temperatures measured at hft above ground level,

Winds measured at 33 ft above ground level.



- 14 -RESTRICTED

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Technical Note No. WE 13 Appendix 1

TABLE 1 Met, conditions for the days of Trials Nos.1-6

		••••••••					Time (1	ocal tim	10 - 3.M.	.T. + 2 h	<b>rs</b> )							·		
00.00	05,00	06.00	07.00	08,00	09,00	10,00	11,00	12.00	13,00	, 14.00	15.00	16.00	17.00	13,00	19.00	20,00	21.00	22,00	23.00	24,00
17.6 Calm Calm 82 1013.7 N11	17.0 Calm Calm 81 1013.7 Nil	16.6 Calm Calm 77 1013.3 Nil	17.8 Calm Calm 61 1013.5 Nil	Caln Calm 53	22,4 Calm Calm 60 1013.9 111 36,3	30.0 02 320 37 1014.2 N11 36.9	31.4 06 350 30 1014.5 N11 71.5	33.2 06 340 26 1014.7 N11 81,6	34.6 08 330 25 1014.4 N31 84.6	55.0 05 010 28 1014.1 111 84	55.9 07 360 26 1013.8 N11 79	35.0 07 020 25 1013.4 Ni1 65.5	35.0 10 28 1013.2 111 49	33.4 13 030 33 1013.1 N11 34.6	30.6 13 040 44 1013.5 N11 8	28.6 10 050 56 1014.1 N11	27.0 10 040 70 1014.7 ¥11	26.3 06 030 71 1015.4 N11	25.2 02 050 76 1015.5 N 11	24,2 Calm Calm 78 1016,1 W11
19.0 03 240 91 1018.0 111	18.8 02 220 92 1018.2 2/8	19.2 02 210 92 1013.5 2/8	19,2 01 .30 92 1019,0 2/8	21.2 05 230 88 1015.6 3/8 10	07 260 75 1019.3 1/8	26.6 03 280 71 1019.7 2/8 60.5/35	28.5 08 290 64 1019.9 7/8 83. U 32	29.8 03 310 54 1019.4 6/8 57/33.6	5/8	32.3 09 030 46 1018,6 3/8 94	32,2 10 020 41 1018,0 4/8 78	<b>52.</b> 0 12 060 44 1017.7 <b>3</b> /8 66	31.2 12 050 48 1017.3 2/8 44.5	30.0 14 040 54 1017.7 N11 33.5	28.6 13 040 61 1018.0 2/8 6.6	27.4 14 060 67 1018.0 2/8	26.4 08 070 75 1018.9 1/8	25.8 09 080 76 1019.5 N11	25.1 07 070 79 1019.5 N\$1 -	24.2 05 060 82 1019.7 N11
22.0 04 090 92 1018.9 N11	21.0 04 060 91 1019.0 N11	20,8 03 090 93 1019,0 1/8 -	20,2 03 110 94 1019,5 1/8	23.2 06 110 85 1019.8 1/8	25.8 12 140 73 1019.9 1/8	27.8 13 120 62 1020.3 1/8	30.0 12 110 50 1020.0 2/8	51.4 08 140 43 1019.0 2/3	31,2 08 060 38 1019,1 3/6	32_4 08 090 39 1018_6 4/8	52,2 13 060 39 1018,0 4/3	51.8 13 070 40 1017.4 2/8	51.2 15 050 12 1018.3 1/8	29.8 16 060 50 1017.0 1/8	28.3 15 060 57 1017.7 1/8	27.1 10 060 65 1014.1 1/8	26,2 11 060 71 1018,4 1/8	25.7 10 070 75 1018.8 N11 -	25,2 08 090 74 1019,2 N11	24.7 08 100 77 1019.4 N11
22.0 06 100 83 016.7 N11	21.6 06 100 82 1016.3 N11	21.0 06 110 84 1016.5 N11	20,8 05 090 36 1016,9 Nil	23.2 06 060 73 1017.4 N11 6.5	25.1 05 090 65 1018.1 N11 39	27.0 07 080 55 1018.6 N11 64	29,2 06 010 41 1013,6 N11 73,2	31,2 06 010 35 1018.L N11 80.5	32.2 05 070 30 1017.8 1/6 83.9	<b>32.8</b> 07 360 28 1017.6 2/8 84.6	33,2 10 350 35 1016,8 3/8 83,9	32,6 10 020 35 1016,6 3/8 65,4	<b>32.8</b> 12 010 <b>32</b> 1016,3 1/8 47.1	<b>30,0</b> 12 060 49 1016, <b>3</b> 1/8 21	29,2 12 050 46 1016,5 1/8 6	27,2 11 060 60 1016,6 N11 -	26.2 07 070 65 1017.0 N11 -	25.1 08 070 71 1017.4 NI1 ~	24,2 05 080 74 1017,8 N11	23. 04 080 78 1017. N11
21.2 02 030 82 017.0 NIL	19.6 02 040 85 1016.7 H11 -	19,1 02 030 85 1016,9 N11	19.7 02 030 84 1017.4 211	21.4 03 070 80 1018.0 N11 7.7	25.0 05 090 64 1018.0 N11 12.6	27.4 08 060 51 1013.7 N11 55.5	29,2 06 050 79 1018,8 N11 58	30.h. 07 030 32 1018.6 1/8 79	32,2 06 260 26 1018,2 1/8 83,6	32,6 11 030 30 1017.2 1/8 83.9	<b>32.0</b> 12 030 <b>32</b> 1017.5 1/8 76 <b>.3</b>	30,2 07 350 38 1017,0 1/8 64	30.0 10 020 33 1017.0 1/8 47.9	30,2 10 010 36 1017,0 N11 34,2	28,9 10 010 16 1017,1 N11 6	27,2 07 030 53 1017,3 W11	25,6 04 060 64 1018,2 111	24,,4 05 060 70 1018,9 111 -	22,6 02 050 75 1019,3 111 -	21.0 02 060 83 1019.1 N11
17.4 Calm Calm 90 018.1 Nil	16,6 Calm Calm 92 1013,1 N11	16,4 Calm S2 1016,3 Nil	17.0 Calm Calm 91 1016,4 H11	19.0 02 160 83 1016.0 1/8	23,2 06 180 68 1019,1 N11	25.5 05 130 22 1019.1 N11 55	27,6 05 120 54 1013,3 1/8 68,8	27.6 07 55 1018.8 5/8 84.6	29.6 11 060 49 1017.9 5/3 89.4	30,2 07 040 12 1017,1 5/8 31,8	29.3 12 040 49 1016,6 5/8	29,2 12 050 52 1016,7 5/8 5/8	29.1 12 070 51 1015.8 3/8 12.1	27.6 12 060 57 1015.8 1/3 55	26,6 12 060 61 1015,7 N11	25,2 10 060 68 1015.7 N11	24,4 06 060 70 1016,3 N11	23,2 05 070 74 1016,6 #11	22,8 05 090 78 1016,6 N11	21.1 02 100 84 1016.5 N11

wred at 4ft above ground level.

above ground level,



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TABLE 1 (Contd.)

Quantity measure	red	Date	01,00	02,00	03.00	04.00	05.00	06.00	07.00	08,00	1 09,00		<b>al the</b> 11.00	= 0.11.T.   12.00
Shade temperature Wind speed Wind direction Humidity Ses-level pressure Low cloud cover Solar intensity (milli	(°C) (knots) (dogreas) (%) (NBS) knatts/cm <sup>2</sup> )	19th Sept, 1962	22,8 05 060 76 1011,0 N11 -	23.2 06 080 65 1011.0 N11	21,8 05 060 77 1010,8 N11 -	20,6 06 030 80 1010,8 N11 -	21.0 05 050 79 1010.9 N11 -	21,2 05 040 86 1011,0 N11 -	20,8 02 070 88 1011.3 N11	21.8 02 090 80 1011.9 N11 -	26,1 06 120 49 1012,1 N11	28,4 04 150 12 1012.5 N11 -	31,8 02 280 36 1012,3 Ni1	32,4 07 060 32 1012,1 N11 -
Shade temperature Wind speed Wind direction Humidity Set-level pressure Low cloud cover Solar intensity (milli	(°C) (ixnots) (degrees) (5) (HB8) Watts/cm <sup>2</sup> )	20th Sept. 1952	24,2 03 050 88 1013.9 111	22,6 03 070 93 1013,9 N11 -	22.0 03 090 95 1013.8 N11 -	21,2 02 090 96 1013.7 N11	21,0 Calm Calm 95 1013.7 3/8	20.0 Calua Calua 97 1013.7 6/8	19.8 Calm Calm 98 1013.7 6/8	20,8 Calm Calm 95 1014,2 6/8	23.8 03 209 90 1014.9 6/8	25.5 05 250 82 1014.7 7/8 -	28,8 05 100 69 1014,8 4/3	29,8 05 210 63 1014,6 2/8

MOTE: Shade temperatures measured at 4 ft above ground level. Minds measured at 33 ft above ground level



Technical Note No. WE 13 Appendix 1 .

