STANDARDIZING SPACE OVENIZED CRYSTAL OSCILLATORS FOR LOWER COST AND FASTER DELIVERY

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Abstract

Building for space is challenging. Program management requirements, parts availability and delivery, parts costs, and testing costs historically make space OCXOs expensive and long-lead. Standardizing specifications based on MIL-PRF-55310, NASA EEE-INST-002, NASA Parts Selection List, and GSFC Preferred Parts Lists, and procuring parts to vendors standard drawings can significantly reduce costs and improve delivery of high-performance OCXOs. Wenzel presents a discussion of parts, materials, and screening specifications for cost and delivery for three grades of OCXOs.

Wenzel Associates builds high-performance, very-low-noise oscillators, frequency standards and synthesizers for military, space, and commercial requirements. Space oscillators have recently been completed for Los Alamos/NNSA, the Lunar Reconnaissance Orbiter, the Aquarius/SAC-D sea surveyor, and TACSAT-4 for NASA. Wenzel’s oscillators were provided to the Mars Lander Mission for JPL, the NEAR Earth Asteroid Rendezvous Satellite for Johns Hopkins APL, and Northrop Grumman and NASA for the National Polar-orbiting Operational Environmental Satellite System (NPOESS). Flight Models of the X-band synthesizer for the Mars Science Laboratory's landing radar for JPL were delivered this year.

INTRODUCTION

“Doing science in space is, above all, expensive, so space researchers often find their careers hanging on the decisions of government ministers rather than grant committees.”

– Daniel Clery - Science, 21 Nov 2008, Cloudy Future for Europe’s Space Plans

Ovenized Crystal Oscillators (OCXOs) are critical components in timing, navigation, ranging, and telemetry systems for space flight. Program management requirements, parts availability and delivery, parts costs, and testing costs historically make space OCXOs expensive with long lead times.

Standardizing specifications based on oscillator specification MIL-PRF-55310, NASA parts specification EEE-INST-002, NASA Parts Selection List, and GSFC Preferred Parts Lists (PPL-21) ensures a clear understanding of the requirements and reduces cost and eliminates errors. Procuring oscillators to vendor’s standard drawings based on these specifications can significantly reduce costs and improve delivery of high performance OCXOs. Also, permitting the use of standard hi-rel and space hardware in the oscillators ensures lower cost for higher reliability oscillators, often with shorter lead times.
# Standardizing Space Ovenized Crystal Oscillators For Lower Cost And Faster Delivery

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Standard oscillator specifications are discussed for three classes of Wenzel's Space VHF OCXO and the High Frequency Space OCXO. A complete data sheet is included for a 10-MHz Space OCXO. The differences between the classes provide insight into the build process vs. cost and delivery options.

**WENZEL’S SPACE HERITAGE**

Wenzel Associates in Austin, Texas, and Croven Crystals, our sister division in Whitby, Ontario, design and build very low phase noise oscillators and crystals for the military and space communities. Some programs for which Wenzel space oscillators were provided are listed:

• 2009 - NPOESS Advanced Technology Microwave Sounder

• 2008 - JPL’s Mars Science Laboratory - Landing Radar

• 2008 - TACSAT-4

• 2007 - Aquarius/SAC Sea-Surveyor

• 2007 - Lunar Reconnaissance Orbiter

• 2003 - NPOESS Advanced Technology Microwave Sounder

• 1996 - NEAR Near Earth Asteroid Rendezvous Satellite

• 1994 - Mars Pathfinder.
The Frequency Synthesizer Assembly delivered to JPL for the Mars Science Laboratory in June, 2008, is part of an innovative landing radar. The synthesizer provides fixed frequencies from 1 GHz to 11.25 GHz.

NPOESS Advanced Technology Microwave Tactical Sounder Oscillator for Northrop/Aerojet.

NPOESS oscillators were first delivered in 2003 and are in current production. The OCXO at 143 MHz, with a built-in odd-order multiplier, is located in the Advanced Technology Microwave Sounder of the National Polar-orbiting Operational Environmental Satellite System.

WHY ARE SPACE OCXOs EXPENSIVE AND LEAD TIMES LONG?

Space is a challenging environment. Launch vibrations are severe and temperatures can be extreme. The spacecraft on which OCXOs are installed can be pelted with radiation, jolted by a pyro shock when a solar panel is deployed, or heated and cooled suddenly as the spacecraft turns toward and away from the sun. Any heat generated in the OCXO must be managed carefully, since most space hardware is in a vacuum. In addition, there is no recall, no easy repair if an OCXO breaks.
There are several cost drivers that make construction of these parts expensive. Program management time can be extensive, parts, and materials are costly and long-lead, and testing and screening costs can accumulate.

