RESEARCH AND DEVELOPMENT ACTIVITIES IN ITALY IN THE FIELD OF AE--ETC(U)

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Research and Development Activities in Italy in the Field of Aerospace Structures and Materials

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RESEARCH AND DEVELOPMENT ACTIVITIES IN ITALY IN THE FIELD OF AEROSPACE STRUCTURES AND MATERIALS

by

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Technical Address given at the 47th Meeting of the Structures and Materials Panel in Florence, Italy on 25 September 1978.
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PREFACE

It is the aim of the AGARD Structures and Materials Panel to stimulate and improve cooperation and exchange of technical information among the NATO member nations with regard to the science and technology of aerospace structures and materials.

Of major importance in fulfilling this task are the bi-annual Meetings of the Structures and Materials Panel, which bring together the leading people of the member nations working in this field. Moreover, the presentations given during these meetings and the ensuing discussions become available to the whole scientific community as AGARD Publications.

It has become a custom that, during the opening session of a Structures and Materials Panel Meeting, the host nation gives a presentation describing their country's activities related to aerospace structures and materials.

The 47th Meeting of the Panel took place from 25—29 September 1978 in Florence, Italy and, at this Meeting, Professor L. Lazzarino of the University of Pisa had the difficult task of presenting in a limited time a concise but complete survey of the aerospace research and development activities in Italy. He succeeded so well in this task that the Structures and Materials Panel decided unanimously to publish Professor Lazzarino's presentation as an AGARD Report. It is hoped that this very interesting survey of current programmes in Italy may stimulate possible cooperation and exchange of information.

J.B. de Jonge
Chairman, Structures and Materials Panel
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RESEARCH AND DEVELOPMENT ACTIVITIES IN ITALY IN THE FIELD OF AEROSPACE STRUCTURES AND MATERIALS

by
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SUMMARY

After a summary of the main programs of the Italian aerospace industries, a review is performed of Research and Development activities, prevalently aimed at supporting these programs in the field of aerospace materials and structures; also, autonomous research work developed by some Italian Universities in this field is indicated.

This review considers the activities concerned with metallic and non metallic aerospace materials, the fields of stress analysis, vibrations and aeroelasticity, fatigue and crack propagation and particularly interesting development work aimed at resolving design problems and at facilitating the introduction of new technologies. Also, the main facilities developed are indicated.

1. INTRODUCTION

The Research and Development activities in Italy in the field of aerospace structures and materials are developed mainly to support the programs of the aerospace industries; moreover some research contracts were stipulated by Universities, both with the C.N.R. (National Research Council) and with foreign research agencies.

The programs of Italian aerospace industries take into consideration both contributions to international military, civil aeronautical and space programs and autonomous industrial activities for obtaining new prototypes. The aerospace production and overhaul activities will be not considered here.

The majority of the research work is performed by industrial laboratories or results from cooperation between Universities, the C.N.R. and aerospace industries; moreover, Universities perform research work independently, mainly in theoretical and computational areas.

Particular development activities are performed mainly by aerospace industries to resolve design and technological problems.

The Research and Development activities of the Laboratories of the Aeronautica Militare (D.L.A.M.) are mainly devoted to making operative testing methods for the acceptance of aerospace materials and of the involved technological procedures.

A first group of Research and Development work is devoted to the study of aerospace materials and of the technological procedures to be applied.

A second group is devoted to the study of problems of aerospace structures in the fields of stress analysis, vibrations, aeroelasticity, fatigue and crack propagation.

A summary of the activities concerning particular development work for the solution of some design and technological problems and for the improvement of tests procedures will then be given.

The Research and Development activities in the field of aerospace materials and structures require the design and the construction of certain facilities, that will also be described briefly.

For a better understanding of the Research and Development activities above indicated a summary of the main programs of the Italian aerospace industries may be useful.

2. SUMMARY OF PROGRAMS OF ITALIAN AEROSPACE INDUSTRIES

2.1 Civil Aircraft

In the field of civil aircraft Aeritalia is developing cooperation with Boeing concerning the advanced commercial aircraft 767 program, for which an extension of many years is foreseen.