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TABLE	4 11	"anta	١.
TROLLE	1 10	Jonta,	• )

					T	ime (100	al time	• 0.J.T.	+ 2 hrs	>										
04.00	05.00	06,00	07.00	08,00	09,00	10,00	11.00	12.00	13,00	14,00	15.00	16.00	17.00	18,00	19.00	20,00	21.00	22.00	23.00	24,00
20.6 06 030 80 1010.8 N11	21,0 05 050 79 1010,9 N11 -	21,2 05 040 86 1011,0 N11	20,8 02 070 88 1011,3 N11 -	21,8 02 090 80 1011,9 N11	26,1 06 120 49 1012,1 N11 -	28,4 04 150 12 1012,5 N11	31.8 02 280 36 1012.7 Ni1	32,4 07 060 32 1012,1 N11 -	35.0 05 050 22 1011.5 N11	35.4 10 050 27 1011.2 N11 -	34, 6 14 010 37 1010,5 N11	33,2 15 040 19 1010,1 N11	32,2 13 040 58 1010,1 N11	30.4 15 040 57 1010.2 N11	28.0 16 040 75 1010.7 N11 -	26.4 17 050 84 1011.2 N11 -	25.8 10 060 80 1012.2 N11 -	25,4 10 070 85 1012,9 N11 -	25.0 08 070 86 1013.6 N12	24, 4 05 060 81 1013,7 1111
21.2 02 090 96 1013.7 N11	21,0 Calm Calm 95 1013.7 3/8	20.0 Calm Calm 97 1013.7 6/8	19.8 Calm Calm 98 1013.7 6/8	20.8 Calm Calm 95 1014.2 6/8	23.8 03 209 90 1014.9 6/8	25.5 05 250 82 1014.7 7/8 -	28,8 05 100 69 101 4,8 4/8	29, 8 05 210 63 1014,6 2/8	30.6 05 030 59 10142 1/8 -	30.6 05 330 62 1014.1 1/8 -	31.2 07 340 58 1013.6 1/8	<b>52.0</b> 06 560 48 1013.5 1/8 -	31.2 07 060 60 1013.3 N11 -	30.1 07 040 55 1013.5 N11 -	28.7 06 060 67 1013.9 N11 -	27.0 06 050 76 1014.3 N11 -	25.8 06 030 81 1014.6 W11 -	24.7 03 040 87 1015.5 N11 -	23.2 02 060 92 1015.9 N11	23,1 Calm 92 1016,1 Nil -

t 4 ft above ground levol.

ground level

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APPENDIX 2
THE FLIGHT RECORDER
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R. S. Howell

#### 1 FLIGHT RECORDER

The flight recorder shown schematically in Fig.1 consists of a Dekko 16 mm camera, modified to single shot, which is controlled and operated by an electronic timing circuit. The camera photographs, through two 45° mirrors, the two instruments mounted above it, namely a temperature indicator and a clock, which are illuminated for the exposure.

#### 2 TIMING AND SELECTOR UNIT

The circuit consists basically of a C.R. time constant and an emitter follower which operates, through a zener diode, two relays, a uniselector and the camera motor. (See Fig.2.)

The circuit supply voltage is stabilised at 16.8 volts by the three zener diodes D.1, D.2 and D.3. This voltage charges C.1 through R.2. The voltage on C.1 is impressed through the emitter follower TR.1, onto the zener diode D.4. At approximately 5.6 volts the diode breaks down and TR.2 conducts allowing relay A to energise. Relay A has four sets of contacts, three of which are used. A.1 provides a hold-on path for the relay, A.4 switches on the instrument lighting and A.3 switches positive volts onto the camera motor. The camera runs until SN-1 is made. This is a commutator switch operated by the camera motor. SN-1completes an energising path for relay B. Relay B has four sets of contacts, all of which are used. B.3 short circuits C.1 reducing the volts across it to zero, B.2 provides a hold on path for the relay through S.V.2, B.4 open circuits the coil of relay A causing the camera to stop, and B.1 switches volts onto the coil of the uniselector allowing it to operate once. In operating, the uniselector open circuits the energising coil of relay B through switch St.2. Relay B deenergising allows the whole cycle of events to be repeated at a period governed by the time constant C.1 R.2.

Fig.3 shows the complete recorder mounted on anti-vibration mountings in a No.114 bomb tail. By unscrewing two knurled nuts, the whole unit can be withdrawn from the tail, and is shown in Fig.4.

#### 3 TESTING

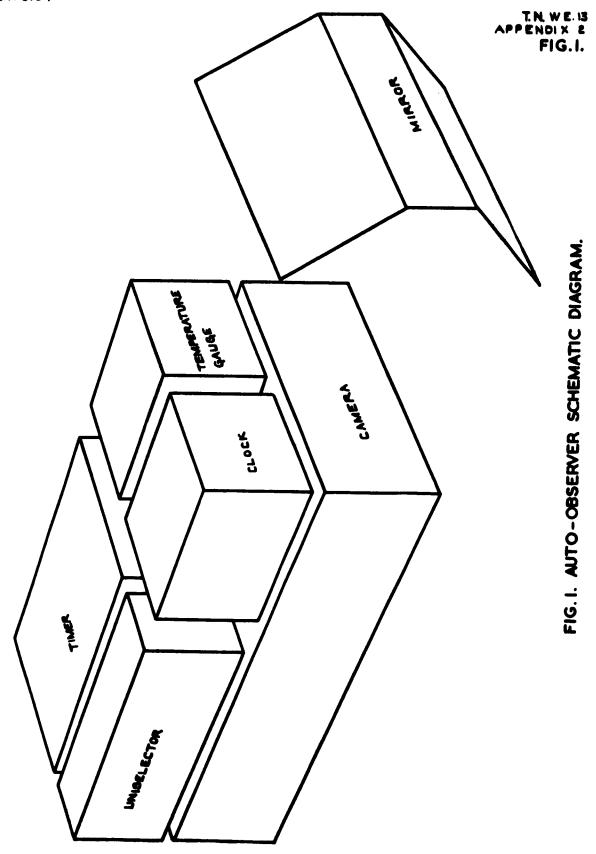
The unit has been tested to operate and give satisfactory photographs under vibration and also at an ambient temperature of 70°C for 1 hour. In flight, the space between bomb tail and recorder was packed with glass-wool to provide some heat insulation, but in fact, under the conditions available for the flights, this was not really necessary.

ATTACHED: Drg. Nos. WE.R.3194-3195 Neg. Nos. 160,661

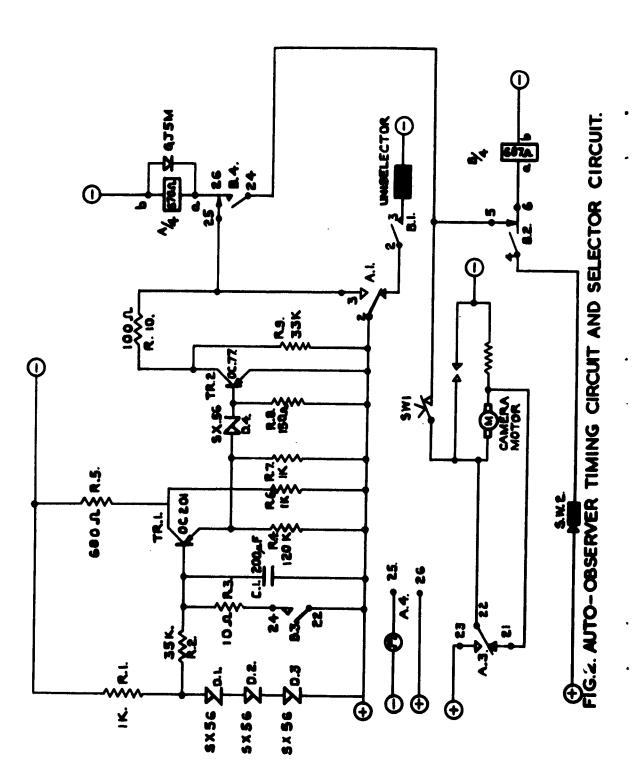
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T.N. WE. 13 APPENDIX.2. FIG. 2.



TECH. NOTE: WE. 13 APPENDIX 2 FIG.3

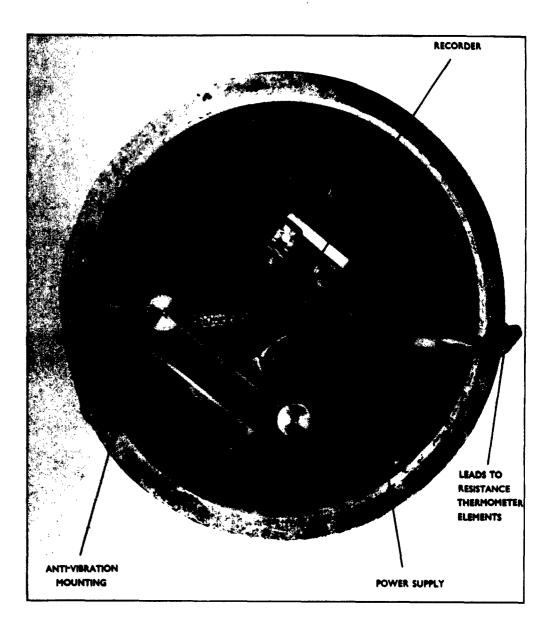


FIG.3. FLIGHT RECORDER MOUNTED IN BOMB TAIL



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## TABLE 1

Bomb temperatures for	ground tr	ial No.1	1000 15 1	ik.10 bomb, no
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					T:	Lmes (G.N	.T. + 2 h	rs)	6 6 6 <b>6</b> 6
	06.00 : 07.00 Daylight 06.30	08.00	09.00	10.00	11.00	12.00	13.00	14.00	15, L
Gauge No.1 2 3 4 5 6 7 8 9 10 11 11 12 13 14 15 16 17	- 20°C 20 20 20 20 20 20 20 20 20 20 20 20 20	20.2°C 20.6 20.6 20 20 20 20 20 20 20 20 20.8 20.8 20.2 21.6 20 20 20 20 20 20 20 20 20 20 20 20 20	- 26.6°C 21.7 24.2 22.6 21.2 22.4 21.4 20 34.4 28.5 22.2 25.6 23.2 20.6 23.2 21.8	- 34°C 27.2 30.4 27.9 24.4 27.8 26 22.4 43.1 35.8 28.1 31.4 27.2 23.7 27.6 26	39.1°C 31.2 36.5 33.2 29 31.4 29.6 23.6 46.2 39.6 31.8 36.6 31.2 27.2 30.4 28.6	42°C 35 41.8 37.8 33 35.2 32.8 28.6 48 41.8 35 42 36.4 30.6 32.6 30.8	35.2 34.4 32.8	43.8°C 39.2 51 46.6 41.4 39.4 38 33.8 46.6 42.8 37.4 52.6 45 39.2 36.4 35	- 44544445443544353
18 22 23	20 - 20	20  20_2	20	21.9 - 35.4	24.3 - 37.4	26.5 - 39.8	28.3 - 40.6	30.3 - 42.4	3
	Su 3	oms into T. fuse:-	filling: filling:	- Nose, - Nose, Power	gauges 1 gauges 2 gauges 3 unit, gau	,5,8 ,6,9	Rear, ga Rear, ga	uges 10,1 uges 11,1 uges 12,1 r, gauge	4,17 5,18
Solar radiation (milliwatts/cm <sup>2</sup>	i	5.6	36.3	36.9	71.5	81.6	84.6	84	7
Shade temperatu (°C)	· ·	21.4	22.4	30.0	31.4	33.2	34.6	35.0	3
Cloud coverage						Nil ala	oud all da	<b>y</b>	