**USING STANDARD VENDOR SPECIFICATIONS**

In order to keep program management costs low, reducing the number and breadth of controlling documents is critical. From the initial quotation to the final end-item data package, evaluating specifications and providing program information to the design and manufacturing teams are important but time consuming tasks. When standard military documents are specified, each member of the build team and customer’s team can be confident that the oscillators will meet the program’s requirements. If specifications are confusing, long conferences between vendor and customer are required and additional screening may be prescribed for added confidence when direction from the documents is not clear. Costs can easily spiral upward.

Complete mission assurance plans are often available on the space-based systems and usually contain detailed information that applies to every section of mission hardware and software in order to assure a consistent build and successful mission. The system Mission Assurance Plan can become one of the controlling specifications for the crystal oscillator. The plan can be helpful to oscillator designers and helps them understand the program’s requirements. But evaluating all sections of the plan for a large system takes time and many of the sections will not apply to the oscillator. If this document is available as a guide, and the actual controlling documents are the vendor’s or customer's specifications, time and cost can be reduced.

**SPECIFYING OSCILLATORS, TESTING, AND ANALYSES**

Generally, components such as chip resistors and capacitors are clearly and simply defined in the military standards that apply to them and rarely require other documents to define their fitness for use. Standard military specifications such as MIL-PRF-55342 for chip resistors and MIL-PRF-55681 for chip ceramic capacitors are readily available. Each military specification includes QPL documentation, which defines detailed electrical specifications, mechanical outlines, screening, testing, and marking requirements. The highest reliability level parts specified in the documents, class T for the resistors and class S for the capacitors, are very expensive, but are considered suitable for all space requirements.

For OCXOs, the military specifications do not provide QPL documentation that specifies the oscillator as a single part number. Instead, these oscillators are purchased to a vendor or customer SCD, often with several other controlling documents.

MIL-PRF-55310 is the general specification for crystal-controlled oscillators, and defines three product levels. Level C is for general-purpose applications, level B is for high reliability applications, and level S for is the highest reliability applications. The document is specific and details the parameters critical to oscillator performance. Inspection, screening, qualification, quality conformance inspection, and packaging details are also listed for each product level. Using this standard specification as a guide when writing the specification-controlled drawing (SCD) for the oscillator, it is straightforward to define processes and testing, simplifying the design and build.
SELECTING COMPONENTS AND COMPONENT TYPES

EEE-INST-002 is NASA’s parts selection, screening, qualification, and derating document. It specifies NASA’s requirements for components typically used in class 1 (discrete) oscillators and class 3 (mixed discrete and hybrid) oscillators, as defined in MIL-PRF-55310. EEE-INST-002 details up-screening options for three levels of high reliability builds, and some military parts may be used for space oscillators without additional screening or testing under most circumstances. Following the parts requirements defined in the EEE document can simplify parts selection and verification and can reduce cost and time. Three levels of high reliability, level 1, level 2, and level 3 reference standard military specifications for various part types. EEE-INST-002 document specifies when a level 1, 2, or 3 part is used:

“Level 1: Parts shall be selected and processed to this level for missions requiring the *highest reliability and lowest level of risk*. Level 1 active parts shall be reviewed for radiation hardness, and radiation testing is required when information is not available. The typical mission duration for level 1 program is 5 years or greater.

Level 2: Parts shall be selected and processed to this level for missions with *low to moderate risk* balanced by cost constraints and mission objectives. Level 2 active parts shall be reviewed for radiation hardness, and radiation testing is required when information is not available. The typical mission duration for level 2 programs varies from 1 to 5 years.

Level 3: Parts represent inherently *high risk or unknown risk* because of the lack of formalized reliability assessment, screening, and qualification. Also, there is little dependable data or flight history available for them as the continuous changes in design, materials, and manufacturing processes may make the data on any particular LDC [lot date code] not applicable to another lot and date code. Level 3 parts are intended for mission applications where the use of high-risk parts is acceptable. Level 3 active parts shall be evaluated for radiation hardness, and radiation testing is required when information is not available. The typical mission duration for level 3 programs varies from less than 1 year to 2 years.”

Some parts classified as level 1 are physically and electrically identical to level 3 parts, but lack additional tests, inspections and certification paperwork. When defining the parts requirements for a space program, the cost of selecting a level 2 part rather than a level 3 part and a level 1 part rather than a level 2 part can be significant. Deliveries can change from weeks to months, causing designers to work with parts specialists on alternatives, under the best circumstances. Under worse circumstances, deliveries will be delayed.