In addition to cooperation for preliminary studies and general design activities, for design calculation, construction and testing of certain aircraft parts and subsystems and the involved tooling and facilities, Aeritalia will be relied upon and therefore involved in the required Research and Development activities.

Aeritalia is also developing a new version of the G.222 aircraft for forest fire quenching. 1

Partenavia 2, Rinaldo Piaggio S.p.A. and SIAI-Marchetti are engaged mainly in autonomous programs concerned with general aviation aircraft and economic problems involved.

The programs of "Costruzioni Aeronautiche Giovanni Agusta S.p.A." (Agusta) are concerned mainly with the development of various types of helicopters, both for civil, Fig. 1, and military use; consequently, research and development activities in the field of structures and materials are aiming at improving design and production of parts and subsystems of helicopters.
2.2 Military Aircraft

In the Tornado-MRCA international program, Aeritalia has the responsibility of the variable geometry wing design, construction and testing and of research and flight test activities concerning flutter, also with external loads. Aeritalia is also developing various new versions of the turboprop military transport aircraft G.222 (Fig. 2) and the new AMX program.

Aeronautica Macchi is developing the new MB 339 trainer, Fig. 3, and has the responsibility of design, construction, testing and development at the external main wing structures of G.222 and of the wing pylon of Tornado-MRCA aircraft; moreover, it contributes to the new AMX program. Rinaldo Piaggio has the responsibility of design, construction, testing and development of the internal main wing structure of the G.222 aircraft.

Military and police versions of some types of helicopters are being developed by Agusta (Fig. 4) also in cooperation with SIAI Marchetti, Fig. 5, that is, moreover, developing military versions of certain general aviation aircraft.

2.3 Space Programs

Various Italian aerospace and avionics industries, led by the Compagnia Nazionale Aerospaziale, are engaged in the national experimental Sirio satellite program, under the supervision of the C.N. S.A.S., as well as in the new national launcher Alfa-program.

The C.R.A. (Centro Ricerche Aerospaziali di Roma), under the framework of the well known "San Marco Project", with the active cooperation of Italian Air Force personnel, has designed, developed and put into orbit (by the C.R.A. launching team, acclaimed by NASA one of the best in the world) from the eccentral Italian Range (Malindi, Kenya) several S. Marco aerosatellites, whose main instrumentation (the so-called "balance") is based upon an original elasto-structural concept.

It should be pointed out that at the C.R.A., with the support of the Italian Air Force, a great complex of experimental means for tests on aerospace structures has been realized.

In particular the C.R.A. has developed original techniques for the measurement of thermal fluxes corresponding to the aerodynamic heating over supersonic and hypersonic speed aircraft or over space-ships during the re-entry phase in the terrestrial atmosphere. Furthermore, the C.R.A. experimental equipment includes big dimensions vibrators, satellite structure balancing machines and a simulator for cosmic vacuum effects and solar radiation tests over satellites, which has been the first to enter into operation in western Europe.

Aeritalia is responsible for the primary and secondary structures of the Spacelab and, in cooperation with Microtectica, for the Spacelab Thermal Control System. This task required remarkable Research and Development activity, also in the field of aerospace materials and structures, and in that of involved sophisticated technological procedures.

Aeritalia also partecipates in O.T.S. international programs (control system and satellite structure), E.C.S. (responsability for design and manufacturing of satellite structure), MAROTS (as well as O.T.S.) and Ariane (design and production of the Capsule Technoloque). Selenia is remarkably active in the design and construction of Satellite antennas, to be used in Intelsat IV, O.T.S., Sirio, Meteosat, Cos B, Intelsat V, ECS, HSat, Sirio 2, Aerosat-ESA and 20/30 GHz ESA programs.

Therefore adequate Research and Development work is required, particularly for the application of new materials, to the solution of the complex antennas problems.

3. RESEARCH AND DEVELOPMENT WORK CONCERNED WITH AEROSPACE MATERIALS AND TECHNOLOGIES

3.1 Metallic Materials and Respective Technologies

The design and the construction of the Spacelab primary and secondary structures required the development by Aeritalia of particular equipment for the application of the Tungsten Inert Gas welding system, for obtaining constant high quality welding in large structures such as those of the Spacelab, and to adequately control the welds obtained. The Centro Ricerche FIAT is developing research work on surface treatments by high power Laser on Nickel superalloys and on heat treatment powder metallurgy Nickel-base superalloys. The results obtained up to now, through this research work, and the methods and procedures used will be explained in detail later in a paper presented at this meeting.