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## TABLE 1

## temperatures for ground trial No.1 1000 lb Mk.10 bomb, nose pointing to south

			T	imes (G.M	.T. + 2 h	rs)						
)	09.00	10,00	11.00	12.00	13.00	14.00	15.00	16.00	17.00	18.00	19.00	20.00 Darkness
C	- 26.6°C 21.7 24.2 22.6 21.2 22.4 21.4 20 34.4 28.5 22.2 25.6 23.2 20.6 23.2 20.6 23.2 20.6 23.2 21.8 20 - 29.2	- 34°C 27.2 30.4 27.9 24.4 27.8 26 22.4 43.1 35.8 28.1 31.4 27.2 23.7 27.6 21.9 - 35.4	- 39.1°C 31.2 36.5 33.2 29 31.4 29.6 23.6 46.2 39.6 31.8 36.6 31.2 27.2 30.4 28.6 24.3 - 37.4	42°C 35 41.8 37.8 33 35.2 32.8 28.6 48 41.8 35 42 36.4 30.6 32.6 30.8 26.5 - 39.8	- 43.6°C 37.7 47.2 42.6 37.8 37.5 35.8 31 47.8 42.4 36.4 47.6 40 35.2 34.4 32.8 28.3 - 40.6	- 43.8°C 39.2 51 46.6 41.4 39.4 38 33.8 46.6 42.8 37.4 52.6 45 39.2 36.4 35.3 30.3 - 42.4	- 1°C 40.2 54.8 40.2 49.6 40.2 36.1 45.8 37.8 56.4 41.8 37.8 36.8 31.3 36.8 31.3 42.6	- 44.4 40.6 56 51.2 47.2 41.8 38.1 44.8 38.1 44.8 58.4 50.8 45 38.4 50.8 45 38.5 - 42.6	- 44"C 41 53 51.4 48.2 42.4 39.6 42.4 39.6 42.4 39.6 42.4 39.6 42.4 38.4 57.8 51.4 46.4 39.5 51.4 40.8 40.8 40.8 40.8 40.8 40.8 40.8 40.8 40.8 40.4 40.8 4	- 42.4°C 41 44.8 45.6 46.7 40.2 41.4 40.4 39.4 37.9 44.1 44.7 38 37.8 37.8 35.4 - 35	- 38.6°C 39.2 38.6 40.1 42.2 36 38.5 38.5 38.6 36.4 37.6 38.8 40.4 35.4 35.4 35.4 35.2 34.4 - 31	
n:- sf to s:-	filling:- filling:-	- Nose, - Nose,	gauges 1, gauges 2, gauges 3, unit, gau	5,8 6,9	Rear, gau Rear, gau	uges 10,13 uges 11,14 uges 12,15 r, gauge 2	,17 (uppe 5,18 (bott	r, starb	ide) Gau oard side	ges 1 and )	22 inopen	rative.
ist	22nd 196	52.										
	36.3	36.9	71.5	81.6	84.6	84	79	65.5	49	34.6	8	-
	22.4	30.0	31.4	33.2	34.6	35.0	35.9	35.8	35.0	33.4	30.6	28.6
				OLO LIN	ud all day	r	r I					



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## TABLE 2

Bomb temperatures for ground trial No.2. 1000 lb kk.10 box

06.00 107.00										+ 2 hr		
1	08.00	09.00	10.00	11.00	12.00	13.00	14.00		16.00			1
06.30 C	°c	°c	°c	°c	°c	°c	°c	°c	°c	°c	°c	
18	18.5	19.4	26 07 F	29.5		-	-	-	- 76 E	-	-	3
											32.6	
			1	33						37.8		
18	19	19	26.5	29			41	42	40.8	39	35	
18	18	18.5	22	25.5	30	33	36.2	38	38.5	37.6	34.6	
		19				31		34•5	34.3		31.3	13
											33	
	1											
											1 -	
				1								
18	18.2			28			40		40.8	38		3
18	18	18.2	22	24.6	28.5	31	35.2	37.5	37.8	37	34	13
18	18		21		26	28	30.4	31.8	32	31	29.8	12
		18										
18	18	18	18	19	21	23	25	26.4	27	27•4	27-5	
•					:	Nose, g	auges 1	, 4, 7				
		-			TUR:	NOSC, E	sauges )	, 0, 7	near,	Rankes	129 ()	<i>)</i> ,
ions - 24-25ti	a August	, 1962										
	10	36.4	35	32	33.6	36.0	94	78	66	44.5	33.5	
/cm <sup>2</sup> )		1		min.	min.	min.				ł		
	Į		-	1		•				ł	1	1
		1	max.	nax.	max.	max.				1		
rature 19.2	21.2	24.8	26.6	28 <b>.5</b>	29.8	29.0	32.3	32.2	32.0	31.2	30.0	2
2/8	3/8	1/8	2/8	7/8	6/8	5/8	3/8	<i>ي</i> 18	3/8	2/8	N11	
	06.30 C 18 18 18 18 18 18 18 18 18 18	06.30   0   0     18   18.5     18   18.5     18   18.5     18   18.5     18   18.5     18   18.6     18   18.18     18   18.3     18   18.3     18   18.18     18   18.19     18   18.18     18   18.18     18   18.18     18   18.18     18   18.18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     18   18     19.2   21.2	$\begin{array}{c c c c c c c c c c c c c c c c c c c $	06.30 C     0 C     0 C	06.30 oC     0 C     0	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	$\begin{array}{ c c c c c c c c c c c c c c c c c c c$

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- 18 -

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TABLE	2

## mb temperatures for ground trial No.2. 1000 lb kk.10 bomb, nose pointing to east

				Times	(G.M.T	+ 2 hr	·s)										
11.00	12.00	13.00	14.00	15.00	16.00	17.00		19.00	20.00 Dark-	21.00	22.00	23.00	24.00	01 .00	02.00	03.00	04.00
°c	°c	°c	°c	°c	°c	°c	°c	°c	ness oc	°c	°c	°c	°c	°c	°c	°c	°c
29.5 27.5 22.5 33 29.5 25.5 24.5 29.5 24.5 24.6 23.5 24.6 23.5 24.6 23.5 24.6 23.5 24.6 23.5 24.6 22.6	29.5 25.5 38.5 35 27.5 24.5 33.5 25 25 40.5 34 28.5 25 24 25 21	- 32.8 27.8 42.4 38 33 31 29.6 26.3 35.6 32 26.6 44.5 37 31 28 27.2 23	- 36.3 31 44 41 36.2 33.5 32 28.3 38 34.6 28.8 45.5 40 35.2 30.4 29 25	- 37 32.6 44.5 42 38 34.5 33.5 30.1 38.8 30.1 38.8 30.1 38.8 30.4 47.5 42 37.5 31.8 30.5 26.4	- 36.5 33.5 41.8 40.8 34.3 34.3 34.3 31.3 38.5 36.4 51.6 45.1 40.8 37.8 32 32 32 32 32 32 32 32 32 32	- 35.5 37.8 39 37.6 33.3 31.3 36.6 35 32 40.1 38 37 31 30.6 27.4	- 33 32.6 33 35 34.6 31.3 32.3 32.3 32.8 31 33.9 34 29.8 29.8 29.8 29.5	- 31.5 30.3 31.5 33.2 30.5 31.3 30.5 31.5 30.5 30.5 30.5 31.5 30.5 30.5 31.5 30.5 31.5 30.5 31.5 30.5 31.5 30.5 31.5 30.5 31.5 30.5 31.5 30.5 31.5 32.5 32.	28.5 29 28 28.5 30.2 28.9 29 29.5 27 28 28 27.5 28.2 30 26.5 27 26.6	26 26.9 28 26.9 28.5 26.5 27.2 27.5 25.4 26.3 26.8 26.8 26.8 27 28 25.5 25.5 25.5 25.2 26	24.6 25.9 26.6 25 25.5 27.3 25.5 25.8 26.3 24.4 25.6 25 26 27 24.5 24 24 24 24	24 7 25.8 24 9 26 25 25.3 23.6 24.5 24.8 25 25.5 24.8 25 25.5 23.8 23.8 23.8 23.4	23 24.1 24.8 23.9 25 24 23 24.5 23.8 24.5 23.8 24.5 23.5 24.4 23.5 24.4 23.5 24.4 23.5	22.4 23.5 23.5 22.4 22.9 24 23.5 23.5 23.5 23.4 22.5 23.5 23.5 23.5 23.5 23.5 23.5 23.5	21.4 22.5 22.8 21.4 21.9 23 22.7 22.5 21.2 22.4 21.5 22.4 21.5 22.4 21.5 22.4 21.5 22.4 21.5 22.4 21.5 22.4 21.4 22.5 22.4 21.4 22.5 22.4 22.4 22.4 22.4 22.4 22.4 22	20.4 21.5 22 20.6 21.5 22.2 21.9 21.4 21.5 20 21.6 21.8 20.7 21.8 20.7 21.8 20.7 21.8 21.4 21.5 21.5	19.8 20.5 21 20 20.5 21.8 21.5 20.8 20.9 19.7 20.4 21 20.4 21.2 20.4 21.2 20.4 21.2 20.4 21.2 20.4 21.2 20.5
o till: t till:	ing: 1	Nose, 6	auges 1 auges 2 auges 3	, 5, 8	Rear,	gauges gauges	11, 14	, 17	No V.T. Gauge 1		itted inoper	ative			ŧ		
52 min. 33.8 max.	33.6 min. 97 max.	36.0 min. 23.5 max.	94	78	"	44.5	33•5	-	6.6	-	-	-	-	-	-	-	-
28 <b>.5</b>	29.8	29.0	32.3	32.2	32.0	31.2	30.0	28.6	27.4	26.4	25.8	25.1	24.2	23.8	22.8	22.4	22.0
7/8	6/8	5/8	3/8	4/8	3/8	2/8	Nil	2/3	2/8	1/8	Nil	Nil	Nil	Nil	Nil	Nil	Nil