Level 2 or 3 parts can be specified in cases where the end use is not long or with additional screening and testing that assures that the oscillator as an individual is documented to be highly reliable.

Several examples of some relative parts costs are listed:

Passive components are numerous and costs are can accumulate. Mean time between failures (MTBF) levels must be carefully reviewed because failure rates can also accumulate in reliability calculations.
Active components can be purchased to the highest level, S, but JANTXV are highly reliable and have more reasonable lead times and costs. DPA and PIND testing for cavity devices can provide additional assurances that the lot is built correctly.

Non-standard parts such as non-QPL passives can be purchased to standard military processes and are readily available. If up-screening is required, an SCD usually must be generated, increasing costs for both customer and vendor.

Non-standard parts such as a passive hybrid semiconductor can be very expensive and long-lead and vendors may perceive cost risk associated with a space program. In the case of this diode-doubler, a standard well-established military diode-doubler was abandoned, because the up-screening costs were prohibitive. A custom-designed alternative was more reasonably priced and performed the same function at a high, but much lower, cost. The cost difference between $45,000 per lot vs. $155,000 per lot was significant for the 20-piece lot. Engineering and program management time to develop custom drawings were high.
### Non-standard Parts, Non QPL Passive Comparison Chart

<table>
<thead>
<tr>
<th>PART TYPE</th>
<th>PART NUMBER</th>
<th>MFG</th>
<th>EACH</th>
<th>LEVEL</th>
</tr>
</thead>
<tbody>
<tr>
<td>TEMPCOMPENSATED ATTENUATOR</td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>COMMERCIAL</td>
<td></td>
<td></td>
<td>$15.00</td>
<td>3</td>
</tr>
<tr>
<td>MIL-PRF-55342</td>
<td></td>
<td></td>
<td>$35.00</td>
<td>2</td>
</tr>
<tr>
<td>MIL-PRF-55342 UP-SCREENED</td>
<td>50 PIECE LOT</td>
<td>$35.00</td>
<td></td>
<td></td>
</tr>
<tr>
<td>GROUP A</td>
<td></td>
<td>$2500 / LOT</td>
<td>$50.00</td>
<td></td>
</tr>
<tr>
<td>GROUP B</td>
<td></td>
<td>$6000 / LOT</td>
<td>$120.00</td>
<td></td>
</tr>
<tr>
<td>TOTAL</td>
<td></td>
<td></td>
<td>$205.00</td>
<td>1</td>
</tr>
</tbody>
</table>

### Non-standard Parts, Passive Hybrid Semiconductor Chart

<table>
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<th>PART NUMBER</th>
<th>MFG</th>
<th>EACH</th>
<th>LEVEL</th>
</tr>
</thead>
<tbody>
<tr>
<td>DIODE DOUBLER</td>
<td></td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>COMMERCIAL</td>
<td></td>
<td></td>
<td>$125.00</td>
<td>COMM</td>
</tr>
<tr>
<td>MIL-PRF-38534</td>
<td></td>
<td></td>
<td>$375.00</td>
<td>2</td>
</tr>
<tr>
<td>MIL-PRF-38534 UP-SCREENED</td>
<td>20 PIECE LOT</td>
<td>$7,750.00</td>
<td>1</td>
<td></td>
</tr>
<tr>
<td>PLUS SCREENING</td>
<td></td>
<td></td>
<td>$7,750.00</td>
<td></td>
</tr>
<tr>
<td>CUSTOM WENZEL SCD</td>
<td></td>
<td></td>
<td>$2,250.00</td>
<td>1</td>
</tr>
</tbody>
</table>

Standard space components are readily available for some parts, such as voltage regulators and feed-thru capacitors.
HS9S-117RH, 5962F9954701VXC voltage regulator, standard space component.

M28861/01-002TB, filter feed-thru, standard space hardware.
STANDARD DATA SHEET DEVELOPMENT

Three data sheets have been developed that define performance, screening, and testing for each of two of Wenzel’s ovenized oscillators, the Space VHF OCXO and the High Frequency Space OCXO.

These differ in the parts requirements, screening, and qualification tests, per the differences detailed in MIL-PRF-55310 for level S and level B oscillators, and use parts defined in EEE-INN-002, level 1, 2, and 3. These specifications can be the basis for defining the requirements for a space OCXO tailored to meet special circumstances.