FIAT Aviazione is developing experimental research work on:

- High temperature creep under fatigue loading of PK 24 and IN 100;
- Slurry techniques to obtain protection against corrosion through various coatings (a patent concerned with aluminium coating of metallic elements was obtained);
- Cooperative research about high temperature behaviour of Ni-MoCast 739;
- Ion implantation and Laser applications to obtain improvement of surface characteristics of metallic materials;
- Development of high temperature sintering procedures (Astroloy);
- NDI of materials applying small angle neutron scattering.

The University of Bologna (Istituto di Metallurgia) began, under the sponsorship of the D.L.A.M., research work on the behaviour at high temperatures of Ni-Alumina composite
materials obtained through electrocodeposition. This work was then continued autonomously and interesting results were obtained; the characteristics of these composite materials were investigated at temperatures up to 1100 °C. Now this research has been extended to Ni-Fe-Alumina, Ni-Cr-Alumina and Ni-Bo systems.

Development work for the technology required for the construction of sophisticated satellite antennas, support frameworks, reflectors and polarizers was carried out by Selenia for the application of several particular kinds of Steel, Aluminium and Magnesium alloys needed for the construction of such structural elements.

Agusta has recently developed experimental research work for the determination of the influence of the forming and casting procedures and of metallurgical defects in aluminium cast alloys on their fatigue behaviour.

The Istituto Sperimentale dei Metalli Leggeri is developing research work, sponsored by the D.L.A.M., to obtain a comparison between the characteristics of Zergal 3 and Zergal 4 forged Aluminium alloys and those of Alcoa X 7050. To improve the characteristics of alloys of this type new compositions are now being subjected to investigation.

Moreover, research work is being carried out at ISML to obtain, through new thermomechanic treatments (TTMF, TTMI) of new Aluminium alloys of the 7000 series, improvement of their maximum strength and fatigue behaviour.

3.2 Non-Metallic Materials, Organic Bonding and Respective Technologies

The possible applications of various reinforced organic fiber materials, both to aircraft and spacecraft structures, attracted the attention of certain Italian aerospace industries. Therefore a remarkable effort, mainly consisting in development work, was recently made by certain industries, sometimes in cooperation with the C.N.R. and Universities.

Aeritalia is now developing an extensive program for the application of carbon fiber reinforced polymerized resin to the construction of movable airplane surfaces (ailerons, flaps, elevators, rudders, spoilers, tabs, etc.) also for large aircraft. After preliminary research to obtain a mathematical model of the behaviour of such structures prototypes structures were designed, calculated and built; certain components are now being subjected to fatigue testing. Moreover, a framework to support a space telescope to be used upon a satellite was obtained by bonding structural elements built in carbon reinforced resin.

SIAI Marchetti is carrying out the development work required to obtain an increasing application of bonding procedures to the construction of wide wing honeycomb panels also of the main wing structure of general aviation aircraft. Bonded structural elements are also adopted for other structures of such aircraft to obtain lighter and cheaper solutions.

The Research and Development work going on aims at obtaining the extensive substitution of the metal skins of honeycomb panels with sheets of reinforced resin.

Agusta and Elicotteri Meridionali are developing a Research and Development program for the extensive application of composite materials to helicopter structures, also with the aim of obtaining remarkable noise reduction because of the good damping characteristics of such materials.

Research work carried out jointly by the Istituto di Tecnologia Aerospaziale of the University of Roma and Selenia, under the sponsorship of the C.N.R.-SAS, is now under way to analyze the possible applications of various composite materials to the construction of directional antennas for telecommunication satellites using frequencies between 10 and 30, GHz; this research requires special test facilities.

Selenia also carried out remarkable development activity for the application of glass, carbon and boron fiber materials and bonding technologies in the construction of satellite antennas.