- 18 -

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## TABLE 3

Bomb temperatures	for ground	trial No.3.	1000 15 Ek.10

	06.00 ; 07.00 Daylight 06.30	08.00	09.00	10.00	11.00			1 <b>4.00</b>	15.00		+ 2 hr 17.00	18.00
		°c	°c	°c	°c	°c	°c	°c	°c	°c	°c	°c
Gauge No.1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18	21 21 21 21 21 21 21 21 21 21 21 21 21 2	21 • 2 21 • 2	22 21.6 21 22.5 22.2 21.8 21.2 21.8 23.2 21.8 23.2 21.8 23.2 22.8 21.4 25 22.8 21.4 21.4 21.4 21.4 21.4 21.4 21.4 21.4	29.4 28.6 23 34.8 30.5 25.8 26.2 23 21.6 29.4 25.8 22.8 35.4 29.4 29.4 29.4 29.4 22.8 35.4 29.4 29.4 24.2 23 24	33.8 30.4 25.9 40.8 36 29.8 29.6 26.4 24.2 33 29 25 41.8 34.4 28.2 26.6 25.8 22.8	38.2 34.2 29.8 46 40.8 35 33.6 29.8 26.8 37.2 31.8 27.4 48 39.4 32.8 29.8 29.8 29.8 29.8 29.8 29.4	41.4 37.8 32 49.6 45 39.4 36.6 33.8 30 40.4 35.8 30.4 52 44.4 37.6 33.2 31.6 27	43.2 39.8 35 50 46 41.8 37.6 35.8 33 42.6 37.8 32.9 53 46.4 39.8 55.2 33.4 29	44.2 41.2 37 50.2 47.4 42.3 39 38 35.4 44.4 39.3 34.8 53.8 53.8 53.8 47.9 41.8 36.4 35.6 30.5	43 41.2 38 47.2 46,2 39 38.4 43.2 39 38.4 43.2 39.8 36.4 43.2 39.8 36.4 43.8 36.4 42.4 36.8 46.4 42.4 36.8 35.8 31.4	41.8 40.6 38 43.4 44.8 38.6 38 37 43 40 36.4 57 44 42 36.8 36.8 32	40.2 39.8 38.2 41 42 41.6 38.2 37.4 41.6 29.8 36.8 42 41.6 29.8 36.8 42 4.1.6 35.6 32.2
	ions - 27-28tr	1	r i		akin we of f	: illing: Nilling:	Nose	, gauge , gauge , gauge	1 181,4, 182,5,	, 7, I , 8, ·I	iear, ga lear, ga lear, ga	uges 11
Solar radia		6.5	39	64	73.8	30.5	83.9	84.6	83.9	65.4	47.1	21
(milliwatts	/om <sup>2</sup> )											-
Shade temper (°C)	rature 20.2	23.2	25.1	27	29.2	31.2	32.2	32.8	33.2	32.6	32.8	30
Cloud cover	Age Nil	Nil	Nil	Nil	Nil	NHI	1/8	2/8	3/8	3/8	1/8	1/8

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- 19 -

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TABLE 3

## temperatures for ground trial No.3. 1000 lb Kk.10 bomb, nose pointing to east

•00 °c	12.00 °c	13.00 °c	14.00 °c	Times 15.00 °C	(G.M.T. 16.00 °c	+ 2 hr 17.00 °c	s) 18.00 <sup>0</sup> c	19.00 °c	20.00 Dark- ness °C	21 .00 <sup>0</sup> c	22.00 °c	23.00 °c	24.00 °c	01.00 °c	02.00	03.00 °c	04.00 °c
.8 .9 .8 .6 .4 .2 .8 .4 .2 .8 .4 .2 .8 .4 .2 .8 .4 .2 .8 .4 .2 .8 .4 .5 .8 .4 .5 .8 .4 .5 .8 .6 .4 .5 .8 .5 .6 .6 .4 .5 .6 .6 .5 .5 .5 .5 .5 .5 .5 .5 .5 .5 .5 .5 .5	38.2 29.8 46 40.8 35 33.6 29.8 26.8 37.2 31.8 27.4 48 39.4 32.8 29.8 29.8 29.8 29.8 29.8 29.8 29.8	41 .4 37.8 32 49.6 45 39.4 36.6 33.8 30 40.4 35.8 30.4 52 44.4 37.6 33.2 31.6 27	43.2 39.3 55 46 41.8 37.6 35.8 37.6 37.8 32.9 53 46.4 39.8 55.2 33.4 29	44.2 41.2 37 50.2 47.4 42.3 39 38 35.4 44.4 39.3 34.8 53.8 47.9 41.8 36.4 35.6 30.5	43 41.2 38 47.2 43.2 39 38.4 43.2 39 36.4 43.2 39 36.4 43.2 39 8 46.4 42.4 36.8 46.8 35.8 31.4	41.8 40.6 38 43.4 44.8 42.8 38.6 38 37 43 40 36.4 57 44.4 42 36.8 36 32	40.2 39.8 38.2 41 42 41.6 38.2 38.2 37.4 41.6 29.8 36.8 42 41.6 29.8 36.8 42 41.6 29.8 36.8 42 35.6 32.2	38 38 37.8 38 40.2 37 39.4 40 38.2 37.2 36.6 38.4 38.4 38.4 38.8 35 35 35 32.8	35.4 35.4 35.2 36.4 37.2 35.2 - 34.8 35.6 35.6 35.6 35.6 35.6 35.8 35.8 35.8 32	33.2 34 35 33.6 35.4 36 33.6 33.6 34.2 33.6 34.2 33.6 34.2 33.6 34.2 33.6 34.2 33.6 34.2 33.6 34.2 35.2 31.6 33.6 34.2 35.6 34.2 35.6	32.2 32.7 33.9 32.2 34.7 32.6 40.4 31.4 31.8 32.2 33.4 31.2 33.4 33 31.2 31 31	31.2 31.4 32 31.2 32 33 31.6 39 40.2 30.4 30.8 31.4 31.4 31.4 31.4 32.8 29.8 29.2 30	30.2 31 31.2 30.2 31.6 32.2 30.6 - 29.4 29.8 30.2 30.6 30.2 29.2 29.8 29.8 29.8	29.2 30 30 29.6 31 30.8 29.6 31 32.4 28.4 28.4 28.4 29.6 29.4 29.4 30 28.2 29 29 29	28.2 29.2 29.6 29 30 30 28.6 29 28.8 27.4 28 28.6 28.2 29.4 29 27.4 28 27.4 28 28.6	27.2 28.2 29 27.6 29.2 28.8 28 27 28.2 26.4 27.2 28.2 27.6 28.4 28.2 27.6 28.4 28.2 26.6 27 27.8	24.2 27 26.8 28.6 27.4 25.6 26.6 27.2 26.2 27 27 27 27 25.6 26 27 27 27 27 27 27 27 27 27 27 27 27
	: illing: illing:	Nose	, gauge , gauge , gauge	в 2, 5,	8, R	ear, ga	uges 10 uges 11 uges 12	, 14, 1	7,	V.T. fu	ze fitt	ed					
.8	30.5	83.9	84.6	83.9	65.4	47.1	21	6	-	-	-	-	-	-	-	-	-
.2	31.2	32.2	32.8	33.2	32.6	32.8	30	29.2	27.2	26.2	25.2	24.2	23.4	22.4	22	21.2	21.2
11	Nil	1/8	2/8	3/8	3/8	1/8	1/8	1/8	Nil	Nil	Nil	Nil	Nil	Nil	Nil	Nil	Nil