<table>
<thead>
<tr>
<th>Part Number</th>
<th>Frequency</th>
<th>Screening Level</th>
<th>Component Level</th>
</tr>
</thead>
<tbody>
<tr>
<td>501-20040</td>
<td>100 MHz</td>
<td>S-Level</td>
<td>Level 1</td>
</tr>
<tr>
<td>501-20041</td>
<td>100 MHz</td>
<td>S-Level</td>
<td>Level 2</td>
</tr>
<tr>
<td>501-20042</td>
<td>100 MHz</td>
<td>B-Level</td>
<td>Level 3</td>
</tr>
<tr>
<td>501-20043</td>
<td>10 MHz</td>
<td>S-Level</td>
<td>Level 1</td>
</tr>
<tr>
<td>501-20044</td>
<td>10 MHz</td>
<td>S-Level</td>
<td>Level 2</td>
</tr>
<tr>
<td>501-20045</td>
<td>10 MHz</td>
<td>B-Level</td>
<td>Level 3</td>
</tr>
</tbody>
</table>

The specifications include:

- General program requirements
- Electrical specifications
- Mechanical specifications and outline
- Environmental requirements
- Model definitions for prototypes, EM, FM and qualification units
- Acceptance testing with screening details.

A data sheet is included as Appendix A for a 10-MHz OCXO, P/N 500-20044, with S-level screening and Type 2 parts. The data sheet references standard military/space documentation and can be tailored to a specific program’s requirements.

CONCLUSION

Designing and building space hardware is challenging. Parts, screening, and program management all require special attention, and costs can be considerable. Since OCXOs are not listed on QPL lists, standard and custom written documents must control the build and testing. With the use of standard military specifications and following MIL-PRF-55310 and EEE-INN-002, costs and lead times can be significantly reduced.

There are many limitations on funding and the cost of a program can determine whether it will be funded or cut. Manufacturers of space OCXOs must look for ways to provide exactly the part that is required by their customers. Performance, reliability, delivery, and cost are all critical.
GENERAL REQUIREMENTS
Material, Design and Construction: MIL-PRF-55010
Parts and Materials List: Supplied
Parts, Materials: EEE-INST-002, JAN/XT Semiconductors, IR passive and discrete ICs, when available.

Crystal:
- 10 MHz, premium 3, Z-swept, synthetic quartz
- TM<1% and CVM<0.1% per SP-P-002A
- Traceability: Semiconductor and passive part and data code tracking
- Derating: per EEE-INST-002 (JPL-D-8545, alternative)
- Soldering: J-STD-001 class 3
- Case: Nickel-plated aluminum housing
- Finish: Electroless nickel per MIL-C-26074

ELECTRICAL PERFORMANCE
RF Output Frequency:
- 10 MHz, sine wave
Frequency Accuracy (initial):
- ±2 x 10^-5 ± 15°C
Frequency Stability:
- ±1 x 10^-5 for -40°C to +60°C (ref +25°C)
- Aging Rate (after 50 days operating): 1 day ±5 x 10^-5

RF Output Power:
- ±15 dBm ±1.5 dB into 50Ω
RF Output 2nd Harmonic:
- ±30 dBc
RF Output Sub-harmonics:
- ≤-40 dBc
RF Output Spurious:
- ≤-100 dBc, 100 KHz to 1 GHz

Phase Noise (Short Term):
- ≤10 dBc
- 1 Hz
- ±100 dBc
- 10 Hz
- ±120 dBc
- 100 Hz
- ±150 dBc
- 1 KHz
- ±150 dBc
- 10 KHz
- ±150 dBc

Supply Voltage:
- ±15 VDC ±5%

Warm-up Power:
- ≤5 watts

Warm-up Time:
- ≤20 minutes at ambient pressure ≤5 x 10^-5 torr

Input Power:
- ≤2.5 watts steady state at ambient pressure ≤5 x 10^-5 torr

ENVIRONMENTAL CONDITIONS
Operating Temperature:
- -40°C to +60°C
Storage Temperature:
- -40°C to +105°C
Ambient Pressure:
- Atmospheric (760 torr), Vacuum (≤5 x 10^-5 torr)

MECHANICAL SPECIFICATIONS
Size:
- 2.25" x 2.25" x 1.0" (57.1 x 57.1 x 25.4 mm)
Weight:
- ≤500 grams
Physical:
- Pressure relief holes

MODEL DEFINITIONS
EM (Engineering Model):
- Design and construction similar in appearance and identical in form, fit, and function to FM. Developed using best commercial practice, including some commercial parts and materials. EM shall be subjected only to electrical tests, with some environmental testing performed.