Bonding centres to obtain the safe industrial bonding of aircraft structures were recently developed by Aeritalia at Pomigliano, by the SIAI Marchetti at Sesto Calende and Agusta at Cascina Costa.

4. AEROSPACE STRUCTURAL PROBLEMS

4.1 Stress Analysis of Aerospace Structures

Research into the stress analysis of aircraft and spacecraft structures was performed both with general methods and with reference to special applications, applying theoretical, computational and experimental procedures.

Such research activity has been considerably developed in Italy over the past decades, obtaining remarkable results, generally reported in reviews and proceedings of conferences, academies and scientific and technical societies, particularly in the review entitled "L'Aerotecnica, Missili e Spazio".

An extensive historical and critical analysis of such long lasting research cannot be given; the more relevant recent results are explained in detail in the papers indicated in the enclosed reference list, (from to 29).

Summarizing, in the field of theoretical methods, Prof. Cicala, in a very remarkable recent paper, has further expanded his asymptotic formulation of linear shell theory, developing a fully rational approach for obtaining the relevant parametric expansions, 13.

Other contributions in this field are the application of variational techniques to the non-linear behaviour of aerospace structural panels and to the development of a
new method in eigensolution search. Remarkable theoretical research on structural thermal problems, particularly concerned with anisotropic aerospace structures, was carried out by the Centro Ricerche Aerospaziali di Roma, solving the equations of heat conduction in anisotropic bodies in an original way. In the computational field, several computer programs have been developed in the Universities to solve certain kinds of problems unsuitable for treatment through general purpose computer programs.

Amongst them we can quote two finite element approaches to obtain the Stress Intensity Factor both in bidimensional and tridimensional cases, and computer programs to determine elastic stress states among contacting bodies, in particular in pin joints. Other contributions concern the application of finite element method to the classic problems of semimonocoque aircraft structures. One of the aims of computational research work is the reduction of the cost and the time required to obtain the necessary quantitative results by applying the various proposed mathematical model and finite element methods.

Other original computational programs or subroutines have been developed by various industries to face particular design problems. Selenia developed some computational programs for the study of the thermoelectric problems of satellite antennas, particularly with reference to the deformation allowed by telecommunication requirements.

Complex structures computer results are generally substantiated by strain measurements. Typical examples are the work on the hub of the main of the Agusta A 109 A helicopter and the calculations carried-out by Aeritalia on the diffusion panel (Fig. 8), which constitutes the rotation hinge of the outer half-wing of the Tornado-MRCA aircraft. Other original computational programs or subroutines have been developed by various industries to face particular design problems.

Selenia developed some computational programs for the study of the thermoelectric problems of satellite antennas, particularly with reference to the deformation allowed by telecommunication requirements.

4.2 Vibrations and Aeroelasticity

As for stress analysis of aircraft and spacecraft structures, remarkable research activity has been going on into vibrations and aeroelastic phenomena in aircraft and spacecraft during the past decades in Italy, mainly in the Universities. The results of this activity can generally be found in reviews and proceedings, and particularly in the review “L’Aeronautica Missili e Spazio”. The more recent papers dealing with this research field are indicated in the enclosed reference list (from 31 to 42), and can be summarized as follows.

The dynamic behaviour and the response to random excitation of thin walled stiffened aircraft and spacecraft structures has been studied by the University of Pisa and by the Technical University of Milano (Politecnico), particularly in the acoustic range because of the remarkable importance of acoustic fatigue in modern aircraft and spacecraft structures.

Computational methods which take into account both bending-torsion displacements and distorsion of the stringer cross-section, or those based on finite strip elements, were developed.

The response of stiffened panels to acoustic excitation, expressed both in displacement and stress terms, was obtained with a procedure based on modal analysis technique. Tests were conducted to obtain the frequencies and the mode shapes of the response. A comparison between the theoretical and experimental results show satisfactory agreement.

In the field of aeroelasticity a new formulation of the flutter equations allowing efficient solutions both by a continuation and a direct method, and extremely effective methods for solving divergence and flutter eigenproblems, have been developed by the Istituto di Ingegneria Aerospaziale of the Technical University of Milano (Politecnico).