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- 19 -

	، مدهد او برمه منگار							والمراجع والمراجع والمراجع	B' 9-8-94
			-		nrs) Sept.	-	40.70	47.00	4 7
والمتعادية ووالات والارتجاز والتحرير	09.30	10,00	10,30	11.00	11.30	12,00	12.30	13.00	13
Geuge No.1 2 3 4 5 6 7 8 9	23.5°C 23.5 23.5 23.5 23.5 23.5 23.5	24.3°C 24.1 23.5 24.1 23.7 24.1	25.3°C 23.9 23.5 24.5 24.1 24.1	25.1°C 25.3 24.5 24.5 24.5 24.5 24.3	25.7°C 25.7 25.1 26 25.5 25.1	25.3°C 25.7 25.1 26 25.3 24.9	25.7°C 26.3 25.5 26.9 25.9 25.3	26.9°C 26.3 25.5 27.1 26.1 25.3	26 25 25 27 27 26
8 9 10 11 12	23.5 23.5 23.5 23.5 23.5 23.5	23.9 23.9 24.3 23.5 23.5	24.9 24.7 24.3 23.5 23.5	24.7 24.7 24.3 23.5 23.5	25.9 25.5 24.7 23.5 23.5	25.9 25.9 24.7 23.5 23.5	26.3 25.9 24.9 23.5 23.5	26.3 25.9 25.1 23.5 23.5	27 26 25 23 23
			Bomb case Surface : 3 cms in:	of filling	gauges 1 gauges 2 gauges 7	5,9			
	Met. con	ditions	September ·	1 <b>5th, 19</b> 62					
Solar radiation (milliwatta/om <sup>2</sup> )	38	55	64.5	68.8	37.0	84.8	85.4	89•4	ı
Shade temperature (°C)	-	25.5	-	27•6	-	27.8	-	29.6	•
Wind speed (kts)	-	5	-	5	-	7	_	17	•
Wind direction	-	<b>13</b> 0	-	120	-	060	-	060	•
Cloud coverage	-	רנא .	-	1/8	-	5/8	-	5/8	•

TABLE 4

Bomb temperatures for ground trial No.4. 1000 lb Mk.10 bomb on Buccaneer pylon, nose pointin



- 20 -

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		Time (G.	M.T. + 2	hra) Sept	15th 1962				ی میں میں اور میں
9.30	10,00	10,30	11,00	11.30	12.00	12.30	13.00	13.30	14.00
3.5°C 3.5 3.5 3.5 3.5 3.5 3.5	24.3°C 24.1 23.5 24.1 23.7 24.1	25.3°C 23.9 23.5 24.5 24.1 24.1	25.1°C 25.3 24.5 24.5 24.5 24.5 24.3	25.7°C 25.7 25.1 26 25.5 25.1	25.3°C 25.7 25.1 26 25.3 24.9	25.7°C 26.3 25.5 26.9 25.9 25.3	26.9°C 26.3 25.5 27.1 26.1 25.3	26.9°0 25.9 25.9 27.9 27 26.1	27.3°C - 27.9 27 26.1
3.5 3.5 3.5 3.5 3.5 3.5	23.9 23.9 24.3 23.5 23.5	24.9 24.7 24.3 23.5 23.5	24.7 24.7 24.3 23.5 23.5	25.9 25.5 24.7 23.5 23.5	25.9 25.9 24.7 23.5 23.5	26.3 25.9 24.9 23.5 23.5	26.3 25.9 25.1 23.5 23.5	27.3 26.5 25.1 23.5 23.5	26.9 26.3 25.1 23.5 23.5
		Bomb case Surface : 3 cms in	of filling	gauges 1 : gauges 2 : gauges 7	5,9				
t. con	ditions	September (	15th, 1962						
8	55	64.5	68.8	37.0	84.8	85.4	89.4	-	31.8
-	25.5	-	27•6	-	27.8	-	29.6	-	30.2
-	5	-	5	-	7	-	11	-	7
-	130	-	120	-	060	-	060	-	040
-	Nil ,	-	1/8	-	5/8	-	5/8	-	5/8

TABLE 4 ares for ground trial No.4. 1000 lb Mk, 10 bomb on Buccaneer pylon, nose pointing west



	ļ			T	lme (mil	utes)				
Gauge No.	0	5	10	15	20	25	30	35	40	
		perature		(°C)						
1	0	4	7.8	10.7	13.3	15.8	17.5	17.5	18.1	<u>Fli</u>
2 3 4 5 6 8 9	0	1.2	3.1	4.2	7.9	9.5	10.6	11.8	12	Hei
3	0	0.2	0.5	0.7	1.1	1.9	2.8	3.7	3.9	Spe
4	0	2.7	6.7	8.9	12.7	14.7	16.6	17.5	17.8	Tak
5	0	0.9	2.7	4	6.5	8.8	10.3	10.9	11.6	Amb
6	0	0	0.3	0,8	0.9	1.5	2.2	2.9	3.5	
8	0	4.1	7.1	10.7	13	16	. 17.8	18•4	18	
	0	1.5	3.2	7.1	8.4	9.9	11.2	12.1	12.4	
10	0	0.4	0.7	0.6	1.3	2	3.1	3.9	4	
11	0	13	21.6	27.4	30.9	33.5	34.0	35.2	35.2	Bom
12	0	9•5	15.6	20.6	24.1	26.3	28.8	30	31	Sur
										3 0
	Mea	n initia	l temps	rature	of whole	e bomb .	32°C			V.T
										V.T
	Ten	perature	s obtai	ned (°C	2					Gau
1	32	36	39.8	42.7	45.3	47.8	49.5	49.5	50.1	
	32	33.2	35.1	36.2	39.9	41.5	42.6	43.8	44	
3	32	32.2	32.5	32.7	33.1	33.9	34.8	35.7	35.9	
<u>i</u>	32	34.7	38.7	40.9	44.7	46.7	48.6	49.5	49.8	Bom
5	32	32.9	34.7	36	38.5	40.8	42.3	42.9	43.6	for
2 3 4 5 6 8	32	32	32.3	32.8	32.9	33.5	34.4	34.9	35.5	
8	32	36.1	39.1	42.7	45	48	49.8	50.4	50	
9	32	33.5	35.2	39.1	40.4	41.9	43.2	44.1	44.4	
10	32	32.4	32.7	32.6	33.3	34	35.1	35.9	36	Теп
11	32	45	53.6	59.4	62.9	65.5	66.6	67.2	67.2	exo
12	32	41.5	47.6	52.6	56.1	58.3	60.8	62	63	
	Mee	n temper	etures	(℃)			•		'	
akin	32	35.6	39.2	42.1	45	47.5	49.3	49.8	50	
ace of filling	32	33.2	35	37.1	39.6	41.4	42.7	43.6	44	
s into filling	32	32,2	32.5	32.7	33.1	33.8	34.7	35.5	35.8	
fuse power unit		45	53.6	59.4	62.9	65.5	66.6	67.2	67.2	
				- <i></i>						

### TABLE 5

1000 1b Mk. 10 bomb carried in Buccaneer bomb bay. Temperatures obtained during flight at Idris. Septem



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- 21 -

#### TABLE 5

## ried in Buccaneer bomb bay. Temperatures obtained during flight at Idris. September 20th 1962

	Time (minutes)							
	5	10	15	20	25	30	35	40
1	perature		°C)				•	
1	4 1	7.8	10.7	13.3	15.8	17.5	17.5	18,1
	1.2	3.1	4.2	7.9	9.5	10.6	11.8	12
	0.2	0.5	0.7	1.1	1.9	2.8	3.7	3.9
	2.7	6.7	8.9	12.7	14.7	16.6	17.5	17.8
	0.9	2.7	4	6.5	8.8	10.3	10.9	11.6
	0	0.3	0.8	0.9	1.5	2.2	2.9	3.5
	4.1	7.1	10.7	13	16	17.8	18.4	18
	1.5	3.2	7.1	8.4	9.9	11.2	12.1	12.4
	0.4	0.7	0.6	1.3	2	3.1	3.9	4
	13	21.6	27.4	30.9	33.5	34.6	35.2	35.2
	9.5	15.6	20,6	24.1	26.3	28.8	30	31
	ean initial temperature of whole bomb 32°C							
	36	39.8	42.7		47.8	49.5	10 F	50.1
				45.3	47.0	49.5	49.5 43.8	44
	33.2 32.2	35.1 32.5	36.2 32.7	39.9		42.8 34.8		44 35•9
	34.7	38.7	40.9	33.1	33.9 46.7	48.6	35•7 49•5	49.8
	32.9		40.9 36	44.7 38.5	40.8	42.3	42.9	43.6
		34.7			40.0 77 E			
	32	32.3	32.8	32.9	33.5	34.4	34.9	35.5
	36.1	39.1	42.7	45	48	49.8	50.4	50
	33.5	35.2	39.1	40.4		43.2	44.1	44.4
	32.4	32.7	32.6	33.3	34	35.1	35.9	36
	45	53.6	59.4	62.9	65.5	66.6	67.2	67.2
	41.5	47.6	52.6	56.1	58.3	60.8	62	63
6	en temperatures (°C)							
1	35.6	39.2	42.1	45	47.5	49.3	49.8	50
	33.2	35	37.1	39.6	41.4	42.7	43.6	44
	32.2	32.5	32.7	33.1	33.8	34.7	35.5	35.8
	45	53.6	59•4	62.9	65.5	66.6	55•5 67•2	67.2
			52.6	56.1	58.3	60.8	62	63
	41.5	47.6	92.00	<b>70</b> •1	20.5	00.0	02	00
-	ليرب ومستعين والمستعين					h	_	

### Flight conditions

Height 500 ft Speed M = 0.81 Take off - 12.00 hrs Ambient temperature 30°C

Bomb case	gauges 1, 4, 8
Surface of filling	gauges 2, 5, 9
3 oms into filling	gauges 3, 6, 10
V.T. fuze power unit	gauge 11
	gauge 12
Gauge 7, reference point,	no recording

Bomb in forward starboard position (doors open) or forward port position when doors closed.

Temperatures of E.R.U. cartridges did not exceed 65°C.