FM (Flight Model):
- Fabricated to meet all design, construction, and test requirements reference MIL-PRF-55010, Class 1.
- Product level 5. FM shall be subjected to the entire compliment of electrical and environmental acceptance tests listed.
  - Flight Model/Space Level, Parts EEC INST-002, Level 1,1
  - MIL-PRF-2887 Level 2 Crystals
  - MIL-PRF-18602 / MIL-STD-750, Semiconductors,
  - JAN/XT with PIN D, JAN/XT with PIN D and CPR, (See)

Qualification Model:
- EM unit, when specified, using EEE-INST-002, Level 1,2 parts where available. Testing per 1 (1 unit).

501-20044-01 EM Engineering Model
501-20044-02 Qualification Model
501-20044-03 FM Flight Model

Wenzel Associates, Inc.
<table>
<thead>
<tr>
<th>Qualification Tests (Non-rigid mode, only)</th>
<th>Revision Record</th>
<th></th>
<th></th>
</tr>
</thead>
<tbody>
<tr>
<td>Group I (1 sample)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Burn-In (operational)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Group II (1 sample)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Aging</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Group III Subgroup 1 (1 sample)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Random Vibration</td>
<td>11.95 Gms, MIL-STD-202, Method 214 D, 50 to 2000 Hz, 5 mm per axis</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Thermal Shock</td>
<td>MIL-STD-202, Method 107, Condition A-1, 25 cycles, -65°C to +85°C</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Ambient Pressure</td>
<td>MIL-STD-202, Method 105, at 5 x 10^3 torr</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Group III Subgroup 2 (1 sample)</td>
<td></td>
<td></td>
<td></td>
</tr>
<tr>
<td>Resistance to Soldering Heat</td>
<td>MIL-STD-202, Method 210, Condition A</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Group III Subgroup 3 (1 sample)</td>
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<td></td>
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</tr>
<tr>
<td>Terminals</td>
<td>MIL-STD-202, Method 211, Condition C, Not applicable for pins &lt;0.25&quot;</td>
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</tr>
<tr>
<td>Solderability</td>
<td>MIL-STD-202, Method 208</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Resistance to solvents</td>
<td>MIL-STD-202, Method 215, Not applicable when marking is electro-etched</td>
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</tr>
<tr>
<td>Electrical Tests*</td>
<td>MIL-STD-202, method 209</td>
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<td></td>
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<tr>
<td>Radiographs</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

Acceptance Tests (Right Mode)

| Electrical Tests*                                | MIL-STD-202, Method 107, Condition A, 5 cycles, -65°C to +85°C |
| Thermal Shock                                    | 7.56 Gms overall, 50 to 2000 Hz, 5 mm per axis |
| Random Vibration (non-operational)               | 50 to 2000 Hz, 5 mm per axis |
| Electrical Tests*                                | 50 to 2000 Hz, 5 mm per axis |
| Burn-In (operational)                            | 240 hours minimum at +60°C |
| Aging Rate                                       | Projected after 30 days operating |
| Electrical Tests*                                | MIL-STD-202, method 209 |

*Electrical Tests*

Tested at ambient pressure x 5 x 10^3 torr and at -40, +25, and +60°C unless otherwise noted

| Warm-Up Power (-40°C only)                        | MIL-STD-202, Method 214 D-I, 50 to 2000 Hz, 5 mm per axis |
| Warm-Up Time (-40°C only)                         | 50 to 2000 Hz, 5 mm per axis |
| Input Power                                       | MIL-STD-202, Method 214 D-I, 50 to 2000 Hz, 5 mm per axis |
| Cold Start (+40°C)                                | 50 to 2000 Hz, 5 mm per axis |
| Hot Start (+45°C)                                 | 50 to 2000 Hz, 5 mm per axis |
| RF Output Power                                   | 50 to 2000 Hz, 5 mm per axis |
| RF Output Harmonics                               | 50 to 2000 Hz, 5 mm per axis |
| IF Output Spurious                                | 50 to 2000 Hz, 5 mm per axis |
| Frequency Accuracy (+25°C only)                   | 50 to 2000 Hz, 5 mm per axis |
| Frequency Stability                               | 50 to 2000 Hz, 5 mm per axis |
| Phase Noise - Static (-25°C only, 760 torr)       | 50 to 2000 Hz, 5 mm per axis |

Analyses

| Thermal Analysis, Component Stress Analysis       | MIL-STD-202, method 209 |