The Istituto di Tecnologie Aerospaziali of the University of Rome has carried out research work on the problem of computing unsteady aerodynamic forces in flutter problems; the results obtained up to now will be presented in this meeting. These original results, obtained thanks to an Italian Air Force Contract, are of a general nature and are presented in operational form.

The Centro Ricerche Aerospaziali of the University of Rome is developing a theoretical and experimental research program about damping in spacecraft structures, using the St. Marco Satellite data. Aeroelastic phenomena require continuous improvement of calculation methods to allow the designer to face design problems effectively, in particular in the case of combat aircraft with different types of external stores.

Aeritalia, with the Tornado-MRCA program, has the primary responsibility of research and experimentation of flutter in flight also with external stores (Fig. 9).

Aeronautica Macchi has developed a mathematical model of a vibrating aircraft to investigate the effect of various mass distribution and of suitably dimensioned recoil dampers on the firing dispersion of 30mm guns carried under the wings. The model was checked by ground vibration tests and the results so obtained were confirmed by ground firing tests (Fig. 10).

Under the sponsorship of the CNR-SAS a cooperative research team, made up of staff from the Universities of Genoa, Pisa, Rome and Naples, as well as from Selenia and Aeritalia, has developed theoretical research work on the attitude control of telecommunication satellites bearing antennas requiring very high directivity.

The Istituto di Tecnologia Aerospaziale of the University of Rome was responsible for...
the part of this research dealing with the elastodynamic phenomena that may influence the attitude of a satellite were carried out by a team made up of a staff from the Technical Universities of Milan and Turin, the University of Pisa and Aeritalia. The items of this research are summarized in Fig. 11. The main results obtained so far are summarized in the paper entitled "Fracture Mechanics Approaches in the Design of Aerospace Vehicles", 47.

In particular, extensive experiments are being carried out to assess the fracture behaviour of several Aluminium alloys, and to evaluate on a common basis the most popular methods for ductile fracture mechanics, Fig. 12. An original model for ductile fracture was developed, implemented through finite element analyses and evaluated by test data, 49, 50, Fig. 13.

Classification of the crack growth behaviour of different stiffened and riveted panel configurations was drawn up on the basis of systematic experimental data, 51, 52, 53, 54. The test data obtained show the remarkable influence of stiffness between the joined elements on the crack growth behaviour of the various types of panel investigated, Fig. 14. Further investigation on this topic is aimed at obtaining an applicable analysis of these influences.

Theoretical and experimental investigation into the reliability of various crack growth computation methods is being carried out following the rationale shown in Fig. 15. An extensive investigation into the impact of the damage tolerant structure requirements on the minimum weight configuration of stiffened structures is being carried out with the sponsorship of the University of Pisa and Aeritalia.

Aeritalia is developing a three-years research program on the fatigue behaviour of structures in the G. 222 aircraft; this program includes the systematic collection and elaboration of flight data and extensive laboratory testing (Fig. 17).

Aeronautica Macchi is collecting and elaborating statistical data on fatigue behaviour of typical aircraft structures in greatly differing flight conditions, in 12 different countries, over a period of 20 years, in which about one million flight hours were recorded.

Laboratory fatigue tests for 120,000 simulated flight hours were carried out. A comparison was made between test results and the theoretically evaluated crack propagation in certain structural elements. 51, 52, 53, 54.

5. PARTICULAR STRUCTURAL DEVELOPMENT ACTIVITIES

To face particular design problems development activities were performed, mainly by aerospace industries, through experimental investigation that required the design and the setting up of special test facilities and devices.

Some of these will be briefly indicated here. Aeritalia set up particular facilities to submit the whole structure of the Spacelab to static and fatigue test (Fig. 18). Previously, vibration systems, environmental chambers and acceleration systems were set up to perform development activities required by satellite design. Agusta developed particular facilities for fatigue testing of the main helicopter structural elements adequately simulating the heavier combinations of foreseen flight conditions (Fig. 19). Aeronautica Macchi set up a facility to simulate the bird impact on aircraft (Fig. 20) and landing gear drop test facilities to improve the shock-absorbing systems.