TABLE	6
	~

1000 1b Mc. 10 bomb carried on Buccaneer port inboard pylon. Temperatures obtained during flight at Idris. Sept

				Time	(minut		الكتابة فبمكالة ويستبت عاملك		
Gauge No.	0	5	10	15	20	25	30	35	40
	Temperature rises (°C)								
	0	12.5	24.7	27	28	28.4	27.8	28.7	27
1	ŏ	3.4	9.5	14.4	18.2	20.4	21.5	21.3	21.6
2 3 4 5 6 8	ŏ	0.3	0.6	1.8	3.5	4.5	6.8	8.6	10.1
4	Õ.	12	24.3	26.4	27	27.6	27.6	27.5	26.4
5	0	2.2	8.6	14	16.8	18.2	19.9	21.7	21.2
6	0	0.1	0.3	1.6	2.9	5	6.4	8 30.8	9.6 28.3
8 9	0	14.5	26 10•4	30.6 15.1	30 <b>.</b> 8 19	31 21.4	31 23.7	23	22.9
10	0	4.6 0.5	2.2	2.6	5.6	7	9	10.4	12.1
10	ŏ	16.1	28.5	33	33.7	33.5	34	33.5	28
12	ō	14	26	31	32	32.5	32.3		30
	Nean	initia	l tempe	ratures	: Bomb	case,	32° <b>C,</b> su	rface	of
	Mean initial temperatures: Bomb case, 32°C, surface of filling 30°C, 3 cms into filling 28°C,V.T. fuse 32°C.								
	Temp	erature	s obtai	ned (°C	)	r		1	
1	32	44.5	56.7	59	60	60.4	59.8	60.7	59
2	30	33.4	39.5	44.4	48.2	50.4	51.5	51.3	51.6
2 3 4 5 6 8	28	28.3	28.6	29.8	31.5	32.5	34.8 59.6	36.6 59.5	38.1 58.4
4	32 30	44 32 <b>.</b> 2	56.3 38.6	58•4 44	59 46 <b>.</b> 8	59.6 48.2	49.9	51.7	51.2
5	28	28.1	28.3	29.6	30.9	33	34.4	36	37.6
8	32	46.5	58	62.6	62.8	63	63	62.8	60.8
9	30	34.6	40.4	45.1	49	51.4	53.7	53	52.9
10	28	28.5	30.2	30.6	33.6	35	37	38.4	40.1
11	32	48.1	60.5	65	65.7	65.5	66	65.5	60
12	32	46	58	63	64	64.5	64.3	! 64	62
	Mean temperatures (°C)								
Bomb case	32	45	57	60	60.6	61	60.8	61	59.4
Surface of filling	30	33.4	39.5	44.5	48	50	51.7	52	51.9
3 oms into filling	28	28.3	29	30	32 65.7	33.5	35•4 66	37 65.5	<b>38.</b> 6
V.T. fuse power unit V.T. fuse amplifier	32 32	48 <b>.</b> 1	60 <b>.</b> 5	65 63	64	65.5 64.5	64•3	64	62
Theoretical temperatures (°C)									
3 cms into filling	28	30.1	32.1		34-5	36.5	38.2	39	39.9

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Bomb or Surface 3 oms : V.T. fi V.T. fi Gauge



- 22 -RESTRICTED

TABLE 6

n Buccaneer port inboard pylon. Temperatures obtained during flight at Idris. September 19th. 1962

Time (minutes)51015202530perature rises (°C)12.524.7272828.427.83.49.514.418.220.421.50.30.61.83.54.56.81224.326.42727.627.62.28.61416.818.219.90.10.31.62.956.414.52630.630.831314.610.415.11921.423.70.52.22.65.679	27.5 21.7 8 30.8 23	26.4 21.2 9.6
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	28.7 21.3 8.6 27.5 21.7 8 30.8 23	27 21.6 10.1 26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	21.3 8.6 27.5 21.7 8 30.8 23	21.6 10.1 26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	21.3 8.6 27.5 21.7 8 30.8 23	21.6 10.1 26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	21.3 8.6 27.5 21.7 8 30.8 23	21.6 10.1 26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	8.6 27.5 21.7 8 30.8 23	10.1 26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	27.5 21.7 8 30.8 23	26.4 21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	21.7 8 30.8 23	21.2 9.6
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	8 30.8 23	9.6
14.5 26 30.6 30.8 31 31 4.6 10.4 15.1 19 21.4 23.7	30.8 23	
4.6 10.4 15.1 19 21.4 23.7	23	
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$		
		22.9
	10,4	
16.1 28.5 33 33.7 33.5 34	33.5	28
14 26 31 32 32.5 32.3	32	30
n initial temperatures: Bomb case, 32°C, su	rface	of
ling 30°C, 3 oms into filling 28°C,V.T. fuze		
TTUR So of S cans the sitting to of the site	52-0.	
perstures obtained (°C)		
	60.7	50
33.4 39.5 44.4 48.2 50.4 51.5		38.1
28.3 28.6 29.8 31.5 32.5 34.8		
2 44 56.3 58.4 59 59.6 59.6		
32.2 38.6 44 46.8 48.2 49.9		51.2
8 28.1 28.3 29.6 30.9 33 34.4		37.6
28.1 28.3 29.6 30.9 33 34.4   2 46.5 58 62.6 62.8 63 63   34.6 40.4 45.1 49 51.4 53.7	62.8	
	53	52.9
8 28.5 30.2 30.6 33.6 35 37	38.4	
2 48.1 60.5 65 65.7 65.5 66	65.5	60
2 46 58 63 64 64.5 64.3	64	62
an temperatures (°C)		
	6	
2 45 57 60 60.6 61 60.8		59.4
9 33.4 39.5 44.5 48 50 51.7		51.9
8 28.3 29 30 32 33.5 35.4		38.6
2 48.1 60.5 65 65.7 65.5 66	65.5	60
2 46 58 63 64 64.5 64.3	64	62
coretical temperatures (°C)	•	•
8 30.1 32.1 33.6 34.5 36.5 38.2	39	39.9

#### Flight conditions

Height 500 ft Speed M = 0.81 Take-off 15.00 hrs Ambient temperature 30°C

Bomb case	gauges 1, 4, 8
Surface of filling	gauges 2, 5, 9
3 oms into filling	gauges 3, 6, 10
V.T. fuze power unit	gauge 11
	gauge 12
Gauge 7, reference point,	no recording.



#### TABLE 7

#### Calculated flight limitations for V.T. fuzes Nos.906 and 907 when their temperatures must not exceed 70°C

Sea level		Lin	niting M	ach numbe	r for hei	ghts of:-	,
temperature °C	0	2000 ft	5000 ft	10,000 ft	20,000 ft	30,000 ft	Tropopause
-26	1.45	1.45	1.45	1.55	1.7	1.85	2.15
-20	1.4	1.4	1.45	1.5	1.65	1.85	2.15
-15	1.35	1.4	1•4	1.45	1.65	1.8	2.1
-10	1.3	1.3	1.3	1.45	1.6	1.75	2.05
- 5	1.25	1.3	1.3	1.4	1:55	1.75	2.0
0	1.2	1.2	1.25	1.35	1.5	1.7	1.95
5	1.15	1.2	1.2	1.3	1.5	1.65	1.9
10	1.1	1.1	1.2	1.25	1.45	1.65	1.85
15	1.0	1.1	1.15	1.25	1.4	1.6	1.8
20	1.0	1.0	1.1	1.2	1.4	1.6	1.75
25	0.9	1.0	1.1	1.15	1.35	1.55	1.75
30	0.9	1.0	1.05	1.15	1.35	1.55	1.75
35	0.8	0.95	1.0	1.1	1.35	1.5	1.7
40	0.7	0.95	1.0	1.1	1.3	1.4.5	1.7
45	0.7	0.9	0.95	1.05	1.3	1.45	1.65
50	0.6	0.9	0.9	1.05	1.25	1.4	1.6

<u>NOTE:</u> (1) For speeds in excess of the above, temperature limits will be exceeded for any flight in excess of 10 minutes duration.

(2) Values above are rounded to the nearest 0.05%.

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#### TABLE 8

Calculated fligh								
filling tempe	ratur	e must	not	exceed	80°C	at a	dept	h of
	3 ome	after	40 m	inutes	fligh	nt		

Sea level	Limiting Mach number for heights of :-						
temperature °C	0	2000 ft	5000 ft	10,000 ft	20,000 ft	30,000 ft	Tropopause
-26	2.5	2.5	2.5	2.6	2.75	2.85	3.15
-20	2.4	2.4	2.4	2.55	2.7	2.8	3.1
-15	2.3	2.3	2.35	2.5	2:.65	2.8	3.05
-10	2.25	2.25	2.3	2.4	2.6	2.75	3.0
-5	2.2	2.2	2.2	2.35	2.55	2.75	2.95
0	2.1	2.1	2.15	2.3	2.5	2.7	2.9
5	2.0	2.05	2.1	2.25	2.45	2.7	2.83
10	1.95	2.0	2.05	2.2	2.4	2.65	2.8
15	1.85	1.9	1.95	2.15	2.35	2.65	2.75
20	1.8	1.85	1.9	2.1	2.3	2.6	2.75
25	1.75	1.8	1.85	2.0	2.25	2.6	2.7
30	1.65	1.75	1.8	1.95	2.2	2.55	2.7
35	1.6	1.65	1.75	1.9	2.15	2.5	2.65
40	1.5	1.6	1.7	1.85	2.1	2.5	2.65
45	1.45	1.5	1.6	1.8	2.05	2.45	2.65
50	1.35	1.45	1.55	1.75	1.0	2.4	2.6

NOTE: (1) These limitations apply only to the bomb itself, or one with fuzes inserted in fuze wells within the bomb nose or tail. They do not apply to a V.T. fuze external to the bomb.

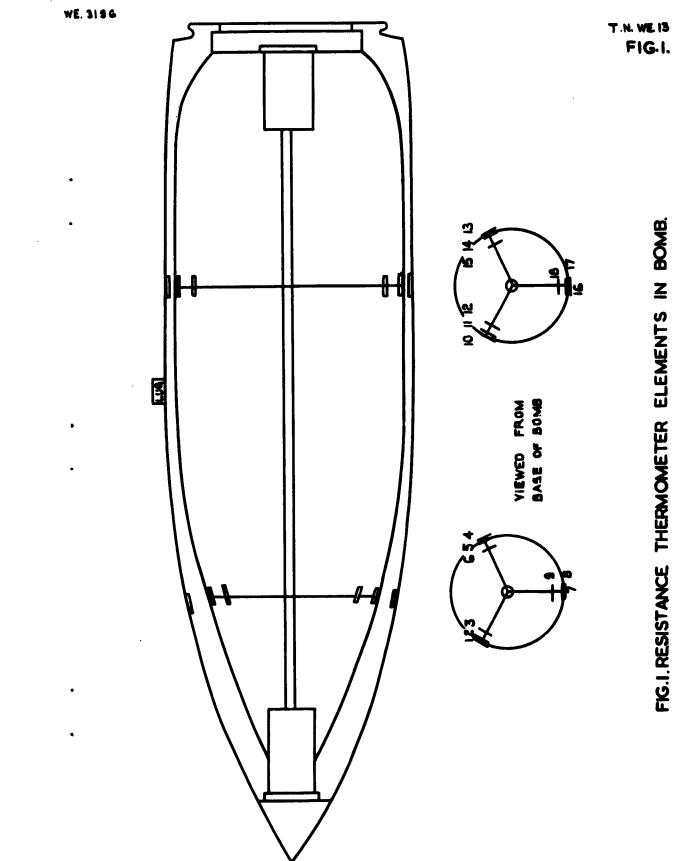
(2) Initial bomb temperatures assumed (see Ref.4):-

Sea level temp.	Initial bomb temp.
$-26$ to $-10^{\circ}$ C	-20°C
-9.9 to $+10%$	000
10.1 to 30°C 30.1 to 50°C	20°C
30.1 to 50°C	30 <sup>0</sup> C

(3) Values are rounded to the nearest 0.05H.

- 24 -

RESTRICTED



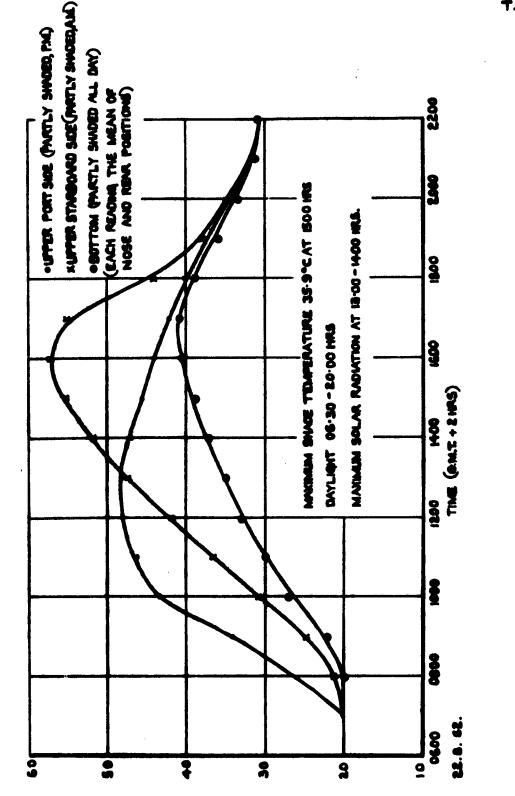
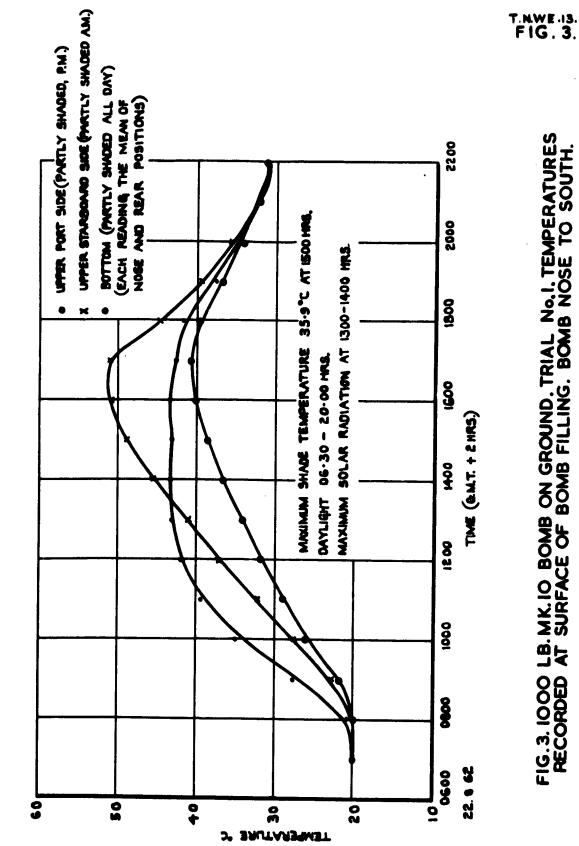


FIG.2. 1000 LE. MK. 10 BOMB ON GROUND. TRIAL No. I. TEMPERATURES RECORDED ON BOMB SKIN. BOMB NOSE TO SOUTH.

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T.N. WE.13. FIG.2.



WE.K. 3190

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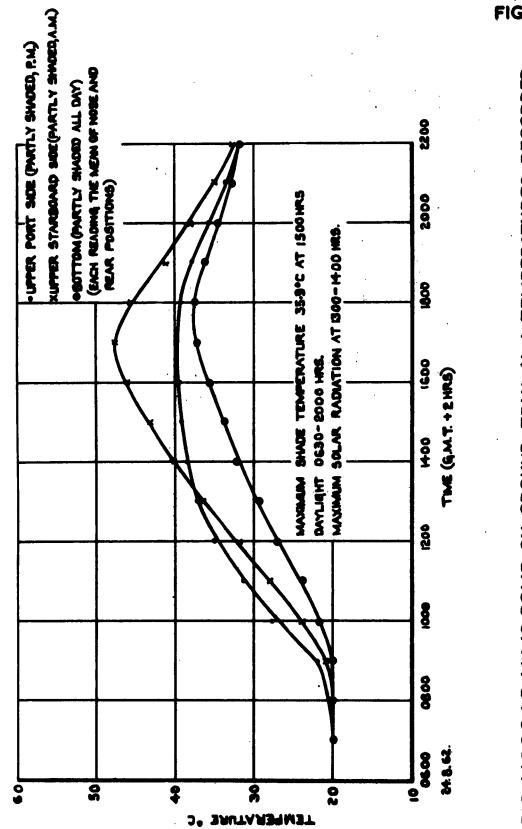
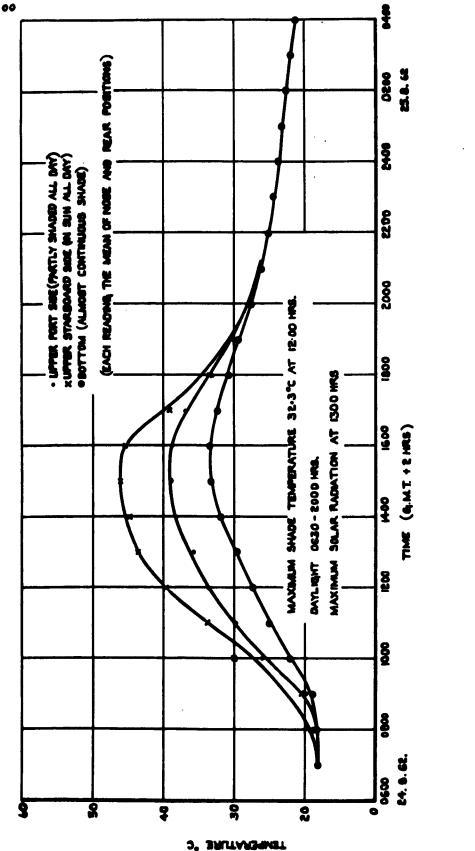


FIG.4.1000 LB. MK. 10 BOMB ON GROUND. TRIAL No.1. TEMPERATURES RECORDED AT A DEPTH OF 3 CMS. INTO THE FILLING. BOMB NOSE TO SOUTH.