Rinaldo Piaggio developed a fail safe fuselage air tank test facilities to check structural solutions aimed at preventing explosive decompression due to unstable propagation of cracks which may occur during the aircraft's life. Interesting results were obtained with the use of these facilities.

The Technical University (Politecnico) of Torino carried out development activity to obtain extruded solutions for wing and aileron structures for various types of aircraft.
The extruded elements can be chemically milled to obtain adequate variation of their thickness. Experiments were performed and others will soon be completed.

6. CONSIDERATIONS ON THE AVAILABLE RESEARCH AND DEVELOPMENT CAPACITIES AND FACILITIES IN ITALY IN THE FIELD OF AEROSPACE STRUCTURES AND MATERIALS

The main programs briefly summarized above of the Italian aerospace industries indicate clearly their more relevant interests in the general field of aerospace structures and materials.

From the foregoing synthetic review of Research and Development activities, both in aerospace industries and Universities, it is possible to obtain general information about the capabilities and facilities available in the field indicated above.

In the past decades, notwithstanding the heavy economic and financial difficulties that troubled the Italian aerospace industries and the well known general University crisis, a remarkable effort has been made to reach a good scientific and technical level both in industries and Universities. Modern computational programs were extensively acquired and then originally developed to face particular problems and to make their use cheaper. Cooperation between Italian aerospace industries and Universities and foreign industries and research agencies has improved and this also produced an improvement in the capabilities of the aerospace people and therefore of autonomously designed aircraft and spacecraft and of the autonomous underdeveloped university research activity.

The Research and Development facilities in the field of aerospace structures were developed, nevertheless following too strictly and sometimes inadequately the few aerospace programs capable of surviving. This inadequacy was particularly heavy in the field of new aerospace materials, the majority of activities in this field being devoted to development with a view to particular applications.

I hope this review has provided sufficient coverage of Italian capacities and interest in the general field of aerospace structures and materials and contributed to assisting this Panel in its objective of encouraging and facilitating cooperation between the larger and smaller nations.

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Fig. 1 - Agusta A 109 Civil Helicopter.

Fig. 2 - New military version of the G. 222 aircraft.
Fig. 3 - Aeronautica Macchi MB.339 trainer.

Fig. 4 - Military version of Agusta A 109 helicopter.
Fig. 5 - Agusta-SIAI Chinook CH 47 helicopter.

Fig. 6 - Spacelab primary structure.
Fig. 7 - MB. 339 aircraft-stress analysis.

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Fig. 10 - MB-339 30 mm. underwing gun pod dynamic structural behaviour.
Fig. 11 - Main items of cooperative research on Fracture Mechanics in the design of aerospace vehicles.
Fig. 12 - Outline of an approach for the critical evaluation of different Fracture Mechanics methods.
EXPERIMENTAL INSTALLATION

- J contour
- extremities of the integration intervals
- T rosette
- single gage

Experimental evaluation of J-Integral from Strain Gage Measurement

Comparison of Experiments and FE Analyses

Fig. 13 - Experimental evaluation of models for ductile fracture.
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RELIABILITY EVALUATION OF CRACK GROWTH PREDICTION METHODS

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Fig. 15 - Flow chart and typical results of a rationale for the evaluation of the crack growth prediction methods.
Fig. 16 - Facilities for fatigue and fracture mechanics tests at the University of Pisa, Institute of Aeronautics.

Fig. 17 - Fatigue Test of G. 222 Aircraft.
Fig. 18 - Static and fatigue test of Spacelab structure.

Fig. 19 - Typical fatigue tests of helicopter dynamic components at Agusta fatigue laboratory.
Fig. 20 - Aeronautica Macchi particular dynamic testing facilities.
After a summary of the main programs of the Italian aerospace industries, a review is performed of Research and Development activities, prevalently aimed at supporting these programs in the field of aerospace materials and structures; also, autonomous research work developed by some Italian Universities in this field is indicated.

This review considers the activities concerned with metallic and non-metallic aerospace materials, the fields of stress analysis, vibrations and aeroelasticity, fatigue and crack propagation and particularly interesting development work aimed at resolving design problems and at facilitating the introduction of new technologies. Also, the main facilities developed are indicated.

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