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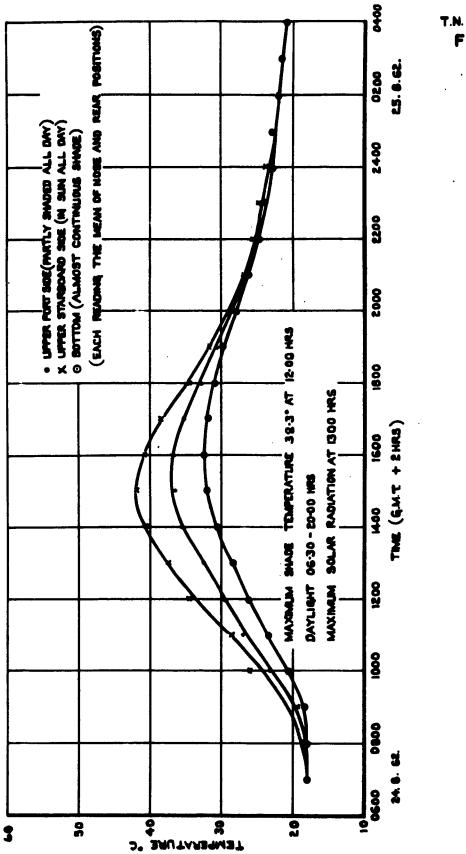


WE.R.3200

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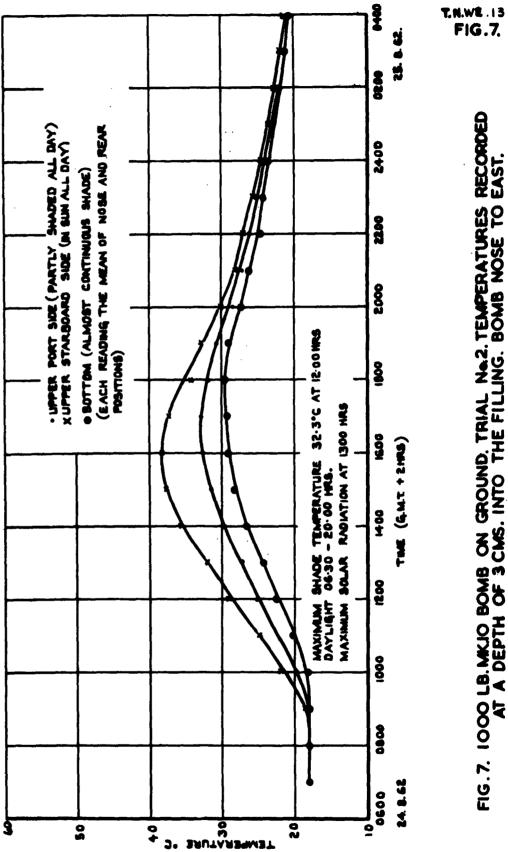
T. N. W E . 13 FIG . 5.

FIG.5. 1000 LB. MK.10 BOMB ON GROUND. TRIAL No.2 TEMPERATURES RECORDED ON BOMB SKIN. BOMB NOSE TO EAST.



T.N.W E. 13 FIG. 6,

FIG.6.1000 LB. MK.10 BOMB ON GROUND. TRIAL No.2. TEMPERATURES RECORDED AT SURFACE OF FILLING. BOMB NOSE TO EAST.



WE R. 3202

T.N. W E. 13 FIG . 8.

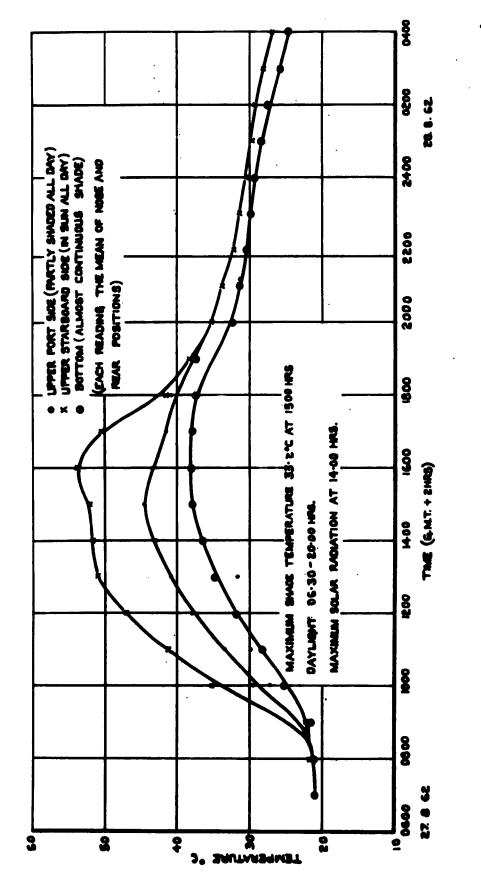
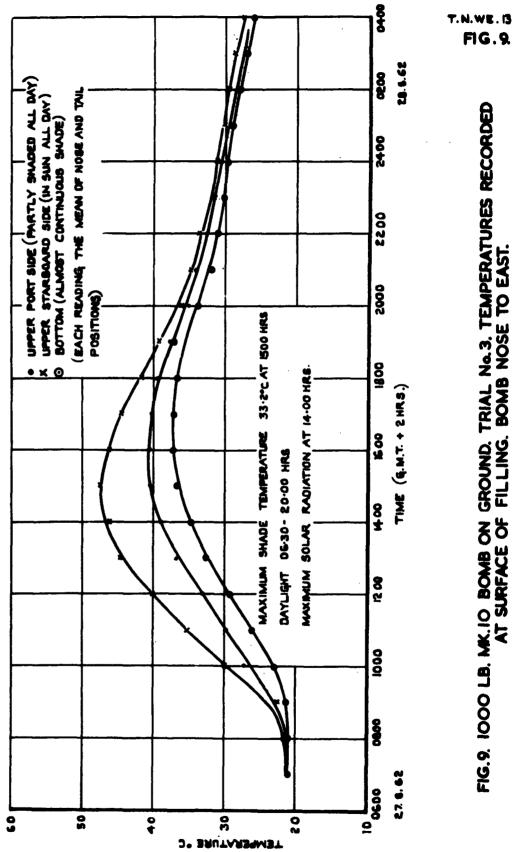
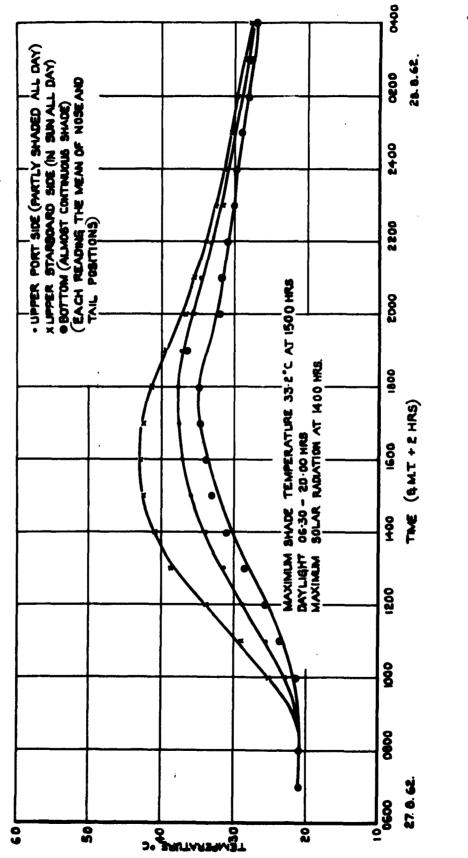


FIG.8. 1000 LB. MK.10 BOMB ON GROUND. TRIAL No.3. TEMPERATURES RECORDED ON BOMB SKIN. BOMB NOSE TO EAST.



WE. R. 3204

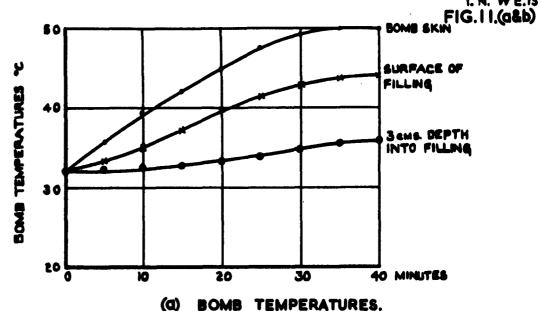


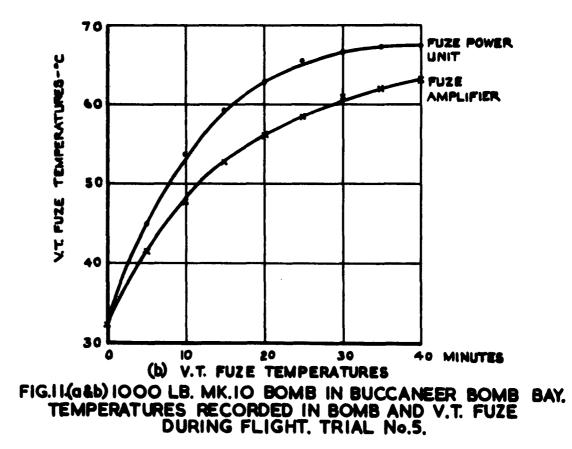


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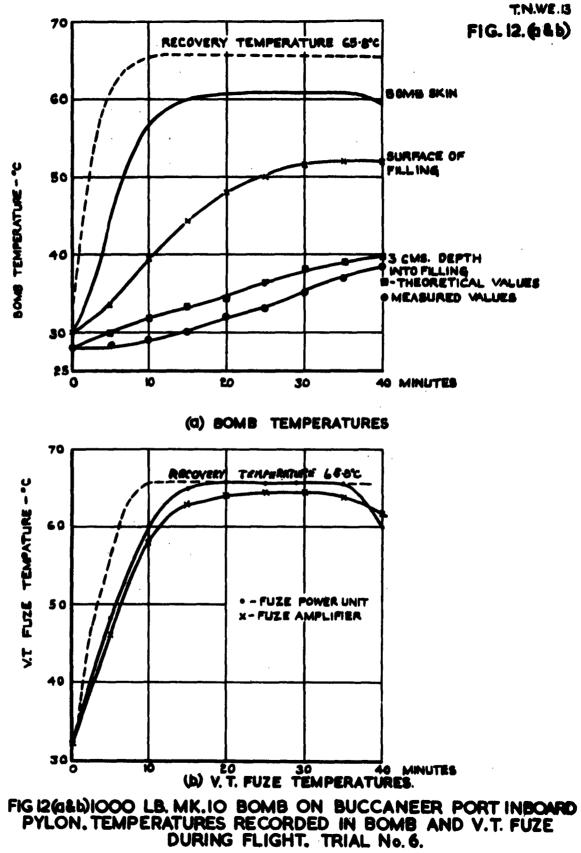
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WE.R. 3206





T. N. WE.13



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